

Table 6 The complete prompting file, **equivellipse.PRO**, corresponding to the generic case called "equivellipse". This file is automatically created by "GENTEXT" once the GENOPT user has provided all variable names, one-line definitions, and "help" paragraphs during the entire "GENTEXT" interactive session. The variable names, the one-line definitions of the variables, and the "help" paragraphs, created by the GENOPT user during the GENTEXT interactive session, **will be seen by the "end" user**. If the GENOPT user has done his or her job well, the program system created by GENOPT (BEGIN, DECIDE, MAINSETUP, OPTIMIZE, CHOOSEPLOT, etc.), that is, the program system for the generic case, "**equivellipse**", to be used later by the "end" user for specific cases (such as a specific case called "eqellipse"), will be user-friendly. The prompting numbers (e.g. 10) corresponding to each prompt for input data (e.g. "number of x coordinates: npoint") are listed with the variable names (e.g. "npoint") and one-line definitions (e.g. "number of x-coordinates") in Table 2.

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5.0

OPTIMUM DESIGN OF ISOGRID-STIFFENED ELLIPSOIDAL HEAD

David Bushnell, retired (formerly with Lockheed Martin)

ABSTRACT: The externally pressurized head is elastic, has internal isogrid stiffening, and is attached to a short, unstiffened cylindrical shell of uniform thickness.

The BIGBOSOR4 computer program is used for the structural analysis and GENOPT is used to set up the user-friendly optimization program. Please read the following papers for descriptions of BIGBOSOR4 and GENOPT:

[1] Bushnell, D., "Automated optimum design of shells of revolution with application to ring-stiffened cylindrical shells with wavy walls", Proc. AIAA 41st SDM Meeting, AIAA Paper No. AIAA-2000-1663, April 2000. (Also see the Lockheed Martin report, LMMS P525674, November, 1999 for more details).

[2] Bushnell, D., "GENOPT - a program that writes user-friendly optimization code", Int. J. Solids Structures, Vol. 26, No. 9/10 pp. 1173-1210, 1990

10.1 number of x-coordinates: npoint

10.2

The ellipse is simulated by a number of shell segments (try 10) each of which has constant meridional curvature (toroidal). npoint is the number of x-coordinates corresponding to the ends of the toroidal segments that make up the equivalent ellipse. You might try to simulate the ellipse by using 10 toroidal segments. Then the value of npoint would be 11 npoint includes the apex of the ellipse ($x = 0$) and the equator of the ellipse ($x = a$, in which $a = \text{semimajor axis length}$).

15.1 Number Ixinput of rows in the array xinput: Ixinput

20.1 x-coordinates for ends of segments: xinput

20.2

Please make sure to include $x = 0$ and $x = a$ (equator) when you provide values for `xinput`.

25.1 length of semi-major axis: `ainput`

25.2

`ainput` is the maximum "x-dimension" of the ellipse.

The equation for the ellipse is $x^2/a^2 + y^2/b^2 = 1.0$

30.1 length of semi-minor axis of ellipse: `binput`

30.2

`binput` is the y-dimension of the ellipse, the equation for which is $x^2/a^2 + y^2/b^2 = 1.0$.

35.1 number of nodal points per segment: `nodes`

35.2

If you have about 10 segments, use a number less than 31. Use an odd number, greater than or equal to 11

40.1 max. x-coordinate for x-coordinate callouts: `xlimit`

40.2

`xlimit` has two functions:

1. a delimiter for the definition of callouts:

for $x < xlimit$ callouts are x-coordinates.

for $x > xlimit$ callouts are y-coordinates.

Set `xlimit` equal to about $a/2$, where a = length of the semi-major axis of the ellipse.

2. a delimiter for the boundary between Region 1

and Region 2, Design margins for maximum stress and minimum buckling load in the shell skin and in the isogrid stiffeners can be computed in two regions,

Region 1: $0 < x < xlimit$, and

Region 2: $xlimit < x < \text{semi-major axis}$.

45.1 skin thickness at `xinput`: `THKSKN`

45.2

`xinput` is the vector of x-coordinate callouts for thickness of the shell skin and height of the isogrid stiffeners.

50.1 height of isogrid members at `xinput`: `HIGHST`

50.2

`xinput` is the vector of x-coordinate callouts for thickness of the shell skin and height of the isogrid stiffeners.

55.1 spacing of the isogrid members: `SPACNG`

55.2

SPACNG = altitude of the equilateral triangle between adjacent isogrid members, measured to middle surfaces of isogrid members.
 $SPACNG = (\text{length of side of triangle}) * \sqrt{3} / 2.$
SPACNG is constant over the entire shell.

60.1 thickness of an isogrid stiffening member: THSTIF

60.2

THSTIF is constant over the entire shell.

65.1 thickness of the cylindrical shell: THKCYL

70.1 radius of the cylindrical shell: RADCYL

75.1 length of the cylindrical segment: LENCYL

80.1 amplitude of the axisymmetric imperfection: WIMP

80.2

Use a positive value greater than zero.
For a perfect shell, use a value of WIMP that is very, very small compared to the skin thickness.
The imperfections are in the shapes of the axisymmetric buckling modes obtained from linear theory for the PERFECT shell. The actual imperfections are equal to $WIMP * WSHAPE(i)$, $i = 1, NUMB$, in which NUMB = number of nodes in a shell segment.
In the paper about optimization of ellipsoidal shells the axisymmetric buckling modal imperfections are called "mode 1", "mode 2", "mode 3", "mode 4", corresponding to the number of the linear buckling eigenvalue corresponding to axisymmetric buckling.
Optimization can be performed with the use of two modes, "mode 1" and "mode 2" or with the use of four modes, "mode 1", "mode 2", "mode 3", "mode 4".
The shell is optimized with the plus and minus version of each axisymmetric buckling modal imperfection present by itself. In other words, the shell is optimized such that it will survive if any ONE of up to eight axisymmetric buckling modal imperfections of amplitude WIMP is present.
The plus and minus versions of the axisymmetric buckling modal imperfections are processed as different load sets "applied" to the shell:
Load set 1 has plus "mode 1" and plus "mode 2";
Load set 2 has minus "mode 1" and minus "mode 2";
Load set 3 has plus "mode 3" and plus "mode 4";
Load set 4 has minus "mode 3" and minus "mode 4".
Usually, optimization should be performed with use of only "mode 1" and "mode 2" imperfection shapes.

85.1 elastic modulus: EMATL

90.1 Poisson ratio of material: NUMATL

95.1 mass density of material: DNMATL

95.2

For example, the mass density of aluminum in English units is 0.000259

100.1 strategy control for imperfection shapes: IMODE

100.2

IMODE governs the strategy used to generate axisymmetric buckling modal imperfection shapes.

IMODE = 1 means use Strategy 1 (Do not use this)

IMODE = 2 means use Strategy 2 (Use this choice)

In Strategy 1 axisymmetric buckling modes are scanned until a mode is found in which the normal modal displacement amplitude at the apex of the shell is at least 0.7. (All buckling modes are normalized so that the maximum buckling modal displacement is 1.0. The buckling modal imperfection is the user-specified amplitude, WIMP, multiplied by the normalized buckling modal displacement distribution WSHAPE along the meridian of the shell.) The remaining n ($n = 2$ or $n = 4$) modes are selected without regard to the imperfection amplitude at the apex.

In Strategy 2 the first n axisymmetric buckling modes ($n = 2$ or $n = 4$) are selected regardless of their amplitude at the apex of the shell.

It is best to try Strategy 2 first.

105.1 Number NCASES of load cases (environments): NCASES

110.1 uniform external pressure: PRESS

115.0 collapse pressure with imperfection mode 1: CLAPS1

120.1 allowable pressure for axisymmetric collapse: CLAPS1A

125.1 factor of safety for axisymmetric collapse: CLAPS1F

130.0 general buckling load factor, mode 1: GENBK1

135.1 allowable general buckling load factor (use 1.0): GENBK1A

135.2

GENBK1 is defined as a "buckling load FACTOR", not as a "buckling LOAD". Therefore, you should always use a value of the "allowable general buckling load factor" equal to unity. This point holds for the treatment of all buckling allowables in this application.

140.1 factor of safety for general buckling: GENBK1F

140.2

Remember, this program already includes the effect of an axisymmetric buckling modal imperfection. If you use an imperfection amplitude, WIMP, significantly greater

than zero you should accordingly use a factor of safety closer to unity than you would for an almost perfect shell.

145.1 Number JSKNBK1 of columns in the array, SKNBK1: JSKNBK1
150.0 local skin buckling load factor, mode 1: SKNBK1
155.1 allowable buckling load factor: SKNBK1A
160.1 factor of safety for skin buckling: SKNBK1F
165.0 buckling load factor, isogrid member, mode 1: STFBK1
170.1 allowable for isogrid stiffener buckling (Use 1.): STFBK1A
175.1 factor of safety for isogrid stiffener buckling: STFBK1F
180.0 maximum stress in the shell skin, mode 1: SKNST1
185.1 allowable stress for the shell skin: SKNST1A
190.1 factor of safety for skin stress: SKNST1F
195.0 maximum stress in isogrid stiffener, mode 1: STFST1
200.1 allowable stress in isogrid stiffeners: STFST1A
205.1 factor of safety for stress in isogrid member: STFST1F
210.0 normal (axial) displacement at apex, mode 1: WAPEX1
215.1 allowable normal (axial) displacement at apex: WAPEX1A
220.1 factor of safety for WAPEX: WAPEX1F
225.0 collapse pressure with imperfection mode 2: CLAPS2
230.1 allowable pressure for axisymmetric collapse: CLAPS2A
235.1 factor of safety for axisymmetric collapse: CLAPS2F
240.0 general buckling load factor, mode 2: GENBK2
245.1 allowable general buckling load factor (use 1.0): GENBK2A
250.1 factor of safety for general buckling: GENBK2F
250.2

Remember, this program already includes the effect of an axisymmetric buckling modal imperfection. If you use an imperfection amplitude, WIMP, significantly greater than zero you should accordingly use a factor of safety closer to unity than you would for an almost perfect shell.

255.1 Number JSKNBK2 of columns in the array, SKNBK2: JSKNBK2
260.0 local skin buckling load factor, mode 2: SKNBK2
265.1 allowable skin buckling load factor (use 1.0): SKNBK2A
270.1 factor of safety for local skin buckling: SKNBK2F
275.0 buckling load factor for isogrid member, mode 2: STFBK2
280.1 allowable for isogrid stiffener buckling (Use 1.): STFBK2A
285.1 factor of safety for isogrid stiffener buckling: STFBK2F
290.0 maximum stress in the shell skin, mode 2: SKNST2
295.1 allowable stress for the shell skin: SKNST2A
300.1 factor of safety for skin stress: SKNST2F
305.0 maximum stress in isogrid stiffener, mode 2: STFST2
310.1 allowable stress in isogrid stiffeners: STFST2A
315.1 factor of safety for stress in isogrid member: STFST2F
320.0 normal (axial) displacement at apex, mode 2: WAPEX2

325.1 allowable normal (axial) displacement at apex: WAPEx2A

330.1 factor of safety for WAPEx: WAPEx2F

335.0 weight of the equivalent ellipsoidal head: WEIGHT

335.2

You can get the weight of just the head (no cylindrical shell
by setting the density of the cylindrical segment equal to 0.

NOTE: This is done in SUBROUTINE BOSDEC for you.

999.0 DUMMY ENTRY TO MARK END OF FILE

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