

# Formalization of Requirements for CPS and NTSS

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## 1 Requirement Formalization

### 1.1 Autopilot

#### **R1.4.1:**

Textual:

Steady state roll commands shall be tracked within 1 degree.

Formalisation:

```
(forall tc in [ 1 , T ]:
(
  ( Tk(tc) != Tk(tc-1) ) implies
  ( exists t2 in [ tc , T ]:
  (
forall t in [ t2 , T ]:
(
  Phi(t) - PhiRef(t) <= 1
)
)
))));
```

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#### **R1.6:**

Textual:

The maximum roll angle (Phi) allowed shall be 30 deg +/-10%

Formalisation:

```
forall t in [ 0 , T ]:
(
  Phi(t) <= 33 and Phi(t) >= -33
)
);
```

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#### **R12.1:**

Textual:

When the autopilot is enabled, the aircraft altitude should reach the desired altitude within 500 seconds in calm air

Formalisation:

```
(forall tc in [ 1 , T ]:
(
  ( APEng(tc) = 1 ) implies
  ( exists t2 in [ tc , T ]:
  (
```

```

    forall t in [ t2 , T ]:
(
  alt(t) - ALTRef(t) >= 0
)
)
)
)
);

```

## 1.2 Finite State Machine

### R1:

Textual:

Exceeding sensor Limits shall latch an autopilot pullup when the pilot is not in control (not Standby) and the system is Supported without failures (not Apfail).

Formalisation:

```

(forall t in [ 0 , T ]:
(
(
  Standby(t) = 0 and Apfail(t) = 0 and Supported(t) = 1 and Limits(t) = 1
) implies
  pullup(t) = 1
)
);

```

## 1.3 Non Linear Guidance

### R6:

Textual:

The change in the magnitude of the output over one frame of execution with T sample period shall not exceed the quantity of the combined velocity of the target plus the velocity of the vehicle multiplied by T

Formalization:

```

(
  forall t in [ 1 , T ]:
(
(
  xtarg[1](t) = xtarg[1](t-1) and xtarg[2](t) = xtarg[2](t-1) and xtarg[3](t) = xtarg[3](t-1)
and yout[1](t) = Xap2[1](t) and yout[2](t) = Xap2[2](t) and yout[3](t) = Xap2[3](t)
and yout[1](t-1) = Xap2[1](t-1) and yout[2](t-1) = Xap2[2](t-1) and yout[3](t-1) = Xap2[3](t-1)
)
implies
(
  ( ( yout[1](t) - yout[1](t-1) ) * ( yout[1](t) - yout[1](t-1) ) ) +
  ( ( yout[2](t) - yout[2](t-1) ) * ( yout[2](t) - yout[2](t-1) ) ) +
  ( ( yout[3](t) - yout[3](t-1) ) * ( yout[3](t) - yout[3](t-1) ) )
)
)
)

```

```

    ) <= 0.0004 * (
        ( vv[1](t) + vt[1](t) ) * ( vv[1](t) + vt[1](t) ) +
        ( vv[2](t) + vt[2](t) ) * ( vv[2](t) + vt[2](t) ) +
        ( vv[3](t) + vt[3](t) ) * ( vv[3](t) + vt[3](t) )
    )
)
);

```

## 1.4 Regulator

**R7:**

Textual:

The Inner Loop Pitch Regulator Shall not command transient changes in angular roll acceleration greater than 50 deg/sec<sup>2</sup>/sec

Formalisation:

```

(
    forall t in [ 1 , T ]:
    (
        mcvdt_cmd_fcs_dps2(t) - mcvdt_cmd_fcs_dps2(t-1) <= 0.5
    )
);

```

## 1.5 Tustin

**R1:**

Textual:

When Reset is True and the Initial Condition (ic) is bounded by the provided Top and Bottom Limits (BL <= ic <= TL),  
the Output (yout) shall equal the Initial Condition (ic).

Formalisation:

```

(
    forall t in [ 0 , T ]:
    (
        ( Reset(t) = 1 and Ic(t) <= tl(t) and Ic(t) >= bl(t) implies yout(t) = Ic(t) ) and
        ( Reset(t) = 1 and Ic(t) <= tl(t) and tl(t) >= bl(t) implies yout(t) = tl(t) ) and
        ( Reset(t) = 1 and Ic(t) <= bl(t) and tl(t) >= bl(t) implies yout(t) = bl(t) ) and
        ( Reset(t) = 1 and Ic(t) <= bl(t) and tl(t) < bl(t) implies yout(t) = bl(t) )
    )
);

```

---

**R2:**

Textual:

The Output (yout) shall be bounded by the provided Top and Bottom limits (TL and BL)  
Formalisation:

```
(
  forall t in [ 0 , T ]:
  (
    ( tl(t) >= bl(t)  implies (  bl(t) <= yout(t)  and  yout(t) <= tl(t)  ) ) and
    ( tl(t) < bl(t)  implies (  tl(t) <= yout(t)  and  yout(t) <= bl(t)  ) )
  )
);
```

---

#### R4a:

Textual:

Over a 10 second computational duration at an execution frequency of 10 hz, the Output should equal the sine of time t, sin, where time is defined as a vector from 0 to 10 by increments of 0.1 seconds within a +/- 0.1 tolerance for an input equal to the cosine of time t, cos, with the sample delta time T = 0.1 seconds when in normal mode of operation.  
Formalisation:

```
(
  forall t in [ 0 , T ]:
  (
    (
      ( tl(t) >= bl(t)  and  ( yout(t) >= bl(t)  and  yout(t) <= tl(t) ) ) or
      ( bl(t) >= tl(t)  and  ( yout(t) >= tl(t)  and  yout(t) <= bl(t) ) )
    ) implies
    |yout(T) - 10 | <= 0.1
  )
);
```

---

#### R4b:

Textual:

Over a 10 second computational duration at an execution frequency of 10 hz, the Output should equal the sine of time t, sin, where time is defined as a vector from 0 to 10 by increments of 0.1 seconds within a +/- 0.1 tolerance for an input equal to the cosine of time t, cos, with the sample delta time T = 0.1 seconds when in normal mode of operation.  
Formalisation:

```
(
  forall t in [ 0 , T ]:
  (
    (
      ( tl(t) >= bl(t)  and  yout(t) >= bl(t)  and  yout(t) <= tl(t) ) or
      ( bl(t) >= tl(t)  and  yout(t) >= tl(t)  and  yout(t) <= bl(t) )
    ) implies
    |yout(t) - sinsig(t) | <= 0.1
  )
);
```

)  
);

## 1.6 NTSS

Textual: Good network connectivity should be maintained even when high traffic flows through different priority classes.

Formalization:

$$\mathcal{R} = \begin{cases} \overline{mos}_n & \text{if } \bigwedge_{i=1..n} \overline{mos}_i < \overline{mosTh}_i, \\ 1 + \overline{mos}_{n-1} & \text{if } \overline{mos}_n \geq \overline{mosTh}_n \wedge \bigwedge_{i=1..n-1} \overline{mos}_i < \overline{mosTh}_i, \\ 2 + \overline{mos}_{n-2} & \text{if } \bigwedge_{i \in \{n-1, n\}} \overline{mos}_i \geq \overline{mosTh}_i \wedge \bigwedge_{i \in \{1..n-2\}} \overline{mos}_i < \overline{mosTh}_i, \\ \dots & \dots \\ n & \text{if } \bigwedge_{i=1..n} \overline{mos}_i \geq \overline{mosTh}_i \end{cases}$$

## 2 Input Specification for CPS and NTSS

Table 1: Input Specification for CPS and NTSS

Subject: TU1...TU9		
Input Name	Type	Range
<b>Xin</b>	Double	[-20,20]
<b>TL</b>	Double	[-10,10]
<b>BL</b>	Double	[-10,10]
<b>IC</b>	Double	[-20,20]

Subject: NLG		
Input Name	Type	Range
$X_{targ}$	Double	[-100,100]
$X_v$	Double	[-100,100]
$V_v$	Double	[-100,100]
$V_{targ}$	Double	[-100,100]
<b>r</b>	Double	[0,100]

Subject: REG		
Input Name	Type	Range
beta_adc_deg	Double	[0,5]
vtas_adc_kts	Double	[0,5]
lev_md_fos_dps	Double	[0,5]
hdg_des_deg	Double	[0,5]
mev_emd_fes_dps	Double	[0,5]
alt_des_ft	Double	[0,5]
nev_cmd_fcs_dps	Double	[0,5]
xev_cmd_fcs_fps	Double	[0,5]
airspeed_des_fps	Double	[0,5]
hcv_cmd_fcs_fps	Double	[0,5]
lcv_fcs_dps	Double	[0,5]
mcv_fcs_dps	Double	[0,5]
ncv_fcs_dps	Double	[0,5]
dcv_fcs_fps	Double	[0,5]
zcv_cmd_fcs_fps	Double	[0,5]
betadot	Double	[0,5]

Subject: FSM		
Input Name	Type	Range
standby	Boolean	{0,1}
apfail	Boolean	{0,1}
supported	Boolean	{0,1}
limits	Boolean	{0,1}

Subject: AP1...AP3		
Input Name	Type	Range
AP Eng	Boolean	{0,1}
HDG Mode	Boolean	{0,1}
ALT Mode	Boolean	{0,1}
HDG Ref	Double	[-180,180]
Turn knob	Double	[0,45]
ALT Ref	Double	[0,1000]
Pitch wheel	Double	[-30,30]
Throttle	Double	[0,1]

Subject: NTSS		
Input Name	Type	Range
Class0	Integer	[0,thresh <sub>0</sub> ]
Class1	Integer	[0,thresh <sub>1</sub> ]
Class2	Integer	[0,thresh <sub>2</sub> ]
Class3	Integer	[0,thresh <sub>3</sub> ]
Class4	Integer	[0,thresh <sub>4</sub> ]
Class5	Integer	[0,thresh <sub>5</sub> ]
Class6	Integer	[0,thresh <sub>6</sub> ]
Class7	Integer	[0,thresh <sub>7</sub> ]