

The 1090MHz Riddle

An open-access book about decoding Mode-S and ADS-B data

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GNU GPL v2 open-source license.

Preface

Begun with a frustration on the lack of technical public information on ADS-B and Mode-S in the year 2015, I created a live online document to record my understanding of ADS-B data. Previously, this was known as “ADS-B Decoding Guide” project. Together with the tutorial, we also developed its related Python library, the pyModeS. With time, I received many feedbacks, compliments, and contributions from open-source community users.

Since the beginning of 2017, the interests of tapping into Mode-S Enhanced Surveillance (EHS) data brought us a whole new chapter of Mode-S inference and decoding into the pyModeS. This also enriches the “ADS-B” guide. With the advance in this area, I am planning to compile a more comprehensive online book to cover both ADS-B and Mode-S decoding and related topic.

That’s the starting of this new repository. I am also starting to host the online book on my own server to allow more flexibility of editing and publishing. You can read the most up-to-date book on mode-s.org.

Oh, it is still GNU GPL. It was great to see the pull requests from different contributors previously. I am looking forward to seeing more comments and pulls from the community. Enjoy!

Contents

- 1. ADS-B 4**
 - 1.1. ADS-B Basics 4
 - 1.1.1. Message structure 4
 - 1.1.2. ICAO address 5
 - 1.1.3. ADS-B message types 5
 - 1.1.4. ADS-B Checksum 6
 - 1.2. Aircraft Identification 6
 - 1.3. Compact Position Reporting 8
 - 1.3.1. The CPR and functions 9
 - 1.4. Airborne Positions 10
 - 1.4.1. Globally unambiguous position (decoding with two messages) 11
 - 1.4.2. Locally unambiguous position (decoding with one message) 15
 - 1.5. Airborne Velocity 17
 - 1.5.1. Subtype 1 (Ground Speed) 18
 - 1.5.2. Subtype 3 (Airspeed) 20
- 2. Advanced ADS-B topics 23**
 - 2.1. ADS-B versions 23
 - 2.1.1. Identify the ADS-B Version 24
 - 2.2. Aircraft Operation Status 24
 - 2.3. Uncertainty and accuracy 27
 - 2.3.1. Version 0 28
 - 2.3.2. Version 1 31
 - 2.3.3. Version 2 34
- 3. Enhanced Mode-S 38**
 - 3.1. Enhanced Mode-S Basics 38
 - 3.1.1. Downlink Format and message structure 38
 - 3.1.2. Parity and ICAO address recovery 39
 - 3.1.3. BDS (Binary Data Selector) 40
 - 3.2. Aircraft identification (BDS 2,0) 40
 - 3.3. Selected intention (BDS 4,0) 41
 - 3.4. Track and turn (BDS 5,0) 43
 - 3.5. Heading and speed (BDS 6,0) 45
- 4. About the book 48**
 - 4.1. Related resources 48
 - 4.2. Contributors 48
 - 4.3. Contact 48
 - 4.4. References 49

Chapter 1

ADS-B

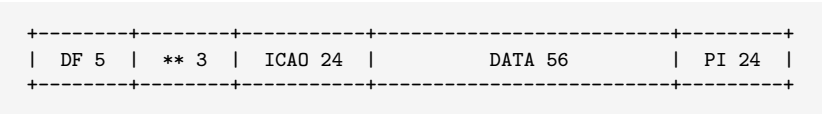
ADS-B is short for Automatic Dependent SurveillanceBroadcast. It is a satellite based surveillance system. Aircraft position, velocity, together with identification are transmitted through Mode-S Extended Squitter (1090 MHz).

Majority of the aircraft nowadays are broadcasting ADS-B messages constantly. There are many ways you can set up you own receiver and antenna to start tapping into those signals (DVB-T USB stick, Mode-S Beast, Raspberry Pi, RadarScape, etc).

1.1. ADS-B Basics

1.1.1. Message structure

An ADS-B message is 112 bits long, and consists of 5 parts.



Any ADS-B must start with the Downlink Format 17, or 18 in case of TIS-B message. They correspond to 10001 or 10010 in binary for the first 5 bits. Bits 6-8 are used as an additional identifier, which has different meanings within each ADS-B subtype.

In following Table 1.1, the key information of a ADS-B message is listed.

Table 1.1. Structure of ADS-B messages

nBits	Bits	Abbr.	Name
5	1 - 5	DF	Downlink Format
3	6 - 8	CA	Capability (additional identifier)
24	9 - 32	ICAO	ICAO aircraft address
56	33 - 88	DATA	Data
	[33 - 37]	[TC]	Type code
24	89 - 112	PI	Parity/Interrogator ID

It is worth noting that the ADS-B Extended Squitter sent from a Mode S transponder use Downlink Format 17 (DF=17). Non-Transponder-Based ADS-B Transmitting Subsystems and TIS-B Transmitting equipment use Downlink Format 18 (DF=18). By using DF=18

instead of `DF=17` , an ADS-B/TIS-B Receiving Subsystem will know that the message comes from equipment that cannot be interrogated.

An example:

Raw message in hexadecimal:
8D4840D6202CC371C32CE0576098

[00100]0000010110011
00001101110001110000
110010110011100000

HEX		8D		4840D6		202CC371C32CE0		576098	
BIN		10001	101		010010000100		[00100]0000010110011		010101110110
					000011010110		00001101110001110000		000010011000
							110010110011100000		
DEC		17	5				[4]		
		DF	CA		ICAO		[TC] --- DATA ----		PI

1.1.2. ICAO address

In each ADS-B message, the sender (originating aircraft) can be identified using the ICAO address. It is located from 9 to 32 bits in binary (or 3 to 8 in hexadecimal). In the example above, it is `4840D6` or `010010000100` .

An unique ICAO address is assigned to each Mode-S transponder of an aircraft. Thus this is a unique identifier for each aircraft. You can use the query tool (World Aircraft Database) from mode-s.org to find out more about the aircraft with a given ICAO address. For instance, using the previous ICAO `4840D6` example, it will return the result of a `Fokker 70` with registration of `PH-KZD` .

In addition, you can download the database from the aforementioned website in CSV format.

1.1.3. ADS-B message types

To identify what information is contained in an ADS-B message, we need to take a look at the `Type Code` of the message, indicated at bits 33 - 37 of the ADS-B message (or first 5 bits of the `DATA` segment).

In following Table 1.2, the relationships between each `Type Code` and its information contained in the `DATA` segment are shown.

Table 1.2. ADS-B Type Code and content

Type Code	Content
1 - 4	Aircraft identification
5 - 8	Surface position
9 - 18	Airborne position (w/ Baro Altitude)
19	Airborne velocities
20 - 22	Airborne position (w/ GNSS Height)
23 - 27	Reserved
28	Aircraft status
29	Target state and status information
31	Aircraft operation status

1.1.4. ADS-B Checksum

ADS-B uses a cyclic redundancy check to validate the correctness of the received message, where the last 24 bits are the parity bits. The following pseudo-code describes the CRC process:

```

GENERATOR = 111111111111010000001001

MSG = binary("8D4840D6202CC371C32CE0576098") # total 112 bits

FOR i FROM 0 TO 88:                                # 112 - 24 parity bits
    if MSG[i] is 1:
        MSG[i:i+24] = MSG[i:i+24] ^ GENERATOR

CRC = MSG[-24:]                                     # last 24 bits

IF CRC not 0:
    MSG is corrupted

```

For the implementation of CRC encoder in Python, refer to the pyModeS library function:

```
pyModeS.crc()
```

A comprehensive documentation on Mode-S parity coding can be found:

Gertz, Jeffrey L. Fundamentals of mode s parity coding. No. ATC-117. MASSACHUSETTS INST OF TECH LEXINGTON LINCOLN LAB, 1984. APA

1.2. Aircraft Identification

An aircraft identification message has **DF: 17 or 18**, and **TC: 1 to 4**, the 56-bit **DATA** field is configured as follows:

```

+-----+-----+-----+-----+-----+-----+-----+-----+
| TC,5 | EC,3 | C1,6 | C2,6 | C3,6 | C4,6 | C5,6 | C6,6 | C7,6 | C8,6 |
+-----+-----+-----+-----+-----+-----+-----+-----+

TC: Type code
EC: Emitter category
C*: A character

```

To decode characters, a lookup table is needed for mapping numbers to characters. It is defined as follows, where the # is not used, and _ represents a separation.

```
#ABCDEFGHIJKLMNOPQRSTUVWXYZ####_#####0123456789#####
```

In summary, characters and their decimal representations are:

```

A - Z :   1 - 26
0 - 9 :  48 - 57
_ :      32

```

The EC value in combination with TC value defines the category of the aircraft (such as: heavy, large, small, light, glider, etc.). When EC is set to zeros, such information is not available.

For example:

```
8D4840D6202CC371C32CE0576098
```

The structure of the message is as follows:

```

      DF... CA.  ICAO.. DATA..... PI....
HEX: 8   D      4840D6 2   0      2CC371C32CE0 576098
BIN: 10001 101  ***** 00100 000 ***** *****
DEC: 17    5              4     0
      TC    *

```

Note that Type Code is inside of the DATA frame (first 5 bits). With DF=17 and TC=4, we can confirm this is an aircraft identification message. Aircraft callsign then can be decoded.

In the previous example message, it is easy to decode the Data segment:

```

HEX: 20          2CC371C32CE0
BIN: 00100000 | 001011 001100 001101 110001 110000 110010 110011 100000
DEC:          |  11    12    13    49    48    50    51    32
LTR:          |  K     L     M     1     0     2     3     _

```

So the final aircraft callsign decoded is: KLM1023

For detailed codes in Python, refer to the pyModeS library function: `pyModeS.adsb.callsign()`

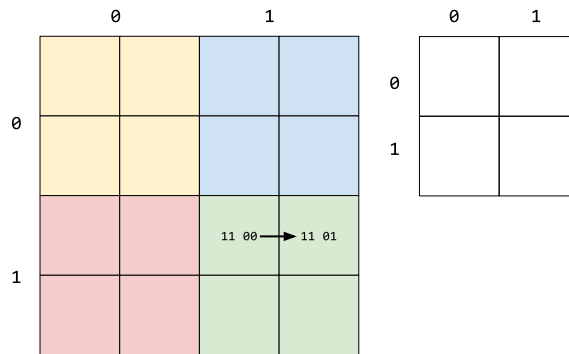
1.3. Compact Position Reporting

The position information in ADS-B messages is encoded in a compact position reporting (CPR) format. The general idea behind CPR is to be able to encode more coordinate decimals using less bits. It is achieved by trading global position ambiguity and time with local position accuracy.

An easy example to understand the principle behind CPR:

Imaging the world is constructed by 16 grid, which we have divided into two levels, each level is encoded with two bits. Higher levels in color are 00 (yellow), 01 (blue), 10 (red), 11 (green). And within each color grid, the lower levels are also encoded similarly.

Then each grid can be represented as 4 digits from 0000 to 1111. Now, we want to describe the movement indicated as the arrows in the green grids 1100 → 1101, but we only have 3 bits to encode each position.

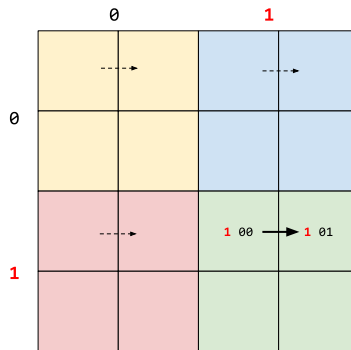


It is easy to see that the high 2 bits appeared in all positions, so we can define a structure to do the following:

1. The last two bits shall represent the local position
2. The combination of first digit from two messages defines the higher grid

Then the two messages can be sent as 1 00 → 1 01.

From lower bits 00 → 01, we have four different possibility of movement as shown in dashed arrows, and from the two first bits combination 11, we know that the arrow shall represent the movement in the green grids:



1.3.1. The CPR and functions

The actual CPR algorithm of course is more complicated, but the principle is very similar to the previous example. If only one message is given, it is possible to find multiple solutions that are spaced around the world. The combination of two (different types of) messages will yield the final result.

In CPR encoding, the Earth is divided in many zones (similar to the grid in the previous example). And the encoding algorithm is also more complicated (described in a later section). First, we will list some of the parameters and common functions used in the decoding process here.

NZ

Number of geographic latitude zones between equator and a pole. It is set to `NZ = 15` for Mode-S CPR encoding.

floor(x)

the floor function `floor(x)` defines as the greatest integer value k , such that $k \leq x$, for example:

```
floor(5.6) = 5
floor(-5.6) = -6
```

mod(x, y)

the modulus function `mod(x, y)` returns:

$$x - y \cdot \text{floor}\left(\frac{x}{y}\right) \quad (1.1)$$

where y can not be zero

NL(lat)

Denotes the “number of longitude zones” function, given the latitude angle lat . The returned integer value is constrained within $[1, 59]$, calculated as:

$$\text{NL}(\text{lat}) = \text{floor}\left(\frac{2\pi}{\arccos\left(1 - \frac{1 - \cos\left(\frac{\pi \cdot \text{NZ}}{2}\right)}{\cos^2\left(\frac{\pi}{180} \cdot \text{lat}\right)}\right)}\right) \quad (1.2)$$

For latitudes that are close to the equator or the poles, one of following values is returned:

```
lat = 0      ->   NL = 59
lat = +87    ->   NL = 2
lat = -87    ->   NL = 2
lat > +87    ->   NL = 1
lat < -87    ->   NL = 1
```

1.4. Airborne Positions

An aircraft airborne position message has downlink format 17 (or 18) with type code from 9 to 18

Messages are composed as shown in following Table 1.3

Table 1.3. Airborne position message bits explained

Data Bits	MSG Bits	N-bit	Abbr	Content
33 - 37	1 - 5	5	TC	Type code
38 - 39	6 - 7	2	SS	Surveillance status
40	8	1	NICsb	NIC supplement-B
41 - 52	9 - 20	12	ALT	Altitude
53	21	1	T	Time
54	22	1	F	CPR odd/even frame flag
55 - 71	23 - 39	17	LAT-CPR	Latitude in CPR format
72 - 88	40 - 56	17	LON-CPR	Longitude in CPR format

Two types of the position messages (odd and even frames) are broadcast alternately. There are two different ways to decode an airborne position based on these messages:

1. Unknown position, using both types of messages (aka globally unambiguous position)

2. Knowing previous position, using only one message (aka locally unambiguous position)

Note: The definition of functions `NL(lat)` , `floor(x)` , and `mod(x,y)` are described in the CPR chapter.

1.4.1. Globally unambiguous position (decoding with two messages)

odd or even message?

For each frame, bit 54 determines whether it is an odd or even frame:

```
0 -> Even frame
1 -> Odd frame
```

For example, the two following messages are received:

```
8D40621D58C382D690C8AC2863A7
8D40621D58C386435CC412692AD6
```

	ICAO24	DATA	CRC
8D	40621D	58C382D690C8AC	2863A7
8D	40621D	58C386435CC412	692AD6

The payload data in binary format:

DATA									
TC	...	ALT	T	F	CPR-LAT	CPR-LON			
01011	000	110000111000	0	0	10110101101001000	01100100010101100			
01011	000	110000111000	0	1	10010000110101110	01100010000010010			

In both messages we can find `DF=17` and `TC=11` , with the same ICAO24 address `40621D` . So, those two frames are valid for decoding the positions of this aircraft. Assume the first message is the newest message received.

The CPR representation of coordinates

F	CPR Latitude	CPR Longitude	
0	10110101101001000	01100100010101100	-> newest
1	10010000110101110	01100010000010010	

In decimal:

0	93000	51372
1	74158	50194

CPR_LAT_EVEN: 93000 / 131072 -> 0.7095
CPR_LON_EVEN: 51372 / 131072 -> 0.3919
CPR_LAT_ODD: 74158 / 131072 -> 0.5658
CPR_LON_ODD: 50194 / 131072 -> 0.3829

Since CPR latitude and longitude are encoded in 17 bits, 131072 (2^{17}) is the maximum value.

Calculate the latitude index j

Use the following equation:

$$j = \text{floor} \left(59 \cdot \text{Lat}_{\text{cprEven}} - 60 \cdot \text{Lat}_{\text{cprOdd}} + \frac{1}{2} \right) \quad (1.3)$$

where j is set 8.

Calculate latitude

First, two constants will be used:

$$\begin{aligned} d\text{Lat}_{\text{even}} &= \frac{360}{4 \cdot NZ} = \frac{360}{60} \\ d\text{Lat}_{\text{odd}} &= \frac{360}{4 \cdot NZ - 1} = \frac{360}{59} \end{aligned} \quad (1.4)$$

Then we can use the following equations to compute the relative latitudes:

$$\begin{aligned} \text{Lat}_{\text{even}} &= \text{dLat}_{\text{even}} \cdot [\text{mod}(j, 60) + \text{Lat}_{\text{cprEven}}] \\ \text{Lat}_{\text{odd}} &= \text{dLat}_{\text{odd}} \cdot [\text{mod}(j, 59) + \text{Lat}_{\text{cprOdd}}] \end{aligned} \quad (1.5)$$

For the southern hemisphere, values will fall from 270 to 360 degrees. We need to make sure the latitude is within the range `[-90, +90]` :

$$\begin{aligned} \text{Lat}_{\text{even}} &= \text{Lat}_{\text{even}} - 360 \quad \text{if } (\text{Lat}_{\text{even}} \geq 270) \\ \text{Lat}_{\text{odd}} &= \text{Lat}_{\text{odd}} - 360 \quad \text{if } (\text{Lat}_{\text{odd}} \geq 270) \end{aligned} \quad (1.6)$$

Final latitude is chosen depending on the time stamp of the frames, the newest one, is used:

$$\text{Lat} = \begin{cases} \text{Lat}_{\text{even}} & \text{if } (T_{\text{even}} \geq T_{\text{odd}}) \\ \text{Lat}_{\text{odd}} & \text{else} \end{cases} \quad (1.7)$$

In the example:

```
Lat_EVEN = 52.25720214843750
Lat_ODD  = 52.26578017412606
Lat = Lat_EVEN = 52.25720
```

Check the latitude zone consistency

Compute `NL(Lat_E)` and `NL(Lat_O)` . If not the same, two positions are located at different latitude zones. Computation of a global longitude is not possible. Exit the calculation and wait for new messages. If two values are the same, we proceed to longitude calculation.

Calculate longitude

If the even frame comes latest `T_EVEN > T_ODD` :

$$\begin{aligned} ni &= \max(NL(\text{Lat}_{\text{even}}), 1) \\ \text{dLon} &= \frac{360}{ni} \\ m &= \text{floor} \left\{ \text{Lon}_{\text{cprEven}} \cdot [NL(\text{Lat}_{\text{even}}) - 1] - \text{Lon}_{\text{cprOdd}} \cdot NL(\text{Lat}_{\text{even}}) + \frac{1}{2} \right\} \\ \text{Lon} &= \text{dLon} \cdot (\text{mod}(m, ni) + \text{Lon}_{\text{cprEven}}) \end{aligned} \quad (1.8)$$

In case where the odd frame comes latest `T_EVEN < T_ODD` :

$$\begin{aligned}
ni &= \max(NL(\text{Lat}_{\text{odd}}) - 1, 1) \\
d\text{Lon} &= \frac{360}{ni} \\
m &= \text{floor} \left\{ \text{Lon}_{\text{cprEven}} \cdot [NL(\text{Lat}_{\text{odd}}) - 1] - \text{Lon}_{\text{cprOdd}} \cdot NL(\text{Lat}_{\text{odd}}) + \frac{1}{2} \right\} \\
\text{Lon} &= d\text{Lon} \cdot (\text{mod}(m, ni) + \text{Lon}_{\text{cprOdd}})
\end{aligned} \tag{1.9}$$

if the result is larger than 180 degrees:

$$\text{Lon} = \text{Lon} - 360 \quad \text{if } (\text{Lon} \geq 180) \tag{1.10}$$

In the example:

```
Lon: 3.91937
```

Here is a Python implementation: <https://github.com/junzis/pyModeS/blob/faf4313/pyModeS/adsb.py#L166>

Calculate altitude

The altitude of the aircraft is much easier to compute from the data frame. The bits in the altitude field (either odd or even frame) are as follows:

```
1100001 1 1000
      ^
      Q-bit
```

This Q-bit (bit 48) indicates whether the altitude is encoded in multiples of 25 or 100 ft (0: 100 ft, 1: 25 ft).

For Q = 1, we can calculate the altitude as follows:

First, remove the Q-bit :

```
N = 1100001 1000 => 1560 (in decimal)
```

The final altitude value will be:

$$\text{Alt} = N \cdot 25 - 1000 \quad (\text{ft.}) \tag{1.11}$$

In this example, the altitude at which the aircraft is flying is:

```
1560 * 25 - 1000 = 38000 ft.
```

Note that the altitude has the accuracy of +/- 25 ft when the Q-bit is 1, and the value can represent altitudes from -1000 to +50175 ft.

The final position

Finally, we have all three components (latitude/longitude/altitude) of the aircraft position:

```
LAT: 52.25720 (degrees N)
LON: 3.91937 (degrees E)
ALT: 38000 ft
```

1.4.2. Locally unambiguous position (decoding with one message)

This method gives the possibility of decoding aircraft using only one message knowing a reference position. This method computes the latitude index (j) and the longitude index (m) based on such reference, and can be used with either type of the messages.

The reference position

The reference position should be close to the actual position (eg. position of aircraft previously decoded, or the location of ADS-B antenna), and must be **within a 180 NM** range.

Calculate $dLat$

$$dLat = \begin{cases} \frac{360}{4 \cdot NZ} = \frac{360}{60} & \text{for even message} \\ \frac{360}{4 \cdot NZ - 1} = \frac{360}{59} & \text{for odd message} \end{cases} \quad (1.12)$$

Calculate the latitude index j

$$j = \text{floor}\left(\frac{Lat_{ref}}{dLat}\right) + \text{floor}\left(\frac{\text{mod}(Lat_{ref}, dLat)}{dLat} - Lat_{cpr} + \frac{1}{2}\right) \quad (1.13)$$

Calculate latitude

$$Lat = dLat \cdot (j + Lat_{cpr}) \quad (1.14)$$

Calculate $dLon$

For even message:

$$dLon = \begin{cases} \frac{360}{NL(Lat)} & \text{if } NL(Lat) > 0 \\ 360 & \text{if } NL(Lat) = 0 \end{cases} \quad (1.15)$$

For odd message:

$$dLon = \begin{cases} \frac{360}{NL(Lat)-1} & \text{if } NL(Lat) - 1 > 0 \\ 360 & \text{if } NL(Lat) - 1 = 0 \end{cases} \quad (1.16)$$

Calculate longitude index m

$$m = floor\left(\frac{Lon_{ref}}{dLon}\right) + floor\left(\frac{mod(Lon_{ref}, dLon)}{dLon} - Lon_{cpr} + \frac{1}{2}\right) \quad (1.17)$$

Calculate longitude

$$Lon = dLon \cdot (m + Lon_{cpr}) \quad (1.18)$$

Example

For the same example message:

```
8D40621D58C382D690C8AC2863A7
```

```
Reference position:
```

```
LAT: 52.258
```

```
LON: 3.918
```

The structure of the message is:


```
8D40621D58C382D690C8AC2863A7
```

ICA024	DATA	CRC
8D	40621D	58C382D690C8AC
		2863A7

Data in binary:

DATA							
TC	...	ALT	T	F	CPR-LAT	CPR-LON	
01011	000	110000111000	0	0	10110101101001000	01100100010101100	

CPR representation:

F	CPR Latitude	CPR Longitude
0	10110101101001000	01100100010101100
	93000 / 131072	51372 / 131072
	0.7095	0.3919

```
d_lat: 6
j: 8
lat: 52.25720
m: 0
d_lon: 10
lon: 3.91937
```

1.5. Airborne Velocity

There are two different types of messages for velocities, determined by **3-bit subtype in the message**. With subtype 1 and 2, surface velocity (ground speed) is reported. And in subtype 3 and 4, aircraft airspeed is reported.

Type 2 and 4 are for supersonic aircraft. So, before we have another commercial supersonic aircraft flying around, you won't see any of those types.

In real world, very few of subtype 3 messages are reported. In our setup, we only received **0.3%** of these message with regards to subtype 1.

An aircraft velocity message has DF: 17 or 18, TC: 19, and the subtype codes are represented in bits 38 to 40. Now, we can decode those messages.

1.5.1. Subtype 1 (Ground Speed)

Subtype 1 (subsonic, ground speed), are broadcast when ground velocity information is available. The aircraft velocity contains speed and heading information. The speed and heading are also decomposed into North-South, and East-West components.

For example, the following message is received:

Message: 8D485020994409940838175B284F

ICA024	DATA	CRC
8D	485020	99440994083817
5B284F		

Convert DATA [99440994083817] into binary:

TC	ST	IC	RESV_A	NAC	S-EW	V-EW
10011	001	0	1	000	1	0000001001

S-NS	V-NS	VrSrc	S-Vr	Vr	RESV_B	S_Dif	Dif
1	0010100000	0	1	000001110	00	0	0010111

There are quite a few parameters in the velocity message. From left to right, the number of bits indicates the contents in following Table 1.4

Table 1.4. Airborne velocity message bits explained - Subtype 1

MSG Bits	Data Bits	Len	Abbr	Content
33 - 37	1 - 5	5	TC	Type code
38 - 40	6 - 8	3	ST	Subtype
41	9	1	IC	Intent change flag
42	10	1	RESV_A	Reserved-A
43 - 45	11 - 13	3	NAC	Velocity uncertainty (NAC)
46	14	1	S_ew	East-West velocity sign
47 - 56	15 - 24	10	V_ew	East-West velocity
57	25	1	S_ns	North-South velocity sign
58 - 67	26 - 35	10	V_ns	North-South velocity
68	36	1	VrSrc	Vertical rate source
69	37	1	S_vr	Vertical rate sign
70 - 78	38 - 46	9	Vr	Vertical rate
79 - 80	47 - 48	2	RESV_B	Reserved-B
81	49	1	S_Dif	Diff from baro alt, sign
82 - 88	50 - 56	7	Dif	Diff from baro alt

Horizontal Velocity

For calculating the horizontal speed and heading we need four values, East-West Velocity V_{ew} , East-West Velocity Sign S_{ew} , North-South Velocity V_{ns} , North-South Velocity Sign S_{ns} . And pay attention on the directions (signs) in the calculation.

```
S_ns:
  1 -> flying North to South
  0 -> flying South to North
S_ew:
  1 -> flying East to West
  0 -> flying West to East
```

The Speed (v) and heading (h) with unit knot and degree can be computed as follows:

$$V_{we} = \begin{cases} -1 \cdot (V_{ew} - 1) & \text{if } s_{ew} = 1 \\ V_{ew} - 1 & \text{if } s_{ew} = 0 \end{cases}$$

$$V_{sn} = \begin{cases} -1 \cdot (V_{ns} - 1) & \text{if } s_{ns} = 1 \\ V_{ns} - 1 & \text{if } s_{ns} = 0 \end{cases}$$

$$v = \sqrt{V_{we}^2 + V_{sn}^2}$$

$$h = \arctan2(V_{we}, V_{sn}) \cdot \frac{360}{2\pi} \quad (\text{deg})$$

In case of a negative value here, we will simply add 360 degrees.

$$h = h + 360 \quad (\text{if } h < 0)$$

So, now we have the speed and heading of our example:

```
V-EW: 0000001001 -> 9
S-EW: 1
V-NS: 0010100000 -> 160
S-NS: 1

V_{we} = - (9 - 1) = -8
V_{sn} = - (160 - 1) = -159

v = 159.20 (kt)
h = 182.88 (deg)
```


Table 1.5. Airborne velocity message bits explained - Subtype 3

DATA Bits	MSG Bits	Len	Abbr	Content
33 - 37	1 - 5	5	TC	Type code
38 - 40	6 - 8	3	ST	Subtype
41	9	1	IC	Intent change flag
42	10	1	RESV_A	Reserved-A
43 - 45	11 - 13	3	NAC	Velocity uncertainty (NAC)
46	14	1	S_hdg	Heading status
47 - 56	15 - 24	10	Hdg	Heading (proportion)
57	25	1	AS-t	Airspeed Type
58 - 67	26 - 35	10	AS	Airspeed
68	36	1	VrSrc	Vertical rate source
69	37	1	S_vr	Vertical rate sign
70 - 78	38 - 46	9	Vr	Vertical rate
79 - 80	47 - 48	2	RESV_B	Reserved-B
81	49	1	S_Dif	Difference from baro alt, sign
82 - 88	50 - 66	7	Dif	Difference from baro alt

Heading

S_hdg makes the status of heading data:

```
0 -> heading data not available
1 -> heading data available
```

10-bits Hdg is the represent the proportion of the degrees of a full circle, i.e. 360 degrees. (Note: 0000000000 - 1111111111 represents 0 - 1023)

$$heading = Decimal(Hdg)/1024 * 360^{\circ}$$

in our example :

```
1010110110 -> 694
heading = 694 / 1024 * 360 = 243.98 (degree)
```

Velocity (Airspeed)

To find out which type of the airspeed (TAS or IAS), first we need to look at the AS-t field:

```
0 -> Indicated Airspeed (IAS)
1 -> True Airspeed (TAS)
```

And then the speed is simply a binary to decimal conversion of AS bits (in knot). In our example:

```
0101111000 -> 376 knot
```

Vertical Rate

The vertical rate decoding remains the same as subtype 1.

Chapter 2

Advanced ADS-B topics

In this chapter, we are going to discuss some advanced topics regarding ADS-B. Rather than position and velocity of aircraft, ADS-B provides many other types of interesting information. Especially with newer versions of the ADS-B implementations (yes, there is more than one version!), more data is downlinked. We will start with an introduction on different ADS-B versions and differences among them. Then, go on with the decoding with some new message types. Lastly, the uncertainty of ADS-B measurements will be discussed in detail.

2.1. ADS-B versions

In this advanced chapter, we are going to look into different versions and evolution of the ADS-B.

Since the beginning of ADS-B, there have been three different versions (to my knowledge) implemented. The major reason for these updates is to enable more information (types of data) in ADS-B. Documentations on these versions and differences are quite far from user friendly. They are always presented in a very scattered fashion. Even the official ICAO.9871 document is confusing to read. I am going to try my best to put the pieces together in this chapter.

There are three versions implemented so far, starting from Version 0, then Version 1 around 2008 and Version 2 around 2012. Major changes in Version 1 and Version 2 are listed as follows:

From Version 0 to Version 1 :

- Added Type Code 28, 29, and 31 messages
 - TC=28 : Aircraft status - Emergency/priority status and ACAS RA Broadcast
 - TC=29 : Target state and status
 - TC=31 : Operational status
- Introduced the “Navigation integrity category (NIC)” and “Surveillance integrity level (SIL)” in addition to the “Navigation accuracy category (NAC)” from the Version 0
 - Type Code and an NIC Supplement bit (NICs) is used to define the NIC

– NIC Supplement bit included in TC=31 messages

- The ADS-B version number is now indicated in operation status message TC=31

From Version 1 to Version 2 :

- Re-defined the structure and content of TC=28 , TC=29 , and TC=31 messages.
- Introduced two additional NIC Supplement Bit
- NICa is defined in operational status messages (TC=31)
- NICb is defined in airborne position messages (TC=9-18)
- NICc is defined in operational status messages (TC=31)
- Introduced an additional “Horizontal Containment Radius (Rc)” within NIC=6 / TC=13

2.1.1. Identify the ADS-B Version

There are two steps to check the ADS-B version, this is due to the fact that ADS-B Version 0 is not included in any message.

1. Step 1: Check whether an aircraft is broadcasting ADS-B messages with TC=31 at all. If no message is ever reported, it is safe to assume that the version is Version 0
2. Step 2: If messages with TC=31 are received, check the version numbers located in the 41-43 bit of the payload (or 73-75 bit of the message).

After identifying the right ADS-B version for an aircraft (which does not change often), one can decode related TC=28 , TC=29 , and TC=31 messages accordingly.

2.2. Aircraft Operation Status

Operation status message is introduced since the Version 1 of ADS-B. And there are also slight differences in the structure of Aircraft Operation Messages between Version 1 and 2.

To understand about these versions, first take a look at the ADS-B version chapter.

The operation status is transmitted with Type Code 31 (TC=31). The structure of the message (in Version 1) is laid out as follows:

FIELD		MSG	MB	N-BITS
Downlink Format = 17		1		8
		8		
ICAO Address		9		24
		32		
Type Code = 31		33	1	5
		37	5	
Subtype Code		38	6	3
- 0: airborne				
- 1: surface				
- 2-7: reserved		40	8	
Airborne capacity class codes	Surface capacity class codes	41	9	16
		52	20	
	Length/width codes	53	21	(4)
56		24		
Operational mode code		57	25	16
		72	40	
ADS-B version number		73	41	3
		75	43	
NIC supplement bit		76	44	1
NACp: Navigation accuracy category		77	45	4
- position		80	48	
BAQ = 0	Reserved	81	49	2
		82	50	
SIL: Surveillance integrity level		83	51	2
		84	52	
NIC-BARO	TRK/HDG	85	53	1
HRD		86	54	1
Reserved		87	55	2
		88	56	

Acronyms:

- BAQ: Barometric Altitude Quality (always set to zero for airborne message subtype=1)
- HRD: Horizontal Reference Direction
 - 0: True North
 - 1: Magnetic North

In ADS-B Version 2 , most part of the message remains the same, we will only address the second half of the message, where the changes have been made.

FIELD		MSG	MB	N-BITS
Airborne operational mode code	Surface operational mode code	57	25	16
		72	40	
ADS-B version number		73	41	3
		75	43	
NIC supplement bit - A		76	44	1
NACp: Navigation accuracy category - position		77	45	4
		80	48	
GVA	Reserved	81	49	2
		82	50	
SIL: Surveillance integrity level		83	51	2
		84	52	
NIC-BARO	TRK/HDG	85	53	1
HRD		86	54	1
SIL supplement bit		87	55	1
Reserved		88	56	1

Acronyms:

- GVA: Geometric Vertical Accuracy - GNSS position source, 95% vertical figure of merit (VFOM)
 - 0: unknown or > 150 meters
 - 1: < 150 meters
 - 2: < 45 meters
 - 3: reserved
- SIL, NIC, NAC are also related to measurement uncertainty or accuracy.

- A lot or more details are given in the uncertainty chapter.

2.3. Uncertainty and accuracy

NIC, NAC, NUC, and SIL, those acronyms do sound confusing. They are categorical numbers for the integrity, accuracy, or uncertainties of the position measurements.

- **NUCp** : Navigation Uncertainty Category - Position
 - Values: 0 - 9
 - Version 0, 1, and 2
- **NUCr** : Navigation Uncertainty Category - Velocity (Rate)
 - Values: 0 - 4
 - Version 0
- **NIC** : Navigation Integrity Category
 - Values: 0 - 11
 - Version 1 and 2
- **NACp** : Navigation Accuracy Category - Position
 - Values: 0 - 11
 - Version 1 and 2
- **NACv** : Navigation Accuracy Category - Velocity
 - Values: 0 - 4
 - Version 1 and 2
- **SIL** : Surveillance Integrity Level
 - Values: 0 - 3
 - Version 1 and 2

For each category name, specific values are given corresponding to the numerical indicators. The relation of the category names and value names are:

- **NUCp** :
 - Horizontal Protection Limit (**HPL**)
 - 95% Containment Radius - Horizontal (**RCu**)
 - 95% Containment Radius - Vertical (**RCv**)
- **NUCr** :

- 95% Horizontal Velocity Error (HVE)
- 95% Vertical Velocity Error (VVE)
- NIC :
 - Horizontal Radius of Containment (RCu)
 - Vertical Protection Limit (VPL)
 - * a.k.a. Integrity Containment Region
- NACp :
 - 95% horizontal accuracy bounds, Estimated Position Uncertainty (EPU)
 - * a.k.a. Horizontal Figure of Merit (HFOM)
 - 95% vertical accuracy bounds, Vertical Estimated Position Uncertainty (VEPU)
 - * a.k.a. Vertical Figure of Merit (VFOM)
- NACv :
 - 95% horizontal accuracy bounds for velocity, Horizontal Figure of Merit (HFOM_r)
 - 95% vertical accuracy bounds for velocity, Vertical Figure of Merit (VFOM_r)
- SIL :
 - Probability of exceeding Horizontal Radius of Containment RCu (PE_RCu)
 - Probability of exceeding Vertical Integrity Containment Region VPL (PE_VPL)

Depending on the ADS-B versions, the bits to uncover these values maybe different. We are going to address the uncertainty measures by ADS-B versions.

To understand about these versions, first take a look at the ADS-B version chapter.

2.3.1. Version 0

NUCp

In ADS-B Version 0 , the accuracy is expressed as **Navigation Uncertainty Category - position** (NUCp). It is directly related (one-to-one relationship) with Type Code , as follows:

For surface position messages:

TC	0	5	6	7	8
NUCp	0	9	8	7	6

For airborne position with barometric altitude:

TC	9	10	11	12	13	14	15	16	17	18
NUCp	9	8	7	6	5	4	3	2	1	0

For airborne position with GNSS altitude:

TC	20	21	22
NUCp	9	8	0

Higher number of NUCp represents a higher confidence of in the position measurement (hence, a lower uncertainty). Horizontal Protection Limit (HPL) and Radius of containment for horizontal (RCu) are used to quantify the uncertainty.

For surface position (TC=5-8):

TC	NUCp	HPL	RCu
5	9	< 7.5 m	< 3 m
6	8	< 25 m	< 10 m
7	7	< 0.1 NM (185 m)	< 0.05 NM (93 m)
8	6	> 0.1 NM (185 m)	> 0.05 NM (93 m)

For airborne position with barometric altitude (TC=9-18):

TC	NUC _p	HPL	RC _u
9	9	< 7.5 m	< 3 m
10	8	< 25 m	< 10 m
11	7	< 0.1 NM (185 m)	< 0.05 NM (93 m)
12	6	< 0.2 NM (370 m)	< 0.1 NM (185 m)
13	5	< 0.5 NM (926 m)	< 0.25 NM (463 m)
14	4	< 1 NM (1852 m)	< 0.5 NM (926 m)
15	3	< 2 NM (3704 m)	< 1 NM (1852 m)
16	2	< 10 NM (18520 m)	< 5 NM (9260 m)
17	1	< 20 NM (37040 m)	< 10 NM (18520 m)
18	0	> 20 NM (37040 m)	> 10 NM (18520 m)

In the case of airborne position with GNSS height (TC=20-22), HPL and RC_u are defined slight differently. In addition, Radius of containment for vertical position (RC_v) is added:

TC	NUC _p	HPL	RC _u	RC _v
20	9	< 7.5 m	< 3 m	< 4 m
21	8	< 25 m	< 10 m	< 15 m
22	0	> 25 m	> 10 m	> 15 m

NUC_v

The **Navigation Uncertainty Category - velocity** (NUC_v) it is used to indicate the uncertainty of the horizontal and vertical speeds. Related bits are located at the airborne velocity message, TC=19 , message bit 43-45 (or payload bit 11-13). It defines the 95% of the error in horizontal and vertical speed.

NUC _p	HVE (95%)	VVE (95%)
0	unknown	unknown
1	< 10 m/s	< 15.2 m/s (50 fps)
2	< 3 m/s	< 4.5 m/s (15 fps)
3	< 1 m/s	< 1.5 m/s (5 fps)
4	< 0.3 m/s	< 0.46 m/s (1.5 fps)

2.3.2. Version 1

NIC

In ADS-B Version 1, **Navigation Integrity Category** (NIC) is introduced to provide more levels of uncertainty definitions. The NUC_p value is still kept, but has been moved to the new “operation status messages” (TC=31).

As for NIC , in Type Code : 7 and 11 , two NIC levels present in each code. In order to distinguish these two different levels, a **NIC Supplement Bit** (NICs) is introduced in “operation status messages” TC=31 (message bit 76 or payload bit 44). The relation of TC, NIC, and Rc are list in following tables.

For surface position (TC=5-8)

TC	NICs	NIC	Rc
5	0	11	< 7.5 m
6	0	10	< 25 m
7	1	9	< 75 m
	0	8	< 0.1 NM (185 m)
8	0	0	> 0.1 NM or Unknown

For airborne position with barometric altitude (TC=9-18):

TC	NICs	NIC	Rc	VPL
9	0	11	< 7.5 m	< 11 m
10	0	10	< 25 m	< 37.5 m
11	1	9	< 75 m	< 112 m
	0	8	< 0.1 NM (185 m)	
12	0	7	< 0.2 NM (370 m)	
13	0	6	< 0.5 NM (926 m)	
	1		< 0.6 NM (1111 m)	
14	0	5	< 1.0 NM (1852 m)	
15	0	4	< 2 NM (3704 m)	
16	1	3	< 4 NM (7408 m)	
	0	2	< 8 NM (14.8 km)	
17	0	1	< 20 NM (37.0 km)	
18	0	0	> 20 NM or Unknown	

In the case of airborne position with GNSS height (TC=20-22), Rc is defined slightly differently:

TC	NIC	Rc	VPL
20	11	< 7.5 m	< 11 m
21	10	< 25 m	< 37.5 m
22	0	> 25 m	> 112 m

NACp

In addition to NIC , **Navigation Accuracy Category - Position** (NACp) is also introduced in ADS-B Version 1 . NACp can be found in:

- TC=29 , target state and status message, message bit 72-75 (or payload bit 40-43)
- TC=31 , operational status message, message bit 77-80 (or payload bit 45-48)

With NACp , one can find out the 95% horizontal and vertical accuracy bounds (EPU and

VEPU, or HFOM and VFOM). NACp and NIC usually share the same value, and their values are closely related.

$$\begin{aligned} \text{EPU} &\approx R_c/2.5 & \text{NAC, NIC} &\geq 9 \\ \text{EPU} &\approx R_c/2.0 & \text{NAC, NIC} &> 9 \end{aligned} \quad (2.1)$$

For ADS-B, NACp is also indicated in the operational status messages (TC=31). NACp and corresponding EPU / VEPu are:

NACp	EPU (HFOM)	VEPU (VFOM)
11	< 3 m	< 4 m
10	< 10 m	< 15 m
9	< 30 m	< 45 m
8	< 0.05 NM (93 m)	
7	< 0.1 NM (185 m)	
6	< 0.3 NM (556 m)	
5	< 0.5 NM (926 m)	
4	< 1.0 NM (1852 m)	
3	< 2 NM (3704 m)	
2	< 4 NM (7408 m)	
1	< 10 NM (18520 km)	
0	> 10 NM or Unknown	

NACv

The **Navigation Accuracy Category - Velocity** (NACv) is introduced in Version 1 to replace the NUCv from Version 0. The bits are located at the same location and have the same definitions of values. It can be found in airborne velocity message, TC=19, message bit 43-45 (or payload bit 11-13). It defines the 95% of the error in horizontal and vertical speed.

NAVc	HFOMr (95%)	VFOMr (95%)
0	unknown	unknown
1	< 10 m/s	< 15.2 m/s (50 fps)
2	< 3 m/s	< 4.5 m/s (15 fps)
3	< 1 m/s	< 1.5 m/s (5 fps)
4	< 0.3 m/s	< 0.46 m/s (1.5 fps)

SIL

On the other hand, **Surveillance Integrity Level** (SIL) is introduced in Version 1 to indicate the probability of measurement exceeding the containment radius horizontally (PE_RCu) and vertically (PE_VPL).

SIL value can be found in two different locations:

- TC=29 , target state and status message, message bit 77-78 (or payload bit 45-46)
- TC=31 , operational status message, message bit 83-84 (or payload bit 51-52)

The definitions are as follows:

SIL	PE_RCu	PE_VPL
0	unknown	unknown
1	< 1 x 10 ⁻³	< 1 x 10 ⁻³
2	< 1 x 10 ⁻⁵	< 1 x 10 ⁻⁵
3	< 1 x 10 ⁻⁷	< 2 x 10 ⁻⁷

The unit for PE_RCu and PE_VPL is per hour or per sample.

2.3.3. Version 2

NIC

In ADS-B Version 2 , levels of accuracies are further divided by two additional NIC supplement bits. In Version 2 , we call them NIC Supplement Bit A (NICa), NIC Supplement Bit B (NICb), and NIC Supplement Bit C (NICc).

- **NICa** is in the same location as in ADS-B **Version 1**, which is located in the operational status message **TC=31** (message bit 76 or payload bit 44).
- **NICb** is located in the airborne position message (**TC=9-18** , message bit 40 or payload bit 8), where the “single antenna flag” was located in previous ADS-B versions.
- **NICc** is also located in the operational status message **TC=31** (message bit 52 or payload bit 20).

Gather the NIC supplement bits, together with the **Type Code** , accuracies are defined.

For surface possible message (**TC=5-8**), using **TC** , **NICa** , and **NICc** :

TC	NICa	NICc	NIC	Rc
5	0	0	11	< 7.5 m
6	0	0	10	< 25 m
7	1	0	9	< 75 m
	0	0	8	< 0.1 NM (185 m)
8	1	1	7	< 0.2 NM (370 m)
	1	0	6	< 0.3 NM (556 m)
	0	1		< 0.6 NM (1111 m)
	0	0	0	> 0.6 NM or Unknown

For airborne possible message (**TC=9-18**), using **TC** , **NICa** , and **NICc** :

TC	NICa	NICb	NIC	Rc
9	0	0	11	< 7.5 m
10	0	0	10	< 25 m
11	1	1	9	< 75 m
	0	0	8	< 0.1 NM (185 m)
12	0	0	7	< 0.2 NM (370 m)
13	0	1	6	< 0.3 NM (556 m)
	0	0		< 0.5 NM (926 m)
	1	1		< 0.6 NM (1111 m)
14	0	0	5	< 1.0 NM (1852 m)
15	0	0	4	< 2 NM (3704 m)
16	1	1	3	< 4 NM (7408 m)
	0	0	2	< 8 NM (14.8 km)
17	0	0	1	< 20 NM (37.0 km)
18	0	0	0	> 20 NM or Unknown

In the case of airborne position with GNSS height (TC=20-22), the table remains the same as previous version.

TC	NIC	Rc
20	11	< 7.5 m
21	10	< 25 m
22	0	> 25 m

NACp

NACp in Version 2 remains the same as in Version 1 .

NACv

NACv in Version 2 remains the same as in Version 1 .

SIL

In Version 2 , an additional **SIL supplement bit** (**SILs**) is introduced to distinguish if the value is based on per flight hour or per sample.

The same as in Version 1 , **SIL** value can be found in two different locations:

- TC=29 , target state and status message, message bit 77-78 (or payload bit 45-46)
- TC=31 , operational status message, message bit 83-84 (or payload bit 51-52)

SILs bit can also be found at two different locations:

- TC=29 , target state and status message, message bit 40 (or payload bit 8)
- TC=31 , operational status message, message bit 87 (or payload bit 55)
- **SILs=0** : probability is “per hour” based
- **SILs=1** : probability is “per sample” based

The same values from Version 1 are kept:

SIL	PE_RCu	PE_VPL
0	unknown	unknown
1	$< 1 \times 10^{-3}$	$< 1 \times 10^{-3}$
2	$< 1 \times 10^{-5}$	$< 1 \times 10^{-5}$
3	$< 1 \times 10^{-7}$	$< 2 \times 10^{-7}$

Chapter 3

Enhanced Mode-S

The Mode-S Enhanced Surveillance (EHS) provides Air Traffic Control (ATC) more information than what is included in the Mode-S Elementary Surveillance (ELS).

There are quite a few very interesting data contained within various types of the EHS messages. Such as: airspeeds (IAS, TAS, Mach), roll angles, track angles, track angle rates, selected altitude, magnetic heading, vertical rate, etc.

There are a few challenges to decode this information:

1. Which aircraft does one message come from?
2. What is the type of one message (aka which BDS code) most likely to be?
3. How reliable is the information that has been decoded?

3.1. Enhanced Mode-S Basics

3.1.1. Downlink Format and message structure

DF 20 and DF 21 are used for downlink messages.

The same as ADS-B, in all Mode-S messages, the first 5 bits contain the Downlink Format. The same identification process can be used to discover EHS messages. So the EHS messages starting bits are:

```
DF20 - 10100  
DF21 - 10101
```

The message is structured as follows, where the digit represents the number of binary digits:


```
Message:      A0001838CA380031440000F24177

Data:         A0001838CA380031440000
Parity:                               F24177

Encode data:  CE2CA7

ICA0:         [F24177] XOR [CE2CA7] => [3C6DD0]
```

For the implementation of the CRC encoder, refer to the pyModeS library `pyModeS.common.crc(msg, e`

3.1.3. BDS (Binary Data Selector)

In simple words, BDS is a number (usually a 2-digit hexadecimal) that defines the type of message we are looking at. Both ADS-B messages and other types of Mods-S messages are all assigned their distinctive BDS number. However, it is **nowhere** to be found in the messages.

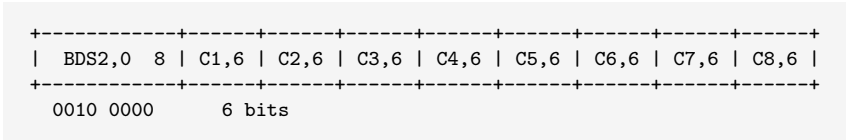
When SSR interrogates an aircraft, a BDS code is included in the request message (Uplink Format - UF 4, 5, 20, or 21). This BDS code is then used by the aircraft transponder to register the type of message to be sent. But when the downlink message is transmitted, its BDS code is not included in the message (because the SSR knows what kind of message it requested). Good news for them, but challenges for us.

Here are some BDS codes that we are interested in, where additional parameters about an aircraft can be found:

```
BDS 2,0  Aircraft identification
BDS 2,1  Aircraft and airline registration markings
BDS 4,0  Selected vertical intention
BDS 4,4  Meteorological routine air report
BDS 5,0  Track and turn report
BDS 6,0  Heading and speed report
```

3.2. Aircraft identification (BDS 2,0)

Similar to an ADS-B aircraft identification message, the callsign of an aircraft can be decoded in the same way. For the 56-bit MB (message, Comm-B) field, information decodes as follows:



Here, 8 bits are 0010 0000 (2,0 in hexadecimal) and the rest of chars are 6 bits each. To decode the chars, the same char map as ADS-B is used:


```
'#ABCDEFGHIJKLMNOPQRSTUVWXYZ####_#####0123456789#####'
```

Example:

```
MSG: A000083E202CC371C31DE0AA1CCF
DATA:      202CC371C31DE0

BIN: 0010 0000 001011 001100 001101 110001 110000 110001 110111 100000
HEX:   2    0
DEC:      11    12    13    49    48    49    55    32
CHR:      K     L     M     1     0     1     7     _

ID: KLM1017
```

3.3. Selected intention (BDS 4,0)

In BDS 4,0, information such as aircraft selected altitude and barometric pressure settings are given. The 56-bit MB field is structured as follows:

FIELD	MB	N-BITS	
Status	1	1	
MCP/FCU selected altitude	2	12	**
range = [0, 65520] ft			
LSB: 16 ft	13		
Status	14	1	
FMS selected altitude	15	12	**
range = [0, 65520] ft			
LSB: 16 ft	26		
Status	27	1	
Barometric pressure setting	28	12	**
-> Note: actual value minus 800			
range = [0, 410] mb			
LSB: 0.1 mb	39		
Reserved	40	8	
-> set to ZEROS	47		
Status	48	1	
-> next 3 fields			
Mode: VNAV	49	1	
Mode: Alt hold	50	1	
Mode: Approach	51	1	
Reserved	52	2	
-> set to ZEROS	53		
Status	54	1	
Target alt source	55	2	
-> 00: Unknown			
-> 01: Aircraft altitude			
-> 10: FCU/MCP selected altitude			
-> 11: FMS selected altitude	56		

An example:

```
MSG:  A000029C85E42F313000007047D3
MB:      85E42F31300000

-----
MB BIN:  1 0000101111100 1 0000101111100 1 100010011000 00000000 0 0 0 0 00 0 00
-----
STATUS:  1
MCP:      188 (x16)
-----
STATUS:  1
FMS:      188 (x16)
-----
STATUS:  1
BARO:      2200 (x0.1 + 800)
-----
FINAL:    3008 ft      3008 ft      1020 mb
-----
```

3.4. Track and turn (BDS 5,0)

Within the BDS 5,0 message, five different types of aircraft states are given, mostly related with the turns:

- roll angle
- true track angle
- ground speed
- track angle rate
- true airspeed

The 56-bit MB field is structured as follows:

FIELD	MB	N-BITS
Status	1	1
Sign, 1 -> Use two's complement	1	1
Roll angle	3	9
range = [-90, 90] degrees		
LSB: 45/256 degree	11	
Status	12	1
Sign, 1 -> Use two's complement	13	1
True track angle	14	10
range = [-180, 180] degrees		
LSB: 90/512 degree	23	
Status	24	1
Ground speed	25	10
range = [0, 2046] knots		
LSB: 2 knots	34	
Status	35	1
Sign, 1 -> Use two's complement	36	1
Track angle rate	37	9
range = [-16, 16] degrees		
LSB: 8/256 degree / second	45	
Status	46	1
True airspeed	47	10
range = [0, 2046] knots		
LSB: 2 knots	56	

When a parameter is signed (SIGN = 1), the two's complement should be used to calculate the value. The value can be calculated as follows:

$$(-2^{*(\text{Number of bits parameter})} + \text{value}) * \text{LSB}.$$

The number of bits for roll angle and track rate is 9, and for true track angle 10. An

example calculation with a signed parameter is given in (Airspeed and Heading) <https://mode-s.org/decode/ehs/bds60-airspeed.html>.

An example:

```
MSG: A000139381951536E024D4CCF6B5
MB: 81951536E024D4
```

```
-----
MB BIN: 1 0 000001100 1 0 1010001010 1 0011011011 1 0 000000100 1 0011010100
-----
```

```
STATUS: 1
SIGN: 0
ROLL: 12 (x45/256)
-----
```

```
STATUS: 1
SIGN: 0
TRACK ANGLE: 650 (x90/512)
-----
```

```
STATUS: 1
GROUND SPEED: 219 (x2)
-----
```

```
STATUS: 1
SIGN: 0
TRACK ANGLE RATE: 4 (x8/256)
-----
```

```
STATUS: 1
TRUE AIRSPEED: 212 (x2)
-----
```

```
FINAL: 2.1 deg 114.3 deg 438 kt 0.1 deg/s 424 kt
-----
```

Of course, all fields are not always available in each of DBS 5,0 message. For the information that is not available, status bits are set to 0.

3.5. Heading and speed (BDS 6,0)

Within the BDS 6,0 message, five different types of aircraft states are given:

- magnetic heading
- indicated airspeed
- Mach number
- barometric altitude rate
- inertial vertical rate

The 56-bit MB field is structured as follows:

FIELD	MB	N-BITS
Status	1	1
Sign, 1 -> Use two's complement	1	1
Magnetic heading	3	10
range = [-180, 180] degrees		
LSB: 90/512 degree	12	
Status	13	1
Indicated airspeed	14	10
range = [0, 1023] knots		
LSB: 1 knots	23	
Status	24	1
Mach number	25	10
range = [0, 4.092] Mach		
LSB: 2.048 / 512 Mach	34	
Status	35	1
SIGN 1 -> Use two's complement	36	1
Barometric altitude rate	37	9
range = [-16384, 16352] ft/min		
LSB: 32 ft/min	45	
Status	46	1
SIGN 1 -> Use two's complement	47	1
Inertial altitude rate	48	9
range = [-16384, 16352] ft/min		
LSB: 32 ft/min	56	

When a parameter is signed (SIGN = 1), the two's complement should be used to calculate the value. The value can be calculated as follows:

$$(-2^{\{\text{Number of bits parameter}\}} + \text{value}) * \text{LSB}.$$

The number of bits for magnetic heading is 10, and for barometric/inertial altitude rate

9.

An example:

```
MSG: A000029CFFBAA11E2004727281F1
MB:   FFBAA11E200472
```

```
-----
MB BIN:  1 1 1111111011 1 0101010000 1 0001111000 1 0 000000000 1 0 001110010
-----
```

```
STATUS:  1
SIGN:    1 (Two's complement  $-2^{\{10\}} + \text{value}$ )
HEADING: 1019 (x90/512)
-----
```

```
STATUS:  1
IAS:     336
-----
```

```
STATUS:  1
MACH:    120 (x2.048/512)
-----
```

```
STATUS:  1
SIGN:    0
VERTIVAL RATE - BARO: 0 (x32)
-----
```

```
STATUS:  1
SIGN:    0
VERTICAL RATE - INERTIAL: 114 (x32)
-----
```

```
FINAL:    -0.88 deg*   336 kt   0.48 Mach   0 ft/min   3648 ft/min
-----
```

```
* Two's complement calculation:  $(-2^{\{10\}} + 1019) * 90/512 = -0.88 \text{ degree } (= 359.12)$ 
```

Chapter 4

About the book

4.1. Related resources

This guide document is shared on GitHub and mode-s.org. Please feel free to help us improving it.

Links to this guide document:

- (Rst source) <https://github.com/junzis/the-1090mhz-riddle>
- (Live book) <http://mode-s.org>

You can download the pyModeS tool from GitHub, which is a Python implementation of all (and more) message types described here:

- (GitHub) <https://github.com/junzis/pyModeS>

4.2. Contributors

From TU Delft:

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- Jacco Hoekstra, Prof.dr.ir, TU Delft
- Joost EllerBroek, Dr.ir, TU Delft

From GitHub community:

- <https://github.com/junzis/the-1090mhz-riddle/graphs/contributors>

4.3. Contact

Since the start of the this project, I have received many questions by email. However, the best way to post your questions is using the **GitHub Issues**. This way, your questions and my answers can help others as well:

- Related with this book: <https://github.com/junzis/the-1090mhz-riddle/issues>

- Related with pyModeS: <https://github.com/junzis/pyModes/issues>

Anyhow, still feel free to drop me a messages at: **j.sun-1[at]tudelft.nl**

4.4. References

- Technical Provisions for Mode S Services and Extended Squitter. International Civil Aviation Organization, 2008.
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- Fundamentals of mode s parity coding, tech. rep., Massachusetts Institute of Technology, Lincoln Laboratory, 1984.
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- A Very Simple ADSB Receiver