

Design References

- LRFD** AASHTO LRFD Bridge Design Specifications, 9th Edition
MAST 2 Lateral Stability of Long Prestressed Concrete Beams - Part 2, PCI Journal, Jan-Feb 1993

LIFTING FROM BED - VERTICAL CABLES

Concrete Properties

Concrete Compressive Strength, $f_{c.lift1} = 5.50$ ksi
 Unit Weight of Unreinforced Girder Concrete, $w_c = 0.150$ kcf
 Correction Factor for Modulus of Elasticity, $K_1 = 1.00$
 Concrete Density Modification Factor, $\lambda = 1.00$
 Concrete Modulus of Elasticity, $E_{c.lift1}$: **LRFD Eq. 5.4.2.4-1**

$$\begin{aligned} E_{c.lift1} &= 120000 K_1 w_c^2 f_{c.lift1}^{0.33} \\ &= 120000(1.00) (0.150 \text{ kcf})^2 (5.50 \text{ ksi})^{0.33} \\ &= 4739 \text{ ksi} \end{aligned}$$

Modulus of Rupture, $f_{r.lift1}$: **LRFD 5.4.2.6**

$$\begin{aligned} f_{r.lift1} &= -0.24 \lambda \sqrt{f_{c.lift1}} \\ &= -0.24(1.00) \sqrt{5.50 \text{ ksi}} \\ &= -0.563 \text{ ksi} \end{aligned}$$

Prestress Force

Effective Prestress Force at Lifting, $P_{eff.lift1} = 1232$ kips
 CG of Strands at Midspan to Bottom of Girder, $y_{cgs.mid.lift1} = 5.000$ in
 Camber, $\Delta_{camb.lift1} = 2.920$ in

Other Configuration Parameters

Lift Connection Locations from End of Girder, $a_{lift1} = 9.00$ ft
 Rigid Extension Lift Connection above Top of Girder, $y_{lift} = 0.000$ in
 Lift Connection Tolerance from Centerline of Girder, $e_{conn} = 0.935$ in

Other Loading Parameters

Lateral Wind Force, $w_{wind.lift1} = 0.000$ klf
 Vertical Wind Uplift Considered Negligible for Lifting
 Lifting Impact Factor, $IM_{lift1} = 0\% = 0.00$
 Unit Weight of Girder, $w_{DC.girde} = 0.876$ klf
 Effective Unit Weight of Girder, $w_{DC.girder.lift1}$:

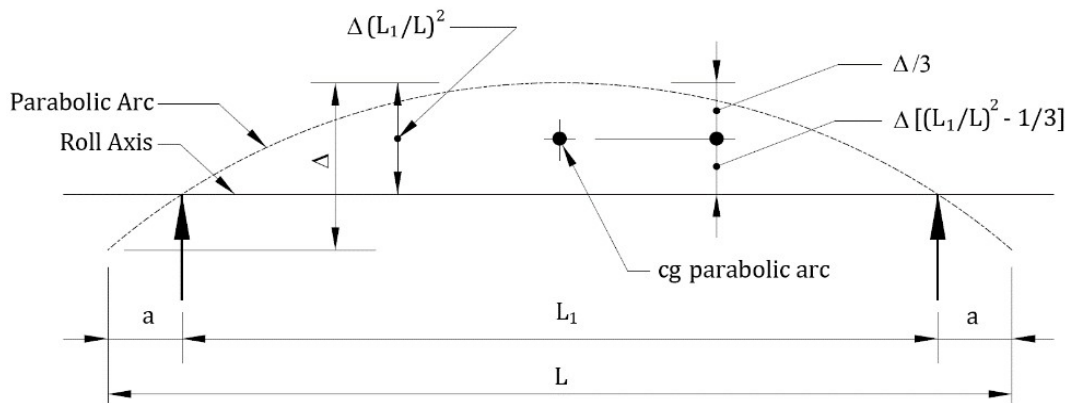
$$\begin{aligned} w_{DC.girder.lift1} &= w_{DC.girder} (1 + IM_{lift1}) \\ &= (0.876 \text{ klf})(1 + 0.00) \\ &= 0.876 \text{ klf} \end{aligned}$$

Girder Eccentricities

Total Lateral Deflection over Girder Length, $e_{i.total.lift}$ 0.850 in (Assumed Parabolic)

Center of Mass Eccentricity Reduction Factor, $offset_{lift1}$:

$$\begin{aligned} L_{1.lift1} &= L_{girder} - 2 a_{lift1} \\ &= 136.00 \text{ ft} - 2(9.00 \text{ ft}) \\ &= 118.00 \text{ ft} \end{aligned}$$



Offset to centroid of a parabolically deflected girder from roll axis (used horizontally and vertically)

based on MAST 2 Figure B1, with generic variables

$$\begin{aligned} offset_{lift1} &= \left(\frac{L_{1.lift1}}{L_{girder}} \right)^2 - \frac{1}{3} \\ &= \left(\frac{118.00 \text{ ft}}{136.00 \text{ ft}} \right)^2 - \frac{1}{3} \\ &= 0.419 \end{aligned}$$

Center of Mass Eccentricity Due to Lateral Deflection, $e_{i.lift1}$:

$$\begin{aligned} e_{i.lift1} &= e_{i.total.lift1} offset_{lift1} \\ &= (0.850 \text{ in})(0.419) \\ &= 0.357 \text{ in} \end{aligned}$$

Distance from the Center of Mass of the Cambered Girder below Roll Axis, $y_{r.lift1}$:

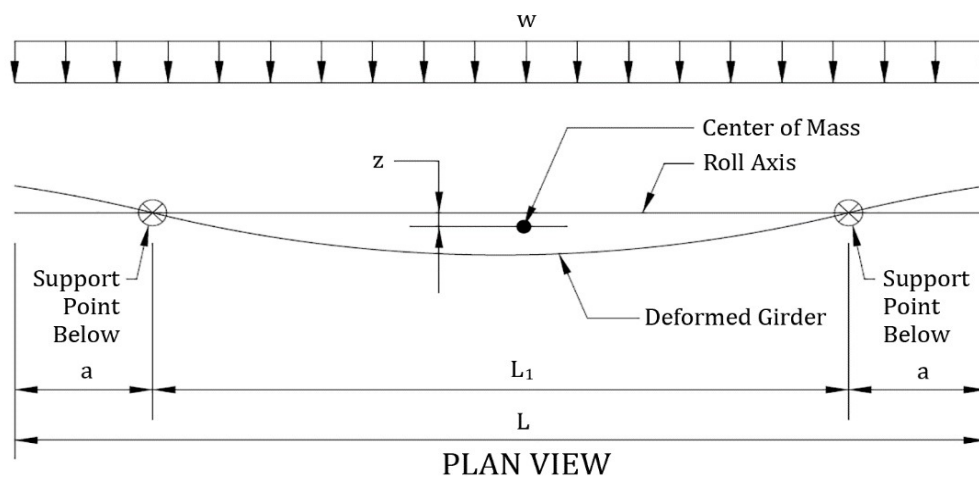
$$\begin{aligned} y_{r.lift1} &= y_t - offset_{lift1} \Delta_{camb.lift1} + y_{lift} \\ &= 35.400 \text{ in} - 0.419(2.920 \text{ in}) + 0.000 \text{ in} \\ &= 34.175 \text{ in} \end{aligned}$$

Center of Mass Eccentricity Due to Wind Deflection, $z_{wind.lift1}$:

$$z_{wind.lift1} = \frac{w_{wind.lift1}}{12 E_{c.lift1} I_y L_{girder}} \left(\frac{L_{1.lift1}^5}{10} - a_{lift1}^2 L_{1.lift1}^3 + 3 a_{lift1}^4 L_{1.lift1} + \frac{6}{5} a_{lift1}^5 \right)$$

$$= \left[\frac{(0.000 \text{ klf})(12 \text{ in/ft})^3}{(12)(4,739 \text{ ksi})(37,634 \text{ in}^4)(136.00 \text{ ft})} \right] \left[\frac{(118.00 \text{ ft})^5}{10} - (9.00 \text{ ft})^2(118.00 \text{ ft})^3 + 3(9.00 \text{ ft})^4(118.00 \text{ ft}) + \frac{6(9.00 \text{ ft})^5}{5} \right]$$

$$= 0.000 \text{ in}$$



Mid-Height of the Cambered Arc Below Roll Axis, $y_{w.lift1}$:

$$y_{w.lift1} = \frac{h_{girder}}{2} + y_{lift} - offset_{lift1} \Delta_{camber.lift1}$$

$$= \frac{72.000 \text{ in} + 0.000 \text{ in} - 0.419(2.920 \text{ in})}{2}$$

$$= 34.775 \text{ in}$$

Center of Mass Eccentricity Due to Girder Weight on Weak Axis, $z_{0.lift1}$:

$$z_{0.lift1} = \frac{w_{DC.girder.lift1}}{12 E_{c.lift1} I_y L_{girder}} \left(\frac{L_{1.lift1}^5}{10} - a_{lift1}^2 L_{1.lift1}^3 + 3 a_{lift1}^4 L_{1.lift1} + \frac{6}{5} a_{lift1}^5 \right)$$

$$= \left[\frac{(0.876 \text{ klf})(12 \text{ in/ft})^3}{(12)(4,739 \text{ ksi})(37,634 \text{ in}^4)(136.00 \text{ ft})} \right] \left[\frac{(118.00 \text{ ft})^5}{10} - (9.00 \text{ ft})^2(118.00 \text{ ft})^3 + 3(9.00 \text{ ft})^4(118.00 \text{ ft}) + \frac{6(9.00 \text{ ft})^5}{5} \right]$$

$$= 11.218 \text{ in}$$

Eccentricity of Girder Dead Load to Equilibrate Wind Load, $e_{wind.lift1}$:

$$e_{wind.lift1} = \frac{w_{wind.lift1} y_{w.lift1}}{w_{DC.girder.lift1}}$$

$$= \frac{(0.000 \text{ klf})(34.775 \text{ in})}{0.876 \text{ klf}} = 0.000 \text{ in}$$

Check Stresses

Moment Due to Gravity Load, $M_{g.lift1}$ at $x = a_{harp}$ from girder end:

$$M_{g.lift1} = \frac{w_{DC.girder.lift1} L_{girder}}{2} (x - a_{lift1}) - \frac{w_{DC.girder.lift1} x^2}{2}$$

$$= \frac{(0.876 \text{ klf})(136.00 \text{ ft})}{2} (54.40 \text{ ft} - 9.00 \text{ ft}) - \frac{(0.876 \text{ klf})(54.40 \text{ ft})^2}{2}$$

$$= 1408 \text{ kip-ft} = 16898 \text{ kip-in}$$

Lateral Moment Due to Wind, $M_{wind.lift1}$ at $x = a_{harp}$ from girder end:

$$M_{wind.lift1} = \frac{w_{wind.lift1} L_{girder}}{2} (x - a_{lift1}) - \frac{w_{wind.lift1} x^2}{2}$$

$$= \frac{(0.000 \text{ klf})(136.00 \text{ ft})}{2} (54.40 \text{ ft} - 9.00 \text{ ft}) - \frac{(0.000 \text{ klf})(54.40 \text{ ft})^2}{2}$$

$$= 0.0 \text{ kip-ft} = 0 \text{ kip-in}$$

Base Concrete Stresses in Girder before Rotation and Wind

$$f_{t.lift1.base} = P_{eff.lift1} \left(\frac{1}{A_{girder}} - \frac{y_b - y_{cgs.mid.lift1}}{S_{x,t}} \right) + \frac{M_{g.lift1}}{S_{x,t}}$$

$$= 1,232 \text{ kips} \left[\frac{1}{767.0 \text{ in}^2} - \frac{36.600 \text{ in} - 5.000 \text{ in}}{15,421 \text{ in}^3} \right] + \frac{16,898 \text{ kip-in}}{15,421 \text{ in}^3}$$

$$= 0.177 \text{ ksi}$$

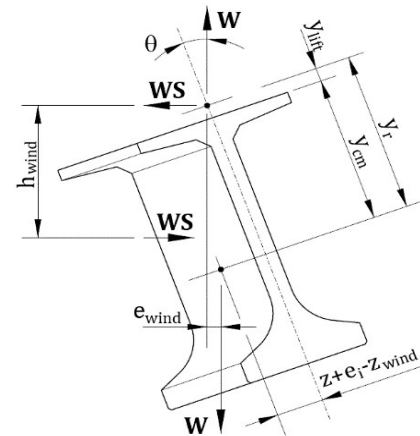
$$f_{b.lift1.base} = P_{eff.lift1} \left(\frac{1}{A_{girder}} + \frac{y_b - y_{cgs.mid.lift1}}{S_{x,b}} \right) - \frac{M_{g.lift1}}{S_{x,b}}$$

$$= 1,232 \text{ kips} \left[\frac{1}{767.0 \text{ in}^2} + \frac{36.600 \text{ in} - 5.000 \text{ in}}{14,915 \text{ in}^3} \right] - \frac{16,898 \text{ kip-in}}{14,915 \text{ in}^3}$$

$$= 3.084 \text{ ksi}$$

Check Compressive and Tensile Stresses at Equilibrium Rotation - Wind Right

$$\begin{aligned}\theta_{eq.lift1.wr} &= \frac{(e_{i.lift1} + e_{conn} - z_{wind.lift1} + e_{wind.lift1})}{y_r.lift1 - z_{0.lift1}} \\ &= \frac{(0.357 \text{ in} + 0.935 \text{ in} - 0.000 \text{ in} + 0.000 \text{ in})}{34.175 \text{ in} - 11.218 \text{ in}} \\ &= 0.05626 \text{ rad}\end{aligned}$$



**Girder Free Body
Diagram, Wind Right**

Check of Bottom Flange Tips - Wind Right

$$\begin{aligned}f_{b.lift1.wr.left} &= f_{b.lift1.base} - \frac{M_{g.lift1} \theta_{eq.lift1.wr}}{S_{y.b}} + \frac{M_{wind.lift1}}{S_{y.b}} \\ &= 3.084 \text{ ksi} - \frac{(16,898 \text{ kip-in})(0.05626 \text{ rad})}{2,895 \text{ in}^3} + \frac{0 \text{ kip-in}}{2,895 \text{ in}^3} \\ &= 2.755 \text{ ksi}\end{aligned}$$

$$\begin{aligned}f_{b.lift1.wr.right} &= f_{b.lift1.base} + \frac{M_{g.lift1} \theta_{eq.lift1.wr}}{S_{y.b}} - \frac{M_{wind.lift1}}{S_{y.b}} \\ &= 3.084 \text{ ksi} + \frac{(16,898 \text{ kip-in})(0.05626 \text{ rad})}{2,895 \text{ in}^3} - \frac{0 \text{ kip-in}}{2,895 \text{ in}^3} \\ &= 3.412 \text{ ksi}\end{aligned}$$

$$\begin{aligned}f_{b.ck.lift1.wr} &= \text{Max}(f_{b.lift1.wr.left}, f_{b.lift1.wr.right}) \\ &= \text{Max}(2.755 \text{ ksi}, 3.412 \text{ ksi}) = 3.412 \text{ ksi}\end{aligned}$$

Check of Top Flange Tips - Wind Right

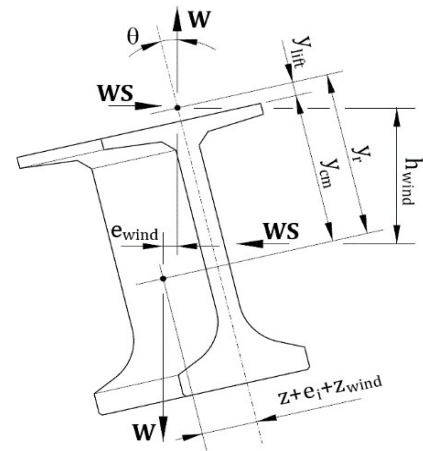
$$\begin{aligned}f_{t.lift1.wr.left} &= f_{t.lift1.base} - \frac{M_{g.lift1} \theta_{eq.lift1.wr}}{S_{y.t}} + \frac{M_{wind.lift1}}{S_{y.t}} \\ &= 0.177 \text{ ksi} - \frac{(16,898 \text{ kip-in})(0.05626 \text{ rad})}{1,792 \text{ in}^3} + \frac{0 \text{ kip-in}}{1,792 \text{ in}^3} \\ &= -0.353 \text{ ksi}\end{aligned}$$

$$\begin{aligned}f_{t.lift1.wr.right} &= f_{t.lift1.base} + \frac{M_{g.lift1} \theta_{eq.lift1.wr}}{S_{y.t}} - \frac{M_{wind.lift1}}{S_{y.t}} \\ &= 0.177 \text{ ksi} + \frac{(16,898 \text{ kip-in})(0.05626 \text{ rad})}{1,792 \text{ in}^3} - \frac{0 \text{ kip-in}}{1,792 \text{ in}^3} \\ &= 0.708 \text{ ksi}\end{aligned}$$

$$\begin{aligned}f_{t.ck.lift1.wr} &= \text{Min}(f_{t.lift1.wr.left}, f_{t.lift1.wr.right}) \\ &= \text{Min}(-0.353 \text{ ksi}, 0.708 \text{ ksi}) = -0.353 \text{ ksi}\end{aligned}$$

Check Compressive and Tensile Stress esat Equilibrium Rotation - Wind Left

$$\begin{aligned}\theta_{eq.lift1.wl} &= \frac{(e_{i.lift1} + e_{conn} + z_{wind.lift1} - e_{wind.lift1})}{y_{r.lift1} - z_{0.lift1}} \\ &= \frac{(0.357 \text{ in} + 0.935 \text{ in} + 0.000 \text{ in} - 0.000 \text{ in})}{34.175 \text{ in} - 11.218 \text{ in}} \\ &= 0.05626 \text{ rad}\end{aligned}$$



**Girder Free Body
Diagram, Wind Left**

Check of Bottom Flange Tips - Wind Left

$$\begin{aligned}f_{b.lift1.wl.left} &= f_{b.lift1.base} - \frac{M_{g.lift1} \theta_{eq.lift1.wl}}{S_{y.b}} - \frac{M_{wind.lift1}}{S_{y.b}} \\ &= 3.084 \text{ ksi} - \frac{(16,898 \text{ kip-in})(0.05626 \text{ rad})}{2,895 \text{ in}^3} - \frac{0 \text{ kip-in}}{2,895 \text{ in}^3} \\ &= 2.755 \text{ ksi}\end{aligned}$$

$$\begin{aligned}f_{b.lift1.wl.right} &= f_{b.lift1.base} + \frac{M_{g.lift1} \theta_{eq.lift1.wl}}{S_{y.b}} + \frac{M_{wind.lift1}}{S_{y.b}} \\ &= 3.084 \text{ ksi} + \frac{(16,898 \text{ kip-in})(0.05626 \text{ rad})}{2,895 \text{ in}^3} + \frac{0 \text{ kip-in}}{2,895 \text{ in}^3} \\ &= 3.412 \text{ ksi}\end{aligned}$$

$$\begin{aligned}f_{b.ck.lift1.wl} &= \text{Max}(f_{b.lift1.wl.left}, f_{b.lift1.wl.right}) \\ &= \text{Max}(2.755 \text{ ksi}, 3.412 \text{ ksi}) = 3.412 \text{ ksi}\end{aligned}$$

Check of Top Flange Tips - Wind Left

$$\begin{aligned}f_{t.lift1.wl.left} &= f_{t.lift1.base} - \frac{M_{g.lift1} \theta_{eq.lift1.wl}}{S_{y.t}} - \frac{M_{wind.lift1}}{S_{y.t}} \\ &= 0.177 \text{ ksi} - \frac{(16,898 \text{ kip-in})(0.05626 \text{ rad})}{1,792 \text{ in}^3} - \frac{0 \text{ kip-in}}{1,792 \text{ in}^3} \\ &= -0.353 \text{ ksi}\end{aligned}$$

$$\begin{aligned}f_{t.lift1.wl.right} &= f_{t.lift1.base} + \frac{M_{g.lift1} \theta_{eq.lift1.wl}}{S_{y.t}} + \frac{M_{wind.lift1}}{S_{y.t}} \\ &= 0.177 \text{ ksi} + \frac{(16,898 \text{ kip-in})(0.05626 \text{ rad})}{1,792 \text{ in}^3} + \frac{0 \text{ kip-in}}{1,792 \text{ in}^3} \\ &= 0.708 \text{ ksi}\end{aligned}$$

$$\begin{aligned}f_{t.ck.lift1.wl} &= \text{Min}(f_{t.lift1.wl.left}, f_{t.lift1.wl.right}) \\ &= \text{Min}(-0.353 \text{ ksi}, 0.708 \text{ ksi}) = -0.353 \text{ ksi}\end{aligned}$$

	Critical	Allowable	LRFD 5.9.2.3.1a	
Compressive Stresses	3.412 ksi	$0.65 f_{c.lift1}$	$= 0.65(5.50 \text{ ksi}) =$	3.58 ksi OK
Tensile Stresses	-0.353 ksi	$f_{r.lift1} =$	-0.563 ksi	OK

Check Factor of Safety Against Cracking

Analysis assumes e_{lift1} is due to form misalignment and not eccentric prestressing.

Base Concrete Stresses in Girder with Wind (Left Top Tip)

$$\begin{aligned}
 f_{t.lift1.wr} &= f_{t.lift1.base} + \frac{M_{wind.lift1}}{S_{y,t}} \\
 &= 0.177 \text{ ksi} + \frac{0 \text{ kip-in}}{1,792 \text{ in}^3} \\
 &= 0.177 \text{ ksi}
 \end{aligned}$$

$$\begin{aligned}
 f_{t.lift1.wl} &= f_{t.lift1.base} - \frac{M_{wind.lift1}}{S_{y,t}} \\
 &= 0.177 \text{ ksi} - \frac{0 \text{ kip-in}}{1,792 \text{ in}^3} \\
 &= 0.177 \text{ ksi}
 \end{aligned}$$

Lateral Moment to Cause Cracking:

Wind Right:

$$\begin{aligned}
 M_{lat.lift1.wr} &= (f_{t.lift1.wr} - f_{r.lift1})S_{y,t} \\
 &= (0.177 \text{ ksi} - (-0.563 \text{ ksi}))(1,792 \text{ in}^3 / 12 \text{ in/ft}) \\
 &= 110.6 \text{ kip-ft}
 \end{aligned}$$

Wind Left:

$$\begin{aligned}
 M_{lat.lift1.wl} &= (f_{t.lift1.wl} - f_{r.lift1})S_{y,t} \\
 &= (0.177 \text{ ksi} - (-0.563 \text{ ksi}))(1,792 \text{ in}^3 / 12 \text{ in/ft}) \\
 &= 110.6 \text{ kip-ft}
 \end{aligned}$$

Tilt Angle at Cracking due to Lateral Deflection:

Wind Right:

$$\begin{aligned}
 \theta_{cr.lift1.wr} &= \frac{M_{lat.lift1.wr}}{M_{g.lift1}} \\
 &= \frac{110.6 \text{ kip-ft}}{1,408 \text{ kip-ft}} = 0.07851 \text{ rad}
 \end{aligned}$$

Wind Left:

$$\begin{aligned}
 \theta_{cr.lift1.wl} &= \frac{M_{lat.lift1.wl}}{M_{g.lift1}} \\
 &= \frac{110.6 \text{ kip-ft}}{1,408 \text{ kip-ft}} = 0.07851 \text{ rad}
 \end{aligned}$$

Factor of Safety Against Cracking:

Wind Right:

$$\begin{aligned}
 FS_{cr.lift1.wr} &= \frac{y_{r.lift1} \theta_{cr.lift1.wr}}{z_{0.lift1} \theta_{cr.lift1.wr} + e_{conn} - z_{wind.lift1} + e_{wind.lift1} + e_{i.lift1}} \\
 &= \frac{(34.175 \text{ in})(0.07851 \text{ rad})}{(11.218 \text{ in})(0.07851 \text{ rad}) + 0.935 \text{ in} - 0.000 \text{ in} + 0.000 \text{ in} + 0.357 \text{ in}} \\
 &= 1.235
 \end{aligned}$$

Wind Left:

$$\begin{aligned}
 FS_{cr.lift1.wl} &= \frac{y_{r.lift1} \theta_{cr.lift1.wl}}{z_{0.lift1} \theta_{cr.lift1.wl} + e_{conn} + z_{wind.lift1} - e_{wind.lift1} + e_{i.lift1}} \\
 &= \frac{(34.175 \text{ in})(0.07851 \text{ rad})}{(11.218 \text{ in})(0.07851 \text{ rad}) + 0.935 \text{ in} + 0.000 \text{ in} - 0.000 \text{ in} + 0.357 \text{ in}} \\
 &= 1.235
 \end{aligned}$$

Critical Factor of Safety Against Cracking:

$$\begin{aligned}
 FS_{cr.lift1} &= \min(FS_{cr.lift1.wr}, FS_{cr.lift1.wl}) \\
 &= 1.235 \geq 1.000
 \end{aligned}$$

OK

Check Factor of Safety Against Failure - Wind Right

Rotation at Maximum Factor of Safety:

$$\begin{aligned}
 \theta_{max.ult.lift1.wr} &= \sqrt{\frac{e_{i.lift1} + e_{conn} - z_{wind.lift1} + e_{wind.lift1}}{2.5 z_{0.lift1}}} \\
 &= \sqrt{\frac{0.357 \text{ in} + 0.935 \text{ in} - 0.000 \text{ in} + 0.000 \text{ in}}{2.5(11.218 \text{ in})}} \\
 &= \sqrt{\frac{1.292 \text{ in}}{28.045 \text{ in}}} \quad \begin{array}{l} \text{Check for rotation of center of mass past vertical, defined with negative numerator.} \\ \text{If negative, wind left is critical case, set } FS_{ult.lift1.wr} \text{ to default value of 99.} \end{array} \\
 &= 0.21460 \text{ rad}
 \end{aligned}$$

Factor of Safety Against Failure:

$$\begin{aligned}
 FS_{ult.lift1.wr}(\theta) &= \frac{y_{r.lift1} \theta}{(z_{0.lift1} \theta - z_{wind.lift1})(1 + 2.5\theta) + e_{wind.lift1} + e_{conn} + e_{i.lift1}} \\
 &= \frac{(34.175 \text{ in})(0.21460 \text{ rad})}{((11.218 \text{ in})(0.21460 \text{ rad}) - 0.000 \text{ in})(1 + 2.5(0.21460 \text{ rad})) + 0.000 \text{ in} + 0.935 \text{ in} + 0.357 \text{ in}} \\
 &= 1.470
 \end{aligned}$$

Check Factor of Safety Against Failure - Wind Left

Rotation at Maximum Factor of Safety:

$$\begin{aligned}\theta_{max.ult.lift1.wl} &= \sqrt{\frac{e_{i.lift1} + e_{conn} + z_{wind.lift1} - e_{wind.lift1}}{2.5 z_{0.lift1}}} \\ &= \sqrt{\frac{0.357 \text{ in} + 0.935 \text{ in} + 0.000 \text{ in} - 0.000 \text{ in}}{2.5(11.218 \text{ in})}} \\ &= \sqrt{\frac{1.292 \text{ in}}{28.045 \text{ in}}} \quad \begin{array}{l} \text{Check for rotation of center of mass past vertical, defined with negative numerator.} \\ \text{If negative, wind right is critical case, set } FS_{ult.lift1.wr} \text{ to default value of 99.} \end{array} \\ &= 0.21460 \text{ rad}\end{aligned}$$

Factor of Safety Against Failure:

$$\begin{aligned}FS_{ult.lift1.wl}(\theta) &= \frac{y_{r.lift1} \theta}{(z_{0.lift1} \theta + z_{wind.lift1})(1 + 2.5\theta) - e_{wind.lift1} + e_{conn} + e_{i.lift1}} \\ &= \frac{(34.175 \text{ in})(0.21460 \text{ rad})}{((11.218 \text{ in})(0.21460 \text{ rad}) + 0.000 \text{ in})(1 + 2.5(0.21460 \text{ rad})) - 0.000 \text{ in} + 0.935 \text{ in} + 0.357 \text{ in}} \\ &= 1.470\end{aligned}$$

Critical Factor of Safety Against Failure:

$$\begin{aligned}FS_{ult.crit.lift1} &= \min(FS_{ult.lift1.wr}, FS_{ult.lift1.wl}) \\ &= 1.470\end{aligned}$$

Check Factor of Safety Against Failure for Critical Case

$$\begin{aligned}FS_{ult.lift1} &= \max(FS_{ult.crit.lift1}, FS_{cr.lift1}) \\ &= 1.470 < 1.500\end{aligned}$$

N.G.

Lateral Ultimate Moment Capacity Required, $M_{ult.y.lift1}$:

$$\begin{aligned}M_{ult.y.lift1} &= \frac{1.5}{FS_{ult.lift1}} (M_{g.lift1} \max(\theta_{max.ult.lift1.wr}, \theta_{max.ult.lift1.wl})) \\ &= \frac{1.5}{1.470} (1,408 \text{ kip-ft})(0.21460 \text{ rad}) \\ &= 308 \text{ kip-ft}\end{aligned}$$