BASIS FOR THE APPROACH

Steps:

- 2 Errors in *p* and *q* arise from error at static sources
- Find Δq required to match LAMS; hence Δp
- Refinements for accuracy
- **5** Δp is a function of measured quantities like p_m , q_m , α_m
- Flight maneuvers: checks and to calibrate T
- Use LAMS with the above results to measure T directly.

Results:

- Calibration of dynamic pressure, hence true airspeed, hence longitudinal component of wind
- Calibration of pressure
- Calibration of temperature via accurate measurements of pressure + GPS

