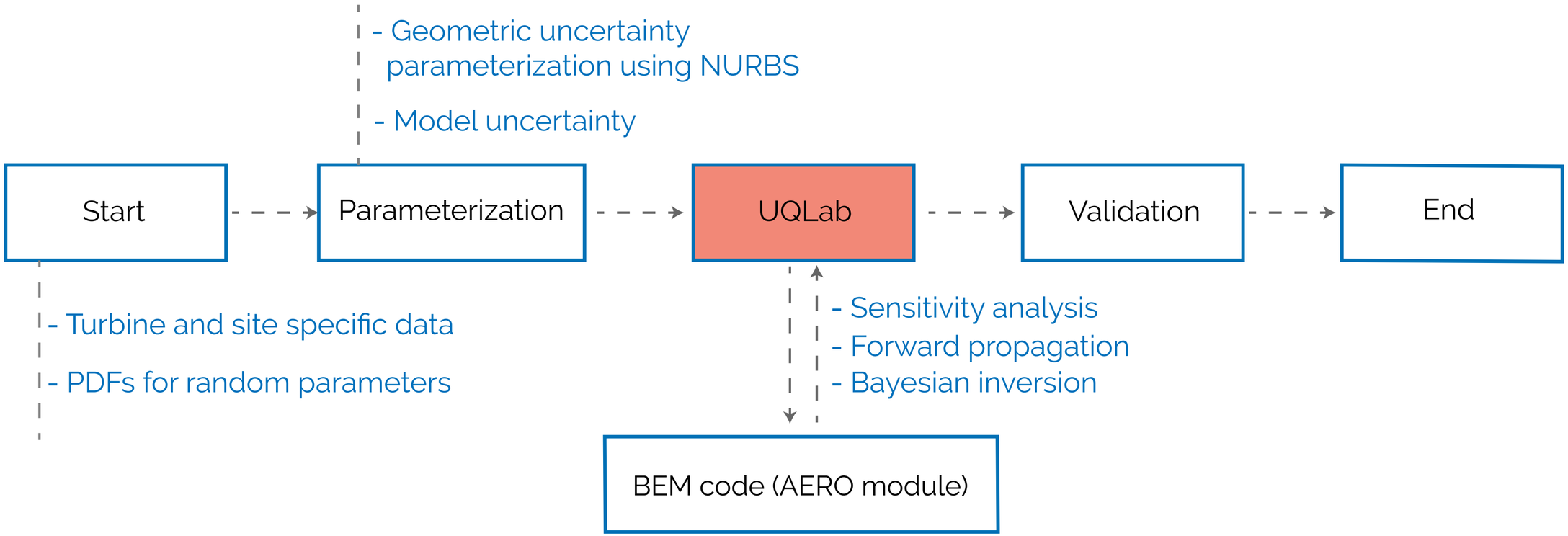
WindTrue: Sensitivity analysis applied to DANAERO wind turbine

Prashant Kumar
Benjamin Sanderse

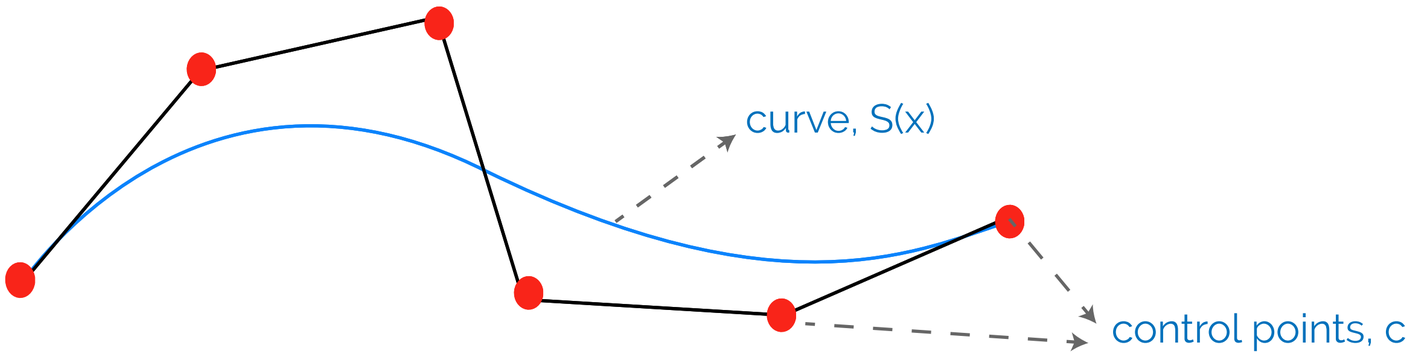
CWI

Workflow


- Geometric uncertainty
parameterization using NURBS
- Model uncertainty
Start
Parameterization
UQLab
Validation
End
- Turbine and site specific data
- PDFs for random parameters
- Sensitivity analysis
- Forward propagation
- Bayesian inversion
BEM code (AERO module)

NURBS based parametrization

Non-Uniform Rational Basis Spline (NURBS)



$$S(x) = \sum_{i=0}^{N-1} c_i B_{i,p}(x)$$

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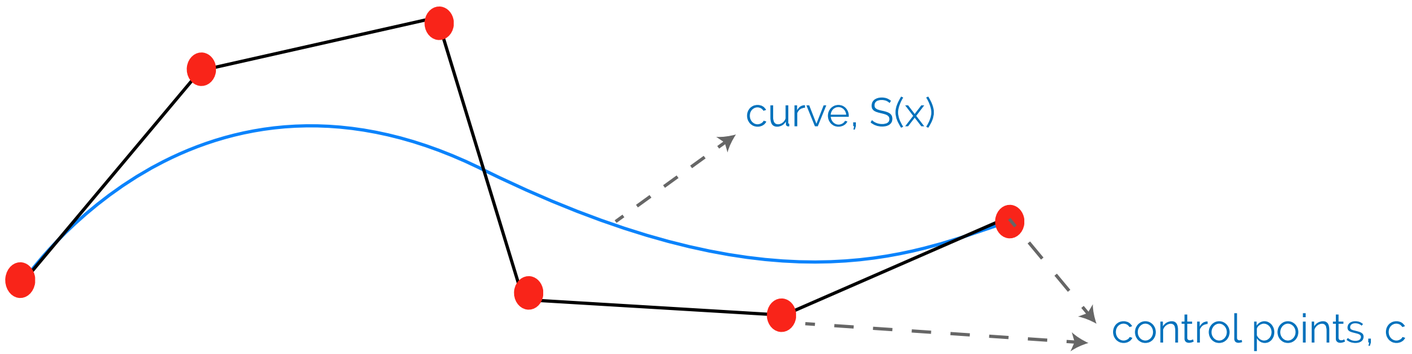

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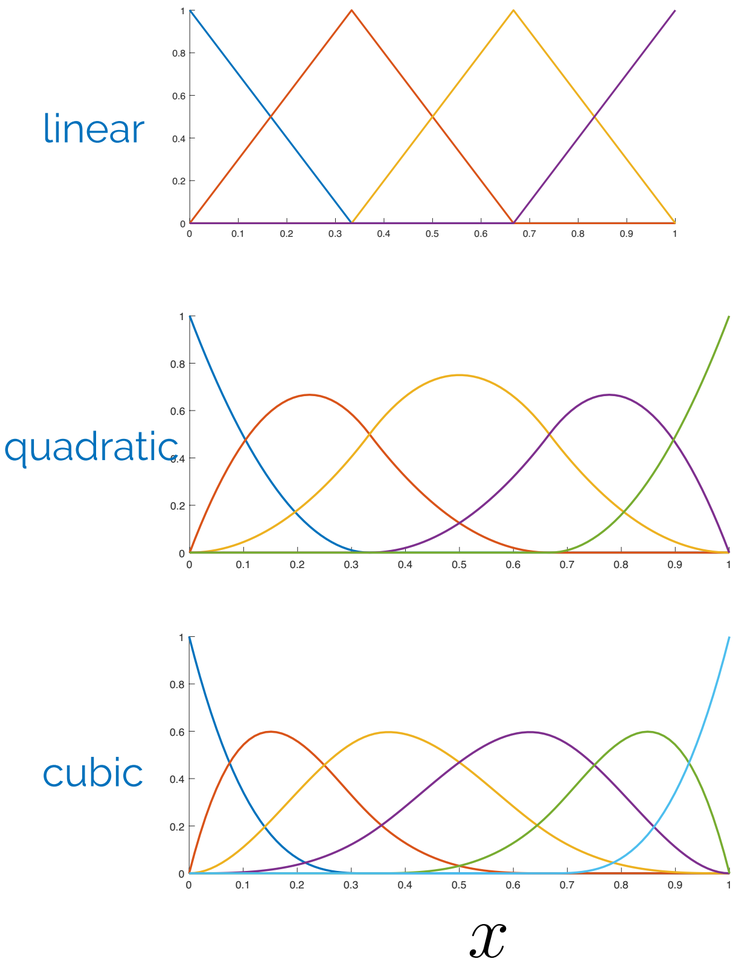
Basis functions, $B_{i,p}(x)$



Why NURBS?

-> Represent complex shapes with very few points

-> Flexibility to design a large variety of shapes

-> Easy to obtain high-order polynomials

Parameterization using NURBS

Goal: Obtain perturbed chord/twist from a given reference curves

Parameterization using NURBS

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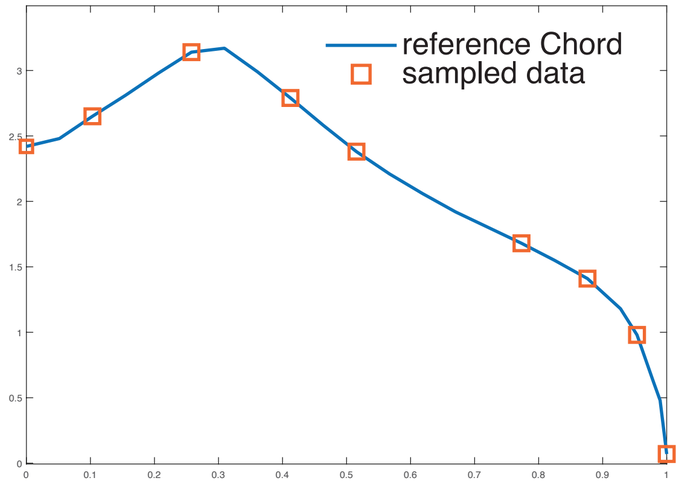
Step 1: Sample locations from the reference curve



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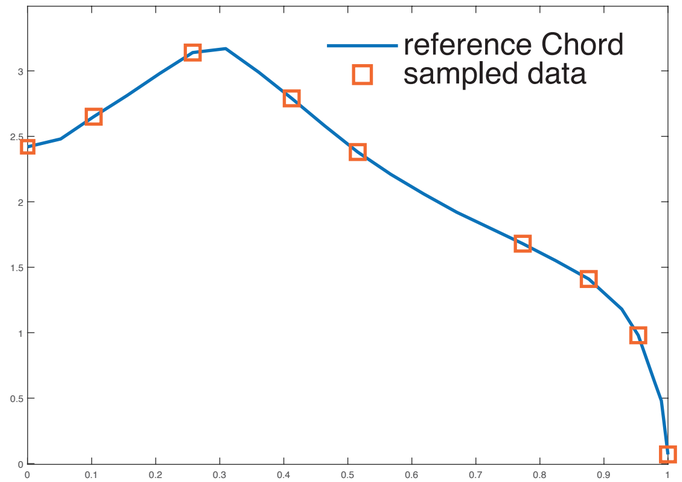

Step 2: Compute control points at sampled location via inversion

$$S(x) = \sum_{i=0}^{N-1} c_i B_{i,p}(x) \quad \Longrightarrow \quad \mathbf{Bc} = \mathbf{S}$$

$S(x)$: known; $B_{i,p}(x)$: known

Parameterization using NURBS

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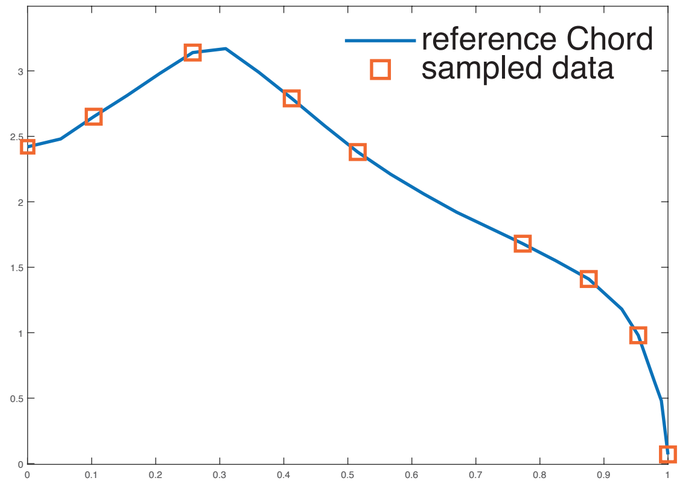

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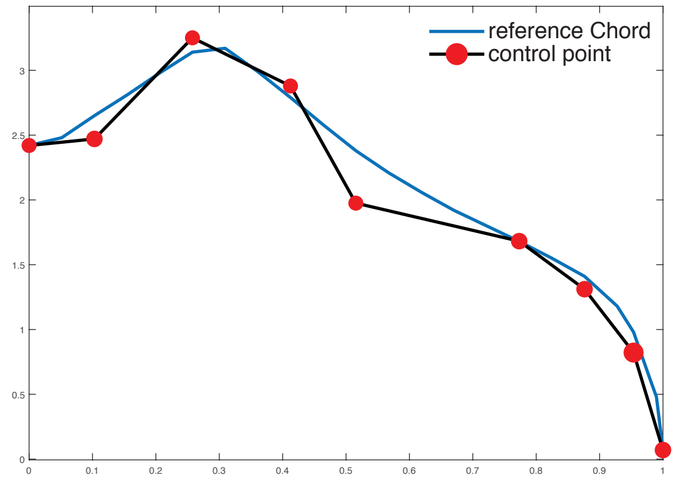
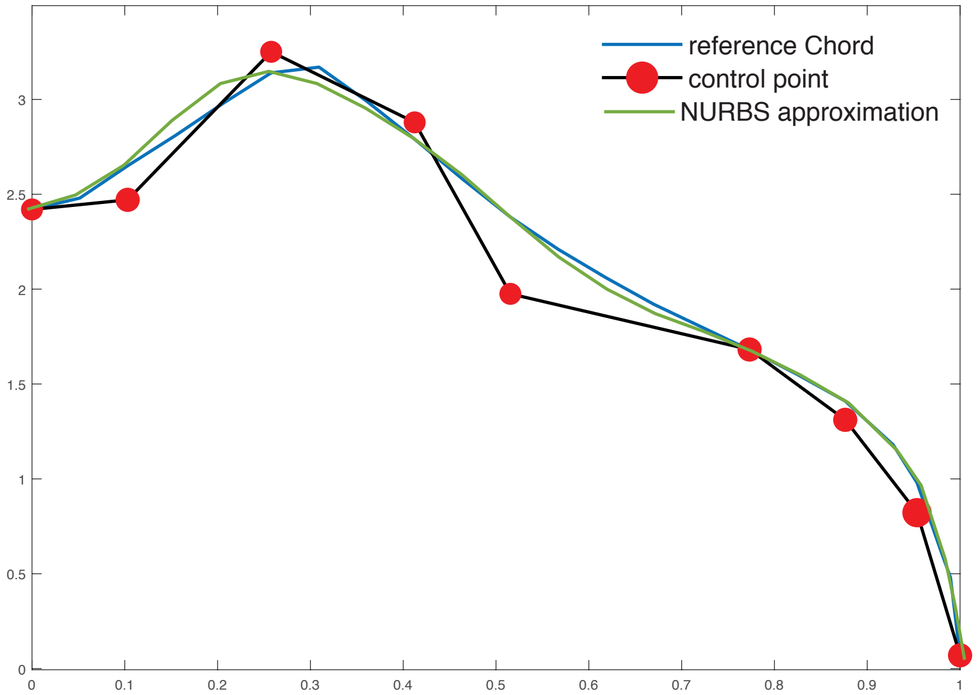
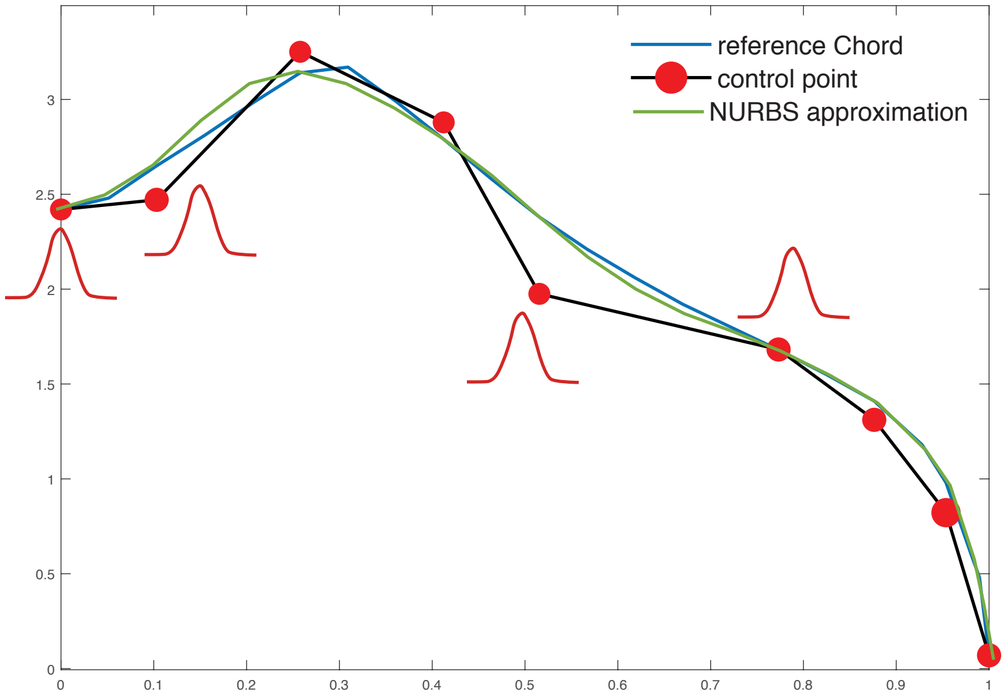


Parameterization using NURBS



Parameterization using NURBS

Step 3: Perturb control point values using some PDF



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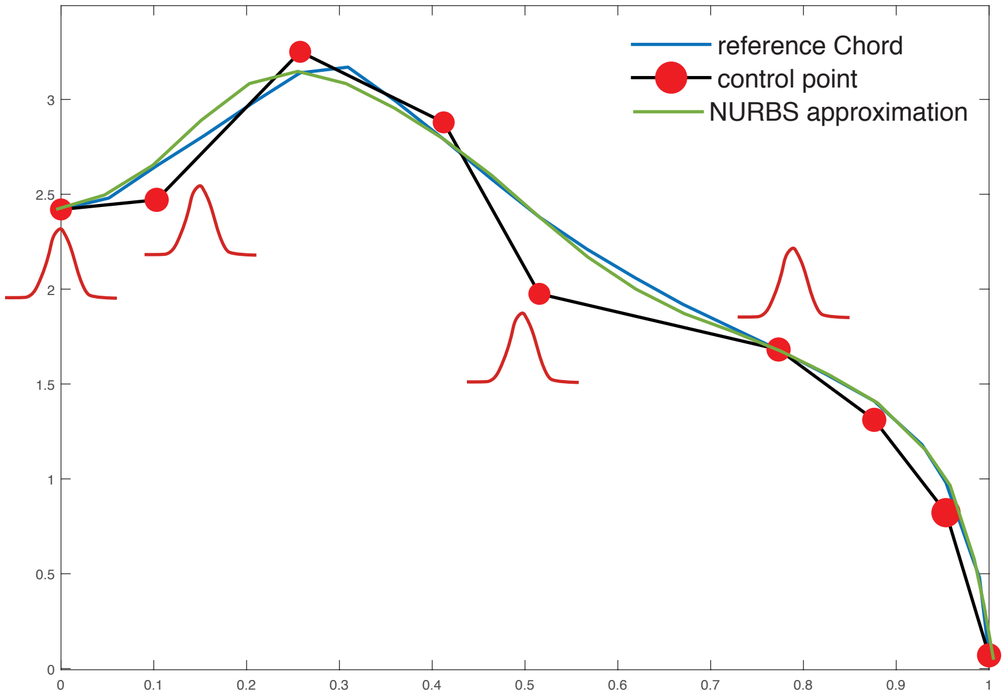
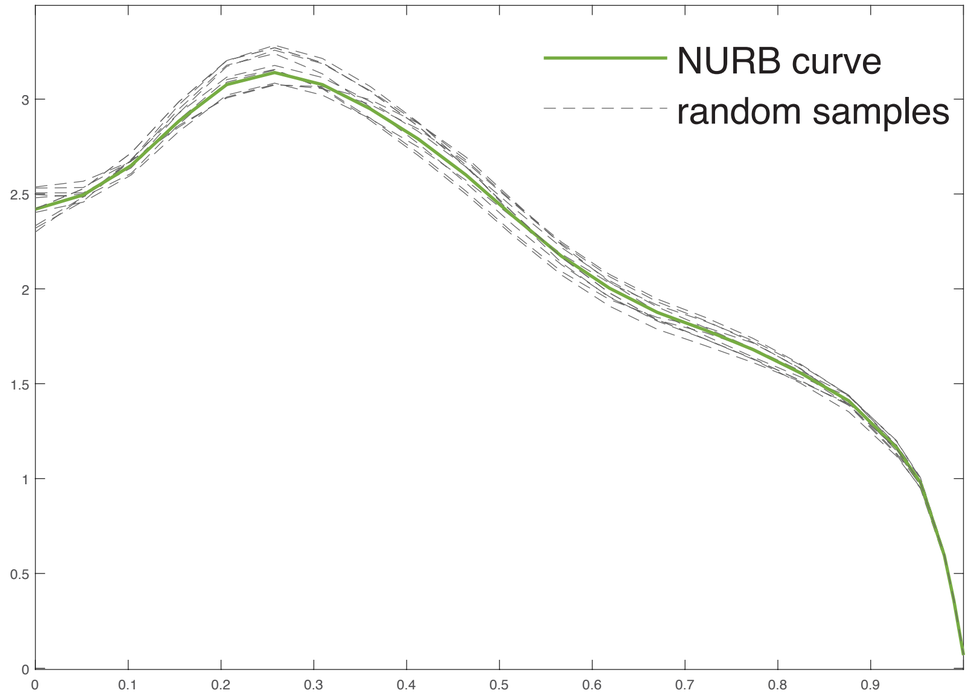


Step 4: Sample perturbed curves



Global sensitivity analysis

Global sensitivity analysis

-> Goal is to rank the uncertain parameters in the order of importance

-> Global approaches cover the uncertainty spaces more exhaustively

-> Better able to capture uncertainty in the model output

Sobol sensitivity indices

Main idea: Decompose the variance of model output in terms of contribution from individual input parameters and their combinations.

$$V(y) = \sum_i V_i + \sum_{i,j} V_{i,j} + \text{higher order terms}$$

First order indices

$$S_1 = \frac{V_1}{V}, S_2 = \frac{V_2}{V}, \ldots$$

Second order indices

$$S_{1,2} = \frac{V_{1,2}}{V}, S_{1,3} = \frac{V_{1,3}}{V}, \ldots$$

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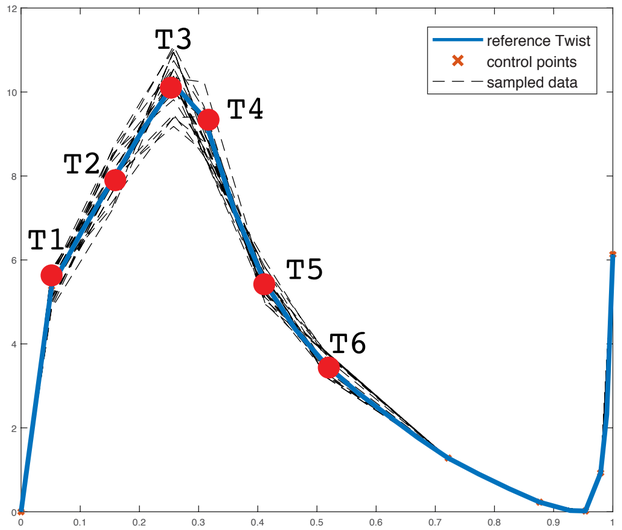
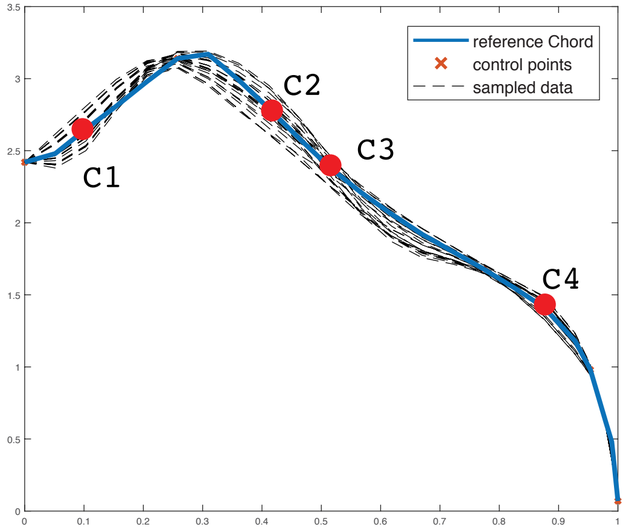
We use **adaptive Polynomial Choas Expansion** (Least angle regression) to compute variances

Sensitivity analysis results

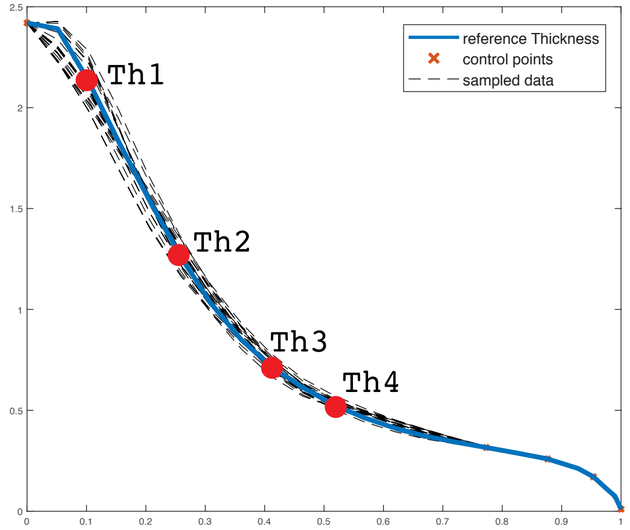
Geometric Uncertainty



Twist



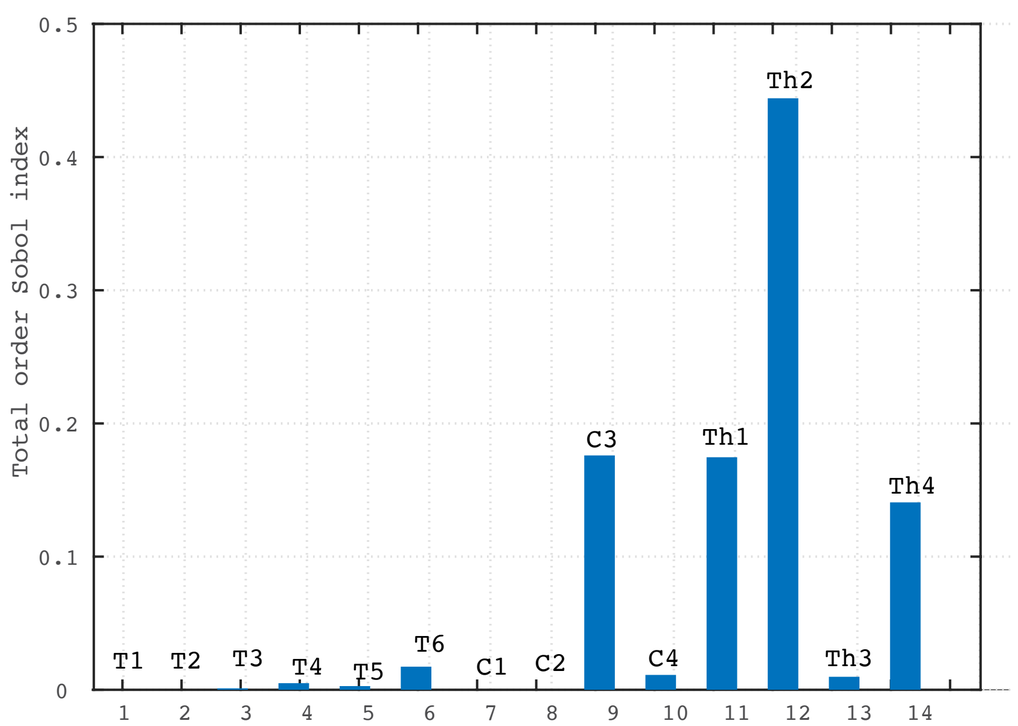
Chord



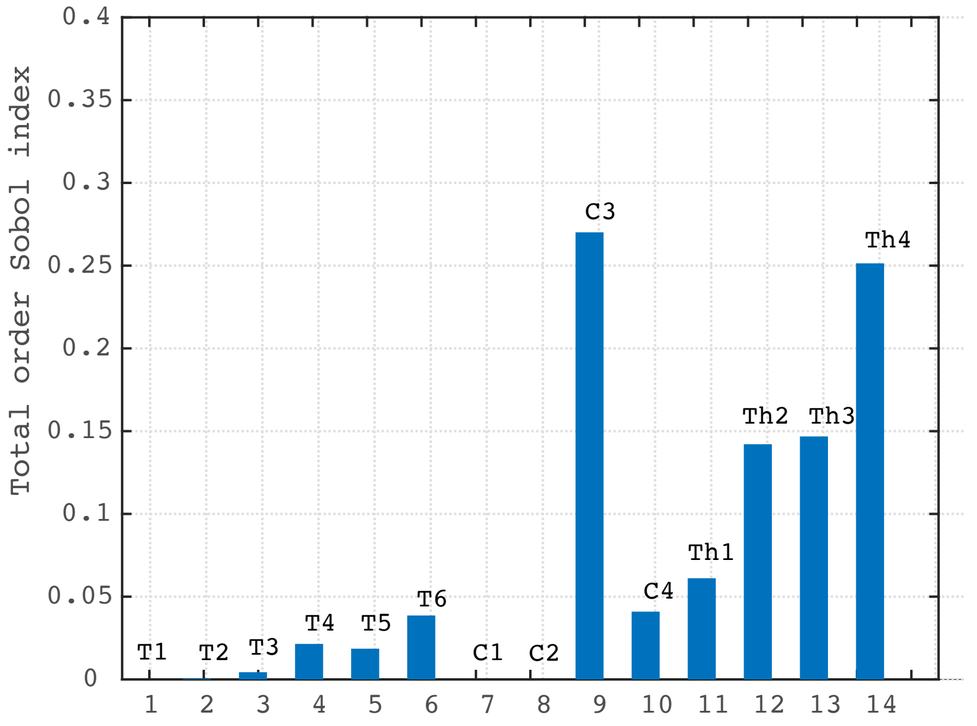
Thickness

10% (uniformly distributed) uncertainty in chosen control point

Sobol indices



Power



Axial Force

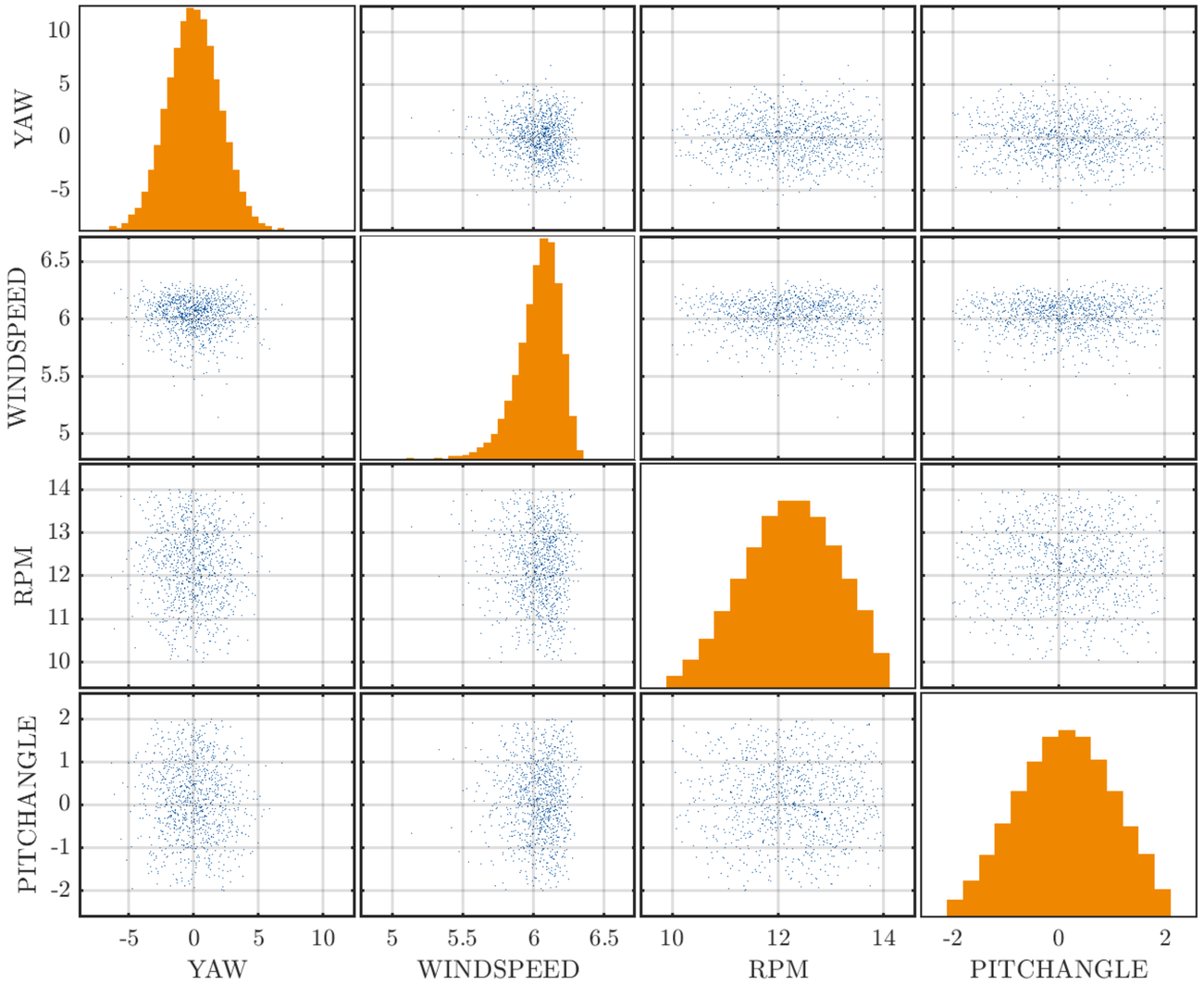
Operation Uncertainty

Yaw
Truncated Gaussian [mean = 0, std = 2, LB = -10, UB = 10]

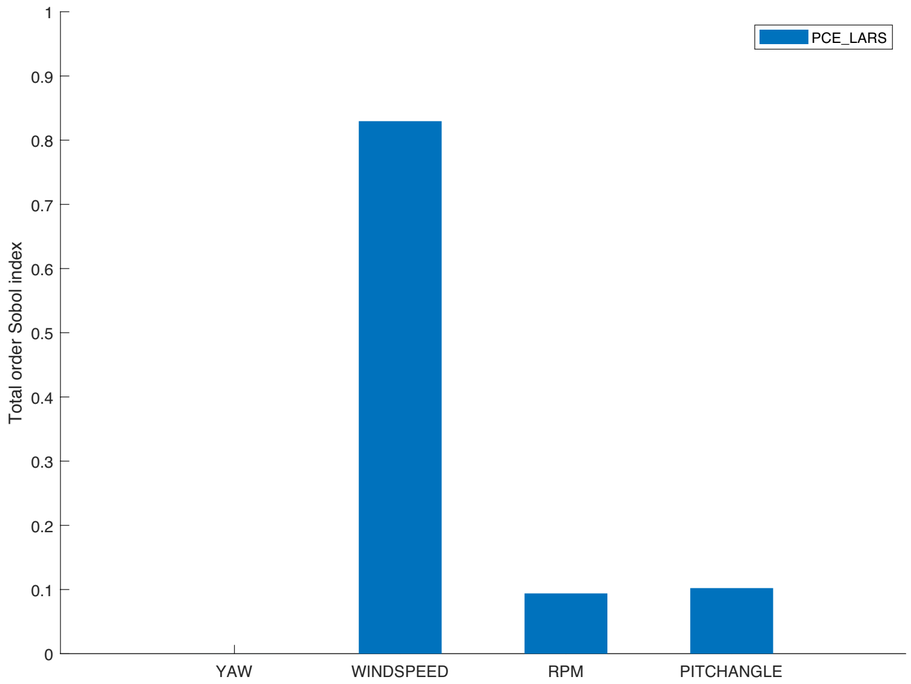
WindSpeed
Weibull distribution [Scale = 6.1, Shape = 50]

RPM
Truncated Gaussian [mean = 12.3, std = 1, LB = 10, UB = 14]

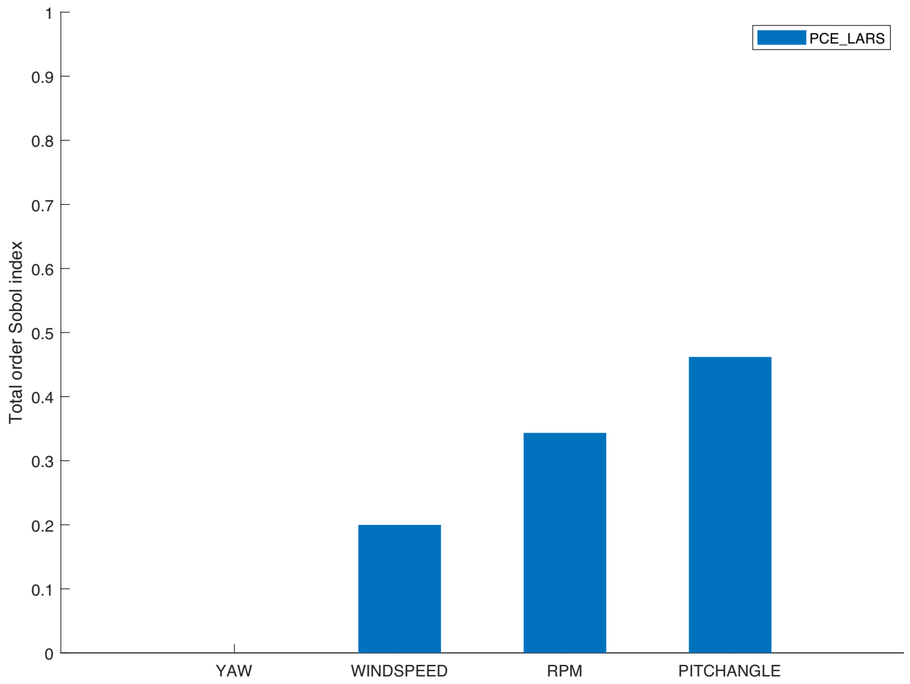
PitchAngle
Truncated Gaussian [mean = 0.15, std = 1, LB = -2, UB = 2]



Sobol indices



Power



Axial Force

-> Yaw angle is least sensitive parameter

Next Steps:

-> Determine realistic amount of perturbations for uncertain parameters

-> Parametrization for other random inputs, model-form uncertainty

-> Include other BEM codes in the workflow and perform comparisons