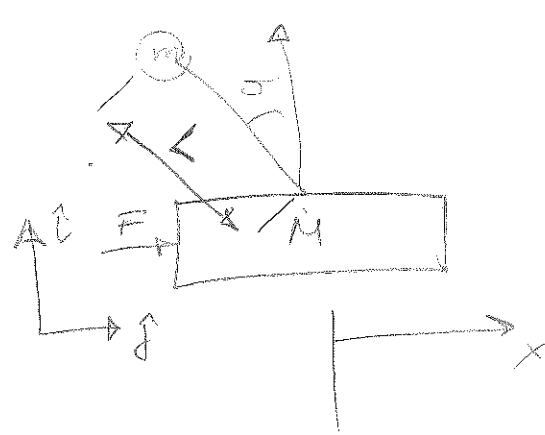
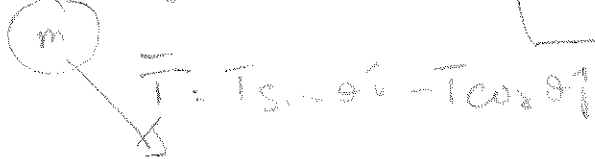
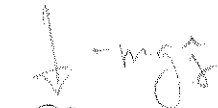
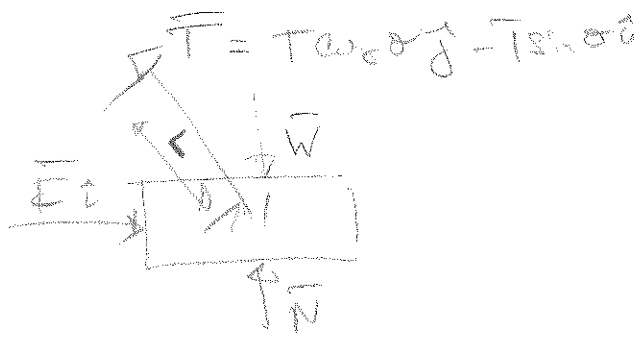


Project Part I

inverted pendulum



Cart:

$$\sum F_x: F - T \sin \theta = M \ddot{x}_M(t)$$

Ball:

$$\sum F_y: -T \cos \theta - mg = m \ddot{y}_m(t)$$

$$\sum F_x: T \sin \theta = m \ddot{x}_m(t)$$

acceleration of the ball:

$$\vec{a}_p = \ddot{x}_M(t) \hat{i} + [L \ddot{\theta} \hat{e}_\theta - L \dot{\theta}^2 \hat{e}_r]$$

$$= \ddot{x}_M(t) \hat{i} + L \ddot{\theta} [-\cos \theta \hat{i} - \sin \theta \hat{j}]$$

$$- L \dot{\theta}^2 [-\sin \theta \hat{i} + \cos \theta \hat{j}]$$

$$\Rightarrow T \sin \theta = m \ddot{x}_M - m L \ddot{\theta} \cos \theta + m L \dot{\theta}^2 \sin \theta \quad \leftarrow \hat{i} \text{ coordinate}$$

$$T \cos \theta + mg = m L \ddot{\theta} \sin \theta + m L \dot{\theta}^2 \cos \theta \quad \leftarrow \hat{j} \text{ coordinate}$$

$$T \cos \theta \sin \theta = m \ddot{x}_M \cos \theta - m L \ddot{\theta} \cos^2 \theta + m L \dot{\theta}^2 \sin \theta \cos \theta$$

$$T \cos \theta \sin \theta + (\sin \theta) mg = m L \ddot{\theta} \sin^2 \theta + m L \dot{\theta}^2 \sin \theta \cos \theta$$

$$\Rightarrow \underline{L \ddot{\theta} = g \sin \theta + \ddot{x}_M \cos \theta}$$

$$\ddot{\theta} L - g \sin \theta + \dot{X}_M \cos \theta = 0$$

$$\ddot{\theta} L = g \sin \theta \rightarrow \ddot{X}_M \cos \theta$$

$$\Rightarrow (M+m) \ddot{X}_M + m(g \sin \theta - \ddot{X}_M \cos \theta) \cos \theta + m \dot{\theta}^2 \sin \theta = f(t)$$

$$(M+m) \ddot{X}_M + mg \sin \theta \cos \theta - m \ddot{X}_M \cos^2 \theta + m \dot{\theta}^2 \sin \theta = f(t)$$

$$(M+m - m \cos^2 \theta) \ddot{X}_M + mg \sin \theta \cos \theta + m \dot{\theta}^2 \sin \theta = f(t)$$

$$(M + m \sin^2 \theta) \ddot{X}_M + mg \sin \theta \cos \theta + m \dot{\theta}^2 \sin \theta = f(t)$$

$$\ddot{X}_M(t) = \frac{-mg \sin \theta \cos \theta - m \dot{\theta}^2 \sin \theta + f(t)}{M + m \sin^2 \theta}$$

$$\ddot{\theta} = \frac{g \sin \theta}{L} + \cos \theta \left(\frac{mg \sin \theta \cos \theta + m \dot{\theta}^2 \sin \theta + f(t)}{M + m \sin^2 \theta} \right)$$

Use to use small angle approximations.

Single input

$$\begin{cases} L \ddot{\theta} - g \sin \theta + \dot{X}_M \cos \theta = 0 \end{cases}$$

$$\begin{cases} (M+m) \ddot{X}_M + m L \ddot{\theta} \cos \theta + m \dot{\theta}^2 \sin \theta = f(t) \end{cases}$$

Double input

$$\begin{cases} L \ddot{\theta} - g \sin \theta + \dot{X}_M \cos \theta = \tilde{u}(t) \end{cases}$$

$$\begin{cases} (M+m) \ddot{X}_M + m L \ddot{\theta} \cos \theta + m \dot{\theta}^2 \sin \theta = f(t) \end{cases}$$

$$\text{Let } x_1(t) = X; \quad x_2(t) = \dot{X}; \quad x_3(t) = \theta; \quad x_4(t) = \dot{\theta}.$$

$$L \dot{\chi}_4(t) - g \sin(\chi_3(t)) - \dot{\chi}_2(t) \cos(\chi_3(t)) = u_2(t)$$

$$(H+m) \dot{\chi}_2(t) + m L \dot{\chi}_4(t) \cos(\chi_3(t)) + m \chi_4^2(t) \sin(\chi_3(t)) = u_1(t)$$

Project part 2 a_i(a):

unit impulse input $f(t)$ and zero initial state.

```
%Yevgen Solodkyy
%CE2.2a part i.a:

clear all;
clc;
m1=2; m2=1; L=.75; g=9.81;

A = [0 1 0 0; 0 0 m2*g/m1 0; 0 0 0 1; 0 0 (m1+m2)*g/(m1*L) 0];
EigenValues = eig(A)

D = 0;

t=[0:.01:7];

x0 = [0;0;0;0];

impulse = t==0;

B = [0; 1/m1; 0; 1/(L*m1)]; % need to verify what this matrix looks
like.
C = [0 0 1 0];

u=[impulse]; %zeros(size(t)) is t x 1 dimentions.

sys_ia = ss(A,B,C,D)% create a state space system based on the system
matrices

[Y0,t,X0] = lsim(sys_ia,u,t,x0); % simulate the system output solution
Y0 & system state solution X0.

% I have 4 state variables and one system solution. So I should have 5
% plots all together for each case.

%PLOTTING PORTION

figure(1);
set(gcf,'numbertitle','on','name','state variable responses i(a)') %
figure title
subplot(411),
plot(t,X0(:,1)); grid; %plot the first state variable
title('state variable responses i(a)')
axis([0 5 -1 100]);
set(gca, 'FontSize',12);
xlabel('\ittime(sec)'); ylabel('{\itx}_1 (\itm)'); % might want to
remove the x-lalel here

subplot(412),
```

```

plot(t,X0(:,2)); grid; %plot the second state variable
axis([0 5 -1 100]);
set(gca, 'FontSize',12);
xlabel('\ittime(sec)'); ylabel('{\itx}_2 (\itm/s)');

subplot(413),
plot(t,X0(:,3)); grid; %plot the third state variable
axis([0 5 -1 100]);
set(gca, 'FontSize',12);
xlabel('\ittime(sec)'); ylabel('{\itx}_3 (\itrad)'); % might want to
remove the x-lalel here

subplot(414),
plot(t,X0(:,4)); grid; %plot the second variable
axis([0 5 -1 100]);
set(gca, 'FontSize',12);
xlabel('\ittime(sec)'); ylabel('{\itx}_4 (\itrad/s)'); % might want to
remove the x-lalel here

```

Command Output:

sys_ia =

```

A =
      x1      x2      x3      x4
x1      0      1      0      0
x2      0      0  4.905      0
x3      0      0      0      1
x4      0      0 19.62      0

```

```

B =
      u1
x1      0
x2     0.5
x3      0
x4  0.6667

```

```

C =
      x1  x2  x3  x4
y1      0   0   1   0

```

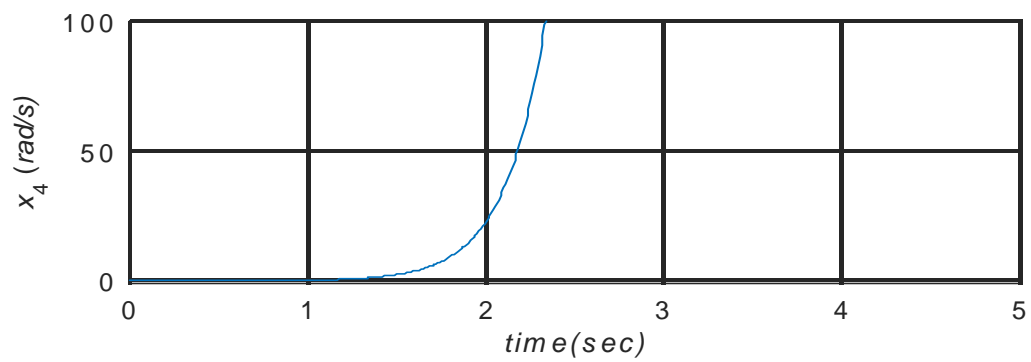
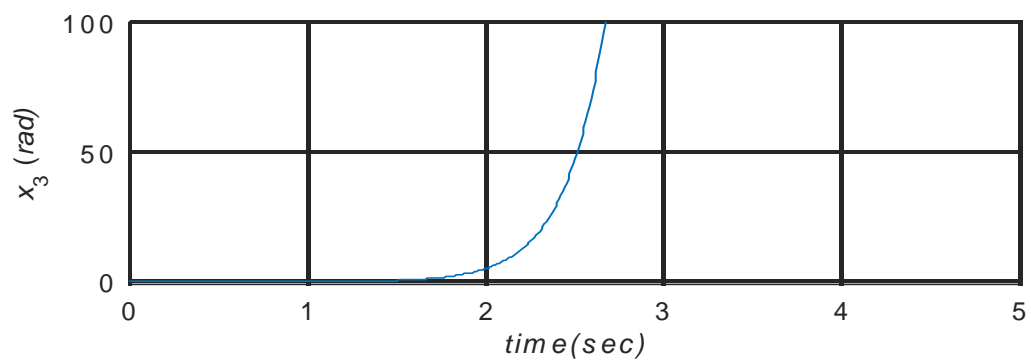
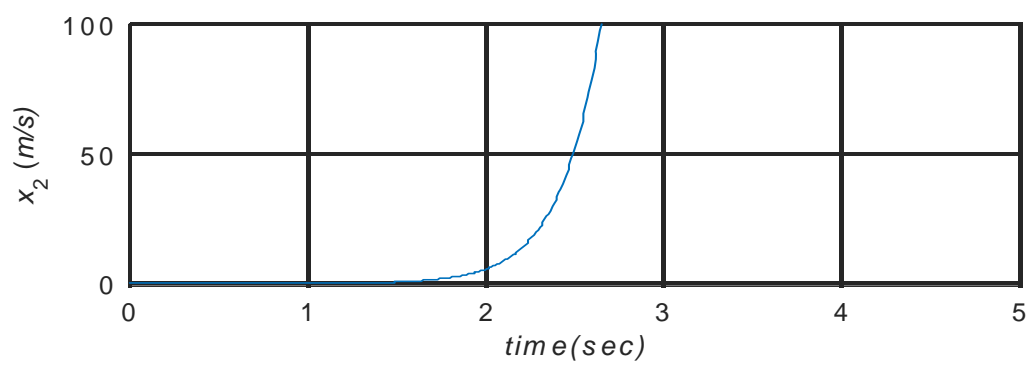
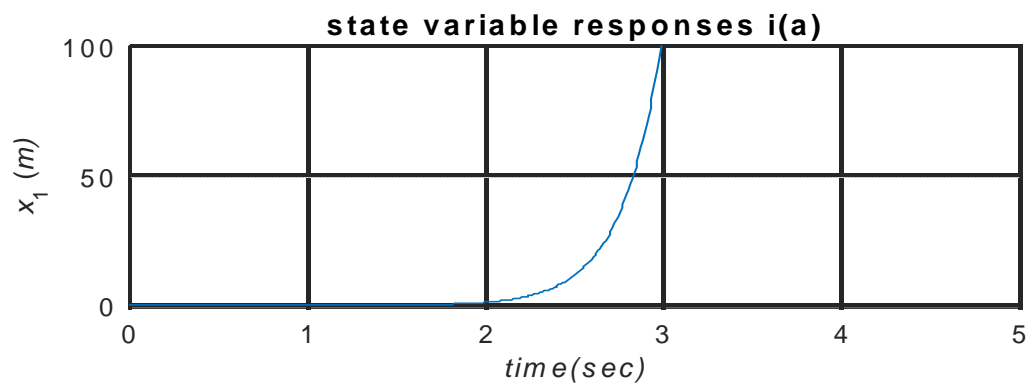
```

D =
      u1
y1      0

```

Continuous-time state-space model.

>>



Project Part 2 a_i(b):

zero input $f(t)$ and initial conditions $\theta(0) = 0.1$ rad

Code:

```
%CE2.2a, part i.b.

clear all;
clc;
m1=2; m2=1; L=.75; g=9.81;

A = [0 1 0 0; 0 0 m2*g/m1 0; 0 0 0 1; 0 0 (m1+m2)*g/(m1*L) 0];
EigenValues = eig(A)

D = 0;

t=[0:.01:7];

impulse = t==0;

B = [0; 1/m1; 0; 1/(L*m1)];
C = [0 0 1 0];
x0_ib = [0;0;.1;0];
u_ib=zeros(size(t));

sys_ib = ss(A,B,C,D)

[Y0_ib,t,X0_ib] = lsim(sys_ib,u_ib,t,x0_ib);

%PLOTting PORTION

figure(2);
set(gcf, 'numbertitle', 'off', 'name', 'state variable responses i(b)') %
figure title
subplot(411),
plot(t,X0_ib(:,1)); grid; %plot the first state variable
title('state variable responses i(b)')
axis([0 5 -1 100]);
set(gca, 'FontSize',12);
xlabel('\it(sec)'); ylabel('{\itx}_1 (\itm)'); % might want to remove
the x-label here

subplot(412), plot(t,X0_ib(:,2)); grid; %plot the second state variable
axis([0 5 -1 100]);
set(gca, 'FontSize',12);
xlabel('\it(sec)'); ylabel('{\itx}_2 (\itm/s)');

subplot(413), plot(t,X0_ib(:,3)); grid; %plot the third state variable
axis([0 5 -1 100]);
set(gca, 'FontSize',12);
```

```
xlabel('\it(sec)'); ylabel('{\itx}_3 (\itrad)'); % might want to remove
the x-label here
```

```
subplot(414), plot(t,X0_ib(:,4)); grid; %plot the second variable
axis([0 5 -1 100]);
set(gca, 'FontSize',12);
xlabel('\it(sec)'); ylabel('{\itx}_4 (\itrad/s)'); % might want to
remove the x-label here
```

Command output:

```
sys_ib =
```

```
A =
      x1      x2      x3      x4
x1      0      1      0      0
x2      0      0  4.905      0
x3      0      0      0      1
x4      0      0 19.62      0
```

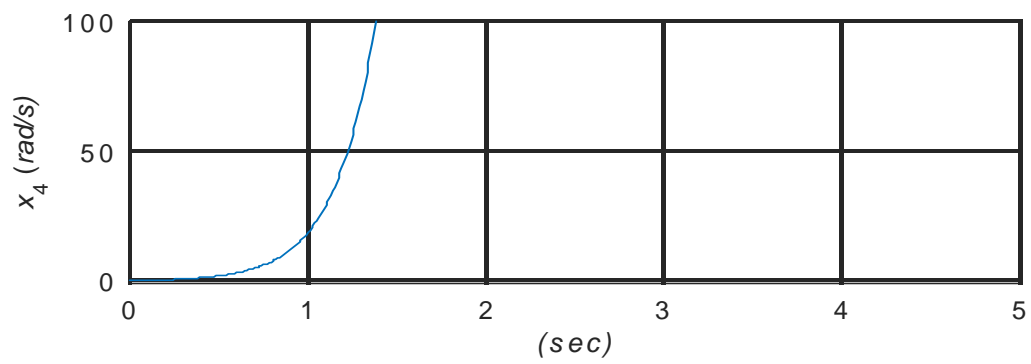
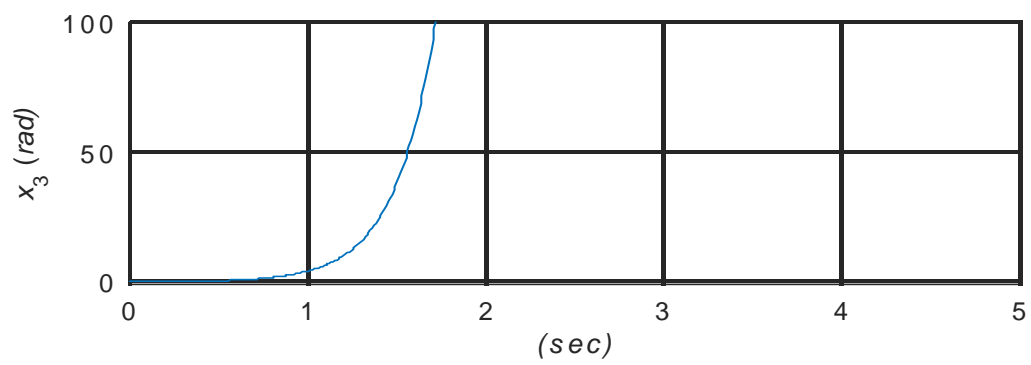
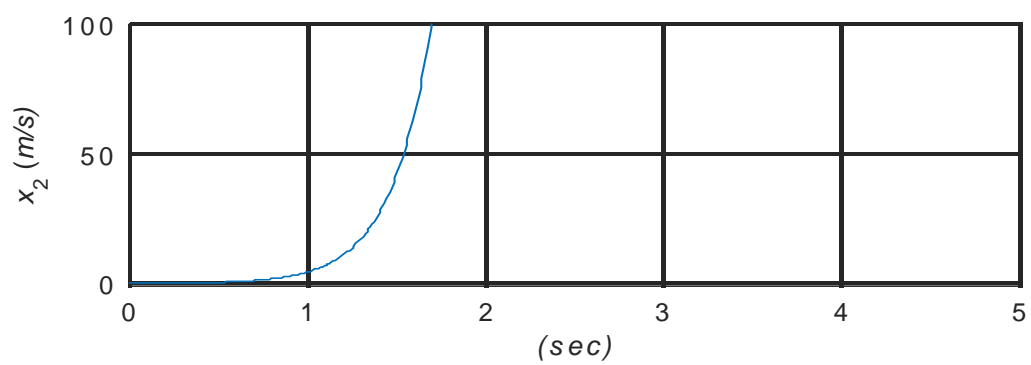
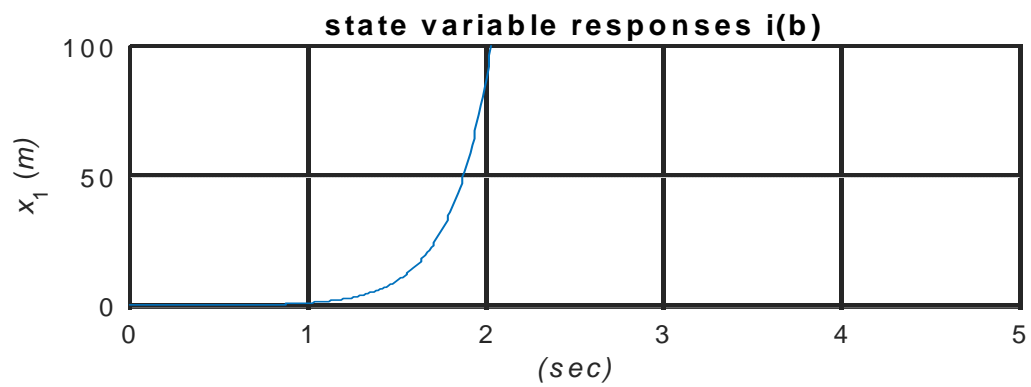
```
B =
      u1
x1      0
x2      0.5
x3      0
x4  0.6667
```

```
C =
      x1  x2  x3  x4
y1      0   0   1   0
```

```
D =
      u1
y1      0
```

Continuous-time state-space model.

```
>>
```

Project Part 2 a_ii:

Single input, multiple output: impulse input $f(t)$ & two outputs

Code:

```
%Yevgen Solodkyy
%CE2.2a, part ii

clear all;
clc;
m1=2; m2=1; L=.75; g=9.81;

A = [0 1 0 0; 0 0 m2*g/m1 0; 0 0 0 1; 0 0 (m1+m2)*g/(m1*L) 0];

D = 0;

t=[0:.01:7];

x0 = [0;0;0;0];

impulse = t==0;

B = [0; 1/m1; 0; 1/(L*m1)]; % need to verify what this matrix looks
like.
C = [1 0 0 0; 0 0 1 0]; % two outputs

u=[impulse];

sys_ii = ss(A,B,C,D)

[Y0,t,X0] = lsim(sys_ii,u,t,x0);

%PLOTting PORTION

figure(3);
set(gcf,'numbertitle','off','name','state variable responses ii') %
figure title
subplot(411),
plot(t,X0(:,1)); grid; %plot the first state variable
title('state variable responses ii')
axis([0 5 -1 100]);
set(gca, 'FontSize',12);
xlabel('\it{time(sec)}'); ylabel('{\it{x}}_1 ({\it{m}})'); % might want to
remove the x-label here

subplot(412), plot(t,X0(:,2)); grid; %plot the second state variable
```

```

axis([0 5 -1 100]);
set(gca,'FontSize',12);
xlabel('\ittime(sec)'); ylabel('{\itx}_2 (\itm/s)');

subplot(413), plot(t,X0(:,3)); grid; %plot the third state variable
axis([0 5 -1 100]);
set(gca, 'FontSize',12);
xlabel('\ittime(sec)'); ylabel('{\itx}_3 (\itrad)'); % might want to
remove the x-lalel here

subplot(414), plot(t,X0(:,4)); grid; %plot the second variable
axis([0 5 -1 100]);
set(gca, 'FontSize',12);
xlabel('\ittime(sec)'); ylabel('{\itx}_4 (\itrad/s)'); % might want to
remove the x-lalel here

```

Command Output:

sys_ii =

```

A =
      x1      x2      x3      x4
x1      0      1      0      0
x2      0      0  4.905      0
x3      0      0      0      1
x4      0      0 19.62      0

```

```

B =
      u1
x1      0
x2     0.5
x3      0
x4  0.6667

```

```

C =
      x1  x2  x3  x4
y1      1   0   0   0
y2      0   0   1   0

```

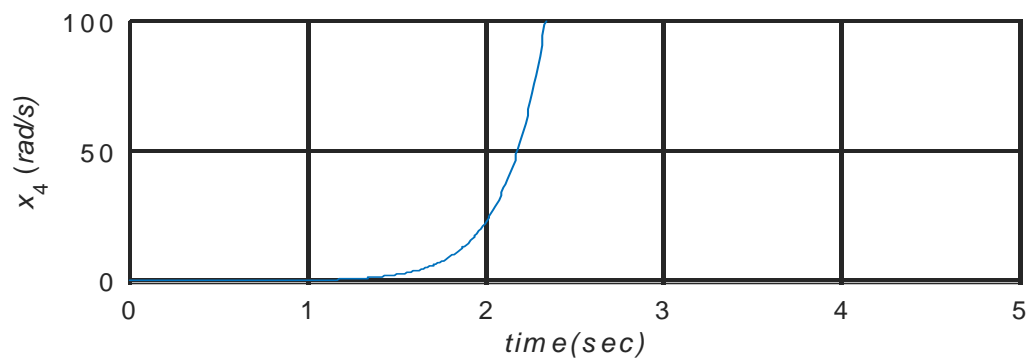
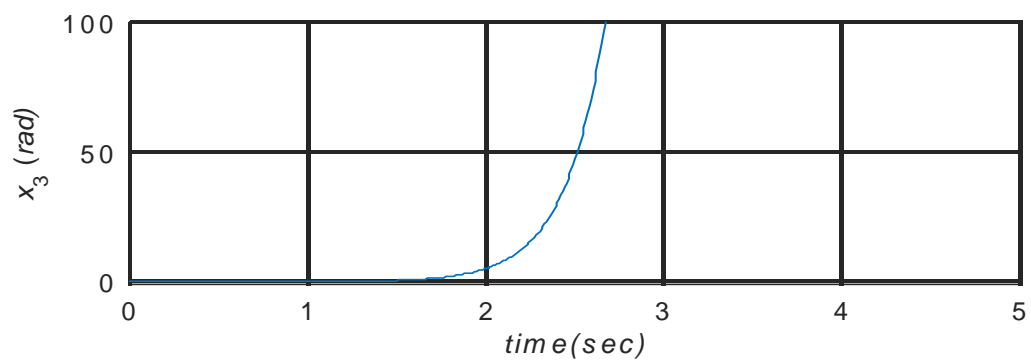
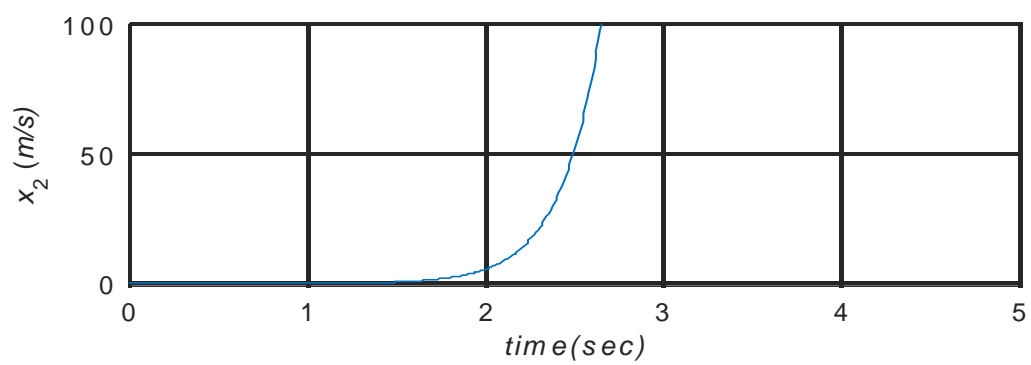
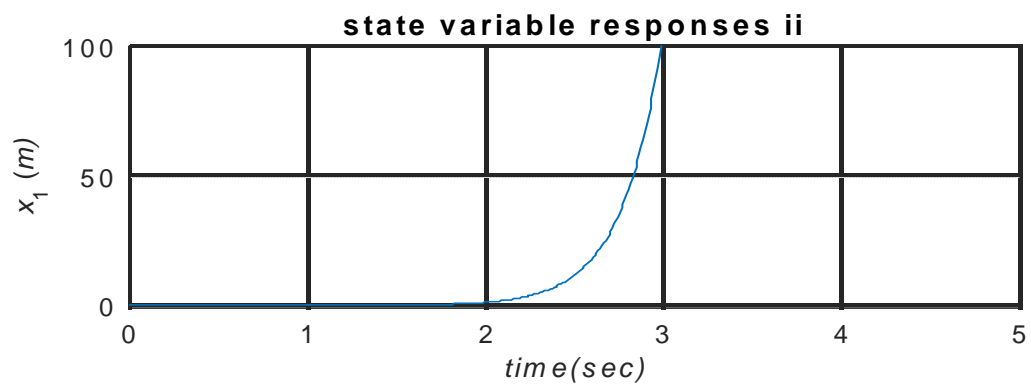
```

D =
      u1
y1      0
y2      0

```

Continuous-time state-space model.

part a_ii plots:



Project Part 2_iii:

Multiple-input multiple-output :
two unit step inputs & two outputs.

Code:

```
%CE2.2a, part III

clear all;
clc;
m1=2; m2=1; L=.75; g=9.81;

A = [0 1 0 0; 0 0 m2*g/m1 0; 0 0 0 1; 0 0 (m1+m2)*g/(m1*L) 0];

D = 0;

t=[0:.01:3600];

x0 = [0;0;0;0];

%impulse = t==0;

B = [0 0; 1/m1 1/(L*m1); 0 0; 1/m1*L (m2+m1)/(m1*m2*L^2)] % need to
verify what this matrix looks like.
C = [1 0 0 0; 0 0 1 0]; % two outputs

u=[ones(size(t)); ones(size(t))];%[zeros(size(t)); zeros(size(t))];

sys_iii = ss(A,B,C,D)

[Y0,t,X0] = lsim(sys_iii,u,t,x0);

%PLOTting PORTION%{
figure(4);
set(gcf,'numbertitle','off','name','state variable responses III') %
figure title
subplot(411),
plot(t,X0(:,1)); grid; %plot the first state variable
title('state variable responses iii')
axis([0 5 -1 100]);
set(gca, 'FontSize',12);
xlabel('\ittime(sec)'); ylabel('{\itx}_1 ({itm}'); % might want to
remove the x-label here

subplot(412), plot(t,X0(:,2)); grid; %plot the second state variable
```

```

axis([0 5 -1 100]);
set(gca,'FontSize',12);
xlabel('\ittime(sec)'); ylabel('{\itx}_2 (\itm/s)');

subplot(413), plot(t,X0(:,3)); grid; %plot the third state variable
axis([0 5 -1 100]);
set(gca, 'FontSize',12);
xlabel('\ittime(sec)'); ylabel('{\itx}_3 (\itrad)'); % might want to
remove the x-lalel here

subplot(414), plot(t,X0(:,4)); grid; %plot the second variable
axis([0 5 -1 100]);
set(gca, 'FontSize',12);
xlabel('\ittime(sec)'); ylabel('{\itx}_4 (\itrad/s)'); % might want to
remove the x-lalel here%}

```

Command Output:

B =

	0	0
0.5000	0.6667	
0	0	
0.3750	2.6667	

sys_iii =

A =

	x1	x2	x3	x4
x1	0	1	0	0
x2	0	0	4.905	0
x3	0	0	0	1
x4	0	0	19.62	0

B =

	u1	u2
x1	0	0
x2	0.5	0.6667
x3	0	0
x4	0.375	2.667

C =

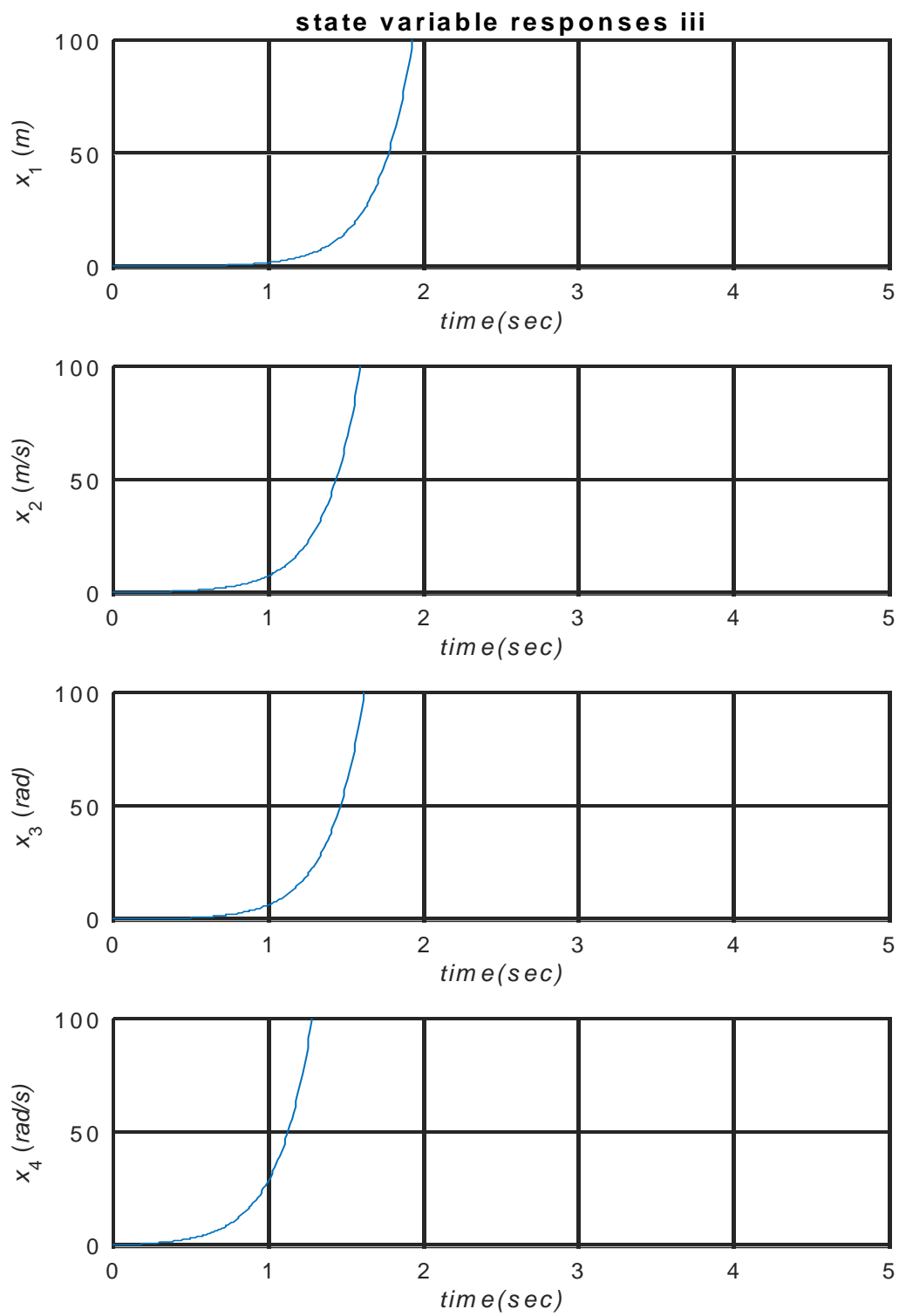
	x1	x2	x3	x4
y1	1	0	0	0
y2	0	0	1	0

D =

	u1	u2
y1	0	0
y2	0	0

Continuous-time state-space model.

Part a_iii Plots:



Project Part 3:

Code:

```
%Yevgen Solodkyy
%Project Part 3

clc;
m1=2; m2=1; L=.75; g=9.81;

A = [0 1 0 0; 0 0 m2*g/m1 0; 0 0 0 1; 0 0 (m1+m2)*g/(m1*L) 0];

D = 0;

t=[0:.01:7];

x0 = [0;0;0;0];

impulse = t==0;

B = [0; 1/m1; 0; 1/(L*m1)]; % need to verify what this matrix looks
like.
C = [0 0 1 0];

u=[impulse; zeros(size(t))]; %zeros(size(t)) is t x 1 dimentions.

sys_ia = ss(A,B,C,D)

det_A= det(A)% |A| =0 -> cannont be diagonalized. The diagonal canonical
form cannot be obtained.

display('|A| =0 -> Diagonal Canonitcal Form cannot be obtained.')
```

Command Output:

```
sys_ia =
```

```
A =
```

	x1	x2	x3	x4
x1	0	1	0	0
x2	0	0	4.905	0
x3	0	0	0	1
x4	0	0	19.62	0

```
B =
```

	u1

```

x1      0
x2      0.5
x3      0
x4      0.6667

C =
      x1  x2  x3  x4
y1      0   0   1   0

D =
      u1
y1      0

```

Continuous-time state-space model.

```

det_A =

      0

|A| =0 -> Diagonal Canonitcal Form cannot be obtained.
>>

```

Project Part 4: assess system controllability

Code:

```

%Yevgen Solodkyy
%project part 4:
clear all;
clc;
m1=2; m2=1; L=.75; g=9.81;
display('is the system controllable? If rank(P) = n yes; if rank(P) < n,
no.')

A = [0 1 0 0; 0 0 m2*g/m1 0; 0 0 0 1; 0 0 (m1+m2)*g/(m1*L) 0];

%part i-ii
B = [0; 1/m1; 0; 1/(L*m1)];

P = [B A*B A^2*B A^3*B]

RANK =rank(P)
display('rank(P) = 4, so the system is controllable')
%part iii
clear B;

B = [0 0; 1/m1 1/(L*m1); 0 0; 1/m1*L (m2+m1)/(m1*m2*L^2)];
P = [B A*B A^2*B A^3*B]

```

```
RANK=rank(P)
display('rank(P) = 4, so the system is controllable')
```

Command Output:

is the system controllable? If rank(P) = n yes; if rank(P) < n, no.

P =

0	0.5000	0	3.2700
0.5000	0	3.2700	0
0	0.6667	0	13.0800
0.6667	0	13.0800	0

RANK =

4

rank(P) = 4, so the system is controllable

P =

0	0	0.5000	0.6667	0	0
1.8394	13.0800				
0.5000	0.6667	0	0	1.8394	13.0800
0	0				
0	0	0.3750	2.6667	0	0
7.3575	52.3200				
0.3750	2.6667	0	0	7.3575	52.3200
0	0				

RANK =

4

rank(P) = 4, so the system is controllable

>>

Project Part 5:

Calculate the CCF for the sytem
Assessing system observability

Code:

```
%Yevgen Solodkyy
%Project part 5;

clear all; clc;

display(' Problem CME3.2b: calculate the CCF for the sytem')
%calculate cotroller canonical form for the system
m1=2; m2=1; L=.75; g=9.81;

A = [0 1 0 0; 0 0 m2*g/m1 0; 0 0 0 1; 0 0 (m1+m2)*g/(m1*L) 0];
B = [0; 1/m1; 0; 1/(L*m1)];
C = [0 0 1 0];
C_ii = [1 0 0 0; 0 0 1 0]
C_iii= [1 0 0 0; 0 0 1 0]

CharPoly = poly(A)
a3 = CharPoly(4)
a2 = CharPoly(3)
a1 = CharPoly(2)
a0 = CharPoly(1)

Pccf_i=[a1 a2 a3 1;a2 a3 1 0; a3 1 0 0; 1 0 0 0]
P=[B A*B A^2*B A^3*B]
Tccf=P*Pccf_i

Accf=Tccf^-1*A*Tccf
Bccf=Tccf^-1*B
Cccf=C*Tccf

%observability

display('CME 4.2a: Assessing system observability')

Q = [C;C*A;C*A^2; C*A^3]

detQ=det(Q)

display('|Q|=0, the system in cae (i) is not observeble. The system can
be reduced to an observable rank(Q)-size system:')
rank_Q = rank(Q)

T_tilda= [0,0,1,0;0,0,0,1;0,1,19.62,0;1,0,0,19.62]
```

```

T =(T_tilda')^-1

A_tilda = T^-1*A*T
C_tilda = C*T
display('The observable portion of the system is:')

A11= [0 19.62; 1 0]
C1 = [0 1]
B1=[0;1]

ObsrSYS = ss(A11,B1,C1,0)

%observability of parts (ii) & (iii)

display('the systems in cases (ii) & (iii) are observable as both Qii &
Qiii are rank 4:')

Qii = [C_ii; C_ii*A; C_ii*A^2; C_ii*A^3]
display('rank(Qii):')
rank(Qii)

Qiii = [C_iii; C_iii*A; C_iii*A^2; C_iii*A^3]
display('rank(Qiii):')
rank(Qiii)

```

command output:

Problem CME3.2b: calculate the CCF for the sytem

C_ii =

1	0	0	0
0	0	1	0

C_iii =

1	0	0	0
0	0	1	0

CharPoly =

1.0000	0	-19.6200	0	0
--------	---	----------	---	---

a3 =

0

a2 =

-19.6200

a1 =

0

a0 =

1

Pccf_i =

0	-19.6200	0	1.0000
-19.6200	0	1.0000	0
0	1.0000	0	0
1.0000	0	0	0

P =

0	0.5000	0	3.2700
0.5000	0	3.2700	0
0	0.6667	0	13.0800
0.6667	0	13.0800	0

Tccf =

-6.5400	0	0.5000	0
0	-6.5400	0	0.5000
0	0	0.6667	0
0	0	0	0.6667

Accf =

0	1.0000	0	0
0	0	1.0000	0
0	0	0	1.0000
0	0	19.6200	0

Bccf =

0
0
0

1

Cccf =

0	0	0.6667	0
---	---	--------	---

CME 4.2a: Assessing system observability

Q =

0	0	1.0000	0
0	0	0	1.0000
0	0	19.6200	0
0	0	0	19.6200

detQ =

0

$|Q|=0$, the system in cae (i) is not observable. The system can be reduced to an observable rank(Q)-size system:

rank_Q =

2

T_tilda =

0	0	1.0000	0
0	0	0	1.0000
0	1.0000	19.6200	0
1.0000	0	0	19.6200

T =

0	-19.6200	1.0000	0
-19.6200	0	0	1.0000
0	1.0000	0	0
1.0000	0	0	0

A_tilda =

0	19.6200	0	0
1.0000	0	0	0
0	0	0	1.0000
0	389.8494	0	0

C_tilda =

0 1 0 0

The observable portion of the system is:

A11 =

0 19.6200
1.0000 0

C1 =

0 1

B1 =

0
1

ObsrSYS =

A =

	x1	x2
x1	0	19.62
x2	1	0

B =

	u1
x1	0
x2	1

C =

	x1	x2
y1	0	1

D =

	u1
y1	0

Continuous-time state-space model.

the systems in cases (ii) & (iii) are observable as both Qii & Qiii are rank 4:

Qii =

1.0000	0	0	0
0	0	1.0000	0
0	1.0000	0	0


```

0      0      0      1.0000
0      0      4.9050      0
0      0     19.6200      0
0      0      0      4.9050
0      0      0     19.6200

```

```
rank(Qii):
```

```
ans =
```

```
4
```

```
Qiii =
```

```

1.0000      0      0      0
0      0      1.0000      0
0     1.0000      0      0
0      0      0      1.0000
0      0      4.9050      0
0      0     19.6200      0
0      0      0      4.9050
0      0      0     19.6200

```

```
rank(Qiii):
```

```
ans =
```

```
4
```

```
>>
```

```
>>
```

Project Part 6:

Assess the stability properties of the CE2.2 system

```

%Yevgen Solodkyy
%project part 6
%assess the stability properties of the CE2.2 system

```

Code:

```

clear all;
clc;

display('assess the stability properties of the system')

```

```

m1=2; m2=1; L=.75; g=9.81;

A = [0 1 0 0; 0 0 m2*g/m1 0; 0 0 0 1; 0 0 (m1+m2)*g/(m1*L) 0]

display('Eigen value approach:')

eig(A)

display('because Re{eigen values} are not strictly negative, the
Lyapunov approach will fail.')

display('Lyapunov approach:')

Q = eye(4)
S=lyap(A',Q)
display(' |s(1,1)| :')
det(S(1,1))
display(' |s(1:2,1:2)| :')
det(S(1:2,1:2))
display(' |s(1:3,1:3)| :')
det(S(1:3,1:3))
display(' |s(1:3,1:3)| :')
det(S)

```

Command Output:

assess the stability properties of the system

A =

```

0  1.0000    0    0
0   0  4.9050    0
0   0    0  1.0000
0   0 19.6200    0

```

Eigen value approach:

ans =

```
0
0
4.4294
-4.4294
```

because **Re{eigen values}** are not strictly negative, the Lyapunov approach will fail.

Lyapunov approach:

Q =

```
1  0  0  0
0  1  0  0
0  0  1  0
0  0  0  1
```

Error using lyap (line 69)

The solution of this Lyapunov equation does not exist or is not unique.

Error in Project_part6 (line 25)

S=lyap(A',Q)

>>

Project part 7:

Design state feedback control laws for closed loop eigenvalues

7-i(a):

code:

```
%Yevgen Solodkyy
%Project part 7;
%design state feedback control laws for closed loop eigenvalues.

clear all; clc;

display('Design state feedback control laws for closed loop
eigenvalues')

clear all;
clc;
m1=2; m2=1; L=.75; g=9.81;

A = [0 1 0 0; 0 0 m2*g/m1 0; 0 0 0 1; 0 0 (m1+m2)*g/(m1*L) 0];
B_i = [0; 1/m1; 0; 1/(L*m1)];
B_iii = [0 0; 1/m1 1/(L*m1); 0 0; 1/m1*L (m2+m1)/(m1*m2*L^2)]; % need
to verify again the last value of B.
C = [0 0 1 0];
C_ii = [1 0 0 0; 0 0 1 0];
C_iii= [1 0 0 0; 0 0 1 0];
D=0;

%desired eigven values
l1= -1.27+3.79i;
l2= -1.27-3.79i;
l3= -1.88+1.24i;
l4= -1.88-1.24i;

DesEig = [l1; l2; l3; l4]

%open loop state space for part i:
Sys_ol_i = ss(A,B_i,C,D)

% K for part i:
K_i=place(A,B_i,DesEig)

Ac = A-B_i*K_i;

%closed loop state space for part i:
Sys_cl_i = ss(Ac,B_i,C,D)

[Yc,t,Xc] = impulse(Sys_cl_i);
[Y0,t0,X0] = impulse(Sys_ol_i);

figure(1)
```

```

subplot(411),
plot(t,Xc(:,1),'green',tO,XO(:,1),'red');
grid;
axis([0 12 -.3 .3]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_1 (\itm)');
legend('Closed-loop','Open-loop');
title('7 i(a): Open vs. closed-loop responses', 'FontSize',14);

subplot(412),
plot(t,Xc(:,2),'green',tO,XO(:,2),'red');
grid;
axis([0 12 -.6 1]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_2 (\itm/s)');
legend('Closed-loop','Open-loop');

subplot(413),
plot(t,Xc(:,3),'green',tO,XO(:,3),'red');
grid;
axis([0 12 -.3 .3]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_3 (\itrad)');
legend('Closed-loop','Open-loop');

subplot(414),
plot(t,Xc(:,3),'green',tO,XO(:,3),'red');
grid;
axis([0 12 -.3 .3]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_3 (\itrad)');
legend('Closed-loop','Open-loop');

figure(2)
%subplot(411),
plot(t,Yc(:,1),'green',tO,YO(:,1),'red');
grid;
axis([0 12 -.3 .3]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\ity}_1 (\itm)');
legend('Closed-loop','Open-loop');
title('7 i(a): Open/closed-loop responses', 'FontSize',14);

```

command output:

DesEig =

-1.2700 + 3.7900i

-1.2700 - 3.7900i

-1.8800 + 1.2400i

-1.8800 - 1.2400i

Sys_ol_i =

A =

	x1	x2	x3	x4
x1	0	1	0	0
x2	0	0	4.905	0
x3	0	0	0	1
x4	0	0	19.62	0

B =

	u1
x1	0
x2	0.5

x3 0

x4 0.6667

C =

x1 x2 x3 x4

y1 0 0 1 0

D =

u1

y1 0

Continuous-time state-space model.

K_i =

-12.3907 -11.1554 84.6221 17.8166

Sys_cl_i =

A =

x1 x2 x3 x4

x1 0 1 0 0

x2 6.195 5.578 -37.41 -8.908

x3 0 0 0 1

x4 8.26 7.437 -36.79 -11.88

B =

u1

x1 0

x2 0.5

x3 0

x4 0.6667

C =

x1 x2 x3 x4

y1 0 0 1 0

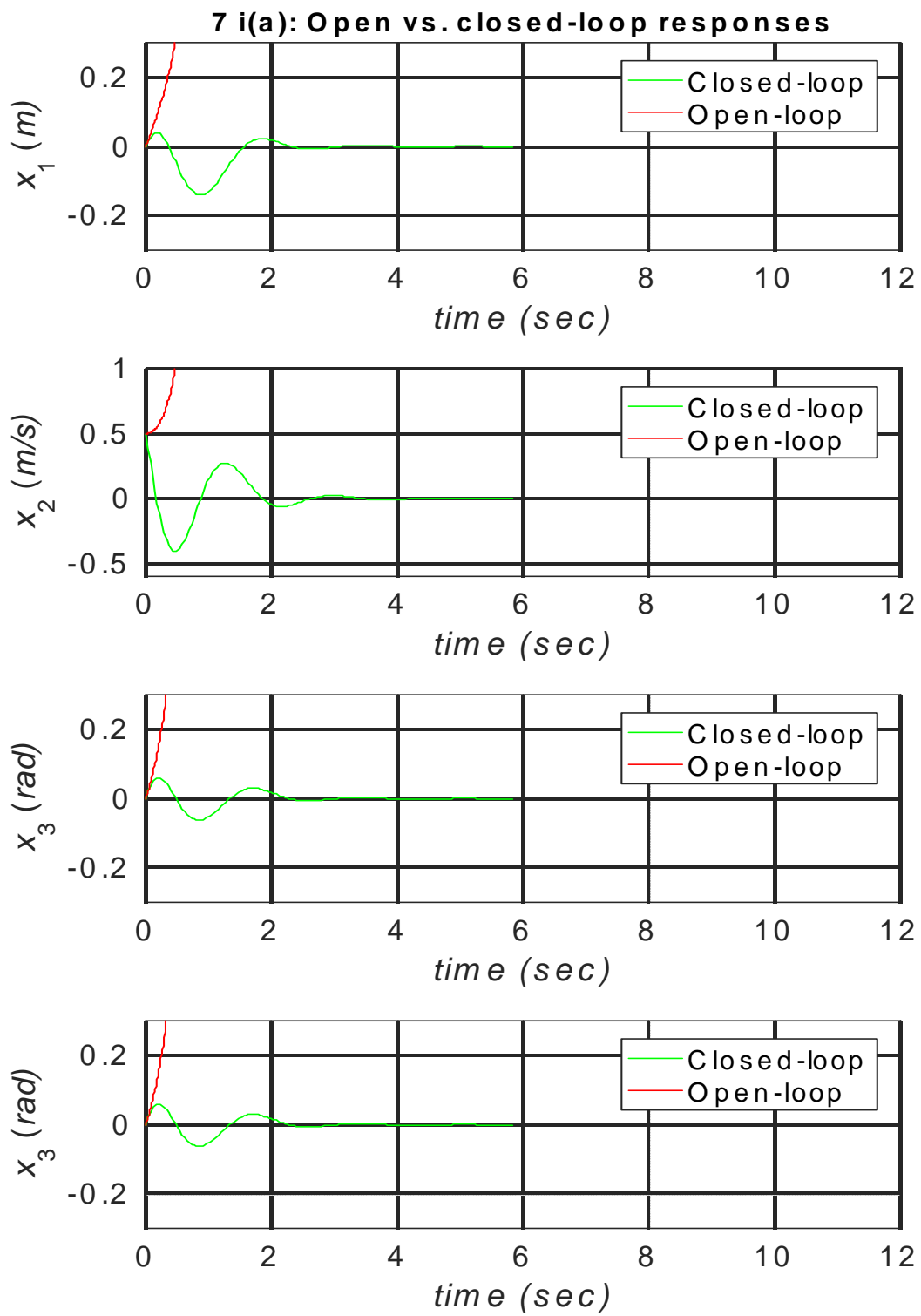
D =

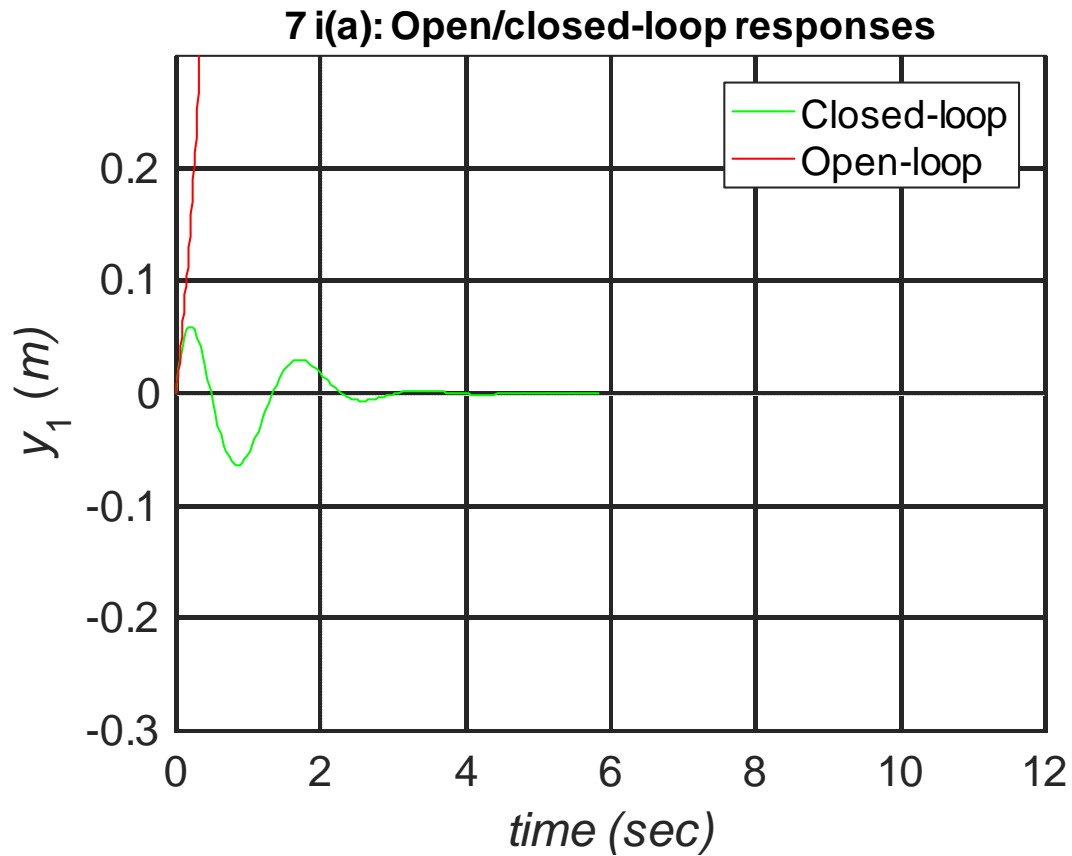
u1

y1 0

Continuous-time state-space model.

7-i(a) Plots:





part 7-i(b):

code:

```
%Yevgen Solodkyy
%Project part 7;
%design state feedback control laws for closed loop eigenvalues.

clear all; clc;

display('Design state feedback control laws for closed loop
eigenvalues')

clear all;
clc;
m1=2; m2=1; L=.75; g=9.81;

A = [0 1 0 0; 0 0 m2*g/m1 0; 0 0 0 1; 0 0 (m1+m2)*g/(m1*L) 0];
B_i = [0; 1/m1; 0; 1/(L*m1)];
%B_iii = [0 0; 1/m1 1/(L*m1); 0 0; 1/m1*L (m2+m1)/(m1*m2*L^2)]; % need
to verify again the last value of B.
C = [0 0 1 0];
```

```

%C_ii = [1 0 0 0;0 0 1 0]
%C_iii= [1 0 0 0;0 0 1 0]
D=0;

%desired eigven values
l1= -1.27+3.79i;
l2= -1.27-3.79i;
l3= -1.88+1.24i;
l4= -1.88-1.24i;

DesEig = [l1; l2; l3; l4]

%open loop state space for part i:
Sys_oli = ss(A,B_i,C,D)

% K for part i:
K_i=place(A,B_i,DesEig)

Ac = A-B_i*K_i;

%closed loop state space for part i:
Sys_cl = ss(Ac,B_i,C,D)

t1 = 0:.1:7;

X0 = [0;0;0.1;0];

u=[zeros(size(t1))];

[Yc,t,Xc] = lsim(Sys_cl,u,t1,X0);

[Y0,t0,X0] = lsim(Sys_oli,u,t1,X0);

figure(1);
subplot(411),
plot(t,Xc(:,1),'green',t0,X0(:,1),'red');
grid;
axis([0 12 -.3 .3]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_1 (\itm)');
legend('Closed-loop','Open-loop');
title('7 i-(b): Open/closed-loop responses', 'FontSize',14);

subplot(412),

```

```

plot(t,Xc(:,2),'green',t0,X0(:,2),'red');
grid;
axis([0 12 -.6 .5]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_2 (\itm/s)');
legend('Closed-loop','Open-loop');

subplot(413),
plot(t,Xc(:,3),'green',t0,X0(:,3),'red');
grid;
axis([0 12 -.3 .3]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_3 (\itrad)');
legend('Closed-loop','Open-loop');

subplot(414),
plot(t,Xc(:,3),'green',t0,X0(:,3),'red');
grid;
axis([0 12 -.3 .3]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_3 (\itrad)');
legend('Closed-loop','Open-loop');

figure(2)

plot(t,Yc(:,1),'green',t0,Y0(:,1),'red');
grid;
axis([0 12 -.3 .3]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\ity}_1 (\itm)');
legend('Closed-loop','Open-loop');
title('7 i(b): Open/closed-loop responses', 'FontSize',14);

```

command output:

DesEig =

-1.2700 + 3.7900i

-1.2700 - 3.7900i

-1.8800 + 1.2400i

-1.8800 - 1.2400i

Sys_oli =

A =

	x1	x2	x3	x4
x1	0	1	0	0
x2	0	0	4.905	0
x3	0	0	0	1
x4	0	0	19.62	0

B =

	u1
x1	0
x2	0.5
x3	0
x4	0.6667

C =

	x1	x2	x3	x4
y1	0	0	1	0

D =

u1

y1 0

Continuous-time state-space model.

K_i =

-12.3907 -11.1554 84.6221 17.8166

Sys_cl =

A =

	x1	x2	x3	x4
x1	0	1	0	0
x2	6.195	5.578	-37.41	-8.908
x3	0	0	0	1
x4	8.26	7.437	-36.79	-11.88

B =

u1

x1 0

x2 0.5

x3 0

x4 0.6667

C =

x1 x2 x3 x4

y1 0 0 1 0

D =

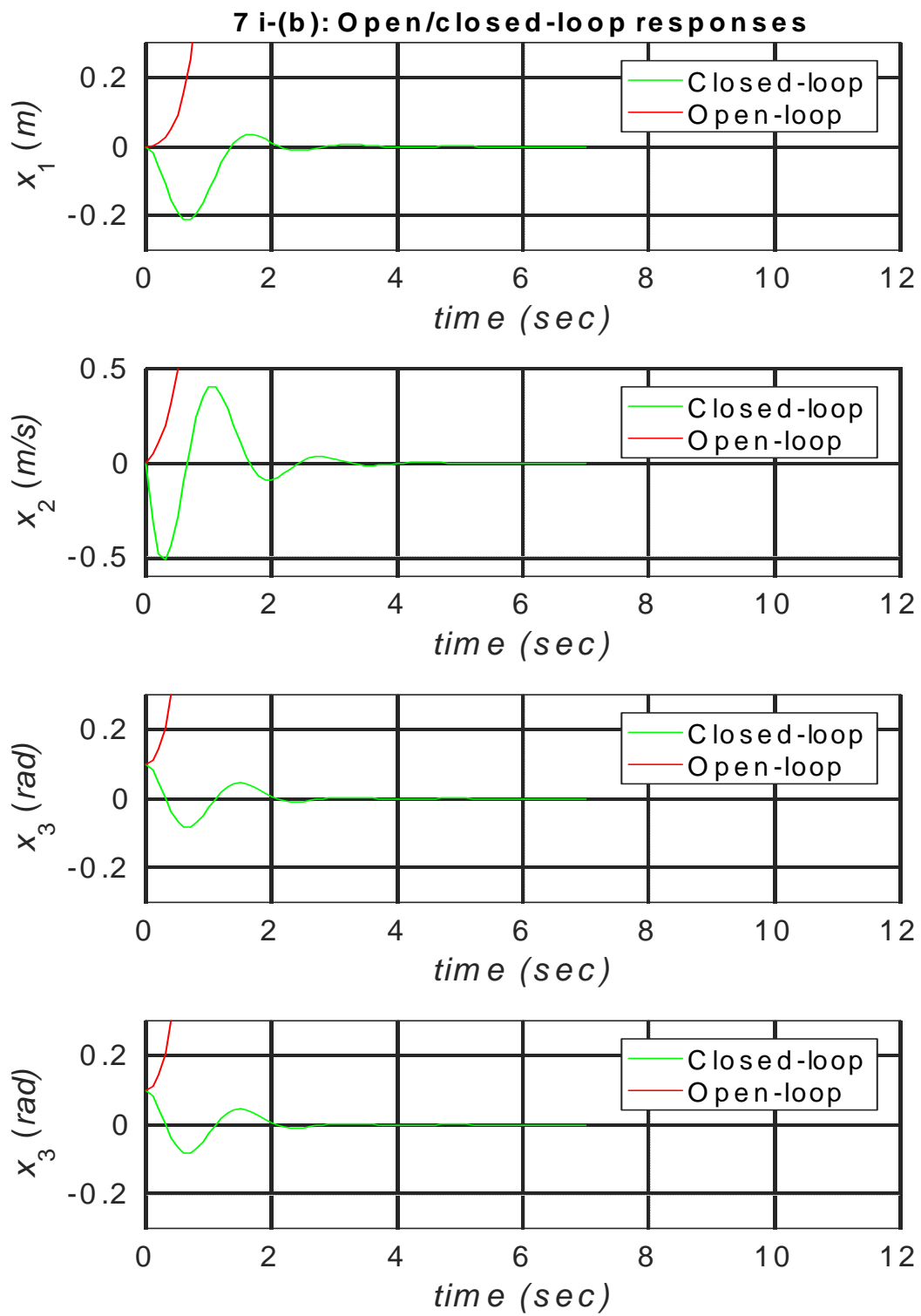
u1

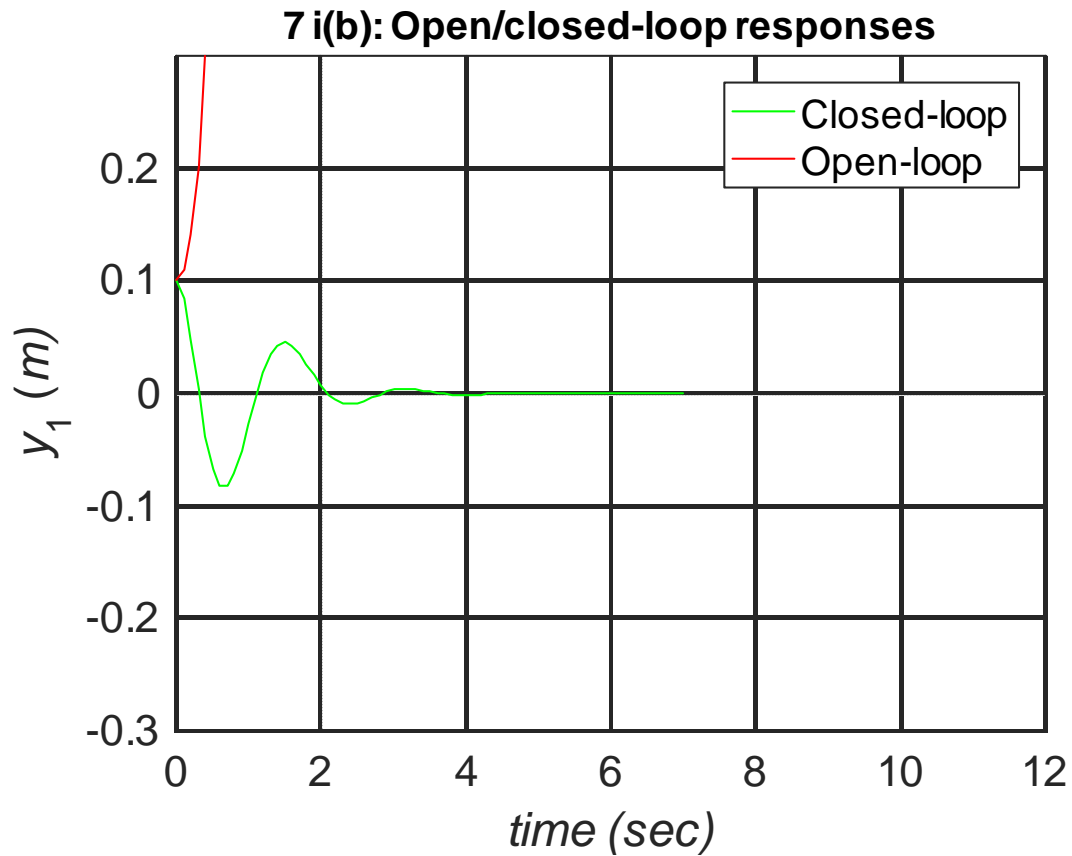
y1 0

Continuous-time state-space model.

>>

Plots:





part 7-ii:

```
%Yevgen Solodkyy
%Project part 7;
%design state feedback control laws for closed loop eigenvalues.

clear all; clc;

display('Design state feedback control laws for closed loop
eigenvalues')

clear all;
clc;
m1=2; m2=1; L=.75; g=9.81;

A = [0 1 0 0; 0 0 m2*g/m1 0; 0 0 0 1; 0 0 (m1+m2)*g/(m1*L) 0];
B_i = [0; 1/m1; 0; 1/(L*m1)];
B_iii = [0 0; 1/m1 1/(L*m1); 0 0; 1/m1*L (m2+m1)/(m1*m2*L^2)]; % need
to verify again the last value of B.
C = [0 0 1 0];
```

```

C_ii = [1 0 0 0;0 0 1 0]
C_iii= [1 0 0 0;0 0 1 0]
D=0;

%desired eigven values
l1= -1.27+3.79i;
l2= -1.27-3.79i;
l3= -1.88+1.24i;
l4= -1.88-1.24i;

DesEig = [l1; l2; l3; l4]

% K for part i:
K_i=place(A,B_i,DesEig)

K_ii = K_i % parts i & ii have the same B matrix.

%open look state space for part ii:
Sys_ol_ii = ss(A,B_i,C_ii,D)

% K for part i:

Ac_ii = A-B_i*K_ii;

Sys_cl_ii = ss(Ac_ii,B_i,C_ii,D)

[Yc,t,Xc] = impulse(Sys_cl_ii);
[Y0,t0,X0] = impulse(Sys_ol_ii);

figure(1)
subplot(411),
plot(t,Xc(:,1), 'green',t0,X0(:,1), 'red');
grid;
axis([0 12 -.3 .3]);
set(gca, 'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_1 (\itm)');
legend('Closed-loop', 'Open-loop');
title('7 (i): Open vs. closed-loop responses', 'FontSize',14);

subplot(412),
plot(t,Xc(:,2), 'green',t0,X0(:,2), 'red');
grid;
axis([-1 12 -.5 1]);
set(gca, 'FontSize',16);
xlabel('\ittime (sec)');

```

```

ylabel('{\itx}_2 (\itm/s)');
legend('Closed-loop', 'Open-loop');

subplot(413),
plot(t,Xc(:,3), 'green', tO,XO(:,3), 'red');
grid;
axis([0 12 -.3 .3]);
set(gca, 'FontSize', 16);
xlabel('\ittime (sec)');
ylabel('{\itx}_3 (\itrad)');
legend('Closed-loop', 'Open-loop');

subplot(414),
plot(t,Xc(:,3), 'green', tO,XO(:,3), 'red');
grid;
axis([0 12 -.3 .3]);
set(gca, 'FontSize', 16);
xlabel('\ittime (sec)');
ylabel('{\itx}_3 (\itrad)');
legend('Closed-loop', 'Open-loop');

figure(2)
subplot(211),
plot(t,Yc(:,1), 'green', tO,YO(:,1), 'red');
grid;
axis([0 12 -.3 .3]);
set(gca, 'FontSize', 16);
xlabel('\ittime (sec)');
ylabel('{\ity}_1 (\itm)');
legend('Closed-loop', 'Open-loop');
title('7 ii: Open/closed-loop responses', 'FontSize', 14);

subplot(212),
plot(t,Yc(:,2), 'green', tO,YO(:,2), 'red');
grid;
axis([0 12 -.3 .3]);
set(gca, 'FontSize', 16);
xlabel('\ittime (sec)');
ylabel('{\ity}_2 (\itm)');
legend('Closed-loop', 'Open-loop');
title('7 ii: Open/closed-loop responses', 'FontSize', 14);

```

command output:

C_{ii} =

1	0	0	0
0	0	1	0

DesEig =

-1.2700 + 3.7900i

-1.2700 - 3.7900i

-1.8800 + 1.2400i

-1.8800 - 1.2400i

K_i =

-12.3907 -11.1554 84.6221 17.8166

K_ii =

-12.3907 -11.1554 84.6221 17.8166

Sys_ol_ii =

A =

	x1	x2	x3	x4
x1	0	1	0	0

x2 0 0 4.905 0

x3 0 0 0 1

x4 0 0 19.62 0

B =

u1

x1 0

x2 0.5

x3 0

x4 0.6667

C =

x1 x2 x3 x4

y1 1 0 0 0

y2 0 0 1 0

D =

u1

y1 0

y2 0

Continuous-time state-space model.

Sys_cl_ii =

A =

	x1	x2	x3	x4
x1	0	1	0	0
x2	6.195	5.578	-37.41	-8.908
x3	0	0	0	1
x4	8.26	7.437	-36.79	-11.88

B =

	u1
x1	0
x2	0.5
x3	0
x4	0.6667

C =

	x1	x2	x3	x4
y1	1	0	0	0
y2	0	0	1	0

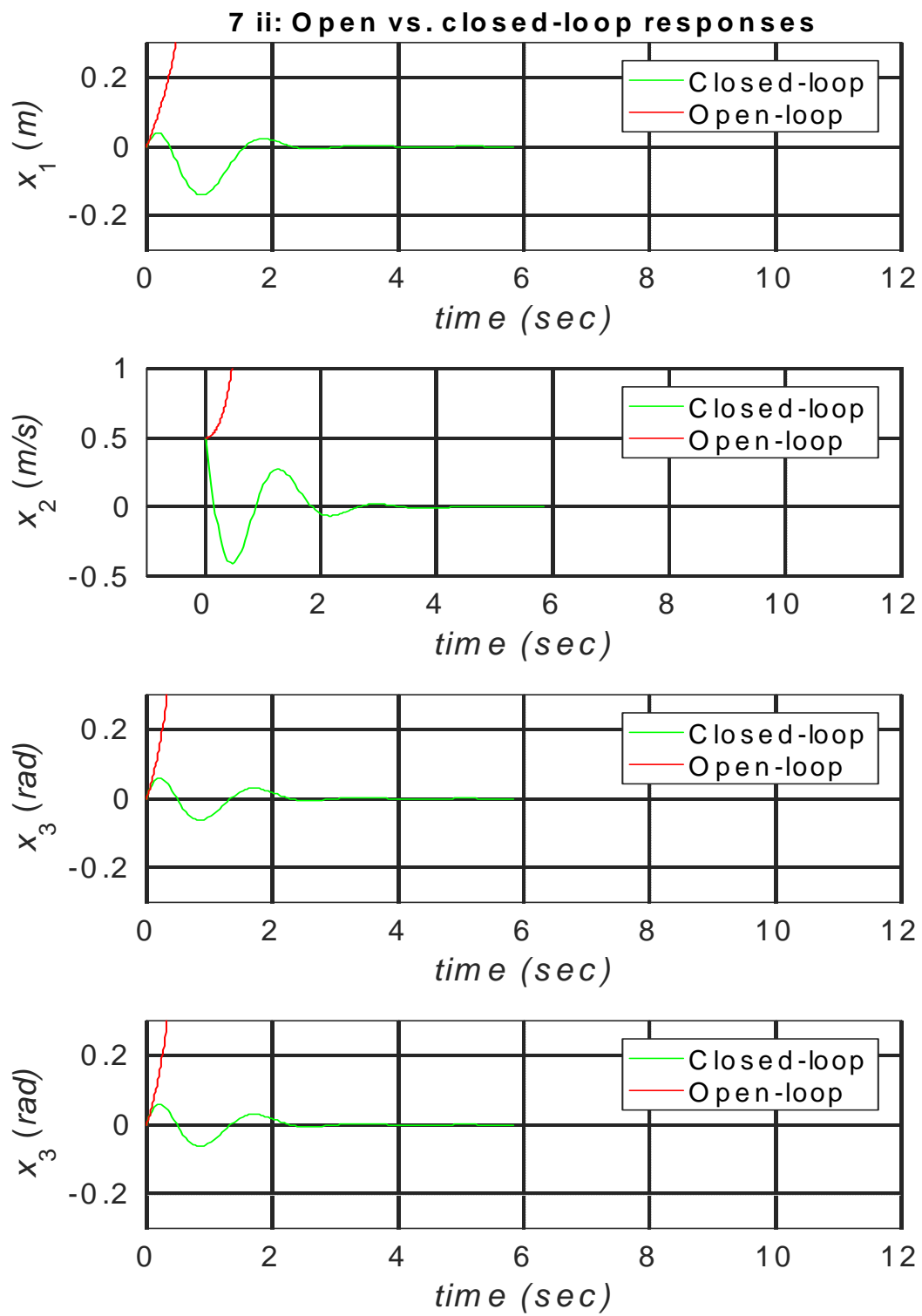
D =

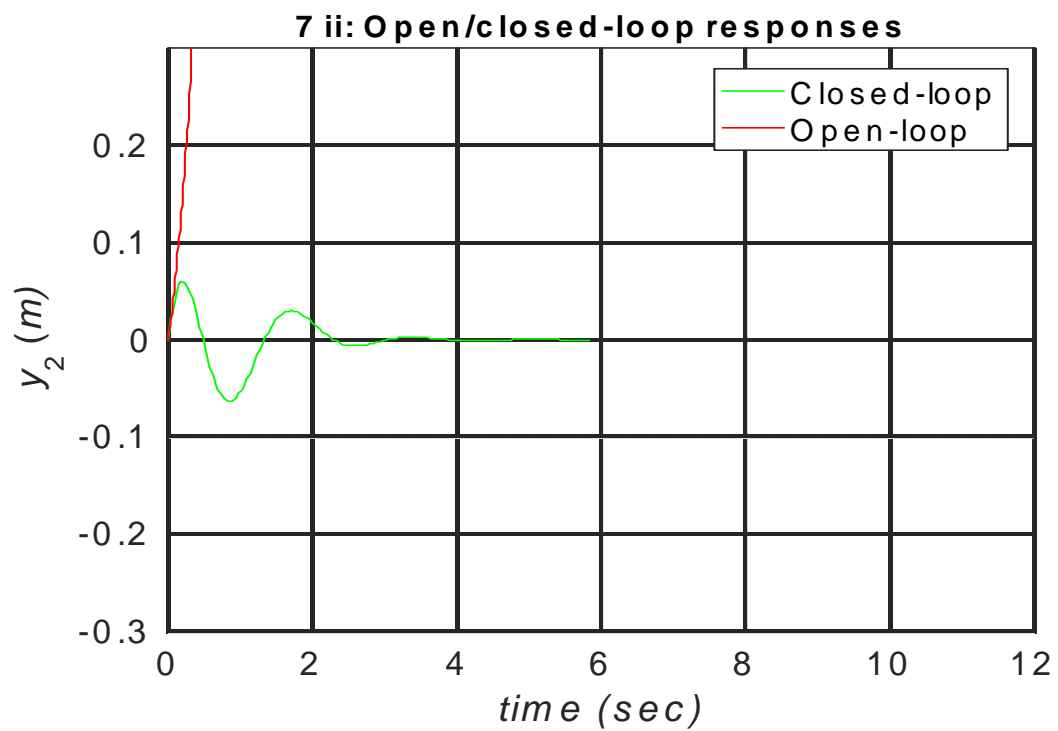
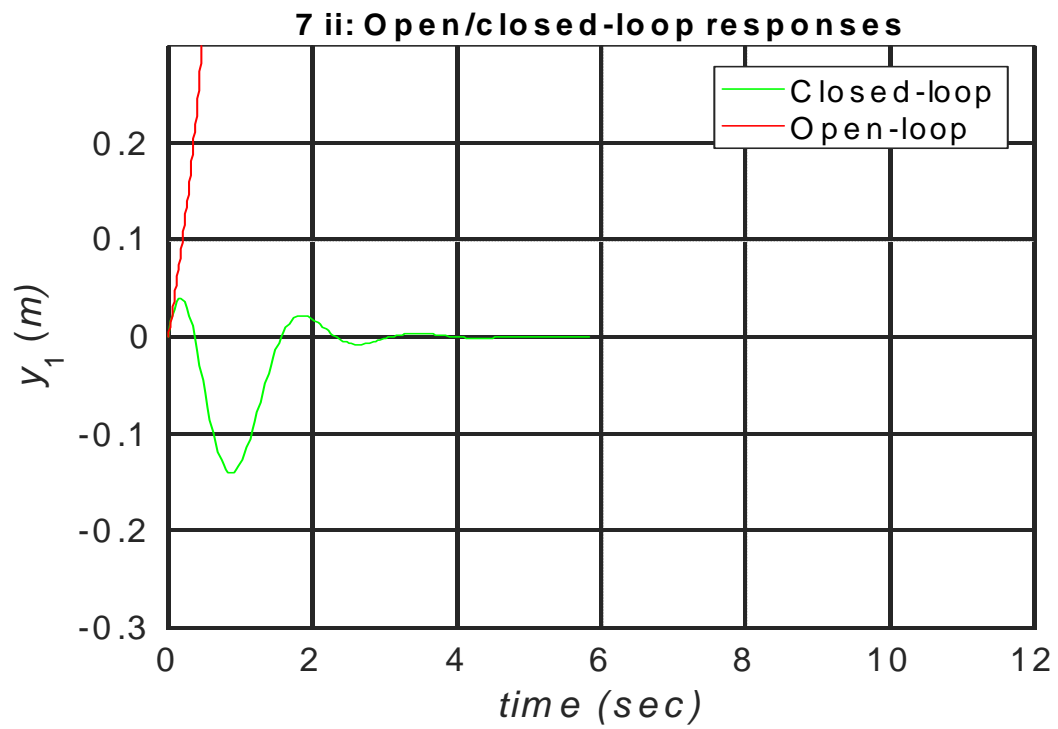
	u1
y1	0
y2	0

Continuous-time state-space model.

>>

Plots:





part 7_iii:

code:

```
%Yevgen Solodkyy
%Project part 7;
%design state feedback control laws for closed loop eigenvalues.

clear all; clc;

display('Design state feedback control laws for closed loop
eigenvalues')

clear all;
clc;
m1=2; m2=1; L=.75; g=9.81;

A = [0 1 0 0; 0 0 m2*g/m1 0; 0 0 0 1; 0 0 (m1+m2)*g/(m1*L) 0];
B_i = [0; 1/m1; 0; 1/(L*m1)];
B_iii = [0 0; 1/m1 1/(L*m1); 0 0; 1/m1*L (m2+m1)/(m1*m2*L^2)]; % need
to verify again the last value of B.
C = [0 0 1 0];
C_ii = [1 0 0 0; 0 0 1 0]
C_iii= [1 0 0 0; 0 0 1 0]
D=0;

%desired eigen values
l1= -1.27+3.79i;
l2= -1.27-3.79i;
l3= -1.88+1.24i;
l4= -1.88-1.24i;

DesEig = [l1; l2; l3; l4]

K_iii=place(A,B_iii,DesEig)

Sys_ol_iii = ss(A,B_iii,C_iii,D)
Ac_iii = A-B_iii*K_iii;
Sys_cl_iii = ss(Ac_iii,B_iii,C_iii,D)

[Yc,t,Xc] = step(Sys_cl_iii);
[Y0,t0,X0] = step(Sys_ol_iii);
```

```

figure()
subplot(411),
plot(t,Xc(:,1),'green',tO,XO(:,1),'red');
grid;
axis([0 12 -.2 .2]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_1 (\itm)');
legend('Closed-loop','Open-loop');
title('7 iii: Open/closed-loop responses', 'FontSize',14);

subplot(412),
plot(t,Xc(:,2),'green',tO,XO(:,2),'red');
grid;
axis([0 12 -.2 .2]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_2 (\itm/s)');
legend('Closed-loop','Open-loop');

subplot(413),
plot(t,Xc(:,3),'green',tO,XO(:,3),'red');
grid;
axis([0 12 -.2 .2]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_3 (\itrad)');
legend('Closed-loop','Open-loop');

subplot(414),
plot(t,Xc(:,3),'green',tO,XO(:,3),'red');
grid;
axis([0 12 -.2 .2]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_3 (\itrad)');
legend('Closed-loop','Open-loop');

figure(2)
subplot(211),
plot(t,Yc(:,1),'green',tO,YO(:,1),'red');
grid;
axis([0 12 -.3 .3]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\ity}_1 (\itm)');
legend('Closed-loop','Open-loop');
title('7 ii: Open/closed-loop responses', 'FontSize',14);

```

```

subplot(212),
plot(t,Yc(:,2),'green',tO,YO(:,2),'red');
grid;
axis([0 12 -.3 .3]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\ity}_2 (\itm)');
legend('Closed-loop','Open-loop');
title('7 iii: Open/closed-loop responses', 'FontSize',14);

```

command output:

C_iii =

```

1  0  0  0
0  0  1  0

```

DesEig =

```

-1.2700 + 3.7900i
-1.2700 - 3.7900i
-1.8800 + 1.2400i
-1.8800 - 1.2400i

```

K_iii =

```

20.8408  9.3247  9.3042  4.3430
-5.0110 -2.2669  8.7096  0.5701

```

Sys_ol_iii =

A =

	x1	x2	x3	x4
x1	0	1	0	0
x2	0	0	4.905	0
x3	0	0	0	1
x4	0	0	19.62	0

B =

	u1	u2
x1	0	0
x2	0.5	0.6667
x3	0	0
x4	0.375	2.667

C =

	x1	x2	x3	x4
y1	1	0	0	0
y2	0	0	1	0

D =

	u1	u2
y1	0	0
y2	0	0

Continuous-time state-space model.

Sys_cl_iii =

A =

	x1	x2	x3	x4
x1	0	1	0	0
x2	-7.08	-3.151	-5.554	-2.552
x3	0	0	0	1
x4	5.547	2.548	-7.095	-3.149

B =

	u1	u2
x1	0	0
x2	0.5	0.6667
x3	0	0
x4	0.375	2.667

C =

	x1	x2	x3	x4
y1	1	0	0	0
y2	0	0	1	0

D =

u1 u2

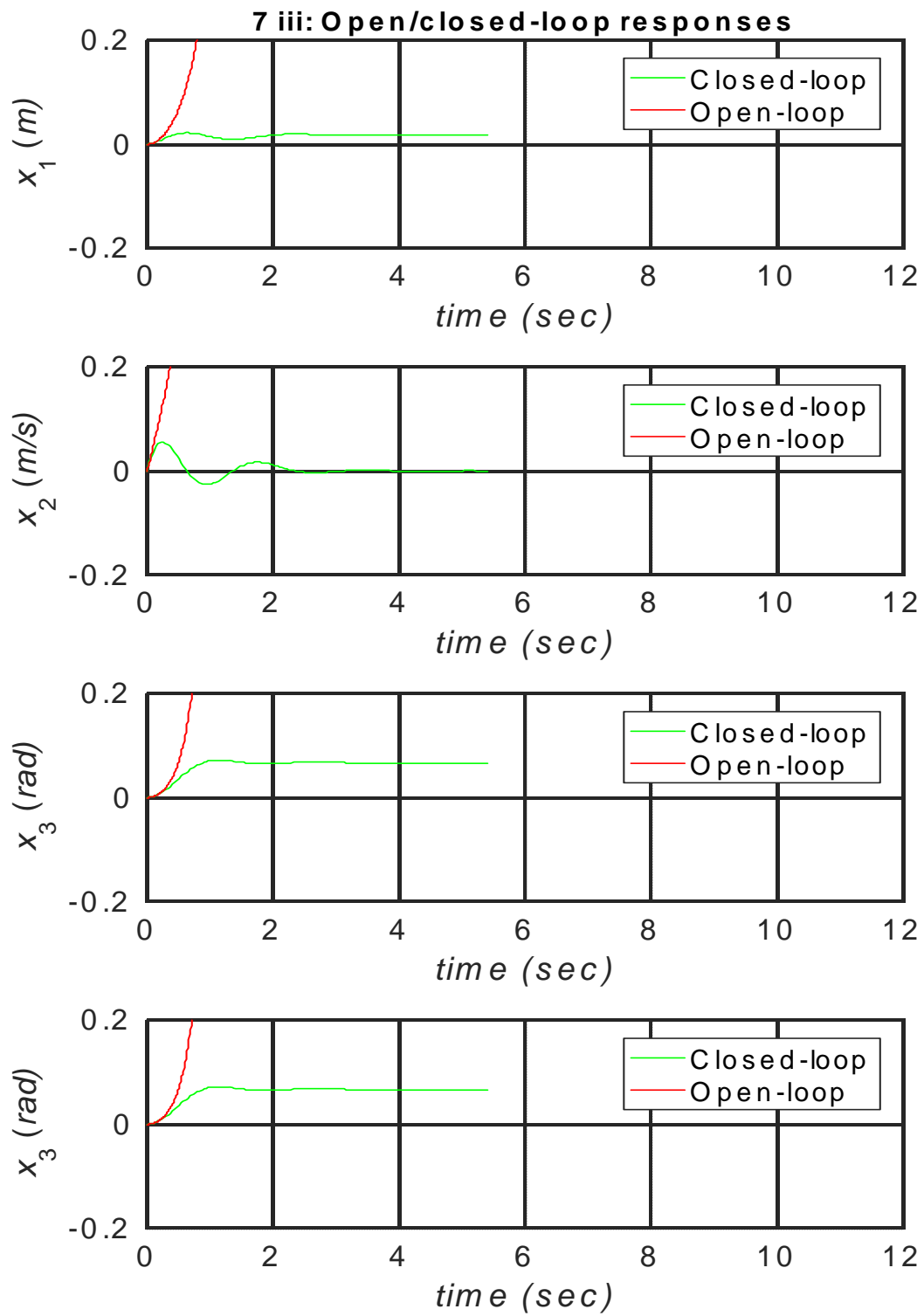
y1 0 0

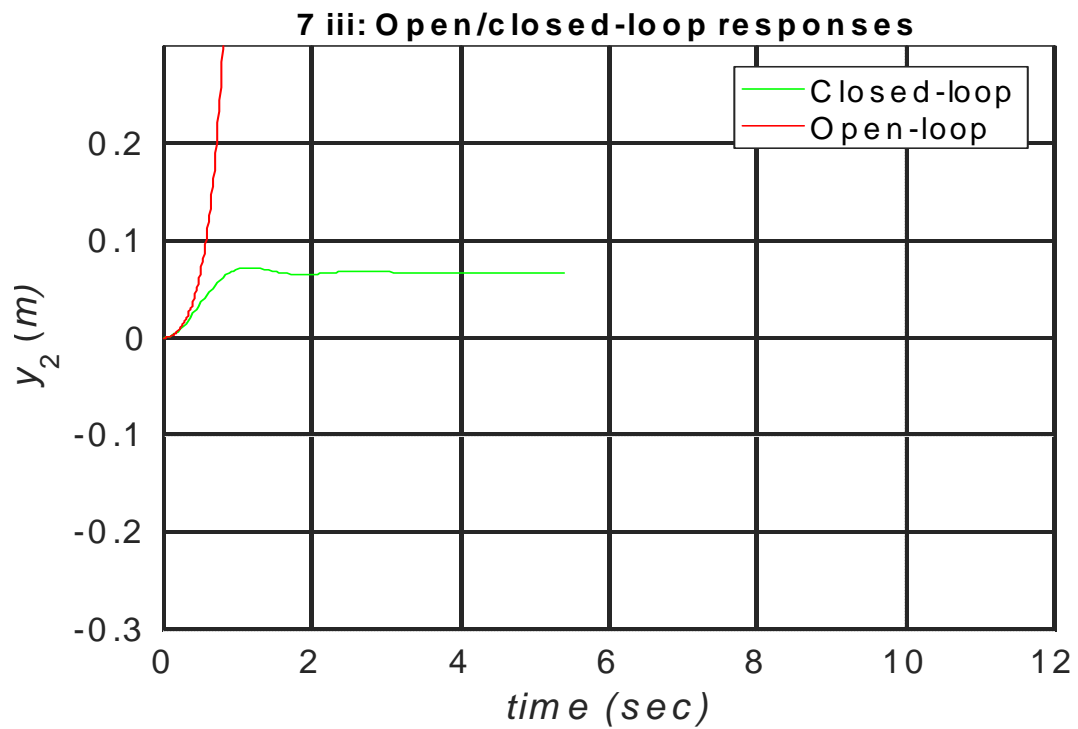
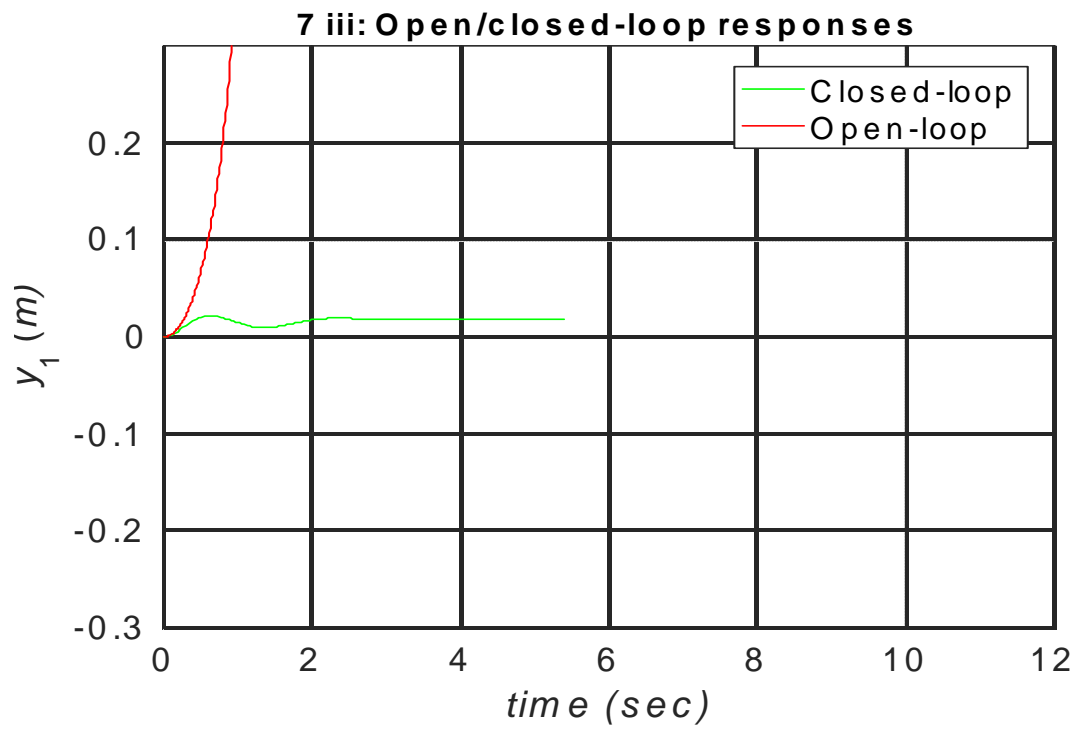
y2 0 0

Continuous-time state-space model.

>>

Plots:





Project part 8: Observer Compensator

part 8(i) system is not observable as a result, I had difficulty creating a reduced observer compensator for the sytem.

Systems 8(ii) & 8(iii) were observable, and the following observer based compensators were created:

Part 8(i):

code:

```
%Yevgen Solodkyy
%Project part 8 ii;
%

clear all; clc;

clear all;
clc;
m1=2; m2=1; L=.75; g=9.81;

A = [0 1 0 0; 0 0 m2*g/m1 0; 0 0 0 1; 0 0 (m1+m2)*g/(m1*L) 0];
B_i = [0; 1/m1; 0; 1/(L*m1)];
B_iii = [0 0; 1/m1 1/(L*m1); 0 0; 1/m1*L (m2+m1)/(m1*m2*L^2)]; % need
to verify again the last value of B.
C = [0 0 1 0];
C_ii = [1 0 0 0; 0 0 1 0]
C_iii= [1 0 0 0; 0 0 1 0]
D=0;

%desired eigven values
l1= -1.27+3.79i;
l2= -1.27-3.79i;
l3= -1.88+1.24i;
l4= -1.88-1.24i;

DesEig = [l1; l2; l3; l4]
```

```

% K for part i:
K_i=place(A,B_i,DesEig)

K_ii = K_i % parts i & ii have the same B matrix.

%open look state space for part ii:
Sys_ol_ii = ss(A,B_i,C_ii,D)

% K for part i:

Ac_ii = A-B_i*K_ii;

Sys_cl_ii = ss(Ac_ii,B_i,C_ii,D)

[Yc,t,Xc] = impulse(Sys_cl_ii);
[Y0,t0,X0] = impulse(Sys_ol_ii);

%% Observer portion

ObsEig = 100*DesEig; % observer eigen values 10x desired eigen values

%Q= [C; C*A; C*A^2; C*A^3];

L=place(A',C_ii',ObsEig)';
Ahat=A-L*C_ii % declaring oberver state matrix

Ar = [(A-B_i*K_i) B_i*K_i;zeros(size(A)) (A-L*C_ii)];
Br = [B_i;zeros(size(B_i))];
Cr = [C_ii zeros(size(C_ii))];
Dr = D;

Sys_r = ss(Ar,Br,Cr,Dr);

tr = [0:0.01:10];

r = [zeros(size(tr))];
Xr0 = [0.1;0.1;0.1;0.1;0.1;0.1;0.1;0.1]; %initial error conditions

```

```
%create vecros like these for the ope and closed loop systems as well.
see
%page 345 for reference
```

```
[Yr,tr,Xr] = lsim(Sys_r,r,tr,Xr0);
```

```
%%
figure(1)
subplot(411),
plot(t,Xc(:,1),'green',t0,X0(:,1),'red',tr,Xr(:,1),'blue') %plot
(t0,Y0(:,1),'r',tc,Yc,'g',t,Yr,'b'); grid;
grid;
axis([0 12 -.3 .3]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_1 (\itm)');
legend('Closed-loop','Open-loop','Observer');
title('8 ii: Open vs. closed-loop responses , & observer, '); %
%legend('open loop','closee loop','w/ Observer');

subplot(412),
plot(t,Xc(:,2),'green',t0,X0(:,2),'red',tr,Xr(:,2),'b'); %plot(t,Xc
(:,2),'green',t0,X0(:,2),'red');
grid;
axis([0 12 -.5 1]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_2 (\itm/s)');
legend('Closed-loop','Open-loop','Observer');

subplot(413),
plot(t,Xc(:,3),'green',t0,X0(:,3),'red',tr,Xr(:,3),'b'); %plot(t,Xc
(:,3),'green',t0,X0(:,3),'red');
grid;
axis([0 12 -.3 .3]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_3 (\itrad)');
legend('Closed-loop','Open-loop','Observer');

subplot(414),
plot(t,Xc(:,3),'green',t0,X0(:,3),'red',tr,Xr(:,3),'b'); %plot(t,Xc
(:,3),'green',t0,X0(:,3),'red');
grid;
axis([0 12 -.3 .3]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_3 (\itrad)');
legend('Closed-loop','Open-loop','Observer');
```

```

figure(2)
subplot(211),
plot(t,Xc(:,1),'green',tO,XO(:,1),'red',tr,Yr(:,1),'b'); %plot(t,Yc
(:,1),'green',tO,YO(:,1),'red');
grid;
axis([0 12 -.3 .3]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\ity}_1 (\itm)');
legend('Closed-loop','Open-loop','Observer');
title('8 ii: Open/closed-loop responses', 'FontSize',14);

subplot(212),
plot(t,Xc(:,2),'green',tO,XO(:,2),'red',tr,Yr(:,2),'b');%plot(t,Yc
(:,2),'green',tO,YO(:,2),'red');
grid;
axis([0 12 -.3 .3]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\ity}_2 (\itm)');
legend('Closed-loop','Open-loop','Observer');
title('8 ii: Open/closed-loop responses', 'FontSize',14);

```

Command Output:

C_{ii} =

```

1  0  0  0
0  0  1  0

```

C_{iii} =

```

1  0  0  0
0  0  1  0

```

DesEig =

-1.2700 + 3.7900i

-1.2700 - 3.7900i

-1.8800 + 1.2400i

-1.8800 - 1.2400i

K_i =

-12.3907 -11.1554 84.6221 17.8166

K_ii =

-12.3907 -11.1554 84.6221 17.8166

Sys_ol_ii =

A =

	x1	x2	x3	x4
x1	0	1	0	0
x2	0	0	4.905	0
x3	0	0	0	1

x4 0 0 19.62 0

B =

u1

x1 0

x2 0.5

x3 0

x4 0.6667

C =

x1 x2 x3 x4

y1 1 0 0 0

y2 0 0 1 0

D =

u1

y1 0

y2 0

Continuous-time state-space model.

Sys_cl_ii =

A =

x1 x2 x3 x4

x1	0	1	0	0
x2	6.195	5.578	-37.41	-8.908
x3	0	0	0	1
x4	8.26	7.437	-36.79	-11.88

B =

	u1
x1	0
x2	0.5
x3	0
x4	0.6667

C =

	x1	x2	x3	x4
y1	1	0	0	0
y2	0	0	1	0

D =

	u1
y1	0
y2	0

Continuous-time state-space model.

Ahat =

1.0e+04 *

-0.0315 0.0001 0.0254 0

-7.1171 0 5.5228 0

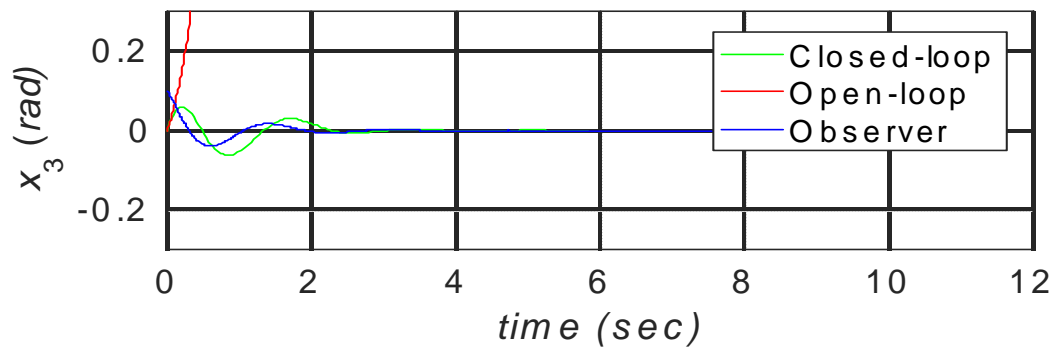
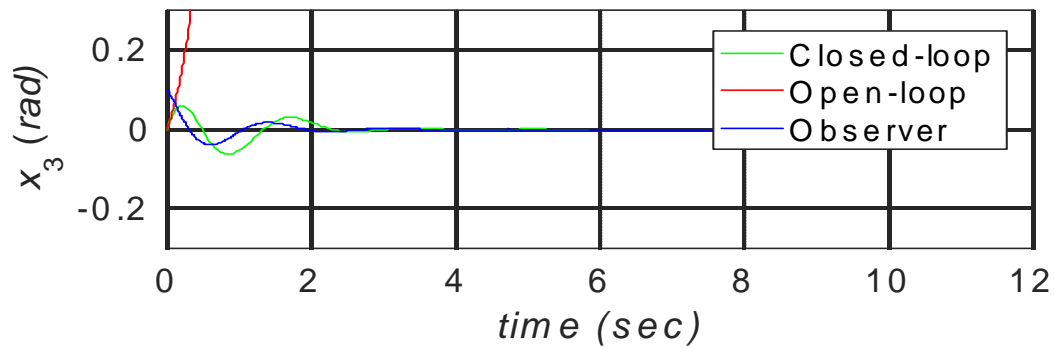
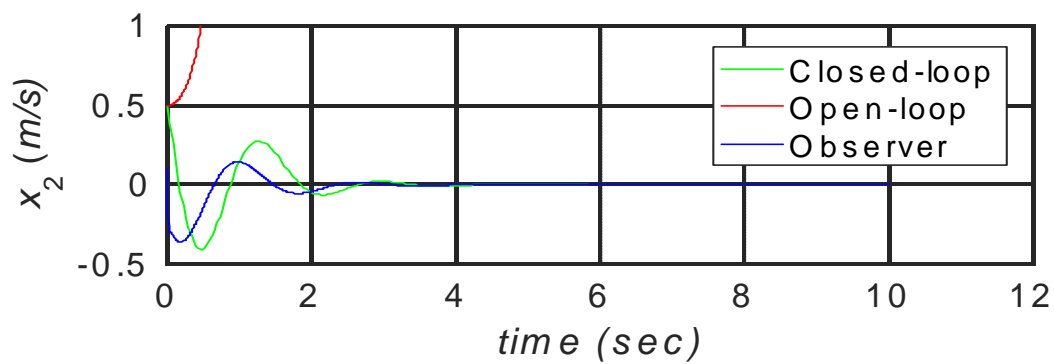
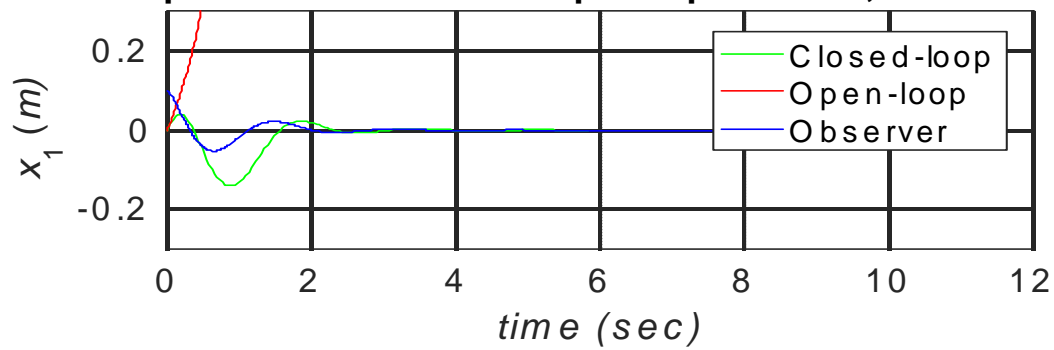
-0.0256 0 -0.0315 0.0001

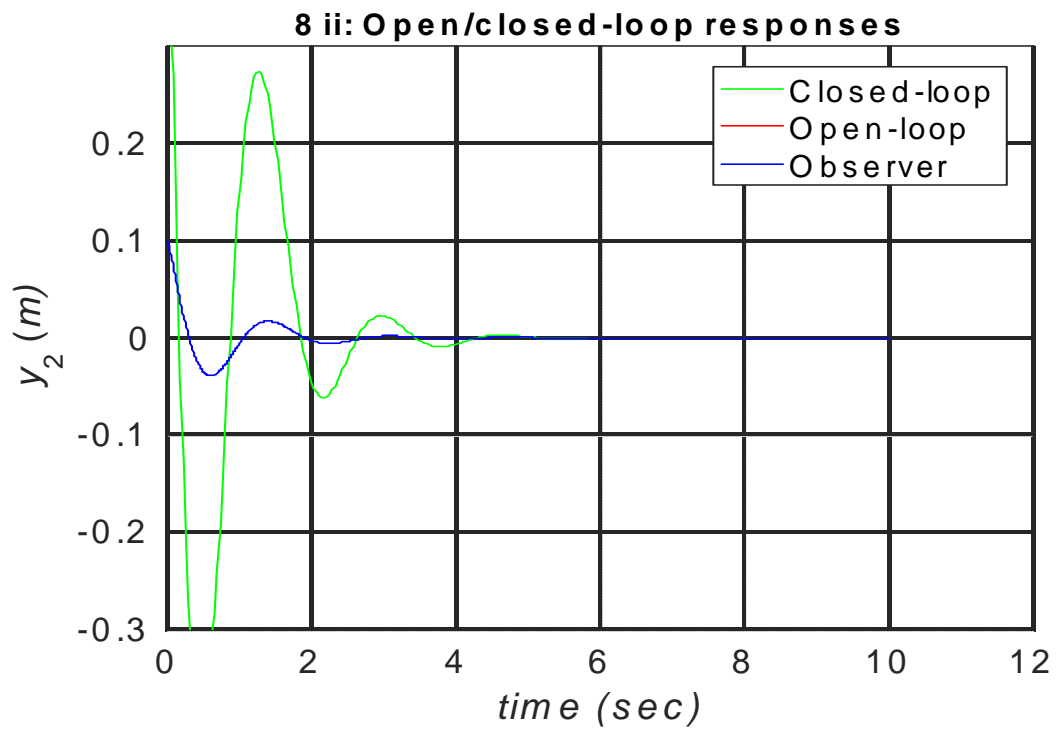
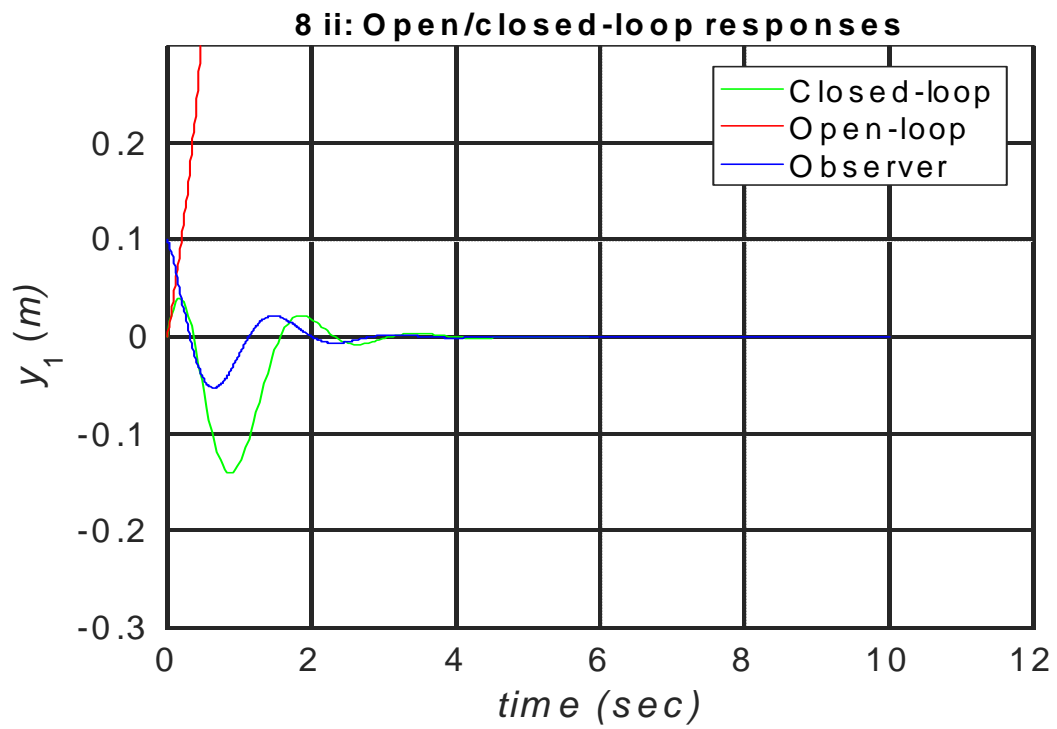
-5.5782 0 -7.0574 0

>>

Plots:

8 ii: Open vs. closed-loop responses , & observer





Project part 8(iii):

code:

```
%Yevgen Solodkyy
%Project part 8 iii;

clear all; clc;

display('Design state feedback control laws for closed loop
eigenvalues')

clear all;
clc;
m1=2; m2=1; L=.75; g=9.81;

A = [0 1 0 0; 0 0 m2*g/m1 0; 0 0 0 1; 0 0 (m1+m2)*g/(m1*L) 0];
B_i = [0; 1/m1; 0; 1/(L*m1)];
B_iii = [0 0; 1/m1 1/(L*m1); 0 0; 1/m1*L (m2+m1)/(m1*m2*L^2)]; % need
to verify again the last value of B.
C = [0 0 1 0];
C_ii = [1 0 0 0; 0 0 1 0]
C_iii= [1 0 0 0; 0 0 1 0]
D=0;

%desired eigen values
l1= -1.27+3.79i;
l2= -1.27-3.79i;
l3= -1.88+1.24i;
l4= -1.88-1.24i;

DesEig = [l1; l2; l3; l4]

K_iii=place(A,B_iii,DesEig)

Sys_ol_iii = ss(A,B_iii,C_iii,D)
Ac_iii = A-B_iii*K_iii;
Sys_cl_iii = ss(Ac_iii,B_iii,C_iii,D)
```

```

[Yc,t,Xc] = step(Sys_cl_iii);
[Y0,t0,X0] = step(Sys_ol_iii);

%% Observer portion

ObsEig = 100*DesEig; % observer eigen values 10x desired eigen values

%Q= [C; C*A; C*A^2; C*A^3];

L=place(A',C_iii',ObsEig)';
Ahat=A-L*C_iii % declaring observer state matrix

Ar = [(A-B_iii*K_iii) B_iii*K_iii;zeros(size(A)) (A-L*C_iii)];
Br = [B_iii;zeros(size(B_iii))];
Cr = [C_iii zeros(size(C_iii))];
Dr = D;

Sys_r = ss(Ar,Br,Cr,Dr);

tr = [0:0.01:10];

r = [zeros(size(tr));zeros(size(tr))];
Xr0 = [0.1;0.1;0.1;0.1;0.1;0.1;0.1;0.1]; %initial error conditions

%create vecros like these for the ope and closed loop systems as well.
see
%page 345 for reference

[Yr,tr,Xr] = lsim(Sys_r,r,tr,Xr0);

%%
figure(1)
subplot(411),
plot(t,Xc(:,1),'green',t0,X0(:,1),'red',tr,Xr(:,1),'blue') %plot
(t0,Y0(:,1),'r',tc,Yc,'g',t,Yr,'b'); grid;
grid;
axis([0 12 -.3 .3]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_1 (\itm)');

```

```

legend('Closed-loop','Open-loop','Observer');
title('8 iii: Open vs. closed-loop responses , & observer, '); %
%legend('open loop','closee loop','w/ Observer');

subplot(412),
plot(t,Xc(:,2),'green',t0,X0(:,2),'red',tr,Xr(:,2),'b'); %plot(t,Xc
(:,2),'green',t0,X0(:,2),'red');
grid;
axis([0 12 -.5 1]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_2 (\itm/s)');
legend('Closed-loop','Open-loop','Observer');

subplot(413),
plot(t,Xc(:,3),'green',t0,X0(:,3),'red',tr,Xr(:,3),'b'); %plot(t,Xc
(:,3),'green',t0,X0(:,3),'red');
grid;
axis([0 12 -.3 .3]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_3 (\itrad)');
legend('Closed-loop','Open-loop','Observer');

subplot(414),
plot(t,Xc(:,3),'green',t0,X0(:,3),'red',tr,Xr(:,3),'b'); %plot(t,Xc
(:,3),'green',t0,X0(:,3),'red');
grid;
axis([0 12 -.3 .3]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\itx}_3 (\itrad)');
legend('Closed-loop','Open-loop','Observer');

figure(2)
subplot(211),
plot(t,Xc(:,1),'green',t0,X0(:,1),'red',tr,Yr(:,1),'b'); %plot(t,Yc
(:,1),'green',t0,Y0(:,1),'red');
grid;
axis([0 12 -.3 .3]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\ity}_1 (\itm)');
legend('Closed-loop','Open-loop','Observer');
title('8 iii: Open/closed-loop responses', 'FontSize',14);

subplot(212),
plot(t,Xc(:,2),'green',t0,X0(:,2),'red',tr,Yr(:,2),'b'); %plot(t,Yc
(:,2),'green',t0,Y0(:,2),'red');
grid;
axis([0 12 -.3 .3]);
set(gca,'FontSize',16);
xlabel('\ittime (sec)');
ylabel('{\ity}_2 (\itm)');

```



```
legend('Closed-loop','Open-loop','Observer');  
title('8 iii: Open/closed-loop responses', 'FontSize',14);
```

command output:

C_ii =

```
1  0  0  0  
0  0  1  0
```

C_iii =

```
1  0  0  0  
0  0  1  0
```

DesEig =

```
-1.2700 + 3.7900i  
-1.2700 - 3.7900i  
-1.8800 + 1.2400i  
-1.8800 - 1.2400i
```

K_iii =

20.8408	9.3247	9.3042	4.3430
-5.0110	-2.2669	8.7096	0.5701

Sys_ol_iii =

A =

	x1	x2	x3	x4
x1	0	1	0	0
x2	0	0	4.905	0
x3	0	0	0	1
x4	0	0	19.62	0

B =

	u1	u2
x1	0	0
x2	0.5	0.6667
x3	0	0
x4	0.375	2.667

C =

	x1	x2	x3	x4
y1	1	0	0	0
y2	0	0	1	0

D =

```

    u1 u2
y1  0  0
y2  0  0

```

Continuous-time state-space model.

Sys_cl_iii =

A =

	x1	x2	x3	x4
x1	0	1	0	0
x2	-7.08	-3.151	-5.554	-2.552
x3	0	0	0	1
x4	5.547	2.548	-7.095	-3.149

B =

	u1	u2
x1	0	0
x2	0.5	0.6667
x3	0	0
x4	0.375	2.667

C =

	x1	x2	x3	x4
y1	1	0	0	0

y2 0 0 1 0

D =

u1 u2

y1 0 0

y2 0 0

Continuous-time state-space model.

Ahat =

1.0e+04 *

-0.0315 0.0001 0.0254 0

-7.1171 0 5.5228 0

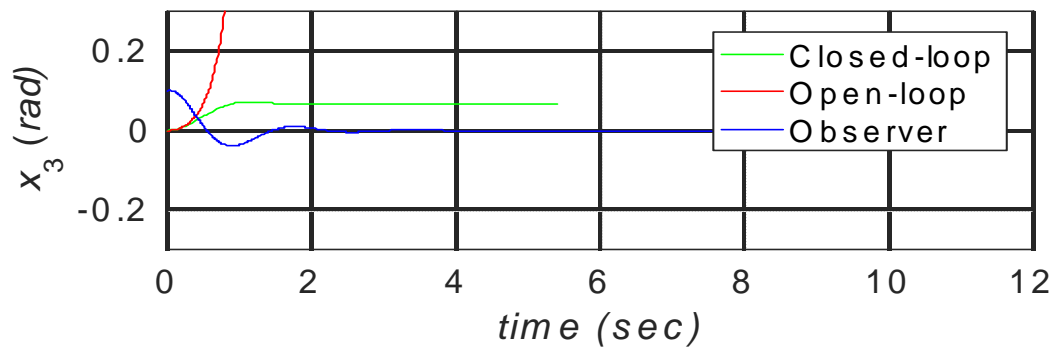
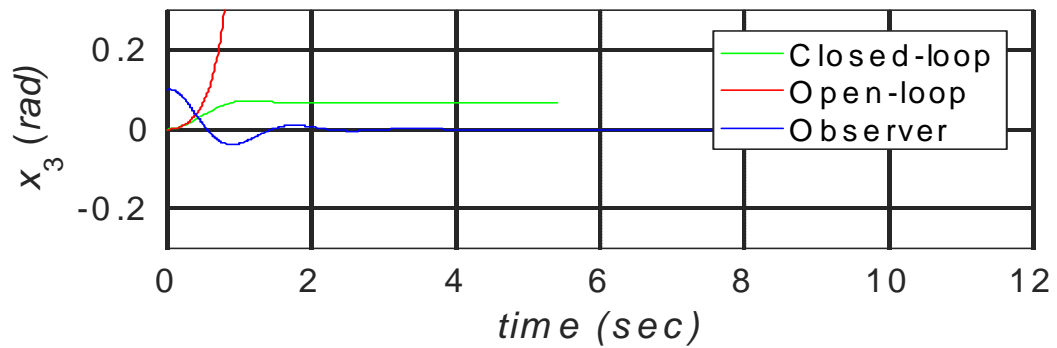
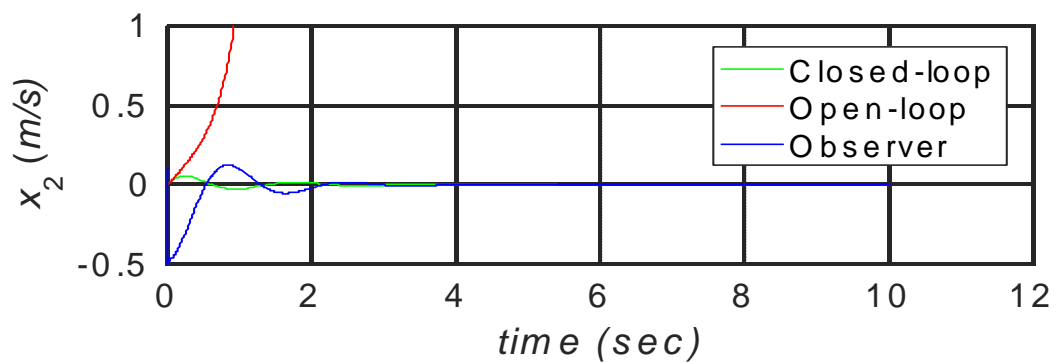
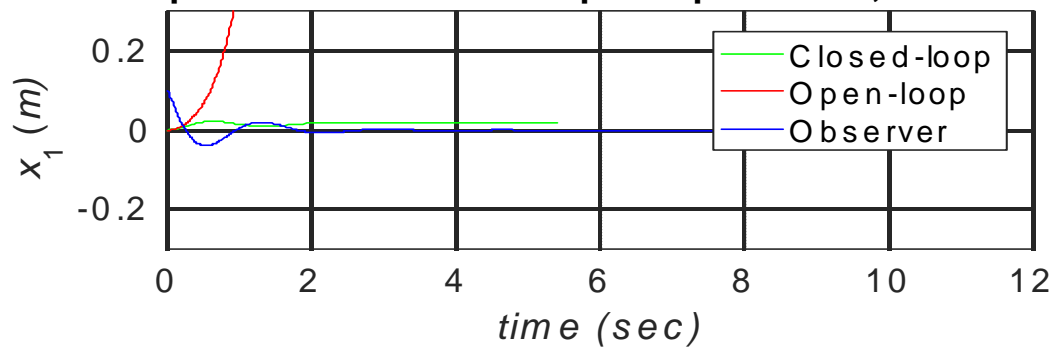
-0.0256 0 -0.0315 0.0001

-5.5782 0 -7.0574 0

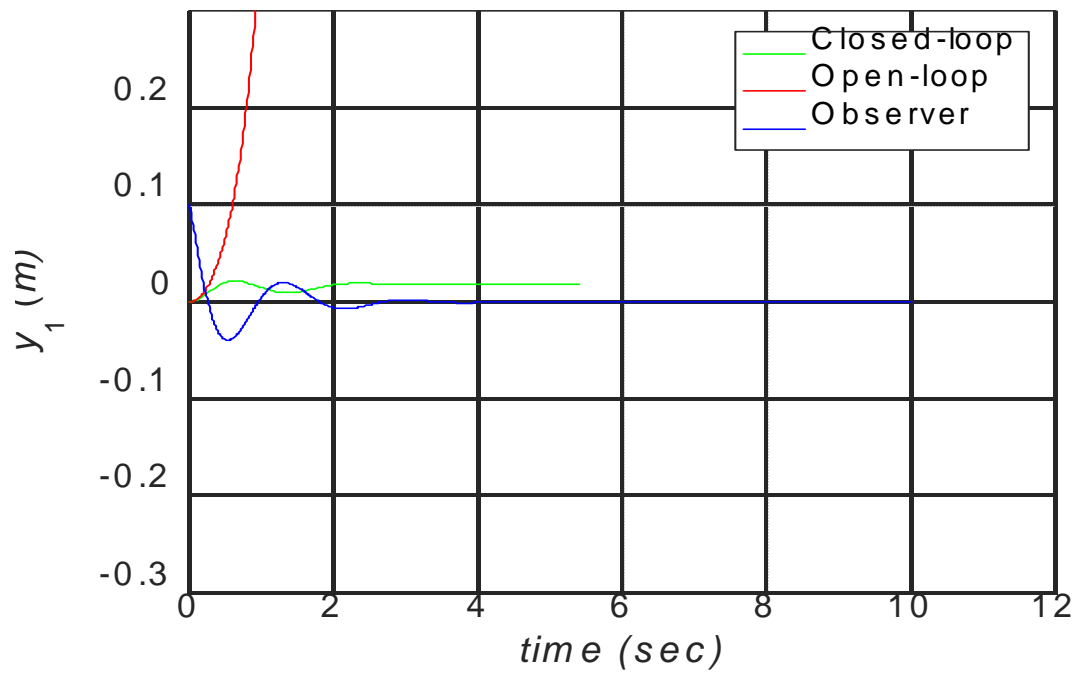
>>

PLots:

8 iii: Open vs. closed-loop responses , & observer



8 iii: Open/closed-loop responses



8 iii: Open/closed-loop responses

