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<github.com/ASU-CompMethodsPhysics-PHY494/final-atlas-mis</pre>

Saturn is **10 times** farther from the Sun than Earth

It orbits once about every **10,000** days

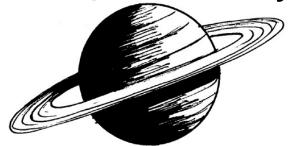


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Titan orbits Saturn

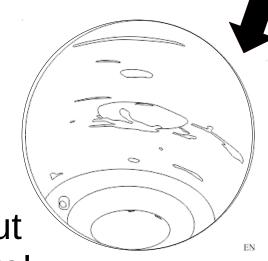
It has **lakes of methane** and a dense atmosphere

Astrobiologists hypothesize that **life** could form here



ATLAS is a robotic scientific explorer made for Titan

It's up to us to figure out how to get **ATLAS** there!



The Mission

Titan represents an incredible opportunity to explore a potentially habitable world outside our own. ATLAS will gather evidence that could answer the question of whether or not we're alone in the universe.

The Problem

Getting a spacecraft all the way out to Saturn is no easy feat. Launch vehicles burn vast amounts of fuel, and even then it can be close to impossible to attain the velocities necessary to reach the outer solar system. If we could utilize gravitational assists from planets to accelerate our spacecraft, that would reduce the need for powered propulsion.

The Solution

We set out to create a dynamic simulation of the solar system based on gravitational interactions between bodies. This required calculating, iteratively,

$$F = \frac{GMm}{r^3} \vec{r}$$

for all masses and their relative distance from each other and plotting the system as it evolves over time.

With that done, we were able to then model ATLAS. The spacecraft is treated like any other body in the simulation, but with additional functions to control aspects of its launch, as well as retain data of its motion throughout the trajectory.

Epilogue

mission would have liked to place ATLAS in objectives, but with more time, the ATLAS Through careful analysis of many ATLAS trajectory simulations, we were able to project initial conditions that lead to a We didn't fulfill all of our hoped for a stable orbit at a Lagrange point. successful intercept with Titan.

Exposition

calculations in the velocity Verlet, making modeling the movement of all bodies in There were a total of 34 simultaneous Implementation of the velocity Verlet algorithm was the key component in equations being solved for the force the solar system (including ATLAS). optimization of the Verlet function necessary.

```
if i == int(launchday/dt):
    r[i+1, 6] = r[i+1, 1] + .01*r[i+1, 1]
    v[i+1, 6] = r[i+1, 1] + velofrac*v[i+1, 1]
    v[i+1, 6] = v[i+1, 1] + velofrac*v[i+1, 1]
    kt 6 = (F gravity(r[i+1, 6]-r[i+1, 0], mass['Atlas'], mass['Yenus']) +
    F gravity(r[i+1, 6]-r[i+1, 1], mass['Atlas'], mass['Mars']) +
    F gravity(r[i+1, 6]-r[i+1, 2], mass['Atlas'], mass['Jupiter']) +
    F gravity(r[i+1, 6]-r[i+1, 4], mass['Atlas'], mass['Saturn']) +
    F gravity(r[i+1, 6]-r[i+1, 5], mass['Atlas'], mass['Titan']))
                                                                                                                                                                                                                                                                                                                         mass['Jupiter'])
mass['Saturn'])
mass['Titan']))
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  #v[i+1, 6] = [-0.4*v[i+1,6][0],v[i+1, 5][1]]
v[i+1, 6] = [0.80866*v[i+1,6][0],2.21166*v[i+1,6][1]]
#if i == int(2535/dt):
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                      6] = [0.2*v[i+1,6][0],0.2*v[i+1,6][1]]
                                                                                                                                                                                                                                                                                                                                                                                                                                        print('launch', v[i, 6])
c int(launchday/dt):
r[i+1, 6] = r[i+1, 1]
v[i+1, 6] = v[i+1, 1]
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                              == int(t_max/dt):
                                                                                                              0
                                                                                                        Ft_next_6[:] =
                                                                           Ft_6[:] =
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