ASTR 119: Session 16

Orbital Mechanics Random Numbers



Outline

- 1) Visualization of the Day
- 2) Final projects
- 3) Orbital Mechanics
- 4) Final project organization time
- 5) Save your work to GitHub

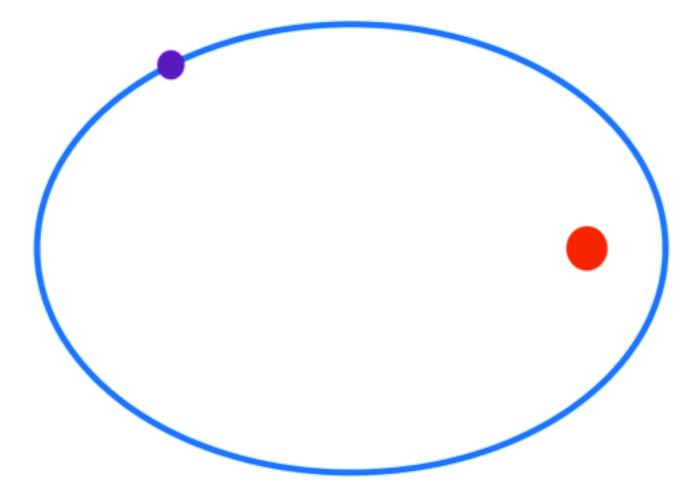


Final Projects

- 1) You will be asked to perform one of four final projects, working in a group of up to 4 people of your choosing from your SECTION. If you would like to be assigned to a group, please let me know immediately.
- 2) Please notify me of your group by **8:00am Thursday Nov 19**. Anyone who does not respond by **8:00am Thursday** letting me know their group or their desire to be assigned to a group will work on their own final project.
- 3) You will choose from one of the following 4 project topics, described in the following slides:
 - 1) Damped, driven pendulum
 - 2) Logistic map and chaos
 - 3) Astronomical source detection
 - 4) Monte Carlo Integration
- 4) Groups must select a final project topic by **Tues Nov 24**. Detailed instructions for each project will be distributed Thursday Nov 19.
- 5) Groups should be organized through GitHub, invite your group members and us (TAs) as collaborators on your project. Tag us (TAs) when your final project is ready to grade.
- 6) Each python module should indicate which student authored which part of the code.
- 7) Final projects are due (tags must happen by) Tuesday, December 15, 2020 at 3pm.

ORBITAL MECHANICS DEFINITIONS:

Consider the solar system, where the Sun effectively sits at the center of the system and the planets orbit about the Sun.



We are going to define a few useful quantities...

We have chosen an integration method, but now we need to consider all the overhead involved in calculating the orbits. We'll want a file to save the information about the solar system over time, and we'll want to write to it in a convenient format. We'll need to evolve the global system many timesteps, and evolve each planet over its own time step potentially many times per global timestep. So, we'll need:

- 1) A driving routine to perform the bookkeeping of the orbital integration.
- 2) A routine to open the file we'll write the data out to.
- 3) A routine to write the data to file each global timestep.
- 4) To update the position and velocity according to the leapfrog integrator.
- 5) Update the planetary time steps according to their velocity and acceleration.

We'd like to save the results of the simulation and use that data to visualize the solar system. We want our data files to be efficient and easy to use (these are sometimes at cross purposes). Also, we want the position and velocities to be written at the same time (but they are offset by 1/2 time step in our integrator). We need to:

- 1) Open a data file.
- 2) Write to the data file in a useful format.
- 3) Synchronize the position and velocity data when writing to the file.

Write a function to save the data to file

```
def SaveSolarSystem(p, n planets, t, dt, istep, ndim):
]:
    3
           #loop over the number of planets
     4
           for i in range(n planets):
     5
                #define a filename
    6
                fname = "planet.%s.txt" % p[i].name
    7
    8
    9
                if(istep==0):
                    #create the file on the first timestep
   10
   11
                    fp = open(fname, "w")
   12
                else:
   13
                    #append the file on subsequent timesteps
                    fp = open(fname, "a")
   14
   15
   16
                #compute the drifted properties of the planet
   17
                v drift = np.zeros(ndim)
   18
   19
                for k in range(ndim):
   20
                    v_{drift[k]} = p[i].v[k] + 0.5*p[i].a_g[k]*p[i].dt
   21
   22
                #write the data to file
   23
                s = "%6d\t%6.5f\t%6.5f\t%6d\t%6.5f\t%6.5f\t% 6.5f\t% 6.5f\t% 6.5f\t% 6.5f\t% 6.5f\t% 6.5f\t% 6.5f\n" % \
   24
                    (istep,t,dt,p[i].istep,p[i].t,p[i].dt,p[i].x[0],p[i].x[1],v drift[0],v drift[1],
   25
                     p[i].a_g[0],p[i].a_g[1])
                fp.write(s)
   26
   27
   28
                #close the file
   29
   30
                fp.close()
```

Write a function to evolve the solar system

```
def EvolveSolarSystem(p,n planets,t max):
]:
           #number of spatial dimensions
            ndim = 2
           #define the first timestep
            dt = 0.5/365.25
           #define the starting time
   10
           t = 0.0
   11
           #define the starting timestep
   12
           istep = 0
   13
   14
   15
            #save the initial conditions
   16
            SaveSolarSystem(p,n planets,t,dt,istep,ndim)
   17
            #begin a loop over the global timescale
   18
           while(t<t_max):</pre>
   19
    20
    21
                #check to see if the next step exceeds the
    22
                #maximum time. If so, take a smaller step
    23
                if(t+dt>t max):
    24
                    dt = t max - t # limit the step to align with t max
    25
    26
                #evolve each planet
                for i in range(n planets):
    27
```

```
26
            #evolve each planet
27
            for i in range(n planets):
28
                                                                          Cell cont
29
                while(p[i].t<t+dt):</pre>
30
31
                    #special case for istep==0
32
                    if(p[i].istep==0):
33
                        #take the first step according to a verlet scheme
34
                        for k in range(ndim):
35
                             p[i].x[k] = x_first_step(p[i].x[k],p[i].v[k],p[i].a_g[k],p[i].dt)
36
37
38
                        #update the acceleration
                        p[i].a_g = SolarGravitationalAcceleration(p[i])
39
40
                        #update the time by 1/2dt
41
42
                        p[i].t += 0.5*p[i].dt
43
                        #update the timestep
44
45
                        p[i].dt = calc dt(p[i])
46
                    #continue with a normal step
47
48
                    #limit to align with the global timestep
49
50
                    if(p[i].t + p[i].dt > t+dt):
51
                        p[i].dt = t+dt-p[i].t
```

```
49
                    #limit to align with the global timestep
50
                    if(p[i].t + p[i].dt > t+dt):
                        p[i].dt = t+dt-p[i].t
51
52
                                                                        Cell cont
                    #evolve the velocity
53
54
                    for k in range(ndim):
                        p[i].v[k] = v_full_step(p[i].v[k],p[i].a_g[k],p[i].dt)
55
56
                    #evolve the position
57
58
                    for k in range(ndim):
59
                        p[i].x[k] = x_{full_step(p[i].x[k],p[i].v[k],p[i].a_g[k],p[i].dt)
60
                    #update the acceleration
61
                    p[i].a g = SolarGravitationalAcceleration(p[i])
62
63
                    #update by dt
64
                    p[i].t += p[i].dt
65
66
                    #compute the new timestep
67
                    p[i].dt = calc dt(p[i])
68
69
70
                    #update the planet's timestep
                    p[i].istep+=1
71
72
73
                #now update the global system time
74
                t+=dt
```

```
73
                #now update the global system time
74
                t+=dt
                                                          Cell cont
75
76
                #update the global step number
                istep += 1
77
78
79
                #output the current state
80
                SaveSolarSystem(p,n planets,t,dt,istep,ndim)
81
       #print the final steps and time
82
       print("Time t = ",t)
83
       print("Maximum t = ",t max)
84
       print("Maximum number of steps = ",istep)
85
86
       #end of evolution
87
```

Create a routine to read in the data

```
def read twelve arrays(fname):
    fp = open(fname, "r")
    fl = fp.readlines()
    n = len(fl)
    a = np.zeros(n)
    b = np.zeros(n)
    c = np.zeros(n)
    d = np.zeros(n)
    f = np.zeros(n)
    g = np.zeros(n)
    h = np.zeros(n)
    j = np.zeros(n)
    k = np.zeros(n)
    1 = np.zeros(n)
    m = np.zeros(n)
    p = np.zeros(n)
    for i in range(n):
        a[i] = float(fl[i].split()[0])
        b[i] = float(fl[i].split()[1])
        c[i] = float(fl[i].split()[2])
        d[i] = float(fl[i].split()[3])
        f[i] = float(fl[i].split()[4])
        g[i] = float(fl[i].split()[5])
        h[i] = float(fl[i].split()[6])
        j[i] = float(fl[i].split()[7])
        k[i] = float(fl[i].split()[8])
        l[i] = float(fl[i].split()[9])
        m[i] = float(fl[i].split()[10])
        p[i] = float(fl[i].split()[11])
    return a,b,c,d,f,g,h,j,k,l,m,p
```

Perform the integration of the solar system

```
451:
         #set the number of planets
         n planets = 3
         #set the maximum time of the simulation
         t max = 2.0
         #create empty list of planets
         p = []
      9
     10
         #set the planets
     11
         for i in range(n planets):
     12
     13
             #create an empty planet
     14
             ptmp = planet(0.0,0.0)
     15
             #set the planet properties
     16
     17
             SetPlanet(ptmp,i)
     18
     19
             #remember the planet
     20
             p.append(ptmp)
     21
         #evolve the solar system
         EvolveSolarSystem(p,n planets,t max)
     Time t = 2.000000000000001
     Maximum t = 2.0
```

Maximum number of steps = 1464

Read the data back in for every planet

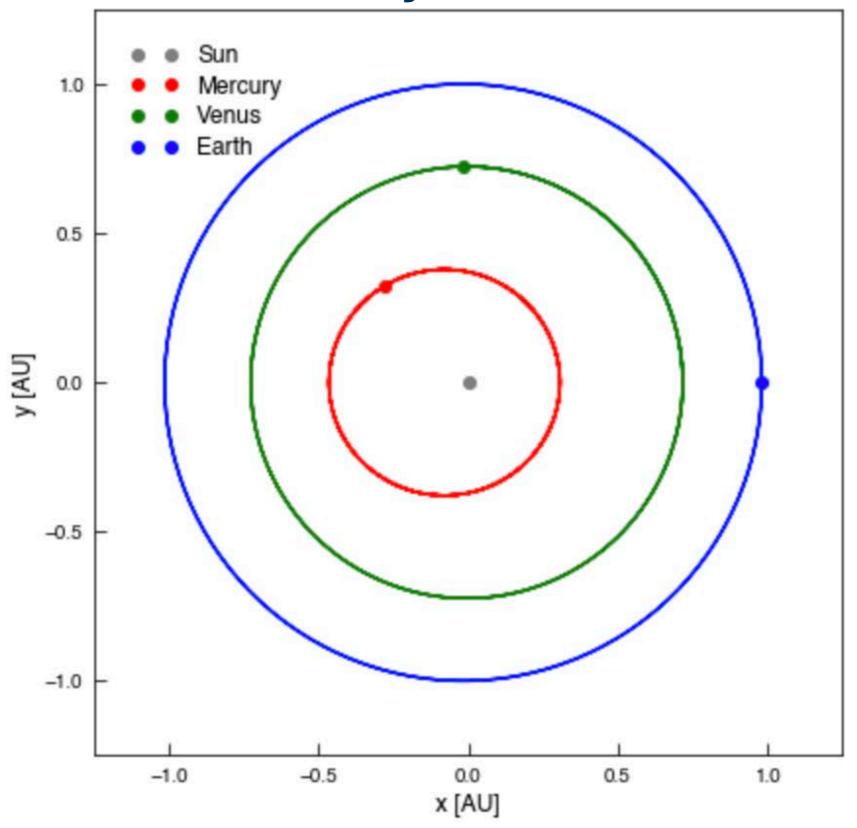
```
if name = "planet.Mercury.txt"
  istepMg,tMg,dtMg,istepM,tM,dtM,xM,yM,vxM,vyM,axM,ayM = read_twelve_arrays(fname)

if name = "planet.Earth.txt"
  istepEg,tEg,dtEg,istepE,tE,dtE,xE,yE,vxE,vyE,axE,ayE = read_twelve_arrays(fname)

if name = "planet.Venus.txt"
  istepVg,tVg,dtVg,istepV,tV,dtV,xV,yV,vx,vyV,axV,ayV = read_twelve_arrays(fname)
```

Plot the data

```
fig = plt.figure(figsize=(7,7))
   xSun = [0.0]
   ySun = [0.0]
   plt.plot(xSun,ySun,'o',color="0.5",label="Sun")
   plt.plot(xM,yM,color="red")
   plt.plot(xM[-1],yM[-1],'o',color="red",label="Mercury")
 9
10
   plt.plot(xV,yV,color="green")
   plt.plot(xV[-1],yV[-1],'o',color="green",label="Venus")
11
12
13
   plt.plot(xE,yE,color="blue")
   plt.plot(xE[-1],yE[-1],'o',color="blue",label="Earth")
14
15
16
17
18
   plt.xlim([-1.25,1.25])
   plt.ylim([-1.25,1.25])
19
20 plt.xlabel('x [AU]')
21 plt.ylabel('y [AU]')
22 plt.axes().set aspect('equal')
   plt.legend(frameon=False,loc=2)
```



In science, we often have to "fake it till we make it", meaning that in order to understand our experiments and their systematic uncertainties we need to model our data. Given both natural randomness and measurement error, these models amount to random number generation.

Often, we would like to generate random numbers distributed from a wide variety of distributions. If we can calculate the integral of the probability density distribution we wish to pull random numbers from, it is possible to use uniformly distributed random numbers to generate random numbers from a general distribution.

Given the complexity of this problem, let's take some time to learn about publicly available libraries to help us with a variety of computational tasks including random number generation. This will be useful for many problems you might encounter in the future.

Use a random number generator to simulate a simple Gaussian process

```
In [2]: %matplotlib inline
import matplotlib.pyplot as plt
import numpy as np
```

Set some properties of the random system

```
In [5]: n_samples = 10000 #number of random samples to use
n_bins = 100 #number of bins for our histogram
sigma = 1.0 #rms width of the gaussian
mu = 0.0 #mean of the gaussian
```

Generate the random numbers using numpy

```
In [15]: x = np.random.normal(mu,sigma,n_samples)
    print(x.min())
    print(x.max())

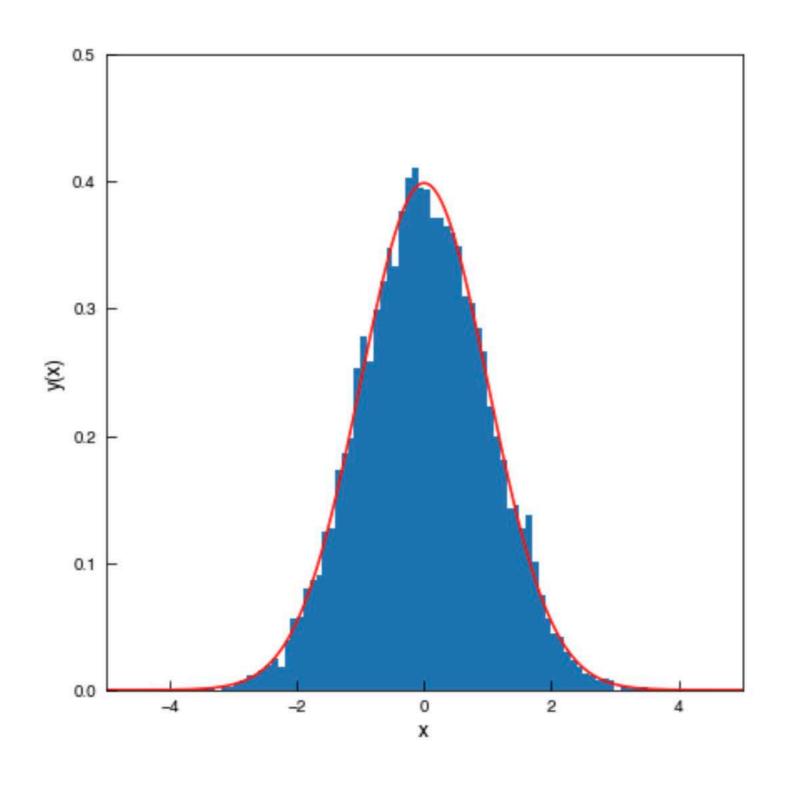
-3.76402898364
3.70499724783
```

Define a function to plot a Gaussian

```
In [19]: def gaussian(x,mu,sigma):
    return 1./(2.0*np.pi*sigma**2)**0.5 * np.exp(-0.5*((x-mu)/sigma)**2)
```

Create a histogram of the data

```
In [24]: fig = plt.figure(figsize=(7,7))
    y_hist, x_hist, ignored = plt.hist(x, bins=n_bins, range=[-5,5], density=True)
    xx = np.linspace(-5.0,5.0,1000)
    plt.plot(xx,gaussian(xx,mu,sigma),color="red")
    plt.ylim([0,0.5])
    plt.xlim([-5,5])
    plt.gca().set_aspect(20)
    plt.xlabel('x')
    plt.ylabel('y(x)')
    plt.show()
```



One very useful application of random number generation is to perform Monte Carlo integration. Monte Carlo integration uses random numbers to sample a space and compute the integral of a given function by finding the ratio of the region under the curve to the total area probed by the samples.

I wanted to provide a simple example of Monte Carlo integration — the computation of Pi.



Perform a simple Monte Carlo integration to compute Pi

```
In [1]: %matplotlib inline
import matplotlib.pyplot as plt
import numpy as np
```

Set some parameters of the integration

```
In [18]: n = 10000 #number of samples for the integration
```

Make some uniformly sampled variables [-1,1]

```
In [19]: x = np.random.uniform(-1,1,n)
y = np.random.uniform(-1,1,n)
```

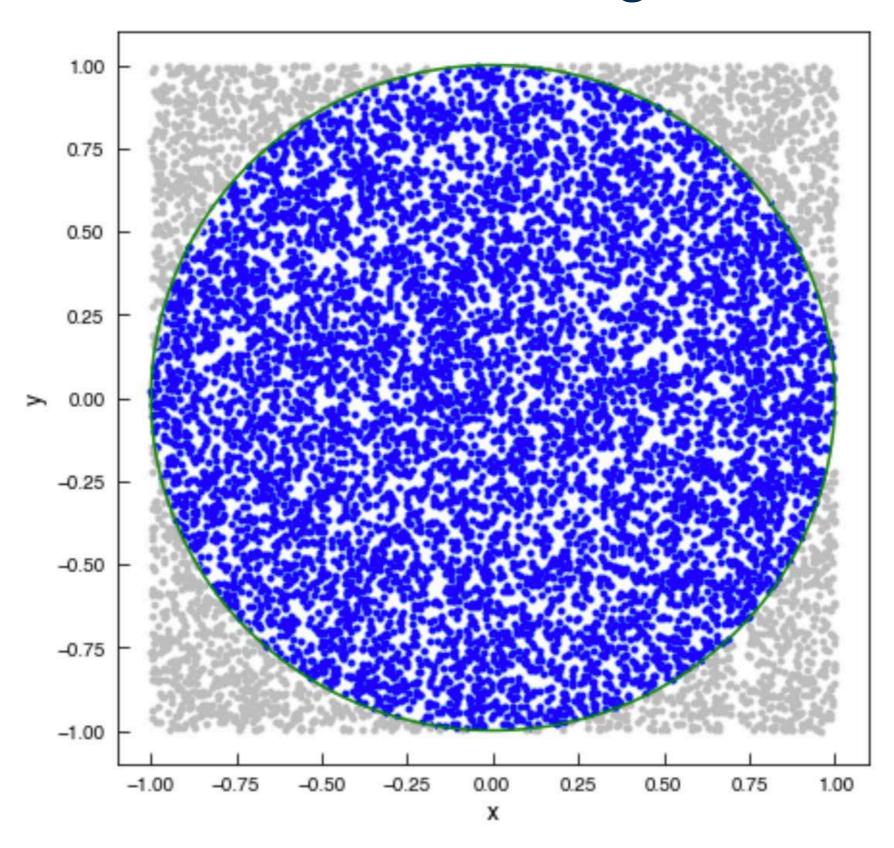
Find the number of samples within the unit circle

```
In [20]: ir = np.where((x**2 + y**2)<1.0)[0]

ur = np.where((x**2 + y**2)>=1.0)[0]
```

Plot the samples and the circle

```
In [21]: fig = plt.figure(figsize=(7,7))
         plt.xlim([-1.1,1.1])
         plt.ylim([-1.1,1.1])
         plt.plot(x[ir],y[ir],'.',color="blue")
         plt.plot(x[ur],y[ur],'.',color="0.75")
         theta = np.linspace(0,2*np.pi,1000)
         xc = np.cos(theta)
         yc = np.sin(theta)
         plt.plot(xc,yc,color="green")
         plt.xlabel('x')
         plt.ylabel('y')
         plt.show()
```





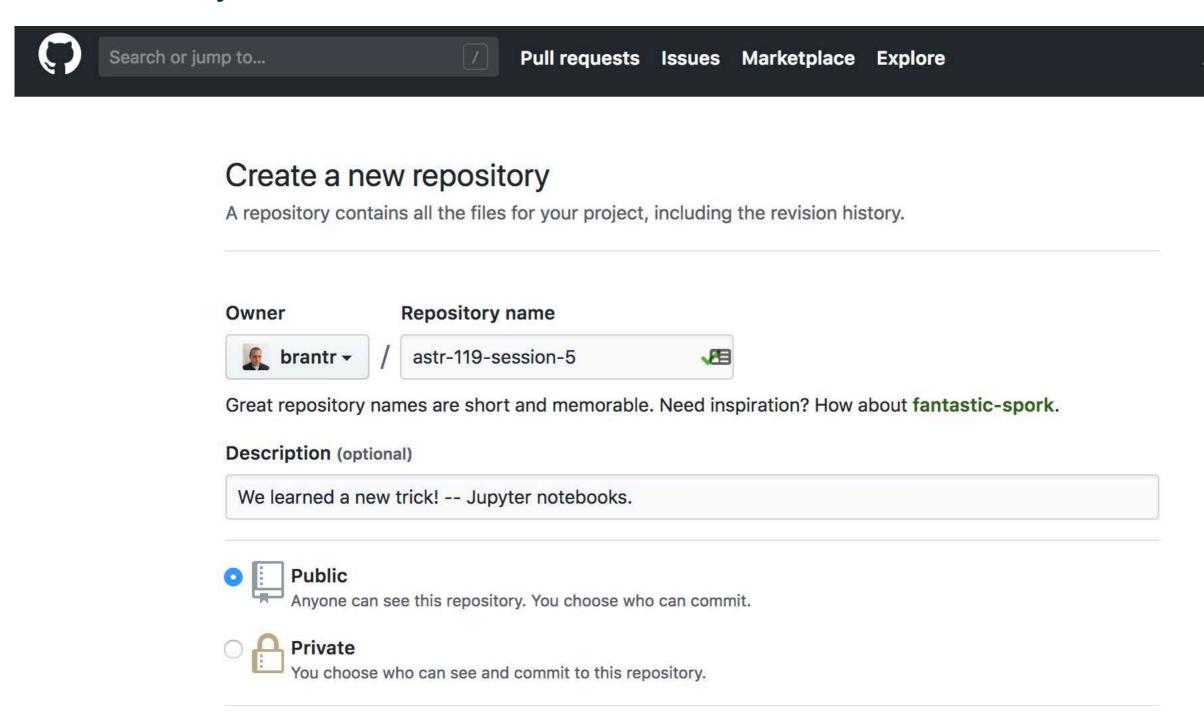
Report the result for Pi

```
In [24]: pi_approx = 4.0*len(ir)/float(n)
    error_pi = (pi_approx-np.pi)/np.pi
    print("Number of samples = ",n)
    print("Approximate pi = ",pi_approx)
    print("Error in approx = ",error_pi)

Number of samples = 10000
    Approximate pi = 3.1476
    Error in approx = 0.0019121977520996127
```

Save Your Work

Make a GitHub project "astr-119-session-17", and commit the programs you made today.



ORBITAL MECHANICS DEFINITIONS:

Consider the solar system, where the Sun effectively sits at the center of the system and the planets orbit about the Sun.

We are going to define a few useful quantities...



Create a simple solar system model

```
In [1]: 1 %matplotlib inline
2 import matplotlib.pyplot as plt
3 import numpy as np
4 from collections import namedtuple
```

Define a planet class

```
In [2]:
           class planet():
               "A planet in our solar system"
               def init (self, semimajor, eccentricity):
                  self.x = np.zeros(2) #x and y position
                  self.v = np.zeros(2) #x and y velocity
                  self.a g = np.zeros(2) #x and y acceleration
                  self.t = 0.0 #current time
                  self.dt = 0.0 #current timestep
                  self.a = semimajor #semimajor axis of the orbit
         9
                  self.e = eccentricity #eccentricity of the orbit
        10
             self.istep = 0
                                        #current integer timestepl
        11
                  self.name
        12
                                        #name for the planet
        13
```

The basic physical model that we have is that the planets orbit about the Sun, with the gravitational force supplying the centripetal acceleration to maintain the orbit. As a result, there is a minimum amount of information about the solar system that we need.

- 1) Mass of the sun
- 2) The gravitational constant G
- 3) The circular velocity about the sun at any location.
- 4) The gravitational acceleration about the sun at any location.

Define a dictionary with some constants

```
In [35]: 1 solar_system = { "M_sun":1.0, "G":39.4784176043574320}
```

Define some functions for setting circular velocity, and acceleration

Write a function to compute the gravitational acceleration on each planet from the Sun

```
def SolarGravitationalAcceleration(p):
5]:
     2
            G = solar system["G"]
            M = solar system["M_sun"]
     4
            r = (p.x[0]**2 + p.x[1]**2)**0.5
     5
     6
     7
            #acceleration in AU/yr/yr
            a grav = -1.0*G*M/r**2
     8
     9
    10
            #find the angle at this position
            if(p.x[0]==0.0):
    11
    12
                 if(p.x[1]>0.0):
                     theta = 0.5*np.pi
    13
    14
                 else:
    15
                     theta = 1.5*np.pi
    16
            else:
    17
                 theta = np.arctan2(p.x[1],p.x[0])
    18
    19
            #set the x and y components of the velocity
    20
            \#p.a \ g[0] = a \ grav * np.cos(theta)
    21
            \#p.a g[1] = a grav * np.sin(theta)
    22
            return a grav*np.cos(theta), a grav*np.sin(theta)
```

Compute the timestep

```
def calc_dt(p):
    #integration tolerance
    #ETA_TIME_STEP = 0.0004

#compute timestep
    eta = ETA_TIME_STEP
    v = (p.v[0]**2 + p.v[1]**2)**0.5
    a = (p.a_g[0]**2 + p.a_g[1]**2)**0.5
    dt = eta * np.fmin(1./np.fabs(v),1./np.fabs(a)**0.5)

return dt
```

As with any differential, we need to set the initial conditions. Since we are treating the planets as independent (currently), we can initialize the planets independently. For each planet, we need to

- 1) Set the semi-major axis of the orbit.
- 2) Set the eccentricity.
- 3) Initialize the position at time t=t_init.
- 4) Initialize the velocity at time t=t_init.
- 5) Calculate the initial acceleration at time t = t_init.
- 6) Use the initial velocity and acceleration to determine the time step.

Define the initial conditions

```
def SetPlanet(p, i):
 2
 3
       AU in km = 1.495979e + 8 \# an AU in km
 4
 5
       #circular velocity
       v c = 0.0
                       #circular velocity in AU/yr
 6
                  #velocity at perihelion in AU/yr
 7
       v = 0.0
 8
9
        #planet-by planet initial conditions
10
11
        #Mercury
12
        if(i==0):
            #semi-major axis in AU
13
14
            p.a = 57909227.0/AU in km
15
            #eccentricity
16
            p.e = 0.20563593
17
18
19
            #name
            p.name = "Mercury"
20
21
22
        #Venus
23
       elif(i==1):
            #semi-major axis in AU
24
25
            p.a = 108209475.0/AU in km
26
            #eccentricity
27
            p.e = 0.00677672
28
29
30
            #name
31
            p.name = "Venus"
```

```
30
            #name
            p.name = "Venus"
31
32
        #Earth
33
       elif(i==2):
34
                                                      Cell cont.
            #semi-major axis in AU
35
36
            p.a = 1.0
37
            #eccentricity
38
39
            p.e = 0.01671123
40
41
            #name
42
            p.name = "Earth"
43
       #set remaining properties
44
       p.t = 0.0
45
       p.x[0] = p.a*(1.0-p.e)
46
47
       p.x[1] = 0.0
48
49
       #get equiv circular velocity
       v c = SolarCircularVelocity(p)
50
51
       #velocity at perihelion
52
       v = v c*(1 + p.e)**0.5
53
54
       #set velocity
55
                         #no x velocity at perihelion
       p.v[0] = 0.0
56
                         #y velocity at perihelion (counter clockwise)
       p.v[1] = v e
57
58
       #calculate gravitational acceleration from Sun
59
60
       p.a g = SolarGravitationalAcceleration(p)
61
62
       #set timestep
       p.dt = calc dt(p)
63
64
```

As we've discussed, for conservative systems it is desirable to use a symplectic integration method that obeys a Hamiltonian of the system. The leapfrog method is one such method, so let's use that. To do so, we'll need:

- 1) A function to take the first position step (special for the first step only).
- 2) A function to take a full step in position.
- 3) A function to take a full step in velocity.

Write leapfrog integrator

```
def x_first_step(x_i, v_i, a_i, dt):
    #x_1/2 = x_0 + 1/2 v_0 Delta_t + 1/4 a_0 Delta t^2
    return x_i + 0.5*v_i*dt + 0.25*a_i*dt**2

def v_full_step(v_i, a_ipoh, dt):
    #v_i+1 = v_i + a_i+1/2 Delta t
    return v_i + a_ipoh*dt;

def x_full_step(x_ipoh, v_ip1, a_ipoh, dt):
    #x_3/2 = x_1/2 + v_i+1 Delta t
    return x_ipoh + v_ip1*dt;
```

We have chosen an integration method, but now we need to consider all the overhead involved in calculating the orbits. We'll want a file to save the information about the solar system over time, and we'll want to write to it in a convenient format. We'll need to evolve the global system many timesteps, and evolve each planet over its own time step potentially many times per global timestep. So, we'll need:

- 1) A driving routine to perform the bookkeeping of the orbital integration.
- 2) A routine to open the file we'll write the data out to.
- 3) A routine to write the data to file each global timestep.
- 4) To update the position and velocity according to the leapfrog integrator.
- 5) Update the planetary time steps according to their velocity and acceleration.

We'd like to save the results of the simulation and use that data to visualize the solar system. We want our data files to be efficient and easy to use (these are sometimes at cross purposes). Also, we want the position and velocities to be written at the same time (but they are offset by 1/2 time step in our integrator). We need to:

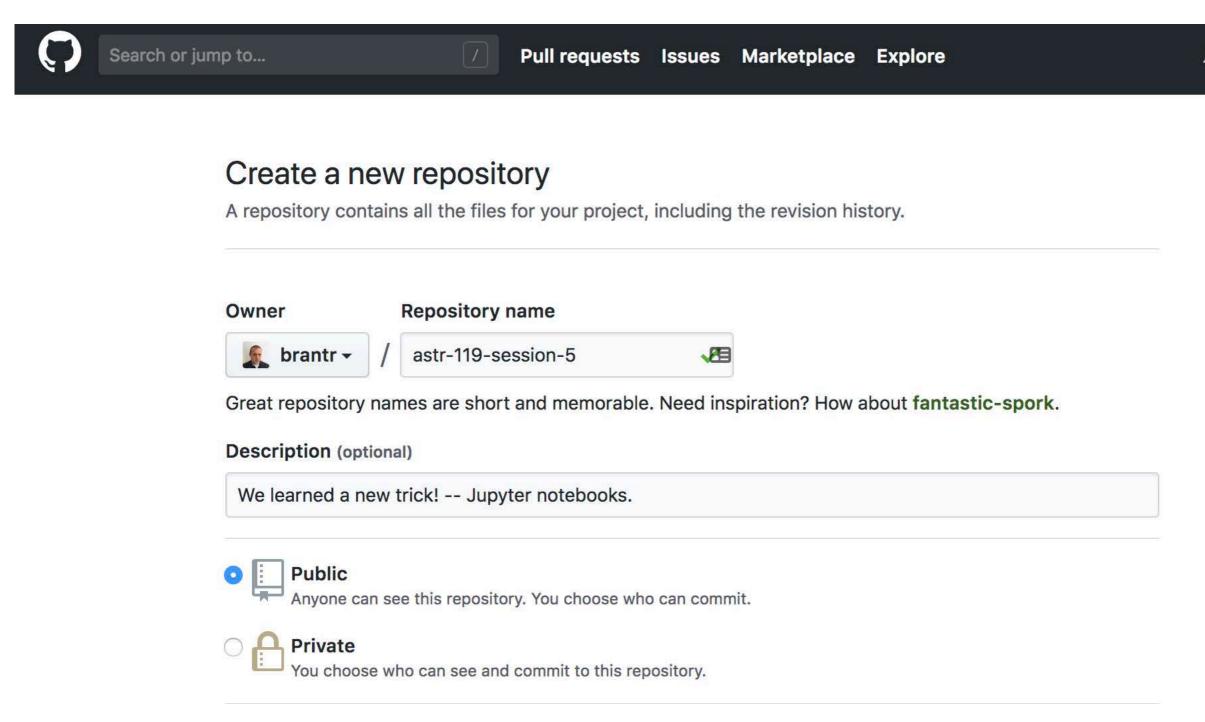
- 1) Open a data file.
- 2) Write to the data file in a useful format.
- 3) Synchronize the position and velocity data when writing to the file.

Write a function to save the data to file

```
def SaveSolarSystem(p, n planets, t, dt, istep, ndim):
]:
    3
           #loop over the number of planets
     4
           for i in range(n planets):
     5
                #define a filename
    6
                fname = "planet.%s.txt" % p[i].name
    7
    8
    9
                if(istep==0):
                    #create the file on the first timestep
   10
   11
                    fp = open(fname, "w")
   12
                else:
   13
                    #append the file on subsequent timesteps
                    fp = open(fname, "a")
   14
   15
   16
                #compute the drifted properties of the planet
   17
                v drift = np.zeros(ndim)
   18
   19
                for k in range(ndim):
   20
                    v_{drift[k]} = p[i].v[k] + 0.5*p[i].a_g[k]*p[i].dt
   21
   22
                #write the data to file
   23
                s = "%6d\t%6.5f\t%6.5f\t%6d\t%6.5f\t%6.5f\t% 6.5f\t% 6.5f\t% 6.5f\t% 6.5f\t% 6.5f\t% 6.5f\t% 6.5f\n" % \
   24
                    (istep,t,dt,p[i].istep,p[i].t,p[i].dt,p[i].x[0],p[i].x[1],v drift[0],v drift[1],
   25
                     p[i].a_g[0],p[i].a_g[1])
                fp.write(s)
   26
   27
   28
                #close the file
   29
   30
                fp.close()
```

Save Your Work

Make a GitHub project "astr-119-session-16", and commit the programs my_first_jupyter_notebook.ipynb and test_matplotlib.ipynb you made today.



Create a simple solar system model

```
In [1]: 1 %matplotlib inline
2 import matplotlib.pyplot as plt
3 import numpy as np
4 from collections import namedtuple
```

Define a planet class

```
In [2]:
           class planet():
               "A planet in our solar system"
               def init (self, semimajor, eccentricity):
                  self.x = np.zeros(2) #x and y position
                  self.v = np.zeros(2) #x and y velocity
                  self.a g = np.zeros(2) #x and y acceleration
                  self.t = 0.0 #current time
                  self.dt = 0.0 #current timestep
                  self.a = semimajor #semimajor axis of the orbit
         9
                  self.e = eccentricity #eccentricity of the orbit
        10
             self.istep = 0
                                       #current integer timestepl
        11
                 self.name
        12
                                       #name for the planet
        13
```

The basic physical model that we have is that the planets orbit about the Sun, with the gravitational force supplying the centripetal acceleration to maintain the orbit. As a result, there is a minimum amount of information about the solar system that we need.

- 1) Mass of the sun
- 2) The gravitational constant G
- 3) The circular velocity about the sun at any location.
- 4) The gravitational acceleration about the sun at any location.

Define a dictionary with some constants

```
In [35]: 1 solar_system = { "M_sun":1.0, "G":39.4784176043574320}
```

Define some functions for setting circular velocity, and acceleration

```
In [4]: 1 def SolarCircularVelocity(p):
2
3    G = solar_system["G"]
4    M = solar_system["M_sun"]
5    r = ( p.x[0]**2 + p.x[1]**2 )**0.5
6
7    #return the circular velocity
8    return (G*M/r)**0.5
```

Write a function to compute the gravitational acceleration on each planet from the Sun

```
def SolarGravitationalAcceleration(p):
5]:
     2
            G = solar system["G"]
            M = solar system["M_sun"]
     4
            r = (p.x[0]**2 + p.x[1]**2)**0.5
     5
     6
     7
            #acceleration in AU/yr/yr
            a grav = -1.0*G*M/r**2
     8
     9
    10
            #find the angle at this position
            if(p.x[0]==0.0):
    11
    12
                 if(p.x[1]>0.0):
                     theta = 0.5*np.pi
    13
    14
                 else:
    15
                     theta = 1.5*np.pi
    16
            else:
    17
                 theta = np.arctan2(p.x[1],p.x[0])
    18
    19
            #set the x and y components of the velocity
    20
            \#p.a \ g[0] = a \ grav * np.cos(theta)
    21
            \#p.a g[1] = a grav * np.sin(theta)
    22
            return a grav*np.cos(theta), a grav*np.sin(theta)
```

Compute the timestep

As with any differential, we need to set the initial conditions. Since we are treating the planets as independent (currently), we can initialize the planets independently. For each planet, we need to

- 1) Set the semi-major axis of the orbit.
- 2) Set the eccentricity.
- 3) Initialize the position at time t=t_init.
- 4) Initialize the velocity at time t=t_init.
- 5) Calculate the initial acceleration at time t = t_init.
- 6) Use the initial velocity and acceleration to determine the time step.

Define the initial conditions

```
def SetPlanet(p, i):
 2
 3
        AU in km = 1.495979e + 8 \# an AU in km
 4
 5
        #circular velocity
        v c = 0.0
                       #circular velocity in AU/yr
 6
                     #velocity at perihelion in AU/yr
 7
        v = 0.0
 8
9
        #planet-by planet initial conditions
10
11
        #Mercury
12
        if(i==0):
            #semi-major axis in AU
13
14
            p.a = 57909227.0/AU in km
15
            #eccentricity
16
            p.e = 0.20563593
17
18
19
            #name
            p.name = "Mercury"
20
21
22
        #Venus
23
        elif(i==1):
            #semi-major axis in AU
24
25
            p.a = 108209475.0/AU in km
26
            #eccentricity
27
            p.e = 0.00677672
28
29
30
            #name
31
            p.name = "Venus"
```

```
30
            #name
            p.name = "Venus"
31
32
        #Earth
33
       elif(i==2):
34
                                                      Cell cont.
            #semi-major axis in AU
35
36
            p.a = 1.0
37
            #eccentricity
38
39
            p.e = 0.01671123
40
41
            #name
            p.name = "Earth"
42
43
       #set remaining properties
44
       p.t = 0.0
45
       p.x[0] = p.a*(1.0-p.e)
46
47
       p.x[1] = 0.0
48
49
       #get equiv circular velocity
       v c = SolarCircularVelocity(p)
50
51
       #velocity at perihelion
52
       v = v c*(1 + p.e)**0.5
53
54
       #set velocity
55
                         #no x velocity at perihelion
       p.v[0] = 0.0
56
                       #y velocity at perihelion (counter clockwise)
       p.v[1] = v e
57
58
       #calculate gravitational acceleration from Sun
59
60
       p.a g = SolarGravitationalAcceleration(p)
61
62
       #set timestep
       p.dt = calc dt(p)
63
64
```

As we've discussed, for conservative systems it is desirable to use a symplectic integration method that obeys a Hamiltonian of the system. The leapfrog method is one such method, so let's use that. To do so, we'll need:

- 1) A function to take the first position step (special for the first step only).
- 2) A function to take a full step in position.
- 3) A function to take a full step in velocity.



Write leapfrog integrator

```
def x_first_step(x_i, v_i, a_i, dt):
    #x_1/2 = x_0 + 1/2 v_0 Delta_t + 1/4 a_0 Delta t^2
    return x_i + 0.5*v_i*dt + 0.25*a_i*dt**2

def v_full_step(v_i, a_ipoh, dt):
    #v_i+1 = v_i + a_i+1/2 Delta t
    return v_i + a_ipoh*dt;

def x_full_step(x_ipoh, v_ip1, a_ipoh, dt):
    #x_3/2 = x_1/2 + v_i+1 Delta t
    return x_ipoh + v_ip1*dt;
```