

Table 3 Input data for the PANDA2 processor MAINSETUP (allenrngs.OPT). This file, named allenrngs.OPT when MAINSETUP is executed, is stored here as the file, allenrngs.opt.table3

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n          $ Do you want a tutorial session and tutorial output?
-1000.00   $ Resultant (e.g. lb/in) normal to the plane of screen, Nx( 1)
0.000000   $ Resultant (e.g. lb/in) in the plane of the screen,   Ny( 1)
0.000000   $ In-plane shear in load set A,                        Nxy( 1)
N          $ Does the axial load vary in the L2 direction?
0.000000   $ Applied axial moment resultant (e.g. in-lb/in), Mx( 1)
0.000000   $ Applied hoop moment resultant (e.g. in-lb/in), My( 1)
Y          $ Want to include effect of transverse shear deformation?
0          $ IQUICK = quick analysis indicator (0 or 1)
Y          $ Do you want to vary M for minimum local buckling load?
N          $ Do you want to choose a starting M for local buckling?
Y          $ Do you want to perform a "low-axial-wavenumber" search?
0.999000   $ Factor of safety for general instability, FSGEN( 1)
0.999000   $ Factor of safety for panel (between rings) instability,FSPAN(1)
0.1000000  $ Minimum load factor for local buckling, FSLOC( 1)
1.000000   $ Minimum load factor for stiffener buckling (Type H), FSBSTR( 1)
1.000000   $ Factor of safety for stress, FSSTR( 1)
n          $ Do you want "flat skin" discretized module for local buckling?
n          $ Do you want to skip the KOITER local postbuckling analysis?
n          $ Do you want wide-column buckling to constrain the design?
0          $ Resultant (e.g. lb/in) normal to the plane of screen, Nx0( 1)
0          $ Resultant (e.g. lb/in) in the plane of the screen,   Ny0( 1)
0          $ Axial load applied along the (0=neutral plane), (1=panel skin)
0.000000   $ Uniform applied pressure [positive upward. See H(elp)], p( 1)
0.000000   $ Out-of-roundness, Wimpg1=(Max.diameter-Min.diam)/4, Wimpg1(1)
0.000000   $ Initial buckling modal general imperfection amplitude,Wimpg2(1)
0.000000   $ Initial buckling modal inter-ring imperfection amplitude,Wpan(1)
0.1000000E-06 $ Initial local imperfection amplitude(must be positive),Wloc(1)
Y          $ Do you want PANDA2 to change imperfection amplitudes?(1)
130        $ Axial halfwavelength of typical general buckling mode,AXLWAV(1)
Y          $ Do you want PANDA2 to find the general imperfection shape?( 1)
1.000000   $ Maximum allowable average axial strain (type H for HELP)( 1)
N          $ Is there any thermal "loading" in this load set (Y/N)?
Y          $ Do you want a "complete" analysis (type H for "Help")?
N          $ Want to provide another load set ?
N          $ Do you want to impose minimum TOTAL thickness of any segment?
N          $ Do you want to impose maximum TOTAL thickness of any segment?
N          $ Do you want to impose minimum TOTAL thickness of any segment?
N          $ Do you want to impose maximum TOTAL thickness of any segment?
N          $ Use reduced effective stiffness in panel skin (H(elp), Y or N)?
0          $ NPRINT= output index (-1=min. 0=good, 1=ok, 2=more, 3=too much)
1          $ Index for type of shell theory (0 or 1 or 2), ISAND
Y          $ Does the postbuckling axial wavelength of local buckles change?
Y          $ Want to suppress general buckling mode with many axial waves?
N          $ Do you want to double-check PANDA-type eigenvalues?
1          $ Choose (0=transverse inextensional; 1=transverse extensional)
1          $ Choose ICONSV = -1 or 0 or 1 or H(elp), ICONSV
1          $ Choose type of analysis (ITYPE = 1 or 2 or 3 or 4 or 5)
Y          $ Do you want to prevent secondary buckling (mode jumping)?
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N	\$ Do you want to use the "alternative" buckling solution?
5	\$ How many design iterations permitted in this run (5 to 25)?
1.000000	\$ MAXMAR. Plot only those margins less than MAXMAR (Type H)
N	\$ Do you want to reset total iterations to zero (Type H)?
1	\$ Index for objective (1=min. weight, 2=min. distortion)
1.000000	\$ FMARG (Skip load case with min. margin greater than FMARG)

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