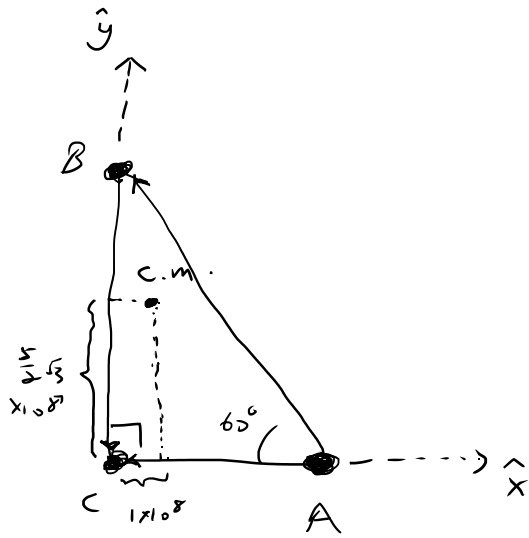


1. a) It is to infer uniformly distributed gravity field around each body, thus allowing us to use the gravity equations.



$$\bar{r}_{AC} = -4 \times 10^8 \hat{x}$$

$$16 + 48 = 64$$

$$\bar{r}_{BC} = 4\sqrt{3} \times 10^8 \hat{y}$$

$$\bar{r}_{AB} = \bar{r}_{AC} - \bar{r}_{BC} = -4 \times 10^8 \hat{x} + 4\sqrt{3} \times 10^8 \hat{y}$$

$$\text{Let } \bar{r}_A = -\bar{r}_{AC}, \bar{r}_B = -\bar{r}_{BC}, \bar{r}_C = 0.$$

(C coincides the origin at this moment)

$$M \bar{r}_{cm} = m_A \bar{r}_A + m_B \bar{r}_B + m_C \bar{r}_C$$

$$\bar{r}_{cm} = \frac{2 \times 4 \times 10^{16} \hat{x} + 5 \times 4\sqrt{3} \times 10^{16} \hat{y}}{8 \times 10^8}$$

$$\bar{r}_{cm} = 1 \times 10^8 \hat{x} + \frac{5}{2} \sqrt{3} \times 10^8 \hat{y} \text{ km}$$

- b) i) motion of C relative to A perturbed by B.

$$\ddot{\bar{r}}_{AC} + \frac{G(m_A + m_C)}{r_{AC}^3} \bar{r}_{AC} = G m_B \left(\frac{\bar{r}_{CB}}{r_{CB}^3} - \frac{\bar{r}_{AB}}{r_{AB}^3} \right)$$

Both formulations are correct, because they are based on different reference points, which will give us different information on the interested body.

$$\text{ii) } m_C \ddot{\bar{r}}_C = -G \frac{m_A m_C}{r_{AC}^3} \bar{r}_{AC} - G \frac{m_B m_C}{r_{BC}^3} \bar{r}_{BC}$$

iii) Independent variable: t , (time)

Dependent variables: $\bar{r}_C, \dot{\bar{r}}_C, \bar{r}_A, \bar{r}_B$ (each with \hat{x}, \hat{y} components, assuming planar motion)

c) Dominant Acceleration;

$$i) \frac{-G(M_A + m_c)}{r_{AC}^3} \vec{r}_{AC} = - \frac{3 \times 10^8}{(4 \times 10^8)^3} \times (-4 \times 10^8 \hat{x}) = \boxed{1.875 \times 10^{-9} \hat{x} \text{ km/s}^2}$$

mag: $1.875 \times 10^{-9} \text{ km/s}^2$

Direct Perturbation:

$$G m_B \frac{\vec{r}_{CB}}{r_{CB}^3} = 5 \times 10^8 \times \frac{4\sqrt{3} \times 10^8 \hat{y}}{(4\sqrt{3} \times 10^8)^3} = \boxed{1.0417 \times 10^{-9} \hat{y} \text{ km/s}^2}$$

mag: $1.0417 \times 10^{-9} \text{ km/s}^2$

Indirect Perturbation:

$$G m_B \frac{\vec{r}_{AB}}{r_{AB}^3} = 5 \times 10^8 \times \frac{(-4 \times 10^8 \hat{x} + 4\sqrt{3} \times 10^8 \hat{y})}{(8 \times 10^8)^3}$$

$$= \boxed{-3.90625 \times 10^{-10} \hat{x} + 6.7658 \times 10^{-10} \hat{y} \text{ km/s}^2}$$

mag: $7.8125 \times 10^{-10} \text{ km/s}^2$

Net Perturbation:

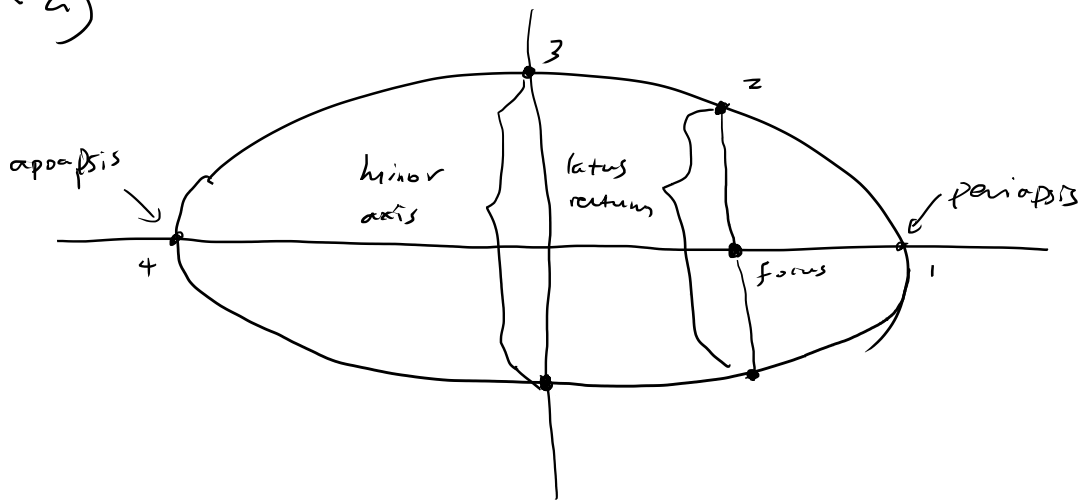
$$G m_B \frac{\vec{r}_{CB}}{r_{CB}^3} - G m_B \frac{\vec{r}_{AB}}{r_{AB}^3} = \boxed{3.90625 \times 10^{-10} \hat{x} + 3.6512 \times 10^{-10} \hat{y} \text{ km/s}^2}$$

mag: $5.3470 \times 10^{-10} \text{ km/s}^2$

(ii) Dominant term has the largest magnitude, followed by direct perturbation and then indirect perturbation. The net perturbation has the smallest magnitude.

(iii) The dominant term is positive in \hat{x} , and the net perturbing term is also positive in both \hat{x} & \hat{y} . The net acceleration in \hat{x} is greater than that in \hat{y} . Since A is on the \hat{x} axis at this instance, the distance between A & C will tend to decrease.

2. a)



b) . periapsis ; $\theta^* = 0^\circ$

$$r_1 = \frac{p}{1+e} = \boxed{a(1-e)}$$

Apoapsis : $\theta^* = 180^\circ$

$$r_4 = \frac{p}{1-e} = \boxed{a(1+e)}$$

Latus rectum.

$$r_2 = p = \boxed{a(1-e^2)}$$

Minor axis

$$r_3 = a(1-e\cos 90^\circ) = \boxed{a}$$

$$c) \quad \frac{1}{r} = \frac{J^2}{2} - \frac{\mu}{r}$$

$$V^2 = 2\tilde{\epsilon} + 2\frac{r}{\mu}$$

$$V^2 = 2\tilde{\epsilon} + \frac{2r}{-2a\tilde{\epsilon}}$$

$$V^2 = 2\tilde{\epsilon} - \frac{r}{a\tilde{\epsilon}}$$

Periapsis:

$$V_1 = \sqrt{2\tilde{\epsilon} - \frac{r_1}{a\tilde{\epsilon}}}$$

$$= \sqrt{2\tilde{\epsilon} - \frac{a(1-e)}{a\tilde{\epsilon}}}$$

$$\boxed{V_1 = \sqrt{2\tilde{\epsilon} - \frac{1-e}{\tilde{\epsilon}}}}$$

$$\tilde{\epsilon} = -\frac{\mu}{2a}$$

$$\mu = -2a\tilde{\epsilon}$$

$$V^2 = 2V_c^2 - \frac{\mu}{a} = 2V_c^2 + 2\tilde{\epsilon}$$

apapsis

$$V_4 = \sqrt{2\tilde{\epsilon} - \frac{r_4}{a\tilde{\epsilon}}}$$

$$= \sqrt{2\tilde{\epsilon} - \frac{a(1+e)}{a\tilde{\epsilon}}}$$

$$\boxed{V_4 = \sqrt{2\tilde{\epsilon} - \frac{1+e}{\tilde{\epsilon}}}}$$

latus rectum:

$$V_2 = \sqrt{2\tilde{\epsilon} - \frac{r_2}{a\tilde{\epsilon}}} = \boxed{\sqrt{2\tilde{\epsilon} - \frac{1-e^2}{\tilde{\epsilon}}}}$$

minor axis:

$$V_3 = \sqrt{2\tilde{\epsilon} - \frac{r_3}{a\tilde{\epsilon}}} = \boxed{\sqrt{2\tilde{\epsilon} - \frac{1}{\tilde{\epsilon}}}}$$

$$1) \begin{cases} r_1 = 0.5 a \\ r_4 = 1.5 a \\ r_2 = 0.75 a \\ r_3 = a \end{cases}$$

$$\begin{aligned} v_1 &= \sqrt{2\epsilon - \frac{1}{2\epsilon}} & v_3 &= \sqrt{2\epsilon - \frac{1}{\epsilon}} \\ v_4 &= \sqrt{2\epsilon - \frac{3}{2\epsilon}} \\ v_2 &= \sqrt{2\epsilon - \frac{3}{4\epsilon}} \end{aligned}$$

2)

3. $m = 8000 \text{ kg}$. $\vec{C}_1 = \vec{0}$, $|\vec{h}| = 2.5 \times 10^4 \text{ km}^2/\text{s}$, $|\vec{e}| = 1.2351 \text{ km}^2/\text{s}$
 $R_\odot = 3400 \text{ km}$ $\mu_\odot = 4.2 \times 10^4 \text{ km}^3/\text{s}^2$

a) $|\vec{C}_1| = 0$ tells us that the center of the mass of the system is not moving in this reference frame, thus we only need to consider rotational motion.

b) $\epsilon = -\frac{\mu}{2a}$
 $1.2351 = \frac{4.2 \times 10^4}{2a}$
 $p = \frac{h^2}{\mu} = \frac{(2.5 \times 10^4)^2}{(4.2 \times 10^4)} = \boxed{1.4881 \times 10^4 \text{ km}} = 4.3768 R_\odot$

$a = 1.7002 \times 10^4 \text{ km} = 5.0006 R_\odot$

$a = \frac{p}{1-e^2}$

$b = a\sqrt{1-e^2}$
 $= 1.7002 \times 10^4 \sqrt{1-0.3532^2}$

$e = \sqrt{1 - \frac{p}{a}} = \sqrt{1 - \frac{1.4881}{1.7002}} = \boxed{0.3532}$
 e

$b = 1.5906 \times 10^4 \text{ km}$
 $= 4.6782 R_\odot$

$r_p = a(1-e) = \boxed{1.0997 \times 10^4 \text{ km}} = 3.2344 R_\odot$

$r_a = a(1+e) = \boxed{2.3007 \times 10^4 \text{ km}} = 6.7668 R_\odot$

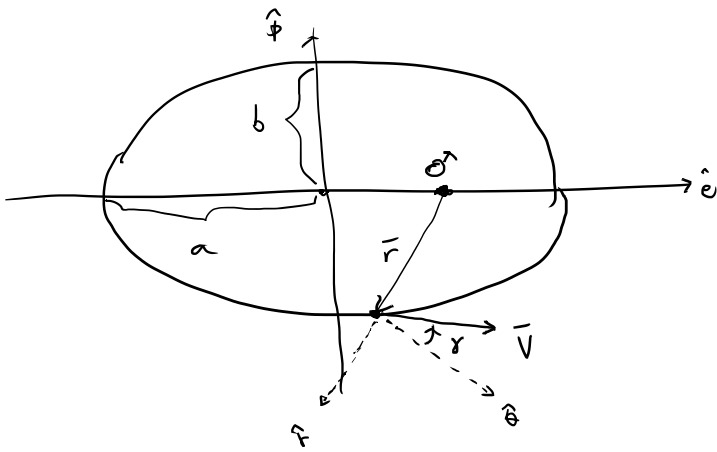
$P = 2\pi \sqrt{\frac{a^3}{\mu}}$
 $= 2\pi \sqrt{\frac{(1.7002 \times 10^4)^3}{4.2 \times 10^4}}$
 $= \boxed{6.7968 \times 10^4 \text{ s}}$

c) $r = \frac{p}{1+e \cos(\theta^*)} = \frac{1.4881 \times 10^4}{1+0.3532 \cdot \cos(110^\circ)} = \boxed{1.6926 \times 10^4 \text{ km}}$

$v = \sqrt{\frac{2\mu}{r} - \left(\frac{\mu}{a}\right)} \rightarrow \frac{h}{2}$

$= \sqrt{\frac{2 \times 4.2 \times 10^4}{1.6926 \times 10^4} - \frac{1.2351}{2}}$

$= 2.0845 \text{ km/s}$



d). If $e > 0$, it is a hyperbolic orbit.

$$a = -\frac{\mu}{2\xi} = -1.7002 \times 10^4 \text{ km}$$

$$p = 1.4881 \times 10^4 \text{ km}$$

$$e = \sqrt{1 + \frac{2\xi h^2}{\mu^2}} = \sqrt{1 + \frac{2 \times 1.2351 \times (25 \times 10^4)^2}{(4.2 \times 10^4)^2}} = 1.3694$$

$$b = |a| \sqrt{e^2 - 1} = 1.5906 \times 10^4 \text{ km}$$

$$r_p = |a| \sqrt{e - 1} = 1.0334 \times 10^4 \text{ km}$$

$$r_a = \infty$$

$$p = \infty$$

p , a & b remained the same for both orbit.

$$r = 1.6926 \times 10^4 \text{ km}$$

$$v = \sqrt{\frac{2 \times 4.2 \times 10^4}{1.6926 \times 10^4} + \frac{1.2351}{2}} = 2.3623 \text{ km/s}$$

The velocity is way faster.

$$\sqrt{\frac{2 \times 4.2 \times 10^4}{1.6926 \times 10^4} + \frac{1.2351}{2}}$$

e) For the elliptic orbit:

For the hyperbolic orbit: $\theta_\infty = \sin^{-1} \frac{1}{e} = 46.91^\circ$ This is the angle of the asymptotes.

