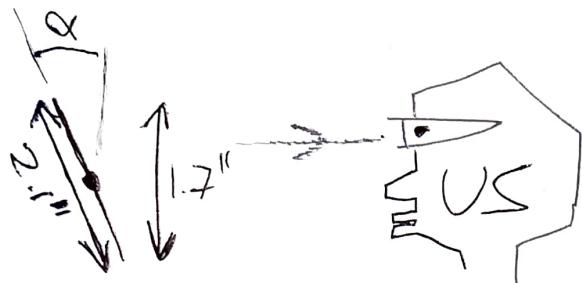


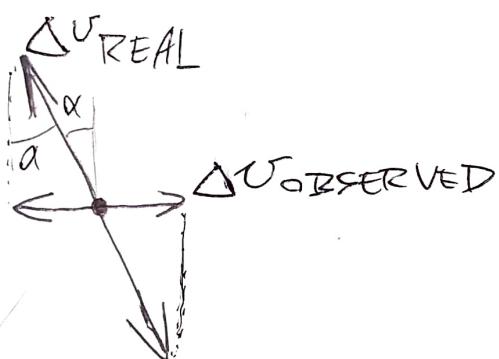
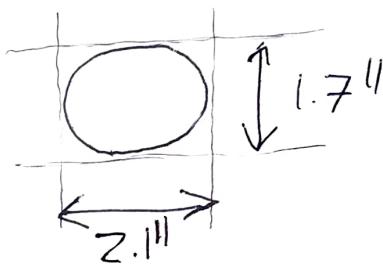
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(i)

SIDE VIEW



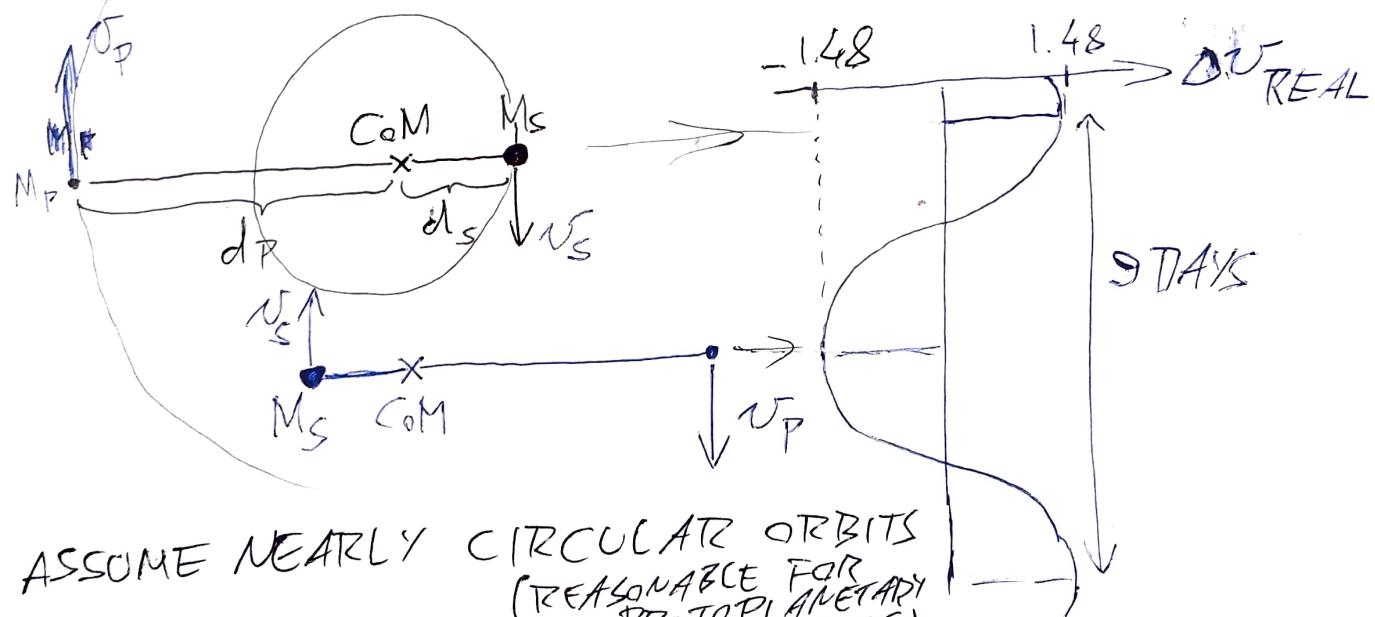
FRONT VIEW



$$\Delta V_{REAL} = \frac{\Delta V_{OBSERVED}}{\cos \alpha}$$

$$= \frac{\Delta V_{OBSERVED}}{\cos(\arccos \frac{1.7}{2.1})}$$

$$= 1.2 \cdot \frac{2.1}{1.7} = 1.48 \frac{\text{km}}{\text{s}}$$



ASSUME NEARLY CIRCULAR ORBITS
(REASONABLE FOR PROTOPLANETARY DISCS)

$$2\pi d_S / v_S = P$$

$$d_S = \frac{P v_S}{2\pi} = \frac{9.24 \cdot 60^2 \cdot 1.48 \cdot 10^3}{2\pi} = 1.83 \cdot 10^8 \text{ m}$$

FROM
DEFINITION
OF C.M =

$$M_p dp = M_s ds \Rightarrow M_p = M_s \frac{ds}{dp}$$

FORCE BALANCE:

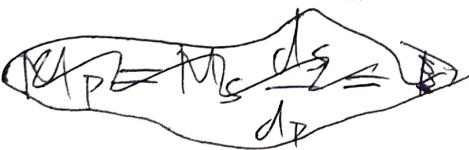
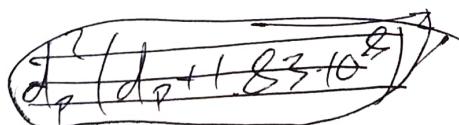
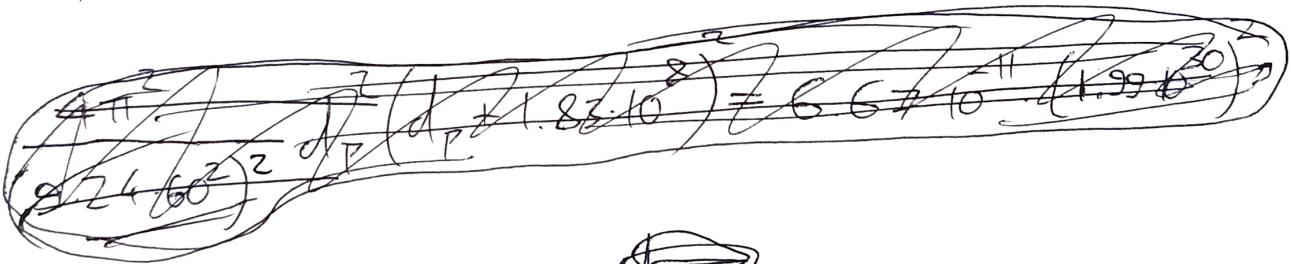
$$\frac{v_p^2}{dp} = G \frac{M_p M_s}{(d_p + d_s)^2}$$

$$v_p = d_p \cdot \omega = d_p \frac{2\pi}{P} \Rightarrow v_p^2 = 4\pi^2 \left(\frac{d_p}{P}\right)^2$$

REWRITE FORCE BALANCE EQUATION:

$$\frac{4\pi^2}{P^2} d_p = G \frac{M_s \frac{ds}{dp} M_s}{(d_p + d_s)^2}$$

$$\frac{4\pi^2}{P^2} d_p^2 (d_p + d_s)^2 = G M_s^2$$



SOLVE NUMERICALLY
OR SAY: $d_p \gg d_s$, THEN

$$d_p \approx M_s \frac{P^2}{4\pi^2}$$

SOLVE NUMERICALLY OR SAY: $d_p \Rightarrow d_s$
(REASONABLE)

$$d_p^4 = \frac{GM_s^2 P^2}{4\pi^2}$$

$$d_p = \sqrt[4]{\frac{GM_s^2 P^2}{4\pi^2}} = 4.8 \cdot 10^{13} \text{ m}$$

(WHICH IS CONVENIENTLY
MUCH LARGER THAN $1.83 \cdot 10^8 \text{ m}$)

$$M_p = M_s \frac{d_s}{d_p} = 2 \cdot 10^{30} \cdot \frac{1.83 \cdot 10^8}{4.8 \cdot 10^{13}} = \underline{\underline{7.6 \cdot 10^{24} \text{ kg}}}$$

$$\text{ORBITAL RADIUS} = \underline{\underline{4.8 \cdot 10^{13} \text{ m}}}$$

Eccentricity deduction:

SINUSOIDAL VARIATION \Rightarrow ORBIT PROBABLY
HIGHLY CIRCULAR
SO $e \ll 1, e \approx 0$.