

$$1. \quad a_{\text{comet}} = 8 \text{ AU} = 8 \times 1.5 \times 10^8 = 1.2 \times 10^9 \text{ km}. \quad \mu_s = 1.3 \times 10^{11} \text{ km}^3/\text{s}^2$$

$$e_{\text{comet}} = 0.8.$$

$$v_J = 13 \quad \theta_s^* = 90^\circ$$

$$r_J = r_{J_{\text{ref}}} = 5.2 \text{ AU} = 7.8 \times 10^8 \text{ km} \quad \gamma_J = 5^\circ$$

$$e_J = \sqrt{\left(\frac{r_J v_J^2}{\mu_s} - 1\right)^2 \cdot \cos^2 \gamma_J + \sin^2 \gamma_J} = \sqrt{\left(\frac{7.8 \times 10^8 \times 13^2}{1.3 \times 10^{11}} - 1\right)^2 \cdot \cos^2(5^\circ) + \sin^2(5^\circ)} = \underline{0.0883}$$

$$a_J = -\mu_s / 2 \left(\frac{v_J^2}{2} - \frac{\mu_s}{r_J} \right) = -1.3 \times 10^{11} / 2 \left(\frac{13^2}{2} - \frac{1.3 \times 10^{11}}{7.8 \times 10^8} \right) = \underline{7.9108 \times 10^8 \text{ km}}.$$

$$25 R_J = 25 \times 7 \times 10^4 = 1.75 \times 10^6 \text{ km} = r_{\text{p.h.}}$$

a) $\boxed{r_c^- = r_J = 7.8 \times 10^8 \text{ km}}$

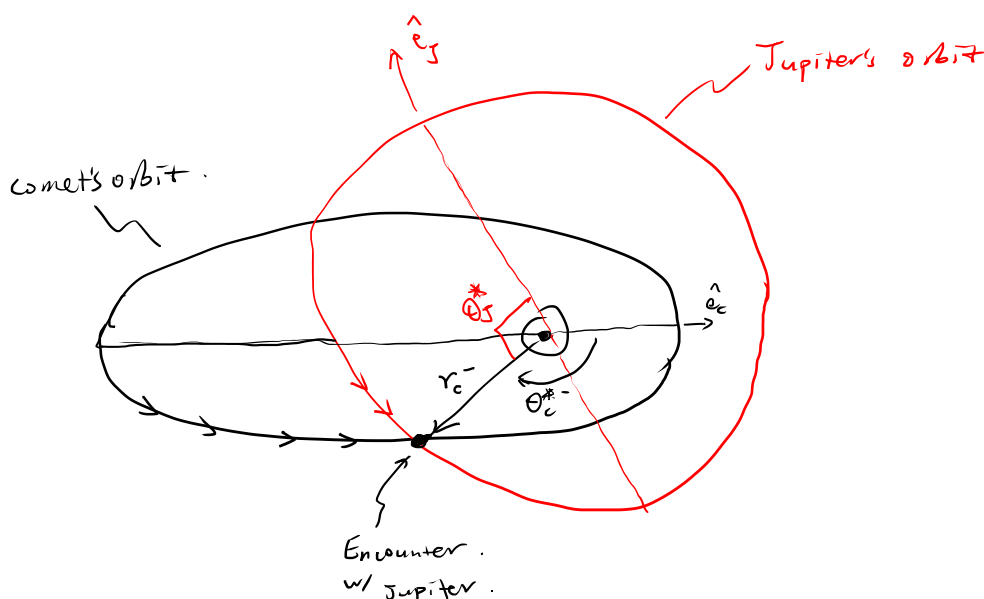
$$P_c = a_c \cdot (1 - e_c^2) = 1.2 \times 10^9 \times (1 - 0.8^2) = 4.32 \times 10^8 \text{ km}.$$

$$\theta_c^* = \cos^{-1} \left(\frac{P_c}{r_c \cdot e_c} - \frac{1}{e_c} \right) = \cos^{-1} \left(\frac{4.32 \times 10^8}{7.8 \times 10^8 \times 0.8} - \frac{1}{0.8} \right) = \boxed{-123.8964^\circ} \text{ inbound/descend.}$$

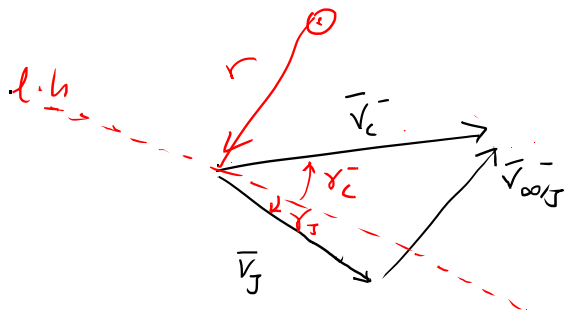
$$v_{\text{comet}} = v_c^- = \sqrt{\frac{2\mu_s}{r_c} - \frac{\mu_s}{a_c}} = \sqrt{\frac{2 \times 1.3 \times 10^{11}}{7.8 \times 10^8} - \frac{1.3 \times 10^{11}}{1.2 \times 10^9}} = \boxed{15 \text{ km/s}}$$

$$\gamma_c^- = \cos^{-1} \left(\frac{\sqrt{\mu_s P_c}}{r_c v_c} \right) = \cos^{-1} \left(\frac{\sqrt{1.3 \times 10^{11} \times 4.32 \times 10^8}}{7.8 \times 10^8 \times 15} \right) = \boxed{-50.1699^\circ} \text{ inbound/descend}$$

$$E_c^- = 2 \tan^{-1} \left(\tan \frac{\theta_c^*}{2} \cdot \sqrt{\frac{1-e_c}{1+e_c}} \right) = -64.0556^\circ$$



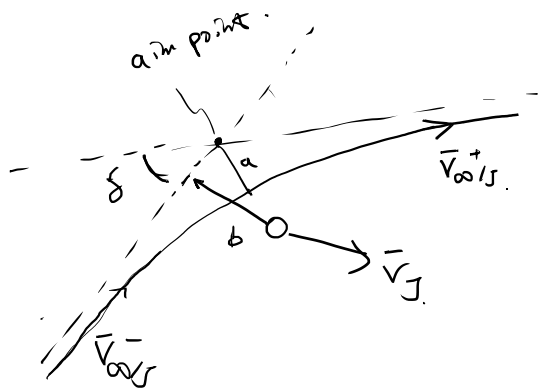
1 b)



$$V_{\infty/J} = \sqrt{V_c^2 + V_J^2 - 2V_c V_J \cos(|\delta_J| + |\delta_c|)}$$

$$= \sqrt{15^2 + 13^2 - 2 \times 15 \times 13 \times \cos(5 + 50.1699)}$$

$$= 13.0864 \text{ km/s.}$$



$$r_{p.h} = 25 R_J = 1.75 \times 10^6 \text{ km.}$$

$$|a_h| = \frac{M_J}{V_{\infty/J}^2} = \frac{1.3 \times 10^8}{13.0864^2} = \boxed{7.5911 \times 10^5 \text{ km}}$$

$$= \boxed{10.8444 R_J}$$

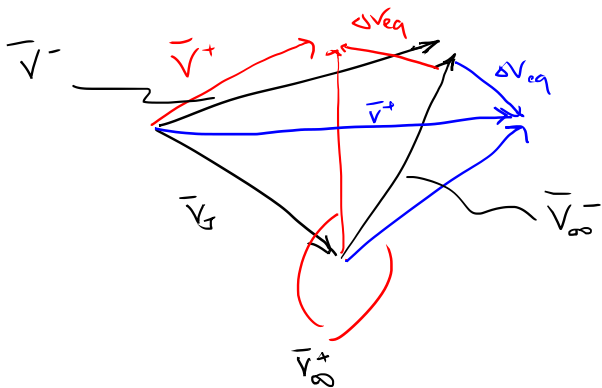
$$e_h = \frac{r_{p.h}}{|a_h|} + 1 = \frac{1.75 \times 10^6}{7.5911 \times 10^5} + 1 = 3.3053$$

$$\delta = 2 \cdot \sin^{-1} \frac{1}{e_h} = \boxed{35.2206^\circ}$$

$$|b_h| = |a_h| \cdot \sqrt{e_h^2 - 1} = \boxed{2.3915 \times 10^6 \text{ km}}$$

$$= \boxed{34.1643 R_J}$$

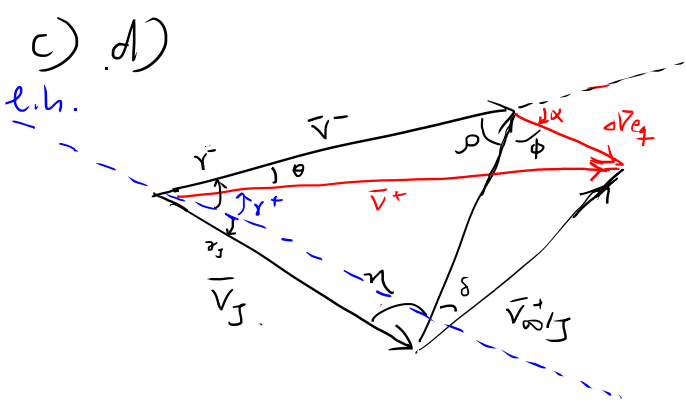
The comet will pass behind Jupiter, because we know the energy of the comet's orbit increases.



As shown here, \bar{V}^+ with the blue ΔV_{eq} has a larger magnitude, indicating increased energy. The blue one corresponds to passing behind Jupiter.

c) d)

l.h.



$$\phi = \frac{180^\circ - \delta}{2} = \frac{180^\circ - 35.2206^\circ}{2} = 72.3897^\circ$$

$$\frac{\sin \rho}{V_J} = \frac{\sin(181 + 18^\circ)}{V_{\infty/J}}$$

$$\rho = \sin^{-1}\left(\frac{13}{13.0864} \cdot \sin(55.1699)\right) = 54.6299^\circ$$

$$\begin{aligned} V^+ &= \sqrt{V^-^2 + \Delta V_{eq}^2 - 2 V^- \Delta V_{eq} \cos(\rho + \phi)} \\ &= \sqrt{15^2 + 7.9184^2 - 2 \times 15 \times 7.9184 \cdot \cos(127.0196)} \\ &= \boxed{20.7540 \text{ km/s}} \end{aligned}$$

$$\begin{aligned} |\Delta V_{eq}| &= \sqrt{2 V_{\infty/J}^2 (1 - \cos \delta)} = \sqrt{2 \times 13.0864^2 \times (1 - 35.2206)} \\ &= \boxed{7.9184 \text{ km/s}} \end{aligned}$$

$$\alpha = 180^\circ - \phi - \rho = \boxed{52.9804^\circ}$$

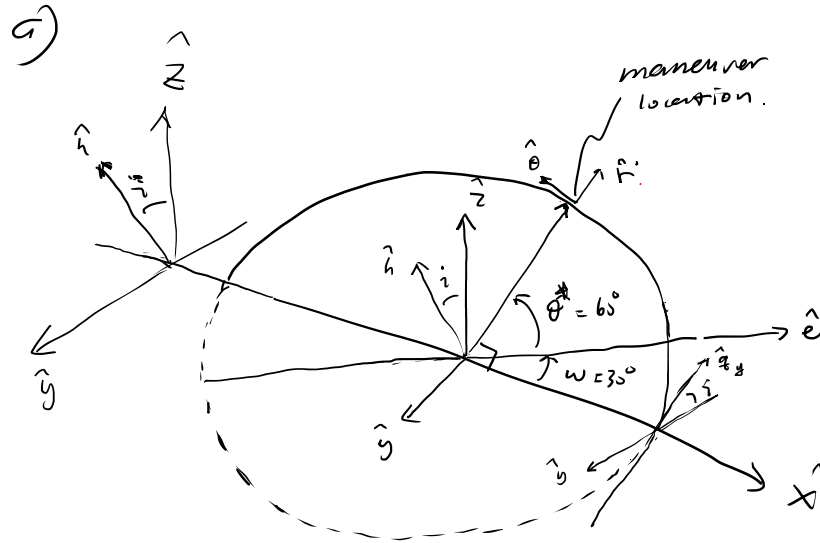
$$\begin{aligned} \theta &= \sin^{-1}\left(\frac{\Delta V_{eq}}{V^+} \sin(\rho + \phi)\right) \\ &= 17.7359^\circ \end{aligned}$$

$$\begin{aligned} \Delta \vec{V}_{eq} &= 7.9184 \cdot (\sin \alpha \hat{V} + \cos \alpha \hat{B}) \\ &= 6.3222 \hat{V} + 4.7675 \hat{B} \text{ km/s} \end{aligned}$$

$$\begin{aligned} |\gamma^+| &= |\gamma^-| - \theta = 32.434^\circ \\ &= \boxed{\gamma^+ = -32.434^\circ} \end{aligned}$$

2. $a = 5 R_{\oplus} = 32000 \text{ km}$
 $e = 0.4 \quad \Omega = 0^\circ \text{ (RAAN)}$
 $i = 30^\circ \quad \omega = 30^\circ \text{ (AOP)}$

Maneuver at $\theta^* = 60^\circ$



\hat{h} is in \hat{y} - \hat{z} plane.

because $\Omega = 0^\circ$, \hat{h} is only rotated by $i = 30^\circ$ about \hat{x} -axis.

b) $\Delta \bar{V} = -\frac{1}{2} \hat{y} \text{ km/s} \quad \theta = 60^\circ + 30^\circ = 90^\circ$

$$C_{313} = \begin{bmatrix} \cancel{C_{\Omega} C_{\theta} - S_{\Omega} C_i S_{\theta}} & -C_{\Omega} S_{\theta} - \cancel{S_{\Omega} C_i C_{\theta}} & \cancel{S_{\Omega} S_i} \\ \cancel{S_{\Omega} C_{\theta} + C_{\Omega} C_i S_{\theta}} & -\cancel{S_{\Omega} S_{\theta} + C_{\Omega} C_i C_{\theta}} & -C_{\Omega} S_i \\ S_i S_{\theta} & S_i C_{\theta} & C_i \end{bmatrix} = \begin{bmatrix} 0 & -1 & 0 \\ C_i & 0 & -S_i \\ S_i & 0 & C_i \end{bmatrix} = \begin{bmatrix} 0 & -1 & 0 \\ 0.866 & 0 & -0.5 \\ 0.5 & 0 & 0.866 \end{bmatrix}$$

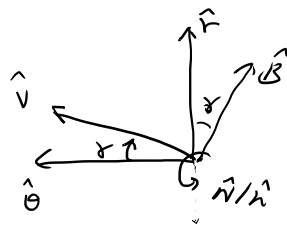
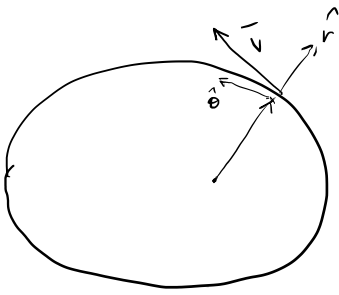
$$\Delta \bar{V}_{xyz} = C_{313} \cdot \Delta \bar{V}_{r\theta h}$$

$$\Delta \bar{V}_{r\theta h} = \begin{bmatrix} 0 & 0.866 & 0.5 \\ -1 & 0 & 0 \\ 0 & -0.5 & 0.866 \end{bmatrix} \cdot \begin{bmatrix} 0 \\ -0.5 \\ 0 \end{bmatrix} = \begin{bmatrix} -0.433 \\ 0 \\ 0.25 \end{bmatrix} = -0.433 \hat{r} + 0.25 \hat{h} \text{ km/s}$$

$$p = a(1-e^2) = 26880 \text{ km.} \quad r = \frac{p}{1+e \cos \theta^*} = \frac{26880}{1+0.4 \cos(60)} = 22400 \text{ km}$$

$$V = \sqrt{\frac{2\mu_E}{r} - \frac{\mu}{a}} = \sqrt{\frac{2 \times 4 \times 10^5}{22400} - \frac{4 \times 10^5}{32000}} = 4.8181 \text{ km/s}$$

$$\gamma = \cos^{-1} \left(\frac{\sqrt{\mu_E p}}{r v} \right) = \cos^{-1} \left(\frac{\sqrt{4 \times 10^5 \times 26880}}{22400 \times 4.8181} \right) = 16.1013^\circ \text{ (ascend).}$$

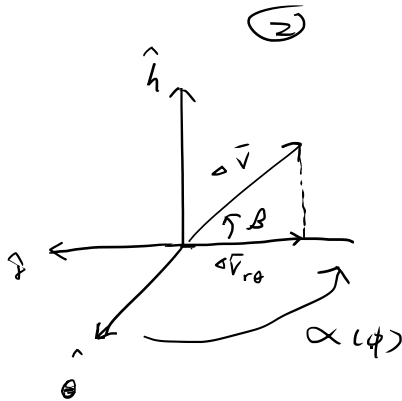
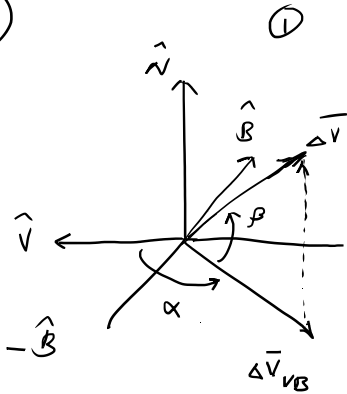


$$\begin{aligned}\hat{r} &= \sin(\gamma)\hat{V} + \cos(\gamma)\hat{B} \\ &= 0.2773\hat{V} + 0.9608\hat{B} \\ \hat{h} &= \hat{N}\end{aligned}$$

$$\Delta \bar{V}_{VNB} = -0.433 \times 0.2773\hat{V} - 0.433 \times 0.9608\hat{B} + 0.25\hat{N}$$

$$= \boxed{-0.1201\hat{V} + 0.25\hat{N} - 0.4160\hat{B} \text{ km/s}}$$

c) d)



$$\textcircled{1}. \alpha = -90^\circ - \gamma = \boxed{-106.1013^\circ}$$

$$\beta = \tan^{-1}\left(\frac{0.25}{0.433}\right) = \boxed{30^\circ}$$

$$|\Delta \bar{V}_{NB}| = \sqrt{0.1201^2 + 0.4160^2} = 0.433 \text{ km/s}$$

$$\textcircled{2}. \alpha = \boxed{-90^\circ} \quad \beta = \boxed{30^\circ}$$

$$|\Delta \bar{V}_{NB}| = 0.433 \text{ km/s}$$

$$|\Delta \bar{V}_N| = 0.25 \text{ km/s}$$

Since VNB is rotn rotated by γ about \hat{h} , there are now two ways of representing $\Delta \bar{V}$.

On the right, (1) shows that α is the angle between $+\hat{V}$ and $\Delta \bar{V}_{NB}$, the in-plane component of $\Delta \bar{V}$. β is the angle between the in-plane component $\Delta \bar{V}_{NB}$ & actual $\Delta \bar{V}$ vector.

On the left, (2) shows α is the angle between $+\hat{\theta}$ & $\Delta \bar{V}_{\theta}$, the in-plane component. β is the angle between the in-plane component $\Delta \bar{V}_{\theta}$ and $\Delta \bar{V}$.

3. a). $c = \sqrt{(3R_{\oplus})^2 + (4R_{\oplus})^2} = 5R_{\oplus} = 32000 \text{ km}$

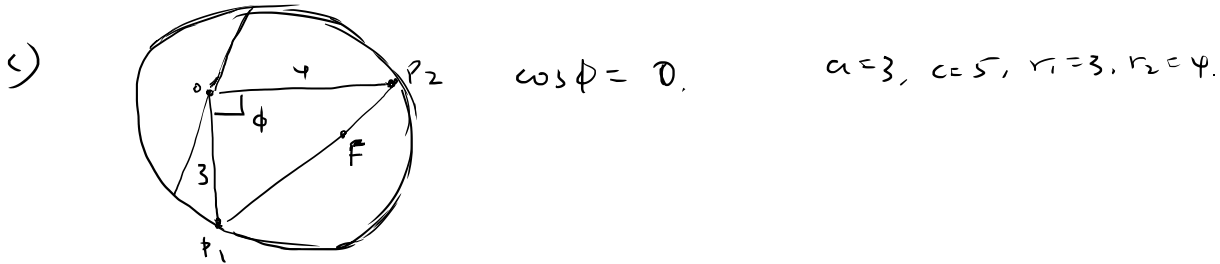
b). $r_1 = 3R_{\oplus} = 19200 \text{ km}$

$r_2 = 4R_{\oplus} = 25600 \text{ km}$

$s = 6R_{\oplus}$

$a_{\min} = \frac{1}{4}(r_1 + r_2 + c) = 3R_{\oplus} = 19200 \text{ km}$

$\zeta_p = -\frac{\mu_E}{2a_m} = -10.4167 \text{ km}^2/\text{s}^2$



$$a^2 p^2 + r_1 r_2 (1-0) [-2a(r_1+r_2) + r_1 r_2 (1+0)] p + a r_1^2 r_2^2 (-1)^2 = 0$$

$$(3 \times 25) R_{\oplus}^3 p^2 + 12 R_{\oplus}^2 [(2 \times 3 \times 7) R_{\oplus}^2 + 12 R_{\oplus}^2] p + 3 \times 9 \times 16 R_{\oplus}^5 = 0.$$

$$75 p^2 - 360 R_{\oplus} p + 432 R_{\oplus}^2 = 0.$$

$$25 p^2 - 120 R_{\oplus} p + 144 R_{\oplus}^2 = 0.$$

$$(5p - 12 R_{\oplus})^2 = 0.$$

$$5p = 12 R_{\oplus}$$

$$p = 2.4 R_{\oplus} \text{ (Proven).}$$

$$e = \sqrt{1 - \frac{p}{a}} = \sqrt{1 - \frac{2.4}{3}} = \sqrt{0.2} = 0.4472$$

$$b = a \cdot \sqrt{1 - e^2} = 3 R_{\oplus} \cdot \sqrt{1 - 0.2} = 2.6833 R_{\oplus} = 17173. \text{ km}$$

$$d). r_p = a(1-e) = \boxed{10613.76 \text{ km}}$$

$$r_a = a(1+e) = \boxed{27786.24 \text{ km}}$$

$$r_{dep} = r_1 = \boxed{19200 \text{ km}}$$

$$v_{dep} = \sqrt{2 \frac{\mu_E}{r_{dep}} - \frac{\mu_E}{a}} = \boxed{4.5644 \text{ km/s}}$$

$$\gamma_{dep} = \cos^{-1} \left(\frac{\sqrt{\mu p}}{r_{dep} v_{dep}} \right) = \boxed{26.5662^\circ} \text{ ascending.}$$

$$E_{dep} = \boxed{90^\circ} \quad (r=a)$$

$$\theta_{dep}^* = \cos^{-1} \left(\frac{p}{r_{dep} e} - \frac{1}{e} \right) = \boxed{116.5659^\circ} \text{ ascending.}$$

$$\theta_{arr}^* = \cos^{-1} \left(\frac{p}{r_{arr} e} - \frac{1}{e} \right) = \boxed{-153.4384^\circ} \text{ descending}$$

e)

