

Finite Element Analysis of Flat Slab with Calcpad

Using Bogner-Fox-Schmit (BFS) plate element [1]

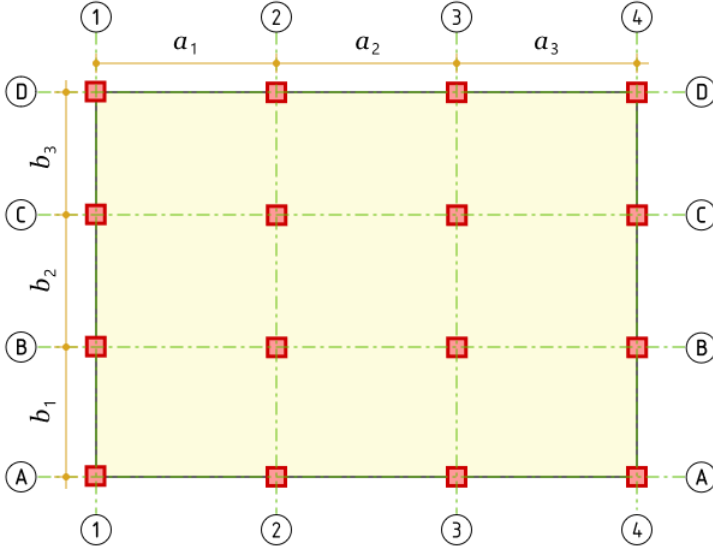
Input data

Span lengths

$$\vec{a} = \text{hp}([3.6; 4.2; 4.2; 3.6]) = [3.6 \ 4.2 \ 4.2 \ 3.6] \text{ m}, \quad \vec{b} = \text{hp}([3; 3.6; 3]) = [3 \ 3.6 \ 3] \text{ m}$$

$$\text{Number of axes} - n_{sa} = \text{len}(\vec{a}) + 1 = 5,$$

$$n_{sb} = \text{len}(\vec{b}) + 1 = 4$$



$$\text{Axis coordinates} - \vec{x}_s = [0 \ 3.6 \ 7.8 \ 12 \ 15.6] \text{ m},$$

$$\vec{y}_s = [0 \ 3 \ 6.6 \ 9.6] \text{ m}$$

$$\text{Slab dimensions} - l_a = \vec{x}_s = 15.6 \text{ m}, \quad l_b = \vec{y}_s = 9.6 \text{ m}$$

$$\text{Thickness} - t = 0.2 \text{ m}$$

$$\text{Load} - q = 10 \text{ kN/m}^2$$

$$\text{Modulus of elasticity} - E = 35000 \text{ MPa}$$

$$\text{Poisson's ratio} - \nu = 0.2$$

Finite element mesh

We will use BFS rectangular finite element with $n_{DOFs} = 16$

$$\text{Element dimensions} - a_1 = 0.6 \text{ m}, \quad b_1 = 0.6 \text{ m}$$

Number of elements and joints along a and b

$$\vec{n}_a = \text{ceiling}\left(\frac{\vec{a}}{a_1}\right) = [6 \ 7 \ 7 \ 6], \quad n_{ea} = \text{sum}(\vec{n}_a) = 26, \quad n_{ja} = n_{ea} + 1 = 27$$

$$\vec{n}_b = \text{ceiling}\left(\frac{\vec{b}}{b_1}\right) = [5 \ 6 \ 5], \quad n_{eb} = \text{sum}(\vec{n}_b) = 16, \quad n_{jb} = n_{eb} + 1 = 17$$

$$\text{Total number of elements} - n_e = n_{ea} \cdot n_{eb} = 416$$

$$\text{Total number of joints} - n_j = n_{ja} \cdot n_{jb} = 459$$

$$\text{Supported joints count} - n_s = n_{sa} \cdot n_{sb} = 20$$

Joint coordinates

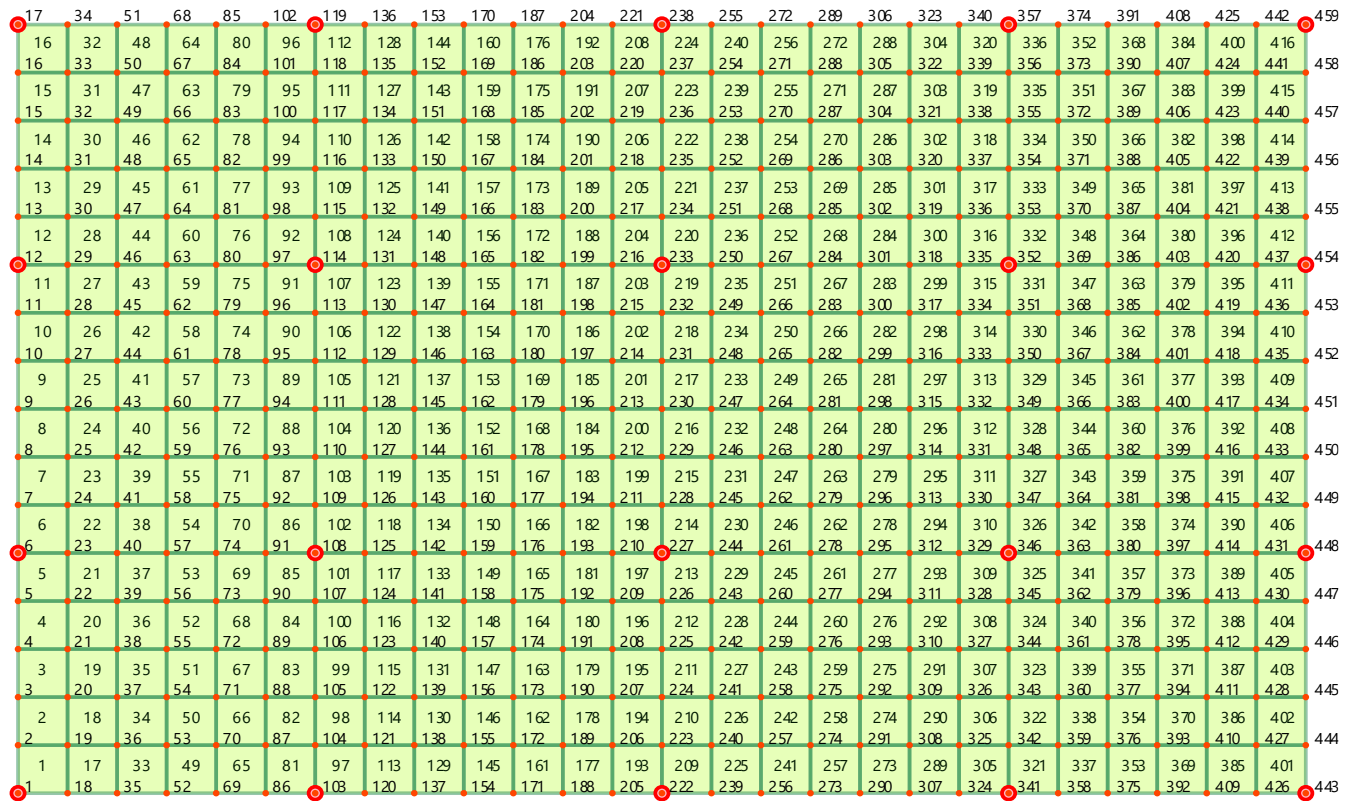
$$\vec{x}_j = [0 \ 0 \ 0 \ 0 \ 0 \ 0 \ 0 \ 0 \ 0 \ 0 \ 0 \ \dots \ 15.6] \text{ m}, \quad \vec{y}_j = [0 \ 0.6 \ 1.2 \ 1.8 \ 2.4 \ 3 \ 3.6 \ 4.2 \ 4.8 \ 5.4 \ \dots \ 9.6] \text{ m}$$

Numbers of joints at elements' corners

$$\text{transp}(e_j) = \begin{bmatrix} 1 & 2 & 3 & 4 & 5 & 6 & 7 & 8 & 9 & 10 & \dots & 441 \\ 18 & 19 & 20 & 21 & 22 & 23 & 24 & 25 & 26 & 27 & \dots & 458 \\ 19 & 20 & 21 & 22 & 23 & 24 & 25 & 26 & 27 & 28 & \dots & 459 \\ 2 & 3 & 4 & 5 & 6 & 7 & 8 & 9 & 10 & 11 & \dots & 442 \end{bmatrix}$$

$$\text{Supported joints} - \vec{s}_j = [1 \ 6 \ 12 \ 17 \ 103 \ 108 \ 114 \ 119 \ 222 \ 227 \ \dots \ 459]$$

$$\text{Joints for element } e - j_e(e) = \text{row}(e_j; e)$$



Finite element formulation

Shape functions

Along dimension a

Base functions

$$\Phi_{1a}(\xi) = 1 - \xi^2 \cdot (3 - 2 \cdot \xi)$$

$$\Phi_{2a}(\xi) = \xi \cdot a_1 \cdot (1 - \xi \cdot (2 - \xi))$$

$$\Phi_{3a}(\xi) = \xi^2 \cdot (3 - 2 \cdot \xi)$$

$$\Phi_{4a}(\xi) = \xi^2 \cdot a_1 \cdot (-1 + \xi)$$

First derivatives

$$\Phi'_{1a}(\xi) = -6 \cdot \frac{\xi}{a_1} \cdot (1 - \xi)$$

$$\Phi'_{2a}(\xi) = 1 - \xi \cdot (4 - 3 \cdot \xi)$$

$$\Phi'_{3a}(\xi) = 6 \cdot \frac{\xi}{a_1} \cdot (1 - \xi)$$

$$\Phi'_{4a}(\xi) = -\xi \cdot (2 - 3 \cdot \xi)$$

Second derivatives

$$\Phi''_{1a}(\xi) = -\frac{6}{a_1^2} \cdot (1 - 2 \cdot \xi)$$

$$\Phi''_{2a}(\xi) = -\frac{2}{a_1} \cdot (2 - 3 \cdot \xi)$$

$$\Phi''_{3a}(\xi) = \frac{6}{a_1^2} \cdot (1 - 2 \cdot \xi)$$

$$\Phi''_{4a}(\xi) = -\frac{2}{a_1} \cdot (1 - 3 \cdot \xi)$$

Along dimension b

Base functions

$$\Phi_{1b}(\eta) = 1 - \eta^2 \cdot (3 - 2 \cdot \eta)$$

$$\Phi_{2b}(\eta) = \eta \cdot b_1 \cdot (1 - \eta \cdot (2 - \eta))$$

$$\Phi_{3b}(\eta) = \eta^2 \cdot (3 - 2 \cdot \eta)$$

$$\Phi_{4b}(\eta) = \eta^2 \cdot b_1 \cdot (-1 + \eta)$$

For vertical displacements

$$N_{1,w}(\xi; \eta) = \Phi_{1a}(\xi) \cdot \Phi_{1b}(\eta)$$

$$N_{2,w}(\xi; \eta) = \Phi_{3a}(\xi) \cdot \Phi_{1b}(\eta)$$

$$N_{3,w}(\xi; \eta) = \Phi_{3a}(\xi) \cdot \Phi_{3b}(\eta)$$

$$N_{4,w}(\xi; \eta) = \Phi_{1a}(\xi) \cdot \Phi_{3b}(\eta)$$

For twist ψ

$$N_{1,\psi}(\xi; \eta) = \Phi_{2a}(\xi) \cdot \Phi_{2b}(\eta)$$

$$N_{2,\psi}(\xi; \eta) = \Phi_{4a}(\xi) \cdot \Phi_{2b}(\eta)$$

$$N_{3,\psi}(\xi; \eta) = \Phi_{4a}(\xi) \cdot \Phi_{4b}(\eta)$$

$$N_{4,\psi}(\xi; \eta) = \Phi_{2a}(\xi) \cdot \Phi_{4b}(\eta)$$

First derivatives

$$\Phi'_{1b}(\eta) = -6 \cdot \frac{\eta}{b_1} \cdot (1 - \eta)$$

$$\Phi'_{2b}(\eta) = 1 - \eta \cdot (4 - 3 \cdot \eta)$$

$$\Phi'_{3b}(\eta) = 6 \cdot \frac{\eta}{b_1} \cdot (1 - \eta)$$

$$\Phi'_{4b}(\eta) = -\eta \cdot (2 - 3 \cdot \eta)$$

For rotations ϑ_x

$$N_{1,\theta_x}(\xi; \eta) = \Phi_{2a}(\xi) \cdot \Phi_{1b}(\eta)$$

$$N_{2,\theta_x}(\xi; \eta) = \Phi_{4a}(\xi) \cdot \Phi_{1b}(\eta)$$

$$N_{3,\theta_x}(\xi; \eta) = \Phi_{4a}(\xi) \cdot \Phi_{3b}(\eta)$$

$$N_{4,\theta_x}(\xi; \eta) = \Phi_{2a}(\xi) \cdot \Phi_{3b}(\eta)$$

Second derivatives

$$\Phi''_{1b}(\eta) = -\frac{6}{b_1^2} \cdot (1 - 2 \cdot \eta)$$

$$\Phi''_{2b}(\eta) = -\frac{2}{b_1} \cdot (2 - 3 \cdot \eta)$$

$$\Phi''_{3b}(\eta) = \frac{6}{b_1^2} \cdot (1 - 2 \cdot \eta)$$

$$\Phi''_{4b}(\eta) = -\frac{2}{b_1} \cdot (1 - 3 \cdot \eta)$$

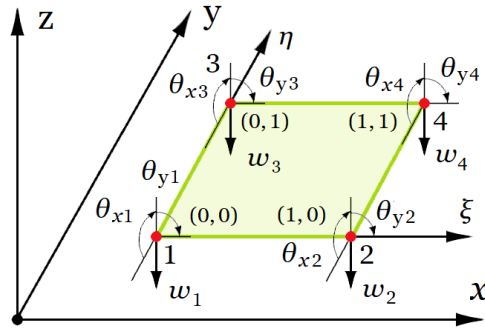
For rotations ϑ_y

$$N_{1,\theta_y}(\xi; \eta) = \Phi_{1a}(\xi) \cdot \Phi_{2b}(\eta)$$

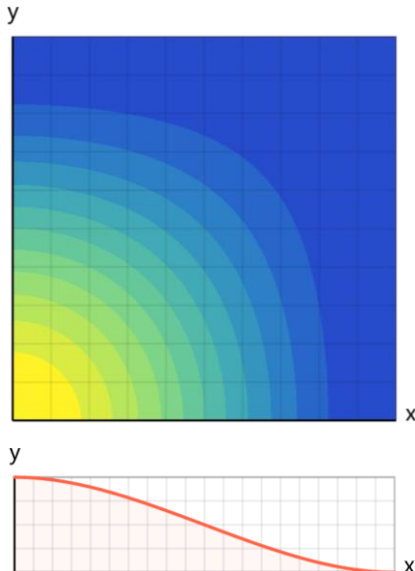
$$N_{2,\theta_y}(\xi; \eta) = \Phi_{3a}(\xi) \cdot \Phi_{2b}(\eta)$$

$$N_{3,\theta_y}(\xi; \eta) = \Phi_{3a}(\xi) \cdot \Phi_{4b}(\eta)$$

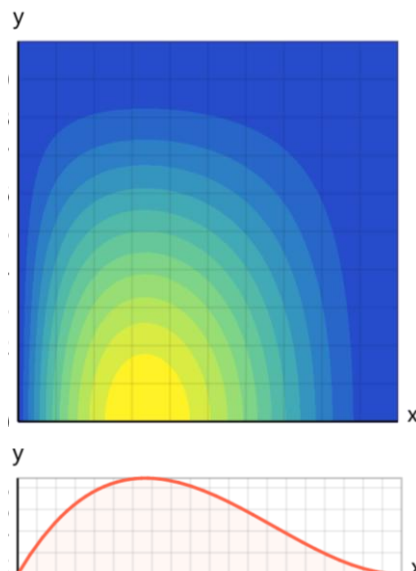
$$N_{4,\theta_y}(\xi; \eta) = \Phi_{1a}(\xi) \cdot \Phi_{4b}(\eta)$$



$N_{1,w}$ shape function plot



N_{1,θ_x} shape function plot



$$\text{Shape functions vector } N(i; \xi; \eta) = \text{take} \begin{pmatrix} i; N_{1,w}(\xi; \eta); N_{1,\theta_x}(\xi; \eta); N_{1,\theta_y}(\xi; \eta); N_{1,\psi}(\xi; \eta); \\ N_{2,w}(\xi; \eta); N_{2,\theta_x}(\xi; \eta); N_{2,\theta_y}(\xi; \eta); N_{2,\psi}(\xi; \eta); \\ N_{3,w}(\xi; \eta); N_{3,\theta_x}(\xi; \eta); N_{3,\theta_y}(\xi; \eta); N_{3,\psi}(\xi; \eta); \\ N_{4,w}(\xi; \eta); N_{4,\theta_x}(\xi; \eta); N_{4,\theta_y}(\xi; \eta); N_{4,\psi}(\xi; \eta) \end{pmatrix}$$

Constitutive matrix (stress - strain relationship)

$$D = \frac{E \cdot t^3}{12 \cdot (1 - \nu^2)} \cdot \text{hp} \left(\left[1; \nu; 0 \mid \nu; 1; 0 \mid 0; 0; \frac{1-\nu}{2} \right] \right) = \begin{bmatrix} 24.31 & 4.86 & 0 \\ 4.86 & 24.31 & 0 \\ 0 & 0 & 9.72 \end{bmatrix} \text{ kNm}$$

Strain-displacement matrix

$$B_1(j; \xi; \eta) = \text{take}(j; \Phi''_{1a}(\xi) \cdot \Phi_{1b}(\eta); \Phi''_{2a}(\xi) \cdot \Phi_{1b}(\eta); \Phi''_{1a}(\xi) \cdot \Phi_{2b}(\eta); \Phi''_{2a}(\xi) \cdot \Phi_{2b}(\eta); \Phi''_{3a}(\xi) \cdot \Phi_{1b}(\eta); \Phi''_{4a}(\xi) \cdot \Phi_{1b}(\eta); \Phi''_{3a}(\xi) \cdot \Phi_{2b}(\eta); \Phi''_{4a}(\xi) \cdot \Phi_{2b}(\eta); \Phi''_{3a}(\xi) \cdot \Phi_{3b}(\eta); \Phi''_{4a}(\xi) \cdot \Phi_{3b}(\eta); \Phi''_{3a}(\xi) \cdot \Phi_{4b}(\eta); \Phi''_{4a}(\xi) \cdot \Phi_{4b}(\eta); \Phi''_{1a}(\xi) \cdot \Phi_{3b}(\eta); \Phi''_{2a}(\xi) \cdot \Phi_{3b}(\eta); \Phi''_{1a}(\xi) \cdot \Phi_{4b}(\eta); \Phi''_{2a}(\xi) \cdot \Phi_{4b}(\eta))$$

$$B_2(j; \xi; \eta) = \text{take}(j; \Phi_{1a}(\xi) \cdot \Phi''_{1b}(\eta); \Phi_{2a}(\xi) \cdot \Phi''_{1b}(\eta); \Phi_{1a}(\xi) \cdot \Phi''_{2b}(\eta); \Phi_{2a}(\xi) \cdot \Phi''_{2b}(\eta); \Phi_{3a}(\xi) \cdot \Phi''_{1b}(\eta); \Phi_{4a}(\xi) \cdot \Phi''_{1b}(\eta); \Phi_{3a}(\xi) \cdot \Phi''_{2b}(\eta); \Phi_{4a}(\xi) \cdot \Phi''_{2b}(\eta); \Phi_{3a}(\xi) \cdot \Phi''_{3b}(\eta); \Phi_{4a}(\xi) \cdot \Phi''_{3b}(\eta); \Phi_{3a}(\xi) \cdot \Phi''_{4b}(\eta); \Phi_{4a}(\xi) \cdot \Phi''_{4b}(\eta); \Phi_{1a}(\xi) \cdot \Phi''_{3b}(\eta); \Phi_{2a}(\xi) \cdot \Phi''_{3b}(\eta); \Phi_{1a}(\xi) \cdot \Phi''_{4b}(\eta); \Phi_{2a}(\xi) \cdot \Phi''_{4b}(\eta))$$

$$B_3(j; \xi; \eta) = 2 \cdot \text{take}(j; \Phi'_{1a}(\xi) \cdot \Phi'_{1b}(\eta); \Phi'_{2a}(\xi) \cdot \Phi'_{1b}(\eta); \Phi'_{1a}(\xi) \cdot \Phi'_{2b}(\eta); \Phi'_{2a}(\xi) \cdot \Phi'_{2b}(\eta); \Phi'_{3a}(\xi) \cdot \Phi'_{1b}(\eta); \Phi'_{4a}(\xi) \cdot \Phi'_{1b}(\eta); \Phi'_{3a}(\xi) \cdot \Phi'_{2b}(\eta); \Phi'_{4a}(\xi) \cdot \Phi'_{2b}(\eta); \Phi'_{3a}(\xi) \cdot \Phi'_{3b}(\eta); \Phi'_{4a}(\xi) \cdot \Phi'_{3b}(\eta); \Phi'_{3a}(\xi) \cdot \Phi'_{4b}(\eta); \Phi'_{4a}(\xi) \cdot \Phi'_{4b}(\eta); \Phi'_{1a}(\xi) \cdot \Phi'_{3b}(\eta); \Phi'_{2a}(\xi) \cdot \Phi'_{3b}(\eta); \Phi'_{1a}(\xi) \cdot \Phi'_{4b}(\eta); \Phi'_{2a}(\xi) \cdot \Phi'_{4b}(\eta))$$

$$B(j; \xi; \eta) = \text{hp}([B_1(j; \xi; \eta); B_2(j; \xi; \eta); B_3(j; \xi; \eta)])$$

The coefficients of the stiffness matrix will be calculated by using the equation

$$K_{e,ij} = a_1 \cdot b_1 \cdot \int_0^1 \int_0^1 B_i(\xi; \eta)^T \cdot D \cdot B_j(\xi; \eta) d\xi d\eta$$

Element stiffness matrix – numerical evaluation

$$BTDB_e(i; j; \xi; \eta) = \text{transp}(B(i; \xi; \eta)) \cdot D \cdot B(j; \xi; \eta)$$

$$K_e(i; j) = a_1 \cdot b_1 \cdot \int_0^1 \int_0^1 BTDB_e(i; j; \xi; \eta) d\eta d\xi$$

$$\$Repeat \{ \$Repeat \{ K_{e,i,j} = K_e(i; j); j = i \dots n \}; i = 1 \dots n \}$$

$$K_e = \begin{bmatrix} 796.3 & 135.19 & 135.19 & 16.74 & -391.2 & 84.95 & -13.66 & 4.1 & -13.89 & 36.57 & \dots & 4.1 \\ 0 & 46.67 & 21.6 & 4.67 & -84.95 & 14.03 & -4.09 & 0.708 & -36.57 & 10.28 & \dots & -0.583 \\ 0 & 0 & 46.67 & 4.67 & -13.66 & 4.09 & 1.94 & -0.583 & -36.57 & 8.54 & \dots & 0.708 \\ 0 & 0 & 0 & 0.978 & -4.1 & 0.708 & 0.583 & -0.161 & -8.54 & 1.62 & \dots & -0.161 \\ 0 & 0 & 0 & 0 & 796.3 & -135.19 & 135.19 & -16.74 & -391.2 & 13.66 & \dots & 8.54 \\ 0 & 0 & 0 & 0 & 0 & 46.67 & -21.6 & 4.67 & 13.66 & 1.94 & \dots & -1.62 \\ 0 & 0 & 0 & 0 & 0 & 0 & 46.67 & -4.67 & -84.95 & 4.1 & \dots & 1.62 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0.978 & 4.1 & 0.583 & \dots & -0.231 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 796.3 & -135.19 & \dots & 4.1 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 46.67 & \dots & -0.708 \\ \vdots & \vdots & \vdots & \vdots & \vdots & \vdots & \vdots & \vdots & \vdots & \vdots & \ddots & \vdots \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & \dots & 0.978 \end{bmatrix}$$

Element load vector

$$F_{e,i} = a_1 \cdot b_1 \cdot \int_0^1 \int_0^1 N_i(\xi; \eta)^T \cdot q \, d\xi \, d\eta$$

$$\vec{F}_e = [0.9 \ 0.09 \ 0.09 \ 0.009 \ 0.9 \ -0.09 \ 0.09 \ -0.009 \ 0.9 \ -0.09 \ \dots \ -0.009] \text{ kN}$$

Solution

Global stiffness matrix

$$K = \begin{bmatrix} 10^{20} & 135.19 & 135.19 & 16.74 & -391.2 & -13.66 & 84.95 & 4.1 & 0 & 0 & \dots & 0 \\ 135.19 & 46.67 & 21.6 & 4.67 & -13.66 & 1.94 & 4.1 & -0.583 & 0 & 0 & \dots & 0 \\ 135.19 & 21.6 & 46.67 & 4.67 & -84.95 & -4.1 & 14.03 & 0.708 & 0 & 0 & \dots & 0 \\ 16.74 & 4.67 & 4.67 & 0.978 & -4.1 & 0.583 & 0.708 & -0.161 & 0 & 0 & \dots & 0 \\ -391.2 & -13.66 & -84.95 & -4.1 & 1592.59 & 270.37 & 0 & 0 & -391.2 & -13.66 & \dots & 0 \\ -13.66 & 1.94 & -4.1 & 0.583 & 270.37 & 93.33 & 0 & 0 & -13.66 & 1.94 & \dots & 0 \\ 84.95 & 4.1 & 14.03 & 0.708 & 0 & 0 & 93.33 & 9.33 & -84.95 & -4.1 & \dots & 0 \\ 4.1 & -0.583 & 0.708 & -0.161 & 0 & 0 & 9.33 & 1.96 & -4.1 & 0.583 & \dots & 0 \\ 0 & 0 & 0 & 0 & -391.2 & -13.66 & -84.95 & -4.1 & 1592.59 & 270.37 & \dots & 0 \\ 0 & 0 & 0 & 0 & -13.66 & 1.94 & -4.1 & 0.583 & 270.37 & 93.33 & \dots & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots & \vdots & \vdots & \vdots & \vdots & \vdots & \ddots & \vdots \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & \dots & 0.978 \end{bmatrix}$$

Global load vector

$$\vec{F} = [0.9 \ 0.09 \ 0.09 \ 0.009 \ 1.8 \ 0.18 \ 0 \ 0 \ 1.8 \ 0.18 \ \dots \ 0.009] \text{ kN}$$

Solution of the system of equations

$$\vec{Z} = \text{solve}(K; \vec{F}) = [0 \ 0.552 \ 0.383 \ -0.416 \ 0.203 \ 0.373 \ 0.265 \ -0.194 \ 0.299 \ 0.309 \ \dots \ -0.416] \text{ mm}$$

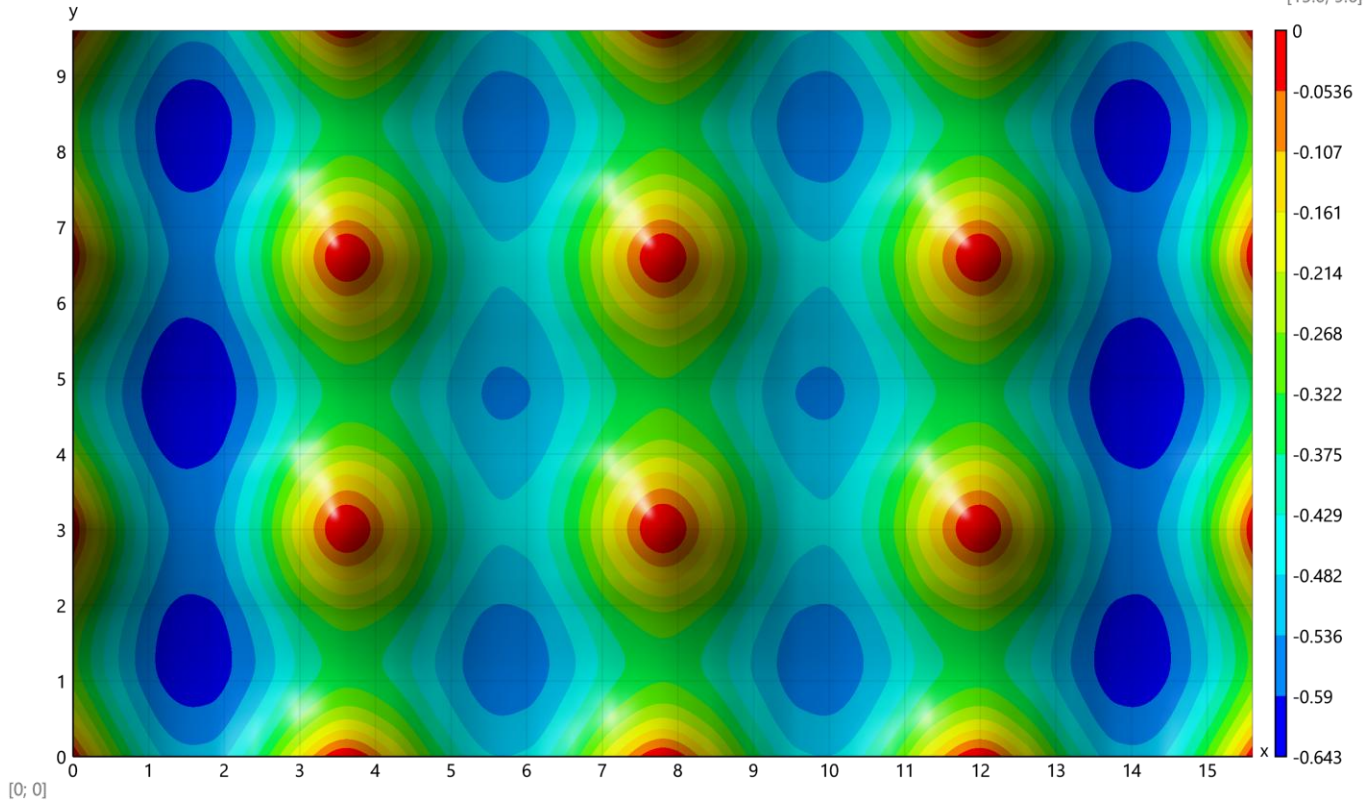
Results

Joint displacements – $\text{transp}(W_z) =$

0	0.303	0.488	0.512	0.383	0.165	0	0.139	0.34	0.469	...	0
0.203	0.419	0.562	0.581	0.485	0.337	0.25	0.31	0.438	0.531	...	0.203
0.299	0.482	0.605	0.62	0.537	0.417	0.35	0.387	0.485	0.562	...	0.299
0.261	0.461	0.593	0.608	0.513	0.375	0.299	0.342	0.455	0.542	...	0.261
0.121	0.386	0.55	0.565	0.443	0.253	0.138	0.217	0.381	0.493	...	0.121
0	0.34	0.527	0.542	0.404	0.173	0	0.135	0.336	0.463	...	0
0.139	0.398	0.556	0.566	0.44	0.247	0.129	0.206	0.367	0.478	...	0.139
0.299	0.487	0.608	0.612	0.511	0.369	0.287	0.325	0.433	0.516	...	0.299
0.363	0.526	0.632	0.635	0.544	0.42	0.35	0.376	0.464	0.536	...	0.363
0.299	0.487	0.608	0.612	0.511	0.369	0.287	0.325	0.433	0.516	...	0.299
⋮	⋮	⋮	⋮	⋮	⋮	⋮	⋮	⋮	⋮	⋮	⋮
0	0.303	0.488	0.512	0.383	0.165	0	0.139	0.34	0.469	...	0

mm

[15.6; 9.6]



Bending moments

$$Z_j(j) = \text{slice}(\vec{Z}; k_1 \cdot (j - 1) + 1; k_1 \cdot j)$$

$$Z_e(e) = \text{hp}([Z_j(e_{j,e,1}); Z_j(e_{j,e,2}); Z_j(e_{j,e,3}); Z_j(e_{j,e,4})])$$

Average bending moments at joints, kNm/m

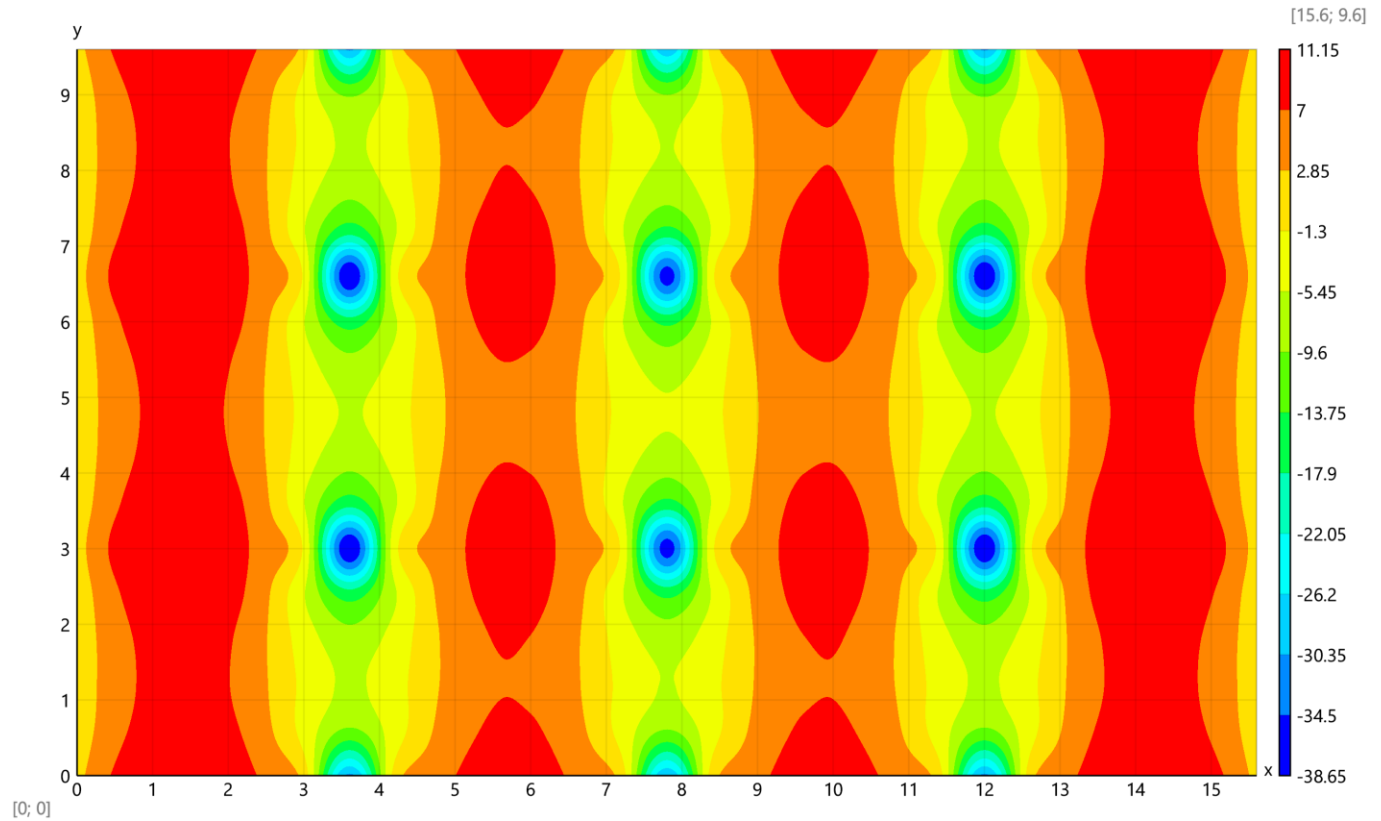
$$M_j = \begin{bmatrix} 1.5 & 0.31 & 0.22 & 0.156 & 0.157 & 0.998 & 0.156 & 0.152 & 0.194 & 0.152 & \dots & 1.5 \\ 1.57 & 7.81 & 9.34 & 7.67 & 3.33 & -28.36 & 3.15 & 7.32 & 8.9 & 7.32 & \dots & 1.57 \\ 8.09 & 3.77 & 0.486 & -2.48 & -4.46 & 0.155 & 4.78 & 2.84 & 1.19 \times 10^{-8} & -2.84 & \dots & 8.09 \end{bmatrix}$$

Bending moments for the plate

Bending moments - M_x - **transp**(M_x) =

1.5	8.5	11.07	10.48	6.8	0.884	-31.24	0.302	5.64	8.78	...	1.5
0.31	6.48	9.43	8.88	4.92	-2.7	-10.24	-3.26	3.81	7.25	...	0.31
0.22	5.78	8.72	8.14	4.04	-2.26	-6.06	-2.82	2.93	6.52	...	0.22
0.156	5.99	9.13	8.49	4.06	-3.26	-7.99	-3.83	2.91	6.8	...	0.156
0.157	7.27	10.37	9.56	4.97	-5.27	-16.12	-5.87	3.77	7.71	...	0.157
0.998	9.04	11.08	10.12	5.76	-2.14	-38.65	-2.75	4.52	8.19	...	0.998
0.156	7.22	10.28	9.44	4.85	-5.36	-16.18	-5.96	3.62	7.56	...	0.156
0.152	5.87	8.93	8.23	3.77	-3.42	-8.03	-4.01	2.59	6.47	...	0.152
0.194	5.51	8.38	7.69	3.47	-2.49	-5.74	-3.07	2.31	5.98	...	0.194
0.152	5.87	8.93	8.23	3.77	-3.42	-8.03	-4.01	2.59	6.47	...	0.152
⋮	⋮	⋮	⋮	⋮	⋮	⋮	⋮	⋮	⋮	⋮	⋮
1.5	8.5	11.07	10.48	6.8	0.884	-31.24	0.302	5.64	8.78	...	1.5

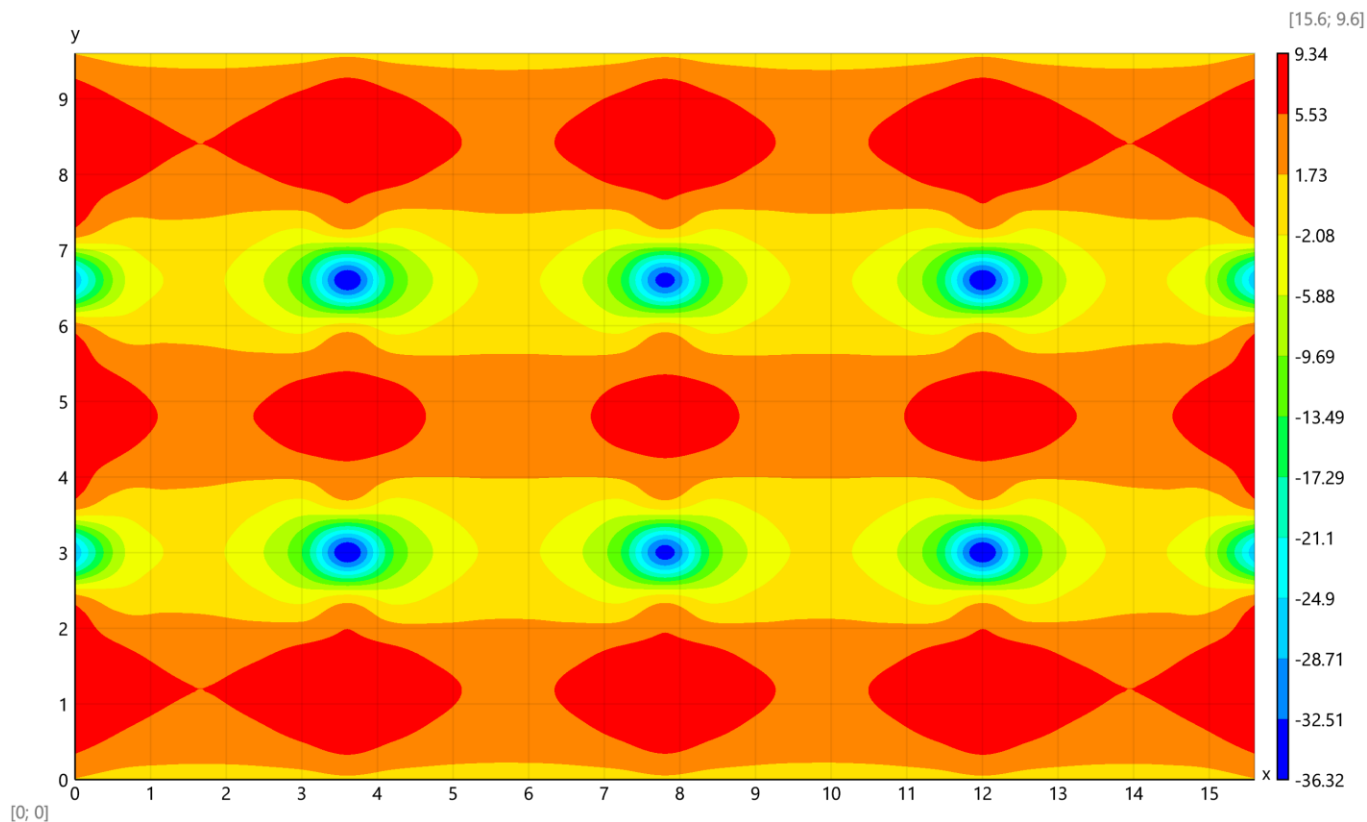
kNm/m



Bending moments M_y – **transp**(M_y) =

1.57	0.321	0.233	0.215	0.168	0.18	1.02	0.179	0.167	0.21	...	1.57
7.81	5.38	4.31	4.13	4.74	6.34	8.23	6.28	4.61	3.89	...	7.81
9.34	7.13	5.8	5.55	6.41	8.12	9.05	8.01	6.19	5.14	...	9.34
7.67	5.42	4.26	3.92	4.23	5.42	6.34	5.3	3.98	3.48	...	7.67
3.33	-0.328	0.54	0.406	-1.17	-3.41	-0.33	-3.53	-1.4	0.0996	...	3.33
-28.36	-7.04	-1.94	-1.65	-4.99	-13.64	-36.32	-13.77	-5.22	-1.86	...	-28.36
3.15	-0.507	0.349	0.18	-1.46	-3.76	-0.723	-3.92	-1.77	-0.252	...	3.15
7.32	5.09	3.91	3.49	3.68	4.69	5.5	4.48	3.25	2.79	...	7.32
8.9	6.74	5.35	4.96	5.61	6.91	7.57	6.69	5.14	4.18	...	8.9
7.32	5.09	3.91	3.49	3.68	4.69	5.5	4.48	3.25	2.79	...	7.32
⋮	⋮	⋮	⋮	⋮	⋮	⋮	⋮	⋮	⋮	⋮	⋮
1.57	0.32	0.233	0.215	0.168	0.18	1.02	0.179	0.167	0.21	...	1.57

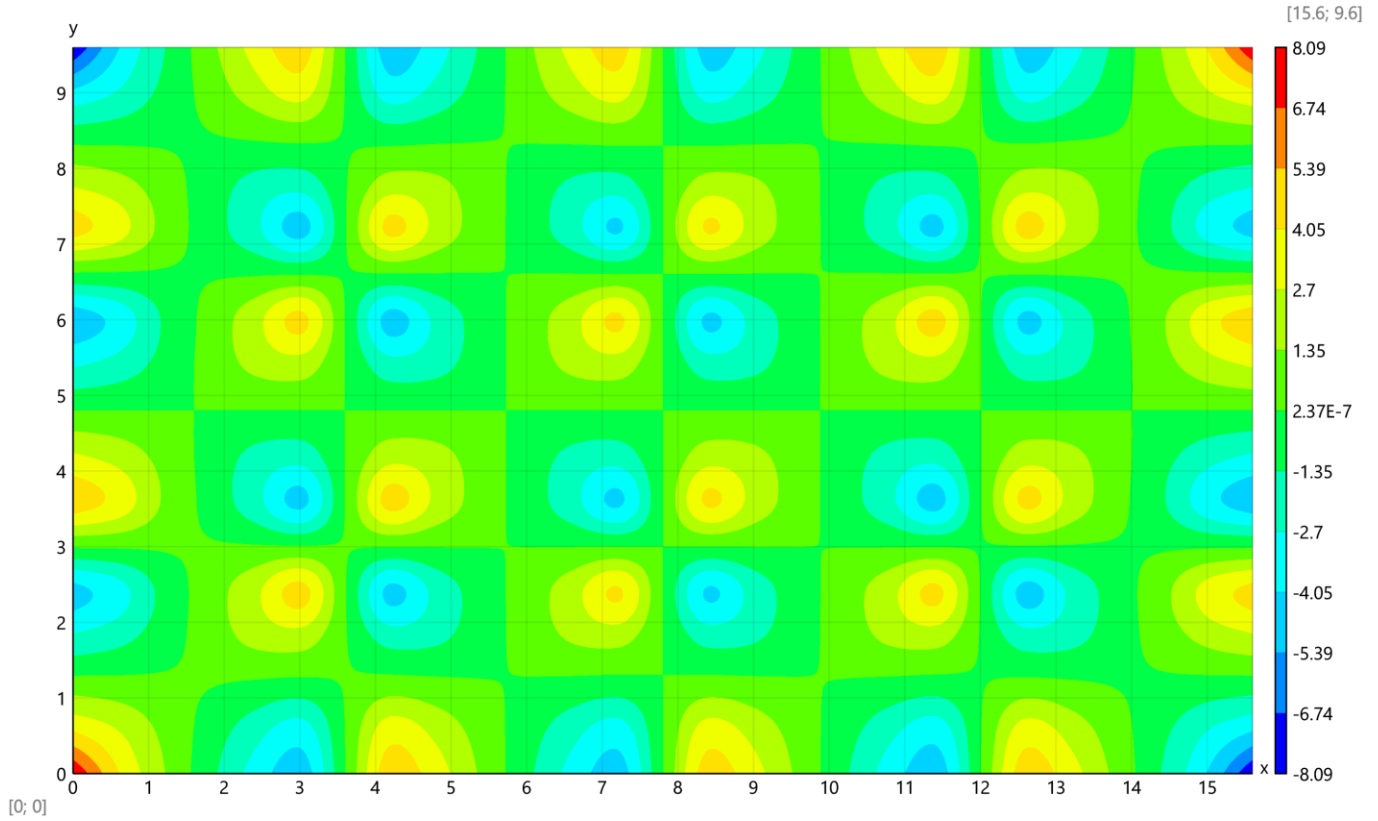
kNm/m



Bending moments $M_{xy} - \text{transp}(M_{xy}) =$

8.09	4.11	1.42	-0.946	-3.23	-4.78	0.0218	4.84	3.34	1.14	...	-8.09
3.77	2.57	0.983	-0.516	-2.06	-3.23	0.0582	3.35	2.21	0.731	...	-3.77
0.486	0.367	0.21	0.0857	-0.12	-0.314	0.0735	0.461	0.267	0.0568	...	-0.486
-2.48	-1.8	-0.601	0.617	1.75	2.09	0.0763	-1.95	-1.63	-0.581	...	2.48
-4.46	-3.24	-0.898	0.683	2.37	4.52	0.0758	-4.38	-2.26	-0.682	...	4.46
0.155	0.161	0.149	0.125	0.0964	0.078	0.0739	0.0699	0.0513	0.0211	...	-0.155
4.78	3.57	1.19	-0.451	-2.21	-4.4	0.0635	4.54	2.38	0.725	...	-4.78
2.84	2.13	0.868	-0.438	-1.68	-2.08	0.0378	2.17	1.8	0.63	...	-2.84
0	0	0	0	0	0	0	0	0	0	...	-0
-2.84	-2.13	-0.868	0.438	1.68	2.08	-0.0378	-2.17	-1.8	-0.63	...	2.84
⋮	⋮	⋮	⋮	⋮	⋮	⋮	⋮	⋮	⋮	⋮	⋮
-8.09	-4.11	-1.42	0.946	3.23	4.78	-0.0218	-4.84	-3.34	-1.14	...	8.09

kNm/m



Element stiffness matrix calculation by analytical expressions (faster)

$$\alpha = \frac{b_1}{a_1} = 1$$

$$K_{e_{1,1}} = \frac{156}{35} \cdot \left(\frac{\alpha}{a_1^2} + \frac{1}{\alpha \cdot b_1^2} \right) + \frac{72}{25 \cdot A_1} = 32.76$$

$$K_{e_{5,5}} = K_{e_{1,1}}, K_{e_{9,9}} = K_{e_{1,1}}, K_{e_{13,13}} = K_{e_{1,1}}$$

$$K_{e_{1,2}} = \frac{2}{35} \cdot \left(\frac{39 \cdot \alpha}{a_1} + \frac{11}{\alpha^2 \cdot b_1} \right) + \frac{30 \cdot \nu + 6}{25 \cdot b_1} = 5.56$$

$$K_{e_{5,6}} = -K_{e_{1,2}}, K_{e_{9,10}} = -K_{e_{1,2}}, K_{e_{13,14}} = K_{e_{1,2}}$$

$$K_{e_{1,3}} = \frac{2}{35} \cdot \left(\frac{11 \cdot \alpha^2}{a_1} + \frac{39}{\alpha \cdot b_1} \right) + \frac{30 \cdot \nu + 6}{25 \cdot a_1} = 5.56$$

$$K_{e_{5,7}} = K_{e_{1,3}}, K_{e_{9,11}} = -K_{e_{1,3}}, K_{e_{13,15}} = -K_{e_{1,3}}$$

$$K_{e_{1,4}} = \frac{11}{35} \cdot \left(\alpha^2 + \frac{1}{\alpha^2} \right) + \frac{10 \cdot \nu + 1}{50} = 0.689$$

$$K_{e_{5,8}} = -K_{e_{1,4}}, K_{e_{9,12}} = K_{e_{1,4}}, K_{e_{13,16}} = -K_{e_{1,4}}$$

$$K_{e_{1,5}} = \frac{2}{35} \cdot \left(\frac{27}{\alpha \cdot b_1^2} - \frac{78 \cdot \alpha}{a_1^2} \right) - \frac{72}{25 \cdot A_1} = -16.1$$

$$K_{e_{9,13}} = K_{e_{1,5}}$$

$$K_{e_{1,6}} = \frac{13}{35} \cdot \left(\frac{6 \cdot \alpha}{a_1} - \frac{1}{\alpha^2 \cdot b_1} \right) + \frac{6}{25 \cdot b_1} = 3.5$$

$$K_{e_{2,5}} = -K_{e_{1,6}}, \quad K_{e_{9,14}} = -K_{e_{1,6}}, \quad K_{e_{10,13}} = K_{e_{1,6}}$$

$$K_{e_{1,7}} = \frac{1}{35} \cdot \left(\frac{27}{\alpha \cdot b_1} - \frac{22 \cdot \alpha^2}{a_1} \right) - \frac{30 \cdot \nu + 6}{25 \cdot a_1} = -0.562$$

$$K_{e_{3,5}} = K_{e_{1,7}}, \quad K_{e_{9,15}} = -K_{e_{1,7}}, \quad K_{e_{11,13}} = -K_{e_{1,7}}$$

$$K_{e_{2,4}} = \frac{2}{35} \cdot \left(\frac{11 \cdot \alpha \cdot b_1}{3} + \frac{a_1}{\alpha^2} \right) + \frac{2 \cdot a_1 \cdot (5 \cdot \nu + 1)}{75} = 0.192$$

$$K_{e_{6,8}} = K_{e_{2,4}}, \quad K_{e_{10,12}} = -K_{e_{2,4}}, \quad K_{e_{14,16}} = -K_{e_{2,4}}$$

$$K_{e_{2,6}} = \frac{1}{35} \cdot \left(26 \cdot \alpha - \frac{3}{\alpha^3} \right) - \frac{2}{25 \cdot \alpha} = 0.577$$

$$K_{e_{10,14}} = K_{e_{2,6}}$$

$$K_{e_{2,8}} = \frac{1}{35} \cdot \left(\frac{11 \cdot \alpha \cdot b_1}{3} - \frac{3 \cdot a_1}{2 \cdot \alpha^2} \right) - \frac{a_1 \cdot (5 \cdot \nu + 1)}{150} = 0.0291$$

$$K_{e_{4,6}} = K_{e_{2,8}}, \quad K_{e_{10,16}} = -K_{e_{2,8}}, \quad K_{e_{12,14}} = -K_{e_{2,8}}$$

$$K_{e_{2,10}} = \frac{3}{35} \cdot \left(\frac{1}{\alpha^3} + 3 \cdot \alpha \right) + \frac{2}{25 \cdot \alpha} = 0.423$$

$$K_{e_{6,14}} = K_{e_{2,10}}$$

$$K_{e_{2,12}} = -\frac{1}{70} \cdot \left(\frac{3 \cdot a_1}{\alpha^2} + \frac{13 \cdot \alpha \cdot b_1}{3} \right) - \frac{a_1}{150} = -0.0669$$

$$K_{e_{4,10}} = -K_{e_{2,12}}, \quad K_{e_{6,16}} = K_{e_{2,12}}, \quad K_{e_{8,14}} = -K_{e_{2,12}}$$

$$K_{e_{2,14}} = \frac{2}{35} \cdot \left(9 \cdot \alpha - \frac{2}{\alpha^3} \right) - \frac{8}{25 \cdot \alpha} = 0.08$$

$$K_{e_{6,10}} = K_{e_{2,14}}$$

$$K_{e_{2,15}} = \frac{1}{35} \cdot \left(\frac{11}{\alpha^2} - \frac{13 \cdot \alpha^2}{2} \right) + \frac{5 \cdot \nu + 1}{50} = 0.169$$

$$K_{e_{3,14}} = -K_{e_{2,15}}, \quad K_{e_{6,11}} = -K_{e_{2,15}}, \quad K_{e_{7,10}} = K_{e_{2,15}}$$

$$K_{e_{1,14}} = \frac{1}{35} \cdot \left(\frac{27 \cdot \alpha}{a_1} - \frac{22}{\alpha^2 \cdot b_1} \right) - \frac{30 \cdot \nu + 6}{25 \cdot b_1} = -0.562$$

$$K_{e_{2,13}} = K_{e_{1,14}}, \quad K_{e_{5,10}} = -K_{e_{1,14}}, \quad K_{e_{6,9}} = -K_{e_{1,14}}$$

$$K_{e_{1,15}} = \frac{13}{35} \cdot \left(\frac{6}{\alpha \cdot b_1} - \frac{\alpha^2}{a_1} \right) + \frac{6}{25 \cdot a_1} = 3.5$$

$$K_{e_{3,13}} = -K_{e_{1,15}}, \quad K_{e_{5,11}} = K_{e_{1,15}}, \quad K_{e_{7,9}} = -K_{e_{1,15}}$$

$$K_{e_{1,16}} = \frac{1}{35} \cdot \left(\frac{11}{\alpha^2} - \frac{13 \cdot \alpha^2}{2} \right) + \frac{5 \cdot \nu + 1}{50} = 0.169$$

$$K_{e_{4,13}} = -K_{e_{1,16}}, K_{e_{8,9}} = K_{e_{1,16}}, K_{e_{5,12}} = -K_{e_{1,16}}$$

$$K_{e_{2,2}} = \frac{4}{35} \cdot \left(13 \cdot \alpha + \frac{1}{\alpha^3} \right) + \frac{8}{25 \cdot \alpha} = 1.92$$

$$K_{e_{6,6}} = K_{e_{2,2}}, K_{e_{10,10}} = K_{e_{2,2}}, K_{e_{14,14}} = K_{e_{2,2}}$$

$$K_{e_{2,3}} = \frac{11}{35} \cdot \left(\alpha^2 + \frac{1}{\alpha^2} \right) + \frac{60 \cdot \nu + 1}{50} = 0.889$$

$$K_{e_{6,7}} = -K_{e_{2,3}}, K_{e_{10,11}} = K_{e_{2,3}}, K_{e_{14,15}} = -K_{e_{2,3}}$$

$$K_{e_{3,16}} = -\frac{1}{35} \cdot \left(\frac{3 \cdot \alpha^2 \cdot b_1}{2} - \frac{11 \cdot a_1}{3 \cdot \alpha} \right) - \frac{b_1 \cdot (5 \cdot \nu + 1)}{150} = 0.0291$$

$$K_{e_{4,15}} = K_{e_{3,16}}, K_{e_{7,12}} = -K_{e_{3,16}}, K_{e_{8,11}} = -K_{e_{3,16}}$$

$$K_{e_{4,4}} = \frac{4}{105} \cdot \left(\alpha \cdot b_1^2 + \frac{a_1^2}{\alpha} \right) + \frac{8 \cdot A_1}{225} = 0.0402$$

$$K_{e_{8,8}} = K_{e_{4,4}}, K_{e_{12,12}} = K_{e_{4,4}}, K_{e_{16,16}} = K_{e_{4,4}}$$

$$K_{e_{4,8}} = \frac{1}{35} \cdot \left(\frac{2 \cdot \alpha \cdot b_1^2}{3} - \frac{a_1^2}{\alpha} \right) - \frac{2 \cdot A_1}{225} = -0.00663$$

$$K_{e_{12,16}} = K_{e_{4,8}}$$

$$K_{e_{4,12}} = -\frac{1}{70} \cdot \left(\frac{a_1^2}{\alpha} + \alpha \cdot b_1^2 \right) + \frac{A_1}{450} = -0.00949$$

$$K_{e_{8,16}} = K_{e_{4,12}}$$

$$K_{e_{4,16}} = \frac{1}{35} \cdot \left(\frac{2 \cdot a_1^2}{3 \cdot \alpha} - \alpha \cdot b_1^2 \right) - \frac{2 \cdot A_1}{225} = -0.00663$$

$$K_{e_{8,12}} = K_{e_{4,16}}$$

$$K_{e_{1,8}} = \frac{1}{35} \cdot \left(11 \cdot \alpha^2 - \frac{13}{2 \cdot \alpha^2} \right) + \frac{5 \cdot \nu + 1}{50} = 0.169$$

$$K_{e_{2,7}} = -K_{e_{1,8}}, K_{e_{3,6}} = K_{e_{1,8}}, K_{e_{4,5}} = -K_{e_{1,8}}, K_{e_{9,16}} = K_{e_{1,8}}$$

$$K_{e_{10,15}} = -K_{e_{1,8}}, K_{e_{11,14}} = K_{e_{1,8}}, K_{e_{12,13}} = -K_{e_{1,8}}$$

$$K_{e_{1,9}} = -\frac{54}{35} \cdot \left(\frac{\alpha}{a_1^2} + \frac{1}{\alpha \cdot b_1^2} \right) + \frac{72}{25 \cdot A_1} = -0.571$$

$$K_{e_{5,13}} = K_{e_{1,9}}$$

$$K_{e_{1,10}} = \frac{1}{35} \cdot \left(\frac{27 \cdot \alpha}{a_1} + \frac{13}{\alpha^2 \cdot b_1} \right) - \frac{6}{25 \cdot b_1} = 1.5$$

$$K_{e_{2,9}} = -K_{e_{1,10}}, K_{e_{5,14}} = -K_{e_{1,10}}, K_{e_{6,13}} = K_{e_{1,10}}$$

$$K_{e_{1,11}} = \frac{1}{35} \cdot \left(\frac{13 \cdot \alpha^2}{a_1} + \frac{27}{\alpha \cdot b_1} \right) - \frac{6}{25 \cdot a_1} = 1.5$$

$$K_{e_{3,9}} = -K_{e_{1,11}}, K_{e_{5,15}} = K_{e_{1,11}}, K_{e_{7,13}} = -K_{e_{1,11}}$$

$$K_{e_{1,12}} = -\frac{13}{70} \cdot \left(\alpha^2 + \frac{1}{\alpha^2} \right) + \frac{1}{50} = -0.351$$

$$K_{e_{2,11}} = -K_{e_{1,12}}, K_{e_{3,10}} = -K_{e_{1,12}}, K_{e_{4,9}} = K_{e_{1,12}}, K_{e_{5,16}} = -K_{e_{1,12}}$$

$$K_{e_{6,15}} = K_{e_{1,12}}, K_{e_{7,14}} = K_{e_{1,12}}, K_{e_{8,13}} = -K_{e_{1,12}}$$

$$K_{e_{1,13}} = \frac{2}{35} \cdot \left(\frac{27 \cdot \alpha}{a_1^2} - \frac{78}{\alpha \cdot b_1^2} \right) - \frac{72}{25 \cdot A_1} = -16.1$$

$$K_{e_{5,9}} = K_{e_{1,13}}$$

$$K_{e_{2,16}} = \frac{1}{35} \cdot \left(\frac{2 \cdot a_1}{\alpha^2} - \frac{13 \cdot \alpha \cdot b_1}{3} \right) + \frac{2 \cdot a_1}{75} = -0.024$$

$$K_{e_{4,14}} = -K_{e_{2,16}}, K_{e_{6,12}} = K_{e_{2,16}}, K_{e_{8,10}} = -K_{e_{2,16}}$$

$$K_{e_{3,3}} = \frac{4}{35} \cdot \left(\alpha^3 + \frac{13}{\alpha} \right) + \frac{8 \cdot \alpha}{25} = 1.92$$

$$K_{e_{7,7}} = K_{e_{3,3}}, K_{e_{11,11}} = K_{e_{3,3}}, K_{e_{15,15}} = K_{e_{3,3}}$$

$$K_{e_{3,4}} = \frac{2}{35} \cdot \left(\alpha^2 \cdot b_1 + \frac{11 \cdot a_1}{3 \cdot \alpha} \right) + \frac{2 \cdot b_1 \cdot (5 \cdot v + 1)}{75} = 0.192$$

$$K_{e_{7,8}} = -K_{e_{3,4}}, K_{e_{11,12}} = -K_{e_{3,4}}, K_{e_{15,16}} = K_{e_{3,4}}$$

$$K_{e_{3,7}} = \frac{2}{35} \cdot \left(-2 \cdot \alpha^3 + \frac{9}{\alpha} \right) - \frac{8 \cdot \alpha}{25} = 0.08$$

$$K_{e_{11,15}} = K_{e_{3,7}}$$

$$K_{e_{3,8}} = \frac{1}{35} \cdot \left(2 \cdot \alpha^2 \cdot b_1 - \frac{13 \cdot a_1}{3 \cdot \alpha} \right) + \frac{2 \cdot b_1}{75} = -0.024$$

$$K_{e_{4,7}} = -K_{e_{3,8}}, K_{e_{11,16}} = -K_{e_{3,8}}, K_{e_{12,15}} = K_{e_{3,8}}$$

$$K_{e_{3,11}} = \frac{3}{35} \cdot \left(\alpha^3 + \frac{3}{\alpha} \right) + \frac{2 \cdot \alpha}{25} = 0.423$$

$$K_{e_{7,15}} = K_{e_{3,11}}$$

$$K_{e_{3,12}} = -\frac{1}{70} \cdot \left(3 \cdot \alpha^2 \cdot b_1 + \frac{13 \cdot a_1}{3 \cdot \alpha} \right) - \frac{b_1}{150} = -0.0669$$

$$K_{e_{4,11}} = -K_{e_{3,12}}, K_{e_{7,16}} = -K_{e_{3,12}}, K_{e_{8,15}} = K_{e_{3,12}}$$

$$K_{e_{3,15}} = \frac{1}{35} \cdot \left(\frac{26}{\alpha} - 3 \cdot \alpha^3 \right) - \frac{2 \cdot \alpha}{25} = 0.577$$

$$K_{e7,11} = K_{e3,15}$$

Element stiffness matrix coefficients (above the main diagonal only)

$$K_e = D_{1,1} \cdot K_e = \begin{bmatrix} 796.3 & 135.19 & 135.19 & 16.74 & -391.2 & 84.95 & -13.66 & 4.1 & -13.89 & 36.57 & \dots & 4.1 \\ 0 & 46.67 & 21.6 & 4.67 & -84.95 & 14.03 & -4.1 & 0.708 & -36.57 & 10.28 & \dots & -0.583 \\ 0 & 0 & 46.67 & 4.67 & -13.66 & 4.1 & 1.94 & -0.583 & -36.57 & 8.54 & \dots & 0.708 \\ 0 & 0 & 0 & 0.978 & -4.1 & 0.708 & 0.583 & -0.161 & -8.54 & 1.63 & \dots & -0.161 \\ 0 & 0 & 0 & 0 & 796.3 & -135.19 & 135.19 & -16.74 & -391.2 & 13.66 & \dots & 8.54 \\ 0 & 0 & 0 & 0 & 0 & 46.67 & -21.6 & 4.67 & 13.66 & 1.94 & \dots & -1.63 \\ 0 & 0 & 0 & 0 & 0 & 0 & 46.67 & -4.67 & -84.95 & 4.1 & \dots & 1.63 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0.978 & 4.1 & 0.583 & \dots & -0.231 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 796.3 & -135.19 & \dots & 4.1 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 46.67 & \dots & -0.708 \\ \vdots & \vdots & \vdots & \vdots & \vdots & \vdots & \vdots & \vdots & \vdots & \vdots & \ddots & \vdots \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & \dots & 0.978 \end{bmatrix}$$

Element load vector

$$\vec{F}_e = \frac{q \cdot A_1}{24} \cdot \left[6; a_1; b_1; \frac{A_1}{6}; 6; -a_1; b_1; \frac{-A_1}{6}; 6; -a_1; -b_1; \frac{A_1}{6}; 6; a_1; -b_1; \frac{-A_1}{6} \right] =$$

$$[0.9 \ 0.09 \ 0.09 \ 0.009 \ 0.9 \ -0.09 \ 0.09 \ -0.009 \ 0.9 \ -0.09 \ \dots \ -0.009] \text{ kN}$$

The obtained element stiffness matrix and load vector are identical to the numerical formulation.

[1] Bogner, F. K., Fox, R. L., and Schmit, L. A. The generation of interelement compatible stiffness and mass matrices by the use of interpolation formulae, *Proceedings of the Conference on Matrix Methods in Structural Mechanics*, 397–444, 1965