

# Dynamics Formula Sheet

## Quiz I

### Kinematics

$$\vec{v} = \frac{dx}{dt}$$

$$\vec{a} = \frac{d\vec{v}}{dt} = \frac{d\vec{x}}{dt}$$

Note: Using the chain rule,  $\vec{a}$  can be rewritten

$$\vec{a} = \vec{v} \frac{d\vec{x}}{dt}$$

### Constant Acceleration Kinematics

$$\vec{v} = \vec{v}_o + \vec{a}_o t$$

$$v_f^2 = v_o^2 + 2\vec{a}_o \Delta \vec{x}$$

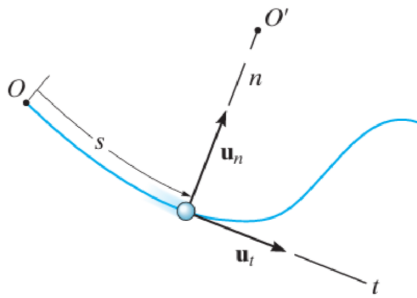
$$\vec{x} = \vec{x}_o + \vec{v}_o t + \frac{1}{2} \vec{a}_o t^2$$

### Relative Motion

$$x_{B/A} = x_B - x_A \quad B \text{ relative to } A$$

Note: Subscripts cancel  $x_B = x_A + x_{B/A}$

### Normal and Tangential



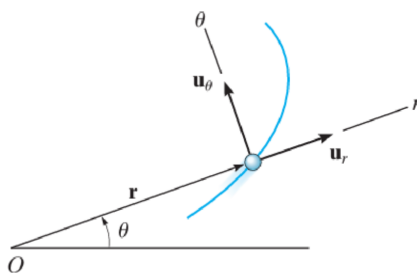
$$\vec{v} = v \hat{u}_t$$

$$\vec{a} = \frac{dv}{dt} \hat{u}_t + \frac{v^2}{\rho} \hat{u}_n$$

$$\rho = \frac{\left[1 + \left(\frac{dy}{dx}\right)^2\right]^{3/2}}{\left|\frac{d^2y}{dx^2}\right|} \quad (\text{Radius of curvature})$$

Note: In 3d, define  $\hat{u}_b = \hat{u}_t \times \hat{u}_n$

### Cylindrical (Radial and Transverse)



$$\vec{v} = \dot{r} \hat{u}_r + r \dot{\theta} \hat{u}_\theta$$

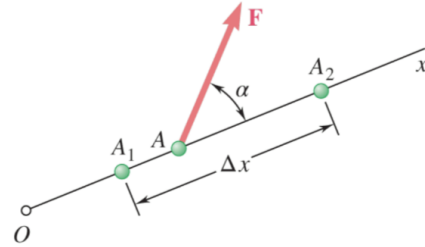
$$\vec{a} = \left[\ddot{r} - r\dot{\theta}^2\right] \hat{u}_r + \left[r\ddot{\theta} + 2\dot{r}\dot{\theta}\right] \hat{u}_\theta$$

Note: In 3d, add a  $\hat{u}_z$  where  $\vec{r}_z = z\hat{u}_z, \vec{v}_z = \dot{z}\hat{u}_z$  and  $\vec{a}_z = \ddot{z}\hat{u}_z$

### Work

$$U_{1 \rightarrow 2} = \int_{r_1}^{r_2} \vec{F} \cdot d\vec{r}$$

Constant force in rectilinear motion:  $U_{1 \rightarrow 2} = F \cos(\alpha) \Delta x$



### Work by gravity

$$U_{g,1 \rightarrow 2} = -W \Delta y \quad (\text{Weight})$$

$$U_{g,1 \rightarrow 2} = \frac{GMm}{r_2} - \frac{GMm}{r_1} \quad (\text{In space})$$

### Work by spring

$$U_{s,1 \rightarrow 2} = -\frac{1}{2}k(x_2^2 - x_1^2) = \frac{1}{2}k(x_1^2 - x_2^2)$$

### Potential Energy

#### Gravitational Potential

$$V_g = Wy = mgh \quad (\text{On Earth})$$

$$V_g = -\frac{GMm}{r} = -\frac{WR^2}{r} \quad (\text{In space})$$

#### Elastic Potential

$$V_e = \frac{1}{2}kx^2$$

### Work and Energy

$$U_{1 \rightarrow 2} = T_2 - T_1$$

$$T_2 = T_1 + U_{1 \rightarrow 2}$$

Work of a force = change in kinetic Energy

Kinetic energy = capacity to do work

Note:  $T = \frac{1}{2}mv^2$

### Conservation of Energy

$$T_1 + V_{g1} + V_{e1} + U_{1 \rightarrow 2}^{NC} = T_2 + V_{g2} + V_{e2}$$

### Power and Efficiency

#### Power

$$P = \frac{dU}{dt} = \vec{F} \cdot \vec{v}$$

#### Efficiency

$$\eta = \frac{P_{out}}{P_{in}}$$

# Orbital Mechanics

## Newton's Law of Gravitation

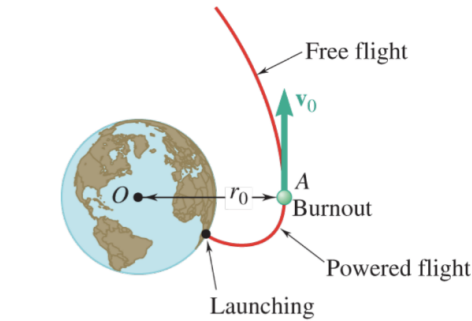
$$F_g = \frac{GMm}{r^2} \quad (G = 6.67 \times 10^{-11} \text{ m}^3/\text{kg s}^2)$$

**Differential equation for shape of orbital trajectory (derived by applying central force motion):**

$$\frac{d^2 u}{d\theta^2} + u = \frac{GM}{h^2}$$

- $u = 1/r$ ,  $r$  is distance from centers of mass
- $h = r^2 \dot{\theta}$ , angular momentum per unit mass (where angular momentum =  $H_o = hm = rmv_\theta$ )
- $M$  is the mass of the central body

**Solutions to this differential equation are conic paths:**



$$u = \frac{1}{r} = \frac{GM}{h^2} + C \cos \theta = \frac{GM}{h^2} (1 + \varepsilon \cos \theta)$$

Where  $C$  and  $h$  are constants defined by the instantaneous  $r$  and  $v$  at any point along the orbit:

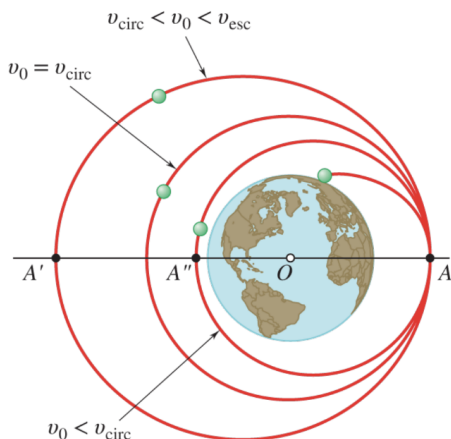
- $h = r^2 \dot{\theta} = r_0 v_0$
- $C = \frac{1}{r_0} - \frac{GM}{h^2} = \frac{1}{r_0} - \frac{GM}{r_0^2 v_0^2}$

With **eccentricity**:

$$\varepsilon = \frac{C}{GM/h^2} = \frac{Ch^2}{GM}$$

Note: Hyperbola ( $\varepsilon > 1$ ), Parabola ( $\varepsilon = 1$ ), Ellipse ( $\varepsilon < 1$ ), Circle ( $\varepsilon = 0$ )

## Initial conditions for each path



**Parabolic path (barely escaping orbit)  $\varepsilon = 1$ :**

At  $\theta = 0$  and  $r = r_o$  (start of free flight):

$$\frac{1}{r_o} = \frac{GM}{r_o^2 v_o^2} (1 + 1) \quad (h = r_o v_o)$$

$$v_{esc} = v_o = \sqrt{\frac{2GM}{r_o}}$$

**Circular orbit  $\varepsilon = 0$**

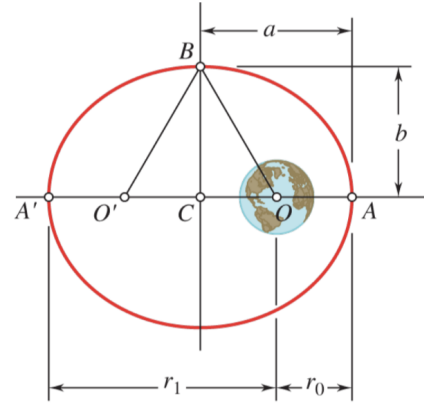
At  $\theta = 0$  and  $r = r_o$  (start of free flight):

$$\frac{1}{r_o} = \frac{GM}{r_o^2 v_o^2} \quad (h = r_o v_o)$$

$$v_{circ} = v_o = \sqrt{\frac{GM}{r_o}}$$

Note:  $GM = gR^2$  where  $R$  is the radius of Earth

## Periodic time ( $\tau$ ):



Time to complete 1 orbit:

$$\tau = \frac{\text{Area of elliptical orbit}}{\text{Areal velocity}}$$

From geometry, the major and minor axes are:

$$a = \frac{1}{2}(r_0 + r_1)$$

$$b = \sqrt{r_0 r_1}$$

The area of an ellipse is:

$$A = \pi ab$$

Areal velocity (constant):

$$dA = \frac{1}{2} r^2 d\theta$$

$$\frac{dA}{dt} = \frac{1}{2} r^2 \frac{d\theta}{dt} = \frac{h}{2} \quad (h = r^2 \dot{\theta})$$

Thus periodic time is:

$$\tau = \frac{A}{h/2} = \frac{2A}{h} = \frac{2\pi ab}{h}$$