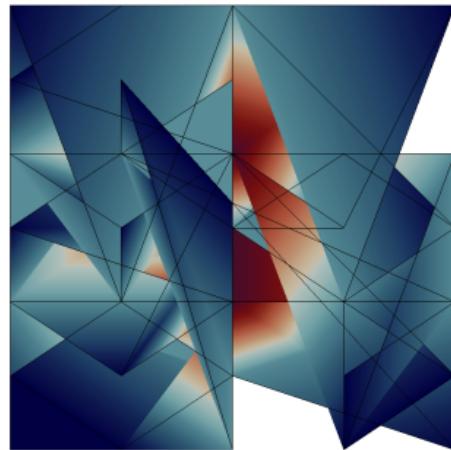


FieldStone

[Finite Elements (duh!) applied to Stones]



C. Thieulot

With contributions from: Wolfgang Bangerth, Eric van den Hoogen, Job Mos, Erik van der Wiel, Lukas van de Wiel, Alraune Zech, Wim Spakman.

If you find anything in this document useful for your research please cite it as follows (*please* include the doi!):

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```

Why do I have to promise where I am going while I am not there yet?

You can't google something you don't know exists.

You can be correct or you can get stuff done.

Computational Geodynamics

C. Thieulot

November 11, 2019

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WARNING: this is work in progress

1 Introduction

1.1 Philosophy

This document was written with my students in mind, i.e. 3rd and 4th year Geology/Geophysics students at Utrecht University. I have chosen to use as little jargon as possible unless it is a term that is commonly found in the geodynamics literature (methods paper as well as application papers). There is no mathematical proof of any theorem that may be mentioned but I will try to refer to the appropriate sources, i.e. generic Numerical Analysis, Finite Element and Linear Algebra books. If you find that this book lacks references to Sobolev spaces, Hilbert spaces, and other spaces, this book is just not for you.

The codes I provide here are by no means optimised as I have chosen code readability over code efficiency. I have also chosen to avoid resorting to multiple code files or even functions in order to favour a sequential reading of the codes. These codes are not designed to form the basis of a real life application: Existing open source highly optimised codes should be preferred, such as ASPECT [1119, 882], CITCOM [2014, 2010], LAMM [1036], PTATIN [1292, 1289], PYLITH[1], ...

All kinds of feedback is welcome on the text (grammar, typos, ...), on the text, the equations or on the code(s). You will have my eternal gratitude if you wish to contribute an example, a benchmark, a cookbook.

All the python scripts and this document are freely available at

<https://github.com/cedrikt/fieldstone>

1.2 ambition & motivation

I wish to provide the community with:

- an enormous bibliography data base - simply search the pdf for keywords. The \LaTeX file is also available next to the manual.tex file on github;
- a go-to document for anybody who wants to know more about a particular topic in computational geodynamics;
- a useful teaching tool for researchers, teachers, students and PhD students alike;
- small, readable, educative codes.

1.3 Acknowledgements

I have benefitted from many discussions, lectures, tutorials, coffee machine discussions, debugging sessions, conference poster sessions, etc ... over the years. I wish to name these instrumental people in particular and in alphabetic order: Wolfgang Bangerth, Jean Braun, Rens Elbertsen, Philippe Fullsack, Menno Fraters, Anne Glerum, Timo Heister, Dave May, Robert Myhill, John Naliboff, E. Gerry Puckett, Melchior Schuh-Senlis, Michael Tetley, Lukas van de Wiel, Arie van den Berg, Eric van den Hoogen, Tom Weir, and the whole ASPECT family/team.

1.4 About the author

I have BSc in mathematics, and an MSc diploma in physics (with a specialization in musical acoustics [515]). I did my PhD at the university of Groningen (The Netherlands) title *Thermodynamically consistent fluid particle modelling of phase separating mixtures*¹. Although half of the thesis deals with the re-derivation of the Navier-Stokes equations for such systems[585], the second half is concerned with the implementation of these equations with the Smoothed Particle Hydrodynamics method [1764, 1765, 1763].

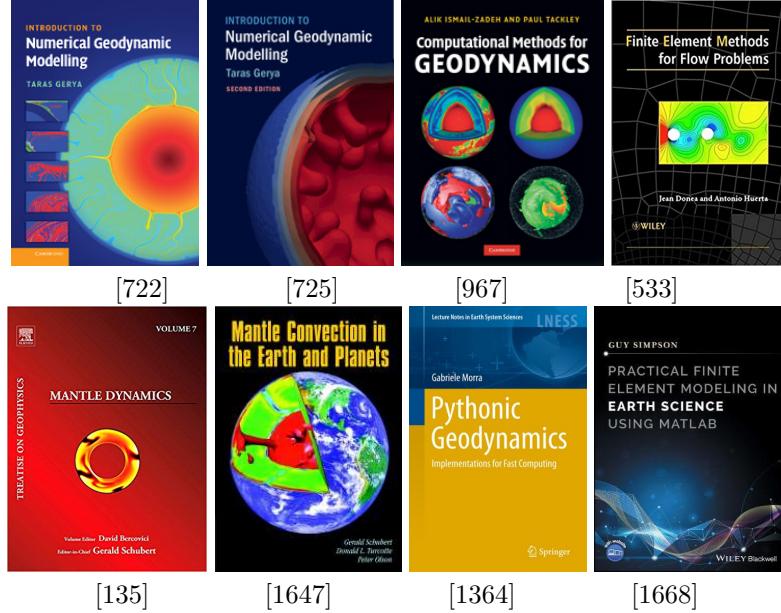
I then taught physics and programming at the University of Rennes (France) for a year, after which I did a 2-year post-doc with Prof. J. Braun² in the Geosciences department. I then did a 4-year post-doc

¹<http://cedriktchielot.net/thesis.html>

²<https://www.gfz-potsdam.de/en/staff/jean-braun/>

with prof. R. Huismans³ at the University of Bergen (Norway), followed by a 3-year post-doc with profs. T. Torsvik and W. Spakman at the Utrecht University (The Netherlands). Since June 2015 I am assistant professor there in the geophysics group.

1.5 Essential/relevant literature



- *Numerical modeling of Earth Systems* by Thorsten W. Becker and Boris J. P. Kaus, <http://www-udc.ig.utexas.edu/external/becker/teaching-557.html>
- *Myths & Methods in Modeling* by M. Spiegelman, <https://www.1deo.columbia.edu/~mspieg/mmm/>
- *Computational Science I* by Matthew G. Knepley, <https://cse.buffalo.edu/~knepley/classes/caam519/Syllabus.html>
- *Introduction to Numerical Methods for Variational Problems* by Hans Petter Langtangen and Kent-Andre Mardal, <https://hplgit.github.io/fem-book/doc/pub/book/pdf/fem-book-4print.pdf>

1.6 Installation

If numpy, scipy or matplotlib are not installed on your machine, here is how you can install them:

```
python3.6 -m pip install --user numpy scipy matplotlib
```

To install the umfpack solver:

```
pip install --upgrade scikit-umfpack --user
```

1.7 What is a fieldstone?

Simply put, it is stone collected from the surface of fields where it occurs naturally. It also stands for the bad acronym: finite element deformation of stones which echoes the primary application of these codes: geodynamic modelling.

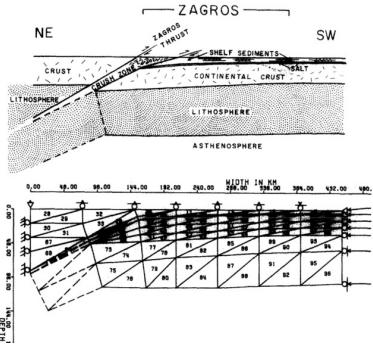
³<https://folk.uib.no/huismans/>

1.8 Why the Finite Element method?

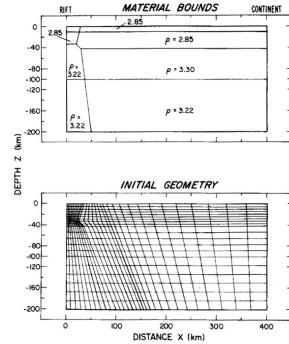
The Finite Element Method (FEM) is by no means the only method to solve PDEs in geodynamics, nor is it necessarily the best one. Other methods are employed very successfully, such as the Finite Difference Method (FDM), the Finite Volume Method (FVM), and to a lesser extent the Discrete Element Method (DEM) [1737, 564, 565, 685], the Lattice-Boltzmann method [934], the Rigid Element Method [1132], or the Element Free Galerkin Method (EFGM) [859]. I have been using FEM since 2008 and I do not have real experience to speak of in FVM or FDM so I concentrate in this book on what I know best.

1.9 Oldies but goodies

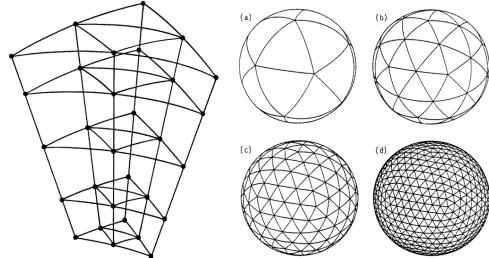
The first papers I could find showcasing the FEM in geodynamics are listed hereafter (I arbitrarily stop at 1995): [695], [27][1309][234] [575] [1751] [576][1361] [2040] [196] [225] [228]. I hereunder show a few plots taken from early geodynamics papers.



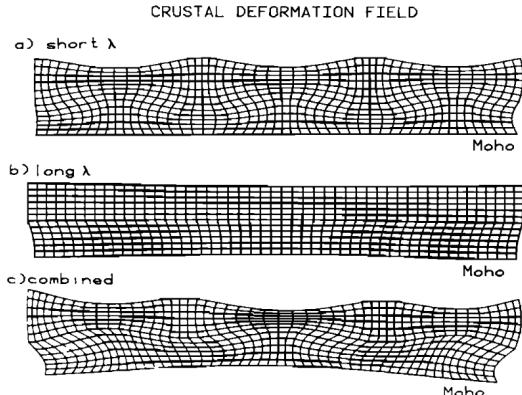
1978: Finite element modelling of lithosphere deformation: the Zagros collision orogeny [167]



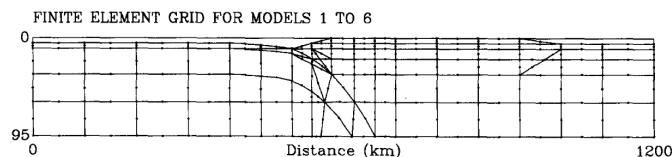
1981: Thermal regimes, mantle diapirs and crustal stresses of continental rifts [235]



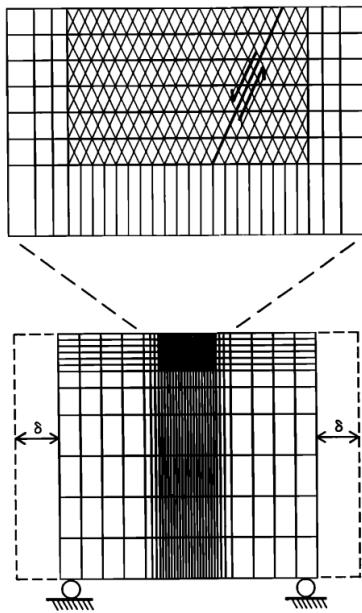
1985: Three-Dimensional Treatment of Convective Flow in the Earth's Mantle. [91]



1986: Lithospheric necking: a dynamic model for rift morphology [2041]

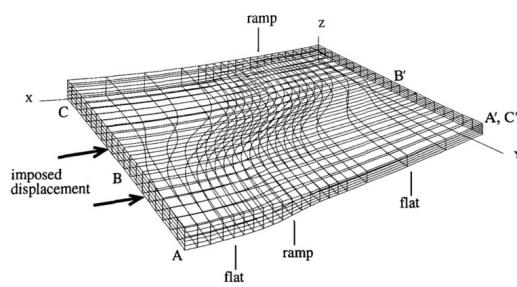


1989: Plate boundary forces at subduction zones and trench-arc compression [196]

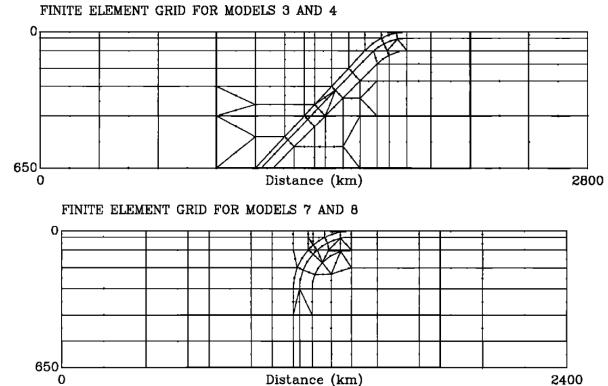


1989: Mechanics of graben formation in crustal rocks

[1310]

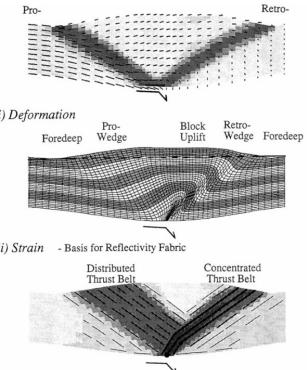


1993: 3D numerical modeling of compressional orogenies: Thrust geometry and oblique convergence [224]

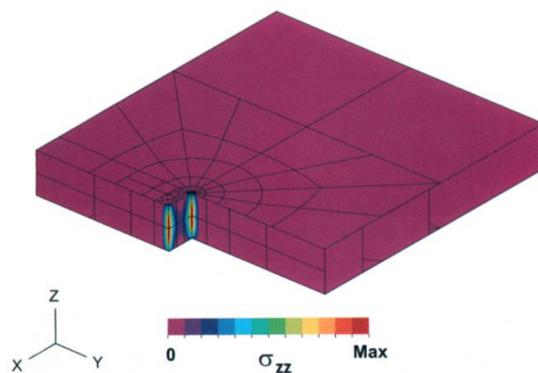


1992: Stresses and plate boundary forces associated with subduction plate margins

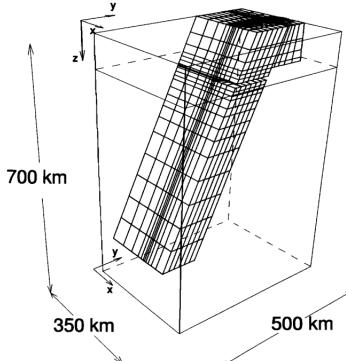
[1929]



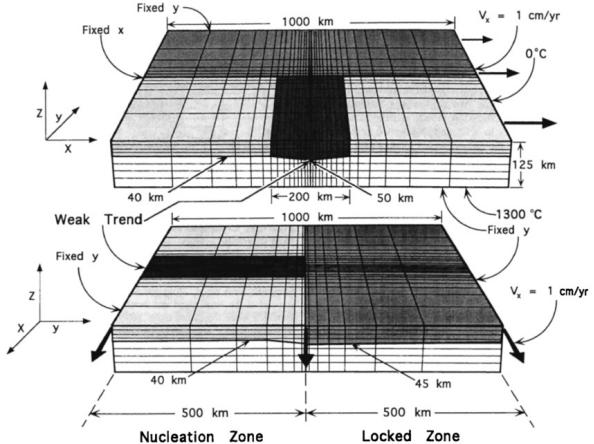
Crustal-scale compressional orogens [103]



1995: modeling of pull-apart basins [1027]



1995: 3D numerical modeling of detachment of subducted lithosphere [1981]



1996: 3D dynamical model of continental rift propagation and margin plateau formation [544]

1.10 Notations

Scalars such as temperature, density, pressure, etc ... are simply obtained in L^AT_EX by using the math mode, e.g. T , ρ , p . Although it is common to lump vectors and matrices/tensors together by using bold fonts, I have decided in the interest of clarity to distinguish between those: vectors are denoted by an arrow atop the quantity, e.g. \vec{v} , \vec{g} , while matrices and tensors are in bold M , σ , etc ...

Also I use the \cdot notation between two vectors to denote a dot product $\vec{u} \cdot \vec{v} = u_i v_i$ or a matrix-vector multiplication $M \cdot \vec{a} = M_{ij} a_j$. If there is no \cdot between vectors, it means that the result $\vec{a}\vec{b} = a_i b_j$ is a matrix (it is a dyadic product⁴. Case in point, $\vec{\nabla} \cdot \vec{v}$ is the velocity divergence while $\vec{\nabla} \vec{v}$ is the velocity gradient tensor.

1.11 Colour maps for visualisation

In an attempt to homogenise the figures obtained with ParaView, I have decided to use a fixed colour scale for each field throughout this document. These colour scales were obtained from this link and are Perceptually Uniform Colour Maps [1112].

Field	colour code
Velocity/displacement	CET-D1A
Pressure	CET-L17
Velocity divergence	CET-L1
Density	CET-D3
Strain rate	CET-R2
Viscosity	CET-R3
Temperature	CET-D9

1.12 Bibliography

If the paper is a single-author paper, say by Garfield, published in 1978, its code in my bibliography file is *garf78* (i.e. the first four letters of the name, followed by the two digits of the publication year).

⁴<https://en.wikipedia.org/wiki/Dyadics>

If the paper was written by two authors, say Garfield and Odie, in 1987, its code will be *gaod87*, i.e. the first two letters of the first author followed by the two first letters of the second author followed by two digits.

If the paper was written by three or more authors, say Garfield, Odie, John and Irene in 2003, its code will be *gaoj03*, i.e. the first two letters of the first author followed by the first letter of the second author, the first letter of the third author and the year.

Remark. Dutch names such as '*van Hunen*' or '*van den Berg*' are classified under letter '*v*', not '*h*' or '*d*' nor '*b*'.

2 List of stones

I AM NOT HAPPY ABOUT THIS ... WILL CHANGE THIS SOON-ISH

stone number	element	outer solver	formulation	physical problem	3D	temperature	time stepping	nonlinear	compressible	analytical benchmark	numerical benchmark	elastomechanics
1	$Q_1 \times P_0$		penalty	analytical benchmark MMS1					†			
2	$Q_1 \times P_0$		penalty	Stokes sphere								
3	$Q_1 \times P_0$		penalty	Blankenbach et al., 1989	†	†						
4	$Q_1 \times P_0$		penalty	Lid driven cavity								
5	$Q_1 \times P_0$		penalty	SolCx benchmark								
6	$Q_1 \times P_0$		penalty	SolKz benchmark								
7	$Q_1 \times P_0$		penalty	SolVi benchmark								
8	$Q_1 \times P_0$		penalty	Indentor				†				
9	$Q_1 \times P_0$		penalty	annulus benchmark								
10	$Q_1 \times P_0$		penalty	Stokes sphere	†							
11	$Q_1 \times P_0$	full matrix	mixed	Stokes sphere	†							
12	$Q_1 \times P_0$		penalty	analytical benchmark + consistent press recovery								
13	$Q_1 \times P_0$		penalty	Stokes sphere + markers averaging								
14	$Q_1 \times P_0$	full matrix	mixed	analytical benchmark								
15	$Q_1 \times P_0$	Schur comp. CG	mixed	analytical benchmark								
16	$Q_1 \times P_0$	Schur comp. PCG	mixed	Stokes sphere								
17	$Q_2 \times Q_1$	full matrix	mixed	Burstedde benchmark	†							
18	$Q_2 \times Q_1$	full matrix	mixed	analytical benchmark								
19	$Q_3 \times Q_2$	full matrix	mixed	analytical benchmark								
20	$Q_1 \times P_0$		penalty	Busse et al., 1993	†	†	†					
21	$Q_1 \times P_0$ R-T		penalty	analytical benchmark								
22	$Q_1 \times Q_1$ -stab	full matrix	mixed	analytical benchmark								
23	$Q_1 \times P_0$		mixed	analytical benchmark					†			
24	$Q_1 \times P_0$		mixed	convection box	†	†			†			

Analytical benchmark means that an analytical solution exists while numerical benchmark means that a comparison with other code(s) has been carried out.

3 The physical equations

Symbol	meaning	unit
t	Time	s
x, y, z	Cartesian coordinates	m
r, θ	Polar coordinates	m,-
r, θ, z	Cylindrical coordinates	m,-,m
r, θ, ϕ	Spherical coordinates	m,-,-
$\vec{v} = (u, v, w)$	velocity vector	$\text{m} \cdot \text{s}^{-1}$
\vec{u}	displacement vector	m
ρ	mass density	kg/m^3
η	dynamic viscosity	$\text{Pa} \cdot \text{s}$
λ	penalty parameter	$\text{Pa} \cdot \text{s}$
T	temperature	K
$\vec{\nabla}$	gradient operator	m^{-1}
$\vec{\nabla} \cdot$	divergence operator	m^{-1}
p	pressure	Pa
$\dot{\varepsilon}(\vec{v})$	strain rate tensor	s^{-1}
α	thermal expansion coefficient	K^{-1}
k	thermal conductivity	$\text{W}/(\text{m} \cdot \text{K})$
C_p	Heat capacity	J/K
H	intrinsic specific heat production	W/kg
β_T	isothermal compressibility	Pa^{-1}
τ	deviatoric stress tensor	Pa
σ	full stress tensor	Pa

3.1 Coordinate systems

3.1.1 Cartesian coordinates

The unit vectors along the x , y and z axis are \vec{e}_x , \vec{e}_y and \vec{e}_z respectively.

3.1.2 Cylindrical coordinates

We have $r > 0$ and $\theta = [0, 2\pi[$, defined in the (x, y) -plane.

The relation between the unit vector in Cartesian and Polar/Cylindrical coordinates is given by:

$$\begin{pmatrix} \vec{e}_r \\ \vec{e}_\theta \end{pmatrix} = \begin{pmatrix} \cos \theta & \sin \theta \\ -\sin \theta & \cos \theta \end{pmatrix} \cdot \begin{pmatrix} \vec{e}_x \\ \vec{e}_y \end{pmatrix}$$

which yields

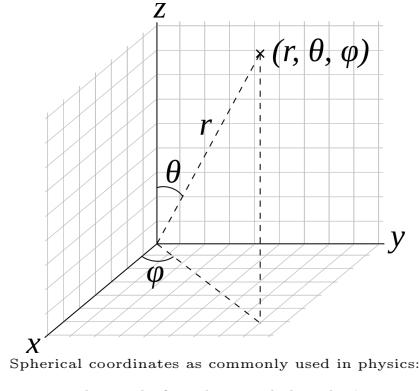
$$\begin{pmatrix} \vec{e}_x \\ \vec{e}_y \end{pmatrix} = \begin{pmatrix} \cos \theta & -\sin \theta \\ \sin \theta & \cos \theta \end{pmatrix} \cdot \begin{pmatrix} \vec{e}_r \\ \vec{e}_\theta \end{pmatrix}$$

so that for any vector \vec{V}

$$\begin{aligned} \vec{V} &= V_x \vec{e}_x + V_y \vec{e}_y \\ &= V_x[(\cos \theta) \vec{e}_r - (\sin \theta) \vec{e}_\theta] + V_y[(\sin \theta) \vec{e}_r + (\cos \theta) \vec{e}_\theta] \\ &= [V_x(\cos \theta) + V_y(\sin \theta)] \vec{e}_r + [-V_x(\sin \theta) + V_y(\cos \theta)] \vec{e}_\theta \end{aligned} \tag{1}$$

3.1.3 Spherical coordinates

On the following figure are represented the three cartesian axis, a point and its spherical coordinates r, θ, ϕ :



In this case $\theta \in [0 : \pi]$ and $\phi \in] -\pi : \pi]$ and we have the following relationships:

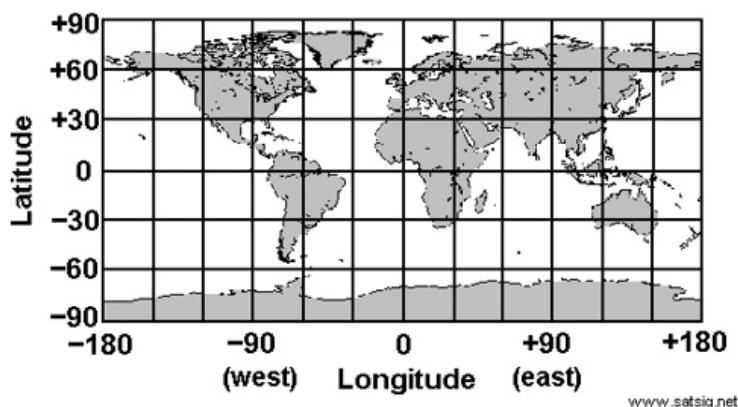
$$r = \sqrt{x^2 + y^2 + z^2} \quad (2)$$

$$\theta = \arccos(z/r) \quad (3)$$

$$\phi = \arctan(y/x) \quad (4)$$

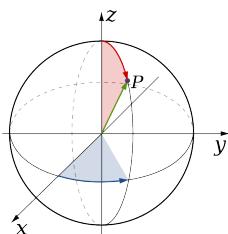
The inverse tangent used to compute ϕ must be suitably defined, taking into account the correct quadrant of (x, y) , which is why the atan2 intrinsic function is used in fortran for example.

In geography one uses latitude and longitude, represented hereunder:



- Latitude $\in [-90 : 90]$, or $\in [-\pi/2 : \pi/2]$
- Longitude $\in] -180 : 180]$, or $\in] -\pi : \pi]$

Since the colatitude is the complementary angle of the latitude, i.e. the difference between 90 and the latitude, where southern latitudes are denoted with a minus sign, θ as shown above is actually the colatitude. The co-latitude is shown in red on the following figure:



3.2 A continuum mechanics primer

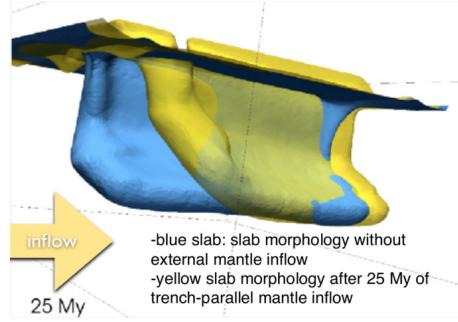
Contains contributions by W. Spakman

In continuum mechanics we make a distinction between two broad classes of forces:

- Body forces defined as force per unit volume (N/m^3): gravity, electro-magnetic forces
- Traction: Surface forces defined as force per unit surface area (N/m^2): Contact forces, elastic forces per unit area, internal flow friction, pressure, ...

A traction is the surface average of all atomic forces exerted by atoms on the one side on atoms on the other side of the surface. For real-Earth processes, internal tractions are ultimately caused by the body forces, usually gravity.

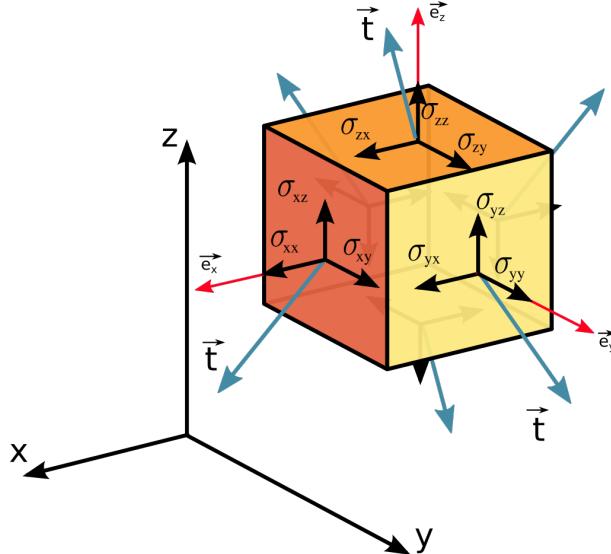
Existing mantle flow(i.e. flow that is forced elsewhere) can exert tractions (shear stresses) on the subducting slab or for instance at the base of lithosphere plates. In HPT-laboratory experiments external tractions (pressure, shear traction) are applied to a rock sample, which cause internal tractions to balance the exerted forces.



3.2.1 Stress tensor and tractions

The Cauchy tensor⁵ consists of nine components σ_{ij} that completely define the state of stress at a point inside a material. The tensor relates a unit-length direction vector \vec{n} to the so-called 'stress vector' (most commonly called 'traction') $\vec{t}(\vec{n})$ across an imaginary surface perpendicular to \vec{n} :

$$\vec{t}(\vec{n}) = \vec{n} \cdot \boldsymbol{\sigma}$$



Modified from original file on Wikipedia⁶

With respect to an orthonormal basis $\{\vec{e}_x, \vec{e}_y, \vec{e}_z\}$, the Cauchy stress tensor is given by:

$$\boldsymbol{\sigma} = \begin{pmatrix} \sigma_{xx} & \sigma_{xy} & \sigma_{xz} \\ \sigma_{yx} & \sigma_{yy} & \sigma_{yz} \\ \sigma_{zx} & \sigma_{zy} & \sigma_{zz} \end{pmatrix} \quad (5)$$

⁵https://en.wikipedia.org/wiki/Cauchy_stress_tensor

⁶https://commons.wikimedia.org/wiki/File:Components_stress_tensor_cartesian.svg

The three diagonal elements are called normal stresses while the off-diagonal terms are called shear stresses.

One can easily prove (see for instance Section 3.3.6 of [815]) that the balance of angular momentum leads reduces to the statement that the Cauchy stress tensor is symmetric, i.e. $\boldsymbol{\sigma} = \boldsymbol{\sigma}^T$. Therefore, the stress state of the medium at any point and instant can be specified by only six independent parameters, rather than nine:

$$\boldsymbol{\sigma} = \begin{pmatrix} \sigma_{xx} & \sigma_{xy} & \sigma_{xz} \\ \sigma_{xy} & \sigma_{yy} & \sigma_{yz} \\ \sigma_{xz} & \sigma_{yz} & \sigma_{zz} \end{pmatrix} \quad \text{or sometimes} \quad \boldsymbol{\sigma} = \begin{pmatrix} \sigma_x & \tau_{xy} & \tau_{xz} \\ \tau_{xy} & \sigma_y & \tau_{yz} \\ \tau_{xz} & \tau_{yz} & \sigma_z \end{pmatrix} \quad (6)$$

where the elements $\sigma_x, \sigma_y, \sigma_z$ are called the orthogonal normal stresses (relative to the chosen coordinate system), and $\tau_{xy}, \tau_{xz}, \tau_{yz}$ the orthogonal shear stresses.

As seen above, the SI units of both stress tensor and traction are N/m².

3.2.2 Strain rate and spin tensor

The velocity gradient is given in Cartesian coordinates by:

$$\vec{\nabla}\vec{v} = \begin{pmatrix} \frac{\partial u}{\partial x} & \frac{\partial v}{\partial x} & \frac{\partial w}{\partial x} \\ \frac{\partial u}{\partial y} & \frac{\partial v}{\partial y} & \frac{\partial w}{\partial y} \\ \frac{\partial u}{\partial z} & \frac{\partial v}{\partial z} & \frac{\partial w}{\partial z} \end{pmatrix} \quad (7)$$

It can be decomposed into its symmetric and skew-symmetric parts according to:

$$\vec{\nabla}\vec{v} = \vec{\nabla}^s\vec{v} + \vec{\nabla}^w\vec{v} \quad (8)$$

The symmetric part is called the strain rate (or rate of deformation):

$$\dot{\boldsymbol{\epsilon}}(\vec{v}) = \frac{1}{2} \left(\vec{\nabla}\vec{v} + (\vec{\nabla}\vec{v})^T \right) \quad (9)$$

The skew-symmetric tensor is called spin tensor (or vorticity tensor):

$$\dot{\boldsymbol{R}}(\vec{v}) = \frac{1}{2} \left(\vec{\nabla}\vec{v} - (\vec{\nabla}\vec{v})^T \right) \quad (10)$$

Remark. In the mathematical literature a different notation for the strain rate tensor is often used, i.e. $D(\vec{v})$, such as for instance in [678].

3.3 Viscous Newtonian rheology

The relationship between velocity-related stresses and velocity derivatives is such that the total stress tensor has the form [135]

$$\boldsymbol{\sigma} = -p\mathbf{1} + \mathbf{A} : \dot{\boldsymbol{\epsilon}} \quad (11)$$

where p is the thermodynamic pressure which is a function of the density ρ and the temperature T (an equation of state is then needed) and \mathbf{A} is the fourth-rank stiffness tensor.

Since both the stress and the strain tensors are symmetric and for isotropic fluids we have [1256]

$$\mathbf{A} : \dot{\boldsymbol{\epsilon}} = \lambda(\vec{\nabla} \cdot \vec{v})\mathbf{1} + 2\eta\dot{\boldsymbol{\epsilon}} \quad (12)$$

where λ is the bulk viscosity and η is the dynamic viscosity⁷. The stress tensor is then

$$\boldsymbol{\sigma} = (-p + \lambda(\vec{\nabla} \cdot \vec{v}))\mathbf{1} + 2\eta\dot{\boldsymbol{\epsilon}} \quad (13)$$

By writing

$$\dot{\boldsymbol{\epsilon}} = \frac{1}{3} \text{tr}(\dot{\boldsymbol{\epsilon}})\mathbf{1} + \dot{\boldsymbol{\epsilon}}^d = \frac{1}{3}(\vec{\nabla} \cdot \vec{v})\mathbf{1} + \dot{\boldsymbol{\epsilon}}^d$$

⁷also sometimes called shear viscosity

where $\dot{\epsilon}^d$ is the deviatoric strain rate tensor and (in Cartesian coordinates)

$$\vec{\nabla} \cdot \vec{v} = \text{div}(\vec{v}) = \text{tr}(\dot{\epsilon}) = \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} \quad (14)$$

where tr is the trace operator, we arrive at

$$\sigma = (-p + \lambda(\vec{\nabla} \cdot \vec{v}))\mathbf{1} + 2\eta \left[\frac{1}{3}(\vec{\nabla} \cdot \vec{v})\mathbf{1} + \dot{\epsilon}^d \right] \quad (15)$$

$$= \left[-p + \left(\lambda + \frac{2}{3}\eta \right) (\vec{\nabla} \cdot \vec{v}) \right] \mathbf{1} + 2\eta \dot{\epsilon}^d \quad (16)$$

Introducing the second viscosity $\zeta = \lambda + \frac{2}{3}\eta$:

$$\sigma = \left[-p + \zeta(\vec{\nabla} \cdot \vec{v}) \right] \mathbf{1} + 2\eta \dot{\epsilon}^d \quad (17)$$

$$= -\bar{p}\mathbf{1} + 2\eta \dot{\epsilon}^d \quad (18)$$

The effect of the volume viscosity ζ is that the mechanical pressure \bar{p} is not equivalent to the thermodynamic pressure p

$$\bar{p} = p - \zeta(\vec{\nabla} \cdot \vec{v}) \quad (19)$$

In other words: the isotropic average of the total stress is *not* the pressure term! This difference is usually neglected (and it is safe to do so, see [135, section 7.02.3.2.2]) by explicitly assuming $\zeta = 0$ (also called the Stokes assumption [1647, p256]), so that one can then refer to pressure as a single well-defined value. Note that in the case of an incompressible Newtonian Fluid, the strain rate tensor is deviatoric ($\text{tr}(\dot{\epsilon}) = \text{div}(\vec{v}) = 0$) and the above considerations vanish.

Finally, for both compressible and incompressible flow, the stress tensor becomes simply

$$\sigma = -p\mathbf{1} + 2\eta \dot{\epsilon}^d = -p\mathbf{1} + \tau \quad (20)$$

where $\tau = 2\eta \dot{\epsilon}^d$ is the deviatoric stress tensor.

Remark. On page 256 of [1647], equation 6.5.3, the authors write $\tau_{ii}/3 = k_B e_{ii}$ while stating that τ is deviatoric in equation 6.4.2. This is an obvious conflict of notations.

3.4 The heat transport equation - energy conservation equation

Let us start from the heat transport equation as shown in Schubert, Turcotte and Olson [1647]:

$$\rho C_p \frac{DT}{Dt} - \alpha T \frac{Dp}{Dt} = \vec{\nabla} \cdot k \vec{\nabla} T + \Phi + \rho H \quad (21)$$

with D/Dt being the total derivatives so that

$$\frac{DT}{Dt} = \frac{\partial T}{\partial t} + \vec{v} \cdot \vec{\nabla} T \quad \frac{Dp}{Dt} = \frac{\partial p}{\partial t} + \vec{v} \cdot \vec{\nabla} p \quad (22)$$

Solving for temperature, this equation is often rewritten as follows:

$$\rho C_p \frac{DT}{Dt} - \vec{\nabla} \cdot k \vec{\nabla} T = \alpha T \frac{Dp}{Dt} + \Phi + \rho H \quad (23)$$

where Φ is the shear heating [1551, p287]. In many publications, Φ is given by $\Phi = \tau_{ij} \partial_j v_i = \tau : \vec{\nabla} \vec{v}$.

In what follows I use the index notation as it makes for easier derivations:

$$\begin{aligned}
\Phi &= \tau_{ij} \partial_j v_i \\
&= 2\eta \dot{\varepsilon}_{ij}^d \partial_j v_i \\
&= 2\eta \frac{1}{2} (\dot{\varepsilon}_{ij}^d \partial_j v_i + \dot{\varepsilon}_{ji}^d \partial_i u_j) \\
&= 2\eta \frac{1}{2} (\dot{\varepsilon}_{ij}^d \partial_j v_i + \dot{\varepsilon}_{ij}^d \partial_i u_j) \\
&= 2\eta \dot{\varepsilon}_{ij}^d \frac{1}{2} (\partial_j v_i + \partial_i u_j) \\
&= 2\eta \dot{\varepsilon}_{ij}^d \dot{\varepsilon}_{ij} \\
&= 2\eta \dot{\varepsilon}^d : \dot{\varepsilon} \\
&= 2\eta \dot{\varepsilon}^d : \left(\dot{\varepsilon}^d + \frac{1}{3} (\vec{\nabla} \cdot \vec{v}) \mathbf{1} \right) \\
&= 2\eta \dot{\varepsilon}^d : \dot{\varepsilon}^d + 2\eta \dot{\varepsilon}^d : \mathbf{1} (\vec{\nabla} \cdot \vec{v}) \\
&= 2\eta \dot{\varepsilon}^d : \dot{\varepsilon}^d
\end{aligned} \tag{24}$$

Finally

$$\Phi = \boldsymbol{\tau} : \vec{\nabla} \vec{v} = 2\eta \dot{\varepsilon}^d : \dot{\varepsilon}^d = 2\eta ((\dot{\varepsilon}_{xx}^d)^2 + (\dot{\varepsilon}_{yy}^d)^2 + 2(\dot{\varepsilon}_{xy}^d)^2) \tag{25}$$

3.5 The momentum conservation equations

Because the Prandlt number is virtually zero in Earth science applications the Navier Stokes equations reduce to the Stokes equation:

$$\vec{\nabla} \cdot \boldsymbol{\sigma} + \rho \vec{g} = \vec{0} \tag{26}$$

Since

$$\boldsymbol{\sigma} = -p \mathbf{1} + \boldsymbol{\tau} \tag{27}$$

it also writes

$$-\vec{\nabla} p + \vec{\nabla} \cdot \boldsymbol{\tau} + \rho \vec{g} = \vec{0} \tag{28}$$

Using the relationship $\boldsymbol{\tau} = 2\eta \dot{\varepsilon}^d$ we arrive at

$$-\vec{\nabla} p + \vec{\nabla} \cdot (2\eta \dot{\varepsilon}^d) + \rho \vec{g} = \vec{0} \tag{29}$$

3.6 The mass conservation equations

The mass conservation equation is given by

$$\frac{D\rho}{Dt} + \rho \vec{\nabla} \cdot \vec{v} = 0$$

or,

$$\frac{\partial \rho}{\partial t} + \vec{\nabla} \cdot (\rho \vec{v}) = 0$$

In the case of an incompressible flow, then $\partial \rho / \partial t = 0$ and $\vec{\nabla} \rho = 0$, i.e. $D\rho/Dt = 0$ and the remaining equation is simply:

$$\vec{\nabla} \cdot \vec{v} = 0$$

A vector field that is divergence-free is also called solenoidal⁸.

⁸https://en.wikipedia.org/wiki/Solenoidal_vector_field

3.7 The equations in ASPECT manual

The following is lifted off the ASPECT manual. We focus on the system of equations in a $d = 2$ - or $d = 3$ -dimensional domain Ω that describes the motion of a highly viscous fluid driven by differences in the gravitational force due to a density that depends on the temperature. In the following, we largely follow the exposition of this material in Schubert, Turcotte and Olson [1647].

Specifically, we consider the following set of equations for velocity \mathbf{u} , pressure p and temperature T :

$$-\vec{\nabla} \cdot \left[2\eta \left(\dot{\varepsilon}(\vec{v}) - \frac{1}{3}(\vec{\nabla} \cdot \vec{v})\mathbf{1} \right) \right] + \vec{\nabla} p = \rho \vec{g} \quad \text{in } \Omega, \quad (30)$$

$$\vec{\nabla} \cdot (\rho \vec{v}) = 0 \quad \text{in } \Omega, \quad (31)$$

$$\begin{aligned} \rho C_p \left(\frac{\partial T}{\partial t} + \vec{v} \cdot \vec{\nabla} T \right) - \vec{\nabla} \cdot k \vec{\nabla} T &= \rho H \\ &+ 2\eta \left(\dot{\varepsilon}(\mathbf{v}) - \frac{1}{3}(\vec{\nabla} \cdot \vec{v})\mathbf{1} \right) : \left(\dot{\varepsilon}(\mathbf{v}) - \frac{1}{3}(\vec{\nabla} \cdot \vec{v})\mathbf{1} \right) \\ &+ \alpha T (\mathbf{v} \cdot \vec{\nabla} p) \end{aligned} \quad \text{in } \Omega, \quad (32)$$

where $\dot{\varepsilon}(\vec{v}) = \frac{1}{2}(\vec{\nabla} \vec{v} + \vec{\nabla} \vec{v}^T)$ is the symmetric gradient of the velocity (often called the *strain rate*).

In this set of equations, (684) and (685) represent the compressible Stokes equations in which $\mathbf{v} = \mathbf{v}(\mathbf{x}, t)$ is the velocity field and $p = p(\mathbf{x}, t)$ the pressure field. Both fields depend on space \mathbf{x} and time t . Fluid flow is driven by the gravity force that acts on the fluid and that is proportional to both the density of the fluid and the strength of the gravitational pull.

Coupled to this Stokes system is equation (686) for the temperature field $T = T(\mathbf{x}, t)$ that contains heat conduction terms as well as advection with the flow velocity \mathbf{v} . The right hand side terms of this equation correspond to

- internal heat production for example due to radioactive decay;
- friction (shear) heating;
- adiabatic compression of material;

In order to arrive at the set of equations that ASPECT solves, we need to

- neglect the $\partial p / \partial t$. **WHY?**
- neglect the $\partial \rho / \partial t$. **WHY?**

from equations above.

Also, their definition of the shear heating term Φ is:

$$\Phi = k_B(\nabla \cdot \mathbf{v})^2 + 2\eta \dot{\varepsilon}^d : \dot{\varepsilon}^d$$

For many fluids the bulk viscosity k_B is very small and is often taken to be zero, an assumption known as the Stokes assumption: $k_B = \lambda + 2\eta/3 = 0$. Note that η is the dynamic viscosity and λ the second viscosity. Also,

$$\boldsymbol{\tau} = 2\eta \dot{\varepsilon} + \lambda(\nabla \cdot \mathbf{v})\mathbf{1}$$

but since $k_B = \lambda + 2\eta/3 = 0$, then $\lambda = -2\eta/3$ so

$$\boldsymbol{\tau} = 2\eta \dot{\varepsilon} - \frac{2}{3}\eta(\nabla \cdot \mathbf{v})\mathbf{1} = 2\eta \dot{\varepsilon}^d$$

3.8 The Boussinesq approximation

As nicely explained in Spiegel & Veronis [1684]: "In the study of problems of thermal convection it is a frequent practice to simplify the basic equations by introducing certain approximations which are attributed to Boussinesq (1903). The Boussinesq approximations can best be summarized by two statements: (1) The fluctuations in density which appear with the advent of motion result principally from thermal (as opposed to pressure) effects. (2) In the equations for the rate of change of momentum and mass, density variations may be neglected except when they are coupled to the gravitational acceleration in the buoyancy force." Note that their paper examines the Boussinesq approximation for compressible fluids.

[from ASPECT manual] The Boussinesq approximation assumes that the density can be considered constant in all occurrences in the equations with the exception of the buoyancy term on the right hand side of (684). The primary result of this assumption is that the continuity equation (685) will now read

$$\nabla \cdot \mathbf{v} = 0$$

This implies that the strain rate tensor is deviatoric. Under the Boussinesq approximation, the equations are much simplified:

$$-\nabla \cdot [2\eta\dot{\epsilon}(\mathbf{v})] + \nabla p = \rho\mathbf{g} \quad \text{in } \Omega, \quad (33)$$

$$\nabla \cdot (\rho\mathbf{v}) = 0 \quad \text{in } \Omega, \quad (34)$$

$$\rho_0 C_p \left(\frac{\partial T}{\partial t} + \mathbf{v} \cdot \nabla T \right) - \nabla \cdot k \nabla T = \rho H \quad \text{in } \Omega \quad (35)$$

Note that all terms on the rhs of the temperature equations have disappeared, with the exception of the source term.

3.9 Stokes equation for elastic medium

What follows is mostly borrowed from Becker & Kaus lecture notes.

The strong form of the PDE that governs force balance in a medium is given by

$$\vec{\nabla} \cdot \boldsymbol{\sigma} + \vec{f} = \vec{0}$$

where $\boldsymbol{\sigma}$ is the stress tensor and \vec{f} is a body force.

The stress tensor is related to the strain tensor through the generalised Hooke's law⁹:

$$\sigma_{ij} = \sum_{kl} C_{ijkl} \varepsilon_{kl} \quad \text{or} \quad \boldsymbol{\sigma} = \mathbf{C} : \boldsymbol{\varepsilon} \quad (36)$$

where \mathbf{C} is the fourth-order elastic tensor.

Due to the inherent symmetries of $\boldsymbol{\sigma}$, $\boldsymbol{\varepsilon}$, and \mathbf{C} , only 21 elastic coefficients of the latter are independent. For isotropic media (which have the same physical properties in any direction), \mathbf{C} can be reduced to only two independent numbers (for example the bulk modulus K and the shear modulus G that quantify the material's resistance to changes in volume and to shearing deformations, respectively).

One often then write Eq. as follows:

$$\boldsymbol{\sigma} = \lambda(\vec{\nabla} \cdot \vec{u})\mathbf{1} + 2\mu\boldsymbol{\varepsilon} \quad (37)$$

where λ is the Lamé parameter and μ is the shear modulus¹⁰. The term $\vec{\nabla} \cdot \vec{u}$ is the isotropic dilation.

The strain tensor is related to the displacement as follows:

$$\boldsymbol{\varepsilon} = \frac{1}{2}(\vec{\nabla}\vec{u} + (\vec{\nabla}\vec{u})^T)$$

The incompressibility (or bulk modulus) K is defined as $p = -K\vec{\nabla} \cdot \vec{u}$ where p is the pressure with

$$\begin{aligned} p &= -\frac{1}{3}\text{tr}(\boldsymbol{\sigma}) \\ &= -\frac{1}{3}[\lambda(\vec{\nabla} \cdot \vec{u})\text{tr}[\mathbf{1}] + 2\mu\text{tr}[\boldsymbol{\varepsilon}]] \\ &= -\frac{1}{3}[\lambda(\vec{\nabla} \cdot \vec{u})3 + 2\mu(\vec{\nabla} \cdot \vec{u})] \\ &= -\left[\lambda + \frac{2}{3}\mu\right](\vec{\nabla} \cdot \vec{u}) \end{aligned} \quad (38)$$

so that $K = \lambda + \frac{2}{3}\mu$.

Remark. Eq. (3.9) and (37) are analogous to the ones that one has to solve in the context of viscous flow using the penalty method. In this case λ is the penalty coefficient, \mathbf{u} is the velocity, and μ is then the dynamic viscosity.

The Lamé parameter and the shear modulus are also linked to ν the poisson ratio, and E , Young's modulus:

$$\lambda = \mu \frac{2\nu}{1-2\nu} = \frac{\nu E}{(1+\nu)(1-2\nu)} \quad \text{with} \quad E = 2\mu(1+\nu)$$

The shear modulus, expressed often in GPa, describes the material's response to shear stress. The poisson ratio describes the response in the direction orthogonal to uniaxial stress. The Young modulus, expressed in GPa, describes the material's strain response to uniaxial stress in the direction of this stress.

⁹[https://en.wikipedia.org/wiki/Hooke's_law](https://en.wikipedia.org/wiki/Hooke%27s_law)

¹⁰It is also sometimes written G

3.10 The strain rate tensor in all coordinate systems

The strain rate tensor $\dot{\varepsilon}$ is given by

$$\dot{\varepsilon} = \frac{1}{2}(\vec{\nabla}\vec{v} + \vec{\nabla}\vec{v}^T) \quad (39)$$

3.10.1 Cartesian coordinates

$$\dot{\varepsilon}_{xx} = \frac{\partial u}{\partial x} \quad (40)$$

$$\dot{\varepsilon}_{yy} = \frac{\partial v}{\partial y} \quad (41)$$

$$\dot{\varepsilon}_{zz} = \frac{\partial w}{\partial z} \quad (42)$$

$$\dot{\varepsilon}_{yx} = \dot{\varepsilon}_{xy} = \frac{1}{2} \left(\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) \quad (43)$$

$$\dot{\varepsilon}_{zx} = \dot{\varepsilon}_{xz} = \frac{1}{2} \left(\frac{\partial u}{\partial z} + \frac{\partial w}{\partial x} \right) \quad (44)$$

$$\dot{\varepsilon}_{zy} = \dot{\varepsilon}_{yz} = \frac{1}{2} \left(\frac{\partial v}{\partial z} + \frac{\partial w}{\partial y} \right) \quad (45)$$

3.10.2 Polar coordinates

$$\dot{\varepsilon}_{rr} = \frac{\partial v_r}{\partial r} \quad (46)$$

$$\dot{\varepsilon}_{\theta\theta} = \frac{v_r}{r} + \frac{1}{r} \frac{\partial v_\theta}{\partial \theta} \quad (47)$$

$$\dot{\varepsilon}_{\theta r} = \dot{\varepsilon}_{r\theta} = \frac{1}{2} \left(\frac{\partial v_\theta}{\partial r} - \frac{v_\theta}{r} + \frac{1}{r} \frac{\partial v_r}{\partial \theta} \right) \quad (48)$$

3.10.3 Cylindrical coordinates

$$\dot{\varepsilon}_{rr} = \frac{\partial v_r}{\partial r} \quad (49)$$

$$\dot{\varepsilon}_{\theta\theta} = \frac{v_r}{r} + \frac{1}{r} \frac{\partial v_\theta}{\partial \theta} \quad (50)$$

$$\dot{\varepsilon}_{\theta r} = \dot{\varepsilon}_{r\theta} = \frac{1}{2} \left(\frac{\partial v_\theta}{\partial r} - \frac{v_\theta}{r} + \frac{1}{r} \frac{\partial v_r}{\partial \theta} \right) \quad (51)$$

$$\dot{\varepsilon}_{zz} = \frac{\partial w}{\partial z} \quad (52)$$

$$\dot{\varepsilon}_{rz} = \dot{\varepsilon}_{zr} = \frac{1}{2} () \quad (53)$$

$$\dot{\varepsilon}_{\theta z} = \dot{\varepsilon}_{z\theta} = \frac{1}{2} \left(\frac{1}{r} \frac{\partial w}{\partial \theta} + \frac{\partial v_\theta}{\partial z} \right) \quad (54)$$

$$(55)$$

CHECK AND HOMOGENIZE NOTATIONS

<http://eml.ou.edu/equation/FLUIDS/STRAIN/STRAIN.HTM>

3.10.4 Spherical coordinates

$$\dot{\varepsilon}_{rr} = \frac{\partial v_r}{\partial r} \quad (56)$$

$$\dot{\varepsilon}_{\theta\theta} = \frac{v_r}{r} + \frac{1}{r} \frac{\partial v_\theta}{\partial \theta} \quad (57)$$

$$\dot{\varepsilon}_{\phi\phi} = \frac{1}{r \sin \theta} \frac{\partial v_\phi}{\partial \phi} \quad (58)$$

$$\dot{\varepsilon}_{\theta r} = \dot{\varepsilon}_{r\theta} = \frac{1}{2} \left(r \frac{\partial}{\partial r} \left(\frac{v_\theta}{r} \right) + \frac{1}{r} \frac{\partial v_r}{\partial \theta} \right) \quad (59)$$

$$\dot{\varepsilon}_{\phi r} = \dot{\varepsilon}_{r\phi} = \frac{1}{2} \left(\frac{1}{r \sin \theta} \frac{\partial v_r}{\partial \phi} + r \frac{\partial}{\partial r} \left(\frac{v_\phi}{r} \right) \right) \quad (60)$$

$$\dot{\varepsilon}_{\phi\theta} = \dot{\varepsilon}_{\theta\phi} = \frac{1}{2} \left(\frac{\sin \theta}{r} \frac{\partial}{\partial \theta} \left(\frac{v_\phi}{\sin \theta} \right) + \frac{1}{r \sin \theta} \frac{\partial v_\theta}{\partial \phi} \right) \quad (61)$$

3.11 Boundary conditions

In mathematics, the Dirichlet (or first-type) boundary condition is a type of boundary condition, named after Peter Gustav Lejeune Dirichlet. When imposed on an ODE or PDE, it specifies the values that a solution needs to take along the boundary of the domain. Note that a Dirichlet boundary condition may also be referred to as a fixed boundary condition.

The Neumann (or second-type) boundary condition is a type of boundary condition, named after Carl Neumann. When imposed on an ordinary or a partial differential equation, the condition specifies the values in which the derivative of a solution is applied within the boundary of the domain.

It is possible to describe the problem using other boundary conditions: a Dirichlet boundary condition specifies the values of the solution itself (as opposed to its derivative) on the boundary, whereas the Cauchy boundary condition, mixed boundary condition and Robin boundary condition are all different types of combinations of the Neumann and Dirichlet boundary conditions.

3.11.1 The Stokes equations

You may find the following terms in the computational geodynamics literature:

- free surface: this means that no force is acting on the surface, i.e. $\sigma \cdot \vec{n} = \vec{0}$. It is usually used on the top boundary of the domain and allows for topography evolution.
- free slip: $\vec{v} \cdot \vec{n} = 0$ and $(\sigma \cdot \vec{n}) \times \vec{n} = \vec{0}$. This condition ensures a frictionless flow parallel to the boundary where it is prescribed.
- no slip: this means that the velocity (or displacement) is exactly zero on the boundary, i.e. $\vec{v} = \vec{0}$.
- prescribed velocity: $\vec{v} = \vec{v}_{bc}$
- stress b.c.:
- open .b.c.: see fieldstone 29.

3.11.2 The heat transport equation

There are two types of boundary conditions for this equation: temperature boundary conditions (Dirichlet boundary conditions) and heat flux boundary conditions (Neumann boundary conditions).

3.12 Meaningful physical quantities

- Velocity \vec{v} (m/s): This is a vector quantity and both magnitude and direction are needed to define it. It is the rate of change of position with respect to a frame of reference.
- Root mean square velocity v_{rms} (m/s):

$$v_{rms} = \left(\frac{\int_{\Omega} |\vec{v}|^2 d\Omega}{\int_{\Omega} d\Omega} \right)^{1/2} = \left(\frac{1}{V_{\Omega}} \int_{\Omega} |\vec{v}|^2 d\Omega \right)^{1/2} \quad (62)$$

Remark. V_{Ω} is usually computed numerically at the same time v_{rms} is computed.

In Cartesian coordinates, for a cuboid domain of size $L_x \times L_y \times L_z$, the v_{rms} is simply given by:

$$v_{rms} = \left(\frac{1}{L_x L_y L_z} \int_0^{L_x} \int_0^{L_y} \int_0^{L_z} (u^2 + v^2 + w^2) dx dy dz \right)^{1/2} \quad (63)$$

In the case of an annulus domain, although calculations are carried out in Cartesian coordinates, it makes sense to look at the radial velocity component v_r and the tangential velocity component v_{θ} , and their respective root mean square averages:

$$v_r|_{rms} = \left(\frac{1}{V_{\Omega}} \int_{\Omega} v_r^2 d\Omega \right)^{1/2} \quad (64)$$

$$v_{\theta}|_{rms} = \left(\frac{1}{V_{\Omega}} \int_{\Omega} v_{\theta}^2 d\Omega \right)^{1/2} \quad (65)$$

- Pressure p (Pa):
- Stress tensor σ (Pa):
- Strain tensor ϵ (dimensionless):
- Strain rate tensor $\dot{\epsilon}$ (s⁻¹):
- Rayleigh number Ra (X): It is a dimensionless number that expresses the ratio of the driving forces to the opposing forces. The buoyancy force comes from the volumetric thermal expansion while the viscous forces and the heat diffusivity oppose convection (the latter one smoothes out thermal gradients).

The Rayleigh number for convection driven by a constant temperature hot base and a cold surface in a domain of thickness D is:

$$Ra = \frac{\rho_0 g \alpha D^3}{\eta \kappa} \cdot \Delta T = \frac{\rho_0^2 C_p g \alpha D^3 \Delta T}{\eta k}$$

The Rayleigh number for convection driven by a hot base (constant basal heat flow q_b) and a colder surface is:

$$Ra = \frac{\rho_0 g \alpha D^3}{\eta \kappa} \cdot \frac{q_b D}{k}$$

The Rayleigh number for convection driven by internal heating H (production per cubic meter) is:

$$Ra = \frac{\rho_0 g \alpha D^3}{\eta \kappa} \cdot \frac{H D^2}{k}$$

The Rayleigh number for convection driven by both basal heat flow and internal heating is:

$$Ra = \frac{\rho_0 g \alpha D^3}{\eta \kappa} \cdot \frac{q_b D + H D^2}{k}$$

For convection to occur, the Rayleigh number must be larger than the so-called critical Rayleigh number, which ranges from 600 to 3000 (it depends on the boundary conditions and the geometry).

- Prandtl number Pr (X): It is named after the German physicist Ludwig Prandtl and is defined as the ratio of momentum diffusivity to thermal diffusivity. It is given as:

$$Pr = \frac{\text{momentum diffusivity}}{\text{thermal diffusivity}} = \frac{\eta/\rho}{k/(\rho C_p)} = \frac{\eta C_p}{k}$$

For Earth materials, we have $Pr \sim (10^{21} 1000)/3 \gg 1$, which means that momentum diffusivity dominates.

- Nusselt number N_u (X): the Nusselt number (Nu) is the ratio of convective to conductive heat transfer across (normal to) the boundary. The conductive component is measured under the same conditions as the heat convection but with a (hypothetically) stagnant (or motionless) fluid.

In practice the Nusselt number Nu of a layer (typically the mantle of a planet) is defined as follows:

$$\text{Nu} = \frac{q}{q_c} \quad (66)$$

where q is the heat transferred by convection while $q_c = k\Delta T/D$ is the amount of heat that would be conducted through a layer of thickness D with a temperature difference ΔT across it with k being the thermal conductivity.

For 2D Cartesian systems of size (L_x, L_y) the Nu is computed [175]

$$\text{Nu} = \frac{\frac{1}{L_x} \int_0^{L_x} k \frac{\partial T}{\partial y}(x, y = L_y) dx}{-\frac{1}{L_x} \int_0^{L_x} k T(x, y = 0) / L_y dx} = -L_y \frac{\int_0^{L_x} \frac{\partial T}{\partial y}(x, y = L_y) dx}{\int_0^{L_x} T(x, y = 0) dx}$$

i.e. it is the mean surface temperature gradient over the mean bottom temperature.

finish, not happy with definition. Look at literature

Note that in the case when no convection takes place then the measured heat flux at the top is the one obtained from a purely conductive profile which yields $\text{Nu}=1$.

Note that a relationship $\text{Ra} \propto \text{Nu}^\alpha$ exists between the Rayleigh number Ra and the Nusselt number Nu in convective systems, see [1949] and references therein.

Turning now to cylindrical geometries with inner radius R_1 and outer radius R_2 , we define $f = R_1/R_2$. A small value of f corresponds to a high degree of curvature. We assume now that $R_2 - R_1 = 1$, so that $R_2 = 1/(1-f)$ and $R_1 = f/(1-f)$. Following [992], the Nusselt number at the inner and outer boundaries are:

$$\text{Nu}_{inner} = \frac{f \ln f}{1-f} \frac{1}{2\pi} \int_0^{2\pi} \left(\frac{\partial T}{\partial r} \right)_{r=R_1} d\theta \quad (67)$$

$$\text{Nu}_{outer} = \frac{\ln f}{1-f} \frac{1}{2\pi} \int_0^{2\pi} \left(\frac{\partial T}{\partial r} \right)_{r=R_2} d\theta \quad (68)$$

Note that a conductive geotherm in such an annulus between temperatures T_1 and T_2 is given by

$$T_c(r) = \frac{\ln(r/R_2)}{\ln(R_1/R_2)} = \frac{\ln(r(1-f))}{\ln f}$$

so that

$$\frac{\partial T_c}{\partial r} = \frac{1}{r \ln f}$$

We then find:

$$\text{Nu}_{inner} = \frac{f \ln f}{1-f} \frac{1}{2\pi} \int_0^{2\pi} \left(\frac{\partial T_c}{\partial r} \right)_{r=R_1} d\theta = \frac{f \ln f}{1-f} \frac{1}{R_1 \ln f} = 1 \quad (69)$$

$$\text{Nu}_{outer} = \frac{\ln f}{1-f} \frac{1}{2\pi} \int_0^{2\pi} \left(\frac{\partial T_c}{\partial r} \right)_{r=R_2} d\theta = \frac{\ln f}{1-f} \frac{1}{R_2 \ln f} = 1 \quad (70)$$

As expected, the recovered Nusselt number at both boundaries is exactly 1 when the temperature field is given by a steady state conductive geotherm.

[derive formula for Earth size R1 and R2](#)

- Temperature (K):
- Viscosity (Pa.s):
- Density (kg/m³):
- Heat capacity C_p (J.K⁻¹): It is the measure of the heat/energy required to increase the temperature of a unit quantity of a substance by unit degree. Note that the specific heat capacity c_p of a substance is the heat capacity of a sample of the substance divided by the mass of the sample, with units $J \cdot K \cdot kg^{-1}$.
- Heat conductivity, or thermal conductivity k (W.m⁻¹.K⁻¹). It is the property of a material that indicates its ability to conduct heat. It appears primarily in Fourier's Law for heat conduction. Note that it is a function of temperature, especially in mantle convection settings [1329]. See also [189] and refs therein. Note also that it can be a tensorial quantity in anisotropic context.
- Heat diffusivity: $\kappa = k/(\rho C_p)$ (m².s⁻¹). Substances with high thermal diffusivity rapidly adjust their temperature to that of their surroundings, because they conduct heat quickly in comparison to their volumetric heat capacity or 'thermal bulk'.
- thermal expansion α (K⁻¹): it is the tendency of a matter to change in volume in response to a change in temperature. Note that it is a function of temperature, especially in mantle convection settings [1329].
- Urey Ratio: mantle heat production divided by heat loss) is a key constraint for thermal history models. Recent Urey ratio estimates are in the range of 0.210.49. [1159]

[check aspect manual The 2D cylindrical shell benchmarks by Davies et al. 5.4.12](#)

3.13 Principal stress

As seen before (see Section 3.2.1) the stress tensor is a symmetric 3×3 real matrix, and linear algebra tells us that it therefore has three mutually orthogonal unit-length eigenvectors $\vec{n}_1, \vec{n}_2, \vec{n}_3$ and three real eigenvalues $\lambda_1, \lambda_2, \lambda_3$ such that $\sigma \cdot \vec{n}_i = \lambda_i \vec{n}_i$.

As a consequence, in a coordinate system with axes $\vec{n}_1, \vec{n}_2, \vec{n}_3$, the stress tensor is a diagonal matrix, and has only the three normal components $\lambda_1, \lambda_2, \lambda_3$ i.e. the principal stresses. If the three eigenvalues are equal, the stress is an isotropic compression or tension, always perpendicular to any surface, there is no shear stress, and the tensor is a diagonal matrix in any coordinate frame.

3.13.1 In two dimensions

We are looking for the stress tensor eigenvector vector $\vec{n} = (n_x, n_y)$ associated to the eigenvalue λ such that

$$\begin{pmatrix} \sigma_{xx} & \sigma_{xy} \\ \sigma_{xy} & \sigma_{yy} \end{pmatrix} \cdot \begin{pmatrix} n_x \\ n_y \end{pmatrix} = \lambda \begin{pmatrix} n_x \\ n_y \end{pmatrix}$$

or,

$$\begin{pmatrix} \sigma_{xx} & \sigma_{xy} \\ \sigma_{xy} & \sigma_{yy} \end{pmatrix} \cdot \begin{pmatrix} n_x \\ n_y \end{pmatrix} - \begin{pmatrix} \lambda & 0 \\ 0 & \lambda \end{pmatrix} \cdot \begin{pmatrix} n_x \\ n_y \end{pmatrix} = \vec{0}$$

i.e.,

$$\begin{pmatrix} \sigma_{xx} - \lambda & \sigma_{xy} \\ \sigma_{xy} & \sigma_{yy} - \lambda \end{pmatrix} \cdot \begin{pmatrix} n_x \\ n_y \end{pmatrix} = \vec{0}$$

which yields

$$(\sigma_{xx} - \lambda)(\sigma_{yy} - \lambda) - \sigma_{xy}^2 = 0$$

or,

$$\lambda^2 - (\sigma_{xx} + \sigma_{yy})\lambda + (\sigma_{xx}\sigma_{yy} - \sigma_{xy}^2) = 0$$

The discriminant Δ is

$$\begin{aligned}\Delta &= (\sigma_{xx} + \sigma_{yy})^2 - 4(\sigma_{xx}\sigma_{yy} - \sigma_{xy}^2) \\ &= (\sigma_{xx} - \sigma_{yy})^2 + 4\sigma_{xy}^2\end{aligned}$$

The roots are given by:

$$\begin{aligned}\lambda_{\pm} &= \frac{(\sigma_{xx} + \sigma_{yy}) \pm \sqrt{(\sigma_{xx} - \sigma_{yy})^2 + 4\sigma_{xy}^2}}{2} \\ &= \frac{\sigma_{xx} + \sigma_{yy}}{2} \pm \sqrt{\left(\frac{\sigma_{xx} - \sigma_{yy}}{2}\right)^2 + \sigma_{xy}^2}\end{aligned}\tag{71}$$

The two principal stresses are then:

$$\begin{aligned}\sigma_1 &= \frac{\sigma_{xx} + \sigma_{yy}}{2} + \sqrt{\left(\frac{\sigma_{xx} - \sigma_{yy}}{2}\right)^2 + \sigma_{xy}^2} \\ \sigma_2 &= \frac{\sigma_{xx} + \sigma_{yy}}{2} - \sqrt{\left(\frac{\sigma_{xx} - \sigma_{yy}}{2}\right)^2 + \sigma_{xy}^2}\end{aligned}\tag{72}$$

with the convention $\sigma_1 > \sigma_2$. The maximum shear stress is defined as one-half the difference between the two principal stresses

$$\tau_{max} = \frac{\sigma_1 - \sigma_2}{2} = \sqrt{\left(\frac{\sigma_{xx} - \sigma_{yy}}{2}\right)^2 + \sigma_{xy}^2}$$

The eigenvector \vec{n}_1 corresponding to σ_1 is obtained by solving

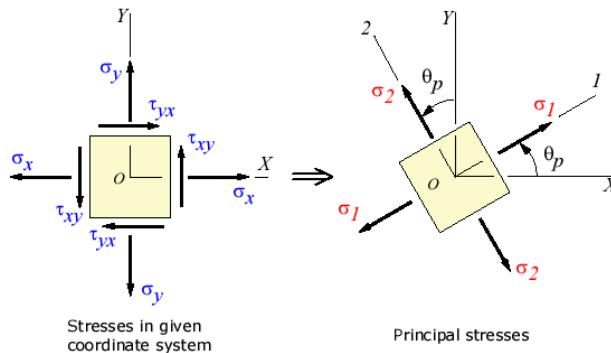
$$\boldsymbol{\sigma} \cdot \vec{n}_1 = \sigma_1 \vec{n}_1$$

and same for the other eigenvalue/vector:

$$\boldsymbol{\sigma} \cdot \vec{n}_2 = \sigma_2 \vec{n}_2$$

Each is a system of two equations with two unknowns. These are not difficult to solve, but can prove cumbersome. Note that linear algebra tells us that $\vec{n}_1 \cdot \vec{n}_2 = 0$, i.e. the eigenvectors form a basis of \mathbb{R}^2 .

This is the reason why often people go another route. One can ask the question: what is the value of the angle θ_p which, if used to perform a rotation of the axis system, yields a stress tensor that is diagonal, with the principal stresses on the diagonal?



Taken from https://www.efunda.com/formulae/solid_mechanics/mat_mechanics/plane_stress_principal.cfm

The rotation matrix is

$$\mathbf{R} = \begin{pmatrix} \cos \theta_p & -\sin \theta_p \\ \sin \theta_p & \cos \theta_p \end{pmatrix}$$

and the image of σ by means of the axis rotation is $\sigma' = \mathbf{R} \cdot \sigma \cdot \mathbf{R}^{-1}$, i.e.

$$\begin{aligned} \sigma' &= \begin{pmatrix} \cos \theta_p & -\sin \theta_p \\ \sin \theta_p & \cos \theta_p \end{pmatrix} \cdot \begin{pmatrix} \sigma_{xx} & \sigma_{xy} \\ \sigma_{xy} & \sigma_{yy} \end{pmatrix} \cdot \begin{pmatrix} \cos \theta_p & \sin \theta_p \\ -\sin \theta_p & \cos \theta_p \end{pmatrix} \\ &= \begin{pmatrix} \cos \theta_p & -\sin \theta_p \\ \sin \theta_p & \cos \theta_p \end{pmatrix} \cdot \begin{pmatrix} \sigma_{xx} \cos \theta_p - \sigma_{xy} \sin \theta_p & \sigma_{xx} \sin \theta_p + \sigma_{xy} \cos \theta_p \\ \sigma_{xy} \cos \theta_p - \sigma_{yy} \sin \theta_p & \sigma_{xy} \sin \theta_p + \sigma_{yy} \cos \theta_p \end{pmatrix} \\ &= \begin{pmatrix} \dots & \cos \theta_p (\sigma_{xx} \sin \theta_p + \sigma_{xy} \cos \theta_p) - \sin \theta_p (\sigma_{xy} \sin \theta_p + \sigma_{yy} \cos \theta_p) \\ \dots & \dots \end{pmatrix} \end{aligned}$$

In the matrix above I have only computed the off diagonal term since we are actually looking for θ_p such that $\sigma'_{xy} = 0$, or

$$\begin{aligned} \cos \theta_p (\sigma_{xx} \sin \theta_p + \sigma_{xy} \cos \theta_p) - \sin \theta_p (\sigma_{xy} \sin \theta_p + \sigma_{yy} \cos \theta_p) &= 0 \\ \sin \theta_p \cos \theta_p (\sigma_{xx} - \sigma_{yy}) + (\cos^2 \theta_p - \sin^2 \theta_p) \sigma_{xy} &= 0 \end{aligned}$$

and then

$$\frac{\sin \theta_p \cos \theta_p}{\cos^2 \theta_p - \sin^2 \theta_p} = \frac{\sigma_{xy}}{\sigma_{xx} - \sigma_{yy}}$$

The left hand term is actually a trigonometric identity¹¹:

$$\frac{\sin \theta_p \cos \theta_p}{\cos^2 \theta_p - \sin^2 \theta_p} = \frac{1}{2} \tan 2\theta_p$$

and finally:

$$\tan 2\theta_p = \frac{2\sigma_{xy}}{\sigma_{xx} - \sigma_{yy}} \quad \text{or} \quad \theta_p = \frac{1}{2} \tan^{-1} \frac{2\sigma_{xy}}{\sigma_{xx} - \sigma_{yy}}$$

Once θ_p has been found the other direction is given by $\theta_p + \pi/2$.

Example: Let us start with

$$\sigma = \begin{pmatrix} a & 0 \\ 0 & b \end{pmatrix}$$

then $\tan 2\theta_p = 0$, and then $\theta_p = 0$. The principal directions are the horizontal and vertical directions, i.e. the Cartesian axis system, which is consistent.

3.13.2 In three dimensions

We are looking for the stress tensor eigenvector vector $\vec{n} = (n_x, n_y, n_z)$ associated to the eigenvalue λ such that

$$\begin{pmatrix} \sigma_{xx} & \sigma_{xy} & \sigma_{xz} \\ \sigma_{xy} & \sigma_{yy} & \sigma_{yz} \\ \sigma_{xz} & \sigma_{yz} & \sigma_{zz} \end{pmatrix} \cdot \begin{pmatrix} n_x \\ n_y \\ n_z \end{pmatrix} = \lambda \begin{pmatrix} n_x \\ n_y \\ n_z \end{pmatrix}$$

or,

$$\begin{aligned} \begin{pmatrix} \sigma_{xx} & \sigma_{xy} & \sigma_{xz} \\ \sigma_{xy} & \sigma_{yy} & \sigma_{yz} \\ \sigma_{xz} & \sigma_{yz} & \sigma_{zz} \end{pmatrix} \cdot \begin{pmatrix} n_x \\ n_y \\ n_z \end{pmatrix} - \begin{pmatrix} \lambda & 0 & 0 \\ 0 & \lambda & 0 \\ 0 & 0 & \lambda \end{pmatrix} \cdot \begin{pmatrix} n_x \\ n_y \\ n_z \end{pmatrix} &= \vec{0} \\ \begin{pmatrix} \sigma_{xx} - \lambda & \sigma_{xy} & \sigma_{xz} \\ \sigma_{xy} & \sigma_{yy} - \lambda & \sigma_{yz} \\ \sigma_{xz} & \sigma_{yz} & \sigma_{zz} - \lambda \end{pmatrix} \cdot \begin{pmatrix} n_x \\ n_y \\ n_z \end{pmatrix} &= \vec{0} \end{aligned}$$

¹¹https://en.wikipedia.org/wiki/List_of_trigonometric_identities

Non-trivial solutions of this equation require

$$\begin{vmatrix} \sigma_{xx} - \lambda & \sigma_{xy} & \sigma_{xz} \\ \sigma_{yx} & \sigma_{yy} - \lambda & \sigma_{yz} \\ \sigma_{zx} & \sigma_{zy} & \sigma_{zz} - \lambda \end{vmatrix} = 0$$

Expanding the determinant results in the cubic equation

$$\lambda^3 - K_1\lambda^2 + K_2\lambda - K_3 = 0 \quad (73)$$

where:

$$\begin{aligned} K_1 &= \sigma_{xx} + \sigma_{yy} + \sigma_{zz} \\ K_2 &= \sigma_{xx}\sigma_{yy} + \sigma_{yy}\sigma_{zz} + \sigma_{zz}\sigma_{xx} - \sigma_{xy}\sigma_{yx} - \sigma_{xz}\sigma_{zx} - \sigma_{yz}\sigma_{zy} \\ K_3 &= \det(\boldsymbol{\sigma}) \end{aligned}$$

Very often we will find ourselves interested in the principal components of the deviatoric stress tensor \mathbf{s} . In this case the cubic equation becomes

$$\lambda^3 - J_2\lambda - J_3 = 0 \quad (74)$$

where J_2 and J_3 are the second and third invariants of deviatoric stress tensor \mathbf{s} .

Noting the trigonometric identity

$$\sin^3 \theta - \frac{3}{4} \sin \theta + \frac{1}{4} \sin 3\theta = 0 \quad (75)$$

and substituting $\lambda = r \sin \theta$ into (74) we have

$$\sin^3 \theta - \frac{J_2}{r^2} \sin \theta - \frac{J_3}{r^3} = 0 \quad (76)$$

Comparing (75) and (76) gives

$$r = \frac{2}{\sqrt{3}} \sqrt{J_2} \quad (77)$$

$$\sin 3\theta = -\frac{4J_3}{r^3} = -\frac{3\sqrt{3}}{2} \frac{J_3}{(J_2)^{3/2}} \quad (78)$$

The first root of (78) with θ determined for 3θ in the range $\pm\pi/2$ is a convenient alternative to the third invariant, J_3 . By noting the cyclic nature of $\sin(3\theta + 2n\pi)$ we have immediately the three (and only three) possible values of $\sin \theta$ which define the three principal stresses. The deviatoric principal stresses are given by $t = r \sin \theta$ on substitution of the three values of $\sin \theta$ in turn. Substituting for r from (77) and adding the mean hydrodynamic stress component gives the total principal stresses to be

The so-called Lode angle [2031] is then given by

$$\theta = \frac{1}{3} \sin^{-1} \left(-\frac{3\sqrt{3}}{2} \frac{J_3}{J_2^{3/2}} \right)$$

with $-\pi/6 < \theta < \pi/6$. Note that the Lode angle is one of the Lode coordinates¹², or HaighWestergaard coordinates.

SOMETHING missing HERE

$$\begin{Bmatrix} \sigma_1 \\ \sigma_2 \\ \sigma_3 \end{Bmatrix} = \sqrt{J_2} \begin{Bmatrix} \cos \theta + \frac{1}{\sqrt{3}} \sin \theta \\ \frac{2}{\sqrt{3}} \sin \theta \\ \frac{1}{\sqrt{3}} \sin \theta - \cos \theta \end{Bmatrix} + \frac{J_1}{3} \begin{Bmatrix} 1 \\ 1 \\ 1 \end{Bmatrix} \quad (79)$$

¹²https://en.wikipedia.org/wiki/Lode_coordinates

with $\sigma_1 > \sigma_2 > \sigma_3$.

We will later need $\sigma_1 - \sigma_3$ and $\sigma_1 + \sigma_3$ so let's compute these quantities now:

$$\begin{aligned}\sigma_1 - \sigma_3 &= \left(\frac{1}{3}J_1 + \sqrt{J_2}(\cos\theta + \frac{1}{\sqrt{3}}\sin\theta) \right) - \left(\frac{1}{3}J_1 + \sqrt{J_2}(\frac{1}{\sqrt{3}}\sin\theta - \cos\theta) \right) \\ &= \sqrt{J_2} \left((\cos\theta + \frac{1}{\sqrt{3}}\sin\theta) - (\frac{1}{\sqrt{3}}\sin\theta - \cos\theta) \right) \\ &= 2\sqrt{J_2} \cos\theta\end{aligned}\tag{80}$$

$$\begin{aligned}\sigma_1 + \sigma_3 &= \left(\frac{1}{3}J_1 + \sqrt{J_2}(\cos\theta + \frac{1}{\sqrt{3}}\sin\theta) \right) + \left(\frac{1}{3}J_1 + \sqrt{J_2}(\frac{1}{\sqrt{3}}\sin\theta - \cos\theta) \right) \\ &= \frac{2}{3}J_1 + \sqrt{J_2} \frac{2}{\sqrt{3}} \sin\theta\end{aligned}\tag{81}$$

Note that the expression for the Lode angle is different in [2035, p101] than in [2031] or [2034, p62]

3.14 The need for numerical modelling

The governing equations we have seen in this chapter require the use of numerical solution techniques for three main reasons:

- the advection term in the energy equation couples velocity and temperature;
- the constitutive law (the relationship between stress and strain rate) often depends on velocity (or rather, strain rate), temperature, pressure, ...
- Even when the coefficients of the PDE's are linear, often their spatial variability, coupled to potentially complex domain geometries prevent arriving at the analytical solution.

3.15 Rheology in geodynamics

The reader is referred to Barnes [79] for a discussion and review of non-linear viscous rheologies and to Coussot [440] for a review of experimental data for yield stress fluid flows.

Here is a quick recap of notations:

σ	(full) stress tensor
$\sigma_1, \sigma_2, \sigma_3$	principal stresses
τ	deviatoric stress tensor
τ_1, τ_2, τ_3	principal stresses
$\mathcal{I}_1(\mathbf{T})$	first invariant of tensor \mathbf{T}
$\mathcal{I}_2(\mathbf{T})$	first invariant of tensor \mathbf{T}
$\mathcal{I}_3(\mathbf{T})$	first invariant of tensor \mathbf{T}
$\tau_e = \sqrt{\mathcal{I}_2(\boldsymbol{\tau})}$	effective deviatoric stress
$\dot{\varepsilon}_e = \sqrt{\mathcal{I}_2(\boldsymbol{\varepsilon}^d)}$	effective deviatoric strain rate

Note that for incompressible flow, the stress tensor is given by $\sigma = -p\mathbf{1} + \tau$ so that $\mathcal{I}_1(\sigma) = -p\mathcal{I}_1(\mathbf{1}) + \mathcal{I}_1(\tau)$. Since τ is deviatoric, its first invariant is zero. We then have $\mathcal{I}_1(\sigma) = -pn_D$ where n_D is the number of dimensions.

3.15.1 Linear viscous aka Newtonian

Simply put, a Newtonian fluid is a fluid in which the viscous stresses at every point are linearly proportional to the local strain rate. Mathematically speaking, this means that the fourth-order tensor \mathbf{C} relating the viscous stress tensor to the strain rate tensor does not depend on the stress state and velocity of the flow.

$$\tau = \mathbf{C} : \dot{\varepsilon}$$

One very often make the assumption that the fluid is isotropic, i.e. its mechanical properties are the same along any direction. As a consequence the fourth order viscosity tensor \mathbf{C} is symmetric and will have only two independent real parameters: a bulk viscosity coefficient, that defines the resistance of the medium to gradual uniform compression; and a dynamic viscosity coefficient η that expresses its resistance to gradual shearing, (we here neglect the so-called rotational viscosity coefficient which results from a coupling between the fluid flow and the rotation of the individual particles).

Rather logically we denote by non-Newtonian fluids with are not Newtonian, i.e. their viscosity (tensor) depends on stress. Such fluids are part of our daily life, e.g. honey, toothpaste, paint, blood, and shampoo. They are also sometimes denoted as Generalized Newtonian Fluid .

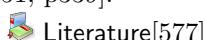
3.15.2 Power-law model

One of the simplest non-Newtonian viscosity model is the power-law model:

$$\eta = K \dot{\varepsilon}_e^{(n-1)/2} \quad (82)$$

where n and K are parameters. n is called the power-law index.

Note that a Newtonian viscosity is recovered when $n = 1$. Also n and K may depend on temperature [1551, p339].



Literature[577]

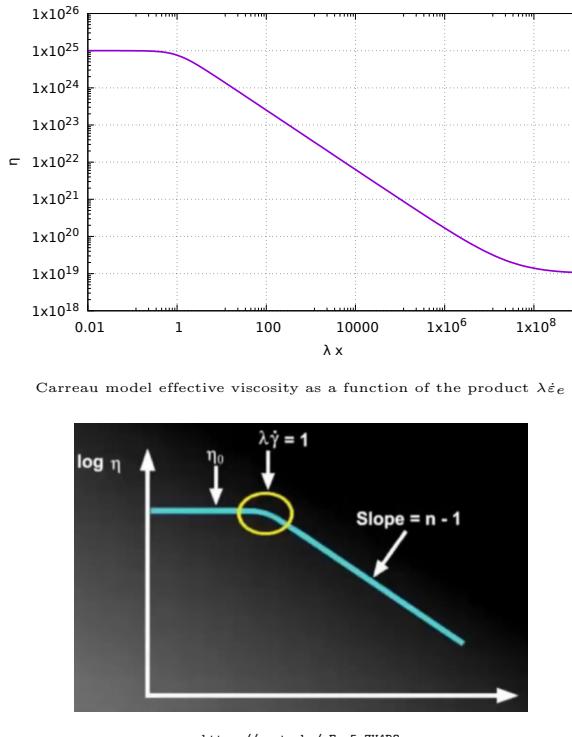
3.15.3 Carreau model

Note that this model is sometimes called Bird-Carreau in the literature. As explained in [1551], the power-law model poses no restriction on how small or large the viscosity may become, which may prove problematic once implemented as it can lead to runaway effects (strain rate becomes large \rightarrow viscosity becomes smaller \rightarrow strain rate becomes larger, etc ...). This problem is alleviated in the so-called Carreau¹³ model [2037]. The viscosity is then given by

$$\eta = \eta_\infty + (\eta_0 - \eta_\infty) (1 + (\lambda \dot{\varepsilon}_e)^2)^{(n-1)/2} \quad (83)$$

where η_0 , η_∞ , λ and $n \in [0, 1]$ are material parameters. λ is called the relaxation time: it is the inverse of the shear rate at which the fluid changes from Newtonian to power-law behavior.

At low strain rate a Carreau fluid behaves as a Newtonian fluid with viscosity η_0 . At intermediate strain rates $\dot{\varepsilon}_e \lambda \sim 1$ a Carreau fluid behaves as a Power-law fluid. At high strain rate, a Carreau fluid behaves as a Newtonian fluid again with viscosity η_∞ .



Note that the (Bird)-Carreau-Yasuda model [1441] is very similar to the standard (Bird)-Carreau:

$$\eta = \eta_\infty + (\eta_0 - \eta_\infty) (1 + (\lambda \dot{\varepsilon}_e)^a)^{(n-1)/a} \quad (84)$$



3.15.4 Bingham model

Bingham fluids can sustain an applied stress without any motion occurring. Only when the applied stress exceeds a yield stress τ_0 then the fluid flows. This translates as follows [1551]:

$$\tau = \left(\frac{\tau_0}{\dot{\varepsilon}} + 2\eta_0 \right) \dot{\varepsilon}^d \quad \text{if } \tau_e > \tau_0 \quad (85)$$

$$\tau = 0 \quad \text{if } \tau_e \leq \tau_0 \quad (86)$$

¹³https://en.wikipedia.org/wiki/Carreau_fluid

When flow occurs, the effective viscosity is then given by:

$$\eta_{eff} = \frac{\tau_0}{\dot{\varepsilon}_e} + 2\eta_0$$

and when the strain rate is large we recover a Newtonian behaviour. Typical Bingham fluids are mud, slurry, toothpaste.

When using a velocity-based FEM code, the implementation of this rheological behaviour is complicated by the no-flow condition under a given stress. However, our codes require a relationship between stress and strain rate in the form of an effective viscosity which cannot be zero. This difficulty can be circumvented by implementing Bingham fluids as follows [1551]:

$$\tau = \left(\frac{\tau_0(1 - \eta/\eta_r)}{\dot{\varepsilon}_e} + 2\eta_0 \right) \dot{\varepsilon} \quad \text{if } \tau_e > \tau_0 \quad (87)$$

$$\tau = 2\eta_r \dot{\varepsilon} \quad \text{if } \tau_e \leq \tau_0 \quad (88)$$

where η_r is a pre-yield viscosity and $\eta/\eta_r \ll 1$ (typically 1% or less). This is a form of regularisation, and we will see a similar one in the next section.

 Literature: [1450, 173, 1327, 1208, 1247, 1709, 165, 75, 147]

3.15.5 Herschel-Bulkley visco-plastic model

The Herschel-Bulkley model is effectively a combination of the power-law model and a simple plastic model:

$$s = 2 \left(K \dot{\varepsilon}_e^{n-1} + \frac{\tau_0}{\dot{\varepsilon}} \right) \dot{\varepsilon} \quad \text{if } \tau_e > \tau_0 \quad (89)$$

$$\dot{\varepsilon} = \mathbf{0} \quad \text{if } \tau_e \leq \tau_0 \quad (90)$$

in which τ_0 is the yield stress, K the consistency, and n is the flow index [501]. The flow index measures the degree to which the fluid is shear-thinning ($n < 1$) or shear-thickening ($n > 1$). If $n = 1$ and $\tau_0 = 0$ the model reduces to the Newtonian model.

The term between parenthesis above is the nonlinear effective viscosity. Concretely, the implementation goes as follows¹⁴:

$$\eta_{eff} = \begin{cases} \eta_0 & \dot{\varepsilon}_e \leq \dot{\varepsilon}_0 \\ K \dot{\varepsilon}_e^{n-1} + \frac{\tau_0}{\dot{\varepsilon}_e} & \dot{\varepsilon}_e \geq \dot{\varepsilon}_0 \end{cases}$$

The limiting viscosity η_0 is chosen such that $\eta_0 = K \dot{\varepsilon}_0^{n-1} + \frac{\tau_0}{\dot{\varepsilon}_0}$

A large limiting viscosity means that the fluid will only flow in response to a large applied force. This feature captures the Bingham-type behaviour of the fluid. Note that when strain rates are large, the power-law behavior dominates.

As we have seen for Bingham fluids, the equations above are not easily amenable to implementation so that one usually resorts to regularisation, which is a modification of the equations by introducing a new material parameter which controls the exponential growth of stress. This way the equation is valid for both yielded and unyielded areas [173, 1450, 2037]:

$$\eta_{eff} = K \dot{\varepsilon}_e^{n-1} + \frac{\tau_0}{\dot{\varepsilon}_e} [1 - \exp(-m\dot{\varepsilon}_e)] \quad (91)$$

When the strain rate becomes (very) small a Taylor expansion of the regularisation term yields $1 - \exp(-m\dot{\varepsilon}) \sim m\dot{\varepsilon}$ so that $\eta_{eff} \rightarrow m\tau_0$.

Channel flow of wikipedia with analytical solution!

 Literature: [1283]. Check dam break problem of [26, 403, 74, 1732, 1326].

(MSc Thesis): redo the sinking sphere in Bingham or herschel-Bulkley fluid experiment [501].

¹⁴https://en.wikipedia.org/wiki/Herschel-Bulkley_fluid

3.15.6 The Casson model

It is described by [79]

$$\sqrt{\sigma} = \sqrt{\sigma_y} + \sqrt{\eta_p \dot{\varepsilon}_e}$$

or, when squaring it:

$$\sigma = \sigma_y + \eta_p \dot{\varepsilon}_e + 2\sqrt{\sigma_y \eta_p \dot{\varepsilon}_e}$$

This model has been found to accurately describe the behaviour of synthetic based muds [4].

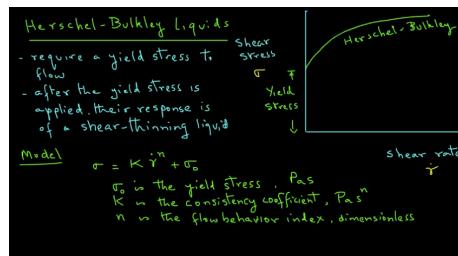
3.15.7 One model to rule them all?

Let us consider the base equation

$$\frac{\eta - \eta_\infty}{\eta_0 - \eta_\infty} = [1 + (K\dot{\varepsilon}_e)^a]^{-(1-n)/a}$$

This equation is purposefully generic and specific parameter combination choices allow to recover any of the above models (and more) [1441]. See also an early paper by Cross [453] for a somewhat similar equation.

Similar conclusions are reached in the following video:



3.15.8 Dislocation and Diffusion creep

The standard dislocation creep effective viscosity is given by:

$$\eta_{eff}^{ds} = \frac{1}{2} f A^{-1/n} \dot{\varepsilon}_e^{(1-n)/n} \exp\left(\frac{Q + pV}{nRT}\right)$$

where A is the pre-exponential scaling factor, f is a scaling factor representing viscous weakening or strengthening, Q is the activation energy, V is the activation volume, T is the absolute temperature, n is the power-law exponent, R is the universal gas constant.

The coefficients A, n, Q, V are material parameters and are obtained in the laboratory by means of high pressure/temperature experiments[1020]. Unfortunately these experiments cannot be run at Earth-like strain rate values ($\sim 10^{-15} \text{s}^{-1}$) so that extrapolations must be carried out over several orders of magnitude to arrive at values we can use in our numerical models. The 1/2 factor arises from the relationship between deviatoric stress and strain rate which involves a factor 2.

The factor f is in fact a tuning parameter used to explore end members (e.g. 'weak crust' vs 'strong crust'), see discussion in the supplementary material in [946]. This approach has been extensively used by the SOPALE users community, see for instance [1910, 1911, 1912, 807].

Furthermore, we know that several other factors will strongly affect the rheology:

- water content, or as often mentioned: 'dry' vs 'wet'. Following [1020], dry means water-free and wet means water-saturated conditions.

Mechanism	Dry	Wet
<i>Dislocation creep</i>		
A (s^{-1})	3.5×10^{22}	2.0×10^{18}
n	3.5	3.0
m	0	0
E^* (kJ mol^{-1})	540	430
V^* ($\text{cm}^3 \text{mol}^{-1}$)	15 to 25†	10 to 20†
<i>Diffusion creep</i>		
A (s^{-1})	8.7×10^{15}	5.3×10^{15}
n	1.0	1.0
m	2.5‡	2.5‡
E^* (kJ mol^{-1})	300	240
V^* ($\text{cm}^3 \text{mol}^{-1}$)	6§	5§

†The activation volume for dislocation creep is not well constrained. Values from $13 \text{ cm}^3 \text{mol}^{-1}$ at wet conditions to $27 \text{ cm}^3 \text{mol}^{-1}$ at nearly dry conditions have been reported [33]. Considering this uncertainty, we used a range of activation volumes, 15 to $25 \text{ cm}^3 \text{mol}^{-1}$, for dry olivine and 10 to $20 \text{ cm}^3 \text{mol}^{-1}$ for wet olivine. ‡The grain-size exponent is fixed at 2 for dry olivine and 3 for wet olivine [16]. However, during extrapolation to a large grain size, the grain-size exponent may change from 2 to 3 or from 3 to 2 [16]. Taking this uncertainty into account, we chose a grain-size exponent of 2.5, and the preexponential factors were modified accordingly. §The activation volume for diffusion is determined as $6 \text{ cm}^3 \text{mol}^{-1}$ at dry conditions [26]. No data are available for the activation volume for diffusion at wet conditions; we assumed that this value is $\sim 80\%$ of the activation volume at dry conditions.

Taken from Karato and Wu [1020].

 Literature: [1528] and refs therein for the effects of water migration on models of subduction dynamics.

- composition: while one typically assigns olivine properties to the mantle in models, the mineral olivine¹⁵ is actually a magnesium iron silicate with the formula $(\text{Mg}^{2+}, \text{Fe}^{2+})_2\text{SiO}_4$. and the ratio of magnesium to iron varies between the two endmembers of the solid solution series: forsterite (Mg-endmember: Mg_2SiO_4) and fayalite (Fe-endmember: Fe_2SiO_4).
- grain size: this only affects diffusion creep mechanisms [1020]. Grain size varies over several orders of magnitude and also evolves over time and its evolution is affected by the ambient deformation and the deformation history. Dannberg et al [470] then used a diffusion creep effective viscosity given by:

$$\eta_{eff}^{df} = \frac{1}{2} A_{df}^{-1} d^m \exp \left(\frac{Q_{df} + pV_{df}}{RT} \right)$$

where d is the (variable) grain size and m the grain size exponent. Grain growth/evolution is usually approximated using semi-empirical expressions [470, section 2.2]. Smaller grains facilitating faster creep.

The evolution of grain size in the lower mantle is discussed in detail by Solomatov [1676, 1679, 1681] and Cerpa et al [346].

 Literature: [1385]

- anisotropy, LPO:
- phase changes:

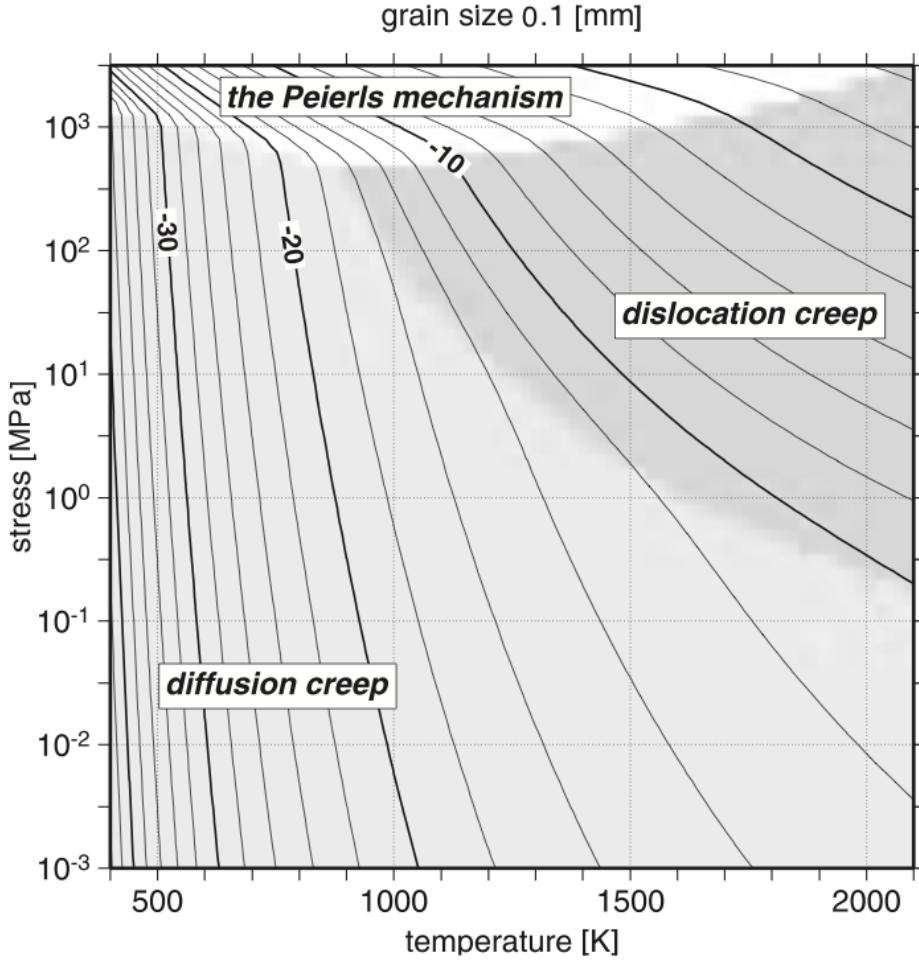
Remark. It is not uncommon to find in the literature effective viscosity formulations written as a function of B with $B = A^{-1/n}$ [1910, 1911, 1912]. Also, this B coefficient often contains the conversion factor of the next remark.

Remark. uniaxial stuff An additional coefficient is added to the effective viscosity formula(s ??) [807, 809].

$$3^{-(1+n)/2n} 2^{(1-n)/n}$$

[1539]

¹⁵<https://en.wikipedia.org/wiki/Olivine>



Deformation mechanism map calculated for grain size $a = 0.1\text{mm}$. The lightly shaded area indicates that deformation mainly occurs by diffusion creep. The densely shaded area indicates that deformation mainly occurs by power-law creep. The white region indicates that deformation mainly occurs by the Peierls mechanism. The solid curves are lines of constant strain rate. The numbers attached to each contour indicate the logarithm of the strain rate in the unit of s^{-1} [1014].

A closer look at the diffusion creep of Karato & Wu (1993) In the article, the following equation is used:

$$\dot{\varepsilon} = A \left(\frac{\tau}{\mu} \right) \left(\frac{b}{d} \right)^m \exp \left(-\frac{Q + pV}{RT} \right)$$

where μ is the shear modulus ($\sim 80\text{GPa}$), b is the length of the Burgers vector ($\sim 0.5\text{nm}$) and d is the grain size. One can express the above equation in terms of second invariants (see Section 8.11.1):

$$\dot{\varepsilon}_e = A \left(\frac{\tau_e}{\mu} \right) \left(\frac{b}{d} \right)^m \exp \left(-\frac{Q + pV}{RT} \right)$$

and assuming a Newtonian linearisation/relation between deviatoric stress and strain rate $\tau_e = 2\eta_{df}\dot{\varepsilon}_e$, one arrive at

$$\eta_{df} = \frac{1}{2} \left(\frac{A}{\mu} \right)^{-1} \left(\frac{b}{d} \right)^{-m} \exp \left(\frac{Q + pV}{RT} \right)$$

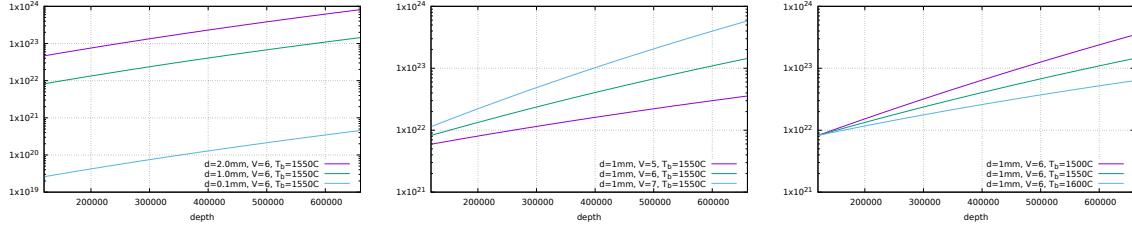
or,

$$\eta_{df} = \frac{1}{2} \left[\frac{A}{\mu} \left(\frac{b}{d} \right)^m \right]^{-1} \exp \left(\frac{Q + pV}{RT} \right)$$

The effective diffusion creep viscosity is independent of strain-rate so that one could substitute the total pressure for lithostatic pressure in the equation, assume a geotherm and easily compute the predicted viscosity as a function of grain size d .

Let us assume that the 1D profile starts from the base of the lithosphere (say 120km depth) and ends at the 660 boundary. Assume the temperature to increase linearly from 1300°C to T_{bottom} . At the bottom of the lithosphere, the lithostatic pressure is of the order of $\rho \cdot g \cdot L \simeq 3000 \cdot 10 \cdot 120e3 \simeq 4\text{GPa}$. At the bottom of the domain, the pressure has increased by $3300 \cdot 10 \cdot 630e3 \simeq 21\text{GPa}$.

The viscosity profile is plotted hereunder for three different grain sizes, bottom temperature and activation volumes ($4,5,6 \text{ cm}^3/\text{mol}$).



Although this exercise only provides us with first-order results, we can conclude that one can essentially change the diffusion creep effective viscosity by up to 2 orders of magnitude simply by choosing key parameters within acceptable ranges.

3.15.9 Combining diffusion and dislocation creep

It is rather common in the computational geodynamics to combine both viscosities into one so-called 'effective viscous viscosity' (as opposed to the so-called 'effective plastic viscosity'). The harmonic average is then used:

$$\eta_{eff} = \frac{\eta_{df}\eta_{ds}}{\eta_{df} + \eta_{ds}} = \left(\frac{1}{\eta_{df}} + \frac{1}{\eta_{ds}} \right)^{-1} \quad (92)$$

See for instance [1056].

However, although this formula seems simple enough, it hides an important fact: strain rates are supposed to be additive, i.e. the total strain rate $\dot{\varepsilon}_T$ of the deformation is the sum of the dislocation strain rate $\dot{\varepsilon}_{ds}$ and the diffusion strain rate $\dot{\varepsilon}_{df}$:

$$\dot{\varepsilon}_T = \dot{\varepsilon}_{df} + \dot{\varepsilon}_{ds}$$

and so are the effective strain rates (the underscore e has been dropped for convenience - see section 8.11.1 for definitions):

$$\dot{\varepsilon}_T = \dot{\varepsilon}_{df} + \dot{\varepsilon}_{ds}$$

i.e.,

$$\dot{\varepsilon}_T = A_{df}\tau \exp\left(-\frac{Q_{df} + pV_{df}}{RT}\right) + A_{ds}\tau^n \exp\left(-\frac{Q_{ds} + pV_{ds}}{RT}\right)$$

where τ is the effective shear stress.

We know all the coefficients in this equation with the exception of the scalar τ (the total strain rate is obtained from the previous nonlinear iteration velocity or previous time-step velocity) so that it can be cast under the form $f(\tau) = 0$

$$f(\tau) = \dot{\varepsilon}_T - A_{df}\tau \exp\left(-\frac{Q_{df} + pV_{df}}{RT}\right) - A_{ds}\tau^n \exp\left(-\frac{Q_{ds} + pV_{ds}}{RT}\right)$$

A Newton method can be used where and the iterations will be of the form:

$$\tau_{n+1} = \tau_n - \frac{f(\tau_n)}{f'(\tau_n)}$$

where the derivative of the function f with respect to τ reads:

$$f'(\tau) = \frac{\partial f}{\partial \tau} = -A_{df} \exp\left(-\frac{Q_{df} + pV_{df}}{RT}\right) - A_{ds}n\tau^{n-1} \exp\left(-\frac{Q_{ds} + pV_{ds}}{RT}\right)$$

Also, from the additivity of strain rates follows the harmonic averaging of viscosities:

$$\dot{\varepsilon}_T = \dot{\varepsilon}_{df} + \dot{\varepsilon}_{ds} \quad \Rightarrow \quad \frac{\tau}{2\eta_{eff}} = \frac{\tau}{2\eta_{df}} + \frac{\tau}{2\eta_{ds}} \quad \Rightarrow \quad \frac{1}{\eta_{eff}} = \frac{1}{\eta_{df}} + \frac{1}{\eta_{ds}}$$

so that the algorithm simply writes:

1. compute τ iteratively
2. compute $\dot{\varepsilon}_{df}$, $\dot{\varepsilon}_{ds}$
3. compute $\eta_{ds} = \tau/(2\dot{\varepsilon}_{ds})$, $\eta_{df} = \tau/(2\dot{\varepsilon}_{df})$
4. compute η_{eff}

Note that one can also directly compute η_{eff} as $\eta_{eff} = \frac{\tau}{2\dot{\varepsilon}_T}$ since

$$\eta_{eff} = \frac{\tau}{2\dot{\varepsilon}_T} = \frac{\tau}{2\dot{\varepsilon}_{ds} + 2\dot{\varepsilon}_{df}} = \frac{1}{2\dot{\varepsilon}_{ds}/\tau + 2\dot{\varepsilon}_{df}/\tau} = \frac{1}{1/\eta_{ds} + 1/\eta_{df}}$$

In the case where many deformation mechanisms are present, the total strain rate is seen as the sum of the strain rates associated to each mechanism:

$$\dot{\varepsilon}_T = \dot{\varepsilon}_1 + \dot{\varepsilon}_2 + \dot{\varepsilon}_3 + \dots$$

and then

$$\dot{\varepsilon}_T = \dot{\varepsilon}_1 + \dot{\varepsilon}_2 + \dot{\varepsilon}_3 + \dots \quad \Rightarrow \quad \frac{\tau}{2\eta_{eff}} = \frac{\tau}{2\eta_1} + \frac{\tau}{2\eta_2} + \frac{\tau}{2\eta_3} + \dots \quad \Rightarrow \quad \frac{1}{\eta_{eff}} = \frac{1}{\eta_1} + \frac{1}{\eta_2} + \frac{1}{\eta_3} + \dots$$

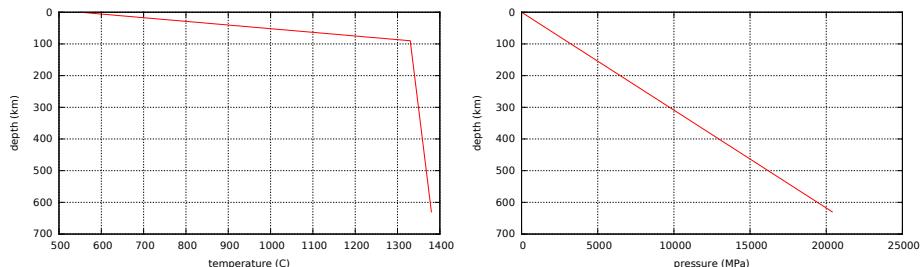
Rather importantly, as we will see hereafter, the following variant is implemented in some codes (e.g. DOUAR, FANTOM, SOPALE) so as to bypass the costly Newton iterations:

1. compute η_{ds} and η_{df} with the *same* strainrate, pressure and temperature values
2. average them by means of an harmonic average

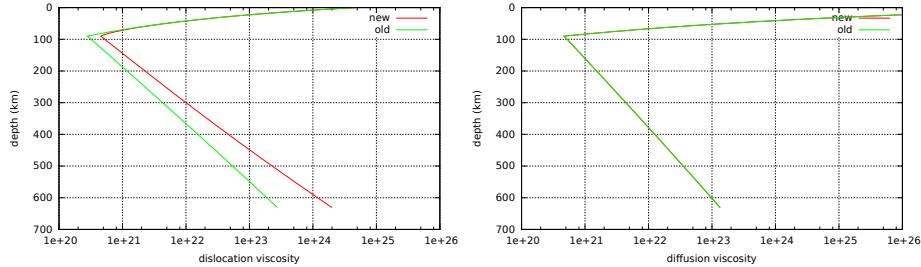
In this case, we have

$$\dot{\varepsilon}_T = A_{df}\tau_{df} \exp\left(-\frac{Q_{df} + pV_{df}}{RT}\right) \quad \dot{\varepsilon}_T = A_{ds}\tau_{ds}^n \exp\left(-\frac{Q_{ds} + pV_{ds}}{RT}\right)$$

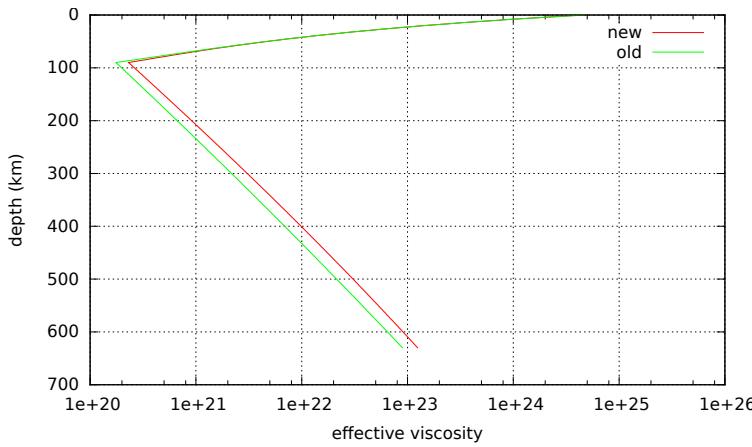
A concrete example Let us consider a vertical section of upper mantle, from 660km depth to 30km depth. The lithosphere is assumed to be 90km thick. The temperature at the moho (the top of the domain) is set to 550C, 1330C at the LMB and 1380C at the bottom. A constant strainrate $\dot{\varepsilon}$ is assumed. The density is set to a constant 3300kg/m³ and we assume a lithostatic pressure. The temperature and pressure fields are shown hereunder:



In what follows, the values obtained with Newton iterations are coined 'new' and those obtained without are coined 'old'. The diffusion and dislocation creep viscosities can be computed for both old and new algorithms and are shown hereunder (the diffusion creep viscosity is independent of strain rate so is the same for both):



Using the algorithm highlighted hereabove we can compute the composite effective viscosity and compare it with the 'standard' effective viscosity:



look at folder, codes, clean up, change labels new and old

3.15.10 The von Mises failure criterion

The von Mises yield criterion suggests that the yielding of materials begins when the second deviatoric stress invariant $\mathcal{I}_2(\boldsymbol{\tau})$ reaches a critical value. For this reason, it is sometimes called the J_2 -plasticity or J_2 flow theory¹⁶. It is part of a plasticity theory that applies best to ductile materials, such as metals.

In material science and engineering the von Mises yield criterion can be also formulated in terms of the von Mises stress or equivalent tensile stress, σ_v , a scalar stress value that can be computed from the stress tensor. In this case, a material is said to start yielding when its von Mises stress reaches a critical value known as the yield strength, σ_Y . The von Mises stress is used to predict yielding of materials under any loading condition from results of simple uniaxial tensile tests. The von Mises stress satisfies the property that two stress states with equal distortion energy have equal von Mises stress.

Because the von Mises yield criterion is independent of the first stress invariant, $\mathcal{I}_1(\boldsymbol{\sigma})$, it is applicable for the analysis of plastic deformation for ductile materials such as metals, as the onset of yield for these materials does not depend on the hydrostatic component of the stress tensor.

Although formulated by Maxwell in 1865, it is generally attributed to von Mises [1889]. Huber (1904), in a paper in Polish, anticipated to some extent this criterion. Heinrich Hencky formulated the same criterion as von Mises independently in 1924 [883, 1738]. This criterion is also referred to as the Maxwell-Huber-Hencky-von Mises theory.

The von Mises yield criterion (also known as Prandtl-Reuss yield criterion) is expressed in the principal stresses as

$$\sigma_v = \sqrt{3J'_2} = \sigma_Y \quad \text{or,} \quad (\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_3 - \sigma_1)^2 = 2\sigma_y^2$$

where σ_Y is the yield stress in uniaxial tension.

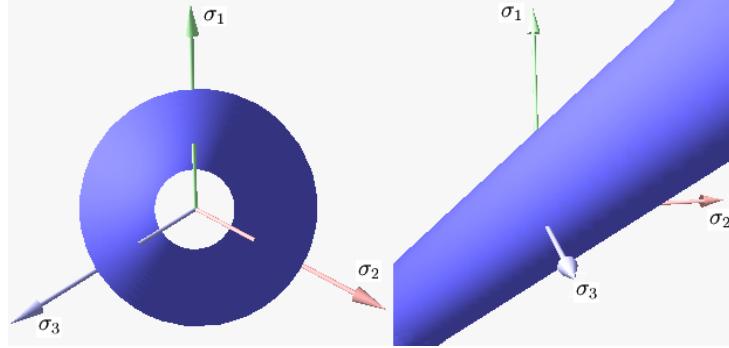
The von Mises yield criterion writes:

¹⁶ J_2 is the common notation for $\mathcal{I}_2(\boldsymbol{\tau})$

$$F^{VM} = \sqrt{J_2} - \frac{2c}{\sqrt{3}} \quad (93)$$

which is purposely the Drucker-Prager criterion with $\phi = 0$.

The following figure shows the von Mises yield surface in the three-dimensional space of principal stresses.



It is a circular cylinder of infinite length with its axis inclined at equal angles to the three principal stresses.

Literature: [1450, 1770]

3.15.11 The Tresca failure criterion

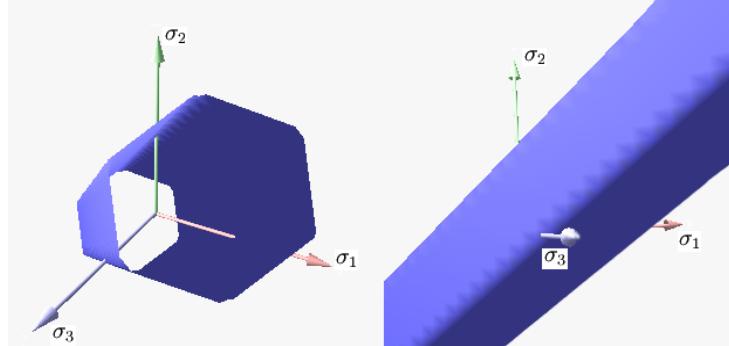
The Tresca or maximum shear stress yield criterion is taken to be the work of Henri Tresca. It is also referred as the Tresca-Guest (TG) criterion. The functional form of this yield criterion is

$$f(\sigma_1, \sigma_2, \sigma_3) = 0$$

In terms of the principal stresses the Tresca criterion is expressed as

$$\max(|\sigma_1 - \sigma_2|, |\sigma_2 - \sigma_3|, |\sigma_3 - \sigma_1|) = \sigma_0$$

The following figure shows the Tresca-Guest yield surface in the three-dimensional space of principal stresses.



It is a prism of six sides and having infinite length. This means that the material remains viscous when all three principal stresses are roughly equivalent (a hydrostatic pressure), no matter how much it is compressed or stretched. However, when one of principal stresses becomes smaller (or larger) than the others the material is subject to shearing. In such situations, if the shear stress reaches the yield limit then the material enters the plastic domain.

Remark. *The yield function is non-continuous, making its numerical implementation more difficult (directional derivatives are needed)*

two-dimensional space Using the values of the principal stresses in a two dimensional space, the criterion becomes

$$|\sigma_1 - \sigma_2| = \sqrt{J'_2} = \sigma_0$$

which is equivalent to the von Mises criterion.

three-dimensional space We have already established that

$$\sigma_1 - \sigma_3 = 2\sqrt{J_2} \cos \theta$$

with $\sigma_1 > \sigma_3$, so that the failure criterion is given by

$$F^{TR,3D} = 2\sqrt{J_2} \cos \theta - c$$

Obviously, the Tresca criterion corresponds to the case where the friction angle $\phi = 0$.

 Literature: [1770]

3.15.12 The Drucker-Prager failure criterion

The von Mises yield criterion is not suitable for modelling the yielding of frictional material as it does not include the effect of mean stress as observed in experiments. To overcome this limitation, Drucker and Prager [539] proposed a revised function for frictional materials.

The Drucker-Prager yield criterion has the function form

$$F(\boldsymbol{\sigma}) = F(\mathcal{I}_1(\boldsymbol{\sigma}), \mathcal{I}_2(\boldsymbol{\sigma})) = 0 \quad (94)$$

This criterion is most often used for concrete where both normal and shear stresses can determine failure. The Drucker-Prager yield criterion may be expressed as

$$F^{DP} = \tau_e - (\alpha \mathcal{I}_1(\boldsymbol{\sigma}) + k) \quad (95)$$

Since the Drucker-Prager yield surface is a smooth version of the Mohr-Coulomb yield surface, it is often expressed in terms of the cohesion c and the angle of internal friction ϕ that are used to describe the Mohr-Coulomb yield surface. One then wishes to relate α and k to c and ϕ : $\alpha = fct(\phi)$ and $k = fct(c, \phi)$.

two-dimensional space By choosing $k = c \cos \phi$ and $\alpha = \frac{1}{2} \sin \phi$ and seeing that that $p = \frac{1}{2} \mathcal{I}_1(\boldsymbol{\sigma})$, we can make the Drucker-Prager yield criterion coincide with the Mohr-Coulomb yield criterion so that

$$F^{DP,2D} = \tau_e - (p \sin \phi + c \cos \phi)$$

three-dimensional space Let us first assume that the Drucker-Prager yield surface circumscribes the Mohr-Coulomb yield surface such that the two surfaces coincide at $\theta = \frac{\pi}{3}$. The expression for the Mohr-Coulomb yield criterion is

$$F^{MC,3D} = -\frac{1}{3} J_1 \sin \phi + \sqrt{J_2} \left(\cos \theta - \frac{1}{\sqrt{3}} \sin \theta \sin \phi \right) - c \cos \phi$$

Taking $\theta = \pi/6$ yields:

$$\begin{aligned} F^{MC,3D} &= -\frac{1}{3} J_1 \sin \phi + \sqrt{J_2} \left(\frac{\sqrt{3}}{2} - \frac{1}{\sqrt{3}} \frac{1}{2} \sin \phi \right) - c \cos \phi \\ &= -\frac{1}{3} J_1 \sin \phi + \frac{1}{2\sqrt{3}} \sqrt{J_2} (3 - \sin \phi) - c \cos \phi \\ &= -\frac{1}{3} J_1 \sin \phi + \frac{1}{6} \sqrt{J_2} \sqrt{3} (3 - \sin \phi) - c \cos \phi \\ &= \frac{\sqrt{3}(3 - \sin \phi)}{6} \left(-\frac{2 \sin \phi}{\sqrt{3}(3 - \sin \phi)} J_1 + \sqrt{J_2} - \frac{6c \cos \phi}{\sqrt{3}(3 - \sin \phi)} \right) \end{aligned}$$

The constant in front of the brackets (which always strictly positive) does not matter since we look at the sign of F . Comparing with

$$F^{DP} = \sqrt{J_2} - (\alpha J_1 + k)$$

we naturally set

$$\alpha = \frac{2 \sin \phi}{\sqrt{3}(3 - \sin \phi)} \quad k = \frac{6c \cos \phi}{\sqrt{3}(3 - \sin \phi)}$$

and since $p = \frac{1}{3}\mathcal{I}_1(\boldsymbol{\sigma})$:

$$F^{DP,3D} = \tau_e - \left[\frac{6 \sin \phi}{\sqrt{3}(3 - \sin \phi)} p + \frac{6c \cos \phi}{\sqrt{3}(3 - \sin \phi)} \right]$$

which is the formula used in [766].

TO VERIFY If the Drucker-Prager surface inscribes the Mohr-Coulomb surface, then matching the two surfaces at $\theta = 0$ gives

$$k = \frac{6 c \cos \phi}{\sqrt{3}(3 - \sin \phi)} ; \quad \alpha = \frac{2 \sin \phi}{\sqrt{3}(3 - \sin \phi)}$$

In the principal stress space, the Drucker-Prager surface has the form of a circular cone, whilst the von Mises yield surface is an infinitely long cylinder.

3.15.13 The Mohr-Coulomb failure criterion

Mohr-Coulomb theory is a model describing the response of a material such as rubble piles or concrete to shear stress as well as normal stress. Most of the classical engineering materials somehow follow this rule in at least a portion of their shear failure envelope. In geology it is used to define shear strength of soils at different effective stresses.

In structural engineering it is used to determine failure load as well as the angle of fracture of a displacement fracture in concrete and similar materials. Coulomb's friction hypothesis is used to determine the combination of shear and normal stress that will cause a fracture of the material. Mohr's circle is used to determine which principal stresses that will produce this combination of shear and normal stress, and the angle of the plane in which this will occur. According to the principle of normality the stress introduced at failure will be perpendicular to the line describing the fracture condition.

The Mohr-Coulomb failure criterion represents the linear envelope that is obtained from a plot of the shear strength of a material versus the applied normal stress. This relation is expressed as

$$\tau = \sigma \tan(\phi) + c$$

where τ is the shear strength, σ is the normal stress, c is the intercept of the failure envelope with the τ axis, and ϕ is the slope of the failure envelope. The quantity c is often called the cohesion and the angle ϕ is called the angle of internal friction. Compression is assumed to be positive in the following discussion. If compression is assumed to be negative then σ should be replaced with $-\sigma$.

From Mohr's circle we have

$$\sigma = \sigma_m - \tau_m \sin \phi \quad \tau = \tau_m \cos \phi$$

where

$$\tau_m = \frac{\sigma_1 - \sigma_3}{2} ; \quad \sigma_m = \frac{\sigma_1 + \sigma_3}{2}$$

and σ_1 is the maximum principal stress and σ_3 is the minimum principal stress. Therefore the Mohr-Coulomb criterion may also be expressed as

$$\tau_m = \sigma_m \sin \phi + c \cos \phi$$

The Mohr-Coulomb failure (yield) surface is often expressed in Haigh-Westergaard coordinates. For example, the last equation rewrites

$$\frac{\sigma_1 - \sigma_3}{2} = \frac{\sigma_1 + \sigma_3}{2} \sin \phi + c \cos \phi$$

two-dimensional space The principal stress values are given by

$$\sigma_{1,2} = \frac{\sigma_{xx} + \sigma_{yy}}{2} \pm \sqrt{\frac{1}{4}(\sigma_{xx} - \sigma_{yy})^2 + \sigma_{xy}^2} = \frac{J_1}{2} \pm \sqrt{J_2}$$

The Mohr-Coulomb criterion simply writes:

$$F^{MC,2D} = -\frac{J_1}{2} \sin \phi + \sqrt{J_2} - c \cos \phi$$

three-dimensional space We have already established that

$$\sigma_1 - \sigma_3 = 2\sqrt{J_2} \cos \theta \quad \sigma_1 + \sigma_3 = \frac{2}{3}J_1 + \sqrt{J_2} \frac{2}{\sqrt{3}} \sin \theta$$

$$\begin{aligned} \frac{\sigma_1 - \sigma_3}{2} &= \frac{\sigma_1 + \sigma_3}{2} \sin \phi + c \cos \phi \\ \Rightarrow \sqrt{J_2} \cos \theta &= \left(\frac{1}{3}J_1 + \sqrt{J_2} \frac{1}{\sqrt{3}} \sin \theta \right) \sin \phi + c \cos \phi \\ \Rightarrow -\frac{1}{3}J_1 \sin \phi + \sqrt{J_2} \left(\cos \theta - \frac{1}{\sqrt{3}} \sin \theta \sin \phi \right) - c \cos \phi &= 0 \end{aligned}$$

$$F^{MC,3D} = -\frac{1}{3}J_1 \sin \phi + \sqrt{J_2} \left(\cos \theta - \frac{1}{\sqrt{3}} \sin \theta \sin \phi \right) - c \cos \phi$$

Note that the expression for F in the Mohr-Coulomb case in [2031] contains errors which is later corrected in [2035, p102].

3.15.14 The Griffith-Murrell failure criterion

The Griffith-Murrell yield criterion [225, 228, 85] is not often used. Extending the work of Griffith (1921) to three dimensional stress distributions, Murrell (1963) suggested the following criterion for rock failure expressed in terms of the principal stresses:

$$(\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_3 - \sigma_1)^2 + 24T_0(\sigma_1 + \sigma_2 + \sigma_3) = 0$$

where T_0 is a material property called the tensile strength. In principal stress space, this criterion is represented by a paraboloid of revolution around the pressure (or hydrostatic) axis.

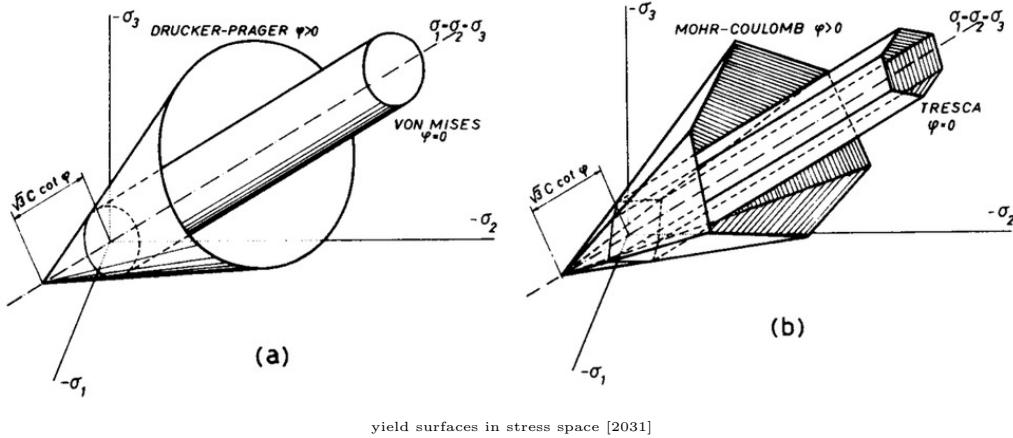
Also found:

$$J_{2D} + 12T_0p = 0$$

3.15.15 The failure envelope, or yield surface

A yield surface is a five-dimensional surface in the six-dimensional space of stresses. The state of stress of inside the yield surface is elastic. When the stress state lies on the surface the material is said to have reached its yield point and the material is said to have become plastic. Further deformation of the material causes the stress state to remain on the yield surface, even though the surface itself may change shape and size as the plastic deformation evolves, this is because stress states that lie outside the yield surface are non-permissible.

The yield surface is usually expressed in terms of (and visualized in) a three-dimensional principal stress space $(\sigma_1, \sigma_2, \sigma_3)$, a two- or three-dimensional space spanned by stress invariants or a version of the three-dimensional Haigh-Westergaard space.



yield surfaces in stress space [2031]

Literature: [1644]

3.15.16 Peierls creep

Looking at the literature, there seems to be many formulations for the Peierls creep deformation mechanism but it appears that a standard formulation for the Peierls creep writes:

$$\dot{\epsilon} = A\sigma^n \exp \left[-\frac{Q+pV}{RT} \left(1 - \left(\frac{\sigma}{\sigma_P} \right)^k \right)^q \right]$$

and it seems common to take $k = 1$, and $n = 2$ [722, 1023]

$$\dot{\epsilon} = A\sigma^2 \exp \left[-\frac{Q+pV}{RT} \left(1 - \frac{\sigma}{\sigma_P} \right)^q \right]$$

In Chenin et al. (2019) [359] the authors state that their Peierls creep implementation relies on parameters from Evans and Goetze (1979) [587] using the approach of Kameyama et al (1999) [1014]:

$$\eta^{pe} = \frac{2}{3} \frac{(1-s)/s}{(1+s)/2s} A (\varepsilon_e^{ds})^{\frac{1}{n}-1}$$

with A for this formulation:

$$A = \left[A_p \exp \left(-\frac{Q(1-\gamma)^2}{RT} \right) \right]^{-1/s} \gamma \sigma_p$$

where s is an effective stress exponent that depends on the temperature:

$$s = 2\gamma \frac{Q}{RT} (1-\gamma)$$

where γ is a fitting parameter.

Literature[56, 303, 609, 693, 722, 770, 1023, 1040, 1017, 1307, 2011, 358]

3.15.17 Stress limiting rheology

Taken from [1838]:

$$\eta_y = \tau_y \dot{\epsilon}_y^{-1/n_y} \dot{\epsilon}^{(1/n_y)-1}$$

where the yield stress τ_y , the yield strain rate $\dot{\epsilon}_y$ and the yield exponent n_y are prescribed parameters. In this article, $n_y = 10$, $\dot{\epsilon}_y = 10^{-15} s^{-1}$. When $n_y = 1$ the viscosity is constant and given by $\tau_y \dot{\epsilon}_y^{-1/n_y}$.

3.15.18 Arrhenius law

A purely temperature-dependent dimensional Arrhenius law that emulates the temperature dependence of viscosity in silicate rock is often employed for mantle rocks [12, 1995, 1834, 207, 1394, 177, 825]:

$$\eta(T) = \eta_0 \exp\left(\frac{Q}{R}\left(\frac{1}{T} - \frac{1}{T_0}\right)\right) \quad (96)$$

where η_0 is a reference viscosity and T_0 its corresponding reference temperature.

It can also account for pressure effects as in [1230] where the diffusion creep viscosity (under the assumption of homogeneous grain size) is temperature- and pressure-dependent:

$$\eta(T) = \eta_0 \exp\left(\frac{1}{R}\left(\frac{Q-pV}{T} - \frac{Q}{T_0}\right)\right)$$

(I find the minus sign rather suspicious)

3.15.19 Simple parametrisation of the mantle

Many CITCOMs-based publications [273, 272] have used the following (dimensionless) viscosity for the mantle:

$$\eta(T, z) = \eta_r(r) \exp(A(0.5 - T))$$

where η_r is a depth dependent viscosity profile (usually defined as discontinuous linear profiles for various shells)

The non-dimensional activation coefficient is chosen to be $A = 9.2103$ in [272] which leads to a temperature-induced viscosity contrast of 10^4 (for $T \in [0, 1]$).

This is also called the Frank-Kamenetskii flow rule, as used in [1141]:

$$\eta' = \eta_0 \exp(-\theta T)$$

where the parameters η_0, θ account for the local chemical composition of the rock.

Another temperature-dependent common expression is as follows [627]:

$$\eta(T) = \eta_\infty \exp\left(\frac{Q}{R}\left(\frac{1}{T} - \frac{1}{T_\infty}\right)\right)$$

Also, following [627]: For studying transient convection in a non-Newtonian rheological fluid, it is expedient from a computational point of view to employ a law which behaves linearly for low stresses initially and becomes gradually non-Newtonian only after a certain threshold stress level has been surpassed [374, 382]:

$$\eta(T, p, \tau_2) = \eta(T, p) \frac{1}{A_2 + A_3 \tau_2^2}$$

where A_2 is a parameter describing the linear creep at low stress levels and A_3 governs the transition stress between Newtonian and non-Newtonian rheologies.

Coltice and Sheppard (2018) [419] use a depth- and temperature-dependent viscosity formulation:

$$\eta(z, T) = \eta_0(z) \exp \frac{Q}{RT}$$

Note that this expression is supplemented with a pseudo-plastic formulation [1593].

 Literature: [1056]

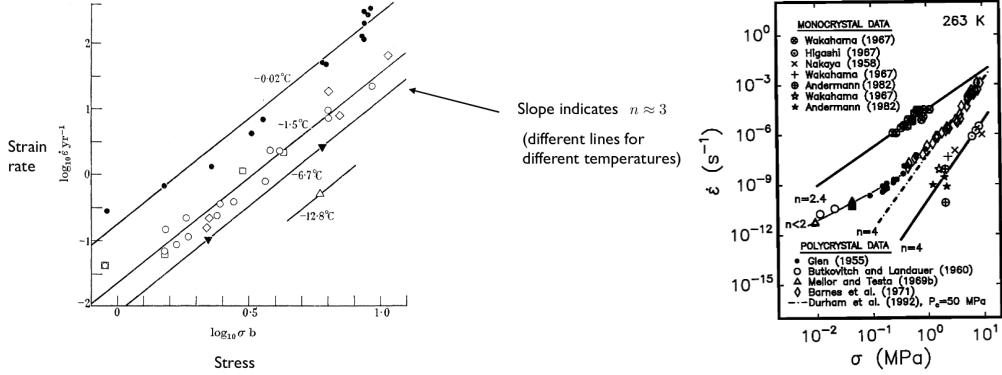
3.15.20 Glen's law for ice

As it turns out, ice and rocks share similarities in terms of rheology. Glens law is the most commonly used flow law for ice in glaciers and ice sheets [765] and it is actually a power-law type rheology:

$$\dot{\varepsilon} = A \tau^n$$

with $n \sim 3$ and $A \sim 2.4 \cdot 10^{-24} \text{ Pa}^{-3} \cdot \text{s}^{-1}$ at 0°C . The effective viscosity is then given by

$$\eta = \frac{1}{2A\tau_e^{n-1}}$$



Left: Taken from Glen [765]; Right: taken from [778].

Most of these studies suggest values of the power-law exponent $n \sim 2 - 4$, and there seems to be a general indication that the exponent is lower at lower stresses.

The A coefficient above has been found to depend on temperature and is reasonably described with an Arrhenius law:

$$A(T) = A_0 \exp\left(-\frac{Q}{RT}\right)$$

A standard formulation is the Paterson-Budd law with a fixed Glen exponent $n = 3$ and a split Arrhenius term [1453]:

$$\begin{aligned} A &= 3.615 \cdot 10^{-13} \text{ Pa}^{-3} \cdot \text{s}^{-1}, & Q &= 60 \text{ kJ/mol}, & \text{if } T < 263 \text{ K} \\ A &= 1.733 \cdot 10^3 \text{ Pa}^{-3} \cdot \text{s}^{-1}, & Q &= 139 \text{ kJ/mol}, & \text{if } T > 263 \text{ K} \end{aligned}$$

Be careful that in these two equations the temperature T is the pressure-adjusted temperature [1453].

Note that A is also affected by the water content and the presence of impurities.

Finally, Glen's law is the standard rheology used for ice-sheet modelling but it does not account for the complex evolution of fabric and resulting anisotropy.

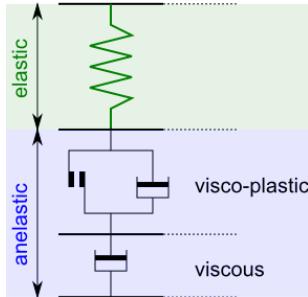
Literature[814, 1114, 815]

3.15.21 Elasto-Visco-Plasticity

Following [359], one can base the rheological model on the additive decomposition of the deviatoric strain rate tensor $\dot{\epsilon}^d$:

$$\dot{\epsilon}^d = \dot{\epsilon}^{el} + \dot{\epsilon}^{pl} + \dot{\epsilon}^{ds} + \dot{\epsilon}^{df} + \dot{\epsilon}^{pe}$$

where the five strain rate terms correspond respectively to the elastic, plastic, and viscous creep (dislocation, diffusion, peierls) contributions. Rather interestingly, Chenin et al [359] state that "this strain rate equation is nonlinear and solved locally on cell centroids and vertices in order to define the current effective viscosity and stress [1498]."



Literature: [911, 1228, 910, 1223, 32, 568]

3.15.22 Anisotropic viscosity

Following the paper by Lev and Hager (2008) [1174], the anisotropic viscosity enters the equation of momentum through a 'correction' term added to the isotropic part of the constitutive equation relating

stress and strain rate [1380]:

$$\sigma_{ij} = -p\delta_{ij} + 2\eta_N \dot{\varepsilon}_{ij} - 2(\eta_N - \eta_S)\Lambda_{ijkl}\dot{\varepsilon}_{kl}$$

where η_N is the normal viscosity and η_S is the shear viscosity. The fourth order tensor Λ reflects the orientation of the directors in space, denoted by \vec{n} :

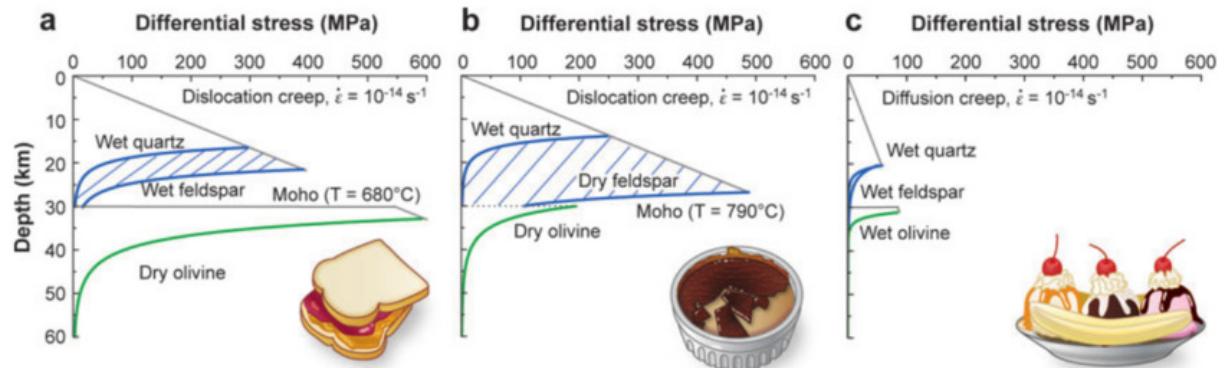
$$\Lambda_{ijkl} = \frac{1}{2}(n_i n_k \delta_{lj} + n_j n_k \delta_{il} + n_i n_l \delta_{kj} n_j n_l \delta_{ik}) - 2n_i n_j n_k n_l$$

Following [1349, 1380], the 'directors' are advected through the model and are analogous to particles. The directors are vector-particles pointing normal to the easy-glide plane or layer, thus defining the directions associated with η_N and η_S . In each time step of the calculation, the directors are advected and rotated by the flow, and in return determine the viscosity structure for the next time step [1379].

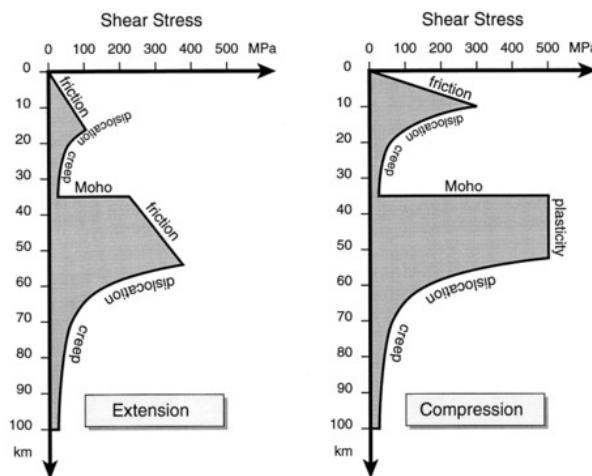
(MSc Thesis): redo the Rayleigh-Taylor instabilities with anisotropic lithospheric viscosity. experiments of Lev & Hager (2008) [1174].

Literature: [1380, 1870, 1379, 1380, 1315, 1579, 1616]

3.15.23 Rheology of the lithosphere



Schematic view of the three most common first order rheological models of the continental lithosphere under a strain rate of 10^{-14}s^{-1} . In all three models the upper crust has its frictional strength increased with pressure and depth. (a) The jelly sandwich model has a weak mid-lower crust and a strong mantle composed of dry olivine. (b) The crème brûlée model assumes that the mantle is weak, due to the presence of water and high temperature deformation, and the dry and brittle crust determines the strength of the lithosphere. (c) The banana split model assumes that the lithosphere as a whole has its strength greatly reduced due to various strain weakening and feedback processes [287]



Taken from [169]. Typical vertical distribution of maximum shear stress in continental lithosphere undergoing compressional (right) or extensional (left) strain at 10^{-15}s . Friction controls level of shear stress in upper part of crust and sometimes in mantle lithosphere; then, below brittle/ductile transition, shear stress is controlled by thermally-activated dislocation creep.



3.16 Moment of inertia

Consider a rigid body rotating with fixed angular velocity ω about an axis which passes through the origin. Let \mathbf{r}_i be the position vector of the i th mass element, whose mass is m_i . We expect this position vector to precess about the axis of rotation (which is parallel to ω) with angular velocity ω .

$$\frac{d\mathbf{r}_i}{dt} = \boldsymbol{\omega} \times \mathbf{r}_i.$$

Thus, the above equation specifies the velocity, $\mathbf{v}_i = d\mathbf{r}_i/dt$, of each mass element as the body rotates with fixed angular velocity ω about an axis passing through the origin.

The total angular momentum of the body (about the origin) is written

$$\mathbf{L} = \sum_{i=1,N} m_i \mathbf{r}_i \times \frac{d\mathbf{r}_i}{dt} = \sum_{i=1,N} m_i \mathbf{r}_i \times (\boldsymbol{\omega} \times \mathbf{r}_i) = \sum_{i=1,N} m_i [r_i^2 \boldsymbol{\omega} - (\mathbf{r}_i \cdot \boldsymbol{\omega}) \mathbf{r}_i]$$

The above formula can be written as a matrix equation of the form

$$\begin{pmatrix} L_x \\ L_y \\ L_z \end{pmatrix} = \begin{pmatrix} I_{xx} & I_{xy} & I_{xz} \\ I_{yx} & I_{yy} & I_{yz} \\ I_{zx} & I_{zy} & I_{zz} \end{pmatrix} \begin{pmatrix} \omega_x \\ \omega_y \\ \omega_z \end{pmatrix}$$

where

$$\begin{aligned} I_{xx} &= + \sum_{i=1,N} (y_i^2 + z_i^2) m_i = \int (y^2 + z^2) dm = \int_V (y^2 + z^2) \rho(x, y, z) dV \\ I_{yy} &= + \sum_{i=1,N} (x_i^2 + z_i^2) m_i = \int (x^2 + z^2) dm = \int_V (x^2 + z^2) \rho(x, y, z) dV \\ I_{zz} &= + \sum_{i=1,N} (x_i^2 + y_i^2) m_i = \int (x^2 + y^2) dm = \int_V (x^2 + y^2) \rho(x, y, z) dV \\ I_{xy} = I_{yx} &= - \sum_{i=1,N} x_i y_i m_i = - \int x y dm = - \int x y \rho(x, y, z) dV \\ I_{yz} = I_{zy} &= - \sum_{i=1,N} y_i z_i m_i = - \int y z dm = - \int y z \rho(x, y, z) dV \\ I_{xz} = I_{zx} &= - \sum_{i=1,N} x_i z_i m_i = - \int x z dm = - \int x z \rho(x, y, z) dV \end{aligned}$$

Here, I_{xx} is called the moment of inertia about the x -axis, I_{yy} the moment of inertia about the y -axis, I_{xy} the xy product of inertia, I_{yz} the yz product of inertia, etc. The matrix of the I_{ij} values is known as the moment of inertia tensor.

In general, the angular momentum vector, \mathbf{L} points in a different direction to the angular velocity vector, $\boldsymbol{\omega}$. In other words, \mathbf{L} is generally not parallel to $\boldsymbol{\omega}$.

Finally, although the above results were obtained assuming a fixed angular velocity, they remain valid at each instant in time if the angular velocity varies.

In the simplified case of a spherically symmetric planet, it is easy to see that $I_{xx} = I_{yy} = I_{zz}$ so that $I = \frac{1}{3}(I_{xx} + I_{yy} + I_{zz})$, and $\rho = \rho(r)$ with $dV = 4\pi r^2 dr$, leading to

$$I = \frac{8\pi}{3} \int_0^R \rho(r) r^4 dr$$

Assuming further that the planet has a constant density ρ_0 , we obtain

$$I = \frac{8\pi}{3}\rho_0 \int_0^R r^4 dr = \frac{8\pi}{3}\rho_0 \frac{R^5}{5} = \frac{2}{5}MR^2$$

where M is the mass of the planet and R is its radius.

Assuming now that the planet is composed of a core of radius R_c and density ρ_c surrounded by a mantle of density ρ_m , we have

$$I = \frac{8\pi}{3} \int_0^R \rho(r)r^4 dr = \frac{8\pi}{3} \left(\int_0^{R_c} \rho_c r^4 dr + \int_{R_c}^R \rho_m r^4 dr \right) = \frac{8\pi}{15} (\rho_c R_c^5 + \rho_m (R^5 - R_c^5))$$

The moment of inertia of the core is given in Table 2 of "Core Dynamics", Treatise on Geophysics, edited by Peter Olson: $I_{core} = 9.2 \times 10^{36} kg.m^2$. The total moment of inertia for the Earth is then given by $I = I_{core} + I_{mantle}$.

4 The building blocks of the Finite Element Method

4.1 Numerical integration

As we will see later, using the Finite Element method to solve problems involves computing integrals which are more often than not too complex to be computed analytically/exactly. We will then need to compute them numerically.

[wiki] In essence, the basic problem in numerical integration is to compute an approximate solution to a definite integral

$$\int_a^b f(x)dx$$

to a given degree of accuracy. This problem has been widely studied and we know that if $f(x)$ is a smooth function, and the domain of integration is bounded, there are many methods for approximating the integral to the desired precision.

There are several reasons for carrying out numerical integration.

- The integrand $f(x)$ may be known only at certain points, such as obtained by sampling. Some embedded systems and other computer applications may need numerical integration for this reason.
- A formula for the integrand may be known, but it may be difficult or impossible to find an antiderivative that is an elementary function. An example of such an integrand is $f(x) = \exp(-x^2)$, the antiderivative of which (the error function, times a constant) cannot be written in elementary form.
- It may be possible to find an antiderivative symbolically, but it may be easier to compute a numerical approximation than to compute the antiderivative. That may be the case if the antiderivative is given as an infinite series or product, or if its evaluation requires a special function that is not available.

4.1.1 in 1D - theory

The simplest method of this type is to let the interpolating function be a constant function (a polynomial of degree zero) that passes through the point $((a+b)/2, f((a+b)/2))$.

This is called the midpoint rule or rectangle rule.

$$\int_a^b f(x)dx \simeq (b-a)f\left(\frac{a+b}{2}\right)$$

insert here figure

The interpolating function may be a straight line (an affine function, i.e. a polynomial of degree 1) passing through the points $(a, f(a))$ and $(b, f(b))$.

This is called the trapezoidal rule.

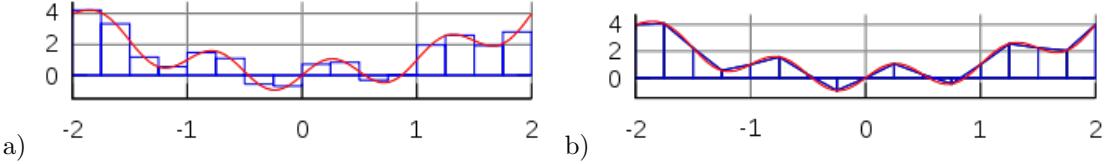
$$\int_a^b f(x)dx \simeq (b-a)\frac{f(a) + f(b)}{2}$$

insert here figure

For either one of these rules, we can make a more accurate approximation by breaking up the interval $[a, b]$ into some number n of subintervals, computing an approximation for each subinterval, then adding up all the results. This is called a composite rule, extended rule, or iterated rule. For example, the composite trapezoidal rule can be stated as

$$\int_a^b f(x)dx \simeq \frac{b-a}{n} \left(\frac{f(a)}{2} + \sum_{k=1}^{n-1} f(a + k \frac{b-a}{n}) + \frac{f(b)}{2} \right)$$

where the subintervals have the form $[kh, (k+1)h]$, with $h = (b-a)/n$ and $k = 0, 1, 2, \dots, n-1$.



The interval $[-2, 2]$ is broken into 16 sub-intervals. The blue lines correspond to the approximation of the red curve by means of a) the midpoint rule, b) the trapezoidal rule.

There are several algorithms for numerical integration (also commonly called 'numerical quadrature', or simply 'quadrature'). Interpolation with polynomials evaluated at equally spaced points in $[a, b]$ yields the Newton-Cotes formulas, of which the rectangle rule and the trapezoidal rule are examples. If we allow the intervals between interpolation points to vary, we find another group of quadrature formulas, such as the Gauss(ian) quadrature formulas. A Gaussian quadrature rule is typically more accurate than a Newton-Cotes rule, which requires the same number of function evaluations, if the integrand is smooth (i.e., if it is sufficiently differentiable).

An n -point Gaussian quadrature rule, named after Carl Friedrich Gauss, is a quadrature rule constructed to yield an exact result for polynomials of degree $2n - 1$ or less by a suitable choice of the points x_i and weights w_i for $i = 1, \dots, n$.

The domain of integration for such a rule is conventionally taken as $[-1, 1]$, so the rule is stated as

$$\int_{-1}^{+1} f(x) dx = \sum_{i_q=1}^n w_{i_q} f(x_{i_q})$$

In this formula the x_{i_q} coordinate is the i -th root of the Legendre polynomial $P_n(x)$.

It is important to note that a Gaussian quadrature will only produce good results if the function $f(x)$ is well approximated by a polynomial function within the range $[-1, 1]$. As a consequence, the method is not, for example, suitable for functions with singularities.

Number of points, n	Points, x_i	Weights, w_i
1	0	2
2	$\pm\sqrt{\frac{1}{3}}$	1
3	0	$\frac{8}{9}$
	$\pm\sqrt{\frac{3}{5}}$	$\frac{5}{9}$
4	$\pm\sqrt{\frac{3}{7} - \frac{2}{7}\sqrt{\frac{6}{5}}}$	$\frac{18+\sqrt{30}}{36}$
	$\pm\sqrt{\frac{3}{7} + \frac{2}{7}\sqrt{\frac{6}{5}}}$	$\frac{18-\sqrt{30}}{36}$
5	0	$\frac{128}{225}$
	$\pm\frac{1}{3}\sqrt{5 - 2\sqrt{\frac{10}{7}}}$	$\frac{322+13\sqrt{70}}{900}$
	$\pm\frac{1}{3}\sqrt{5 + 2\sqrt{\frac{10}{7}}}$	$\frac{322-13\sqrt{70}}{900}$

Gauss-Legendre points and their weights.

n	x_{iq}	w_{iq}	x_{iq} (approx)	w_{iq} (approx)
1	0	2	0	2
2	$\pm\sqrt{1/3}$	1	$\pm 0.577\ 350\ 269\ 189\ 626$	1
3	0	$8/9$	0	0.888 888 888 888 888
	$\pm\sqrt{3/5}$	$5/9$	$\pm 0.774\ 596\ 669\ 241\ 483$	0.555 555 555 555 555
4	$\pm\sqrt{\frac{3}{7} - \frac{2}{7}\sqrt{6/5}}$	$\frac{18+\sqrt{30}}{36}$	$\pm 0.339\ 981\ 043\ 584\ 856$	0.652 145 154 862 546
	$\pm\sqrt{\frac{3}{7} + \frac{2}{7}\sqrt{6/5}}$	$\frac{18-\sqrt{30}}{36}$	$\pm 0.861\ 136\ 311\ 594\ 953$	0.347 854 845 137 454
5	0	$128/225$	0	0.568 888 888 888 889
	$\pm\frac{1}{3}\sqrt{5 - 2\sqrt{\frac{10}{7}}}$	$\frac{322+13\sqrt{70}}{900}$	$\pm 0.538\ 469\ 310\ 105\ 683$	0.478 628 670 499 366
	$\pm\frac{1}{3}\sqrt{5 + 2\sqrt{\frac{10}{7}}}$	$\frac{322-13\sqrt{70}}{900}$	$\pm 0.906\ 179\ 845\ 938\ 664$	0.236 926 885 056 189
6	?	?	$\pm 0.238\ 619\ 186\ 083\ 197$ $\pm 0.661\ 209\ 386\ 466\ 265$ $\pm 0.932\ 469\ 514\ 203\ 152$	0.467 913 934 572 691 0.360 761 573 048 139 0.171 324 492 379 170
7			$\pm 0.946\ 107\ 912\ 342\ 759$ $\pm 0.741\ 531\ 185\ 599\ 394$ $\pm 0.405\ 845\ 151\ 377\ 397$ $0.000\ 000\ 000\ 000\ 000$	0.129 484 966 168 870 0.279 705 391 489 277 0.381 830 050 505 119 0.417 959 183 673 469
8			$\pm 0.960\ 289\ 856\ 497\ 536$ $\pm 0.796\ 666\ 477\ 413\ 627$ $\pm 0.525\ 532\ 409\ 916\ 329$ $\pm 0.183\ 434\ 642\ 495\ 650$	0.101 228 536 290 376 0.222 381 034 453 374 0.313 706 645 877 887 0.362 683 783 378 362
9			$\pm 0.968\ 160\ 239\ 507\ 626$ $\pm 0.836\ 031\ 107\ 326\ 636$ $\pm 0.613\ 371\ 432\ 700\ 590$ $\pm 0.324\ 253\ 423\ 403\ 809$ $0.000\ 000\ 000\ 000\ 000$	0.081 274 388 361 574 0.180 648 160 694 857 0.260 610 696 402 935 0.312 347 077 040 003 0.330 239 355 001 260
10			$\pm 0.973\ 906\ 528\ 517\ 172$ $\pm 0.865\ 063\ 366\ 688\ 985$ $\pm 0.679\ 409\ 568\ 299\ 024$ $\pm 0.433\ 395\ 394\ 129\ 247$ $\pm 0.148\ 874\ 338\ 981\ 631$	0.066 671 344 308 688 0.149 451 349 150 581 0.219 086 362 515 982 0.269 266 719 309 996 0.295 524 224 714 753

Abscissae and weights for Gauss quadratures up to $n = 10$. See [1176, p89]

As shown in the above table, it can be shown that the weight values must fulfill the following condition:

$$\sum_{iq} w_{iq} = 2 \quad (97)$$

and it is worth noting that all quadrature point coordinates are symmetrical around the origin.

Since most quadrature formulae are only valid on a specific interval, we now must address the problem of their use outside of such intervals. The solution turns out to be quite simple: one must carry out a change of variables from the interval $[a, b]$ to $[-1, 1]$.

We then consider the reduced coordinate $r \in [-1, 1]$ such that

$$r = \frac{2}{b-a}(x-a) - 1$$

This relationship can be reversed such that when r is known, its equivalent coordinate $x \in [a, b]$ can be computed:

$$x = \frac{b-a}{2}(1+r) + a$$

From this it follows that

$$dx = \frac{b-a}{2}dr$$

and then

$$\int_a^b f(x)dx = \frac{b-a}{2} \int_{-1}^{+1} f(r)dr \simeq \frac{b-a}{2} \sum_{i_q=1}^{n_q} w_{i_q} f(r_{i_q})$$

4.1.2 in 1D - examples

example 1 Since we know how to carry out any required change of variables, we choose for simplicity $a = -1$, $b = +1$. Let us take for example $f(x) = \pi$. Then we can compute the integral of this function over the interval $[a, b]$ exactly:

$$I = \int_{-1}^{+1} f(x)dx = \pi \int_{-1}^{+1} dx = 2\pi$$

We can now use a Gauss-Legendre formula to compute this same integral:

$$I_{gq} = \int_{-1}^{+1} f(x)dx = \sum_{i_q=1}^{n_q} w_{i_q} f(x_{i_q}) = \sum_{i_q=1}^{n_q} w_{i_q} \pi = \pi \underbrace{\sum_{i_q=1}^{n_q} w_{i_q}}_{=2} = 2\pi$$

where we have used the property of the weight values of Eq.(97). Since the actual number of points was never specified, this result is valid for all quadrature rules.

example 2 Let us now take $f(x) = mx + p$ and repeat the same exercise:

$$I = \int_{-1}^{+1} f(x)dx = \int_{-1}^{+1} (mx + p)dx = [\frac{1}{2}mx^2 + px]_{-1}^{+1} = 2p$$

$$I_{gq} = \int_{-1}^{+1} f(x)dx = \sum_{i_q=1}^{n_q} w_{i_q} f(x_{i_q}) = \sum_{i_q=1}^{n_q} w_{i_q} (mx_{i_q} + p) = m \underbrace{\sum_{i_q=1}^{n_q} w_{i_q} x_{i_q}}_{=0} + p \underbrace{\sum_{i_q=1}^{n_q} w_{i_q}}_{=2} = 2p$$

since the quadrature points are symmetric w.r.t. to zero on the x-axis. Once again the quadrature is able to compute the exact value of this integral: this makes sense since an n -point rule exactly integrates a $2n - 1$ order polynomial such that a 1 point quadrature exactly integrates a first order polynomial like the one above.

example 3 Let us now take $f(x) = x^2$. We have

$$I = \int_{-1}^{+1} f(x)dx = \int_{-1}^{+1} x^2 dx = [\frac{1}{3}x^3]_{-1}^{+1} = \frac{2}{3}$$

and

$$I_{gq} = \int_{-1}^{+1} f(x)dx = \sum_{i_q=1}^{n_q} w_{i_q} f(x_{i_q}) = \sum_{i_q=1}^{n_q} w_{i_q} x_{i_q}^2$$

- $n_q = 1$: $x_{i_q}^{(1)} = 0$, $w_{i_q} = 2$. $I_{gq} = 0$
- $n_q = 2$: $x_q^{(1)} = -1/\sqrt{3}$, $x_q^{(2)} = 1/\sqrt{3}$, $w_q^{(1)} = w_q^{(2)} = 1$. $I_{gq} = \frac{2}{3}$
- It also works $\forall n_q > 2$!

4.1.3 in 2D/3D - theory

Let us now turn to a two-dimensional integral of the form

$$I = \int_{-1}^{+1} \int_{-1}^{+1} f(x, y) dxdy$$

The equivalent Gaussian quadrature writes:

$$I_{gq} \simeq \sum_{i_q=1}^{n_q} \sum_{j_q=1}^{n_q} f(x_{i_q}, y_{j_q}) w_{i_q} w_{j_q}$$

4.1.4 quadrature on triangles

Quadrature rules for triangles can be found in Dunavant, 1985 [543]. The following ones are identical to those in *ip_triangle.m* file of the MILAMIN code [466].

	r_q	s_q	w_q	
$iq = 1$	1/3	1/3	1/2	
$iq = 1$	1/6	1/6	1/6	
$iq = 2$	2/3	1/6	1/6	
$iq = 3$	1/6	2/3	1/6	
$iq = 1$	1/3	1/3	-27/96	
$iq = 2$	0.6	0.2	25/96	
$iq = 3$	0.2	0.6	25/96	
$iq = 4$	0.2	0.2	25/96	
$iq = 1$	$1 - 2g_1$	g_1	$w_1/2$	0.108103018168070
$iq = 2$	g_1	$1 - 2g_1$	$w_1/2$	0.445948490915965
$iq = 3$	g_1	g_1	$w_1/2$	0.445948490915965
$iq = 4$	$1 - 2g_2$	g_2	$w_2/2$	0.816847572980459
$iq = 5$	g_2	$1 - 2g_2$	$w_2/2$	0.091576213509771
$iq = 6$	g_2	g_2	$w_2/2$	0.091576213509771
$iq = 1$				0.091576213509771
$iq = 2$				0.816847572980459
$iq = 3$				0.091576213509771
$iq = 4$				0.445948490915965
$iq = 5$				0.108103018168070
$iq = 6$				0.445948490915965
$iq = 1$				0.091576213509771
$iq = 2$				0.091576213509771
$iq = 3$				0.816847572980459
$iq = 4$				0.091576213509771
$iq = 5$				0.445948490915965
$iq = 6$				0.108103018168070
$iq = 7$				0.223381589678011/2.0
$iq = 1$				0.091576213509771
$iq = 2$				0.091576213509771
$iq = 3$				0.091576213509771
$iq = 4$				0.091576213509771
$iq = 5$				0.091576213509771
$iq = 6$				0.091576213509771
$iq = 7$				0.091576213509771
$iq = 1$				0.1012865073235
$iq = 2$				0.7974269853531
$iq = 3$				0.1012865073235
$iq = 4$				0.4701420641051
$iq = 5$				0.4701420641051
$iq = 6$				0.0597158717898
$iq = 7$				0.3333333333333
$iq = 1$				0.1012865073235
$iq = 2$				0.7974269853531
$iq = 3$				0.1012865073235
$iq = 4$				0.4701420641051
$iq = 5$				0.4701420641051
$iq = 6$				0.0597158717898
$iq = 7$				0.3333333333333
$iq = 1$				0.3333333333333
$iq = 1$				5.01426509658179E - 01
$iq = 2$				2.49286745170910E - 01
$iq = 3$				2.49286745170910E - 01
$iq = 4$				8.73821971016996E - 01
$iq = 5$				6.30890144915020E - 02
$iq = 6$				6.30890144915020E - 02
$iq = 7$				5.31450498448170E - 02
$iq = 8$				6.36502499121399E - 01
$iq = 9$				3.10352451033784E - 01
$iq = 10$				5.31450498448170E - 02
$iq = 11$				6.36502499121399E - 01
$iq = 12$				3.10352451033784E - 01
$iq = 1$				5.83931378631895E - 02
$iq = 2$				5.83931378631895E - 02
$iq = 3$				5.83931378631895E - 02
$iq = 4$				2.54224531851035E - 02
$iq = 5$				2.54224531851035E - 02
$iq = 6$				2.54224531851035E - 02
$iq = 7$				2.54224531851035E - 02
$iq = 8$				2.54224531851035E - 02
$iq = 9$				2.54224531851035E - 02
$iq = 10$				2.54224531851035E - 02
$iq = 11$				2.54224531851035E - 02
$iq = 12$				2.54224531851035E - 02
$iq = 1$				5.83931378631895E - 02
$iq = 2$				5.83931378631895E - 02
$iq = 3$				5.83931378631895E - 02
$iq = 4$				0.109951743655322/2.0
$iq = 5$				0.109951743655322/2.0
$iq = 6$				0.109951743655322/2.0
$iq = 7$				0.109951743655322/2.0
$iq = 8$				0.109951743655322/2.0
$iq = 9$				0.109951743655322/2.0
$iq = 10$				0.109951743655322/2.0
$iq = 11$				0.109951743655322/2.0
$iq = 12$				0.109951743655322/2.0

where

$$g_1 = \left(8 - \sqrt{10} + \sqrt{38 - 44\sqrt{2/5}} \right) / 18 \quad g_2 = \left(8 - \sqrt{10} - \sqrt{38 - 44\sqrt{2/5}} \right) / 18$$

$$w_1 = \left(620 + \sqrt{213125 - 53320\sqrt{10}} \right) / 3720 \quad w_2 = \left(620 - \sqrt{213125 - 53320\sqrt{10}} \right) / 3720$$

4.1.5 quadrature on tetrahedra

Remark. In what follows the coefficients in the tables are not the reduced coordinates of the quadrature points but the coefficients corresponding to the 4 nodes.

Quadrature rules on tetrahedra take the form:

$$\int \int \int_{el} f(x, y, z) dx dy dz = V_{el} \sum_{iq=1}^{nqel} w_{iq} f(\xi_1^{iq}, \xi_2^{iq}, \xi_3^{iq}, \xi_4^{iq})$$

or, that is to say:

$$\int \int \int_{el} f(x, y, z) dx dy dz = \sum_{iq=1}^{nqel} (w_{iq} V_{el}) f(\xi_1^{iq}, \xi_2^{iq}, \xi_3^{iq}, \xi_4^{iq})$$

with in our case $V_{el} = 1/6$.

In the literature it can be found that a one point quadrature is characterised by

$$w_{iq} = 1 \quad \xi_1^{iq} = \xi_2^{iq} = \xi_3^{iq} = \xi_4^{iq} = 0.25$$

i.e, the coordinates of the single point are given by:

$$x_{iq} = \sum_{i=1}^4 \xi_i^{iq} x_i = \frac{1}{4}(x_1 + x_2 + x_3 + x_4)$$

Same for y and z coordinates.

A four-point quadrature rule is characterised by $w_{iq} = V_{el} * 0.25 = 1/24 \simeq 04166666666666667$ and

	ξ_1	ξ_2	ξ_3	ξ_4
$iq=1$	0.585410196624969	0.138196601125011	0.138196601125011	0.138196601125011
$iq=2$	0.138196601125011	0.585410196624969	0.138196601125011	0.138196601125011
$iq=3$	0.138196601125011	0.138196601125011	0.585410196624969	0.138196601125011
$iq=4$	0.138196601125011	0.138196601125011	0.138196601125011	0.585410196624969

We then have:

$$r_{iq} = \sum_{i=1}^4 \xi_i^{iq} x_i = (\xi_1^{iq}, \xi_2^{iq}, \xi_3^{iq}, \xi_4^{iq}) \cdot (r_1, r_2, r_3, r_4) = (\xi_1^{iq}, \xi_2^{iq}, \xi_3^{iq}, \xi_4^{iq}) \cdot (0, 1, 0, 0) = \xi_2^{iq}$$

$$s_{iq} = \sum_{i=1}^4 \xi_i^{iq} y_i = (\xi_1^{iq}, \xi_2^{iq}, \xi_3^{iq}, \xi_4^{iq}) \cdot (s_1, s_2, s_3, s_4) = (\xi_1^{iq}, \xi_2^{iq}, \xi_3^{iq}, \xi_4^{iq}) \cdot (0, 0, 1, 0) = \xi_3^{iq}$$

$$t_{iq} = \sum_{i=1}^4 \xi_i^{iq} z_i = (\xi_1^{iq}, \xi_2^{iq}, \xi_3^{iq}, \xi_4^{iq}) \cdot (t_1, t_2, t_3, t_4) = (\xi_1^{iq}, \xi_2^{iq}, \xi_3^{iq}, \xi_4^{iq}) \cdot (0, 0, 0, 1) = \xi_4^{iq}$$

Finally:

	r_q	s_q	t_q	w_q
$iq = 1$	0.138196601125011	0.138196601125011	0.138196601125011	0.04166666666666667
$iq = 2$	0.585410196624969	0.138196601125011	0.138196601125011	0.04166666666666667
$iq = 3$	0.138196601125011	0.585410196624969	0.138196601125011	0.04166666666666667
$iq = 4$	0.138196601125011	0.138196601125011	0.585410196624969	0.04166666666666667

4.1.6 The Gauss-Lobatto approach

All what we have seen above falls under the Gauss-Legendre quadrature method. There is however another somewhat common quadrature method: the Gauss-Lobatto quadrature. . It is similar to Gaussian quadrature with the following important differences: 1) There are integration points in the interval but they also always include the end points of the integration interval; 2) It is accurate for polynomials up to degree $2n - 3$, where n is the number of integration points.

In 1D, it reads:

$$\int_{-1}^{+1} f(x) dx = \frac{2}{n(n-1)} [f(-1) + f(1)] + \sum_{i=2}^{n-1} w_i f(x_i)$$

The locations and weights of the integration points are as follows:

n	x_{iq}	w_{iq}	x_{iq} (approx)	w_{iq} (approx)
3	0	4/3	1/3	
	± 1	1/3		
4	$\pm \sqrt{\frac{1}{5}}$	5/6	1/6	
	± 1	1/6		
5	0	32/45	1/10	
	$\pm \sqrt{\frac{3}{7}}$	49/90		
	± 1	1/10		
6	$\pm \sqrt{\frac{1}{3} - \frac{2\sqrt{7}}{21}}$	$\frac{14+\sqrt{7}}{30}$	1/15	
	$\pm \sqrt{\frac{1}{3} + \frac{2\sqrt{7}}{21}}$	$\frac{14-\sqrt{7}}{30}$		
	± 1	1/15		

4.2 The mesh

4.3 A bit of FE terminology

We introduce here some terminology for efficient element descriptions [813]:

- For triangles/tetrahedra, the designation $P_m \times P_n$ means that each component of the velocity is approximated by continuous piecewise complete Polynomials of degree m and pressure by continuous piecewise complete Polynomials of degree n . For example $P_2 \times P_1$ means

$$u \sim a_1 + a_2x + a_3y + a_4xy + a_5x^2 + a_6y^2$$

with similar approximations for v , and

$$p \sim b_1 + b_2x + b_3y$$

Both velocity and pressure are continuous across element boundaries, and each triangular element contains 6 velocity nodes and three pressure nodes.

- For the same families, $P_m \times P_{-n}$ is as above, except that pressure is approximated via piecewise discontinuous polynomials of degree n . For instance, $P_2 \times P_{-1}$ is the same as P_2P_1 except that pressure is now an independent linear function in each element and therefore discontinuous at element boundaries.
- For quadrilaterals/hexahedra, the designation $Q_m \times Q_n$ means that each component of the velocity is approximated by a continuous piecewise polynomial of degree m in each direction on the quadrilateral and likewise for pressure, except that the polynomial is of degree n . For instance, $Q_2 \times Q_1$ means

$$u \sim a_1 + a_2x + a_3y + a_4xy + a_5x^2 + a_6y^2 + a_7x^2y + a_8xy^2 + a_9x^2y^2$$

and

$$p \sim b_1 + b_2x + b_3y + b_4xy$$

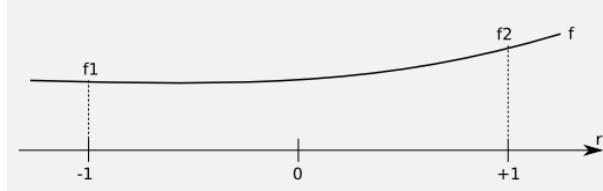
- For these same families, $Q_m \times Q_{-n}$ is as above, except that the pressure approximation is not continuous at element boundaries.
- Again for the same families, $Q_m \times P_{-n}$ indicates the same velocity approximation with a pressure approximation that is a discontinuous complete piecewise polynomial of degree n (not of degree n in each direction !)
- The designation P_m^+ or Q_m^+ means that some sort of bubble function was added to the polynomial approximation for the velocity. You may also find the term 'enriched element' in the literature.
- Finally, for $n = 0$, we have piecewise-constant pressure, and we omit the minus sign for simplicity.

Another point which needs to be clarified is the use of so-called 'conforming elements' (or 'non-conforming elements'). Following again [813], conforming velocity elements are those for which the basis functions for a subset of H^1 for the continuous problem (the first derivatives and their squares are integrable in Ω). For instance, the rotated $Q_1 \times P_0$ element of Rannacher and Turek (see section ??) is such that the velocity is discontinuous across element edges, so that the derivative does not exist there. Another typical example of non-conforming element is the Crouzeix-Raviart element [457].

4.4 Elements and basis functions in 1D

4.4.1 Linear basis functions (Q_1)

Let $f(r)$ be a C^1 function on the interval $[-1 : 1]$ with $f(-1) = f_1$ and $f(1) = f_2$.



Let us assume that the function $f(r)$ is to be approximated on $[-1, 1]$ by the first order polynomial

$$f(r) = a + br \quad (98)$$

Then it must fulfill

$$\begin{aligned} f(r = -1) &= a - b = f_1 \\ f(r = +1) &= a + b = f_2 \end{aligned}$$

This leads to

$$a = \frac{1}{2}(f_1 + f_2) \quad b = \frac{1}{2}(-f_1 + f_2)$$

and then replacing a, b in Eq. (98) by the above values on gets

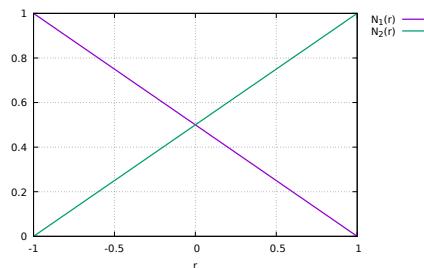
$$f(r) = \left[\frac{1}{2}(1 - r) \right] f_1 + \left[\frac{1}{2}(1 + r) \right] f_2$$

or

$$f(r) = \sum_{i=1}^2 N_i(r) f_i$$

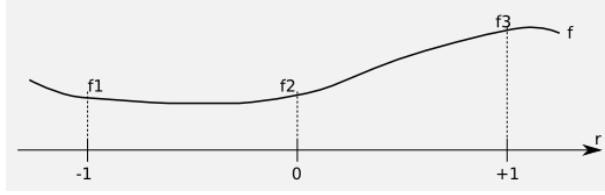
with

$N_1(r) = \frac{1}{2}(1 - r)$ $N_2(r) = \frac{1}{2}(1 + r)$	(99)
--	------



4.4.2 Quadratic basis functions (Q_2)

Let $f(r)$ be a C^1 function on the interval $[-1 : 1]$ with $f(-1) = f_1$, $f(0) = f_2$ and $f(1) = f_3$.



Let us assume that the function $f(r)$ is to be approximated on $[-1, 1]$ by the second order polynomial

$$f(r) = a + br + cr^2 \quad (100)$$

Then it must fulfill

$$\begin{aligned} f(r = -1) &= a - b + c = f_1 \\ f(r = 0) &= a = f_2 \\ f(r = +1) &= a + b + c = f_3 \end{aligned}$$

This leads to

$$a = f_2 \quad b = \frac{1}{2}(-f_1 + f_3) \quad c = \frac{1}{2}(f_1 + f_3 - 2f_2)$$

and then replacing a, b, c in Eq. (100) by the above values one gets

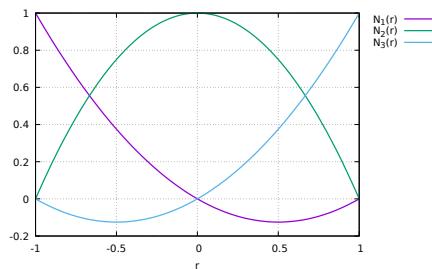
$$f(r) = \left[\frac{1}{2}r(r-1) \right] f_1 + (1-r^2)f_2 + \left[\frac{1}{2}r(r+1) \right] f_3$$

or,

$$f(r) = \sum_{i=1}^3 N_i(r) f_i$$

with

$\begin{aligned} N_1(r) &= \frac{1}{2}r(r-1) \\ N_2(r) &= (1-r^2) \\ N_3(r) &= \frac{1}{2}r(r+1) \end{aligned} \quad (101)$



4.4.3 Cubic basis functions (Q_3)

The 1D basis polynomial is given by

$$f(r) = a + br + cr^2 + dr^3$$

with the nodes at position $-1, -1/3, +1/3$ and $+1$.

$$\begin{aligned}
f(-1) &= a - b + c - d = f_1 \\
f(-1/3) &= a - \frac{b}{3} + \frac{c}{9} - \frac{d}{27} = f_2 \\
f(+1/3) &= a - \frac{b}{3} + \frac{c}{9} - \frac{d}{27} = f_3 \\
f(+1) &= a + b + c + d = f_4
\end{aligned}$$

Adding the first and fourth equation and the second and third, one arrives at

$$f_1 + f_4 = 2a + 2c \quad f_2 + f_3 = 2a + \frac{2c}{9}$$

and finally:

$$\begin{aligned}
a &= \frac{1}{16}(-f_1 + 9f_2 + 9f_3 - f_4) \\
c &= \frac{9}{16}(f_1 - f_2 - f_3 + f_4)
\end{aligned}$$

Combining the original 4 equations in a different way yields

$$2b + 2d = f_4 - f_1 \quad \frac{2b}{3} + \frac{2d}{27} = f_3 - f_2$$

so that

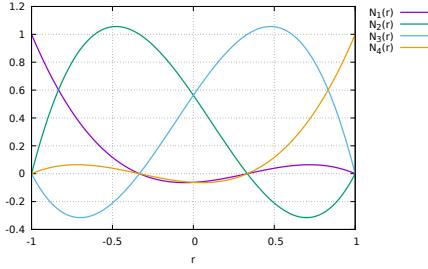
$$\begin{aligned}
b &= \frac{1}{16}(f_1 - 27f_2 + 27f_3 - f_4) \\
d &= \frac{9}{16}(-f_1 + 3f_2 - 3f_3 + f_4)
\end{aligned}$$

Finally,

$$\begin{aligned}
f(r) &= a + b + cr^2 + dr^3 \\
&= \frac{1}{16}(-1 + r + 9r^2 - 9r^3)f_1 \\
&\quad + \frac{1}{16}(9 - 27r - 9r^2 + 27r^3)f_2 \\
&\quad + \frac{1}{16}(9 + 27r - 9r^2 - 27r^3)f_3 \\
&\quad + \frac{1}{16}(-1 - r + 9r^2 + 9r^3)f_4 \\
&= \sum_{i=1}^4 N_i(r)f_i
\end{aligned}$$

where

$ \begin{aligned} N_1 &= \frac{1}{16}(-1 + r + 9r^2 - 9r^3) \\ N_2 &= \frac{1}{16}(9 - 27r - 9r^2 + 27r^3) \\ N_3 &= \frac{1}{16}(9 + 27r - 9r^2 - 27r^3) \\ N_4 &= \frac{1}{16}(-1 - r + 9r^2 + 9r^3) \end{aligned} $
--



These are identical to [1176, p49]

Verification:

- Let us assume $f(r) = C$, then

$$\hat{f}(r) = \sum_i N_i(r) f_i = \sum_i N_i C = C \sum_i N_i = C$$

so that a constant function is exactly reproduced, as expected.

- Let us assume $f(r) = r$, then $f_1 = -1$, $f_2 = -1/3$, $f_3 = 1/3$ and $f_4 = +1$. We then have

$$\begin{aligned}
\hat{f}(r) &= \sum_i N_i(r) f_i \\
&= -N_1(r) - \frac{1}{3}N_2(r) + \frac{1}{3}N_3(r) + N_4(r) \\
&= [-(-1 + r + 9r^2 - 9r^3) \\
&\quad - \frac{1}{3}(9 - 27r - 9r^2 - 27r^3) \\
&\quad + \frac{1}{3}(9 + 27r - 9r^2 + 27r^3) \\
&\quad + (-1 - r + 9r^2 + 9r^3)]/16 \\
&= [-r + 9r + 9r - r]/16 + \dots 0\dots \\
&= r
\end{aligned} \tag{102}$$

The basis functions derivative are given by

$\frac{\partial N_1}{\partial r}$	$=$	$\frac{1}{16}(1 + 18r - 27r^2)$
$\frac{\partial N_2}{\partial r}$	$=$	$\frac{1}{16}(-27 - 18r + 81r^2)$
$\frac{\partial N_3}{\partial r}$	$=$	$\frac{1}{16}(+27 - 18r - 81r^2)$
$\frac{\partial N_4}{\partial r}$	$=$	$\frac{1}{16}(-1 + 18r + 27r^2)$

Verification:

- Let us assume $f(r) = C$, then

$$\begin{aligned}
\frac{\partial \hat{f}}{\partial r} &= \sum_i \frac{\partial N_i}{\partial r} f_i \\
&= C \sum_i \frac{\partial N_i}{\partial r} \\
&= \frac{C}{16} [(1 + 18r - 27r^2) \\
&\quad + (-27 - 18r + 81r^2) \\
&\quad + (+27 - 18r - 81r^2) \\
&\quad + (-1 + 18r + 27r^2)] \\
&= 0
\end{aligned}$$

- Let us assume $f(r) = r$, then $f_1 = -1$, $f_2 = -1/3$, $f_3 = 1/3$ and $f_4 = +1$. We then have

$$\begin{aligned}
\frac{\partial \hat{f}}{\partial r} &= \sum_i \frac{\partial N_i}{\partial r} f_i \\
&= \frac{1}{16} [-(1 + 18r - 27r^2) \\
&\quad - \frac{1}{3}(-27 - 18r + 81r^2) \\
&\quad + \frac{1}{3}(+27 - 18r - 81r^2) \\
&\quad + (-1 + 18r + 27r^2)] \\
&= \frac{1}{16} [-2 + 18 + 54r^2 - 54r^2] \\
&= 1
\end{aligned}$$

4.4.4 Quartic basis functions (Q_4)

The 1D basis polynomial is given by

$$f(r) = a + br + cr^2 + dr^3 + er^4$$

with the nodes at position -1,-1/2, 0, +1/2 and +1.

$$\begin{aligned}
f(-1) &= a - b + c - d + e = f_1 \\
f(-1/2) &= a - \frac{b}{2} + \frac{c}{4} - \frac{d}{8} + \frac{e}{16} = f_2 \\
f(0) &= a = f_3 \\
f(+1/2) &= a - \frac{b}{2} + \frac{c}{4} - \frac{d}{8} + \frac{e}{16} = f_4 \\
f(+1) &= a + b + c + d + e = f_5
\end{aligned}$$

or,

$$\begin{pmatrix} 1 & -1 & 1 & -1 & 1 \\ 1 & -1/2 & 1/4 & -1/8 & 1/16 \\ 1 & 0 & 0 & 0 & 0 \\ 1 & 1/2 & 1/4 & 1/8 & 1/16 \\ 1 & 1 & 1 & 1 & 1 \end{pmatrix} \begin{pmatrix} a \\ b \\ c \\ d \\ e \end{pmatrix} = \begin{pmatrix} f_1 \\ f_2 \\ f_3 \\ f_4 \\ f_5 \end{pmatrix} \quad (103)$$

The third line gives $a = f_3$ so that

$$\underbrace{\begin{pmatrix} -1 & 1 & -1 & 1 \\ -1/2 & 1/4 & -1/8 & 1/16 \\ 1/2 & 1/4 & 1/8 & 1/16 \\ 1 & 1 & 1 & 1 \end{pmatrix}}_A \begin{pmatrix} b \\ c \\ d \\ e \end{pmatrix} = \begin{pmatrix} f_1 - f_3 \\ f_2 - f_3 \\ f_4 - f_3 \\ f_5 - f_3 \end{pmatrix} \quad (104)$$

The inverse of the matrix A is:

$$A^{-1} = \frac{1}{6} \begin{pmatrix} 1 & -8 & 8 & -1 \\ -1 & 16 & 16 & -1 \\ -4 & 8 & -8 & 4 \\ 4 & -16 & -16 & 4 \end{pmatrix}$$

so that

$$\begin{pmatrix} b \\ c \\ d \\ e \end{pmatrix} = \frac{1}{6} \begin{pmatrix} 1 & -8 & 8 & -1 \\ -1 & 16 & 16 & -1 \\ -4 & 8 & -8 & 4 \\ 4 & -16 & -16 & 4 \end{pmatrix} \cdot \begin{pmatrix} f_1 - f_3 \\ f_2 - f_3 \\ f_4 - f_3 \\ f_5 - f_3 \end{pmatrix}$$

and then

$$b = \frac{1}{6} (f_1 - 8f_2 + 8f_4 - f_5) \quad (105)$$

$$c = \frac{1}{6} (-f_1 + 16f_2 - 30f_3 + 16f_4 - f_5) \quad (106)$$

$$d = \frac{1}{6} (-4f_1 + 8f_2 - 8f_4 + 4f_5) \quad (107)$$

$$e = \frac{1}{6} (4f_1 - 16f_2 + 24f_3 - 16f_4 + 4f_5) \quad (108)$$

$$f(r) = a + br + cr^2 + dr^3 + er^4 \quad (109)$$

$$= f_3 + \frac{1}{6} (f_1 - 8f_2 + 8f_4 - f_5) r + \frac{1}{6} (-f_1 + 16f_2 - 30f_3 + 16f_4 - f_5) r^2 + \quad (110)$$

$$+ \frac{1}{6} (-4f_1 + 8f_2 - 8f_4 + 4f_5) r^3 + \frac{1}{6} (4f_1 - 16f_2 + 24f_3 - 16f_4 + 4f_5) r^4 \quad (111)$$

$$= \frac{1}{6} (r - r^2 - 4r^3 + 4r^4) f_1 \quad (112)$$

$$+ \frac{1}{6} (-8r + 16r^2 + 8r^3 - 16r^4) f_2 \quad (113)$$

$$+ (1 - 5r^2 + 4r^4) f_3 \quad (114)$$

$$+ \frac{1}{6} (8r + 16r^2 - 8r^3 - 16r^4) f_4 \quad (115)$$

$$+ \frac{1}{6} (-r - r^2 + 4r^3 + 4r^4) f_5 \quad (116)$$

Finally

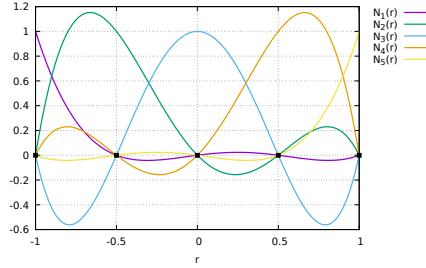
$$N_1(r) = \frac{1}{6} (r - r^2 - 4r^3 + 4r^4) \quad (117)$$

$$N_2(r) = \frac{1}{6} (-8r + 16r^2 + 8r^3 - 16r^4) \quad (118)$$

$$N_3(r) = (1 - 5r^2 + 4r^4) \quad (119)$$

$$N_4(r) = \frac{1}{6} (8r + 16r^2 - 8r^3 - 16r^4) \quad (120)$$

$$N_5(r) = \frac{1}{6} (-r - r^2 + 4r^3 + 4r^4) \quad (121)$$



The basis functions derivative are given by

$$\frac{\partial N_1}{\partial r} = \frac{1}{6} (1 - 2r - 12r^2 + 16r^3) \quad (122)$$

$$\frac{\partial N_2}{\partial r} = \frac{1}{6} (-8 + 32r + 24r^2 - 64r^3) \quad (123)$$

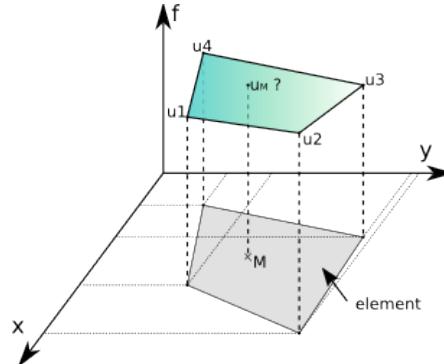
$$\frac{\partial N_3}{\partial r} = -10r + 16r^3 \quad (124)$$

$$\frac{\partial N_4}{\partial r} = \frac{1}{6} (8 + 32r - 24r^2 - 64r^3) \quad (125)$$

$$\frac{\partial N_5}{\partial r} = \frac{1}{6} (-1 - 2r + 12r^2 + 16r^3) \quad (126)$$

4.5 Elements and basis functions in 2D

Let us for a moment consider a single quadrilateral element in the xy -plane, as shown on the following figure:



Let us assume that we know the values of a given field u at the vertices. For a given point M inside the element in the plane, what is the value of the field u at this point? It makes sense to postulate that $u_M = u(x_M, y_M)$ will be given by

$$u_M = \phi(u_1, u_2, u_3, u_4, x_M, y_M)$$

where ϕ is a function to be determined. Although ϕ is not unique, we can decide to express the value u_M as a weighed sum of the values at the vertices u_i . One option could be to assign all four vertices the same weight, say $1/4$ so that $u_M = (u_1 + u_2 + u_3 + u_4)/4$, i.e. u_M is simply given by the arithmetic mean of the vertices values. This approach suffers from a major drawback as it does not use the location of point M inside the element. For instance, when $(x_M, y_M) \rightarrow (x_2, y_2)$ we expect $u_M \rightarrow u_2$.

In light of this, we could now assume that the weights would depend on the position of M in a continuous fashion:

$$u(x_M, y_M) = \sum_{i=1}^4 N_i(x_M, y_M) u_i$$

where the N_i are continuous ("well behaved") functions which have the property:

$$N_i(x_j, y_j) = \delta_{ij}$$

or, in other words:

$$N_3(x_1, y_1) = 0 \quad (127)$$

$$N_3(x_2, y_2) = 0 \quad (128)$$

$$N_3(x_3, y_3) = 1 \quad (129)$$

$$N_3(x_4, y_4) = 0 \quad (130)$$

The functions N_i are commonly called basis functions.

Omitting the M subscripts for any point inside the element, the velocity components u and v are given by:

$$\hat{u}(x, y) = \sum_{i=1}^4 N_i(x, y) u_i \quad (131)$$

$$\hat{v}(x, y) = \sum_{i=1}^4 N_i(x, y) v_i \quad (132)$$

Rather interestingly, one can now easily compute velocity gradients (and therefore the strain rate tensor) since we have assumed the basis functions to be "well behaved" (in this case differentiable):

$$\dot{\epsilon}_{xx}(x, y) = \frac{\partial u}{\partial x} = \sum_{i=1}^4 \frac{\partial N_i}{\partial x} u_i \quad (133)$$

$$\dot{\epsilon}_{yy}(x, y) = \frac{\partial v}{\partial y} = \sum_{i=1}^4 \frac{\partial N_i}{\partial y} v_i \quad (134)$$

$$\dot{\epsilon}_{xy}(x, y) = \frac{1}{2} \frac{\partial u}{\partial y} + \frac{1}{2} \frac{\partial v}{\partial x} = \frac{1}{2} \sum_{i=1}^4 \frac{\partial N_i}{\partial y} u_i + \frac{1}{2} \sum_{i=1}^4 \frac{\partial N_i}{\partial x} v_i \quad (135)$$

How we actually obtain the exact form of the basis functions is explained in the coming section.

4.5.1 Bilinear basis functions in 2D (Q_1)

In this section, we place ourselves in the most favorable case, i.e. the element is a square defined by $-1 < r < 1$, $-1 < s < 1$ in the Cartesian coordinates system (r, s) :

```
3=====2
|           |   (r_0,s_0)=(-1,-1)
|           |   (r_1,s_1)=(+1,-1)
|           |   (r_2,s_2)=(+1,+1)
|           |   (r_3,s_3)=(-1,+1)
|
0=====1
```

This element is commonly called the reference element. How we go from the (x, y) coordinate system to the (r, s) once and vice versa will be dealt later on. For now, the basis functions in the above reference element and in the reduced coordinates system (r, s) are given by:

$$\begin{aligned} N_1(r, s) &= 0.25(1 - r)(1 - s) \\ N_2(r, s) &= 0.25(1 + r)(1 - s) \\ N_3(r, s) &= 0.25(1 + r)(1 + s) \\ N_4(r, s) &= 0.25(1 - r)(1 + s) \end{aligned}$$

The partial derivatives of these functions with respect to r and s automatically follow:

$$\begin{aligned} \frac{\partial N_1}{\partial r}(r, s) &= -0.25(1 - s) & \frac{\partial N_1}{\partial s}(r, s) &= -0.25(1 - r) \\ \frac{\partial N_2}{\partial r}(r, s) &= +0.25(1 - s) & \frac{\partial N_2}{\partial s}(r, s) &= -0.25(1 + r) \\ \frac{\partial N_3}{\partial r}(r, s) &= +0.25(1 + s) & \frac{\partial N_3}{\partial s}(r, s) &= +0.25(1 + r) \\ \frac{\partial N_4}{\partial r}(r, s) &= -0.25(1 + s) & \frac{\partial N_4}{\partial s}(r, s) &= +0.25(1 - r) \end{aligned}$$

Let us go back to Eq.(132). And let us assume that the function $v(r, s) = C$ so that $v_i = C$ for $i = 1, 2, 3, 4$. It then follows that

$$\hat{v}(r, s) = \sum_{i=1}^4 N_i(r, s) v_i = C \sum_{i=1}^4 N_i(r, s) = C[N_1(r, s) + N_2(r, s) + N_3(r, s) + N_4(r, s)] = C$$

This is a very important property: if the v function used to assign values at the vertices is constant, then the value of \hat{v} anywhere in the element is exactly C . If we now turn to the derivatives of v with respect to r and s :

$$\frac{\partial \hat{v}}{\partial r}(r, s) = \sum_{i=1}^4 \frac{\partial N_i}{\partial r}(r, s) v_i = C \sum_{i=1}^4 \frac{\partial N_i}{\partial r}(r, s) = C[-0.25(1 - s) + 0.25(1 - s) + 0.25(1 + s) - 0.25(1 + s)] = 0$$

$$\frac{\partial \hat{v}}{\partial s}(r, s) = \sum_{i=1}^4 \frac{\partial N_i}{\partial s}(r, s) v_i = C \sum_{i=1}^4 \frac{\partial N_i}{\partial s}(r, s) = C[-0.25(1 - r) - 0.25(1 + r) + 0.25(1 + r) + 0.25(1 - r)] = 0$$

We reassuringly find that the derivative of a constant field anywhere in the element is exactly zero.

If we now choose $v(r, s) = ar + bs$ with a and b two constant scalars, we find:

$$\hat{v}(r, s) = \sum_{i=1}^4 N_i(r, s) v_i \tag{136}$$

$$= \sum_{i=1}^4 N_i(r, s)(ar_i + bs_i) \tag{137}$$

$$= \underbrace{a \sum_{i=1}^4 N_i(r, s)r_i}_{r} + \underbrace{b \sum_{i=1}^4 N_i(r, s)s_i}_{s} \tag{138}$$

$$\begin{aligned} &= a[0.25(1 - r)(1 - s)(-1) + 0.25(1 + r)(1 - s)(+1) + 0.25(1 + r)(1 + s)(+1) + 0.25(1 - r)(1 + s)(-1)] \\ &+ b[0.25(1 - r)(1 - s)(-1) + 0.25(1 + r)(1 - s)(-1) + 0.25(1 + r)(1 + s)(+1) + 0.25(1 - r)(1 + s)(+1)] \\ &= a[-0.25(1 - r)(1 - s) + 0.25(1 + r)(1 - s) + 0.25(1 + r)(1 + s) - 0.25(1 - r)(1 + s)] \\ &+ b[-0.25(1 - r)(1 - s) - 0.25(1 + r)(1 - s) + 0.25(1 + r)(1 + s) + 0.25(1 - r)(1 + s)] \\ &= ar + bs \end{aligned} \tag{139}$$

verify above eq. This set of bilinear shape functions is therefore capable of exactly representing a bilinear field. The derivatives are:

$$\frac{\partial \hat{v}}{\partial r}(r, s) = \sum_{i=1}^4 \frac{\partial N_i}{\partial r}(r, s) v_i \quad (140)$$

$$= a \sum_{i=1}^4 \frac{\partial N_i}{\partial r}(r, s) r_i + b \sum_{i=1}^4 \frac{\partial N_i}{\partial r}(r, s) s_i \quad (141)$$

$$= a [-0.25(1-s)(-1) + 0.25(1-s)(+1) + 0.25(1+s)(+1) - 0.25(1+s)(-1)]$$

$$+ b [-0.25(1-s)(-1) + 0.25(1-s)(-1) + 0.25(1+s)(+1) - 0.25(1+s)(+1)]$$

$$= \frac{a}{4} [(1-s) + (1-s) + (1+s) + (1+s)]$$

$$+ \frac{b}{4} [(1-s) - (1-s) + (1+s) - (1+s)]$$

$$= a \quad (142)$$

Here again, we find that the derivative of the bilinear field inside the element is exact: $\frac{\partial \hat{v}}{\partial r} = \frac{\partial v}{\partial r}$.

However, following the same methodology as above, one can easily prove that this is no more true for polynomials of degree strivtly higher than 1. This fact has serious consequences: if the solution to the problem at hand is for instance a parabola, the Q_1 shape functions cannot represent the solution properly, but only by approximating the parabola in each element by a line. As we will see later, Q_2 basis functions can remedy this problem by containing themselves quadratic terms.

4.5.2 Biquadratic basis functions in 2D (Q_2)

This element is part of the so-called LAgrange family.

citation needed

Inside an element the local numbering of the nodes is as follows:

3=====6=====2							
			(r_0,s_0)=(-1,-1)	(r_4,s_4)=(0,-1)			
			(r_1,s_1)=(+1,-1)	(r_5,s_5)=(+1, 0)			
7=====8=====5			(r_2,s_2)=(+1,+1)	(r_6,s_6)=(0,+1)			
			(r_3,s_3)=(-1,+1)	(r_7,s_7)=(-1, 0)			
				(r_8,s_8)=(0, 0)			
0=====4=====1							

Note that this numbering is also employed in [1176, 56]. The basis polynomial is then

$$f(r, s) = a + br + cs + drs + er^2 + fs^2 + gr^2s + hrs^2 + ir^2s^2$$

The velocity shape functions are given by:

$N_0(r, s) = \frac{1}{2}r(r-1)\frac{1}{2}s(s-1)$
$N_1(r, s) = \frac{1}{2}r(r+1)\frac{1}{2}s(s-1)$
$N_2(r, s) = \frac{1}{2}r(r+1)\frac{1}{2}s(s+1)$
$N_3(r, s) = \frac{1}{2}r(r-1)\frac{1}{2}s(s+1)$
$N_4(r, s) = (1-r^2)\frac{1}{2}s(s-1)$
$N_5(r, s) = \frac{1}{2}r(r+1)(1-s^2)$
$N_6(r, s) = (1-r^2)\frac{1}{2}s(s+1)$
$N_7(r, s) = \frac{1}{2}r(r-1)(1-s^2)$
$N_8(r, s) = (1-r^2)(1-s^2)$

These are identical to [1176, p57]. Their derivatives are given by:

$\frac{\partial N_0}{\partial r} = \frac{1}{2}(2r-1)\frac{1}{2}s(s-1)$	$\frac{\partial N_0}{\partial s} = \frac{1}{2}r(r-1)\frac{1}{2}(2s-1)$
$\frac{\partial N_1}{\partial r} = \frac{1}{2}(2r+1)\frac{1}{2}s(s-1)$	$\frac{\partial N_1}{\partial s} = \frac{1}{2}r(r+1)\frac{1}{2}(2s-1)$
$\frac{\partial N_2}{\partial r} = \frac{1}{2}(2r+1)\frac{1}{2}s(s+1)$	$\frac{\partial N_2}{\partial s} = \frac{1}{2}r(r+1)\frac{1}{2}(2s+1)$
$\frac{\partial N_3}{\partial r} = \frac{1}{2}(2r-1)\frac{1}{2}s(s+1)$	$\frac{\partial N_3}{\partial s} = \frac{1}{2}r(r-1)\frac{1}{2}(2s+1)$
$\frac{\partial N_4}{\partial r} = (-2r)\frac{1}{2}s(s-1)$	$\frac{\partial N_4}{\partial s} = (1-r^2)\frac{1}{2}(2s-1)$
$\frac{\partial N_5}{\partial r} = \frac{1}{2}(2r+1)(1-s^2)$	$\frac{\partial N_5}{\partial s} = \frac{1}{2}r(r+1)(-2s)$
$\frac{\partial N_6}{\partial r} = (-2r)\frac{1}{2}s(s+1)$	$\frac{\partial N_6}{\partial s} = (1-r^2)\frac{1}{2}(2s+1)$
$\frac{\partial N_7}{\partial r} = \frac{1}{2}(2r-1)(1-s^2)$	$\frac{\partial N_7}{\partial s} = \frac{1}{2}r(r-1)(-2s)$
$\frac{\partial N_8}{\partial r} = (-2r)(1-s^2)$	$\frac{\partial N_8}{\partial s} = (1-r^2)(-2s)$

4.5.3 Eight node serendipity basis functions in 2D ($Q_2^{(8)}$)

The serendipity elements are those rectangular elements which have no interior nodes [1551, p65].

Inside an element the local numbering of the nodes is as follows:

```
3=====6=====2
|       |       |   (r_0,s_0)=(-1,-1)   (r_4,s_4)=( 0,-1)
|       |       |   (r_1,s_1)=(+1,-1)   (r_5,s_5)=(+1, 0)
7=====+=====5   (r_2,s_2)=(+1,+1)   (r_6,s_6)=( 0,+1)
|       |       |   (r_3,s_3)=(-1,+1)   (r_7,s_7)=(-1, 0)
|       |       |
0=====4=====1
```

The main difference with the Q_2 element resides in the fact that there is no node in the middle of the element. The basis polynomial is then

$$f(r, s) = a + br + cs + drs + er^2 + fs^2 + gr^2s + hrs^2$$

Note that absence of the r^2s^2 term which was previously associated to the center node. We find that

$$N_0(r, s) = \frac{1}{4}(1-r)(1-s)(-r-s-1) \quad (143)$$

$$N_1(r, s) = \frac{1}{4}(1+r)(1-s)(r-s-1) \quad (144)$$

$$N_2(r, s) = \frac{1}{4}(1+r)(1+s)(r+s-1) \quad (145)$$

$$N_3(r, s) = \frac{1}{4}(1-r)(1+s)(-r+s-1) \quad (146)$$

$$N_4(r, s) = \frac{1}{2}(1-r^2)(1-s) \quad (147)$$

$$N_5(r, s) = \frac{1}{2}(1+r)(1-s^2) \quad (148)$$

$$N_6(r, s) = \frac{1}{2}(1-r^2)(1+s) \quad (149)$$

$$N_7(r, s) = \frac{1}{2}(1-r)(1-s^2) \quad (150)$$

The shape functions at the mid side nodes are products of a second order polynomial parallel to side and a linear function perpendicular to the side while shape functions for corner nodes are modifications of the bilinear quadrilateral element.

$$\frac{\partial N_0}{\partial r}(r, s) = -\frac{1}{4}(s-1)(2r+s) \quad (151)$$

$$\frac{\partial N_1}{\partial r}(r, s) = -\frac{1}{4}(s-1)(2r-s) \quad (152)$$

$$\frac{\partial N_2}{\partial r}(r, s) = \frac{1}{4}(s+1)(2r+s) \quad (153)$$

$$\frac{\partial N_3}{\partial r}(r, s) = \frac{1}{4}(s+1)(2r-s) \quad (154)$$

$$\frac{\partial N_4}{\partial r}(r, s) = r(s-1) \quad (155)$$

$$\frac{\partial N_5}{\partial r}(r, s) = \frac{1}{2}(1-s^2) \quad (156)$$

$$\frac{\partial N_6}{\partial r}(r, s) = -r(s+1) \quad (157)$$

$$\frac{\partial N_7}{\partial r}(r, s) = -\frac{1}{2}(1-s^2) \quad (158)$$

$$\frac{\partial N_0}{\partial s}(r, s) = -\frac{1}{4}(r-1)(r+2s) \quad (159)$$

$$\frac{\partial N_1}{\partial s}(r, s) = -\frac{1}{4}(r+1)(r-2s) \quad (160)$$

$$\frac{\partial N_2}{\partial s}(r, s) = \frac{1}{4}(r+1)(r+2s) \quad (161)$$

$$\frac{\partial N_3}{\partial s}(r, s) = \frac{1}{4}(r-1)(r-2s) \quad (162)$$

$$\frac{\partial N_4}{\partial s}(r, s) = -\frac{1}{2}(1-r^2) \quad (163)$$

$$\frac{\partial N_5}{\partial s}(r, s) = -(r+1)s \quad (164)$$

$$\frac{\partial N_6}{\partial s}(r, s) = \frac{1}{2}(1-r^2) \quad (165)$$

$$\frac{\partial N_7}{\partial s}(r, s) = (r-1)s \quad (166)$$

4.5.4 Bicubic basis functions in 2D (Q_3)

Inside an element the local numbering of the nodes is as follows:

12==13==14==15	$(r,s)_{\{00\}}=(-1,-1)$	$(r,s)_{\{08\}}=(-1,+1/3)$
	$(r,s)_{\{01\}}=(-1/3,-1)$	$(r,s)_{\{09\}}=(-1/3,+1/3)$
08==09==10==11	$(r,s)_{\{02\}}=(+1/3,-1)$	$(r,s)_{\{10\}}=(+1/3,+1/3)$
	$(r,s)_{\{03\}}=(+1,-1)$	$(r,s)_{\{11\}}=(+1,+1/3)$
04==05==06==07	$(r,s)_{\{04\}}=(-1,-1/3)$	$(r,s)_{\{12\}}=(-1,+1)$
	$(r,s)_{\{05\}}=(-1/3,-1/3)$	$(r,s)_{\{13\}}=(-1/3,+1)$
00==01==02==03	$(r,s)_{\{06\}}=(+1/3,-1/3)$	$(r,s)_{\{14\}}=(+1/3,+1)$
	$(r,s)_{\{07\}}=(+1,-1/3)$	$(r,s)_{\{15\}}=(+1,+1)$

The velocity shape functions are given by:

$$N_1(r) = (-1 + r + 9r^2 - 9r^3)/16$$

$$N_1(t) = (-1 + t + 9t^2 - 9t^3)/16$$

$$N_2(r) = (+9 - 27r - 9r^2 + 27r^3)/16$$

$$N_2(t) = (+9 - 27t - 9t^2 + 27t^3)/16$$

$$N_3(r) = (+9 + 27r - 9r^2 - 27r^3)/16$$

$$N_3(t) = (+9 + 27t - 9t^2 - 27t^3)/16$$

$$N_4(r) = (-1 - r + 9r^2 + 9r^3)/16$$

$$N_4(t) = (-1 - t + 9t^2 + 9t^3)/16$$

$$\begin{aligned}
N_{01}(r, s) &= N_1(r)N_1(s) = (-1 + r + 9r^2 - 9r^3)/16 * (-1 + t + 9s^2 - 9s^3)/16 \\
N_{02}(r, s) &= N_2(r)N_1(s) = (+9 - 27r - 9r^2 + 27r^3)/16 * (-1 + t + 9s^2 - 9s^3)/16 \\
N_{03}(r, s) &= N_3(r)N_1(s) = (+9 + 27r - 9r^2 - 27r^3)/16 * (-1 + t + 9s^2 - 9s^3)/16 \\
N_{04}(r, s) &= N_4(r)N_1(s) = (-1 - r + 9r^2 + 9r^3)/16 * (-1 + t + 9s^2 - 9s^3)/16 \\
N_{05}(r, s) &= N_1(r)N_2(s) = (-1 + r + 9r^2 - 9r^3)/16 * (9 - 27s - 9s^2 + 27s^3)/16 \\
N_{06}(r, s) &= N_2(r)N_2(s) = (+9 - 27r - 9r^2 + 27r^3)/16 * (9 - 27s - 9s^2 + 27s^3)/16 \\
N_{07}(r, s) &= N_3(r)N_2(s) = (+9 + 27r - 9r^2 - 27r^3)/16 * (9 - 27s - 9s^2 + 27s^3)/16 \\
N_{08}(r, s) &= N_4(r)N_2(s) = (-1 - r + 9r^2 + 9r^3)/16 * (9 - 27s - 9s^2 + 27s^3)/16 \\
N_{09}(r, s) &= N_1(r)N_3(s) = \dots \tag{167} \\
N_{10}(r, s) &= N_2(r)N_3(s) = \dots \tag{168} \\
N_{11}(r, s) &= N_3(r)N_3(s) = \dots \tag{169} \\
N_{12}(r, s) &= N_4(r)N_3(s) = \dots \tag{170} \\
N_{13}(r, s) &= N_1(r)N_4(s) = \dots \tag{171} \\
N_{14}(r, s) &= N_2(r)N_4(s) = \dots \tag{172} \\
N_{15}(r, s) &= N_3(r)N_4(s) = \dots \tag{173} \\
N_{16}(r, s) &= N_4(r)N_4(s) = \dots \tag{174}
\end{aligned}$$

4.5.5 Biquartic basis functions in 2D (Q_4)

Inside an element the local numbering of the nodes is as follows:

```

20====21====22====23====24
||    ||    ||    ||
15====16====17====18====19
||    ||    ||    ||
10====11====12====13====14
||    ||    ||    ||
05====06====07====08====09
||    ||    ||    ||
00====01====02====03====04

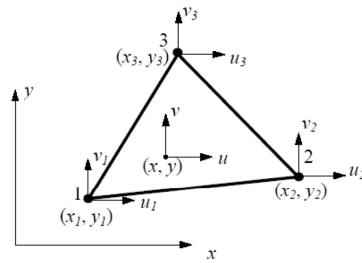
```

4.5.6 Linear basis functions for triangles in 2D (P_1)

Velocities (or displacements) (u, v) in a plane element are interpolated from nodal velocities (u_i, v_i) using shape functions N_i as follows,

$$\begin{pmatrix} u \\ v \end{pmatrix} = \begin{pmatrix} N_1 & 0 & N_2 & 0 & N_3 & 0 \\ 0 & N_1 & 0 & N_2 & 0 & N_3 \end{pmatrix} \cdot \begin{pmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \\ u_3 \\ v_3 \end{pmatrix}$$

This is the simplest 2D element, which is also called linear triangular element.



For this element, we have three nodes at the vertices of the triangle, which are numbered around the element in the counterclockwise direction. Each node has two degrees of freedom (can move in the x and y directions). The velocities u and v are assumed to be linear functions within the element, that is,

$$u = b_1 + b_2x + b_3y \quad v = b_4 + b_5x + b_6y$$

where b_i are constants to be determined and which depend on the triangle shape. Note that the strain rate components are then given by

$$\dot{\varepsilon}_{xx} = b_2 \quad \dot{\varepsilon}_{yy} = b_6 \quad \dot{\varepsilon}_{xy} = (b_3 + b_5)/2$$

and are constant throughout the element.

The velocities should satisfy the following six equations:

$$\begin{aligned} u_1 &= b_1 + b_2x_1 + b_3y_1 \\ u_2 &= b_1 + b_2x_2 + b_3y_2 \\ u_3 &= b_1 + b_2x_3 + b_3y_3 \\ v_1 &= b_4 + b_5x_1 + b_6y_1 \\ v_2 &= b_4 + b_5x_2 + b_6y_2 \\ v_3 &= b_4 + b_5x_3 + b_6y_3 \end{aligned}$$

This can be re-written:

$$\begin{pmatrix} u_1 \\ u_2 \\ u_3 \end{pmatrix} = \begin{pmatrix} 1 & x_1 & y_1 \\ 1 & x_2 & y_2 \\ 1 & x_3 & y_3 \end{pmatrix} \cdot \begin{pmatrix} b_1 \\ b_2 \\ b_3 \end{pmatrix}$$

In order to obtain b_1, b_2, b_3 we need to solve this system, or simply to compute the inverse of the 3×3 \mathbf{M} matrix, as explained in G.2.

We define $D = \det(\mathbf{M})$ and we get

$$\begin{aligned} \begin{pmatrix} b_1 \\ b_2 \\ b_3 \end{pmatrix} &= \frac{1}{D} \tilde{\mathbf{M}} \cdot \begin{pmatrix} u_1 \\ u_2 \\ u_3 \end{pmatrix} \\ \begin{pmatrix} b_4 \\ b_5 \\ b_6 \end{pmatrix} &= \frac{1}{D} \tilde{\mathbf{M}} \cdot \begin{pmatrix} v_1 \\ v_2 \\ v_3 \end{pmatrix} \end{aligned}$$

The matrix $\tilde{\mathbf{M}}$ writes:

$$\tilde{\mathbf{M}} = \begin{pmatrix} x_2y_3 - x_3y_2 & -(y_3 - y_2) & x_3 - x_2 \\ -(x_1y_3 - x_3y_1) & y_3 - y_1 & -(x_3 - x_1) \\ x_1y_2 - x_2y_1 & -(y_2 - y_1) & x_2 - x_1 \end{pmatrix}$$

ie,

$$\tilde{\mathbf{M}} = \begin{pmatrix} x_2y_3 - x_3y_2 & y_2 - y_3 & x_3 - x_2 \\ x_3y_1 - x_1y_3 & y_3 - y_1 & x_1 - x_3 \\ x_1y_2 - x_2y_1 & y_1 - y_2 & x_2 - x_1 \end{pmatrix}$$

and finally the linear shape functions are given by:

$$\begin{aligned} N_1(x, y) &= \frac{1}{D}[(x_2y_3 - x_3y_2) + (y_2 - y_3)x + (x_3 - x_2)y] \\ N_2(x, y) &= \frac{1}{D}[(x_3y_1 - x_1y_3) + (y_3 - y_1)x + (x_1 - x_3)y] \\ N_3(x, y) &= \frac{1}{D}[(x_1y_2 - x_2y_1) + (y_1 - y_2)x + (x_2 - x_1)y] \end{aligned}$$

Note that the area A of the triangle is given by:

$$A = \frac{1}{2}D = \frac{1}{2} \begin{vmatrix} 1 & x_1 & y_1 \\ 1 & x_2 & y_2 \\ 1 & x_3 & y_3 \end{vmatrix}$$

If we now consider the reference element in the reduced coordinates space (r, s) :

```

2
|\ \
| \ \
|   \ \
|     \ \
(r_0,s_0)=(0,0)
(r_1,s_1)=(1,0)
(r_2,s_2)=(0,2)
0=====1

```

The basis polynomial is then

$$f(r,s) = a + br + cs$$

and the shape functions:

$$N_0(r,s) = 1 - r - s \quad (175)$$

$$N_1(r,s) = r \quad (176)$$

$$N_2(r,s) = s \quad (177)$$

4.5.7 Linear basis functions for quadrilaterals in 2D (P_1)

```

.=====.
|       |       |
|       3       |   (r_1,s_1)=(0,0)
|       |       |   (r_2,s_2)=(1/2,0)
.=====1==2==.   (r_3,s_3)=(0,1/2)
|       |       |
|       |       |
|       |       |
.=====.

```

Let us assume that the function $f(r,s)$ is to be approximated on $[-1,1] \times [-1,1]$ by

$$f(r,s) = a + br + cs$$

The function f then must fulfil:

$$\begin{aligned} f(r_1, s_1) &= a = f_1 \\ f(r_2, s_2) &= a + \frac{b}{2} = f_2 \\ f(r_3, s_3) &= a + \frac{c}{2} = f_3 \end{aligned} \quad (178)$$

This leads to :

$$a = f_1 \quad b = 2(f_2 - f_1) \quad c = 2(f_3 - f_1)$$

Then

$$f(r,s) = f_1 + 2(f_2 - f_1)r + 2(f_3 - f_1)s$$

or,

$$f(r) = \sum_{i=1}^3 N_i(r,s) f_i$$

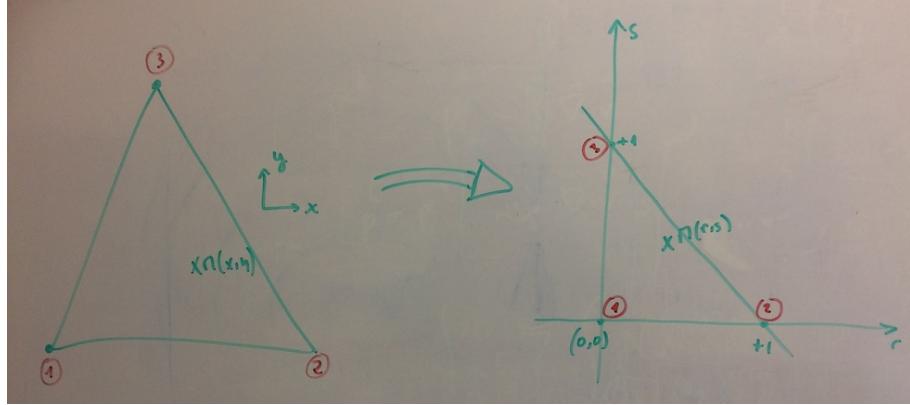
with

$$\begin{aligned} N_1(r) &= 1 - 2(r+s) \\ N_2(r) &= 2r \\ N_3(r) &= 2s \end{aligned} \quad (179)$$

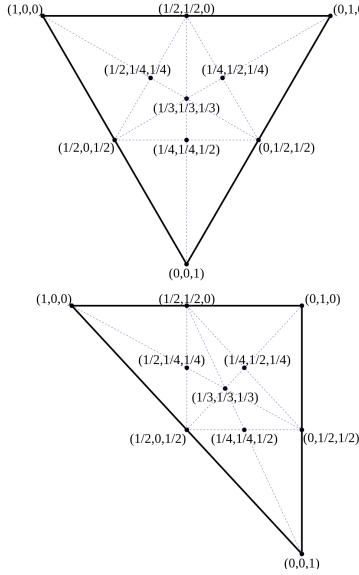
4.5.8 Enriched linear basis functions in triangles (P_1^+)

As we will see in Section 6.2.8 the above P_1 can be enriched with a so-called bubble function. The bubble function of the MINI element is described in [41] as being $\lambda_1\lambda_2\lambda_3$ where λ_i are the so-called barycentric coordinates¹⁷.

$$\begin{aligned}\lambda_1 &= \frac{(y_2 - y_3)(x - x_3) + (x_3 - x_2)(y - y_3)}{(y_2 - y_3)(x_1 - x_3) + (x_3 - x_2)(y_1 - y_3)} \\ \lambda_2 &= \frac{(y_3 - y_1)(x - x_3) + (x_1 - x_3)(y - y_3)}{(y_2 - y_3)(x_1 - x_3) + (x_3 - x_2)(y_1 - y_3)} \\ \lambda_3 &= 1 - \lambda_1 - \lambda_2\end{aligned}$$



representation of the element in the real coordinate system (x, y) and in the reduced coordinate system (r, s)



Barycentric coordinates $(\lambda_1, \lambda_2, \lambda_3)$ on an equilateral triangle and on a right triangle.

In the reference triangle, the barycentric coordinates write

$$\begin{aligned}\lambda_1 &= \frac{(s_2 - s_3)(r - r_3) + (r_3 - r_2)(s - s_3)}{(s_2 - s_3)(r_1 - r_3) + (r_3 - r_2)(s_1 - s_3)} = \frac{(-1)(r) + (-1)(s - 1)}{(-1)(0) + (-1)(-1)} = -r - s + 1 \\ \lambda_2 &= \frac{(s_3 - s_1)(r - r_3) + (r_1 - r_3)(s - s_3)}{(s_2 - s_3)(r_1 - r_3) + (r_3 - r_2)(s_1 - s_3)} = \frac{(1)(r) + (0)(s - 1)}{(-1)(0) + (-1)(-1)} = r \\ \lambda_3 &= 1 - \lambda_1 - \lambda_2 = 1 - (-r - s + 1) - r = s\end{aligned}$$

¹⁷https://en.wikipedia.org/wiki/Barycentric_coordinate_system

As we have seen before the bubble function is given by $\lambda_1\lambda_2\lambda_3 = (1 - r - s)rs$ and the polynomial form for the shape functions is given by:

$$f(r, s) = a + br + cs + d(1 - r - s)rs$$

Setting the location of the bubble at $r = s = 1/3$, i.e. $\lambda_1\lambda_2\lambda_3 = 1/3$, we then have

$$\begin{aligned} f(r_1, s_1) &= f_1 = a + br_1 + cs_1 + d(1 - r_1 - s_1)r_1s_1 = a \\ f(r_2, s_2) &= f_2 = a + br_2 + cs_2 + d(1 - r_2 - s_2)r_2s_2 = a + b \\ f(r_3, s_3) &= f_3 = a + br_3 + cs_3 + d(1 - r_3 - s_3)r_3s_3 = a + c \\ f(r_4, s_4) &= f_4 = a + br_4 + cs_4 + d(1 - r_4 - s_4)r_4s_4 = a + \frac{b}{3} + \frac{c}{3} + \frac{1}{27} \end{aligned}$$

where point 4 is the location of the bubble. This yields

$$a = f_1 \quad b = f_2 - a = f_2 - f_1 \quad c = f_3 - a = f_3 - f_1$$

and

$$d = 27(f_4 - a - \frac{b}{3} - \frac{c}{3}) = 27(f_4 - f_1 - \frac{f_2 - f_1}{3} - \frac{f_3 - f_1}{3}) = 27(f_4 - \frac{f_1}{3} - \frac{f_2}{3} - \frac{f_3}{3})$$

Finally

$$\begin{aligned} f(r, s) &= a + br + cs + d(1 - r - s)rs \\ &= f_1 + (f_2 - f_1)r + (f_3 - f_1)s + 27(f_4 - \frac{f_1}{3} - \frac{f_2}{3} - \frac{f_3}{3})(1 - r - s)rs \\ &= [1 - r - s - 9(1 - r - s)rs]f_1 + [r - 9(1 - r - s)rs]f_2 + [s - 9(1 - r - s)rs]f_3 + [27(1 - r - s)rs]f_4 \end{aligned}$$

so that

$$f(r, s) = \sum_{i=1}^4 N_i(r, s)f_i$$

with

$$\begin{aligned} N_1(r, s) &= 1 - r - s - 9(1 - r - s)rs \\ N_2(r, s) &= r - 9(1 - r - s)rs \\ N_3(r, s) &= s - 9(1 - r - s)rs \\ N_4(r, s) &= 27(1 - r - s)rs \end{aligned}$$

It is trivial to verify that $\sum_i N_i = 1$ for all values of r, s and the gradients of the shape functions are:

$$\frac{\partial N_1}{\partial r}(r, s) = -1 - 9(1 - 2r - s)s \quad (180)$$

$$\frac{\partial N_2}{\partial r}(r, s) = +1 - 9(1 - 2r - s)s \quad (181)$$

$$\frac{\partial N_3}{\partial r}(r, s) = -9(1 - 2r - s)s \quad (182)$$

$$\frac{\partial N_4}{\partial r}(r, s) = 27(1 - 2r - s)s \quad (183)$$

$$(184)$$

$$\frac{\partial N_1}{\partial s}(r, s) = -1 - 9(1 - r - 2s)r \quad (185)$$

$$\frac{\partial N_2}{\partial s}(r, s) = -9(1 - r - 2s)r \quad (186)$$

$$\frac{\partial N_3}{\partial s}(r, s) = +1 - 9(1 - r - 2s)r \quad (187)$$

$$\frac{\partial N_4}{\partial s}(r, s) = 27(1 - r - 2s)r \quad (188)$$

We have two coordinate systems for the element: the global coordinates (x, y) and the natural coordinates (r, s) . Inside the element, the relation between the two is given by

$$\begin{aligned} x &= N_1x_1 + N_2x_2 + N_3x_3 + N_4x_4 = \sum_i N_i(r, s)x_i \\ y &= N_1y_1 + N_2y_2 + N_3y_3 + N_4y_4 = \sum_i N_i(r, s)y_i \end{aligned} \quad (189)$$

or,

$$\begin{aligned} x &= [1 - r - s - 9(1 - r - s)rs]x_1 + [r - 9(1 - r - s)rs]x_2 + [s - 9(1 - r - s)rs]x_3 + [27(1 - r - s)rs]x_4 \\ &= x_1 - r(x_1 - x_2) - s(x_1 - x_3) + (1 - r - s)rs(-9x_1 - 9x_2 - 9x_3 + 27x_4) \\ &= x_1 - r(x_1 - x_2) - s(x_1 - x_3) + (1 - r - s)rs(-9x_1 - 9x_2 - 9x_3 + 27(x_1 + x_2 + x_3)/3) \\ &= x_1 - r(x_1 - x_2) - s(x_1 - x_3) \\ &= x_1 - rx_{12} - sx_{13} \\ y &= [1 - r - s - 9(1 - r - s)rs]y_1 + [r - 9(1 - r - s)rs]y_2 + [s - 9(1 - r - s)rs]y_3 + [27(1 - r - s)rs]y_4 \\ &= y_1 - r(y_1 - y_2) - s(y_1 - y_3) + (1 - r - s)rs(-9y_1 - 9y_2 - 9y_3 + 27y_4) \\ &= y_1 - r(y_1 - y_2) - s(y_1 - y_3) + (1 - r - s)rs(-9y_1 - 9y_2 - 9y_3 + 27(y_1 + y_2 + y_3)/3) \\ &= y_1 - r(y_1 - y_2) - s(y_1 - y_3) \\ &= y_1 - ry_{12} - sy_{13} \end{aligned}$$

4.5.9 Quadratic basis functions for triangles in 2D (P_2)

```

2
|\ \
| \   (r_0,s_0)=(0,0) (r_3,s_3)=(1/2,0)
5   4   (r_1,s_1)=(1,0) (r_4,s_4)=(1/2,1/2)
|     \   (r_2,s_2)=(0,1) (r_5,s_5)=(0,1/2)
|       \
0====3====1

```

The basis polynomial is then

$$f(r, s) = c_1 + c_2r + c_3s + c_4r^2 + c_5rs + c_6s^2$$

We have

$$\begin{aligned} f_1 = f(r_1, s_1) &= c_1 \\ f_2 = f(r_2, s_2) &= c_1 + c_2 + c_4 \\ f_3 = f(r_3, s_3) &= c_1 + c_3 + c_6 \\ f_4 = f(r_4, s_4) &= c_1 + c_2/2 + c_4/4 \\ f_5 = f(r_5, s_5) &= c_1 + c_2/2 + c_3/2 \\ &\quad + c_4/4 + c_5/4 + c_6/4 \\ f_6 = f(r_6, s_6) &= c_1 + c_3/2 + c_6/4 \end{aligned}$$

This can be cast as $\mathbf{f} = \mathbf{A} \cdot \mathbf{c}$ where \mathbf{A} is a 6x6 matrix:

$$\mathbf{A} = \begin{pmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ 1 & 1 & 0 & 1 & 0 & 0 \\ 1 & 0 & 1 & 0 & 0 & 1 \\ 1 & 1/2 & 0 & 1/4 & 0 & 0 \\ 1 & 1/2 & 1/2 & 1/4 & 1/4 & 1/4 \\ 1 & 0 & 1/2 & 0 & 0 & 1/4 \end{pmatrix}$$

It is rather trivial to compute the inverse of this matrix:

$$\mathbf{A}^{-1} = \begin{pmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ -3 & -1 & 0 & 4 & 0 & 0 \\ -3 & 0 & -1 & 0 & 0 & 4 \\ 2 & 2 & 0 & -4 & 0 & 0 \\ 4 & 0 & 0 & -4 & 4 & -4 \\ 2 & 0 & 2 & 0 & 0 & -4 \end{pmatrix}$$

In the end, one obtains:

$$\begin{aligned} f(r, s) &= f_1 + (-3f_1 - f_2 + 4f_4)r + (-3f_1 - f_3 + 4f_6)s \\ &\quad + (2f_1 + 2f_2 - 4f_4)r^2 + (4f_1 - 4f_4 + 4f_5 - 4f_6)rs \\ &\quad + (2f_1 + 2f_3 - 4f_6)s^2 \\ &= \sum_{i=1}^6 N_i(r, s)f_i \end{aligned} \tag{190}$$

with

$N_1(r, s)$	$=$	$1 - 3r - 3s + 2r^2 + 4rs + 2s^2$
$N_2(r, s)$	$=$	$-r + 2r^2$
$N_3(r, s)$	$=$	$-s + 2s^2$
$N_4(r, s)$	$=$	$4r - 4r^2 - 4rs$
$N_5(r, s)$	$=$	$4rs$
$N_6(r, s)$	$=$	$4s - 4rs - 4s^2$

4.5.10 Enriched quadratic basis functions in triangles (P_2^+)

This is used by the Crouzeix-Raviart element, see Section 6.2.10.

```

03          (r_1,s_1)=(0,0)
|| \\
||  \\
||  \\
||  \\
06  05      (r_5,s_5)=(1/2,1/2)
|| 07  \\
||    \\
01==04==02

```

The shape functions are given by:

[find reference](#)

$$N_1(r, s) = (1 - r - s)(1 - 2r - 2s + 3rs) \tag{191}$$

$$N_2(r, s) = r(2r - 1 + 3s - 3rs - 3s^2) \tag{192}$$

$$N_3(r, s) = s(2s - 1 + 3r - 3r^2 - 3rs) \tag{193}$$

$$N_4(r, s) = 4(1 - r - s)r(1 - 3s) \tag{194}$$

$$N_5(r, s) = 4rs[-2 + 3r + 3s] \tag{195}$$

$$N_6(r, s) = 4(1 - r - s)s(1 - 3r) \tag{196}$$

$$N_7(r, s) = 27(1 - r - s)rs \tag{197}$$

It is then easy to verify that for all shape functions we have $N_i(r_j, s_j) = \delta_{ij}$ where j denotes one of the seven nodes.

The derivatives are as follows:

$$\frac{\partial N_1}{\partial r}(r, s) = r(4 - 6s) - 3s^2 + 7s - 3 \quad (198)$$

$$\frac{\partial N_2}{\partial r}(r, s) = r(4 - 6s) - 3s^2 + 3s - 1 \quad (199)$$

$$\frac{\partial N_3}{\partial r}(r, s) = -3s(2r + s - 1) \quad (200)$$

$$\frac{\partial N_4}{\partial r}(r, s) = 4(3s - 1)(2r + s - 1) \quad (201)$$

$$\frac{\partial N_5}{\partial r}(r, s) = 4s(6r + 3s - 2) \quad (202)$$

$$\frac{\partial N_6}{\partial r}(r, s) = 4s(6r + 3s - 4) \quad (203)$$

$$\frac{\partial N_7}{\partial r}(r, s) = -27s(2r + s - 1) \quad (204)$$

$$\frac{\partial N_1}{\partial s}(r, s) = -3r^2 + r(7 - 6s) + 4s - 3 \quad (205)$$

$$\frac{\partial N_2}{\partial s}(r, s) = -3r(r + 2s - 1) \quad (206)$$

$$\frac{\partial N_3}{\partial s}(r, s) = -3r^2 + r(3 - 6s) + 4s - 1 \quad (207)$$

$$\frac{\partial N_4}{\partial s}(r, s) = 4r(3r + 6s - 4) \quad (208)$$

$$\frac{\partial N_5}{\partial s}(r, s) = 4r(3r + 6s - 2) \quad (209)$$

$$\frac{\partial N_6}{\partial s}(r, s) = 4(3r - 1)(r + 2s - 1) \quad (210)$$

$$\frac{\partial N_7}{\partial s}(r, s) = -27r(r + 2s - 1) \quad (211)$$

Note that the shape functions can also be expressed as a function of the barycentric coordinates, as in the MILAMIN code [466] or in Cuvelier et al, 1986 [465]¹⁸

```

03
|| \\
|| \\
||  \\
05  04
|| 07 \\
||      \\
01==06==02

```

$$N_1(\lambda_1, \lambda_2, \lambda_3) = \eta_1(2\eta_1 - 1) + 3\eta_1\eta_2\eta_3 \quad (212)$$

$$N_2(\lambda_1, \lambda_2, \lambda_3) = \eta_2(2\eta_2 - 1) + 3\eta_1\eta_2\eta_3 \quad (213)$$

$$N_3(\lambda_1, \lambda_2, \lambda_3) = \eta_3(2\eta_3 - 1) + 3\eta_1\eta_2\eta_3 \quad (214)$$

$$N_4(\lambda_1, \lambda_2, \lambda_3) = 4\eta_2\eta_3 - 12\eta_1\eta_2\eta_3 \quad (215)$$

$$N_5(\lambda_1, \lambda_2, \lambda_3) = 4\eta_1\eta_3 - 12\eta_1\eta_2\eta_3 \quad (216)$$

$$N_6(\lambda_1, \lambda_2, \lambda_3) = 4\eta_1\eta_2 - 12\eta_1\eta_2\eta_3 \quad (217)$$

$$N_7(\lambda_1, \lambda_2, \lambda_3) = 27\eta_1\eta_2\eta_3 \quad (218)$$

¹⁸Note that the numbering of the nodes in the book is different with respect to the one above.

VERIFY that when $\eta_1 = 1 - r - s$, $\eta_2 = r$ and $\eta_3 = s$ we find the above r, s shape functions

4.5.11 Cubic basis functions for triangles (P_3)

```

2
|\          (r_0,s_0)=(0,0)  (r_5,s_5)=(2/3,1/3)
| \        (r_1,s_1)=(1,0)  (r_6,s_6)=(1/3,2/3)
7   6      (r_2,s_2)=(0,1)  (r_7,s_7)=(0,2/3)
|   \
8   9   5    (r_3,s_3)=(1/3,0) (r_8,s_8)=(0,1/3)
|       \
0==3==4==1

```

The basis polynomial is then

$$f(r, s) = c_1 + c_2r + c_3s + c_4r^2 + c_5rs + c_6s^2 + c_7r^3 + c_8r^2s + c_9rs^2 + c_{10}s^3$$

$$N_0(r, s) = \frac{9}{2}(1 - r - s)(1/3 - r - s)(2/3 - r - s) \quad (219)$$

$$N_1(r, s) = \frac{9}{2}r(r - 1/3)(r - 2/3) \quad (220)$$

$$N_2(r, s) = \frac{9}{2}s(s - 1/3)(s - 2/3) \quad (221)$$

$$N_3(r, s) = \frac{27}{2}(1 - r - s)r(2/3 - r - s) \quad (222)$$

$$N_4(r, s) = \frac{27}{2}(1 - r - s)r(r - 1/3) \quad (223)$$

$$N_5(r, s) = \frac{27}{2}rs(r - 1/3) \quad (224)$$

$$N_6(r, s) = \frac{27}{2}rs(r - 2/3) \quad (225)$$

$$N_7(r, s) = \frac{27}{2}(1 - r - s)s(s - 1/3) \quad (226)$$

$$N_8(r, s) = \frac{27}{2}(1 - r - s)s(2/3 - r - s) \quad (227)$$

$$N_9(r, s) = 27rs(1 - r - s) \quad (228)$$

verify those

4.6 Elements and basis functions in 3D

4.6.1 Linear basis functions in tetrahedra (P_1)

$$(r_0, s_0) = (0, 0, 0)$$

$$(r_1, s_1) = (1, 0, 0)$$

$$(r_2, s_2) = (0, 2, 0)$$

$$(r_3, s_3) = (0, 0, 1)$$

The basis polynomial is given by

$$f(r, s, t) = c_0 + c_1r + c_2s + c_3t$$

$$f_1 = f(r_1, s_1, t_1) = c_0 \quad (229)$$

$$f_2 = f(r_2, s_2, t_2) = c_0 + c_1 \quad (230)$$

$$f_3 = f(r_3, s_3, t_3) = c_0 + c_2 \quad (231)$$

$$f_4 = f(r_4, s_4, t_4) = c_0 + c_3 \quad (232)$$

which yields:

$$c_0 = f_1 \quad c_1 = f_2 - f_1 \quad c_2 = f_3 - f_1 \quad c_3 = f_4 - f_1$$

$$\begin{aligned} f(r, s, t) &= c_0 + c_1 r + c_2 s + c_3 t \\ &= f_1 + (f_2 - f_1)r + (f_3 - f_1)s + (f_4 - f_1)t \\ &= f_1(1 - r - s - t) + f_2 r + f_3 s + f_4 t \\ &= \sum_i N_i(r, s, t) f_i \end{aligned}$$

Finally,

$N_1(r, s, t) = 1 - r - s - t$
$N_2(r, s, t) = r$
$N_3(r, s, t) = s$
$N_4(r, s, t) = t$

4.6.2 Enriched linear in tetrahedra(P_1^+)

These shape functions would be used in the MINI element, see Section 6.2.8.

In 3D the bubble function looks like $rst(1 - r - s - t)$ so that

$$f(r, s, t) = a + b r + c s + d t + e rst(1 - r - s - t)$$

We have node 1 at location $(r, s, t) = (0, 0, 0)$, node 2 at $(r, s, t) = (1, 0, 0)$, node 3 at $(r, s, t) = (0, 1, 0)$, node 4 at $(r, s, t) = (0, 0, 1)$ and we set the location of the bubble (node 5) at $r = s = t = 1/4$ so that

$$f(r_1, s_1, t_1) = f_1 = a + b r_1 + c s_1 + d t_1 + e r_1 s_1 t_1 (1 - r_1 - s_1 - t_1) \quad (233)$$

$$f(r_2, s_2, t_2) = f_2 = a + b r_2 + c s_2 + d t_2 + e r_2 s_2 t_2 (1 - r_2 - s_2 - t_2) \quad (234)$$

$$f(r_3, s_3, t_3) = f_3 = a + b r_3 + c s_3 + d t_3 + e r_3 s_3 t_3 (1 - r_3 - s_3 - t_3) \quad (235)$$

$$f(r_4, s_4, t_4) = f_4 = a + b r_4 + c s_4 + d t_4 + e r_4 s_4 t_4 (1 - r_4 - s_4 - t_4) \quad (236)$$

$$f(r_5, s_5, t_5) = f_5 = a + b r_5 + c s_5 + d t_5 + e r_5 s_5 t_5 (1 - r_5 - s_5 - t_5) \quad (237)$$

i.e.,

$$f_1 = a \quad (238)$$

$$f_2 = a + b \quad (239)$$

$$f_3 = a + c \quad (240)$$

$$f_4 = a + d \quad (241)$$

$$f_5 = a + b/4 + c/4 + d/4 + e/64(1 - 1/4 - 1/4 - 1/4) \quad (242)$$

$$= a + b/4 + c/4 + d/4 + e/256 \quad (243)$$

Then

$$a = f_1 \quad (244)$$

$$b = f_2 - f_1 \quad (245)$$

$$c = f_3 - f_1 \quad (246)$$

$$d = f_4 - f_1 \quad (247)$$

$$e = 256(f_5 - a - b/4 - c/4 - d/4) \quad (248)$$

$$= 256(f_5 - f_1 - (f_2 - f_1)/4 - (f_3 - f_1)/4 - (f_4 - f_1)/4) \quad (249)$$

$$= 256(-f_1/4 - f_2/4 - f_3/4 - f_4/4 + f_5) \quad (250)$$

$$= 64(-f_1 - f_2 - f_3 - f_4 + 4f_5) \quad (251)$$

Finally:

$$\begin{aligned}
f(r, s, t) &= a + br + cs + dt + erst(1 - r - s - t) \\
&= f_1 + (f_2 - f_1)r + (f_3 - f_1)s + (f_4 - f_1)t + 64(-f_1 - f_2 - f_3 - f_4 + 4f_5)rst(1 - r - s - t) \\
&= f_1[1 - r - s - t - 64rst(1 - r - s - t)] \\
&\quad + f_2[r - 64rst(1 - r - s - t)] \\
&\quad + f_3[s - 64rst(1 - r - s - t)] \\
&\quad + f_4[t - 64rst(1 - r - s - t)] \\
&\quad + f_5[256rst(1 - r - s - t)] \\
&= \sum_{i=1}^5 N_i(r, s, t) f_i
\end{aligned} \tag{252}$$

with

$$N_1(r, s, t) = 1 - r - s - t - 64rst(1 - r - s - t) \tag{253}$$

$$N_2(r, s, t) = r - 64rst(1 - r - s - t) \tag{254}$$

$$N_3(r, s, t) = s - 64rst(1 - r - s - t) \tag{255}$$

$$N_4(r, s, t) = t - 64rst(1 - r - s - t) \tag{256}$$

$$N_5(r, s, t) = +256rst(1 - r - s - t) \tag{257}$$

The derivatives are given by:

$$\frac{\partial N_1}{\partial r}(r, s, t) = -1 - 64st(1 - 2r - s - t)$$

$$\frac{\partial N_2}{\partial r}(r, s, t) = +1 - 64st(1 - 2r - s - t)$$

$$\frac{\partial N_3}{\partial r}(r, s, t) = -64st(1 - 2r - s - t)$$

$$\frac{\partial N_4}{\partial r}(r, s, t) = -64st(1 - 2r - s - t)$$

$$\frac{\partial N_5}{\partial r}(r, s, t) = 256st(1 - 2r - s - t)$$

$$\frac{\partial N_1}{\partial s}(r, s, t) = -1 - 64rt(1 - r - 2s - t)$$

$$\frac{\partial N_2}{\partial s}(r, s, t) = -64rt(1 - r - 2s - t)$$

$$\frac{\partial N_3}{\partial s}(r, s, t) = +1 - 64rt(1 - r - 2s - t)$$

$$\frac{\partial N_4}{\partial s}(r, s, t) = -64rt(1 - r - 2s - t)$$

$$\frac{\partial N_5}{\partial s}(r, s, t) = 256rt(1 - r - 2s - t)$$

$$\frac{\partial N_1}{\partial t}(r, s, t) = -1 - 64rs(1 - r - s - 2t)$$

$$\frac{\partial N_2}{\partial t}(r, s, t) = -64rs(1 - r - s - 2t)$$

$$\frac{\partial N_3}{\partial t}(r, s, t) = -64rs(1 - r - s - 2t)$$

$$\frac{\partial N_4}{\partial t}(r, s, t) = +1 - 64rs(1 - r - s - 2t)$$

$$\frac{\partial N_5}{\partial t}(r, s, t) = 256rs(1 - r - s - 2t)$$

4.6.3 Triquadratic basis functions in 3D (Q_2)

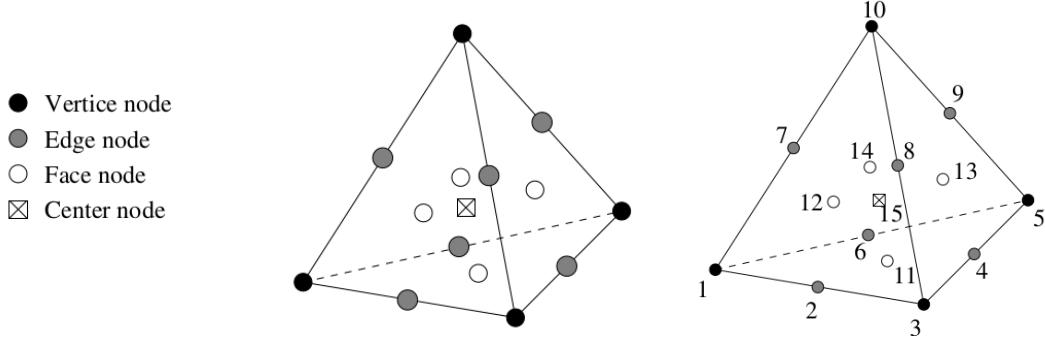
```

t
|
. --s
/
r
          05=====16=====08
          |       |       |
          |       |       |
13=====26=====15   17=====25=====20
          |       |       |
          |       |       |
06=====14=====07   22=====27=====24   01=====12=====04 @ r=-1
          |       |       |
          |       |       |
18=====23=====14   09=====21=====11 @ r=0
          |       |       |
          |       |       |
02=====10=====03 @ r=+1

```

$$\begin{aligned}
N_1 &= 0.5r(r-1) 0.5s(s-1) 0.5t(t-1) \\
N_2 &= 0.5r(r+1) 0.5s(s-1) 0.5t(t-1) \\
N_3 &= 0.5r(r+1) 0.5s(s+1) 0.5t(t-1) \\
N_4 &= 0.5r(r-1) 0.5s(s+1) 0.5t(t-1) \\
N_5 &= 0.5r(r-1) 0.5s(s-1) 0.5t(t+1) \\
N_6 &= 0.5r(r+1) 0.5s(s-1) 0.5t(t+1) \\
N_7 &= 0.5r(r+1) 0.5s(s+1) 0.5t(t+1) \\
N_8 &= 0.5r(r-1) 0.5s(s+1) 0.5t(t+1) \\
N_9 &= (1 - r^2) 0.5s(s-1) 0.5t(t-1) \\
N_{10} &= 0.5r(r+1) (1 - s^2) 0.5t(t-1) \\
N_{11} &= (1 - r^2) 0.5s(s+1) 0.5t(t-1) \\
N_{12} &= 0.5r(r-1) (1 - s^2) 0.5t(t-1) \\
N_{13} &= (1 - r^2) 0.5s(s-1) 0.5t(t+1) \\
N_{14} &= 0.5r(r+1) (1 - s^2) 0.5t(t+1) \\
N_{15} &= (1 - r^2) 0.5s(s+1) 0.5t(t+1) \\
N_{16} &= 0.5r(r-1) (1 - s^2) 0.5t(t+1) \\
N_{17} &= 0.5r(r-1) 0.5s(s-1) (1 - t^2) \\
N_{18} &= 0.5r(r+1) 0.5s(s-1) (1 - t^2) \\
N_{19} &= 0.5r(r+1) 0.5s(s+1) (1 - t^2) \\
N_{20} &= 0.5r(r-1) 0.5s(s+1) (1 - t^2) \\
N_{21} &= (1 - r^2) (1 - s^2) 0.5t(t-1) \\
N_{22} &= (1 - r^2) 0.5s(s-1) (1 - t^2) \\
N_{23} &= 0.5r(r+1) (1 - s^2) (1 - t^2) \\
N_{24} &= (1 - r^2) 0.5s(s+1) (1 - t^2) \\
N_{25} &= 0.5r(r-1) (1 - s^2) (1 - t^2) \\
N_{26} &= (1 - r^2) (1 - s^2) 0.5t(t+1) \\
N_{27} &= (1 - r^2) (1 - s^2) (1 - t^2)
\end{aligned}$$

4.6.4 Enriched quadratic basis functions in tetrahedra (P_2^+)



The velocity shape functions are:

$$\phi_i = \lambda_i(2\lambda_i - 1) + 3(\lambda_i\lambda_j\lambda_k + \lambda_i\lambda_j\lambda_l + \lambda_i\lambda_k\lambda_l) - 4\lambda_i\lambda_j\lambda_k\lambda_l \quad (258)$$

$$\phi_{ij} = 4\lambda_i\lambda_j - 12(\lambda_i\lambda_j\lambda_k + \lambda_i\lambda_j\lambda_l) + 32\lambda_i\lambda_j\lambda_k\lambda_l \quad (259)$$

$$\phi_{ijk} = 27\lambda_i\lambda_j\lambda_k - 108\lambda_i\lambda_j\lambda_k\lambda_l \quad (260)$$

$$\phi_c = 256\lambda_i\lambda_j\lambda_k\lambda_l \quad (261)$$

REFS ??? better definition of functions !

4.6.5 Linear basis functions for tetrahedra (P_1)

This is essentially in the $Q_2 \times P_{-1}$ element.

I choose the reduced coordinates of the pressure nodes to be :

point	r	s	t
1	1/2	-1/2	-1/2
2	-1/2	1/2	-1/2
3	-1/2	-1/2	1/2
4	1/2	1/2	1/2

Inside the element the pressure is given as a linear function of the reduced coordinates r, s, t :

$$p(r, s, t) = a + br + cs + dt$$

This expression must exactly interpolate the pressure at all four pressure nodes:

$$\begin{aligned} p_1 &= p(r_1, s_1, t_1) = a + br_1 + cs_1 + dt_1 = a + b/2 - c/2 - d/2 \\ p_2 &= p(r_2, s_2, t_2) = a + br_2 + cs_2 + dt_2 = a - b/2 + c/2 - d/2 \\ p_3 &= p(r_3, s_3, t_3) = a + br_3 + cs_3 + dt_3 = a - b/2 - c/2 + d/2 \\ p_4 &= p(r_4, s_4, t_4) = a + br_4 + cs_4 + dt_4 = a + b/2 + c/2 + d/2 \end{aligned}$$

or,

$$\begin{pmatrix} 1 & 1/2 & -1/2 & -1/2 \\ 1 & -1/2 & +1/2 & -1/2 \\ 1 & -1/2 & -1/2 & +1/2 \\ 1 & 1/2 & +1/2 & +1/2 \end{pmatrix} \begin{pmatrix} a \\ b \\ c \\ d \end{pmatrix} = \begin{pmatrix} p_1 \\ p_2 \\ p_3 \\ p_4 \end{pmatrix}$$

The matrix is invertible and we get:

$$\begin{pmatrix} a \\ b \\ c \\ d \end{pmatrix} = \begin{pmatrix} 1/4 & 1/4 & 1/4 & 1/4 \\ 1/2 & -1/2 & -1/2 & 1/2 \\ -1/2 & 1/2 & -1/2 & 1/2 \\ -1/2 & -1/2 & 1/2 & 1/2 \end{pmatrix} \begin{pmatrix} p_1 \\ p_2 \\ p_3 \\ p_4 \end{pmatrix}$$

so

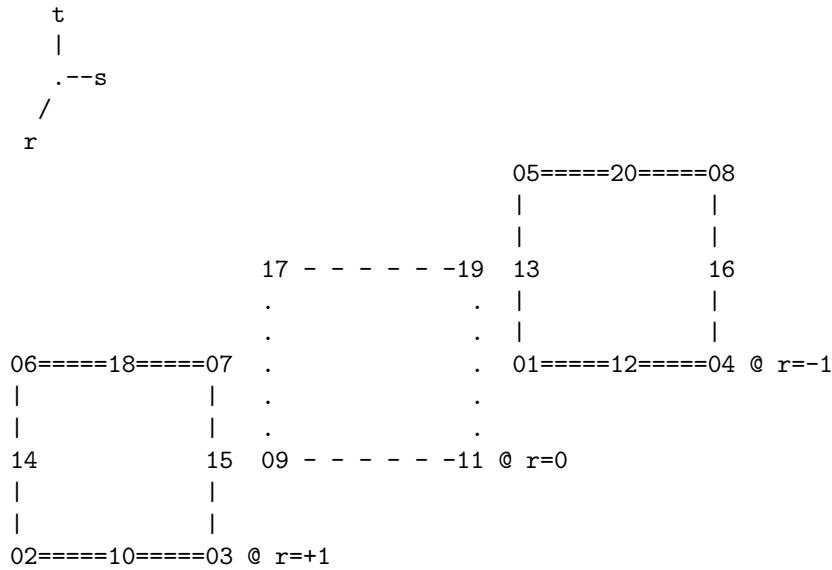
$$\begin{aligned}
 p(r, s, t) &= a + br + cs + dt \\
 &= \frac{1}{4}(p_1 + p_2 + p_3 + p_4) + \frac{1}{2}(p_1 - p_2 - p_3 + p_4)r + \frac{1}{2}(-p_1 + p_2 - p_3 + p_4)s + \frac{1}{2}(-p_1 - p_2 + p_3 + p_4)t \\
 &= \frac{1}{4}(1 + 2r - 2s - 2t)p_1 + \frac{1}{4}(1 - 2r + 2s - 2t)p_2 + \frac{1}{4}(1 - 2r - 2s + 2t)p_3 + \frac{1}{4}(1 + 2r + 2s + 2t)p_4 \\
 &= \sum_{i=1}^4 N_i(r, s, t)p_i
 \end{aligned} \tag{262}$$

with

$$\begin{aligned}
 N_1(r, s, t) &= \frac{1}{4}(1 + 2r - 2s - 2t) \\
 N_2(r, s, t) &= \frac{1}{4}(1 - 2r + 2s - 2t) \\
 N_3(r, s, t) &= \frac{1}{4}(1 - 2r - 2s + 2t) \\
 N_4(r, s, t) &= \frac{1}{4}(1 + 2r + 2s + 2t)
 \end{aligned}$$

4.6.6 20-node serendipity basis functions in 3D ($Q_2^{(20)}$)

The serendipity elements are those rectangular elements which have no interior nodes [1551, p91].

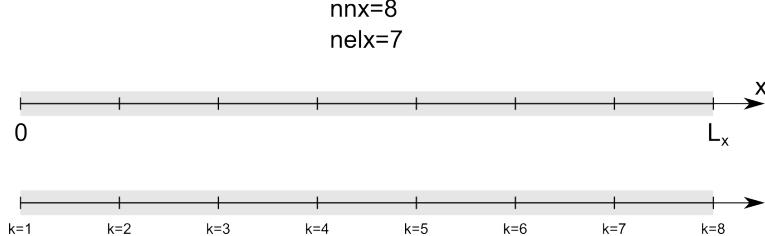


find/build shape functions!

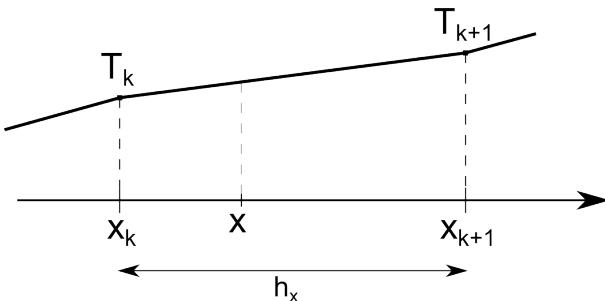
5 Solving the heat transport equation with linear Finite Elements

5.1 The diffusion equation in 1D

Let us consider the following one-dimensional grid:



Its spans the domain Ω of length L_x . It is discretised by means of n_{nx} nodes and $n_{elx} = n_{nx} - 1$ elements. Zooming in on element which is bounded by two nodes k and $k + 1$, its size (also sometimes called diameter) is $h_x = x_{k+1} - x_k$, and the temperature field we wish to compute is located on those nodes so that they are logically called T_k and T_{k+1} :



We focus here on the 1D diffusion equation (no advection, no heat sources):

$$\rho C_p \frac{\partial T}{\partial t} = \frac{\partial}{\partial x} \left(k \frac{\partial T}{\partial x} \right) \quad (263)$$

This is the **strong form** of the ODE to solve. I can multiply this equation by a function¹⁹ $f(x)$ and integrate it over Ω :

$$\int_{\Omega} f(x) \rho C_p \frac{\partial T}{\partial t} dx = \int_{\Omega} f(x) \frac{\partial}{\partial x} \left(k \frac{\partial T}{\partial x} \right) dx \quad (264)$$

Looking at the right hand side, it is of the form $\int uv'$ so that I naturally integrate it by parts:

$$\int_{\Omega} f(x) \frac{\partial}{\partial x} \left(k \frac{\partial T}{\partial x} \right) dx = \left[f(x) k \frac{\partial T}{\partial x} \right]_{\partial\Omega} - \int_{\Omega} \frac{\partial f}{\partial x} k \frac{\partial T}{\partial x} dx \quad (265)$$

Assuming there is no heat flux prescribed on the boundary (i.e. $q_x = -k \partial T / \partial x = 0$),

NOT happy with this statement!!

then:

$$\int_{\Omega} f(x) \frac{\partial}{\partial x} \left(k \frac{\partial T}{\partial x} \right) dx = - \int_{\Omega} \frac{\partial f}{\partial x} k \frac{\partial T}{\partial x} dx \quad (266)$$

We then obtain the **weak form** of the diffusion equation in 1D:

$$\int_{\Omega} f(x) \rho C_p \frac{\partial T}{\partial t} dx + \int_{\Omega} \frac{\partial f}{\partial x} k \frac{\partial T}{\partial x} dx = 0$$

(267)

¹⁹This function should be well-behaved with special properties, but we here assume it is a polynomial function.

We then use the additive property of the integral $\int_{\Omega} \dots = \sum_{elts} \int_{\Omega_e} \dots$ so that

$$\sum_{elts} \left(\underbrace{\int_{\Omega_e} f(x) \rho C_p \frac{\partial T}{\partial t} dx}_{\Lambda_f^e} + \underbrace{\int_{\Omega_e} \frac{\partial f}{\partial x} k \frac{\partial T}{\partial x} dx}_{\Upsilon_f^e} \right) = 0 \quad (268)$$

In order to compute these integrals (analytically or by means of a numerical quadrature), we will need to evaluate T inside the element. However, inside the element, the temperature is not known: all we have is the temperature at the nodes. For $x \in [x_k, x_{k+1}]$ we need to come up with a way to compute the temperature at this location. It makes sense to think that $T(x)$ will then be a function of the temperature at the nodes, i.e. $T(x) = \alpha T_k + \beta T_{k+1}$ where α and β are coefficients. One over-simplified approach would be to assign $T(x) = (T_k + T_{k+1})/2$ but this would make the temperature discontinuous from element to element. The rather logical solution to this problem is a linear temperature field between T_k and T_{k+1} :

$$T(x) = \underbrace{\frac{x_{k+1} - x}{h_x} T_k}_{N_k^\theta(x)} + \underbrace{\frac{x - x_k}{h_x} T_{k+1}}_{N_{k+1}^\theta(x)}$$

where $N_k^\theta(x)$ is the (temperature) shape function associated to node k and $N_{k+1}^\theta(x)$ is the shape function associated to node $k+1$.

Rather reassuringly, we have:

- $x = x_k$ yields $T(x) = T_k$
- $x = x_{k+1}$ yields $T(x) = T_{k+1}$
- $x = (x_k + x_{k+1})/2$ yields $T(x) = (T_k + T_{k+1})/2$

In what follows we abbreviate $\partial T / \partial x$ by \dot{T} . Let us compute Λ_f^e and Υ_f^e separately.

$$\begin{aligned} \Lambda_f^e &= \int_{x_k}^{x_{k+1}} f(x) \rho C_p \dot{T}(x) dx \\ &= \int_{x_k}^{x_{k+1}} f(x) \rho C_p [N_k^\theta(x) \dot{T}_k + N_{k+1}^\theta(x) \dot{T}_{k+1}] dx \\ &= \int_{x_k}^{x_{k+1}} f(x) \rho C_p N_k^\theta(x) \dot{T}_k dx + \int_{x_k}^{x_{k+1}} f(x) \rho C_p N_{k+1}^\theta(x) \dot{T}_{k+1} dx \\ &= \left(\int_{x_k}^{x_{k+1}} f(x) \rho C_p N_k^\theta(x) dx \right) \dot{T}_k + \left(\int_{x_k}^{x_{k+1}} f(x) \rho C_p N_{k+1}^\theta(x) dx \right) \dot{T}_{k+1} \end{aligned}$$

Taking $f(x) = N_k^\theta(x)$ and omitting ' (x) ' in the rhs:

$$\Lambda_{N_k^\theta}^e = \left(\int_{x_k}^{x_{k+1}} \rho C_p N_k^\theta N_k^\theta dx \right) \dot{T}_k + \left(\int_{x_k}^{x_{k+1}} \rho C_p N_k^\theta N_{k+1}^\theta dx \right) \dot{T}_{k+1}$$

Taking $f(x) = N_{k+1}^\theta(x)$ and omitting ' (x) ' in the rhs:

$$\Lambda_{N_{k+1}^\theta}^e = \left(\int_{x_k}^{x_{k+1}} \rho C_p N_{k+1}^\theta N_k^\theta dx \right) \dot{T}_k + \left(\int_{x_k}^{x_{k+1}} \rho C_p N_{k+1}^\theta N_{k+1}^\theta dx \right) \dot{T}_{k+1}$$

We can rearrange these last two equations as follows:

$$\begin{pmatrix} \Lambda_{N_k^\theta}^e \\ \Lambda_{N_{k+1}^\theta}^e \end{pmatrix} = \begin{pmatrix} \int_{x_k}^{x_{k+1}} N_k^\theta \rho C_p N_k^\theta dx & \int_{x_k}^{x_{k+1}} N_k^\theta \rho C_p N_{k+1}^\theta dx \\ \int_{x_k}^{x_{k+1}} N_{k+1}^\theta \rho C_p N_k^\theta dx & \int_{x_k}^{x_{k+1}} N_{k+1}^\theta \rho C_p N_{k+1}^\theta dx \end{pmatrix} \cdot \begin{pmatrix} \dot{T}_k \\ \dot{T}_{k+1} \end{pmatrix}$$

and we can take the integrals outside of the matrix:

$$\begin{pmatrix} \Lambda_{N_k^\theta}^e \\ \Lambda_{N_{k+1}^\theta}^e \end{pmatrix} = \left[\int_{x_k}^{x_{k+1}} \rho C_p \begin{pmatrix} N_k^\theta N_k^\theta & N_k^\theta N_{k+1}^\theta \\ N_{k+1}^\theta N_k^\theta & N_{k+1}^\theta N_{k+1}^\theta \end{pmatrix} dx \right] \cdot \begin{pmatrix} \dot{T}_k \\ \dot{T}_{k+1} \end{pmatrix}$$

Finally, we can define the vectors

$$\vec{N}^T = \begin{pmatrix} N_k^\theta(x) \\ N_{k+1}^\theta(x) \end{pmatrix}$$

and

$$\vec{T}^e = \begin{pmatrix} T_k \\ T_{k+1} \end{pmatrix} \quad \dot{\vec{T}}^e = \begin{pmatrix} \dot{T}_k \\ \dot{T}_{k+1} \end{pmatrix}$$

so that

$$\begin{pmatrix} \Lambda_{N_k^\theta}^e \\ \Lambda_{N_{k+1}^\theta}^e \end{pmatrix} = \left(\int_{x_k}^{x_{k+1}} \vec{N}^T \rho C_p \vec{N} dx \right) \cdot \dot{\vec{T}}^e$$

Back to the diffusion term:

$$\begin{aligned} \Upsilon_f^e &= \int_{x_k}^{x^{k+1}} \frac{\partial f}{\partial x} k \frac{\partial T}{\partial x} dx \\ &= \int_{x_k}^{x^{k+1}} \frac{\partial f}{\partial x} k \frac{\partial (N_k^\theta(x) T_k + N_{k+1}^\theta(x) T_{k+1})}{\partial x} dx \\ &= \left(\int_{x_k}^{x^{k+1}} \frac{\partial f}{\partial x} k \frac{\partial N_k^\theta}{\partial x} dx \right) T_k + \left(\int_{x_k}^{x^{k+1}} \frac{\partial f}{\partial x} k \frac{\partial N_{k+1}^\theta}{\partial x} dx \right) T_{k+1} \end{aligned}$$

Taking $f(x) = N_k^\theta(x)$

$$\Upsilon_{N_k^\theta}^e = \left(\int_{x_k}^{x^{k+1}} k \frac{\partial N_k^\theta}{\partial x} \frac{\partial N_k^\theta}{\partial x} dx \right) T_k + \left(\int_{x_k}^{x^{k+1}} k \frac{\partial N_k^\theta}{\partial x} \frac{\partial N_{k+1}^\theta}{\partial x} dx \right) T_{k+1}$$

Taking $f(x) = N_{k+1}^\theta(x)$

$$\begin{aligned} \Upsilon_{N_{k+1}^\theta}^e &= \left(\int_{x_k}^{x^{k+1}} k \frac{\partial N_{k+1}^\theta}{\partial x} \frac{\partial N_k^\theta}{\partial x} dx \right) T_k + \left(\int_{x_k}^{x^{k+1}} k \frac{\partial N_{k+1}^\theta}{\partial x} \frac{\partial N_{k+1}^\theta}{\partial x} dx \right) T_{k+1} \\ \begin{pmatrix} \Upsilon_{N_k^\theta}^e \\ \Upsilon_{N_{k+1}^\theta}^e \end{pmatrix} &= \begin{pmatrix} \int_{x_k}^{x^{k+1}} \frac{\partial N_k^\theta}{\partial x} k \frac{\partial N_k^\theta}{\partial x} dx & \int_{x_k}^{x^{k+1}} \frac{\partial N_k^\theta}{\partial x} k \frac{\partial N_{k+1}^\theta}{\partial x} dx \\ \int_{x_k}^{x^{k+1}} \frac{\partial N_{k+1}^\theta}{\partial x} k \frac{\partial N_k^\theta}{\partial x} dx & \int_{x_k}^{x^{k+1}} \frac{\partial N_{k+1}^\theta}{\partial x} k \frac{\partial N_{k+1}^\theta}{\partial x} dx \end{pmatrix} \cdot \begin{pmatrix} T_k \\ T_{k+1} \end{pmatrix} \end{aligned}$$

or,

$$\begin{pmatrix} \Upsilon_{N_k^\theta}^e \\ \Upsilon_{N_{k+1}^\theta}^e \end{pmatrix} = \left[\int_{x_k}^{x^{k+1}} k \begin{pmatrix} \frac{\partial N_k^\theta}{\partial x} \frac{\partial N_k^\theta}{\partial x} & \frac{\partial N_k^\theta}{\partial x} \frac{\partial N_{k+1}^\theta}{\partial x} \\ \frac{\partial N_{k+1}^\theta}{\partial x} \frac{\partial N_k^\theta}{\partial x} & \frac{\partial N_{k+1}^\theta}{\partial x} \frac{\partial N_{k+1}^\theta}{\partial x} \end{pmatrix} dx \right] \cdot \begin{pmatrix} T_k \\ T_{k+1} \end{pmatrix}$$

Finally, we can define the vector

$$\vec{B}^T = \begin{pmatrix} \frac{\partial N_k^\theta}{\partial x} \\ \frac{\partial N_{k+1}^\theta}{\partial x} \end{pmatrix}$$

so that

$$\begin{pmatrix} \Upsilon_{N_k^\theta}^e \\ \Upsilon_{N_{k+1}^\theta}^e \end{pmatrix} = \left(\int_{x_k}^{x_{k+1}} \vec{B}^T k \vec{B} dx \right) \cdot \vec{T}^e$$

The weak form discretised over 1 element becomes

$$\underbrace{\left(\int_{x_k}^{x_{k+1}} \vec{N}^T \rho C_p \vec{N} dx \right) \cdot \dot{\vec{T}}^e}_{\mathbf{M}^e} + \underbrace{\left(\int_{x_k}^{x_{k+1}} \vec{B}^T k \vec{B} dx \right) \cdot \vec{T}^e}_{\mathbf{K}_d^e} = 0$$

or,

$$\boxed{\mathbf{M}^e \cdot \dot{\vec{T}}^e + \mathbf{K}_d^e \cdot \vec{T}^e = 0}$$

or,

$$\boxed{\mathbf{M}^e \cdot \frac{\partial \vec{T}^e}{\partial t} + \mathbf{K}_d^e \cdot \vec{T}^e = 0}$$

\mathbf{M}^e is commonly called the mass matrix, or capacitance matrix [1551, p103].

Using a backward first order in time discretisation for the time derivative:

$$\dot{\vec{T}} = \frac{\partial \vec{T}}{\partial t} = \frac{\vec{T}^{new} - \vec{T}^{old}}{\delta t}$$

we get

$$\mathbf{M}^e \cdot \frac{\vec{T}^{new} - \vec{T}^{old}}{\delta t} + \mathbf{K}_d^e \cdot \vec{T}^{new} = 0$$

or,

$$\boxed{(\mathbf{M}^e + \mathbf{K}_d^e \delta t) \cdot \vec{T}^{new} = \mathbf{M}^e \cdot \vec{T}^{old}}$$

with

$$\mathbf{M}^e = \int_{x_k}^{x_{k+1}} \vec{N}^T \rho C_p \vec{N} dx \quad \mathbf{K}_d^e = \int_{x_k}^{x_{k+1}} \vec{B}^T k \vec{B} dx$$

Let us compute \mathbf{M} for an element:

$$\mathbf{M}^e = \int_{x_k}^{x_{k+1}} \vec{N}^T \rho C_p \vec{N} dx$$

with

$$\vec{N}^T = \begin{pmatrix} N_k(x) \\ N_{k+1}(x) \end{pmatrix} = \begin{pmatrix} \frac{x_{k+1}-x}{h_x} \\ \frac{x-x_k}{h_x} \end{pmatrix}$$

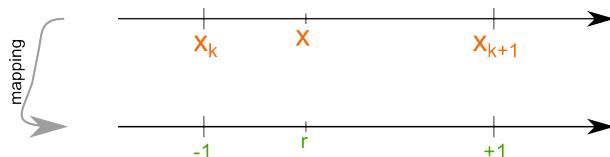
Then

$$\mathbf{M}^e = \begin{pmatrix} M_{11} & M_{12} \\ M_{21} & M_{22} \end{pmatrix} = \begin{pmatrix} \int_{x_k}^{x_{k+1}} \rho C_p N_k^\theta N_k^\theta dx & \int_{x_k}^{x_{k+1}} \rho C_p N_k^\theta N_{k+1}^\theta dx \\ \int_{x_k}^{x_{k+1}} \rho C_p N_{k+1}^\theta N_k^\theta dx & \int_{x_k}^{x_{k+1}} \rho C_p N_{k+1}^\theta N_{k+1}^\theta dx \end{pmatrix}$$

I only need to compute 3 integrals since $M_{12} = M_{21}$. Let us start with M_{11} :

$$M_{11} = \int_{x_k}^{x_{k+1}} \rho C_p N_k^\theta(x) N_k^\theta(x) dx = \int_{x_k}^{x_{k+1}} \rho C_p \frac{x_{k+1}-x}{h_x} \frac{x_{k+1}-x}{h_x} dx$$

It is then customary to carry out the change of variable $x \rightarrow r$ where $r \in [-1 : 1]$ as shown hereunder:



The relationships between x and r are:

$$r = \frac{2}{h_x}(x - x_k) - 1 \quad x = \frac{h_x}{2}(1 + r) + x_k$$

In what follows we assume for simplicity that ρ and C_p are constant within each element.

$$M_{11} = \rho C_p \int_{x_k}^{x_{k+1}} \frac{x_{k+1} - x}{h_x} \frac{x_{k+1} - x}{h_x} dx = \frac{\rho C_p h_x}{8} \int_{-1}^{+1} (1 - r)(1 - r) dr = \frac{h_x}{3} \rho C_p$$

Similarly we arrive at

$$M_{12} = \rho C_p \int_{x_k}^{x_{k+1}} \frac{x_{k+1} - x}{h_x} \frac{x - x_k}{h_x} dx = \frac{\rho C_p h_x}{8} \int_{-1}^{+1} (1 - r)(1 + r) dr = \frac{h_x}{6} \rho C_p$$

and

$$M_{22} = \rho C_p \int_{x_k}^{x_{k+1}} \frac{x - x_k}{h_x} \frac{x - x_k}{h_x} dx = \frac{\rho C_p h_x}{8} \int_{-1}^{+1} (1 + r)(1 + r) dr = \frac{h_x}{3} \rho C_p$$

Finally

$$\boxed{\mathbf{M}^e = \frac{h_x}{3} \rho C_p \begin{pmatrix} 1 & 1/2 \\ 1/2 & 1 \end{pmatrix}}$$

In the new coordinate system, the shape functions

$$N_k^\theta(x) = \frac{x_{k+1} - x}{h_x} \quad N_{k+1}^\theta(x) = \frac{x - x_k}{h_x}$$

become

$$N_k^\theta(r) = \frac{1}{2}(1 - r) \quad N_{k+1}^\theta(r) = \frac{1}{2}(1 + r)$$

Also,

$$\frac{\partial N_k^\theta}{\partial x} = -\frac{1}{h_x} \quad \frac{\partial N_{k+1}^\theta}{\partial x} = \frac{1}{h_x}$$

so that

$$\vec{B}^T = \begin{pmatrix} \frac{\partial N_k^\theta}{\partial x} \\ \frac{\partial N_{k+1}^\theta}{\partial x} \end{pmatrix} = \begin{pmatrix} -\frac{1}{h_x} \\ \frac{1}{h_x} \end{pmatrix}$$

We here also assume that k is constant within the element:

$$\mathbf{K}_d = \int_{x_k}^{x_{k+1}} \vec{B}^T k \vec{B} dx = k \int_{x_k}^{x_{k+1}} \vec{B}^T \vec{B} dx$$

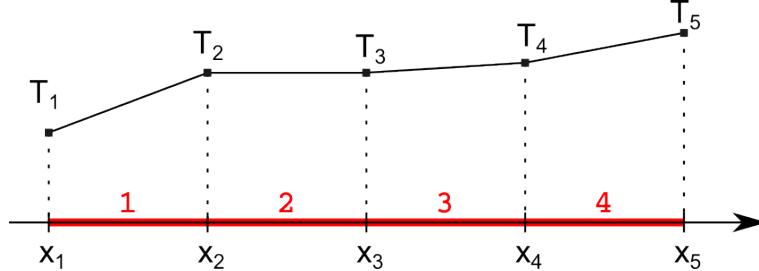
simply becomes

$$\mathbf{K}_d = k \int_{x_k}^{x_{k+1}} \frac{1}{h_x^2} \begin{pmatrix} 1 & -1 \\ -1 & 1 \end{pmatrix} dx$$

and then

$$\boxed{\mathbf{K}_d = \frac{k}{h_x} \begin{pmatrix} 1 & -1 \\ -1 & 1 \end{pmatrix}}$$

Let us consider this very simple grid consisting of 4 elements/5 nodes:



For each element we have

$$\underbrace{(\mathbf{M}^e + \mathbf{K}_d^e \delta t)}_{\mathbf{A}^e} \cdot \vec{\mathbf{T}}^{new} = \underbrace{\mathbf{M}^e \cdot \vec{\mathbf{T}}^{old}}_{\vec{\mathbf{b}}^e}$$

We can write this equation very explicitly for each element:

- element 1

$$\mathbf{A}^1 \cdot \begin{pmatrix} T_1 \\ T_2 \end{pmatrix} = \vec{\mathbf{b}}^1$$

$$\begin{cases} A_{11}^1 T_1 + A_{12}^1 T_2 = b_x^1 \\ A_{21}^1 T_1 + A_{22}^1 T_2 = b_y^1 \end{cases}$$

- element 2

$$\mathbf{A}^2 \cdot \begin{pmatrix} T_2 \\ T_3 \end{pmatrix} = \vec{\mathbf{b}}^2$$

$$\begin{cases} A_{11}^2 T_2 + A_{12}^2 T_3 = b_1^2 \\ A_{21}^2 T_2 + A_{22}^2 T_3 = b_2^2 \end{cases}$$

- element 3

$$\mathbf{A}^3 \cdot \begin{pmatrix} T_3 \\ T_4 \end{pmatrix} = \vec{\mathbf{b}}^3$$

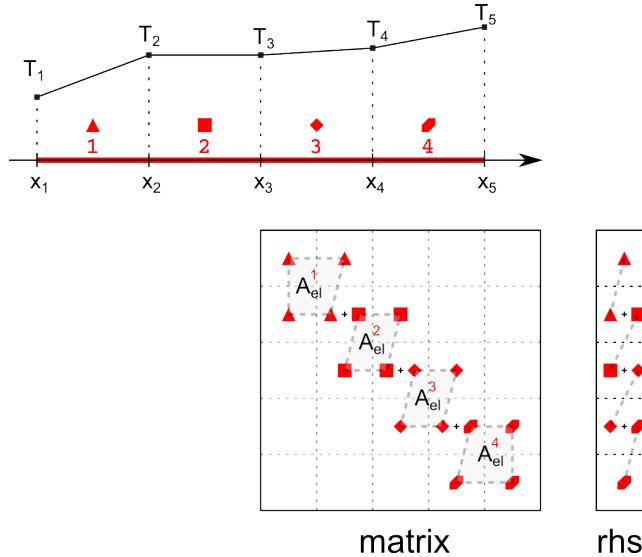
$$\begin{cases} A_{11}^3 T_3 + A_{12}^3 T_4 = b_1^3 \\ A_{21}^3 T_3 + A_{22}^3 T_4 = b_2^3 \end{cases}$$

- element 4

$$\mathbf{A}^4 \cdot \begin{pmatrix} T_4 \\ T_5 \end{pmatrix} = \vec{\mathbf{b}}^4$$

$$\begin{cases} A_{11}^4 T_4 + A_{12}^4 T_5 = b_1^4 \\ A_{21}^4 T_4 + A_{22}^4 T_5 = b_2^4 \end{cases}$$

All equations can be cast into a single linear system: this is the **assembly** phase. The process can also be visualised as shown hereunder. Because nodes 2,3,4 belong to two elements elemental contributions will be summed in the matrix and the rhs:



The assembled matrix and rhs are then:

$$\begin{pmatrix} A_{11}^1 & A_{12}^1 & 0 & 0 & 0 \\ A_{21}^1 & A_{22}^1 + A_{11}^2 & A_{12}^2 & 0 & 0 \\ 0 & A_{21}^2 & A_{22}^2 + A_{11}^3 & A_{12}^3 & 0 \\ 0 & 0 & A_{21}^3 & A_{22}^3 + A_{11}^4 & A_{12}^4 \\ 0 & 0 & 0 & A_{21}^4 & A_{22}^4 \end{pmatrix} \begin{pmatrix} T_1 \\ T_2 \\ T_3 \\ T_4 \\ T_5 \end{pmatrix} = \begin{pmatrix} b_1^1 \\ b_2^1 + b_1^2 \\ b_2^2 + b_1^3 \\ b_2^3 + b_1^4 \\ b_2^4 \end{pmatrix}$$

Ultimately the assembled matrix system also takes the form

$$\begin{pmatrix} A_{11} & A_{12} & 0 & 0 & 0 \\ A_{21} & A_{22} & A_{23} & 0 & 0 \\ 0 & A_{32} & A_{33} & A_{34} & 0 \\ 0 & 0 & A_{43} & A_{44} & A_{45} \\ 0 & 0 & 0 & A_{54} & A_{55} \end{pmatrix} \begin{pmatrix} T_1 \\ T_2 \\ T_3 \\ T_4 \\ T_5 \end{pmatrix} = \begin{pmatrix} b_1 \\ b_2 \\ b_3 \\ b_4 \\ b_5 \end{pmatrix}$$

and we see that it is sparse. Its sparsity structure is easy to derive: each row corresponds to a dof, and since nodes 1 and 2 'see' each other (they belong to the same element) there will be non-zero entries in the first and second column. Likewise, node 2 'sees' node 1 (in other words, there is an edge linking nodes 1 and 2), itself, and node 3, so that there are non-zero entries in the second row at columns 1, 2, and 3.

Before we solve the system, we need to take care of boundary conditions. Let us assume that we wish to fix the temperature at node 2, or in other words we wish to set

$$T_2 = T^{bc}$$

This equation can be cast as

$$(0 \ 1 \ 0 \ 0 \ 0) \begin{pmatrix} T_1 \\ T_2 \\ T_3 \\ T_4 \\ T_5 \end{pmatrix} = \begin{pmatrix} 0 \\ T^{bc} \\ 0 \\ 0 \\ 0 \end{pmatrix}$$

This replaces the second line in the previous matrix equation:

$$\begin{pmatrix} A_{11} & A_{12} & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 \\ 0 & A_{32} & A_{33} & A_{34} & 0 \\ 0 & 0 & A_{43} & A_{44} & A_{45} \\ 0 & 0 & 0 & A_{54} & A_{55} \end{pmatrix} \begin{pmatrix} T_1 \\ T_2 \\ T_3 \\ T_4 \\ T_5 \end{pmatrix} = \begin{pmatrix} b_1 \\ T^{bc} \\ b_3 \\ b_4 \\ b_5 \end{pmatrix}$$

That's it, we have a linear system of equations which can be solved!

5.2 The advection-diffusion equation in 1D

We start with the 1D advection-diffusion equation

$$\rho C_p \left(\frac{\partial T}{\partial t} + u \frac{\partial T}{\partial x} \right) = \frac{\partial}{\partial x} \left(k \frac{\partial T}{\partial x} \right) + H \quad (269)$$

This is the **strong form** of the ODE to solve. As in the previous section, I multiply this equation by a function $f(x)$ and integrate it over the domain Ω :

$$\int_{\Omega} f(x) \rho C_p \frac{\partial T}{\partial t} dx + \int_{\Omega} f(x) \rho C_p u \frac{\partial T}{\partial x} dx = \int_{\Omega} f(x) \frac{\partial}{\partial x} \left(k \frac{\partial T}{\partial x} \right) dx + \int_{\Omega} f(x) H dx$$

As in the previous section I integrate the r.h.s. by parts:

$$\int_{\Omega} f(x) \frac{\partial}{\partial x} \left(k \frac{\partial T}{\partial x} \right) dx = \left[f(x) k \frac{\partial T}{\partial x} \right]_{\partial\Omega} - \int_{\Omega} \frac{\partial f}{\partial x} k \frac{\partial T}{\partial x} dx$$

Disregarding the boundary term for now, we then obtain the **weak form** of the diffusion equation in 1D:

$$\boxed{\int_{\Omega} f(x) \rho C_p \frac{\partial T}{\partial t} dx + \int_{\Omega} f(x) \rho C_p u \frac{\partial T}{\partial x} dx + \int_{\Omega} \frac{\partial f}{\partial x} k \frac{\partial T}{\partial x} dx = \int_{\Omega} f(x) H dx}$$

We then use the additive property of the integral $\int_{\Omega} \dots = \sum_{elts} \int_{\Omega_e} \dots$

$$\sum_{elts} \left(\underbrace{\int_{\Omega_e} f(x) \rho C_p \frac{\partial T}{\partial t} dx}_{\Lambda_f^e} + \underbrace{\int_{\Omega_e} f(x) \rho C_p u \frac{\partial T}{\partial x} dx}_{\Sigma_f^e} + \underbrace{\int_{\Omega_e} \frac{\partial f}{\partial x} k \frac{\partial T}{\partial x} dx}_{\Upsilon_f^e} - \underbrace{\int_{\Omega_e} f(x) H dx}_{\Omega_f^e} \right) = 0$$

In the element, we have seen that the temperature can be written:

$$T(x) = N_k^\theta(x) T_k + N_{k+1}^\theta(x) T_{k+1}$$

In the previous presentation we have computed Λ_f^e and Υ_f^e . Let us now turn to Σ_f^e and Ω_f^e .

$$\begin{aligned} \Sigma_f^e &= \int_{x_k}^{x_{k+1}} f(x) \rho C_p u \frac{\partial T}{\partial x} dx \\ &= \int_{x_k}^{x_{k+1}} f(x) \rho C_p u \frac{\partial [N_k^\theta(x) T_k + N_{k+1}^\theta(x) T_{k+1}]}{\partial x} dx \\ &= \int_{x_k}^{x_{k+1}} f(x) \rho C_p u \frac{\partial N_k^\theta}{\partial x} T_k dx + \int_{x_k}^{x_{k+1}} f(x) \rho C_p u \frac{\partial N_{k+1}^\theta}{\partial x} T_{k+1} dx \\ &= \left(\int_{x_k}^{x_{k+1}} f(x) \rho C_p u \frac{\partial N_k^\theta}{\partial x} dx \right) T_k + \left(\int_{x_k}^{x_{k+1}} f(x) \rho C_p u \frac{\partial N_{k+1}^\theta}{\partial x} dx \right) T_{k+1} \end{aligned}$$

Taking $f(x) = N_k^\theta(x)$ and omitting '(x)' in the rhs:

$$\Sigma_{N_k^\theta}^e = \left(\int_{x_k}^{x_{k+1}} \rho C_p u N_k^\theta \frac{\partial N_k^\theta}{\partial x} dx \right) T_k + \left(\int_{x_k}^{x_{k+1}} \rho C_p u N_{k+1}^\theta \frac{\partial N_{k+1}^\theta}{\partial x} dx \right) T_{k+1}$$

Taking $f(x) = N_{k+1}^\theta(x)$ and omitting '(x)' in the rhs:

$$\Sigma_{N_{k+1}^\theta}^e = \left(\int_{x_k}^{x_{k+1}} \rho C_p u N_{k+1}^\theta \frac{\partial N_k^\theta}{\partial x} dx \right) T_k + \left(\int_{x_k}^{x_{k+1}} \rho C_p u N_{k+1}^\theta \frac{\partial N_{k+1}^\theta}{\partial x} dx \right) T_{k+1}$$

$$\begin{pmatrix} \Sigma_{N_k^\theta} \\ \Sigma_{N_{k+1}^\theta} \end{pmatrix} = \begin{pmatrix} \int_{x_k}^{x_{k+1}} \rho C_p u N_k^\theta \frac{\partial N_k^\theta}{\partial x} dx & \int_{x_k}^{x_{k+1}} \rho C_p u N_k^\theta \frac{\partial N_{k+1}^\theta}{\partial x} dx \\ \int_{x_k}^{x_{k+1}} \rho C_p u N_{k+1}^\theta \frac{\partial N_k^\theta}{\partial x} dx & \int_{x_k}^{x_{k+1}} \rho C_p u N_{k+1}^\theta \frac{\partial N_{k+1}^\theta}{\partial x} dx \end{pmatrix} \cdot \begin{pmatrix} T_k \\ T_{k+1} \end{pmatrix}$$

or,

$$\begin{pmatrix} \Sigma_{N_k^\theta} \\ \Sigma_{N_{k+1}^\theta} \end{pmatrix} = \left[\int_{x_k}^{x_{k+1}} \rho C_p u \begin{pmatrix} N_k^\theta \frac{\partial N_k^\theta}{\partial x} & N_k^\theta \frac{\partial N_{k+1}^\theta}{\partial x} \\ N_{k+1}^\theta \frac{\partial N_k^\theta}{\partial x} & N_{k+1}^\theta \frac{\partial N_{k+1}^\theta}{\partial x} \end{pmatrix} dx \right] \cdot \begin{pmatrix} T_k \\ T_{k+1} \end{pmatrix}$$

Finally, we have already defined the vectors

$$\vec{N}^T = \begin{pmatrix} N_k^\theta(x) \\ N_{k+1}^\theta(x) \end{pmatrix} \quad \vec{B}^T = \begin{pmatrix} \frac{\partial N_k^\theta}{\partial x} \\ \frac{\partial N_{k+1}^\theta}{\partial x} \end{pmatrix} \quad \vec{T}^e = \begin{pmatrix} T_k \\ T_{k+1} \end{pmatrix}$$

so that

$$\begin{pmatrix} \Sigma_{N_k^\theta} \\ \Sigma_{N_{k+1}^\theta} \end{pmatrix} = \left(\int_{x_k}^{x_{k+1}} \vec{N}^T \rho C_p u \vec{B} dx \right) \cdot \vec{T}^e = \mathbf{K}_a \cdot \vec{T}^e$$

One can easily show that

$$\mathbf{K}_a^e = \rho C_p u \begin{pmatrix} -1/2 & 1/2 \\ -1/2 & 1/2 \end{pmatrix}$$

Note that the matrix \mathbf{K}_a^e is not symmetric.

Let us now look at the source term:

$$\Omega_f^e = \int_{x_k}^{x_{k+1}} f(x) H(x) dx$$

Taking $f(x) = N_k^\theta(x)$:

$$\Omega_{N_k^\theta} = \int_{x_k}^{x_{k+1}} N_k^\theta(x) H(x) dx$$

Taking $f(x) = N_{k+1}^\theta(x)$:

$$\Omega_{N_{k+1}^\theta} = \int_{x_k}^{x_{k+1}} N_{k+1}^\theta(x) H(x) dx$$

We can rearrange both equations as follows:

$$\begin{pmatrix} \Omega_{N_k^\theta} \\ \Omega_{N_{k+1}^\theta} \end{pmatrix} = \begin{pmatrix} \int_{x_k}^{x_{k+1}} N_k^\theta(x) H(x) dx \\ \int_{x_k}^{x_{k+1}} N_{k+1}^\theta(x) H(x) dx \end{pmatrix}$$

or,

$$\begin{pmatrix} \Omega_{N_k^\theta} \\ \Omega_{N_{k+1}^\theta} \end{pmatrix} = \left[\int_{x_k}^{x_{k+1}} \begin{pmatrix} N_k^\theta(x) H(x) \\ N_{k+1}^\theta(x) H(x) \end{pmatrix} dx \right]$$

so that

$$\begin{pmatrix} \Omega_{N_k^\theta} \\ \Omega_{N_{k+1}^\theta} \end{pmatrix} = \left(\int_{x_k}^{x_{k+1}} \vec{N}^T H(x) dx \right)$$

The weak form discretised over 1 element becomes

$$\underbrace{\left(\int_{x_k}^{x_{k+1}} \vec{N}^T \rho C_p \mathbf{N} dx \right)}_{\mathbf{M}^e} \cdot \dot{\vec{T}}^e + \underbrace{\left(\int_{x_k}^{x_{k+1}} \vec{N}^T \rho C_p u \mathbf{B} dx \right)}_{\mathbf{K}_a^e} \cdot \vec{T}^e + \underbrace{\left(\int_{x_k}^{x_{k+1}} \vec{B}^T k \mathbf{B} dx \right)}_{\mathbf{K}_d^e} \cdot \vec{T}^e = \underbrace{\left(\int_{x_k}^{x_{k+1}} \vec{N}^T H(x) dx \right)}_{\vec{F}^e}$$

or,

$$\mathbf{M}^e \cdot \dot{\vec{T}}^e + (\mathbf{K}_a^e + \mathbf{K}_d^e) \cdot \vec{T}^e = \vec{F}^e$$

or,

$$\mathbf{M}^e \cdot \frac{\partial \vec{T}^e}{\partial t} + (\mathbf{K}_a^e + \mathbf{K}_d^e) \cdot \vec{T}^e = \vec{F}^e$$

5.3 The advection-diffusion equation in 2D

We start from the 'bare-bones' heat transport equation (source terms are omitted):

$$\rho C_p \left(\frac{\partial T}{\partial t} + \vec{v} \cdot \vec{\nabla} T \right) = \vec{\nabla} \cdot k \vec{\nabla} T \quad (270)$$

In what follows we assume that the velocity field \vec{v} is known so that temperature is the only unknown. Let N^θ be the temperature basis functions so that the temperature inside an element is given by²⁰:

$$T^h(\vec{r}) = \sum_{i=1}^{m_T} N_i^\theta(\vec{r}) T_i = \vec{N}^\theta \cdot \vec{T} \quad (271)$$

where \vec{T} is a vector of length m_T . The weak form is then

$$\begin{aligned} \int_{\Omega} N_i^\theta \left[\rho C_p \left(\frac{\partial T}{\partial t} + \vec{v} \cdot \vec{\nabla} T \right) \right] d\Omega &= \int_{\Omega} N_i^\theta \vec{\nabla} \cdot k \vec{\nabla} T d\Omega \\ \underbrace{\int_{\Omega} N_i^\theta \rho C_p \frac{\partial T}{\partial t} d\Omega}_{I} + \underbrace{\int_{\Omega} N_i^\theta \rho C_p \vec{v} \cdot \vec{\nabla} T d\Omega}_{II} &= \underbrace{\int_{\Omega} N_i^\theta \vec{\nabla} \cdot k \vec{\nabla} T d\Omega}_{III} \quad i = 1, m_T \end{aligned} \quad (272)$$

Looking at the first term:

$$\int_{\Omega} N_i^\theta \rho C_p \frac{\partial T}{\partial t} d\Omega = \int_{\Omega} N_i^\theta \rho C_p \vec{N}^\theta \cdot \dot{\vec{T}} d\Omega \quad (273)$$

$$(274)$$

so that when we assemble all contributions for $i = 1, m_T$ we get:

$$I = \int_{\Omega} \vec{N}^\theta \rho C_p \vec{N}^\theta \cdot \dot{\vec{T}} d\Omega = \left(\int_{\Omega} \rho C_p \vec{N}^\theta \vec{N}^\theta d\Omega \right) \cdot \dot{\vec{T}} = \mathbf{M}^T \cdot \dot{\vec{T}}$$

where \mathbf{M}^T is the mass matrix of the system of size $(m_T \times m_T)$ with

$$M_{ij}^T = \int_{\Omega} \rho C_p N_i^\theta N_j^\theta d\Omega$$

Turning now to the second term:

$$\int_{\Omega} N_i^\theta \rho C_p \vec{v} \cdot \vec{\nabla} T d\Omega = \int_{\Omega} N_i^\theta \rho C_p (u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial y}) d\Omega \quad (275)$$

$$= \int_{\Omega} N_i^\theta \rho C_p (u \frac{\partial \vec{N}^\theta}{\partial x} + v \frac{\partial \vec{N}^\theta}{\partial y}) \cdot \vec{T} d\Omega \quad (276)$$

$$(277)$$

²⁰the θ superscript has been chosen to denote temperature so as to avoid confusion with the transpose operator

so that when we assemble all contributions for $i = 1, m_T$ we get:

$$II = \left(\int_{\Omega} \rho C_p \vec{N}^{\theta} (u \frac{\partial \vec{N}^{\theta}}{\partial x} + v \frac{\partial \vec{N}^{\theta}}{\partial y}) d\Omega \right) \cdot \vec{T} = \mathbf{K}_a \cdot \vec{T}$$

where \mathbf{K}_a is the advection term matrix of size $(m_T \times m_T)$ with

$$(K_a)_{ij} = \int_{\Omega} \rho C_p N_i^{\theta} \left(u \frac{\partial N_j^{\theta}}{\partial x} + v \frac{\partial N_j^{\theta}}{\partial y} \right) d\Omega$$

Now looking at the third term, we carry out an integration by part and neglect the surface term for now, so that

$$\int_{\Omega} N_i^{\theta} \vec{\nabla} \cdot k \vec{\nabla} T d\Omega = - \int_{\Omega} k \vec{\nabla} N_i^{\theta} \cdot \vec{\nabla} T d\Omega \quad (278)$$

$$= - \int_{\Omega} k \vec{\nabla} N_i^{\theta} \cdot \vec{\nabla} (\vec{N}^{\theta} \cdot \vec{T}) d\Omega \quad (279)$$

(280)

with

$$\vec{\nabla} \vec{N}^{\theta} = \begin{pmatrix} \partial_x N_1^{\theta} & \partial_x N_2^{\theta} & \dots & \partial_x N_{m_T}^{\theta} \\ \partial_y N_1^{\theta} & \partial_y N_2^{\theta} & \dots & \partial_y N_{m_T}^{\theta} \end{pmatrix}$$

so that finally:

$$III = - \left(\int_{\Omega} k (\vec{\nabla} \vec{N}^{\theta})^T \cdot \vec{\nabla} \vec{N}^{\theta} d\Omega \right) \cdot \vec{T} = - \mathbf{K}_d \cdot \vec{T}$$

where \mathbf{K}_d is the diffusion term matrix:

$$\mathbf{K}_d = \int_{\Omega} k (\vec{\nabla} \vec{N}^{\theta})^T \cdot \vec{\nabla} \vec{N}^{\theta} d\Omega$$

Ultimately terms I, II, III together yield:

$$\boxed{\mathbf{M}^{\theta} \cdot \dot{\vec{T}} + (\mathbf{K}_a + \mathbf{K}_d) \cdot \vec{T} = \vec{0}}$$

add source term!!

5.3.1 Dealing with the time discretisation

Essentially we have to solve a PDE of the type:

$$\frac{\partial T}{\partial t} = \mathcal{F}(\vec{v}, T, \vec{\nabla} T, \Delta T)$$

with $\mathcal{F} = \frac{1}{\rho C_p} (-\vec{v} \cdot \vec{\nabla} T + \vec{\nabla} \cdot k \vec{\nabla} T)$.

The (explicit) forward Euler method is:

$$\frac{T^{n+1} - T^n}{\delta t} = \mathcal{F}^n(T, \vec{\nabla} T, \Delta T)$$

The (implicit) backward Euler method is:

$$\frac{T^{n+1} - T^n}{\delta t} = \mathcal{F}^{n+1}(T, \vec{\nabla} T, \Delta T)$$

and the (implicit) Crank-Nicolson algorithm is:

$$\frac{T^{n+1} - T^n}{\delta t} = \frac{1}{2} [\mathcal{F}^n(T, \vec{\nabla} T, \Delta T) + \mathcal{F}^{n+1}(T, \vec{\nabla} T, \Delta T)]$$

where the superscript n indicates the time step. The Crank-Nicolson is obviously based on the trapezoidal rule, with second-order convergence in time.

In what follows, I omit the superscript on the mass matrix to simplify notations: $\mathbf{M}^{\theta} = \mathbf{M}$. In terms of Finite Elements, these become:

- Explicit Forward euler:

$$\frac{1}{\delta t}(\mathbf{M}^{n+1} \cdot \vec{T}^{n+1} - \mathbf{M}^n \cdot \vec{T}^n) = -(\mathbf{K}_a^n + \mathbf{K}_d^n) \cdot \vec{T}^n$$

or,

$$\boxed{\mathbf{M}^{n+1} \cdot \vec{T}^{n+1} = (\mathbf{M}^n + (\mathbf{K}_a^n + \mathbf{K}_d^n)\delta t) \cdot \vec{T}^n}$$

- Implicit Backward euler:

$$\frac{1}{\delta t}(\mathbf{M}^{n+1} \cdot \vec{T}^{n+1} - \mathbf{M}^n \cdot \vec{T}^n) = -(\mathbf{K}_a^{n+1} + \mathbf{K}_d^{n+1}) \cdot \vec{T}^{n+1}$$

or,

$$\boxed{(\mathbf{M}^{n+1} + (\mathbf{K}_a^{n+1} + \mathbf{K}_d^{n+1})\delta t) \cdot \vec{T}^{n+1} = \mathbf{M}^n \cdot \vec{T}^n}$$

- Crank-Nicolson

$$\frac{1}{\delta t}(\mathbf{M}^{n+1} \cdot \vec{T}^{n+1} - \mathbf{M}^n \cdot \vec{T}^n) = \frac{1}{2} [-(\mathbf{K}_a^{n+1} + \mathbf{K}_d^{n+1}) \cdot \vec{T}^{n+1} - (\mathbf{K}_a^n + \mathbf{K}_d^n) \cdot \vec{T}^n]$$

or,

$$\boxed{\left(\mathbf{M}^{n+1} + (\mathbf{K}_a^{n+1} + \mathbf{K}_d^{n+1}) \frac{\delta t}{2} \right) \cdot \vec{T}^{n+1} = \left(\mathbf{M}^n + (\mathbf{K}_a^n + \mathbf{K}_d^n) \frac{\delta t}{2} \right) \cdot \vec{T}^n}$$

Note that in benchmarks where the domain/grid does not deform, the coefficients do not change in space and the velocity field is constant in time, or in practice out of convenience, the \mathbf{K} and \mathbf{M} matrices do not change and the r.h.s. can be constructed with the same matrices as the FE matrix.

The Backward differentiation formula (see for instance [850] or Wikipedia²¹. The second-order BDF (or BDF-2) as shown in [1119] is as follows: it is a finite-difference quadratic interpolation approximation of the $\partial T / \partial t$ term which involves t^n , t^{n-1} and t^{n-2} :

$$\frac{\partial T}{\partial t}(t^n) = \frac{1}{\tau_n} \left(\frac{2\tau_n + \tau_{n-1}}{\tau_n + \tau_{n-1}} T(t^n) - \frac{\tau_n + \tau_{n-1}}{\tau_{n-1}} T(t^{n-1}) + \frac{\tau_n^2}{\tau_{n-1}(\tau_n + \tau_{n-1})} T(t^{n-2}) \right) \quad (281)$$

where $\tau_n = t^n - t^{n-1}$. Starting again from $\mathbf{M}^\theta \cdot \dot{\vec{T}} + (\mathbf{K}_a + \mathbf{K}_d) \cdot \vec{T} = \vec{0}$, we write

$$\mathbf{M}^\theta \cdot \frac{1}{\tau_n} \left(\frac{2\tau_n + \tau_{n-1}}{\tau_n + \tau_{n-1}} \vec{T}^n - \frac{\tau_n + \tau_{n-1}}{\tau_{n-1}} \vec{T}^{n-1} + \frac{\tau_n^2}{\tau_{n-1}(\tau_n + \tau_{n-1})} \vec{T}^{n-2} \right) + (\mathbf{K}_a + \mathbf{K}_d) \cdot \vec{T}^n = \vec{0}$$

and finally:

$$\left[\frac{2\tau_n + \tau_{n-1}}{\tau_n + \tau_{n-1}} \mathbf{M}^\theta + \tau_n (\mathbf{K}_a + \mathbf{K}_d) \right] \cdot \vec{T}^n = \frac{\tau_n + \tau_{n-1}}{\tau_{n-1}} \mathbf{M}^\theta \cdot \vec{T}^{n-1} - \frac{\tau_n^2}{\tau_{n-1}(\tau_n + \tau_{n-1})} \mathbf{M}^\theta \cdot \vec{T}^{n-2}$$

Note that if all timesteps are equal, i.e. $\tau_n = \tau_{n-1} = \delta t$, this equation becomes:

$$\left[\frac{3}{2} \mathbf{M}^\theta + \delta t (\mathbf{K}_a + \mathbf{K}_d) \right] \cdot \vec{T}^n = \mathbf{M}^\theta \cdot \left(2\vec{T}^{n-1} - \frac{1}{2}\vec{T}^{n-2} \right)$$

or,

$$\left[\mathbf{M}^\theta + \frac{2}{3} \delta t (\mathbf{K}_a + \mathbf{K}_d) \right] \cdot \vec{T}^n = \mathbf{M}^\theta \cdot \left(\frac{4}{3} \vec{T}^{n-1} - \frac{1}{3} \vec{T}^{n-2} \right)$$

As mentioned before the backward differentiation formula (BDF) is a family of implicit methods for the integration of ODEs. Each BDF- s method achieves order s . The BDF-1 is simply the backward Euler method as seen above:

$$T^{n+1} - T^n = \delta t \mathcal{F}^{n+1}$$

²¹https://en.wikipedia.org/wiki/Backward_differentiation_formula

The BDF-2 is given by

$$T^{n+2} - \frac{4}{3}T^{n+1} + \frac{1}{3}T^n = \frac{2}{3}\delta t\mathcal{F}^{n+2}$$

The BDF-3 is given by

$$T^{n+3} - \frac{18}{11}T^{n+2} + \frac{9}{11}T^{n+1} - \frac{2}{11}T^n = \frac{6}{11}\delta t\mathcal{F}^{n+3}$$

The BDF-4 is given by

$$T^{n+4} - \frac{48}{25}T^{n+1} + \frac{36}{25}T^{n+1} - \frac{16}{25}T^{n+1} + \frac{3}{25}T^n = \frac{12}{25}\delta t\mathcal{F}^{n+4}$$

5.3.2 On steady states

It is said that a system is in a steady state if the (state) variables which define the behavior of the system are unchanging in time. In continuous time, this means that the partial derivative with respect to time is zero and remains so:

$$\frac{\partial}{\partial t} = 0 \quad \forall t$$

This is irrelevant for the Stokes equations which do not contain an explicit time dependence but the heat transport equation can reach a steady state. Note that if one is only interested in the steady state solution (and not how the system gets there in time) then the heat transport equation should be solved with $\partial T / \partial t$ set to zero.

5.3.3 Anisotropic heat conduction

It is most often assumed that the heat conductivity is isotropic so that one speaks of heat conductivity as a scalar k . However many materials are orthotropic and in that case the heat conductivity is a tensor \mathbf{k} which (in 2D) writes [1551, p121]:

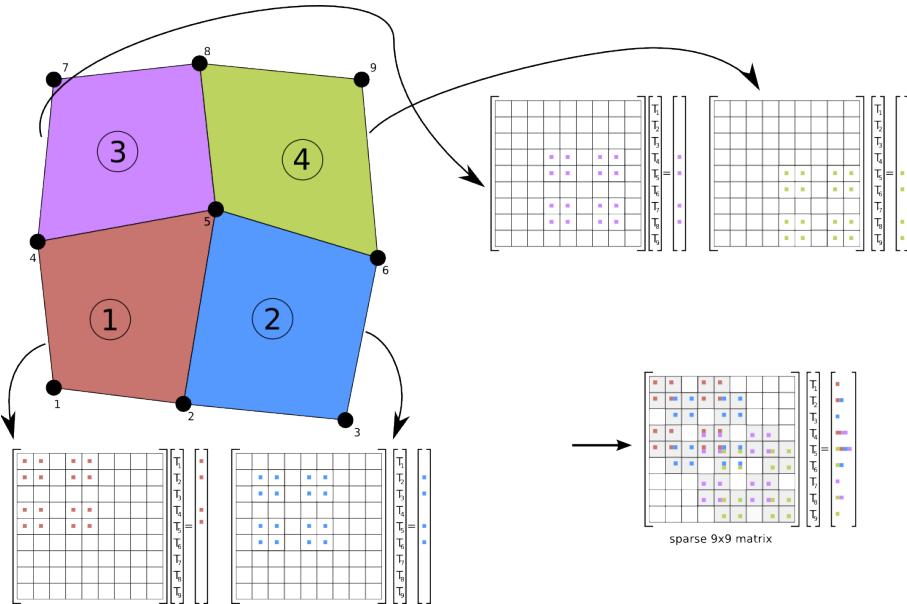
$$\mathbf{k} = \begin{pmatrix} k_{xx} & k_{xy} \\ k_{yx} & k_{yy} \end{pmatrix} = \begin{pmatrix} \cos \theta & \sin \theta \\ -\sin \theta & \cos \theta \end{pmatrix} \cdot \begin{pmatrix} k_1 & 0 \\ 0 & k_2 \end{pmatrix} \cdot \begin{pmatrix} \cos \theta & -\sin \theta \\ \sin \theta & \cos \theta \end{pmatrix}$$

where k_1 and k_2 are the conductivities in the principal axes system and θ is the local orientation. In that case the diffusion term in the heat trasport equation becomes $\vec{\nabla} \cdot (\mathbf{k} \cdot \vec{\nabla} T)$.

(MSc Thesis): [1551, p121], [1551, p143]

5.3.4 About the assembly

Let us consider for simplicity the following grid composed of 9 nodes and 4 Q_1 elements. Each node carries a single degree of freedom.



There are four elements:

- element 1 is composed of nodes $(1, 2, 5, 4) = \vec{T}^{el1}$
- element 2 is composed of nodes $(2, 3, 6, 5) = \vec{T}^{el2}$
- element 3 is composed of nodes $(4, 5, 8, 7) = \vec{T}^{el3}$
- element 4 is composed of nodes $(5, 6, 9, 8) = \vec{T}^{el4}$

For each element one has computed an elemental matrix \mathbf{A}^{el} and a right hand side \vec{b}^{el} .

$$\mathbf{A}^{el1} \cdot \mathbf{T}^{el1} = \mathbf{b}^{el1}$$

$$\mathbf{A}^{el2} \cdot \mathbf{T}^{el2} = \mathbf{b}^{el2}$$

$$\mathbf{A}^{el3} \cdot \mathbf{T}^{el3} = \mathbf{b}^{el3}$$

$$\mathbf{A}^{el4} \cdot \mathbf{T}^{el4} = \mathbf{b}^{el4}$$

As seen in the 1D case, these four linear systems must be assembled in a single large matrix of size 9×9 as shown in the figure above.

6 Solving the flow equations with the FEM

In the case of an incompressible flow, we have seen that the continuity (mass conservation) equation takes the simple form $\vec{\nabla} \cdot \vec{v} = 0$. In other word flow takes place under the constraint that the divergence of its velocity field is exactly zero everywhere (solenoidal constraint), i.e. it is divergence free.

We see that the pressure in the momentum equation is then a degree of freedom which is needed to satisfy the incompressibility constraint (and it is not related to any constitutive equation) [533]. In other words the pressure is acting as a Lagrange multiplier of the incompressibility constraint.

Various approaches have been proposed in the literature to deal with the incompressibility constraint but we will only focus on the penalty method (section 6.3) and the so-called mixed finite element method 6.4.

6.1 Strong and weak forms

The strong form consists of the governing equation and the boundary conditions, i.e. the mass, momentum and energy conservation equations supplemented with Dirichlet and/or Neumann boundary conditions on (parts of) the boundary.

To develop the finite element formulation, the partial differential equations must be restated in an integral form called the weak form. In essence the PDEs are first multiplied by an arbitrary function and integrated over the domain.

6.2 Which velocity-pressure pair for Stokes?

The success of a mixed finite element formulation crucially depends on a proper choice of the local interpolations of the velocity and the pressure.

6.2.1 The compatibility condition (or LBB condition)

'LBB stable' elements assure the existence of a unique solution and assure convergence at the optimal rate.

6.2.2 Families

The family of Taylor-Hood finite element spaces on triangular/tetrahedral grids is given by $P_k \times P_{k-1}$ with $k \geq 2$, and on quadrilateral/hexahedral grids by $Q_k \times Q_{k-1}$ with $k \geq 2$. This means that the pressure is then approximated by continuous functions.

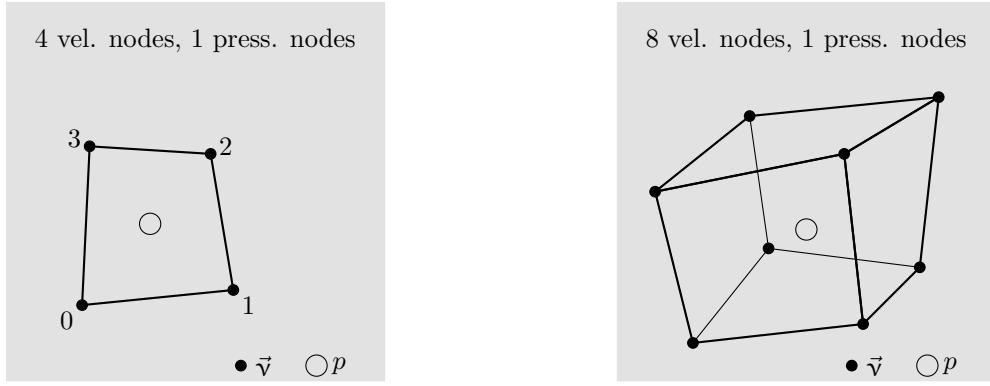
These finite elements are very popular, in particular the pairs for $k = 2$, i.e. $Q_2 \times Q_1$ and $P_2 \times P_1$. The reason why $k \geq 2$ comes from the fact that the $Q_1 \times Q_0$ (i.e. $Q_1 \times P_0$) and $P_2 \times P_1$ are not stable elements (they are not inf-sup stable).

Remark. Note that a similar element to $Q_2 \times Q_1$ has been proposed and used successfully used [1744, 920]: it is denoted by $Q_2^{(8)} \times Q_1$ since the center node (' x^2y^2 ') and its associated degrees of freedom have been removed. It has also been proved to be LBB stable.

The Raviart-Thomas family on triangles and quadrilaterals.

find literature

6.2.3 The bi/tri-linear velocity - constant pressure element ($Q_1 \times P_0$)



discussed in example 3.71 of [998]

However simple it may look, the element is one of the hardest elements to analyze and many questions are still open about its properties. The element does not satisfy the inf-sup condition [941]p211. In [813] it is qualified as follows: slightly unstable but highly usable.

The $Q_1 \times P_0$ mixed approximation is the lowest order conforming approximation method defined on a rectangular grid. It also happens to be the most famous example of an unstable mixed approximation method. [573, p235].

This element is discussed in [634], [636] and in [1482] in the context of multigrid use.

This element is plagued by so-called pressure checkerboard modes which have been thoroughly analysed [817], [350], [1622, 1623]. These can be filtered out [350]. Smoothing techniques are also discussed in [1147].

Literature[635]

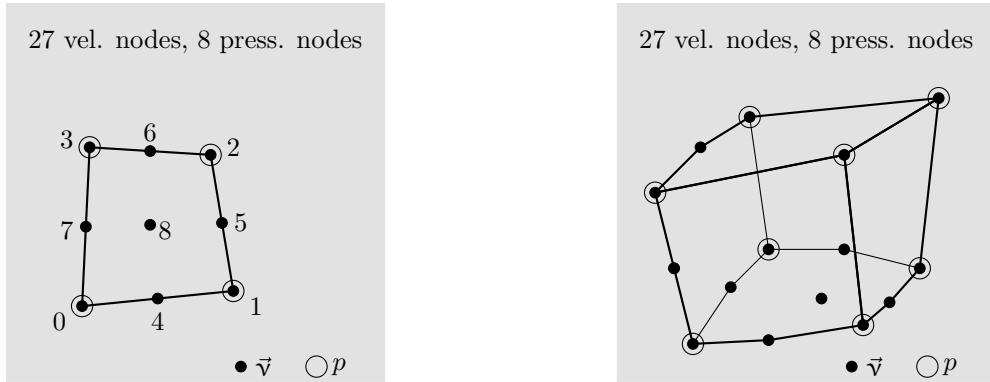
6.2.4 The bi/tri-quadratic velocity - discontinuous linear pressure element ($Q_2 \times P_{-1}$)

This element is crowned "probably the most accurate 2D element" in [813].

Piecewise Biquadratic velocities, piecewise linear discontinuous polynomial pressure. The element satisfies the inf-sup condition [941]p211. It is used in [1818]. See [183] over the two possible choices for the definition of the pressure space.. It is mentioned in [1030], [183], [1456]. It is used in [658] to study 3D fold growth rates (see online supplementary material) and in [1636].

Note that the serendipity version of this pair, i.e. $Q_2^{(20)} \times P_{-1}$ is also LBB stable [1551, p180].

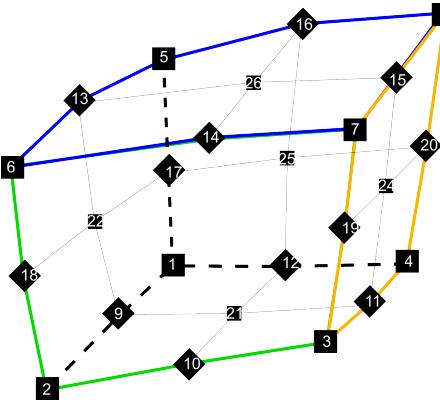
6.2.5 The bi/tri-quadratic velocity - bi/tri-linear pressure element ($Q_2 \times Q_1$)



In [813] Gresho & Sani write that in their opinion $\text{div}(\vec{v}) = 0$ is not strong enough.

This element, implemented in penalised form, is discussed in [146] and the follow-up paper [147].
CHECK

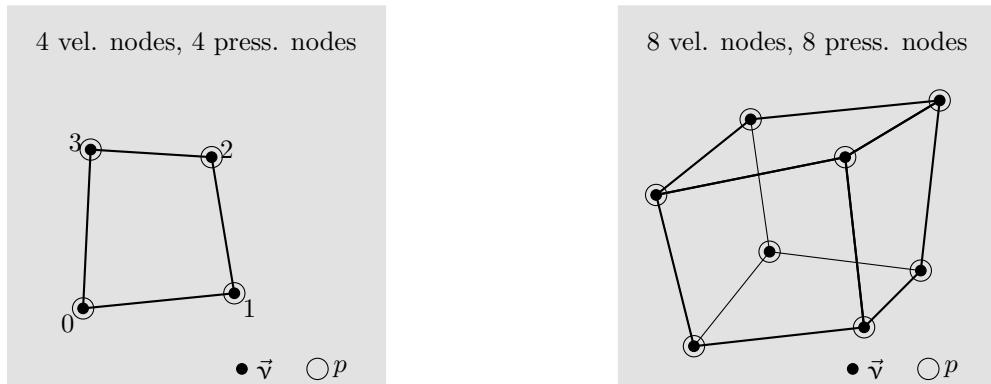
Biquadratic velocities, bilinear pressure. See Hood and Taylor. The element satisfies the inf-sup condition [941]p215.



6.2.6 The stabilised bi/tri-linear velocity - constant pressure element ($Q_1 \times P_0$ -stab)

Literature: [1197]

6.2.7 The stabilised bi/tri-linear velocity - bi/tri-linear pressure element ($Q_1 \times Q_1$ -stab)



See [1424] for a fourier analysis of the normal and stabilised (a la [943]) $Q_1 - Q_1$ element. This element is used in [307, 308] in conjunction with AMR. Stabilisation is worked out in [529, 181].

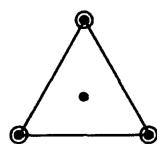
$Q_1 \times P_0$ -stab. Pro: stabilisation can be switched off; Con: stabilisation for deformed elements? problem near boundaries: incomplete stencil? choice of parameter β .

$Q_1 \times Q_1$ -stab. Pro: easier to implement than $Q_1 \times P_0$ -stab, stabilisation local to element, easier when elements are not rectangular, no free parameter; Con: stabilisation cannot be switched off.

Literature: [1750, 810, 1079, 644]

6.2.8 The MINI triangular element ($P_1^+ \times P_1$) in 2D

The MINI element was first introduced in Arnold et al, 1984 [41]. It is also discussed in section 3.6.1 of [998]. It is schematically represented hereunder:



Mini element:
Velocity: continuous linear
+ cubic bubble function,
Pressure: continuous linear,
Satisfies LBB condition,
Linear convergence.

Nodes: • Velocity

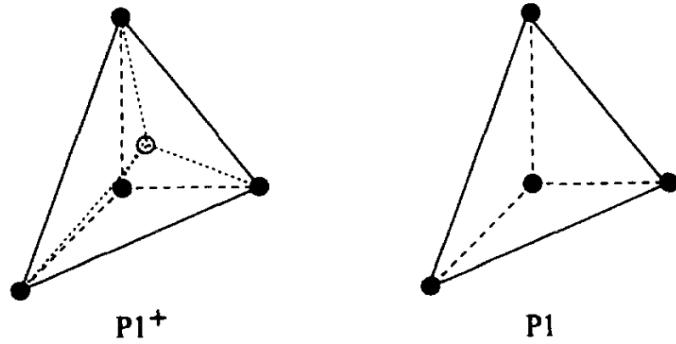
○ Pressure

Figure taken from Donea and Huerta [533]

Remark. Note that [643] propose an equal-order-linear-continuous velocity-pressure variables which is enriched with velocity and pressure bubble functions to model the Stokes problem. They show by static condensation that these bubble functions give rise to a stabilized method involving least-squares forms of the momentum and of the continuity equations. In some cases their approach recovers the MINI element. Also check [689].

Remark. According to Braess[209], since the support of the bubble is restricted to the element, the associated variable (dofs living on the bubble) can be eliminated from the resulting system of linear equations by static condensation. Also, the MINI element is cheaper than the Taylor-Hood element but it is commonly accepted that it yields a poorer approximation of the pressure.

The 3D MINI element is not very common but it is used for instance in [1474]. It is also said to be LBB stable in [1551, p180].



Velocity and pressure nodes for the 3D MINI element, taken from [1474]

Note that this element is used in [210] in the context of Arbitrary Lagrangian Eulerian finite element analysis of free surface flows.

6.2.9 The quadratic velocity - linear pressure triangle ($P_2 \times P_1$)

From [1653]: “Taylor-Hood elements [1744] are characterized by the fact that the pressure is continuous in the region Ω . A typical example is the quadratic triangle (P_2P_1 element). In this element the velocity is approximated by a quadratic polynomial and the pressure by a linear polynomial. One can easily verify that both approximations are continuous over the element boundaries.” It can be shown, Segal (1979), that this element is admissible if at least 3 elements are used. The quadrilateral counterpart of this triangle is the $Q_2 \times Q_1$ element. Reddy and Gartling [1551, p179] also report this element to be LBB stable.

6.2.10 The Crouzeix-Raviart triangle ($P_2^+ \times P_{-1}$)

Since the $P_2 \times P_{-1}$ pair is not LBB stable [1551, p179], it is enhanced by a cubic bubble and is therefore called $P_2^+ \times P_{-1}$.

This element was first introduced in [457]. It is the element used in the MILAMIN code [466]. It is a seven-node triangle with quadratic velocity shape functions enhanced by a cubic bubble function and discontinuous linear interpolation for the pressure field [465]. This element is LBB stable and no additional stabilization techniques are required[573]. The '+' in its name stands for the bubble while the '-' stands for the discontinuous character of the pressure field: once again, it is P_1 over the element, but discontinuous across element edges.

Remark. Cuvelier et al, 1986 [465] recommend a 6-point or 7-point quadrature rule for this element.

Remark. Segal [1653] explains for output purposes (printing, plotting etc.) the discontinuous pressures are averaged in vertices for all the adjoining elements. See also Fig. 7.3 of [465].

Remark. The simplest Crouzeix-Raviart element is the non-conforming linear triangle with constant pressure ($P_1 \times P_0$) [465].

It is worth noting that this element has more degrees of freedom than the Taylor-Hood element for the same order of accuracy. However, since the bubble can be eliminated, one can design a modified version of this element.

Check Cuvelier book chapter 8 for modified element

Remark. *I have once asked the (main) author of MILAMIN why he chose this element, for example over the $P_2 \times P_1$. His answer is as follows: "Elements with continuous pressure are incapable of converging in the L^∞ norm for mechanical problems exhibiting pressure jumps such as the inclusion-host setup. During my MSc and PhD I was focusing on sharp heterogeneities, so this is why I decided to choose $P_2^+ \times P_{-1}$. You will see that it is also easy to invert the pressure mass matrix for such elements, which is really useful (both for the augmentation and preconditioning)."*

This element is used by Poliakov and Podlachikov [1491] to study the deformation of the surface above a rising diapir. Note that they actually use a "13 point integration formula (Hughes 1987) for calculation of the stiffness matrix was used in order to conserve detailed information from the marker field in the coarse FEM mesh". It is also used in [31] in the context of a new free-surface stabilization scheme. It is the element used in LaCoDe [508].

6.2.11 The Rannacher-Turek element - rotated $Q_1 \times P_0$

p. 722 of [998]
[922, 1078, 1197]

6.2.12 Other elements

- $P_1 \times P_0$: example 3.70 in [998], also [997].
- $P_1 \times P_1$ stabilised [1423]
- Q2P0: : Quadratic velocities, constant pressure. The element satisfies the inf-sup condition, but the constant pressure assumption may require fine discretisation.
- Q2Q2: This element is never used, probably because a) it is unstable, b) it is very costly. There is one reference to it in [943].
- P2P2
- the MINI quadrilateral element $Q_1^+ \times Q_1$. [62, 1126]
- Q1P-1 Bilinear velocities, piecewise linear discontinuous polynomial pressure.

6.2.13 A note about incompressibility and standard mixed methods

What follows is nicely explained and demonstrated in John et al [1000]. In their example 1.1 they look at the velocity error of benchmark VJ2 (see Section 8.6.9) which analytical solution is a zero velocity field. They show that for the MINI, Taylor-Hood and Crouzeix-Raviart triangular elements the velocity error grows with the magnitude of the rhs. They also make this statement: "there are important applications, e.g., natural convection problems, where the pressure is larger than the velocity by orders of magnitude. In such situations, one cannot expect to compute accurate velocity fields with classical mixed methods, at least for low order methods."

6.3 The penalty approach for viscous flow

In order to impose the incompressibility constraint, two widely used procedures are available, namely the Lagrange multiplier method and the penalty method [83, 941]. The latter is implemented in ELEFANT, which allows for the elimination of the pressure variable from the momentum equation (resulting in a reduction of the matrix size).

Mathematical details on the origin and validity of the penalty approach applied to the Stokes problem can for instance be found in [465], [1550] or [830].

The penalty formulation of the mass conservation equation is based on a relaxation of the incompressibility constraint and writes

$$\vec{\nabla} \cdot \vec{v} + \frac{p}{\lambda} = 0 \quad (282)$$

where λ is the penalty parameter, that can be interpreted (and has the same dimension) as a bulk viscosity. It is equivalent to say that the material is weakly compressible. It can be shown that if one chooses λ to be a sufficiently large number, the continuity equation $\vec{\nabla} \cdot \vec{v} = 0$ will be approximately satisfied in the finite element solution. The value of λ is often recommended to be 6 to 7 orders of magnitude larger than the shear viscosity [533, 944].

Equation (282) can be used to eliminate the pressure in the momentum equation so that the mass and momentum conservation equations fuse to become :

$$\vec{\nabla} \cdot (2\eta \dot{\epsilon}(\vec{v})) + \lambda \vec{\nabla}(\vec{\nabla} \cdot \vec{v}) = \rho \mathbf{g} = 0 \quad (283)$$

[1253] have established the equivalence for incompressible problems between the reduced integration of the penalty term and a mixed Finite Element approach if the pressure nodes coincide with the integration points of the reduced rule.

In the end, the elimination of the pressure unknown in the Stokes equations replaces the original saddle-point Stokes problem [132] by an elliptical problem, which leads to a symmetric positive definite (SPD) FEM matrix. This is the major benefit of the penalized approach over the full indefinite solver with the velocity-pressure variables. Indeed, the SPD character of the matrix lends itself to efficient solving strategies and is less memory-demanding since it is sufficient to store only the upper half of the matrix including the diagonal [779] .

[list codes which use this approach](#)

Since the penalty formulation is only valid for incompressible flows, then $\dot{\epsilon} = \dot{\epsilon}^d$ so that the d superscript is omitted in what follows. Because the stress tensor is symmetric one can also rewrite it the following vector format:

$$\begin{aligned} \begin{pmatrix} \sigma_{xx} \\ \sigma_{yy} \\ \sigma_{zz} \\ \sigma_{xy} \\ \sigma_{xz} \\ \sigma_{yz} \end{pmatrix} &= \begin{pmatrix} -p \\ -p \\ -p \\ 0 \\ 0 \\ 0 \end{pmatrix} + 2\eta \begin{pmatrix} \dot{\epsilon}_{xx} \\ \dot{\epsilon}_{yy} \\ \dot{\epsilon}_{zz} \\ \dot{\epsilon}_{xy} \\ \dot{\epsilon}_{xz} \\ \dot{\epsilon}_{yz} \end{pmatrix} \\ &= \lambda \begin{pmatrix} \dot{\epsilon}_{xx} + \dot{\epsilon}_{yy} + \dot{\epsilon}_{zz} \\ \dot{\epsilon}_{xx} + \dot{\epsilon}_{yy} + \dot{\epsilon}_{zz} \\ \dot{\epsilon}_{xx} + \dot{\epsilon}_{yy} + \dot{\epsilon}_{zz} \\ 0 \\ 0 \\ 0 \end{pmatrix} + 2\eta \begin{pmatrix} \dot{\epsilon}_{xx} \\ \dot{\epsilon}_{yy} \\ \dot{\epsilon}_{zz} \\ \dot{\epsilon}_{xy} \\ \dot{\epsilon}_{xz} \\ \dot{\epsilon}_{yz} \end{pmatrix} \\ &= \underbrace{\lambda \begin{pmatrix} 1 & 1 & 1 & 0 & 0 & 0 \\ 1 & 1 & 1 & 0 & 0 & 0 \\ 1 & 1 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \end{pmatrix}}_{\mathbf{K}} \underbrace{+ \eta \begin{pmatrix} 2 & 0 & 0 & 0 & 0 & 0 \\ 0 & 2 & 0 & 0 & 0 & 0 \\ 0 & 0 & 2 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{pmatrix}}_{\mathbf{C}} \cdot \begin{pmatrix} \frac{\partial u}{\partial x} \\ \frac{\partial v}{\partial y} \\ \frac{\partial w}{\partial z} \\ \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \\ \frac{\partial u}{\partial z} + \frac{\partial w}{\partial x} \\ \frac{\partial v}{\partial z} + \frac{\partial w}{\partial y} \end{pmatrix} \end{aligned}$$

Remember that

$$\frac{\partial u}{\partial x} = \sum_{i=1}^4 \frac{\partial N_i}{\partial x} u_i \quad \frac{\partial v}{\partial y} = \sum_{i=1}^4 \frac{\partial N_i}{\partial y} v_i \quad \frac{\partial w}{\partial z} = \sum_{i=1}^4 \frac{\partial N_i}{\partial z} w_i$$

and

$$\begin{aligned}\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} &= \sum_{i=1}^4 \frac{\partial N_i}{\partial y} u_i + \sum_{i=1}^4 \frac{\partial N_i}{\partial x} v_i \\ \frac{\partial u}{\partial z} + \frac{\partial w}{\partial x} &= \sum_{i=1}^4 \frac{\partial N_i}{\partial z} u_i + \sum_{i=1}^4 \frac{\partial N_i}{\partial x} w_i \\ \frac{\partial v}{\partial z} + \frac{\partial w}{\partial y} &= \sum_{i=1}^4 \frac{\partial N_i}{\partial z} v_i + \sum_{i=1}^4 \frac{\partial N_i}{\partial y} w_i\end{aligned}$$

so that

$$\left(\begin{array}{c} \frac{\partial u}{\partial x} \\ \frac{\partial v}{\partial y} \\ \frac{\partial w}{\partial z} \\ \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \\ \frac{\partial u}{\partial z} + \frac{\partial w}{\partial x} \\ \frac{\partial v}{\partial z} + \frac{\partial w}{\partial y} \end{array} \right) = \underbrace{\left(\begin{array}{cccccccccccccccc} \frac{\partial N_1}{\partial x} & 0 & 0 & \frac{\partial N_2}{\partial x} & 0 & 0 & \frac{\partial N_3}{\partial x} & 0 & 0 & \dots & \frac{\partial N_4}{\partial x} & 0 & 0 \\ 0 & \frac{\partial N_1}{\partial y} & 0 & 0 & \frac{\partial N_2}{\partial y} & 0 & 0 & \frac{\partial N_3}{\partial y} & 0 & \dots & 0 & \frac{\partial N_4}{\partial y} & 0 \\ 0 & 0 & \frac{\partial N_1}{\partial z} & 0 & 0 & \frac{\partial N_2}{\partial z} & 0 & 0 & \frac{\partial N_3}{\partial z} & \dots & 0 & 0 & \frac{\partial N_4}{\partial z} \\ \frac{\partial N_1}{\partial y} & \frac{\partial N_1}{\partial x} & 0 & \frac{\partial N_2}{\partial y} & \frac{\partial N_2}{\partial x} & 0 & \frac{\partial N_3}{\partial y} & \frac{\partial N_3}{\partial x} & 0 & \dots & \frac{\partial N_4}{\partial y} & \frac{\partial N_4}{\partial x} & 0 \\ \frac{\partial N_1}{\partial z} & 0 & \frac{\partial N_1}{\partial x} & \frac{\partial N_2}{\partial z} & 0 & \frac{\partial N_2}{\partial x} & \frac{\partial N_3}{\partial z} & 0 & \frac{\partial N_3}{\partial x} & \dots & \frac{\partial N_4}{\partial z} & 0 & \frac{\partial N_4}{\partial x} \\ 0 & \frac{\partial N_1}{\partial z} & \frac{\partial N_1}{\partial y} & 0 & \frac{\partial N_2}{\partial z} & \frac{\partial N_2}{\partial y} & 0 & \frac{\partial N_3}{\partial z} & \frac{\partial N_3}{\partial y} & \dots & 0 & \frac{\partial N_4}{\partial z} & \frac{\partial N_4}{\partial y} \end{array} \right)}_{B(6 \times 24)} \cdot \underbrace{\left(\begin{array}{c} u1 \\ v1 \\ w1 \\ u2 \\ v2 \\ w2 \\ u3 \\ v3 \\ w3 \\ \dots \\ u8 \\ v8 \\ w8 \end{array} \right)}_{\vec{V}(24 \times 1)}.$$

Finally,

$$\vec{\sigma} = \begin{pmatrix} \sigma_{xx} \\ \sigma_{yy} \\ \sigma_{zz} \\ \sigma_{xy} \\ \sigma_{xz} \\ \sigma_{yz} \end{pmatrix} = (\lambda \mathbf{K} + \eta \mathbf{C}) \cdot \mathbf{B} \cdot \vec{V}$$

We will now establish the weak form of the momentum conservation equation. We start again from

$$\vec{\nabla} \cdot \boldsymbol{\sigma} + \vec{b} = \vec{0}$$

For the N_i 's 'regular enough', we can write:

$$\int_{\Omega_e} N_i \vec{\nabla} \cdot \boldsymbol{\sigma} d\Omega + \int_{\Omega_e} N_i \vec{b} d\Omega = 0$$

We can integrate by parts and drop the surface term²²:

$$\int_{\Omega_e} \vec{\nabla} N_i \cdot \boldsymbol{\sigma} d\Omega = \int_{\Omega_e} N_i \vec{b} d\Omega$$

or,

$$\int_{\Omega_e} \begin{pmatrix} \frac{\partial N_i}{\partial x} & 0 & 0 & \frac{\partial N_i}{\partial y} & \frac{\partial N_i}{\partial z} & 0 \\ 0 & \frac{\partial N_i}{\partial y} & 0 & \frac{\partial N_i}{\partial x} & 0 & \frac{\partial N_i}{\partial z} \\ 0 & 0 & \frac{\partial N_i}{\partial z} & 0 & \frac{\partial N_i}{\partial x} & \frac{\partial N_i}{\partial y} \end{pmatrix} \cdot \begin{pmatrix} \sigma_{xx} \\ \sigma_{yy} \\ \sigma_{zz} \\ \sigma_{xy} \\ \sigma_{xz} \\ \sigma_{yz} \end{pmatrix} d\Omega = \int_{\Omega_e} N_i \vec{b} d\Omega$$

²²We will come back to this at a later stage

Let $i = 1, 2, 3, 4, \dots, 8$ and stack the resulting eight equations on top of one another.

$$\begin{aligned}
& \int_{\Omega_e} \left(\begin{array}{cccccc} \frac{\partial N_i}{\partial x} & 0 & 0 & \frac{\partial N_i}{\partial y} & \frac{\partial N_i}{\partial z} & 0 \\ 0 & \frac{\partial N_i}{\partial y} & 0 & \frac{\partial N_i}{\partial x} & 0 & \frac{\partial N_i}{\partial z} \\ 0 & 0 & \frac{\partial N_i}{\partial z} & 0 & \frac{\partial N_i}{\partial x} & \frac{\partial N_i}{\partial y} \end{array} \right) \cdot \begin{pmatrix} \sigma_{xx} \\ \sigma_{yy} \\ \sigma_{zz} \\ \sigma_{xy} \\ \sigma_{xz} \\ \sigma_{yz} \end{pmatrix} d\Omega = \int_{\Omega_e} N_1 \begin{pmatrix} b_x \\ b_y \\ b_z \end{pmatrix} d\Omega \\
& \int_{\Omega_e} \left(\begin{array}{cccccc} \frac{\partial N_i}{\partial x} & 0 & 0 & \frac{\partial N_i}{\partial y} & \frac{\partial N_i}{\partial z} & 0 \\ 0 & \frac{\partial N_i}{\partial y} & 0 & \frac{\partial N_i}{\partial x} & 0 & \frac{\partial N_i}{\partial z} \\ 0 & 0 & \frac{\partial N_i}{\partial z} & 0 & \frac{\partial N_i}{\partial x} & \frac{\partial N_i}{\partial y} \end{array} \right) \cdot \begin{pmatrix} \sigma_{xx} \\ \sigma_{yy} \\ \sigma_{zz} \\ \sigma_{xy} \\ \sigma_{xz} \\ \sigma_{yz} \end{pmatrix} d\Omega = \int_{\Omega_e} N_2 \begin{pmatrix} b_x \\ b_y \\ b_z \end{pmatrix} d\Omega \\
& \dots \\
& \int_{\Omega_e} \left(\begin{array}{cccccc} \frac{\partial N_8}{\partial x} & 0 & 0 & \frac{\partial N_8}{\partial y} & \frac{\partial N_8}{\partial z} & 0 \\ 0 & \frac{\partial N_8}{\partial y} & 0 & \frac{\partial N_8}{\partial x} & 0 & \frac{\partial N_8}{\partial z} \\ 0 & 0 & \frac{\partial N_8}{\partial z} & 0 & \frac{\partial N_8}{\partial x} & \frac{\partial N_8}{\partial y} \end{array} \right) \cdot \begin{pmatrix} \sigma_{xx} \\ \sigma_{yy} \\ \sigma_{zz} \\ \sigma_{xy} \\ \sigma_{xz} \\ \sigma_{yz} \end{pmatrix} d\Omega = \int_{\Omega_e} N_8 \begin{pmatrix} b_x \\ b_y \\ b_z \end{pmatrix} d\Omega \quad (284)
\end{aligned}$$

We easily recognize B^T inside the integrals! Let us define

$$\vec{N}_b^T = (N_1 b_x, N_1 b_y, N_1 b_z \dots N_8 b_x, N_8 b_y, N_8 b_z)$$

then we can write

$$\int_{\Omega_e} \boldsymbol{B}^T \cdot \begin{pmatrix} \sigma_{xx} \\ \sigma_{yy} \\ \sigma_{zz} \\ \sigma_{xy} \\ \sigma_{xz} \\ \sigma_{yz} \end{pmatrix} d\Omega = \int_{\Omega_e} \vec{N}_b d\Omega$$

and finally:

$$\int_{\Omega_e} \boldsymbol{B}^T \cdot [\lambda \boldsymbol{K} + \eta \boldsymbol{C}] \cdot \boldsymbol{B} \cdot \vec{\boldsymbol{V}} d\Omega = \int_{\Omega_e} \vec{\boldsymbol{N}}_b d\Omega$$

Since \vec{V} contains the vector of unknowns (i.e. the velocities at the corners), it does not depend on the x or y coordinates so it can be taken outside of the integral:

$$\underbrace{\left(\int_{\Omega_e} \mathbf{B}^T \cdot [\lambda \mathbf{K} + \eta \mathbf{C}] \cdot \mathbf{B} d\Omega \right)}_{\mathbf{A}_{el}(24 \times 24)} \cdot \underbrace{\vec{V}}_{(24 \times 1)} = \underbrace{\int_{\Omega_e} \vec{N}_b b d\Omega}_{\vec{B}_{el}(24 \times 1)}$$

or,

$$\left[\underbrace{\left(\int_{\Omega_e} \lambda \mathbf{B}^T \cdot \mathbf{K} \cdot \mathbf{B} d\Omega \right)}_{\mathbf{A}_{el}^\lambda (24 \times 24)} + \underbrace{\left(\int_{\Omega_e} \eta \mathbf{B}^T \cdot \mathbf{C} \cdot \mathbf{B} d\Omega \right)}_{\mathbf{A}_{el}^\eta (24 \times 24)} \right] \cdot \underbrace{\vec{V}_{(24 \times 1)}}_{\vec{B}_{el}(24 \times 1)} = \underbrace{\int_{\Omega_e} \vec{N}_b d\Omega}_{\vec{B}_{el}(24 \times 1)}$$

reduced integration

reduced integration [944]

write about 3D to 2D

6.4 The mixed FEM for viscous flow

6.4.1 in three dimensions

In what follows the flow is assumed to be incompressible, isoviscous and isothermal.

The methodology to derive the discretised equations of the mixed system is quite similar to the one we have used in the case of the penalty formulation. The big difference comes from the fact that we are now solving for both velocity and pressure at the same time, and that we therefore must solve the mass and momentum conservation equations together. As before, velocity inside an element is given by

$$\vec{v}^h(\vec{r}) = \sum_{i=1}^{m_v} N_i^v(\vec{r}) \vec{v}_i \quad (285)$$

where N_i^v are the polynomial basis functions for the velocity, and the summation runs over the m_v nodes composing the element. A similar expression is used for pressure:

$$p^h(\vec{r}) = \sum_{i=1}^{m_p} N_i^p(\vec{r}) p_i \quad (286)$$

Note that the velocity is a vector of size while pressure (and temperature) is a scalar. There are then $ndof_v$ velocity degrees of freedom per node and $ndof_p$ pressure degrees of freedom. It is also very important to remember that the numbers of velocity nodes and pressure nodes for a given element are more often than not different and that velocity and pressure nodes need not be colocated. Indeed, unless co-called 'stabilised elements' are used, we have $m_v > m_p$, which means that the polynomial order of the velocity field is higher than the polynomial order of the pressure field (usually by value 1).

insert here link(s) to manual and literature

Other notations are sometimes used for Eqs.(285) and (286):

$$u^h(\vec{r}) = \vec{N}^v \cdot \vec{u} \quad v^h(\vec{r}) = \vec{N}^v \cdot \vec{v} \quad w^h(\vec{r}) = \vec{N}^v \cdot \vec{w} \quad p^h(\vec{r}) = \vec{N}^p \cdot \vec{p} \quad (287)$$

where $\vec{v} = (u, v, w)$ and \vec{N}^v is the vector containing all basis functions evaluated at location \vec{r} :

$$\vec{N}^v = (N_1^v(\vec{r}), N_2^v(\vec{r}), N_3^v(\vec{r}), \dots, N_{m_v}^v(\vec{r})) \quad (288)$$

$$\vec{N}^p = (N_1^p(\vec{r}), N_2^p(\vec{r}), N_3^p(\vec{r}), \dots, N_{m_p}^p(\vec{r})) \quad (289)$$

and with

$$\vec{u} = (u_1, u_2, u_3, \dots, u_{m_v}) \quad (290)$$

$$\vec{v} = (v_1, v_2, v_3, \dots, v_{m_v}) \quad (291)$$

$$\vec{w} = (w_1, w_2, w_3, \dots, w_{m_v}) \quad (292)$$

$$\vec{p} = (p_1, p_2, p_3, \dots, p_{m_p}) \quad (293)$$

We will now establish the weak form of the momentum conservation equation. We start again from

$$\vec{\nabla} \cdot \boldsymbol{\sigma} + \vec{b} = \vec{0} \quad (294)$$

$$\vec{\nabla} \cdot \vec{v} = 0 \quad (295)$$

For the N_i^v 's and N_i^p 'regular enough', we can write:

$$\int_{\Omega_e} N_i^v \vec{\nabla} \cdot \boldsymbol{\sigma} d\Omega + \int_{\Omega_e} N_i^v \vec{b} d\Omega = \vec{0} \quad (296)$$

$$\int_{\Omega_e} N_i^p \vec{\nabla} \cdot \vec{v} d\Omega = 0 \quad (297)$$

We can integrate by parts and drop the surface term²³:

$$\int_{\Omega_e} \vec{\nabla} N_i^v \cdot \boldsymbol{\sigma} d\Omega = \int_{\Omega_e} N_i^v \vec{b} d\Omega \quad (298)$$

$$\int_{\Omega_e} N_i^p \vec{\nabla} \cdot \vec{v} d\Omega = 0 \quad (299)$$

²³We will come back to this at a later stage

or,

$$\int_{\Omega_e} \begin{pmatrix} \frac{\partial N_i^\gamma}{\partial x} & 0 & 0 & \frac{\partial N_i^\gamma}{\partial y} & \frac{\partial N_i^\gamma}{\partial z} & 0 \\ 0 & \frac{\partial N_i^\gamma}{\partial y} & 0 & \frac{\partial N_i^\gamma}{\partial x} & 0 & \frac{\partial N_i^\gamma}{\partial z} \\ 0 & 0 & \frac{\partial N_i^\gamma}{\partial z} & 0 & \frac{\partial N_i^\gamma}{\partial x} & \frac{\partial N_i^\gamma}{\partial y} \end{pmatrix} \cdot \begin{pmatrix} \sigma_{xx} \\ \sigma_{yy} \\ \sigma_{zz} \\ \sigma_{xy} \\ \sigma_{xz} \\ \sigma_{yz} \end{pmatrix} d\Omega = \int_{\Omega_e} N_i^\gamma \vec{b} d\Omega \quad (300)$$

The above equation can ultimately be written:

$$\int_{\Omega_e} \mathbf{B}^T \cdot \begin{pmatrix} \sigma_{xx} \\ \sigma_{yy} \\ \sigma_{zz} \\ \sigma_{xy} \\ \sigma_{xz} \\ \sigma_{yz} \end{pmatrix} d\Omega = \int_{\Omega_e} \vec{N}_b d\Omega \quad (301)$$

We have previously established that the strain rate vector $\vec{\varepsilon}$ is:

$$\vec{\varepsilon} = \begin{pmatrix} \frac{\partial u}{\partial x} \\ \frac{\partial v}{\partial y} \\ \frac{\partial w}{\partial z} \\ \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \\ \frac{\partial u}{\partial z} + \frac{\partial w}{\partial x} \\ \frac{\partial v}{\partial z} + \frac{\partial w}{\partial y} \end{pmatrix} = \begin{pmatrix} \sum_i \frac{\partial N_i^\gamma}{\partial x} u_i \\ \sum_i \frac{\partial N_i^\gamma}{\partial y} v_i \\ \sum_i \frac{\partial N_i^\gamma}{\partial z} w_i \\ \sum_i \left(\frac{\partial N_i^\gamma}{\partial y} u_i + \frac{\partial N_i^\gamma}{\partial x} v_i \right) \\ \sum_i \left(\frac{\partial N_i^\gamma}{\partial z} u_i + \frac{\partial N_i^\gamma}{\partial x} w_i \right) \\ \sum_i \left(\frac{\partial N_i^\gamma}{\partial z} v_i + \frac{\partial N_i^\gamma}{\partial y} w_i \right) \end{pmatrix} = \underbrace{\begin{pmatrix} \frac{\partial N_1^\gamma}{\partial x} & 0 & 0 & \dots & \frac{\partial N_{m_v}^\gamma}{\partial x} & 0 & 0 \\ 0 & \frac{\partial N_1^\gamma}{\partial y} & 0 & \dots & 0 & \frac{\partial N_{m_v}^\gamma}{\partial y} & 0 \\ 0 & 0 & \frac{\partial N_1^\gamma}{\partial z} & \dots & 0 & 0 & \frac{\partial N_{m_v}^\gamma}{\partial z} \\ \frac{\partial N_1^\gamma}{\partial y} & \frac{\partial N_1^\gamma}{\partial x} & 0 & \dots & \frac{\partial N_{m_v}^\gamma}{\partial x} & \frac{\partial N_{m_v}^\gamma}{\partial x} & 0 \\ \frac{\partial N_1^\gamma}{\partial z} & 0 & \frac{\partial N_1^\gamma}{\partial x} & \dots & \frac{\partial N_{m_v}^\gamma}{\partial z} & 0 & \frac{\partial N_{m_v}^\gamma}{\partial x} \\ 0 & \frac{\partial N_1^\gamma}{\partial z} & \frac{\partial N_1^\gamma}{\partial y} & \dots & 0 & \frac{\partial N_{m_v}^\gamma}{\partial z} & \frac{\partial N_{m_v}^\gamma}{\partial y} \end{pmatrix}}_B \begin{pmatrix} u_1 \\ v_1 \\ w_1 \\ u_2 \\ v_2 \\ w_2 \\ u_3 \\ v_3 \\ \dots \\ u_{m_v} \\ v_{m_v} \\ w_{m_v} \end{pmatrix} \underbrace{\vec{V}}$$
(302)

or, $\vec{\varepsilon} = \mathbf{B} \cdot \vec{V}$ where \mathbf{B} is the gradient matrix and \vec{V} is the vector of all vector degrees of freedom for the element. The matrix \mathbf{B} is then of size $3 \times m_v \times ndim$ and the vector \vec{V} is $m_v * ndof$ long. we have

$$\sigma_{xx} = -p + 2\eta\dot{\varepsilon}_{xx}^d \quad (303)$$

$$\sigma_{yy} = -p + 2\eta\dot{\varepsilon}_{yy}^d \quad (304)$$

$$\sigma_{zz} = -p + 2\eta\dot{\varepsilon}_{zz}^d \quad (305)$$

$$\sigma_{xy} = 2\eta\dot{\varepsilon}_{xy}^d \quad (306)$$

$$\sigma_{xz} = 2\eta\dot{\varepsilon}_{xz}^d \quad (307)$$

$$\sigma_{yz} = 2\eta\dot{\varepsilon}_{yz}^d \quad (308)$$

Since we here only consider incompressible flow, we have $\dot{\varepsilon}^d = \dot{\varepsilon}$ so

$$\vec{\sigma} = - \begin{pmatrix} 1 \\ 1 \\ 1 \\ 0 \\ 0 \\ 0 \end{pmatrix} p + \mathbf{C} \cdot \vec{\varepsilon} = - \begin{pmatrix} 1 \\ 1 \\ 1 \\ 0 \\ 0 \\ 0 \end{pmatrix} \vec{N}^p \cdot \vec{P} + \mathbf{C} \cdot \mathbf{B} \cdot \vec{V} \quad (309)$$

with

$$\mathbf{C} = \eta \begin{pmatrix} 2 & 0 & 0 & 0 & 0 & 0 \\ 0 & 2 & 0 & 0 & 0 & 0 \\ 0 & 0 & 2 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{pmatrix} \quad \vec{\varepsilon} = \begin{pmatrix} \dot{\varepsilon}_{xx} \\ \dot{\varepsilon}_{yy} \\ \dot{\varepsilon}_{zz} \\ 2\dot{\varepsilon}_{xy} \\ 2\dot{\varepsilon}_{xz} \\ 2\dot{\varepsilon}_{yz} \end{pmatrix} \quad (310)$$

Let us define matrix \mathbf{N}^p of size $6 \times m_p$:

$$\mathbf{N}^p = \begin{pmatrix} 1 \\ 1 \\ 1 \\ 0 \\ 0 \\ 0 \end{pmatrix} \quad \vec{N}^p = \begin{pmatrix} \vec{N}^p \\ \vec{N}^p \\ \vec{N}^p \\ 0 \\ 0 \\ 0 \end{pmatrix} \quad (311)$$

so that

$$\vec{\sigma} = -\mathbf{N}^p \cdot \vec{P} + \mathbf{C} \cdot \mathbf{B} \cdot \vec{V} \quad (312)$$

finally

$$\int_{\Omega_e} \mathbf{B}^T \cdot [-\mathbf{N}^p \cdot \vec{P} + \mathbf{C} \cdot \mathbf{B} \cdot \vec{V}] d\Omega = \int_{\Omega_e} \mathbf{N}_b d\Omega \quad (313)$$

or,

$$\underbrace{\left(- \int_{\Omega_e} \mathbf{B}^T \cdot \mathbf{N}^p d\Omega \right)}_{\mathbb{G}} \cdot \vec{P} + \underbrace{\left(\int_{\Omega_e} \mathbf{B}^T \cdot \mathbf{C} \cdot \mathbf{B} d\Omega \right)}_{\mathbb{K}} \cdot \vec{V} = \underbrace{\int_{\Omega_e} \vec{N}_b d\Omega}_{\vec{f}} \quad (314)$$

where the matrix \mathbb{K} is of size $(m_v * ndof_v \times m_v * ndof_v)$, and matrix \mathbb{G} is of size $(m_v * ndof_v \times m_p * ndof_p)$.

Turning now to the mass conservation equation:

$$\begin{aligned}
\vec{0} &= \int_{\Omega_e} \vec{N}^p \vec{\nabla} \cdot \vec{v} d\Omega \\
&= \int_{\Omega_e} \vec{N}^p \sum_{i=1}^{m_v} \left(\frac{\partial N_i^v}{\partial x} u_i + \frac{\partial N_i^v}{\partial y} v_i + \frac{\partial N_i^v}{\partial z} w_i \right) d\Omega \\
&= \int_{\Omega_e} \begin{pmatrix} N_1^p \left(\sum_{i=1}^{m_v} \frac{\partial N_i^v}{\partial x} u_i + \sum_{i=1}^{m_v} \frac{\partial N_i^v}{\partial y} v_i + \sum_{i=1}^{m_v} \frac{\partial N_i^v}{\partial z} w_i \right) \\ N_2^p \left(\sum_{i=1}^{m_v} \frac{\partial N_i^v}{\partial x} u_i + \sum_{i=1}^{m_v} \frac{\partial N_i^v}{\partial y} v_i + \sum_{i=1}^{m_v} \frac{\partial N_i^v}{\partial z} w_i \right) \\ N_3^p \left(\sum_{i=1}^{m_v} \frac{\partial N_i^v}{\partial x} u_i + \sum_{i=1}^{m_v} \frac{\partial N_i^v}{\partial y} v_i + \sum_{i=1}^{m_v} \frac{\partial N_i^v}{\partial z} w_i \right) \\ \vdots \\ N_{m_p}^p \left(\sum_{i=1}^{m_v} \frac{\partial N_i^v}{\partial x} u_i + \sum_{i=1}^{m_v} \frac{\partial N_i^v}{\partial y} v_i + \sum_{i=1}^{m_v} \frac{\partial N_i^v}{\partial z} w_i \right) \end{pmatrix} d\Omega \\
&= \int_{\Omega_e} \begin{pmatrix} N_1^p & N_1^p & N_1^p & 0 & 0 & 0 \\ N_2^p & N_2^p & N_2^p & 0 & 0 & 0 \\ N_3^p & N_3^p & N_3^p & 0 & 0 & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots & \vdots \\ N_{m_p}^p & N_{m_p}^p & N_{m_p}^p & 0 & 0 & 0 \end{pmatrix} \cdot \begin{pmatrix} \sum_i \frac{\partial N_i^v}{\partial x} u_i \\ \sum_i \frac{\partial N_i^v}{\partial y} v_i \\ \sum_i \frac{\partial N_i^v}{\partial z} w_i \\ \sum_i (\frac{\partial N_i^v}{\partial y} u_i + \frac{\partial N_i^v}{\partial x} v_i) \\ \sum_i (\frac{\partial N_i^v}{\partial z} u_i + \frac{\partial N_i^v}{\partial x} w_i) \\ \sum_i (\frac{\partial N_i^v}{\partial z} v_i + \frac{\partial N_i^v}{\partial y} w_i) \end{pmatrix} d\Omega \\
&= \int_{\Omega_e} \underbrace{\begin{pmatrix} N_1^p & N_1^p & N_1^p & 0 & 0 & 0 \\ N_2^p & N_2^p & N_2^p & 0 & 0 & 0 \\ N_3^p & N_3^p & N_3^p & 0 & 0 & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots & \vdots \\ N_{m_p}^p & N_{m_p}^p & N_{m_p}^p & 0 & 0 & 0 \end{pmatrix}}_{\mathbf{N}^p} \cdot \vec{\varepsilon} d\Omega \\
&= \left(\int \mathbf{N}^p \cdot \mathbf{B} d\Omega \right) \cdot \vec{V} \\
&= -\mathbb{G}_e^T \cdot \vec{V} \tag{315}
\end{aligned}$$

Note that it is common to actually start from $-\vec{\nabla} \cdot \vec{v} = 0$ (see Eq.(3) in [1292]) so as to arrive at $\mathbb{G}_e^T \cdot \vec{V} = \vec{0}$

Ultimately we obtain the following system for each element:

$$\begin{pmatrix} \mathbb{K}_e & \mathbb{G}_e \\ -\mathbb{G}_e^T & 0 \end{pmatrix} \cdot \begin{pmatrix} \vec{V} \\ \vec{P} \end{pmatrix} = \begin{pmatrix} \vec{f}_e \\ 0 \end{pmatrix}$$

Such a matrix is then generated for each element and then must be assembled into the global F.E. matrix. Note that in this case the elemental Stokes matrix is antisymmetric. One can also define the following symmetric modified Stokes matrix:

$$\begin{pmatrix} \mathbb{K}_e & \mathbb{G}_e \\ \mathbb{G}_e^T & 0 \end{pmatrix} \cdot \begin{pmatrix} \vec{V} \\ \vec{P} \end{pmatrix} = \begin{pmatrix} \vec{f}_e \\ 0 \end{pmatrix}$$

This matrix is symmetric, but indefinite. It is non-singular if $\ker(\mathbb{G}^T) = 0$, which is the case if the compatibility condition holds.

CHECK: Matrix \mathbb{K} is the viscosity matrix. Its size is $(ndof_v * N_v) \times (ndof_v * N_v)$ where $ndof_v$ is the number of velocity degrees of freedom per node (typically 1,2 or 3) and N_v is the number of velocity nodes. The size of matrix \mathbb{G} is $(ndof_v * N_v) \times (ndof_p * N_p)$ where $ndof_p (= 1)$ is the number of velocity degrees of freedom per node and N_p is the number of pressure nodes. Conversely, the size of matrix \mathbb{G}^T is $(ndof_p * N_p) \times (ndof_v * N_v)$. The size of the global FE matrix is $N = ndof_v * N_v + ndof_p * N_p$. Note that matrix \mathbb{K} is analogous to a discrete Laplacian operator, matrix \mathbb{G} to a discrete gradient operator, and matrix \mathbb{G}^T to a discrete divergence operator.

On the physical dimensions of the Stokes matrix blocks We start from the Stokes equations:

$$-\vec{\nabla} p + \vec{\nabla} \cdot (2\eta \dot{\boldsymbol{\varepsilon}}) + \rho \mathbf{g} = 0 \quad (316)$$

$$\vec{\nabla} \cdot \vec{v} = 0 \quad (317)$$

The dimensions of the terms in the first equation are: $ML^{-2}T^{-2}$. The blocks \mathbb{K} and \mathbb{G} stem from the weak form which obtained by multiplying the strong form equations by the (dimensionless) basis functions and integrating over the domain, so that it follows that

$$[\mathbb{K} \cdot \vec{V}] = [\mathbb{G} \cdot \vec{P}] = [\vec{f}] = ML^{-2}T^{-2}L^3 = MLT^{-2}$$

We can then easily deduce:

$$[\mathbb{K}] = MT^{-1} \quad [\mathbb{G}] = L^2$$

On elemental level mass balance. Note that in what is above no assumption has been made about whether the pressure basis functions are continuous or discontinuous from one element to another.

Indeed, as mentioned in [813], since the weak formulation of the momentum equation involves integration by parts of $\vec{\nabla} p$, the resulting weak form contains no derivatives of pressure. This introduces the possibility of approximating it by functions (piecewise polynomials, of course) that are not C^0 -continuous, and indeed this has been done and is quite popular/useful.

It is then worth noting that *only* discontinuous pressure elements assure an element-level mass balance [813]: if for instance N_i^p is piecewise-constant on element e (of value 1), the elemental weak form of the mass conservation equation is

$$\int_{\Omega_e} N_i^p \vec{\nabla} \cdot \vec{v} = \int_{\Omega_e} \vec{\nabla} \cdot \vec{v} = \int_{\Gamma_e} \vec{n} \cdot \vec{v} = 0$$

One potentially unwelcome consequence of using discontinuous pressure elements is that they do not possess uniquely defined pressure on the element boundaries; they are dual valued there, and often multi-valued at certain velocity nodes.

On the \mathbf{C} matrix The relationship between deviatoric stress and deviatoric strain rate tensor is

$$\boldsymbol{\tau} = 2\eta \dot{\boldsymbol{\varepsilon}}^d \quad (318)$$

$$= 2\eta \left(\dot{\boldsymbol{\varepsilon}} - \frac{1}{3} (\vec{\nabla} \cdot \vec{v}) \mathbf{1} \right) \quad (319)$$

$$= 2\eta \left[\begin{pmatrix} \dot{\varepsilon}_{xx} & \dot{\varepsilon}_{xy} & \dot{\varepsilon}_{xz} \\ \dot{\varepsilon}_{yx} & \dot{\varepsilon}_{yy} & \dot{\varepsilon}_{yz} \\ \dot{\varepsilon}_{zx} & \dot{\varepsilon}_{zy} & \dot{\varepsilon}_{zz} \end{pmatrix} - \frac{1}{3} (\dot{\varepsilon}_{xx} + \dot{\varepsilon}_{yy} + \dot{\varepsilon}_{zz}) \begin{pmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{pmatrix} \right] \quad (320)$$

$$= \frac{2}{3} \eta \begin{pmatrix} 2\dot{\varepsilon}_{xx} - \dot{\varepsilon}_{yy} - \dot{\varepsilon}_{zz} & 3\dot{\varepsilon}_{xy} & 3\dot{\varepsilon}_{xz} \\ 3\dot{\varepsilon}_{yx} & -\dot{\varepsilon}_{yy} + 2\dot{\varepsilon}_{yy} - \dot{\varepsilon}_{yy} & 3\dot{\varepsilon}_{yz} \\ 3\dot{\varepsilon}_{zx} & 3\dot{\varepsilon}_{zy} & -\dot{\varepsilon}_{xx} - \dot{\varepsilon}_{yy} - 2\dot{\varepsilon}_{zz} \end{pmatrix} \quad (321)$$

so that

$$\vec{\tau} = \frac{2}{3}\eta \begin{pmatrix} 2\dot{\varepsilon}_{xx} - \dot{\varepsilon}_{yy} - \dot{\varepsilon}_{zz} \\ -\dot{\varepsilon}_{yy} + 2\dot{\varepsilon}_{yy} - \dot{\varepsilon}_{yy} \\ -\dot{\varepsilon}_{xx} - \dot{\varepsilon}_{yy} + 2\dot{\varepsilon}_{zz} \\ 3\dot{\varepsilon}_{xy} \\ 3\dot{\varepsilon}_{xz} \\ 3\dot{\varepsilon}_{yz} \end{pmatrix} = \underbrace{\frac{\eta}{3} \begin{pmatrix} 4 & -2 & -2 & 0 & 0 & 0 \\ -2 & 4 & -2 & 0 & 0 & 0 \\ -2 & -2 & 4 & 0 & 0 & 0 \\ 0 & 0 & 0 & 3 & 0 & 0 \\ 0 & 0 & 0 & 0 & 3 & 0 \\ 0 & 0 & 0 & 0 & 0 & 3 \end{pmatrix}}_{C^d} \cdot \begin{pmatrix} \dot{\varepsilon}_{xx} \\ \dot{\varepsilon}_{yy} \\ \dot{\varepsilon}_{zz} \\ 2\dot{\varepsilon}_{xy} \\ 2\dot{\varepsilon}_{xz} \\ 2\dot{\varepsilon}_{yz} \end{pmatrix} = \mathbf{C}^d \cdot \vec{\varepsilon} \quad (322)$$

which is identical to the one in the Appendix A of [1636]. In two dimensions, we have

$$\vec{\tau} = \frac{1}{3}\eta \underbrace{\begin{pmatrix} 4 & -2 & 0 \\ -2 & 4 & 0 \\ 0 & 0 & 3 \end{pmatrix}}_{C^d}.$$

see for instance [31].

In the case where we assume incompressible flow from the beginning, i.e. $\dot{\varepsilon} = \dot{\varepsilon}^d$, then

$$\vec{\tau} = \eta \underbrace{\begin{pmatrix} 2 & 0 & 0 & 0 & 0 & 0 \\ 0 & 2 & 0 & 0 & 0 & 0 \\ 0 & 0 & 2 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{pmatrix}}_C \cdot \begin{pmatrix} \dot{\varepsilon}_{xx} \\ \dot{\varepsilon}_{yy} \\ \dot{\varepsilon}_{zz} \\ 2\dot{\varepsilon}_{xy} \\ 2\dot{\varepsilon}_{xz} \\ 2\dot{\varepsilon}_{yz} \end{pmatrix} = \mathbf{C} \cdot \vec{\varepsilon} \quad (323)$$

Two slightly different formulations The momentum conservation equation can be written as follows:

$$\vec{\nabla} \cdot (2\eta \vec{\varepsilon}) - \vec{\nabla} p + \vec{b} = \vec{0}$$

When the viscosity η is constant this equation becomes

$$\eta \Delta \vec{v} - \vec{\nabla} p + \vec{b} = \vec{0}$$

In this case the matrix \mathbf{B} takes a different form [533, Eq. 6.24] and one should be aware that this can have consequences for the Neumann boundary conditions.

In [307] the authors state that when the Laplacian formulation is used it has the computational advantage that the velocity components are coupled only through the incompressibility condition. While the two formulations are equivalent only for constant viscosity, they state that they employ the Laplacian approach formulation as a preconditioner for the viscous term.

Concretely, we apply the same method as above, i.e. we reorganise the terms of the velocity gradient

tensor in a vector:

$$\vec{\nabla} \vec{v} \rightarrow \begin{pmatrix} \partial_x u \\ \partial_y u \\ \partial_z u \\ \partial_x v \\ \partial_y v \\ \partial_z v \\ \partial_x w \\ \partial_y w \\ \partial_z w \end{pmatrix} = \begin{pmatrix} \sum_i \partial_x N_i u_i \\ \sum_i \partial_y N_i u_i \\ \sum_i \partial_z N_i u_i \\ \sum_i \partial_x N_i v_i \\ \sum_i \partial_y N_i v_i \\ \sum_i \partial_z N_i v_i \\ \sum_i \partial_x N_i w_i \\ \sum_i \partial_y N_i w_i \\ \sum_i \partial_z N_i w_i \end{pmatrix}$$

$$= \underbrace{\begin{pmatrix} \partial_x N_1^Y & 0 & 0 & \partial_x N_2^Y & 0 & 0 & \dots & \partial_x N_{m_v}^Y & 0 & 0 \\ \partial_y N_1^Y & 0 & 0 & \partial_y N_2^Y & 0 & 0 & \dots & \partial_y N_{m_v}^Y & 0 & 0 \\ \partial_z N_1^Y & 0 & 0 & \partial_z N_2^Y & 0 & 0 & \dots & \partial_z N_{m_v}^Y & 0 & 0 \\ 0 & \partial_x N_1^Y & 0 & 0 & \partial_x N_2^Y & 0 & \dots & 0 & \partial_x N_{m_v}^Y & 0 \\ 0 & \partial_y N_1^Y & 0 & 0 & \partial_y N_2^Y & 0 & \dots & 0 & \partial_y N_{m_v}^Y & 0 \\ 0 & \partial_z N_1^Y & 0 & 0 & \partial_z N_2^Y & 0 & \dots & 0 & \partial_z N_{m_v}^Y & 0 \\ 0 & 0 & \partial_x N_1^Y & 0 & 0 & \partial_x N_2^Y & \dots & 0 & 0 & \partial_x N_{m_v}^Y \\ 0 & 0 & \partial_y N_1^Y & 0 & 0 & \partial_y N_2^Y & \dots & 0 & 0 & \partial_y N_{m_v}^Y \\ 0 & 0 & \partial_z N_1^Y & 0 & 0 & \partial_z N_2^Y & \dots & 0 & 0 & \partial_z N_{m_v}^Y \end{pmatrix}}_B \cdot \underbrace{\begin{pmatrix} u_1 \\ v_1 \\ w_1 \\ u_2 \\ v_2 \\ w_2 \\ u_3 \\ v_3 \\ \dots \\ u_{m_v} \\ v_{m_v} \\ w_{m_v} \end{pmatrix}}_{\vec{V}}$$

and in two dimensions:

$$\vec{\nabla} \vec{v} \rightarrow \begin{pmatrix} \partial_x u \\ \partial_y u \\ \partial_x v \\ \partial_y v \end{pmatrix} = \begin{pmatrix} \sum_i \partial_x N_i u_i \\ \sum_i \partial_y N_i u_i \\ \sum_i \partial_x N_i v_i \\ \sum_i \partial_y N_i v_i \end{pmatrix} = \underbrace{\begin{pmatrix} \partial_x N_1^Y & 0 & \partial_x N_2^Y & 0 & \dots & \partial_x N_{m_v}^Y & 0 \\ \partial_y N_1^Y & 0 & \partial_y N_2^Y & 0 & \dots & \partial_y N_{m_v}^Y & 0 \\ 0 & \partial_x N_1^Y & 0 & \partial_x N_2^Y & \dots & 0 & \partial_x N_{m_v}^Y \\ 0 & \partial_y N_1^Y & 0 & \partial_y N_2^Y & \dots & 0 & \partial_y N_{m_v}^Y \end{pmatrix}}_B \cdot \underbrace{\begin{pmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \\ u_3 \\ v_3 \\ \dots \\ u_{m_v} \\ v_{m_v} \end{pmatrix}}_{\vec{V}}$$

On the 'forgotten' surface terms

6.4.2 Going from 3D to 2D

The world is three-dimensional. However, for many different reasons one may wish to solve problems which are two-dimensional.

Following ASPECT manual, we will think of two-dimensional models in the following way:

- We assume that the domain we want to solve on is a two-dimensional cross section (in the $x - y$ plane) that extends infinitely far in both negative and positive z direction.
- We assume that the velocity is zero in the z direction and that all variables have no variation in the z direction.

As a consequence, two-dimensional models are three-dimensional ones in which the z component of the velocity is zero and so are all z derivatives. This allows to reduce the momentum conservation equations from 3 equations to 2 equations. However, contrarily to what is often seen, the 3D definition of the deviatoric strain rate remains, i.e. in other words:

$$\dot{\epsilon}^d = \dot{\epsilon} - \frac{1}{3}(\vec{\nabla} \cdot \vec{v})\mathbf{1} \quad (324)$$

and not $1/2$. In light of all this, the full strain rate tensor and the deviatoric strain rate tensor in 2D are given by:

$$\boldsymbol{\varepsilon} = \begin{pmatrix} \dot{\varepsilon}_{xx} & \dot{\varepsilon}_{xy} & \dot{\varepsilon}_{xz} \\ \dot{\varepsilon}_{yx} & \dot{\varepsilon}_{yy} & \dot{\varepsilon}_{yz} \\ \dot{\varepsilon}_{zx} & \dot{\varepsilon}_{zy} & \dot{\varepsilon}_{zz} \end{pmatrix} = \begin{pmatrix} \frac{\partial u}{\partial x} & \frac{1}{2} \left(\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) & 0 \\ \frac{1}{2} \left(\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) & \frac{\partial v}{\partial y} & 0 \\ 0 & 0 & 0 \end{pmatrix} \quad (325)$$

$$\dot{\boldsymbol{\varepsilon}}^d = \frac{1}{3} \begin{pmatrix} 2 \frac{\partial u}{\partial x} - \frac{\partial v}{\partial y} & \frac{1}{2} \left(\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) & 0 \\ \frac{1}{2} \left(\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) & -\frac{\partial u}{\partial x} + 2 \frac{\partial v}{\partial y} & 0 \\ 0 & 0 & -\frac{\partial u}{\partial x} - \frac{\partial v}{\partial y} \end{pmatrix} \quad (326)$$

Although the bottom right term may be surprising, it is of no consequence when this expression of the deviatoric strain rate is used in the Stokes equation:

$$\vec{\nabla} \cdot 2\eta \dot{\boldsymbol{\varepsilon}}^d =$$

FINISH!

In two dimensions the velocity is then $\vec{v} = (u, v)$ and the FEM building blocks and matrices are simply:

$$\vec{\dot{\boldsymbol{\varepsilon}}} = \begin{pmatrix} \dot{\varepsilon}_{xx} \\ \dot{\varepsilon}_{yy} \\ 2\dot{\varepsilon}_{xy} \end{pmatrix} = \begin{pmatrix} \frac{\partial u}{\partial x} \\ \frac{\partial v}{\partial y} \\ \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \end{pmatrix} = \underbrace{\begin{pmatrix} \frac{\partial N_1^\gamma}{\partial x} & 0 & \frac{\partial N_2^\gamma}{\partial x} & 0 & \frac{\partial N_3^\gamma}{\partial x} & 0 & \dots & \frac{\partial N_{m_v}^\gamma}{\partial x} & 0 \\ 0 & \frac{\partial N_1^\gamma}{\partial y} & 0 & \frac{\partial N_2^\gamma}{\partial y} & 0 & \frac{\partial N_3^\gamma}{\partial y} & \dots & 0 & \frac{\partial N_{m_v}^\gamma}{\partial x} \\ \frac{\partial N_1^\gamma}{\partial y} & \frac{\partial N_1^\gamma}{\partial x} & \frac{\partial N_2^\gamma}{\partial y} & \frac{\partial N_2^\gamma}{\partial x} & \frac{\partial N_3^\gamma}{\partial y} & \frac{\partial N_3^\gamma}{\partial x} & \dots & \frac{\partial N_{m_v}^\gamma}{\partial y} & \frac{\partial N_{m_v}^\gamma}{\partial x} \end{pmatrix}}_B \cdot \underbrace{\begin{pmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \\ u_3 \\ v_3 \\ \dots \\ u_{m_v} \\ v_{m_v} \end{pmatrix}}_{\vec{V}} \quad (327)$$

we have

$$\sigma_{xx} = -p + 2\eta \dot{\varepsilon}_{xx} \quad (328)$$

$$\sigma_{yy} = -p + 2\eta \dot{\varepsilon}_{yy} \quad (329)$$

$$\sigma_{xy} = +2\eta \dot{\varepsilon}_{xy} \quad (330)$$

so

$$\vec{\sigma} = - \begin{pmatrix} 1 \\ 1 \\ 0 \end{pmatrix} p + \mathbf{C} \cdot \vec{\dot{\boldsymbol{\varepsilon}}} = - \begin{pmatrix} 1 \\ 1 \\ 0 \end{pmatrix} \vec{N}^p \cdot \vec{P} + \mathbf{C} \cdot \mathbf{B} \cdot \vec{V} \quad (331)$$

with

$$\mathbf{C} = \eta \begin{pmatrix} 2 & 0 & 0 \\ 0 & 2 & 0 \\ 0 & 0 & 1 \end{pmatrix} \quad \text{or} \quad \mathbf{C} = \frac{\eta}{3} \begin{pmatrix} 4 & -2 & 0 \\ -2 & 4 & 0 \\ 0 & 0 & 3 \end{pmatrix} \quad (332)$$

check the right C

Finally the matrix \mathbf{N}^p is of size $3 \times m_p$:

$$\mathbf{N}^p = \begin{pmatrix} 1 \\ 1 \\ 0 \end{pmatrix} \vec{N}^p = \begin{pmatrix} \vec{N}^p \\ \vec{N}^p \\ 0 \end{pmatrix} \quad (333)$$

6.5 Solving the elastic equations

6.6 A quick tour of similar literature

- *Treatise on Geophysics*, Volume 7, Edited by D. Bercovici and G. Schubert: "Numerical Methods for Mantle Convection", by S.J. Zhong, D.A. Yuen, L.N. Moresi and M.G. Knepley. Note that it is a revision of the previous edition chapter by S.J. Zhong, D.A. Yuen and L.N. Moresi, Volume 7, pp. 227-252, 2007.
- *Computational Science I*, Lecture Notes for CAAM 519, M.G. Knepley, 2017. <https://cse.buffalo.edu/~knepley/classes/caam519/>
- *Numerical Modeling of Earth Systems - An introduction to computational methods with focus on-solid Earth applications of continuum mechanics*, Th.W. Becker and B.J.P. Kaus, 2018. <http://www-udc.ig.utexas.edu/external/becker/Geodynamics557.pdf>
- *Myths and Methods in Modeling*, M. Spiegelman, 2000. https://earth.usc.edu/~becker/teaching/557/reading/spiegelman_mmm.pdf

6.7 The case against the $Q_1 \times P_0$ element

What follows was written by Dave May and sent to me by email in May 2014. It captures so well the problem at hand that I have decided to reproduce it hereunder.

In the case of the incompressible Stokes equations, we would like to solve

$$\begin{pmatrix} \mathbb{K} & \mathbb{G} \\ \mathbb{G}^T & 0 \end{pmatrix} \begin{pmatrix} \vec{\mathcal{V}} \\ \vec{\mathcal{P}} \end{pmatrix} = \begin{pmatrix} \vec{f} \\ 0 \end{pmatrix}$$

with an iterative method which is algorithmically scalable and optimal. Scalable here would mean that the number of iterations doesn't grow as the mesh is refined. Optimal means the solution time varies linearly with the total number of unknowns. When using a stable element, If we right precondition the above system with

$$P = \begin{pmatrix} \mathbb{K} & \mathbb{G} \\ 0 & -\mathbb{S} \end{pmatrix}$$

then convergence will occur in 2 iterations, however this requires an exact solve on \mathbb{K} and on $\mathbb{S} = \mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot \mathbb{G}$ (\mathbb{S} is the pressure schur complement). In practice, people relax the ideal "two iteration" scenario by first replacing \mathbb{S} via $\mathbb{S}^* = \int \eta^{-1} \vec{N}^T \vec{N} dv$ (e.g. the pressure mass matrix scaled by the local inverse of viscosity).

$$P^* = \begin{pmatrix} \mathbb{K} & \mathbb{G} \\ 0 & -\mathbb{S}^* \end{pmatrix}$$

Using P^* , we obtain iteration counts which are larger than 2, but likely less than 10 - *however*, the number of iterations is independent of the mesh size. Replacing the exact \mathbb{K} solve in P^* again increases the iterations required to solve Stokes, but it's still independent of the number of elements. When you have this behaviour, we say the preconditioner (P^*) is spectrally equivalent to the operator (which here is Stokes)

The problem with $Q_1 \times P_0$ is that there are no approximations for \mathbb{S} which can be generated that ensure a spectrally equivalent P^* . Thus, as you refine the mesh using $Q_1 \times P_0$ elements, the iteration count ALWAYS grows. I worked on this problem during my thesis, making some improvements to the situation - however the problem still remains, it cannot be completely fixed and stems entirely from using unstable elements.

Citcom solvers works like this:

1. Solve $\mathbb{S} \cdot \mathcal{P} = \vec{f}'$ for pressure
2. Solve $\mathbb{K} \cdot \mathcal{V} = \vec{f} - \mathbb{G} \cdot \mathcal{P}$ for velocity

To obtain a scalable method, we need the number of iterations performed in (1) and (2) to be independent of the mesh. This means we need a spectrally equivalent preconditioner for \mathbb{S} and \mathbb{K} . Thus, we have the same issue as when you iterate on the full stokes system.

When we don't have a scalable method, it means increasing the resolution requires more cpu time in a manner which cannot be predicted. The increase in iteration counts as the mesh is refined can be dramatic.

If we can bound the number of iterations, AND ensure that the cost per iteration is linearly related to the number of unknowns, then we have a good method which can run on any mesh resolution with a predictable cpu time. Obtaining scalable and optimal preconditioners for \mathbb{K} is somewhat easier. Multigrid will provide us with this.

The reason citcom doesn't run with 400^3 elements is exactly due to this issue. I've added petsc support in citcom (when i was young and naive) - but the root cause of the non-scalable solve is directly caused by the element choice. Note that many of the high resolution citcom jobs are single time step calculations— there is a reason for that.

For many lithosphere dynamics problems, we need a reasonable resolution (at least 200^3 and realistically 400^3 to 800^3). Given the increase in cost which occurs when using Q1P0, this is not achievable, as the citcom code has demonstrated. Note that citcom is 20 years old now and for its time, it was great, but we know much more now and we know how to improve on it. As a result of this realization, I dumped all my old Q1P0 codes (and Q1Q1 codes, but for other reasons) in the trash and started from scratch. The only way to make something like 800^3 tractable is via iterative, scalable and optimal methods and that mandates stable elements. I can actually run at something like 1000^3 (nodal points) these days because of such design choices.

6.8 Isoviscous Stokes for incompressible flow

We start from the momentum equation:

$$-\vec{\nabla}p + \vec{\nabla} \cdot (2\eta\dot{\varepsilon}^d(\vec{v})) + \rho\vec{g} = \vec{0} \quad (334)$$

When the viscosity is constant in space, it can be taken out of the divergence operator:

$$-\vec{\nabla}p + 2\eta\vec{\nabla} \cdot \dot{\varepsilon}^d(\vec{v}) + \rho\vec{g} = \vec{0} \quad (335)$$

Let us for simplicity look at a 2D Cartesian formulation of this equation and for incompressible flow:

$$2\vec{\nabla} \cdot \dot{\varepsilon}^d(\vec{v}) = \vec{\nabla} \cdot (\vec{\nabla}\vec{v} + \vec{\nabla}\vec{v}^T) \quad (336)$$

$$= (\partial_x \partial_y) \cdot \begin{pmatrix} \partial_x u & \partial_x v \\ \partial_y u & \partial_y v \end{pmatrix} + (\partial_x \partial_y) \cdot \begin{pmatrix} \partial_x u & \partial_y u \\ \partial_x v & \partial_y v \end{pmatrix} \quad (337)$$

$$= (\partial_x^2 u + \partial_y^2 u, \partial_x^2 v + \partial_y^2 v) + (\partial_x \partial_x u + \partial_y \partial_x v, \partial_x \partial_y u + \partial_y \partial_y v) \quad (338)$$

$$= (\partial_x^2 u + \partial_y^2 u, \partial_x^2 v + \partial_y^2 v) + (\partial_x \underbrace{(\partial_x u + \partial_y v)}_{=0}, \partial_y \underbrace{(\partial_x u + \partial_y v)}_{=0}) \quad (339)$$

$$= (\partial_x^2 u + \partial_y^2 u, \partial_x^2 v + \partial_y^2 v) \quad (340)$$

and then finally the Stokes equation is:

$$-\vec{\nabla}p + \eta\Delta\vec{v} + \rho\vec{g} = \vec{0} \quad (341)$$

The mass conservation equation remains unchanged and so does the pressure gradient term. We shall then focus on the weak form of the previously obtained term. We multiply it by a velocity test function

N_i^ν and integrate over an element:

$$\begin{aligned}
& \int_{\Omega_e} N_i^\nu \Delta \vec{v}^h dV \\
&= \int_{\Omega_e} \begin{pmatrix} N_i^\nu \Delta u^h \\ N_i^\nu \Delta v^h \end{pmatrix} dV \\
&= \int_{\Omega_e} \begin{pmatrix} N_i^\nu \vec{\nabla} \cdot \vec{\nabla} u^h \\ N_i^\nu \vec{\nabla} \cdot \vec{\nabla} v^h \end{pmatrix} dV \\
&= \int_{\Omega_e} \begin{pmatrix} \vec{\nabla} N_i^\nu \cdot \vec{\nabla} u^h \\ \vec{\nabla} N_i^\nu \cdot \vec{\nabla} v^h \end{pmatrix} dV \\
&= \int_{\Omega_e} \begin{pmatrix} \partial_x N_i^\nu \partial_x u^h + \partial_y N_i^\nu \partial_y u^h \\ \partial_x N_i^\nu \partial_x v^h + \partial_y N_i^\nu \partial_y v^h \end{pmatrix} dV \\
&= \int_{\Omega_e} \begin{pmatrix} \partial_x N_i^\nu & \partial_y N_i^\nu & 0 & 0 \\ 0 & 0 & \partial_x N_i^\nu & \partial_y N_i^\nu \end{pmatrix} \cdot \begin{pmatrix} \partial_x u^h \\ \partial_y u^h \\ \partial_x v^h \\ \partial_y v^h \end{pmatrix} dV \\
&= \int_{\Omega_e} \begin{pmatrix} \partial_x N_i^\nu & \partial_y N_i^\nu & 0 & 0 \\ 0 & 0 & \partial_x N_i^\nu & \partial_y N_i^\nu \end{pmatrix} \cdot \begin{pmatrix} \partial_x N_1^\nu & 0 & \partial_x N_2^\nu & 0 & \cdots & \partial_x N_{m_\nu}^\nu & 0 \\ \partial_y N_1^\nu & 0 & \partial_y N_2^\nu & 0 & \cdots & \partial_y N_{m_\nu}^\nu & 0 \\ 0 & \partial_x N_1^\nu & 0 & \partial_x N_2^\nu & \cdots & 0 & \partial_x N_{m_\nu}^\nu \\ 0 & \partial_y N_1^\nu & 0 & \partial_y N_2^\nu & \cdots & 0 & \partial_y N_{m_\nu}^\nu \end{pmatrix} \cdot \underbrace{\begin{pmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \\ \vdots \\ u_{m_\nu} \\ v_{m_\nu} \end{pmatrix}}_{\vec{v}} dV
\end{aligned}$$

Writing this equation for $i = 1, \dots, m_\nu$, we obtain:

$$\int \underbrace{\begin{pmatrix} \partial_x N_1^\nu & \partial_y N_1^\nu & 0 & 0 \\ 0 & 0 & \partial_x N_1^\nu & \partial_y N_1^\nu \\ \partial_x N_2^\nu & \partial_y N_2^\nu & 0 & 0 \\ 0 & 0 & \partial_x N_2^\nu & \partial_y N_2^\nu \\ \vdots & \vdots & \vdots & \vdots \\ \partial_x N_{m_\nu}^\nu & \partial_y N_{m_\nu}^\nu & 0 & 0 \\ 0 & 0 & \partial_x N_{m_\nu}^\nu & \partial_y N_{m_\nu}^\nu \end{pmatrix}}_{\mathbf{B}} \cdot \underbrace{\begin{pmatrix} \partial_x N_1^\nu & 0 & \partial_x N_2^\nu & 0 & \cdots & \partial_x N_{m_\nu}^\nu & 0 \\ \partial_y N_1^\nu & 0 & \partial_y N_2^\nu & 0 & \cdots & \partial_y N_{m_\nu}^\nu & 0 \\ 0 & \partial_x N_1^\nu & 0 & \partial_x N_2^\nu & \cdots & 0 & \partial_x N_{m_\nu}^\nu \\ 0 & \partial_y N_1^\nu & 0 & \partial_y N_2^\nu & \cdots & 0 & \partial_y N_{m_\nu}^\nu \end{pmatrix}}_{\mathbf{B}^T} \cdot \underbrace{\begin{pmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \\ \vdots \\ u_{m_\nu} \\ v_{m_\nu} \end{pmatrix}}_{\vec{v}} dV$$

or,

$$\mathbf{K}_\eta = \eta \int_{\Omega_e} \mathbf{B}^T \cdot \mathbf{B} dV$$

where \mathbf{B} is a $(ndim * ndim) \times (m_v * ndofV)$ matrix (see also Eq. 6.24 of [533]).

In three dimensions, the matrix \mathbf{B} is given by

$$\begin{pmatrix} \partial_x N_1^\nu & 0 & \partial_x N_2^\nu & 0 & \cdots & \partial_x N_{m_\nu}^\nu & 0 \\ \partial_y N_1^\nu & 0 & \partial_y N_2^\nu & 0 & \cdots & \partial_y N_{m_\nu}^\nu & 0 \\ \partial_z N_1^\nu & 0 & \partial_z N_2^\nu & 0 & \cdots & \partial_z N_{m_\nu}^\nu & 0 \\ 0 & \partial_x N_1^\nu & 0 & \partial_x N_2^\nu & \cdots & 0 & \partial_x N_{m_\nu}^\nu \\ 0 & \partial_y N_1^\nu & 0 & \partial_y N_2^\nu & \cdots & 0 & \partial_y N_{m_\nu}^\nu \\ 0 & \partial_z N_1^\nu & 0 & \partial_z N_2^\nu & \cdots & 0 & \partial_z N_{m_\nu}^\nu \end{pmatrix}$$

7 The Discontinuous Galerkin Finite Element Method (DG-FEM)

7.1 First-order advection ODE in 1D

What follows is borrowed from the book "Discontinuous finite elements in fluid dynamics and heat transfer" by Ben Q. Li [1176].

To illustrate the basic ideas of the discontinuous finite element method, we consider a simple, one-dimensional, first order differential equation with u specified at one of the boundaries:

$$\frac{du}{dx} + g = 0 \quad x \in [a, b] \quad \text{and} \quad u(x = a) = u_a \quad (342)$$

where g is a constant (for simplicity). The domain is discretized such that : $\Omega_j = [x_j, x_{j+1}]$ with $j = 1, 2, \dots, nel$. Then, integrating the above equation over the element j with respect to a weighting function $f(x)$

$$\int_{x_j}^{x_{j+1}} \left(\frac{du}{dx} + g \right) f(x) dx = 0 \quad (343)$$

Remembering that $\int_c^d u(x)v'(x)dx = [u(x)v(x)]_c^d - \int_c^d u'(x)v(x)dx$, we can now perform an integration by parts on the differential operator and we obtain:

$$[u(x)f(x)]_{x_j}^{x_{j+1}} - \int_{x_j}^{x_{j+1}} \left(u \frac{df}{dx} - gf(x) \right) dx = 0 \quad (344)$$

or,

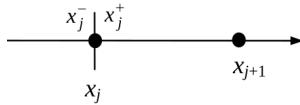
$$u(x_{j+1})f(x_{j+1}) - u(x_j)f(x_j) - \int_{x_j}^{x_{j+1}} \left(u \frac{df}{dx} - gf(x) \right) dx = 0 \quad (345)$$

On Ω_j u is approximated by $u_h \in H$, H being an appropriate function space of finite dimension, and f by f_h taken from the same function space as u_h . Upon substituting (u_h, f_h) for (u, f) in the equation above, we have the discontinuous Galerkin finite element formulation:

$$u_h(x_{j+1})f_h(x_{j+1}) - u_h(x_j)f_h(x_j) - \int_{x_j}^{x_{j+1}} \left(u_h \frac{df_h}{dx} - gf_h(x) \right) dx = 0 \quad (346)$$

In the continuous finite element approach, the field variable u_h is forced to be continuous across the boundary. The essential idea for the discontinuous method is that u_h is allowed to be discontinuous across the boundary. Therefore, across the element, the following two different values are defined at the two sides of the boundary:

$$u_j^+ = \lim_{x \searrow x_j^+} u_h(x) \quad u_j^- = \lim_{x \nearrow x_j^-} u_h(x) \quad (347)$$



An illustration of the jump across x_j of element j : x_j and x_{j+1} mark the boundaries of the element

Conversely, we also have:

$$u_{j+1}^+ = \lim_{x \searrow x_{j+1}^+} u_h(x) \quad u_{j+1}^- = \lim_{x \nearrow x_{j+1}^-} u_h(x) \quad (348)$$

It is key to remember that 1) u_h is discontinuous only at the element boundaries; 2) the solution u is smooth within (but excluding) the boundary. By this definition, the above equation contains the variables only within the integral limits of Ω_j . As a consequence, there is no direct coupling with other intervals or other elements. *The field values at a node, or the interface between two elements, are not unique.* They are calculated using the two limiting values approaching the interface from the two adjacent elements. This feature is certainly desirable for problems with internal discontinuities.

We can now write CHECK CHECK

$$u_{j+1}^- f_h(x_{j+1}) - u_j^+ f_h(x_j) - \int_{x_j}^{x_{j+1}} \left(u_h \frac{df_h}{dx} - g f_h(x) \right) dx = 0 \quad (349)$$

and we can integrate by parts again the term which contains a derivative:

$$\int_{x_j}^{x_{j+1}} u_h(x) \frac{df_h}{dx} dx = [u_h f_h] - \int_{x_j}^{x_{j+1}} f_h(x) \frac{du_h}{dx} dx$$

and then

$$u_{j+1}^- f_h(x_{j+1}) - u_j^+ f_h(x_j) - \int_{x_j}^{x_{j+1}} \left(u_h \frac{df_h}{dx} - g f_h(x) \right) dx = 0 \quad (350)$$

We start from the simplest ODE:

$$\frac{du}{dx} = 1 \quad x \in [0, 2] \quad \text{and} \quad u(x=0) = 0$$

As in the Continuous Galerkin case, the function f is replaced by a shape function $N_i(x)$:

How the hell do I arrive at:

$$\int_{x_j}^{x_{j+1}} \left(\frac{du_h}{dx} + g(u_h) \right) N_i(x) dx + (u_j^+ - u_j^-) N_i(x_j) = 0$$

7.2 Steady state diffusion in 1D

Let us start simple with the 1D steady state heat conduction problem in 1D, given by the following equation:

$$\frac{d^2T}{dx^2} = 0 \quad T(x=0) = 0 \quad T(x=1) = 1 \quad \text{on } x \in [0, 1] \quad (351)$$

Although this equation is usually solved as is with its second-order derivative, it can also be written in a mixed form, using the heat flux q (a scalar in 1D):

$$-\frac{dq}{dx} = 0 \quad q - \frac{dT}{dx} = 0 \quad x \in [0, 1]$$

and the boundary conditions remain unchanged.

We apply the standard approach to establish the weak forms of these two first-order ODEs, and we do so on an element e bound by nodes k and $k+1$ with coordinates x_k and x_{k+1}

$$\begin{aligned} -\int_{x_k}^{x_{k+1}} \frac{dq}{dx} \tilde{f}(x) dx &= -[q \tilde{f}]_{x_k}^{x_{k+1}} + \int_{x_k}^{x_{k+1}} \frac{d\tilde{f}}{dx} q(x) dx = 0 \\ \int_{x_k}^{x_{k+1}} \left(q - \frac{dT}{dx} \right) \bar{f}(x) dx &= \int_{x_k}^{x_{k+1}} q(x) \bar{f}(x) dx - [T \bar{f}]_{x_k}^{x_{k+1}} + \int_{x_k}^{x_{k+1}} \frac{d\bar{f}}{dx} T(x) dx = 0 \end{aligned}$$

where \tilde{f} and \bar{f} are test functions. We now must examine the term between square brackets. Inside the element, the test functions \tilde{f} and \bar{f} are well defined polynomials and we we coin:

$$\tilde{f}_k^+ = \tilde{f}(x_k^+) \quad \tilde{f}_{k+1}^- = \tilde{f}(x_{k+1}^-) \quad \bar{f}_k^+ = \bar{f}(x_k^+) \quad \bar{f}_{k+1}^- = \bar{f}(x_{k+1}^-)$$

Concerning q and T , we will for now give them values \hat{q}_k and \hat{T}_k at node k and \hat{q}_{k+1} and \hat{T}_{k+1} at node $k+1$, and we will specify the hat quantities as follows:

$$\begin{aligned} \hat{T}_k &= \begin{cases} T_k^- & k = 1 \\ \frac{1}{2}(T_k^- + T_k^+) + \mathcal{C}(T_k^- - T_k^+) & k = 2, \dots, N-1 \\ T_k^+ & k = N \end{cases} \\ \hat{q}_k &= \begin{cases} q_k^+ - \mathcal{E}(T_k^- - T_k^+) & k = 1 \\ \frac{1}{2}(q_k^+ + q_k^-) - \mathcal{E}(T_k^- - T_k^+) - \mathcal{C}(q_k^- - q_k^+) & k = 2, \dots, N-1 \\ q_k^- - \mathcal{E}(T_k^- - T_k^+) & k = N \end{cases} \end{aligned} \quad (352)$$

where N is the number of nodes and where \mathcal{C} and \mathcal{E} are two constants.

Remark. Note that $\hat{T}_k = T_1^-$ on the left boundary is consistent with $\hat{T}_k = \frac{1}{2}(T_k^- + T_k^+) + \mathcal{C}(T_k^- - T_k^+)$ provided $T_1^- = T_1^+$. The same goes for the right boundary, and the same reasoning applies for the heat flux terms \hat{q}_k .

Inside an element bounded by nodes k and $k+1$, the temperature T and heat flux q are interpolated over an isoparametric linear element:

$$T_h(x) = N_k(x)T_k^+ + N_{k+1}(x)T_{k+1}^-$$

$$q_h(x) = N_k(x)q_k^+ + N_{k+1}(x)q_{k+1}^-$$

As in the (Continuous) Galerkin case of section 5.1, the test functions are taken to be the shape functions, and in this case for both temperature and flux.

$$\begin{aligned}
0 &= -\hat{q}_{k+1}\tilde{f}(x_{k+1}^-) + \hat{q}_k\tilde{f}(x_k^+) + \int_{x_k}^{x_{k+1}} \frac{d\tilde{f}}{dx} q_h(x) dx \\
&= -\hat{q}_{k+1}\tilde{f}_{k+1}^- + \hat{q}_k\tilde{f}_k^+ + \int_{x_k}^{x_{k+1}} \frac{d\tilde{f}}{dx} (N_k(x)q_k^+ + N_{k+1}(x)q_{k+1}^-) dx \\
&= -\hat{q}_{k+1}\tilde{f}_{k+1}^- + \hat{q}_k\tilde{f}_k^+ + \int_{x_k}^{x_{k+1}} \frac{d\tilde{f}}{dx} N_k(x) dx \cdot q_k^+ + \int_{x_k}^{x_{k+1}} \frac{d\tilde{f}}{dx} N_{k+1}(x) dx \cdot q_{k+1}^- \\
&= -\left(\frac{1}{2}(q_{k+1}^+ + \textcolor{blue}{q}_{k+1}^-) - \mathcal{E}(\textcolor{blue}{T}_{k+1}^- - T_{k+1}^+) - \mathcal{C}(\textcolor{blue}{q}_{k+1}^- - q_{k+1}^+)\right)\tilde{f}_{k+1}^- + \left(\frac{1}{2}(\textcolor{blue}{q}_k^+ + q_k^-) - \mathcal{E}(T_k^- - \textcolor{blue}{T}_k^+) - \mathcal{C}(q_k^- - \textcolor{blue}{q}_k^+)\right)\tilde{f}_k^+ \\
&\quad + \int_{x_k}^{x_{k+1}} \frac{d\tilde{f}}{dx} N_k(x) dx \cdot \textcolor{blue}{q}_k^+ + \int_{x_k}^{x_{k+1}} \frac{d\tilde{f}}{dx} N_{k+1}(x) dx \cdot \textcolor{blue}{q}_{k+1}^- \\
&= \left[-(0.5 + \mathcal{C})\tilde{f}_{k+1}^- + \int_{x_k}^{x_{k+1}} \frac{d\tilde{f}}{dx} N_{k+1}(x) dx\right] \textcolor{blue}{q}_{k+1}^- - \mathcal{E}\tilde{f}_{k+1}^-\textcolor{blue}{T}_{k+1}^- \\
&\quad + \left[(0.5 + \mathcal{C})\tilde{f}_k^+ + \int_{x_k}^{x_{k+1}} \frac{d\tilde{f}}{dx} N_{k+1}(x) dx\right] \textcolor{blue}{q}_k^+ + \mathcal{E}\tilde{f}_k^+\textcolor{blue}{T}_k^+ \\
&\quad - \left(\frac{1}{2}q_{k+1}^+ + \mathcal{E}T_{k+1}^+ + \mathcal{C}q_{k+1}^+\right)\tilde{f}_{k+1}^- + \left(\frac{1}{2}q_k^- - \mathcal{E}T_k^- - \mathcal{C}q_k^-\right)\tilde{f}_k^+
\end{aligned} \tag{353}$$

$$\begin{aligned}
0 &= -[T\bar{f}]_{x_k}^{x_{k+1}} + \int_{x_k}^{x_{k+1}} q_h(x)\bar{f}(x) dx + \int_{x_k}^{x_{k+1}} \frac{d\bar{f}}{dx} T_h(x) dx \\
&= -\hat{T}_{k+1}\bar{f}(x_{k+1}^-) + \hat{T}_k\bar{f}(x_k^+) + \int_{x_k}^{x_{k+1}} q_h(x)\bar{f}(x) dx + \int_{x_k}^{x_{k+1}} \frac{d\bar{f}}{dx} T_h(x) dx \\
&= -\left(\frac{1}{2}(\textcolor{blue}{T}_{k+1}^- + T_{k+1}^+) + \mathcal{C}(\textcolor{blue}{T}_{k+1}^- - T_{k+1}^+)\right)\bar{f}_{k+1}^- + \left(\frac{1}{2}(T_k^- + \textcolor{blue}{T}_k^+) + \mathcal{C}(T_k^- - \textcolor{blue}{T}_k^+)\right)\bar{f}_k^+ \\
&\quad + \int_{x_k}^{x_{k+1}} (N_k(x)\textcolor{blue}{q}_k^+ + N_{k+1}(x)\textcolor{blue}{q}_{k+1}^-)\bar{f}(x) dx + \int_{x_k}^{x_{k+1}} \frac{d\bar{f}}{dx} (N_k(x)\textcolor{blue}{T}_k^+ + N_{k+1}(x)\textcolor{blue}{T}_{k+1}^-) dx
\end{aligned}$$

If I now rewrite both equations above by taking $\tilde{f} = \{N_k, N_{k+1}\}$ and $\bar{f} = \{N_k, N_{k+1}\}$ then we obtain four equations. For each element with the four unknowns $\textcolor{blue}{q}_k^+$, $\textcolor{blue}{q}_{k+1}^-$, $\textcolor{blue}{T}_k^+$ and $\textcolor{blue}{T}_{k+1}^-$. All other q and T quantities in the above equations will need to find their way to the rhs.

We will also use the results obtained in Appendix I:

$$\mathbf{M}^e = \int_{\Omega_e} \vec{N}^T \vec{N} dV = \int_{\Omega_e} \begin{pmatrix} N_k N_k & N_k N_{k+1} \\ N_{k+1} N_k & N_{k+1} N_{k+1} \end{pmatrix} dV = \frac{h}{2} \frac{1}{3} \begin{pmatrix} 2 & 1 \\ 1 & 2 \end{pmatrix} = \frac{h}{6} \begin{pmatrix} 2 & 1 \\ 1 & 2 \end{pmatrix}$$

and also

$$\int_{\Omega_e} \begin{pmatrix} \frac{dN_k}{dx} N_k & \frac{dN_k}{dx} N_{k+1} \\ \frac{dN_{k+1}}{dx} N_k & \frac{dN_{k+1}}{dx} N_{k+1} \end{pmatrix} dV = \frac{1}{2} \begin{pmatrix} -1 & -1 \\ 1 & 1 \end{pmatrix}$$

7.3 Time-dependent diffusion PDE in 1D

7.4 Time-dependent (pure) advection PDE in 1D

7.5 Steady-state diffusion in 2D

7.6 Time-dependent diffusion PDE in 2D

8 Additional techniques, features, measurements

Solving the Stokes equations and the energy equations is one thing. Doing it in a geodynamical context requires a lot of additional techniques.

8.1 Dealing with a free surface (and mesh deformation)

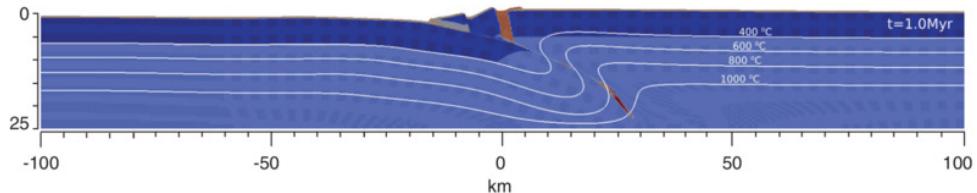
When carrying out global models, typically mantle convection, the effect of the free surface is often neglected/negligible: topography ranges from $\sim 10\text{km}$ depth to $\sim 10\text{km}$ height, which is very small compared to the depth of the mantle ($\sim 3000\text{km}$).

However, it has long been recognised that there is a feedback between topography and crust/lithosphere deformation: the surface of the Earth reflects the deeper processes, from orogeny, back-arc basins, rifts, mid-ocean ridges, etc ... (see for instance [227]).

Remark. *Free surface flows are not unique to Earth sciences, and their modelling has given rise to many studies and textbooks. A typical free-surface flow problem in the CFD literature is the so-called 'dam break' problem [1332, 73, 1198, 1146, 912, 26]. Other occurrences involve sea waves, flow over structures, flow around ships, mould filling, flow with bubbles [1198].*

What distinguishes geodynamics free surface modelling from its engineering counterpart is (i) the absence of surface tension, (ii) the fact that the fluids under consideration are Stokesian, (iii) their rheology is complex (the elastic and plastic components can be dominant at the surface).

The problem of dealing with a free surface can be deceptively simple at first glance: as mentioned before the amplitude of surface movement is often less than 1% of the domain size. Isostacy-driven movements are easy to deal with since the movement is vertical (and often characterised by long wavelength). However, computational problems quickly arise in subduction modelling: the downgoing lithosphere subducts below the overriding plate and the relative convergence of the two is likely to generate a cusp at the trench. The presence of shear bands intersecting the surface accentuates the problem:

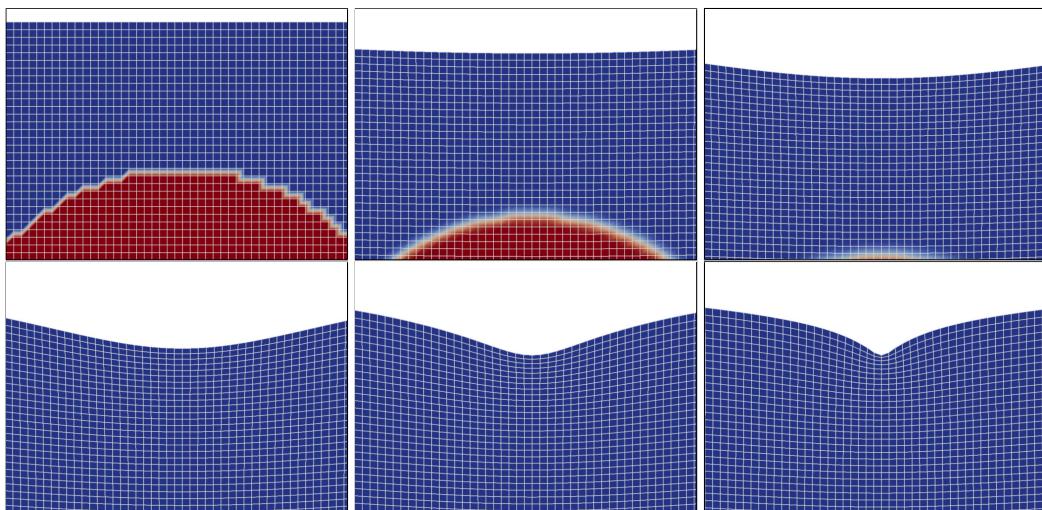


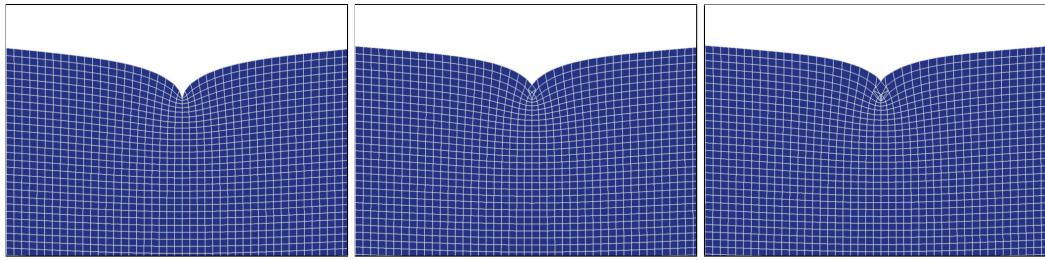
Taken from Maffione et al [1242]. Example of free surface deformation above intra-oceanic subduction initiation

Remark. *It is difficult to talk about free surface without including the underlying mesh. What follows should be read alongside Section 8.10.*

8.1.1 The fully Lagrangian approach

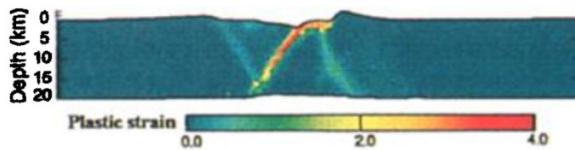
In this case the mesh is deformed with the velocity (or displacement) computed on its nodes. It is sometimes called 'body fitting' [442] or 'boundary fitted'. In the case when large deformation occurs (which is rather frequent in geodynamics - think about subduction or rifting processes where materials end up moving 100's or 1000's of km, horizontally and/or vertically), it leads to highly deformed elements, and in some cases even bow-tied:



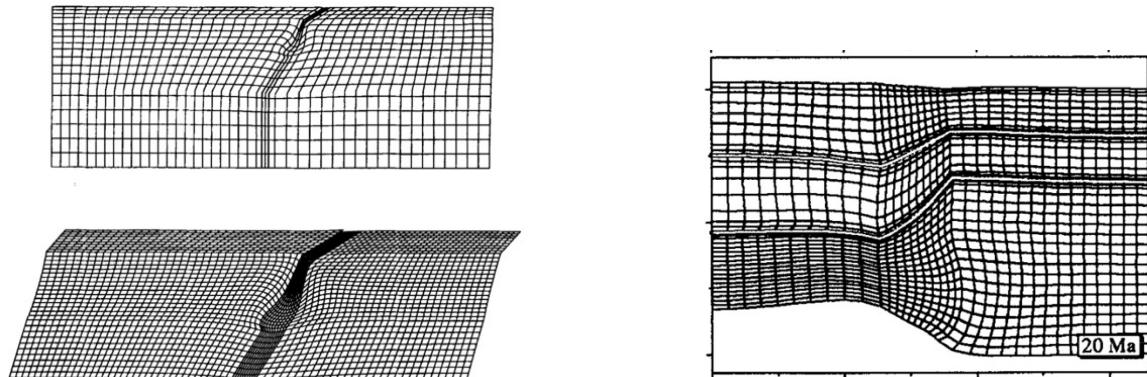


Example of a free surface evolution above a sinking sphere. The isostatic rebound above the sphere generates a cusp which, if no special measure is taken, ultimately leads to a bow-tied element. Once this occurs the simulation stops since the mapping of the bow-tied element to the reference element yields to wrong elemental matrix. Courtesy of M. Fraters

In the mildest cases this does not occur but it has long been established that large mesh deformation yields low accuracy calculations, especially when angles between edges become small or large. One way to overcome this problem is to remesh, i.e. generate a better mesh based on the available information on the deformed one. In 2D this is routinely done, especially when triangular elements are used. In 3D, multiple remeshing are very costly and it is generally avoided. Note also that re-meshing often involves some form of interpolation and therefore some unwanted numerical diffusion. When deformation is reasonably small, fully lagrangian methods work and have been used in geodynamics [875, 1309, 1137].



Taken from [1137]. Upper-crustal faulting, note that the bottom and the top surface are deformed.



Taken from [837]. Subduction model, topographic expression is shown without vertical exaggeration.

Taken from [792]. Asymmetric lithospheric extension.

GET: Crook et al. 2006, and references therein [452]) Beaumont et al. 1994; [97]

8.1.2 The Eulerian approach: using sticky air

Sticky air is the default option for numerical methods which mesh cannot be deformed (typically the finite difference method). In this case, the air above the crust/sediments is modelled as a zero-density fluid with very low viscosity (see for instance the early article by Zaleski and Julien [1990]). One problem quickly arises when one realises that the viscosity of the air ($\sim 18.5 \cdot 10^{-6}$ Pa·s²⁴) is almost 25-30 orders of magnitude lower than the (effective) viscosity of Earth materials. Real air viscosity cannot therefore be used because of 1) round-off errors, 2) extremely poorly-conditioned matrices. Low viscosities around $10^{16} - 10^{19}$ Pa·s are then commonly used as they are still negligible next to those of the (plastic) crust, and the flow of air parallel to Earth materials only generates extremely small shear and normal stress values (thereby approaching the true nature of a free surface). This approach is the one employed in all

²⁴<https://en.wikipedia.org/wiki/Viscosity>

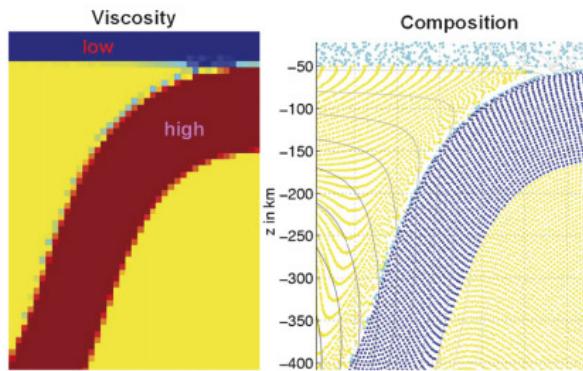
the papers based on the I2/I3(EL)VIS code (see Appendix B) and has been benchmarked in Crameri et al. [442].

This approach has a few advantages:

1. it is simple to implement
2. it is compatible with all the standard numerical methods (FEM, FDM,FVM)
3. it avoids (potentially complicated) remeshing

and quite a few drawbacks:

1. it increases the size of the computational domain, thereby adding more unknowns to the linear system: in [1638] the air layer is set to 50km while the lithospheric domain underneath is 700km thick;
2. it requires the use of averaging all along the free-surface where very large viscosity contrasts are present. Here is what Poliakov and Podlachikov [1491] say about the sticky air method: "Zaleski & Julien [1990] used a top layer with a very low viscosity and density to represent air or water above the surface. This allows a simple representation of the free surface. However, due to the very high viscosity and density contrast and diffusion between the top layer and the underlying layers, calculations sometimes become unstable and give significant errors."
3. it can showcase air entrainment:



Taken from [1638]. Details of the entrainment and lubrication of the soft surface layer. Light blue particles are sticky air particle and are found to greatly alter the viscosity of the subduction channel.

4. it is not clear how thick the air layer must be
5. it often requires to ascribe thermal parameters to the air;
6. it makes the implementation of Dirichlet or Neuman boundary conditions for temperature at the surface less obvious.
7. it makes the coupling with surface processes codes less straightforward.
8. its accuracy depends on the method used to track materials in the rest of the code (markers, level sets, ...). If markers are used, the free surface position is then known up to the average distance between markers.
9. it negatively impacts the condition number of the matrix.

The sticky air approach is employed by various codes in the subduction benchmark study [1638]

The term 'sticky water' is sometimes employed too. The dynamic viscosity of water is about 10^{-3} Pa·s so that it is also negligible compared to the viscosity of Earth materials and the same reasoning as air applies. However, in such a case a density of about 1000kg/m^3 is then assigned to the layer. REF?

In conclusion, as stated in [442]: "the sticky air method is a good way to simulate a free surface for Eulerian approaches, provided that its parameters are chosen carefully."

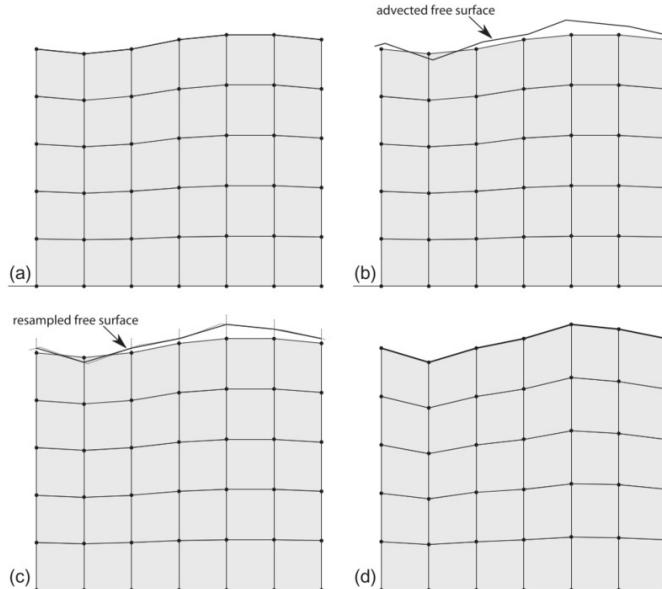
8.1.3 The Arbitrary Lagrangian Eulerian (ALE) approach

It is a very widely used approach in FEM-based geodynamics codes but originates in the field of CFD [904, 939] and is described at length in [1683, 532, 533]. To put it very simply, the key idea in the ALE formulation is the introduction of a computational mesh which can move and deform with a velocity independent of the velocity carried by the material particles.

The simple approach in [1756]. What follows is written with a 2D Cartesian model in mind ($Q_1 \times P_0$ elements are used). The computational domain is a rectangle of size $L_x \times L_y$ and a $n_{nx} \times n_{ny}$ rectangular grid spanning the simulation domain is generated. The grid points constituting the top row of the grid define the discrete free surface of the domain. Once the Eulerian velocity field has been computed on these, their position is first updated using a simple Eulerian advection step (see a,b on figure hereunder):

$$\vec{r}_i(t + \delta t) = \vec{r}_i(t) + \vec{v}_i \cdot \delta t \quad i = 1, \dots, n_{nx}$$

The other boundaries of the system remain fixed at locations $x = 0$, $x = L_x$ and $y = 0$. Even though the Eulerian grid must conform to the current domain shape, only vertical motion of grid nodes is allowed. It is therefore necessary to resample the predicted free surface given by \vec{r}_i at equidistant positions between $x = 0$ and $x = L_x$. The resampling is carried out either with Spline functions or a moving least square algorithm. Finally, the vertical position of all the nodes corresponding to column $i \in [1, n_{nx}]$ is recalculated so that they are equidistant, as sketched in Figure d. This has the advantage of keeping the mesh distortion to a minimum in the case of large deformation.



The ALE algorithm of [1756] in 2D. (a) Grid and free surface at a given time t ; (b) advection of the free surface; (c) resampling of the free surface at equidistant abscissae; (d) vertical adjustment of grid nodes in each column at equidistant ordinates.

The ALE method is used in the SOPALE, SULEC, FANTOM, ELEFANT, and ASPECT codes to name a few (see Appendix B).

The not-so-simple but rather elegant approach of ASPECT What follows is mostly borrowed from Rose et al [1599]. Their approach has the advantage that it does not presuppose a geometry (Cartesian, Spherical, ...) nor a number of dimensions. It is also designed to work in parallel and on octree-based meshes, and with various combinations of boundary conditions. Note that the authors specify that "for moderate mesh deformation, the mesh stays smooth and well conditioned, though it breaks down for large deformations".

This approach is obtained by simply imposing the obvious condition that no particle (fluid parcel) can cross the free surface (because it is a material surface). This can be imposed in a straightforward

manner by using a Lagrangian description along this surface. However, this condition may be relaxed by imposing only the necessary condition: \vec{v} equal to zero along the normal to the boundary (ie. $\vec{n} \cdot \vec{v} = 0$, where \vec{n} is the outward unit normal to the fluid domain). The mesh position, normal to the free surface, is determined from the normal component of the particle velocity and remeshing can be performed along the tangent; see, for instance Huerta and Liu, 1989 [937] or Braess and Wriggers, 2000 [210].

As mentioned above the mesh velocity in normal direction at the free surface (with unit normal \vec{n}) has to be consistent with the velocity of the Stokes velocity solution $\vec{v}(t)$:

$$\vec{v}_{\text{mesh}}(t) \cdot \vec{n} = \vec{v}(t) \cdot \vec{n} \quad \text{on } \Gamma_F \quad (354)$$

In ALE calculations the internal mesh velocity is usually undetermined, but one wants to smoothly deform the mesh so as to preserve its regularity, avoiding inverted or otherwise poorly conditioned cells. The mesh deformation can be calculated in many different ways, including algebraic (as mentioned in the previous paragraph) and PDE based approaches. The latter is chosen here. The Laplace equation is solved where the unknown is the mesh velocity, i.e. one must solve:

$$\Delta \vec{v}_{\text{mesh}} = 0 \quad (355)$$

subjected to the following boundary conditions:

$$\vec{v}_{\text{mesh}} = \vec{0} \quad \text{on } \Gamma_0 \quad (356)$$

$$\vec{v}_{\text{mesh}} = (\vec{v} \cdot \vec{n})\vec{n} \quad \text{on } \Gamma_F \quad (357)$$

$$\vec{v}_{\text{mesh}} \cdot \vec{n} = 0 \quad \text{on } \Gamma_{FS} \quad (358)$$

where Γ_{FS} is the part of the boundary with free slip boundary conditions, Γ_0 is the no-slip part and Γ_{FS} is the free slip part.

Once the mesh velocity has been obtained for all mesh points, these can be moved with said velocity. However, it must be noted that the multiple occurrences of the normal vector in the above equations is not without problem as the normal vectors are not well defined on the mesh vertices, which is where the mesh velocity is defined.

This yields what the author coin the 'quasi-implicit' scheme (we have so far neglected any kind of stabilisation):

1. Solve the Stokes system;
2. Solve for the surface mesh velocity using Equation 359;
3. Solve for the internal mesh velocity using Equations 355, 358;
4. Advect the mesh forward in time using displacements determined by the forward Euler scheme:
 $\vec{x}(t^{n+1}) = \vec{x}(t^n) + \vec{v}_{\text{mesh}}\delta t$.

Note that Rose et al (2017) [1599] go further than this, propose a 'nonstandard finite difference scheme' and make a link with the stabilisation presented in Kaus et al (2010) [1032].

The authors list two simple methods of computing the normals:

- one can take \vec{n} as the direction of the local vertical,
- one could compute \vec{n} as some weighted average of the cell normals adjacent to a given vertex

but conclude that they have found that these schemes do not necessarily have good mass conservation properties.

A better approach is proposed in the form of an L_2 projection of the normal velocity $\vec{v} \cdot \vec{n}$ onto the free surface Γ_F . Multiplying the boundary conditions

$$\vec{v}_{\text{mesh}} = (\vec{v} \cdot \vec{n})\vec{n}$$

by a test function \vec{w} and integrating over the free surface part of the boundary, we find:

$$\int_{\Gamma_F} \vec{w} \cdot \vec{v}_{\text{mesh}} d\Gamma = \int_{\Gamma_F} \vec{w} \cdot (\vec{v} \cdot \vec{n})\vec{n} d\Gamma = \int_{\Gamma_F} (\vec{w} \cdot \vec{n})(\vec{v} \cdot \vec{n}) d\Gamma \quad (359)$$

When discretized, this forms a linear system which can be solved for the mesh velocity \vec{v}_{mesh} at the free surface. This system, being nonzero over only the free surface, is relatively computationally inexpensive to solve. The authors unfortunately fail to mention that this approach is particularly interesting since the numerical quadrature used to compute the above integrals require the normal \vec{n} between the nodes and these normals are well defined over each segment joining two nodes!²⁵

In what follows I present in some detail how to carry out the L_2 projection to arrive at the surface velocity for both Q_1 and Q_2 elements.

I start from the following integral over a Q_1 element:

$$\int_{\Gamma_e} N_i \vec{v}_{\text{mesh}} d\Gamma = \int N_i \begin{pmatrix} u_{\text{mesh}} \\ v_{\text{mesh}} \end{pmatrix} d\Gamma \quad (360)$$

$$= \int N_i \begin{pmatrix} N_1 & 0 & N_2 & 0 \\ 0 & N_1 & 0 & N_2 \end{pmatrix} \cdot \begin{pmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \end{pmatrix} d\Gamma \quad (361)$$

(362)

Writing this equation alternatively for $N_i = N_1, N_2$ yields:

$$\int_{\Gamma_e} \begin{pmatrix} N_1 N_1 & 0 & N_1 N_2 & 0 \\ 0 & N_1 N_1 & 0 & N_1 N_2 \\ N_2 N_1 & 0 & N_2 N_2 & 0 \\ 0 & N_2 N_1 & 0 & N_2 N_2 \end{pmatrix} \cdot \begin{pmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \end{pmatrix} d\Gamma = \int_{\Gamma_e} \begin{pmatrix} N_1 N_1 & 0 & N_1 N_2 & 0 \\ 0 & N_1 N_1 & 0 & N_1 N_2 \\ N_2 N_1 & 0 & N_2 N_2 & 0 \\ 0 & N_2 N_1 & 0 & N_2 N_2 \end{pmatrix} d\Gamma \cdot \begin{pmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \end{pmatrix} \quad (362)$$

Turning now to the right hand side $\int_{\Gamma_e} N_i (\vec{v} \cdot \vec{n}_e) \vec{n}_e d\Gamma$, it yields the following rhs:

$$\int_{\Gamma_e} (\vec{v} \cdot \vec{n}_e) \begin{pmatrix} N_1 n_x \\ N_1 n_y \\ N_2 n_x \\ N_2 n_y \end{pmatrix} d\Gamma$$

The elemental matrix and rhs must be built for each element and assembled in a global matrix and rhs. The solution is the mesh velocity vector at all surface nodes. the same approach can be taken for Q_2 elements:

$$\int_{\Gamma_e} N_i \vec{v}_{\text{mesh}} d\Gamma = \int N_i \begin{pmatrix} u_{\text{mesh}} \\ v_{\text{mesh}} \end{pmatrix} d\Gamma \quad (363)$$

$$= \int N_i \begin{pmatrix} N_1 & 0 & N_2 & 0 & N_3 & 0 \\ 0 & N_1 & 0 & N_2 & 0 & N_3 \end{pmatrix} \cdot \begin{pmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \\ u_3 \\ v_3 \end{pmatrix} d\Gamma \quad (364)$$

Writing this equation alternatively for $N_i = N_1, N_2, N_3$ yields:

$$\int_{\Gamma_e} \begin{pmatrix} N_1 N_1 & 0 & N_1 N_2 & 0 & N_1 N_3 & 0 \\ 0 & N_1 N_1 & 0 & N_1 N_2 & 0 & N_1 N_3 \\ N_2 N_1 & 0 & N_2 N_2 & 0 & N_2 N_3 & 0 \\ 0 & N_2 N_1 & 0 & N_2 N_2 & 0 & N_2 N_3 \\ N_3 N_1 & 0 & N_3 N_2 & 0 & N_3 N_3 & 0 \\ 0 & N_3 N_1 & 0 & N_3 N_2 & 0 & N_3 N_3 \end{pmatrix} \cdot \begin{pmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \\ u_3 \\ v_3 \end{pmatrix} d\Gamma$$

$$= \int_{\Gamma_e} \begin{pmatrix} N_1 N_1 & 0 & N_1 N_2 & 0 & N_1 N_3 & 0 \\ 0 & N_1 N_1 & 0 & N_1 N_2 & 0 & N_1 N_3 \\ N_2 N_1 & 0 & N_2 N_2 & 0 & N_2 N_3 & 0 \\ 0 & N_2 N_1 & 0 & N_2 N_2 & 0 & N_2 N_3 \\ N_3 N_1 & 0 & N_3 N_2 & 0 & N_3 N_3 & 0 \\ 0 & N_3 N_1 & 0 & N_3 N_2 & 0 & N_3 N_3 \end{pmatrix} d\Gamma \cdot \begin{pmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \\ u_3 \\ v_3 \end{pmatrix} \quad (365)$$

²⁵what if Q_k with $k > 1$ elements are used and nodes on the surface no more form a line?

The right hand side is then

$$\int_{\Gamma_e} (\vec{v} \cdot \vec{n}_e) \begin{pmatrix} N_1 n_x \\ N_1 n_y \\ N_2 n_x \\ N_2 n_y \\ N_3 n_x \\ N_3 n_y \end{pmatrix} d\Gamma$$

Having obtained the boundary condition velocity for the Laplace equation, we can now turn our attention to solving this ODE.

Note that Rose et al (2017) [1599] go further than this and propose a 'nonstandard finite difference scheme' and make a link with the stabilisation presented in Kaus et al (2010) [1032].

In what follows I omit the subscript 'mesh' and focus on the 2D case. The components of the (mesh) velocity are given by

$$u^h = \sum_{i=1}^{m_v} N_i^\gamma u_i \quad v^h = \sum_{i=1}^{m_v} N_i^\gamma v_i \quad \vec{v}^h = \begin{pmatrix} u^h \\ v^h \end{pmatrix}$$

We start from the ODE to solve in its strong form:

$$\Delta \vec{v}^h = \vec{0}$$

We multiply it by a velocity test function N_i^γ and integrate over an element:

$$\begin{aligned} & \vec{0} \\ &= \int_{\Omega_e} N_i^\gamma \Delta \vec{v}^h \\ &= \int_{\Omega_e} N_i^\gamma \Delta \vec{v}^h dV \\ &= \int_{\Omega_e} \begin{pmatrix} N_i^\gamma \Delta u^h \\ N_i^\gamma \Delta v^h \end{pmatrix} dV \\ &= \int_{\Omega_e} \begin{pmatrix} N_i^\gamma \vec{\nabla} \cdot \vec{\nabla} u^h \\ N_i^\gamma \vec{\nabla} \cdot \vec{\nabla} v^h \end{pmatrix} dV \\ &= \int_{\Omega_e} \begin{pmatrix} \vec{\nabla} N_i^\gamma \cdot \vec{\nabla} u^h \\ \vec{\nabla} N_i^\gamma \cdot \vec{\nabla} v^h \end{pmatrix} dV \\ &= \int_{\Omega_e} \begin{pmatrix} \partial_x N_i^\gamma \partial_x u^h + \partial_y N_i^\gamma \partial_y u^h \\ \partial_x N_i^\gamma \partial_x v^h + \partial_y N_i^\gamma \partial_y v^h \end{pmatrix} dV \\ &= \int_{\Omega_e} \begin{pmatrix} \partial_x N_i^\gamma & \partial_y N_i^\gamma & 0 & 0 \\ 0 & 0 & \partial_x N_i^\gamma & \partial_y N_i^\gamma \end{pmatrix} \cdot \begin{pmatrix} \partial_x u^h \\ \partial_y u^h \\ \partial_x v^h \\ \partial_y v^h \end{pmatrix} dV \\ &= \int_{\Omega_e} \begin{pmatrix} \frac{\partial N_i^\gamma}{\partial x} & 0 & \frac{\partial N_1^\gamma}{\partial x} & 0 & \dots & \frac{\partial N_{m_v}^\gamma}{\partial x} & 0 \\ \frac{\partial N_1^\gamma}{\partial y} & 0 & \frac{\partial N_2^\gamma}{\partial y} & 0 & \dots & \frac{\partial N_{m_v}^\gamma}{\partial y} & 0 \\ 0 & \frac{\partial N_1^\gamma}{\partial x} & 0 & \frac{\partial N_2^\gamma}{\partial x} & \dots & 0 & \frac{\partial N_{m_v}^\gamma}{\partial x} \\ 0 & \frac{\partial N_1^\gamma}{\partial y} & 0 & \frac{\partial N_2^\gamma}{\partial y} & \dots & 0 & \frac{\partial N_{m_v}^\gamma}{\partial y} \end{pmatrix} \cdot \begin{pmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \\ \dots \\ u_{m_v} \\ v_{m_v} \end{pmatrix} dV \end{aligned}$$

Writing this equation for $i = 1, \dots, m_v$, we obtain:

$$\int \begin{pmatrix} \frac{\partial N_1^v}{\partial x} & \frac{\partial N_1^v}{\partial y} & 0 & 0 \\ 0 & 0 & \frac{\partial N_2^v}{\partial x} & \frac{\partial N_1^v}{\partial y} \\ \frac{\partial N_2^v}{\partial x} & \frac{\partial N_2^v}{\partial y} & 0 & 0 \\ 0 & 0 & \frac{\partial N_2^v}{\partial x} & \frac{\partial N_2^v}{\partial y} \\ \vdots & \vdots & \vdots & \vdots \\ \vdots & \vdots & \vdots & \vdots \\ \frac{\partial N_{m_v}^v}{\partial x} & \frac{\partial N_{m_v}^v}{\partial y} & 0 & 0 \\ 0 & 0 & \frac{\partial N_{m_v}^v}{\partial x} & \frac{\partial N_{m_v}^v}{\partial y} \end{pmatrix} \cdot \begin{pmatrix} \frac{\partial N_1^v}{\partial x} & 0 & \frac{\partial N_2^v}{\partial x} & 0 & \dots & \frac{\partial N_{m_v}^v}{\partial x} & 0 \\ \frac{\partial N_1^v}{\partial y} & 0 & \frac{\partial N_2^v}{\partial y} & 0 & \dots & \frac{\partial N_{m_v}^v}{\partial y} & 0 \\ 0 & \frac{\partial N_1^v}{\partial x} & 0 & \frac{\partial N_2^v}{\partial x} & \dots & 0 & \frac{\partial N_{m_v}^v}{\partial x} \\ 0 & \frac{\partial N_1^v}{\partial y} & 0 & \frac{\partial N_2^v}{\partial y} & \dots & 0 & \frac{\partial N_{m_v}^v}{\partial y} \end{pmatrix} \cdot \underbrace{\begin{pmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \\ \dots \\ u_{m_v} \\ v_{m_v} \end{pmatrix}}_{\vec{V}} dV = \vec{0}$$

or,

$$\left(\int_{\Omega_e} \mathbf{B}^T \cdot \mathbf{B} dV \right) \cdot \vec{V} = \vec{0}$$

where \mathbf{B} is a $(ndim * ndim) \times (m_v * ndofV)$ matrix. This is implemented in Stone 54.9.

Remark. The integration by parts should have a minus appear but since the left hand side is 0, it is not taken into account.

surface terms arising from the integration by parts are neglected. EXPLAIN WHY!

Yet another approach [532] The unknown position of free surfaces can be computed using the following approach: for the simple case of a single-valued function $h = h(x, y, t)$, a hyperbolic equation must be solved,

$$\frac{\partial h}{\partial t} + (\vec{v} \cdot \vec{\nabla}) h = 0 \quad (366)$$

This is the kinematic equation of the surface and has been used, for instance, by Ramaswamy and Kawahara (1987), Huerta and Liu, 1988b, 1990; Souli and Zolesio (2001).

Idea: Eq. (366) is a simple advection equation. One could also add a diffusion operator with a diffusion coefficient D . Low values of D could be used to stabilise the surface while higher values (possibly nonlinear ones) could be used to account for simple surface processes.

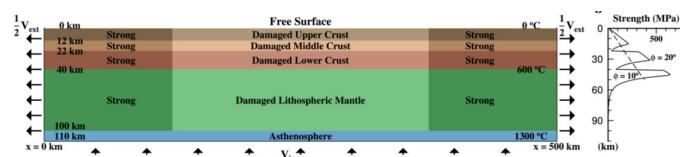
$$\frac{\partial h}{\partial t} + (\vec{v} \cdot \vec{\nabla}) h = D \Delta h \quad (367)$$

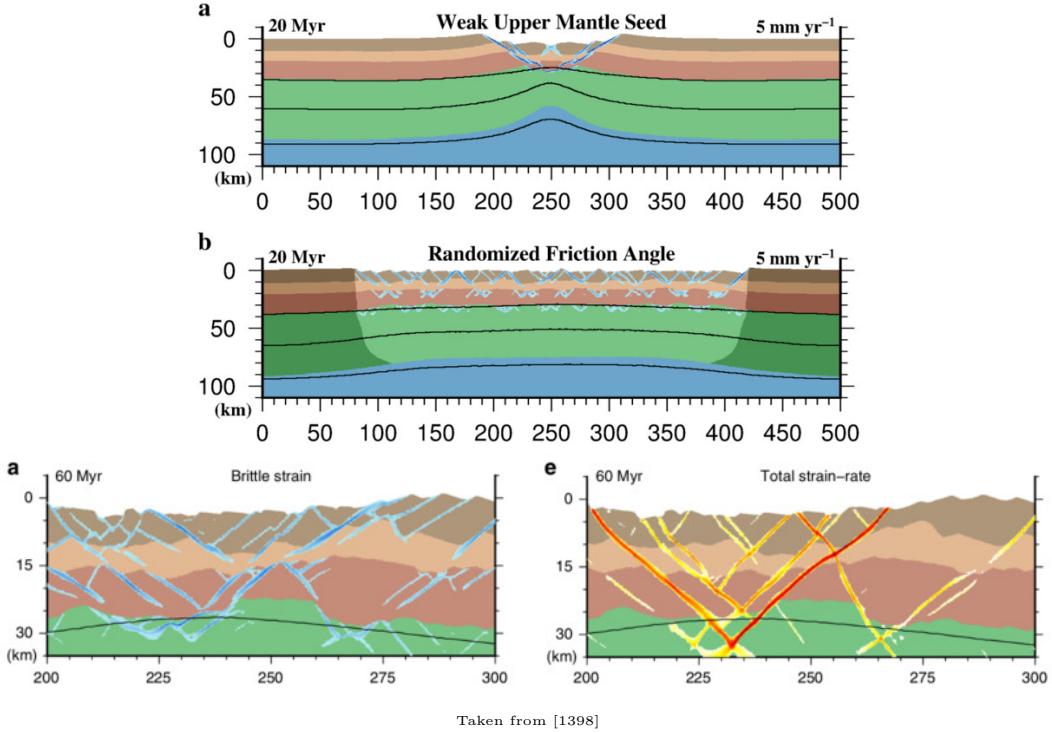
Also, Hansen and Nielsen [862, 861] write: During the entire model evolution surface processes act to re-distribute sediments. These processes are modelled by a diffusion equation with a source term enabling the transport of sediments to and from the model profile. The transport equation is written

$$\dot{h} = \nabla \cdot (\kappa \nabla h) + \dot{s}(w)$$

where $\kappa = 200 \text{ km}^2/\text{Ma}$ is the diffusivity of topography and $\dot{s}(w)$ is a linear function of water depth.

The following pictures are taken from Naliboff et al [1398] on the topic of how complex fault interaction controls continental rifting. It is a beautiful example (among many) of the importance of free surface geodynamical expression and large deformation:





8.1.4 On the topic of moving internal nodes

Braess & Wriggers [210] propose the following interesting algorithm: "A measure of the quality of a triangular mesh is the quotient of the outer radius r_{out} and the inner radius r_{in} of each element. This quotient is important because it plays a certain role in a priori error estimates. If an element degenerates this quotient will approach infinity. Another important feature of good mesh is that no element becomes very large. With these considerations in mind the penalty function W is defined:

$$W = \sum_{elts} \left(\frac{r_{out}}{r_{in}} \right)^m \left(\frac{r_{out}}{r_0} \right)^n \quad (368)$$

where m , n and r_0 are positive constants. For our calculations we chose $m = 3$, $n = 1$ and $r_0 = 1$, but the results seem to depend only slightly on this choice. Whenever a triangle is distorted or very large, this function becomes very large. A similar penalty function was presented in [1004] for four-node elements. In that case the angles of the elements are used to construct the penalty function. In order to regularize a distorted mesh the coordinates of the internal nodes will be chosen such that W is minimized. It is not necessary to reach the global minimum, a rough approximation is sufficient. Therefore the minimization of the potential can be done efficiently with standard procedures and will not be discussed in any detail. This algorithm can also be applied to h -adaptive mesh-generation by choosing appropriate constants r_0 for each triangle."²⁶

This is still WORK IN PROGRESS. I Need to look at those papers: [937][1497] [131] [550] [555] [31] [1115] [1689] [1248] [837][2008] [572] [210] [1451] and talk about free surface stabilisation [1032, 1527, 550,

²⁶Indeed, if r_0 is the same for all elements this parameter will not play any role at all in the minimisation process.

1599].

8.2 Convergence criterion for nonlinear iterations

MEGA WORK in PROGRESS!!

Following [1685], one can monitor the relative changes in the solution from iteration to iteration. For instance,

$$\frac{\|\Delta \vec{V}\|_{L2}}{\|\vec{V}\|_{L2}} = \left(\frac{\int_{\Omega} (\vec{V}_i - \vec{V}_{i-1}) \cdot (\vec{V}_i - \vec{V}_{i-1}) dV}{\int_{\Omega} \vec{V}_i \cdot \vec{V}_i dV} \right)^{1/2}$$

is a measure of the relative change in the velocity field from iteration $i - 1$ to i . The same monitoring can be done for pressure:

$$\frac{\|\Delta \vec{P}\|_{L2}}{\|\vec{P}\|_{L2}} = \left(\frac{\int_{\Omega} (\vec{P}_i - \vec{P}_{i-1}) \cdot (\vec{P}_i - \vec{P}_{i-1}) dV}{\int_{\Omega} \vec{P}_i \cdot \vec{P}_i dV} \right)^{1/2}$$

Convergence is reached when both are below 0.001 [1150] or 0.0001 [1030].

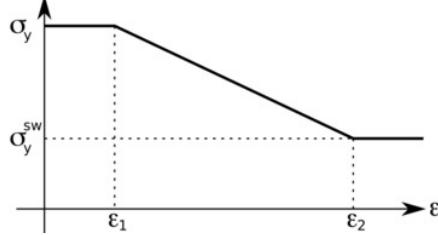
nonlinear residual ... see [1685] p2222.

check correlation of [1756]

8.3 Strain weakening

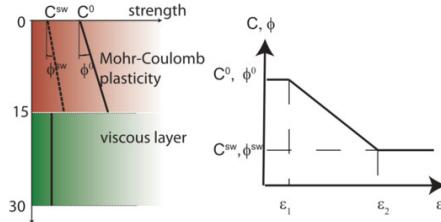
Several mechanisms may contribute to strain or strain rate dependent weakening but their relative and absolute importance is poorly constrained. Furthermore, weakening mechanisms are often crudely parameterised in geodynamical codes with simple mathematical functions and a limited number of parameters.

For example, in [19] the authors use a von Mises plasticity formulation so that the rheology is parameterised by the cohesion c , or $c = \sigma_y$ in their notations. The yield strength σ_y starts constant until the strain ε reaches the threshold value ε_1 . It then decreases linearly from σ_y to σ_y^{sw} between ε_1 and ε_2 . For strain values $\varepsilon > \varepsilon_2$, the yield strength remains constant at σ_y^{sw} .



Taken from [19]

The same authors in a subsequent study use a Drucker-Prager rheology parameterised by cohesion c and friction angle ϕ . They use the same approach as before but now both parameters are subjected to strain weakening:



Taken from [20], see also [1756]

They further define the factor $R = C^0/C^{sw} = \phi^0/\phi^{sw} \geq 1$ which is a proxy for the ratio σ_y/σ_y^{sw} where $\sigma_y = p \sin \phi + c \cos \phi$, and carry out 3D crustal extensional models for R between 2 and 5.

In [1141] the authors also define

$$\tau_y = p \sin(\phi(\varepsilon^p)) + c_0 \cos(\phi(\varepsilon^p))$$

but the cohesion is regarded to be constant. The angle of friction ϕ is assumed to decrease as a function of the accumulated plastic strain ε^p to

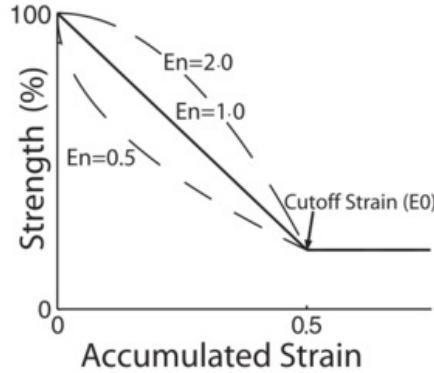
$$\phi(\varepsilon^p) = \max \left(\phi_\infty, \phi_0 - \frac{\varepsilon^p(\phi_0 - \phi_\infty)}{\varepsilon_\infty^p} \right)$$

This equation defines an empirical softening relation which reduces the friction angle linearly with accumulated plastic strain. ϕ_0 defines the initial friction angle, ε_∞^p represents the measure of plastic strain after which complete softening is achieved and internal friction angle reaches ϕ_∞ . Plastic strain represents an integrated, tensorial invariant measure of the deformation which has occurred due to plastic yielding. Thus, the quantity ε^p can be regarded as a simplified measure of material damage.

In Dyksterhuis et al [557] a variant of the above formulation is used:

$$f(\varepsilon) = \begin{cases} 1 - (1-a)(\varepsilon/\varepsilon_0)^n & \varepsilon \leq \varepsilon_0 \\ a & \varepsilon \geq \varepsilon_0 \end{cases}$$

where ε is the accumulated plastic strain, ε_0 is the saturation strain beyond which no further weakening takes place, n is an exponent that controls the shape of the function and a is a maximum value of strain weakening beyond which no further weakening occurs. This equation leads to the following plot:



Strain-softening behaviour showing strength weakening from 100 to 20% after an accumulated strain of 0.5, after which no further weakening occurs.

Dashed lines show the effect of the exponential parameter (E_n) on the curve. Taken from [557]

Although it is not specified in [557] what f is, other users of the code specify that the yield strength is given by

$$\sigma_y = (B_0 + B_1 p) f(\varepsilon)$$

where p is the pressure, B_0 is the cohesion, or yield stress at zero pressure, and B_p is the pressure dependence of the yield stress, equivalent to the friction coefficient in Byerlees law.

In [1971] the authors take a different approach:

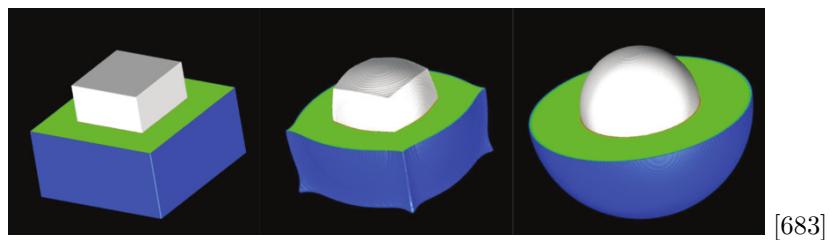
$$C = C_0 + C_1 \exp\left(-\frac{\varepsilon_{plast}}{\varepsilon_{ref}}\right)$$

$$\mu = \mu_0 + \mu_1 \exp\left(-\frac{\varepsilon_{plast}}{\varepsilon_{ref}}\right)$$

where C_0 and $C_0 + C_1$ represent the minimum and maximum cohesions, respectively; μ_0 and $\mu_0 + \mu_1$ represent the minimum and maximum frictional coefficients, respectively. ε_{plast} and ε_{ref} represent accumulated plastic strain and reference strain, respectively.

Literature: [1411]

8.4 The gravity vector



[683]

8.5 The SUPG formulation for the energy equation

As abundantly documented in the literature advection needs to be stabilised as it otherwise showcases non-negligible under- and overshoots. A standard approach is the Streamline Upwind Petrov Galerkin (SUPG) method.

 Literature[238][940]

8.5.1 Linear elements

When using linear elements, its implementation is rather trivial, as shown in the DOUAR paper [221] or the FANTOM paper [1756]. The advection matrix is simply modified and computed as follows:

$$(\mathbf{K}_a^e)_{SUPG} = \int_{x_k}^{x_{k+1}} (\mathbf{N}^*)^T \rho C_p \vec{v} \cdot \mathbf{B} dx \quad \text{with} \quad \mathbf{N}^* = \mathbf{N} + \tau \vec{v} \cdot \mathbf{B}$$

Note that we can also write

$$(\mathbf{K}_a^e)_{SUPG} = \int_{x_k}^{x_{k+1}} \mathbf{N}^T \rho C_p \vec{v} \cdot \mathbf{B} dx + \int_{x_k}^{x_{k+1}} \tau (\vec{v} \cdot \mathbf{B})^T \rho C_p (\vec{v} \cdot \mathbf{B}) dx$$

and we see that the SUPG method introduces an additional term that is akin to a diffusion term in the direction of the flow. This can be seen by looking at the advection matrix a regular grid of 1D elements of size h :

$$(\mathbf{K}_a^e)_{SUPG} = \mathbf{K}_a^e + \rho C_p \frac{\tau u^2}{h} \begin{pmatrix} 1 & -1 \\ -1 & 1 \end{pmatrix}$$

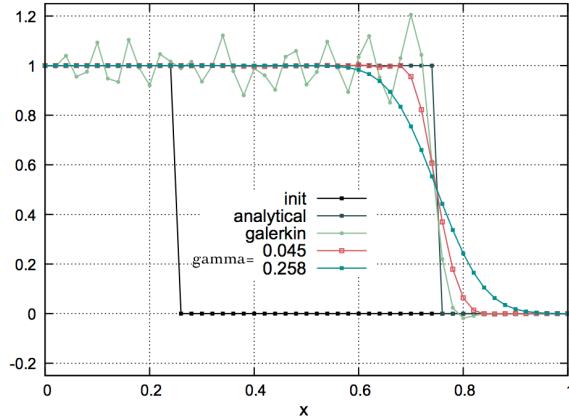
The additional matrix has the same structure as the 1D diffusion matrix matrix in 5.1.

The parameter τ is chosen as follows:

$$\tau = \gamma \frac{h}{\nu} \quad (369)$$

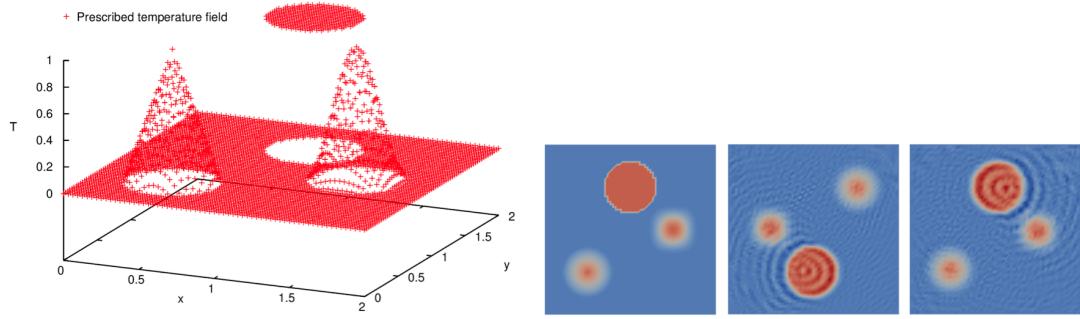
where γ is a user chosen parameter (see Appendix A of [1756]).

A typical test case for testing an advection scheme is the step advection benchmark (see for instance [533]). At $t = 0$, a field $T(x)$ is prescribed in a 1D domain of unit length. For $x \leq 1/4$ we have $T(x) = 1$ and $T(x) = 0$ everywhere else as shown on the following figure:



The prescribed velocity is $\nu = 1$, 50 elements are used and 250 time steps are carried out with $\delta t = 0.1h/\nu = 0.002$. As discussed in [1756], using Equation 369, one arrives to $\gamma = 0.045$, which leads to a desired removal of the oscillations through a small amount of numerical diffusion. Braun [216] argues for a constant $\gamma = 1/\sqrt{15} = 0.258$ (after [942]), which effect is also shown in the figure above. This value is arguably too large and introduces undesirable diffusion.

Another classic example of advection testing is a 2D problem where (for example) a cylinder, a Gaussian and a cone are prescribed and advected with a velocity field (see for instance [533]).



After a 2π rotation and in the absence of stabilisation we see that the temperature field showcases clearly visible ripples.

Remark. Note that ASPECT originally did not rely on the SUPG formulation to stabilise the advection(-diffusion) equations[1119]. It instead relied on the Entropy Viscosity formulation [823]. It is only during the 6th Hackathon in May 2019 that the SUPG was introduced on the code. Note that the ASPECT implementation is based on the deal.II step 63²⁷.

²⁷https://www.dealii.org/developer/doxygen/deal.II/step_63.html

8.6 The method of manufactured solutions

The method of manufactured solutions is a relatively simple way of carrying out code verification. In essence, one postulates a solution for the PDE at hand (as well as the proper boundary conditions), inserts it in the PDE and computes the corresponding source term. The same source term and boundary conditions will then be used in a numerical simulation so that the computed solution can be compared with the (postulated) true analytical solution.

Examples of this approach are to be found in [533, 308, 181, 1501, 1499, 1222, 176].

8.6.1 Analytical benchmark I - "DH"

Taken from [533]. We consider a two-dimensional problem in the square domain $\Omega = [0, 1] \times [0, 1]$, which possesses a closed-form analytical solution. The problem consists of determining the velocity field $\vec{v} = (u, v)$ and the pressure p such that

$$\eta \Delta \vec{v} - \vec{\nabla} p + \vec{b} = \vec{0} \quad \text{in } \Omega \quad (370)$$

$$\vec{\nabla} \cdot \vec{v} = 0 \quad \text{in } \Omega \quad (371)$$

$$\vec{v} = \vec{0} \quad \text{on } \Gamma_D \quad (372)$$

where the fluid viscosity is taken as $\eta = 1$. The components of the body force \vec{b} are prescribed as

$$\begin{aligned} b_x &= (12 - 24y)x^4 + (-24 + 48y)x^3 + (-48y + 72y^2 - 48y^3 + 12)x^2 \\ &\quad + (-2 + 24y - 72y^2 + 48y^3)x + 1 - 4y + 12y^2 - 8y^3 \\ b_y &= (8 - 48y + 48y^2)x^3 + (-12 + 72y - 72y^2)x^2 \\ &\quad + (4 - 24y + 48y^2 - 48y^3 + 24y^4)x - 12y^2 + 24y^3 - 12y^4 \end{aligned}$$

With this prescribed body force, the exact solution is

$$\begin{aligned} u(x, y) &= x^2(1-x)^2(2y-6y^2+4y^3) \\ v(x, y) &= -y^2(1-y)^2(2x-6x^2+4x^3) \\ p(x, y) &= x(1-x) - 1/6 \end{aligned}$$

Note that the pressure obeys $\int_{\Omega} p d\Omega = 0$. One can turn to the spatial derivatives of the fields:

$$\dot{\varepsilon}_{xx} = \frac{\partial u}{\partial x} = (2x - 6x^2 + 4x^3)(2y - 6y^2 + 4y^3) \quad (373)$$

$$\dot{\varepsilon}_{yy} = \frac{\partial v}{\partial y} = -(2x - 6x^2 + 4x^3)(2y - 6y^2 + 4y^3) \quad (374)$$

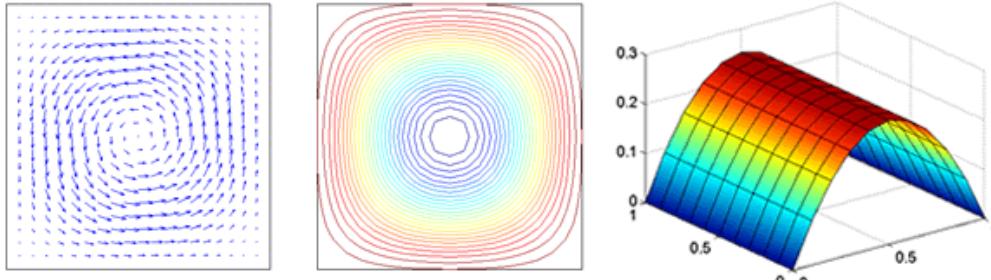
$$\dot{\varepsilon}_{xy} = \frac{1}{2} \left(\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) = \frac{1}{2} (x^2(1-x)^2(2-12y+12y^2) - y^2(1-y)^2(2-12x+12x^2)) \quad (375)$$

with of course $\vec{\nabla} \cdot \vec{v} = 0$ and

$$\frac{\partial p}{\partial x} = 1 - 2x \quad (376)$$

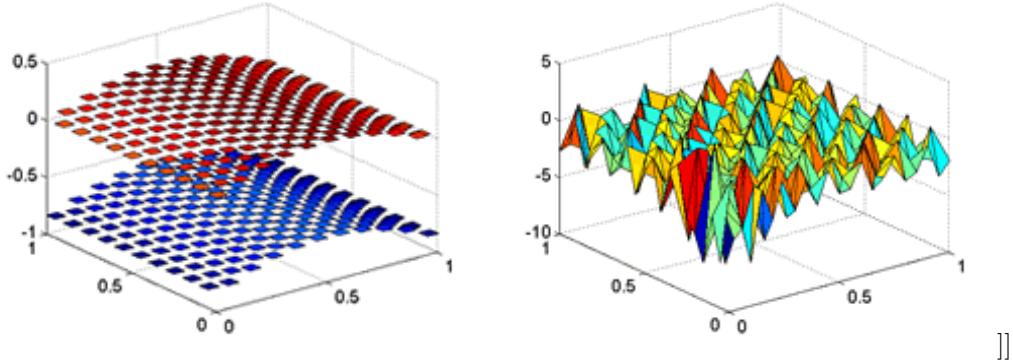
$$\frac{\partial p}{\partial y} = 0 \quad (377)$$

The velocity and pressure fields look like:



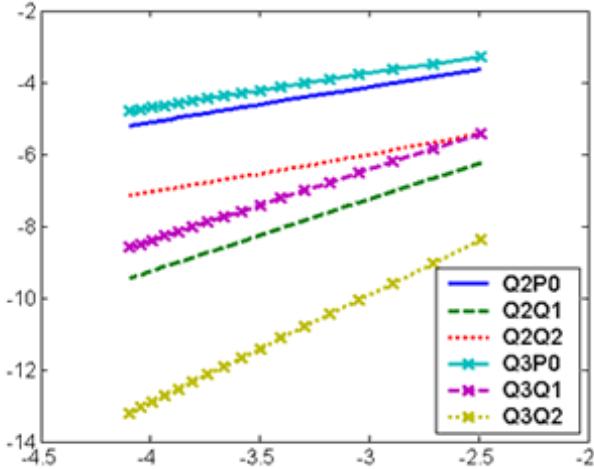
http://ww2.lacan.upc.edu/huerta/exercises/Incompressible/Incompressible_Ex1.htm

As shown in [533], If the LBB condition is not satisfied, spurious oscillations spoil the pressure approximation. Figures below show results obtained with a mesh of 20x20 Q1P0 (left) and P1P1 (right) elements:



http://ww2.lacan.upc.edu/huerta/exercises/Incompressible/Incompressible_Ex1.htm

Taking into account that the proposed problem has got analytical solution, it is easy to analyze convergence of the different pairs of elements:



http://ww2.lacan.upc.edu/huerta/exercises/Incompressible/Incompressible_Ex1.htm

One can also compute the stress components:

$$\sigma_{xx} = 2x^2(2x - 2)(4y^3 - 6y^2 + 2y) + 4x(-x + 1)^2 * (4y^3 - 6y^2 + 2y) - x(-x + 1) + 1/6 \quad (378)$$

$$\sigma_{xy} = x^2(-x + 1)^2 * (12y^2 - 12y + 2) - y^2(-y + 1)^2 * (12x^2 - 12x + 2) \quad (379)$$

$$\sigma_{yy} = -x(-x + 1) - 2y^2(2y - 2)(4x^3 - 6x^2 + 2x) - 4y(-y + 1)^2(4x^3 - 6x^2 + 2x) + 1/6 \quad (380)$$

All the necessary functions to do this benchmark are in `mms/dh.py`:

```
# functions for the Donea & Huerta benchmark (dh)

def u_th(x,y):
    return x**2*(1.-x)**2*(2*y-6*y**2+4*y**3)

def v_th(x,y):
    return -y**2*(1.-y)**2*(2*x-6*x**2+4*x**3)

def p_th(x,y):
    return x*(1-x)-1./6.

def dpdx_th(x,y):
    return 1.-2.*x
```

```

def dpdy_th(x,y):
    return 0.

def exx_th(x,y):
    return x**2*(2*x-2)*(4*y**3-6*y**2+2*y)+2*x*(-x+1)**2*(4*y**3-6*y**2+2*y)

def eyy_th(x,y):
    return (x**2*(-x+1)**2*(12*y**2-12*y+2)-y**2*(-y+1)**2*(12*x**2-12*x+2))/2

def bx(x,y):
    return ((12.-24.*y)*x**4+(-24.+48.*y)*x*x*x +
           (-48.*y+72.*y*y-48.*y*y*y+12.)*x*x +
           (-2.+24.*y-72.*y*y+48.*y*y*y)*x +
           1.-4.*y+12.*y*y-8.*y*y*y)

def by(x,y):
    return ((8.-48.*y+48.*y*y)*x*x*x+
           (-12.+72.*y-72.*y*y)*x*x+
           (4.-24.*y+48.*y*y-48.*y*y*y+24.*y**4)*x -
           12.*y*y+24.*y*y*y-12.*y**4)

```

This benchmark is implemented in ASPECT [77].

8.6.2 Analytical benchmark II - "DB2D"

Taken from [529, 181]. It is for a unit square with $\nu = \mu/\rho = 1$ and the smooth exact solution is

$$u(x, y) = x + x^2 - 2xy + x^3 - 3xy^2 + x^2y \quad (381)$$

$$v(x, y) = -y - 2xy + y^2 - 3x^2y + y^3 - xy^2 \quad (382)$$

$$p(x, y) = xy + x + y + x^3y^2 - 4/3 \quad (383)$$

Note that the pressure obeys $\int_{\Omega} p \, d\Omega = 0$

$$b_x = -(1 + y - 3x^2y^2) \quad (384)$$

$$b_y = -(1 - 3x - 2x^3y) \quad (385)$$

This benchmark is also used in [1954].

8.6.3 Analytical benchmark III - "DB3D"

This benchmark begins by postulating a polynomial solution to the 3D Stokes equation [529]:

$$\mathbf{v} = \begin{pmatrix} x + x^2 + xy + x^3y \\ y + xy + y^2 + x^2y^2 \\ -2z - 3xz - 3yz - 5x^2yz \end{pmatrix} \quad (386)$$

and

$$p = xyz + x^3y^3z - 5/32 \quad (387)$$

While it is then trivial to verify that this velocity field is divergence-free (see here under), the corresponding body force of the Stokes equation can be computed by inserting this solution into the momentum equation with a given viscosity $\eta(x, y, z)$ (constant or position/velocity/strain rate dependent). The domain is a unit cube and velocity boundary conditions simply use Eq. (657). Note that the pressure fulfills

$$\int_{\Omega} p(x, y, z) dV = 0.$$

Following [308], the viscosity is given by the smoothly varying function

$$\eta(x, y, z) = \exp(1 - \beta(x(1-x) + y(1-y) + z(1-z))) \quad (388)$$

Choosing $\beta = 0$ yields a constant velocity $\eta = e^1$ (and greatly simplifies the right-hand side). One can easily show that the ratio of viscosities η^* in the system follows $\eta^* = \exp(-3\beta/4)$ so that choosing $\beta = 10$ yields $\eta^* \simeq 1808$ and $\beta = 20$ yields $\eta^* \simeq 3.269 \times 10^6$.

$$\frac{\partial \mu}{\partial x} = -4\beta(1-2x)\mu(x, y, z) \quad (389)$$

$$\frac{\partial \mu}{\partial y} = -4\beta(1-2y)\mu(x, y, z) \quad (390)$$

$$\frac{\partial \mu}{\partial z} = -4\beta(1-2z)\mu(x, y, z) \quad (391)$$

We can compute the components of the strainrate tensor:

$$\dot{\varepsilon}_{xx} = 1 + 2x + y + 3x^2y \quad (392)$$

$$\dot{\varepsilon}_{yy} = 1 + x + 2y + 2x^2y \quad (393)$$

$$\dot{\varepsilon}_{zz} = -2 - 3x - 3y - 5x^2y \quad (394)$$

$$\dot{\varepsilon}_{xy} = \frac{1}{2}(x + y + 2xy^2 + x^3) \quad (395)$$

$$\dot{\varepsilon}_{xz} = \frac{1}{2}(-3z - 10xyz) \quad (396)$$

$$\dot{\varepsilon}_{yz} = \frac{1}{2}(-3z - 5x^2z) \quad (397)$$

and recover $\dot{\varepsilon}_{xx} + \dot{\varepsilon}_{yy} + \dot{\varepsilon}_{zz} = 0$ as expected.

We further have

$$\frac{\partial}{\partial x} 2\dot{\varepsilon}_{xx} = 2(2 + 6xy)$$

$$\frac{\partial}{\partial y} 2\dot{\varepsilon}_{xy} = 1 + 4xy$$

$$\frac{\partial}{\partial z} 2\dot{\varepsilon}_{xz} = -3 - 10xy$$

$$\frac{\partial}{\partial x} 2\dot{\varepsilon}_{xy} = 1 + 2y^2 + 3x^2$$

$$\frac{\partial}{\partial y} 2\dot{\varepsilon}_{yy} = 2(2 + 2x^2)$$

$$\frac{\partial}{\partial z} 2\dot{\varepsilon}_{yz} = -3 - 5x^2$$

$$\frac{\partial}{\partial x} 2\dot{\varepsilon}_{xz} = -10yz$$

$$\frac{\partial}{\partial y} 2\dot{\varepsilon}_{yz} = 0$$

$$\frac{\partial}{\partial z} 2\dot{\varepsilon}_{zz} = 2(0)$$

$$\frac{\partial p}{\partial x} = yz + 3x^2y^3z \quad (398)$$

$$\frac{\partial p}{\partial y} = xz + 3x^3y^2z \quad (399)$$

$$\frac{\partial p}{\partial z} = xy + x^3y^3 \quad (400)$$

Constant viscosity In this case, the right hand side writes:

$$f_x = -\frac{\partial p}{\partial x} + 2\eta \frac{\partial}{\partial x} \dot{\epsilon}_{xx} + 2\eta \frac{\partial}{\partial y} \dot{\epsilon}_{xy} + 2\eta \frac{\partial}{\partial z} \dot{\epsilon}_{xz} \quad (401)$$

$$= -(yz + 3x^2y^3z) + 2(2 + 6xy) + (1 + 4xy) + (-3 - 10xy) \quad (402)$$

$$f_y = -\frac{\partial p}{\partial y} + 2\mu \frac{\partial}{\partial x} \dot{\epsilon}_{xy} + 2\mu \frac{\partial}{\partial y} \dot{\epsilon}_{yy} + 2\mu \frac{\partial}{\partial z} \dot{\epsilon}_{yz} \quad (403)$$

$$= -(xz + 3x^3y^2z) + \mu(1 + 2y^2 + 3x^2) + \mu(2 + 2x^2) + \mu(-3 - 5x^2) \quad (404)$$

$$f_z = -\frac{\partial p}{\partial z} + 2\mu \frac{\partial}{\partial x} \dot{\epsilon}_{xz} + 2\mu \frac{\partial}{\partial y} \dot{\epsilon}_{yz} + 2\mu \frac{\partial}{\partial z} \dot{\epsilon}_{zz} \quad (405)$$

$$= -(xy + x^3y^3) + \mu(-10yz) + 0 + 0 \quad (406)$$

and finally

$$\begin{aligned} \mathbf{f} &= -\nabla p + \eta \begin{pmatrix} 2 + 6xy \\ 2 + 2x^2 + 2y^2 \\ -10yz \end{pmatrix} \\ &= - \begin{pmatrix} yz + 3x^2y^3z \\ xz + 3x^3y^2z \\ xy + x^3y^3 \end{pmatrix} + \eta \begin{pmatrix} 2 + 6xy \\ 2 + 2x^2 + 2y^2 \\ -10yz \end{pmatrix} \end{aligned}$$

Variable viscosity In this case, the right hand side is obtained through

$$\begin{aligned} \mathbf{f} &= -\nabla p + \eta \begin{pmatrix} 2 + 6xy \\ 2 + 2x^2 + 2y^2 \\ -10yz \end{pmatrix} \\ &+ \begin{pmatrix} 2\dot{\epsilon}_{xx} \\ 2\dot{\epsilon}_{xy} \\ 2\dot{\epsilon}_{xz} \end{pmatrix} \frac{\partial \eta}{\partial x} + \begin{pmatrix} 2\dot{\epsilon}_{xy} \\ 2\dot{\epsilon}_{yy} \\ 2\dot{\epsilon}_{yz} \end{pmatrix} \frac{\partial \eta}{\partial y} + \begin{pmatrix} 2\dot{\epsilon}_{xz} \\ 2\dot{\epsilon}_{yz} \\ 2\dot{\epsilon}_{zz} \end{pmatrix} \frac{\partial \eta}{\partial z} \end{aligned} \quad (407)$$

[308] has carried out this benchmark for $\beta = 4$, i.e.:

$$\mu(x, y, z) = \exp(1 - 4(x(1-x) + y(1-y) + z(1-z)))$$

In a unit cube, this yields a variable viscosity such that $0.1353 < \mu < 2.7182$, i.e. a ratio of approx. 20 within the domain. We then have:

$$\frac{\partial \mu}{\partial x} = -4(1 - 2x)\mu(x, y, z) \quad (408)$$

$$\frac{\partial \mu}{\partial y} = -4(1 - 2y)\mu(x, y, z) \quad (409)$$

$$\frac{\partial \mu}{\partial z} = -4(1 - 2z)\mu(x, y, z) \quad (410)$$

sort out mess wrt Eq 26 of busa13

8.6.4 Analytical benchmark IV - "Bercovier & Engelman"

From [146]. The two-dimensional domain is a unit square. The body forces are:

$$\begin{aligned} f_x &= 128[x^2(x-1)^2(2y-1) + 2(y-1)(2y-1)y(12x^2 - 12x + 2)] \\ f_y &= 128[y^2(y-1)^2(2x-1) + 2(x-1)(2x-1)y(12y^2 - 12y + 2)] \end{aligned} \quad (411)$$

The solution is

$$\begin{aligned} u &= -256x^2(x-1)^2y(y-1)(2y-1) \\ v &= 256y^2(y-1)^2x(x-1)(2x-1) \\ p &= 0 \end{aligned} \quad (412)$$

$$du/dx = 512(1-2x)(-1+x)x(-1+y)y(-1+2y) \quad (413)$$

$$du/dy = -256(-1+x)^2x^2(1-6y+6y^2) \quad (414)$$

$$dv/dx = 256y^2(y-1)^2x(x-1)(2x-1) \quad (415)$$

$$dv/dy = -512(-1+x)x(1-2x)(-1+y)y(-1+2y) \quad (416)$$

$$(417)$$

and we can easily verify that $\vec{\nabla} \cdot \vec{v} = du/dx + dv/dy = 0$.

CHECK RHS !

Another choice with a non-zero pressure:

$$\begin{aligned} f_x &= 128[x^2(x-1)^212(2y-1) + 2(y-1)(2y-1)y(12x^2-12x+2)] + y - 1/2 \\ f_y &= 128[y^2(y-1)^212(2x-1) + 2(x-1)(2x-1)y(12y^2-12y+2)] + x - 1/2 \end{aligned} \quad (418)$$

The solution is

$$\begin{aligned} u &= -256x^2(x-1)^2y(y-1)(2y-1) \\ v &= 256y^2(y-1)^2x(x-1)(2x-1) \\ p &= (x-1/2)(y-1/2) \end{aligned} \quad (419)$$

8.6.5 Analytical benchmark V - "VJ1"

This is taken from Appendix D1 of [998].

The domain Ω is a unit square. We consider the stream function

$$\phi(x, y) = 1000x^2(1-x)^4y^3(1-y)^2$$

The velocity field is defined by

$$u(x, y) = \partial_y \phi = 1000(x^2(1-x)^4y^2(1-y)(3-5y)) \quad (420)$$

$$v(x, y) = -\partial_x \phi = 1000(-2x(1-x)^3(1-3x)y^3(1-y)^2) \quad (421)$$

and it is easy to verify that $\vec{\nabla} \cdot \vec{v} = 0$.

The pressure is given by:

$$p(x, y) = \pi^2(xy^3 \cos(2\pi x^2 y) - x^2 y \sin(2\pi x y)) + \frac{1}{8}$$

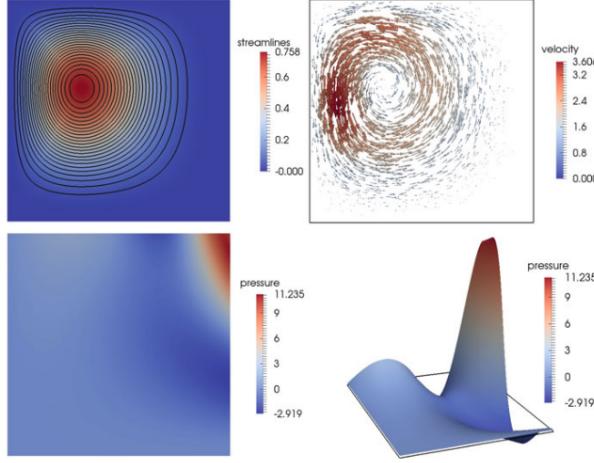


Fig. D.1 Example D.3. Stream function (top left) velocity (top right) and pressure (bottom). These plots are based on results obtained with numerical simulations

Taken from [998].

(BSc Thesis)

8.6.6 Analytical benchmark VI - "Ilinca & Pelletier"

This is taken from [960].

Let us consider the Poiseuille flow of a Newtonian fluid. The channel has isothermal flat walls located at $y = \pm h$. The velocity distribution is parabolic:

$$u = u_0 \left(1 - \frac{y^2}{h^2} \right) \quad v = 0$$

where u_0 is the maximum velocity. The (steady state) temperature field is the solution of the advection-diffusion equation:

$$\rho C_p \vec{v} \cdot \vec{\nabla} T = k \Delta T + \Phi$$

where Φ is the dissipation function given by

$$\Phi = \eta \left[2 \left(\frac{\partial u}{\partial x} \right)^2 + 2 \left(\frac{\partial v}{\partial y} \right)^2 + \left(\frac{\partial v}{\partial x} + \frac{\partial u}{\partial y} \right)^2 \right] = \eta \left(\frac{\partial u}{\partial y} \right)^2 = 4\eta \frac{u_0^2 y^2}{h^4}$$

We logically assume that $T = T(y)$ so that $\partial T / \partial x = 0$ and $\vec{v} \cdot \vec{\nabla} T = 0$. We then have to solve:

$$k \frac{\partial^2 T}{\partial y^2} + 4\eta \frac{u_0^2 y^2}{h^4} = 0$$

We can integrate twice and use the boundary conditions $T(y = \pm h) = T_0$ to arrive at:

$$T(y) = T_0 + \frac{1}{3} \frac{\eta u_0^2}{k} \left[1 - \left(\frac{y}{h} \right)^4 \right]$$

with a maximum temperature

$$T_M = T(y = 0) = T_0 + \frac{1}{3} \frac{\eta u_0^2}{k}$$

8.6.7 Analytical benchmark VII - "grooves"

This benchmark was designed by Dave May. The velocity and pressure fields are given by

$$\begin{aligned} u(x, y) &= x^3 y + x^2 + x y + x \\ v(x, y) &= -\frac{3}{2} x^2 y^2 - 2 x y - \frac{1}{2} y^2 - y \\ p(x, y) &= x^2 y^2 + x y + 5 + p_0 \end{aligned} \tag{422}$$

where p_0 is a constant to be determined based on the type of pressure normalisation. The viscosity is chosen to be

$$\eta(x, y) = -\sin(p) + 1 + \epsilon = -\sin(x^2y^2 + xy + 5) + 1 + \epsilon \quad (423)$$

where ϵ actually controls the viscosity contrast. Note that inserting the polynomial expression of the pressure inside the viscosity expression makes the problem linear. We have

$$\begin{aligned} \dot{\varepsilon}_{xx} &= \frac{\partial u}{\partial x} = 3x^2y + 2x + y + 1 \\ \dot{\varepsilon}_{yy} &= \frac{\partial v}{\partial y} = -3x^2y - 2x - y - 1 \\ \dot{\varepsilon}_{xy} &= \frac{1}{2} \left(\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) = \frac{1}{2} (x^3 + x - 3xy^2 - 2y) \end{aligned} \quad (424)$$

and we can verify that the velocity field is incompressible since $\vec{\nabla} \cdot \vec{v} = \dot{\varepsilon}_{xx} + \dot{\varepsilon}_{yy} = 0$. The pressure gradient is given by

$$\begin{aligned} \frac{\partial p}{\partial x} &= 2xy^2 + y \\ \frac{\partial p}{\partial y} &= 2x^2y + x \end{aligned}$$

The right hand side term of the Stokes equation is such that

$$\begin{aligned} -\frac{\partial p}{\partial x} + \frac{\partial s_{xx}}{\partial x} + \frac{\partial s_{yx}}{\partial y} + f_x &= 0 \\ -\frac{\partial p}{\partial y} + \frac{\partial s_{xy}}{\partial x} + \frac{\partial s_{yy}}{\partial y} + f_y &= 0 \end{aligned} \quad (425)$$

with

$$\begin{aligned} \frac{\partial s_{xx}}{\partial x} &= \frac{\partial(2\eta\dot{\varepsilon}_{xx})}{\partial x} = 2\eta \frac{\partial\dot{\varepsilon}_{xx}}{\partial x} + 2\frac{\partial\eta}{\partial x}\dot{\varepsilon}_{xx} \\ \frac{\partial s_{zx}}{\partial z} &= \frac{\partial(2\eta\dot{\varepsilon}_{zx})}{\partial z} = 2\eta \frac{\partial\dot{\varepsilon}_{zx}}{\partial z} + 2\frac{\partial\eta}{\partial z}\dot{\varepsilon}_{zx} \\ \frac{\partial s_{xz}}{\partial x} &= \frac{\partial(2\eta\dot{\varepsilon}_{xz})}{\partial x} = 2\eta \frac{\partial\dot{\varepsilon}_{xz}}{\partial x} + 2\frac{\partial\eta}{\partial x}\dot{\varepsilon}_{xz} \\ \frac{\partial s_{zz}}{\partial z} &= \frac{\partial(2\eta\dot{\varepsilon}_{zz})}{\partial z} = 2\eta \frac{\partial\dot{\varepsilon}_{zz}}{\partial z} + 2\frac{\partial\eta}{\partial z}\dot{\varepsilon}_{zz} \\ \frac{\partial\eta}{\partial x} &= -z(2xz + 1) \cos(x^2z^2 + xz + 5) \\ \frac{\partial\eta}{\partial z} &= -x(2xz + 1) \cos(x^2z^2 + xz + 5) \\ \frac{\partial\dot{\varepsilon}_{xx}}{\partial x} &= 6xz + 2 \\ \frac{\partial\dot{\varepsilon}_{zx}}{\partial z} &= -3xz - 1 \\ \frac{\partial\dot{\varepsilon}_{xz}}{\partial x} &= \frac{1}{2}(3x^2 + 1 - 3z^2) \\ \frac{\partial\dot{\varepsilon}_{zz}}{\partial z} &= -3x^2 - 1 \end{aligned}$$

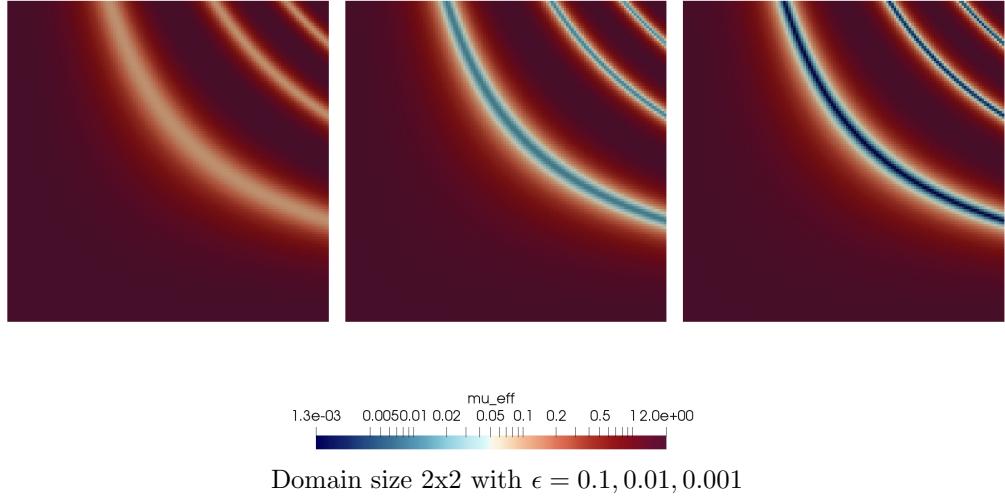
Velocity boundary conditions are prescribed on all four boundaries so that the pressure is known up to a constant (the pressure solution has a nullspace), and the p_0 constant can be determined by requiring that

$$\int_0^L \int_0^L p(x, y) dx dy = \int_0^L \int_0^L (x^2y^2 + xy + 5) dx dy + \int_0^L \int_0^L p_0 dx dy = \int_0^L \int_0^L (x^2y^2 + xy + 5) dx dy + p_0 L^2 = 0$$

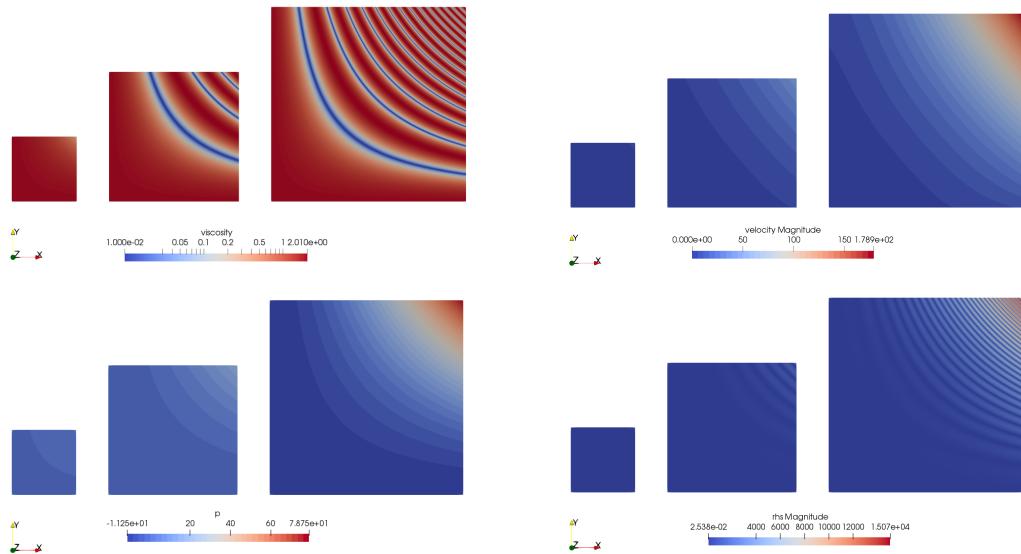
where L is the size of the square domain. Then

$$p_0 = -\frac{1}{L^2} \int_0^L \int_0^L (x^2y^2 + xy + 5) dx dy = -\frac{L^4}{9} - \frac{L^2}{4} - 5$$

As seen in the following figure, the value of ϵ controls the viscosity field amplitude. This is simply explained by the fact that when the sin term of the viscosity takes value 1, the viscosity is then equal to ϵ .



Another interesting aspect of this benchmark is the fact that increasing the domain size adds complexity to it as it increases the number of low viscosity zones and the spacing between them also decreases:



Three different domain sizes (1x1, 2x2, 3x3) with $\epsilon = 0.001$.

Finally, because the analytical expression for both components of the velocity is a polynomial, we can also compute the root mean square velocity exactly. For instance, for a 2x2 domain:

and we end up with (for $L = 2$)

$$v_{rms} = \sqrt{\frac{1}{L^2} \frac{861752}{1575}} = \sqrt{\frac{215438}{1575}} \simeq 11.6955560683$$

(BSc Thesis)

8.6.8 Analytical benchmark VIII - "Kovasznay"

This flow was published by L.I.G. Kovasznay in 1948 [1111]. This paper presents an exact two-dimensional solution of the Navier-Stokes equations with a periodicity in the vertical direction, gives an analytical solution to the steady-state Navier-Stokes equations that is similar which is a flow-field behind a periodic array of cylinders.

$$u(x, y) = 1 - \exp(\lambda x) \cos(2\pi y) \quad v(x, y) = \frac{\lambda}{2\pi} \exp(\lambda x) \sin(2\pi y) \quad \lambda = \frac{Re}{2} - \sqrt{\frac{Re^2}{4} + 4\pi^2}$$

Following step-55 of deal.II²⁸ we have to 'cheat' here since we are not solving the non-linear Navier-Stokes equations, but the linear Stokes system without convective term. Therefore, to recreate the exact same solution we move the convective term into the right-hand side.

The analytical solution is prescribed left and right, while free/no (?) slip is prescribed at top and bottom.

Solution as implemented in step-55:

```
const double pi2 = pi*pi;
u = -exp(x*(-sqrt(25.0 + 4*pi2) + 5.0))*cos(2*y*pi) + 1;
v = (1.0L/2.0L)*(-sqrt(25.0 + 4*pi2) + 5.0)*exp(x*(-sqrt(25.0 + 4*pi2) + 5.0))*sin(2*y*pi)/pi;
p = -1.0L/2.0L*exp(x*(-2*sqrt(25.0 + 4*pi2) + 10.0))
- 2.0*(-6538034.74494422 + 0.0134758939981709*exp(4*sqrt(25.0 + 4*pi2)))/(-80.0*exp(3*sqrt(25.0 + 4*pi2)))
+ 16.0*sqrt(25.0 + 4*pi2)*exp(3*sqrt(25.0 + 4*pi2)))
- 1634508.68623606*exp(-3.0*sqrt(25.0 + 4*pi2))/(-10.0 + 2.0*sqrt(25.0 + 4*pi2))
+ (-0.00673794699908547*exp(sqrt(25.0 + 4*pi2)))
+ 3269017.37247211*exp(-3*sqrt(25.0 + 4*pi2))/(-8*sqrt(25.0 + 4*pi2) + 40.0)
+ 0.00336897349954273*exp(1.0*sqrt(25.0 + 4*pi2))/(-10.0 + 2.0*sqrt(25.0 + 4*pi2));
```

8.6.9 Analytical benchmark IX - "VJ2"

It is presented in [1000] and meant to be a peculiar case where the velocity solution is exactly zero. The viscosity is 1, the domain is a unit square, no-slip boundary conditions are prescribed everywhere. The buoyancy force is given by $\vec{b} = (0, Ra(1 - y + 3y^2))$ where $Ra > 0$ is a parameter. The flow is incompressible and the analytical pressure solution is given by $p = Ra(y^3 - y^2/2 + y - 7/12)$.

8.6.10 Analytical benchmark X - "VJ3"

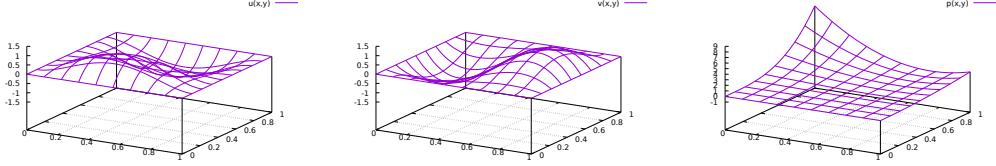
This benchmark comes from John et al. [1000]. The domain is once again the unit square. The velocity field has the form of a large vortex.

²⁸https://www.dealii.org/current/doxygen/deal.II/step_55.html

$$u(x, y) = 200x^2(1-x)^2y(1-y)(1-2y) \quad (426)$$

$$v(x, y) = -200x(1-x)(1-2x)y^2(1-y)^2 \quad (427)$$

$$p(x, y) = 10 [(x-1/2)^3y^2 + (1-x)^3(y-1/2)^3] \quad (428)$$



$$\dot{\varepsilon}_{xx} = \frac{\partial u}{\partial x} = -400(1-x)x(2x-1)(y-1)y(2y-1) \quad (429)$$

$$\frac{\partial u}{\partial y} = 200(1-x)^2x^2(6y^2-6y+1) \quad (430)$$

$$\frac{\partial v}{\partial x} = -200(6x^2-6x+1)(1-y)^2y^2 \quad (431)$$

$$\dot{\varepsilon}_{yy} = \frac{\partial v}{\partial y} = 400(x-1)x(2x-1)(1-y)y(2y-1) \quad (432)$$

so that

$$\begin{aligned} \dot{\varepsilon}_{xy} &= \frac{1}{2} [200(1-x)^2x^2(6y^2-6y+1) - 200(6x^2-6x+1)(1-y)^2y^2] \\ &= 100(1-x)^2x^2(6y^2-6y+1) - 100(6x^2-6x+1)(1-y)^2y^2 \end{aligned} \quad (433)$$

Also

$$\begin{aligned} \frac{\partial \dot{\varepsilon}_{xx}}{\partial x} &= 400(6x^2-6x+1)y(2y^2-3y+1) \\ \frac{\partial \dot{\varepsilon}_{xy}}{\partial x} &= 200(-2x^2(1-x)(6y^2-6y+1) + 2x(1-x)^2(6y^2-6y+1) - 6(2x-1)(1-y)^2y^2) \\ &= 100(-2x^2(1-x)(6y^2-6y+1) + 2x(1-x)^2(6y^2-6y+1) - 6(2x-1)(1-y)^2y^2) \\ \frac{\partial \dot{\varepsilon}_{xy}}{\partial y} &= 400(6x^2-6x+1)(1-y)y^2 + 200(1-x)^2x^2(12y-6) - 400(6x^2-6x+1)(1-y)^2y \\ \frac{\partial \dot{\varepsilon}_{yy}}{\partial y} &= -400x(2x^2-3x+1)(6y^2-6y+1) \end{aligned} \quad (434)$$

$$\frac{\partial p}{\partial x} = 30(x-1/2)^2y^2 - 30(1-x)^2(y-1/2)^3 \quad (435)$$

$$\frac{\partial p}{\partial y} = 20(x-1/2)^3y + 30(1-x)^3(y-1/2)^2 \quad (436)$$

From $\vec{\nabla} \cdot \boldsymbol{\sigma} + \vec{b} = \vec{0}$ we can obtain the rhs as follows:

$$\begin{aligned} \vec{b} &= -\vec{\nabla} \cdot \boldsymbol{\sigma} \\ &= \vec{\nabla} p - \vec{\nabla} \cdot \boldsymbol{s} \\ &= \vec{\nabla} p - \vec{\nabla} \cdot (2\eta \dot{\varepsilon}) \end{aligned} \quad (437)$$

Assuming $\eta = 1$ we arrive at:

$$b_x = \frac{\partial p}{\partial x} - 2 \frac{\partial \dot{\varepsilon}_{xx}}{\partial x} - 2 \frac{\partial \dot{\varepsilon}_{xy}}{\partial y} \quad (438)$$

$$b_y = \frac{\partial p}{\partial y} - 2 \frac{\partial \dot{\varepsilon}_{xy}}{\partial x} - 2 \frac{\partial \dot{\varepsilon}_{yy}}{\partial y} \quad (439)$$

All the necessary functions to do this benchmark are in `mms/vj3.py`:

```

# functions for the Volker John III benchmark (vj3)

def u_th(x,y):
    return 200*x**2*(1-x)**2*y*(1-y)*(1-2*y)

def v_th(x,y):
    return -200*x*(1-x)*(1-2*x)*y**2*(1-y)**2

def p_th(x,y):
    return 10*( (x-1./2.)**3*y**2+(1-x)**3*(y-1./2.)**3 )

def dpdx_th(x,y):
    return 30*(x-1./2.)**2*y**2-30*(1-x)**2*(y-1./2.)**3

def dpdy_th(x,y):
    return 20*(x-1./2.)**3*y + 30*(1-x)**3*(y-1./2.)**2

def exx_th(x,y):
    return -400*(1-x)*x*(2*x-1)*(y-1)*y*(2*y-1)

def exy_th(x,y):
    return 100*(1-x)**2*x**2*(6*y**2-6*y+1)-100*(6*x**2-6*x+1)*(1-y)**2*y**2

def eyy_th(x,y):
    return 400*(x-1)*x*(2*x-1)*(1-y)*y*(2*y-1)

def dexxdx(x,y):
    return 400*(6*x**2-6*x+1)*y*(2*y**2-3*y+1)

def dexydx(x,y):
    return 100*(-2*x**2*(1-x)*(6*y**2-6*y+1) + 2*x*(1-x)**2*(6*y**2-6*y+1) - 6*(2*x-1)*(1-y)**2*y**2)

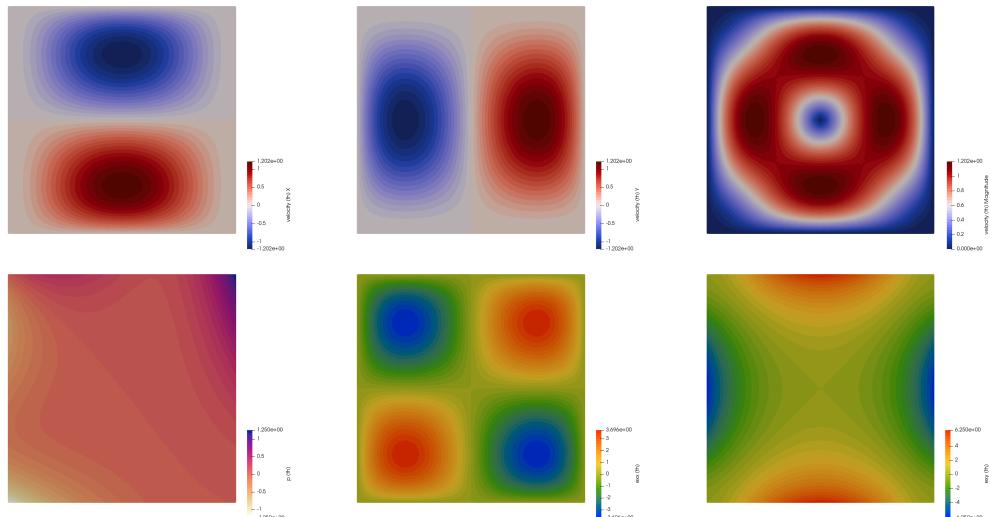
def dexydy(x,y):
    return 200*(6*x**2-6*x+1)*(1-y)*y**2 + 100*(1-x)**2*x**2*(12*y-6) - 200*(6*x**2-6*x+1)*(1-y)**2*y

def deyydy(x,y):
    return -400*x*(2*x**2-3*x+1)*(6*y**2-6*y+1)

def bx(x,y):
    return dpdx_th(x,y)-2*dexxdx(x,y)-2*dexydy(x,y)

def by(x,y):
    return dpdy_th(x,y)-2*dexydx(x,y)-2*deyydy(x,y)

```



8.6.11 Analytical benchmark XI - "PPC1"

8.6.12 Analytical benchmark XII - "PPC2"

8.6.13 Annulus with kinematical b.c.

The domain is a hollow cylinder or inner radius $R_i = 1$ and outside radius $R_o = 1$. Boundary conditions are prescribed both on the inside and the outside with $\vec{v} = (u, v) = (-y, x)$, or in polar coordinates $\vec{v} = r\vec{e}_\theta$.

The gravity is radial and is set to

$$g_x = -x/r \quad g_z = -y/r$$

where $r = \sqrt{x^2 + z^2}$, which in polar coordinates is $\vec{g} = -\vec{e}_r$. The viscosity is also set to 1, and the density is given by

$$\rho(r) = r^n$$

where n is a positive or null integer. The pressure is set to zero at the outer boundary.

The gradient operator in polar coordinates writes:

$$\vec{\nabla} = \frac{\partial}{\partial r}\vec{e}_r + \frac{1}{r}\frac{\partial}{\partial \theta}\vec{e}_\theta$$

and the Laplacian operator:

$$\Delta = \frac{\partial^2}{\partial r^2} + \frac{1}{r}\frac{\partial}{\partial r} + \frac{1}{r^2}\frac{\partial^2}{\partial \theta^2}$$

Note that in our case we need to take the Laplacian of a vector, and unfortunately the Laplacian of a vector is not the Laplacian of the vector's coordinates in polar coordinates (unlike cartesian coordinates). The Laplacian of a vector is given by²⁹

$$\nabla^2 \vec{A} = \nabla(\nabla \cdot \vec{A}) - \nabla \times (\nabla \times \vec{A}) = \begin{pmatrix} \frac{\partial^2 A_r}{\partial r^2} + \frac{1}{r}\frac{\partial A_r}{\partial r} - \frac{1}{r^2}A_r + \frac{1}{r^2}\frac{\partial^2 A_r}{\partial \theta^2} - \frac{2}{r^2}\frac{\partial A_\theta}{\partial \theta} \\ \frac{\partial^2 A_\theta}{\partial r^2} + \frac{1}{r}\frac{\partial A_\theta}{\partial r} - \frac{1}{r^2}A_\theta + \frac{1}{r^2}\frac{\partial^2 A_\theta}{\partial \theta^2} + \frac{2}{r^2}\frac{\partial A_r}{\partial \theta} \end{pmatrix} = \begin{pmatrix} \Delta A_r \\ \Delta A_\theta \end{pmatrix}$$

The Stokes equation writes:

$$\mu \Delta \vec{v} + \rho \vec{g} = \vec{0}$$

The velocity solution is expected to be $\vec{v} = r\vec{e}_\theta$. The Stokes equation in polar coordinates then writes:

$$\begin{aligned} -\frac{\partial p}{\partial r} + \Delta v_r + \rho(r)(-1) &= 0 \\ -\frac{1}{r}\frac{\partial p}{\partial \theta} + \Delta v_\theta &= 0 \end{aligned}$$

Since $\Delta v_\theta = 0$, then $\frac{\partial p}{\partial \theta} = 0$ and then the pressure is independent of θ , which is what we expect since the density distribution is radial. We then focus on the first equation, and since $v_r = 0$, we then obtain:

$$\frac{\partial p}{\partial r} = -\rho(r)$$

- If $\rho(r) = 1$, then

$$\frac{\partial p}{\partial r} = -1$$

yields $p(r) = -r + C$ where C is a constant determined by means of b.c. ($p(r=1) = 0$) so finally

$$p(r) = 1 - r$$

²⁹https://en.wikipedia.org/wiki/Vector_Laplacian

- If $\rho(r) = r$, then

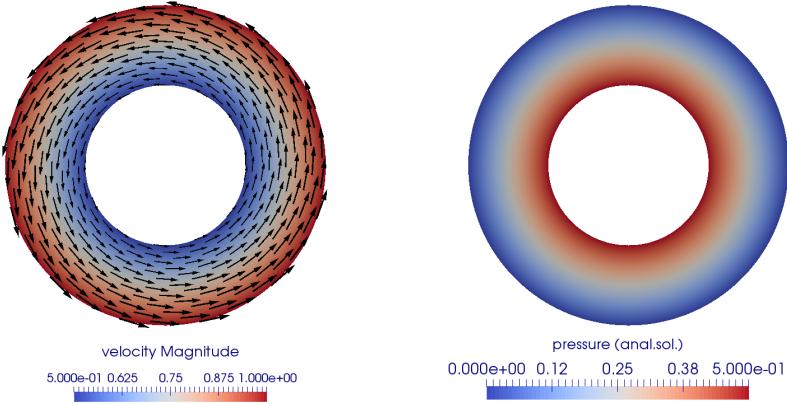
$$\frac{\partial p}{\partial r} = -r$$

so that $p(r) = -\frac{1}{2}r^2 + C$ and likewise

$$p(r) = \frac{1}{2}(1 - r^2)$$

In general, by taking $\rho(r) = r^n$ with $n = 0, 1, \dots$ one arrives to a pressure field given by

$$p(r) = \frac{1}{n+1}(1 - r^{n+1})$$



This benchmark is of course very simple and the fact that the solution is independent of θ renders it not so useful. It has successfully been implemented in ELEFANT.

8.7 Geodynamical benchmarks

Some published numerical experiments have over time become benchmarks for other codes, while some others showcased comparisons between codes. Here is a short list of 'famous' benchmarks' in the computational geodynamics community.

- the plastic brick [1150, 1030, 1529, 1323, 1381, 1248, 1685, 766, 651]
- 2D Rayleigh-Benard convection (Blankenbach) [175, 1794, 363, 1060, 1169, 1892, 1791, 151, 385]
- 2D Rayleigh-Benard convection with nonlinear rheology [1784, 77]
- 2D Rayleigh-Benard laminar plumes [1869]
- 2D Rayleigh-Taylor convection/instability [1505, 1791, 1922, 1491, 1674, 57, 1728, 200] [87, 1529, 1619, 423, 1169, 1225, 1892, 1847, 200, 366, 524, 1323, 1248] [670, 510, 487, 508, 77]
- Thin layer entrainment (see Section 8.7.12)
- 3D Rayleigh-Taylor instability [684, 1890]
- subduction problems [1638, 1845, 345]
- numerical sandbox [260, 1248, 268, 766]
- the Stokes sphere [1129, 77], in visco-plastic fluid [1208, 501]. Finite deformation in and around a fluid sphere [1641, 458].
- the sinking block (sinker) [1756, 345, 722, 742, 1290, 1323, 686, 1248] (see Section 8.7.7)
- multiple sinkers [1292, 1289]
- 1D compression [1348]
- 2D compressible Stokes flow problem [968, 1734, 1167, 1054, 1219]
- 3D convection at infinite Prandtl number (Busse) [313, 1794, 1434, 1119]
- Free surface evolution [442, 77]
- Love's problem [105]
- Poiseuille flow [633, 674, 1732]
- Couette flow with temperature dependent viscosity [568, 508]
- Couette flow with shear heating [568]
- Poiseuille-Couette flow [670]
- Lid driven cavity [637, 750, 195, 1039, 251, 1109, 582]
- Lid driven cavity with analytical solution (see Section 8.7.10)
- Lid driven cavity with nonlinear rheology [147]
- Wannier flow [1905, 1976, 345]
- bending of elastic plate/beam [345, 186, 1890, 568, 508, 1348, 1212]
- flexure of finite length elastic plate [366]
- thermal diffusion of half-cooling space (see Section 8.7.14)
- thermal diffusion of Gaussian distribution (see compgeo notes, elefant manual)
- stress build-up in Maxwell visco-elastic material [721, 366, 568, 508]

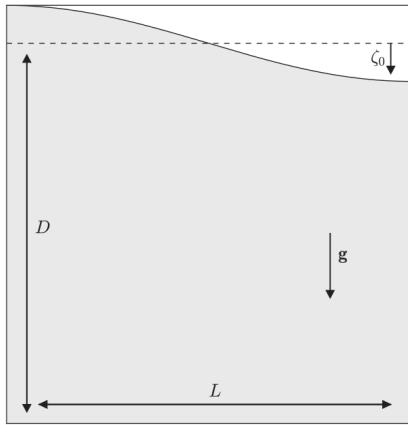
- plastic oedometer test [366]
- SolCx [1290, 508, 77]
- SolKz [1290, 508, 77]
- SolVi, inclusion [1034, 1248, 524, 151, 1890, 508, 77, 1212]
- channel flow (nonlinear) [1248, 651, 722, 568] ((**BSc Thesis**))
- indentor, punch problem (see Section 8.7.9) See also [935, 642, 710] for application.
- relaxation of sinusoidal topography [442, 1599]
- single layer visco-elastic folding [1890]
- Three-dimensional folding of an embedded viscous layer in pure shear [628]
- dam-break problem [1332, 73, 1198, 1146, 912, 26, 816, 905, 87]
- hot blob problem [307, 686] (see Section 8.7.8)
- viscous(-elastic) flow around a cylinder in a channel (see Section ??)
- Infinite plate with a circular hole [1534]
- Slope stability for elasto-plastic materials [1534]
- Time-dependent flow in an annulus [699] (see Section 8.7.5)
- Convection in 2D-box [699] (see Section 8.7.6)
- Onset of convection [77]
- Polydiapirism [1922, 77]
- Slab detachment benchmark (see Section 8.7.16)
- Hollow sphere benchmark [1758]
- Annulus benchmark [77], [1487]
- Viscosity grooves benchmark [77]
- Latent heat benchmark [77]
- Layered flow with viscosity contrast [77] (see Section 8.7.17)
- Brittle thrust wedges benchmark [268, 77]
- 2D linear viscous subduction [1638, 766]
- mantle convection in 3D spherical shell [1546, 2014, 1978, 364, 2010, 1013, 486, 46, 1218]
- Benchmark of 3D numerical models of subduction against a laboratory experiment [1311]
- 3D subduction [1444]
- heat flow around a cylinder (see Section 8.7.13)
- Laplace equation on a semi infinite plate (see Section 8.7.15)
- 2D Stokes flow over cavity [1500]
- fractal networks of shear bands [1492]
- square plate with a crack subjected to a horizontal tensile traction [1212]

- analytical solution for solitary porosity waves [421]
- analytical solution for solitary wave of magma [472] and refs therein
- Stokes flow caused by the motion of a rigid sphere close to a viscous interface [475]

go through my papers and add relevant ones here

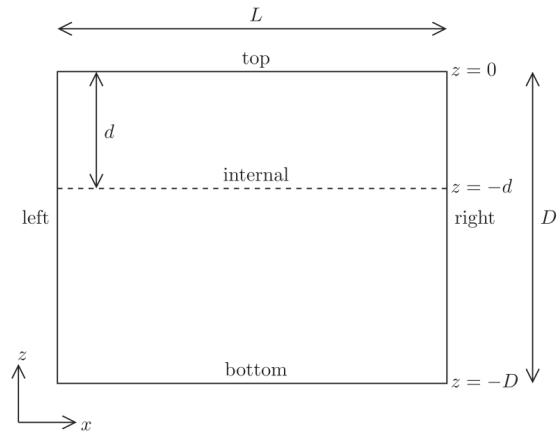
8.7.1 Relaxation of sinusoidal topography

Following Kramer et al. [1115, Section 3.1.1] and [1599] the benchmark consists of the relaxation of surface topography in a two-dimensional Cartesian box with an isoviscous fluid. Free slip boundary conditions are imposed on the sides and bottom of the domain. The setup is as follows:



Taken from [1599]. Setup for the free surface relaxation benchmark.

For the tests $\rho = \eta = g = L = D = 1$ and $\xi_0 = 0.005$.



Taken from [1115]. $D = 3 \cdot 10^6$, $\eta = 10^{21}$, $\rho = 4500$, $g = 10$,

$\xi_0 = 10^3$ m, and $L = D/4, D/2, D, 2D, 4D$.

and the infinitesimal sinusoidal perturbations to the free surface is given by

$$\xi(x, t = 0) = \xi_0 \cos\left(\frac{2\pi n x}{L}\right)$$

where n is a wavenumber which is an integer multiple of $1/2$ (taken to be $1/2$ exactly in both cases).

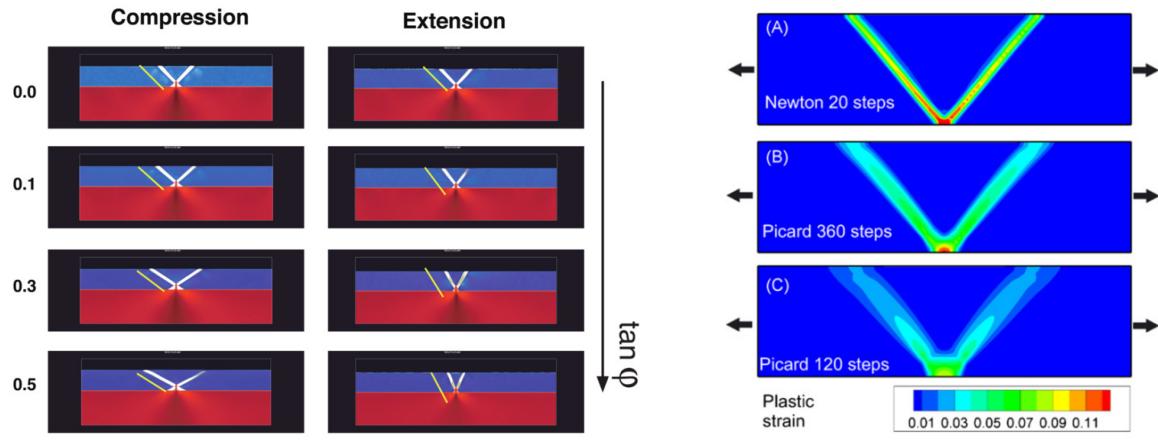
8.7.2 the plastic brick

Literature[859, 1352, 1150, 1030, 568, 1529, 1323, 1248, 1685, 766, 651, 77]

Pretty much all of the brick-type (elasto-)visco-plastic experiments in the literature introduce a weak seed at the bottom of the domain to seed deformation (the shear bands will ultimately stem from it). Dimensioned and dimensionless experiments have been carried out, with or without elastic behaviour, with or without adaptive mesh refinement, with first order and second order quadrilateral elements or Taylor-Hood triangles, with or without Newton algorithm, in extension and compression, with or without time-stepping, with or without viscous lower layer.

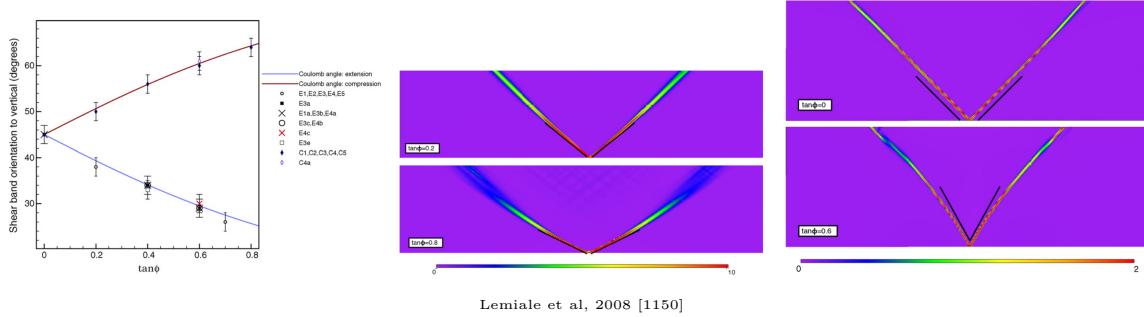


Moresi & Mühlhaus, 2006 [1351]

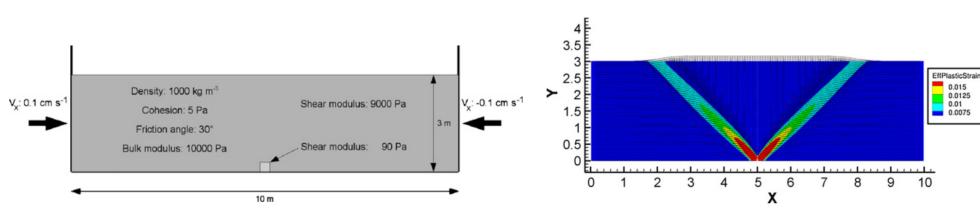


Moresi et al, 2007 [1352]

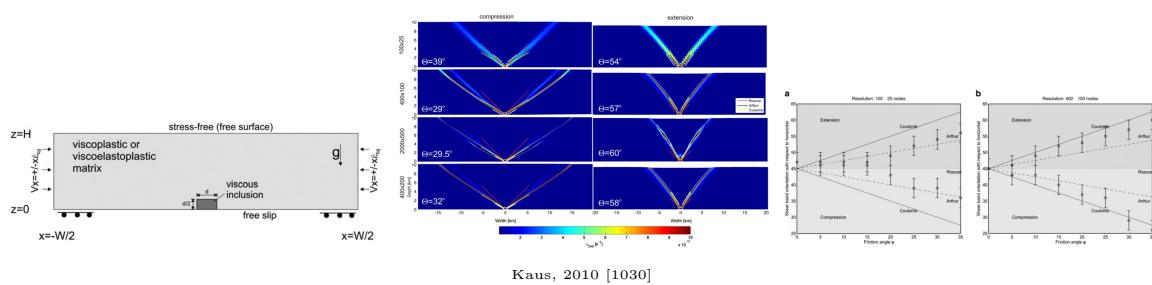
Popov et al, 2008 [1498]



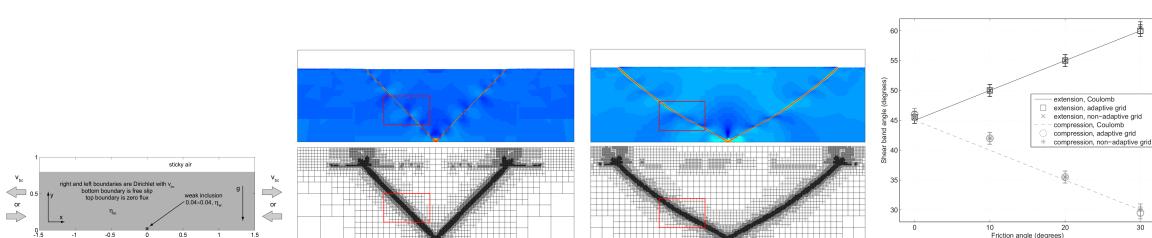
Lemiale et al, 2008 [1150]



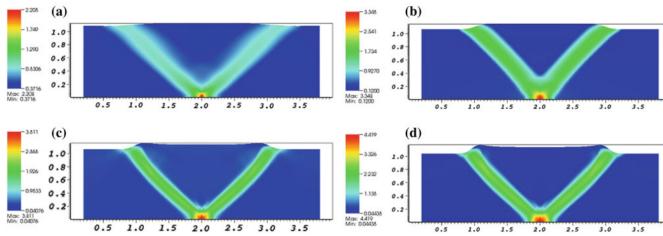
Quinteros et al., 2009 [1529]



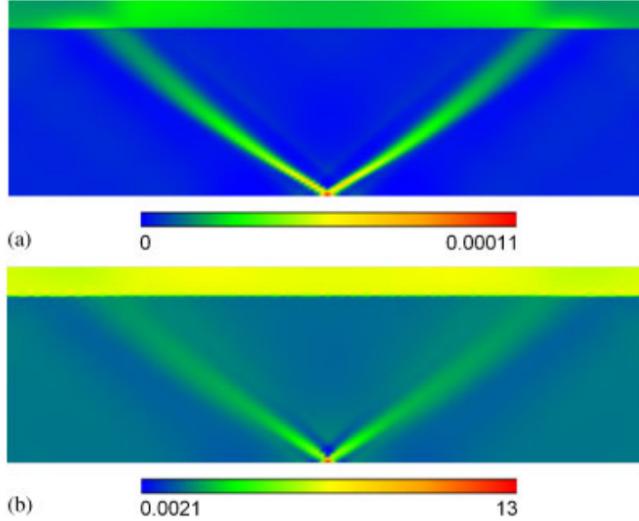
Kaus, 2010 [1030]



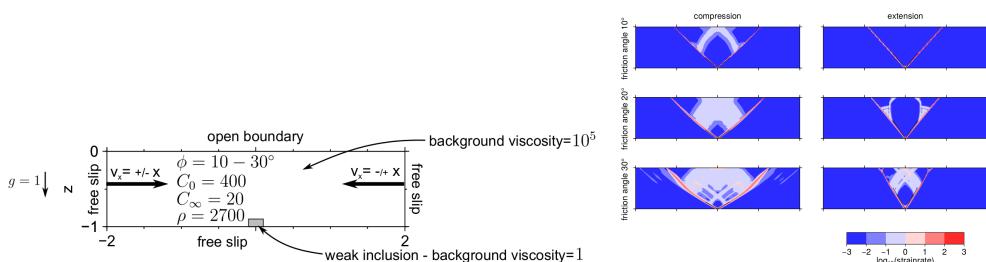
Mishin, phd thesis, 2011 [1323]



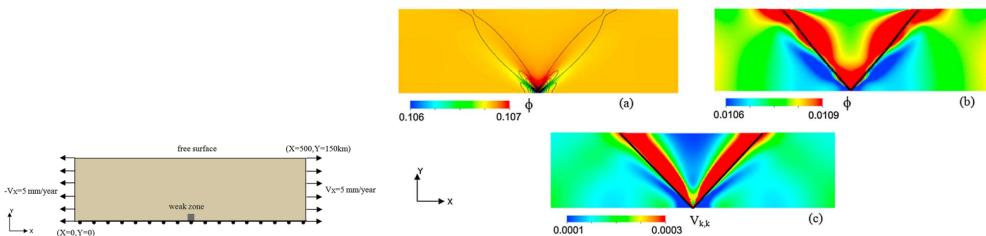
Mühlhaus et al, 2011 [1381].



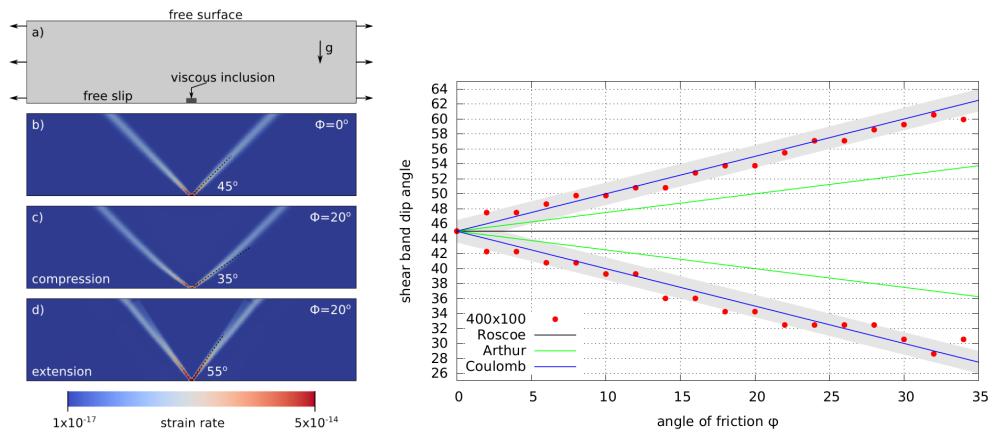
Lemiale et al, 2011 [1151].



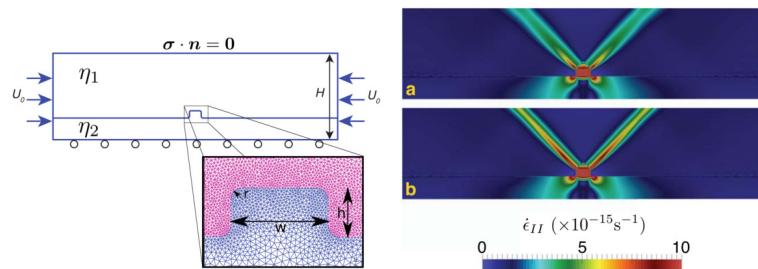
Maierova, phd thesis, 2012 [1248]



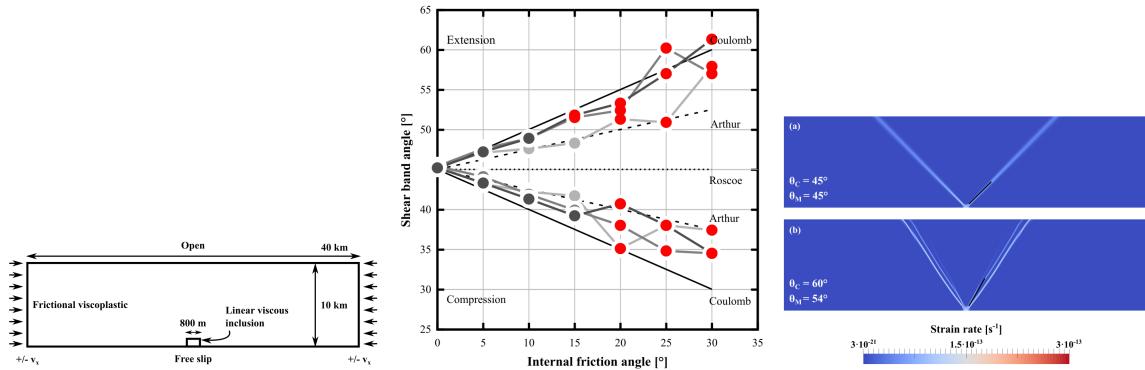
Mohajeri et al, 2013 [1331].



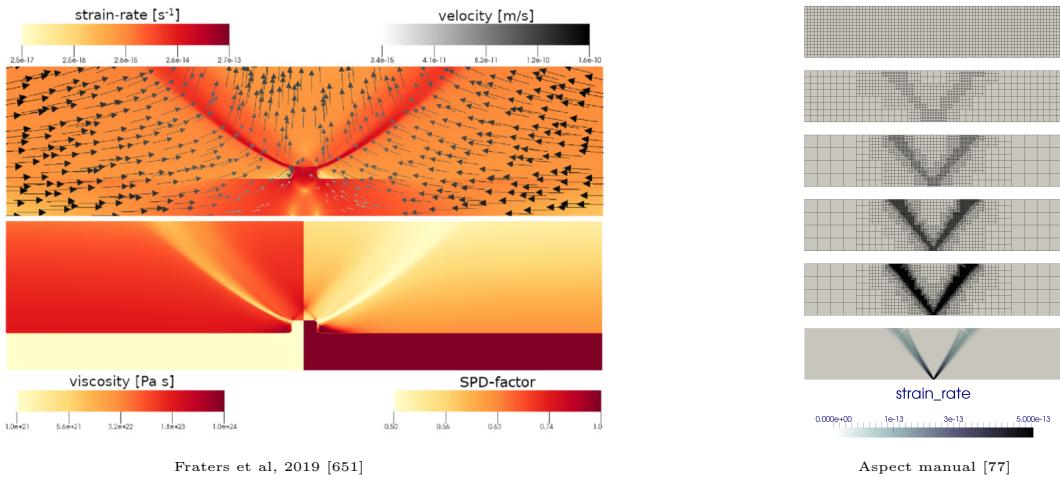
Thieulot, 2014 [1757].



Spiegelman et al, 2016 [1685]



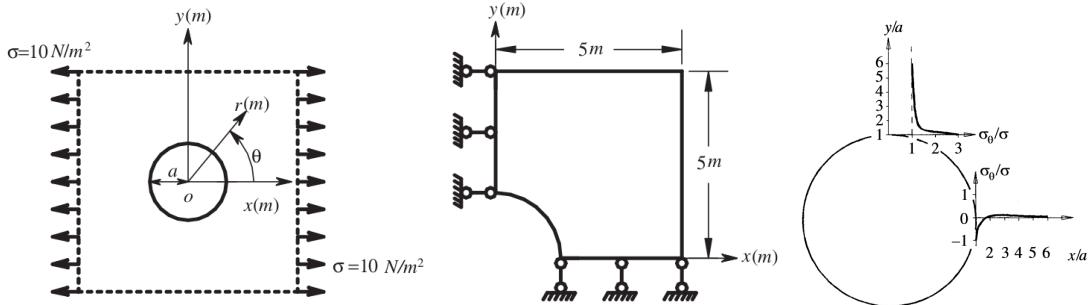
Glerum et al, 2018 [766]



8.7.3 Infinite plate with a circular hole [1534]

(MSc Thesis)

An infinite plate with a circular hole of radius a is subjected to a unidirectional tensile load of σ in the x direction as shown in the figure. In this case, only one quarter of the domain is analysed due to symmetry along x and y axis.



Left: An infinite plate with a circular hole subjected to unidirectional tension and its quarter model with symmetric conditions imposed on the left and bottom edges. Right: Tangential stress distribution for $\theta = 0$ and $\theta = \pi/2$. [1982]

The inner boundary of the hole is traction free and the right edge was imposed with the tractions based on the analytical solutions. The left edge is constrained in the x direction and the bottom edge is constrained in the y direction, respectively. The plane stress condition is considered and the parameters are: Young modulus $E = 3e7$ MPa, Poisson Ratio $\nu = 0.3$, Load $\sigma = 10$ N/m², $a = 1$ m.

The analytical stress components for this problem are

$$\sigma_{xx}(x, y) = \sigma \left(1 - \frac{a^2}{r^2} \left(\frac{3}{2} \cos 2\theta + \cos 4\theta \right) + \frac{3a^4}{2r^4} \cos 4\theta \right) \quad (440)$$

$$\sigma_{yy}(x, y) = -\sigma \left(\frac{a^2}{r^2} \left(\frac{1}{2} \cos 2\theta - \cos 4\theta \right) - \frac{3a^4}{2r^4} \cos 4\theta \right) \quad (441)$$

$$\sigma_{xy}(x, y) = -\sigma \left(\frac{a^2}{r^2} \left(\frac{1}{2} \sin 2\theta + \sin 4\theta \right) + \frac{3a^4}{2r^4} \sin 4\theta \right) \quad (442)$$

Note that [1534] cites [351] which cites the book [1982, p772] which cites [257] for the solution!

there are discrepancies between [1534] and [351]

Following [1982], it can be shown, from linear elasticity, that the tangential stress throughout the plate is given by

$$\sigma_\theta = \frac{\sigma}{2} \left[1 + \frac{a^2}{r^2} - \left(1 + 3 \frac{a^4}{r^4} \right) \cos 2\theta \right]$$

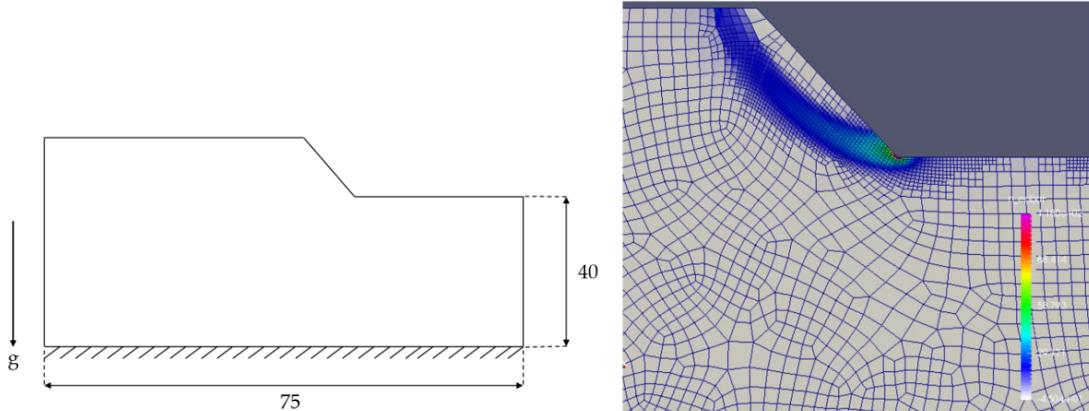
The maximum stress is $\sigma_\theta = 3\sigma$ at $r = a$ and $\theta = \pm\pi/2$. Along the surface of the hole, the tangential stress is $-\sigma$ at $\theta = 0$ and $\theta = \pi$, and increases, as θ increases, to 3σ at $\theta = \pi/2$ and $\theta = 3\pi/2$.

8.7.4 Slope stability for elasto-plastic materials a la [1534]

(MSc Thesis)

The bottom of the domain is constrained and the model is subjected to gravitational load. The material properties considered are

Young modulus $20e3$ MPa, Poisson Ratio 0.49, Constitutive law Mohr-Coulomb (friction angle $\phi = 20^\circ$, dilatancy angle $\phi = 20^\circ$, cohesion $c=50$ MPa).



Left: Slope stability problem setup; Right: Adaptive Refinement based on Plasticity Indicator

8.7.5 Time-dependent benchmark in an annulus

This benchmark is presented in Gassmöller et al [699]. The domain is a 2D annulus with inner and outer radii $R_1 = 1$ and $R_2 = 2$, respectively. In this situation, the incompressible isothermal Stokes equations and their solution can be expressed in a cylindrical coordinate system in terms of the radius r and the azimuthal angle θ . The viscosity is set to $\eta = 1$, and the density is given by

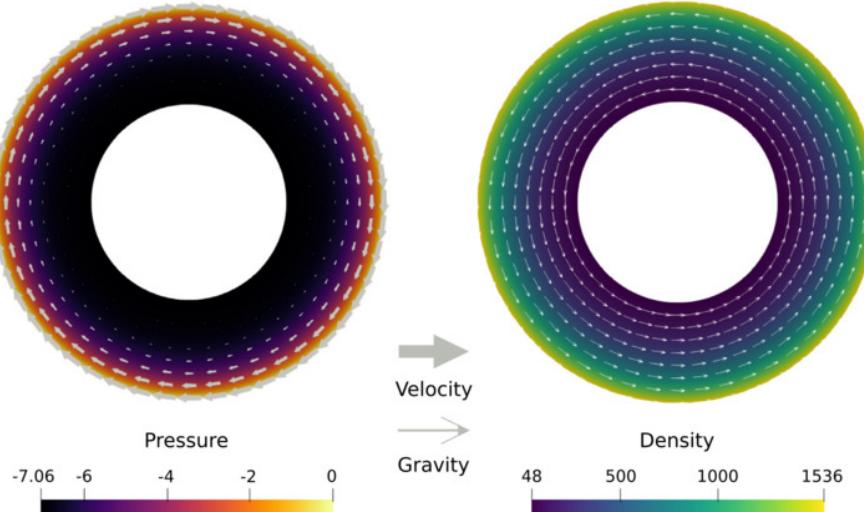
$$\rho(r, \theta) = 48r^5 \quad (443)$$

The gravity vector is set to

$$\vec{g}(r, \theta) = \frac{r^3}{384} \vec{e}_r + \vec{e}_\theta \quad (444)$$

Note that this gravity vector is not the gradient of a gravity potential and consequently not physical. The Stokes system can then be solved using a separation of variables approach and yields

$$\vec{v} = -r^7 \vec{e}_\theta \quad p(r, \theta) = \frac{r^9}{72} - \frac{512}{72} \quad (445)$$



Taken from [699]

Rather importantly, this benchmark was arrived at by means of a stream function (see Section 8.34) $\psi(r, \theta) = F(r)G(\theta)$ with $F(r) = r^8/8$ and $G(\theta) = 1$.

8.7.6 Convection in 2D-box

We start from the following stream function (see Section 8.34):

$$\psi(x, y) = \frac{1}{\pi} \sin \pi x \sin \pi y \quad (446)$$

which yields:

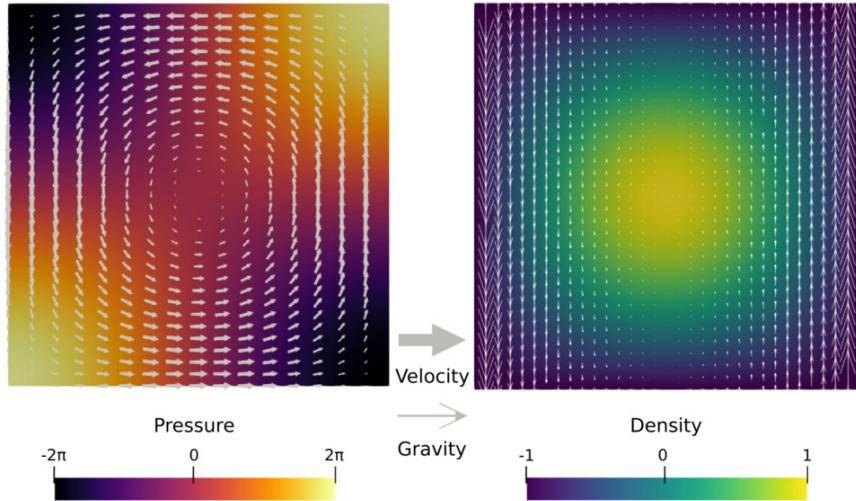
$$\begin{aligned} u(x, y) &= \frac{\partial \psi}{\partial y} = \sin \pi x \cos \pi y \\ v(x, y) &= -\frac{\partial \psi}{\partial x} = -\cos \pi x \sin \pi y \end{aligned} \quad (447)$$

The pressure field is

$$p(x, y) = 2\pi \cos(\pi x) \cos(\pi y) \quad (448)$$

with

$$\rho(x, y) = \sin(\pi x) \sin(\pi y) \quad g_y = -4\pi^2 \frac{\cos(\pi x)}{\sin(\pi x)} \quad (449)$$



Taken from [699]

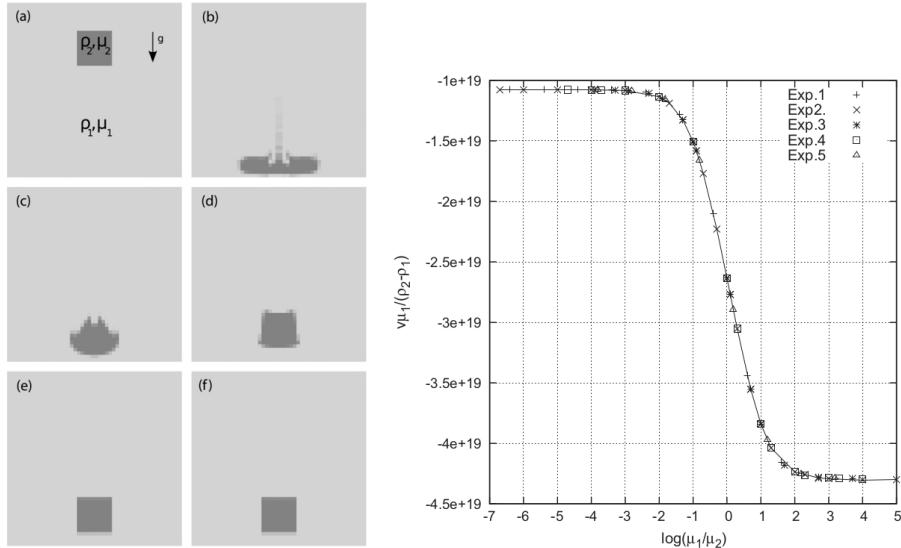
$$\begin{aligned} v_{rms} &= \sqrt{\frac{1}{L_x L_y} \int_0^1 \int_0^1 (u^2 + v^2) dx dy} \\ &= \sqrt{\int_0^1 \int_0^1 (\sin^2(\pi x) \cos^2(\pi y) + \cos^2(\pi x) \sin^2(\pi y)) dx dy} \\ &= \sqrt{\int_0^1 \sin^2(\pi x) dx \cdot \int_0^1 \cos^2(\pi y) dy + \int_0^1 \cos^2(\pi x) dx \cdot \int \sin^2(\pi y) dy} \\ &= \sqrt{\frac{1}{2} \frac{1}{2} + \frac{1}{2} \frac{1}{2}} \\ &= \frac{\sqrt{2}}{2} \\ &\simeq 0.70711... \end{aligned} \quad (450)$$

8.7.7 The sinker problem

This experiment is not a benchmark stricto sensu since there is no analytical solution. However, it is widely used in the technical literature because of its simple setup and since it allows to test solving strategies. Also, it can conveniently be carried out in both two and three dimensions.

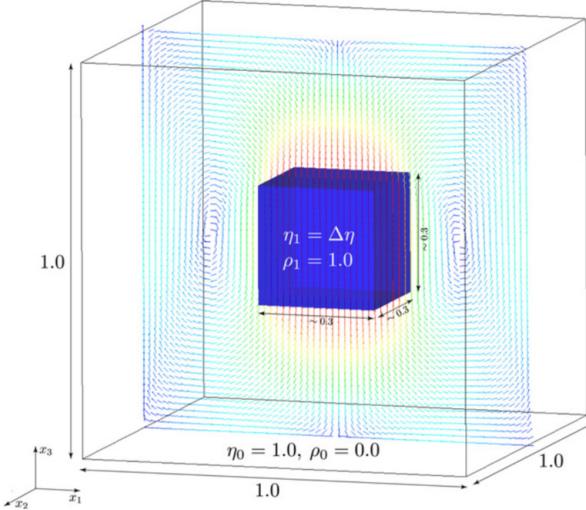
In two dimensions The time dependent version of the experiment is for instance to be found in Gerya [722] and the same is repeated in Thieulot [1756].

This simple benchmark provides challenging numerical experiments dealing with large viscosity variations within the simulation domain. It consists of a bulk of fluid 1 (η_1, ρ_1) in which a block of fluid 2 (η_2, ρ_2) falls under its own weight. The domain is a square of size $L_x = L_y = 500$ km and the block is initially centred at point ($x = 250$ km, $y = 400$ km) with size 100×100 km. Free slip boundary conditions are imposed on all sides of the domain. In [1756] five experiments have been conducted: $\eta_1 = 10^{20}$ Pa.s, $\rho_2 = 3220$ kg/m³; $\eta_1 = 10^{21}$ Pa.s, $\rho_2 = 3300$ kg/m³; $\eta_1 = 10^{22}$ Pa.s, $\rho_2 = 6600$ kg/m³; $\eta_1 = 10^{23}$ Pa.s, $\rho_2 = 3300$ kg/m³; $\eta_1 = 10^{24}$ Pa.s, $\rho_2 = 9900$ kg/m³, while in all experiments the density of the surrounding fluid is $\rho_1 = 3200$ kg/m³ and the viscosity of the block is varied between 10^{19} and $5 \cdot 10^{27}$ Pa.s.



Left: $\eta_1 = 10^{21}$ Pa.s, $\rho_2 = 3300$ kg/m³. (a) Initial setup; (b) $\eta_1 = 10^{21}$ Pa.s at time $t = 10$ Myrs; (c) $\eta_1 = 10^{22}$ Pa.s at time $t = 20$ Myrs; (d) $\eta_1 = 10^{23}$ Pa.s at time $t = 20$ Myrs; (e) $\eta_1 = 10^{25}$ Pa.s at time $t = 20$ Myrs; (f) $\eta_1 = 10^{27}$ Pa.s at time $t = 20$ Myrs. Right: Velocity measurements as a function of the viscosity contrast between surrounding medium and block for all experiments. Taken from [1756]

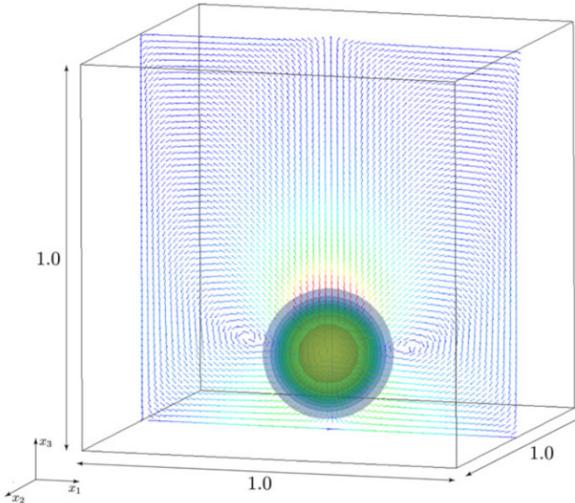
In three dimensions Let us look at the sinker experiment from Furuichi et al [686]: The domain is the unit box the origin at the center of the box. A cube with a viscosity $\eta_1 = \Delta\eta$ and density $\rho_1 = 1$ was placed at the middle of the domain defined by $-0.15 \leq x, y, z \leq 0.15$. The material surrounding the cube has the properties $\eta_0 = 1$ and $\rho_0 = 0$. The body force of the momentum equation was taken as $(0, 0, -\rho g)$ with $g = 1$. Along all walls on the domain, free-slip boundary conditions were employed.



Simulation setup for the 3D falling block (SINKER) problem. The vectors represent computed flow. Taken from [686]

8.7.8 The hot blob problem

This is a very similar setup as the 3D sinker from the same authors with higher but more diffusive variation of viscosity. The body force is given by $(0, 0, \beta T)$ and where the temperature field T is defined by $T = \exp(-\gamma(x^2 + y^2 + (z - 0.3)^2))$ with the constant parameters $\beta = 10^6$ and $\gamma = 200$. The temperature-dependent viscosity $\eta = \exp(-\alpha T)$ is employed with the parameter for viscosity contrast α .

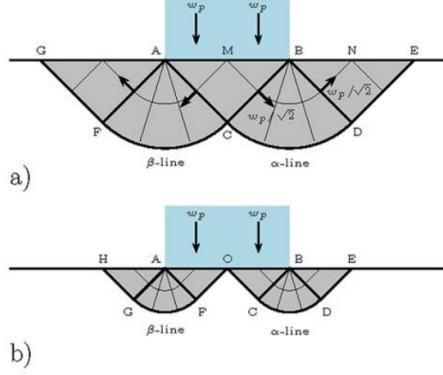


Simulation setting of BLOB problem. Isosurface and vectors represent temperature field and computed flow respectively. Taken from [686]

8.7.9 The punch/indentor problem in 2D

The punch benchmark is one of the few boundary value problems involving plastic solids for which there exists an exact solution. Such solutions are usually either for highly simplified geometries (spherical or axial symmetry, for instance) or simplified material models (such as rigid plastic solids) [1008].

In this experiment, a rigid punch indents a rigid plastic half space; the slip line field theory gives exact solutions as shown hereunder:

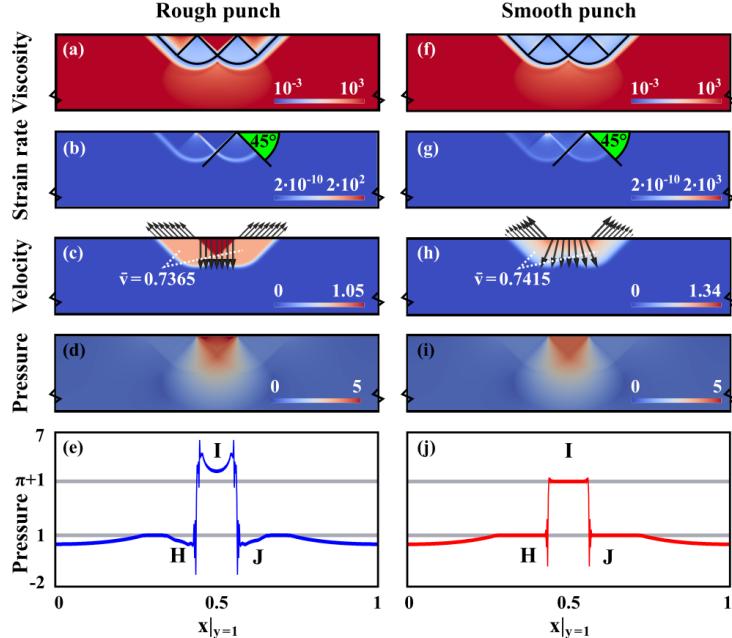


Two-dimensional rigid punch indenting a rigid plastic half space. (a) Prandtl's rigid plastic solution; (b) Hills solution. Taken from [1761]

The plane strain formulation of the equations and the detailed solution to the problem were derived in the Appendix of [1761] and are also presented in [715]. The two dimensional punch problem has been extensively studied numerically for the past 40 years [2033, 1503, 2032, 384, 383, 945, 1983, 259, 1531, 766] and has been used to draw a parallel with the tectonics of eastern China in the context of the India-Eurasia collision [1741, 1336, 575]. It is also worth noting that it has been carried out in one form or another in series of analogue modelling articles concerning the same region, with a rigid indenter colliding with a rheologically stratified lithosphere [1458, 498, 1002].

Numerically, the one-time step punch experiment is performed on a two-dimensional domain of purely plastic von Mises material. Given that the von Mises rheology yield criterion does not depend on pressure, the density of the material and/or the gravity vector is set to zero. Sides are set to free slip boundary conditions, the bottom to no slip, while a vertical velocity $(0, -v_p)$ is prescribed at the top boundary for nodes whose x coordinate is within $[L_x/2 - \delta/2, L_x/2 + \delta/2]$.

The analytical solution predicts that the angle of the shear bands stemming from the sides of the punch is $\pi/4$, that the pressure right under the punch is $1 + \pi$, and that the velocity of the rigid blocks on each side of the punch is $v_p/\sqrt{2}$ (this is simply explained by invoking conservation of mass).



The punch benchmark results after 500 nonlinear iterations for a rough punch (left column) and a smooth punch (right column). (a,f) Viscosity field with analytical slip lines. (b,g) Strain rate norm $\dot{\varepsilon}_e$ with measured shear band angles. (c,h) Velocity magnitude with velocity vectors along the surface of the domain. (d,i) Pressure field. (e,j) Pressure along the surface of the domain (colored line) and analytical solution values $\pi + 1$ and 1 (grey lines).

Taken from [766]

8.7.10 Lid driven cavity with analytical solution

This comes from [573](section 3.1.4). The velocity is prescribed to be

$$\vec{v} = (2y(1-x^2); -2x(1-y^2))$$

with a domain given by $\Omega = [-1 : 1] \times [-1 : 1]$. The strainrate tensor is then given by:

$$\dot{\epsilon} = \begin{pmatrix} -4xy & -x^2 + y^2 \\ -x^2 + y^2 & 4xy \end{pmatrix}$$

The Stokes equation is then:

$$-\frac{\partial p}{\partial x} + 2\eta(-4y + 2y) = \rho g_x \quad (451)$$

$$-\frac{\partial p}{\partial y} + 2\eta(-2x + 4x) = \rho g_y \quad (452)$$

where we assume the viscosity $\eta = 1$ to be constant in space. Assuming $g_x = 0$, the first equation is

$$\frac{\partial p}{\partial x} = -4y$$

i.e.

$$p(x, y) = -4yx + f(y)$$

Inserting this in the second equation:

$$4x - f'(y) + 4x = \rho g_y$$

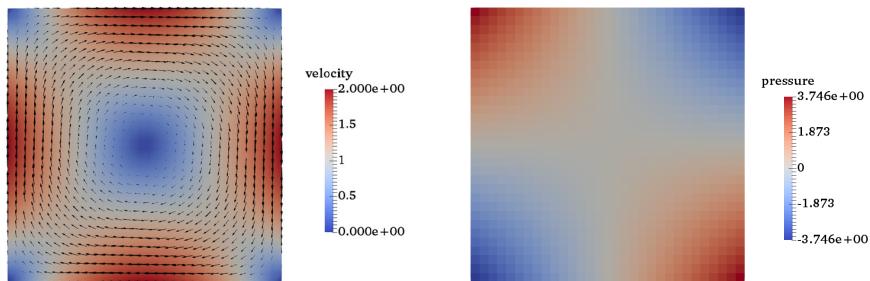
or,

$$-f'(y) + 8x = \rho g_y$$

Assuming $g_y = -1$, we get $\rho = -8x$ and then $f'(y) = 0$ so $f(y) = C$ where C is a constant. Finally the pressure is given by:

$$p(x, y) = -4yx + C$$

We add the following requirement: $\int_{\Omega} p(x, y) d\Omega = 0$ so that $C = 0$.



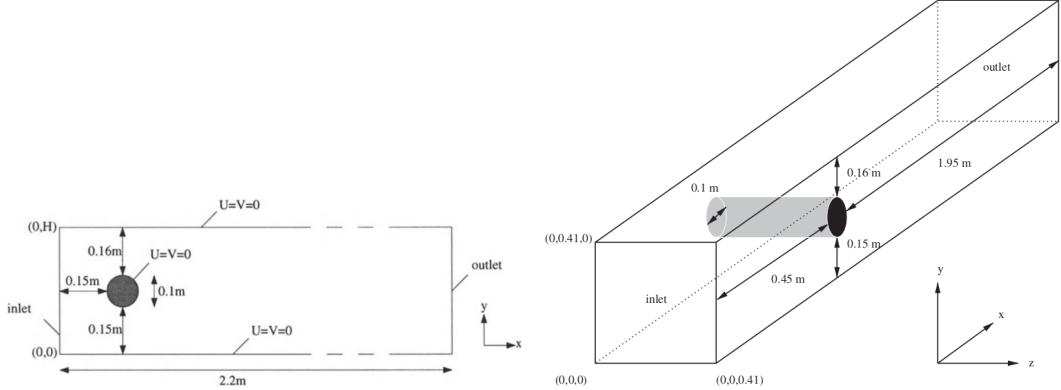
$$\begin{aligned}
 v_{rms}^2 &= \frac{1}{\Omega} \int_{\Omega} (u^2 + v^2) d\Omega \\
 &= \frac{1}{4} \int_{-1}^{+1} \int_{-1}^{+1} (u^2 + v^2) dx dy \\
 &= \frac{1}{4} \int_{-1}^{+1} \int_{-1}^{+1} [4y^2(1-x^2)^2 + 4x^2(1-y^2)^2] dx dy \\
 &=
 \end{aligned} \tag{453}$$

finish vrms calculation of benchmark

8.7.11 Flow around a cylinder

??

There are many variants of this problem: 2D [1807], 3D [999]. Many studies focus on Navier-Stokes flow since the cylinder generates vortices at high Reynolds numbers. Steady state solutions at low Re are shown here³⁰. Note the interesting benchmark for 2D visco-elastic flow in [151].

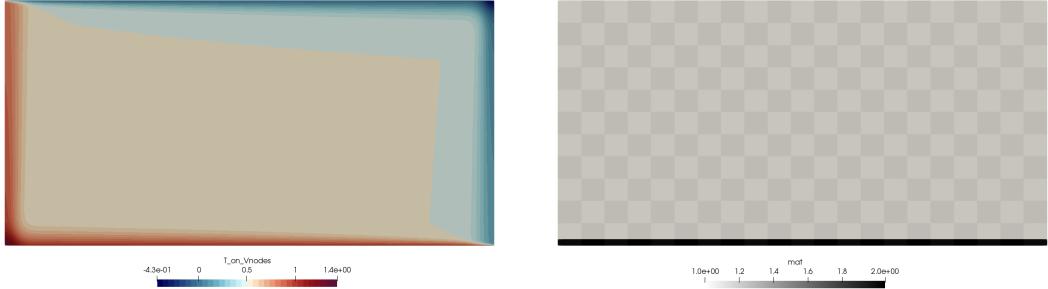


Left: taken from [1807]; Right: taken from [999]

8.7.12 Thin layer entrainment

This problem is a simulation to study the amount of entrainment by thermal convection of a dense, thin layer at the bottom of the model [1847]. To the author's knowledge only two other publications [1728, 1854] have presented results pertaining to this benchmark. The results shown here after are obtained with my ELEFANT code using the particle-in-cell technique.

The box is 2×1 , and contains two fluids:



Fluid 1 has a density $\rho_1 = 1$ and a viscosity $\eta = 1$. Fluid 2 is heavier ($\rho_2 = \rho_1 + \Delta\rho$) but has the same viscosity. Both fluids have a thermal expansion coefficient $\alpha = 10^{-10}$, a thermal conductivity $k = 1$, and a heat capacity coefficient $c_p = 1$. Fluid 2 is placed at the bottom of the box ($0 \leq y \leq 0.025$).

This experiment is parameterised by the thermal Rayleigh number $Ra = 300,000$ and the compositional Rayleigh number $Ra_c = 450,000$ which are defined as follows:

$$Ra_T = \frac{\alpha \rho g \Delta T L_y^3}{\kappa \eta} = \frac{\alpha \rho^2 g \Delta T L_y^3 c_p}{k \eta} = \alpha g \quad (454)$$

$$Ra_c = \frac{\Delta \rho g L_y^3}{\kappa \eta} = \frac{\rho \Delta \rho g L_y^3 c_p}{k \eta} = \Delta \rho g \quad (455)$$

where I have used the relationship $\kappa = k/\rho c_p$. B is defined as $B = Ra_T/Ra_c$ so The gravity acceleration is therefore set to $g = Ra/\alpha$ and this yields $\Delta \rho = Ra_c/g = BRa_T/g = B \times \alpha$.

³⁰upofthetestanddatameasurement

Free-slip boundary conditions are imposed on all sides of the domain. Temperature boundary conditions are $T(x, y = 0) = 1$ and $T(x, y = 1) = 0$. The analytical initial temperature field is given by

$$T(x, y) = T_u(x, y) + T_l(x, y) + T_r(x, y) + T_s(x, y) - \frac{3}{2} \quad (456)$$

where

$$\begin{aligned} T_u(x, y) &= \frac{1}{2} \operatorname{erf} \left(\frac{1-y}{2} \sqrt{\frac{u_0}{x}} \right) \\ T_l(x, y) &= 1 - \frac{1}{2} \operatorname{erf} \left(\frac{y}{2} \sqrt{\frac{u_0}{L_x - x}} \right) \\ T_r(x, y) &= \frac{1}{2} + \frac{Q}{2\sqrt{\pi}} \sqrt{\frac{u_0}{y+1}} \exp \left(-\frac{x^2 u_0}{4y+4} \right) \\ T_s(x, y) &= \frac{1}{2} - \frac{Q}{2\sqrt{\pi}} \sqrt{\frac{u_0}{2-y}} \exp \left(-\frac{(L_x-x)^2 u_0}{8-4y} \right) \end{aligned} \quad (457)$$

with

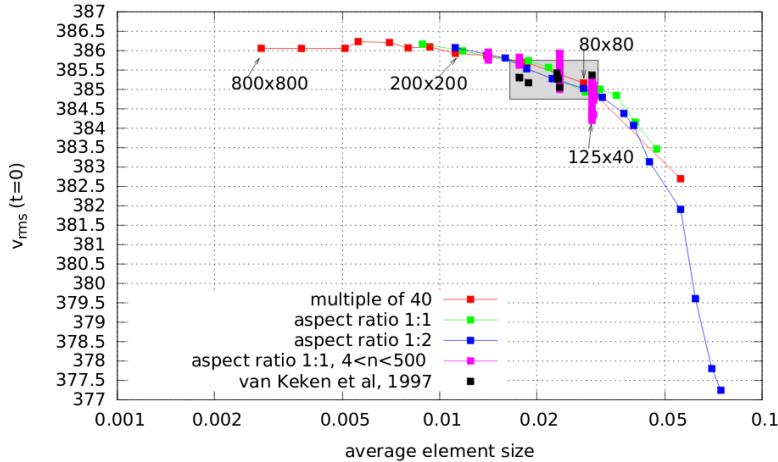
$$u_0 = \frac{L_x^{7/3}}{(1+L_x^4)^{2/3}} \left(\frac{Ra}{2\sqrt{\pi}} \right)^{2/3} \quad Q = 2\sqrt{\frac{L_x}{\pi u_0}} \quad (458)$$

Using $L_x = 2$, $Ra = 3 \times 10^5$, one gets $u_0 \simeq 1469.315$ and $Q \simeq 0.0416305$.

Given the small thickness of the bottom layer, it seems quite legitimate to investigate the influence of grid resolution on the simulation. I have therefore looked at the initial root mean square velocity measurement as a function of the element diagonal value (a proxy for the average resolution in the case where elements are not square).

Results are confirm that the element size plays a non negligible role at startup on the dynamics of the system. Superimposed on the figure are the measurements provided by Prof. van Keken (black squares in the gray box). They agree well with my measurements but also indicate that none of the authors in the original study ran the experiment at a high-enough resolution to start with (their results were therefore most likely resolution dependent).

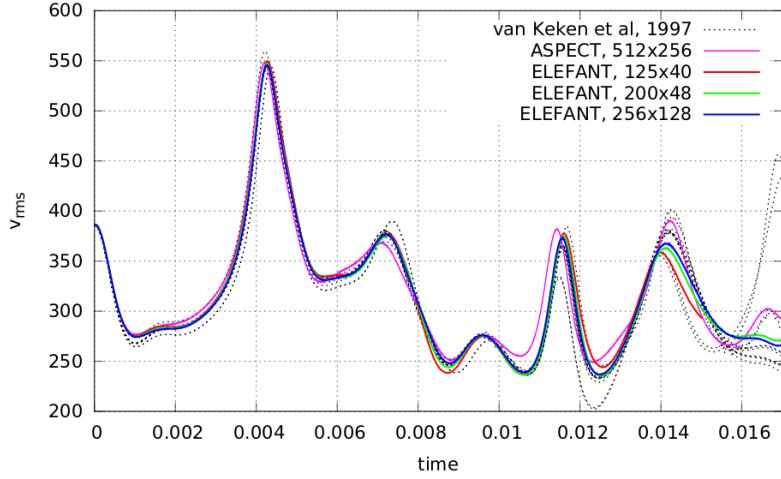
We see that the number of markers per element at startup is critical at (very) low resolution but that it does not lead to significant velocity variations at high resolution.



Thin layer entrainment experiment: root mean square velocity measurements at $t = 0$ as a function of the element diagonal size. The red square points correspond to resolutions where the number of elements in each direction is a multiple of 40 (i.e. L_y/d), so that no element would contain a mix of fluids 1 and 2. Pink points correspond to cases where the number of markers within each element was varied between 4 and 500 (random spatial distribution). Taken from [1757]

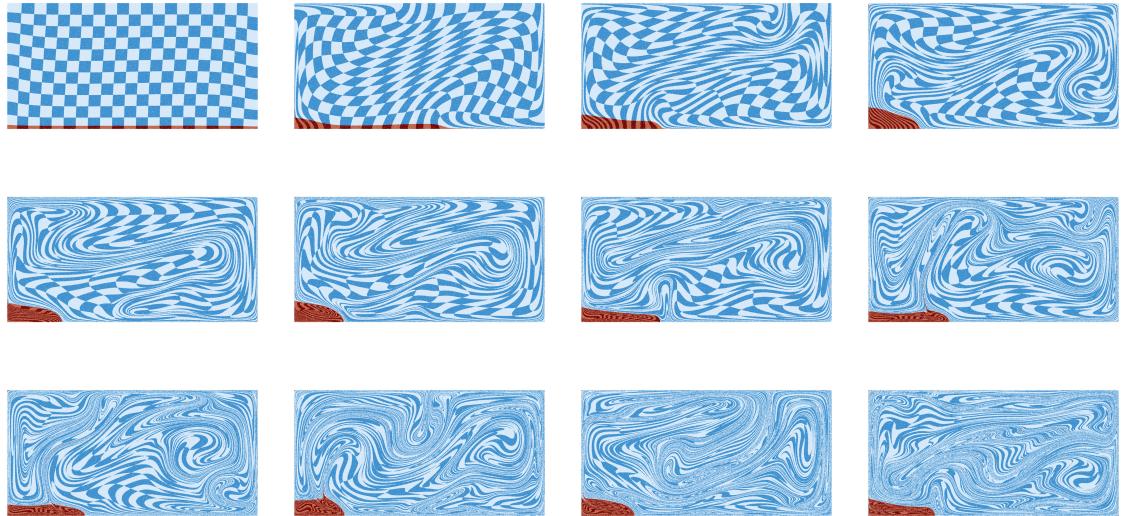
Looking at the root mean square velocity measurements, we see that the measurements done with ELEFANT agree nicely with those presented in [1847]. Past $t \sim 0.015$, the curves diverge clearly across all codes and authors, so I only need to focus the comparison for times $t < 0.015$. For the three tested

resolutions, measurements agree well and fall within the grey curves representing all results of van Keken et al. Additional tests have been carried out concerning the value of the Courant number (0.1 to 0.25) and the initial number of markers per element (100 or 200) and these parameters led to extremely similar results.



Thin layer entrainment experiment. Root mean square velocity as a function of time. All results presented in van Keken et al.(1997) are collapsed in dashed lines. All simulations were run with an initial marker density of 100 markers per element and with a Courant number of 0.25. Taken from [175].

As observed in van Keken et al., the dense layer is first swept into the lower left corner. Thermal instabilities then further develop in an asymmetrical way and entrain the dense material. Past $t \simeq 0.015$ the system becomes more and more chaotic with markers being randomly mixed in the system in a non-orderly fashion.



marker distribution as obtained with ELEFANT for grid 240x120, init_marker_density=7, random distribution, CFL=0.25, rkmethod=2, m_to_q=2. Unpublished.

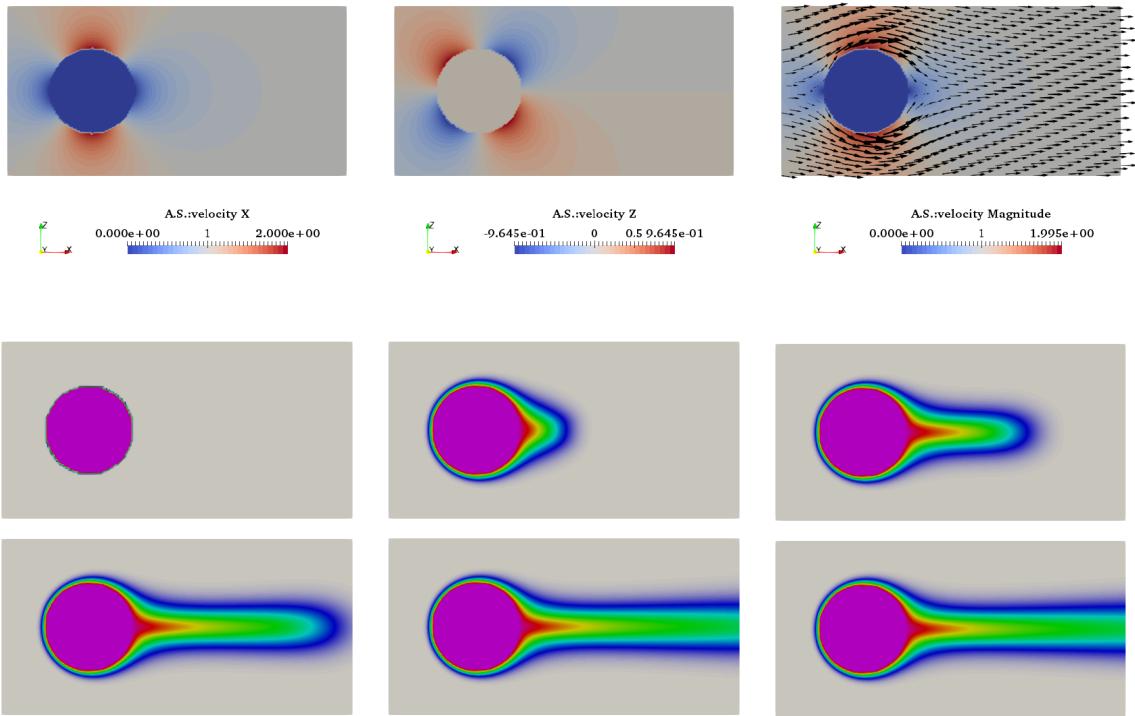
8.7.13 Heat flow around a cylinder

The domain is a 2D Cartesian box of size 8x4. The Stokes equations are not solved and the following velocity is prescribed:

$$u(x, y) = U_\infty \left(1 - \frac{x^2 - y^2}{(x^2 + y^2)^2} \right) \quad (459)$$

$$v(x, y) = -2U_\infty \frac{xy}{(x^2 + y^2)^2} \quad (460)$$

Boundary conditions are as follows: $T = 0$ is imposed at the top and bottom of the domain. $T = 1$ is imposed inside a disc centered at (2,2) with radius 1. Further: $k = 0.01$, $c_p = 1$, $\rho = 1$, CFL number is 0.1.



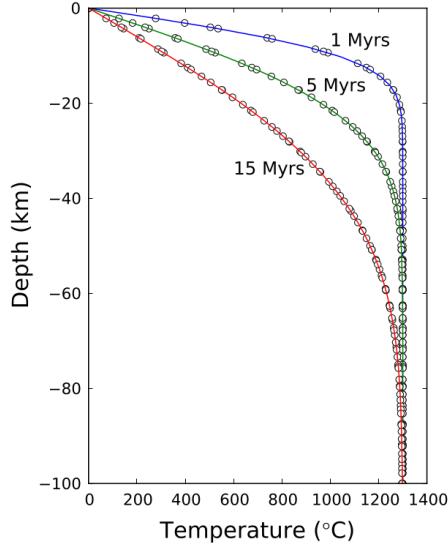
Time evolution of the temperature field. Results obtained with ELEFANT (unpublished)

8.7.14 Thermal diffusion of half-cooling space

This is a simple 1D experiment which solution is (for instance) available in Turcotte & Schubert [1806] and is also presented in [366].

The domain is 100km deep. $T_0 = 0^\circ\text{C}$ is prescribed at the surface and $T_m = 1300^\circ\text{C}$ is prescribed at the bottom. The initial temperature is $T(y) = 1300^\circ\text{C}$. The material is characterised by $\rho = 1000\text{kg/m}^3$, $C_p = 1000\text{J/kg/K}$, $k = 1\text{J/m/K}$. The time-dependent solution is given by:

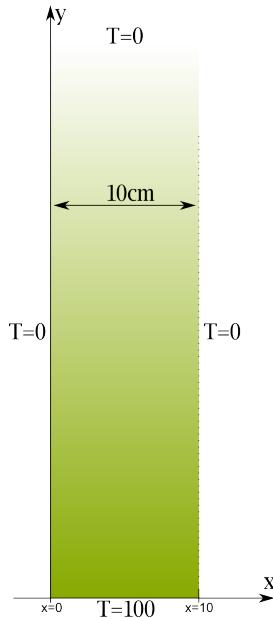
$$T(y, t) = T_0 + (T_0 - T_m) \operatorname{erf} \left(\frac{y}{2\sqrt{kt/\rho C_p}} \right) \quad (461)$$



Thermal diffusion of half space cooling plate. The temperature profiles in the analytical solution at 1, 5, and 15 Myrs are plotted in solid lines. The results from DynEarthSol2D are plotted in circles. Taken from [366]

8.7.15 Laplace equation on a semi infinite plate

) This experiment is based on a 2nd year mathematics lecture I give at Utrecht University. One wishes to solve the Laplace equation for temperature on the following plate subject to the indicated boundary conditions:



The temperature satisfies the 2D Laplace equation inside the plate:

$$\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} = 0 \quad (462)$$

We could try to solve the equation by using a tentative solution of the form:

$$T(x, y) = \theta(x)\Phi(y) \quad (463)$$

We do not *know* the solution is of this form.

We substitute (2) into (1) and obtain:

$$\Phi \frac{\partial^2 \theta}{\partial x^2} + \theta \frac{\partial^2 \Phi}{\partial y^2} = 0$$

Dividing by $\theta\Phi$ gives:

$$\frac{1}{\theta} \frac{\partial^2 \theta}{\partial x^2} + \frac{1}{\Phi} \frac{\partial^2 \Phi}{\partial y^2} = 0$$

Separation of variables: we say that each term is a constant because the first term is a function of x only and the second a function of y only. We then write

$$\frac{1}{\theta} \frac{\partial^2 \theta}{\partial x^2} = - \frac{1}{\Phi} \frac{\partial^2 \Phi}{\partial y^2} = -k^2$$

where k is called the separation constant. This leads to

$$\begin{aligned} \frac{\partial^2 \theta}{\partial x^2} + k^2 \theta &= 0 \\ \frac{\partial^2 \Phi}{\partial y^2} - k^2 \Phi &= 0 \end{aligned}$$

- The solution to the first one is $\theta(x) = \sin kx$ or $\theta(x) = \cos kx$
- The solution to the second one is $\Phi(y) = e^{ky}$ or $\Phi(y) = e^{-ky}$

The general solution writes:

$$T(x, y) = \theta(x)\Phi(y) = \left\{ \begin{array}{l} \sin kx \\ \cos kx \end{array} \right\} \left\{ \begin{array}{l} e^{ky} \\ e^{-ky} \end{array} \right\}$$

We can now use the b.c. to find the solution to the Laplace equation.

- Since $T \rightarrow 0$ when $y \rightarrow \infty$ then e^{ky} unacceptable.
- Since $T = 0$ when $x = 0$ then $\cos kx$ unacceptable.

so

$$T(x, y) = \sin(kx) e^{-ky}$$

We finally use $T = 0$ at $x = 10$ which leads to $10k = n\pi$, i.e.:

$$T(x, y) = \sin\left(\frac{n\pi x}{10}\right) e^{-n\pi y/10}$$

 Problem: the solution does not satisfy $T(x, 0) = 100$. However, a linear combination of solutions is still a solution ! Let's find such a combination which satisfies the b.c. at $y = 0$:

$$T(x, y) = \sum_{n=1}^{\infty} b_n \sin\left(\frac{n\pi x}{10}\right) e^{-n\pi y/10}$$

We impose then $T(x, 0) = 100$:

$$100 = \sum_{n=1}^{\infty} b_n \sin\left(\frac{n\pi x}{10}\right)$$

This is the Fourier sine series of $f(x) = 100$ with $l = 10$ (chapter 7.9 of Boas).

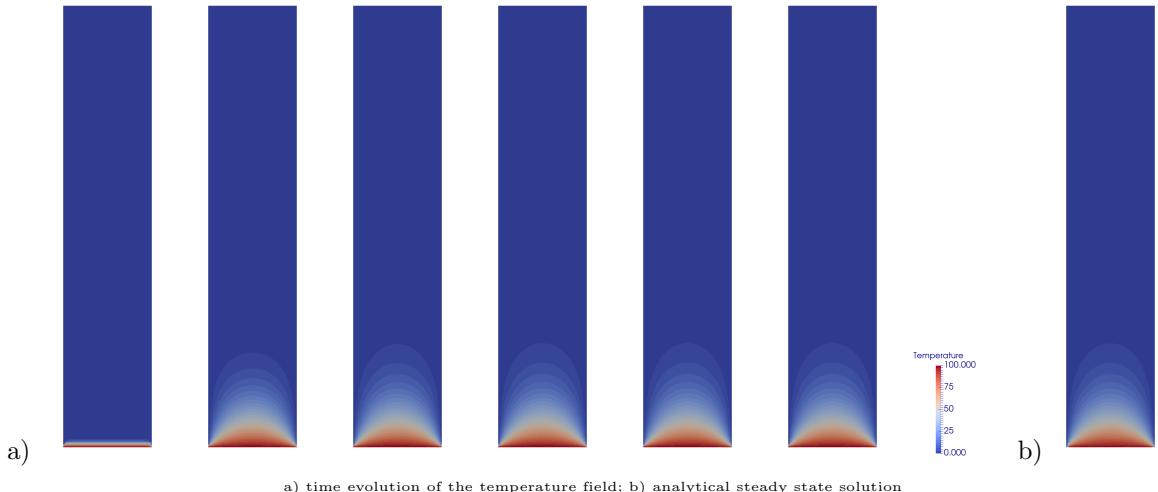
The coefficient b_n is then given by

$$b_n = \frac{2}{l} \int_0^l f(x) \sin \frac{n\pi x}{l} dx = \frac{2}{10} \int_0^{10} 100 \sin \frac{n\pi x}{10} dx = \begin{cases} 400/n\pi & \text{odd } n \\ 0 & \text{even } n \end{cases}$$

Finally (!):

$$T(x, y) = \frac{400}{\pi} \left(e^{-\pi y/10} \sin\left(\frac{\pi x}{10}\right) + \frac{1}{3} \sin\left(\frac{3\pi x}{10}\right) e^{-3\pi y/10} + \dots \right)$$

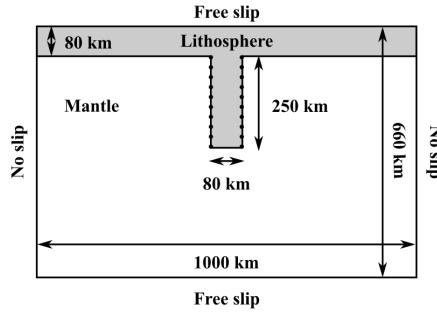
The simulation has been run with a 10x50 domain. All coefficients of the temperature equation are set to 1, and the Stokes equation is not solved. The timestep is fixed to $dt = 0.1$. Resolution is 32x160.



a) time evolution of the temperature field; b) analytical steady state solution

8.7.16 Slab detachment benchmark

[1633, 77, 766]



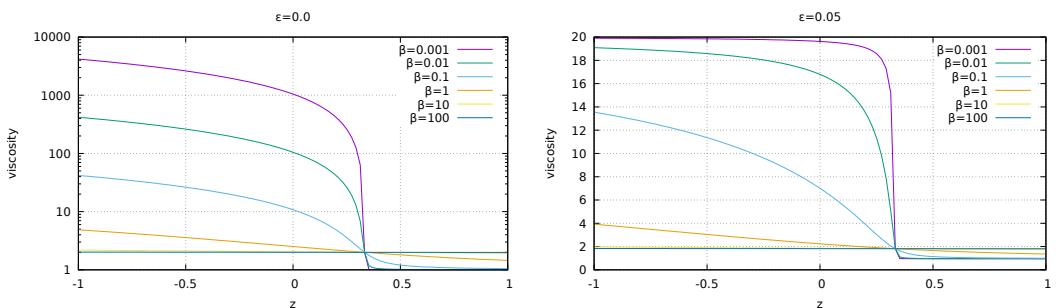
The detachment benchmark model setup of Schmalholz [1633]: a symmetric system of nonlinear viscous lithosphere with a vertical slab extending into a linear viscous mantle. The top and bottom boundaries are free slip, while the vertical boundaries are no slip. Taken from [766].

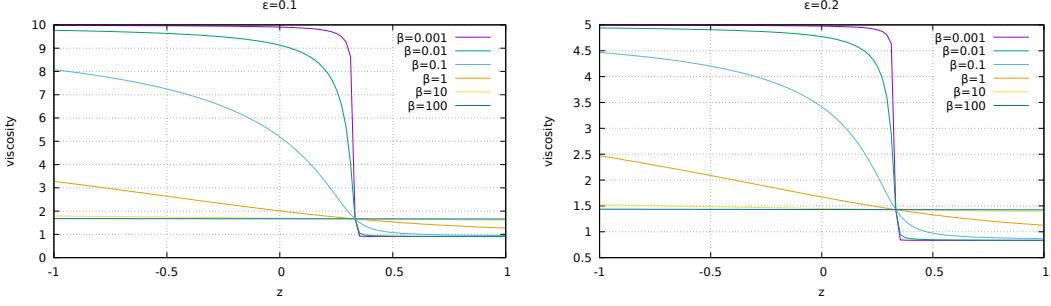
8.7.17 Layered flow with viscosity contrast

The idea behind this benchmark is to construct an analytical solution to the incompressible Stokes equation in the case where the viscosity field showcases a viscosity contrast at location $y = y_0$ whose amplitude and width can be controlled. The viscosity is defined as

$$\eta(y) = \frac{1}{\frac{1}{\pi} \tan^{-1}\left(\frac{y-y_0}{\beta}\right) + 1/2 + \epsilon}$$

where β and ϵ are parameters.





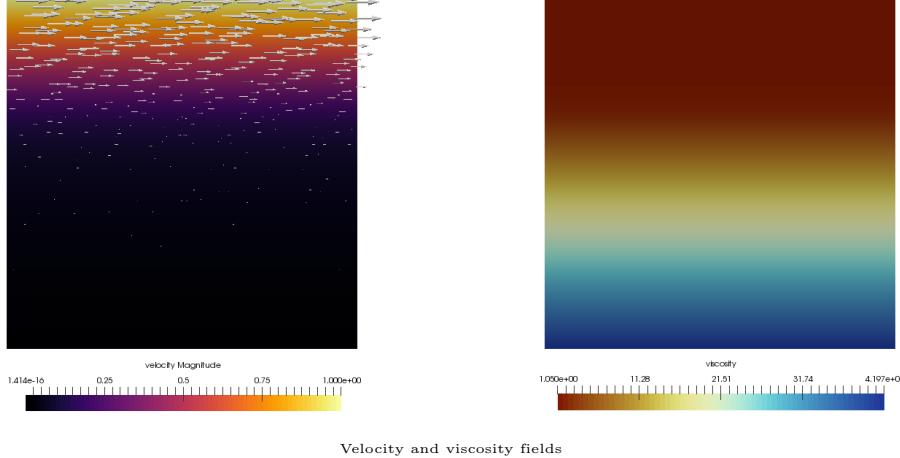
Viscosity profiles for different values of β and ϵ for $y_0 = 1/3$. When β is very large, the viscosity essentially converges to $\sim (1/2 + \epsilon)^{-1}$. β controls the width of the transition while ϵ controls the amplitude of the viscosity variation.

The flow is assumed to take place in an infinitely long pipe (in the horizontal direction) and bound by $y = -1$ and $y + 1$. At the bottom we impose $v_x(y = -1) = 0$ while we impose $v_x(y = +1) = 1$ at the top. The density is set to 1 while the gravity is set to zero. Under these assumptions, the flow velocity and pressure fields are given by:

$$\begin{aligned} v_x(x, y) &= \frac{1}{2\pi} \left(-\beta C_1 \log[\beta^2 + (z - y_0)^2] + 2(z - y_0)C_1 \tan^{-1} \frac{z - y_0}{\beta} + \pi(1 + 2\epsilon)zC_1 + C_2 \right) \\ v_y(x, y) &= 0 \\ p(x, y) &= 0 \end{aligned} \quad (464)$$

where C_1 and C_2 are integration constants:

$$\begin{aligned} C_1 &= 2\pi \left[\beta \log[\beta^2 + (1 + y_0)^2] - 2(1 + y_0) \tan^{-1} \frac{1 + y_0}{\beta} - \beta \log[\beta^2 + (1 - y_0)^2] + 2(1 - y_0) \tan^{-1} \frac{1 - y_0}{\beta} + 2\pi(1 + 2\epsilon) \right] \\ C_2 &= \left[\beta \log[\beta^2 + (1 + y_0)^2] - 2(1 + y_0) \tan^{-1} \frac{1 + y_0}{\beta} + \pi(1 + 2\epsilon) \right] C_1, \end{aligned} \quad (464)$$



Analytical derivations The flow takes place in the horizontal direction and is infinite in the this direction too so that:

$$\vec{v} = (u(y), 0)$$

The strain rate tensor is then given by:

$$\dot{\epsilon} = \frac{1}{2} \begin{pmatrix} 0 & du/dy \\ du/dy & 0 \end{pmatrix}$$

The momentum equation then becomes:

$$\vec{\nabla} \cdot (2\eta\dot{\epsilon}) - \vec{\nabla}p = \vec{\nabla} \cdot \left[\eta(y) \begin{pmatrix} 0 & du/dy \\ du/dy & 0 \end{pmatrix} \right] - \vec{\nabla}p = \rho\vec{g}$$

On the vertical axis, when the gravity is zero, the equation is automatically verified when the pressure is zero. On the horizontal axis:

$$\frac{d}{dy} \left(\eta(y) \frac{du}{dy} \right) = 0$$

Then

$$\eta(y) \frac{du}{dy} = C_1$$

or,

$$\frac{du}{dy} = \frac{C_1}{\eta(y)} = C_1 \left(\frac{1}{\pi} \tan^{-1} \frac{y - y_0}{\beta} + 1/2 + \epsilon \right)$$

so that the velocity is given by:

$$u(y) = \frac{1}{\pi} (y \tan^{-1}((y - y_0)/\beta) - y_0 \tan^{-1}((y - y_0)/\beta) - 0.5 * \beta \log(\beta^2 + y^2 - 2yy_0 + y_0^2) + \pi y(\epsilon + 0.5))$$

$$u(z) = \frac{1}{2\pi} \left(-\beta C_1 \log[\beta^2 + (z - y_0)^2] + 2(z - y_0)C_1 \tan^{-1} \frac{z - y_0}{\beta} + \pi(1 + 2\epsilon)zC_1 + C_2 \right)$$

where C_1 and C_2 are integration constants. I wish to impose $u(z = -1) = 0$ and $u(z = +1) = 1$:

$$\frac{1}{2\pi} \left(-\beta C_1 \log[\beta^2 + (-1 - y_0)^2] + 2(-1 - y_0)C_1 \tan^{-1} \frac{-1 - y_0}{\beta} - \pi(1 + 2\epsilon)C_1 + C_2 \right) = 0$$

$$\frac{1}{2\pi} \left(-\beta C_1 \log[\beta^2 + (1 - y_0)^2] + 2(1 - y_0)C_1 \tan^{-1} \frac{1 - y_0}{\beta} + \pi(1 + 2\epsilon)C_1 + C_2 \right) = 1$$

or,

$$-\beta C_1 \log[\beta^2 + (-1 - y_0)^2] + 2(-1 - y_0)C_1 \tan^{-1} \frac{-1 - y_0}{\beta} - \pi(1 + 2\epsilon)C_1 + C_2 = 0$$

$$-\beta C_1 \log[\beta^2 + (1 - y_0)^2] + 2(1 - y_0)C_1 \tan^{-1} \frac{1 - y_0}{\beta} + \pi(1 + 2\epsilon)C_1 + C_2 = 2\pi$$

or,

$$-\beta C_1 \log[\beta^2 + (-1 - y_0)^2] + 2(1 + y_0)C_1 \tan^{-1} \frac{1 + y_0}{\beta} - \pi(1 + 2\epsilon)C_1 + C_2 = 0$$

$$-\beta C_1 \log[\beta^2 + (1 - y_0)^2] + 2(1 - y_0)C_1 \tan^{-1} \frac{1 - y_0}{\beta} + \pi(1 + 2\epsilon)C_1 + C_2 = 2\pi$$

or,

$$-\beta C_1 \log(\beta^2 + (1 + y_0)^2) + 2(1 + y_0)C_1 \tan^{-1}((1 + y_0)/\beta) - \pi(1 + 2\epsilon)C_1 + C_2 = 0$$

$$-\beta C_1 \log(\beta^2 + (1 - y_0)^2) + 2(1 - y_0)C_1 \tan^{-1}((1 - y_0)/\beta) + \pi(1 + 2\epsilon)C_1 + C_2 = 2\pi$$

I can now subtract the first line from the second line:

$$\beta C_1 \log(\beta^2 + (1 + y_0)^2) - 2(1 + y_0)C_1 \tan^{-1}((1 + y_0)/\beta) - \beta C_1 \log(\beta^2 + (1 - y_0)^2) + 2(1 - y_0)C_1 \tan^{-1}((1 - y_0)/\beta) + 2\pi(1 + 2\epsilon)C_1 = 0$$

i.e.,

$$C_1 = 2\pi \left[\beta \log[\beta^2 + (1 + y_0)^2] - 2(1 + y_0) \tan^{-1} \left[\frac{1 + y_0}{\beta} \right] - \beta \log[\beta^2 + (1 - y_0)^2] + 2(1 - y_0) \tan^{-1} \left[\frac{1 - y_0}{\beta} \right] + 2\pi(1 + 2\epsilon) \right]$$

and then

$$C_2 = \beta C_1 \log(\beta^2 + (1 + y_0)^2) - 2(1 + y_0)C_1 \tan^{-1}((1 + y_0)/\beta) + \pi(1 + 2\epsilon)C_1$$

8.8 Assigning values to quadrature points

As we have seen in Section 6, the building of the elemental matrix and rhs requires (at least) to assign a density and viscosity value to each quadrature point inside the element. Depending on the type of modelling, this task can prove more complex than one might expect and have large consequences on the solution accuracy.

Here are several options:

- The simplest way (which is often used for benchmarks) consists in computing the 'real' coordinates (x_q, y_q, z_q) of a given quadrature point based on its reduced coordinates (r_q, s_q, t_q) , and passing these coordinates to a function which returns density and/or viscosity at this location. For instance, for the Stokes sphere:

```
def rho(x,y):
    if (x-.5)**2+(y-0.5)**2<0.123**2:
        val=2.
    else:
        val=1.
    return val

def mu(x,y):
    if (x-.5)**2+(y-0.5)**2<0.123**2:
        val=1.e2
    else:
        val=1.
    return val
```

This is very simple, but it has been shown to potentially be problematic. In essence, it can introduce very large contrasts inside a single element and perturb the quadrature. Please read section 3.3 of [882] and/or have a look at the section titled "Averaging material properties" in the ASPECT manual.

- another similar approach consists in assigning a density and viscosity value to the nodes of the FE mesh first, and then using these nodal values to assign values to the quadrature points. Very often ,and quite logically, the shape functions are used to this effect. Indeed we have seen before that for any point (r, s, t) inside an element we have

$$f_h(r, s, t) = \sum_i^m f_i N_i(r, s, t)$$

where the f_i are the nodal values and the N_i the corresponding basis functions.

In the case of linear elements (Q_1 basis functions), this is straightforward. In fact, the basis functions N_i can be seen as moving weights: the closer the point is to a node, the higher the weight (basis function value).

However, this is quite another story for quadratic elements (Q_2 basis functions). In order to illustrate the problem, let us consider a 1D problem. The basis functions are

$$N_1(r) = \frac{1}{2}r(r-1) \quad N_2(r) = 1 - r^2 \quad N_3(r) = \frac{1}{2}r(r+1)$$

Let us further assign: $\rho_1 = \rho_2 = 0$ and $\rho_3 = 1$. Then

$$\rho_h(r) = \sum_i^m \rho_i N_i(r) = N_3(r)$$

There lies the core of the problem: the $N_3(r)$ basis function is negative for $r \in [-1, 0]$. This means that the quadrature point in this interval will be assigned a negative density, which is nonsensical and numerically problematic!

use 2X Q1. write about it !

The above methods work fine as long as the domain contains a single material. As soon as there are multiple fluids in the domain a special technique is needed to track either the fluids themselves or their interfaces. Let us start with markers. We are then confronted to the infernal trio (a *menage à trois*?) which is present for each element, composed of its nodes, its markers and its quadrature points.

Each marker carries the material information (density and viscosity). This information must ultimately be projected onto the quadrature points. Two main options are possible: an algorithm is designed and projects the marker-based fields onto the quadrature points directly or the marker fields are first projected onto the FE nodes and then onto the quadrature points using the techniques above.

At a given time, every element e contains n^e markers. During the FE matrix building process, viscosity and density values are needed at the quadrature points. One therefore needs to project the values carried by the markers at these locations. Several approaches are currently in use in the community and the topic has been investigated by [524] and [550] for instance.

ELEFANT adopts a simple approach: viscosity and density are considered to be elemental values, i.e. all the markers within a given element contribute to assign a unique constant density and viscosity value to the element by means of an averaging scheme.

While it is common in the literature to treat the so-called arithmetic, geometric and harmonic means as separate averagings, I hereby wish to introduce the notion of generalised mean, which is a family of functions for aggregating sets of numbers that include as special cases the arithmetic, geometric and harmonic means.

If p is a non-zero real number, we can define the generalised mean (or power mean) with exponent p of the positive real numbers a_1, \dots, a_n as:

$$M_p(a_1, \dots, a_n) = \left(\frac{1}{n} \sum_{i=1}^n a_i^p \right)^{1/p} \quad (466)$$

and it is trivial to verify that we then have the special cases:

$$M_{-\infty} = \lim_{p \rightarrow -\infty} M_p = \min(a_1, \dots, a_n) \quad (\text{minimum}) \quad (467)$$

$$M_{-1} = \frac{n}{\frac{1}{a_1} + \frac{1}{a_2} + \dots + \frac{1}{a_n}} \quad (\text{harm. avrg.}) \quad (468)$$

$$M_0 = \lim_{p \rightarrow 0} M_p = \left(\prod_{i=1}^n a_i \right)^{1/n} \quad (\text{geom. avrg.}) \quad (469)$$

$$M_{+1} = \frac{1}{n} \sum_{i=1}^n a_i \quad (\text{arithm. avrg.}) \quad (470)$$

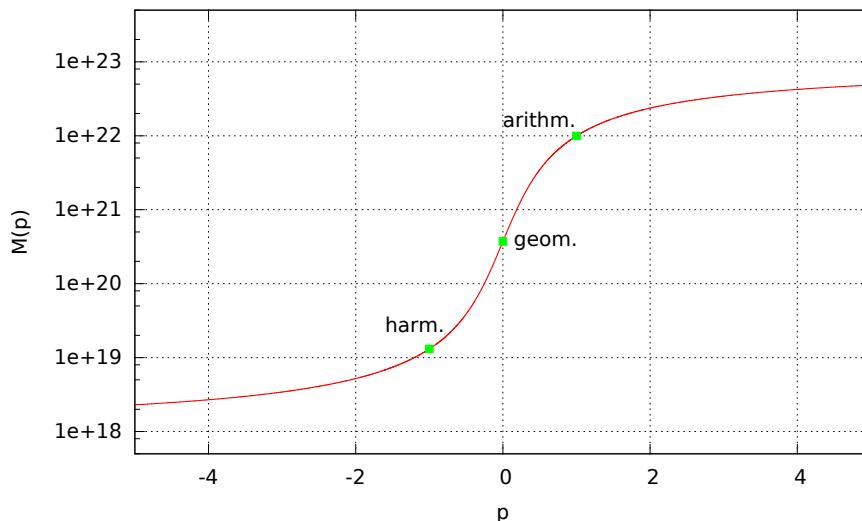
$$M_{+2} = \sqrt{\frac{1}{n} \sum_{i=1}^n a_i^2} \quad (\text{root mean square}) \quad (471)$$

$$M_{+\infty} = \lim_{p \rightarrow +\infty} M_p = \max(a_1, \dots, a_n) \quad (\text{maximum}) \quad (472)$$

Note that the proofs of the limit convergence are given in [274].

An interesting property of the generalised mean is as follows: for two real values p and q , if $p < q$ then $M_p \leq M_q$. This property has for instance been illustrated in Fig. 20 of [1638].

One can then for instance look at the generalised mean of a randomly generated set of 1000 viscosity values within $10^{18} Pa.s$ and $10^{23} Pa.s$ for $-5 \leq p \leq 5$. Results are shown in the figure hereunder and the arithmetic, geometric and harmonic values are indicated too. The function M_p assumes an arctangent-like shape: very low values of p will ultimately yield the minimum viscosity in the array while very high values will yield its maximum. In between, the transition is smooth and occurs essentially for $|p| \leq 5$.



```
▷ python_codes/fieldstone_markers_avrg
```

8.9 Matrix (Sparse) storage

The FE matrix is the result of the assembly process of all elemental matrices. Its size can become quite large when the resolution is being increased (from thousands of lines/columns to tens of millions).

One important property of the matrix is its sparsity. Typically less than 1% of the matrix terms is not zero and this means that the matrix storage can and should be optimised. Clever storage formats were designed early on since the amount of RAM memory in computers was the limiting factor 3 or 4 decades ago. [1613]

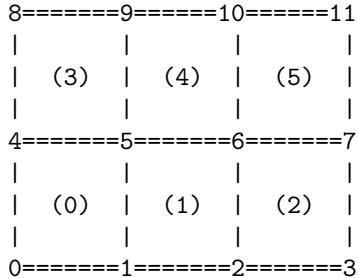
There are several standard formats:

- compressed sparse row (CSR) format
- compressed sparse column format (CSC)
- the Coordinate Format (COO)
- Skyline Storage Format
- ...

I focus on the CSR format in what follows.

8.9.1 2D domain - One degree of freedom per node

Let us consider again the 3×2 element grid which counts 12 nodes.



In the case there is only a single degree of freedom per node, the assembled FEM matrix will look like this:

$$\left(\begin{array}{cccccc|cc} X & X & & X & X & & & \\ X & X & X & X & X & X & & \\ & X & X & X & X & X & X & \\ & & X & X & & X & X & \\ X & X & & X & X & & X & X \\ X & X & X & X & X & X & X & X \\ & X & X & X & X & X & X & X \\ & & X & X & & X & X & \\ & & & X & X & X & X & X \\ & & & & X & X & X & X \\ & & & & & X & X & X \end{array} \right)$$

where the X stand for non-zero terms. This matrix structure stems from the fact that

- node 0 sees nodes 0,1,4,5
- node 1 sees nodes 0,1,2,4,5,6
- node 2 sees nodes 1,2,3,5,6,7
- ...

- node 5 sees nodes 0,1,2,4,5,6,8,9,10
- ...
- node 10 sees nodes 5,6,7,9,10,11
- node 11 sees nodes 6,7,10,11

In light thereof, we have

- 4 corner nodes which have 4 neighbours (counting themselves)
- $2(nnx-2)$ nodes which have 6 neighbours
- $2(nny-2)$ nodes which have 6 neighbours
- $(nnx-2) \times (nny-2)$ nodes which have 9 neighbours

In total, the number of non-zero terms in the matrix is then:

$$NZ = 4 \times 4 + 4 \times 6 + 2 \times 6 + 2 \times 9 = 70$$

In general, we would then have:

$$NZ = 4 \times 4 + [2(nnx - 2) + 2(nny - 2)] \times 6 + (nnx - 2)(nny - 2) \times 9$$

Let us temporarily assume $nnx = nny = n$. Then the matrix size (total number of unknowns) is $N = n^2$ and

$$NZ = 16 + 24(n - 2) + 9(n - 2)^2$$

A full matrix array would contain $N^2 = n^4$ terms. The ratio of NZ (the actual number of reals to store) to the full matrix size (the number of reals a full matrix contains) is then

$$R = \frac{16 + 24(n - 2) + 9(n - 2)^2}{n^4}$$

It is then obvious that when n is large enough $R \sim 1/n^2$.

CSR stores the nonzeros of the matrix row by row, in a single indexed array A of double precision numbers. Another array COLIND contains the column index of each corresponding entry in the A array. A third integer array RWPTR contains pointers to the beginning of each row, which an additional pointer to the first index following the nonzeros of the matrix A. A and COLIND have length NZ and RWPTR has length N+1.

In the case of the here-above matrix, the arrays COLIND and RWPTR will look like:

$$COLIND = (0, 1, 4, 5, 0, 1, 2, 4, 5, 6, 1, 2, 3, 5, 6, 7, \dots, 6, 7, 10, 11)$$

$$RWPTR = (0, 4, 10, 16, \dots)$$

8.9.2 2D domain - Two degrees of freedom per node

When there are now two degrees of freedom per node, such as in the case of the Stokes equation in two-dimensions, the size of the \mathbb{K} matrix is given by

<code>NfemV=nnp*ndofV</code>

In the case of the small grid above, we have `NfemV=24`. Elemental matrices are now 8×8 in size.

We still have

- 4 corner nodes which have 4 neighbours
- $2(nnx-2)$ nodes which have 6 neighbours
- $2(nny-2)$ nodes which have 6 neighbours

- $(nnx-2) \times (nny-2)$ nodes which have 9 neighbours,

but now each degree of freedom from a node sees the other two degrees of freedom of another node too. In that case, the number of nonzeros has been multiplied by four and the assembled FEM matrix looks like:

$$\left(\begin{array}{cccc|cccc|cccc|cccc|cccc|cccc} X & X & X & X & & X & X & X & X & & X & X & X & X & & X & X & X & X & & X & X & X \\ X & X & X & X & & X & X & X & X & & X & X & X & X & & X & X & X & X & & X & X & X \\ X & X \\ X & X \\ & X & X & X & X & X & X & X & X & X & & X & X & X & X & X & X & X & X & X & X & X \\ & X & X & X & X & X & X & X & X & X & & X & X & X & X & X & X & X & X & X & X & X \\ & & X & X & X & X & X & & & & & & X & X & X & X & X & & & & X & X \\ & & X & X & X & X & X & & & & & & X & X & X & X & X & & & & X & X \\ & & X & X & X & X & X & & & & & & X & X & X & X & X & & & & X & X \\ & & X & X & X & X & X & & & & & & X & X & X & X & X & & & & X & X \\ & & X & X & X & X & X & & & & & & X & X & X & X & X & & & & X & X \\ X & X & X & X & & X & X & X & X & & X & X & X & X & & X & X & X & X & & X & X & X \\ X & X & X & X & & X & X & X & X & & X & X & X & X & & X & X & X & X & & X & X & X \\ X & X \\ X & X \\ & X & X & X & X & X & X & X & X & X & & X & X & X & X & X & X & X & X & X & X & X & X \\ & X & X & X & X & X & X & X & X & X & & X & X & X & X & X & X & X & X & X & X & X & X \\ & & X & X & X & X & X & & & & & & X & X & X & X & X & & & & X & X \\ & & X & X & X & X & X & & & & & & X & X & X & X & X & & & & X & X \\ & & X & X & X & X & X & & & & & & X & X & X & X & X & & & & X & X \\ X & X & X & X & X & X & X & X & X & X & & X & X & X & X & X & X & X & X & X & X & X & X \end{array} \right)$$

Note that the degrees of freedom are organised as follows:

$$(u_0, v_0, u_1, v_1, u_2, v_2, \dots, u_{11}, v_{11})$$

In general, we would then have:

$$NZ = 4[4 \times 4 + [2(nnx - 2) + 2(nny - 2)] \times 6 + (nnx - 2)(nny - 2) \times 9]$$

and in the case of the small grid, the number of non-zero terms in the matrix is then:

$$NZ = 4[4 \times 4 + 4 \times 6 + 2 \times 6 + 2 \times 9] = 280$$

In the case of the here-above matrix, the arrays COLIND and RWPTR will look like:

$$COLIND = (0, 1, 2, 3, 8, 9, 10, 11, 0, 1, 2, 3, 8, 9, 10, 11, \dots)$$

$$RWPTR = (0, 8, 16, 28, \dots)$$

8.9.3 Matrix Storage in fieldstone

The majority of the codes have the FE matrix being a full array

```
a_mat = np.zeros((Nfem, Nfem), dtype=np.float64)
```

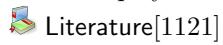
and it is converted to CSR format on the fly in the solve phase:

```
sol = sps.linalg.spsolve(sps.csr_matrix(a_mat), rhs)
```

Note that linked list storages can be used (lil_matrix). Substantial memory savings but much longer compute times.

8.9.4 Sparse Matrix-Vector multiplication

see 9.4 of Kepley's book. [1077]



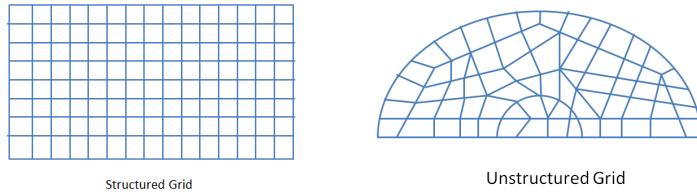
Literature[1121]

8.10 Mesh generation

Before basis functions can be defined and PDEs can be discretised and solved we must first tessellate the domain with polygons, e.g. triangles and quadrilaterals in 2D, tetrahedra, prisms and hexahedra in 3D.

When the domain is itself simple (e.g. a rectangle, a sphere, ...) the mesh (or grid) can be (more or less) easily produced and the connectivity array filled with straightforward algorithms [1759]. However, real life applications can involve extremely complex geometries (e.g. a bridge, a human spine, a car chassis and body, etc ...) and dedicated algorithms/softwares must be used (see [1767, 665, 1960]).

We usually distinguish between two broad classes of grids: structured grids (with a regular connectivity) and unstructured grids (with an irregular connectivity).



Remark. Various families of so-called meshless methods exist and are commonly employed in Computational Fluid Dynamics [1210, 1182, 1209, 1211]. They are however very rarely used in Computational geodynamics, with a noticeable exception [859].

8.10.1 Quadrilateral-based meshes

Let us now focus on the case of a rectangular computational domain of size $L_x \times L_y$ with a regular mesh composed of $\text{nelx} \times \text{nely} = \text{nel}$ quadrilaterals. There are then $\text{nmx} \times \text{nny} = \text{nnp}$ grid points. The elements are of size $hx \times hy$ with $hx = L_x / \text{nelx}$.

We have no reason to come up with an irregular/illogical node numbering so we can number nodes row by row or column by column as shown on the example hereunder of a 3×2 grid:

$\begin{array}{ccccccc} 8 & ===== & 9 & ===== & 10 & ===== & 11 \\ & & & & & & \\ & (3) & & (4) & & (5) & \\ & & & & & & \\ 4 & ===== & 5 & ===== & 6 & ===== & 7 \\ & & & & & & \\ & (0) & & (1) & & (2) & \\ & & & & & & \\ 0 & ===== & 1 & ===== & 2 & ===== & 3 \end{array}$	$\begin{array}{ccccccc} 2 & ===== & 5 & ===== & 8 & ===== & 11 \\ & & & & & & \\ & (1) & & (3) & & (5) & \\ & & & & & & \\ 1 & ===== & 4 & ===== & 7 & ===== & 10 \\ & & & & & & \\ & (0) & & (2) & & (4) & \\ & & & & & & \\ 0 & ===== & 3 & ===== & 6 & ===== & 9 \end{array}$
"row by row"	"column by column"

The numbering of the elements themselves could be done in a somewhat chaotic way but we follow the numbering of the nodes for simplicity. The row by row option is the adopted one in **fieldstone** and the coordinates of the points are computed as follows:

```
x = np.empty(nnp, dtype=np.float64)
y = np.empty(nnp, dtype=np.float64)
counter = 0
for j in range(0,nny):
    for i in range(0,nmx):
        x[counter]=i*hx
        y[counter]=j*hy
        counter += 1
```

The inner loop has i ranging from 0 to $\text{nmx}-1$ first for $j=0, 1, \dots$ up to $\text{nny}-1$ which indeed corresponds to the row by row numbering.

We now turn to the connectivity. As mentioned before, this is a structured mesh so that the so-called connectivity array, named `icon` in our case, can be filled easily. For each element we need to store the node identities of its vertices. Since there are `nel` elements and $m=4$ corners, this is a $m \times nel$ array. The algorithm goes as follows:

```
icon = np.zeros((m, nel), dtype=np.int16)
counter = 0
for j in range(0, nely):
    for i in range(0, nelx):
        icon[0, counter] = i + j * nnx
        icon[1, counter] = i + 1 + j * nnx
        icon[2, counter] = i + 1 + (j + 1) * nnx
        icon[3, counter] = i + (j + 1) * nnx
        counter += 1
```

In the case of the 3×2 mesh, the `icon` is filled as follows:

	element id→	0	1	2	3	4	5
node id↓							
0		0	1	2	4	5	6
1		1	2	3	5	6	7
2		5	6	7	9	10	11
3		4	5	6	8	9	10

It is to be understood as follows: element #4 is composed of nodes 5, 6, 10 and 9. Note that nodes are always stored in a counter clockwise manner, starting at the bottom left. This is very important since the corresponding basis functions and their derivatives will be labelled accordingly.

In three dimensions things are very similar. The mesh now counts `nelx×nely×nelz=nel` elements which represent a cuboid of size `Lx×Ly×Lz`. The position of the nodes is obtained as follows:

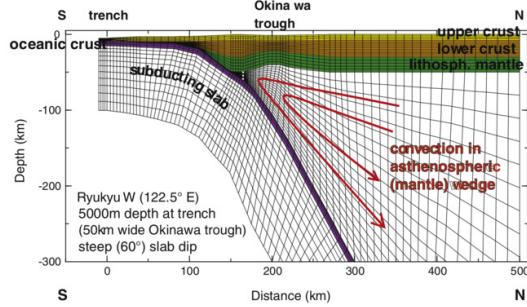
```
x = np.empty(nnp, dtype=np.float64)
y = np.empty(nnp, dtype=np.float64)
z = np.empty(nnp, dtype=np.float64)
counter=0
for i in range(0,nnx):
    for j in range(0,nny):
        for k in range(0,nnz):
            x[counter]=i*hx
            y[counter]=j*hy
            z[counter]=k*hz
            counter += 1
```

The connectivity array is now of size `m×nel` with `m=8`:

```
icon = np.zeros((m, nel), dtype=np.int16)
counter = 0
for i in range(0, nelx):
    for j in range(0, nely):
        for k in range(0, nelz):
            icon[0, counter]=nny*nnz*(i    )+nnz*(j    )+k
            icon[1, counter]=nny*nnz*(i+1)+nnz*(j    )+k
            icon[2, counter]=nny*nnz*(i+1)+nnz*(j+1)+k
            icon[3, counter]=nny*nnz*(i    )+nnz*(j+1)+k
            icon[4, counter]=nny*nnz*(i    )+nnz*(j    )+k+1
            icon[5, counter]=nny*nnz*(i+1)+nnz*(j    )+k+1
            icon[6, counter]=nny*nnz*(i+1)+nnz*(j+1)+k+1
            icon[7, counter]=nny*nnz*(i    )+nnz*(j+1)+k+1
            counter += 1
```

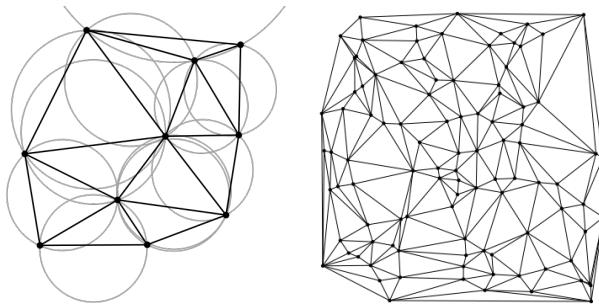
produce drawing of node numbering

Although it is not very common in geosciences, quadrilateral meshes are sometimes employed in a boundary-fitted way, as shown hereunder:



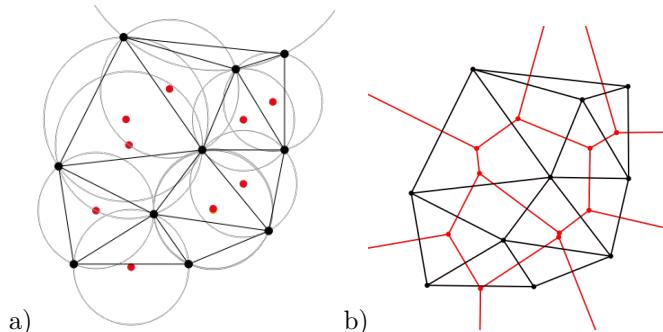
8.10.2 Delaunay triangulation and Voronoi cells, and triangle-based meshes

The topic of Delaunay³¹ triangulation is vast, but a simple definition can be written as follows: "a Delaunay triangulation for a set P of points in a plane is a triangulation $DT(P)$ such that no point in P is inside the circumcircle of any triangle in $DT(P)$." [wikipedia] Other properties of such triangulations are that they maximize the minimum angle of all the angles of the triangles in the triangulation. Note that for four or more points on the same circle (e.g., the vertices of a rectangle) the Delaunay triangulation is not unique and that points on a line also cannot yield a valid triangulation (for the simple reason that they do not form a triangle).



a) A Delaunay triangulation in the plane with circumcircles shown. b) The Delaunay triangulation of a random set of 100 points in a plane.

The Delaunay triangulation of a discrete point set P in general corresponds to the dual graph of the Voronoi diagram for P . A Voronoi diagram is composed of non-overlapping Voronoi cells which make a partition of the plane. For each point there is a corresponding region consisting of all points closer to that point than to any other: this region is the Voronoi cell of that point.



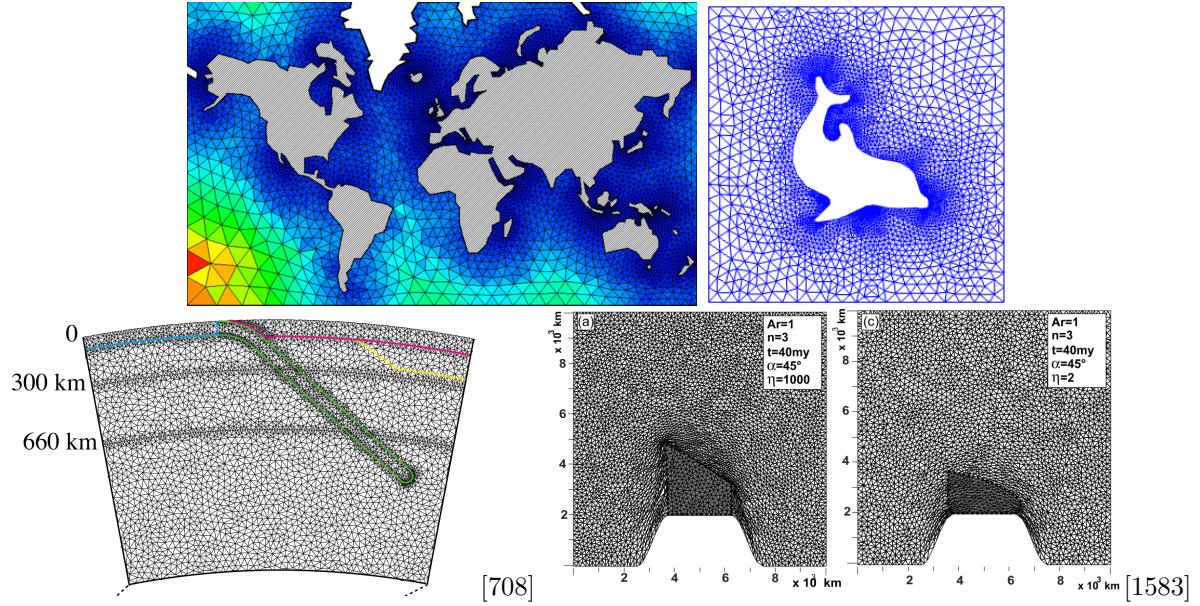
a) The Delaunay triangulation with all the circumcircles and their centers (in red). b) Connecting the centers of the circumcircles produces the Voronoi diagram (in red).

The Delaunay triangulation is used in the DOUAR code which is based on a particle levelset function to track materials. These particles are connected by means of a Delaunay triangulation (usually in a plane at startup, and then in a local Euclidean geometry once the surface is deformed) [221].

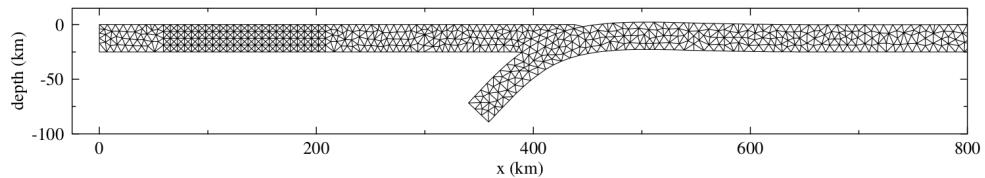
Literature: [705].

³¹The triangulation is named after Boris Delaunay for his work on this topic from 1934.

Once a Delaunay triangulation has been obtained it can be used as a FEM mesh. Triangle-based meshes are obviously better suited for simulations of complex geometries:

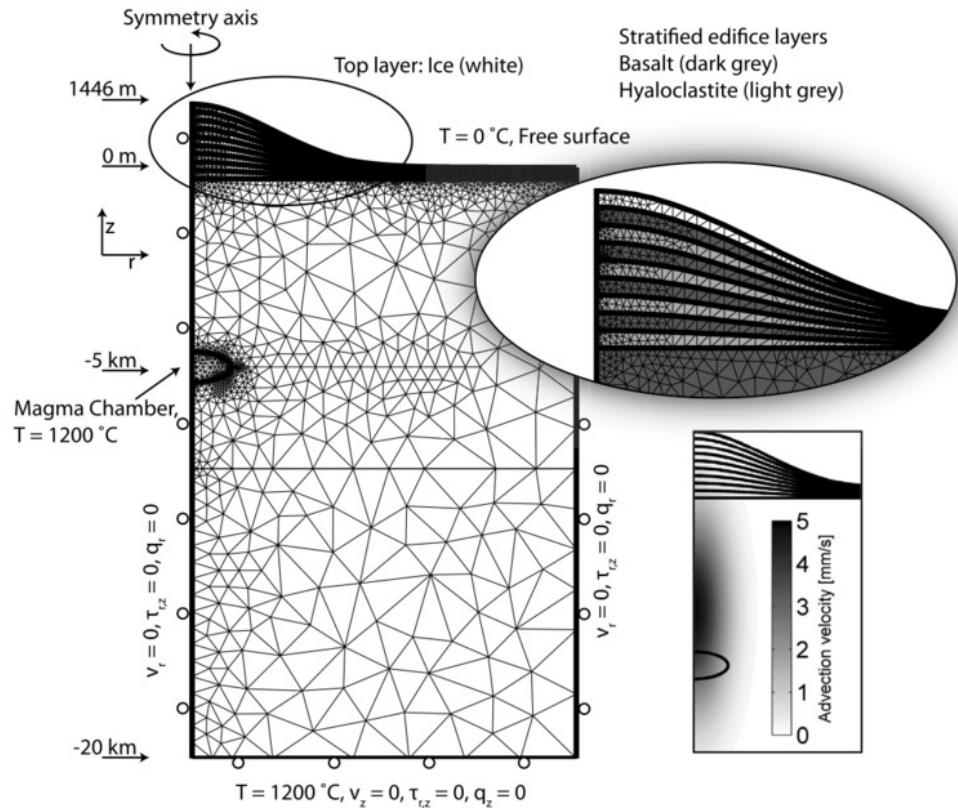


A very practical 2D triangle mesher is the code *Triangle*³² written by J.R. Shewchuk [1665, 1663, 1664]. *Triangle* is specialized for creating two-dimensional finite element meshes, but can also perform simpler related tasks such as forming Delaunay triangulations under various assumptions.



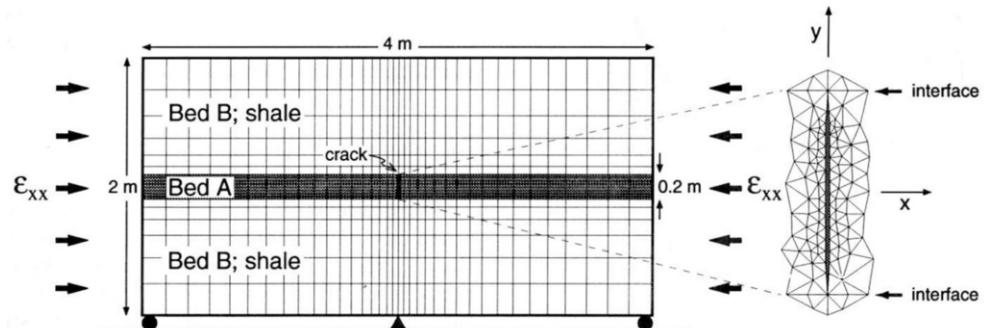
Taken from Buiter et al [263]. Finite element grid. The subducting plate initially extends to 1226 km in the horizontal direction and is not completely shown here. Discretization in the subducting plate is slightly coarser towards the right edge.

³²<https://www.cs.cmu.edu/~quake/triangle.html>



Numerical model setup of the 2D axisymmetric half-space with all applied boundary conditions to study the effects of ice-cap unloading on shallow volcanic systems [65]

Although it is rarely used in practice it is possible to produce meshes which contain both quadrilateral and triangular elements:



Mesh used to analyse the stress distribution around a pressurized crack in a layered elastic medium [623]

Literature[1386][1871]

mention stripack, plus paper moretti with it, and lithos1.0

write about gmsh

8.10.3 Tetrahedra

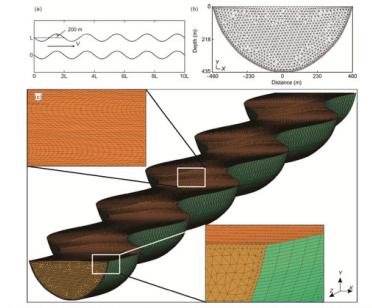
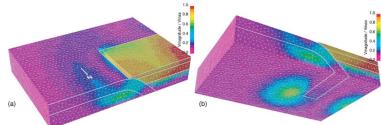


Figure 4 The top view of the initial geometric model (a), the cross section of the initial geometric model (b), the initial geometric model and its grid (c).

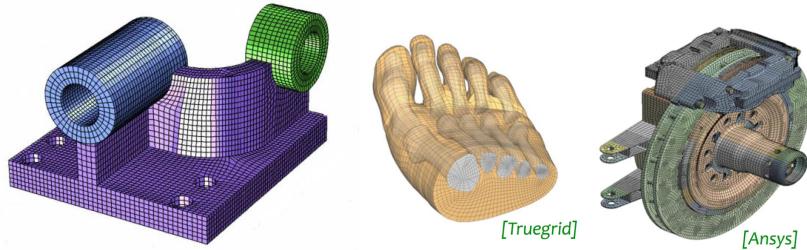
Example of 3D mesh [1968].



Normalized velocities of a STEP subduction model [795].

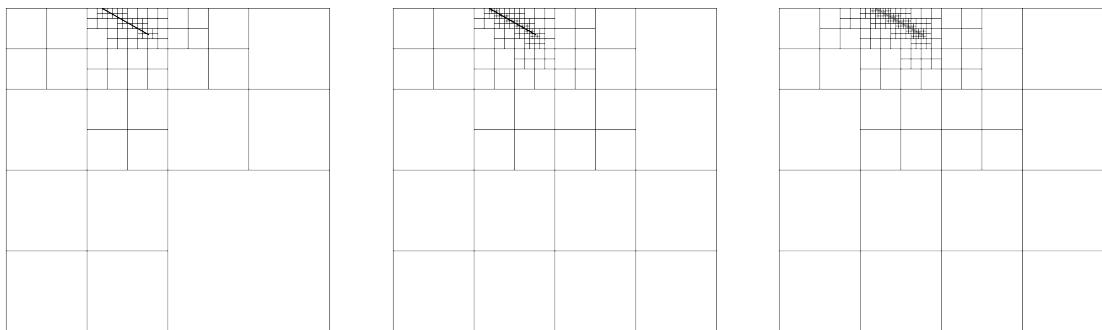
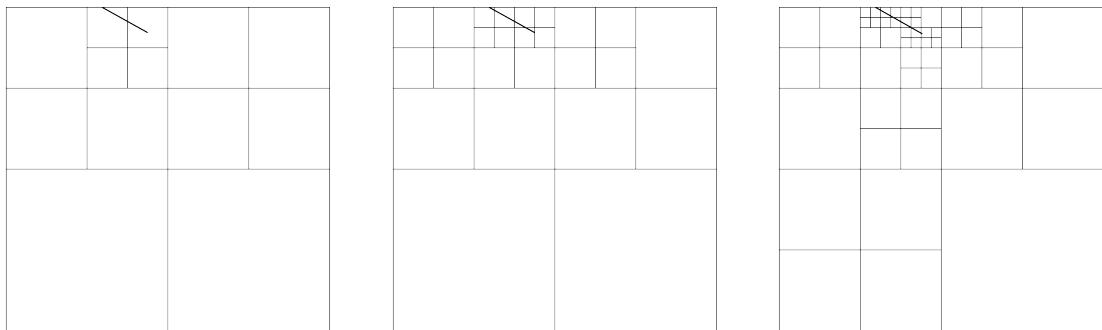
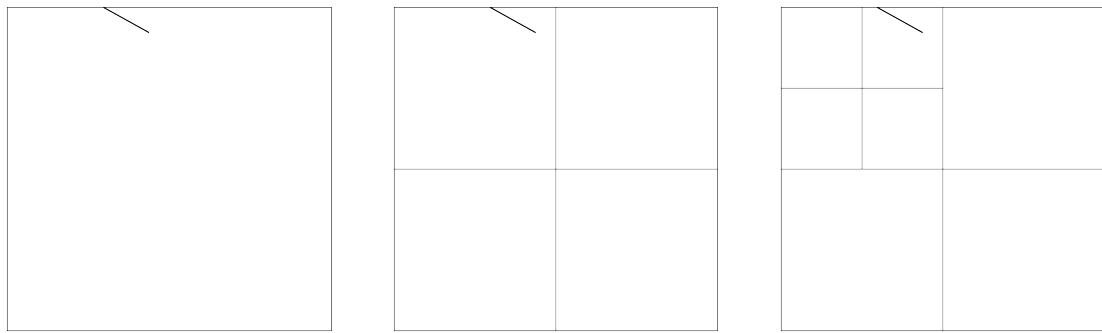
8.10.4 Hexahedra

A hexahedron is a convex polytope isomorphic to the cube $[0, 1]^3$. Edges are line segments, facets are strictly **planar** convex polygons.

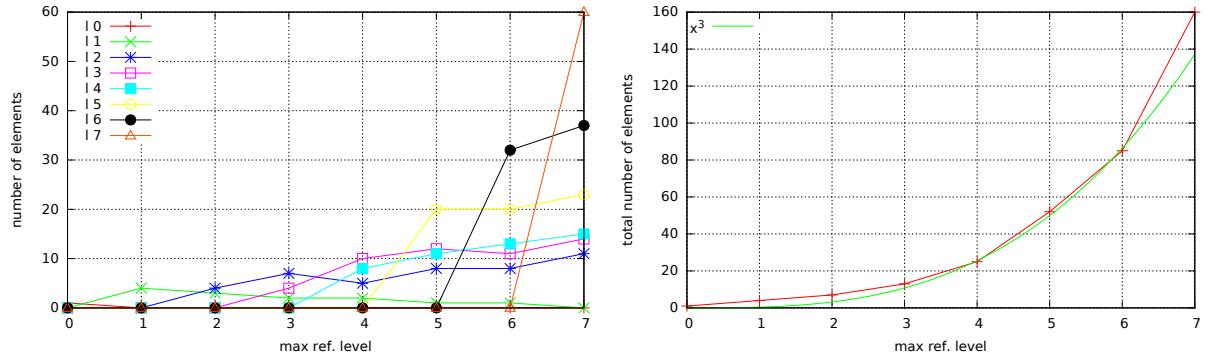


8.10.5 Adaptive Mesh Refinement

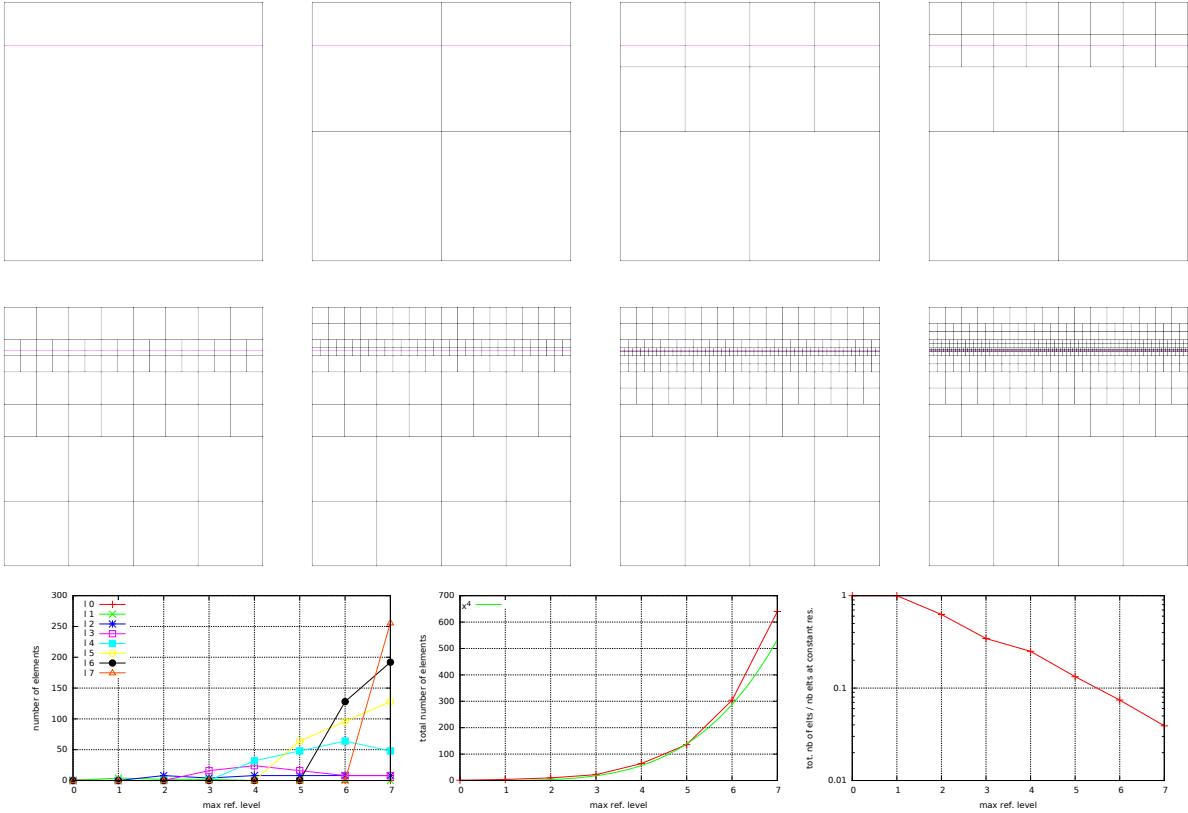
Literature: [310][107][1169] [307, sect 3] [107]



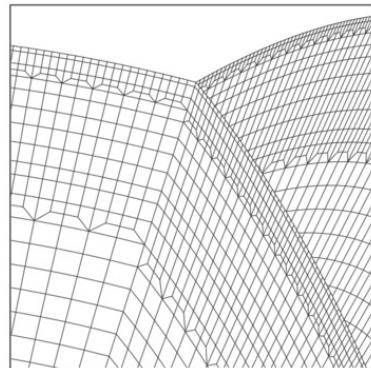
	# 10	# 11	# 12	# 13	# 14	# 15	# 16	# 17	# 18
max level= 0	1								
max level= 1	0	4							
max level= 2	0	3	4						
max level= 3	0	2	7	4					
max level= 4	0	2	5	10	8				
max level= 5	0	1	8	12	11	20			
max level= 6	0	1	8	11	13	20	32		
max level= 7	0	0	11	14	15	23	37	60	
max level= 8	0	0	11	13	17	27	43	72	116



In the particular case presented here, even though the inclusion in a short two-dimensional line, the total number of elements grows faster than the third power of the refinement level. While of course the total number of elements remains much smaller than the constant resolution counterpart, this observation tells us that authorising a unit increase of the maximum refinement level can have a substantial effect on the total number of elements.



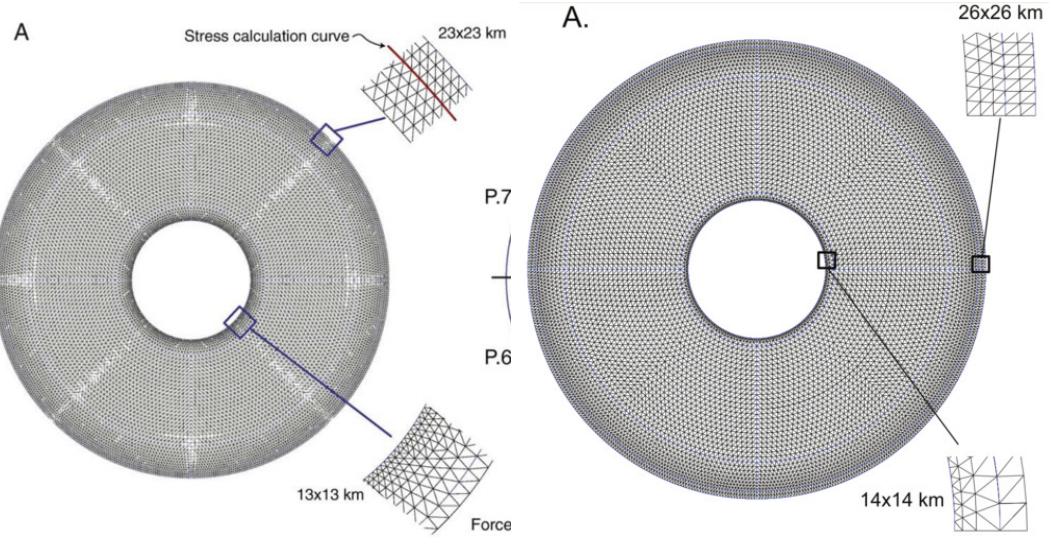
8.10.6 Conformal Mesh Refinement



Close-up of the two geometrical mesh doubling regions in the mantle. Taken from [1088]

Literature: [537]

8.10.7 Meshes in an annulus



The quadratic finite element mesh as used in [211, 213]

8.10.8 Meshes in a hollow sphere

The following is for the most part published in [1759].

To a first approximation the Earth is a sphere: the Earth's polar diameter is about 43 kilometers shorter than its equatorial diameter, a negligible difference of about 0.3%. As a consequence, modelling physical processes which take place in the planet require the discretisation of a sphere. Furthermore, because core dynamics occur on vastly different time scales than mantle dynamics, mantle modelling usually leaves the core out, thereby requiring simulations to be run on a hollow sphere mesh (with the noticeable exception of [721]).

Although so-called latitude-longitude grids would seem appealing, they suffer from the convergence of meridians at the poles (resulting in over sampling at poles) and the juxtaposition of triangles near the poles and quadrilaterals elsewhere. As a consequence more regular, but more complex, grids have been designed over the years which tessellate the surface of the sphere into triangles or quadrilaterals (sometimes overlapping). There is the 'cubed sphere' [1597, 84, 364], the ying-yang grid [1009, 1978, 1013, 1726, 443, 444], the spiral grid [956], an icosahedron-based grid [92, 1710], or a grid composed of 12 blocks further subdivided into quadrilaterals [2014] as used in the CitcomS code. Note that [1426] have also presented a method for generating a numerical grid on a spherical surface which allows the grid to be based on several different regular polyhedrons (including octahedron, cube, icosahedron, and rhombic dodecahedron).

How such meshes are built is often not discussed in the literature. It is a tedious exercise of three-dimensional geometry and it can be time-consuming, especially the connectivity array generation. In [1759] I present an open source mesh generator for three hollow sphere meshes: the 'cubed sphere' mesh, the CitcomS mesh and the icosahedral mesh.

The open source code library GHOST[1759] allows three different types of hollow sphere meshes to be built , i.e. meshes bounded by two concentric spheres:

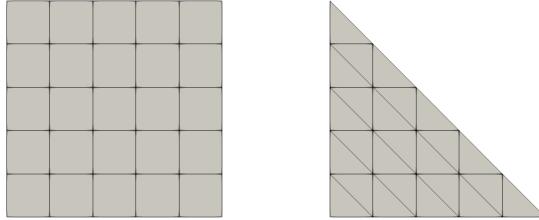
- The cubed sphere ('HS06'), composed of 6 blocks which are themselves subdivided into $N_b \times N_b$ quadrilateral shaped cells [1615, 1597, 84, 308]. Four types of cubed spheres meshes have been proposed: the conformal, elliptic, gnomonic and spring types [1510]. However only gnomonic meshes are considered here: these are obtained by inscribing a cube within a sphere and expanding to the surface of the sphere. The cubed sphere has recently been used in large-scale mantle convection simulation in conjunction with Adaptive Mesh Refinement [15, 308].
- The CitcomS mesh ('HS12') composed of 12 blocks also subdivided into $N_b \times N_b$ quadrilateral shaped cells [2014, 1699, 2010, 46]. Note that ASPECT [1119, 882], a relatively new code aimed at superseeding CitcomS can generate and use this type of mesh [1758] but is not limited to it.

- The icosahedral mesh ('HS20') composed of 20 triangular blocks [92, 91] subdivided into triangles, which is used in the TERRA code [276, 277, 275, 486].

Given the regularity and symmetry of these meshes determining the location of the mesh nodes in space is a relatively straightforward task. Building the mesh connectivity in an efficient manner is where the difficulty lies.

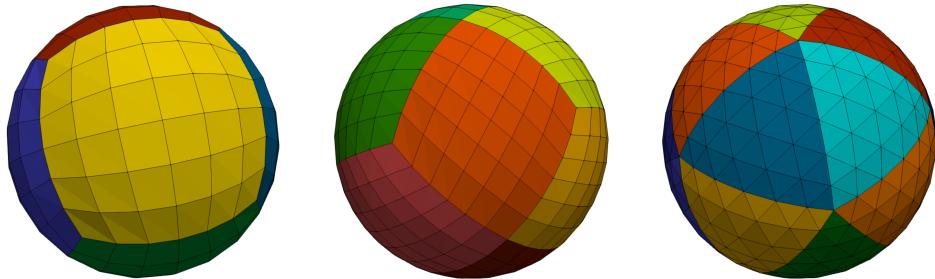
The approach to building all three meshes is identical:

1. A reference square or triangle is populated with cells, parametrised by a level l : the square is subdivided into $l \times l$ quadrilaterals while the triangle is subdivided into l^2 triangles.



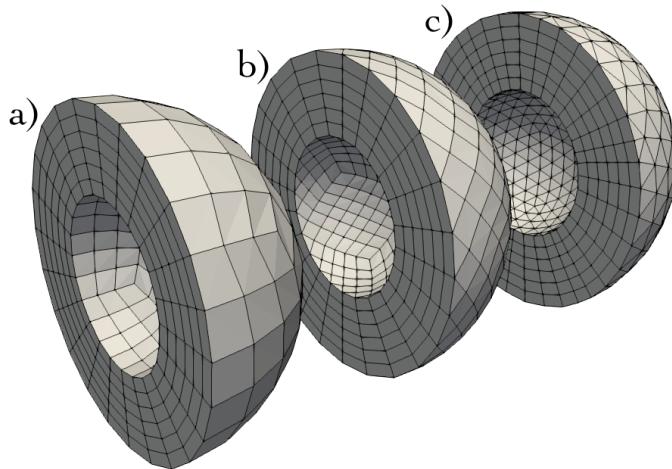
Reference square and triangles meshes at level 5.

2. This reference square or triangle is then replicated n_{block} times (6, 12 or 20) and mapped onto a portion of a unit sphere. The blocks are such that their union covers a full sphere but they cannot overlap except at the edges:



From left to right: HS06, HS12 and HS20 shells coloured by block number.

3. All block meshes are then merged together to generate a shell mesh. This task is rather complex as duplicate nodes must be removed and all connectivity arrays of the blocks must then be mended accordingly.
4. Shell meshes are replicated $n_{layer}+1$ times outwards with increasing radii. The n_{layer} shells are then merged together to form a hollow sphere mesh:



a) HS06 mesh composed of 6 blocks containing each 6^3 cells; b) HS12 mesh composed of 12 blocks containing each 6^3 cells; e) HS20 mesh composed of 20 blocks containing each 6^3 cells.

More information on these steps is available in the manual of the code. In the following table the number of nodes and cells for a variety of resolutions for all three mesh types is reported. Looking at the CitcomS literature of the past 20 years, we find that the mesh data presented in this table cover the various resolutions used, e.g. 12×48^3 [1302, 46], 12×64^3 [272] 12×96^3 [273], 12×128^3 [110, 1924, 1925]. Note that in the case of the HS06 and HS12 meshes the mesh nodes are mapped out to the 6 or 12 blocks following either an equidistant or equiangle approach (see [1510] for details on both approaches).

type	level	N	N_{el}	structure
HS06	2	78	48	6×2^3
HS06	4	490	384	6×4^3
HS06	8	3,474	3,072	6×8^3
HS06	16	26,146	24,576	6×16^3
HS06	32	202,818	196,608	6×32^3
HS06	64	1,597,570	1,572,864	6×64^3
HS06	128	12,681,474	12,582,912	6×128^3
HS06	256	101,057,026	100,663,296	6×256^3
HS12	2	150	96	12×2^3
HS12	4	970	768	12×4^3
HS12	8	6,930	6,144	12×8^3
HS12	16	52,258	49,152	12×16^3
HS12	32	405,570	393,216	12×32^3
HS12	48	1,354,850	1,327,104	12×48^3
HS12	64	3,195,010	3,145,728	12×64^3
HS12	128	25,362,690	25,165,824	12×128^3
HS12	256	202,113,538	201,326,592	12×256^3
HS20	2	126	160	20×2^3
HS20	4	810	1,280	20×4^3
HS20	8	5,778	10,240	20×8^3
HS20	16	43,554	81,920	20×16^3
HS20	32	337,986	655,360	20×32^3
HS20	64	2,662,530	5,242,880	20×64^3
HS20	128	21,135,618	41,943,040	20×128^3
HS20	256	168,428,034	335,544,320	20×256^3

Number of nodes N and elements/cells N_{el} for the three types of meshes and for various levels.

HS06: cubed sphere; HS12: CitcomS mesh; HS20: icosahedral mesh.

8.11 Visco-Plasticity

 Literature: [1378, 365, 1351, 1381]

8.11.1 Tensor invariants

Before we dive into the world of nonlinear rheologies it is necessary to introduce the concept of tensor invariants since they are needed further on. Unfortunately there are many different notations used in the literature and these can prove to be confusing. Note that we only consider symmetric tensors in what follows.

Given a tensor \mathbf{T} , one can compute its (moment) invariants as follows [1551, p.339]:

- first invariant:

$$\begin{aligned}\mathcal{I}_1(\mathbf{T})|^{2D} &= \text{Tr}[\mathbf{T}] = T_{xx} + T_{yy} \\ \mathcal{I}_1(\mathbf{T})|^{3D} &= \text{Tr}[\mathbf{T}] = T_{xx} + T_{yy} + T_{zz}\end{aligned}\quad (473)$$

- second invariant:

$$\begin{aligned}\mathcal{I}_2(\mathbf{T})|^{2D} &= \frac{1}{2} \text{Tr}[\mathbf{T}^2] = \frac{1}{2} \sum_{ij} T_{ij} T_{ji} = \frac{1}{2} (T_{xx}^2 + T_{yy}^2) + T_{xy}^2 \\ \mathcal{I}_2(\mathbf{T})|^{3D} &= \frac{1}{2} \text{Tr}[\mathbf{T}^2] = \frac{1}{2} \sum_{ij} T_{ij} T_{ji} = \frac{1}{2} (T_{xx}^2 + T_{yy}^2 + T_{zz}^2) + T_{xy}^2 + T_{xz}^2 + T_{yz}^2\end{aligned}\quad (474)$$

- third invariant:

$$\mathcal{I}_3(\mathbf{T}) = \frac{1}{3} \text{Tr}[\mathbf{T}^3] = \frac{1}{3} \sum_i \sum_j \sum_k T_{ij} T_{jk} T_{ki}\quad (475)$$

The implementation of the plasticity criterions relies essentially on the second invariants of the (deviatoric) stress $\boldsymbol{\tau}$ and the (deviatoric) strainrate tensors $\dot{\boldsymbol{\varepsilon}}$:

$$\begin{aligned}\mathcal{I}_2(\boldsymbol{\tau})|^{2D} &= \frac{1}{2} (\tau_{xx}^2 + \tau_{yy}^2) + \tau_{xy}^2 \\ &= \frac{1}{4} (\sigma_{xx} - \sigma_{yy})^2 + \sigma_{xy}^2 \\ &= \frac{1}{4} (\sigma_1 - \sigma_2)^2 \\ \mathcal{I}_2(\boldsymbol{\tau})|^{3D} &= \frac{1}{2} (\tau_{xx}^2 + \tau_{yy}^2 + \tau_{zz}^2) + \tau_{xy}^2 + \tau_{xz}^2 + \tau_{yz}^2 \\ &= \frac{1}{6} [(\sigma_{xx} - \sigma_{yy})^2 + (\sigma_{yy} - \sigma_{zz})^2 + (\sigma_{xx} - \sigma_{zz})^2] + \sigma_{xy}^2 + \sigma_{xz}^2 + \sigma_{yz}^2 \\ &= \frac{1}{6} [(\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_1 - \sigma_3)^2] \\ \mathcal{I}_2(\dot{\boldsymbol{\varepsilon}}^d)|^{2D} &= \frac{1}{2} [(\dot{\varepsilon}_{xx}^d)^2 + (\dot{\varepsilon}_{yy}^d)^2] + (\dot{\varepsilon}_{xy}^d)^2 \\ &= \frac{1}{2} \left[\frac{1}{4} (\dot{\varepsilon}_{xx} - \dot{\varepsilon}_{yy})^2 + \frac{1}{4} (\dot{\varepsilon}_{yy} - \dot{\varepsilon}_{xx})^2 \right] + \dot{\varepsilon}_{xy}^2 \\ &= \frac{1}{4} (\dot{\varepsilon}_{xx} - \dot{\varepsilon}_{yy})^2 + \dot{\varepsilon}_{xy}^2 \\ \mathcal{I}_2(\dot{\boldsymbol{\varepsilon}}^d)|^{3D} &= \frac{1}{2} [(\dot{\varepsilon}_{xx}^d)^2 + (\dot{\varepsilon}_{yy}^d)^2 + (\dot{\varepsilon}_{zz}^d)^2] + (\dot{\varepsilon}_{xy}^d)^2 + (\dot{\varepsilon}_{xz}^d)^2 + (\dot{\varepsilon}_{yz}^d)^2 \\ &= \frac{1}{6} [(\dot{\varepsilon}_{xx} - \dot{\varepsilon}_{yy})^2 + (\dot{\varepsilon}_{yy} - \dot{\varepsilon}_{zz})^2 + (\dot{\varepsilon}_{xx} - \dot{\varepsilon}_{zz})^2] + \dot{\varepsilon}_{xy}^2 + \dot{\varepsilon}_{xz}^2 + \dot{\varepsilon}_{yz}^2\end{aligned}$$

Note that these (second) invariants are almost always used under a square root so we define:

$$\tau_e = \sqrt{\mathcal{I}_2(\boldsymbol{\tau})} \quad \dot{\varepsilon}_e = \sqrt{\mathcal{I}_2(\dot{\boldsymbol{\varepsilon}}^d)}$$

Note that these quantities have the same dimensions as their tensor counterparts, i.e. Pa for stresses and s^{-1} for strain rates.

The third (deviatoric) stress invariant is given by: (VERIFY!!)

$$\begin{aligned} \mathcal{I}_3(\boldsymbol{\tau})|^{3D} &= \frac{1}{3}s_{xx}(s_{xx}^2 + 3s_{xy}^2 + 3s_{xz}^2) \\ &+ \frac{1}{3}s_{yy}(3s_{xy}^2 + s_{yy}^2 + 3s_{yz}^2) \\ &+ \frac{1}{3}s_{zz}(3s_{xz}^2 + 3s_{yz}^2 + s_{zz}^2) \\ &+ 2s_{xy}s_{xz}s_{yz} \\ &= s_1s_2s_3 \end{aligned}$$

8.11.2 Scalar viscoplasticity

This formulation is quite easy to implement. It is widely used, e.g. [1935, 1761, 1685], and relies on the assumption that a scalar quantity η_p (the 'effective plastic viscosity') exists such that the deviatoric stress tensor

$$\boldsymbol{\tau} = 2\eta_p \dot{\boldsymbol{\varepsilon}} \quad (476)$$

is bounded by some yield stress value Y . From Eq. (476) it follows that $\underline{\tau}_{II} = 2\eta_p \dot{\underline{\varepsilon}}_{II} = Y$ which yields

$$\eta_p = \frac{Y}{2\underline{\dot{\varepsilon}}_{II}}$$

This approach has also been coined the Viscosity Rescaling Method (VRM) [1008].

insert here the rederivation 2.1.1 of spmw16

It is at this stage important to realise that (i) in areas where the strainrate is low, the resulting effective viscosity will be large, and (ii) in areas where the strainrate is high, the resulting effective viscosity will be low. This is not without consequences since (effective) viscosity contrasts up to 8-10 orders of magnitude have been observed/obtained with this formulation and it makes the FE matrix very stiff, leading to (iterative) solver convergence issues. In order to contain these viscosity contrasts one usually resorts to viscosity limiters η_{min} and η_{max} such that

$$\eta_{min} \leq \eta_p \leq \eta_{max}$$

Caution must be taken when choosing both values as they may influence the final results.

▷ `python_codes/fieldstone_indentor`

8.11.3 About the yield stress value Y

In geodynamics the yield stress value is often given as a simple function. It can be constant (in space and time) and in this case we are dealing with a von Mises plasticity yield criterion. . We simply assume $Y_{vM} = C$ where C is a constant cohesion independent of pressure, strainrate, deformation history, etc ...

Another model is often used: the Drucker-Prager plasticity model. A friction angle ϕ is then introduced and the yield value Y takes the form

$$Y_{DP} = p \sin \phi + C \cos \phi$$

and therefore depends on the pressure p . Because ϕ is with the range $[0^\circ, 45^\circ]$, Y is found to increase with depth (since the lithostatic pressure often dominates the overpressure).

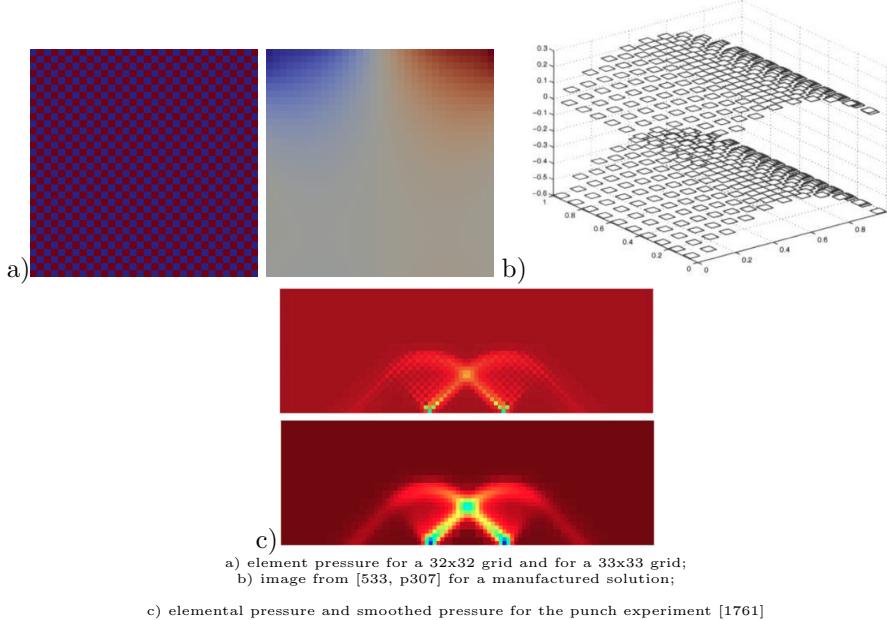
Note that a slightly modified version of this plasticity model has been used: the total pressure p is then replaced by the lithostatic pressure p_{lith} .

8.12 Pressure smoothing

It has been widely documented that the use of the $Q_1 \times P_0$ element is not without problems. Aside from the consequences it has on the FE matrix properties, we will here focus on another unavoidable side effect: the spurious pressure checkerboard modes.

These modes have been thoroughly analysed [817, 350, 1622, 1623]. They can be filtered out [350] or simply smoothed [1147].

On the following figure (a,b), pressure fields for the lid driven cavity experiment are presented for both an even and un-even number of elements. We see that the amplitude of the modes can sometimes be so large that the 'real' pressure is not visible and that something as simple as the number of elements in the domain can trigger those or not at all.



The easiest post-processing step that can be used (especially when a regular grid is used) is explained in [1761]: "The element-to-node interpolation is performed by averaging the elemental values from elements common to each node; the node-to-element interpolation is performed by averaging the nodal values element-by-element. This method is not only very efficient but produces a smoothing of the pressure that is adapted to the local density of the octree. Note that these two steps can be repeated until a satisfying level of smoothness (and diffusion) of the pressure field is attained."

In the codes which rely on the $Q_1 \times P_0$ element, the (elemental) pressure is simply defined as

```
p=np.zeros(nel, dtype=np.float64)
```

while the nodal pressure is then defined as

```
q=np.zeros(nnp, dtype=np.float64)
```

The element-to-node algorithm is then simply (in 2D):

```
count=np.zeros(nnp, dtype=np.int16)
for iel in range(0,nel):
    q[icon[0,iel]]+=p[iel]
    q[icon[1,iel]]+=p[iel]
    q[icon[2,iel]]+=p[iel]
    q[icon[3,iel]]+=p[iel]
    count[icon[0,iel]]+=1
    count[icon[1,iel]]+=1
    count[icon[2,iel]]+=1
    count[icon[3,iel]]+=1
q=q/count
```

Pressure smoothing is further discussed in [944].

[produce figure to explain this](#)

[link to proto paper](#)

[link to least square and nodal derivatives](#)

8.13 Pressure scaling

As perfectly explained in the step 32 of deal.ii³³, we often need to scale the \mathbb{G} term since it is many orders of magnitude smaller than \mathbb{K} , which introduces large inaccuracies in the solving process to the point that the solution is nonsensical. This scaling coefficient is η/L where η and L are representative viscosities and lengths. We start from

$$\begin{pmatrix} \mathbb{K} & \mathbb{G} \\ \mathbb{G}^T & -\mathbb{C} \end{pmatrix} \cdot \begin{pmatrix} \vec{\mathcal{V}} \\ \vec{\mathcal{P}} \end{pmatrix} = \begin{pmatrix} \vec{f} \\ \vec{h} \end{pmatrix}$$

and introduce the scaling coefficient as follows (which in fact does not alter the solution at all):

$$\begin{pmatrix} \mathbb{K} & \frac{\eta}{L}\mathbb{G} \\ \frac{\eta}{L}\mathbb{G}^T & -\frac{\eta^2}{L^2}\mathbb{C} \end{pmatrix} \cdot \begin{pmatrix} \vec{\mathcal{V}} \\ \underline{\mathcal{P}} \end{pmatrix} = \begin{pmatrix} \vec{f} \\ \underline{h} \end{pmatrix}$$

We then end up with the modified Stokes system:

$$\begin{pmatrix} \mathbb{K} & \mathbb{G} \\ \underline{\mathbb{G}}^T & \underline{\mathbb{C}} \end{pmatrix} \cdot \begin{pmatrix} \vec{\mathcal{V}} \\ \underline{\mathcal{P}} \end{pmatrix} = \begin{pmatrix} \vec{f} \\ \underline{h} \end{pmatrix}$$

where

$$\underline{\mathbb{G}} = \frac{\eta}{L}\mathbb{G} \quad \underline{\mathcal{P}} = \frac{L}{\eta}\vec{\mathcal{P}} \quad \underline{\mathbb{C}} = \frac{\eta^2}{L^2}\mathbb{C} \quad \underline{h} = \frac{\eta}{L}\vec{h}$$

After the solve phase, we recover the real pressure with $\vec{\mathcal{P}} = \frac{\eta}{L}\underline{\mathcal{P}}$.

³³https://www.dealii.org/9.0.0/doxygen/deal.II/step_32.html

8.14 Pressure normalisation

8.14.1 Basic idea and naive implementation

When Dirichlet boundary conditions are imposed everywhere on the boundary, pressure is only present by its gradient in the equations. It is thus determined up to an arbitrary constant (one speaks then of a nullspace of size 1). In such a case, one commonly impose the average of the pressure over the whole domain or on a subset of the boundary to have a zero average, i.e.

$$\int_{\Omega} p dV = 0 \quad (477)$$

Another possibility is to impose the pressure value at a single node.

Let us assume for example that we are using $Q_1 \times P_0$ elements. Then the pressure is constant inside each element. The integral above becomes:

$$\int_{\Omega} p dV = \sum_e \int_{\Omega_e} p dV = \sum_e p_e \int_{\Omega_e} dV = \sum_e p_e A_e = 0 \quad (478)$$

where the sum runs over all elements e of area A_e . This can be rewritten

$$\mathbb{L}^T \cdot \vec{\mathcal{P}} = 0$$

and it is a constraint on the pressure solution which couples *all* pressure dofs. We can associate to it a Lagrange multiplier λ so that we must solve the modified Stokes system:

$$\begin{pmatrix} \mathbb{K} & \mathbb{G} & 0 \\ \mathbb{G}^T & 0 & \mathbb{L} \\ 0 & \mathbb{L}^T & 0 \end{pmatrix} \cdot \begin{pmatrix} \vec{\mathcal{V}} \\ \vec{\mathcal{P}} \\ \lambda \end{pmatrix} = \begin{pmatrix} \vec{f} \\ \vec{h} \\ 0 \end{pmatrix}$$

When higher order spaces are used for pressure (continuous or discontinuous) one must then carry out the above integration numerically by means of (usually) a Gauss-Legendre quadrature.

Although valid, this approach has one main disadvantage: it makes the Stokes matrix larger (although marginally so – only one row and column are added), but more importantly it prevents the use of some of the solving strategies of Section 8.15.

8.14.2 Implementation – the real deal

The idea is actually quite simple and requires two steps:

1. remove the null space by prescribing the pressure at one location and solve the system;
2. post-process the pressure so as to arrive at a pressure field which fulfills the required normalisation (surface, volume, ...)

The reason why it works is as follows: a constant pressure value lies in the null space, so that one can add or delete any value to the pressure field without consequence. As such I can choose said constant such that the pressure at a given node/element is zero. All other computed pressures are then relative to that one. The post-processing step will redistribute a constant value to all pressures (it will shift them up or down) so that the normalising condition is respected.

8.15 Solving the Stokes system

Let us start again from the (full) Stokes system:

$$\begin{pmatrix} \mathbb{K} & \mathbb{G} \\ \mathbb{G}^T & -\mathbb{C} \end{pmatrix} \cdot \begin{pmatrix} \vec{\mathcal{V}} \\ \vec{\mathcal{P}} \end{pmatrix} = \begin{pmatrix} \vec{f} \\ \vec{h} \end{pmatrix} \quad (479)$$

We need to solve this system in order to obtain the solution, i.e. the $\vec{\mathcal{V}}$ and $\vec{\mathcal{P}}$ vectors. But how? Unfortunately, this question is not simple to answer and the appropriate method depends on many parameters, but mainly on how big the matrix blocks are and what the condition number of the matrix \mathbb{K} is.

In what follow I cover:

- solving when the penalty approach is used
- the Schur complement approach
- the FGMRES approach [?]



8.15.1 when using the penalty formulation

In this case we are only solving for velocity since pressure is recovered in a post-processing step:

$$(\mathbb{K}_\eta + \mathbb{K}_\lambda) \cdot \vec{\mathcal{V}} = \vec{f}$$

We also know that the penalty factor is many orders of magnitude higher than the viscosity and in combination with the use of the $Q_1 \times P_0$ element the resulting matrix condition number is very high so that the use of iterative solvers is precluded. Indeed codes such as SOPALE [678], DOUAR [221], or FANTOM [1756] relying on the penalty formulation all use direct solvers. The most popular are BLKFCT³⁴, MUMPS³⁵ [22, 24, 23, 25], WSMP³⁶ [831, 832], UMFPACK and CHOLMOD³⁷, SuperLU, PARDISO³⁸ [504, 1873, 1110], or those inside PETSc ??.

Braun et al [221] list the following features of such solvers:

- Robust
- Black-box operation
- Difficult to parallelize
- Memory consumption
- Limited scalability

The main advantage of direct solvers is used in this case: They can solve ill-conditioned matrices. However memory requirements for the storage of number of nonzeros in the Cholesky matrix grow very fast as the number of equations/grid size increases, especially in 3D, to the point that even modern computers with tens of Gb of RAM cannot deal with a 100^3 element mesh. This explains why direct solvers are often used for 2D problems and rarely in 3D with noticeable exceptions [1761, 1964, 223, 1226, 19, 20, 21, 1926, 1405].

³⁴<http://dm.unife.it/blkfclt/>

³⁵<http://mumps.enseeicht.fr/>

³⁶<http://www.research.ibm.com/projects/wsmp>

³⁷<http://faculty.cse.tamu.edu/davis/suitesparse.html>

³⁸<https://www.pardiso-project.org/>

8.15.2 Conjugate gradient and the Schur complement approach

Let us write the above system as two equations:

$$\mathbb{K} \cdot \vec{\mathcal{V}} + \mathbb{G} \cdot \vec{\mathcal{P}} = \vec{f} \quad (480)$$

$$\mathbb{G}^T \cdot \vec{\mathcal{V}} = \vec{h} \quad (481)$$

The first line can be re-written $\vec{\mathcal{V}} = \mathbb{K}^{-1} \cdot (\vec{f} - \mathbb{G} \cdot \vec{\mathcal{P}})$ and can be inserted in the second:

$$\mathbb{G}^T \cdot \vec{\mathcal{V}} = \mathbb{G}^T \cdot [\mathbb{K}^{-1} \cdot (\vec{f} - \mathbb{G} \cdot \vec{\mathcal{P}})] = \vec{h} \quad (482)$$

or,

$$(\mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot \mathbb{G}) \cdot \vec{\mathcal{P}} = \mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot \vec{f} - \vec{h} \quad (483)$$

The matrix $\mathbb{S} = \mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot \mathbb{G}$ is called the Schur complement. It is Symmetric (since \mathbb{K} is symmetric) and Positive-Definite³⁹ (SPD) if $\text{Ker}(\mathbb{G}) = 0$. [look in donea-huerta book for details](#) Having solved this equation (we have obtained $\vec{\mathcal{P}}$), the velocity can be recovered by solving $\mathbb{K} \cdot \vec{\mathcal{V}} = \vec{f} - \mathbb{G} \cdot \vec{\mathcal{P}}$.

For now, let us assume that we have built the \mathbb{S} matrix and the right hand side $\vec{f} = \mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot \vec{f} - \vec{h}$. We must solve $\mathbb{S} \cdot \vec{\mathcal{P}} = \vec{f}$.

One can resort to so-called Richardson iterations, defined as follows (e.g., see [1867], p141): in solving the matrix equation $\mathbf{A} \cdot \vec{X} = \vec{b}$, the Richardson iterative method is defined by:

$$\vec{X}_{k+1} = \vec{X}_k + \alpha_k (-\mathbf{A} \cdot \vec{X}_k + \vec{b}) \quad m \geq 0 \quad (484)$$

where the α_k 's are real scalars. It is easy to see that when the method converges then $\vec{X}_{k+1} \simeq \vec{X}_k$ and then $\mathbf{A} \cdot \vec{X} = \vec{b}$ is satisfied. In our case, it writes:

$$\begin{aligned} \vec{\mathcal{P}}_{k+1} &= \vec{\mathcal{P}}_k + \alpha_k (-\mathbb{S} \cdot \vec{\mathcal{P}}_k + \vec{f}) \\ &= \vec{\mathcal{P}}_k + \alpha_k (-\mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot \mathbb{G} \cdot \vec{\mathcal{P}}_k + \mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot \vec{f} - \vec{h}) \\ &= \vec{\mathcal{P}}_k + \alpha_k [\mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot (-\mathbb{G} \cdot \vec{\mathcal{P}}_k + \vec{f}) - \vec{h}] \\ &= \vec{\mathcal{P}}_k + \alpha_k [\mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot (\mathbb{K} \cdot \vec{\mathcal{V}}_k) - \vec{h}] \\ &= \vec{\mathcal{P}}_k + \alpha_k (\mathbb{G}^T \cdot \vec{\mathcal{V}}_k - \vec{h}) \end{aligned} \quad (485)$$

The above iterations are then carried out and for each new pressure field the associated velocity field is computed. The method of using Richardson iterations applied to the Schur complement is commonly called the Uzawa algorithm [209, p221].

Uzawa algorithm (1):

$$\text{solve } \mathbb{K} \cdot \vec{\mathcal{V}}_k = \vec{f} - \mathbb{G} \cdot \vec{\mathcal{P}}_{k-1} \quad (486)$$

$$\vec{\mathcal{P}}_k = \vec{\mathcal{P}}_{k-1} + \alpha_k (\mathbb{G}^T \cdot \vec{\mathcal{V}}_k - \vec{h}) \quad k = 1, 2, \dots \quad (487)$$

This method is rather simple to implement, although what makes an appropriate set of α_k values is not straightforward, which is why the conjugate gradient is often preferred, as detailed in the next subsection.

It is known that such iterations will converge for $0 < \alpha < \rho(\mathbb{S}) = \lambda_{max}(\mathbb{S})$ where $\rho(\mathbb{S})$ is the spectral radius of the matrix \mathbb{S} which is essentially the largest, in absolute value, eigenvalue of \mathbb{S} (neither of which can be computed easily). It can also be proven that the rate of convergence depends on the condition number of the matrix.

Richardson iterations are part of the family of stationary iterative methods, since it can be rewritten

$$\vec{X}_{k+1} = (\mathbf{I} - \alpha_k \mathbf{A}) \cdot \vec{X}_k + \alpha_k \vec{b} \quad (488)$$

³⁹ M positive definite $\iff x^T M x > 0 \forall x \in \mathbb{R}^n \setminus \mathbf{0}$

which is the definition of a stationary method.

Since the α parameter is the key to a successful Uzawa algorithm, this issue has of course been looked into. What follows is presented in [209, p221]. For the analysis of the Uzawa algorithm, we define the residue

$$\vec{\mathcal{R}}_k = \vec{h} - \mathbb{G}^T \cdot \vec{\mathcal{V}}_k$$

In addition, suppose the solution of the saddle point problem is denoted by $(\mathcal{V}^*, \mathcal{P}^*)$. Now substituting the iteration formula for \mathcal{V}_k , we get

$$\mathcal{R}_k = \mathbb{G}^T \cdot \vec{\mathcal{V}}^* - \mathbb{G}^T \cdot \mathbb{K}^{-1}(\vec{f} - \mathbb{G} \cdot \mathcal{P}_{k-1}) \quad (489)$$

$$= \mathbb{G}^T \cdot \vec{\mathcal{V}}^* - \mathbb{G}^T \cdot \mathbb{K}^{-1}(\mathbb{K} \cdot \vec{\mathcal{V}}^* + \mathbb{G} \cdot \vec{\mathcal{P}}^* - \mathbb{G} \cdot \mathcal{P}_{k-1}) \quad (490)$$

$$= \mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot \mathbb{G} \cdot (\vec{\mathcal{P}}_{k-1} - \vec{\mathcal{P}}^*) \quad (491)$$

From Eq. 487 it follows that:

$$\mathcal{P}_k - \mathcal{P}_{k-1} = \alpha(\mathbb{G}^T \cdot \vec{\mathcal{V}}_k - \vec{h}) \quad (492)$$

$$= -\alpha \vec{\mathcal{R}}_k \quad (493)$$

$$= -\alpha \mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot \mathbb{G} \cdot (\vec{\mathcal{P}}_{k-1} - \vec{\mathcal{P}}^*) \quad (494)$$

$$= \alpha \mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot \mathbb{G} \cdot (\vec{\mathcal{P}}^* - \vec{\mathcal{P}}_{k-1}) \quad (495)$$

Thus the Uzawa algorithm is equivalent to applying the gradient method to the reduced equation using a fixed step size. In particular, the iteration converges for $\alpha < 2\|\mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot \mathbb{G}\|^{-1}$ and one can show that the good step size α_k is given by

$$\alpha_k = \frac{\mathcal{R}_k \cdot \mathcal{R}_k}{(\mathbb{G} q_k) \cdot (\mathbb{K}^{-1} \mathbb{G} q_k)} \quad (496)$$

However, if we were to use this rule formally, we would need an additional multiplication by \mathbb{K}^{-1} in every step of the iteration. This can be avoided by storing an auxiliary vector.

Note that in [767] it is stated: the convergence of this algorithm is proved for $\alpha \in (0, 2\mu/d)$ (where d is the number of dimensions).

check this, and report page number

Note that this algorithm is presented in [2036] in the context of viscoplastic flow.

As mentioned above, there is a way to rework the original Uzawa algorithm to include Eq. (496). It yields a modified Uzawa algorithm [209, p221]:

Uzawa algorithm (2): Solve $\mathbb{K} \cdot \vec{\mathcal{V}}_1 = \vec{f} - \mathbb{G} \cdot \vec{\mathcal{P}}_0$. For $k = 1, 2, \dots$, compute

$$\vec{q}_k = \vec{h} - \mathbb{G}^T \cdot \vec{\mathcal{V}}_k \quad (497)$$

$$\vec{p}_k = \mathbb{G} \cdot q_k \quad (498)$$

$$\vec{H}_k = \mathbb{K}^{-1} \cdot \vec{p}_k \quad (499)$$

$$\alpha_k = \frac{\vec{q}_k \cdot \vec{q}_k}{\vec{p}_k \cdot \vec{H}_k} \quad (500)$$

$$\vec{\mathcal{P}}_k = \vec{\mathcal{P}}_{k-1} - \alpha_k \vec{q}_k \quad (501)$$

$$\vec{\mathcal{V}}_{k+1} = \vec{\mathcal{V}}_k + \alpha_k \vec{H}_k \quad (502)$$

 Literature[323, 326]

8.15.3 Conjugate gradient and the Schur complement approach

Since \mathbb{S} is SPD, the Conjugate Gradient (CG) method is very appropriate to solve this system. Indeed, looking at the definition of Wikipedia: "In mathematics, the conjugate gradient method is an algorithm for the numerical solution of particular systems of linear equations, namely those whose matrix is symmetric and positive-definite. The conjugate gradient method is often implemented as an iterative algorithm, applicable to sparse systems that are too large to be handled by a direct implementation or other direct

methods such as the Cholesky decomposition. Large sparse systems often arise when numerically solving partial differential equations or optimization problems.”

A simple Google search tells us that the Conjugate Gradient algorithm is as follows:

```

 $\mathbf{r}_0 := \mathbf{b} - \mathbf{Ax}_0$ 
if  $\mathbf{r}_0$  is sufficiently small, then return  $\mathbf{x}_0$  as the result
 $\mathbf{p}_0 := \mathbf{r}_0$ 
 $k := 0$ 
repeat
   $\alpha_k := \frac{\mathbf{r}_k^\top \mathbf{r}_k}{\mathbf{p}_k^\top \mathbf{Ap}_k}$ 
   $\mathbf{x}_{k+1} := \mathbf{x}_k + \alpha_k \mathbf{p}_k$ 
   $\mathbf{r}_{k+1} := \mathbf{r}_k - \alpha_k \mathbf{Ap}_k$ 
  if  $\mathbf{r}_{k+1}$  is sufficiently small, then exit loop
   $\beta_k := \frac{\mathbf{r}_{k+1}^\top \mathbf{r}_{k+1}}{\mathbf{r}_k^\top \mathbf{r}_k}$ 
   $\mathbf{p}_{k+1} := \mathbf{r}_{k+1} + \beta_k \mathbf{p}_k$ 
   $k := k + 1$ 
end repeat
return  $\mathbf{x}_{k+1}$  as the result

```

Algorithm as obtained from Wikipedia⁴⁰

This algorithm is of course explained in detail in many textbooks such as [1613]

[add biblio](#)

Let us look at this algorithm up close. The parts which may prove to be somewhat tricky are those involving the matrix inverse (in our case the Schur complement). We start the iterations with a guess pressure \vec{P}_0 (and an initial guess velocity which could be obtained by solving $\mathbb{K} \cdot \vec{V}_0 = \vec{f} - \mathbb{G} \cdot \vec{P}_0$).

$$\vec{r}_0 = \vec{f} - \mathbb{S} \cdot \vec{P}_0 \quad (503)$$

$$= \mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot \vec{f} - \vec{h} - (\mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot \mathbb{G}) \cdot \vec{P}_0 \quad (504)$$

$$= \mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot (\vec{f} - \mathbb{G} \cdot \vec{P}_0) - \vec{h} \quad (505)$$

$$= \mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot \mathbb{K} \cdot \vec{V}_0 - \vec{h} \quad (506)$$

$$= \mathbb{G}^T \cdot \vec{V}_0 - \vec{h} \quad (507)$$

$$(508)$$

We now turn to the α_k coefficient:

$$\alpha_k = \frac{\vec{r}_k^T \cdot \vec{r}_k}{\vec{p}_k \cdot \mathbb{S} \cdot \vec{p}_k} = \frac{\vec{r}_k^T \cdot \vec{r}_k}{\vec{p}_k \cdot \mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot \mathbb{G} \cdot \vec{p}_k} = \frac{\vec{r}_k^T \cdot \vec{r}_k}{(\mathbb{G} \cdot \vec{p}_k)^T \cdot \mathbb{K}^{-1} \cdot (\mathbb{G} \cdot \vec{p}_k)}$$

We then define $\tilde{\vec{p}}_k = \mathbb{G} \cdot \vec{p}_k$, so that α_k can be computed as follows:

1. compute $\tilde{\vec{p}}_k = \mathbb{G} \cdot \vec{p}_k$
2. solve $\mathbb{K} \cdot \vec{d}_k = \tilde{\vec{p}}_k$
3. compute $\alpha_k = (\vec{r}_k^T \cdot \vec{r}_k) / (\tilde{\vec{p}}_k^T \cdot \vec{d}_k)$

Then we need to look at the term $\mathbb{S} \cdot \vec{p}_k$:

$$\mathbb{S} \cdot \vec{p}_k = \mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot \mathbb{G} \cdot \vec{p}_k = \mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot \tilde{\vec{p}}_k = \mathbb{G}^T \cdot \vec{d}_k$$

We can then rewrite the CG algorithm as follows [2018]:

- $\vec{r}_0 = \mathbb{G}^T \cdot \vec{V}_0 - \vec{h}$

⁴⁰https://en.wikipedia.org/wiki/Conjugate_gradient_method

- if \vec{r}_0 is sufficiently small, then return $(\vec{\mathcal{V}}_0, \vec{\mathcal{P}}_0)$ as the result
- $\vec{p}_0 = \vec{r}_0$
- $k = 0$
- repeat
 - compute $\tilde{\vec{p}}_k = \mathbb{G} \cdot \vec{p}_k$
 - solve $\mathbb{K} \cdot \vec{d}_k = \tilde{\vec{p}}_k$
 - compute $\alpha_k = (\vec{r}_k^T \cdot \vec{r}_k) / (\tilde{\vec{p}}_k^T \cdot \vec{d}_k)$
 - $\vec{\mathcal{P}}_{k+1} = \vec{\mathcal{P}}_k + \alpha_k \vec{p}_k$
 - $\vec{r}_{k+1} = \vec{r}_k - \alpha_k \mathbb{G}^T \cdot \vec{d}_k$
 - if \vec{r}_{k+1} is sufficiently small, then exit loop
 - $\beta_k = (\vec{r}_{k+1}^T \cdot \vec{r}_{k+1}) / (\vec{r}_k^T \cdot \vec{r}_k)$
 - $\vec{p}_{k+1} = \vec{r}_{k+1} + \beta_k \vec{p}_k$
 - $k = k + 1$
- return $\vec{\mathcal{P}}_{k+1}$ as result

We see that we have managed to solve the Schur complement equation with the Conjugate Gradient method without ever building the matrix \mathbb{S} . Having obtained the pressure solution, we can easily recover the corresponding velocity with $\mathbb{K} \cdot \vec{\mathcal{V}}_{k+1} = \vec{f} - \mathbb{G} \cdot \vec{\mathcal{P}}_{k+1}$. However, this is rather unfortunate because it requires yet another solve with the \mathbb{K} matrix. As it turns out, we can slightly alter the above algorithm to have it update the velocity as well so that this last solve is unnecessary.

We have

$$\vec{\mathcal{V}}_{k+1} = \mathbb{K}^{-1} \cdot (\vec{f} - \mathbb{G} \cdot \vec{\mathcal{P}}_{p+1}) \quad (509)$$

$$= \mathbb{K}^{-1} \cdot (\vec{f} - \mathbb{G} \cdot (\vec{\mathcal{P}}_k + \alpha_k \vec{p}_k)) \quad (510)$$

$$= \mathbb{K}^{-1} \cdot (\vec{f} - \mathbb{G} \cdot \vec{\mathcal{P}}_k) - \alpha_k \mathbb{K}^{-1} \cdot \mathbb{G} \cdot \vec{p}_k \quad (511)$$

$$= \vec{\mathcal{V}}_k - \alpha_k \mathbb{K}^{-1} \cdot \tilde{\vec{p}}_k \quad (512)$$

$$= \vec{\mathcal{V}}_k - \alpha_k \vec{d}_k \quad (513)$$

and we can insert this minor extra calculation inside the algorithm and get the velocity solution nearly for free. The final CG algorithm is then

solver_cg:

- compute $\vec{\mathcal{V}}_0 = \mathbb{K}^{-1} \cdot (\vec{f} - \mathbb{G} \cdot \vec{\mathcal{P}}_0)$
- $\vec{r}_0 = \mathbb{G}^T \cdot \vec{\mathcal{V}}_0 - \vec{h}$
- if \vec{r}_0 is sufficiently small, then return $(\vec{\mathcal{V}}_0, \vec{\mathcal{P}}_0)$ as the result
- $\vec{p}_0 = \vec{r}_0$
- $k = 0$
- repeat
 - compute $\tilde{\vec{p}}_k = \mathbb{G} \cdot \vec{p}_k$
 - solve $\mathbb{K} \cdot \vec{d}_k = \tilde{\vec{p}}_k$
 - compute $\alpha_k = (\vec{r}_k^T \cdot \vec{r}_k) / (\tilde{\vec{p}}_k^T \cdot \vec{d}_k)$
 - $\vec{\mathcal{P}}_{k+1} = \vec{\mathcal{P}}_k + \alpha_k \vec{p}_k$
 - $\vec{\mathcal{V}}_{k+1} = \vec{\mathcal{V}}_k - \alpha_k \vec{d}_k$
 - $\vec{r}_{k+1} = \vec{r}_k - \alpha_k \mathbb{G}^T \cdot \vec{d}_k$

- if \vec{r}_{k+1} is sufficiently small ($\|\vec{r}_{k+1}\|_2 / \|\vec{r}_0\|_2 < tol$), then exit loop
- $\beta_k = (r_{k+1}^T r_{k+1}) / (r_k^T r_k)$
- $\vec{p}_{k+1} = \vec{r}_{k+1} + \beta_k \vec{p}_k$
- $k = k + 1$
- return $\vec{\mathcal{P}}_{k+1}$ as result

This iterative algorithm will converge to the solution with a rate which depends on the condition number of the \mathbb{S} matrix, which is not easy to compute since \mathbb{S} is never built. However, it has been established that large viscosity contrasts in the domain will have a negative impact on the convergence.

Remark. This algorithm requires one solve with matrix \mathbb{K} per iteration but says nothing about the method employed to do so (direct solver, iterative solver, ...)

One thing we know improves the convergence of any iterative solver is the use of a preconditioner matrix and therefore now focus on the Preconditioned Conjugate Gradient (PCG) method. Once again a quick Google search yields:

```

r0 := b - Ax0
z0 := M-1r0
p0 := z0
k := 0
repeat
     $\alpha_k := \frac{\mathbf{r}_k^T \mathbf{z}_k}{\mathbf{p}_k^T \mathbf{A} \mathbf{p}_k}$ 
     $\mathbf{x}_{k+1} := \mathbf{x}_k + \alpha_k \mathbf{p}_k$ 
     $\mathbf{r}_{k+1} := \mathbf{r}_k - \alpha_k \mathbf{A} \mathbf{p}_k$ 
    if  $\mathbf{r}_{k+1}$  is sufficiently small then exit loop end if
     $\mathbf{z}_{k+1} := \mathbf{M}^{-1} \mathbf{r}_{k+1}$ 
     $\beta_k := \frac{\mathbf{z}_{k+1}^T \mathbf{r}_{k+1}}{\mathbf{z}_k^T \mathbf{r}_k}$ 
     $\mathbf{p}_{k+1} := \mathbf{z}_{k+1} + \beta_k \mathbf{p}_k$ 
    k := k + 1
end repeat
The result is xk+1

```

Algorithm obtained from Wikipedia⁴¹.

Note that in the algorithm above the preconditioner matrix M has to be symmetric positive-definite and fixed, i.e., cannot change from iteration to iteration. We see that this algorithm introduces an additional vector \vec{z} and a solve with the matrix M at each iteration, which means that M must be such that solving $M \cdot \vec{x} = \vec{f}$ where \vec{f} is a given rhs vector must be cheap. Ultimately, the PCG algorithm applied to the Schur complement equation takes the form:

solver_pcg:

- compute $\mathcal{V}_0 = \mathbb{K}^{-1}(f - \mathbb{G}\mathcal{P}_0)$
- $r_0 = \mathbb{G}^T \mathcal{V}_0 - h$
- if \vec{r}_0 is sufficiently small, then return $(\vec{\mathcal{V}}_0, \vec{\mathcal{P}}_0)$ as the result
- $\vec{z}_0 = M^{-1} \cdot \vec{r}_0$
- $\vec{p}_0 = \vec{z}_0$
- $k = 0$

⁴¹https://en.wikipedia.org/wiki/Conjugate_gradient_method

- repeat
 - compute $\tilde{\vec{p}}_k = \mathbb{G} \cdot \vec{p}_k$
 - solve $\mathbb{K} \cdot \vec{d}_k = \tilde{\vec{p}}_k$
 - compute $\alpha_k = (\vec{r}_k^T \cdot \vec{z}_k) / (\tilde{\vec{p}}_k^T \cdot \vec{d}_k)$
 - $\vec{\mathcal{P}}_{k+1} = \mathcal{P}_k + \alpha_k \vec{p}_k$
 - $\vec{\mathcal{V}}_{k+1} = \mathcal{V}_k - \alpha_k \vec{d}_k$
 - $\vec{r}_{k+1} = \vec{r}_k - \alpha_k \mathbb{G}^T \cdot \vec{d}_k$
 - if r_{k+1} is sufficiently small ($\|r_{k+1}\|_2 / \|r_0\|_2 < tol$), then exit loop
 - $\vec{z}_{k+1} = M^{-1} \cdot r_{k+1}$
 - $\beta_k = (\vec{z}_{k+1}^T \cdot \vec{r}_{k+1}) / (\vec{z}_k^T \cdot \vec{r}_k)$
 - $\vec{p}_{k+1} = \vec{z}_{k+1} + \beta_k \vec{p}_k$
 - $k = k + 1$
- return $\vec{\mathcal{P}}_{k+1}$ as result

Following [2018] one can define the following matrix as preconditioner:

$$M = \text{diag} [\mathbb{G}^T (\text{diag}[\mathbb{K}])^{-1} \mathbb{G}]$$

which is the preconditioner used for the Citcom codes (see appendix ??). It can be constructed while the FEM matrix is being built/assembled and it is trivial to invert.

how to compute M for the Schur complement ?

8.15.4 The Augmented Lagrangian approach

see LaCoDe paper [508].

We start from the saddle point Stokes system:

$$\begin{pmatrix} \mathbb{K} & \mathbb{G} \\ \mathbb{G}^T & 0 \end{pmatrix} \cdot \begin{pmatrix} \vec{\mathcal{V}} \\ \vec{\mathcal{P}} \end{pmatrix} = \begin{pmatrix} \vec{f} \\ \vec{h} \end{pmatrix} \quad (514)$$

The AL method consists of subtracting $\lambda^{-1} \mathbb{M}_p \cdot \vec{\mathcal{P}}$ from the left and right-side of the mass conservation equation (where \mathbb{M}_p is the pressure mass matrix) and introducing the following iterative scheme:

$$\begin{pmatrix} \mathbb{K} & \mathbb{G} \\ \mathbb{G}^T & -\lambda^{-1} \mathbb{M}_p \end{pmatrix} \cdot \begin{pmatrix} \vec{\mathcal{V}}^{k+1} \\ \vec{\mathcal{P}}^{k+1} \end{pmatrix} = \begin{pmatrix} \vec{f} \\ \vec{h} - \lambda^{-1} \mathbb{M}_p \cdot \vec{\mathcal{P}}^k \end{pmatrix} \quad (515)$$

where k is the iteration counter and λ is an artificial compressibility term which has the dimensions of dynamic viscosity. The choice of λ can be difficult as too low or too high a value yields either erroneous results and/or terribly ill-conditioned matrices. LaCoDe paper (!! use such a method and report that $\lambda = \max_\Omega(\eta)$) works well. Note that at convergence we have $\|\vec{\mathcal{P}}^{k+1} - \vec{\mathcal{P}}^k\| < \epsilon$ and then Eq.(515) converges to Eq.(514) and the velocity and pressure fields are solution of the unmodified system Eq.(514).

The introduction of this term serves one purpose: allowing us to solve the system in a segregated manner (i.e. computing successive iterates of the velocity and pressure fields until convergence is reached). The second line of Eq. (515) is

$$\mathbb{G}^T \cdot \vec{\mathcal{V}}^{k+1} - \lambda^{-1} \mathbb{M}_p \cdot \vec{\mathcal{P}}^{k+1} = \vec{h} - \lambda^{-1} \mathbb{M}_p \cdot \vec{\mathcal{P}}^k$$

and can therefore be rewritten

$$\vec{\mathcal{P}}^{k+1} = \vec{\mathcal{P}}^k + \lambda \mathbb{M}_p^{-1} \cdot (\mathbb{G}^T \cdot \vec{\mathcal{V}}^{k+1} - \vec{h})$$

We can then substitute this expression of $\vec{\mathcal{P}}^{k+1}$ in the first equation. This yields:

$$\mathbb{K} \cdot \vec{\mathcal{V}}^{k+1} = \vec{f} - \mathbb{G} \cdot \vec{\mathcal{P}}^{k+1}) \quad (516)$$

$$\mathbb{K} \cdot \vec{\mathcal{V}}^{k+1} = \vec{f} - \mathbb{G} \cdot (\vec{\mathcal{P}}^k + \lambda \mathbb{M}_p^{-1} \cdot (\mathbb{G}^T \cdot \vec{\mathcal{V}}^{k+1} - \vec{h})) \quad (517)$$

$$\mathbb{K} \cdot \vec{\mathcal{V}}^{k+1} + \lambda \mathbb{G} \cdot \mathbb{M}_p^{-1} \cdot \mathbb{G}^T \cdot \vec{\mathcal{V}}^{k+1} = \vec{f} - \mathbb{G} \cdot (\vec{\mathcal{P}}^k - \lambda \mathbb{M}_p^{-1} \vec{h}) \quad (518)$$

$$\underbrace{(\mathbb{K} + \lambda \mathbb{G} \cdot \mathbb{M}_p^{-1} \cdot \mathbb{G}^T)}_{\tilde{\mathbb{K}}} \cdot \vec{\mathcal{V}}^{k+1} = \underbrace{\vec{f} - \mathbb{G} \cdot (\vec{\mathcal{P}}^k - \lambda \mathbb{M}_p^{-1} \vec{h})}_{\vec{f}^{k+1}} \quad (519)$$

(520)

The iterative algorithm goes as follows:

1. if it is the first timestep, set $\vec{\mathcal{P}}^0 = 0$, otherwise set it to the pressure of the previous timestep.
2. calculate $\tilde{\mathbb{K}}$
3. calculate \vec{f}^{k+1}
4. solve $\tilde{\mathbb{K}} \cdot \vec{\mathcal{V}}^{k+1} = \vec{f}^{k+1}$
5. update pressure with $\vec{\mathcal{P}}^{k+1} = \vec{\mathcal{P}}^k + \lambda \mathbb{M}_p^{-1} \cdot (\mathbb{G}^T \cdot \vec{\mathcal{V}}^{k+1} - \vec{h})$

Remark. If discontinuous pressures are used, the pressure mass matrix can be inverted element by element which is cheaper than inverting \mathbb{M}_p as a whole.

Remark. This method has obvious ties with the penalty method.

Remark. If $\lambda >> \max_{\Omega} \eta$ then the matrix $\tilde{\mathbb{K}}$ is ill-conditioned and an iterative solver must be used.

8.15.5 The GMRES approach

The Generalized Minimal Residual method [1614] is an extension of MINRES (which is only applicable to symmetric systems) to unsymmetric systems. Like MINRES, it generates a sequence of orthogonal vectors and combines these through a least-squares solve and update. However, in the absence of symmetry this can no longer be done with short recurrences. As a consequence, all previously computed vectors in the orthogonal sequence have to be retained and for this reason "restarted" versions of the method are used.

It must be said that the (preconditioned) GMRES method is actually much more difficult to implement than the (preconditioned) Conjugate Gradient method. However, since it can deal with unsymmetric matrices, it means that it can be applied directly to the Stokes system matrix (as opposed to the CG method which is used on the Schur complement equation).

 Literature: [567, p208] [1613] [81] [53]

finish GMRES algo description, not sure what to do, hard to explain, not easy to code.

8.16 The consistent boundary flux (CBF)

The Consistent Boundary Flux technique was devised to alleviate the problem of the accuracy of primary variables derivatives (mainly velocity and temperature) on boundaries. These derivatives are important since they are needed to compute the heat flux (and therefore the Nusselt number) or dynamic topography and geoid.

The idea was first introduced in [1330] and later used in geodynamics [2007]. It was finally implemented in the CitcomS code [2010, 1356] and more recently in the ASPECT code (dynamic topography postprocessor). Note that the CBF should be seen as a post-processor step as it does not alter the primary variables values.

The CBF method is implemented and used in Stone ???. It is also discussed but not explicitly named in [1551, p309]. Also see [1130, 812, 1278].

8.16.1 The CBF applied to the Stokes equation

We start from the strong form:

$$\vec{\nabla} \cdot \boldsymbol{\sigma} + \vec{b} = \vec{0} \quad (521)$$

and then write the weak form on an element e :

$$\int_{\Omega_e} N_i^y \vec{\nabla} \cdot \boldsymbol{\sigma} d\Omega + \int_{\Omega_e} N_i^y \vec{b} d\Omega = \vec{0} \quad (522)$$

We then use the two equations:

$$\boldsymbol{\nabla} \cdot (N\boldsymbol{\sigma}) = N\boldsymbol{\nabla} \cdot \boldsymbol{\sigma} + \boldsymbol{\nabla}N \cdot \boldsymbol{\sigma} \quad (\text{chain rule})$$

$$\int_{\Omega} (\boldsymbol{\nabla} \cdot \boldsymbol{\sigma}) dV = \int_{\Gamma} \boldsymbol{\sigma} \cdot \mathbf{n} dS \quad (\text{divergence theorem})$$

and integrate by parts in order to obtain:

$$\int_{\Gamma} N_i^y \boldsymbol{\sigma} \cdot \mathbf{n} dS - \int_{\Omega_e} \vec{\nabla} N_i^y \cdot \boldsymbol{\sigma} d\Omega + \int_{\Omega_e} N_i^y \vec{b} d\Omega = \vec{0} \quad (523)$$

and since the traction vector \vec{t} is given by $\vec{t} = \boldsymbol{\sigma} \cdot \mathbf{n}$ we have:

$$\int_{\Gamma_e} N_i^y t dS = \int_{\Omega_e} \vec{\nabla} N_i^y \cdot \boldsymbol{\sigma} d\Omega - \int_{\Omega_e} N_i^y \vec{b} d\Omega \quad (524)$$

The core idea of the method lies in considering the traction vector as an unknown living on the nodes on the boundary, and assuming we have already solved the Stokes equation and therefore have obtained the velocity and pressure.

Finally, since the traction vector can be expressed as a function of the velocity shape functions on the edges i.e.

$$\vec{t} = \sum_{i=1}^m N_i^y \vec{t}_i$$

the left hand term yields an edge (1D) mass matrix M' (see Section I).

Remark. In Stone ??? an alternative to equation 524 is used. Although somewhat inefficient, the elemental matrices \mathbb{K} and \mathbb{G} and the corresponding body force rhs are built and the rhs of the traction equation is computed as follows:

$$M' \cdot \mathcal{T} = -\mathbb{K}\mathcal{V} - \mathbb{G}\mathcal{P} + f$$

where \mathcal{T} is the vector of assembled tractions which we want to compute and \mathcal{V} and \mathcal{T} are the solutions of the Stokes problem.

Remark. The assembled mass matrix is tri-diagonal and can be easily solved with a Conjugate Gradient method.

Remark. With a trapezoidal integration rule (i.e. Gauss-Lobatto - see Section 4.1.6) the matrix can even be diagonalised and the resulting matrix is simply diagonal, which results in a very cheap solve [2007].

8.16.2 The CBF applied to the heat transport equation

We start from the strong form of the heat transfer equation (without the source terms for simplicity):

$$\rho C_p \left(\frac{\partial T}{\partial t} + \vec{v} \cdot \vec{\nabla} T \right) = \vec{\nabla} \cdot k \vec{\nabla} T$$

The weak form then writes:

$$\int_{\Omega} N^{\theta} \rho C_p \frac{\partial T}{\partial t} dV + \rho C_p \int_{\Omega} N^{\theta} \vec{v} \cdot \vec{\nabla} T dV = \int_{\Omega} N^{\theta} \vec{\nabla} \cdot k \vec{\nabla} T dV$$

Using once again integration by parts and divergence theorem:

$$\int_{\Omega} N \rho C_p \frac{\partial T}{\partial t} dV + \rho C_p \int_{\Omega} N \vec{v} \cdot \nabla T dV = \int_{\Gamma} N k \nabla T \cdot \mathbf{n} d\Gamma - \int_{\Omega} \nabla N \cdot k \nabla T dV$$

On the boundary we are interested in the heat flux $\mathbf{q} = -k \nabla T$

$$\int_{\Omega} N \rho C_p \frac{\partial T}{\partial t} dV + \rho C_p \int_{\Omega} N \vec{v} \cdot \nabla T dV = - \int_{\Gamma} N \mathbf{q} \cdot \mathbf{n} d\Gamma - \int_{\Omega} \nabla N \cdot k \nabla T dV$$

or,

$$\int_{\Gamma} N \mathbf{q} \cdot \mathbf{n} d\Gamma = - \int_{\Omega} N \rho C_p \frac{\partial T}{\partial t} dV - \rho C_p \int_{\Omega} N \vec{v} \cdot \nabla T dV - \int_{\Omega} \nabla N \cdot k \nabla T dV$$

Considering the normal heat flux $q_n = \mathbf{q} \cdot \mathbf{n}$ as an unknown living on the nodes on the boundary,

$$q_n = \sum_{i=1}^2 q_{n|i} N_i$$

so that the left hand term becomes a mass matrix for the shape functions living on the boundary. We have already covered the right hand side terms when building the FE system to solve the heat transport equation, so that in the end

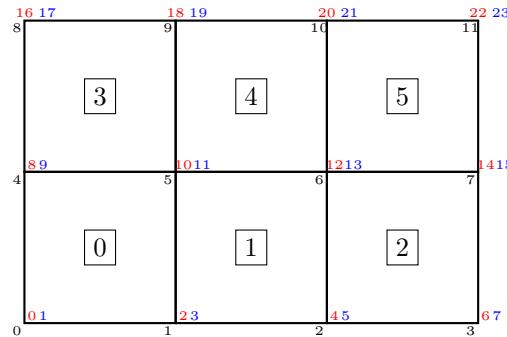
$$M' \cdot \mathcal{Q}_n = -M \cdot \frac{\partial \mathbf{T}}{\partial t} - K_a \cdot \mathbf{T} - K_d \cdot \mathbf{T}$$

where \mathcal{Q}_n is the assembled vector of normal heat flux components. Note that in all terms the assembly only takes place over the elements along the boundary.

Note that the resulting matrix is symmetric.

8.16.3 Some implementation details for the Stokes equation

What follows is relevant for Stone ?? which relies on Q_1 shape functions for the velocity. Let us start with a small example, a 3x2 element FE grid:



Red color corresponds to the dofs in the x direction, blue color indicates a dof in the y direction.

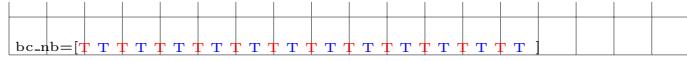
We have nnp=12, nel=6, NfemV=24. Let us assume that free slip boundary conditions are applied. The boundary conditions `fix_bc` array is then:

```
bc_fix=[T T T T T T T T T T T T T T T T T T ]
```

Note that since corners belong to two edges, we effectively prescribed no-slip boundary conditions on those.

[why does array contain only T??](#)

We wish to compute the tractions on the boundaries, and more precisely for the dofs for which a Dirichlet velocity boundary condition has been prescribed. The number of (traction) unknowns NfemTr is then the number of T in the bc_fix array. In our specific case, we have NfemTr=. This means that we need for each targeted dof to be able to find its identity/number between 0 and NfemTr-1. We therefore create the array bc_nb which is filled as follows:



This translates as follows in the code:

```
NfemTr=np.sum(bc_fix)
bc_nb=np.zeros(NfemV,dtype=np.int32)
counter=0
for i in range(0,NfemV):
    if (bc_fix[i]):
        bc_nb[i]=counter
        counter+=1
```

The algorithm is then as follows

- A Prepare two arrays to store the matrix M_{cbf} and its right hand side rhs_{cbf}
- B Loop over all elements
- C For each element touching a boundary, compute the residual vector $R_{el} = -f_{el} + \mathbb{K}_{el}\mathcal{V}_{el} + \mathbb{G}_{el}\mathcal{P}_{el}$
- D Loop over the four edges of the element using the connectivity array
- E For each edge loop over the number of degrees of freedom (2 in 2D)
- F For each edge assess whether the dofs on both ends are target dofs.
- G If so, compute the mass matrix M_{edge} for this edge
- H Extract the 2 values off the element residual vector and assemble these in rhs_{cbf}
- I Assemble M_{edge} into NfemTrxNfemTr matrix using bc_nb

```
M_cbf = np.zeros((NfemTr,NfemTr),np.float64) # A
rhs_cbf = np.zeros(NfemTr,np.float64)

for iel in range(0,nel): # B
    ... compute elemental residual ... # C

    #boundary 0-1 # D
    for i in range(0,ndofV): # E
        idof0=2*icon[0,iel]+i
        idof1=2*icon[1,iel]+i
        if (bc_fix[idof0] and bc_fix[idof1]): # F
            idofTr0=bc_nb[idof0]
            idofTr1=bc_nb[idof1]
            rhs_cbf[idofTr0]+=res_el[0+i] # H
            rhs_cbf[idofTr1]+=res_el[2+i]
            M_cbf[idofTr0,idofTr0]+=M_edge[0,0] # I
            M_cbf[idofTr0,idofTr1]+=M_edge[0,1]
            M_cbf[idofTr1,idofTr0]+=M_edge[1,0]
            M_cbf[idofTr1,idofTr1]+=M_edge[1,1] # I

    #boundary 1-2 # [D]
```

...	
#boundary 2-3	$\#[D]$
...	
#boundary 3-0	$\#[D]$
...	

8.17 The value of the timestep

The chosen time step δt used for time integration is chosen to comply with the Courant-Friedrichs-Lowy condition [30].

$$\delta t = C \min \left(\frac{h}{\max |\mathbf{v}|}, \frac{h^2}{\kappa} \right) \quad (525)$$

where h is a measure of the element size, $\kappa = k/\rho C_p$ is the thermal diffusivity and C is the so-called CFL number chosen in $[0, 1]$.

In essence the CFL condition arises when solving hyperbolic PDEs . It limits the time step in many explicit time-marching computer simulations so that the simulation does not produce incorrect results.

This condition is not needed when solving the Stokes equation but it is mandatory when solving the heat transport equation or any kind of advection-diffusion equation. Note that any increase of grid resolution (i.e. h becomes smaller) yields an automatic decrease of the time step value.

8.18 Mappings

The name isoparametric derives from the fact that the same ('iso') functions are used as basis functions and for the mapping to the reference element.

More generally, if n_e denotes the number of nodes of an element and n_g denotes the number of nodes describing the geometry of the element, then the element is termed subparametric when $n_g < n_e$ and superparametric when $n_g > n_e$.

8.18.1 Linear mapping on a triangle

```

2
|\      s
| \    |_x
|  \
3==1

```

Let us assume that the coordinates of the vertices are (x_1, y_1) , (x_2, y_2) , and (x_3, y_3) . The coordinates inside the reference element are (r, s) . We then simply have the following relationship, i.e. any point of the reference element can be mapped to the physical triangle as follows:

$$x = rx_1 + sx_2 + (1 - r - s)x_3 \quad (526)$$

$$y = ry_1 + sy_2 + (1 - r - s)y_3 \quad (527)$$

There is also an inverse map, which is easily computed:

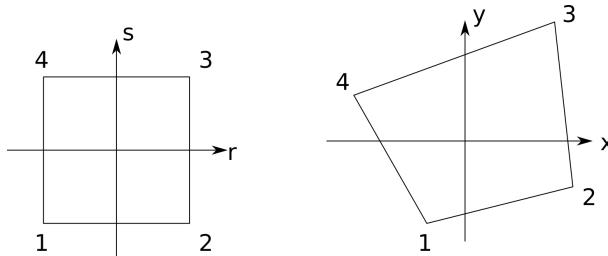
$$r = \frac{(y_2 - y_3)(x - x_3) - (x_2 - x_3)(y - y_3)}{(x_1 - x_3)(y_2 - y_3) - (y_1 - y_3)(x_2 - x_3)} \quad (528)$$

$$s = \frac{-(y_1 - y_3)(x - x_3) + (x_1 - x_3)(y - y_3)}{(x_1 - x_3)(y_2 - y_3) - (y_1 - y_3)(x_2 - x_3)} \quad (529)$$

Remark. The denominator will not vanish, because it is a multiple of the area of the triangle.

8.18.2 Bilinear mapping on a linear quadrilateral

The is in the (r, s) space. It is a square of size 2×2 centered around the origin. We wish to map it to the quadrilateral in the (x, y) space:



The coordinates of the vertices are (x_1, y_1) , (x_2, y_2) , (x_3, y_3) and (x_4, y_4) . We then simply have the following relationship, i.e. any point of the reference element can be mapped to the physical quadrilateral as follows:

$$x = N_1(r, s)x_1 + N_2(r, s)x_2 + N_3(r, s)x_3 + N_4(r, s)x_4 \quad (530)$$

$$y = N_1(r, s)y_1 + N_2(r, s)y_2 + N_3(r, s)y_3 + N_4(r, s)y_4 \quad (531)$$

where the shape functions $N_i(r, s)$ are defined in section 4.4.

In the following example the program randomly generates 10000 points inside the reference element and computes their mapping into the (x, y) space.

```

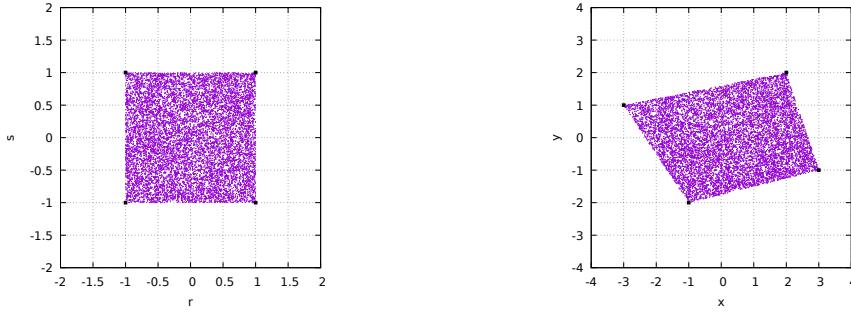
x1=-1 ; y1=-2
x2=3 ; y2=-1
x3=2 ; y3=2
x4=-3 ; y4=1

npts=10000
r=np.zeros(npts,dtype=np.float64)
s=np.zeros(npts,dtype=np.float64)
x=np.zeros(npts,dtype=np.float64)
y=np.zeros(npts,dtype=np.float64)

for i in range(0,npts):
    # compute random r,s coordinates
    r[i]=random.uniform(-1.,+1)
    s[i]=random.uniform(-1.,+1)
    # compute basis function values at r,s
    N1=0.25*(1-r[i])*(1-s[i])
    N2=0.25*(1+r[i])*(1-s[i])
    N3=0.25*(1+r[i])*(1+s[i])
    N4=0.25*(1-r[i])*(1+s[i])
    # compute x,y coordinates
    x[i]=N1*x1+N2*x2+N3*x3+N4*x4
    y[i]=N1*y1+N2*y2+N3*y3+N4*y4

np.savetxt('rs.ascii',np.array([r,s]).T)
np.savetxt('xy.ascii',np.array([x,y]).T)

```



There is also an inverse map, which is not so easily computed (see section 8.21). However, if the quadrilateral in the (x, y) space is a rectangle of size (h_x, h_y) , the inverse mapping is trivial:

$$r = \frac{x - x_1}{x_2 - x_1} \quad (532)$$

$$s = \frac{y - y_1}{y_4 - y_1} \quad (533)$$

Also in this case the shape functions can easily be written as functions of (x, y) :

$$\begin{aligned} N_1(x, y) &= \left(\frac{x_3 - x}{h_x} \right) \left(\frac{y_3 - y}{h_y} \right) \\ N_2(x, y) &= \left(\frac{x - x_1}{h_x} \right) \left(\frac{y_3 - y}{h_y} \right) \\ N_3(x, y) &= \left(\frac{x - x_1}{h_x} \right) \left(\frac{y - y_1}{h_y} \right) \\ N_4(x, y) &= \left(\frac{x_3 - x}{h_x} \right) \left(\frac{y - y_1}{h_y} \right) \end{aligned}$$

On the one hand, any variable defined on the element can be approximated using the shape functions:

$$f_h(r, s) = \sum_i N_i(r, s) f_i. \quad (534)$$

If we treat the coordinate variables x and y themselves as functions, then the shape functions can be used to construct the mapping:

$$x(r, s) = \sum_i N_i(r, s)x_i \quad y(r, s) = \sum_i N_i(r, s)y_i, \quad (535)$$

leading to write

$$\frac{\partial x}{\partial r} = \sum_i \frac{\partial N_i}{\partial r} x_i \quad (536)$$

$$\frac{\partial x}{\partial s} = \sum_i \frac{\partial N_i}{\partial s} x_i \quad (537)$$

$$\frac{\partial y}{\partial r} = \sum_i \frac{\partial N_i}{\partial r} y_i \quad (538)$$

$$\frac{\partial y}{\partial s} = \sum_i \frac{\partial N_i}{\partial s} y_i \quad (539)$$

On the other hand we also have

$$\frac{\partial f}{\partial r} = \frac{\partial f}{\partial x} \frac{\partial x}{\partial r} + \frac{\partial f}{\partial y} \frac{\partial y}{\partial r} \quad (540)$$

$$\frac{\partial f}{\partial s} = \frac{\partial f}{\partial x} \frac{\partial x}{\partial s} + \frac{\partial f}{\partial y} \frac{\partial y}{\partial s} \quad (541)$$

or in matrix form:

$$\begin{pmatrix} \frac{\partial f}{\partial r} \\ \frac{\partial f}{\partial s} \end{pmatrix} = \underbrace{\begin{pmatrix} \frac{\partial x}{\partial r} & \frac{\partial y}{\partial r} \\ \frac{\partial x}{\partial s} & \frac{\partial y}{\partial s} \end{pmatrix}}_{J} \cdot \begin{pmatrix} \frac{\partial f}{\partial x} \\ \frac{\partial f}{\partial y} \end{pmatrix}$$

where J is called the Jacobian of the transformation. By inverting the Jacobian matrix, the desired derivatives with respect to x and y can be obtained:

We have:

$$\begin{pmatrix} \frac{\partial f}{\partial x} \\ \frac{\partial f}{\partial y} \end{pmatrix} = J^{-1} \cdot \begin{pmatrix} \frac{\partial f}{\partial r} \\ \frac{\partial f}{\partial s} \end{pmatrix}$$

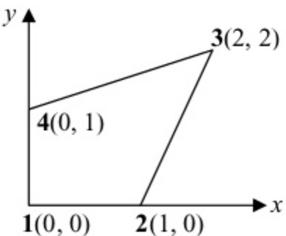
The inverse of the Jacobian matrix can be simply obtained in 2D (Kramer's rule for 2×2 matrices):

$$J^{-1} = \frac{1}{|J|} \begin{pmatrix} \frac{\partial y}{\partial s} & -\frac{\partial y}{\partial r} \\ -\frac{\partial x}{\partial s} & \frac{\partial x}{\partial r} \end{pmatrix}$$

The presence of the determinant in the denominator implies that it cannot be zero anywhere, or in other words: the mapping is not valid if $|J|$ is zero anywhere over the element.

Note that Hua [932] published analytical inverse transformation for quadrilateral isoparametric elements, i.e. how to compute J^{-1} as a function of space coordinates and not just at the quadrature points.

Let us look at this by means of a simple example and let us consider the following element:



Then a Q_1 mapping yields:

$$x(r, s) = \sum_i N_i(r, s)x_i = N_2 + 2N_3 = \frac{1}{4}(3 + 3r + s + rt) \quad (542)$$

$$y(r, s) = \sum_i N_i(r, s)y_i = 2N_3 + N_4 = \frac{1}{4}(3 + r + 3s + rt) \quad (543)$$

The Jacobian matrix is then

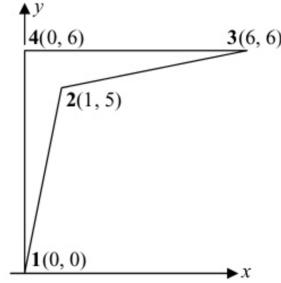
$$\mathbf{J} = \begin{pmatrix} \frac{\partial x}{\partial r} & \frac{\partial x}{\partial s} \\ \frac{\partial y}{\partial r} & \frac{\partial y}{\partial s} \end{pmatrix} = \frac{1}{4} \begin{pmatrix} 3+s & 1+s \\ 1+r & 3+r \end{pmatrix}$$

and its determinant is

$$|\mathbf{J}| = \frac{1}{4}[(3+s)(3+r) - (1+s)(1+r)] = \frac{1}{2} + \frac{1}{8}r + \frac{1}{8}s \quad (544)$$

It is clear that $|\mathbf{J}| > 0$ for $-1 \leq r \leq +1$ and $-1 \leq s \leq +1$.

Let us now consider another example, the following element:



It follows that

$$x(r, s) = \sum_i N_i(r, s)x_i = \frac{1}{4}(1+r)(7+5s) \quad (545)$$

$$y(r, s) = \sum_i N_i(r, s)y_i = \frac{1}{4}(17+5r+7s-5rs) \quad (546)$$

and the determinant:

$$|\mathbf{J}| = \frac{3}{2} - \frac{15r}{4} + \frac{15s}{4}$$

is zero for $r - s = 2/5$. This mapping is invalid!

Remark. Problems also arise when the Jacobian matrix is nearly singular due to round-off errors. To avoid problems linked to badly shaped elements, it is recommended that the inside angles of an element are larger than 15° and less than 165° .

From Eq. 535, we can also write:

$$dx = \frac{\partial x}{\partial r}dr + \frac{\partial x}{\partial s}ds \quad (547)$$

$$dy = \frac{\partial y}{\partial r}dr + \frac{\partial y}{\partial s}ds \quad (548)$$

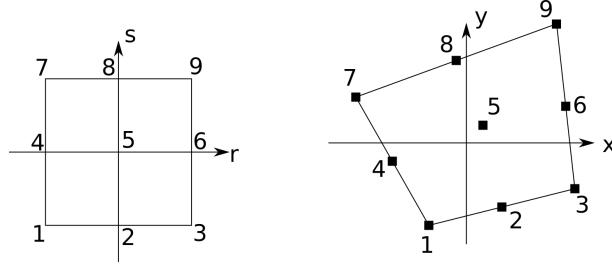
, or,

$$\begin{pmatrix} dx \\ dy \end{pmatrix} = \mathbf{J} \cdot \begin{pmatrix} dr \\ ds \end{pmatrix} \quad (549)$$

This means that

$$\int \int \dots dx dy = \int \int \dots |\mathbf{J}| dr ds \quad (550)$$

8.18.3 biquadratic mapping of a straight-line face Q_2 element



The reference element now contains 9 nodes: 1,3,7,9 are the corners, nodes 2,4,6,8 are the mid-face points and node 5 is in the middle. The mapping from the (r, s) space to the (x, y) space is then as follows:

$$\begin{aligned} \begin{pmatrix} x(r, s) \\ y(r, s) \end{pmatrix} &= N_1(r, s) \begin{pmatrix} x_1 \\ y_1 \end{pmatrix} + N_2(r, s) \begin{pmatrix} x_2 \\ y_2 \end{pmatrix} + N_3(r, s) \begin{pmatrix} x_3 \\ y_3 \end{pmatrix} + N_4(r, s) \begin{pmatrix} x_4 \\ y_4 \end{pmatrix} \\ &+ N_5(r, s) \begin{pmatrix} x_5 \\ y_5 \end{pmatrix} + N_6(r, s) \begin{pmatrix} x_6 \\ y_6 \end{pmatrix} + N_7(r, s) \begin{pmatrix} x_7 \\ y_7 \end{pmatrix} + N_8(r, s) \begin{pmatrix} x_8 \\ y_8 \end{pmatrix} \\ &+ N_9(r, s) \begin{pmatrix} x_9 \\ y_9 \end{pmatrix} \end{aligned}$$

where

$$\begin{aligned} N_1(r, t) &= 0.5r(r-1)0.5t(t-1) \\ N_2(r, t) &= (1-r^2)0.5t(t-1) \\ N_3(r, t) &= 0.5r(r+1)0.5t(t-1) \\ N_4(r, t) &= 0.5r(r-1)(1-t^2) \\ N_5(r, t) &= (1-r^2)(1-t^2) \\ N_6(r, t) &= 0.5r(r+1)(1-t^2) \\ N_7(r, t) &= 0.5r(r-1)0.5t(t+1) \\ N_8(r, t) &= (1-r^2)0.5t(t+1) \\ N_9(r, t) &= 0.5r(r+1)0.5t(t+1) \end{aligned}$$

```

x1=-1 ; y1=-2
x3=3 ; y3=-1
x9=2 ; y9=2
x7=-3 ; y7=1
x2=0.5*(x1+x3) ; y2=0.5*(y1+y3)
x4=0.5*(x1+x7) ; y4=0.5*(y1+y7)
x6=0.5*(x3+x9) ; y6=0.5*(y3+y9)
x8=0.5*(x7+x9) ; y8=0.5*(y7+y9)
x5=0.25*(x1+x3+x7+x9) ; y5=0.25*(y1+y3+y7+y9)

npts=10000
r=np.zeros( npts , dtype=np.float64 )
s=np.zeros( npts , dtype=np.float64 )
xQ1=np.zeros( npts , dtype=np.float64 )
yQ1=np.zeros( npts , dtype=np.float64 )
xQ2=np.zeros( npts , dtype=np.float64 )
yQ2=np.zeros( npts , dtype=np.float64 )

for i in range(0,npts):
    # compute random r, s coordinates
    r[ i ]=random.uniform(-1.,+1)
    s[ i ]=random.uniform(-1.,+1)
    # compute Q2 basis function values at r, s
    N1= 0.5*r[ i ]*( r[ i ]-1. ) * 0.5*s[ i ]*( s[ i ]-1. )

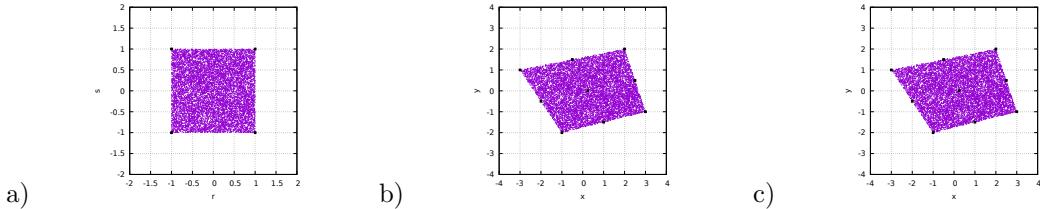
```

```

N2=      (1.-r[i]**2) * 0.5*s[i]*(s[i]-1.)
N3= 0.5*r[i]*(r[i]+1.) * 0.5*s[i]*(s[i]-1.)
N4= 0.5*r[i]*(r[i]-1.) *          (1.-s[i]**2)
N5=      (1.-r[i]**2) *          (1.-s[i]**2)
N6= 0.5*r[i]*(r[i]+1.) *          (1.-s[i]**2)
N7= 0.5*r[i]*(r[i]-1.) * 0.5*s[i]*(s[i]+1.)
N8=      (1.-r[i]**2) * 0.5*s[i]*(s[i]+1.)
N9= 0.5*r[i]*(r[i]+1.) * 0.5*s[i]*(s[i]+1.)
# compute x, y coordinates
xQ2[i]=N1*x1+N2*x2+N3*x3+N4*x4+N5*x5+N6*x6+N7*x7+N8*x8+N9*x9
yQ2[i]=N1*y1+N2*y2+N3*y3+N4*y4+N5*y5+N6*y6+N7*y7+N8*y8+N9*y9
# compute Q1 basis function values at r, s
N1=0.25*(1-r[i])*(1-s[i])
N2=0.25*(1+r[i])*(1-s[i])
N3=0.25*(1+r[i])*(1+s[i])
N4=0.25*(1-r[i])*(1+s[i])
# compute x, y coordinates
xQ1[i]=N1*x1+N2*x3+N3*x9+N4*x7
yQ1[i]=N1*y1+N2*y3+N3*y9+N4*y7

np.savetxt('rs.ascii',np.array([r,s]).T)
np.savetxt('xyQ1.ascii',np.array([xQ1,yQ1]).T)
np.savetxt('xyQ2.ascii',np.array([xQ2,yQ2]).T)

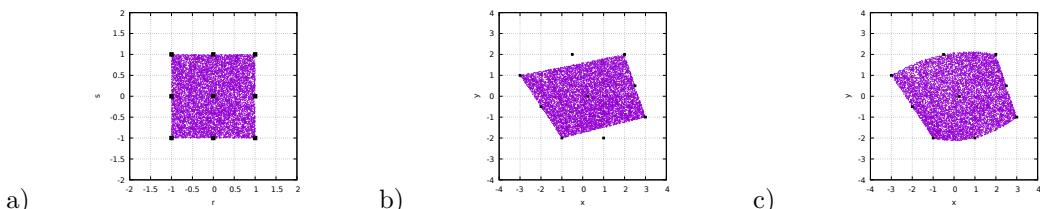
```



a) 10,000 random points in the reference element; b,c) image of these points by means of a bilinear and biquadratic mapping respectively. When the sides of the element are straight we see that a Q_1 mapping is sufficient.

8.18.4 biquadratic mapping of a not-so straight-line face Q_2 element

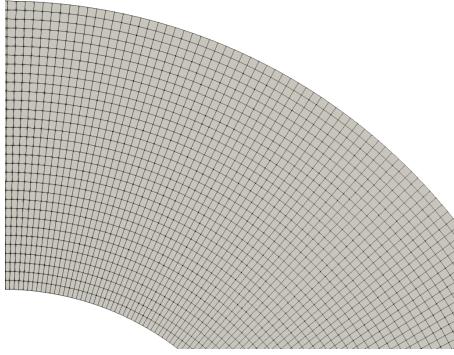
We now carry out the same exercise as before but nodes 2 and 8 are no more in the middle of nodes 1-3 and 7-9 respectively.



a) 10,000 random points in the reference element; b,c) image of these points by means of a bilinear and biquadratic mapping respectively. In this case we see that the Q_2 mapping manages to capture the 'real' shape of the element.

8.18.5 bilinear, biquadratic and bicubic mapping in an annulus

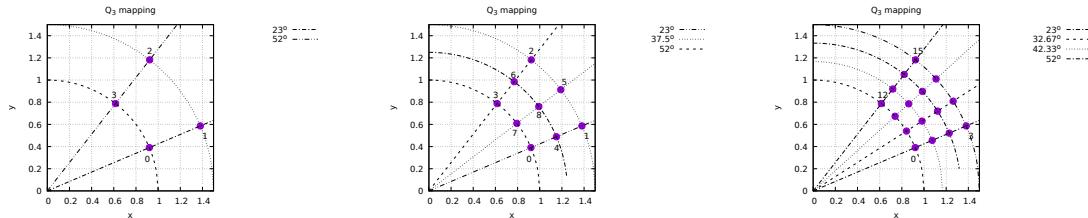
In the light of what precedes, we can now ask ourselves how this translates to a real geodynamic cas. Let us then consider the case of an annular domain, a cross section of a hollow sphere. When using quadrilateral elements, the mesh will look similar to this:



We here focus on Q_1 , Q_2 and Q_3 mappings. We single out an element, and arbitrarily define it as follows in polar coordinates:

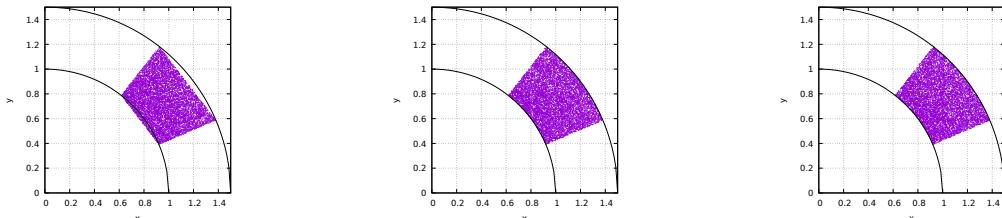
```
theta1=23./180.*np.pi
theta2=52./180.*np.pi
R1=1.
R2=1.5
```

The Q_1 mapping requires four points, the Q_2 nine points and the Q_3 sixteen points. These are placed equidistantly in the r, θ coordinate system, as shown hereunder:



Left to right: position of the nodes for the Q_1 , Q_2 and Q_3 mappings.

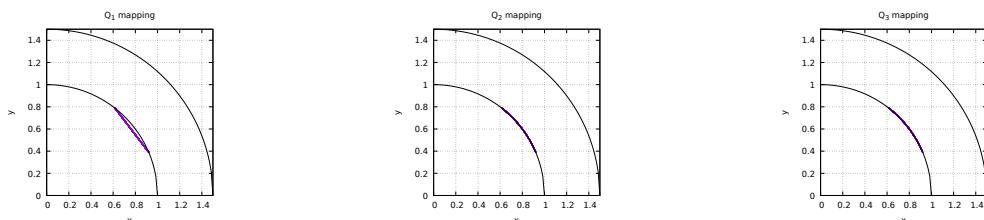
As before, we randomly shoot 10,000 points inside the reference element and map these out in the x, y space. Resulting swarms of points are shown in the following figures:



Left to right: position of the mapped points for the Q_1 , Q_2 and Q_3 mappings.

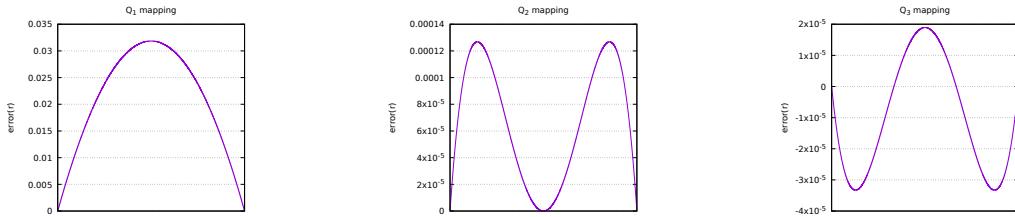
The image of a square with a Q_1 mapping is obviously a quadrilateral so that it looks like quite a few points land outside of the domain $R_1 \leq r \leq R_2$. Note that points are well within $23^\circ \leq \theta \leq 52^\circ$, which can simply be explained by the fact that the faces of the element are straight lines.

However, it looks like the biquadratic and bicubic mappings are doing a much better job at mapping the region of space $R_1 \leq r \leq R_2$. In order to characterise this better, we now place 10,000 points on the bottom face of the reference element (i.e. $s = -1$) and once again compute their coordinates in the the x, y space:



Left to right: position of the mapped points for the Q_1 , Q_2 and Q_3 mappings.

For each point i we now compute ist distance r_i to the origin, which, if the mapping was perfect, whould be exactly equal to $R_1 = 1$. On the following plots are shown the error $r_i - 1$ for all points, from $r = -1$ to $r = +1$.



Left to right: radius error of the mapped points for the Q_1 , Q_2 and Q_3 mappings.

We see that the amplitude of the error decreases with the order of the mapping used, which is why for instance ASPECT uses a Q_4 mapping by default. Actually, in this particular case, the equation which describes the cirle is not a polynomial so that no high-order mapping will ever be able to *exactly* represent the curved boundary of the element!

Another interesting point to keep in mind is that the location of the quadrature points in the x, y space is also determined by the mapping used, which can have consequences on the accuracy of the integration and it will be reflected (for instance) on the error convergence rate.

Finally, the coordinates of the nodes of the element in the x, y are uniquely determined when they are on the convex hull of the element (for instance nodes 0-7 for Q_2) but we need to choose the position of the last nodes which are inside the element. Unfortunately, this choice is not neutral.

re ask Wolfgang about this - correlate with deal.II

8.19 Exporting data to vtk format

This format seems to be the universally accepted format for 2D and 3D visualisation in Computational Geodynamics. Such files can be opened with free softwares such as Paraview⁴², MayaVi⁴³ or Visit⁴⁴.

Unfortunately it is my experience that no simple tutorial exists about how to build such files. There is an official document which describes the vtk format⁴⁵ but it delivers the information in a convoluted way. I therefore describe hereafter how **fieldstone** builds the vtk files.

I hereunder show vtk file corresponding to the 3x2 grid presented earlier ???. In this particular example there are:

- 12 nodes and 6 elements
- 1 elemental field: the pressure p)
- 2 nodal fields: 1 scalar (the smoothed pressure q), 1 vector (the velocity field $u, v, 0$)

Note that vtk files are inherently 3D so that even in the case of a 2D simulation the z -coordinate of the points and for instance their z -velocity component must be provided. The file, usually called *solution.vtu* starts with a header:

```
<VTKFile type='UnstructuredGrid' version='0.1' byte_order='BigEndian'>
<UnstructuredGrid>
<Piece NumberOfPoints='12' NumberOfCells='6'>
```

We then proceed to write the node coordinates as follows:

```
<Points>
<DataArray type='Float32' NumberOfComponents='3' Format='ascii'>
0.000000e+00 0.000000e+00 0.000000e+00
3.333333e-01 0.000000e+00 0.000000e+00
6.666667e-01 0.000000e+00 0.000000e+00
1.000000e+00 0.000000e+00 0.000000e+00
0.000000e+00 5.000000e-01 0.000000e+00
3.333333e-01 5.000000e-01 0.000000e+00
6.666667e-01 5.000000e-01 0.000000e+00
1.000000e+00 5.000000e-01 0.000000e+00
0.000000e+00 1.000000e+00 0.000000e+00
3.333333e-01 1.000000e+00 0.000000e+00
6.666667e-01 1.000000e+00 0.000000e+00
1.000000e+00 1.000000e+00 0.000000e+00
</DataArray>
</Points>
```

These are followed by the elemental field(s):

```
<CellData Scalars='scalars'>
<DataArray type='Float32' Name='p' Format='ascii'>
-1.333333e+00
-3.104414e-10
1.333333e+00
-1.333333e+00
8.278417e-17
1.333333e+00
</DataArray>
</CellData>
```

Nodal quantities are written next:

```
<PointData Scalars='scalars'>
<DataArray type='Float32' NumberOfComponents='3' Name='velocity' Format='ascii'>
0.000000e+00 0.000000e+00 0.000000e+00
0.000000e+00 0.000000e+00 0.000000e+00
0.000000e+00 0.000000e+00 0.000000e+00
0.000000e+00 0.000000e+00 0.000000e+00
```

⁴²<https://www.paraview.org/>

⁴³<https://docs.enthought.com/mayavi/mayavi/>

⁴⁴<https://wci.llnl.gov/simulation/computer-codes/visit/>

⁴⁵<https://www.vtk.org/wp-content/uploads/2015/04/file-formats.pdf>

```

0.000000e+00 0.000000e+00 0.000000e+00
8.888885e-08 -8.278405e-24 0.000000e+00
8.888885e-08 1.655682e-23 0.000000e+00
0.000000e+00 0.000000e+00 0.000000e+00
1.000000e+00 0.000000e+00 0.000000e+00
</DataArray>
<DataArray type='Float32' NumberOfComponents='1' Name='q' Format='ascii'>
-1.333333e+00
-6.666664e-01
6.666664e-01
1.333333e+00
-1.333333e+00
-6.666664e-01
6.666664e-01
1.333333e+00
-1.333333e+00
-6.666664e-01
6.666664e-01
1.333333e+00
</DataArray>
</PointData>

```

To these informations we must append 3 more datasets. The first one is the connectivity, the second one is the offsets and the third one is the type. The first one is trivial since said connectivity is needed for the Finite Elements. The second must be understood as follows: when reading the connectivity information in a linear manner the offset values indicate the beginning of each element (omitting the zero value). The third simply is the type of element as given in the vtk format document (9 corresponds to a generic quadrilateral with an internal numbering consistent with ours).

```

<Cells>
<DataArray type='Int32' Name='connectivity' Format='ascii'>
0 1 5 4
1 2 6 5
2 3 7 6
4 5 9 8
5 6 10 9
6 7 11 10
</DataArray>
<DataArray type='Int32' Name='offsets' Format='ascii'>
4
8
12
16
20
24
</DataArray>
<DataArray type='Int32' Name='types' Format='ascii'>
9
9
9
9
9
9
</DataArray>
</Cells>

```

The file is then closed with

```

</Piece>
</UnstructuredGrid>
</VTKFile>

```

The *solution.vtu* file can then be opened with ParaView, MayaVi or Visit and the reader is advised to find tutorials online on how to install and use these softwares.

8.20 Runge-Kutta methods

These methods were developed around 1900 by the German mathematicians Carl Runge and Martin Kutta. The RK methods are methods for the numerical integration of ODEs⁴⁶. These methods are well documented in any numerical analysis textbook and the reader is referred to [722, 967]. Any Runge-Kutta method is uniquely identified by its Butcher tableau (REF?) which contains all necessary coefficients to build the algorithm.

The simplest RungeKutta method is the (forward) Euler method. Its tableau is:

0				
	1			

The standard second-order RK method method (also called midpoint method) is:

0				
1/2	1/2			
		0	1	

Another second-order RK method, called Heun's method⁴⁷ is follows:

0				
1	1			
		1/2	1/2	

A third-order RK method is as follows:

0				
1/2	1/2			
1	-1	2		
			1/6	4/6
				1/6

The RK4 method falls in this framework. Its tableau is:

0				
1/2	1/2			
1/2	0	1/2		
1	0	0	1	
	1/6	1/6	1/3	1/6

A slight variation of the standard RK4 method is also due to Kutta in 1901 and is called the 3/8-rule. Almost all of the error coefficients are smaller than in the standard method but it requires slightly more FLOPs per time step. Its Butcher tableau is

0				
1/3	1/3			
2/3	-1/3	1		
1	1	-1	1	
	1/8	3/8	3/8	1/8

The following method is called the Runge-Kutta-Fehlberg method and is commonly abbreviated RKF45⁴⁸. Its Butcher tableau is as follows:

⁴⁶https://en.wikipedia.org/wiki/Runge-Kutta_methods

⁴⁷https://en.wikipedia.org/wiki/Heun%27s_method

⁴⁸https://en.wikipedia.org/wiki/Runge-Kutta-Fehlberg_method

0						
1/4	1/4					
3/8	3/32	9/32				
12/13	1932/2197	-7200/2197	7296/2197			
1	439/216	-8	3680/513	-845/4104		
1/2	-8/27	2	-3544/2565	1859/4104	-11/40	
	16/135	0	6656/12825	28561/56430	-9/50	2/55
	25/216	0	1408/2565	2197/4104	-1/5	0

The first row of coefficients at the bottom of the table gives the fifth-order accurate method, and the second row gives the fourth-order accurate method.

 Literature[612, 849, 534, 535, 1504]

8.20.1 Using RK methods to advect particles/markers

In the context of geodynamical modelling, one is usually confronted to the following problem: now that I have a velocity field on my FE mesh, how can I use it to advect the Lagrangian markers?

Runge-Kutta methods are used to this effect but only their spatial component is used: the velocity solution is not recomputed at the intermediate fractional timesteps, i.e. only the coefficients of the right hand side of the tableaus is used.

The RK1 method is simple. Carry out a loop over markers and

1. interpolate velocity \vec{v}_m onto each marker m
2. compute new position as follows: $\vec{r}_m(t + \delta t) = \vec{r}_m(t) + \vec{v}_m \delta t$

The RK2 method is also simple but requires a bit more work. Carry out a loop over markers and

1. interpolate velocity \vec{v}_m onto each marker m at position \vec{r}_m
2. compute new intermediate position as follows: $\vec{r}_m^{(1)}(t + \delta t) = \vec{r}_m(t) + \vec{v}_m \delta t / 2$
3. compute velocity $\vec{v}_m^{(1)}$ at position $\vec{r}_m^{(1)}$
4. compute new position: $\vec{r}_m(t + \delta t) = \vec{r}_m(t) + \vec{v}_m^{(1)} \delta t$

Note that the intermediate positions could be in a different element of the mesh so extra care must be taken when computing intermediate velocities.

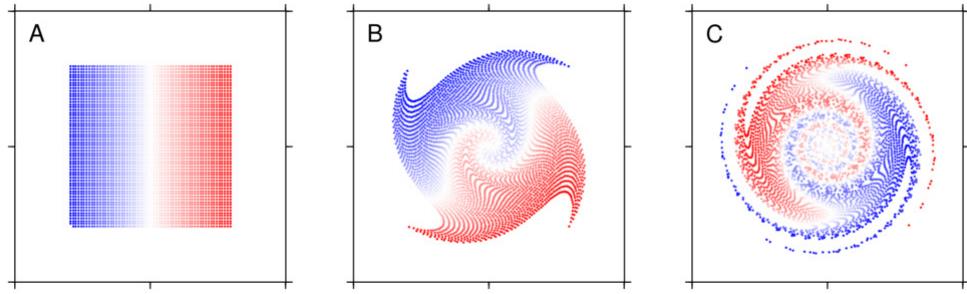
The RK3 method introduces two intermediate steps. Carry out a loop over markers and

1. interpolate velocity \vec{v}_m onto each marker m at position \vec{r}_m
2. compute new intermediate position as follows: $\vec{r}_m^{(1)}(t + \delta t) = \vec{r}_m(t) + \vec{v}_m \delta t / 2$
3. compute velocity $\vec{v}_m^{(1)}$ at position $\vec{r}_m^{(1)}$
4. compute new intermediate position as follows: $\vec{r}_m^{(2)}(t + \delta t) = \vec{r}_m(t) + (2\vec{v}_m^{(1)} - \vec{v}_m) \delta t / 2$
5. compute velocity $\vec{v}_m^{(2)}$ at position $\vec{r}_m^{(2)}$
6. compute new position: $\vec{r}_m(t + \delta t) = \vec{r}_m(t) + (\vec{v}_m + 4\vec{v}_m^{(1)} + \vec{v}_m^{(2)}) \delta t / 6$

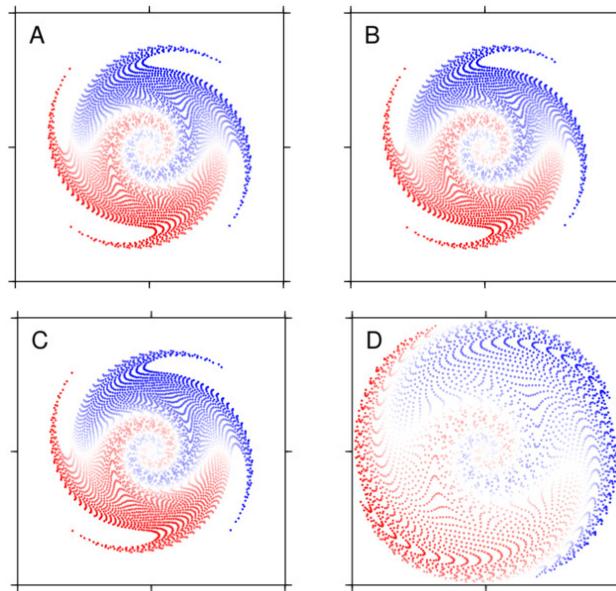
The following example is borrowed from [1248], itself borrowed from Fullsack [678, Section 5.4]. It is a whirl flow [1442], a flow with rotational symmetry in which concentric layers of material rotate around a centre with an angular velocity:

$$\omega(r) = \omega_0 \frac{r}{r_0} \exp\left(-\frac{r}{r_0}\right)$$

The box is $[-0.5, 0.5] \times [-0.5, 0.5]$, $r_0 = 0.25$, $\omega_0 = 0.3$ and $\delta t = 1$. 60×60 particles are regularly positioned inside the $[-0.3, 0.3] \times [-0.3, 0.3]$ square. Maierova [1248] has carried out this experiment for the above Runge-Kutta methods.



Model domain with particles colored at three different time-steps: (A) $t = 0$ (initial position of particles), (B) $t = 50$, and (C) $t = 200$. The advection is computed using the fourth-order Runge-Kutta scheme. Taken from [1248]



The same plot as above, but for different advection schemes at $t = 100$. Advection was computed using (A) the fourth-order Runge-Kutta scheme, (B) the mid-point method, (C) Heuns method and (D) the explicit Euler method. Taken from [1248]

(BSc Thesis)

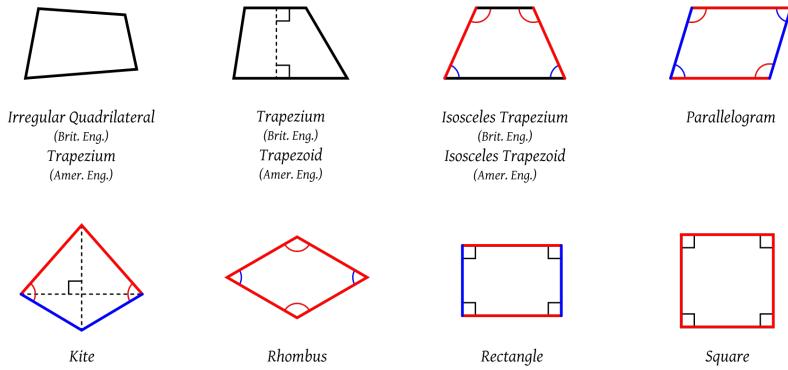
8.21 Am I in or not? - finding reduced coordinates

It is quite common that at some point one must answer the question: "Given a mesh and its connectivity on the one hand, and the coordinates of a point on the other, how do I accurately and quickly determine in which element the point resides?"

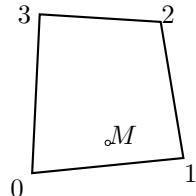
One typical occurrence of such a problem is linked to the use of the Particle-In-Cell technique: particles are advected and move through the mesh, and need to be localised at every time step. This question could arise in the context of a benchmark where certain quantities need to be measured at specific locations inside the domain.

8.21.1 Two-dimensional space

We shall first focus on quadrilaterals. There are many kinds of quadrilaterals as shown hereunder:



I wish to arrive at a single algorithm which is applicable to all quadrilaterals and therefore choose an irregular quadrilateral. For simplicity, let us consider a Q_1 element, with a single node at each corner.



Several rather simple options exist:

- we could subdivide the quadrilateral into two triangles and check whether point M is inside any of them (as it turns out, this problem is rather straightforward for triangles. Simply google it.)
- We could check that point M is always on the left side of segments $0 \rightarrow 1$, $1 \rightarrow 2$, $2 \rightarrow 3$, $3 \rightarrow 0$.
- ...

Any of these approaches will work although some might be faster than others. In three-dimensions all will however become cumbersome to implement and might not even work at all. Fortunately, there is an elegant way to answer the question, as detailed in the following subsection.

8.21.2 Three-dimensional space

If point M is inside the quadrilateral, there exist a set of reduced coordinates $r, s, t \in [-1 : 1]^3$ such that

$$\sum_{i=1}^4 N_i(r_M, s, t) x_i = x_M \quad \sum_{i=1}^4 N_i(r_M, s, t) y_i = y_M \quad \sum_{i=1}^4 N_i(r_M, s, t) z_i = z_M$$

This can be cast as a system of three equations and three unknowns. Unfortunately, each shape function N_i contains a term rst (as well as rs , rt , and st) so that it is not a linear system and standard techniques are not applicable. We must then use an iterative technique: the algorithm starts with a guess for values r, s, t and improves on their value iteration after iteration.

The classical way of solving nonlinear systems of equations is Newton's method. We can rewrite the equations above as $\mathbf{F}(r, s, t) = 0$:

$$\begin{aligned} \sum_{i=1}^8 N_i(r, s, t)x_i - x_M &= 0 \\ \sum_{i=1}^8 N_i(r, s, t)y_i - y_M &= 0 \\ \sum_{i=1}^8 N_i(r, s, t)z_i - z_M &= 0 \end{aligned} \quad (551)$$

or,

$$\begin{aligned} F_r(r, s, t) &= 0 \\ F_s(r, s, t) &= 0 \\ F_t(r, s, t) &= 0 \end{aligned}$$

so that we now have to find the zeroes of continuously differentiable functions $\mathbf{F} : \mathbb{R} \rightarrow \mathbb{R}$. The recursion is simply:

$$\begin{pmatrix} r_{k+1} \\ s_{k+1} \\ t_{k+1} \end{pmatrix} = \begin{pmatrix} r_k \\ s_k \\ t_k \end{pmatrix} - J_F(r_k, s_k, t_k)^{-1} \begin{pmatrix} F_r(r_k, s_k, t_k) \\ F_s(r_k, s_k, t_k) \\ F_t(r_k, s_k, t_k) \end{pmatrix}$$

where J the Jacobian matrix:

$$\begin{aligned} J_F(r_k, s_k, t_k) &= \begin{pmatrix} \frac{\partial F_r}{\partial r}(r_k, s_k, t_k) & \frac{\partial F_r}{\partial s}(r_k, s_k, t_k) & \frac{\partial F_r}{\partial t}(r_k, s_k, t_k) \\ \frac{\partial F_s}{\partial r}(r_k, s_k, t_k) & \frac{\partial F_s}{\partial s}(r_k, s_k, t_k) & \frac{\partial F_s}{\partial t}(r_k, s_k, t_k) \\ \frac{\partial F_t}{\partial r}(r_k, s_k, t_k) & \frac{\partial F_t}{\partial s}(r_k, s_k, t_k) & \frac{\partial F_t}{\partial t}(r_k, s_k, t_k) \end{pmatrix} \\ &= \begin{pmatrix} \sum_{i=1}^8 \frac{\partial N_i}{\partial r}(r_k, s_k, t_k)x_i & \sum_{i=1}^8 \frac{\partial N_i}{\partial s}(r_k, s_k, t_k)x_i & \sum_{i=1}^8 \frac{\partial N_i}{\partial t}(r_k, s_k, t_k)x_i \\ \sum_{i=1}^8 \frac{\partial N_i}{\partial r}(r_k, s_k, t_k)y_i & \sum_{i=1}^8 \frac{\partial N_i}{\partial s}(r_k, s_k, t_k)y_i & \sum_{i=1}^8 \frac{\partial N_i}{\partial t}(r_k, s_k, t_k)y_i \\ \sum_{i=1}^8 \frac{\partial N_i}{\partial r}(r_k, s_k, t_k)z_i & \sum_{i=1}^8 \frac{\partial N_i}{\partial s}(r_k, s_k, t_k)z_i & \sum_{i=1}^8 \frac{\partial N_i}{\partial t}(r_k, s_k, t_k)z_i \end{pmatrix} \end{aligned}$$

In practice, we solve the following system:

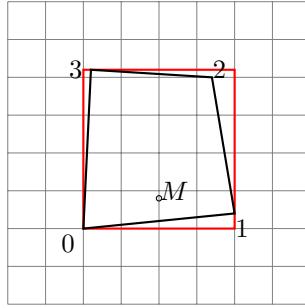
$$J_F(r_k, s_k, t_k) \left[\begin{pmatrix} r_{k+1} \\ s_{k+1} \\ t_{k+1} \end{pmatrix} - \begin{pmatrix} r_k \\ s_k \\ t_k \end{pmatrix} \right] = - \begin{pmatrix} F_r(r_k, s_k, t_k) \\ F_s(r_k, s_k, t_k) \\ F_t(r_k, s_k, t_k) \end{pmatrix}$$

Finally, the algorithm goes as follows:

- set guess values for r, s, t (typically 0)
- loop over $k=0, \dots$
- Compute $\text{rhs} = -\mathbf{F}(r_k, s_k, t_k)$
- Compute matrix $J_F(r_k, s_k, t_k)$

- solve system for (dr_k, ds_k, dt_k)
- update $r_{k+1} = r_k + dr_k, s_{k+1} = s_k + ds_k, t_{k+1} = t_k + dt_k$
- stop iterations when (dr_k, ds_k, dt_k) is small
- if $r_k, s_k, t_k \in [-1, 1]^3$ then M is inside.

This method converges quickly but involves iterations, and multiple solves of 3×3 systems which, when carried out for each marker and at each time step can prove to be expensive. A simple modification can be added to the above algorithm: iterations should be carried out *only* when the point M is inside of a cuboid of size $[\min_i x_i : \max_i x_i] \times [\min_i y_i : \max_i y_i] \times [\min_i z_i : \max_i z_i]$ where the sums run over the vertices of the element. In 2D this translates as follows: only carry out Newton iterations when M is inside the red rectangle!



Note that the algorithm above extends to high degree elements such as Q_2 and higher, even with curved sides.

write about case when element is rectangle/cuboid

8.22 Error measurements and convergence rates

What follows is written in the case of a two-dimensional model. Generalisation to 3D is trivial. What follows is mostly borrowed from [1754].

When measuring the order of accuracy of the primitive variables \vec{v} and p , it is standard to report errors in both the L_1 and the L_2 norm. For a scalar quantity Ψ , the L_1 and L_2 norms are computed as

$$\|\Psi\|_1 = \int_V |\Psi| dV \quad \|\Psi\|_2 = \sqrt{\int_V \Psi^2 dV} \quad (552)$$

For a vector quantity $\vec{k} = (k_x, k_y)$ in a two-dimensional space, the L_1 and L_2 norms are defined as:

$$\|\vec{k}\|_1 = \int_V (|k_x| + |k_y|) dV \quad \|\vec{k}\|_2 = \sqrt{\int_V (k_x^2 + k_y^2) dV} \quad (553)$$

To compute the respective norms the integrals in the above norms can be approximated by splitting them into their element-wise contributions. The element volume integral can then be easily computed by numerical integration using Gauss-Legendre quadrature.

The respective L_1 and L_2 norms for the pressure error can be evaluated via

$$e_p^h|_1 = \sum_{i=1}^{n_e} \sum_{q=1}^{n_q} |e_p^h(\vec{r}_q)| w_q |J_q| \quad e_p^h|_2 = \sqrt{\sum_{i=1}^{n_e} \sum_{q=1}^{n_q} |e_p^h(\vec{r}_q)|^2 w_q |J_q|} \quad (554)$$

where $e_p^h(\vec{r}_q) = p^h(\vec{r}_q) - p(\vec{r}_q)$ is the pressure error evaluated at the q -th quadrature associated with the i th element. n_e and n_q refer to the number of elements and the number of quadrature points per element. w_q and J_q are the quadrature weight and the Jacobian associated with point q .

The velocity error $e_{\vec{v}}^h$ is evaluated using the following two norms

$$e_{\vec{v}}^h|_1 = \sum_{i=1}^{n_e} \sum_{q=1}^{n_q} [|e_u^h(\vec{r}_q)| + |e_v^h(\vec{r}_q)|] w_q |J_q| \quad e_{\vec{v}}^h|_2 = \sqrt{\sum_{i=1}^{n_e} \sum_{q=1}^{n_q} [|e_u^h(\vec{r}_q)|^2 + |e_v^h(\vec{r}_q)|^2] w_q |J_q|} \quad (555)$$

where $e_u^h(\vec{r}_q) = u^h(\vec{r}_q) - u(\vec{r}_q)$ and $e_v^h(\vec{r}_q) = v^h(\vec{r}_q) - v(\vec{r}_q)$.

Another norm is very rarely used in the geodynamics literature but is preferred in the Finite Element literature: the H^1 norm. The mathematical basis for this norm and the nature of the $H^1(\Omega)$ Hilbert space is to be found in many FE books [533, 998, 941]. This norm is expressed as follows for a function f such that $f, |\nabla f| \in L^2(\Omega)$ ⁴⁹

$$\|f\|_{H^1} = \left(\int_{\Omega} (|f|^2 + |\nabla f|^2) d\Omega \right)^{1/2} \quad (556)$$

We then have

$$e_{\vec{v}}^h|_{H^1} = \|\vec{v}^h - \vec{v}\|_{H^1} = \sqrt{\sum_{i=1}^d \int_{\Omega} [(v_i^h - v_i)^2 + \vec{\nabla}(v_i^h - v_i) \cdot \vec{\nabla}(v_i^h - v_i)] d\Omega} \quad (557)$$

where d is the number of dimensions. Note that sometimes the following semi-norm is used [529, 181]:

$$e_{\vec{v}}^h|_{H^1} = \|\vec{v}^h - \vec{v}\|_{H^1} = \sqrt{\sum_{i=1}^d \int_{\Omega} [\vec{\nabla}(v_i^h - v_i) \cdot \vec{\nabla}(v_i^h - v_i)] d\Omega} \quad (558)$$

When computing the different error norms for e_p and $e_{\vec{v}}$ for a set of numerical experiments with varying resolution h we expect the error norms to follow the following relationships:

$$e_{\vec{v}}^h|_1 = Ch^{rvL_1} \quad e_{\vec{v}}^h|_2 = Ch^{rvL_2} \quad e_{\vec{v}}^h|_{H^1} = Ch^{rvH^1} \quad (559)$$

⁴⁹https://en.wikipedia.org/wiki/Sobolev_space

$$e_p^h|_1 = Ch^{rpL_1} \quad e_p^h|_2 = Ch^{rpL_2} \quad (560)$$

where C is a resolution-independent constant and $rpXX$ and $rvXX$ are the convergence rates for pressure and velocity in various norms, respectively. Using linear regression on the logarithm of the respective error norm and the resolution h , one can compute the convergence rates of the numerical solutions.

As mentioned in [529], when finite element solutions converge at the same rates as the interpolants we say that the method is optimal, i.e.:

$$e_v^h|_{L_2} = \mathcal{O}(h^3) \quad e_v^h|_{H^1} = \mathcal{O}(h^2) \quad e_p^h|_{L_2} = \mathcal{O}(h^2) \quad (561)$$

We note that when using discontinuous pressure space (e.g., P_0, P_{-1}), these bounds remain valid even when the viscosity is discontinuous provided that the element boundaries conform to the discontinuity.

8.22.1 About extrapolation

Section contributed by W. Bangerth and part of Thieulot & Bangerth [in prep.]

In a number of numerical benchmarks we want to estimate the error $X_h - X^*$ between a quantity X_h computed from the numerical solution \vec{u}_h, p_h and the corresponding value X computed from the exact solution \vec{u}, p . Examples of such quantities X are the root mean square velocity v_{rms} , but it could also be a mass flux across a boundary, an average horizontal velocity at the top boundary, or any other scalar quantity.

If the exact solution is known, then one can of course compute X from it. On the other hand, we would of course like to assess convergence also in cases where the exact solution is not known. In that case, one can compute an *estimate* X^* for X by way of *extrapolation*. To this end, we make the assumption that asymptotically, X_h converges to X at a fixed (but unknown) rate r , so that

$$e_h = |X_h - X| \approx Ch^r. \quad (562)$$

Here, X , C and r are all unknown constants to be determined, although we are not really interested in C . We can evaluate X_h from the numerical solution on successively refined meshes with mesh sizes h , $h/2$, and $h/4$. Then, in addition to (562) we also have

$$e_{h/2} = |X_{h/2} - X| \approx C \left(\frac{h}{2} \right)^r, \quad (563)$$

$$e_{h/4} = |X_{h/4} - X| \approx C \left(\frac{h}{4} \right)^r. \quad (564)$$

Taking ratios of equations (562)–(564), and replacing the unknown X by an *estimate* X^* , we then arrive at the following equation:

$$\frac{|X_h - X^*|}{|X_{h/2} - X^*|} = \frac{|X_{h/2} - X^*|}{|X_{h/4} - X^*|} = 2^r.$$

If one assumes that X_h converges to X uniformly either from above or below (rather than oscillate around X), then this equation allows us to solve for X^* and r :

$$X^* = \frac{X_h X_{h/2} - X_{h/2}^2}{X_h - 2X_{h/2} + X_{h/4}}, \quad r = \log_2 \frac{X_{h/2} - X^*}{X_{h/4} - X^*}.$$

In the determination of r , we could also have used X_h and $X_{h/2}$, but using $X_{h/2}$ and $X_{h/4}$ is generally more reliable because the higher order terms we have omitted in (562) are less visible on finer meshes.

8.23 The initial temperature field

8.23.1 Single layer with imposed temperature b.c.

Let us take a single layer of material characterised by a heat capacity C_p , a heat conductivity k and a heat production term H .



The Heat transport equation writes

$$\rho C_p \left(\frac{\partial T}{\partial t} + \vec{v} \cdot \vec{\nabla} T \right) = \vec{\nabla} \cdot (k \vec{\nabla} T) + \rho H \quad (565)$$

At steady state and in the absence of a velocity field, assuming that the material properties to be independent of time and space, and assuming that there is no heat production ($H = 0$), this equation simplifies to

$$\Delta T = 0 \quad (566)$$

Assuming the layer to be parallel to the x -axis, the temperature is $T(x, y) = T(y) = \alpha T + \beta$. In order to specify the constants α and β , we need two constraints.

At the bottom of the layer $y = y_b$ a temperature T_b is prescribed while a temperature T_t is prescribed at the top with $y = y_t$. This ultimately yields a temperature field in the layer given by

$$T(y) = \frac{T_t - T_b}{y_t - y_b}(y - y_b) + T_b$$

If now the heat production coefficient is not zero, the differential equation reads

$$k \Delta T + H = 0 \quad (567)$$

The temperature field is then expected to be of the form

$$T(y) = -\frac{H}{2k}y^2 + \alpha y + \beta \quad (568)$$

Supplied again with the same boundary conditions, this leads to

$$\beta = T_b + \frac{H}{2k}y_b^2 - \alpha y_b$$

ie,

$$T(y) = -\frac{H}{2k}(y^2 - y_b^2) + \alpha(y - y_b) + T_b$$

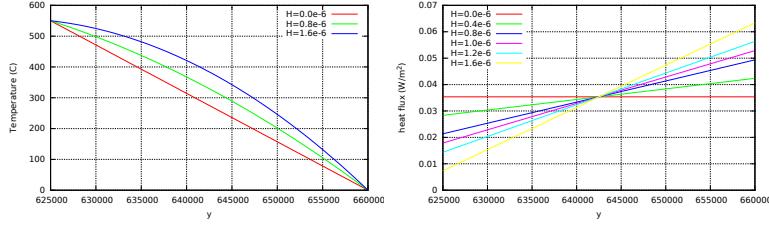
and finally

$$\alpha = \frac{T_t - T_b}{y_t - y_b} + \frac{H}{2k}(y_b + y_t)$$

or,

$$T(y) = -\frac{H}{2k}(y^2 - y_b^2) + \left(\frac{T_t - T_b}{y_t - y_b} + \frac{H}{2k}(y_b + y_t) \right)(y - y_b) + T_b$$

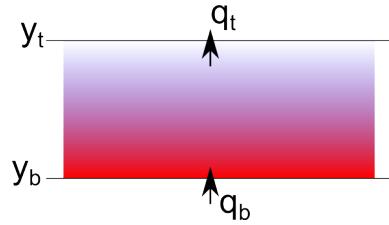
Taking $H = 0$ in this equation obviously yields the temperature field obtained previously. Taking $k = 2.25$, $T_t = 0C$, $T_b = 550C$, $y_t = 660km$, $y_b = 630km$ yields the following temperature profiles and heat fluxes when the heat production H varies:



Looking at the values at the top, which are somewhat estimated to be about $55 - 65 \text{ mW/m}^2$ [993, table 8.6], one sees that value $H = 0.8e-6$ yields a very acceptable heat flux. Looking at the bottom, the heat flux is then about 0.03 W/m^2 which is somewhat problematic since the heat flux at the Moho is reported to be somewhere between 10 and 20 mW/m^2 in [993, table 7.1].

8.23.2 Single layer with imposed heat flux b.c.

Let us now assume that heat fluxes are imposed at the top and bottom of the layer:



We start again from the ODE

$$k\Delta T + H = 0$$

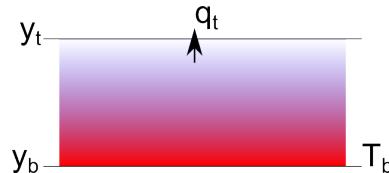
but only integrate it once:

$$k \frac{dT}{dy} + Hy + \alpha = 0$$

At the bottom $q = k(dT/dy)|_{y=y_b} = q_b$ and at the top $q = k(dT/dy)|_{y=y_t} = q_t$ so that

[to finish](#)

8.23.3 Single layer with imposed heat flux and temperature b.c.



[to finish](#)

8.23.4 Half cooling space

TODO.



8.23.5 Plate model

8.23.6 McKenzie slab

When doing thermo-mechanical modelling, the initial temperature field in the domain is of prime importance. This is especially true for the temperature in the slab for subduction modelling as its rheological behaviour is strongly temperature-dependent. One could easily design a simple geometrical initial field but it is unlikely to be close to the field of a slowly subducting slab at an angle in a hot mantle.

McKenzie [1300] derived such approximate initial field from the steady-state energy equation in two dimensions:

$$\rho C_p \vec{v} \cdot \vec{\nabla} T = k \vec{\nabla}^2 T \quad (569)$$

We denote by T_l the temperature at the base of the lithosphere and l its thickness (i.e. the thickness of the slab).

Assuming $\vec{v} = (v_x, 0)$ yields

$$\rho C_p v_x \frac{\partial T}{\partial x} = k \frac{\partial^2 T}{\partial x^2}$$

and substitution of $T' = T/T_l$, $x' = x/l$ and $z' = z/l \in [0, 1]$ in this equation leads to

$$\rho C_p v_x \frac{T_l}{l} \frac{\partial T'}{\partial x'} = k \frac{T_l}{l^2} \left(\frac{\partial^2 T'}{\partial x'^2} + \frac{\partial^2 T'}{\partial z'^2} \right)$$

or

$$\frac{\rho C_p v_x l}{k} \frac{\partial T'}{\partial x'} = \frac{\partial^2 T'}{\partial x'^2} + \frac{\partial^2 T'}{\partial z'^2}$$

and finally (see Eq. 2.3 of [1300]):

$$\frac{\partial^2 T'}{\partial x'^2} - 2R \frac{\partial T'}{\partial x'} + \frac{\partial^2 T'}{\partial z'^2} = 0$$

where R is the thermal Reynolds number

$$R = \frac{\rho C_p v_x l}{2k}$$

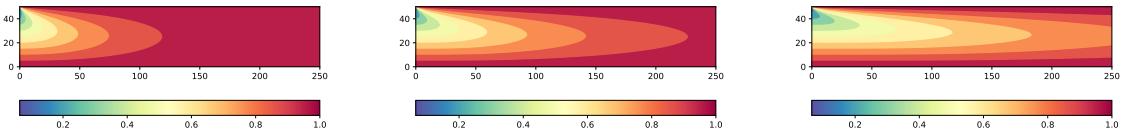
The general solution to this PDE with $T' = 1$ on the top, left and right boundary is

$$T'(x', z') = 1 + \sum_n C_n \exp \left[\left(R - (R^2 + n^2 \pi^2)^{1/2} \right) x' \right] \sin(n \pi z')$$

We now must make an assumption about the temperature on the left boundary ($x' = 0$), which is the temperature of the lithosphere. For simplicity McKenzie assumes that $T'(x' = 0, z') = 1 - z'$ so that $C_n = 2(-1)^n/n\pi$ and finally

$$T'(x', z') = 1 + 2 \sum_n \frac{(-1)^n}{n\pi} \exp \left[\left(R - (R^2 + n^2 \pi^2)^{1/2} \right) x' \right] \sin(n \pi z') \quad (570)$$

Let us build a simple temperature model for a $250\text{km} \times 50\text{km}$ slab, with $\rho = 3000$, $C_p = 1250$, $k = 3$. The python code is available in `images/mckenzie/mckenzie1.py`.

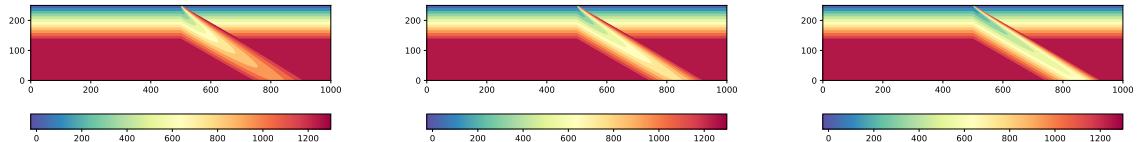


Left to right: Dimensionless temperature T' in a $250\text{km} \times 50\text{km}$ slab for $v_x = 0.5, 1, 2\text{cm/year}$

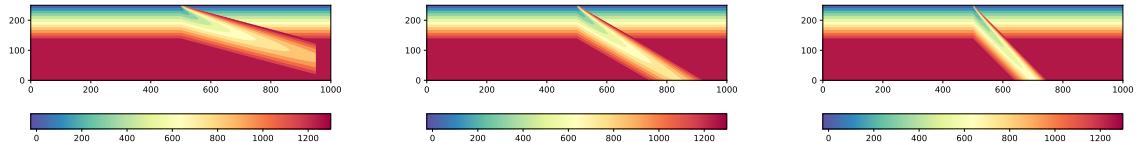
We logically recover the fact that the slower the slab penetrates the mantle the more temperature diffusion dominates over temperature advection. For $v = 0.5\text{cm/year}$ we see that that the slab assumes a constant temperature $T' = 1$ at all depths $0 \leq z' \leq 1$ for $x' \geq 125\text{km}$.

Note that this field is a steady-state field, valid for a constant density, heat conductivity and heat capacity, zero heat production, that it implies that the velocity is constant and that the lithosphere temperature is linear.

One can also embed the slab in a more realistic context, a subduction zone, involving a subducting lithosphere, an over-riding plate and a mantle. The domain is $1000\text{km} \times 250\text{km}$. The mantle temperature is set to 1300° . The slab dip can be varied and so can the velocity. The python code is available in `images/mckenzie/mckenzie2.py`.



Left to right: temperature T for $v_x = 0.5, 1, 2 \text{ cm/year}$ and $\phi = 30^\circ$.



Left to right: temperature T for $v_x = 1 \text{ cm/year}$ and $\phi = 15, 30, 45^\circ$.

8.23.7 Initial temperature for global mantle convection models

This is a difficult topic, and Gottschaldt et al [788] list a few issues or facts to take into account:

- Frequent impacts may have determined the heat structure of the outer layers (Arrhenius and Lepland 2000), leading to an early thermally stable stratification.
- A global magma ocean (Solomatov 2000) or several large scale melting events (Kleine et al. 2004) are also conceivable.
- Fractional crystallisation and subsequent overturn has the potential to result in compositionally or thermally stable layering, too (Elkins-Tanton et al. 2003; Zaranek and Parmentier 2004)

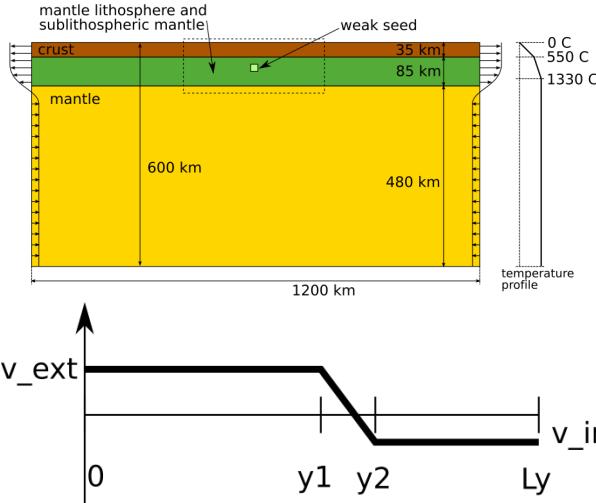
8.24 Kinematic boundary conditions

Boundary conditions come in two basic flavors: essential and natural.

- Essential bcs directly affect DOFs, and are imposed on the FEM matrix.
- Natural bcs do not directly affect DOFs and are imposed on the right-hand side vector.

8.24.1 In-out flux boundary conditions for lithospheric models

C. Thieulot / Physics of the Earth and Planetary Interiors 188 (2011) 47–68



The velocity on the side is given by

$$\begin{aligned} u(y) &= v_{ext} \quad y < L_y \\ u(y) &= \frac{v_{in} - v_{ext}}{y_2 - y_1} (y - y_1) + v_{ext} \quad y_1 < y < y_2 \\ u(y) &= v_{in} \quad y > y_2 \end{aligned}$$

The requirement for volume conservation is:

$$\Phi = \int_0^{L_y} u(y) dy = 0$$

Having chosen v_{in} (the velocity of the plate), one can then compute v_{ext} as a function of y_1 and y_2 .

$$\begin{aligned} \Phi &= \int_0^{y_1} u(y) dy + \int_{y_1}^{y_2} u(y) dy + \int_{y_2}^{L_y} u(y) dy \\ &= v_{ext} y_1 + \frac{1}{2} (v_{in} + v_{ext}) (y_2 - y_1) + (L_y - y_2) v_{in} \\ &= v_{ext} [y_1 + \frac{1}{2} (y_2 - y_1)] + v_{in} [\frac{1}{2} (y_2 - y_1) + (L_y - y_2)] \\ &= v_{ext} \frac{1}{2} (y_1 + y_2) + v_{in} [L_y - \frac{1}{2} (y_1 + y_2)] \end{aligned}$$

and finally

$$v_{ext} = -v_{in} \frac{\frac{1}{2} (y_1 + y_2)}{\frac{1}{2} (y_1 + y_2)}$$

8.25 Computing gradients - the recovery process

write about recovering accurate strain rate components and heat flux components on the nodes.

Let $\vec{g}(\vec{r})$ be the desired nodal field which we want to be the continuous Q_1 representation of the field $\vec{\nabla}f^h$. Since the derivative of the shape function does not exist on the nodes we need to design an algorithm do do so. This problem is well known and has been investigated

refs!

. The main standard techniques are listed hereafter.

8.25.1 Global recovery

The global recovery approach is rather simple: we wish to find \vec{g}^h such that it satisfies

$$\int_{\Omega} \phi \vec{g}^h \, d\Omega = \int_{\Omega} \phi \vec{\nabla} f^h \, d\Omega \quad \forall \phi$$

We will then successively replace ϕ by all the shape functions N_i and since we have $g^h = \sum_j N_i g_i$ we then obtain

$$\sum_j \int N_i N_j d\Omega g_i = \int N_i \vec{\nabla} f^h \, d\Omega$$

or,

$$\mathbb{M} \cdot \vec{\mathcal{G}} = \vec{f}$$

8.25.2 Local recovery - centroid average over patch

8.25.3 Local recovery - nodal average over patch

Let j be the node at which we want to compute \vec{g} . Then

$$\vec{g}_j = \vec{g}(\vec{r}_j) = \frac{\sum_{e \text{ adj. to } j} |\Omega_e| (\vec{\nabla} f)_e(\vec{r}_j)}{\sum |\Omega_e|}$$

where $|\Omega_e|$ is the volume of the element and $(\vec{\nabla} f^h)_e(\vec{r}_j)$ is the gradient of f as obtained with the shape functions inside element e and computed at location \vec{r}_j .

8.25.4 Local recovery - least squares over patch

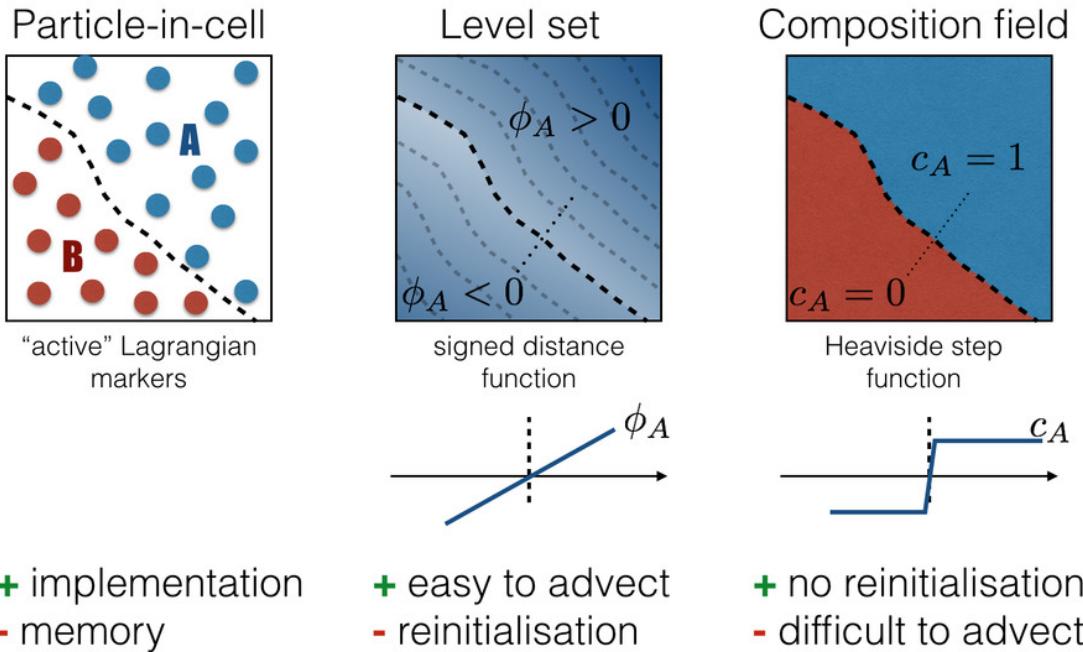
8.25.5 Link to pressure smoothing

When the penalty method is used to solve the Stokes equation, the pressure is then given by $p = -\lambda \vec{\nabla} \cdot \vec{v}$. As explained in section 6.3, the velocity is first obtained and the pressure is recovered by using this equation as a postprocessing step. Since the divergence cannot be computed easily at the nodes, the pressure is traditionally computed in the middle of the elements, yielding an elemental pressure field (remember, we are talking about $Q_1 P_0$ elements here – bi/tri-linear velocity, discontinuous constant pressure)

tie to fieldstone 12

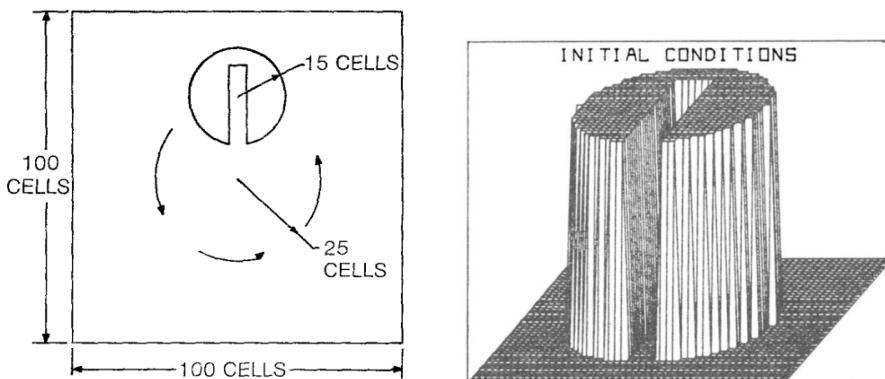
8.26 Tracking materials and/or interfaces

Unless using a fully Lagrangian formulation, one needs an additional numerical method to represent/track the various materials present in an undeformable (Eulerian) mesh. The figure below (by B. Hillebrand) illustrates the three main methods used in geodynamics.



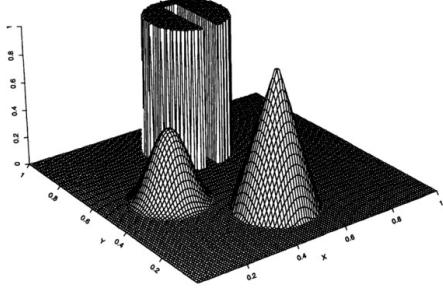
Note that what follows is applicable to FEM, FDM, etc ...

A typical test for advection algorithm is the Zalesak disk [1989]. It is a two dimensional test problem of solid body rotation with a constant angular velocity ω (in rad/sec):



Taken from [1989]. Left: Schematic representation of two dimensional solid body rotation problem. The field inside the cut out has value 3 and it is 1 outside. The rotational speed is such that one full revolution is effected in 628 cycles. The width of the gap separating the two halves of the cylinder, as well as the maximum extent of the “bridge” connecting the two halves, is 5 cells. Right: Perspective view of initial conditions for the two dimensional solid body rotation problem. Note that only a 50×50 portion of the mesh centered on the cylinder is displayed.

This benchmark is widely used in the literature [1707, 1840, 1476, 87, 2002]. Note that the Zalesak disc is often supplemented with a cone and a Gaussian features:



Taken from [1175]. Initial data for solid rotation tests

8.26.1 The Particle-in-cell technique

Remark. The terms 'particle' and 'marker' are commonly (and unfortunately) interchangeably used in the literature in the context of the particle-in-cell technique. However, one should be aware that the marker-and-cell (MAC) technique is something different: it was invented in the early 60's at the Los Alamos Laboratories by Harlow and Welch [870]. For more information on MAC see the review paper by McKee et al [1295].

The Particle-in-cell method is by far the most widely used in computational geodynamics. In its most basic form it is a rather simple method to implement and this probably owes to its success and early adoption [1491] in non-parallel codes such as SOPALE [678], I2VIS [742] or CITCOM [1302] (Appendix B). It has been implemented in ASPECT [698] and the inherent load balancing issues arising from the parallel implementation as well as from the use of Adaptive Mesh Refinement are discussed. It has also been implemented in the MILAMIN code [466] to study LLSVPs [1386].

The basic methodology goes as follows:

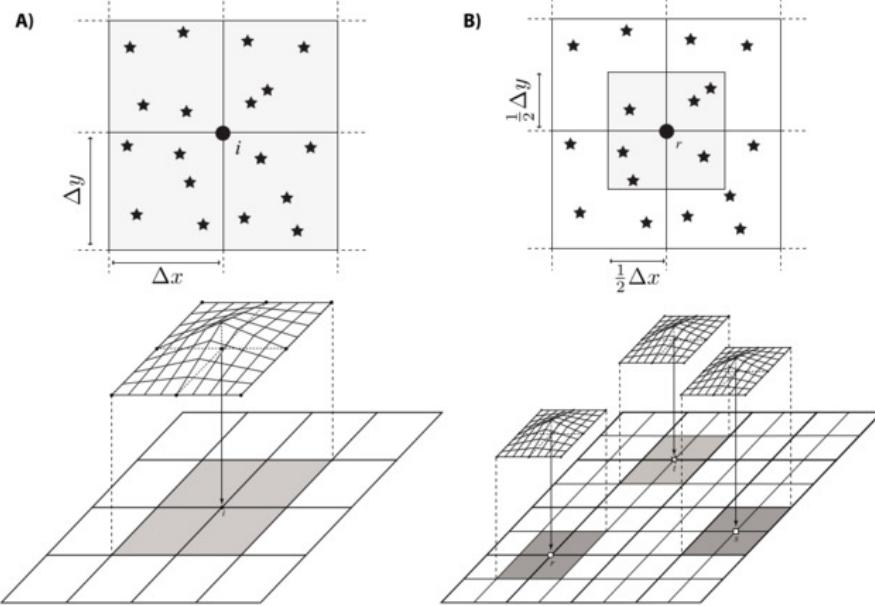
1. distribute particles in the domain
2. assign a material identity (and/or any other quantity) to each of them
3. project particle quantities of the Eulerian nodes of the mesh
4. solve the Stokes equations for a new velocity field
5. interpolate the velocity onto the particles
6. move the particles with their respective velocities
7. go back to step 3

As it turns out each step above needs to be carefully executed and is more difficult than it first looks.

Distributing particles in the domain. Let us assume we wish to distribute N_p particles in the domain. How large must N_p be? To simplify, one end member could be 'as many particles as possible that fit in memory' while the other end member could be 'one per element/cell on average'. While the former does not necessarily guarantee a desired accuracy while being CPU and memory intensive, the latter will certainly lead to zones in the domain void of particles which will be problematic since the projection onto the mesh might yield zero values or very inaccurate values. How many particles (per element/cell) will be enough? Also, should the particles be randomly distributed in the domain or on some kind of regular grid? See fieldstone 13 (Section 9).

Averaging and projection. This is a very critical step. Unfortunately, there is no community-wide agreed-upon method. The problem at hand boils down to: at a given location (\vec{r}) in space I need a quantity which is carried by the particles. The first step is to find the particle(s) close to this point. If done naively, this is a very costly affair, and begs the question what 'close' means. Finding all particles within a radius R of point \vec{r} can be done very efficiently (e.g. with linked lists, Verlet lists, ...) but the choice of R proves to be critical: if too small, there may not be any particle inside the circle, and if too

large there may be many particles inside the circle and the averaging over so many particles in space will prove to be over diffusive. In practice, the FD or FE mesh is used to provide an indication of R . In FDM, the four cells (or quarter cells) around a node represent the volume of space containing the particles whose properties are to be averaged [550] as illustrated in the following figure:



Taken from [550]. The "4-cell" and "1-cell" schemes for projecting properties defined on the markers (denoted by stars) onto a node (denoted by the solid circle). (A) The 4-cell scheme. The support of the interpolating function N_i associated with node i is indicated by the shaded region. Only markers within the support of node i contribute to the projection operation used to define the nodal value at i . The shape of the bilinear interpolation function for node i is indicated in the lower frame. (B) The 1-cell scheme. The thick lines in the lower frame indicate the grid used to discretize the Stokes equations, while the thin lines indicate the grid onto which marker properties are projected. The 1-cell scheme utilizes a compact support of size $\Delta x \times \Delta y$. The support for nodes r, s, t are indicated by the shaded regions. Only markers within the nodal support contribute to the projection operation for that node.

Given that the FEM requires to compute integrals over each element, only the particles inside the element will contribute to the average values assigned to the quadrature points. However, one could also decide to first average the properties onto the nodes before using these nodal values to assign values to the quadrature points. In this case the FDM approach applies.

Finally, in both FDM and FEM bi/trilinear shape functions are used for the interpolation as they can be interpreted as weighing functions. Higher order shape functions could also be used but the standard Q_2 shape functions (Section 4.5) are 2-nd order polynomials which can take negative values (as opposed to the Q_1 shape functions which are strictly positive) and this can pose problems: in some cases, although all values to be averaged are positive, their weighed average can be negative. Q1 projection PUCKETT

it would be nice to have a Q1 and Q2 drawing of a 1D element and show that indeed negative values arise

Assuming that we have established a list of particles, all tracking a field $f(\vec{r})$ and that each particle has an associated weight N_i (function of the location where the average is to be computed or not), we must now compute their average value $\langle f \rangle$. The simplest approach which comes to mind is the (weighted) arithmetic mean (am):

$$\langle f \rangle_{am} = \frac{\sum_{i=1}^n N_i f_i}{\sum_{i=1}^n N_i}$$

In the case where f is the (mass) density ρ , it is indeed what should be used. However, turning now to viscosity η , we know that its value can vary by many orders of magnitude over very short distances. It is then likely that the average runs over values spanning values between 10^{18}Pa s and 10^{25}Pa s . As explained in [1638] the arithmetic averaging tends to 'favour' large values: if the sum runs over 10

particles, 9 carrying the value 10^{25} and 1 carrying the value 10^{19} , the average value (assuming $N_i = 1$ for simplicity) is then

$$\langle \eta \rangle = \frac{9 \cdot 10^{25} + 1 \cdot 10^{19}}{10} \simeq 0.9 \cdot 10^{25}$$

which is much closer to 10^{25} than to 10^{19} . Other averagings are then commonly used, namely the geometric mean (gm) and the harmonic mean (hm), defined as follows:

$$\langle f \rangle_{gm} = \left(\prod_i f_i^{N_i} \right)^{1/\sum_i N_i} \quad \text{or,} \quad \log_{10} \langle f \rangle_{gm} = \frac{\sum_i N_i \log_{10} f_i}{\sum_i N_i}$$

and

$$\langle f \rangle_{hm} = \left(\frac{\sum_{i=1}^n N_i \frac{1}{f_i}}{\sum_i N_i} \right)^{-1} \quad \text{or,} \quad \frac{1}{\langle f \rangle_{hm}} = \frac{\sum_{i=1}^n N_i \frac{1}{f_i}}{\sum_i N_i}$$

The geometric mean can be seen as a form of arithmetic mean of \log_{10} values, while the harmonic mean can be seen as a form of arithmetic mean of the inverse values.

Looking back at the above example, the geometric mean of the viscosities is given by

$$\log \langle \eta \rangle_{gm} = \frac{9 \cdot 25 + 1 \cdot 19}{10} = 24.4 \quad \text{or,} \quad \langle \eta \rangle_{gm} \simeq 2.5 \cdot 10^{24}$$

and the harmonic mean:

$$\langle \eta \rangle_{hm} \simeq \left(\frac{1}{10 \cdot 10^{19}} \right)^{-1} = 10^{20}$$

We see that the harmonic mean tends to favour the small values. Also we recover the known property:

$$\langle f \rangle_{am} \geq \langle f \rangle_{gm} \geq \langle f \rangle_{hm} \tag{571}$$

When all f_i are equal to f_0 their computed average should also be equal to f_0 . As a consequence the weights N_i should fulfill the condition $\sum_{i=1}^n N_i = 1$. If all weights are equal, then $N_i = 1/n$ and the averagings become:

$$\langle f \rangle_{am} = \frac{1}{n} \sum_{i=1}^n f_i \quad \langle f \rangle_{gm} = \prod_i f_i^{1/n} \quad \langle f \rangle_{hm} = \left(\frac{1}{n} \sum_i \frac{1}{f_i} \right)^{-1} \tag{572}$$

There are many papers which have looked at particle averagings and projections. I will for now simply point to the following ones: [1638] [524] [550] [1349] [1498] [1754] [698] [699]. Also check the multiscale PIC in [48]

write more about particle averaging and projection

Interpolation of the velocity onto particles .

Once the particle i has been localised inside a given element (Section 8.21) and its reduced coordinates (r, s, t) determined, the velocity at this location can be computed through the shape functions:

$$\vec{v}_i = \sum_{k=1}^m N_i(r, s, t) \vec{v}_k$$

This approach is not without problem: while the nodal velocities \vec{v}_k are such that⁵⁰ $\vec{\nabla} \cdot \vec{v} = 0$ (in the weak sense), the computed velocity \vec{v}_i is not necessarily divergence-free! In order to remedy this, a Conservative Velocity Interpolation (CVI) has been proposed in [1895].

Moving the particles This is discussed in the context of the Runge-Kutta Methods, see Section 8.20.1.

⁵⁰for incompressible flows, of course

8.26.2 The level set function technique

This method was developed in the 80's by Stanley Osher and James Sethian [1229]

The Level-set Method (LSM), as it is commonly used in Computational Fluid Dynamics – and especially in Computational Geodynamics – represents a close curve Γ (say, in our case, the interface between two fluids or layers) by means of a function ϕ (called the level-set function, or LSF). Γ is then the zero level-set of ϕ :

$$\Gamma = \{(x, y) \mid \phi(x, y) = 0\} \quad (573)$$

The convention is that $\phi > 0$ inside the region delimited by Γ and $\phi < 0$ outside. The function value indicates on which side of the interface a point is located (negative or positive) and this is used to identify materials.

Furthermore, if the curve Γ moves with a velocity \vec{v} , then it satisfies the following equation:

$$\frac{\partial \phi}{\partial t} + \vec{v} \cdot \vec{\nabla} \phi = 0 \quad (574)$$

The level set function is generally chosen to be a signed distance function, i.e. $|\vec{\nabla} \phi| = 1$ everywhere and its value is also the distance to the interface.

As explained in [901], the level-set function ϕ is advected with the velocity \vec{v} which is obtained by solving the Stokes equations. This velocity does not guarantee that after an advection step the signed distance quality of the LSF is preserved. The LSF then needs to be corrected, which is also called reinitialisation. Finally, solving the advection equation must be done in an accurate manner both in time and space, so that so-called ENO (essentially non-oscillatory) schemes are often employed for the space derivative [1440, 1619].

The level set method has not often been used in the geodynamics community with some notable exceptions [200, 201, 851, 822, 2038, 852, 1706, 1705, 901] An overview of the method and applications can be found in [1439].

Several improvements upon the original LSM have been proposed, such as for instance the conservative level set of [2002]. The most notable difference between CLS method originally proposed by Olsson et al. [1431, 1432] and standard LS method lies in the choice of LS function. Instead of the signed distance function, the CLS methods employ the Heaviside function $H(\phi)$

$$H(\phi) = \begin{cases} 1 & \phi > 0 \\ 1/2 & \phi = 0 \\ 0 & \phi < 0 \end{cases}$$

where ϕ is the signed distance function as in the LSM. In practice, a hyperbolic tangent function is used:

$$H(\phi) = \frac{1}{2}(1 + \tanh(\phi/2\epsilon))$$

where ϵ defines the spreading width of H . In the case where there are only two fluids (i.e. a single level set is sufficient), the material properties such as density and viscosity are computed as follows:

$$\rho = \rho_1 + (\rho_2 - \rho_1)H(\phi)$$

$$\eta = \eta_1 + (\eta_2 - \eta_1)H(\phi)$$

8.26.3 The field/composition technique

This is the approach taken by the ASPECT developers [1119, 882]. Each material i is represented by a compositional field c_i , which takes values between 0 and 1. Each compositional field is then advected with the (prescribed or computed) Stokes velocity [372]:

$$\frac{\partial c_i}{\partial t} + \mathbf{v} \cdot \nabla c_i = 0 \quad (575)$$

The value at a point (Finite element node or quadrature point) is 1 if it is in the domain covered by the material i , and 0 otherwise. In one dimension, each compositional field is a Heavyside function. This

approach is somewhat similar to the LSM but the field is essentially discontinuous across the interface, which makes it very difficult to advect. On the plus side, compositional fields need not be reinitialised, as opposed to LSF's.

Accurate numerical advection is a notoriously difficult problem. Unless very specialised techniques are used it often yields undershoot ($c_i < 0$) and overshoot ($c_i > 0$), which ultimately yields mass conservation issues. Also, unless special care is taken, compositional fields tend to become more and more diffuse over time: the SUPG method (Section 8.5) and the entropy viscosity method add small amounts of diffusion to dampen the under- and overshoots. This means that at a given point two or more compositions may have values, which require some form of averaging. If under- and overshoots are present, these averagings can become very problematic and even yield meaningless quantities (e.g. negative viscosities).

One rather old and popular filtering approach is the so-called Lenardic and Kaula filter [1153]:

The filtering algorithm for two-component flow is as follows. An initial step distribution in C is assumed with $C = 0$ and $C = 1$ used to distinguish distinct materials. A high-order upwind solution scheme is applied to equation (2), with prescribed initial conditions, resulting in an uncorrected C field. The field is corrected via the following filtering algorithm:

1. The initial sum of all nodal C values is calculated and is assigned to the variable C_{sum0} .
2. Nodal C values below 0 are set to 0 and the peak value below 0 is assigned to the variable C_{min} .
3. Nodal C values above 1 are set to 1 and the peak value above 1 is assigned to the variable C_{max} .
4. Nodal C values less than or equal to the absolute value of C_{min} are set to 0.
5. Nodal C values greater than or equal to $2 - C_{max}$ are set to 1.
6. The sum of all nodal C values is calculated and assigned to the variable C_{sum1} .
7. The number of nodal C values not 1 or 0 is assigned to the variable NUM.
8. The variable DIST is defined as $(C_{sum0} - C_{sum1})/NUM$ and is added to all C values not 1 or 0.

Taken from Lenardic and Kaula [1153]

The basic idea of the filtering algorithm is to ensure that ϕ remains within the bounds $0 \leq \phi \leq 1$, and to minimize dispersion error. We refer the reader to Lenardic and Kaula (1993) for the detailed explanation and here give the outline of the algorithm for a discrete property field $\phi = \{\phi_i\}$.

Algorithm 10 A property filtering algorithm

- (1) Compute the initial sum S_0 of all values of ϕ .
 - (2) Find the minimal value ϕ_{min} below 0.
 - (3) Find the maximal value ϕ_{max} above 1.
 - (4) Set $\phi_i = 0$ if $\phi_i \leq |\phi_{min}|$.
 - (5) Set $\phi_i = 1$ if $\phi_i \geq 2 - \phi_{max}$.
 - (6) Compute the sum S_1 of all values of ϕ .
 - (7) Compute the number num of $0 < \phi_i < 1$.
 - (8) Add $dist = (S_1 - S_0)/num$ to all $0 < \phi_i < 1$.
-

From FENICS book



Literature: [1892]

Entropy viscosity method [824]

[write about DG approach](#)

8.26.4 The Volume-of-Fluid method

[905]

8.26.5 The method of characteristics

[ask Arie to write something](#)

[510]

8.26.6 The Marker Chain method

In two dimensions, the idea is quite simple: each interface is discretised by means of a number of Lagrangian points (which may or may not vary in time). The points are numbered and connected (think of the connectivity array of a 1D FEM code). In the case of small deformations, and in the absence of in/out-flow boundaries, the method is reasonably trivial to implement, and each couple of point defines a segment (and therefore its normal vector too) which can then be used to answer the question: "at this location, am I above or below this interface" or "am I this domain our outside this domain" (in the case that the interface does not reach any of the boundaries).

This method becomes somewhat impractical when large deformation occurs or, for example, when a domain splits into two (think: slab break off). One interface must then become two, and it requires an algorithm capable of detecting the breakup of the surface and capable of rebuilding/patching the new ones so that they can still be used further. Note that in case of large deformation some markers may get further and further apart from each other which makes for a poor representation of the surface. New markers should then be added but the question of when and where must then be addressed.

Also, switching to three dimensions can prove to be very difficult or simply very costly: the generation of the initial marker position is trivial but their connectivity can be complicated to establish at startup: for instance, a Stokes sphere will require a mesh made of triangles which maps exactly the surface of the sphere (see [1759, 1360] for methods on how to efficiently produce such meshes). In the case of more complex 3D geometries this may prove nearly impossible to do. So will the problem of splitting a surface into two (or merging two domains).

This method is usually coupled to Eulerian meshes (typically with FDM, but not only). It was used in [1946] in the context of salt domes analysis and later in [376, 382]. It is also used in [1847] but little details are given about the algorithms used to track and update the chain in the presence of such large deformation. It is also used (although coupled to level set functions) in the DOUAR code[221] (see Section B). Having worked myself on this code and having had to produce complex initial triangulated surfaces for simulations (see for example [1226]) it is easy to understand why later users of this code did implement the marker-in-cell technique. More recently, it is used to track the free surface position in a FDM code [555, 359].

Finally, Christensen [372] makes the following interesting comment: "One might assume that different methods of representing the discontinuity, for example, by a tracer chain [382] or a cloud of tracers, would solve these problems. However, the difficulties arise not only from the way in which material boundaries are represented. Physically, the rate of shear strain parallel to a rheological boundary is discontinuous. Within the finite element scheme such jump can only be realized at an element boundary. In an Eulerian scheme, where the discontinuity will crosscut the elements, the jump in strain rate must be approximated by a continuous variation, and effectively, the rheological properties on both sides of the discontinuity will be averaged in some way within the element."

8.26.7 Hybrid methods

In Braun et al. [221] a level set method is presented which is based on a 3-D set of triangulated points, which makes it a hybrid between tracers and level set functions: in the DOUAR code (Appendix B) the interface is then explicitly tracked by means of the tracers while the LSF is computed on the FE nodes. Although very promising in theory, this method proved to be difficult to use in practice since it requires a) a triangulation of the interfaces at $t = 0$ which is not trivial if the geometries are complex (think about a slab in 3D); b) the addition or removal of tracers because of the interface deformation and the patching of the triangulation; c) the calculation of the distance to the interfaces for each FE node based on the triangle normal vectors. This probably explains why the Particle-In-Cell method was later implemented in this code (pers. comm.). Note that another very similar approach is used in [1619].

8.26.8 Boundary fitted mesh

This method is rather simple to implement and works well for small deformations. It is for instance used by Frehner [658] (see online supplementary material) in which it is stated: "The numerical grid is set up in such a way that the interface between different material phases (two layers in this case) coincides with element boundaries. Hence, each element belongs to a unique material phase and no interpolation

I still have
pics from the
old days using
DOUAR- in-
clude

is necessary.” With such a method, each element is initially attributed a material phase/number and its material properties do not change.

8.27 Static condensation

The idea behind static condensation is quite simple: in some cases, there are dofs belonging to an element which only belong to that element. For instance, the so-called MINI element ($P_1^+ \times P_1$) showcases a bubble function in the middle (see section ??). In the following, \vec{V}^* corresponds to the list of such dofs inside an element. The discretised Stokes equations on any element looks like:

$$\begin{pmatrix} \mathbb{K} & L & \mathbb{G} \\ L^T & \mathbb{K}^* & H \\ \mathbb{G}^T & H^T & 0 \end{pmatrix}_e \begin{pmatrix} \vec{\mathcal{V}} \\ \vec{V}^* \\ \vec{\mathcal{P}} \end{pmatrix}_e = \begin{pmatrix} \vec{f} \\ \vec{f}^* \\ \vec{h} \end{pmatrix}_e \quad (576)$$

This is only a re-writing of the elemental Stokes matrix where the matrix \mathbb{K} has been split in four parts. Note that the matrix \mathbb{K}^* is diagonal.

This can also be re-written in non-matrix form:

$$\mathbb{K} \cdot \vec{\mathcal{V}} + L \cdot \vec{V}^* + \mathbb{G} \cdot \vec{\mathcal{P}} = \vec{f} \quad (577)$$

$$L^T \vec{V} + K^* \cdot \vec{V}^* + H \cdot \vec{\mathcal{P}} = \vec{f}^* \quad (578)$$

$$\mathbb{G}^T \cdot \vec{\mathcal{V}} + H^T \vec{V}^* = \vec{h} \quad (579)$$

The V^* in the second equation can be isolated:

$$\vec{V}^* = \mathbb{K}^{-*} \cdot (\vec{f}^* - L^T \cdot \vec{\mathcal{V}} - H \cdot \vec{\mathcal{P}})$$

and inserted in the first and third equations:

$$\mathbb{K} \cdot \vec{\mathcal{V}} + L \left[\mathbb{K}^{-*} (\vec{f}^* - L^T \cdot \vec{\mathcal{V}} - H \cdot \vec{\mathcal{P}}) \right] + \mathbb{G} \cdot \vec{\mathcal{P}} = \vec{f} \quad (580)$$

$$\mathbb{G}^T \cdot \vec{\mathcal{V}} + H^T \left[\mathbb{K}^{-*} (\vec{f}^* - L^T \cdot \vec{\mathcal{V}} - H \cdot \vec{\mathcal{P}}) \right] = \vec{h} \quad (581)$$

or,

$$(\mathbb{K} - L \cdot \mathbb{K}^{-*} \cdot L^T) \cdot \vec{\mathcal{V}} + (G - L \cdot \mathbb{K}^{-*} \cdot H) \cdot \vec{\mathcal{P}} = \vec{f} - L \cdot \mathbb{K}^{-*} \cdot \vec{f}^* \quad (582)$$

$$(G^T - H^T \cdot \mathbb{K}^{-*} \cdot L^T) \cdot \vec{\mathcal{V}} - (H^T \cdot \mathbb{K}^{-*} \cdot H) \cdot \vec{\mathcal{P}} = \vec{h} - H^T \cdot \mathbb{K}^{-*} \cdot \vec{f}^* \quad (583)$$

i.e.

$$\underline{\mathbb{K}} \cdot \vec{\mathcal{V}} + \underline{\mathbb{G}} \cdot \vec{\mathcal{P}} = \underline{\vec{f}} \quad (584)$$

$$\underline{\mathbb{G}}^T \cdot \vec{\mathcal{V}} - \underline{\mathbb{C}} \cdot \vec{\mathcal{P}} = \underline{\vec{h}} \quad (585)$$

with

$$\underline{\mathbb{K}} = K - L \cdot \mathbb{K}^{-*} \cdot L^T \quad (586)$$

$$\underline{\mathbb{G}} = G - L \cdot \mathbb{K}^{-*} \cdot H \quad (587)$$

$$\underline{\mathbb{C}} = H^T \cdot \mathbb{K}^{-*} \cdot H \quad (588)$$

$$\underline{\vec{f}} = \vec{f} - L \cdot \mathbb{K}^{-*} \cdot \vec{f}^* \quad (589)$$

$$\underline{\vec{h}} = \vec{h} - H^T \cdot \mathbb{K}^{-*} \cdot \vec{f}^* \quad (590)$$

Note that $\underline{\mathbb{K}}$ is symmetric, and so is the Stokes matrix.

For instance, in the case of the MINI element, the dofs corresponding to the bubble could be eliminated at the elemental level, which would make the Stokes matrix smaller. However, it is then important to note that static condensation introduces a pressure-pressure term which was not there in the original formulation.

8.28 Measuring incompressibility

The velocity divergence error integrated over the whole element is given by

$$e_{div} = \int_{\Omega} (\vec{\nabla} \cdot \vec{v}^h - \underbrace{\vec{\nabla} \cdot \vec{v}}_{=0}) d\Omega = \int_{\Omega} (\vec{\nabla} \cdot \vec{v}^h) d\Omega \quad (591)$$

where Γ_e is the boundary of element e and \vec{n} is the unit outward normal of Γ_e .

Furthermore, one can show that [529]:

$$e_{div} = \int_{\Gamma_e} \vec{v}^h \cdot \vec{n} d\Gamma$$

The reason is as follows and is called the divergence theorem⁵¹: suppose a volume V subset of \mathbb{R}^d which is compact and has a piecewise smooth boundary S , and if \vec{F} is a continuously differentiable vector field then

$$\int_V (\vec{\nabla} \cdot \vec{F}) dV = \int_S (\vec{F} \cdot \vec{n}) dS$$

The left side is a volume integral while the right side is a surface integral. Note that sometimes the notation $d\vec{S} = \vec{n} dS$ is used so that $\vec{F} \cdot \vec{n} dS = \vec{F} \cdot d\vec{S}$.

The average velocity divergence over an element can be defined as

$$\langle \vec{\nabla} \cdot \vec{v} \rangle_e = \frac{1}{V_e} \int_{\Omega_e} (\vec{\nabla} \cdot \vec{v}) d\Omega = \frac{1}{V_e} \int_{\Gamma_e} \vec{v} \cdot \vec{n} d\Gamma$$

Note that for elements using discontinuous pressures we shall recover a zero divergence element per element (local mass conservation) while for continuous pressure elements the mass conservation is guaranteed only globally (i.e. over the whole domain), see section 3.13.2 of [813].

Note that one could instead compute $\langle |\vec{\nabla} \cdot \vec{v}| \rangle_e$. Either volume or surface integral can be computed by means of an appropriate Gauss-Legendre quadrature algorithm.

[implement and report](#)

⁵¹https://en.wikipedia.org/wiki/Divergence_theorem

8.29 Periodic boundary conditions

This type of boundary conditions can be handy in some specific cases such as infinite domains. The idea is simple: when material leaves the domain through a boundary it comes back in through the opposite boundary (which of course presupposes a certain topology of the domain).

For instance, if one wants to model a gas at the molecular level and wishes to avoid interactions of the molecules with the walls of the container, such boundary conditions can be used, mimicking an infinite domain in all directions.

Let us consider the small mesh depicted hereunder:

missing picture

We wish to implement horizontal boundary conditions so that

$$u_5 = u_1 \quad u_{10} = u_6 \quad u_{15} = u_{11} \quad u_{20} = u_{16}$$

One could of course rewrite these conditions as constraints and extend the Stokes matrix but this approach turns out to be not practical at all.

Instead, the method is rather simple: replace in the connectivity array the dofs on the right side (nodes 5, 10, 15, 20) by the dofs on the left side. In essence, we wrap the system upon itself in the horizontal direction so that elements 4, 8 and 12 'see' and are 'made of' the nodes 1, 6, 11 and 16. In fact, this is only necessary during the assembly. Everywhere in the loops nodes 5, 10, 15 and 20 appear one must replace them by their left pendants 1, 6, 11 and 16. This automatically generates a matrix with lines and columns corresponding to the u_5 , u_{10} , u_{15} and u_{20} being exactly zero. The Stokes matrix is the same size, the blocks are the same size and the symmetric character of the matrix is respected. However, there remains a problem. There are zeros on the diagonal of the above mentioned lines and columns. One must then place there 1 or a more appropriate value.

Another way of seeing this is as follows: let us assume we have built and assembled the Stokes matrix, and we want to impose periodic b.c. so that dof j and i are the same. The algorithm is composed of four steps:

1. add col j to col i
2. add row j to row i (including rhs)
3. zero out row j , col j
4. put average diagonal value on diagonal (j, j)

Remark. *Unfortunately the non-zero pattern of the matrix with periodic b.c. is not the same as the matrix without periodic b.c.*

8.30 Removing rotational nullspace

November 11, 2019 - C.T.

When free slip boundary conditions are prescribed in an annulus or hollow sphere geometry there exists a rotational nullspace, or in other words there exists a tangential velocity field ('pure rotation') which, if added or subtracted to the solution, generates a solution which is still the solution of the PDEs.

As in the pressure normalisation case (see section 8.14), the solution is simple:

1. fix the tangential velocity at *one* node on a boundary, and solve the system (the nullspace has been removed)
2. post-process the solution to have the velocity field fulfill the required conditions, i.e. either a zero net angular momentum or a zero net angular velocity of the domain.

Remark. In ASPECT this is available under the option "Remove nullspace = angular momentum" and "Remove nullspace = net rotation". The "angular momentum" option removes a rotation such that the net angular momentum is zero. The "net rotation" option removes the net rotation of the domain.

Angular momentum approach In order to remove the angular momentum, we search for a rotation vector $\vec{\omega}$ such that

$$\int_{\Omega} \rho[\vec{r} \times (\vec{v} - \vec{\omega} \times \vec{r})] dV = \vec{0} \quad (592)$$

The angular momentum of a rigid body can be obtained from the sum of the angular momentums of the particles forming the body⁵²:

$$\vec{H} = \sum_i \vec{L}_i \quad (593)$$

$$= \sum_i \vec{r}_i \times m_i \vec{v}_i \quad (594)$$

$$= \sum_i \vec{r}_i \times m_i (\vec{\omega}_i \times \vec{r}_i) \quad (595)$$

$$= \sum_i m_i \begin{pmatrix} \sum_i m_i(y_i^2 + z_i^2) & -\sum_i m_i x_i y_i & -\sum_i m_i x_i z_i \\ -\sum_i m_i x_i y_i & \sum_i m_i(x_i^2 + z_i^2) & -\sum_i m_i y_i z_i \\ -\sum_i m_i x_i z_i & -\sum_i m_i y_i z_i & \sum_i m_i(x_i^2 + y_i^2) \end{pmatrix} \cdot \begin{pmatrix} \omega_x \\ \omega_y \\ \omega_z \end{pmatrix} \quad (596)$$

In the continuum limit, we have:

$$\vec{H} = \int_{\Omega} \rho(\vec{r}) \vec{r} \times \vec{v} dV \quad (597)$$

and the 3×3 moment of inertia tensor \mathbf{I} (also called inertia tensor) is given by⁵³

$$\mathbf{I} = \int_{\Omega} \rho(\vec{r}) [\vec{r} \cdot \vec{r} \mathbf{1} - \vec{r} \times \vec{r}] dV \quad (598)$$

so that the above equation writes: $\vec{H} = \mathbf{I} \cdot \vec{\omega}$ and then $\vec{\omega} = \mathbf{I}^{-1} \cdot \vec{H}$.

Ultimately, at each velocity node a rotation about the rotation vector $\vec{\omega}$ is then subtracted from the velocity solution [2010, eq. 26]:

$$\vec{v}_{new} = \vec{v}_{old} - \vec{\omega} \times \vec{r} \quad (599)$$

Angular velocity approach The angular velocity⁵⁴ vector is given by $\vec{\omega} = \frac{\vec{r} \times \vec{v}}{r^2}$ so that the volume-averaged angular velocity of the cylindrical shell is:

$$\vec{\omega} = \frac{1}{|\Omega|} \int_{\Omega} \frac{\vec{r} \times \vec{v}}{r^2} dV \quad (600)$$

⁵²<http://www.kwon3d.com/theory/moi/iten.html>

⁵³https://en.wikipedia.org/wiki/Moment_of_inertia

⁵⁴https://en.wikipedia.org/wiki/angular_velocity

8.30.1 Three dimensions

The angular momentum vector is given by:

$$\vec{H} = \int_{\Omega} \rho(\vec{r}) \begin{pmatrix} yw - zv \\ zu - xw \\ xv - yu \end{pmatrix} d\vec{r} = \begin{pmatrix} \int_{\Omega} \rho(\vec{r})(yw - zv) d\vec{r} \\ \int_{\Omega} \rho(\vec{r})(zu - xw) d\vec{r} \\ \int_{\Omega} \rho(\vec{r})(xv - yu) d\vec{r} \end{pmatrix} = \begin{pmatrix} H_x \\ H_y \\ H_z \end{pmatrix} \quad (601)$$

while the inertia tensor for a continuous body is given by

$$\mathbf{I} = \int_{\Omega} \rho(\vec{r}) [\vec{r} \cdot \vec{r} \mathbf{1} - \vec{r} \times \vec{r}] d\vec{r} \quad (602)$$

$$= \int_{\Omega} \rho(\vec{r}) \left[\begin{pmatrix} x^2 + y^2 + z^2 & 0 & 0 \\ 0 & x^2 + y^2 + z^2 & 0 \\ 0 & 0 & x^2 + y^2 + z^2 \end{pmatrix} - \begin{pmatrix} xx & xy & xz \\ yx & yy & yz \\ zx & zy & zz \end{pmatrix} \right] d\vec{r} \quad (603)$$

$$= \int_{\Omega} \rho(\vec{r}) \begin{pmatrix} y^2 + z^2 & -xy & -xz \\ -yx & x^2 + z^2 & -yz \\ -zx & -zy & x^2 + y^2 \end{pmatrix} d\vec{r} \quad (604)$$

$$= \begin{pmatrix} \int_{\Omega} \rho(\vec{r})(y^2 + z^2) d\vec{r} & -\int_{\Omega} \rho(\vec{r})xy d\vec{r} & -\int_{\Omega} \rho(\vec{r})xz d\vec{r} \\ -\int_{\Omega} \rho(\vec{r})yx d\vec{r} & \int_{\Omega} \rho(\vec{r})(x^2 + z^2) d\vec{r} & -\int_{\Omega} \rho(\vec{r})yz d\vec{r} \\ -\int_{\Omega} \rho(\vec{r})zx d\vec{r} & -\int_{\Omega} \rho(\vec{r})zy d\vec{r} & \int_{\Omega} \rho(\vec{r})(x^2 + y^2) d\vec{r} \end{pmatrix} \quad (605)$$

$$= \begin{pmatrix} I_{xx} & I_{xy} & I_{xz} \\ I_{yx} & I_{yy} & I_{yz} \\ I_{zx} & I_{zy} & I_{zz} \end{pmatrix} \quad (606)$$

8.30.2 Two dimensions

In two dimensions, flow is taking place in the (x, y) plane. This means that $\vec{r} = (x, y, 0)$ and $\vec{v} = (u, v, 0)$ are coplanar, and therefore that $\vec{\omega}$ is perpendicular to the plane. We have then

$$\vec{H} = \int_{\Omega} \rho(\vec{r}) \begin{pmatrix} 0 \\ 0 \\ xv - yu \end{pmatrix} d\vec{r} = \begin{pmatrix} 0 \\ 0 \\ \int_{\Omega} \rho(\vec{r})(xv - yu) d\vec{r} \end{pmatrix} \quad (607)$$

and

$$\mathbf{I} = \begin{pmatrix} I_{xx} & I_{xy} & I_{xz} \\ I_{yx} & I_{yy} & I_{yz} \\ I_{zx} & I_{zy} & I_{zz} \end{pmatrix} = \begin{pmatrix} I_{xx} & I_{xy} & 0 \\ I_{yx} & I_{yy} & 0 \\ 0 & 0 & I_{zz} \end{pmatrix} \quad (608)$$

since $I_{xz} = I_{yz} = 0$ as $z = 0$, and with $I_{xx} = \int_{\Omega} \rho(\vec{r})y^2 d\vec{r}$ and $I_{yy} = \int_{\Omega} \rho(\vec{r})x^2 d\vec{r}$. The solution to $\mathbf{I} \cdot \vec{\omega} = \vec{H}$ can be easily obtained (see Appendix G.2):

$$\omega_x = \frac{1}{\det(\mathbf{I})} \begin{vmatrix} 0 & I_{xy} & 0 \\ 0 & I_{yy} & 0 \\ H_3 & 0 & I_{zz} \end{vmatrix} = 0 \quad (609)$$

$$\omega_y = \frac{1}{\det(\mathbf{I})} \begin{vmatrix} I_{xx} & 0 & 0 \\ I_{yx} & 0 & 0 \\ 0 & H_z & I_{zz} \end{vmatrix} = 0 \quad (610)$$

$$\omega_z = \frac{1}{\det(\mathbf{I})} \begin{vmatrix} I_{xx} & I_{xy} & 0 \\ I_{yx} & I_{yy} & 0 \\ 0 & 0 & H_z \end{vmatrix} \quad (611)$$

$$= \frac{1}{\det(\mathbf{I})} (I_{xx}I_{yy}H_z - I_{yx}I_{xy}H_z) \quad (612)$$

$$= \frac{1}{\det(\mathbf{I})} (I_{xx}I_{yy} - I_{yx}I_{xy}) H_z \quad (613)$$

with $\det(\mathbf{I}) = I_{xx}I_{yy}I_{zz} - I_{yx}I_{xy}I_{zz}$. Concretely, this means that in 2D one does not need to solve the system $\mathbf{I} \cdot \vec{\omega} = \vec{H}$ since only ω_z is not zero.

Likewise, the volume-averaged angular velocity is then simply:

$$\omega_z = \frac{1}{|\Omega|} \int_{\Omega} \frac{xv - yu}{r^2} d\vec{r} \quad (614)$$

8.31 Picard and Newton

explain why our eqs are nonlinear

8.31.1 Picard iterations

Let us consider the following system of nonlinear algebraic equations:

$$\mathbb{A}(\vec{X}) \cdot \vec{X} = \vec{b}(\vec{X})$$

Both matrix and right hand side depend on the solution vector \vec{X} .

For many mildly nonlinear problems, a simple successive substitution iteration scheme (also called Picard method) will converge to the solution and it is given by the simple relationship:

$$\mathbb{A}(\vec{X}^n) \cdot \vec{X}^{n+1} = \vec{b}(\vec{X}^n)$$

where n is the iteration number. It is easy to implement:

1. guess \vec{X}^0 or use the solution from previous time step
2. compute \mathbb{A} and \vec{b} with current solution vector \vec{X}^{old}
3. solve system, obtain \vec{T}^{new}
4. check for convergence (are \vec{X}^{old} and \vec{X}^{new} close enough?)
5. $\vec{X}^{old} \leftarrow \vec{X}^{new}$
6. go back to 2.

There are various ways to test whether iterations have converged. The simplest one is to look at $\|\vec{X}^{old} - \vec{X}^{new}\|$ (in the L_1 , L_2 or maximum norm) and assess whether this term is smaller than a given tolerance ϵ . However this approach poses a problem: in geodynamics, if two consecutively obtained temperatures do not change by more than a thousandth of a Kelvin (say $\epsilon = 10^{-3}$ K) we could consider that iterations have converged but looking now at velocities which are of the order of a cm/year (i.e. $\sim 3 \cdot 10^{-11}$ m/s) we would need a tolerance probably less than 10^{-13} m/s. We see that using absolute values for a convergence criterion is a potentially dangerous affair, which is why one uses a relative formulation (thereby making ϵ a dimensionless parameter):

$$\frac{\|\vec{X}^{old} - \vec{X}^{new}\|}{\|\vec{X}^{new}\|} < \epsilon$$

Another convergence criterion is proposed by Reddy (section 3.7.2) [1551]:

$$\left(\frac{(\vec{X}^{old} - \vec{X}^{new}) \cdot (\vec{X}^{old} - \vec{X}^{new})}{\vec{X}^{new} \cdot \vec{X}^{new}} \right)^{1/2} < \epsilon$$

Yet another convergence criterion is used in [1756]: the means $\langle \vec{X}^{old} \rangle$, $\langle \vec{X}^{new} \rangle$ as well as the variances σ^{old} and σ^{new} are computed, followed by the correlation factor R :

$$R = \frac{\langle (\vec{X}^{old} - \langle \vec{X}^{old} \rangle) \cdot (\vec{X}^{new} - \langle \vec{X}^{new} \rangle) \rangle}{\sqrt{\sigma^{old} \sigma^{new}}}$$

Since the correlation is normalised, it takes values between 0 (very dissimilar velocity fields) and 1 (very similar fields). The following convergence criterion is then used: $1 - R < \epsilon$.

write about nonlinear residual

Note that in some instances and improvement in convergence rate can be obtained by use of a relaxation formula where one first solves

$$\mathbb{A}(\vec{X}^n) \cdot \vec{X}^* = \vec{b}(\vec{X}^n)$$

and then updates \vec{X}^n as follows:

$$\vec{X}^n = \gamma \vec{X}^n + (1 - \gamma) \vec{X}^* \quad 0 < \gamma \leq 1$$

When $\gamma = 1$ we recover the standard Picard iterations formula above.

8.32 Defect correction formulation

Work in progress.

We start from the system to solve:

$$\mathbf{A}(\vec{X}) \cdot \vec{X} = \vec{b}(\vec{X})$$

with the associated residual vector \vec{F}

$$\vec{F}(\vec{X}) = \mathbf{A}(\vec{X}) \cdot \vec{X} - \vec{b}(\vec{X})$$

The Newton-Raphson algorithm consists of two steps:

1. solve $\mathbf{J}_k \cdot \delta \vec{X}_k = -\vec{F}(\vec{X}_k)$, or in the case of the incompressible Stokes equation FEM system:

$$\begin{pmatrix} \mathbf{J}_k^{\mathcal{V}\mathcal{V}} & \mathbf{J}_k^{\mathcal{V}\mathcal{P}} \\ \mathbf{J}_k^{\mathcal{P}\mathcal{V}} & 0 \end{pmatrix} \cdot \begin{pmatrix} \delta \vec{\mathcal{V}}_k \\ \delta \vec{\mathcal{P}}_k \end{pmatrix} = \begin{pmatrix} -\vec{F}_k^{\mathcal{V}} \\ -\vec{F}_k^{\mathcal{P}} \end{pmatrix}$$

2. update $\vec{X}_{k+1} = \vec{X}_k + \alpha_k \delta \vec{X}_k$

The defect correction Picard approach consists of neglecting the derivative terms present in the J terms (Eqs. 16,17,18 of [651]) so that

$$\mathbf{J}_k^{\mathcal{V}\mathcal{V}} \simeq \mathbb{K}_k \quad \mathbf{J}_k^{\mathcal{V}\mathcal{P}} \simeq \mathbb{G} \quad \mathbf{J}_k^{\mathcal{P}\mathcal{V}} \simeq \mathbb{G}^T$$

and step 1 of the above iterations become:

$$\begin{pmatrix} \mathbb{K}_k & \mathbb{G} \\ \mathbb{G}^T & 0 \end{pmatrix} \cdot \begin{pmatrix} \delta \vec{\mathcal{V}}_k \\ \delta \vec{\mathcal{P}}_k \end{pmatrix} = \begin{pmatrix} -\vec{F}_k^{\mathcal{V}} \\ -\vec{F}_k^{\mathcal{P}} \end{pmatrix}$$

(MSc Thesis): implement a simple Newton solver and apply it to a few nonlinear benchmarks.

8.33 Parallel or not?

Let us assume that we want to run a simulation of the whole Earth mantle with a constant resolution of 5km. The volume of the mantle is

$$V_{mantle} = \frac{4}{3}\pi(R_{out}^3 - R_{in}^3) \simeq 10^{12} km^3$$

while the volume of an element is $V_e = 125 km^3$ (this is only an average since the tessellation of a hollow sphere with hexahedra yields elements which are not all similar [1759]). Consequently, the number of cells needed to discretise the mantle is

$$N_{el} = \frac{V_{mantle}}{V_e} \simeq 8 \times 10^9$$

We know that the matrix size is approx. 4 times the number of elements in 3D:

$$N \simeq 25 \times 10^9$$

Using between 9 and 125 particles per element (a very conservative number), the total number of particles is then

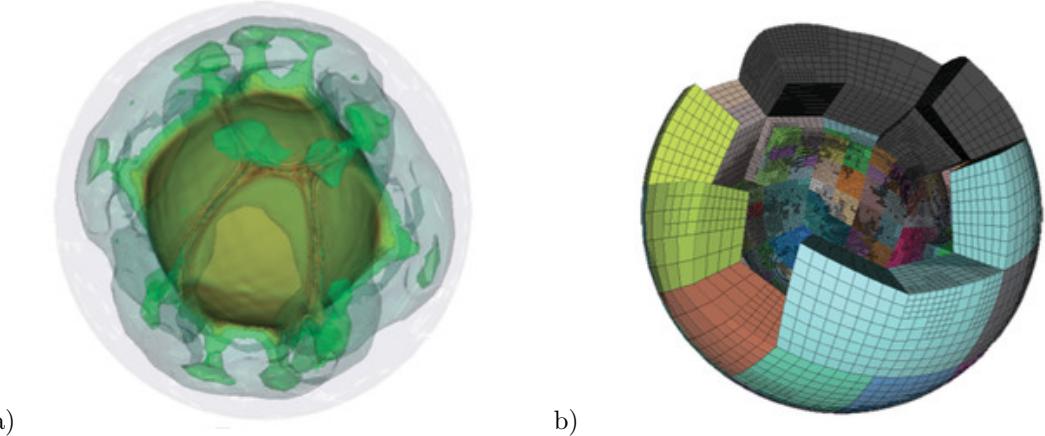
$$N_{particles} \geq 10^{10}$$

The unescapable conclusion is that high-resolution 3D calculations have a very large memory footprint and require extremely long computational times.

The only way to overcome this problem is by resorting to using supercomputers with many processors and large memory capacities.

The idea behind parallel programming is to have each processor carry out only a subset of the total number of operations required. In order to reduce the memory footprint on each processor, only a subset of the computational mesh is known by each: one speaks then of domain decomposition.

An example of such a large parallel calculation of 3D convection with domain decomposition in a spherical shell can be found in [1119]:



a) Isocontours of the temperature field; b) Partitioning of the domain onto 512 proc. The mesh counts 1,424,176 cells. The solution has approximately 54 million unknowns (39 million vel., 1.7 million press., and 13 million temp.)

8.34 Stream function

 Literature[758][1639][382][374]

8.34.1 In Cartesian coordinates

The Stream function (commonly denoted by Φ or Ψ) approach is a useful approach in fluid dynamics as it can provide relatively quick solutions to 2D incompressible flow problems. Lines of constant Φ are called stream lines and give a useful representation of the flow. The definition of the stream function is such that

$$u = -\frac{\partial \Phi}{\partial y} \quad (615)$$

$$v = \frac{\partial \Phi}{\partial x} \quad (616)$$

It then follows that the velocity field based on the above equations automatically fulfills the continuity equation:

$$\vec{\nabla} \cdot \vec{v} = \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = -\frac{\partial^2 \Phi}{\partial x \partial y} + \frac{\partial^2 \Phi}{\partial y \partial x} = 0$$

The stream function can also be substituted into the (constant viscosity) Stokes equation $-\vec{\nabla} p + \eta \Delta \vec{v} = \vec{0}$:

$$-\frac{\partial p}{\partial x} - \eta \left(\frac{\partial^3 \Phi}{\partial^2 x \partial y} + \frac{\partial^3 \Phi}{\partial^3 y} \right) = 0 \quad (617)$$

$$-\frac{\partial p}{\partial y} - \eta \left(\frac{\partial^3 \Phi}{\partial^3 x} + \frac{\partial^3 \Phi}{\partial x \partial^2 y} \right) = 0 \quad (618)$$

We can now eliminate the pressure term by taking the partial derivative of the first equation with respect to y and the partial derivative of the second one with respect to x , and subtracting both. We get:

$$\frac{\partial^4 \Phi}{\partial x^4} + \frac{\partial^4 \Phi}{\partial x^2 \partial y^2} + \frac{\partial^4 \Phi}{\partial y^4} = 0 \quad (619)$$

or,

$$\left(\frac{\partial^2}{\partial x^2} + \frac{\partial^2}{\partial y^2} \right) \left(\frac{\partial^2}{\partial x^2} + \frac{\partial^2}{\partial y^2} \right) \Phi = 0 \quad (620)$$

or,

$$\vec{\nabla}^2 \vec{\nabla}^2 \Phi = \vec{\nabla}^4 \Phi = 0$$

which is known as the Biharmonic operator.

8.34.2 In Cylindrical coordinates

TODO

VERIFY THOSE! minus signs ?

$$\mathbf{v}_r = \frac{1}{r} \frac{\partial \Phi}{\partial \theta}$$

$$\mathbf{v}_\theta = -\frac{\partial \Phi}{\partial r}$$

8.35 Corner flow

The mantle wedge comprised between the downgoing slab and the overriding plate has been extensively studied since very important geodynamical processes take place in it or right above it (slab dehydration and water transport, melting, over-riding plate deformation, vulcanism, ...).

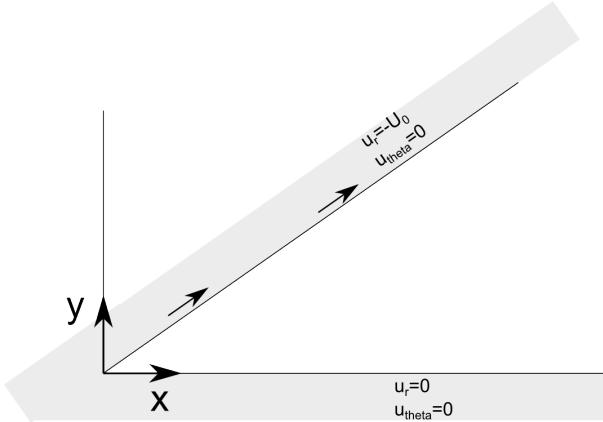
To first approximation one can approach the problem and simplify it greatly by assuming that both plates kinematic behaviour are independent of what happens in the wedge, that the wedge geometry does not change over time, that the problem is essentially 2D, and that the mantle extends very far away from the actual wedge (plates are infinite).

Under such assumptions, it is possible to derive an analytical solution for incompressible Stokes flow in the wedge as documented at p. 224 in Batchelor [82].

Literature: [1790]

FIND refs. check new version of Vol7 theoretical geophys

A corner flow setup is shown hereunder:



The solution to this problem is arrived at by means of the stream function Φ , defined as $u = -\partial\Phi/\partial y$ and $v = \partial\Phi/\partial x$, so that we automatically have $\vec{\nabla} \cdot \vec{v} = 0$. As shown in Section 8.34, the stream function Φ is then the solution to the biharmonic equation

$$\vec{\nabla}^2 \vec{\nabla}^2 \Phi = \vec{\nabla}^4 \Phi = 0$$

Considering the geometry of the problem has plates of infinite extent with constant relative velocity, the solution for velocity everywhere is expected to be independent of r . This means the equation is separable and we will use a solution of the form

$$\Phi(r, \theta) = R(r)f(\theta)$$

However, given the infinite extent of the domain, the velocity is expected to be independent of r , so we postulate $R(r) = r$ (look at the relationship between velocity components and stream function), or:

$$\Phi(r, \theta) = rf(\theta)$$

and we then have to solve

$$\Delta \left(\frac{1}{r}(f + f'') \right) = \frac{1}{r^3}(f + 2f'' + f''') = 0.$$

The solution of this equation for f is:

$$\begin{aligned} f(\theta) &= A \sin \theta + B \cos \theta + C \theta \sin \theta + D \theta \cos \theta \\ f'(\theta) &= A \cos \theta - B \sin \theta + C(\sin \theta + \theta \cos \theta) + D(\cos \theta - \theta \sin \theta) \end{aligned}$$

with

$$\begin{aligned} v_r &= \frac{1}{r} \frac{\partial \Phi}{\partial \theta} = f'(\theta) \\ v_\theta &= -\frac{\partial \Phi}{\partial r} = -f(\theta) \end{aligned}$$

A, B, C and D are four constants to be determined by means of the boundary conditions which are as follows:

$$\begin{aligned}\mathbf{v}_r(\theta = 0) &= 0 \\ \mathbf{v}_\theta(\theta = 0) &= 0 \\ \mathbf{v}_r(\theta = \theta_0) &= -U_0 \\ \mathbf{v}_\theta(\theta = \theta_0) &= 0\end{aligned}$$

or,

$$f'(0) = A + D = 0 \quad (621)$$

$$f(0) = B = 0 \quad (622)$$

$$f'(\theta_0) = -U_0 \quad (623)$$

$$f(\theta_0) = 0 \quad (624)$$

From the second equation it is trivial to see that $B = 0$, so that:

$$f(\theta) = A \sin \theta + C\theta \sin \theta + D\theta \cos \theta$$

$$f'(\theta) = A \cos \theta + C(\sin \theta + \theta \cos \theta) + D(\cos \theta - \theta \sin \theta)$$

From the first one we obtain $D = -A$ so that

$$f(\theta) = A(\sin \theta - \theta \cos \theta) + C\theta \sin \theta$$

$$f'(\theta) = A(\theta \sin \theta) + C(\sin \theta + \theta \cos \theta)$$

The last two boundary conditions yield:

$$0 = A(\sin \theta_0 - \theta_0 \cos \theta_0) + C\theta_0 \sin \theta_0$$

$$-U_0 = A(\theta_0 \sin \theta_0) + C(\sin \theta_0 + \theta_0 \cos \theta_0)$$

or,

$$A = -U_0 \frac{\theta_0 \sin \theta_0}{\theta_0^2 - \sin^2 \theta_0} \quad C = U_0 \frac{\sin \theta_0 - \theta_0 \cos \theta_0}{\theta_0^2 - \sin^2 \theta_0}$$

Finally:

$$(A, B, C, D) = (-\theta_0 \sin \theta_0, 0, \sin \theta_0 - \theta_0 \cos \theta_0, \theta_0 \sin \theta_0) \frac{U_0}{\theta_0^2 - \sin^2 \theta_0}$$

We have

$$\mathbf{e}_r = \cos \theta \mathbf{e}_x + \sin \theta \mathbf{e}_y \quad (625)$$

$$\mathbf{e}_\theta = -\sin \theta \mathbf{e}_x + \cos \theta \mathbf{e}_y \quad (626)$$

so that the velocity field can be expressed in cartesian coordinates:

$$\begin{aligned}\mathbf{v} &= \mathbf{v}_r \mathbf{e}_r + \mathbf{v}_\theta \mathbf{e}_\theta \\ &= \mathbf{v}_r (\cos \theta \mathbf{u}_x + \sin \theta \mathbf{u}_y) + \mathbf{v}_\theta (-\sin \theta \mathbf{u}_x + \cos \theta \mathbf{u}_y) \\ &= (\mathbf{v}_r \cos \theta - \mathbf{v}_\theta \sin \theta) \mathbf{u}_x + (\mathbf{v}_r \sin \theta + \mathbf{v}_\theta \cos \theta) \mathbf{u}_y\end{aligned} \quad (627)$$

8.36 Surface processes

8.36.1 In 1D - simple nonlinear diffusion a la [293]

The tectonic-scale transport equations describe long term changes in topography $h(x, y, t)$ as a result of simultaneous short- and long-range mass transport processes [96, 1092].

The short-range surface processes are represented by cumulative effects of hillslope processes (soil creep, rainsplash, slides) that remove material from uplifted areas down to the valleys. It is then assumed that the horizontal material flux \vec{q}_s is related to local slope $\vec{\nabla}h$ by $\vec{q}_s = -K_s \vec{\nabla}h$ where K_s is the effective diffusivity. Assumption of conservation of mass volume leads to the linear diffusion equation for erosion:

$$\frac{\partial h}{\partial t} = K_s \Delta h$$

This equation can be solved with constant-elevation (fixed h value) boundary conditions simulating local base levels of erosion.

Note that in practice the coefficient K_s might depend on slope and curvature, i.e.

$$\frac{\partial h}{\partial t} = K_s(x, y, h, \vec{\nabla}h) \Delta h$$

Following [786], Burov & Cloetingh use an empirical non linear expression $K_s = k_s(x)(\vec{\nabla}h)^n$.

8.36.2 In 1D - not so simple, a la [32]

The change in surface elevation rate due to surface processes is equal to the divergence of the sediment flux (assuming there is no density difference between the bedrock and sediment and ignoring the effects of compaction):

$$\frac{\partial h}{\partial t} = -\frac{\partial q_s}{\partial x}$$

where h is the topography, t is the time, q_s represents the sediment flux, and x is the horizontal coordinate.

The next step consists in a formulation for the sediment flux. Still following [32], in the subaerial environment, it is possible to define the sediment transport flux q_s in terms of the water flux q_w as

$$q_s = -(K + cq_w^n) \frac{\partial h}{\partial x}$$

where K is the slope diffusivity, c is the transport coefficient, and $n \geq 1$ is the power law that defines the type of relationship between the sediment transport and the water flux (Simpson & Schlunegger, 2003; Smith & Bretherton, 1972).

[get these papers](#)

This model accounts for hillslope diffusion processes where the topography will tend to a dispersive diffusion (Culling, 1960) and fluvial transport processes that result in concentrative diffusion due to water run off (Graf, 1984). For a simple parameterization we choose a linear relationship between sediment transport and water flux ($n = 1$).

The water flux can be related to the water discharge/effective rainfall α as

$$\frac{\partial}{\partial x}(\vec{n}q_w) = -\alpha$$

where \vec{n} is a unit vector directed down the surface gradient (Smith & Bretherton, 1972). By assuming a constant α and integrating equation (12) over the surface in the downstream direction, we obtain

$$q_w = \alpha x_d$$

where x_d is the downstream distance from the drainage divide. By substituting equations (11) and (13) into (10) we obtain the 1-D sediment mass conservation equation for combined hillslope and discharge-dependent fluvial transport

$$\frac{\partial h}{\partial t} = \frac{\partial}{\partial x} \left((K + k\alpha x_d) \frac{\partial h}{\partial x} \right)$$

where the downstream distance x_d is calculated at each time step as the distance from the topographic highs to the valley floors. Because q_w is dependent on the length of the drainage, the model mimics 1-D landscapes similar to river profiles in which fluvial processes are dominant.

8.37 Geometric multigrid

The following is mostly borrowed from the Wikipedia page on multigrid methods⁵⁵.

There are many types of (geometric) multigrid algorithms, but the common features are that a hierarchy of grids is considered. The important steps are:

- *Smoothing*: reducing high frequency errors, for example using a few iterations of the Gauss-Seidel method.
- *Residual Computation*: computing residual error after the smoothing operation(s).
- *Restriction*: downsampling the residual error to a coarser grid.
- *Interpolation or prolongation*: interpolating a correction computed on a coarser grid into a finer grid.
- *Correction*: Adding prolonged coarser grid solution onto the finer grid.

There are many choices of multigrid methods with varying trade-offs between speed of solving a single iteration and the rate of convergence with said iteration. The 3 main types are V-Cycle, F-Cycle, and W-Cycle.

Any geometric multigrid cycle iteration is performed on a hierarchy of grids and hence it can be coded using recursion. Since the function calls itself with smaller sized (coarser) parameters, the coarsest grid is where the recursion stops.

Note that the ratio of the number of nodes between two consecutive levels has to be constant between all the levels. Often powers of 2 are used (especially if the grids are based on quad/octrees) but it is not a requirement.

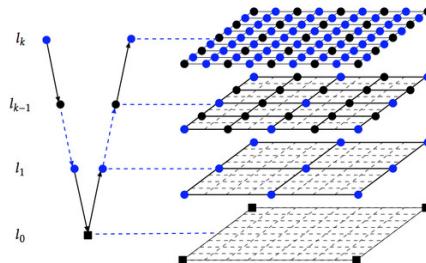


Image from <http://web.utk.edu/~wfeng1/research.html>

What follows is a pseudo-code example of a recursive V-Cycle Multigrid for solving the Poisson equation ($\nabla^2\phi = f$) on a uniform grid of spacing h :

```
function phi = V_Cycle(phi,f,h)
% Pre-Smoothing
phi = smoothing(phi,f,h);
% Compute Residual Errors
r = residual(phi,f,h);
% Restriction
rhs = restriction(r);
eps = zeros(size(rhs));
% stop recursion at smallest grid size
if smallest_grid_size_is_achieved
    eps = smoothing(eps,rhs,2*h);
else
    eps = V_Cycle(eps,rhs,2*h);
end
% Prolongation and Correction
```

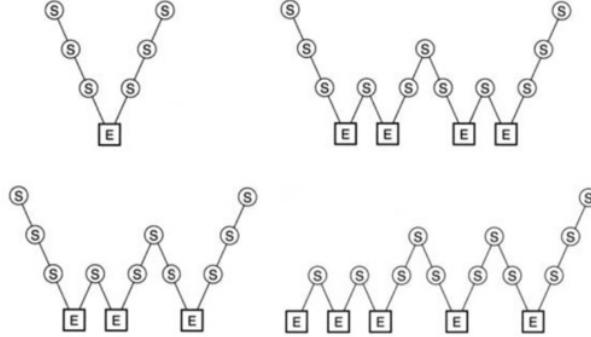
⁵⁵https://en.wikipedia.org/wiki/Multigrid_method

```

phi = phi + prolongation(eps);
% Post-Smoothing
phi = smoothing(phi,f,h);
end

```

A multigrid method with an intentionally reduced tolerance can be used as an efficient preconditioner for an external iterative solver. The solution may still be obtained in $\mathcal{O}(N)$ time as well as in the case where the multigrid method is used as a solver. Multigrid preconditioning is used in practice even for linear systems, typically with one cycle per iteration.



Taken from [967]: Different types of multigrid cycle with four grid levels: (top left) V-cycle, (top right) W-cycle, (bottom left) F-cycle and (bottom right) full multigrid. S denotes smoothing while E denotes exact coarse-grid solution.

Check Kaus BEcker syllabus!

Literature: [1482, 1205, 967, 722, 1289, 1222, 1796, 1372, 1973, 1975] Book [236]

8.38 Computing depth

In the case of a perfectly rectangular, cylindrical or spherical domain, computing the depth of any given point inside the domain is trivial. However, when the free surface becomes somewhat distorted, the concept of depth needs to be refined. What follows is an attempt at bringing clarity as to how to compute depth in all cases.

The depth $d(\mathbf{r})$ satisfies the equation:

$$\frac{\mathbf{g}}{|\mathbf{g}|} \cdot \nabla d = 1$$

with $d = 0$ at the surface.

This is a form of steady-state advection equation (the time derivative is zero, there is no diffusion, nor any source term).

Given the boundary conditions, one could solve this equation over the whole domain.

Note that in the case of a cartesian box, $\mathbf{g} = -g\mathbf{u}_z$, we need to solve

$$-\frac{\partial}{\partial z} d = 1$$

For a flat top surface at $d(z = L_z) = 0$ so that in the end

$$d(z) = L_z - z$$

8.39 Imposing boundary conditions

Let us consider a quadrilateral element with one degree of freedom per node and let us assume that we are solving the temperature equation. The local matrix and right-hand side vector are given by

$$A_{el}(4 \times 4) \quad \text{and} \quad B_{el}(4)$$

Let us assume that we want to impose $\tilde{T} = 10$ on the third node (local coordinates numbering). For instance, having built A_{el} and B_{el} , the system looks like :

$$\begin{pmatrix} 3 & 1 & 6 & 9 \\ 5 & 2 & 2 & 8 \\ 7 & 4 & 11 & 2 \\ 9 & 6 & 4 & 3 \end{pmatrix} \begin{pmatrix} T_1 \\ T_2 \\ T_3 \\ T_4 \end{pmatrix} = \begin{pmatrix} 4 \\ 3 \\ 1 \\ 2 \end{pmatrix}$$

which can be rewritten

$$3T_1 + T_2 + 6T_3 + 9T_4 = 4$$

$$5T_1 + 2T_2 + 2T_3 + 8T_4 = 3$$

$$7T_1 + 4T_2 + 11T_3 + 2T_4 = 1$$

$$9T_1 + 6T_2 + 4T_3 + 3T_4 = 2$$

or,

$$3T_1 + T_2 + +9T_4 = 4 - 6T_3$$

$$5T_1 + 2T_2 + +8T_4 = 3 - 2T_3$$

$$7T_1 + 4T_2 + 11T_3 + 2T_4 = 1$$

$$9T_1 + 6T_2 + +3T_4 = 2 - 4T_3$$

- Technique 1: Replace the hereabove system by

$$\begin{pmatrix} 3 & 1 & 6 & 9 \\ 5 & 2 & 2 & 8 \\ 7 & 4 & 11 + 10^{12} & 2 \\ 9 & 6 & 4 & 3 \end{pmatrix} \begin{pmatrix} T_1 \\ T_2 \\ T_3 \\ T_4 \end{pmatrix} = \begin{pmatrix} 4 \\ 3 \\ \tilde{T} \times (11 + 10^{12}) \\ 2 \end{pmatrix}$$

- Technique 2: One can choose not to solve for T_3 anymore, i.e. not to consider it as a degree of freedom and therefore write:

$$3T_1 + T_2 + 9T_4 = 4 - 6T_3$$

$$5T_1 + 2T_2 + 8T_4 = 3 - 2T_3$$

$$9T_1 + 6T_2 + 3T_4 = 2 - 4T_3$$

- Technique 3: Since we want to impose $T_3 = 10$, then we can write

$$3T_1 + T_2 + +9T_4 = 4 - 6T_3$$

$$5T_1 + 2T_2 + +8T_4 = 3 - 2T_3$$

$$0 + 0 + T_3 + 0 = 10$$

$$9T_1 + 6T_2 + +3T_4 = 2 - 4T_3$$

and in matrix form :

$$\begin{pmatrix} 3 & 1 & 0 & 9 \\ 5 & 2 & 0 & 8 \\ 0 & 0 & 1 & 0 \\ 9 & 6 & 0 & 3 \end{pmatrix} \begin{pmatrix} T_1 \\ T_2 \\ T_3 \\ T_4 \end{pmatrix} = \begin{pmatrix} 4 - A_{13}T_3 \\ 3 - A_{23}T_3 \\ 10 \\ 2 - A_{43}T_3 \end{pmatrix}$$

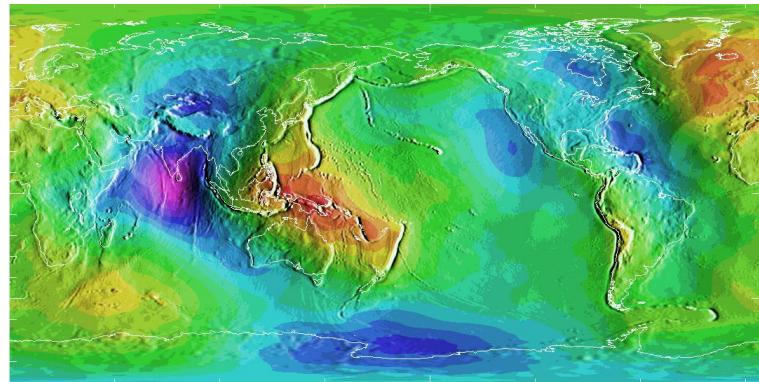
The first technique is not a good idea in practice as it introduces very large values and will likely derail the solver. The second option is somewhat difficult to implement as it means that elemental matrix and rhs sizes will change from element to element and it therefore requires more book-keeping. The third technique is the one adopted throughout this document.

8.40 The Geoid

8.40.1 What is the geoid?

The geoid is usually defined in two ways:

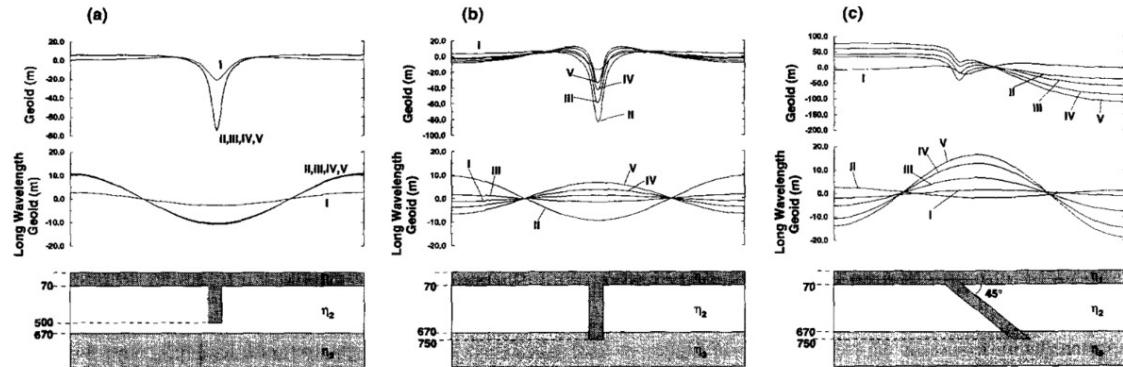
- Mean sea level (easy to define in the oceans, but harder on land)
- A gravitational equipotential surface. This means that everywhere at sea level experiences the same value of gravity potential, so there is no tendency for water to flow downhill since all points in the vicinity have the same value of gravity potential, pointed toward the center of the earth.



Data Max value: 85.4 meters, east of New Guinea. Data Min value:-107.0 meters, south of India. This image shows 15'x15' geoid undulations covering the planet Earth from the NIMA/GSFC WGS-84 EGM96 15' Geoid Height File. The undulations refer to the differences from the WGS-84(G873) reference ellipsoid. Map and description from National Geodetic Survey.

8.40.2 How to compute it?

8.40.3 Interesting modelling



Idealized 2D slab calculations for each viscosity model: geoid and geoid filtered to pass only the longest wavelengths (~ 4000 km). (a) Cold slab extends to 500 km depth in the upper mantle, (b) Slab extends to 750 km so that it is partly supported by the high viscosity lower mantle at 670 km. (c) Slab tilted at 45° to the vertical extending to the top of the lower mantle. Taken from [1350]

8.41 The Lyapunov time/exponent, mixing stirring

Simply put, the Lyapunov time is the characteristic timescale on which a dynamical system is chaotic. It is defined as the inverse of a system's largest Lyapunov exponent.

The Lyapunov time mirrors the limits of the predictability of the system. By convention, it is defined as the time for the distance between nearby trajectories of the system to increase by a factor of e . However, measures in terms of 2-foldings and 10-foldings are sometimes found, since they correspond to the loss of one bit of information or one digit of precision respectively.

The Lyapunov exponent or Lyapunov characteristic exponent of a dynamical system is a quantity that characterizes the rate of separation of infinitesimally close trajectories. Quantitatively, two trajectories in phase space with initial separation $\delta\mathbf{Z}_0$ diverge (provided that the divergence can be treated within the linearized approximation) at a rate given by

$$|\delta\mathbf{Z}(t)| \approx e^{\lambda t} |\delta\mathbf{Z}_0|$$

where λ is the Lyapunov exponent.

Measuring the Lyapunov exponent or time (or related quantities) is relevant in the context of mantle stirring. On the one hand it is argued that the mantle is convecting and very efficient at mixing resulting in a somewhat homogenous composition. On the other hand, there are modeling studies that suggest that whole-mantle convection can preserve heterogeneity in the presence of well-mixed mantle.

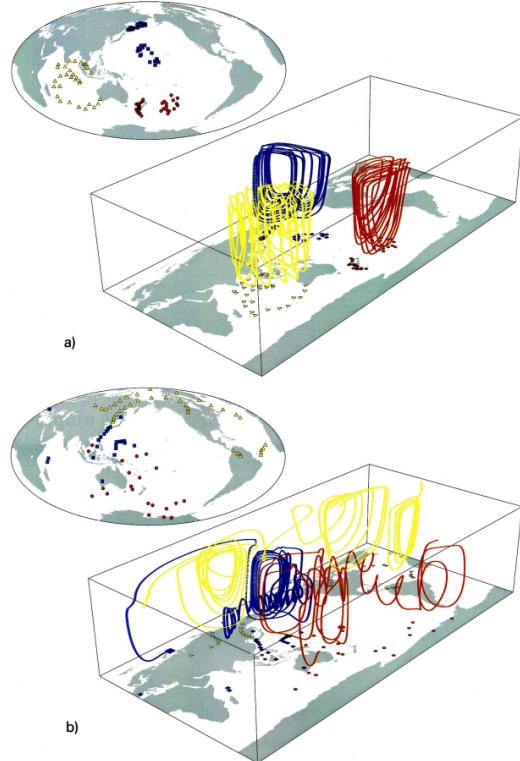
Mixing takes place by the repeated stretching and folding of interfaces. A measure of the mixing efficiency is the time evolution of the area of the mixing surface. Maximum efficiency of mixing is reached with turbulent mixing behavior where one can formally show whether mixing is laminar or turbulent by evaluating the Luyaponov exponents σ . These are of the form:

$$\sigma = \lim_{t \rightarrow \infty} \lim_{X \rightarrow 0} \left[\frac{1}{t} \ln \left(\frac{X(t)}{X(t=0)} \right) \right]$$

where $X(t)$ is the length of this segment at time t . Non-zero Luyaponov exponents indicate that stretching is exponential and the larger the exponent, the more efficient mixing is. However, the limits in the above equation are difficult to evaluate and the interpretation of the 'finite-time' Luyaponov exponent, where both limits are truncated, is difficult to formalize.

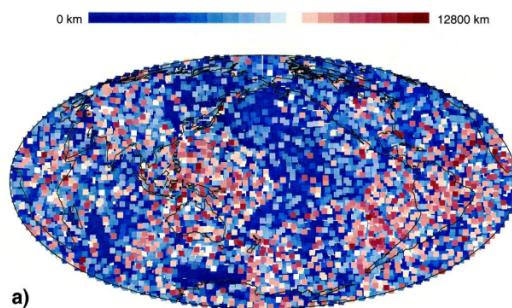
Two approaches are taken in the literature when it comes to studying mixing/stirring and/or measuring Lyapunov quantities::

- using marker advection: in van Keken and Zhong [1843] the authors use a steady state velocity pattern obtained for a model of present-day mantle convection. The velocity model is based on the solution of the Stokes equations in a 3D spherical model with variable rheology. To study mixing, they release particles in the velocity model and follow these by numerical integration.



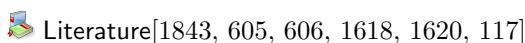
a) The three particles in this plot were selected for their relatively regular pattern. b) Three other particles that traverse a large portion of the model. These particles feel the strong toroidal motion and their paths form corkscrew-like patterns. They indicate that certain parts of the model can exhibit strong mixing. Taken from [1843].

Rather than calculating the exponents explicitly, the authors use an approximation to the finite-time, finite-length Luyapponov exponent by evaluating the distance between two points that are closely spaced at time $t = 0$. For this they compute the advection of a large number of 10 km long line segments that were originally at 1500 km depth. The length of these segments is approximated by the distance between the endpoints and the results are summarized in the following figure:



Length of the line segment after 4 billion years. Approximately 14,000 line segments were released with regular spacing at 1500 km depth. The length of the segment is indicated by the colored symbols that are plotted at the initial position. The results indicate that there is a strong diversity in mixing behavior. In some regions (north Pacific, parts under the Indian/Australian plate) stretching is very limited, indicating laminar and consequently inefficient mixing. Regions that are under strong toroidal surface motion (western Pacific, Nazca and South America) show very efficient stretching of up to the maximum length of the diameter of the Earth. Taken from [1843].

- twin experiments [117]



Literature[1843, 605, 606, 1618, 1620, 117]

9 Gravity and co

WORK in PROGRESS. DUH.

We start from the Poisson equation for the gravity potential:

$$\Delta U = 4\pi\rho\mathcal{G} \quad (628)$$

As a consequence, inside a domain where $\rho = 0$, the equation becomes $\Delta U = 0$.

Let us assume that the spherical coordinates are appropriate for the problem at hand, and that the potential can be decomposed as follows:

$$U(r, \theta, \phi) = U_r(r)U_\perp(\theta, \phi)$$

The full Laplacian operator in spherical coordinates is given by⁵⁶:

$$\Delta U = \underbrace{\frac{1}{r^2}\frac{\partial}{\partial r}\left(r^2\frac{\partial U}{\partial r}\right)}_{\Delta_r} + \underbrace{\frac{1}{r^2\sin\theta}\frac{\partial}{\partial\theta}\left(\sin\theta\frac{\partial U}{\partial\theta}\right)}_{\Delta_\perp} + \frac{1}{r^2\sin^2\theta}\frac{\partial^2 U}{\partial\phi^2}$$

we then have:

$$(\Delta_r + \Delta_\perp)(U_r U_\perp) = 0$$

i.e.,

$$U_\perp \Delta_r U_r + U_r \Delta_\perp U_\perp = 0$$

Assuming $U_\perp = \sum_l \sum_m U_{lm} Y_{lm}$, knowing that spherical harmonics functions verify

$$r^2 \Delta_\perp Y_l^m(\theta, \phi) = -l(l+1)Y_l^m(\theta, \phi)$$

and assuming for now that the problem at hand is 1st degree ($l=1$), then

$$\Delta_\perp Y_l^m(\theta, \phi) = -\frac{2}{r^2} Y_l^m(\theta, \phi)$$

and then

$$\Delta_r U_r - U_r \frac{2}{r^2} = 0$$

make a link with my 2018 paper.

⁵⁶https://en.wikipedia.org/wiki/Laplace_operator

In spherical coordinates, the Laplacian is given by

$$\Delta = \frac{1}{r^2} \frac{\partial}{\partial r} \left(r^2 \frac{\partial}{\partial r} \right) + \frac{1}{r^2 \sin^2 \theta} \frac{\partial}{\partial \theta} \left(\sin \theta \frac{\partial}{\partial \theta} \right) + \frac{1}{r^2 \sin^2 \theta} \frac{\partial^2}{\partial \phi^2}$$

We wish to solve Laplace's equation $\Delta T(r, \theta, \phi) = 0$ using the method of separation of variables:

$$T(r, \theta, \phi) = R(r)\Theta(\theta)\Phi(\phi)$$

We can insert this decomposition into the Laplace equation and multiply it by $r^2/R\Theta\Phi$ to obtain

$$\frac{1}{R} \frac{d}{dr} \left(r^2 \frac{dR}{dr} \right) + \frac{1}{\Theta} \frac{1}{\sin \theta} \frac{d}{d\theta} \left(\sin \theta \frac{d\Theta}{d\theta} \right) + \frac{1}{\Phi} \frac{1}{\sin^2 \theta} \frac{d^2\Phi}{d\phi^2} = 0$$

For reasons that will become clear later, the separation constant is taken to be $-m^2$:

$$\frac{1}{\Phi} \frac{d^2\Phi}{d\phi^2} = -m^2 \quad (629)$$

and

$$-\frac{\sin \theta}{\Theta} \frac{d}{d\theta} \left(\sin \theta \frac{d\Theta}{d\theta} \right) - \frac{\sin^2 \theta}{R} \frac{d}{dr} \left(r^2 \frac{dR}{dr} \right) = -m^2 \quad (630)$$

The first equation yields

$$\Phi(\phi) = \begin{cases} e^{im\phi} \\ e^{-im\phi} \end{cases} \quad \text{for } m = 0, 1, 2, 3, \dots$$

Note that m must be an integer since ϕ is a periodic variable and $\Phi(\phi + 2\pi) = \Phi(\phi)$. In the case of $m = 0$, the general solution is $\Phi(\phi) = a\phi + b$, but we must choose $a = 0$ to be consistent with $\Phi(\phi + 2\pi) = \Phi(\phi)$. Hence in the case of $m = 0$, only one solution is allowed.

Equation 630 can now be recast in the following form:

$$\frac{1}{R} \frac{d}{dr} \left(r^2 \frac{dR}{dr} \right) = -\frac{1}{\Theta} \frac{1}{\sin \theta} \frac{d}{d\theta} \left(\sin \theta \frac{d\Theta}{d\theta} \right) + \frac{m^2}{\sin^2 \theta} = l(l+1) \quad (631)$$

where the separation variable at this step is denoted by $l(l+1)$ for reasons that will shortly become clear. The resulting radial equation is

$$\frac{1}{R} \frac{d}{dr} \left(r^2 \frac{dR}{dr} \right) = l(l+1)$$

or,

$$r^2 \frac{d^2R}{dr^2} + 2r \frac{dR}{dr} - l(l+1)R = 0$$

The solution is of the form $R = r^s$. To determine the exponent s , we insert this solution back into the above ODE. The end result is

$$s(s+1) = l(l+1) \quad \Rightarrow \quad s = l \text{ or } s = -l-1$$

or,

$$R(r) = \begin{cases} r^l \\ r^{-(l+1)} \end{cases}$$

Eq. (631) also yields:

$$\frac{1}{\sin \theta} \frac{d}{d\theta} \left(\sin \theta \frac{d\Theta}{d\theta} \right) + \left[l(l+1) - \frac{m^2}{\sin^2 \theta} \right] \Theta = 0$$

One can then carry out the following change of variables $x = \cos \theta$ and $y = \Theta(\theta)$ so that the above equation reduces to:

$$(1-x^2) \frac{d^2y}{dx^2} - 2x \frac{dy}{dx} + \left[l(l+1) - \frac{m^2}{\sin^2 \theta} \right] y = 0$$

This equation is the differential equation for associated Legendre polynomials⁵⁷. We then have

$$y = P_l^m(x) \quad \text{for } l = 0, 1, 2, 3, \dots \quad \text{and } m = -l, -l+1, \dots, 0, \dots, l = 1, l$$

and

$$P_l^m(x) = \frac{(-1)^m}{2^l l!} (1-x^2)^{m/2} \frac{d^{l+m}}{dx^{l+m}} (x^2 - 1)^l$$

with $m \geq 0$ and $l \geq 0$.

In our case the differential equation for the associated Legendre polynomials, given above, depends on m^2 and is therefore not sensitive to the sign of m . Consequently, $P_l^m(x)$ and $P_l^{-m}(x)$ must be equivalent solutions and hence proportional to each other, and one can show that

$$P_l^{-m}(\cos \theta) = (-1)^m \frac{(l-m)!}{(l+m)!} P_l^m(\cos \theta)$$

Combining all the results obtained above, we have found that the general solution to Laplaces equation is of the form

$$T(r, \theta, \phi) = \left\{ \begin{array}{c} r^l \\ r^{-(l+1)} \end{array} \right\} P_l^m(\cos \theta) \left\{ \begin{array}{c} e^{im\phi} \\ e^{-im\phi} \end{array} \right\}$$

where $l = 0, 1, 2, 3, \dots$ and $m = -l, -l+1, \dots, l-1, l$.

When solving the Laplaces equation in spherical coordinates, it is traditional to introduce the spherical harmonics, $Y_l^m(\theta, \phi)$:

$$Y_l^m(\theta, \phi) = (-1)^m \sqrt{\frac{2l+1}{4\pi} \frac{(l-m)!}{(l+m)!}} P_l^m(\cos \theta) e^{im\phi}$$

⁵⁷https://en.wikipedia.org/wiki/Associated_Legendre_polynomials

Stone 01: simple analytical solution (D&H)

This fieldstone was developed in collaboration with Job Mos.

This benchmark is taken from [533] and is described fully in section 8.6. In order to illustrate the behavior of selected mixed finite elements in the solution of stationary Stokes flow, we consider a two-dimensional problem in the square domain $\Omega = [0, 1] \times [0, 1]$, which possesses a closed-form analytical solution. The problem consists of determining the velocity field $\mathbf{v} = (u, v)$ and the pressure p such that

$$\eta \Delta \vec{v} - \vec{\nabla} p + \vec{b} = \vec{0} \quad \text{in } \Omega$$

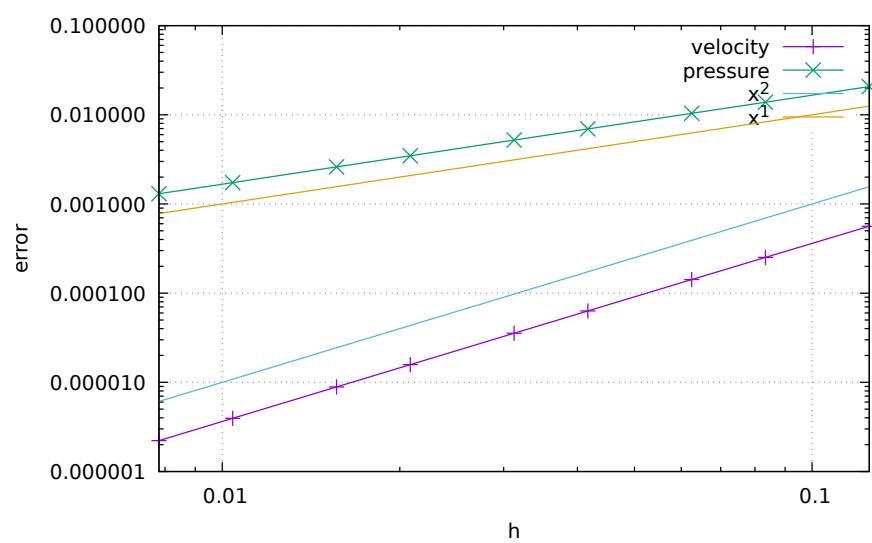
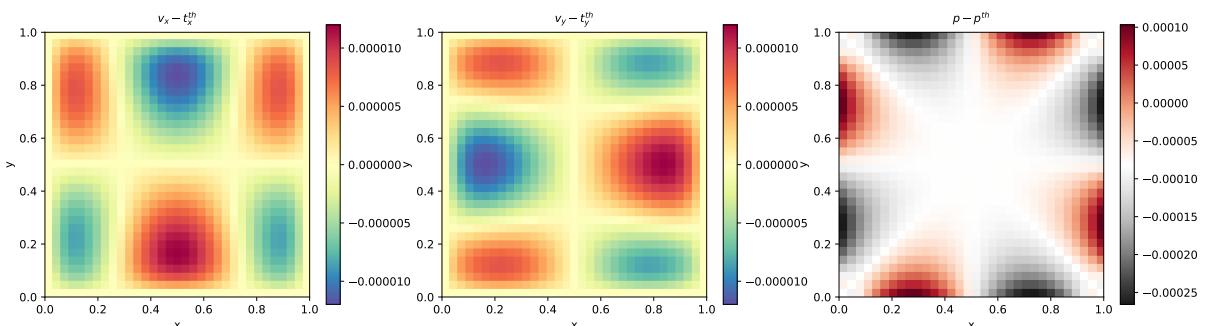
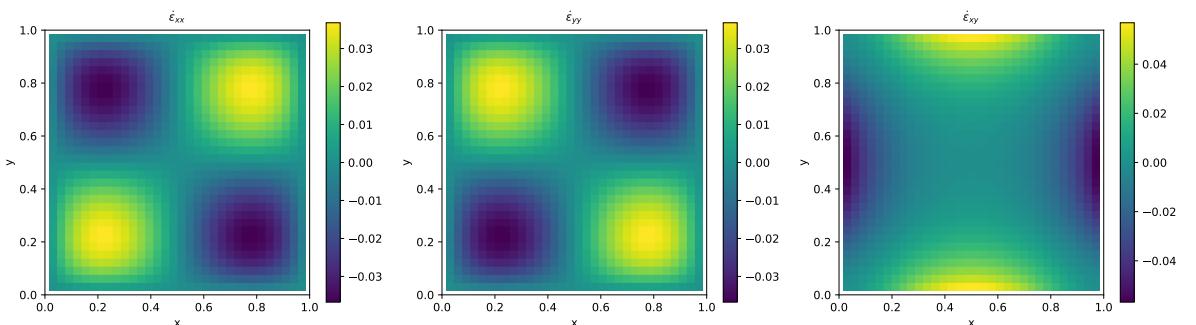
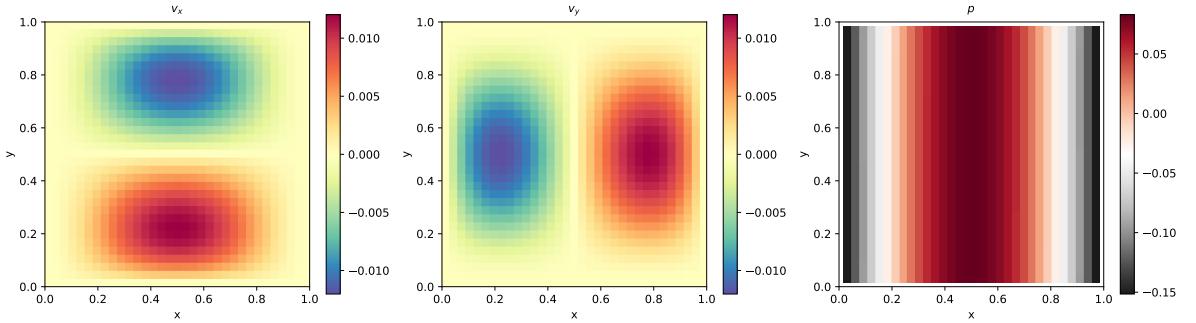
$$\vec{\nabla} \cdot \vec{v} = 0 \quad \text{in } \Omega$$

$$\vec{v} = \vec{0} \quad \text{on } \Gamma_D$$

where the fluid viscosity is taken as $\eta = 1$.

features

- $Q_1 \times P_0$ element
- incompressible flow
- penalty formulation
- Dirichlet boundary conditions (no-slip)
- direct solver
- isothermal
- isoviscous
- analytical solution



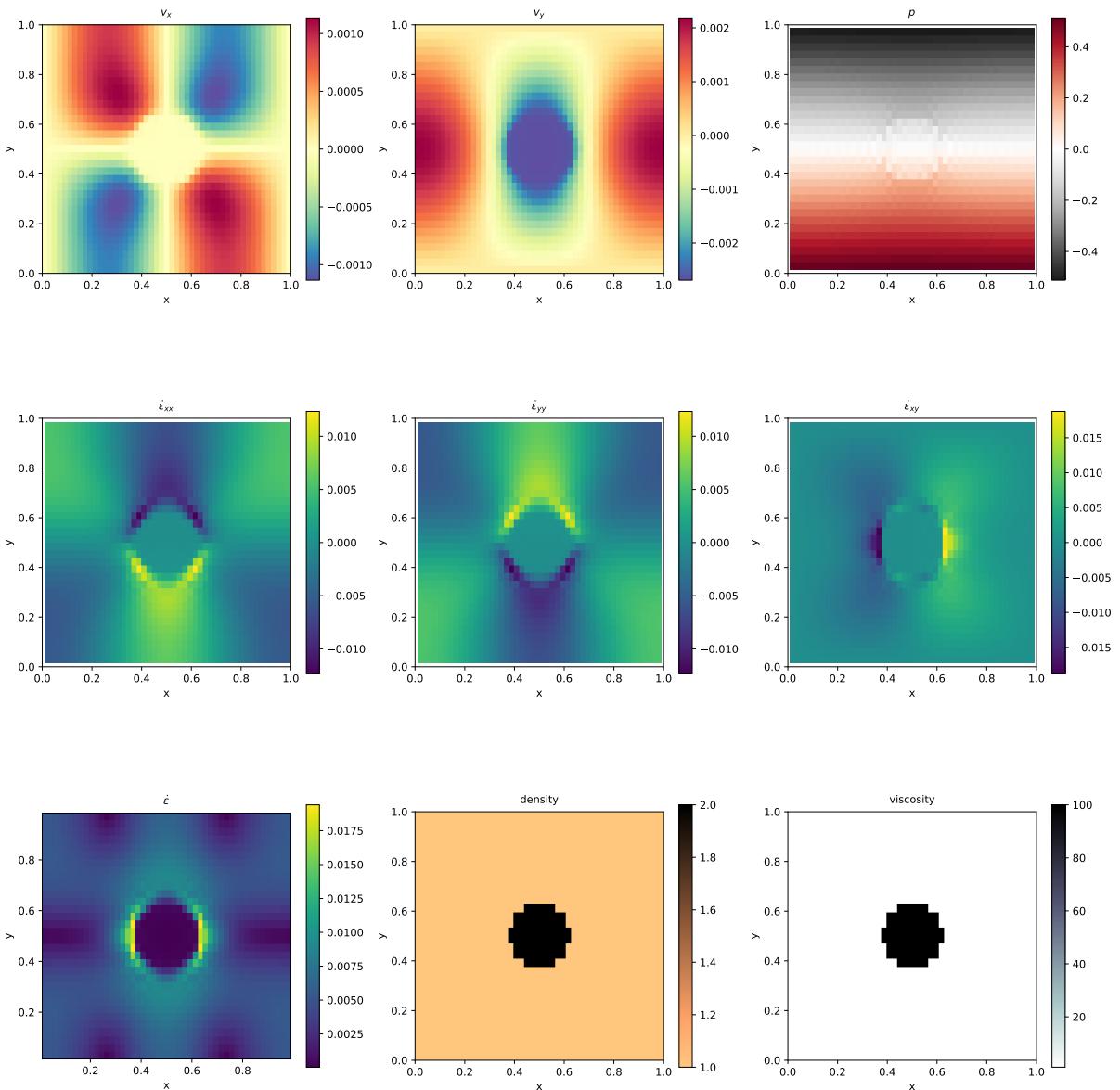
Quadratic convergence for velocity error, linear convergence for pressure error, as expected.

Stone 02: Stokes sphere

Viscosity and density directly computed at the quadrature points.

features

- $Q_1 \times P_0$ element
- incompressible flow
- penalty formulation
- Dirichlet boundary conditions (free-slip)
- isothermal
- non-isoviscous
- buoyancy-driven flow
- Stokes sphere



Stone 03: Convection in a 2D box

This benchmark deals with the 2-D thermal convection of a fluid of infinite Prandtl number in a rectangular closed cell. In what follows, I carry out the case 1a, 1b, and 1c experiments as shown in [175]: steady convection with constant viscosity in a square box.

The temperature is fixed to zero on top and to ΔT at the bottom, with reflecting symmetry at the sidewalls (i.e. $\partial_x T = 0$) and there are no internal heat sources. Free-slip conditions are implemented on all boundaries.

The Rayleigh number is given by

$$Ra = \frac{\alpha g_y \Delta T h^3}{\kappa \nu} = \frac{\alpha g_y \Delta T h^3 \rho^2 C_p}{k \mu} \quad (632)$$

In what follows, I use the following parameter values: $L_x = L_y = 1, \rho_0 = c_P = k = \mu = 1, T_0 = 0, \alpha = 10^{-2}, g = 10^2 Ra$ and I run the model with $Ra = 10^4, 10^5$ and 10^6 .

The initial temperature field is given by

$$T(x, y) = (1 - y) - 0.01 \cos(\pi x) \sin(\pi y) \quad (633)$$

The perturbation in the initial temperature fields leads to a perturbation of the density field and sets the fluid in motion.

Depending on the initial Rayleigh number, the system ultimately reaches a steady state after some time.

The Nusselt number (i.e. the mean surface temperature gradient over mean bottom temperature) is computed as follows [175]:

$$Nu = L_y \frac{\int \frac{\partial T}{\partial y} (y = L_y) dx}{\int T(y = 0) dx} \quad (634)$$

Note that in our case the denominator is equal to 1 since $L_x = 1$ and the temperature at the bottom is prescribed to be 1.

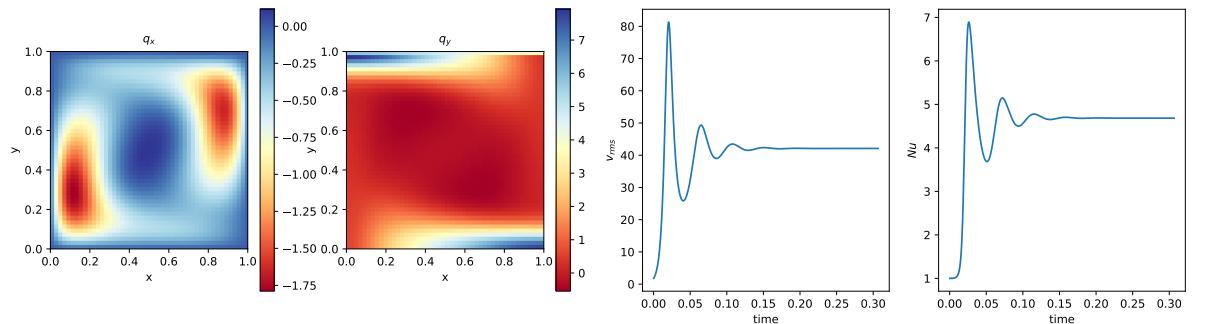
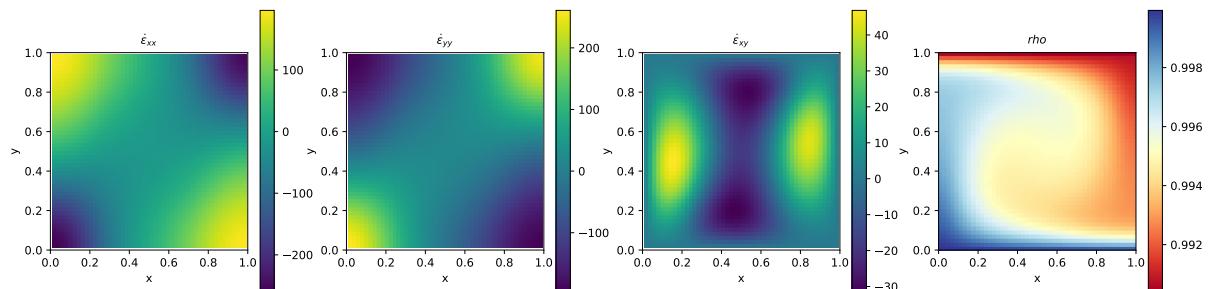
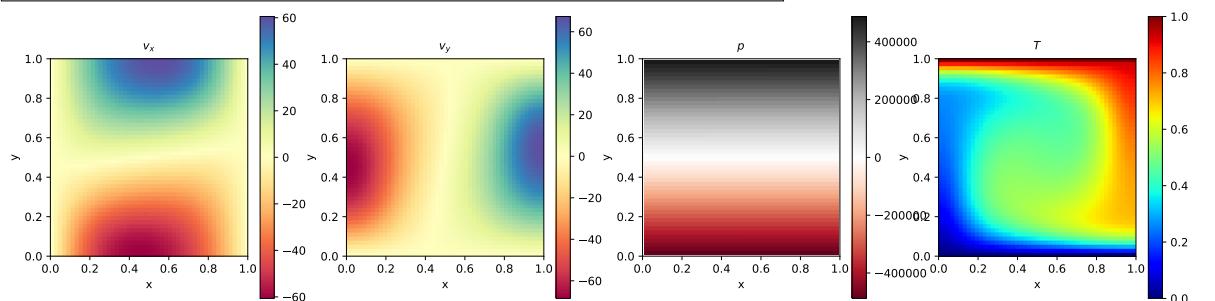
Finally, the steady state root mean square velocity and Nusselt number measurements are indicated in Table ?? alongside those of [175] and [1712]. (Note that this benchmark was also carried out and published in other publications [1794, 12, 722, 488, 1169] but since they did not provide a complete set of measurement values, they are not included in the table.)

		Blankenbach et al	Tackley [1712]
$Ra = 10^4$	V_{rms}	42.864947 ± 0.000020	42.775
	Nu	4.884409 ± 0.000010	4.878
$Ra = 10^5$	V_{rms}	193.21454 ± 0.00010	193.11
	Nu	10.534095 ± 0.000010	10.531
$Ra = 10^6$	V_{rms}	833.98977 ± 0.00020	833.55
	Nu	21.972465 ± 0.000020	21.998

Steady state Nusselt number Nu and V_{rms} measurements as reported in the literature.

features

- $Q_1 \times P_0$ element
- incompressible flow
- penalty formulation
- Dirichlet boundary conditions (free-slip)
- Boussinesq approximation
- direct solver
- non-isothermal
- buoyancy-driven flow
- isoviscous
- CFL-condition



ToDo:

implement steady state criterion

reach steady state

do $\text{Ra}=1\text{e}4, 1\text{e}5, 1\text{e}6$

plot against blankenbach paper and aspect

look at critical Ra number

This benchmark has been carried out in many scientific papers [151].

Stone 04: The lid driven cavity

The lid driven cavity is a famous Computational Fluid Dynamics test case [1039, 750, 1447, 195, 251, 816, 1660] and has been studied in countless publications with a wealth of numerical techniques (see [582] for a succinct review) and also in the laboratory [1109].

It models a plane flow of an isothermal isoviscous fluid in a rectangular (usually square) lid-driven cavity. The boundary conditions are indicated in the Fig. ??a. The gravity is set to zero.

the lid driven cavity problem (ldc=0)

In the standard case, the upper side of the cavity moves in its own plane at unit speed, while the other sides are fixed. This thereby introduces a discontinuity in the boundary conditions at the two upper corners of the cavity and yields an uncertainty as to which boundary (side or top) the corner points belong to. In this version of the code the top corner nodes are considered to be part of the lid. If these are excluded the recovered pressure showcases an extremely large checkboard pattern.

This benchmark is usually discussed in the context of low to very high Reynolds number with the full Navier-Stokes equations being solved (with the noticeable exception of [1622, 1623, 350, 566] which focus on the Stokes equation). In the case of the incompressible Stokes flow, the absence of inertia renders this problem instantaneous so that only one time step is needed.

the lid driven cavity problem - regularisation I (ldc=1)

We avoid the top corner nodes issue altogether by prescribing the horizontal velocity of the lid as follows:

$$u(x) = x^2(1-x)^2. \quad (635)$$

In this case the velocity and its first derivative is continuous at the corners. This is the so-called regularised lid-driven cavity problem [1477].

the lid driven cavity problem - regularisation II (ldc=2)

Another regularisation was presented in [507]. Also in Appendix D.4 of [998]. Here, a regularized lid driven cavity is studied which is consistent in the sense that $\nabla \cdot \mathbf{v} = 0$ holds also at the corners of the domain. There are no-slip conditions at the boundaries $x = 0$, $x = 1$, and $y = 0$.

The velocity at $y = 1$ is given by

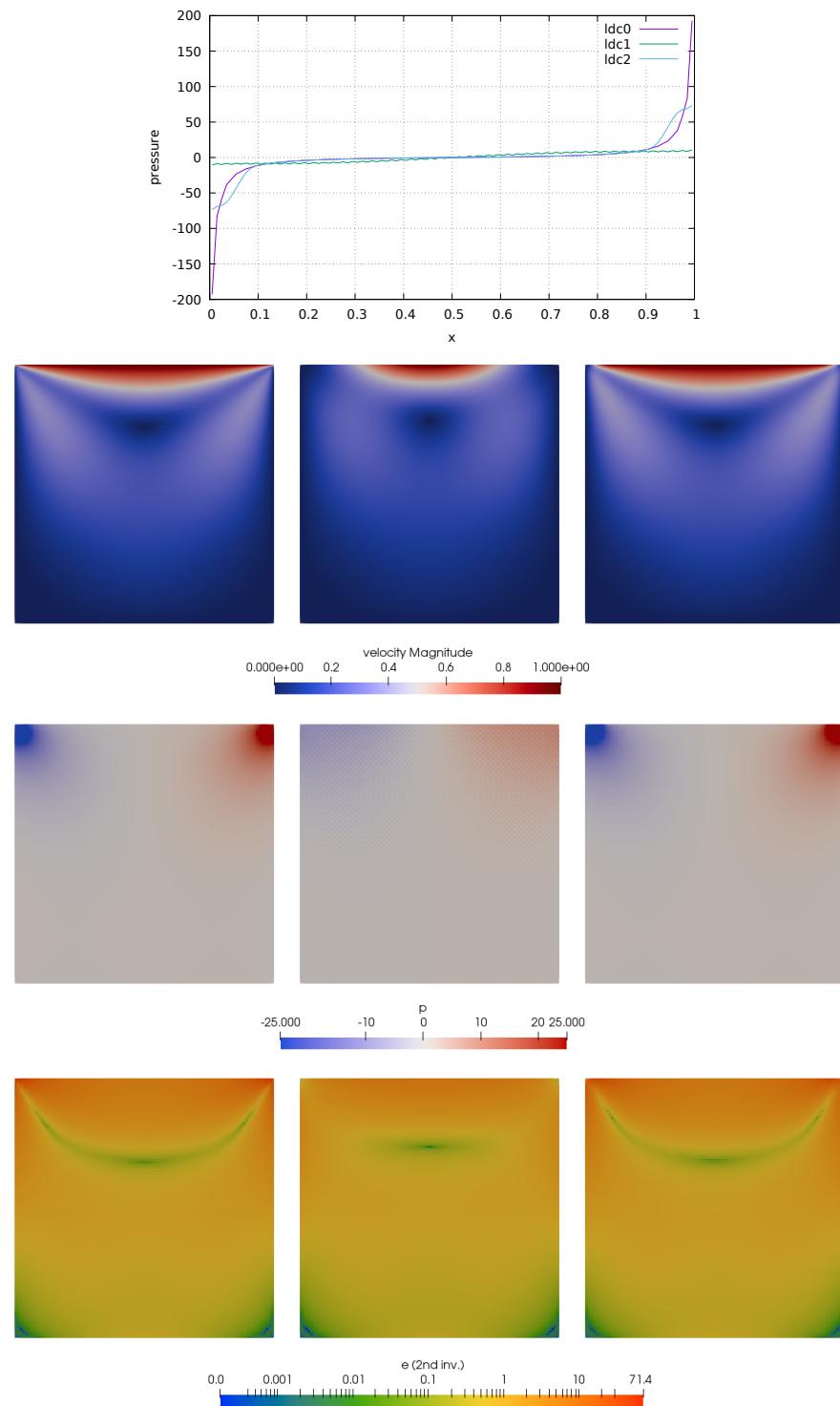
$$\begin{aligned} u(x) &= 1 - \frac{1}{4} \left(1 - \cos\left(\frac{x_1 - x}{x_1}\pi\right) \right)^2 & x \in [0, x_1] \\ u(x) &= 1 & x \in [x_1, 1 - x_1] \\ u(x) &= 1 - \frac{1}{4} \left(1 - \cos\left(\frac{x - (1 - x_1)}{x_1}\pi\right) \right)^2 & x \in [1 - x_1, 1] \end{aligned} \quad (636)$$

Results are obtained with $x_1 = 0.1$.

features

- $Q_1 \times P_0$ element
- incompressible flow
- penalty formulation
- isothermal
- isoviscous

A 100x100 element grid is used. No-slip boundary conditions are prescribed on sides and bottom. A zero vertical velocity is prescribed at the top and the exact form of the prescribed horizontal velocity is controlled by the ldc parameter.



Stone 05: SolCx benchmark

The SolCx benchmark is intended to test the accuracy of the solution to a problem that has a large jump in the viscosity along a line through the domain. Such situations are common in geophysics: for example, the viscosity in a cold, subducting slab is much larger than in the surrounding, relatively hot mantle material.

The SolCx benchmark computes the Stokes flow field of a fluid driven by spatial density variations, subject to a spatially variable viscosity. Specifically, the domain is $\Omega = [0, 1]^2$, gravity is $\mathbf{g} = (0, -1)^T$ and the density is given by

$$\rho(x, y) = \sin(\pi y) \cos(\pi x) \quad (637)$$

Boundary conditions are free slip on all of the sides of the domain and the temperature plays no role in this benchmark. The viscosity is prescribed as follows:

$$\mu(x, y) = \begin{cases} 1 & \text{for } x < 0.5 \\ 10^6 & \text{for } x > 0.5 \end{cases} \quad (638)$$

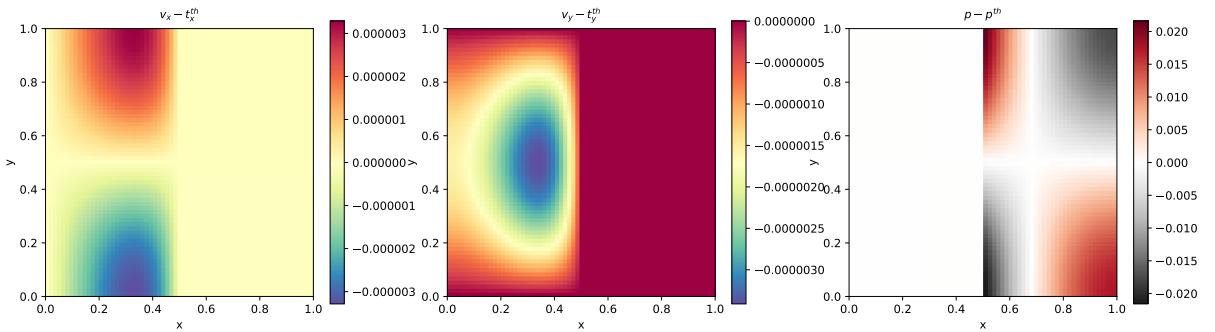
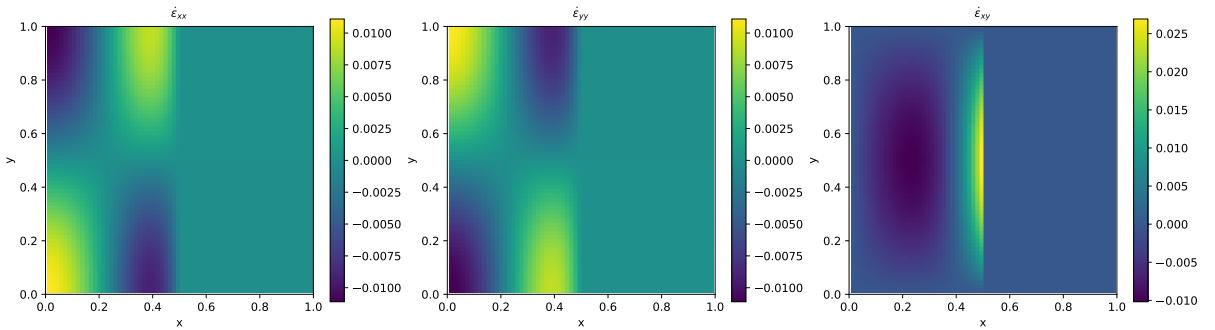
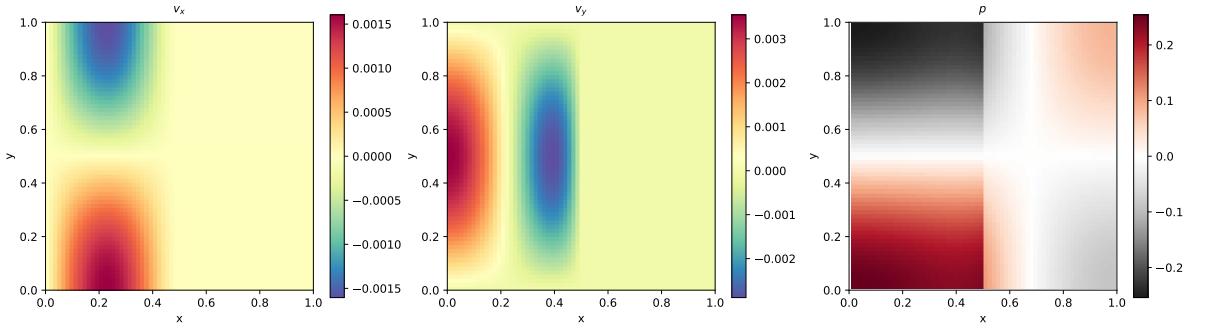
Note the strongly discontinuous viscosity field yields a stagnant flow in the right half of the domain and thereby yields a pressure discontinuity along the interface.

The SolCx benchmark was previously used in [550] (references to earlier uses of the benchmark are available there) and its analytic solution is given in [2017]. It has been carried out in [1119] and [736]. Note that the source code which evaluates the velocity and pressure fields for both SolCx and SolKz is distributed as part of the open source package Underworld ([1353], <http://underworldproject.org>).

In this particular example, the viscosity is computed analytically at the quadrature points (i.e. tracers are not used to attribute a viscosity to the element). If the number of elements is even in any direction, all elements (and their associated quadrature points) have a constant viscosity(1 or 10^6). If it is odd, then the elements situated at the viscosity jump have half their integration points with $\mu = 1$ and half with $\mu = 10^6$ (which is a pathological case since the used quadrature rule inside elements cannot represent accurately such a jump).

features

- $Q_1 \times P_0$ element
- incompressible flow
- penalty formulation
- Dirichlet boundary conditions (free-slip)
- direct solver
- isothermal
- non-isoviscous
- analytical solution



What we learn from this

Stone 06: SolKz benchmark

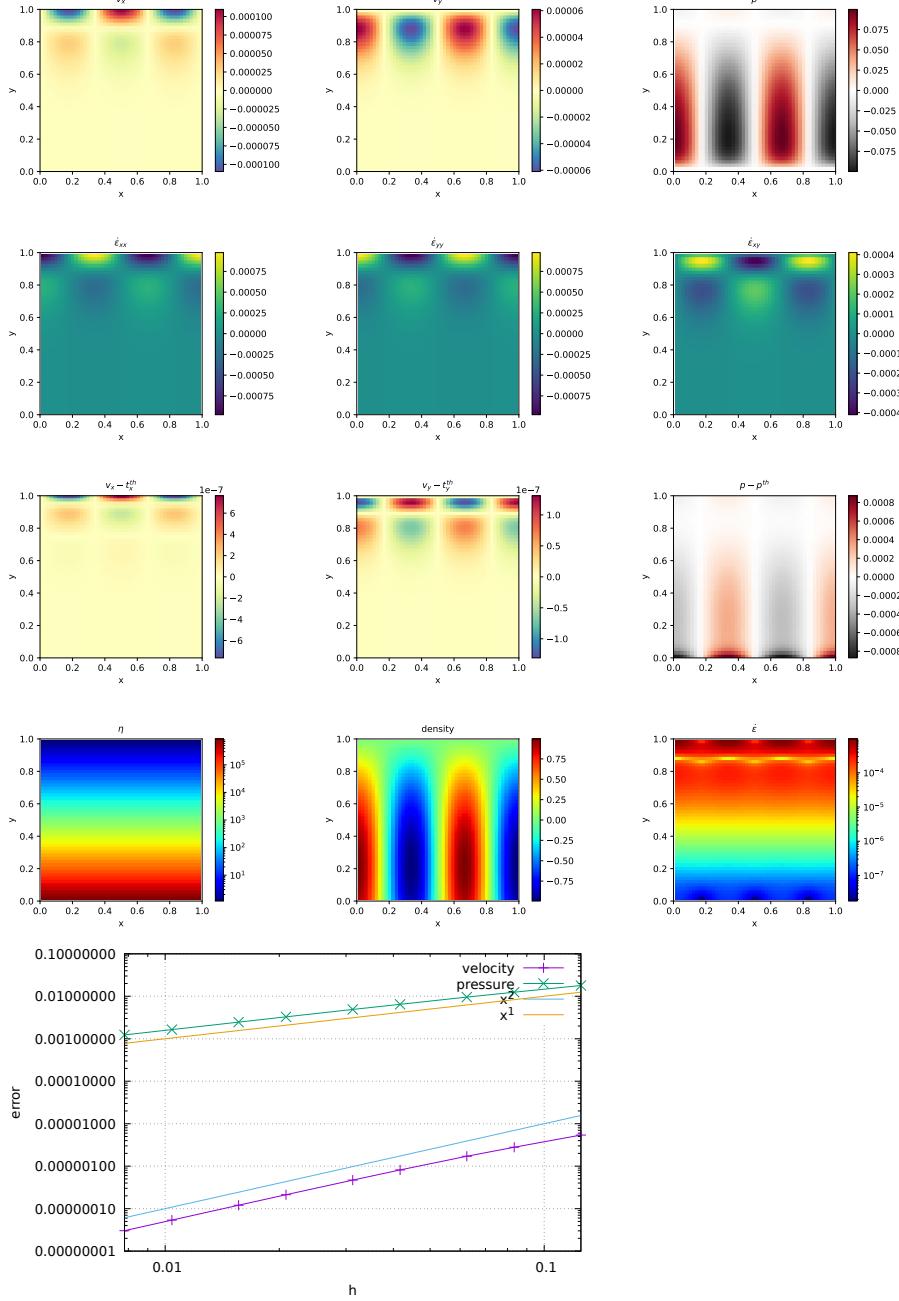
The SolKz benchmark [1567] is similar to the SolCx benchmark, but the viscosity is now a function of the space coordinates:

$$\mu(y) = \exp(By) \quad \text{with} \quad B = 13.8155 \quad (639)$$

It is however not a discontinuous function but grows exponentially with the vertical coordinate so that its overall variation is again 10^6 . The forcing is again chosen by imposing a spatially variable density variation as follows:

$$\rho(x, y) = \sin(2y) \cos(3\pi x) \quad (640)$$

Free slip boundary conditions are imposed on all sides of the domain. This benchmark is presented in [2017] as well and is studied in [550] and [736].



Stone 07: SolVi benchmark

Following SolCx and SolKz, the SolVi inclusion benchmark solves a problem with a discontinuous viscosity field, but in this case the viscosity field is chosen in such a way that the discontinuity is along a circle. Given the regular nature of the grid used by a majority of codes and the present one, this ensures that the discontinuity in the viscosity never aligns to cell boundaries. This in turns leads to almost discontinuous pressures along the interface which are difficult to represent accurately. [1642] derived a simple analytic solution for the pressure and velocity fields for a circular inclusion under simple shear and it was used in [524], [1706], [550], [1119] and [736].

Because of the symmetry of the problem, we only have to solve over the top right quarter of the domain (see Fig. ??a).

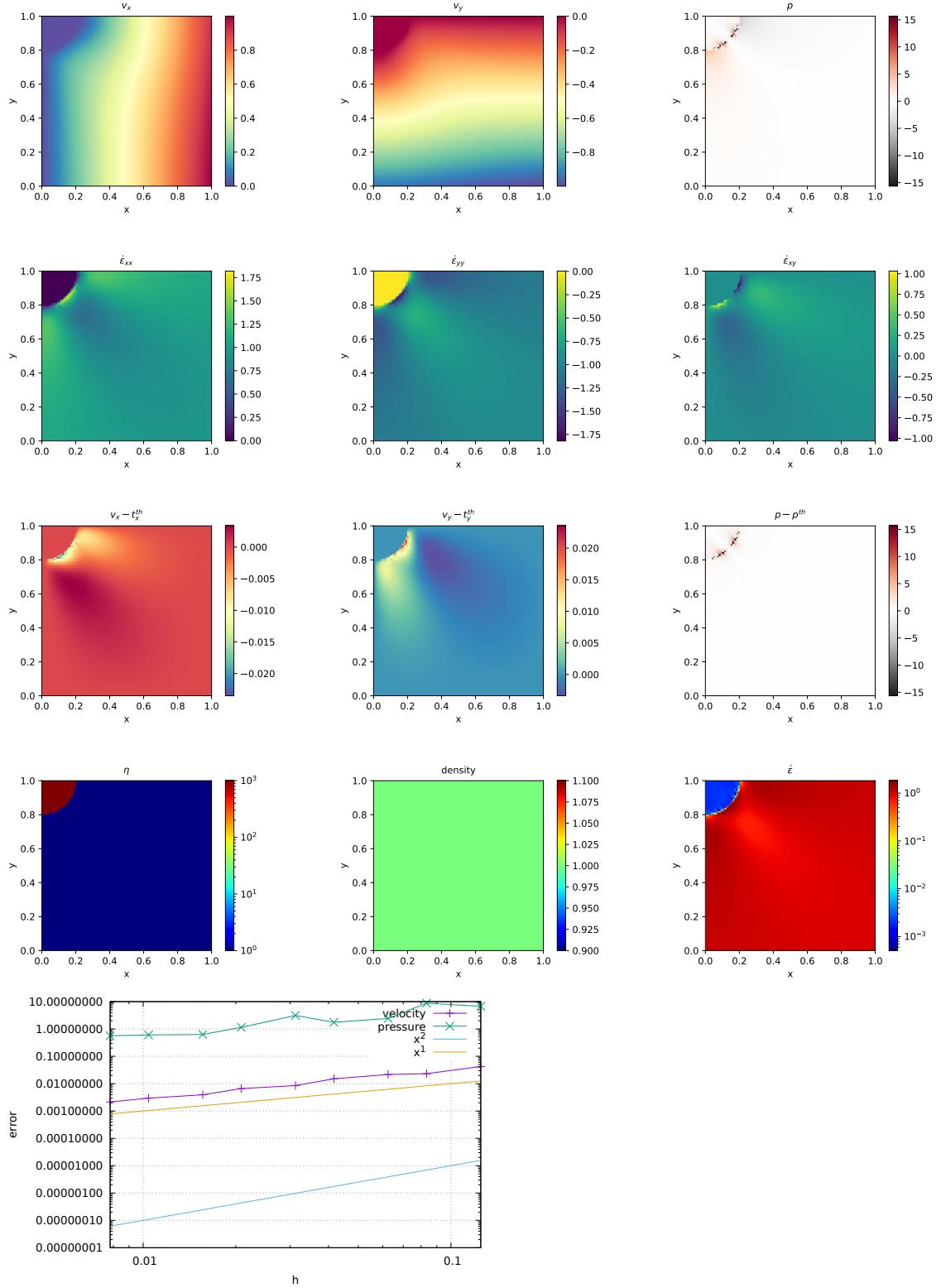
The analytical solution requires a strain rate boundary condition (e.g., pure shear) to be applied far away from the inclusion. In order to avoid using very large domains and/or dealing with this type of boundary condition altogether, the analytical solution is evaluated and imposed on the boundaries of the domain. By doing so, the truncation error introduced while discretizing the strain rate boundary condition is removed.

A characteristic of the analytic solution is that the pressure is zero inside the inclusion, while outside it follows the relation

$$p_m = 4\dot{\epsilon} \frac{\mu_m(\mu_i - \mu_m)}{\mu_i + \mu_m} \frac{r_i^2}{r^2} \cos(2\theta) \quad (641)$$

where $\mu_i = 10^3$ is the viscosity of the inclusion and $\mu_m = 1$ is the viscosity of the background media, $\theta = \tan^{-1}(y/x)$, and $\dot{\epsilon} = 1$ is the applied strain rate.

[524] thoroughly investigated this problem with various numerical methods (FEM, FDM), with and without tracers, and conclusively showed how various averagings lead to different results. [550] obtained a first order convergence for both pressure and velocity, while [1119] and [736] showed that the use of adaptive mesh refinement in respectively the FEM and FDM yields convergence rates which depend on refinement strategies.



Stone 08: the indentor benchmark

The punch benchmark is one of the few boundary value problems involving plastic solids for which there exists an exact solution. Such solutions are usually either for highly simplified geometries (spherical or axial symmetry, for instance) or simplified material models (such as rigid plastic solids) [1008].

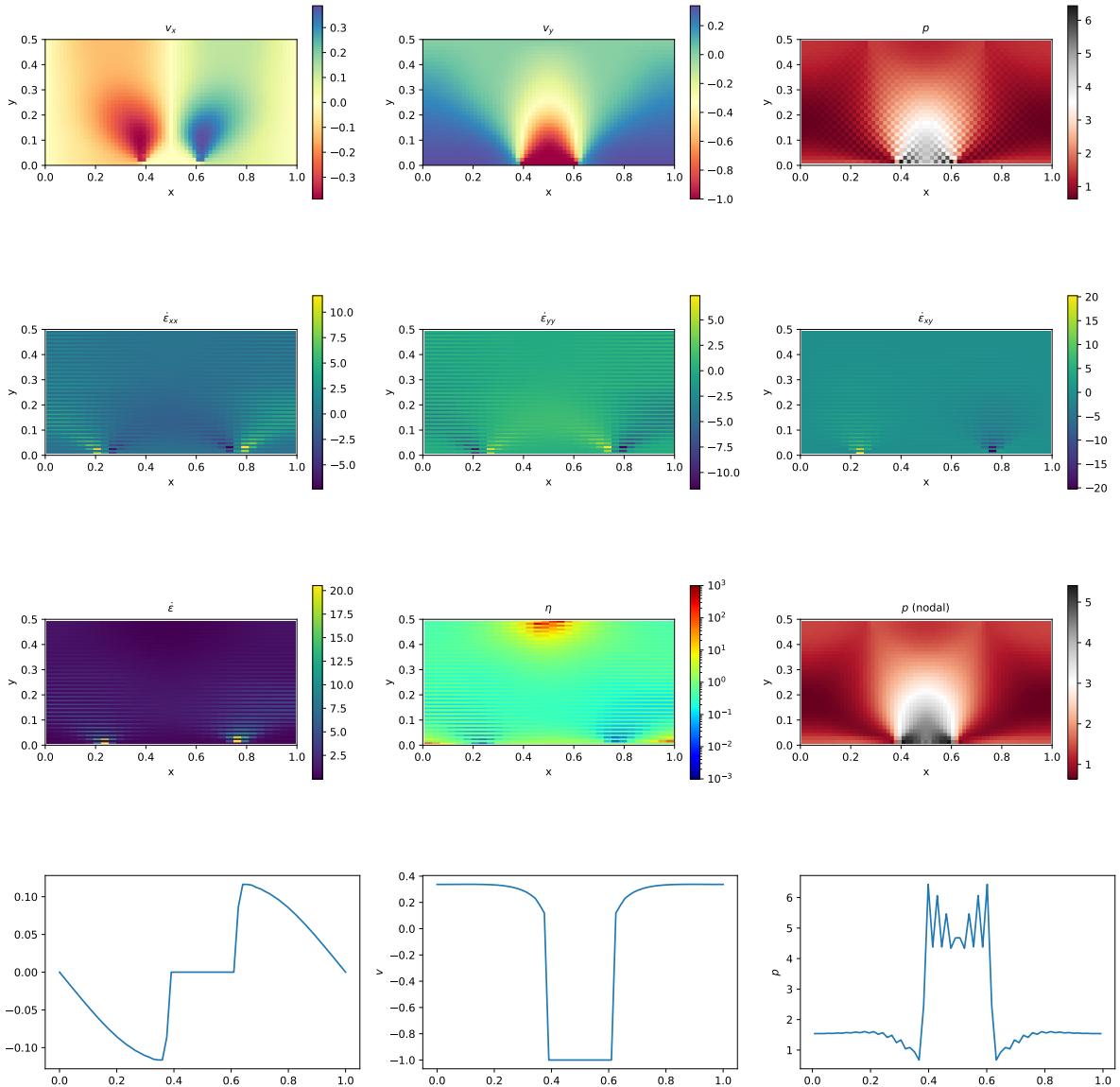
In this experiment, a rigid punch indents a rigid plastic half space; the slip line field theory gives exact solutions as shown in Fig. ??a. The plane strain formulation of the equations and the detailed solution to the problem were derived in the Appendix of [1761] and are also presented in [715].

The two dimensional punch problem has been extensively studied numerically for the past 40 years [2033, 2032, 384, 383, 945, 1983, 259, 1531] and has been used to draw a parallel with the tectonics of eastern China in the context of the India-Eurasia collision [1741, 1336]. It is also worth noting that it has been carried out in one form or another in series of analogue modelling articles concerning the same region, with a rigid indenter colliding with a rheologically stratified lithosphere [1458, 498, 1002].

Numerically, the one-time step punch experiment is performed on a two-dimensional domain of purely plastic von Mises material. Given that the von Mises rheology yield criterion does not depend on pressure, the density of the material and/or the gravity vector is set to zero. Sides are set to free slip boundary conditions, the bottom to no slip, while a vertical velocity $(0, -v_p)$ is prescribed at the top boundary for nodes whose x coordinate is within $[L_x/2 - \delta/2, L_x/2 + \delta/2]$.

The following parameters are used: $L_x = 1$, $L_y = 0.5$, $\mu_{min} = 10^{-3}$, $\mu_{max} = 10^3$, $v_p = 1$, $\delta = 0.123456789$ and the yield value of the material is set to $k = 1$.

The analytical solution predicts that the angle of the shear bands stemming from the sides of the punch is $\pi/4$, that the pressure right under the punch is $1 + \pi$, and that the velocity of the rigid blocks on each side of the punch is $v_p/\sqrt{2}$ (this is simply explained by invoking conservation of mass).



ToDo: smooth punch

features

- $Q_1 \times P_0$ element
- incompressible flow
- penalty formulation
- Dirichlet boundary conditions (no-slip)
- isothermal
- non-isoviscous
- nonlinear rheology

Stone 09: the annulus benchmark

keywords: annulus, $Q_1 \times P_0$, penalty, manufactured solution, incompressible flow, Dirichlet boundary conditions, direct solver, isothermal, isoviscous

This fieldstone was developed in collaboration with Prof. E.G.P. Puckett.

This benchmark is based on Thieulot & Puckett [Subm.] in which an analytical solution to the isoviscous incompressible Stokes equations is derived in an annulus geometry. The velocity and pressure fields are as follows:

$$v_r(r, \theta) = g(r)k \sin(k\theta), \quad (642)$$

$$v_\theta(r, \theta) = f(r) \cos(k\theta), \quad (643)$$

$$p(r, \theta) = kh(r) \sin(k\theta), \quad (644)$$

$$\rho(r, \theta) = \aleph(r)k \sin(k\theta), \quad (645)$$

with

$$f(r) = Ar + B/r, \quad (646)$$

$$g(r) = \frac{A}{2}r + \frac{B}{r} \ln r + \frac{C}{r}, \quad (647)$$

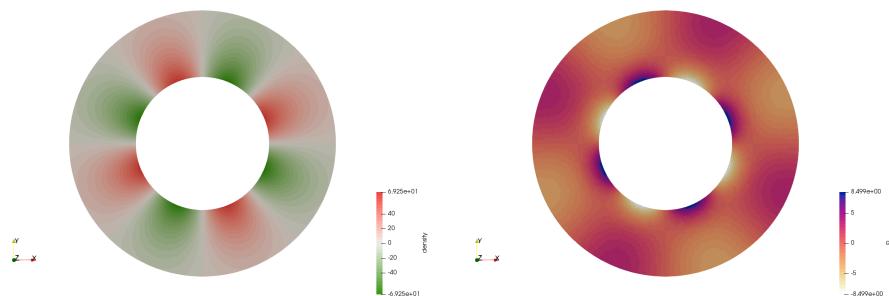
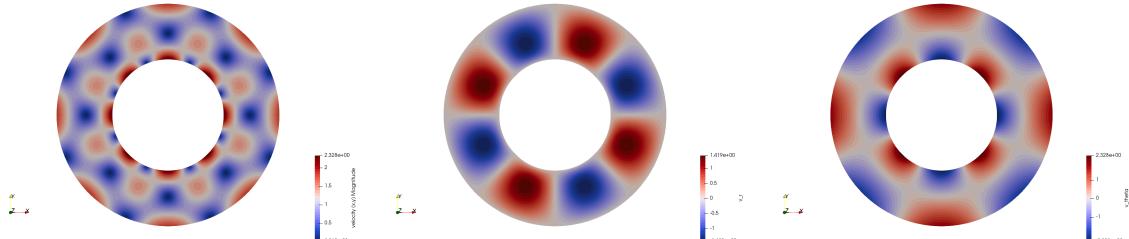
$$h(r) = \frac{2g(r) - f(r)}{r}, \quad (648)$$

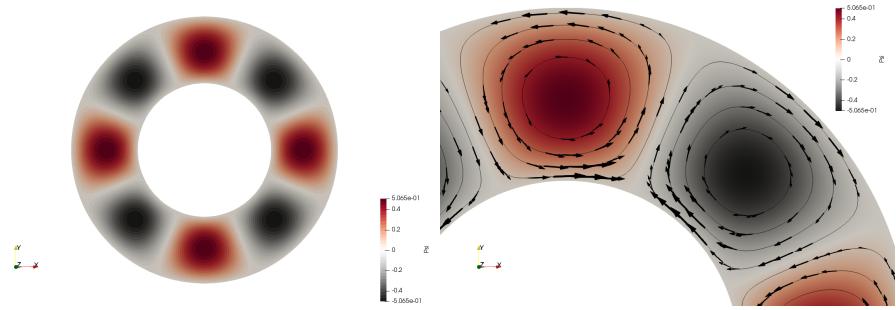
$$\aleph(r) = g'' - \frac{g'}{r} - \frac{g}{r^2}(k^2 - 1) + \frac{f}{r^2} + \frac{f'}{r}, \quad (649)$$

$$A = -C \frac{2(\ln R_1 - \ln R_2)}{R_2^2 \ln R_1 - R_1^2 \ln R_2}, \quad (650)$$

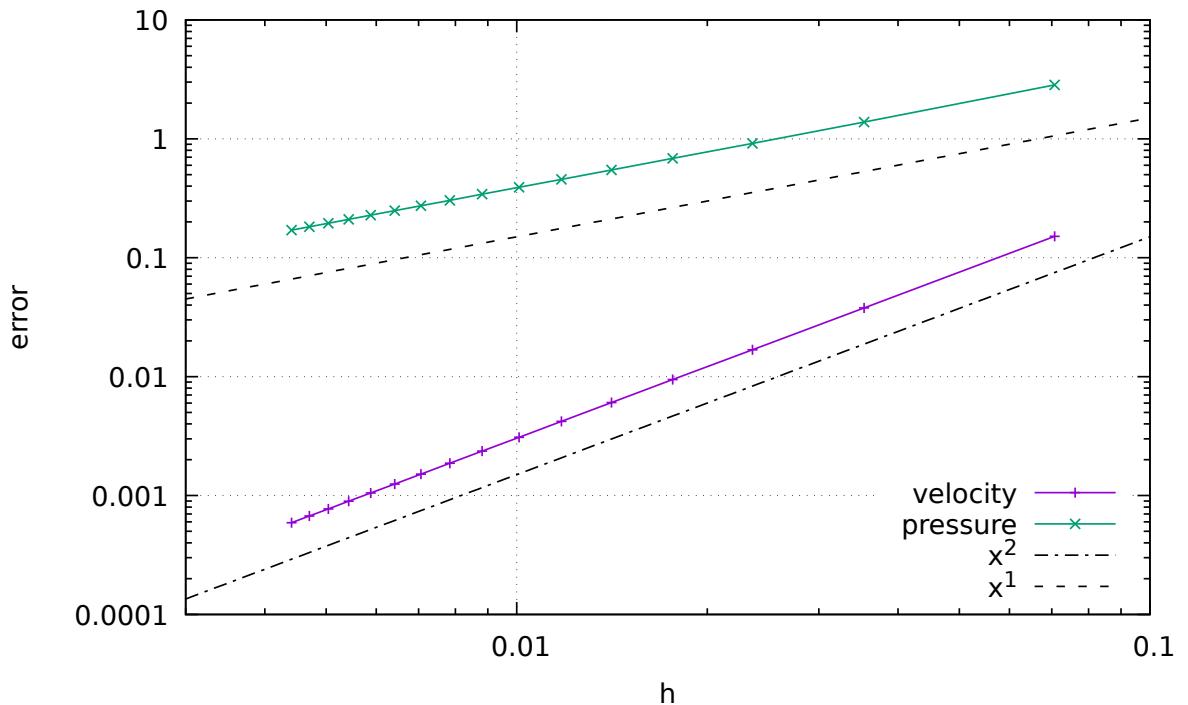
$$B = -C \frac{R_2^2 - R_1^2}{R_2^2 \ln R_1 - R_1^2 \ln R_2}. \quad (651)$$

The parameters A and B are chosen so that $v_r(R_1) = v_r(R_2) = 0$, i.e. the velocity is tangential to both inner and outer surfaces. The gravity vector is radial and of unit length. In the present case, we set $R_1 = 1$, $R_2 = 2$ and $C = -1$.





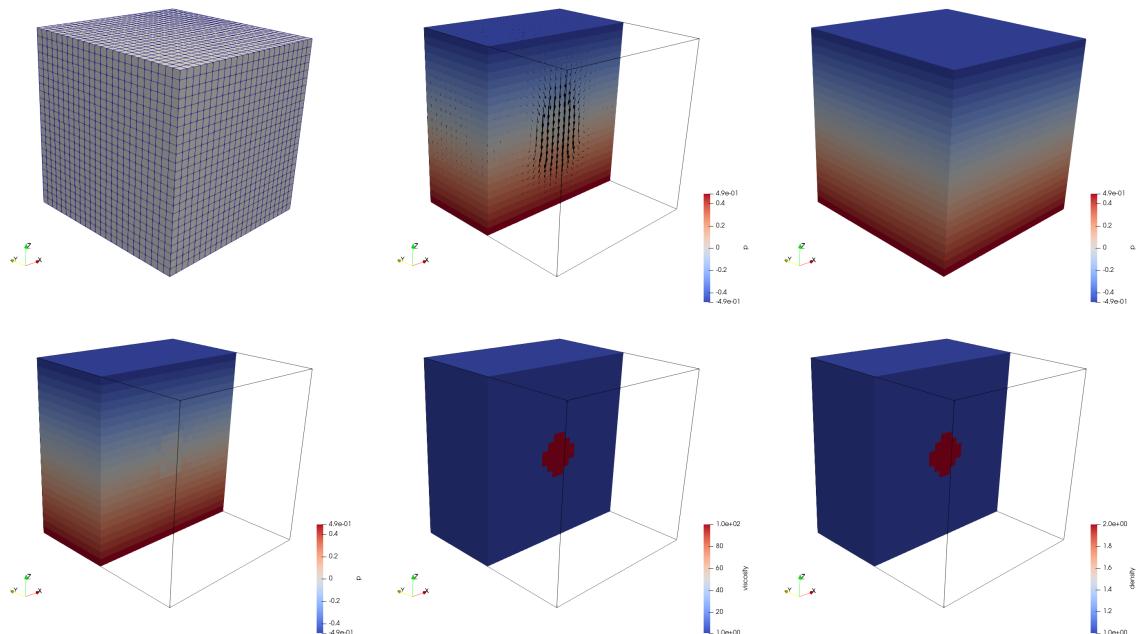
Left: ψ field; right: ψ isolines with velocity arrows.



Stone 10: Stokes sphere (3D) - penalty

features

- $Q_1 \times P_0$ element
- incompressible flow
- penalty formulation
- Dirichlet boundary conditions (free-slip)
- direct solver
- isothermal
- non-isoviscous
- 3D
- elemental b.c.
- buoyancy driven



Stone 11: stokes sphere (3D) - mixed formulation

This is the same setup as Section 9.

features

- $Q_1 \times P_0$ element
- incompressible flow
- mixed formulation
- Dirichlet boundary conditions (free-slip)
- direct solver
- isothermal
- non-isoviscous
- 3D
- elemental b.c.
- buoyancy driven

Stone 12: consistent pressure recovery

What follows is presented in [2030]. The second part of their paper wishes to establish a simple and effective numerical method to calculate variables eliminated by the penalisation process. The method involves an additional finite element solution for the nodal pressures using the same finite element basis and numerical quadrature as used for the velocity.

Let us start with:

$$p = -\lambda \nabla \cdot \mathbf{v}$$

which lead to

$$(q, p) = -\lambda(q, \nabla \cdot \mathbf{v})$$

and then

$$\left(\int \mathbf{N} \mathbf{N} d\Omega \right) \cdot \mathbf{P} = - \left(\lambda \int \mathbf{N} \nabla \mathbf{N} d\Omega \right) \cdot \mathbf{V}$$

or,

$$\mathbf{M} \cdot \mathbf{P} = -\mathbf{D} \cdot \mathbf{V}$$

and finally

$$\mathbf{P} = -\mathbf{M}^{-1} \cdot \mathbf{D} \cdot \mathbf{V}$$

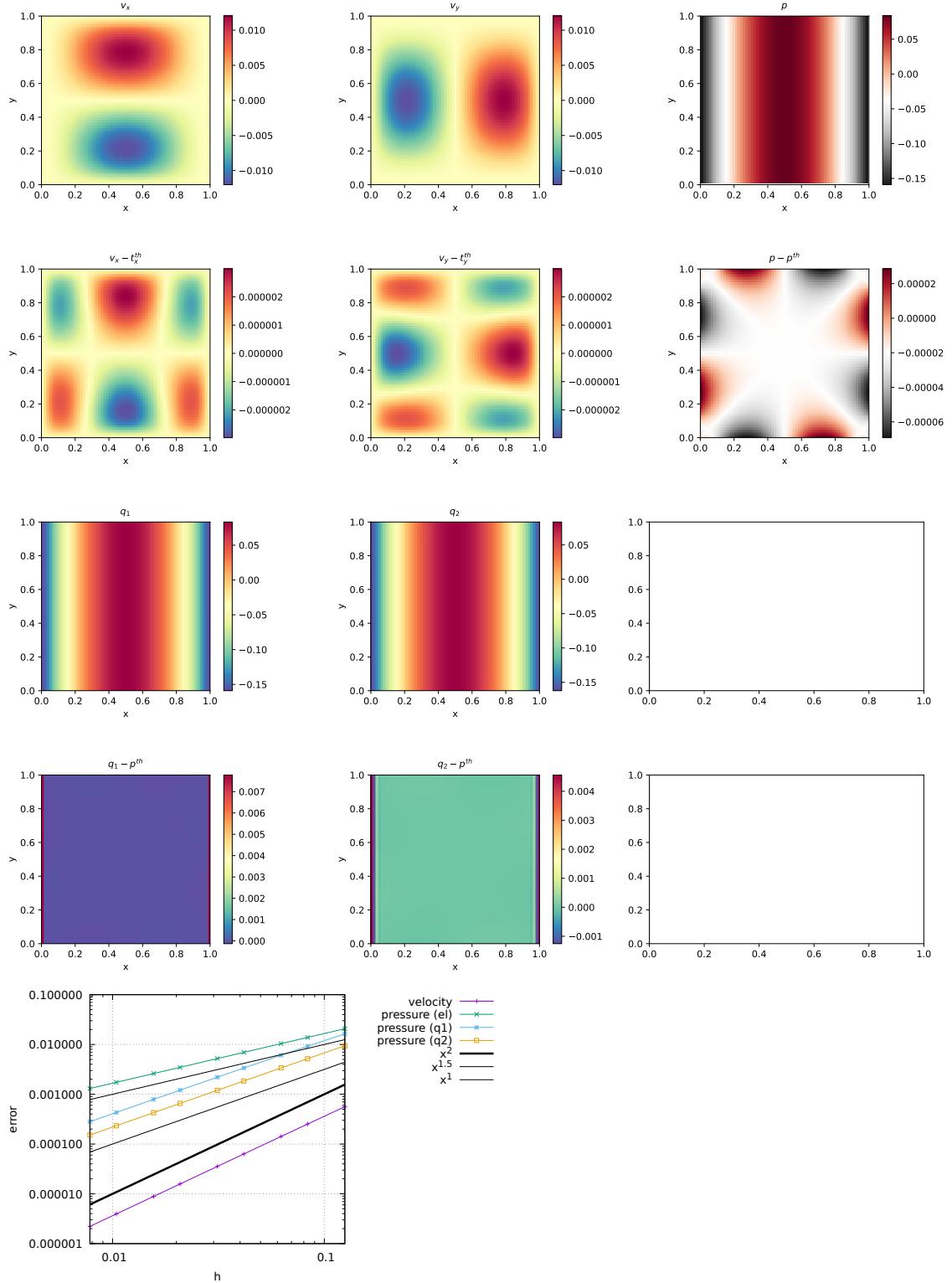
with \mathbf{M} of size $(np \times np)$, \mathbf{D} of size $(np * ndof \times np * ndof)$ and \mathbf{V} of size $(np * ndof)$. The vector \mathbf{P} contains the np nodal pressure values directly, with no need for a smoothing scheme. The mass matrix \mathbf{M} is to be evaluated at the full integration points, while the constraint part (the right hand side of the equation) is to be evaluated at the reduced integration point.

As noted by [2030], it is interesting to note that when linear elements are used and the lumped matrices are used for the \mathbf{M} the resulting algebraic equation is identical to the smoothing scheme based on the averaging method only if the uniform square finite element mesh is used. In this respect this method is expected to yield different results when elements are not square or even rectangular.

q_1 is smoothed pressure obtained with the center-to-node approach.

q_2 is recovered pressure obtained with [2030].

All three fulfill the zero average condition: $\int p d\Omega = 0$.



In terms of pressure error, q_2 is better than q_1 which is better than elemental.

QUESTION: why are the averages exactly zero ?!

TODO:

- add randomness to internal node positions.
- look at elefant algorithms

Stone 13: the Particle in Cell technique (1) - the effect of averaging

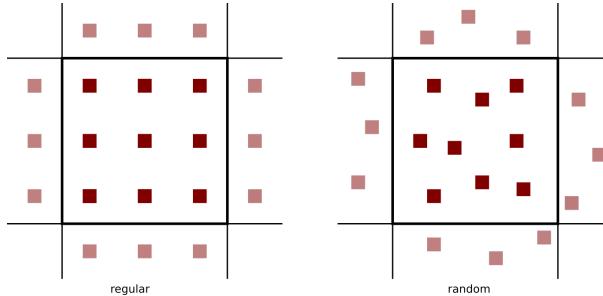
This fieldstone is being developed in collaboration with BSc student Eric Hoogen.

features

- $Q_1 \times P_0$ element
- incompressible flow
- penalty formulation
- Dirichlet boundary conditions (no-slip)
- isothermal
- non-isoviscous
- particle-in-cell

After the initial setup of the grid, markers can then be generated and placed in the domain. One could simply randomly generate the marker positions in the whole domain but unless a *very* large number of markers is used, the chance that an element does not contain any marker exists and this will prove problematic. In order to get a better control over the markers spatial distribution, one usually generates the marker per element, so that the total number of markers in the domain is the product of the number of elements times the user-chosen initial number of markers per element.

Our next concern is how to actually place the markers inside an element. Two methods come to mind: on a regular grid, or in a random manner, as shown on the following figure:



In both cases we make use of the basis shape functions: we generate the positions of the markers (random or regular) in the reference element first (r_{im}, s_{im}), and then map those out to the real element as follows:

$$x_{im} = \sum_i^m N_i(r_{im}, s_{im}) x_i \quad y_{im} = \sum_i^m N_i(r_{im}, s_{im}) y_i \quad (652)$$

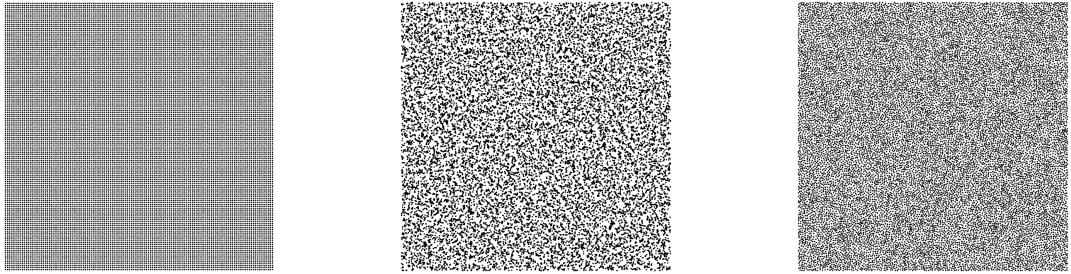
where x_i, y_i are the coordinates of the vertices of the element.

A third option consists in the use of the so-called Poisson-disc sampling which produces points that are tightly-packed, but no closer to each other than a specified minimum distance, resulting in a more natural pattern⁵⁸. Note that the Poisson-disc algorithm fills the whole domain at once, not element after element.

say smthg about avrg dist

insert here theory and link about Poisson disc

⁵⁸<https://en.wikipedia.org/wiki/Supersampling>



Left: regular distribution, middle: random, right: Poisson disc.
16384 markers (32x32 grid, 16 markers per element).

When using *active* markers, one is faced with the problem of transferring the properties they carry to the mesh on which the PDEs are to be solved. As we have seen, building the FE matrix involves a loop over all elements, so one simple approach consists of assigning each element a single property computed as the average of the values carried by the markers in that element. Often in colloquial language "average" refers to the arithmetic mean:

$$\langle \phi \rangle_{am} = \frac{1}{n} \sum_k^n \phi_i \quad (653)$$

where $\langle \phi \rangle_{am}$ is the arithmetic average of the n numbers ϕ_i . However, in mathematics other means are commonly used, such as the geometric mean:

$$\langle \phi \rangle_{gm} = \left(\prod_i^n \phi_i \right) \quad (654)$$

PROBLEM with this formula!!!! and the harmonic mean:

$$\langle \phi \rangle_{hm} = \left(\frac{1}{n} \sum_i^n \frac{1}{\phi_i} \right)^{-1} \quad (655)$$

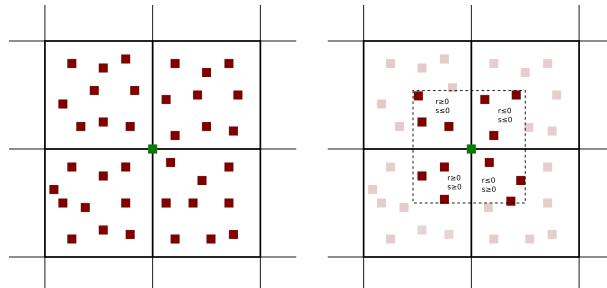
Furthermore, there is a well known inequality for any set of positive numbers,

$$\langle \phi \rangle_{am} \geq \langle \phi \rangle_{gm} \geq \langle \phi \rangle_{hm} \quad (656)$$

which will prove to be important later on.

Let us now turn to a simple concrete example: the 2D Stokes sphere. There are two materials in the domain, so that markers carry the label "mat=1" or "mat=2". For each element an average density and viscosity need to be computed. The majority of elements contains markers with a single material label so that the choice of averaging does not matter (it is trivial to verify that if $\phi_i = \phi_0$ then $\langle \phi \rangle_{am} = \langle \phi \rangle_{gm} = \langle \phi \rangle_{hm} = \phi_0$). Remain the elements crossed by the interface between the two materials: they contain markers of both materials and the average density and viscosity inside those depends on 1) the total number of markers inside the element, 2) the ratio of markers 1 to markers 2, 3) the type of averaging.

This averaging problem has been studied and documented in the literature [1638, 524, 1754, 1508]



Nodal projection. Left: all markers inside elements to which the green node belongs to are taken into account.
Right: only the markers closest to the green node count.

Let k be the green node of the figures above. Let (r, s) denote the coordinates of a marker inside its element. For clarity, we define the follow three nodal averaging schemes:

- nodal type A:

$$f_k = \frac{\text{sum of values carried by markers in 4 neighbour elements}}{\text{number of markers in 4 neighbour elements}}$$

- nodal type B:

$$f_k = \frac{\text{sum of values carried by markers inside dashed line}}{\text{number of markers in area delimited by the dashed line}}$$

- nodal type C

$$f_k = \frac{\text{sum of values carried by markers in 4 neighbour elements} * N_p(r, s)}{\text{sum of } N_p(r, s)}$$

where N_p is the Q_1 basis function corresponding to node p defined on each element. Since these functions are 1 on node k and then linearly decrease and become zero on the neighbouring nodes, this effectively gives more weight to those markers closest to node k .

This strategy is adopted in [1292, 1289] (although it is used to interpolate onto the nodes of Q_2P_{-1} elements. It is formulated as follows:

"We assume that an arbitrary material point property f , is discretized via $f(\mathbf{x}) \simeq \delta(\mathbf{x} - \mathbf{x}_p)f_p$. We then utilize an approximate local L_2 projection of f_p onto a continuous Q_1 finite element space. The corner vertices of each Q_2 finite element define the mesh f_p is projected onto. The local reconstruction for a node i is defined by

$$\hat{f}_i = \frac{\int_{\Omega_i} N_i(\mathbf{x}) f(\mathbf{x})}{\int_{\Omega_i} N_i(\mathbf{x})} \simeq \frac{\sum_p N_i(\mathbf{x}_p) f_p}{\sum_p N_i(\mathbf{x}_p)}$$

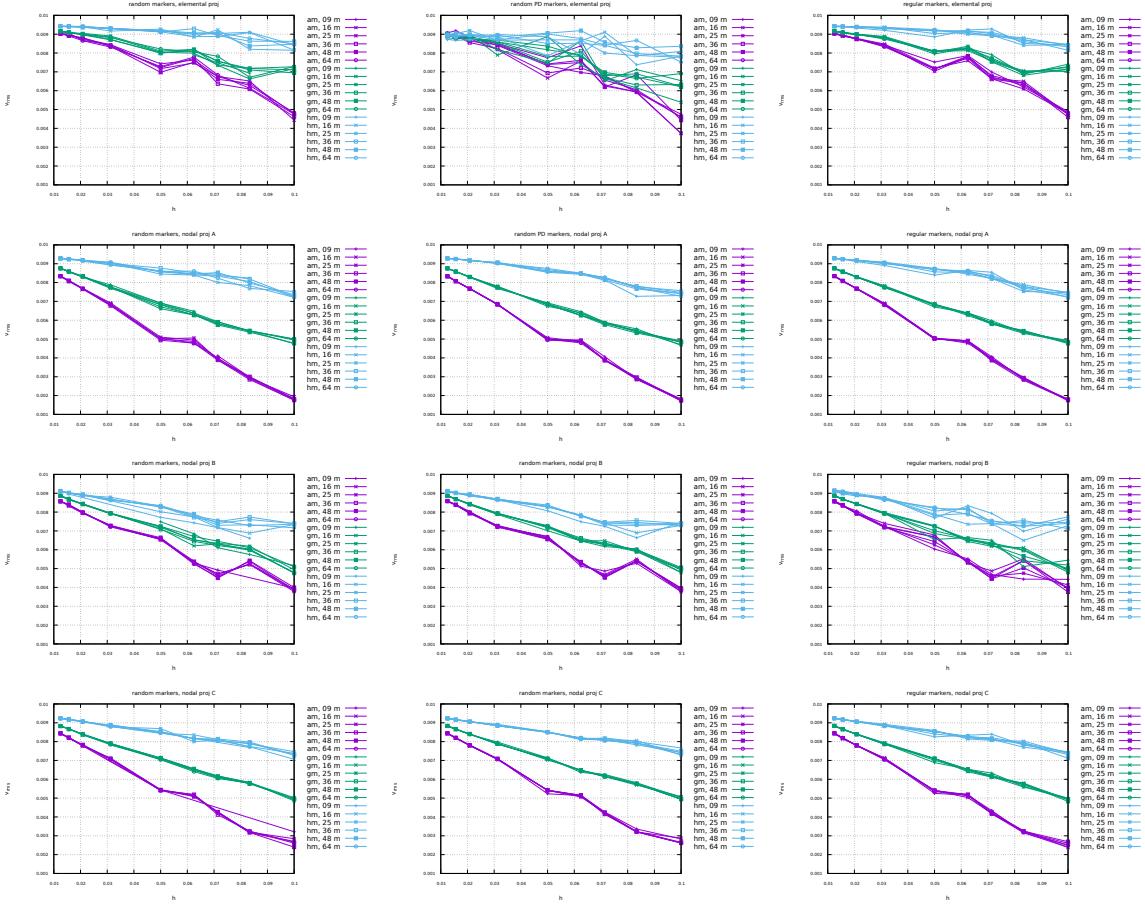
where the summation over p includes all material points contained within the support Ω_i of the trilinear interpolant N_i ".

The setup is identical to the Stokes sphere experiment. The bash script `script_runall` runs the code for many resolutions, both initial marker distribution and all four averaging types. The viscosity of the sphere has been set to 10^3 while the viscosity of the surrounding fluid is 1. The average density is always computed with an arithmetic mean.

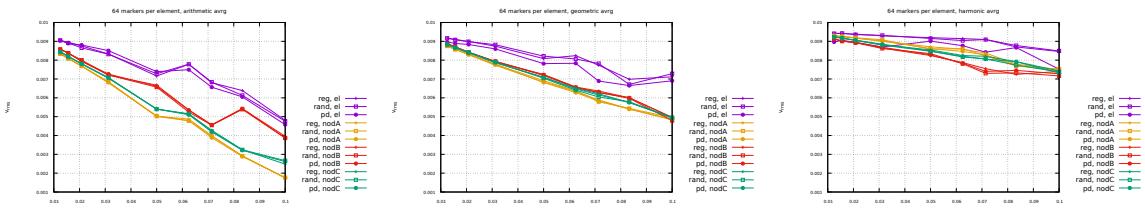
Conclusions:

- With increasing resolution ($h \rightarrow 0$) vrms values seem to converge towards a single value, irrespective of the number of markers.
- At low resolution, say 32x32 (i.e. $h=0.03125$), vrms values for the three averagings differ by about 10%. At higher resolution, say 128x128, vrms values are still not converged.
- The number of markers per element plays a role at low resolution, but less and less with increasing resolution.
- Results for random and regular marker distributions are not identical but follow a similar trend and seem to converge to the same value.
- elemental values yield better results (espcecially at low resolutions)
- harmonic mean yields overal the best results

Root mean square velocity results are shown hereunder:



Left column: random markers, middle column: Poisson disc, right column: regular markers. First row: elemental projection, second row: nodal 1 projection, third row: nodal 2 projection, fourth row: nodal 3 projection.



Left to right: arithmetic, geometric, harmonic averaging for viscosity.

Stone 14: solving the full saddle point problem

The details of the numerical setup are presented in Section ??.

The main difference is that we no longer use the penalty formulation and therefore keep both velocity and pressure as unknowns. Therefore we end up having to solve the following system:

$$\begin{pmatrix} \mathbb{K} & \mathbb{G} \\ \mathbb{G}^T & 0 \end{pmatrix} \cdot \begin{pmatrix} V \\ P \end{pmatrix} = \begin{pmatrix} f \\ h \end{pmatrix} \quad \text{or,} \quad \mathbb{A} \cdot X = rhs$$

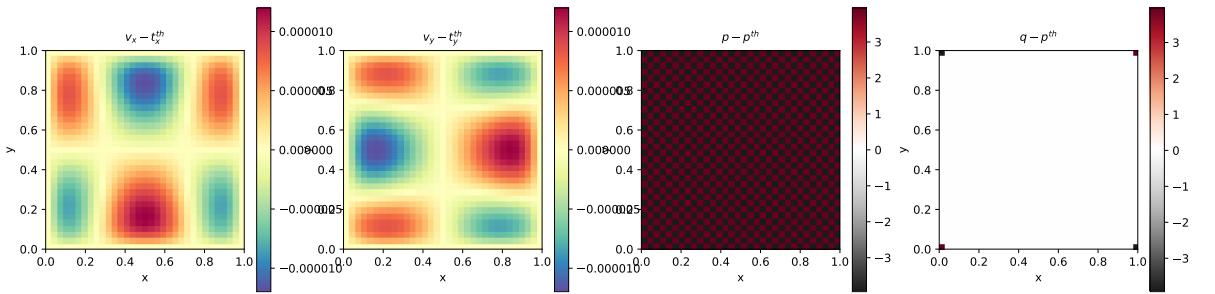
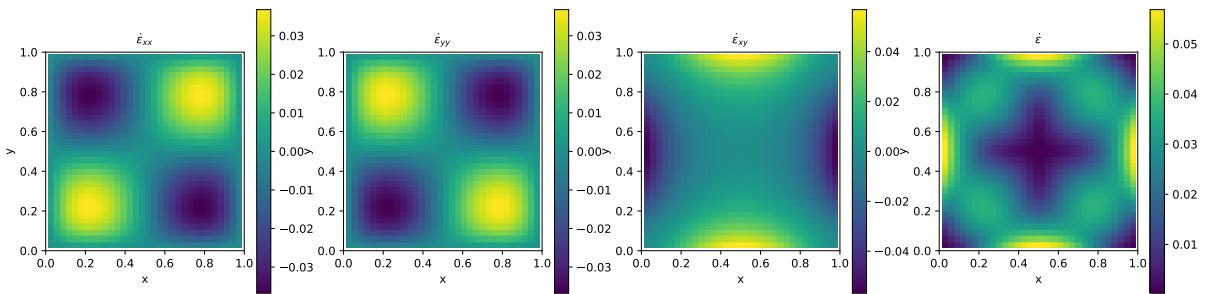
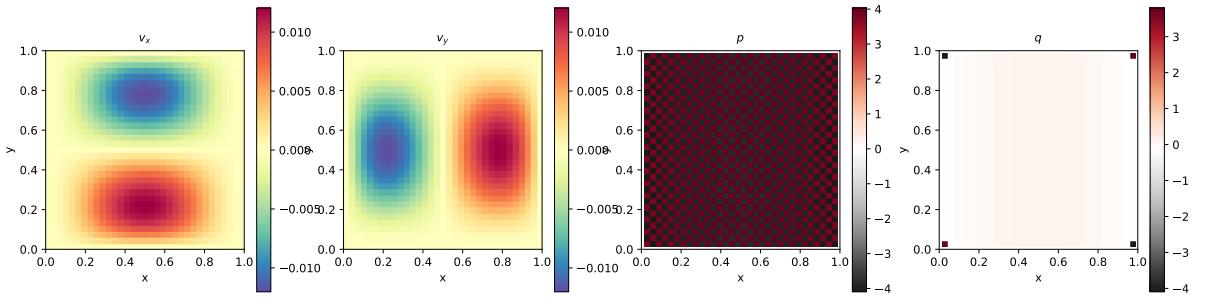
Each block \mathbb{K} , \mathbb{G} and vector f , h are built separately in the code and assembled into the matrix \mathbb{A} and vector rhs afterwards. \mathbb{A} and rhs are then passed to the solver. We will see later that there are alternatives to solve this approach which do not require to build the full Stokes matrix \mathbb{A} .

Each element has $m = 4$ vertices so in total $ndofV \times m = 8$ velocity dofs and a single pressure dof, commonly situated in the center of the element. The total number of velocity dofs is therefore $NfemV = nnp \times ndofV$ while the total number of pressure dofs is $NfemP = nel$. The total number of dofs is then $Nfem = NfemV + NfemP$.

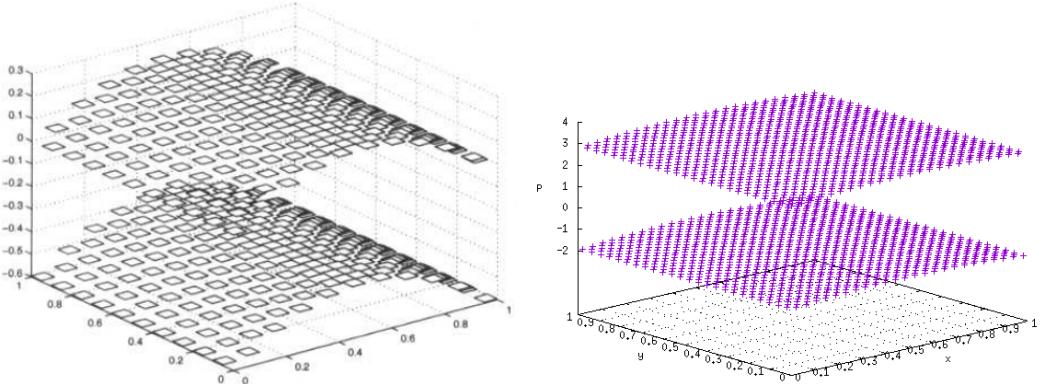
As a consequence, matrix \mathbb{K} has size $NfemV, NfemV$ and matrix \mathbb{G} has size $NfemV, NfemP$. Vector f is of size $NfemV$ and vector h is of size $NfemP$.

features

- $Q_1 \times P_0$ element
- incompressible flow
- mixed formulation
- Dirichlet boundary conditions (no-slip)
- direct solver (?)
- isothermal
- isoviscous
- analytical solution
- pressure smoothing



Unlike the results obtained with the penalty formulation (see Section ??), the pressure showcases a very strong checkerboard pattern, similar to the one in [533].



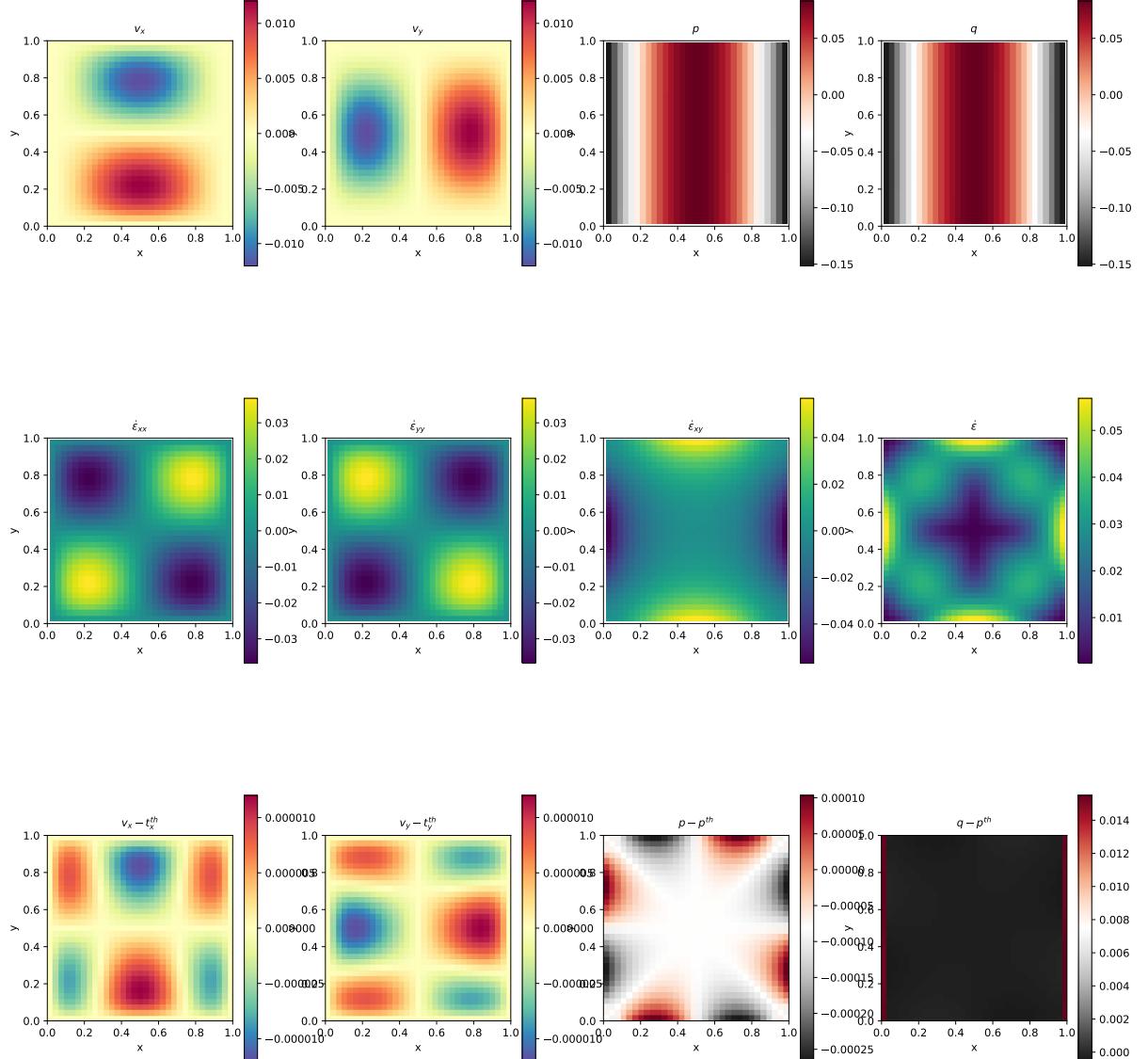
Left: pressure solution as shown in [533]; Right: pressure solution obtained with fieldstone.

Rather interestingly, the nodal pressure (obtained with a simple center-to-node algorithm) fails to recover a correct pressure at the four corners.

Note that the umfpack solver complains a lot about the matrix condition number, even at (very) low resolutions. I believe it does not like the zeros on the (2,2) block of the assembled Stokes matrix.

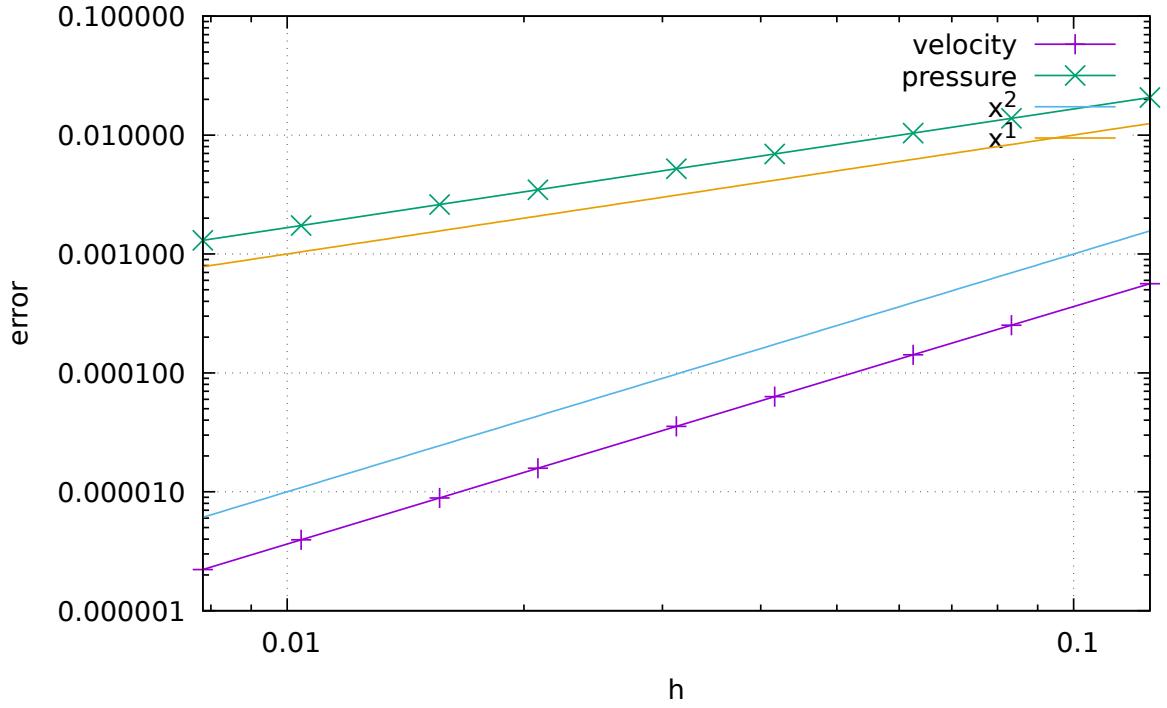
Stone 15: saddle point problem with Schur complement approach - benchmark

The details of the numerical setup are presented in Section ???. The main difference resides in the Schur complement approach to solve the Stokes system, as presented in Section ??? (see `solver_cg`). This iterative solver is very easy to implement once the blocks \mathbb{K} and \mathbb{G} , as well as the rhs vectors f and h have been built.

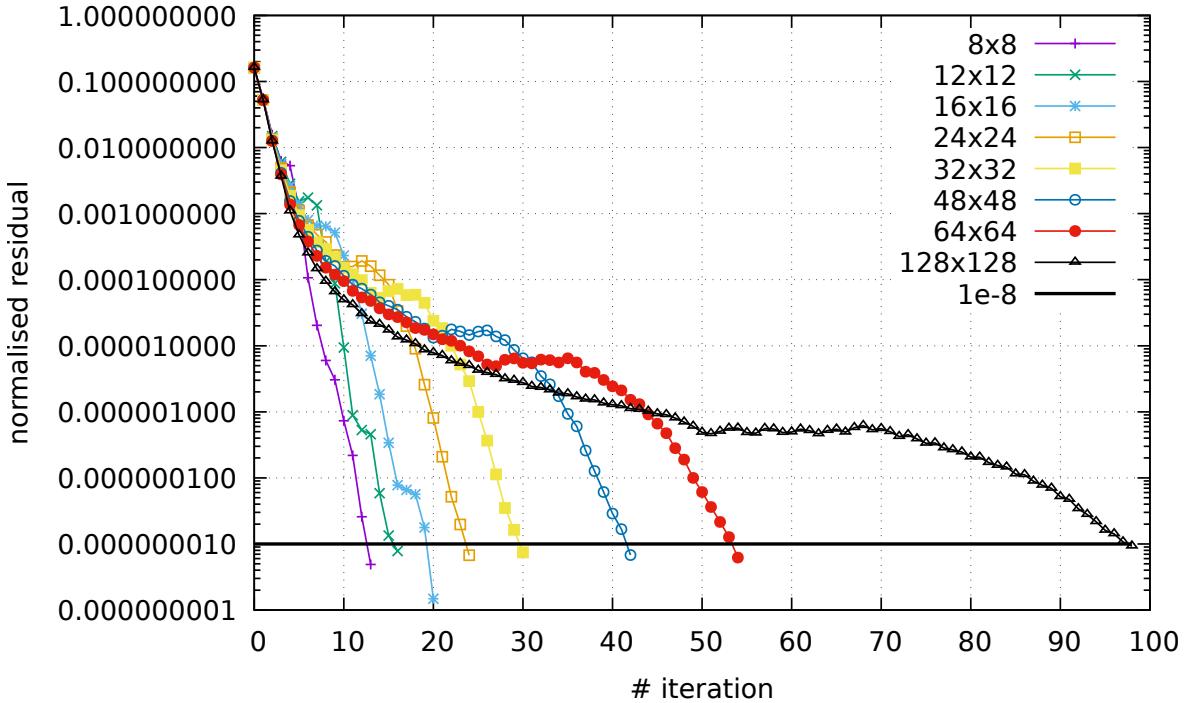


Rather interestingly the pressure checkerboard modes are not nearly as present as in Section ??? which uses a full matrix approach.

Looking at the discretisation errors for velocity and pressure, we of course recover the same rates and values as in the full matrix case.



Finally, for each experiment the normalised residual (see `solver_cg`) was recorded. We see that all things equal the resolution has a strong influence on the number of iterations the solver must perform to reach the required tolerance. This is one of the manifestations of the fact that the $Q_1 \times P_0$ element is not a stable element: the condition number of the matrix increases with resolution. We will see that this is not the case of stable elements such as $Q_2 \times Q_1$.



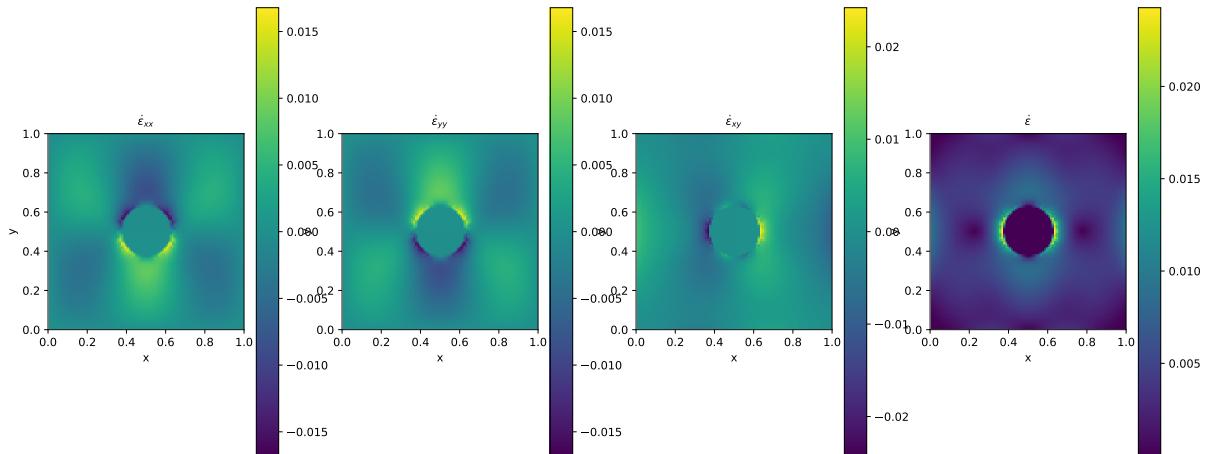
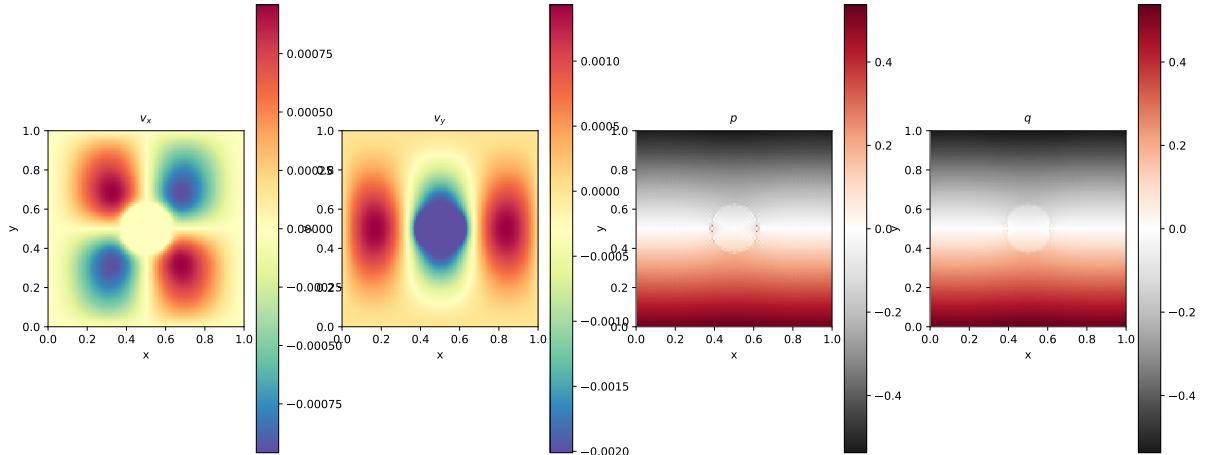
features

- $Q_1 \times P_0$ element
- incompressible flow
- mixed formulation
- Schur complement approach
- isothermal
- isoviscous
- analytical solution

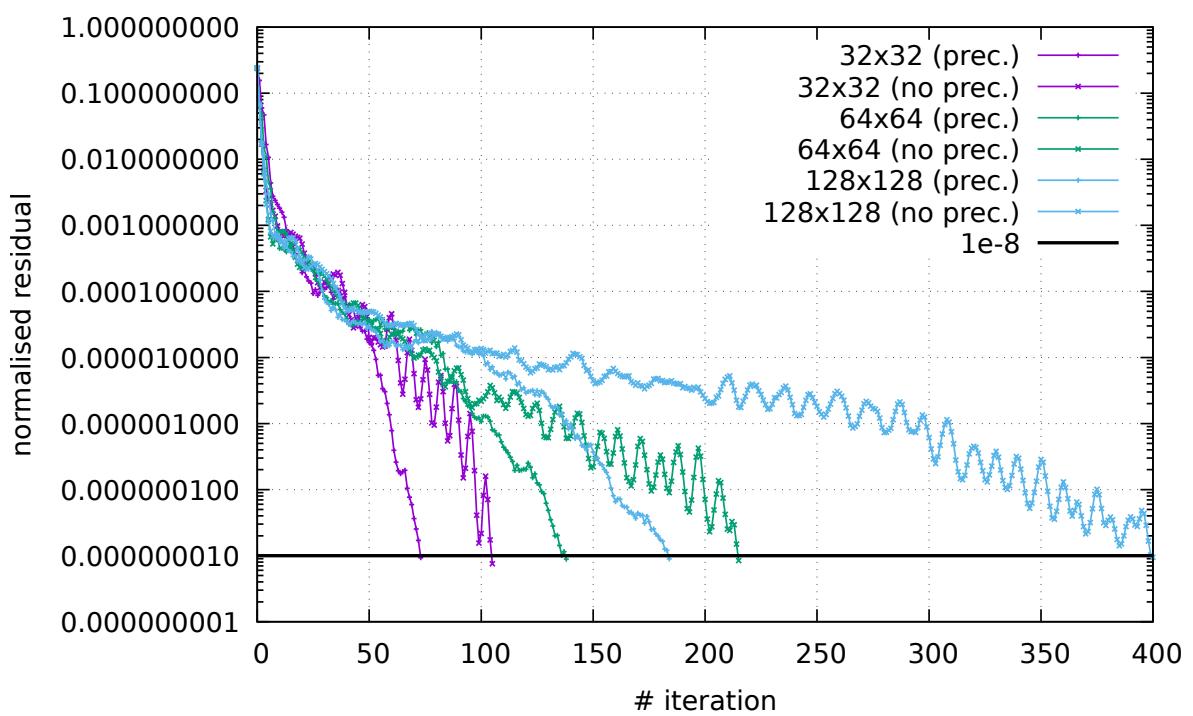
build S and have python compute its smallest and largest eigenvalues as a function of resolution?

Stone 16: saddle point problem with Schur complement approach - Stokes sphere

We are revisiting the 2D Stokes sphere problem, but this time we use the Schur complement approach to solve the Stokes system. Because there are viscosity contrasts in the domain, it is advisable to use the Preconditioned Conjugate Gradient as presented in Section ?? (see `solver_pcg`).



The normalised residual (see `solver_pcg`) was recorded. We see that all things equal the resolution has a strong influence on the number of iterations the solver must perform to reach the required tolerance. However, we see that the use of the preconditioner can substantially reduce the number of iterations inside the Stokes solver. At resolution 128x128, this number is halved.



features

- $Q_1 \times P_0$ element
- incompressible flow
- mixed formulation
- Schur complement approach
- isothermal
- non-isoviscous
- Stokes sphere

Stone 17: solving the full saddle point problem in 3D

When using $Q_1 \times P_0$ elements, this benchmark fails because of the Dirichlet b.c. on all 6 sides and all three components. However, as we will see, it does work well with $Q_2 \times Q_1$ elements. .

This benchmark begins by postulating a polynomial solution to the 3D Stokes equation [529]:

$$\mathbf{v} = \begin{pmatrix} x + x^2 + xy + x^3y \\ y + xy + y^2 + x^2y^2 \\ -2z - 3xz - 3yz - 5x^2yz \end{pmatrix} \quad (657)$$

and

$$p = xyz + x^3y^3z - 5/32 \quad (658)$$

While it is then trivial to verify that this velocity field is divergence-free, the corresponding body force of the Stokes equation can be computed by inserting this solution into the momentum equation with a given viscosity μ (constant or position/velocity/strain rate dependent). The domain is a unit cube and velocity boundary conditions simply use Eq. (657). Following [308], the viscosity is given by the smoothly varying function

$$\mu = \exp(1 - \beta(x(1-x) + y(1-y) + z(1-z))) \quad (659)$$

One can easily show that the ratio of viscosities μ^* in the system follows $\mu^* = \exp(-3\beta/4)$ so that choosing $\beta = 10$ yields $\mu^* \simeq 1808$ and $\beta = 20$ yields $\mu^* \simeq 3.269 \times 10^6$.

We start from the momentum conservation equation:

$$-\nabla p + \nabla \cdot (2\mu \dot{\epsilon}) = \mathbf{f}$$

The x -component of this equation writes

$$f_x = -\frac{\partial p}{\partial x} + \frac{\partial}{\partial x}(2\mu \dot{\epsilon}_{xx}) + \frac{\partial}{\partial y}(2\mu \dot{\epsilon}_{xy}) + \frac{\partial}{\partial z}(2\mu \dot{\epsilon}_{xz}) \quad (660)$$

$$= -\frac{\partial p}{\partial x} + 2\mu \frac{\partial}{\partial x} \dot{\epsilon}_{xx} + 2\mu \frac{\partial}{\partial y} \dot{\epsilon}_{xy} + 2\mu \frac{\partial}{\partial z} \dot{\epsilon}_{xz} + 2 \frac{\partial \mu}{\partial x} \dot{\epsilon}_{xx} + 2 \frac{\partial \mu}{\partial y} \dot{\epsilon}_{xy} + 2 \frac{\partial \mu}{\partial z} \dot{\epsilon}_{xz} \quad (661)$$

Let us compute all the block separately:

$$\begin{aligned} \dot{\epsilon}_{xx} &= 1 + 2x + y + 3x^2y \\ \dot{\epsilon}_{yy} &= 1 + x + 2y + 2x^2y \\ \dot{\epsilon}_{zz} &= -2 - 3x - 3y - 5x^2y \\ 2\dot{\epsilon}_{xy} &= (x + x^3) + (y + 2xy^2) = x + y + 2xy^2 + x^3 \\ 2\dot{\epsilon}_{xz} &= (0) + (-3z - 10xyz) = -3z - 10xyz \\ 2\dot{\epsilon}_{yz} &= (0) + (-3z - 5x^2z) = -3z - 5x^2z \end{aligned}$$

In passing, one can verify that $\dot{\epsilon}_{xx} + \dot{\epsilon}_{yy} + \dot{\epsilon}_{zz} = 0$. We further have

$$\begin{aligned}\frac{\partial}{\partial x} 2\dot{\epsilon}_{xx} &= 2(2 + 6xy) \\ \frac{\partial}{\partial y} 2\dot{\epsilon}_{xy} &= 1 + 4xy \\ \frac{\partial}{\partial z} 2\dot{\epsilon}_{xz} &= -3 - 10xy \\ \frac{\partial}{\partial x} 2\dot{\epsilon}_{xy} &= 1 + 2y^2 + 3x^2 \\ \frac{\partial}{\partial y} 2\dot{\epsilon}_{yy} &= 2(2 + 2x^2) \\ \frac{\partial}{\partial z} 2\dot{\epsilon}_{yz} &= -3 - 5x^2 \\ \frac{\partial}{\partial x} 2\dot{\epsilon}_{xz} &= -10yz \\ \frac{\partial}{\partial y} 2\dot{\epsilon}_{yz} &= 0 \\ \frac{\partial}{\partial z} 2\dot{\epsilon}_{zz} &= 2(0)\end{aligned}$$

$$\frac{\partial p}{\partial x} = yz + 3x^2y^3z \quad (662)$$

$$\frac{\partial p}{\partial y} = xz + 3x^3y^2z \quad (663)$$

$$\frac{\partial p}{\partial z} = xy + x^3y^3 \quad (664)$$

Pressure normalisation Here again, because Dirichlet boundary conditions are prescribed on all sides the pressure is known up to an arbitrary constant. This constant can be determined by (arbitrarily) choosing to normalised the pressure field as follows:

$$\int_{\Omega} p \, d\Omega = 0 \quad (665)$$

This is a single constraint associated to a single Lagrange multiplier λ and the global Stokes system takes the form

$$\begin{pmatrix} \mathbb{K} & \mathbb{G} & 0 \\ \mathbb{G}^T & 0 & \mathcal{C} \\ 0 & \mathcal{C}^T & 0 \end{pmatrix} \begin{pmatrix} V \\ P \\ \lambda \end{pmatrix}$$

In this particular case the constraint matrix \mathcal{C} is a vector and it only acts on the pressure degrees of freedom because of Eq.(665). Its exact expression is as follows:

$$\int_{\Omega} p \, d\Omega = \sum_e \int_{\Omega_e} p \, d\Omega = \sum_e \int_{\Omega_e} \sum_i N_i^p p_i \, d\Omega = \sum_e \sum_i \left(\int_{\Omega_e} N_i^p \, d\Omega \right) p_i = \sum_e \mathcal{C}_e \cdot \mathbf{p}_e$$

where \mathbf{p}_e is the list of pressure dofs of element e . The elemental constraint vector contains the corresponding pressure basis functions integrated over the element. These elemental constraints are then assembled into the vector \mathcal{C} .

Constant viscosity

Choosing $\beta = 0$ yields a constant velocity $\mu(x, y, z) = \exp(1) \simeq 2.718$ (and greatly simplifies the right-hand side) so that

$$\frac{\partial}{\partial x} \mu(x, y, z) = 0 \quad (666)$$

$$\frac{\partial}{\partial y} \mu(x, y, z) = 0 \quad (667)$$

$$\frac{\partial}{\partial z} \mu(x, y, z) = 0 \quad (668)$$

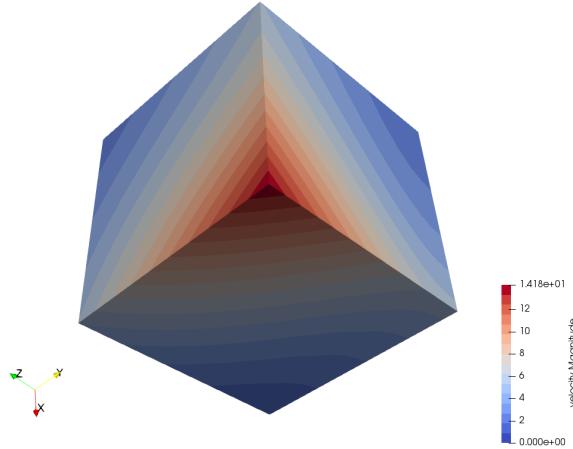
and

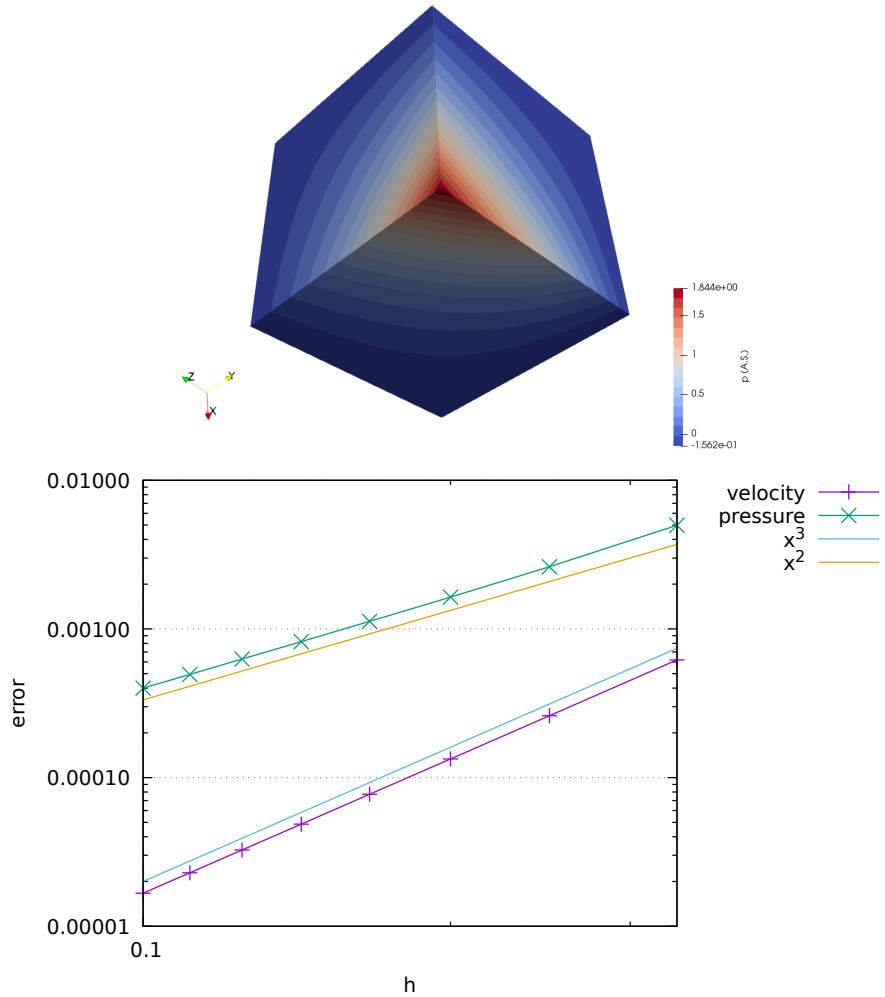
$$\begin{aligned} f_x &= -\frac{\partial p}{\partial x} + 2\mu \frac{\partial}{\partial x} \dot{\epsilon}_{xx} + 2\mu \frac{\partial}{\partial y} \dot{\epsilon}_{xy} + 2\mu \frac{\partial}{\partial z} \dot{\epsilon}_{xz} \\ &= -(yz + 3x^2y^3z) + 2(2 + 6xy) + (1 + 4xy) + (-3 - 10xy) \\ &= -(yz + 3x^2y^3z) + \mu(2 + 6xy) \\ f_y &= -\frac{\partial p}{\partial y} + 2\mu \frac{\partial}{\partial x} \dot{\epsilon}_{xy} + 2\mu \frac{\partial}{\partial y} \dot{\epsilon}_{yy} + 2\mu \frac{\partial}{\partial z} \dot{\epsilon}_{yz} \\ &= -(xz + 3x^3y^2z) + \mu(1 + 2y^2 + 3x^2) + \mu(2(2 + 2x^2) + \mu(-3 - 5x^2)) \\ &= -(xz + 3x^3y^2z) + \mu(2 + 2x^2 + 2y^2) \\ f_z &= -\frac{\partial p}{\partial z} + 2\mu \frac{\partial}{\partial x} \dot{\epsilon}_{xz} + 2\mu \frac{\partial}{\partial y} \dot{\epsilon}_{yz} + 2\mu \frac{\partial}{\partial z} \dot{\epsilon}_{zz} \\ &= -(xy + x^3y^3) + \mu(-10yz) + 0 + 0 \\ &= -(xy + x^3y^3) + \mu(-10yz) \end{aligned}$$

Finally

$$\mathbf{f} = - \begin{pmatrix} yz + 3x^2y^3z \\ xz + 3x^3y^2z \\ xy + x^3y^3 \end{pmatrix} + \mu \begin{pmatrix} 2 + 6xy \\ 2 + 2x^2 + 2y^2 \\ -10yz \end{pmatrix}$$

Note that there seems to be a sign problem with Eq.(26) in [308].





Variable viscosity

The spatial derivatives of the viscosity are then given by

$$\begin{aligned}\frac{\partial}{\partial x} \mu(x, y, z) &= -(1 - 2x)\beta\mu(x, y, z) \\ \frac{\partial}{\partial y} \mu(x, y, z) &= -(1 - 2y)\beta\mu(x, y, z) \\ \frac{\partial}{\partial z} \mu(x, y, z) &= -(1 - 2z)\beta\mu(x, y, z)\end{aligned}$$

and the right-hand side by

$$\begin{aligned}
\mathbf{f} &= - \begin{pmatrix} yz + 3x^2y^3z \\ xz + 3x^3y^2z \\ xy + x^3y^3 \end{pmatrix} + \mu \begin{pmatrix} 2 + 6xy \\ 2 + 2x^2 + 2y^2 \\ -10yz \end{pmatrix} \\
&\quad - (1 - 2x)\beta\mu(x, y, z) \begin{pmatrix} 2\dot{\epsilon}_{xx} \\ 2\dot{\epsilon}_{xy} \\ 2\dot{\epsilon}_{xz} \end{pmatrix} - (1 - 2y)\beta\mu(x, y, z) \begin{pmatrix} 2\dot{\epsilon}_{xy} \\ 2\dot{\epsilon}_{yy} \\ 2\dot{\epsilon}_{yz} \end{pmatrix} - (1 - 2z)\beta\mu(x, y, z) \begin{pmatrix} 2\dot{\epsilon}_{xz} \\ 2\dot{\epsilon}_{yz} \\ 2\dot{\epsilon}_{zz} \end{pmatrix} \\
&= - \begin{pmatrix} yz + 3x^2y^3z \\ xz + 3x^3y^2z \\ xy + x^3y^3 \end{pmatrix} + \mu \begin{pmatrix} 2 + 6xy \\ 2 + 2x^2 + 2y^2 \\ -10yz \end{pmatrix} \\
&\quad - (1 - 2x)\beta\mu \begin{pmatrix} 2 + 4x + 2y + 6x^2y \\ x + y + 2xy^2 + x^3 \\ -3z - 10xyz \end{pmatrix} - (1 - 2y)\beta\mu \begin{pmatrix} x + y + 2xy^2 + x^3 \\ 2 + 2x + 4y + 4x^2y \\ -3z - 5x^2z \end{pmatrix} - (1 - 2z)\beta\mu \begin{pmatrix} -3z - 10xyz \\ -3z - 5x^2z \\ -4 - 6x - 6y - 10xz \end{pmatrix}
\end{aligned}$$

Note that at $(x, y, z) = (0, 0, 0)$, $\mu = \exp(1)$, and at $(x, y, z) = (0.5, 0.5, 0.5)$, $\mu = \exp(1 - 3\beta/4)$ so that the maximum viscosity ratio is given by

$$\mu^* = \frac{\exp(1 - 3\beta/4)}{\exp(1)} = \exp(-3\beta/4)$$

By varying β between 1 and 22 we can get up to 7 orders of magnitude viscosity difference.

features

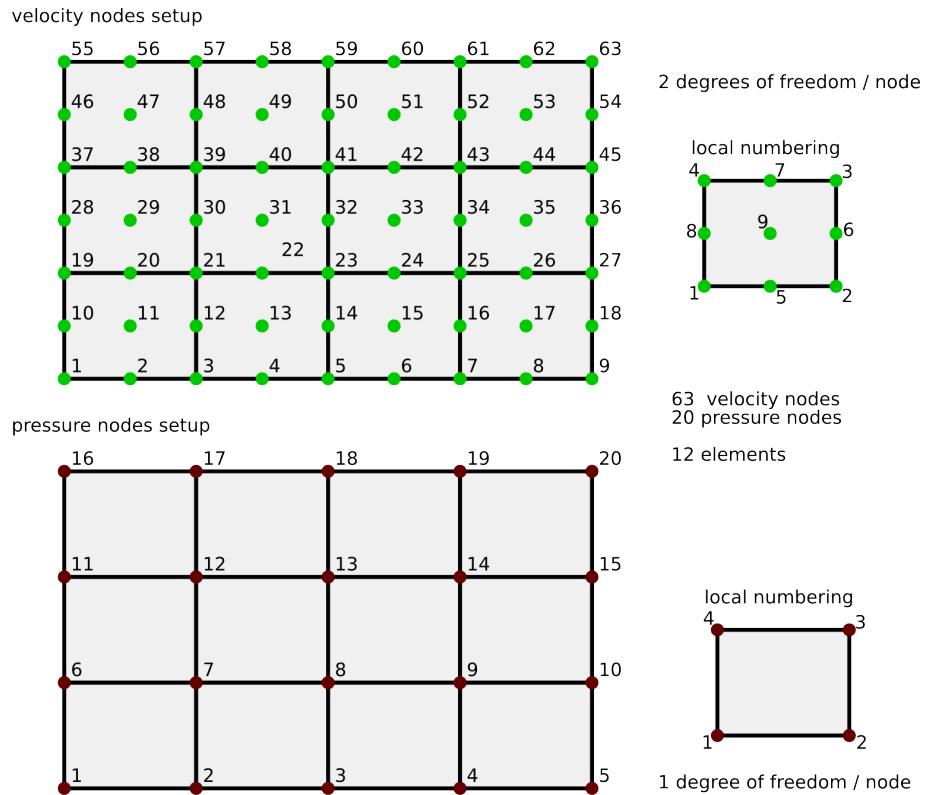
- $Q_1 \times P_0$ element
- incompressible flow
- saddle point system
- Dirichlet boundary conditions (free-slip)
- direct solver
- isothermal
- non-isoviscous
- 3D
- elemental b.c.
- analytical solution

Stone 18: solving the full saddle point problem with $Q_2 \times Q_1$ elements

The details of the numerical setup are presented in Section ??.

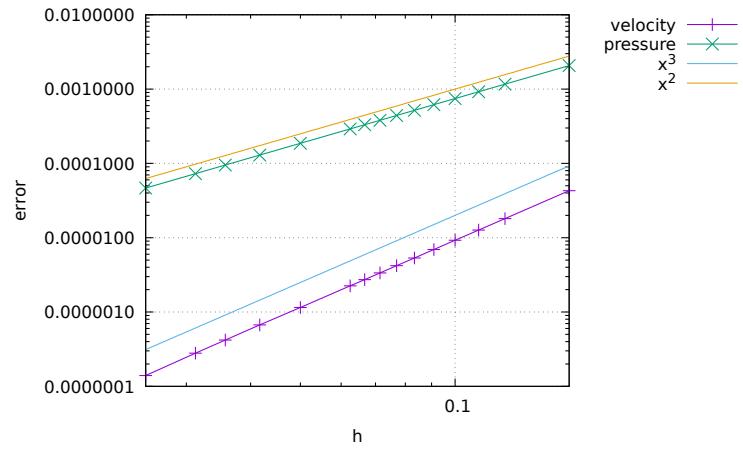
Each element has $m_V = 9$ vertices so in total $ndof_V \times m_V = 18$ velocity dofs and $ndof_P \times m_P = 4$ pressure dofs. The total number of velocity dofs is therefore $NfemV = nnp \times ndofV$ while the total number of pressure dofs is $NfemP = nel$. The total number of dofs is then $Nfem = NfemV + NfemP$.

As a consequence, matrix \mathbb{K} has size $NfemV, NfemV$ and matrix \mathbb{G} has size $NfemV, NfemP$. Vector f is of size $NfemV$ and vector h is of size $NfemP$.



features

- $Q_2 \times Q_1$ element
- incompressible flow
- mixed formulation
- Dirichlet boundary conditions (no-slip)
- isothermal
- isoviscous
- analytical solution



Stone 19: solving the full saddle point problem with $Q_3 \times Q_2$ elements

The details of the numerical setup are presented in Section ??.

Each element has $m_V = 16$ vertices so in total $ndof_V \times m_V = 32$ velocity dofs and $ndof_P \times m_P = 9$ pressure dofs. The total number of velocity dofs is therefore $NfemV = nnp \times ndofV$ while the total number of pressure dofs is $NfemP = nel$. The total number of dofs is then $Nfem = NfemV + NfemP$.

As a consequence, matrix \mathbb{K} has size $NfemV, NfemV$ and matrix \mathbb{G} has size $NfemV, NfemP$. Vector f is of size $NfemV$ and vector h is of size $NfemP$.

```

60====61====62====63====64====65====66====67====68====70
||          ||          ||          ||
50    51    52    53    54    55    56    57    58    59
||          ||          ||          ||
40    41    42    43    44    45    46    47    48    49
||          ||          ||          ||
30====31====32====33====34====35====36====37====38====39
||          ||          ||          ||
20    21    22    23    24    25    26    27    28    29
||          ||          ||          ||
10    11    12    13    14    15    16    17    18    19
||          ||          ||          ||
00====01====02====03====04====05====06====07====08====09

```

Example of 3x2 mesh. $n_{nx}=10$, $n_{ny}=7$, $nnp=70$, $nelx=3$, $nely=2$, $nel=6$

```

12====13====14====15          06=====07=====08
||  ||  ||  ||          ||  ||  ||  ||
08====09====10====11          ||  ||  ||  ||
||  ||  ||  ||          03=====04=====05
04====05====06====07          ||  ||  ||  ||
||  ||  ||  ||          00=====01=====02
00====01====02====03

```

Velocity (Q3)

```

(r,s)_{00}=(-1,-1)          (r,s)_{00}=(-1,-1)
(r,s)_{01}=(-1/3,-1)         (r,s)_{01}=(0,-1)
(r,s)_{02}=(+1/3,-1)         (r,s)_{02}=(+1,-1)
(r,s)_{03}=(+1,-1)           (r,s)_{03}=(-1,0)
(r,s)_{04}=(-1,-1/3)         (r,s)_{04}=(0,0)
(r,s)_{05}=(-1/3,-1/3)       (r,s)_{05}=(+1,0)
(r,s)_{06}=(+1/3,-1/3)       (r,s)_{06}=(-1,+1)
(r,s)_{07}=(+1,-1/3)          (r,s)_{07}=(0,+1)
(r,s)_{08}=(-1,+1/3)          (r,s)_{08}=(+1,+1)
(r,s)_{09}=(-1/3,+1/3)
(r,s)_{10}=(+1/3,+1/3)
(r,s)_{11}=(+1,+1/3)
(r,s)_{12}=(-1,+1)
(r,s)_{13}=(-1/3,+1)
(r,s)_{14}=(+1/3,+1)
(r,s)_{15}=(+1,+1)

```

Pressure (Q2)

```

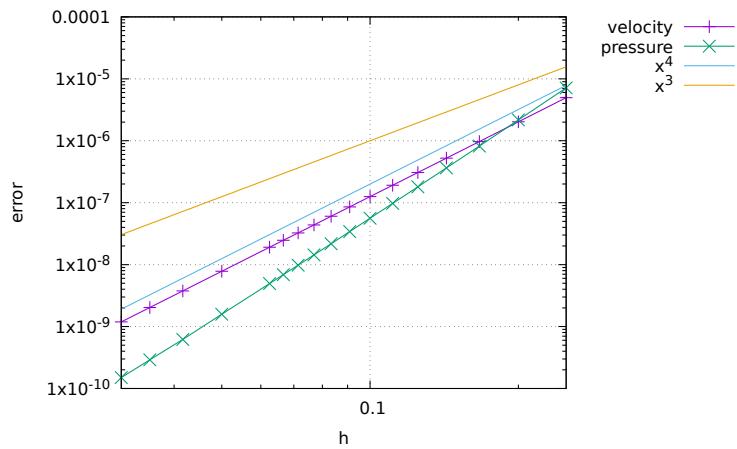
(r,s)_{00}=(-1,-1)
(r,s)_{01}=(0,-1)
(r,s)_{02}=(+1,-1)
(r,s)_{03}=(-1,0)
(r,s)_{04}=(0,0)
(r,s)_{05}=(+1,0)
(r,s)_{06}=(-1,+1)
(r,s)_{07}=(0,+1)
(r,s)_{08}=(+1,+1)

```

Write about 4 point quadrature.

features

- $Q_3 \times Q_2$ element
- incompressible flow
- mixed formulation
- isothermal
- isoviscous
- analytical solution



velocity error rate is cubic, pressure superconvergent since the pressure field is quadratic and therefore lies into the Q_2 space.

Stone 20: the Busse benchmark

This three-dimensional benchmark was first proposed by [313]. It has been subsequently presented in [1712, 1794, 12, 1434, 488, 1119]. We here focus on Case 1 of [313]: an isoviscous bimodal convection experiment at $Ra = 3 \cdot 10^5$.

The domain is of size $a \times b \times h$ with $a = 1.0079h$, $b = 0.6283h$ with $h = 2700\text{km}$. It is filled with a Newtonian fluid characterised by $\rho_0 = 3300\text{kg.m}^{-3}$, $\alpha = 10^{-5}\text{K}^{-1}$, $\mu = 8.0198 \times 10^{23}\text{Pa.s}$, $k = 3.564\text{W.m}^{-1}\text{.K}^{-1}$, $C_p = 1080\text{J.K}^{-1}\text{.kg}^{-1}$. The gravity vector is set to $\mathbf{g} = (0, 0, -10)^T$. The temperature is imposed at the bottom ($T = 3700^\circ\text{C}$) and at the top ($T = 0^\circ\text{C}$).

Note that using these numbers (as provided in the original paper), we arrive at $\text{Ra}=29967.01$, which is not exactly $3 \cdot 10^5$ as announced. Also, the heat diffusivity $\kappa = k/\rho_0 C_p$ is *exactly* 10^{-6} .

The various measurements presented in [313] are listed hereafter:

- The Nusselt number Nu computed at the top surface following Eq. (634):

$$Nu = L_z \frac{\int \int_{z=L_z} \frac{\partial T}{\partial y} dx dy}{\int \int_{z=0} T dx dy}$$

- the root mean square velocity v_{rms} and the temperature mean square velocity T_{rms}
- The vertical velocity w and temperature T at points $\mathbf{x}_1 = (0, 0, L_z/2)$, $\mathbf{x}_2 = (L_x, 0, L_z/2)$, $\mathbf{x}_3 = (0, L_y, L_z/2)$ and $\mathbf{x}_4 = (L_x, L_y, L_z/2)$;
- the vertical component of the heat flux Q at the top surface at all four corners.

Methodology

In what follows I highlight a few important points which are key to understanding how the code is put together and works.

```

load needed modules and functions
define parameters
build V grid (xV,yV,zV)
build V connectivity (iconV)
define b.c. for velocity (bc_fixV,bc_valV)
build T grid (xT,yT,zT)
build T connectivity (iconT)
define b.c. for temperature (bc_fixT,bc_valT)
initial temperature field
.-----> istep -----
| build K,G,f,h
| assemble them in A,rhs
| solve
| split solution vector in u,v,w,p
| {u,v,w}=relax*{u,v,w}+(1-relax)*{u,v,w}
| compute vrms
| build A, rhs for temperature
| solve for temperature T
| T=relax*T+(1-relax)*T
| compute elemental strainrate
| compute nodal strainrate
| compute nodal pressure
| measure V and T at mid side edges, Nu ...
| export to vtu and ascii files
.-----<-----.

```

I first load the shape functions which are in two separate files:

```

from shape_functionsV import NNV, dNNVdr, dNNVds, dNNVdt
from shape_functionsT import NNT, dNNTdr, dNNTds, dNNTdt

```

There are NV=nnx*nny*nnz velocity nodes and NT=NV temperature nodes.

The velocity grid is built: xV, yV, zV, iconV, and these are copied in xT, yT, zT and iconT for the temperature grid.

The initial temperature field is built as follows:

```

for i in range(0,NT):
    T[i] = (Temperature2-Temperature1)/Lz*zT[i]+Temperature1 \
        + 100*(np.pi*cos(np.pi*xT[i]/Lx) + np.pi*cos(np.pi*yT[i]/Ly))*np.sin(np.pi*zT[i]/Lz)

```

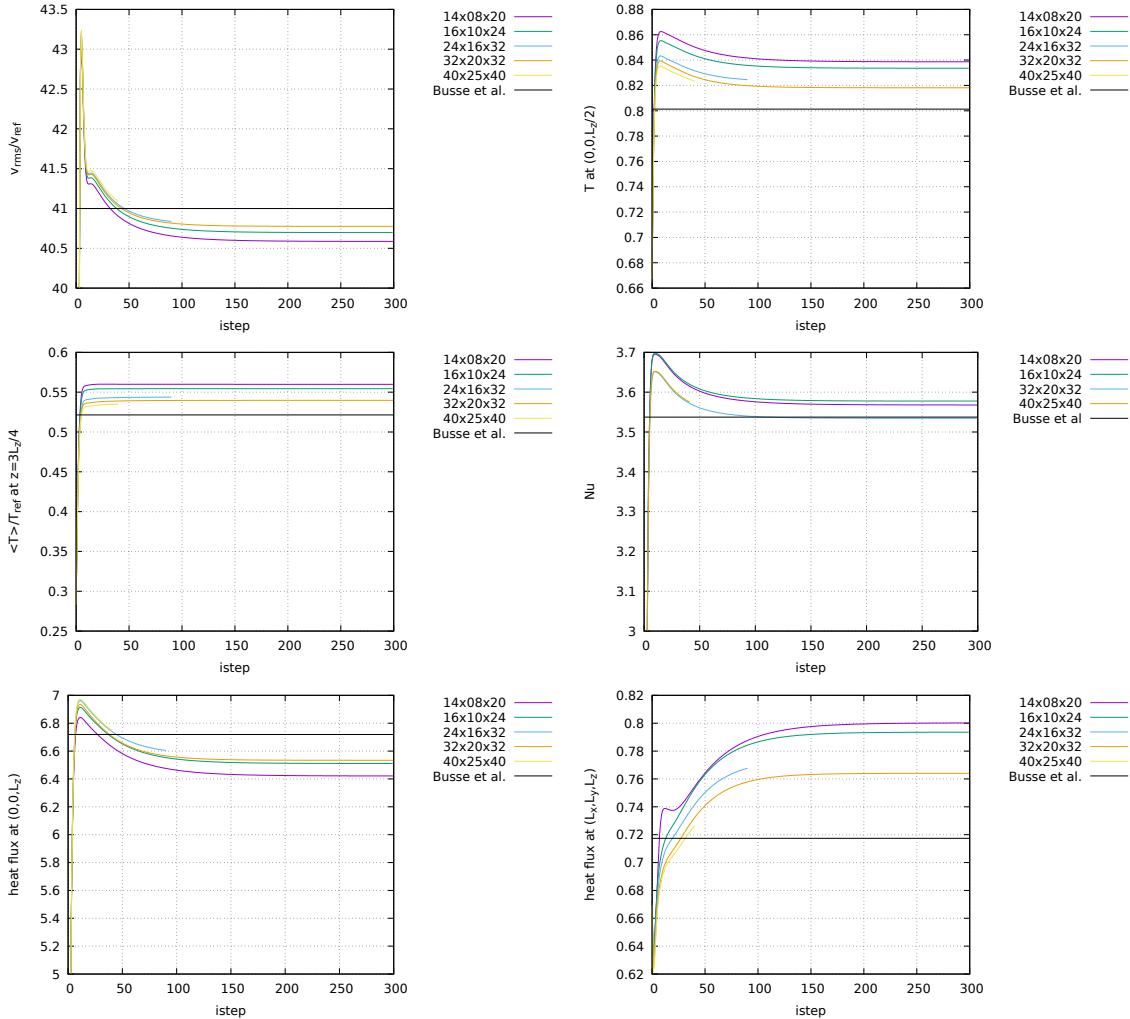
The C matrix of Eq. 310 is then built:

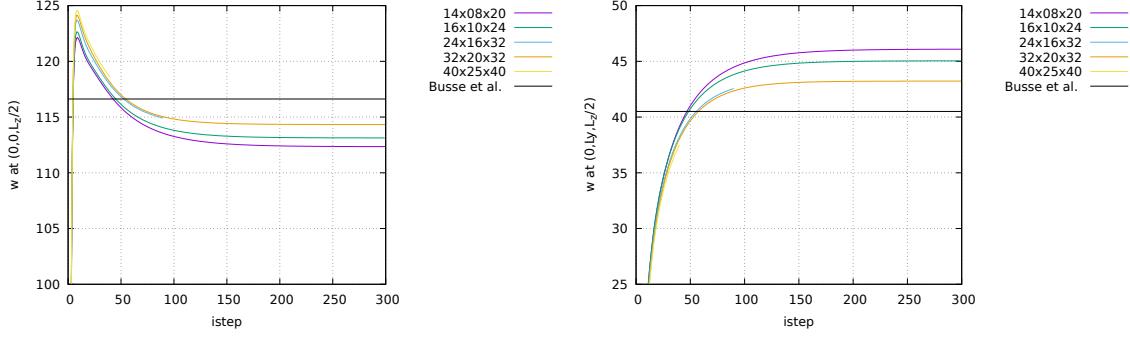
```

c_mat = np.array([[2,0,0,0,0,0], \
                  [0,2,0,0,0,0], \
                  [0,0,2,0,0,0], \
                  [0,0,0,1,0,0], \
                  [0,0,0,0,1,0], \
                  [0,0,0,0,0,1]], dtype=np.float64)

```

Results





The reported values for Busse et al. in the following table are taken from Table 3 of [313]. The reported values for fieldstone are adimensionalised by means of a reference temperature (3700K), a reference lengthscale 2700km, and a reference time $L_z^2/\kappa \sim 7.29e + 18$ s. The steady state is arrived at by solving the steady state Stokes and temperature equations with a relaxation parameter of 1/2.

Mesh size	ASPECT			Busse et al [313]	fieldstone
	$L_z/24$	$L_z/32$	$L_z/48$	(best results)	$14 \times 8 \times 20$
Nu	3.5539	3.5447	3.5397	3.5374 ± 0.0005	
v_{rms}	40.997	40.999	40.999	40.999 ± 0.004	
$\langle T \rangle$ at $0.75 * L_z$	0.52148	0.52148	0.52148	0.52148 ± 0.00003	
$w(0, 0, L_z/2)$	116.605	116.618	116.623	116.625 ± 0.030	
$w(L_x, 0, L_z/2)$	-	-	-	-	
$w(L_x, L_y, L_z/2)$	-	-	-	-	
$w(0, L_y, L_z/2)$				40.500 ± 0.030	
$T(0, 0, L_z/2)$	0.80126	0.80128	0.80129	0.80130 ± 0.00005	
$T(L_x, 0, L_z/2)$	-	-	-	-	
$T(L_x, L_y, L_z/2)$	-	-	-	-	
$T(0, L_y, L_z/2)$				0.61876 ± 0.00005	
$dTdz(0, 0, L_z)$	6.7679	6.7357	6.7189	6.7127 ± 0.0500	
$dTdz(L_x, 0, L_z)$				1.5080 ± 0.0500	
$dTdz(L_x, L_y, L_z)$	0.7237	0.7205	0.7174	0.7140 ± 0.0500	
$dTdz(0, L_y, L_z)$				3.1740 ± 0.0500	

THIS IS NOT FINISHED: I need to run the model at higher resolutions, which will take a few days.

Stone 22: The stabilised $Q_1 \times Q_1$ element

The details of the numerical setup are presented in Section 8.6.

We wish to use $Q_1 \times Q_1$ element, which, unless stabilised, violates the LBB stability condition and therefore is unusable. Stabilisation can be of two types: least-squares [533, 1750, 1050, 182], or by means of an additional term in the weak form as first introduced in [529, 181], which is appealing since there is no explicit stabilisation parameter. It is further analysed in [1424, 1179, 943, 1643, 810]. Note that an equal-order velocity-pressure formulation that does not exhibit spurious pressure modes (without stabilisation) has been presented in [1573].

This element corresponds to bilinear velocities, bilinear pressure (equal order interpolation for both velocity and pressure) which is very convenient in terms of data structures since all dofs are colocated.

In geodynamics, it is used in the Rhea code [1688, 308] and in Gale [45]. It is also used in [1169] in its stabilised form, in conjunction with AMR. This element is quickly discussed at page 217 of Volker John's book [998].

The stabilisation term \mathbb{C} enters the Stokes matrix in the (2,2) position:

$$\begin{pmatrix} \mathbb{K} & \mathbb{G} \\ \mathbb{G}^T & -\mathbb{C} \end{pmatrix} \cdot \begin{pmatrix} \mathcal{V} \\ \mathcal{P} \end{pmatrix} = \begin{pmatrix} f \\ h \end{pmatrix}$$

The purpose of the \mathbb{C} term is to stabilise the linear system. It is given by:

$$\mathbb{C}(p, q) = \sum_e \int_{\Omega_e} \frac{1}{\eta} (p - \Pi p)(q - \Pi q) d\Omega$$

where Π is the L^2 -projection onto the space of element-wise constant functions:

$$\Pi p = \frac{1}{|\Omega_e|} \int_{\Omega_e} p d\Omega$$

Because of the stabilisation matrix \mathbb{C} , the numerical solution satisfies the incompressibility condition only approximately. Local mesh refinement helps to control these unwanted effects [307, 308]. Since \mathbb{K} and \mathbb{C} are symmetric matrices, the Stokes system is then an indefinite symmetric system. The Schur complement matrix \mathbb{S} is then given by

$$\mathbb{S} = \mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot \mathbb{G} + \mathbb{C}$$

One can further expand the above expression for the \mathbb{C} term:

$$\begin{aligned} \mathbb{C}(p, q) &= \sum_e \int_{\Omega_e} \frac{1}{\eta} (p - \Pi p)(q - \Pi q) d\Omega \\ &= \sum_e \int_{\Omega_e} \frac{1}{\eta} [pq - (\Pi p)q - (\Pi q)p + (\Pi p)(\Pi q)] d\Omega \\ &= \sum_e \frac{1}{\eta_e} \left[\int_{\Omega_e} pq d\Omega - \int_{\Omega_e} (\Pi p)qd\Omega - \int_{\Omega_e} (\Pi q)pd\Omega + \int_{\Omega_e} (\Pi p)(\Pi q)d\Omega \right] \\ &= \sum_e \frac{1}{\eta_e} \left[\int_{\Omega_e} pq d\Omega - (\Pi p) \int_{\Omega_e} q d\Omega - (\Pi q) \int_{\Omega_e} p d\Omega + (\Pi p)(\Pi q) \int_{\Omega_e} d\Omega \right] \\ &= \sum_e \frac{1}{\eta_e} \left[\int_{\Omega_e} pq d\Omega - (\Pi p)|\Omega_e|(\Pi q) - (\Pi q)|\Omega_e|(\Pi p) + (\Pi p)(\Pi q)|\Omega_e| \right] \\ &= \sum_e \frac{1}{\eta_e} \left[\int_{\Omega_e} pq d\Omega - |\Omega_e|(\Pi p)(\Pi q) \right] \end{aligned} \tag{670}$$

where we have used the fact that on each element Πp^h is constant. The left term will obviously yield a Q_1 mass matrix (scaled by the elemental viscosities). Note that this approach is not used in practice as we'll see hereafter.

The pressure inside an element is given by

$$p^h(\vec{x}) = \sum_k N_k^p(\vec{x}) p_k$$

so that

$$\Pi p^h = \frac{1}{|\Omega_e|} \int_{\Omega_e} \sum_k N_k^p p_k d\Omega = \sum_k \left(\underbrace{\frac{1}{|\Omega_e|} \int_{\Omega_e} N_k^p d\Omega}_{\tilde{N}_k^p} \right) p_k \quad (671)$$

and then

$$p^h - \Pi p^h = \sum_k N_k^p(\vec{x}) p_k - \sum_k \tilde{N}_k^p p_k = \sum_k (N_k^p(\vec{x}) - \tilde{N}_k^p) p_k$$

The algorithm is straightforward and as follows: In the loop over elements, a) Compute the average of each shape function $N_k^p(\vec{x})$ over the element; b) Subtract this average to the shape function; c) Build mass matrix with modified/offset shape functions (taking in account the viscosity).

In the case of rectangular elements of size (h_x, h_y) , \tilde{N}_k^p simplifies even more:

$$\tilde{N}_k^p = \frac{1}{|\Omega_e|} \int_{\Omega_e} N_k^p(\vec{x}) d\Omega = \frac{1}{h_x h_y} \frac{h_x h_y}{4} \int_{-1}^{+1} \int_{-1}^{+1} N_k^p(r, s) dr ds = \frac{1}{4} \int_{-1}^{+1} \int_{-1}^{+1} N_k^p(r, s) dr ds \quad (672)$$

It is easy to show that the average of the Q_1 shape functions over the reference element is 1, so that $\tilde{N}_k^p = 1/4$. This explains why in the code we have:

```
Navrg = np.zeros(m, dtype=np.float64)
Navrg[0]=0.25
Navrg[1]=0.25
Navrg[2]=0.25
Navrg[3]=0.25
```

This also means that $\Pi p^h = (p_1 + p_2 + p_3 + p_4)/4$, i.e. the projected pressure is the mean of the vertex values. It follows, as shown on p.244 of [573] that the elemental \mathbb{C} matrix is (omitting the viscosity term)

$$\mathbb{C}_{el} = \mathbb{M}_{el} - \vec{q}^T \vec{q} |\Omega_e| \quad \vec{q} = \left(\frac{1}{4}, \frac{1}{4}, \frac{1}{4}, \frac{1}{4} \right)$$

The nullspace of \mathbb{C} consists of constant vectors, i.e. $\vec{1} \in \text{null}(\mathbb{C})$ which means that the assembled stabilisation operator is consistent.

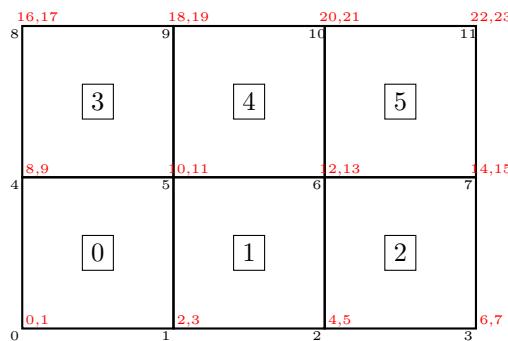
The elemental \mathbb{C}_{el} matrix is then computed like a mass matrix, although with modified shape function vectors. Inside the loop over quadrature points, we do:

```
Nvect[0:,0:m]=N[0:m]-Navrg[0:m]
C_el+=Nvect.T.dot(Nvect)*jacob*weightq/viscosity(xq,yq,case)
```

It is then assembled inside the big FEM matrix

```
for k1 in range(0,m):
    for k2 in range(0,m):
        C_mat[icon[k1, ie1], icon[k2, ie1]]+=C_el[k1, k2]
```

Non-zero pattern of the \mathbb{G} matrix: Let us take a simple example: a 3x2 element grid.



The \mathbb{K} matrix is of size $NfemV \times NfemV$ with $NfemV = dofV \times nnp = 2 \times 12 = 24$. The \mathbb{G} matrix is of size $NfemV \times NfemP$ with $NfemP = dofP \times nnp = 1 \times 12 = 12$. The \mathbb{C} matrix is of size $NfemP \times NfemP$.

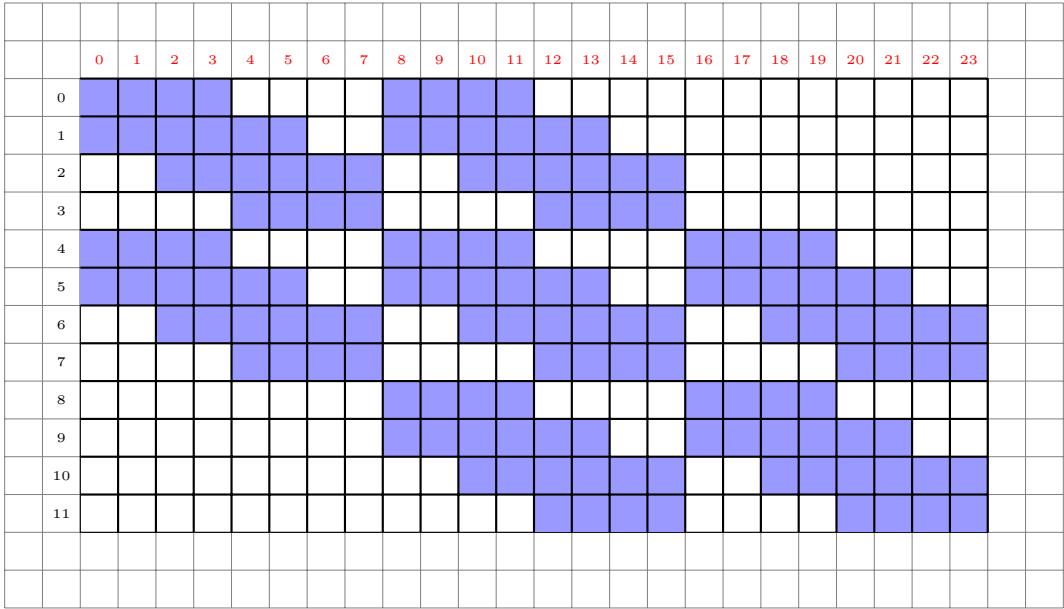
A corner pdof sees 4 vdofs, a side pdof sees 12 vdofs and an inside pdof sees 18 vdofs, so that the total number of nonzeros in \mathbb{G} can be computed as follows:

$$NZ_{\mathbb{G}} = \underbrace{4}_{\text{corners}} + \underbrace{2(nnx - 2) * 12}_{\text{2hor.sides}} + \underbrace{2(nny - 2) * 12}_{\text{2vert.sides}} + \underbrace{(nnx - 2)(nny - 2) * 18}_{\text{insidenodes}}$$

Concretely,

- pdof #0 sees vdofs 0,1,2,3,8,9,10,11
- pdof #1 sees vdofs 0,1,2,3,4,5,8,9,10,11,12,13
- pdof #5 sees vdofs 0,1,2,3,4,5,8,9,10,11,12,13,16,17,18,19,20,21

so that the \mathbb{G}^T matrix non-zero structure then is as follows:



Non-zero pattern of the \mathbb{C} matrix: Let us take a simple example: a 3x2 element grid.

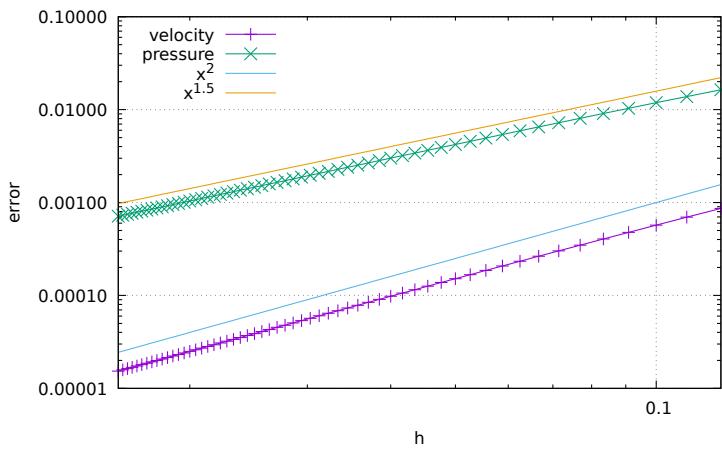
`finish structure of C matrix for q1q1`

We impose $\int pdV = 0$ which means that the following constraint is added to the Stokes matrix:

$$\begin{pmatrix} \mathbb{K} & \mathbb{G} & 0 \\ \mathbb{G}^T & \mathbb{C} & \mathbb{L} \\ 0 & \mathbb{L}^T & 0 \end{pmatrix} \cdot \begin{pmatrix} \mathcal{V} \\ \mathcal{P} \\ \lambda \end{pmatrix} = \begin{pmatrix} f \\ h \\ 0 \end{pmatrix}$$

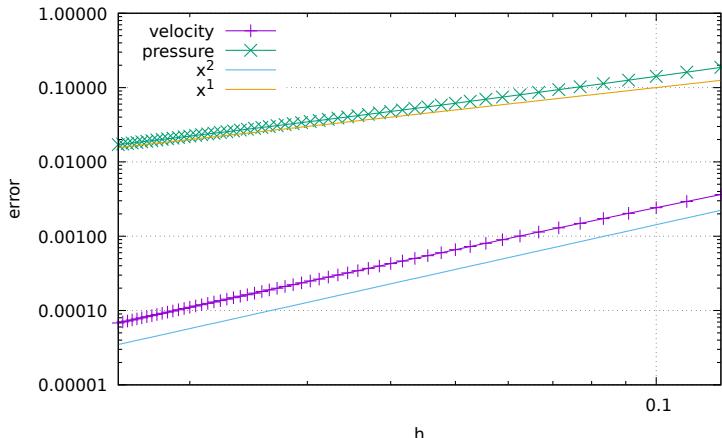
The Donea & Huerta benchmark

As in [533] we solve the benchmark problem presented in section 8.6.1.



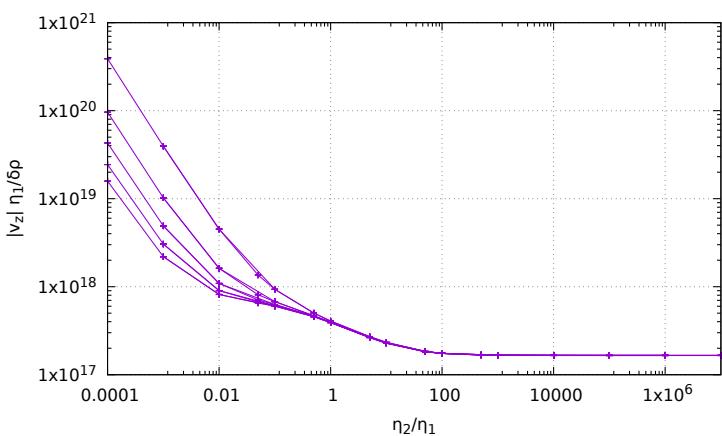
The Dohrmann & Bochev benchmark

As in [529] we solve the benchmark problem presented in section 8.6.2.



The falling block experiment

The setup is described in [1760].



Stone 23: compressible flow (1) - analytical benchmark

This work is part of the MSc thesis of T. Weir (2018).

We first start with an isothermal Stokes flow, so that we disregard the heat transport equation and the equations we wish to solve are simply:

$$-\nabla \cdot \left[2\eta \left(\dot{\epsilon}(\mathbf{v}) - \frac{1}{3}(\nabla \cdot \mathbf{v})\mathbf{1} \right) \right] + \nabla p = \rho \mathbf{g} \quad \text{in } \Omega, \quad (673)$$

$$\nabla \cdot (\rho \mathbf{v}) = 0 \quad \text{in } \Omega \quad (674)$$

The second equation can be rewritten $\nabla \cdot (\rho \mathbf{v}) = \rho \nabla \cdot \mathbf{v} + \mathbf{v} \cdot \nabla \rho = 0$ or,

$$\nabla \cdot \mathbf{v} + \frac{1}{\rho} \mathbf{v} \cdot \nabla \rho = 0$$

Note that this presupposes that the density is not zero anywhere in the domain.

We use a mixed formulation and therefore keep both velocity and pressure as unknowns. We end up having to solve the following system:

$$\begin{pmatrix} \mathbb{K} & \mathbb{G} \\ \mathbb{G}^T + \mathbb{Z} & 0 \end{pmatrix} \cdot \begin{pmatrix} \mathcal{V} \\ \mathcal{P} \end{pmatrix} = \begin{pmatrix} f \\ h \end{pmatrix} \quad \text{or,} \quad \mathbb{A} \cdot X = rhs$$

Where \mathbb{K} is the stiffness matrix, \mathbb{G} is the discrete gradient operator, \mathbb{G}^T is the discrete divergence operator, \mathcal{V} the velocity vector, \mathcal{P} the pressure vector. Note that the term $\mathbb{Z}\mathcal{V}$ derives from term $\mathbf{v} \cdot \nabla \rho$ in the continuity equation.

Each block \mathbb{K} , \mathbb{G} , \mathbb{Z} and vectors f and h are built separately in the code and assembled into the matrix \mathbb{A} and vector rhs afterwards. \mathbb{A} and rhs are then passed to the solver. We will see later that there are alternatives to solve this approach which do not require to build the full Stokes matrix \mathbb{A} .

Remark: the term $\mathbb{Z}\mathcal{V}$ is often put in the rhs (i.e. added to h) so that the matrix \mathbb{A} retains the same structure as in the incompressible case. This is indeed how it is implemented in ASPECT. This however requires more work since the rhs depends on the solution and some form of iterations is needed.

In the case of a compressible flow the strain rate tensor and the deviatoric strain rate tensor are no more equal (since $\nabla \cdot \mathbf{v} \neq 0$). The deviatoric strainrate tensor is given by⁵⁹

$$\dot{\epsilon}^d(\mathbf{v}) = \dot{\epsilon}(\mathbf{v}) - \frac{1}{3} Tr(\dot{\epsilon})\mathbf{1} = \dot{\epsilon}(\mathbf{v}) - \frac{1}{3}(\nabla \cdot \mathbf{v})\mathbf{1}$$

In that case:

$$\dot{\epsilon}_{xx}^d = \frac{\partial u}{\partial x} - \frac{1}{3} \left(\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} \right) = \frac{2}{3} \frac{\partial u}{\partial x} - \frac{1}{3} \frac{\partial v}{\partial y} \quad (675)$$

$$\dot{\epsilon}_{yy}^d = \frac{\partial v}{\partial y} - \frac{1}{3} \left(\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} \right) = -\frac{1}{3} \frac{\partial u}{\partial x} + \frac{2}{3} \frac{\partial v}{\partial y} \quad (676)$$

$$2\dot{\epsilon}_{xy}^d = \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \quad (677)$$

and then

$$\dot{\epsilon}^d(\mathbf{v}) = \begin{pmatrix} \frac{2}{3} \frac{\partial u}{\partial x} - \frac{1}{3} \frac{\partial v}{\partial y} & \frac{1}{2} \frac{\partial u}{\partial y} + \frac{1}{2} \frac{\partial v}{\partial x} \\ \frac{1}{2} \frac{\partial u}{\partial y} + \frac{1}{2} \frac{\partial v}{\partial x} & -\frac{1}{3} \frac{\partial u}{\partial x} + \frac{2}{3} \frac{\partial v}{\partial y} \end{pmatrix}$$

From $\vec{\tau} = 2\eta \vec{\epsilon}^d$ we arrive at:

$$\begin{pmatrix} \tau_{xx} \\ \tau_{yy} \\ \tau_{xy} \end{pmatrix} = 2\eta \begin{pmatrix} \dot{\epsilon}_{xx}^d \\ \dot{\epsilon}_{yy}^d \\ \dot{\epsilon}_{xy}^d \end{pmatrix} = 2\eta \begin{pmatrix} 2/3 & -1/3 & 0 \\ -1/3 & 2/3 & 0 \\ 0 & 0 & 1/2 \end{pmatrix} \cdot \begin{pmatrix} \frac{\partial u}{\partial x} \\ \frac{\partial v}{\partial y} \\ \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \end{pmatrix} = \eta \begin{pmatrix} 4/3 & -2/3 & 0 \\ -2/3 & 4/3 & 0 \\ 0 & 0 & 1 \end{pmatrix} \cdot \begin{pmatrix} \frac{\partial u}{\partial x} \\ \frac{\partial v}{\partial y} \\ \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \end{pmatrix}$$

or,

$$\vec{\tau} = \mathbf{C}_\eta \mathbf{B} \mathbf{V}$$

⁵⁹See the ASPECT manual for a justification of the 3 value in the denominator in 2D and 3D.

In order to test our implementation we have created a few manufactured solutions:

- benchmark #1 (ibench=1): Starting from a density profile of:

$$\rho(x, y) = xy \quad (678)$$

We derive a velocity given by:

$$v_x(x, y) = \frac{C_x}{x}, v_y(x, y) = \frac{C_y}{y} \quad (679)$$

With $g_x(x, y) = \frac{1}{x}$ and $g_y(x, y) = \frac{1}{y}$, this leads us to a pressure profile:

$$p = -\eta \left(\frac{4C_x}{3x^2} + \frac{4C_y}{3y^2} \right) + xy + C_0 \quad (680)$$

This gives us a strain rate:

$$\dot{\epsilon}_{xx} = \frac{-C_x}{x^2} \quad \dot{\epsilon}_{yy} = \frac{-C_y}{y^2} \quad \dot{\epsilon}_{xy} = 0$$

In what follows, we choose $\eta = 1$ and $C_x = C_y = 1$ and for a unit square domain $[1 : 2] \times [1 : 2]$ we compute C_0 so that the pressure is normalised to zero over the whole domain and obtain $C_0 = -1$.

- benchmark #2 (ibench=2): Starting from a density profile of:

$$\rho = \cos(x) \cos(y) \quad (681)$$

We derive a velocity given by:

$$v_x = \frac{C_x}{\cos(x)}, v_y = \frac{C_y}{\cos(y)} \quad (682)$$

With $g_x = \frac{1}{\cos(y)}$ and $g_y = \frac{1}{\cos(x)}$, this leads us to a pressure profile:

$$p = \eta \left(\frac{4C_x \sin(x)}{3 \cos^2(x)} + \frac{4C_y \sin(y)}{3 \cos^2(y)} \right) + (\sin(x) + \sin(y)) + C_0 \quad (683)$$

$$\dot{\epsilon}_{xx} = C_x \frac{\sin(x)}{\cos^2(x)} \quad \dot{\epsilon}_{yy} = C_y \frac{\sin(y)}{\cos^2(y)} \quad \dot{\epsilon}_{xy} = 0$$

We choose $\eta = 1$ and $C_x = C_y = 1$. The domain is the unit square $[0 : 1] \times [0 : 1]$ and we obtain C_0 as before and obtain

$$C_0 = 2 - 2 \cos(1) + 8/3 \left(\frac{1}{\cos(1)} - 1 \right) \simeq 3.18823730$$

(thank you WolframAlpha)

- benchmark #3 (ibench=3)
- benchmark #4 (ibench=4)
- benchmark #5 (ibench=5)

features

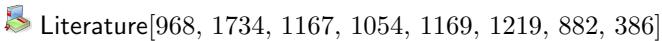
- $Q_1 \times P_0$ element
- incompressible flow
- mixed formulation
- Dirichlet boundary conditions (no-slip)
- isothermal
- isoviscous
- analytical solution
- pressure smoothing

ToDo:

- pbs with odd vs even number of elements
- q is 'fine' everywhere except in the corners - revisit pressure smoothing paper?
- redo A v d Berg benchmark (see Tom Weir thesis)

Stone 24: compressible flow (2) - convection box

This work is part of the MSc thesis of T. Weir (2018).



Literature[968, 1734, 1167, 1054, 1169, 1219, 882, 386]

The physics

Let us start with some thermodynamics. Every material has an equation of state. The equilibrium thermodynamic state of any material can be constrained if any two state variables are specified. Examples of state variables include the pressure p and specific volume $\nu = 1/\rho$, as well as the temperature T .

After linearisation, the density depends on temperature and pressure as follows:

$$\rho(T, p) = \rho_0 ((1 - \alpha(T - T_0) + \beta_T p)$$

where α is the coefficient of thermal expansion, also called thermal expansivity:

$$\alpha = -\frac{1}{\rho} \left(\frac{\partial \rho}{\partial T} \right)_p$$

α is the percentage increase in volume of a material per degree of temperature increase; the subscript p means that the pressure is held fixed.

β_T is the isothermal compressibility of the fluid, which is given by

$$\beta_T = \frac{1}{K} = \frac{1}{\rho} \left(\frac{\partial \rho}{\partial P} \right)_T$$

with K the bulk modulus. Values of $\beta_T = 10^{-12} - 10^{-11} \text{ Pa}^{-1}$ are reasonable for Earth's mantle, with values decreasing by about a factor of 5 between the shallow lithosphere and core-mantle boundary. This is the percentage increase in density per unit change in pressure at constant temperature. Both the coefficient of thermal expansion and the isothermal compressibility can be obtained from the equation of state.

The full set of equations we wish to solve is given by

$$-\nabla \cdot [2\eta\dot{\epsilon}^d(\mathbf{v})] + \nabla p = \rho_0 ((1 - \alpha(T - T_0) + \beta_T p) \mathbf{g} \quad \text{in } \Omega \quad (684)$$

$$\nabla \cdot \mathbf{v} + \frac{1}{\rho} \mathbf{v} \cdot \nabla \rho = 0 \quad \text{in } \Omega \quad (685)$$

$$\rho C_p \left(\frac{\partial T}{\partial t} + \mathbf{v} \cdot \nabla T \right) - \nabla \cdot k \nabla T = \rho H + 2\eta\dot{\epsilon}^d : \dot{\epsilon}^d + \alpha T \left(\frac{\partial p}{\partial t} + \mathbf{v} \cdot \nabla p \right) \quad \text{in } \Omega, \quad (686)$$

Note that this presupposes that the density is not zero anywhere in the domain.

The numerics

We use a mixed formulation and therefore keep both velocity and pressure as unknowns. We end up having to solve the following system:

$$\begin{pmatrix} \mathbb{K} & \mathbb{G} + \mathbb{W} \\ \mathbb{G}^T + \mathbb{Z} & 0 \end{pmatrix} \cdot \begin{pmatrix} \mathcal{V} \\ \mathcal{P} \end{pmatrix} = \begin{pmatrix} f \\ h \end{pmatrix} \quad \text{or,} \quad \mathbb{A} \cdot X = rhs$$

Where \mathbb{K} is the stiffness matrix, \mathbb{G} is the discrete gradient operator, \mathbb{G}^T is the discrete divergence operator, \mathcal{V} the velocity vector, \mathcal{P} the pressure vector. Note that the term $\mathbb{Z}\mathcal{V}$ derives from term $\mathbf{v} \cdot \nabla \rho$ in the continuity equation.

As perfectly explained in the step 32 of deal.II⁶⁰, we need to scale the \mathbb{G} term since it is many orders of magnitude smaller than \mathbb{K} , which introduces large inaccuracies in the solving process to the point that

⁶⁰https://www.dealii.org/9.0.0/doxygen/deal.II/step_32.html

the solution is nonsensical. This scaling coefficient is η/L . After building the \mathbb{G} block, it is then scaled as follows: $\mathbb{G}' = \frac{\eta}{L}\mathbb{G}$ so that we now solve

$$\begin{pmatrix} \mathbb{K} & \mathbb{G}' + \mathbb{W} \\ \mathbb{G}'^T + \mathbb{Z} & 0 \end{pmatrix} \cdot \begin{pmatrix} \mathcal{V} \\ \mathcal{P}' \end{pmatrix} = \begin{pmatrix} f \\ h \end{pmatrix}$$

After the solve phase, we recover the real pressure with $\mathcal{P} = \frac{\eta}{L}\mathcal{P}'$.

adapt notes since I should scale \mathbb{W} and \mathbb{Z} too. h should be caled too !!!!!!!!

Each block \mathbb{K} , \mathbb{G} , \mathbb{Z} and vectors f and h are built separately in the code and assembled into the matrix \mathbb{A} and vector rhs afterwards. \mathbb{A} and rhs are then passed to the solver. We will see later that there are alternatives to solve this approach which do not require to build the full Stokes matrix \mathbb{A} .

Remark 1: the terms $\mathbb{Z}\mathcal{V}$ and $\mathbb{W}\mathcal{P}$ are often put in the rhs (i.e. added to h) so that the matrix \mathbb{A} retains the same structure as in the incompressible case. This is indeed how it is implemented in ASPECT, see also appendix A of [1167]. This however requires more work since the rhs depends on the solution and some form of iterations is needed.

Remark 2: Very often the adiabatic heating term $\alpha T(\mathbf{v} \cdot \nabla p)$ is simplified as follows: If you assume the vertical component of the gradient of the dynamic pressure to be small compared to the gradient of the total pressure (in other words, the gradient is dominated by the gradient of the hydrostatic pressure), then $-\rho\mathbf{g} \simeq \nabla p$ and then $\alpha T(\mathbf{v} \cdot \nabla p) \simeq -\alpha\rho T\mathbf{v} \cdot \mathbf{g}$. We will however not be using this approximation in what follows.

We have already established that

$$\vec{\tau} = \mathbf{C}_\eta \mathbf{B} \mathbf{V}$$

The following measurements are carried out:

- The root mean square velocity (**vrms**):

$$v_{rms} = \sqrt{\frac{1}{V} \int_V v^2 dV}$$

- The average temperature (**Tavrg**):

$$\langle T \rangle = \frac{1}{V} \int_V T dV$$

- The total mass (**mass**):

$$M = \int_V \rho dV$$

- The Nusselt number (**Nu**):

$$Nu = -\frac{1}{Lx} \frac{1}{\Delta T} \int_0^{Lx} \frac{\partial T(x, y = L_y)}{\partial y} dx$$

- The kinetic energy (**EK**):

$$E_K = \int_V \frac{1}{2} \rho v^2 dV$$

- The work done against gravity

$$\langle W \rangle = - \int_V \rho g_y v_y dV$$

- The total viscous dissipation (**visc_diss**)

$$\langle \Phi \rangle = \int \Phi dV = \frac{1}{V} \int 2\eta \dot{\epsilon} : \dot{\epsilon} dV$$

- The gravitational potential energy (**EG**)

$$E_G = \int_V \rho g_y (L_y - y) dV$$

- The internal thermal energy (ET)

$$E_T = \int_V \rho_{(0)} C_p T dV$$

Remark 3: Measuring the total mass can be misleading: indeed because $\rho = \rho_0(1 - \alpha T)$, then measuring the total mass amounts to measuring a constant minus the volume-integrated temperature, and there is no reason why the latter should be zero, so that there is no reason why the total mass should be zero...!

The experimental setup

The setup is as follows: the domain is $Lx = Ly = 3000\text{km}$. Free slip boundary conditions are imposed on all four sides. The initial temperature is given by:

$$T(x, y) = \left(\frac{L_y - y}{L_y} - 0.01 \cos\left(\frac{\pi x}{L_x}\right) \sin\left(\frac{\pi y}{L_y}\right) \right) \Delta T + T_{surf}$$

with $\Delta T = 4000\text{K}$, $T_{surf} = T_0 = 273.15\text{K}$. The temperature is set to $\Delta T + T_{surf}$ at the bottom and T_{surf} at the top. We also set $k = 3$, $C_p = 1250$, $|g| = 10$, $\rho_0 = 3000$ and we keep the Rayleigh number Ra and dissipation number Di as input parameters:

$$Ra = \frac{\alpha g \Delta T L^3 \rho_0^2 C_p}{\eta k} \quad Di = \frac{\alpha g L}{C_p}$$

From the second equation we get $\alpha = \frac{Di C_p}{gL}$, which we can insert in the first one:

$$Ra = \frac{Di C_p^2 \Delta T L^2 \rho_0^2}{\eta k} \quad \text{or}, \quad \eta = \frac{Di C_p^2 \Delta T L^2 \rho_0^2}{Ra k}$$

For instance, for $Ra = 10^4$ and $Di = 0.75$, we obtain $\alpha \simeq 3 \cdot 10^{-5}$ and $\eta \simeq 10^{25}$ which are quite reasonable values.

Scaling

Following [1054], we non-dimensionalize the equations using the reference values for density ρ_r , thermal expansivity α_r , temperature contrast ΔT_r (`refTemp`), thermal conductivity k_r , heat capacity C_p , depth of the fluid layer L and viscosity η_r . The non-dimensionalization for velocity, u_r , pressure p_r and time, t_r become

$$u_r = \frac{k_r}{\rho_r C_p L} \quad (\text{refvel})$$

$$p_r = \frac{\eta_r k_r}{\rho_r C_p L^2} \quad (\text{refpress})$$

$$t_r = \frac{\rho_r C_p L^2}{k_r} \quad (\text{reftime})$$

In the case of the setup described hereabove, and when choosing $Ra = 10^4$ and $Di = 0.5$, we get:

```
alphaT 2.08333e-05
eta 8.437500e+24
reftime 1.125000e+19
refvel 2.666667e-13
refPress 7.500000e+05
```

Conservation of energy 1

under BA and EBA approximations

Following [1167], we take the dot product of the momentum equation with the velocity \mathbf{v} and integrate over the whole volume⁶¹:

$$\int_V [-\nabla \cdot \boldsymbol{\tau} + \nabla p] \cdot \mathbf{v} dV = \int_V \rho \mathbf{g} \cdot \mathbf{v} dV$$

or,

$$-\int_V (\nabla \cdot \boldsymbol{\tau}) \cdot \mathbf{v} dV + \int_V \nabla p \cdot \mathbf{v} dV = \int_V \rho \mathbf{g} \cdot \mathbf{v} dV$$

Let us look at each block separately:

$$-\int_V (\nabla \cdot \boldsymbol{\tau}) \cdot \mathbf{v} dV = -\int_S \boldsymbol{\tau} \underbrace{\mathbf{v} \cdot \mathbf{n}}_{=0 \text{ (b.c.)}} dS + \int_V \boldsymbol{\tau} : \nabla \mathbf{v} dV = \int_V \boldsymbol{\tau} : \dot{\boldsymbol{\epsilon}} dV = \int_V \Phi dV$$

which is the volume integral of the shear heating. Then,

$$\int_V \nabla p \cdot \mathbf{v} dV = \int_S p \underbrace{\mathbf{v} \cdot \mathbf{n}}_{=0 \text{ (b.c.)}} dS - \int_V \underbrace{\nabla \cdot \mathbf{v}}_{=0 \text{ (incomp.)}} pdV = 0$$

which is then zero in the case of an incompressible flow. And finally

$$\int_V \rho \mathbf{g} \cdot \mathbf{v} dV = W$$

which is the work against gravity.

Conclusion for an *incompressible* fluid: we should have

$$\int_V \Phi dV = \int_V \rho \mathbf{g} \cdot \mathbf{v} dV \quad (687)$$

This formula is hugely problematic: indeed, the term ρ in the rhs is the full density. We know that to the value of ρ_0 corresponds a lithostatic pressure gradient $p_L = \rho_0 gy$. In this case one can write $\rho = \rho_0 + \rho'$ and $p = p_L + p'$ so that we also have

$$\int_V [-\nabla \cdot \boldsymbol{\tau} + \nabla p'] \cdot \mathbf{v} dV = \int_V \rho' \mathbf{g} \cdot \mathbf{v} dV$$

which will ultimately yield

$$\int_V \Phi dV = \int_V \rho' \mathbf{g} \cdot \mathbf{v} dV = \int_V (\rho - \rho_0) \mathbf{g} \cdot \mathbf{v} dV \quad (688)$$

Obviously Eqs.(687) and (688) cannot be true at the same time. The problem comes from the nature of the (E)BA approximation: $\rho = \rho_0$ in the mass conservation equation but it is not constant in the momentum conservation equation, which is of course inconsistent. Since the mass conservation equation is $\nabla \cdot \mathbf{v} = 0$ under this approximation then the term $\int_V \nabla p \cdot \mathbf{v} dV$ is always zero for any pressure (full pressure p , or overpressure $p - p_L$), hence the paradox. This paradox will be lifted when a consistent set of equations will be used (compressible formulation). On a practical note, Eqs.(687) is not verified by the code, while (688) is.

In the end:

$$\int_V \Phi dV = \underbrace{\int_V}_{\text{visc.diss}} (\rho - \rho_0) \underbrace{\mathbf{g} \cdot \mathbf{v} dV}_{\text{work_grav}}$$

(689)

⁶¹Check: this is akin to looking at the power, force*velocity, says Arie

under no approximation at all

$$\int_V \nabla p \cdot \mathbf{v} dV = \int_S p \underbrace{\mathbf{v} \cdot \mathbf{n}}_{=0 \text{ (b.c.)}} dS - \int_V \nabla \cdot \mathbf{v} pdV = 0 \quad (690)$$

$$= \int_V \frac{1}{\rho} \mathbf{v} \cdot \nabla \rho pdV = 0 \quad (691)$$

(692)

ToDo: see section 3 of [1167] where this is carried out with the Adams-Williamson eos.

Conservation of energy 2

Also, following the Reynold's transport theorem [1256], p210, we have for a property A (per unit mass)

$$\frac{d}{dt} \int_V A \rho dV = \int_V \frac{\partial}{\partial t} (A \rho) dV + \int_S A \rho \mathbf{v} \cdot \mathbf{n} dS$$

Let us apply to this to $A = C_p T$ and compute the time derivative of the internal energy:

$$\frac{d}{dt} \int_V \rho C_p T dV = \int_V \frac{\partial}{\partial t} (\rho C_p T) dV + \int_S \rho \underbrace{\mathbf{v} \cdot \mathbf{n}}_{=0 \text{ (b.c.)}} dS = \underbrace{\int_V C_p T \frac{\partial \rho}{\partial t} dV}_I + \underbrace{\int_V \rho C_p \frac{\partial T}{\partial t} dV}_{II} \quad (693)$$

In order to expand I , the mass conservation equation will be used, while the heat transport equation will be used for II :

$$I = \int_V C_p T \frac{\partial \rho}{\partial t} dV = - \int_V C_p T \nabla \cdot (\rho \mathbf{v}) dV = - \int_V C_p T \rho \underbrace{\mathbf{v} \cdot \mathbf{n}}_{=0 \text{ (b.c.)}} dS + \int_V \rho C_p \nabla T \cdot \mathbf{v} dV \quad (694)$$

$$II = \int_V \rho C_p \frac{\partial T}{\partial t} dV = \int_V \left[-\rho C_p \mathbf{v} \cdot \nabla T + \nabla \cdot k \nabla T + \rho H + \Phi + \alpha T \left(\frac{\partial p}{\partial t} + \mathbf{v} \cdot \nabla p \right) \right] dV \quad (695)$$

$$= \int_V \left[-\rho C_p \mathbf{v} \cdot \nabla T + \rho H + \Phi + \alpha T \left(\frac{\partial p}{\partial t} + \mathbf{v} \cdot \nabla p \right) \right] dV + \int_V \nabla \cdot k \nabla T dV \quad (696)$$

$$= \int_V \left[-\rho C_p \mathbf{v} \cdot \nabla T + \rho H + \Phi + \alpha T \left(\frac{\partial p}{\partial t} + \mathbf{v} \cdot \nabla p \right) \right] dV + \int_S k \nabla T \cdot \mathbf{n} dS \quad (697)$$

$$= \int_V \left[-\rho C_p \mathbf{v} \cdot \nabla T + \rho H + \Phi + \alpha T \left(\frac{\partial p}{\partial t} + \mathbf{v} \cdot \nabla p \right) \right] dV - \int_S \mathbf{q} \cdot \mathbf{n} dS \quad (698)$$

Finally:

$$I + II = \frac{d}{dt} \underbrace{\int_V \rho C_p T dV}_{\text{ET}} = \int_V \left[\rho H + \Phi + \alpha T \left(\frac{\partial p}{\partial t} + \mathbf{v} \cdot \nabla p \right) \right] dV - \int_S \mathbf{q} \cdot \mathbf{n} dS \quad (699)$$

$$= \int_V \rho H dV + \underbrace{\int_V \Phi dV}_{\text{visc.diss}} + \underbrace{\int_V \alpha T \frac{\partial p}{\partial t} dV}_{\text{extra}} + \underbrace{\int_V \alpha T \mathbf{v} \cdot \nabla p dV}_{\text{adiab_heating}} - \underbrace{\int_S \mathbf{q} \cdot \mathbf{n} dS}_{\text{heatflux_boundary}} \quad (700)$$

This was of course needlessly complicated as the term $\partial \rho / \partial t$ is always taken to be zero, so that $I = 0$ automatically. The mass conservation equation is then simply $\nabla \cdot (\rho \mathbf{v}) = 0$. Then it follows that

$$0 = \int_V C_p T \nabla \cdot (\rho \mathbf{v}) dV = - \int_V C_p T \rho \underbrace{\mathbf{v} \cdot \mathbf{n}}_{=0 \text{ (b.c.)}} dS + \int_V \rho C_p \nabla T \cdot \mathbf{v} dV \quad (701)$$

$$= \int_V \rho C_p \nabla T \cdot \mathbf{v} dV \quad (702)$$

so that the same term in Eq.(698) vanishes too, and then Eq.(700) is always valid, although one should be careful when computing E_T in the BA and EBA cases as it should use ρ_0 and not ρ .

The problem of the onset of convection

[wiki] In geophysics, the Rayleigh number is of fundamental importance: it indicates the presence and strength of convection within a fluid body such as the Earth's mantle. The mantle is a solid that behaves as a fluid over geological time scales.

The Rayleigh number essentially is an indicator of the type of heat transport mechanism. At low Rayleigh numbers conduction processes dominate over convection ones. At high Rayleigh numbers it is the other way around. There is a so-called critical value of the number with delineates the transition from one regime to the other.

This problem has been studied and approached both theoretically and numerically [1806, e.g.] and it was found that the critical Rayleigh number Ra_c is

$$Ra_c = (27/4)\pi^4 \simeq 657.5$$

in setups similar to ours.

VERY BIG PROBLEM

The temperature setup is built as follows: T_{surf} is prescribed at the top, $T_{surf} + \Delta T$ is prescribed at the bottom. The initial temperature profile is linear between these two values. In the case of BA, the actual value of T_{surf} is of no consequence. However, for the EBA the full temperature is present in the adiabatic heating term on the rhs of the hte, and the value of T_{surf} will therefore influence the solution greatly. This is very problematic as there is no real way to arrive at the surface temperature from the King paper. On top of this, the density uses a reference temperature T_0 which too will influence the solution without being present in the controlling Ra and Di numbers!!

In light thereof, it will be very difficult to recover the values of King et al for EBA!

features

- $Q_1 \times P_0$ element
- compressible flow
- mixed formulation
- Dirichlet boundary conditions (no-slip)
- isoviscous
- analytical solution
- pressure smoothing

Relevant literature: [136, 968, 1734, 1167, 1054, 1169, 1219, 882]

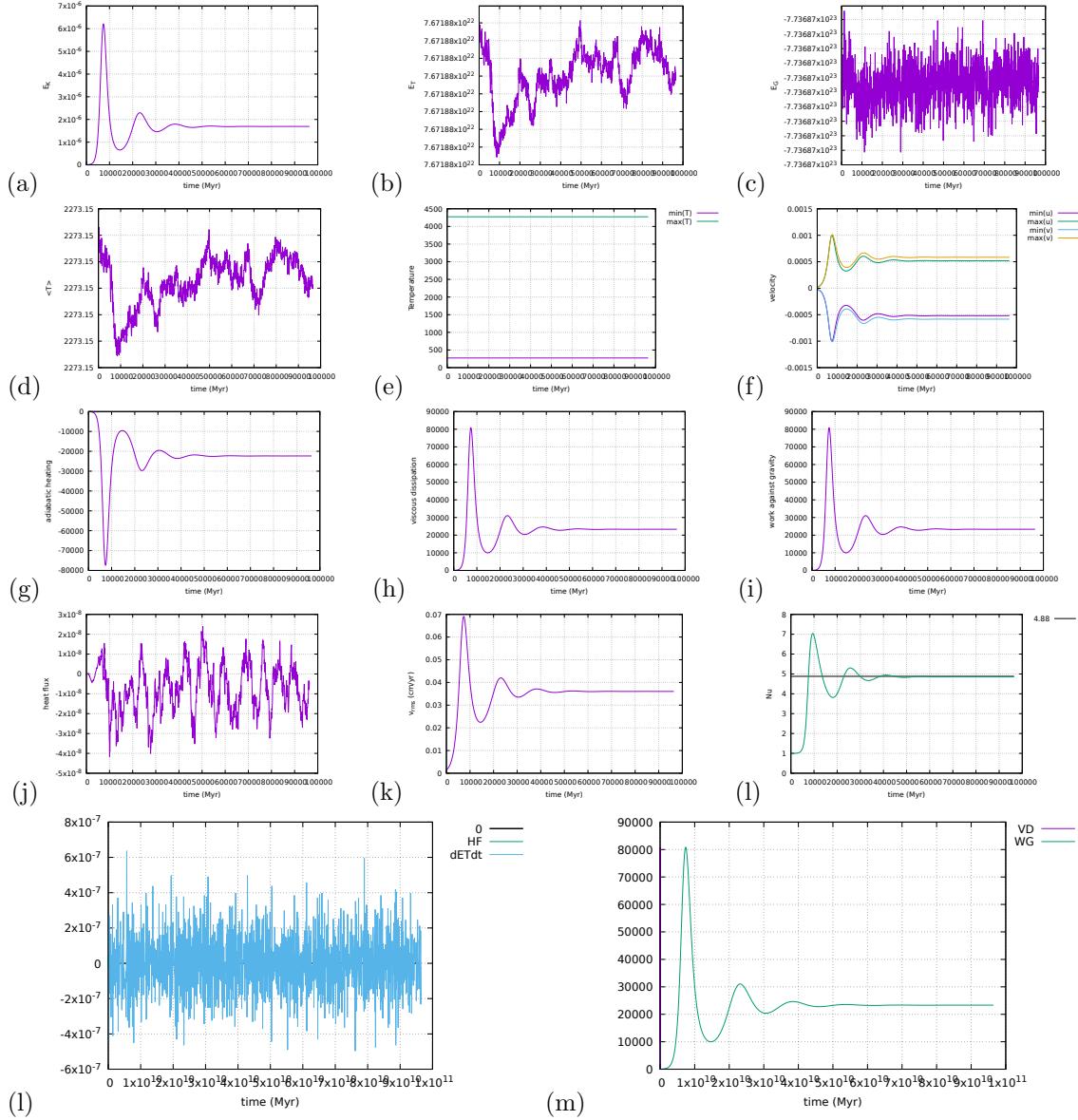
ToDo:

- heat flux is at the moment elemental, so Nusselt and heat flux on boundaries measurements not as accurate as could be.

- implement steady state detection
- do $Ra = 10^5$ and $Ra = 10^6$
- velocity average at surface
- non dimensional heat flux at corners [175]
- depth-dependent viscosity (case 2 of [175])

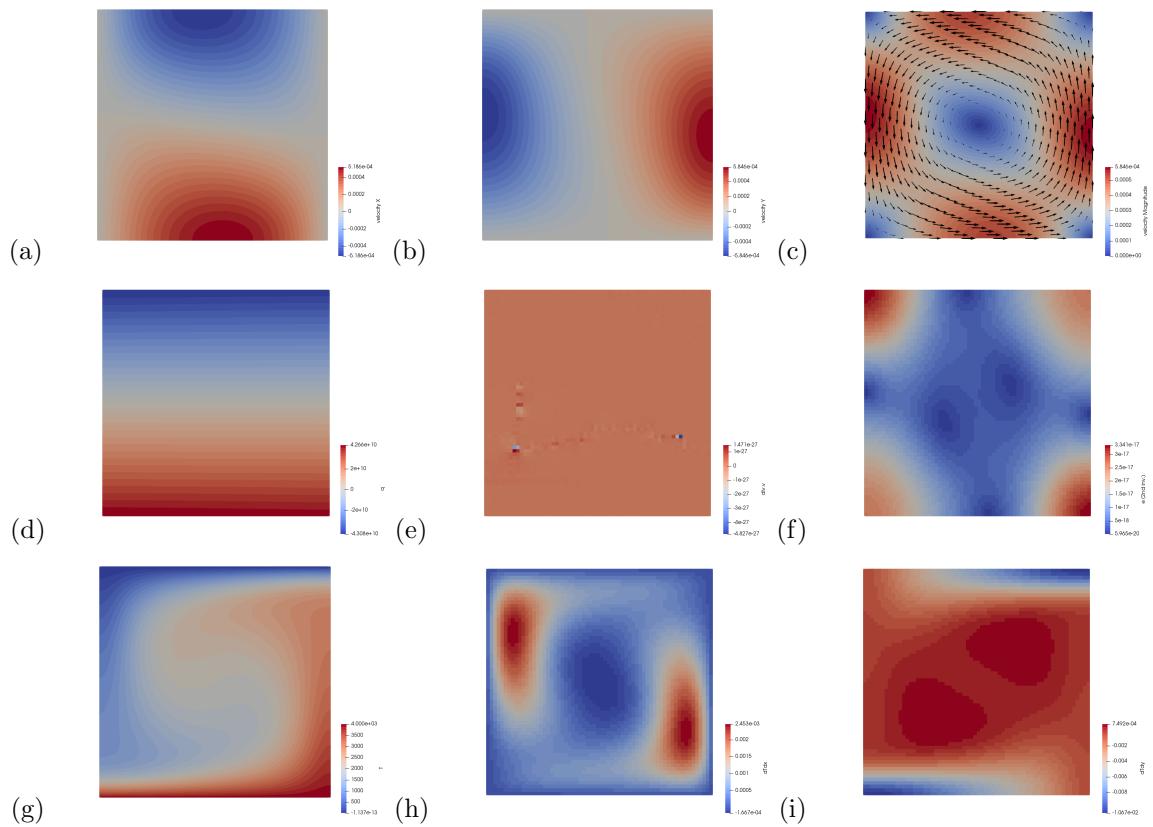
results - BA - $Ra = 10^4$

These results were obtained with a 64x64 resolution, and CFL number of 1. Steady state was reached after about 1250 timesteps.



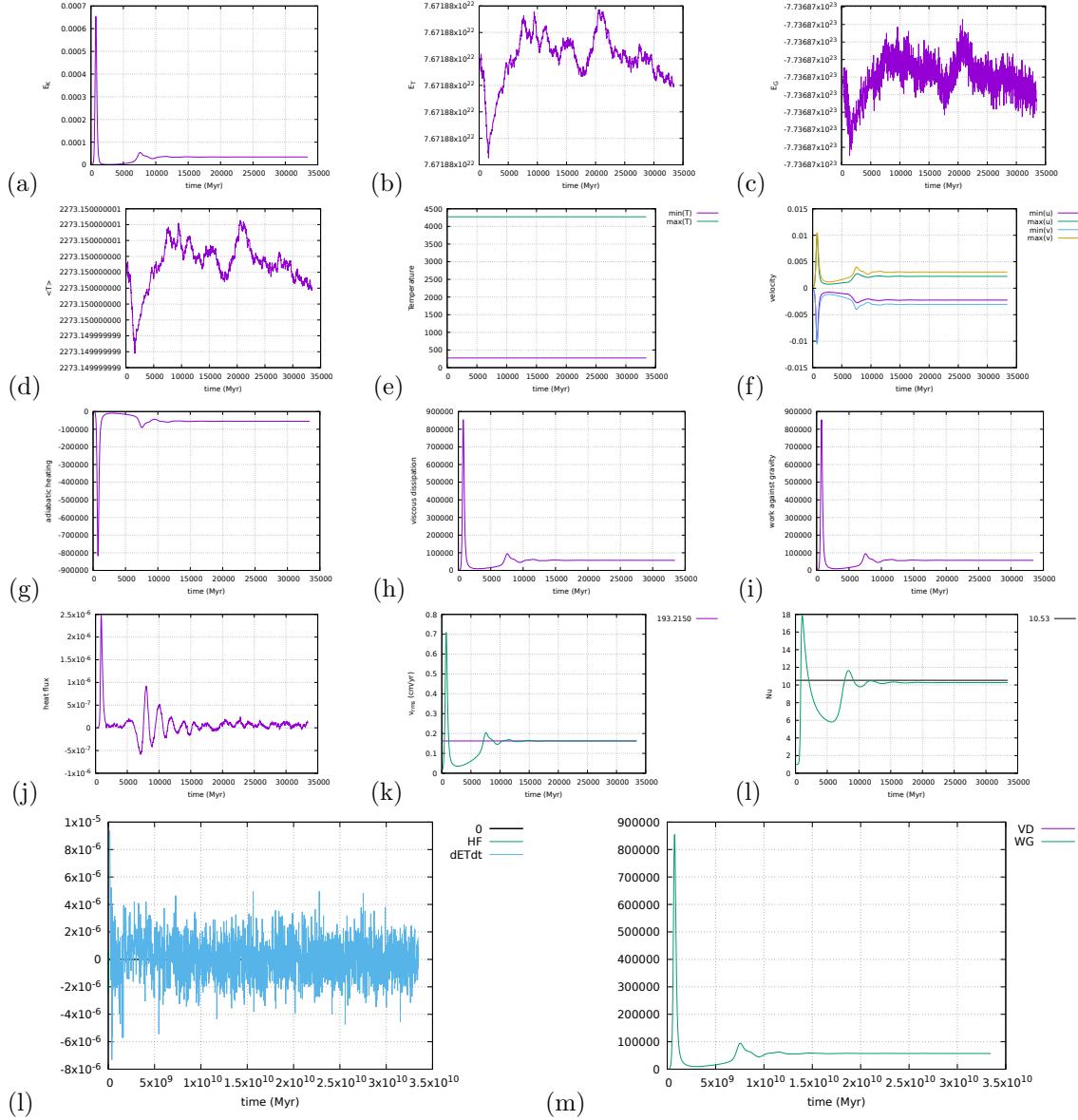
AH: adiabatic heating, VD: viscous dissipation, HF: heat flux, WG: work against gravity

Eq.(700) is verified by (l) and Eq.(689) is verified by (m).



results - BA - $Ra = 10^5$

These results were obtained with a 64x64 resolution, and CFL number of 1. Steady state was reached after about 1250 timesteps.



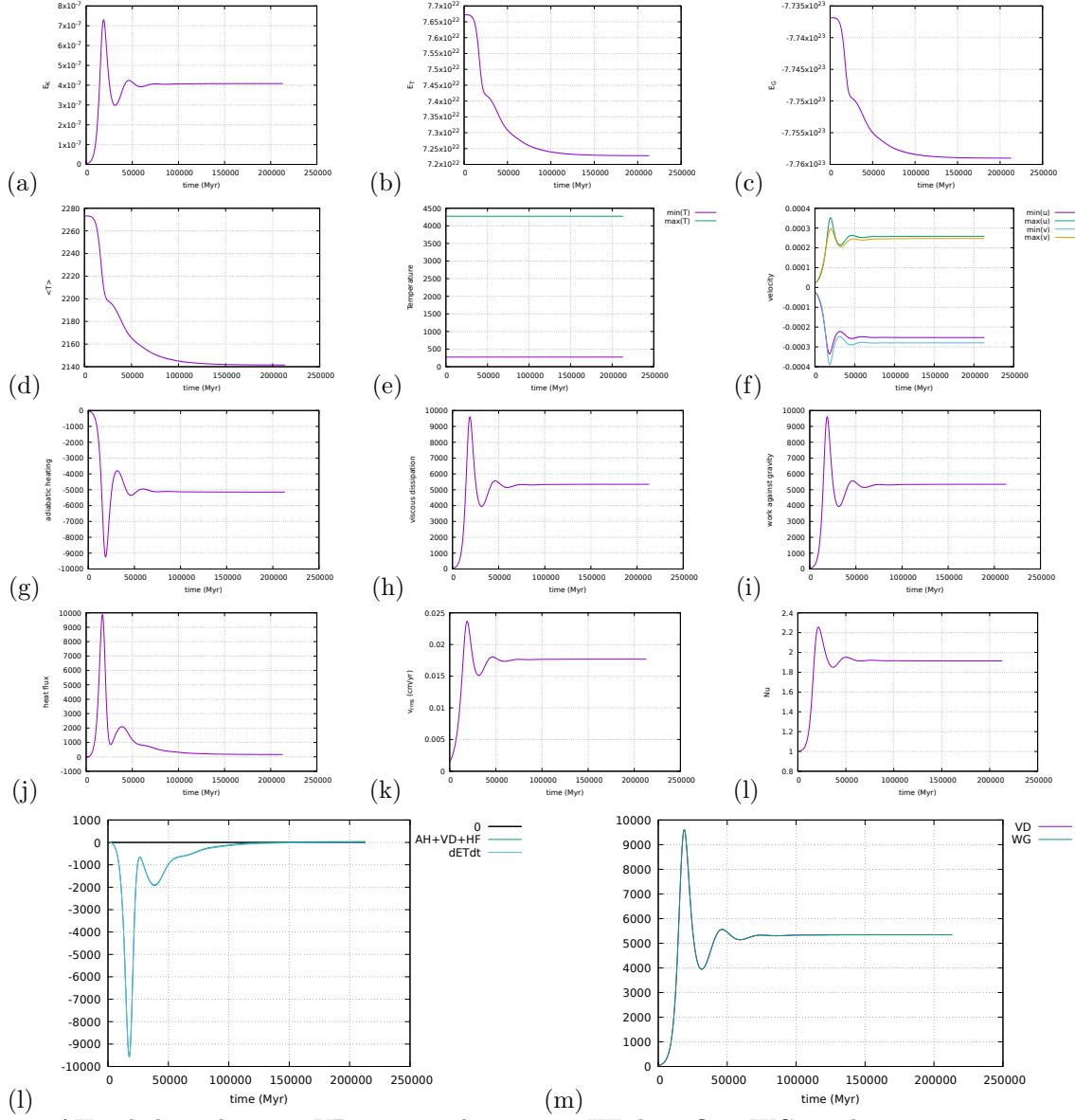
AH: adiabatic heating, VD: viscous dissipation, HF: heat flux, WG: work against gravity

Eq.(700) is verified by (l) and Eq.(689) is verified by (m).

results - BA - $Ra = 10^6$

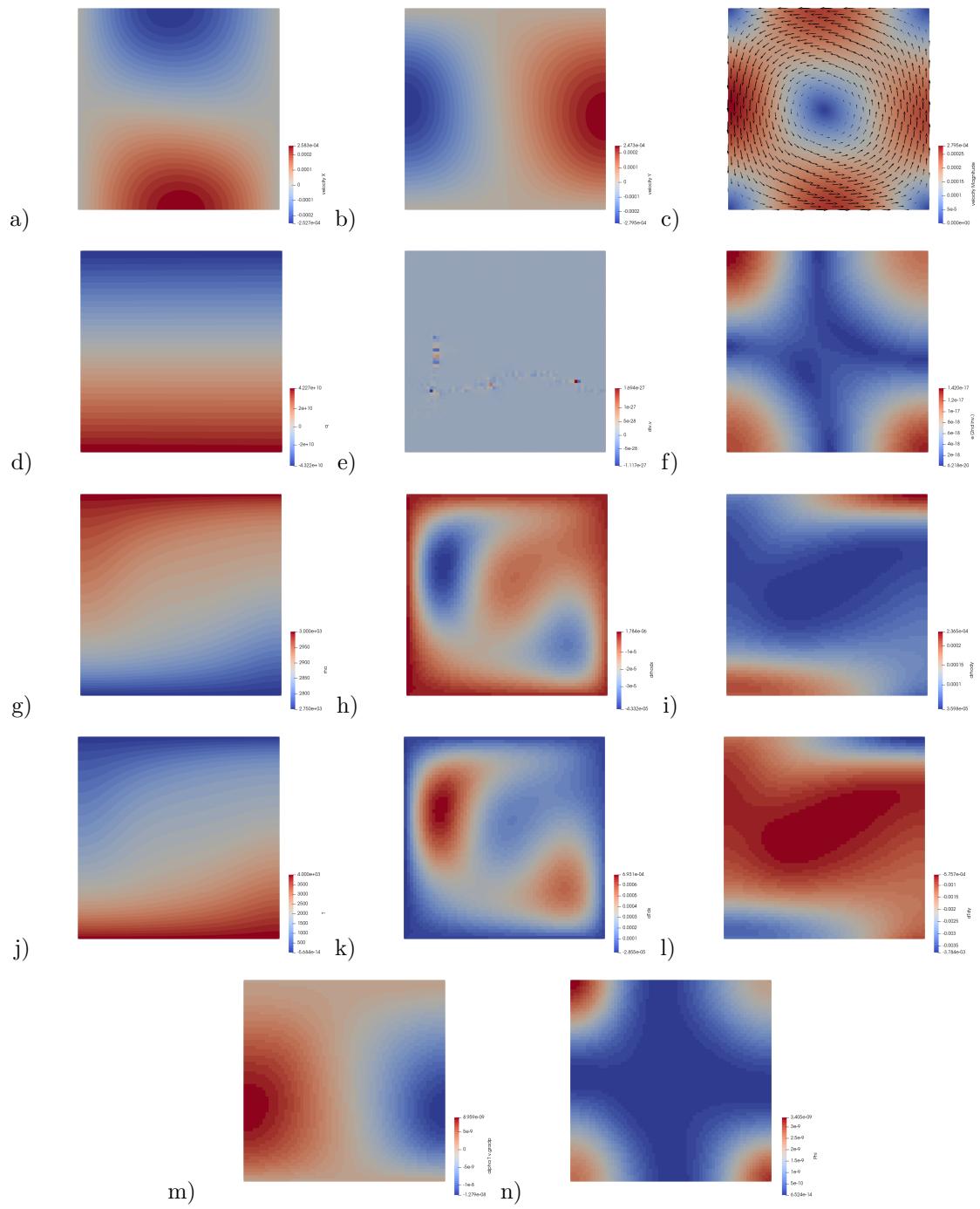
results - EBA - $Ra = 10^4$

These results were obtained with a 64x64 resolution, and CFL number of 1. Steady state was reached after about 2500 timesteps



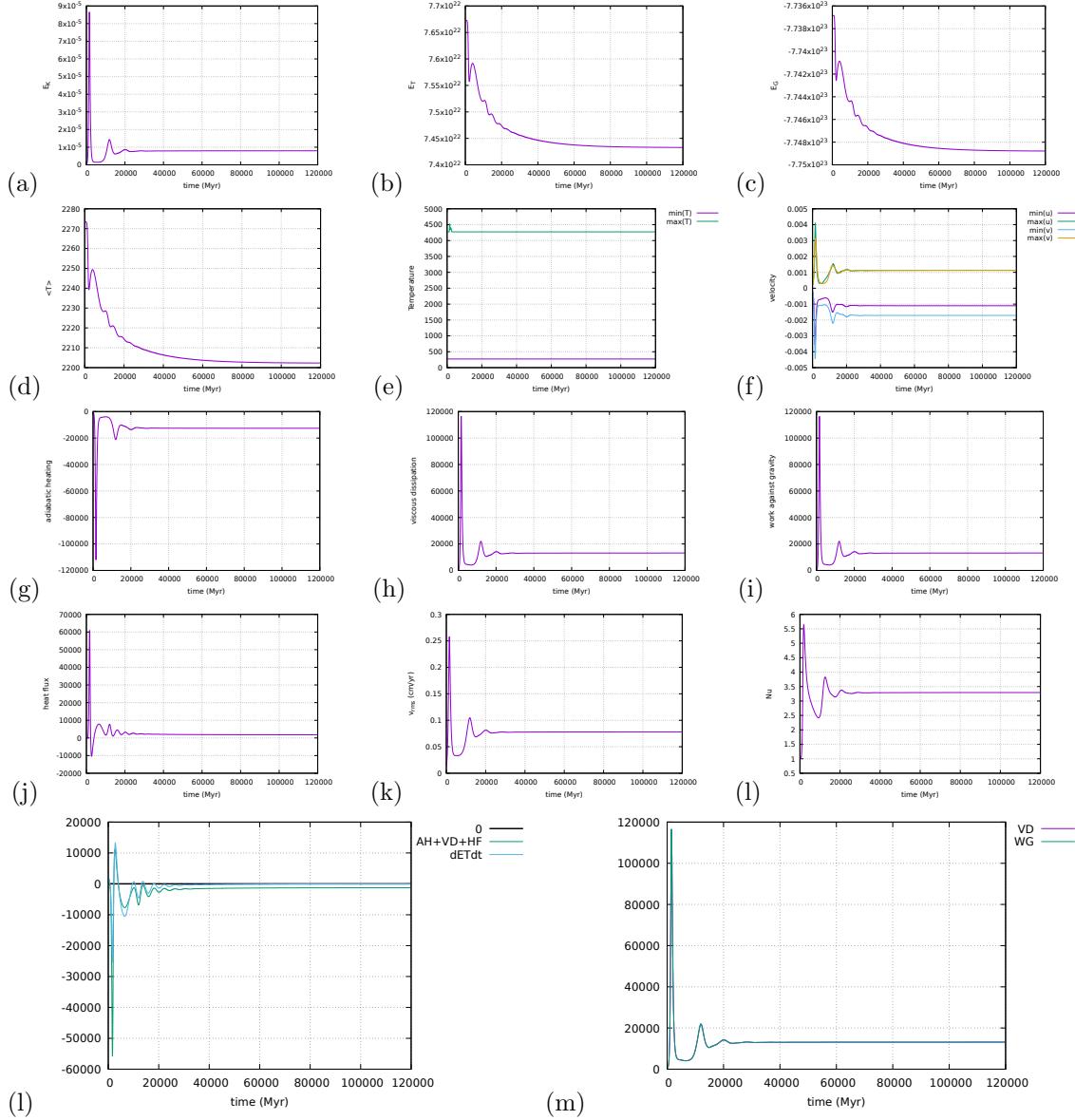
AH: adiabatic heating, VD: viscous dissipation, HF: heat flux, WG: work against gravity

Eq.(700) is verified by (l) and Eq.(689) is verified by (m).



results - EBA - $Ra = 10^5$

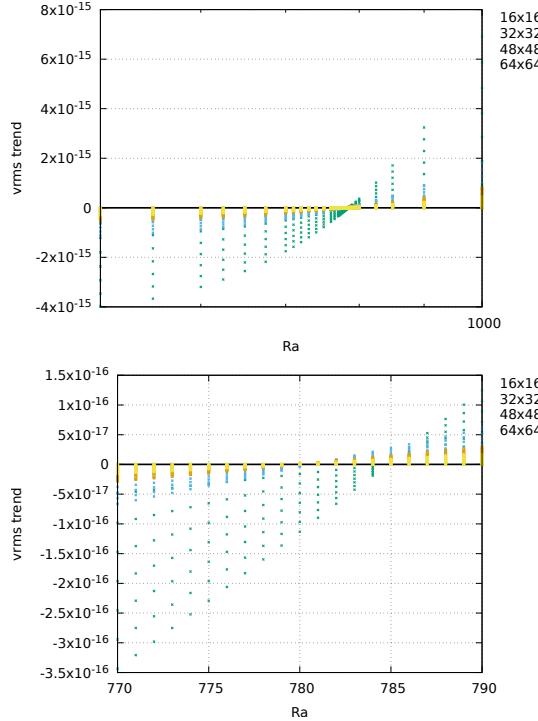
These results were obtained with a 64x64 resolution, and CFL number of 1. Simulation was stopped after about 4300 timesteps.



AH: adiabatic heating, VD: viscous dissipation, HF: heat flux, WG: work against gravity

Onset of convection

The code can be run for values of Ra between 500 and 1000, at various resolutions for the BA formulation. The value $v_{rms}(t) - v_{rms}(0)$ is plotted as a function of Ra and for the 10 first timesteps. If the v_{rms} is found to decrease, then the Rayleigh number is not high enough to allow for convection and the initial temperature perturbation relaxes by diffusion (and then $v_{rms}(t) - v_{rms}(0) < 0$). If the v_{rms} is found to increase, then $v_{rms}(t) - v_{rms}(0) > 0$ and the system is going to showcase convection. The zero value of $v_{rms}(t) - v_{rms}(0)$ gives us the critical Rayleigh number, which is found between 775 and 790.

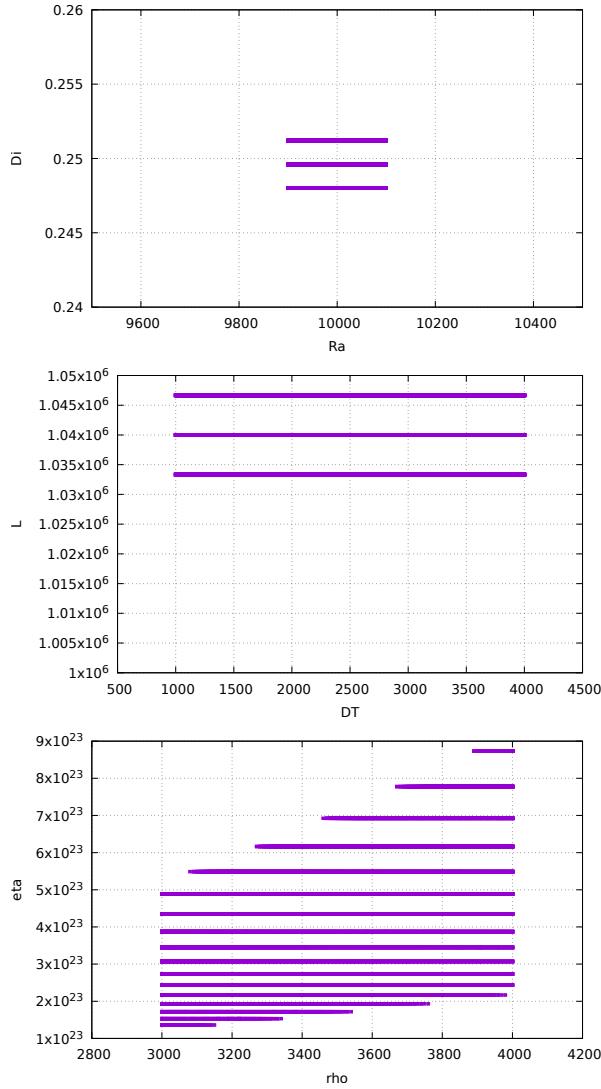


Appendix: Looking for the right combination of parameters for the King benchmark.

I run a quadruple do loop over L , ΔT , ρ_0 and η_0 between plausible values (see code targets.py) and write in a file only the combination which yields the required Rayleigh and Dissipation number values (down to 1% accuracy).

```
alpha=3e-5
g=10
hcapa=1250
hcond=3
DTmin=1000 ; DTmax=4000 ; DTnpts=251
Lmin=1e6 ; Lmax=3e6 ; Lnpts=251
rhomin=3000 ; rhomax=3500 ; rhonpts=41
etamin=19 ; etamax=25 ; etanpts=100
```

On the following plots the 'winning' combinations of these four parameters are shown:



We see that:

- the parameter L (being to the 3rd power in the Ra number) cannot vary too much. Although it is varied between 1000 and 3000km there seems to be a 'right' value at about 1040 km. (why?)
- viscosities are within 10^{23} and 10^{24} which are plausible values (although a bit high?).
- densities can be chosen freely between 3000 and 3500
- ΔT seems to be the most problematic value since it can range from 1000 to 4000K ...

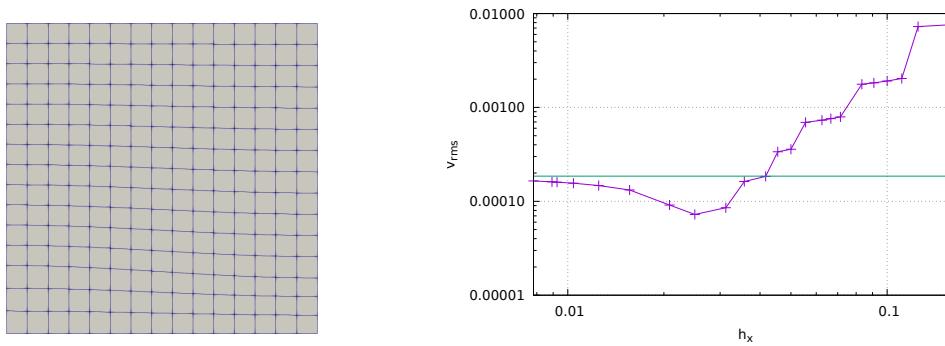
Stone 25: Rayleigh-Taylor instability (1) - instantaneous

This numerical experiment was first presented in [1847]. It consists of an isothermal Rayleigh-Taylor instability in a two-dimensional box of size $L_x = 0.9142$ and $L_y = 1$. Two Newtonian fluids are present in the system: the buoyant layer is placed at the bottom of the box and the interface between both fluids is given by $y(x) = 0.2 + 0.02 \cos\left(\frac{\pi x}{L_x}\right)$. The bottom fluid is parametrised by its mass density ρ_1 and its viscosity μ_1 , while the layer above is parametrised by ρ_2 and μ_2 .

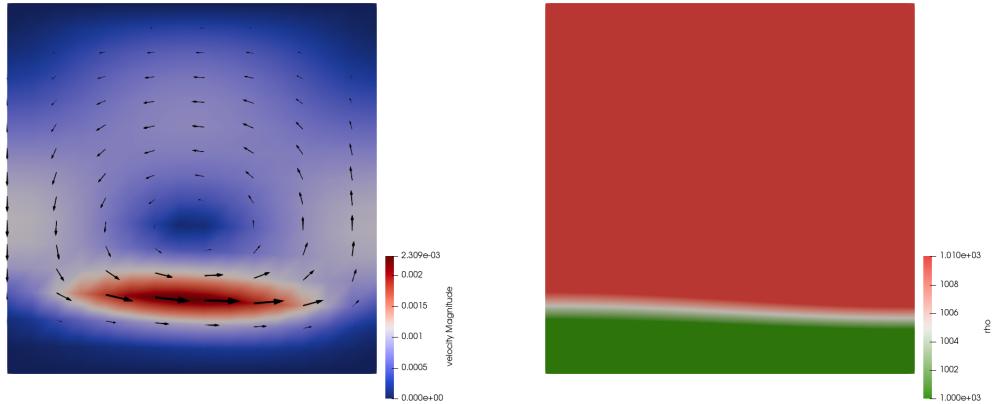
No-slip boundary conditions are applied at the bottom and at the top of the box while free-slip boundary conditions are applied on the sides.

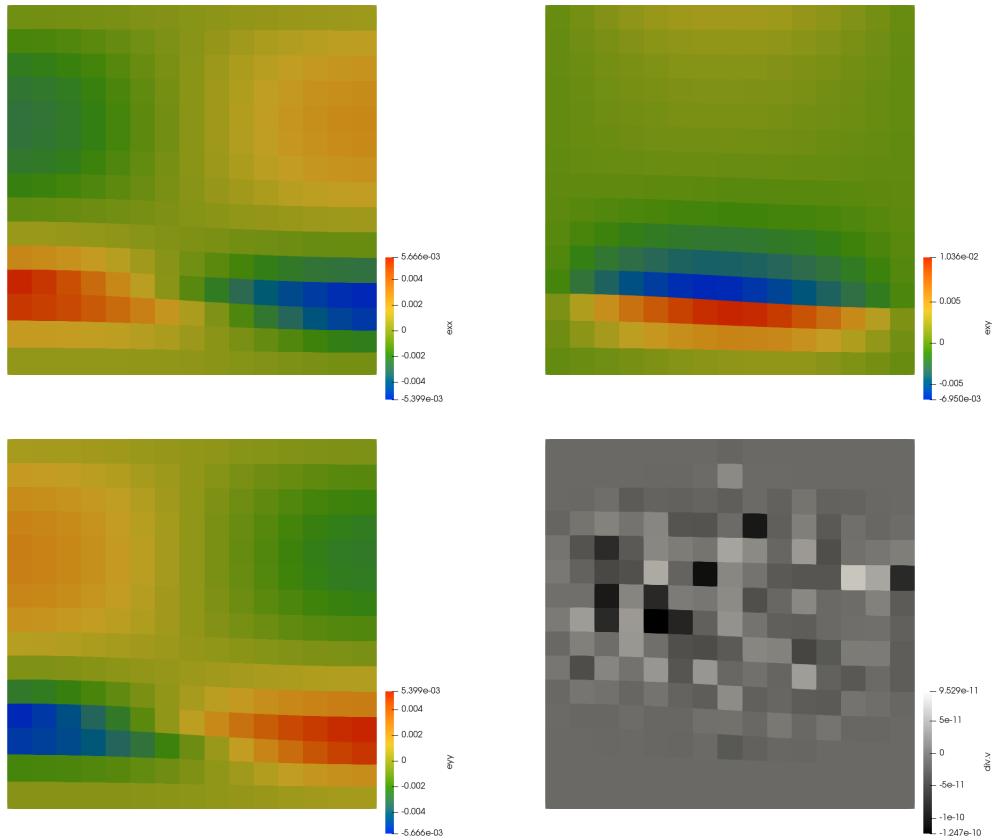
In the original benchmark the system is run over 2000 units of dimensionless time and the timing and position of various upwellings/downwellings is monitored. In this present experiment only the root mean square velocity is measured at $t = 0$: the code is indeed not yet foreseen of any algorithm capable of tracking deformation.

Another approach than the ones presented in the extensive literature which showcases results of this benchmark is taken. The mesh is initially fitted to the fluids interface and the resolution is progressively increased. This results in the following figure:



The green line indicates results obtained with my code ELEFANT with grids up to 2000x2000 with the exact same methodology.





features

- $Q_1 \times P_0$ element
- incompressible flow
- mixed formulation
- isothermal
- numerical benchmark

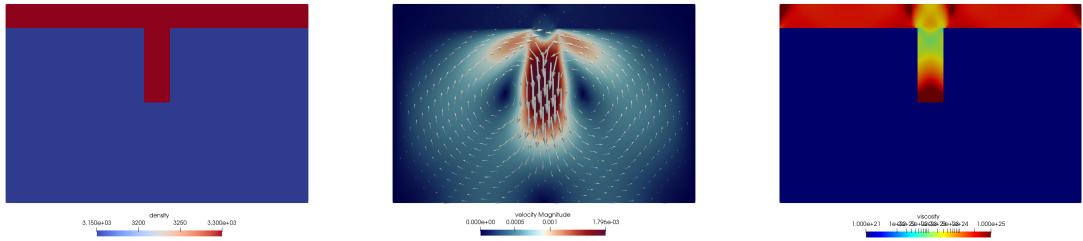
Stone 26: Slab detachment benchmark (1) - instantaneous

As in [1633], the computational domain is $1000\text{km} \times 660\text{km}$. No-slip boundary conditions are imposed on the sides of the system while free-slip boundary conditions are imposed at the top and bottom. Two materials are present in the domain: the lithosphere (mat.1) and the mantle (mat.2). The overriding plate (mat.1) is 80km thick and is placed at the top of the domain. An already subducted slab (mat.1) of 250km length hangs vertically under this plate. The mantle occupies the rest of the domain.

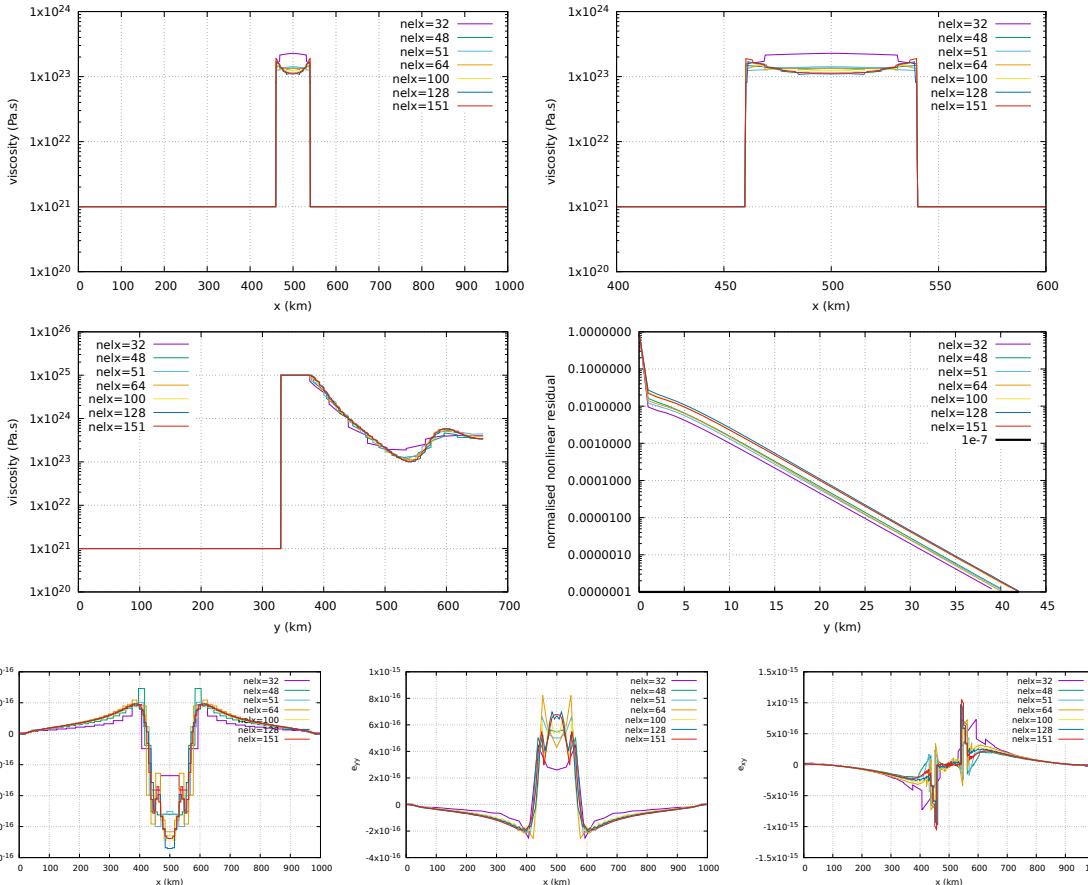
The mantle has a constant viscosity $\eta_0 = 10^{21}\text{Pa.s}$ and a density $\rho = 3150\text{kg/m}^3$. The slab has a density $\rho = 3300\text{kg/m}^3$ and is characterised by a power-law flow law so that its effective viscosity depends on the second invariant of the strainrate I_2 as follows:

$$\eta_{eff} = \frac{1}{2}A^{-1/n_s}I_2^{1/n_s-1} = \frac{1}{2}[(2 \times 4.75 \times 10^{11})^{-n_s}]^{-1/n_s}I_2^{1/n_s-1} = 4.75 \times 10^{11}I_2^{1/n_s-1} = \eta_0 I_2^{1/n_s-1} \quad (703)$$

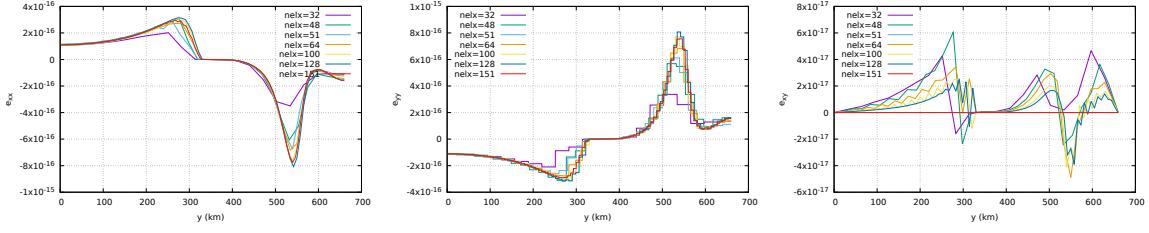
with $n_s = 4$ and $A = (2 \times 4.75 \times 10^{11})^{-n_s}$, or $\eta_0 = 4.75 \times 10^{11}$.



Fields at convergence for 151x99 grid.



Along the horizontal line



Along the vertical line

features

- $Q_1 \times P_0$ element
- incompressible flow
- mixed formulation
- isothermal
- nonlinear rheology
- nonlinear residual

Todo: nonlinear mantle, pressure normalisation

Also check Bellas et al, 2018 [116].

Stone 27: Consistent Boundary Flux

In what follows we will be re-doing the numerical experiments presented in Zhong et al. [2007].

The first benchmark showcases a unit square domain with free slip boundary conditions prescribed on all sides. The resolution is fixed to $64 \times 64 Q_1 \times P_0$ elements. The flow is isoviscous and the buoyancy force \mathbf{f} is given by

$$\begin{aligned} f_x &= 0 \\ f_y &= \rho_0 \alpha T(x, y) \end{aligned}$$

with the temperature field given by

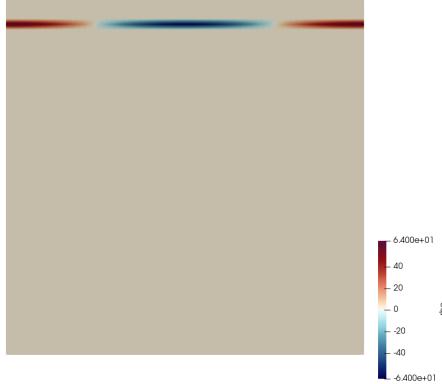
$$T(x, y) = \cos(kx)\delta(y - y_0)$$

where $k = 2\pi/\lambda$ and λ is a wavelength, and y_0 represents the location of the buoyancy strip. We set $g_y = -1$ and prescribe $\rho(x, y) = \rho_0 \alpha \cos(kx)\delta(y - y_0)$ on the nodes of the mesh.

One can prove ([2007] and refs. therein) that there is an analytic solution for the surface stress σ_{zz} ⁶²

$$\frac{\sigma_{yy}}{\rho \alpha g h} = \frac{\cos(kx)}{\sinh^2(k)} [k(1 - y_0) \sinh(k) \cosh(ky_0) - k \sinh(k(1 - y_0)) + \sinh(k) \sinh(ky_0)]$$

We choose $\rho_0 \alpha = 64$, $\eta = 1$ (note that in this case the normalising coefficient of the stress is exactly 1 (since $h = L_x/nelx = 1/64$) so it is not implemented in the code). $\lambda = 1$ is set to 1 and we explore $y_0 = \frac{63}{64}, \frac{62}{64}, \frac{59}{64}$ and $y_0 = 32/64$. Under these assumptions the density field for $y_0 = 59/64$ is:

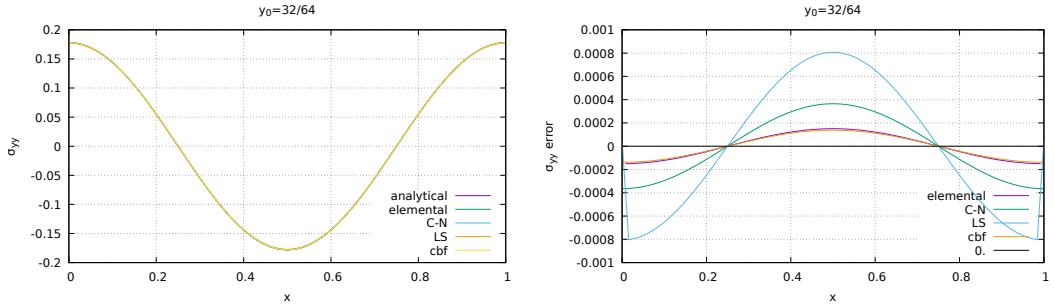


We can recover the stress at the boundary by computing the yy component of the stress tensor in the top row of elements:

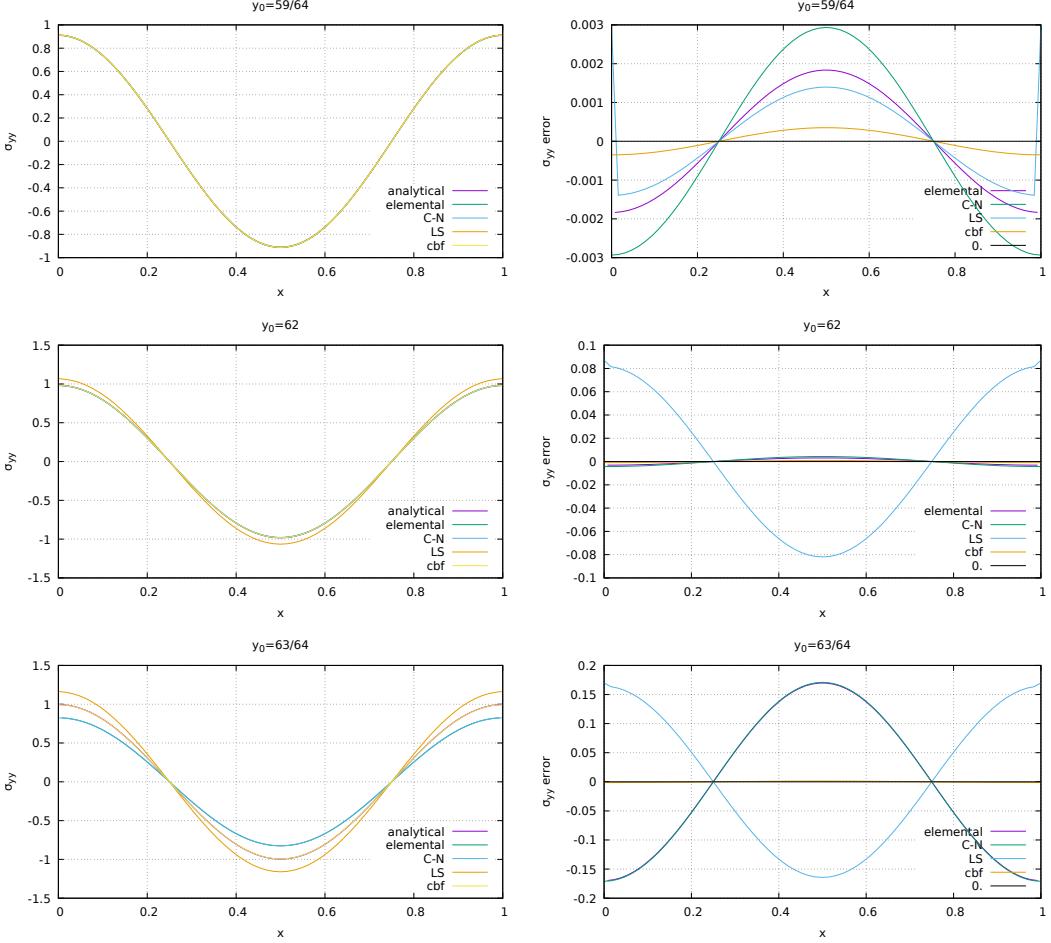
$$\sigma_{yy} = -p + 2\eta\dot{\epsilon}_{yy}$$

Note that pressure is by definition elemental, and that strain rate components are then also computed in the middle of each element.

These elemental quantities can be projected onto the nodes (see section ??) by means of the C→N algorithm or a least square algorithm (LS).



⁶²Note that in the paper the authors use $\rho \alpha g$ which does not have the dimensions of a stress



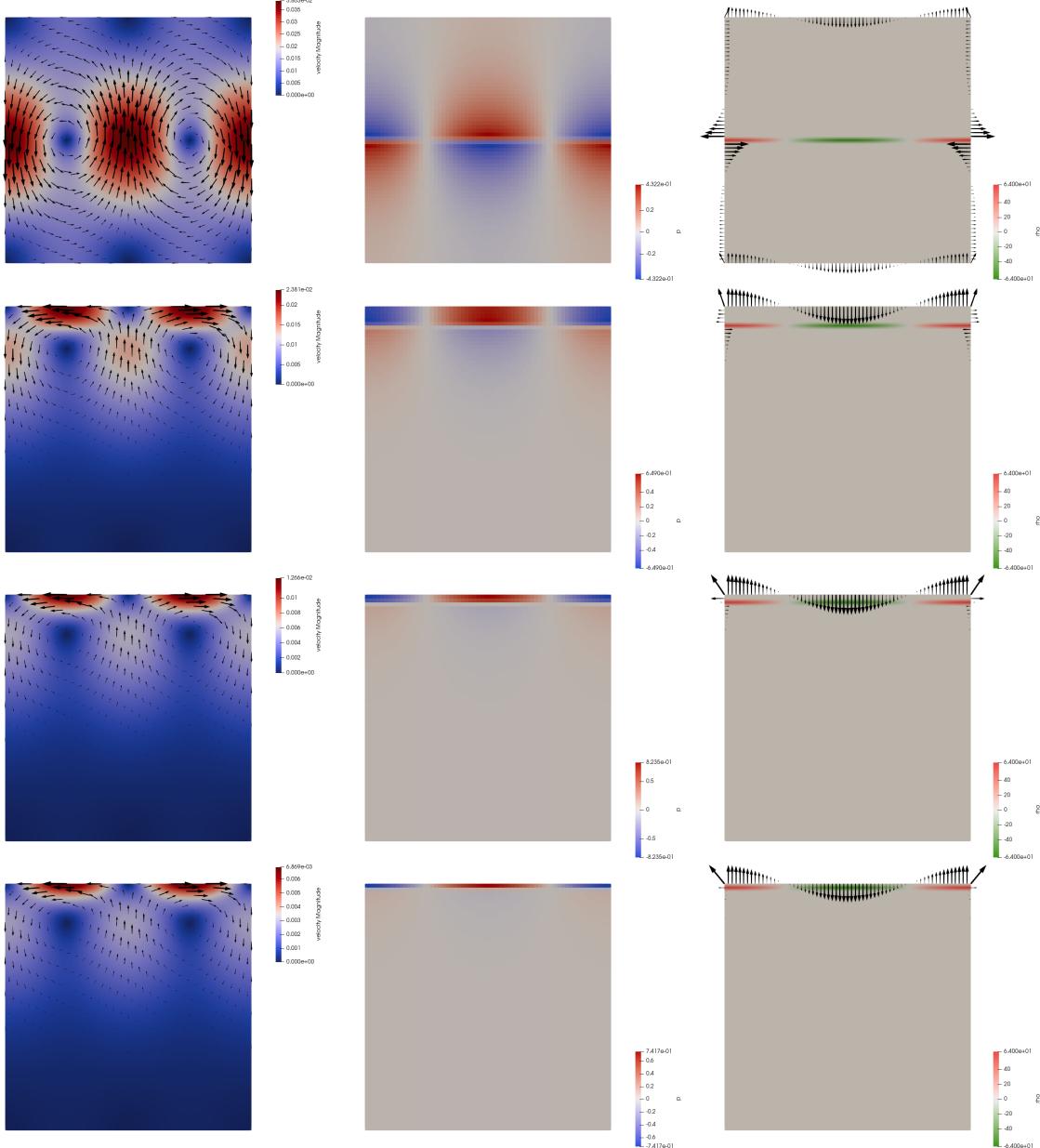
The consistent boundary flux (CBF) method allows us to compute traction vectors $\mathbf{t} = \boldsymbol{\sigma} \cdot \mathbf{n}$ on the boundary of the domain. On the top boundary, $\mathbf{n} = (0, 1)$ so that $\mathbf{t} = (\sigma_{xy}, \sigma_{yy})^T$ and t_y is the quantity we need to consider and compare to other results.

In the following table are shown the results presented in [2007] alongside the results obtained with Fieldstone:

Method	$y_0 = 63/64$	$y_0 = 62/64$	$y_0 = 59/64$	$y_0 = 32/64$
Analytic solution	0.995476	0.983053	0.912506	0.178136
Pressure smoothing [2007]	1.15974	1.06498	0.911109	n.a.
CBF [2007]	0.994236	0.982116	0.912157	n.a.
fieldstone: elemental	0.824554 (-17.17 %)	0.978744 (-0.44%)	0.909574 (-0.32 %)	0.177771 (-0.20 %)
fieldstone: nodal (C→N)	0.824554 (-17.17 %)	0.978744 (-0.44%)	0.909574 (-0.32 %)	0.177771 (-0.20 %)
fieldstone: LS	1.165321 (17.06 %)	1.070105 (8.86%)	0.915496 (0.33 %)	0.178182 (0.03 %)
fieldstone: CBF	0.994236 (-0.13 %)	0.982116 (-0.10%)	0.912157 (-0.04 %)	0.177998 (-0.08 %)

We see that we recover the published results with the same exact accuracy, thereby validating our implementation. Also rather fascinating is the fact that the original paper carries out the whole study without showing any image of the 2D domain ever.

On the following figures are shown the velocity, pressure and traction fields for two cases $y_0 = 32/64$ and $y_0 = 63/64$.



Here lies the superiority of our approach over the one presented in the original article: our code computes all traction vectors on all boundaries at once.

[explain how Medge is arrived at!](#)

[compare with ASPECT ???](#)

[pressure average on surface instead of volume ?](#)

Remark. The original article on the CBF [2007] uses a penalty-based formulation (see Section 6.3) so that they do not have to worry about pressure normalisation. However, since constant pressure fields lie in the nullspace of \mathbb{G} the pressure normalisation constant does not play a role.

As shown in Appendix I, the Q_1 mass matrix for the reference cell/element is given by:

$$\mathbf{M}^e = \frac{1}{3} \begin{pmatrix} 2 & 1 \\ 1 & 2 \end{pmatrix}$$

This matrix needs to be multiplied by $h/2$ for an element of size h . Following the methodology presented in [2007], one can also use the Gauss-Lobatto quadrature method to arrive at the mass matrix, and in

this case the simple trapezoidal integration rule variant thereof (see Section 4.1.1). We then have:

$$\mathbf{M}_e = \int_{\Omega_e} \vec{N}^T \vec{N} dV = \int_{-1}^{+1} \vec{N}^T \vec{N} dr \quad (704)$$

on the reference element, with

$$\vec{N}^T = \begin{pmatrix} N_1(r) \\ N_2(r) \end{pmatrix} = \frac{1}{2} \begin{pmatrix} 1-r \\ 1+r \end{pmatrix}$$

We know compute the following integrals with the trapezoidal rule:

$$\int_{-1}^{+1} N_1(r) N_1(r) dr = (1 - -1) \frac{N_1(-1) N_1(-1) + N_1(+1) N_1(+1)}{2} = 1 \quad (705)$$

$$\int_{-1}^{+1} N_1(r) N_2(r) dr = (1 - -1) \frac{N_1(-1) N_2(-1) + N_1(+1) N_2(+1)}{2} = 0 \quad (706)$$

$$\int_{-1}^{+1} N_2(r) N_2(r) dr = (1 - -1) \frac{N_2(-1) N_2(-1) + N_2(+1) N_2(+1)}{2} = 1 \quad (707)$$

and finally

$$\mathbf{M}^e = \begin{pmatrix} 1 & 0 \\ 0 & 1 \end{pmatrix}$$

The resulting matrix is diagonal and it is simply the lumped version of the exact mass matrix.

features

- $Q_1 \times P_0$ element
- incompressible flow
- mixed formulation
- isothermal
- isoviscous
- analytical solution
- pressure smoothing
- consistent boundary flux

Stone 28: convection 2D box - Tosi et al, 2015

This fieldstone was developed in collaboration with Rens Elbertsen.

The viscosity field μ is calculated as the harmonic average between a linear part μ_{lin} that depends on temperature only or on temperature and depth d , and a non-linear, plastic part μ_{plast} dependent on the strain rate:

$$\mu(T, z, \dot{\epsilon}) = 2 \left(\frac{1}{\mu_{lin}(T, z)} + \frac{1}{\mu_{plast}(\dot{\epsilon})} \right)^{-1}. \quad (708)$$

The linear part is given by the linearized Arrhenius law (the so-called Frank-Kamenetskii approximation [648]):

$$\mu_{lin}(T, z) = \exp(-\gamma_T T + \gamma_z z), \quad (709)$$

where $\gamma_T = \ln(\Delta\mu_T)$ and $\gamma_z = \ln(\Delta\mu_z)$ are parameters controlling the total viscosity contrast due to temperature ($\Delta\mu_T$) and pressure ($\Delta\mu_z$). The non-linear part is given by [1794, 1793]:

$$\mu_{plast}(\dot{\epsilon}) = \mu^* + \frac{\sigma_Y}{\sqrt{\dot{\epsilon} : \dot{\epsilon}}}, \quad (710)$$

where μ^* is a constant representing the effective viscosity at high stresses [1694] and σ_Y is the yield stress, also assumed to be constant. In 2-D, the denominator in the second term of equation (710) is given explicitly by

$$\sqrt{\dot{\epsilon} : \dot{\epsilon}} = \sqrt{\dot{\epsilon}_{ij}\dot{\epsilon}_{ij}} = \sqrt{\left(\frac{\partial u_x}{\partial x}\right)^2 + \frac{1}{2} \left(\frac{\partial u_x}{\partial y} + \frac{\partial u_y}{\partial x}\right)^2 + \left(\frac{\partial u_y}{\partial y}\right)^2}. \quad (711)$$

The viscoplastic flow law (equation 708) leads to linear viscous deformation at low stresses (equation (709)) and to plastic deformation for stresses that exceed σ_Y (equation (710)), with the decrease in viscosity limited by the choice of μ^* [1694].

In all cases that we present, the domain is a two-dimensional square box. The mechanical boundary conditions are for all boundaries free-slip with no flux across, i.e. $\tau_{xy} = \tau_{yx} = 0$ and $\mathbf{u} \cdot \mathbf{n} = 0$, where \mathbf{n} denotes the outward normal to the boundary. Concerning the energy equation, the bottom and top boundaries are isothermal, with the temperature T set to 1 and 0, respectively, while side-walls are assumed to be insulating, i.e. $\partial T / \partial x = 0$. The initial distribution of the temperature field is prescribed as follows:

$$T(x, y) = (1 - y) + A \cos(\pi x) \sin(\pi y), \quad (712)$$

where $A = 0.01$ is the amplitude of the initial perturbation.

In the following Table ??, we list the benchmark cases according to the parameters used.

Case	Ra	$\Delta\mu_T$	$\Delta\mu_y$	μ^*	σ_Y	Convective regime
1	10^2	10^5	1	—	—	Stagnant lid
2	10^2	10^5	1	10^{-3}	1	Mobile lid
3	10^2	10^5	10	—	—	Stagnant lid
4	10^2	10^5	10	10^{-3}	1	Mobile lid
5a	10^2	10^5	10	10^{-3}	4	Periodic
5b	10^2	10^5	10	10^{-3}	3 – 5	Mobile lid – Periodic – Stagnant lid

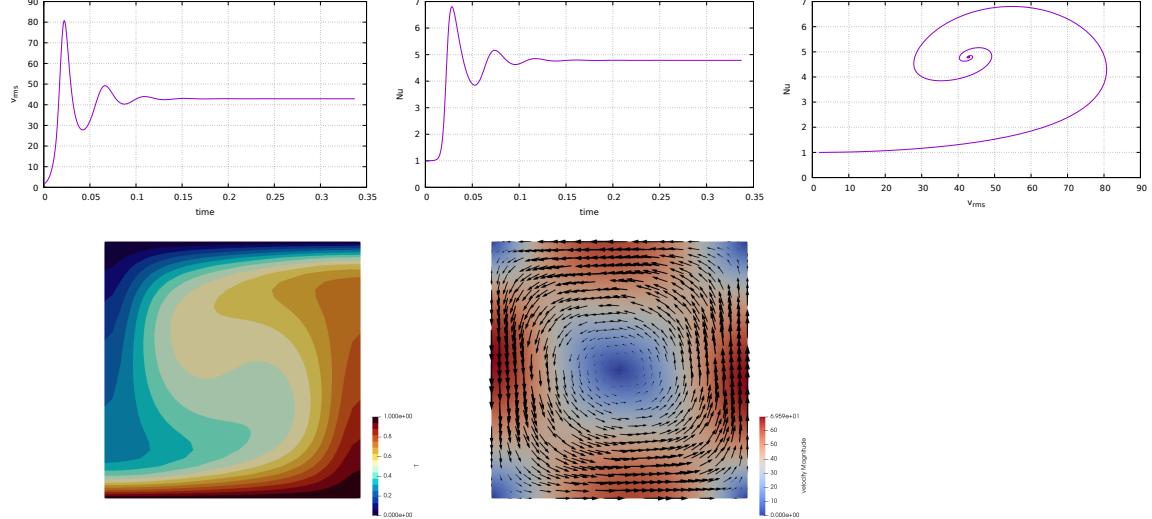
Benchmark cases and corresponding parameters.

In Cases 1 and 3 the viscosity is directly calculated from equation (709), while for Cases 2, 4, 5a, and 5b, we used equation (708). For a given mesh resolution, Case 5b requires running simulations with yield stress varying between 3 and 5

In all tests, the reference Rayleigh number is set at the surface ($y = 1$) to 10^2 , and the viscosity contrast due to temperature $\Delta\mu_T$ is 10^5 , implying therefore a maximum effective Rayleigh number of 10^7 for $T = 1$. Cases 3, 4, 5a, and 5b employ in addition a depth-dependent rheology with viscosity contrast $\Delta\mu_z = 10$. Cases 1 and 3 assume a linear viscous rheology that leads to a stagnant lid regime.

Cases 2 and 4 assume a viscoplastic rheology that leads instead to a mobile lid regime. Case 5a also assumes a viscoplastic rheology but a higher yield stress, which ultimately causes the emergence of a strictly periodic regime. The setup of Case 5b is identical to that of Case 5a but the test consists in running several simulations using different yield stresses. Specifically, we varied σ_Y between 3 and 5 in increments of 0.1 in order to identify the values of the yield stress corresponding to the transition from mobile to periodic and from periodic to stagnant lid regime.

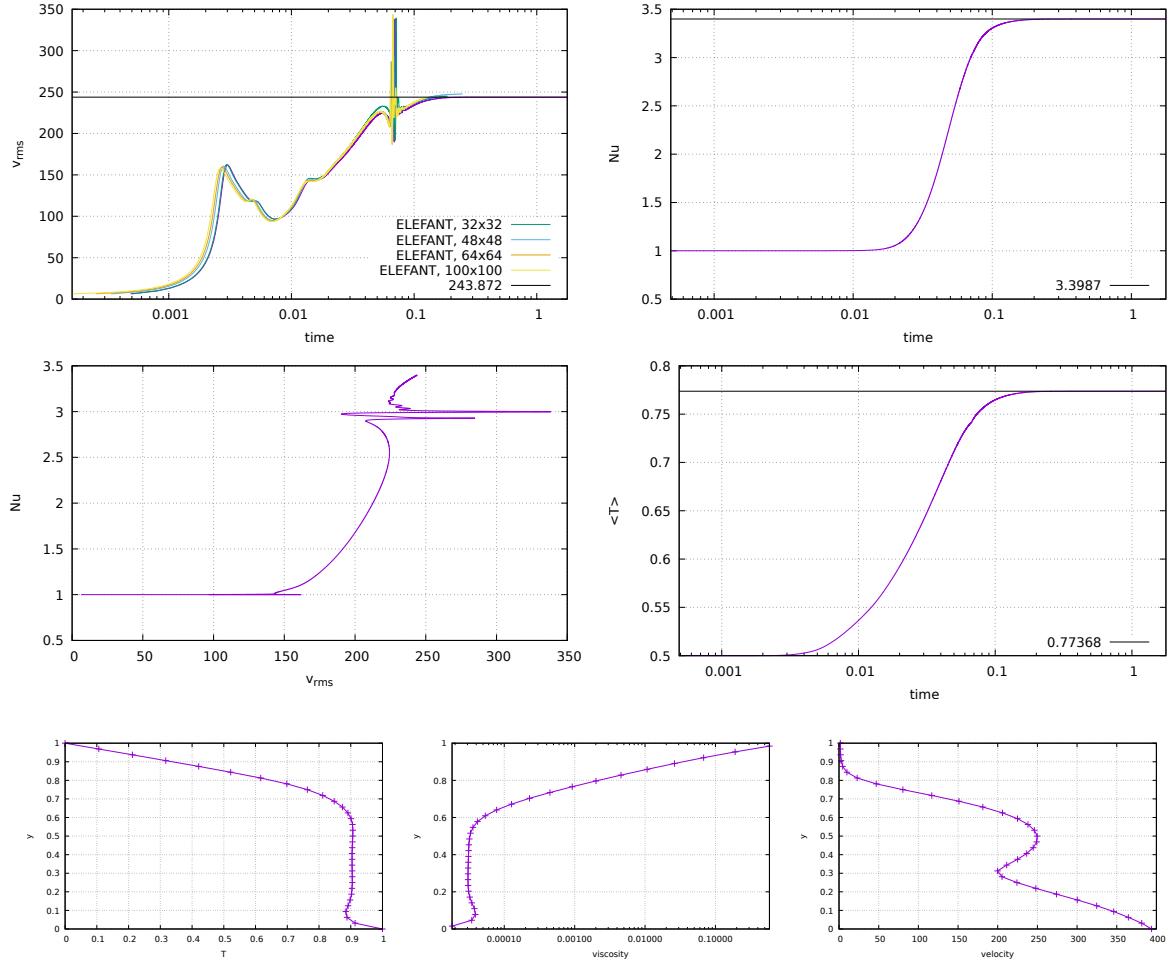
Case 0: Newtonian case, a la Blankenbach et al., 1989

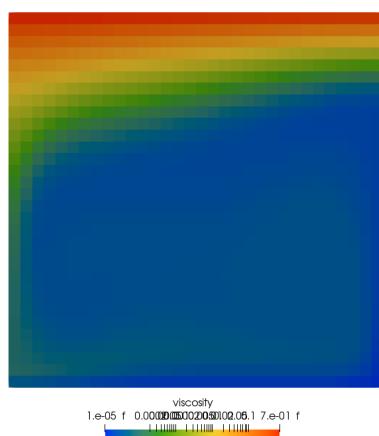
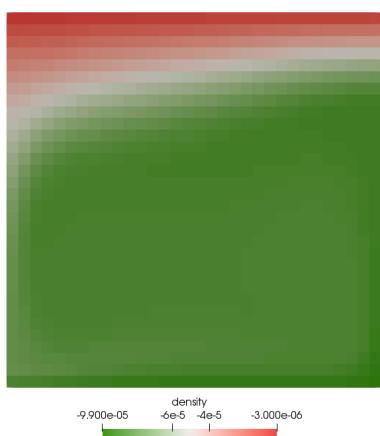
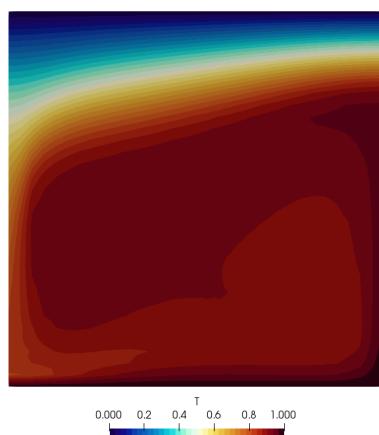
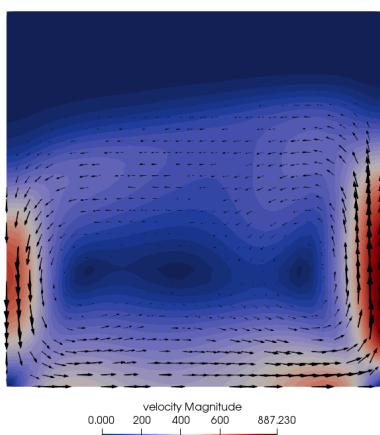
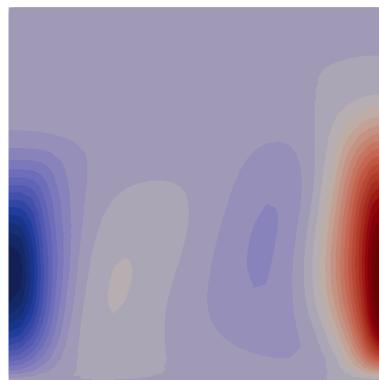
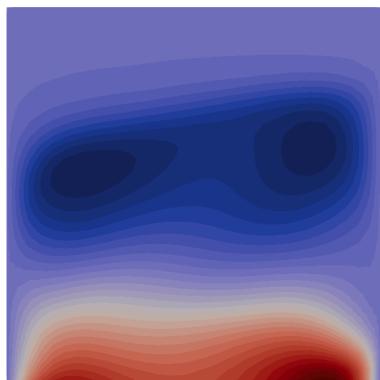


Case 1

In this case $\mu^* = 0$ and $\sigma_Y = 0$ so that μ_{plast} can be discarded. The CFL number is set to 0.5 and the viscosity is given by $\mu(T, z, \dot{\epsilon}) = \mu_{lin}(T, z)$. And since $\Delta\mu_z = 1$ then $\gamma_z = 0$ so that $\mu_{lin}(T, z) = \exp(-\gamma_T T)$

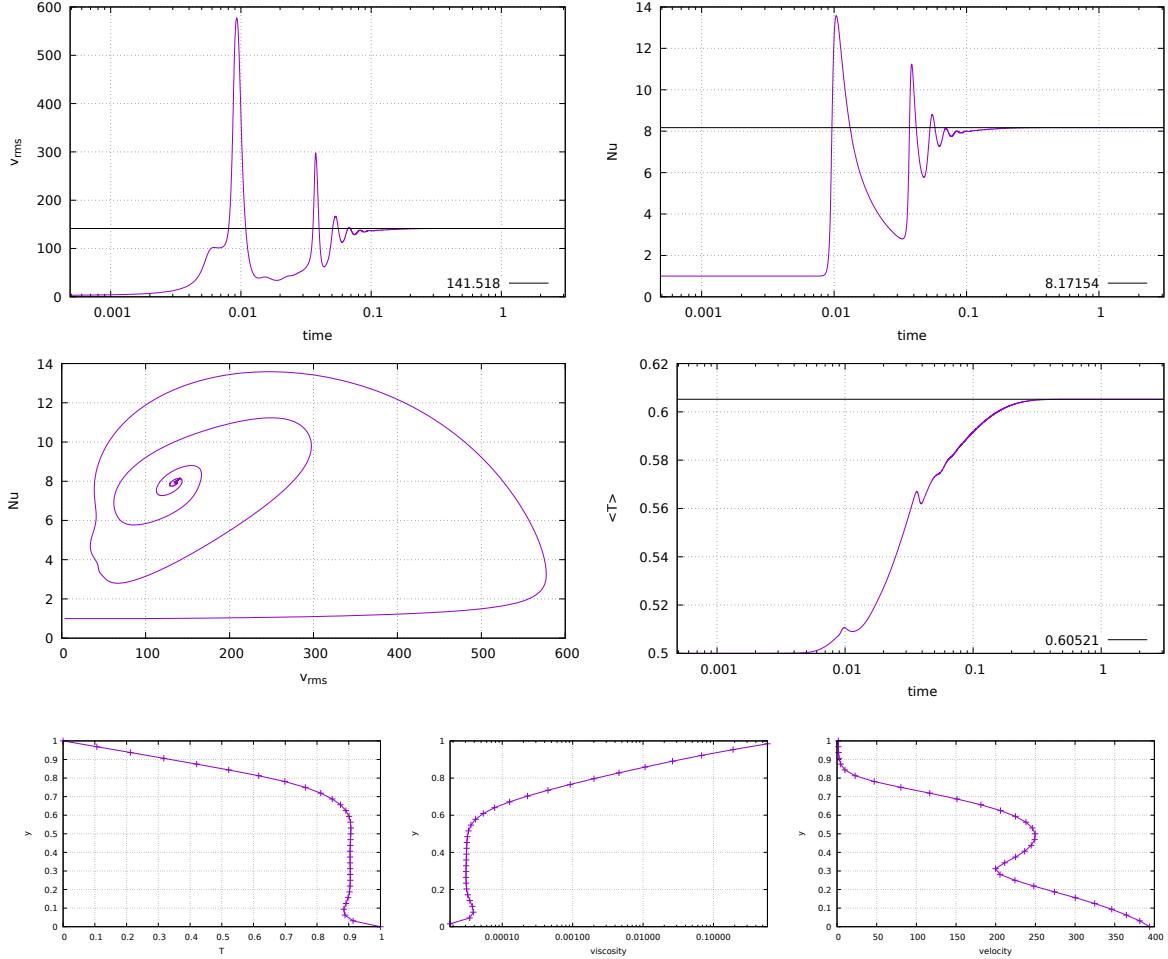
Code	YACC 100 × 100	Plaatjes 128×128r	CHIC 80 × 80	GAIA 100r×100r	StreamV 80 × 80	StagYY 128×128r	FEniCS 80 × 80	Fluidity 128 × 128	ELEFANT 100 × 100	ASPECT 64 × 64	MC3D 100 × 100
<i>Case 1</i>											
$\langle T \rangle$	0.7767	0.7759	0.7758	0.7759	0.776	0.776	0.7759	0.7758	0.7758	0.7768	0.779
Nu_{top}	3.4298	3.4159	3.4260	3.4213	3.4091	3.419	3.5889	3.4253	3.4214	3.4305	3.3129
Nu_{bot}	3.3143	3.4159	3.4259	3.4213	3.4091	3.419	3.4231	3.3795	3.313	3.4142	3.3139
u_{rms}	251.7997	249.54	249.2985	250.0738	252.0906	249.541	249.5730	248.9252	249.134	251.3069	296.6156
u_{surf}^{rms}	1.8298	1.878	1.8999	1.8836	1.8823	1.8723	1.8698	1.8474	1.8642	1.8695	1.1114
u_{surf}^{max}	2.5516	2.618	2.6477	2.6254	2.64	2.6104	2.6066	2.5761	2.6119	2.6064	1.5329
$\langle W \rangle$	2.4583	2.369	2.431	2.4121	2.4071	2.4189	2.4246	2.4148	2.4316	2.4282	2.5548
$\langle \Phi \rangle / Ra$	2.4333	2.4119	2.4189	2.4165	2.392	2.4182	2.4246	2.4148	2.4276	2.4281	2.31916
δ	1.02%	1.78%	0.50%	0.18%	0.63%	0.03%	< 0.01%	< 0.01%	0.16%	< 0.01%	9.22%

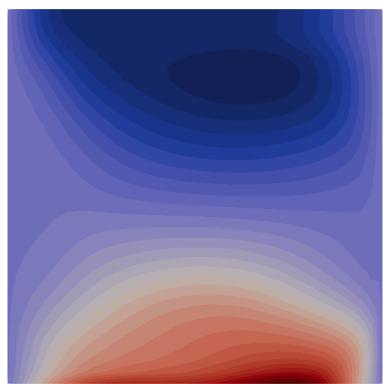




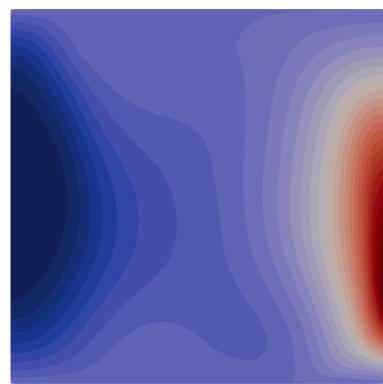
Case 2

$\langle T \rangle$	0.5289	0.5276	0.5276	0.5289	0.5283	0.527304	0.527521	0.5274	0.5277	0.5278	0.5364
Nu_{top}	6.5572	6.6156	6.6074	6.5913	6.6356	6.61082	6.68224	6.6401	6.5912	6.6249	7.4376
Nu_{bot}	6.5243	6.6158	6.6073	6.5913	6.6356	6.61082	6.66899	6.6326	6.5834	6.6267	7.4376
U_{RMS}	79.6202	79.1358	79.0181	78.6652	79.4334	78.9903	79.0684	79.0318	79.1105	79.1996	98.8912
U_{RMS}^{surf}	75.4814	75.1727	75.0434	74.1719	74.8587	75.0606	75.0975	75.0827	74.7596	75.1903	93.5057
U_{\max}^{surf}	89.2940	88.9715	88.8130	87.6118	89.1444	88.823	88.8753	88.85	88.9146	88.9848	123.046
η_{\min}	1.9174×10^{-4}	1.9220×10^{-4}	1.9448×10^{-4}	1.9204×10^{-4}	1.9900×10^{-4}	1.9574×10^{-4}	1.9167×10^{-4}	1.9200×10^{-4}	1.9860×10^{-4}	1.9178×10^{-4}	1.93×10^{-4}
η_{\max}	1.6773	1.9834	1.6508	1.9670	1.1800	1.6665	1.7446	1.8891	1.5200	1.8831	5.67×10^{-3}
$\langle W \rangle$	5.6512	5.6251	5.6076	5.5903	5.629	5.61012	5.61425	5.6136	5.6216	5.6235	6.751
$\langle \Phi \rangle / \text{Ra}$	5.6463	5.6174	5.6024	5.5434	5.6325	5.60152	5.61425	5.6136	5.6182	5.6235	7.786
δ	0.09%	0.14%	0.09%	0.84%	0.06%	0.15%	<0.01%	<0.01%	0.06%	<0.01%	13.29%

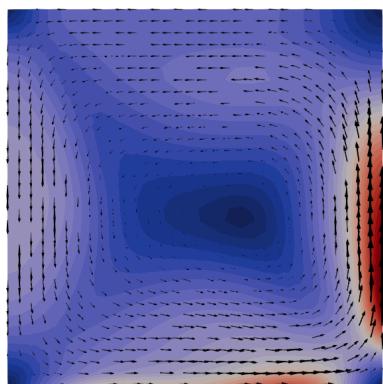




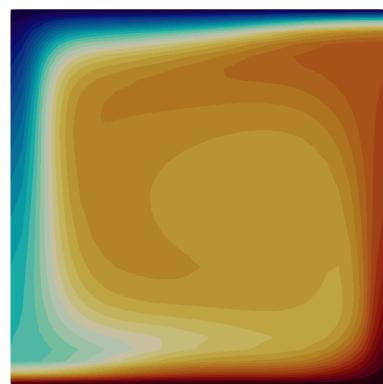
velocity X
-1.437e+02 0 100 200 3.247e+02



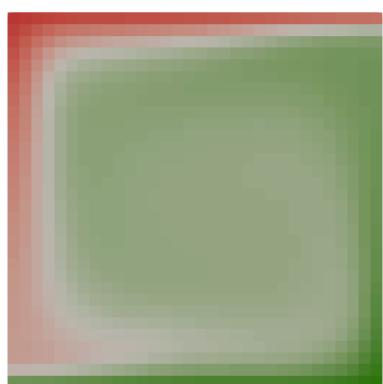
velocity Y
-1.764e+02 0 200 4.549e+02



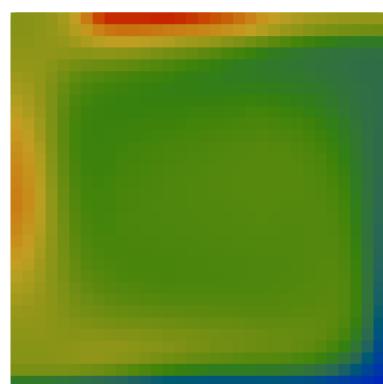
velocity Magnitude
0.000e+00 100 200 300 4.549e+02



T
0.000 0.2 0.4 0.6 0.8 1.000



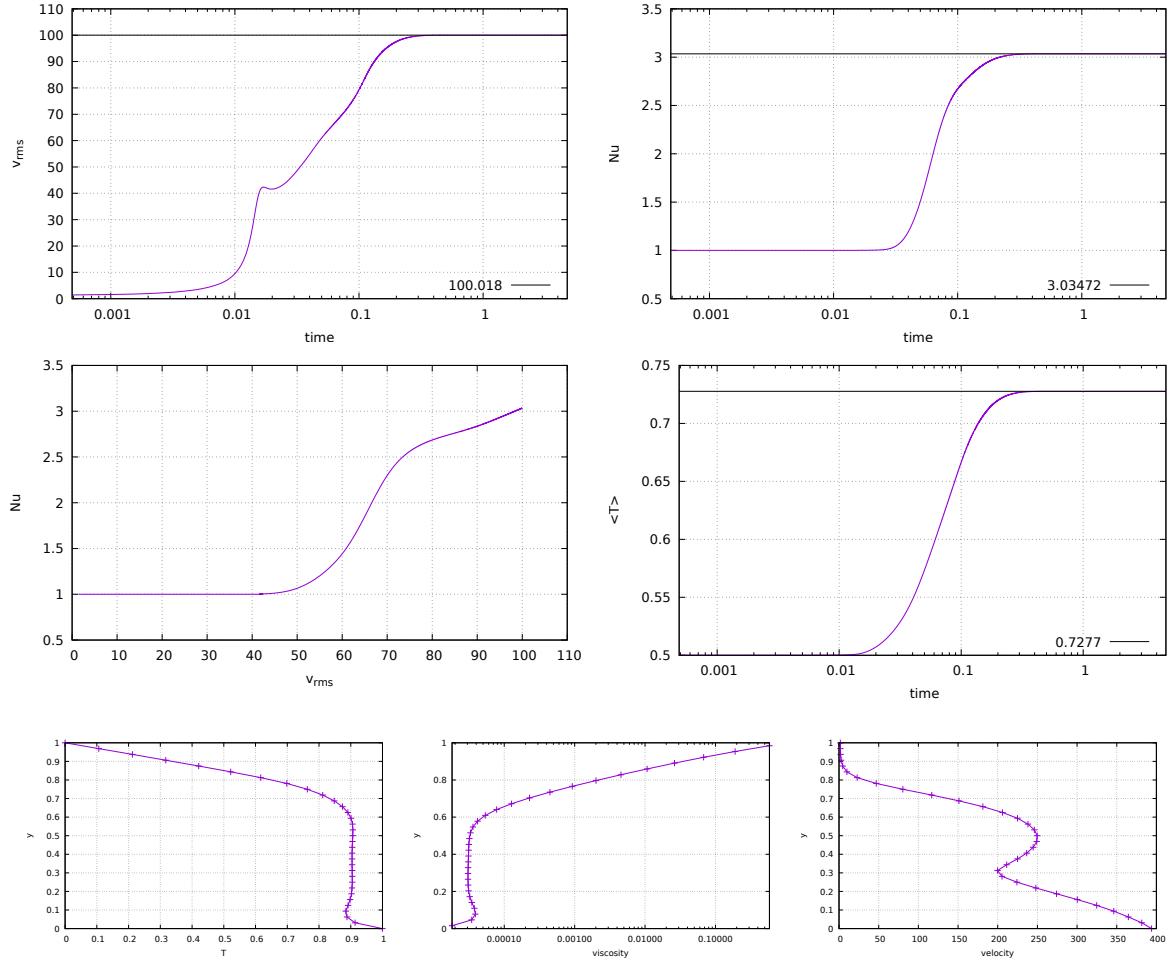
density
-9.900e-05 -6e-5 -4e-5 -1e-5 -2.000e-06

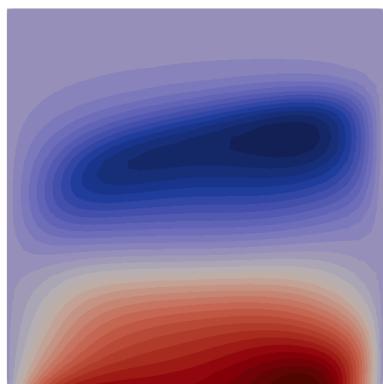


viscosity
2.300e-05 0.00002 0.00003 0.00004 0.00005 6.326e-01

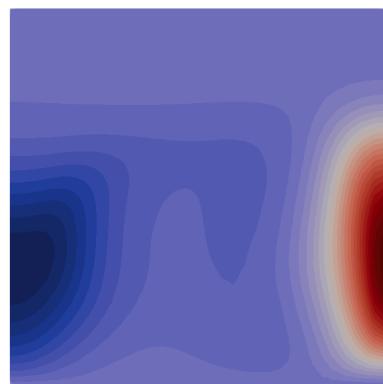
Case 3

$\langle T \rangle$	0.7286	0.7275	0.7271	0.7272	0.7241	0.7274	0.727464	0.7275	0.7275	0.7278	0.7305
Nu_{top}	3.0374	3.0298	3.0324	3.0314	3.0253	3.03025	3.0918	3.0399	3.0347	3.0371	2.9311
Nu_{bot}	2.9628	3.0298	3.0323	3.0314	3.0253	3.03025	3.03487	3.0376	2.9908	3.0410	2.9311
u_{RMS}	100.9467	100.024	99.8701	99.9917	100.197	100.018	100.127	100.0396	100.1208	100.3368	111.6121
u_{surf}^{surf}	2.0374	2.0785	2.0916	2.0835	2.0789	2.07299	2.07301	2.0569	2.0652	2.0727	1.356
u_{max}	2.8458	2.9029	2.9201	2.9094	2.9207	2.89495	2.89501	2.873	2.9019	2.8946	1.8806
η_{min}	4.7907×10^{-5}	4.8140×10^{-5}	4.8014×10^{-5}	4.8047×10^{-5}	4.8400×10^{-5}	4.7951×10^{-5}	4.8081×10^{-5}	4.8000×10^{-5}	4.8080×10^{-5}	4.7972×10^{-5}	10^{-4}
η_{max}	1	0.9987	0.9857	0.9988	0.9010	0.9637	0.9999	1	0.9023	1	1
$\langle W \rangle$	2.0400	2.0028	2.0340	2.0269	2.0235	2.03002	2.03482	2.0298	2.0384	2.0362	2.056
$\langle \Phi \rangle / Ra$	2.0335	2.0277	2.0304	2.0286	2.0164	2.0302	2.03482	2.0298	2.037	2.0362	1.865
δ	0.32%	1.23%	0.18%	0.08%	0.35%	0.01%	<0.01%	<0.01%	0.07%	<0.01%	9.29%

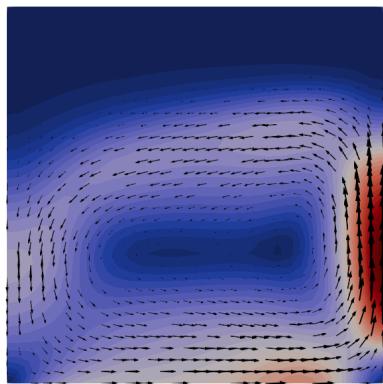




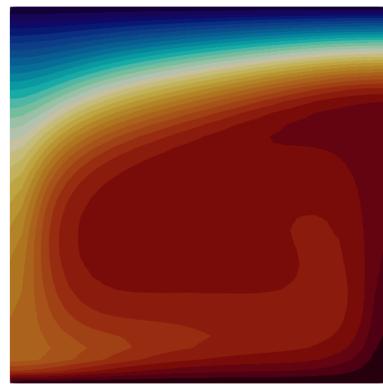
-133.934 0 100 204.210
velocity X



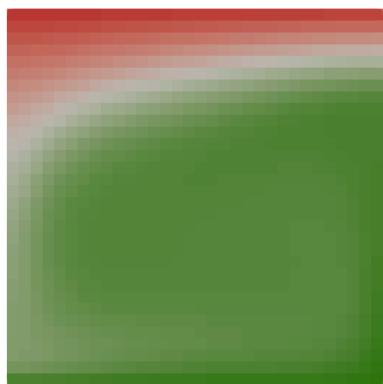
-137.969 0 100 200 330.512
velocity Y



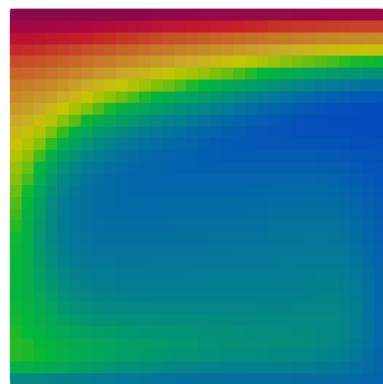
0.000 100 200 330.512
velocity Magnitude



0.000 0.2 0.4 0.6 0.8 1.000
T



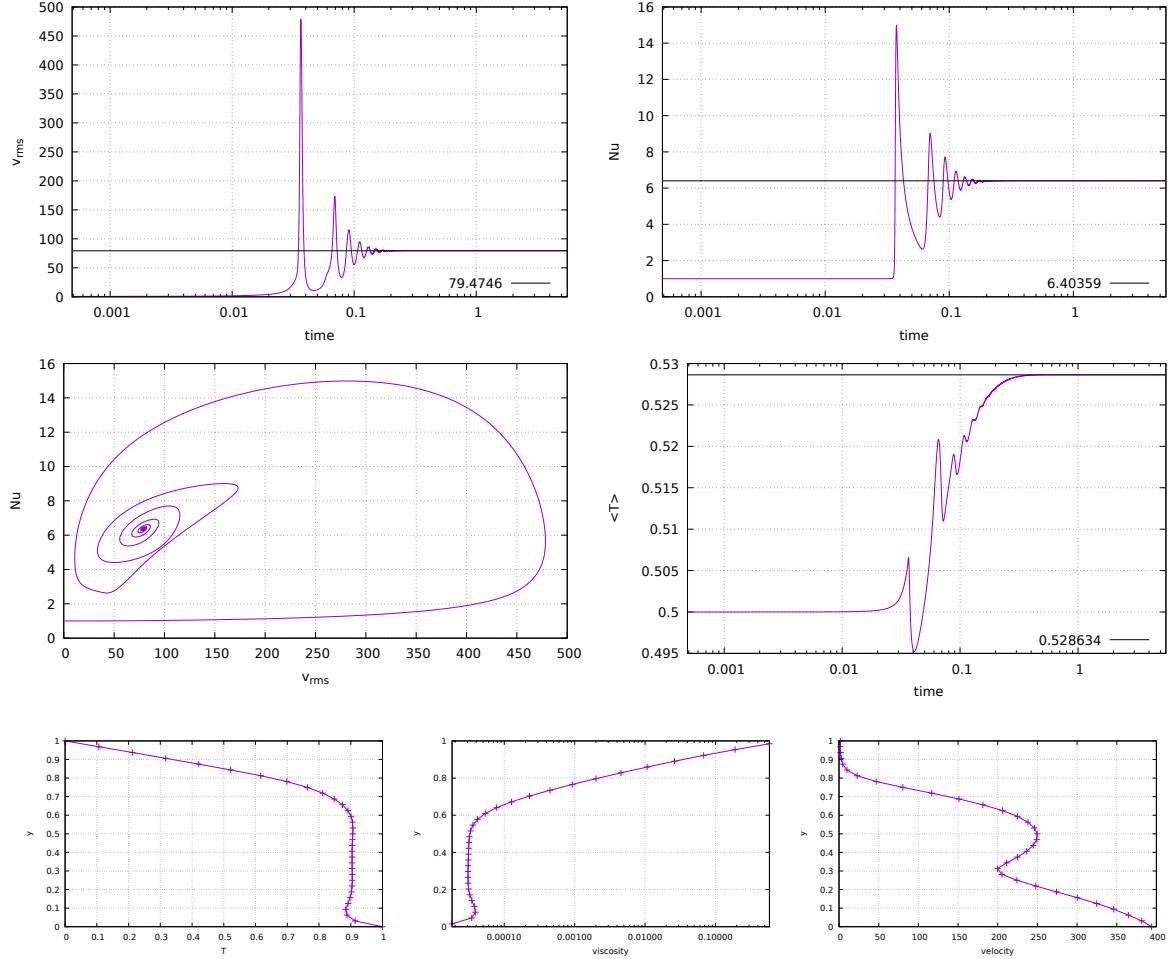
-1.000e-04 -6e-5 -4e-5 -3.000e-06
density

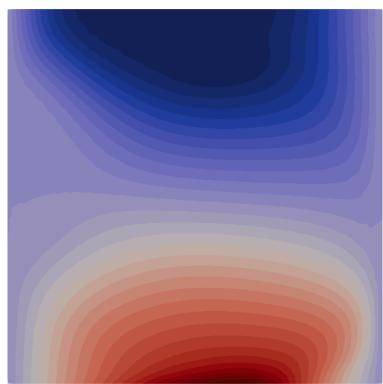


4.800e-05 0.000020000020.001 7.802e-01
viscosity

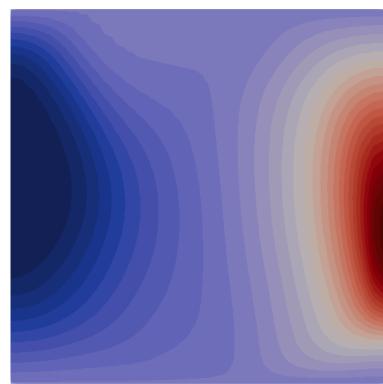
Case 4

$\langle T \rangle$	0.5289	0.5276	0.5276	0.5289	0.5283	0.527304	0.527521	0.5274	0.5277	0.5278	0.5364
Nu_{top}	6.5572	6.6156	6.6074	6.5913	6.6356	6.61082	6.68224	6.6401	6.5912	6.6249	7.4376
Nu_{bot}	6.5243	6.6158	6.6073	6.5913	6.6356	6.61082	6.66899	6.6326	6.5834	6.6267	7.4376
U_{RMS}	79.6202	79.1358	79.0181	78.6652	79.4334	78.9903	79.0684	79.0318	79.1105	79.1996	98.8912
U_{RMS}^{surf}	75.4814	75.1727	75.0434	74.1719	74.8587	75.0606	75.0975	75.0827	74.7596	75.1903	93.5057
U_{max}^{surf}	89.2940	88.9715	88.8130	87.6118	89.1444	88.823	88.8753	88.85	88.9146	88.9848	123.046
η_{min}	1.9174×10^{-4}	1.9220×10^{-4}	1.9448×10^{-4}	1.9204×10^{-4}	1.9900×10^{-4}	1.9574×10^{-4}	1.9167×10^{-4}	1.9200×10^{-4}	1.9860×10^{-4}	1.9178×10^{-4}	1.93×10^{-4}
η_{max}	1.6773	1.9834	1.6508	1.9670	1.1800	1.6665	1.7446	1.8891	1.5200	1.8831	5.67×10^{-3}
$\langle W \rangle$	5.6512	5.6251	5.6076	5.5903	5.629	5.61012	5.61425	5.6136	5.6216	5.6235	6.751
$\langle \Phi \rangle / Ra$	5.6463	5.6174	5.6024	5.5434	5.6325	5.60152	5.61425	5.6136	5.6182	5.6235	7.786
δ	0.09%	0.14%	0.09%	0.84%	0.06%	0.15%	<0.01%	<0.01%	0.06%	<0.01%	13.29%

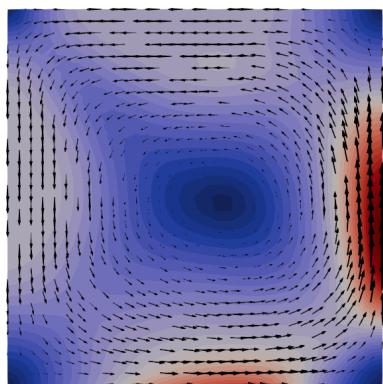




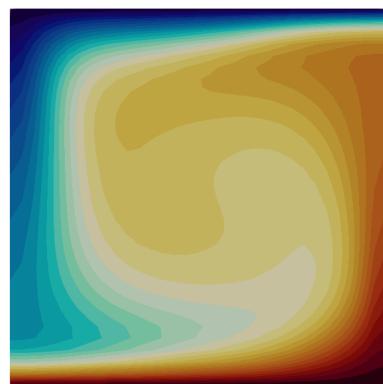
velocity X
-92.305 -50 0 50 100 153.873



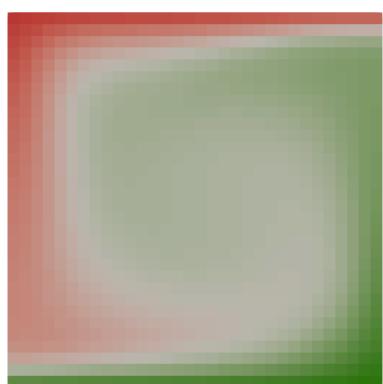
velocity Y
-97.677 0 100 209.053



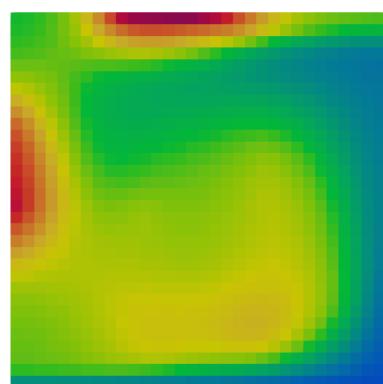
velocity Magnitude
0.000 50 100 150 209.053



T
0.000 0.2 0.4 0.6 0.8 1.000

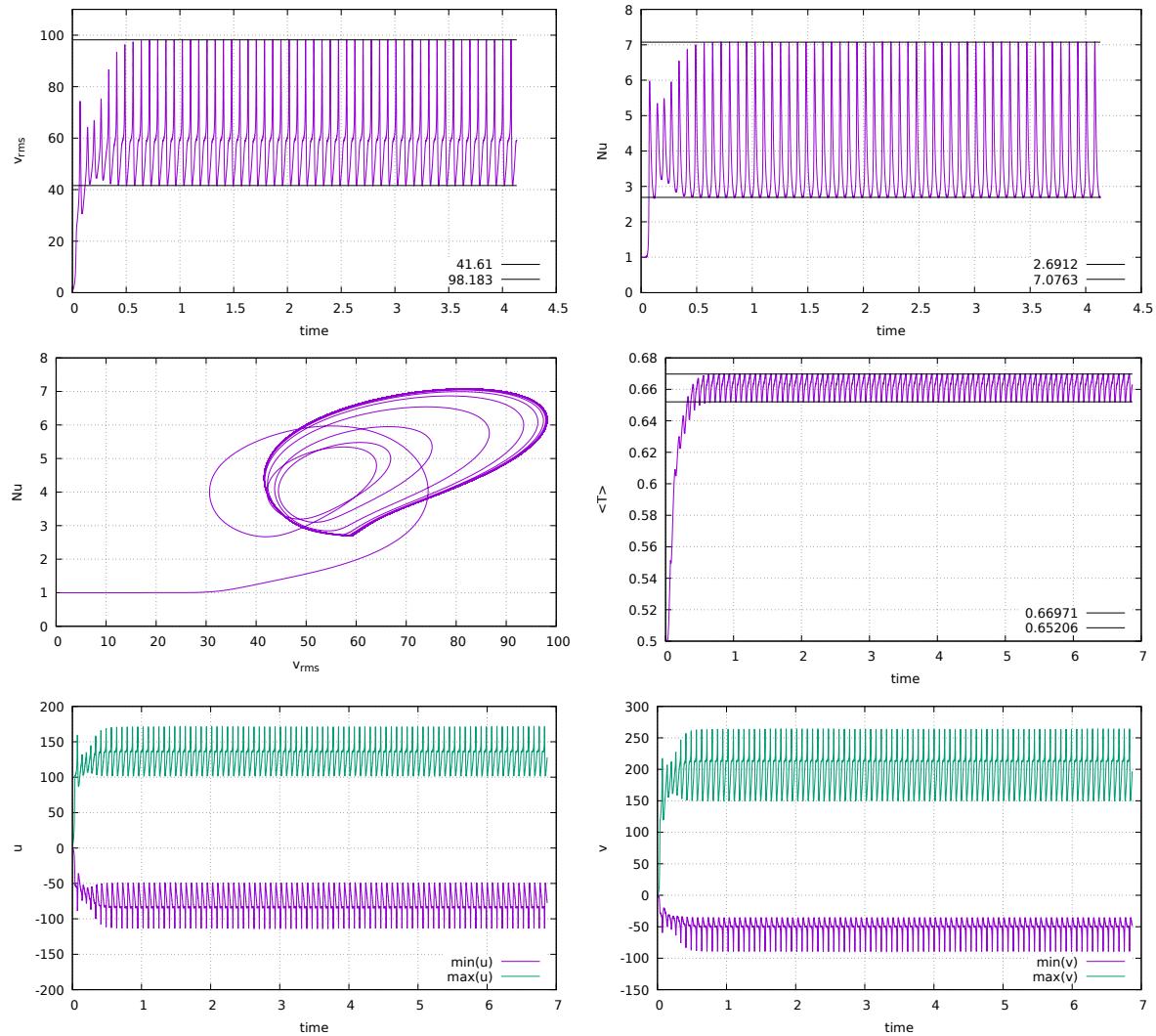


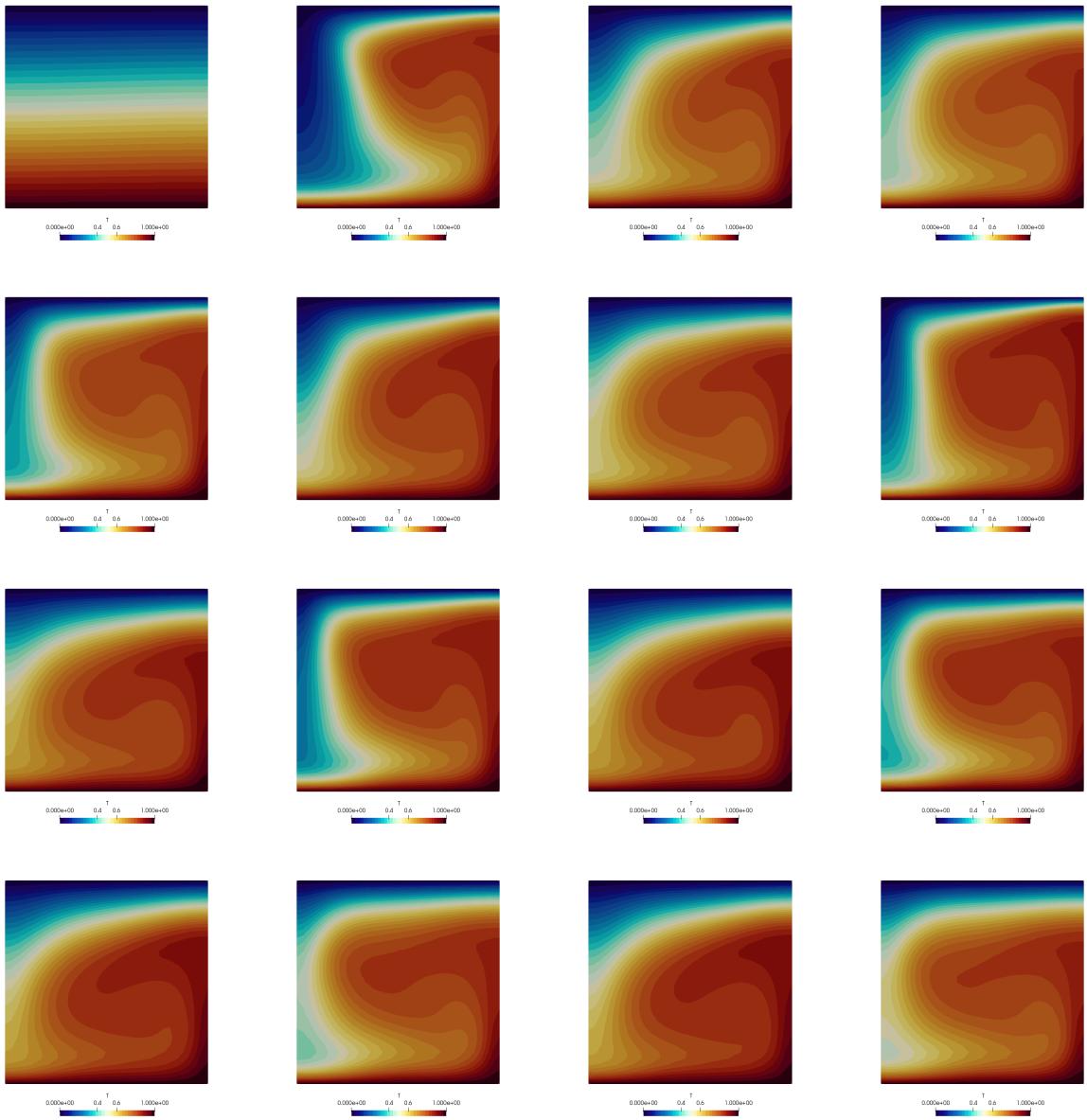
density
-9.900e-05 -5e-05 -4e-05 -1.000e-06



viscosity
0.00021 0.00203 0.00205 0.1 0.75499

Case 5

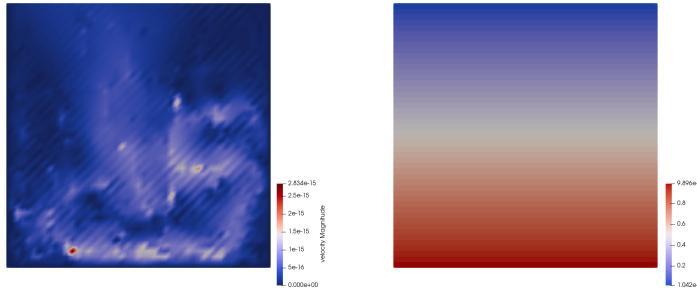




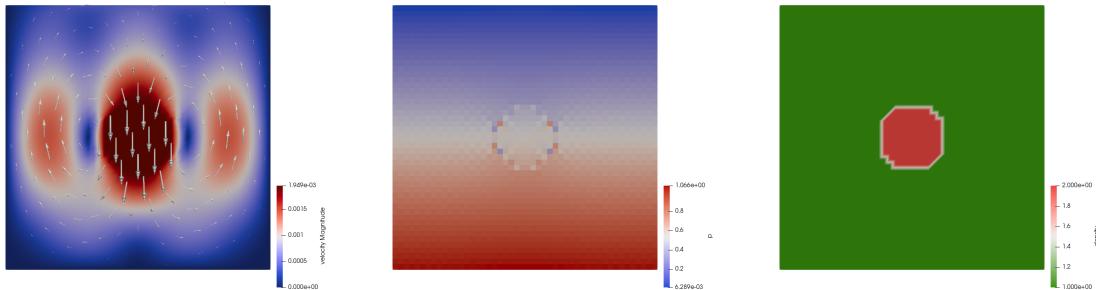
Stone 29: open boundary conditions

In what follows we will investigate the use of the so-called open boundary conditions in the very simple context of a 2D Stokes sphere experiment.

We start with a domain without the sphere. Essentially, it is what people would call an aquarium. Free slip boundary conditions are prescribed on the sides and no-slip conditions at the bottom. The top surface is left free. The fluid has a density $\rho_0 = 1$ and a viscosity $\eta_0 = 1$. In the absence of any density difference in the domain there is no active buoyancy force so that we expect a zero velocity field and a lithostatic pressure field. This is indeed what we recover:



If we now implement a sphere parametrised by its density $\rho_s = \rho_0 + 1$, its viscosity $\eta_s = 10^3$ and its radius $R_s = 0.123$ in the middle of the domain, we see clear velocity field which logically shows the sphere falling downward and a symmetric return flow of the fluid on each side:

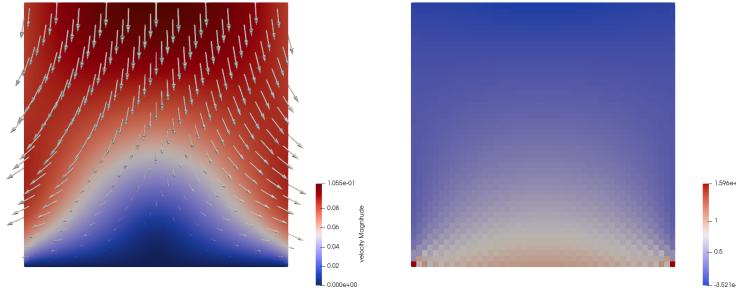


Unfortunately it has been widely documented that the presence of free-slip boundary conditions affects the evolution of subduction [360], even when these are placed rather far from the subduction zone. A proposed solution to this problem is the use of 'open boundary conditions' which are in fact stress boundary conditions. The main idea is to prescribe a stress on the lateral boundaries (instead of free slip) so that it balances out exactly the existing lithostatic pressure inside the domain along the side walls. Only pressure deviations with respect to the lithostatic are responsible for flow and such boundary conditions allow flow across the boundaries.

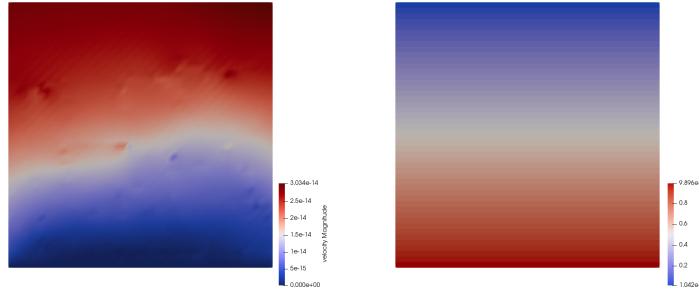
We need the lithostatic pressure and compute it before hand (which is trivial in our case but can prove to be a bit more tedious in real life situations when for instance density varies in the domain as a function of temperature and/or pressure).

```
plith = np.zeros(nnp, dtype=np.float64)
for i in range(0,nnp):
    plith[i]=(Ly-y[i])*rho0*abs(gy)
```

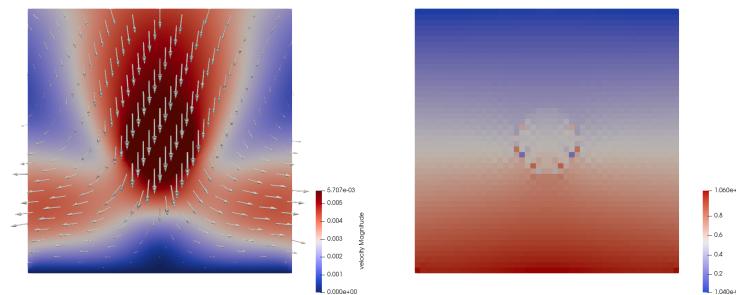
Let us start with a somewhat pathological case: even in the absence of the sphere, what happens when no boundary conditions are prescribed on the sides? The answer is simple: think about an aquarium without side walls, or a broken dam. The velocity field indeed shows a complete collapse of the fluid left and right of the bottom.



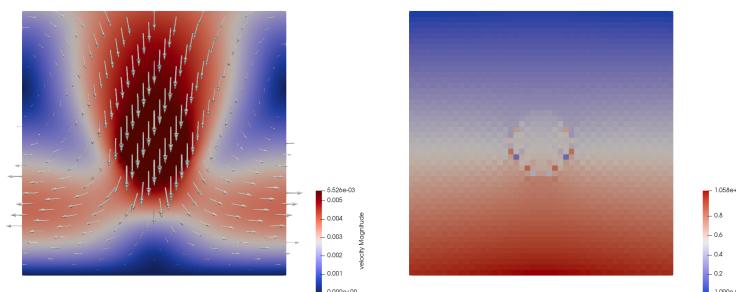
Let us then continue (still with no sphere) but let us now switch on the open boundary conditions. Since the side boundary conditions match the lithostatic pressure we expect no flow at all in the absence of any density perturbation in the system. This is indeed what is recovered:



Finally, let us reintroduce the sphere. This time flow is allowed through the left and right side boundaries:



Finally, although horizontal velocity Dirichlet boundary conditions and open boundary conditions are not compatible, the same is not true for the vertical component of the velocity: the open b.c. implementation acts on the horizontal velocity dofs only, so that one can fix the vertical component to zero, as is shown hereunder:



We indeed see that the in/outflow on the sides is perpendicular to the boundaries.

Turning now to the actual implementation, we see that it is quite trivial, since all element edges are vertical, and all have the same vertical dimension h_x . Since we use a Q_0 approximation for the pressure

we need to prescribe a single pressure value in the middle of the element. Finally because of the sign of the normal vector projection onto the x -axis, we obtain:

```
if open_bc_left and x[icon[0, iel]]<eps: # left side
    pmid=0.5*(plith[icon[0, iel]]+plith[icon[3, iel]])
    f_el[0]+=0.5*hy*pmid
    f_el[6]+=0.5*hy*pmid
if open_bc_right and x[icon[1, iel]]>Lx-eps: # right side
    pmid=0.5*(plith[icon[1, iel]]+plith[icon[2, iel]])
    f_el[2]=-0.5*hy*pmid
    f_el[4]=-0.5*hy*pmid
```

These few lines of code are added after the elemental matrices and rhs are built, and before the application of other Dirichlet boundary conditions, and assembly.

features

- $Q_1 \times P_0$ element
- incompressible flow
- mixed formulation
- open boundary conditions
- isoviscous

Stone 30: conservative velocity interpolation

In this the Stokes equations are not solved. It is a 2D implementation of the cvi algorithm as introduced in [1895] which deals with the advection of markers. Q_1 and Q_2 basis functions are used and in both cases the cvi algorithm can be toggled on/off. Markers can be distributed regularly or randomly at startup.

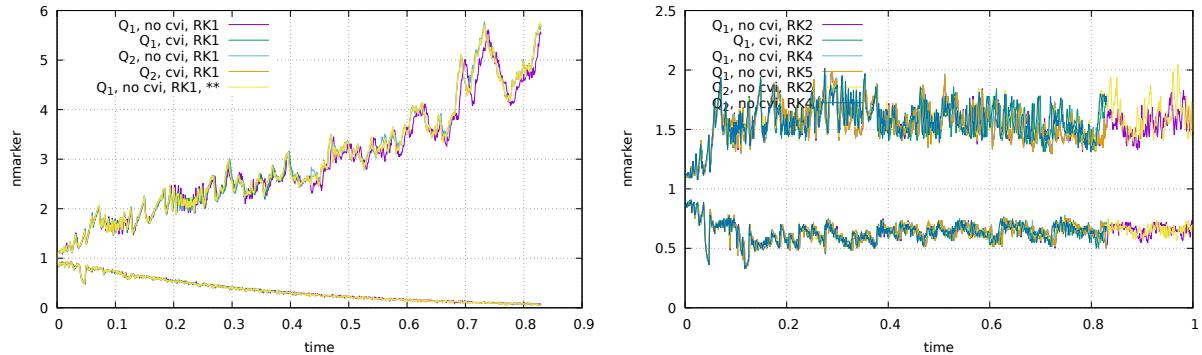
Three velocity fields are prescribed on the mesh:

- the so-called Couette flow of [1895]
- the SolCx solution
- a flow created by means of a stream line function (see fieldstone 32)

Couette flow

SolCx

Streamline flow



In this case RK order seems to be more important than cvi.

Explore why ?!

features

- $Q_1 \times P_0$ element
- incompressible flow
- penalty formulation
- Dirichlet boundary conditions (free-slip)
- direct solver
- isothermal
- non-isoviscous
- analytical solution

Stone 31: conservative velocity interpolation 3D

Stone 32: 2D analytical sol. from stream function

Background theory

The stream function is a function of coordinates and time of an inviscid liquid. It allows to determine the components of velocity by differentiating the stream function with respect to the space coordinates. A family of curves $\Psi = \text{const}$ represent *streamlines*, i.e. the stream function remains constant along a streamline. Although also valid in 3D, this approach is mostly used in 2D because of its relative simplicity

REFERENCES.

In two dimensions the velocity is obtained as follows:

$$\mathbf{v} = \left(\frac{\partial \Psi}{\partial y}, -\frac{\partial \Psi}{\partial x} \right) \quad (713)$$

Provided the function Ψ is a smooth enough function, this automatically insures that the flow is incompressible:

$$\nabla \cdot \mathbf{v} = \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = \frac{\partial^2 \Psi}{\partial x \partial y} - \frac{\partial^2 \Psi}{\partial y \partial x} = 0 \quad (714)$$

Assuming constant viscosity, the Stokes equation writes:

$$-\nabla p + \mu \Delta \mathbf{v} = \rho \mathbf{g} \quad (715)$$

Let us introduce the vector \mathbf{W} for convenience such that in each dimension:

$$W_x = -\frac{\partial p}{\partial x} + \mu \left(\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial x \partial y} \right) = \rho g_x \quad (716)$$

$$W_y = -\frac{\partial p}{\partial y} + \mu \left(\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial x \partial y} \right) = \rho g_y \quad (717)$$

Taking the curl of the vector \mathbf{W} and only considering the component perpendicular to the xy -plane:

$$\frac{\partial V_y}{\partial x} - \frac{\partial V_x}{\partial y} = \frac{\partial \rho g_y}{\partial x} - \frac{\partial \rho g_x}{\partial y} \quad (718)$$

The advantage of this approach is that the pressure terms cancel out (the curl of a gradient is always zero), so that:

$$\frac{\partial}{\partial x} \mu \left(\left(\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial x \partial y} \right) - \frac{\partial}{\partial y} \mu \left(\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial x \partial y} \right) \right) = \frac{\partial \rho g_y}{\partial x} - \frac{\partial \rho g_x}{\partial y} \quad (719)$$

and then replacing u, v by their stream function derivatives yields (for a constant viscosity):

$$\mu \left(\frac{\partial^4 \Psi}{\partial x^4} + \frac{\partial^4 \Psi}{\partial y^4} + 2 \frac{\partial^4 \Psi}{\partial x^2 \partial y^2} \right) = \frac{\partial \rho g_y}{\partial x} - \frac{\partial \rho g_x}{\partial y} \quad (720)$$

or,

$$\nabla^4 \Psi = \left(\frac{\partial^2}{\partial x^2} + \frac{\partial^2}{\partial y^2} \right) \left(\frac{\partial^2}{\partial x^2} + \frac{\partial^2}{\partial y^2} \right) \Psi = \frac{\partial \rho g_y}{\partial x} - \frac{\partial \rho g_x}{\partial y} \quad (721)$$

These equations are also to be found in the geodynamics literature, eee Eq. 1.43 of Tackley book, p 70-71 of Gerya book.

A simple application

I wish to arrive at an analytical formulation for a 2D incompressible flow in the square domain $[-1 : 1] \times [-1 : 1]$. The fluid has constant viscosity $\mu = 1$ and is subject to free slip boundary conditions on all sides. For reasons that will become clear in what follows I postulate the following stream function:

$$\Psi(x, y) = \sin(m\pi x) \sin(n\pi y) \quad (722)$$

We have the velocity being defined as:

$$\mathbf{v} = (u, v) = \left(\frac{\partial \Psi}{\partial y}, -\frac{\partial \Psi}{\partial x} \right) = (n\pi \sin(m\pi x) \cos(n\pi y), -m\pi \cos(m\pi x) \sin(n\pi y)) \quad (723)$$

The strain rate components are then:

$$\dot{\varepsilon}_{xx} = \frac{\partial u}{\partial x} = mn\pi^2 \cos(m\pi x) \cos(n\pi y) \quad (724)$$

$$\dot{\varepsilon}_{yy} = \frac{\partial v}{\partial y} = -mn\pi^2 \cos(m\pi x) \cos(n\pi y) \quad (725)$$

$$2\dot{\varepsilon}_{xy} = \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \quad (726)$$

$$= \frac{\partial^2 \Psi}{\partial y^2} - \frac{\partial^2 \Psi}{\partial x^2} \quad (727)$$

$$= -n^2\pi^2\Psi + m^2\pi^2\Psi \quad (728)$$

$$= (m^2 - n^2)\pi^2 \sin(m\pi x) \sin(n\pi y) \quad (729)$$

Note that if $m = n$ the last term is identically zero, which is not desirable (flow is too 'simple') so in what follows we will prefer $m \neq n$.

It is also easy to verify that $u = 0$ on the sides and $v = 0$ at the top and bottom and that the term $\dot{\varepsilon}_{xy}$ is null on all four sides, thereby guaranteeing free slip.

Our choice of stream function yields:

$$\nabla^4 \Psi = \frac{\partial^4 \Psi}{\partial x^4} + \frac{\partial^4 \Psi}{\partial y^4} + 2 \frac{\partial^2 \Psi}{\partial x^2 \partial y^2} = \pi^4 (m^4 \Psi + n^4 \Psi + 2m^2 n^2 \Psi) = (m^4 + n^4 + 2m^2 n^2) \pi^4 \Psi$$

We assume $g_x = 0$ and $g_y = -1$ so that we simply have

$$(m^4 + n^4 + 2m^2 n^2) \pi^4 \Psi = -\frac{\partial \rho}{\partial x} \quad (730)$$

so that (assuming the integration constant to be zero):

$$\rho(x, y) = \frac{m^4 + n^4 + 2m^2 n^2}{m} \pi^3 \cos(m\pi x) \sin(n\pi y)$$

The x -component of the momentum equation is

$$-\frac{\partial p}{\partial x} + \frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} = -\frac{\partial p}{\partial x} - m^2 n \pi^3 \sin(m\pi x) \cos(n\pi y) - n^3 \pi^3 \sin(m\pi x) \cos(n\pi y) = 0$$

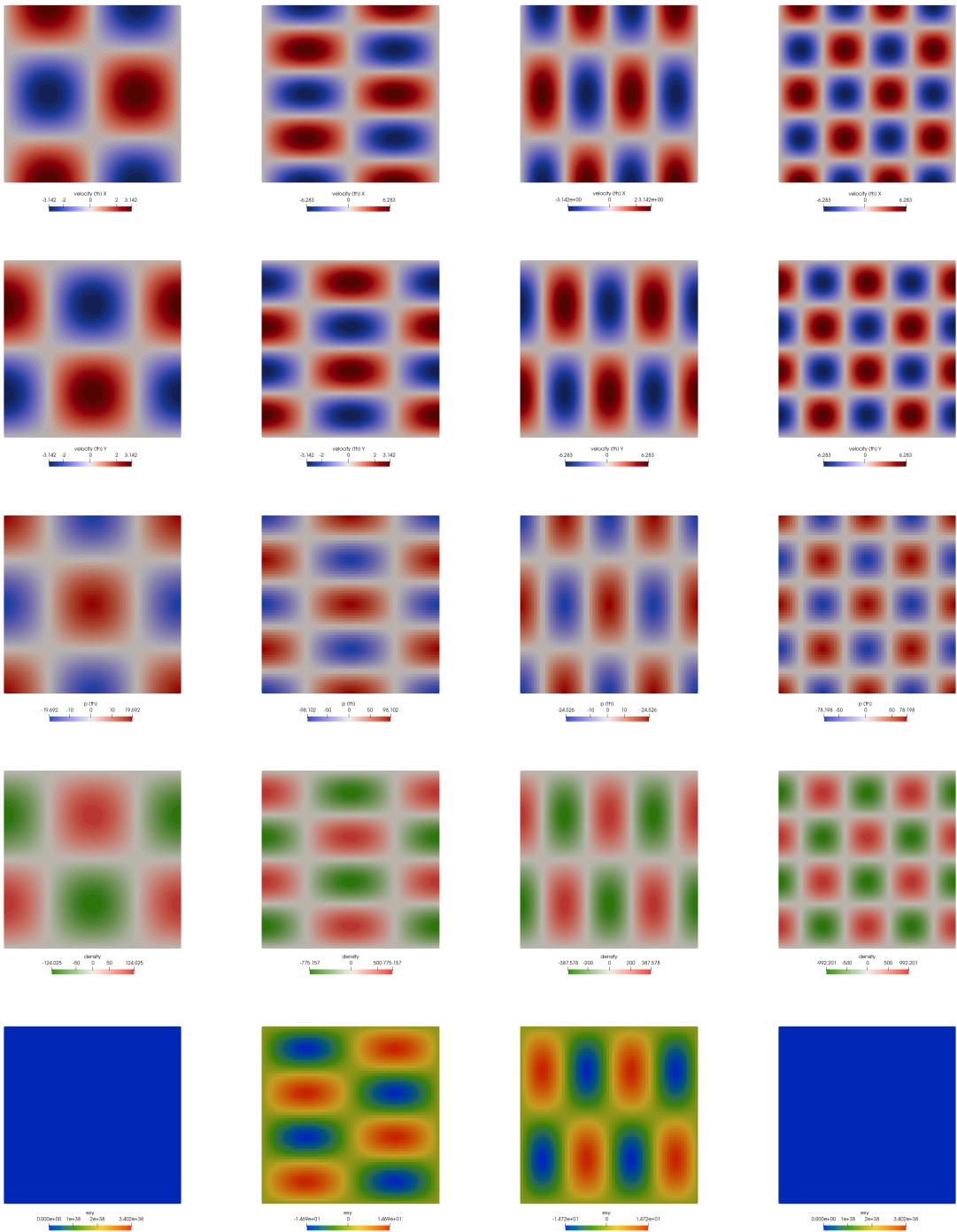
so

$$\frac{\partial p}{\partial x} = -(m^2 n + n^3) \pi^3 \sin(m\pi x) \cos(n\pi y)$$

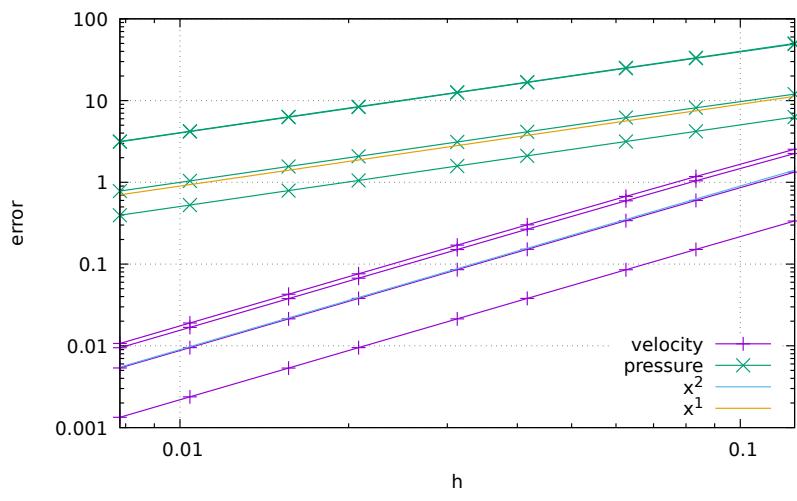
and the pressure field is then (once again neglecting the integration constant):

$$p(x, y) = \frac{m^2 n + n^3}{m} \pi^2 \cos(\pi x) \cos(\pi y)$$

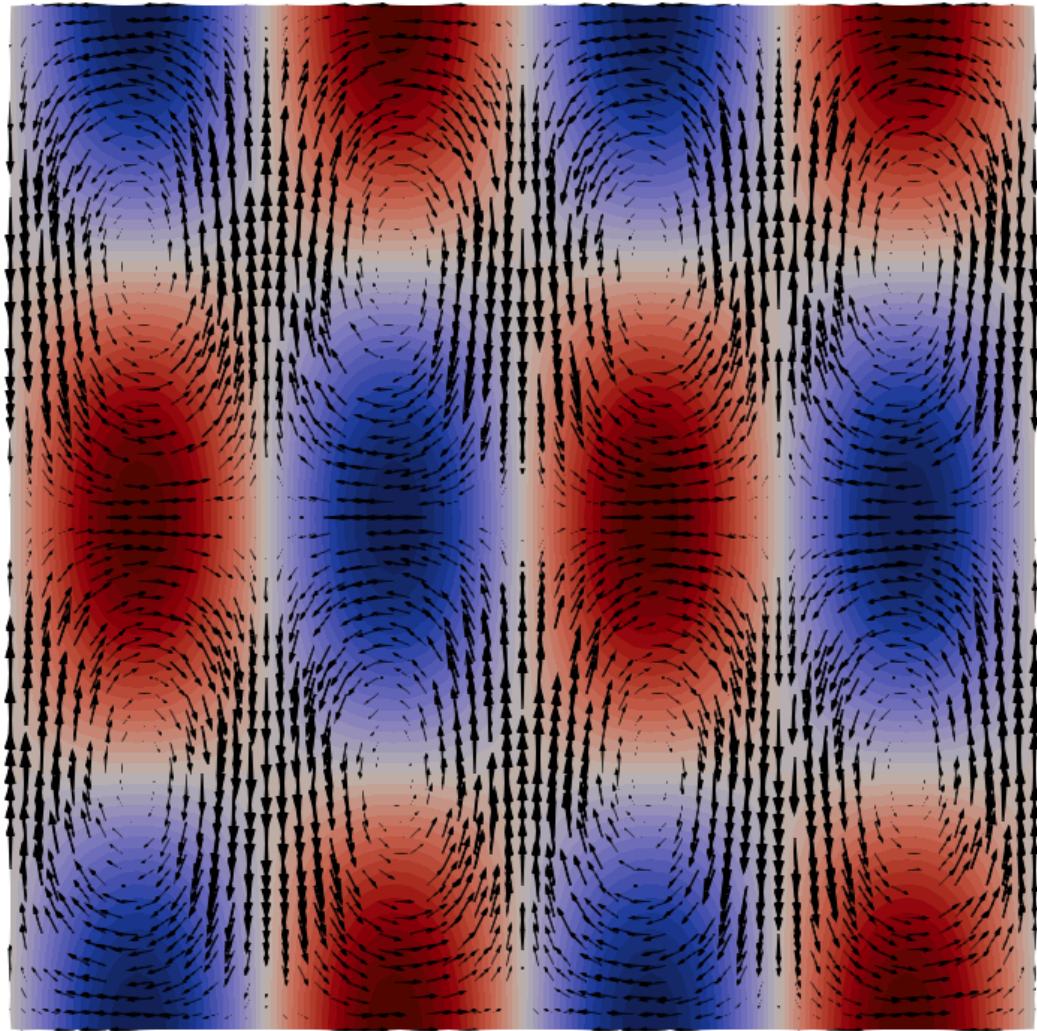
Note that in this case $\int p dV = 0$ so that volume normalisation of the pressure is turned on (when free slip boundary conditions are prescribed on all sides the pressure is known up to a constant and this undeterminacy can be lifted by adding an additional constraint to the pressure field).



Top to bottom: Velocity components u and v , pressure p , density ρ and strain rate $\dot{\epsilon}_{xy}$. From left to right:
 $(m, n) = (1, 1), (m, n) = (1, 2), (m, n) = (2, 1), (m, n) = (2, 2)$



Errors for velocity and pressure for $(m, n) = (1, 1), (1, 2), (2, 1), (2, 2)$



Velocity arrows for $(m, n) = (2, 1)$

Stone 33: Convection in an annulus



This fieldstone was developed in collaboration with Rens Elbertsen.

This is based on the community benchmark for viscoplastic thermal convection in a 2D square box [1784] as already carried out in ??.

In this experiment the geometry is an annulus of inner radius $R_1 = 1.22$ and outer radius $R_2 = 2.22$. The rheology and buoyancy forces are identical to those of the box experiment. The initial temperature is now given by:

$$T(r, \theta) = T_c(r) + A s(1 - s) \cos(N_0 \theta) \quad s = \frac{R_2 - r}{R_2 - R_1} \in [0, 1]$$

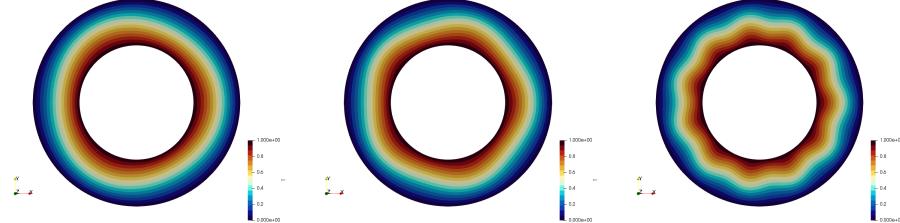
where s is the normalised depth, A is the amplitude of the perturbation and N_0 the number of lobes. In this equation $T_c(r)$ stands for the steady state purely conductive temperature solution which is obtained by solving the Laplace's equation in polar coordinates (all terms in θ are dropped because of radial symmetry) supplemented with two boundary conditions:

$$\Delta T_c = \frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial T}{\partial r} \right) = 0 \quad T(r = R_1) = T_1 = 1 \quad T(r = R_2) = T_2 = 0$$

We obtain

$$T_c(r) = \frac{\log(r/R_2)}{\log(R_1/R_2)}$$

Note that this profile differs from the straight line that is used in [1784] and in section 9.



Examples of initial temperature fields for $N_0 = 3, 5, 11$

Boundary conditions can be either no-slip or free-slip on both inner and outer boundary. However, when free-slip is used on both a velocity null space exists and must be filtered out. In other words, the solver may be able to come up with a solution to the Stokes operator, but that solution plus an arbitrary rotation is also an equally valid solution. This additional velocity field can be problematic since it is used for advecting temperature (and/or compositions) and it also essentially determines the time step value for a chosen mesh size (CFL condition).

For these reasons the nullspace must be removed from the obtained solution after every timestep. There are two types of nullspace removal: removing net angular momentum, and removing net rotations.

We calculate the following output parameters:

- the average temperature $\langle T \rangle$

$$\langle T \rangle = \frac{\int_{\Omega} T d\Omega}{\int_{\Omega} d\Omega} = \frac{1}{V_{\Omega}} \int_{\Omega} T d\Omega \quad (731)$$

- the root mean square velocity v_{rms} as given by equation (62).
- the root mean square of the radial and tangential velocity components as given by equations (64) and (65).

- the heat transfer through both boundaries Q :

$$Q_{inner,outer} = \int_{\Gamma_{i,o}} \mathbf{q} \cdot \mathbf{n} \, d\Gamma \quad (732)$$

- the Nusselt number at both boundaries Nu as given by equations (67) and (68).
- the power spectrum of the temperature field:

$$PS_n(T) = \left| \int_{\Omega} T(r, \theta) e^{in\theta} d\Omega \right|^2. \quad (733)$$

features

- $Q_1 \times P_0$ element
- incompressible flow
- penalty formulation
- Dirichlet boundary conditions
- non-isothermal
- non-isoviscous
- annulus geometry

Stone 34: the Cartesian geometry elastic aquarium

This fieldstone was developed in collaboration with Lukas van de Wiel.

The setup is as follows: a 2D square of elastic material of size L is subjected to the following boundary conditions: free slip on the sides, no slip at the bottom and free at the top. It has a density ρ and is placed in a gravity field $\mathbf{g} = -ge_y$. For an isotropic elastic medium the stress tensor is given by:

$$\boldsymbol{\sigma} = \lambda(\nabla \cdot \mathbf{u})\mathbf{1} + 2\mu\boldsymbol{\varepsilon}$$

where λ is the Lamé parameter and μ is the shear modulus. The displacement field is $\mathbf{u} = (0, u_y(y))$ because of symmetry reasons (we do not expect any of the dynamic quantities to depend on the x coordinate and also expect the horizontal displacement to be exactly zero). The velocity divergence is then $\nabla \cdot \mathbf{u} = \partial u_y / \partial y$ and the strain tensor:

$$\boldsymbol{\varepsilon} = \begin{pmatrix} 0 & 0 \\ 0 & \frac{\partial u_y}{\partial y} \end{pmatrix}$$

so that the stress tensor is:

$$\begin{aligned} \boldsymbol{\sigma} &= \begin{pmatrix} \lambda \frac{\partial u_y}{\partial y} & 0 \\ 0 & (\lambda + 2\mu) \frac{\partial u_y}{\partial y} \end{pmatrix} \\ \nabla \cdot \boldsymbol{\sigma} &= (\partial_x \quad \partial_y) \cdot \begin{pmatrix} \lambda \frac{\partial u_y}{\partial y} & 0 \\ 0 & (\lambda + 2\mu) \frac{\partial u_y}{\partial y} \end{pmatrix} = \begin{pmatrix} 0 \\ (\lambda + 2\mu) \frac{\partial^2 u_y}{\partial y^2} \end{pmatrix} = \begin{pmatrix} 0 \\ \rho g \end{pmatrix} \end{aligned}$$

so that the vertical displacement is then given by:

$$u_y(y) = \frac{1}{2} \frac{\rho g}{\lambda + 2\mu} y^2 + \alpha y + \beta$$

where α and β are two integration constants. We need now to use the two boundary conditions: the first one states that the displacement is zero at the bottom, i.e. $u_y(y=0) = 0$ which immediately implies $\beta = 0$. The second states that the stress at the top is zero (free surface), which implies that $\partial u_y / \partial y(y=L) = 0$ which allows us to compute α . Finally:

$$u_y(y) = \frac{\rho g}{\lambda + 2\mu} \left(\frac{y^2}{2} - Ly \right)$$

The pressure is given by

$$p = -(\lambda + \frac{2}{3}\mu)\nabla \cdot \mathbf{u} = (\lambda + \frac{2}{3}\mu) \frac{\rho g}{\lambda + 2\mu} (L - y) = \frac{\lambda + \frac{2}{3}\mu}{\lambda + 2\mu} \rho g (L - y) = \frac{1 + \frac{2\mu}{3\lambda}}{1 + 2\mu/\lambda} \rho g (L - y)$$

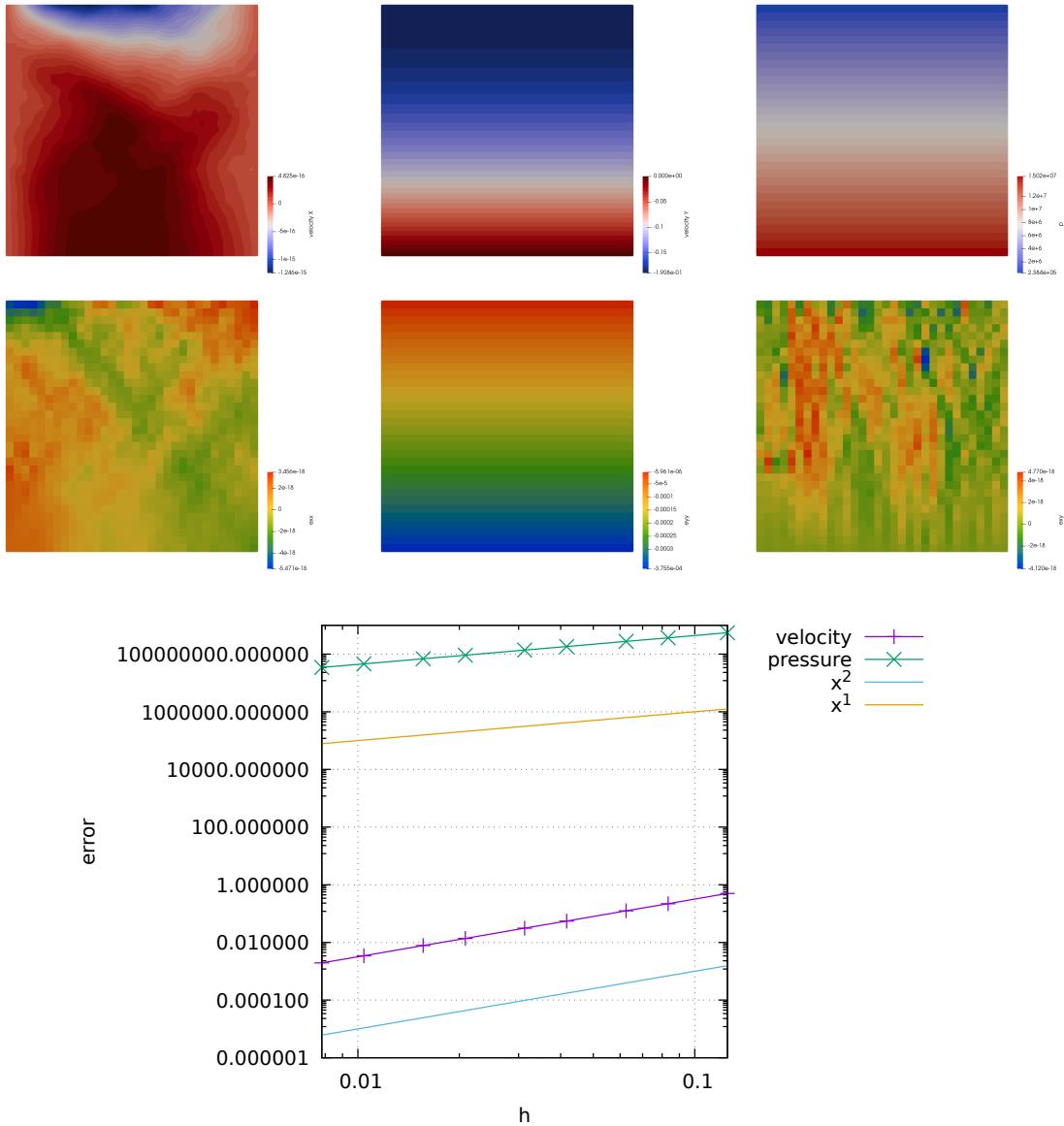
In the incompressible limit, the poisson ratio is $\nu \sim 0.5$. Materials are characterised by a finite Young's modulus E , which is related to ν and λ :

$$\lambda = \frac{E\nu}{(1+\nu)(1-2\nu)} \quad \mu = \frac{E}{2(1+\nu)}$$

It is then clear that for incompressible parameters λ becomes infinite while μ remains finite. In that case the pressure then logically converges to the well known formula:

$$p = \rho g(L - y)$$

In what follows we set $L = 1000\text{m}$, $\rho = 2800$, $\nu = 0.25$, $E = 6 \cdot 10^{10}$, $g = 9.81$.



Stone 35: 2D analytical sol. in annulus from stream function



We seek an exact solution to the incompressible Stokes equations for an isoviscous, isothermal fluid in an annulus. Given the geometry of the problem, we work in polar coordinates. We denote the orthonormal basis vectors by \mathbf{e}_r and \mathbf{e}_θ , the inner radius of the annulus by R_1 and the outer radius by R_2 . Further, we assume that the viscosity μ is constant, which we set to $\mu = 1$ we set the gravity vector to $\mathbf{g} = -g_r \mathbf{e}_r$ with $g_r = 1$.

Given these assumptions, the incompressible Stokes equations in the annulus are [1647]

$$A_r = \frac{\partial^2 v_r}{\partial r^2} + \frac{1}{r} \frac{\partial v_r}{\partial r} + \frac{1}{r^2} \frac{\partial^2 v_r}{\partial \theta^2} - \frac{v_r}{r^2} - \frac{2}{r^2} \frac{\partial u_\theta}{\partial \theta} - \frac{\partial p}{\partial r} = \rho g_r \quad (734)$$

$$A_\theta = \frac{\partial^2 v_\theta}{\partial r^2} + \frac{1}{r} \frac{\partial v_\theta}{\partial r} + \frac{1}{r^2} \frac{\partial^2 v_\theta}{\partial \theta^2} + \frac{2}{r^2} \frac{\partial v_r}{\partial \theta} - \frac{v_\theta}{r^2} - \frac{1}{r} \frac{\partial p}{\partial \theta} = 0 \quad (735)$$

$$\frac{1}{r} \frac{\partial(rv_r)}{\partial r} + \frac{1}{r} \frac{\partial v_\theta}{\partial \theta} = 0 \quad (736)$$

Equations (734) and (735) are the momentum equations in polar coordinates while Equation (736) is the incompressibility constraint. The components of the velocity are obtained from the stream function as follows:

$$v_r = \frac{1}{r} \frac{\partial \Psi}{\partial \theta} \quad v_\theta = - \frac{\partial \Psi}{\partial r}$$

where v_r is the radial component and v_θ is the tangential component of the velocity vector.

The stream function is defined for incompressible (divergence-free) flows in 2D (as well as in 3D with axisymmetry). The stream function can be used to plot streamlines, which represent the trajectories of particles in a steady flow. From calculus it is known that the gradient vector $\nabla \Psi$ is normal to the curve $\Psi = C$. It can be shown that everywhere $\mathbf{u} \cdot \nabla \Psi = 0$ using the formula for \mathbf{u} in terms of Ψ which proves that level curves of Ψ are streamlines:

$$\mathbf{u} \cdot \nabla \Psi = v_r \frac{\partial \Psi}{\partial r} + v_\theta \frac{1}{r} \frac{\partial \Psi}{\partial \theta} = \frac{1}{r} \frac{\partial \Psi}{\partial \theta} \frac{\partial \Psi}{\partial r} - \frac{\partial \Psi}{\partial r} \frac{1}{r} \frac{\partial \Psi}{\partial \theta} = 0$$

In polar coordinates the curl of a vector \mathbf{A} is:

$$\nabla \times \mathbf{A} = \frac{1}{r} \left(\frac{\partial(rA_\theta)}{\partial r} - \frac{\partial A_r}{\partial \theta} \right)$$

Taking the curl of vector \mathbf{A} yields:

$$\frac{1}{r} \left(\frac{\partial(rA_\theta)}{\partial r} - \frac{\partial A_r}{\partial \theta} \right) = \frac{1}{r} \left(- \frac{\partial(\rho g_r)}{\partial \theta} \right)$$

Multiplying on each side by r

$$\frac{\partial(rA_\theta)}{\partial r} - \frac{\partial A_r}{\partial \theta} = - \frac{\partial(\rho g_r)}{\partial \theta}$$

If we now replace A_r and A_θ by their expressions as a function of velocity and pressure, we see that the pressure terms cancel out and assuming the viscosity and the gravity vector to be constant we get: Let us assume the following separation of variables $\boxed{\Psi(r, \theta) = \phi(r)\xi(\theta)}$. Then

$$v_r = \frac{1}{r} \frac{\partial \Psi}{\partial \theta} = \frac{\phi \xi'}{r} \quad v_\theta = - \frac{\partial \Psi}{\partial r} = -\phi' \xi$$

Let us first express A_r and A_θ as functions of Ψ and

$$A_r = \frac{\partial^2 v_r}{\partial r^2} + \frac{1}{r} \frac{\partial v_r}{\partial r} + \frac{1}{r^2} \frac{\partial^2 v_r}{\partial \theta^2} - \frac{v_r}{r^2} - \frac{2}{r^2} \frac{\partial u_\theta}{\partial \theta} \quad (737)$$

$$= \frac{\partial^2}{\partial r^2} \left(\frac{\phi \xi'}{r} \right) + \frac{1}{r} \frac{\partial}{\partial r} \left(\frac{\phi \xi'}{r} \right) + \frac{1}{r^2} \frac{\partial^2}{\partial \theta^2} \left(\frac{\phi \xi'}{r} \right) - \frac{1}{r^2} \left(\frac{\phi \xi'}{r} \right) - \frac{2}{r^2} \frac{\partial}{\partial \theta} (-\phi' \xi) \quad (738)$$

$$= \left(\frac{\phi''}{r} - 2 \frac{\phi'}{r^2} + 2 \frac{\phi}{r^3} \right) \xi' + \left(\frac{\phi'}{r^2} - \frac{\phi}{r^3} \right) \xi' + \frac{\phi}{r^3} \xi''' - \frac{\phi \xi'}{r^3} + \frac{2}{r^2} \phi' \xi' \quad (739)$$

$$= \frac{\phi'' \xi'}{r} + \frac{\phi' \xi'}{r^2} + \frac{\phi \xi'''}{r^3} \quad (740)$$

$$\frac{\partial A_r}{\partial \theta} = \frac{\phi'' \xi''}{r} + \frac{\phi' \xi''}{r^2} + \frac{\phi \xi''''}{r^3} \quad (741)$$

$$(742)$$

$$A_\theta = \frac{\partial^2 v_\theta}{\partial r^2} + \frac{1}{r} \frac{\partial v_\theta}{\partial r} + \frac{1}{r^2} \frac{\partial^2 v_\theta}{\partial \theta^2} + \frac{2}{r^2} \frac{\partial v_r}{\partial \theta} - \frac{v_\theta}{r^2} \quad (743)$$

$$= \frac{\partial^2}{\partial r^2} (-\phi' \xi) + \frac{1}{r} \frac{\partial}{\partial r} (-\phi' \xi) + \frac{1}{r^2} \frac{\partial^2}{\partial \theta^2} (-\phi' \xi) + \frac{2}{r^2} \frac{\partial}{\partial \theta} \left(\frac{\phi \xi'}{r} \right) - \frac{1}{r^2} (-\phi' \xi) \quad (744)$$

$$= -\phi''' \xi - \frac{\phi'' \xi}{r} - \frac{\phi' \xi''}{r^2} + \frac{2\phi \xi''}{r^2} + \frac{\phi' \xi}{r^2} \quad (745)$$

$$\frac{\partial(rA_\theta)}{\partial r} = \quad (746)$$

WRONG:

$$\frac{\partial(r\Delta v)}{\partial r} = \frac{\partial}{\partial r} \left(\frac{\partial}{\partial r} \left(r \frac{\partial v}{\partial r} \right) + \frac{1}{r} \frac{\partial^2 v}{\partial \theta^2} \right) \quad (747)$$

$$= \frac{\partial^2}{\partial r^2} \left(r \frac{\partial v}{\partial r} \right) + \frac{\partial}{\partial r} \left(\frac{1}{r} \frac{\partial^2 v}{\partial \theta^2} \right) \quad (748)$$

$$= \frac{\partial^2}{\partial r^2} \left(r \frac{\partial}{\partial r} \left(-\frac{\partial \Psi}{\partial r} \right) \right) + \frac{\partial}{\partial r} \left(\frac{1}{r} \frac{\partial^2}{\partial \theta^2} \left(-\frac{\partial \Psi}{\partial r} \right) \right) \quad (749)$$

$$= -\frac{\partial^2}{\partial r^2} \left(r \frac{\partial^2 \Psi}{\partial r^2} \right) - \frac{\partial}{\partial r} \left(\frac{1}{r} \frac{\partial^3 \Psi}{\partial \theta^2 \partial r} \right) \quad (750)$$

$$= -2 \frac{\partial^3 \Psi}{\partial r^3} - r \frac{\partial^4 \Psi}{\partial r^4} + \frac{1}{r^2} \frac{\partial^3 \Psi}{\partial \theta^2 \partial r} - \frac{1}{r} \frac{\partial^4 \Psi}{\partial \theta^2 \partial r^2} \quad (751)$$

$$\frac{\partial \Delta u}{\partial \theta} = \frac{\partial}{\partial \theta} \left(\frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial u}{\partial r} \right) + \frac{1}{r^2} \frac{\partial^2 u}{\partial \theta^2} \right) \quad (752)$$

$$= \frac{\partial}{\partial \theta} \left(\frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial u}{\partial r} \right) \right) + \frac{1}{r^2} \frac{\partial^3 u}{\partial \theta^3} \quad (753)$$

$$= \frac{\partial}{\partial \theta} \left(\frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial}{\partial r} \left(\frac{1}{r} \frac{\partial \Psi}{\partial \theta} \right) \right) \right) + \frac{1}{r^2} \frac{\partial^3}{\partial \theta^3} \left(\frac{1}{r} \frac{\partial \Psi}{\partial \theta} \right) \quad (754)$$

$$= \frac{1}{r^3} \frac{\partial^2 \Psi}{\partial \theta^2} - \frac{1}{r^2} \frac{\partial^3 \Psi}{\partial r \partial \theta^2} + \frac{1}{r} \frac{\partial^4 \Psi}{\partial r^2 \partial \theta^2} + \frac{1}{r^3} \frac{\partial^4 \Psi}{\partial \theta^4} \quad (755)$$

Assuming the following separation of variables $\boxed{\Psi(r, \theta) = \phi(r)\xi(\theta)}$:

$$\frac{\partial(r\Delta v)}{\partial r} = -2\phi'''\xi - r\phi''''\xi + \frac{1}{r^2}\phi'\xi'' - \frac{1}{r}\phi''\xi'' \quad (756)$$

$$\frac{\partial \Delta u}{\partial \theta} = \frac{1}{r^3}\phi\xi'' - \frac{1}{r^2}\phi'\xi'' + \frac{1}{r}\phi''\xi'' + \frac{1}{r^3}\phi\xi''' \quad (757)$$

so that

$$\frac{\partial(r\Delta v)}{\partial r} - \frac{\partial \Delta u}{\partial \theta} = -2\phi'''\xi - r\phi''''\xi + \frac{1}{r^2}\phi'\xi'' - \frac{1}{r}\phi''\xi'' - \frac{1}{r^3}\phi\xi'' + \frac{1}{r^2}\phi'\xi'' - \frac{1}{r}\phi''\xi'' - \frac{1}{r^3}\phi\xi'''$$

Further assuming $\boxed{\xi(\theta) = \cos(k\theta)}$, then $\xi'' = -k^2\xi$ and $\xi''' = k^4\xi$ then

$$\frac{\partial(r\Delta v)}{\partial r} - \frac{\partial \Delta u}{\partial \theta} = -2\phi'''\xi - r\phi''''\xi - k^2 \frac{1}{r^2} \phi' \xi + k^2 \frac{1}{r} \phi'' \xi + k^2 \frac{1}{r^3} \phi \xi - k^2 \frac{1}{r^2} \phi' \xi + k^2 \frac{1}{r} \phi'' \xi - k^4 \frac{1}{r^3} \phi \xi$$

By choosing ρ such that $\rho = \lambda(r)\Upsilon(\theta)$ and such that $\partial_\theta \Upsilon = \xi = \cos(k\theta)$ then we have

$$-2\phi'''\xi - r\phi''''\xi - k^2 \frac{1}{r^2} \phi' \xi + k^2 \frac{1}{r} \phi'' \xi + k^2 \frac{1}{r^3} \phi \xi - k^2 \frac{1}{r^2} \phi' \xi + k^2 \frac{1}{r} \phi'' \xi - k^4 \frac{1}{r^3} \phi \xi = -\frac{1}{\eta} \lambda \xi g_r$$

and then dividing by ξ : (IS THIS OK ?)

$$-2\phi'''' - r\phi''''' - k^2 \frac{1}{r^2} \phi' + k^2 \frac{1}{r} \phi'' + k^2 \frac{1}{r^3} \phi - k^2 \frac{1}{r^2} \phi' + k^2 \frac{1}{r} \phi'' - k^4 \frac{1}{r^3} \phi = -\frac{1}{\eta} \lambda g_r$$

$$-2\phi'''' - r\phi''''' - 2k^2 \frac{1}{r^2} \phi' + 2k^2 \frac{1}{r} \phi'' + (k^2 - k^4) \frac{1}{r^3} \phi = -\frac{1}{\eta} \lambda g_r$$

so

$$\lambda(r) = \frac{\eta}{g_r} \left(2\phi'''' + r\phi''''' + 2k^2 \frac{1}{r^2} \phi' - 2k^2 \frac{1}{r} \phi'' - (k^2 - k^4) \frac{1}{r^3} \phi \right)$$

Also not forget $\Upsilon = \frac{1}{k} \sin(k\theta)$

Linking with our paper

We have

$$\phi(r) = -rg(r) \quad (758)$$

$$\phi'(r) = -g(r) - rg'(r) = -f(r) \quad (759)$$

$$\phi''(r) = -f'(r) \quad (760)$$

$$\phi'''(r) = -f''(r) \quad (761)$$

$$\phi''''(r) = -f'''(r) \quad (762)$$

$$f(r) = \frac{\eta_0}{g_0} \left(2f''(r) + rf'''(r) + 2k^2 \frac{1}{r^2} f(r) - 2k^2 \frac{1}{r} f'(r) + (k^2 - k^4) \frac{1}{r^2} g(r) \right)$$

No slip boundary conditions

No-slip boundary conditions inside and outside impose that all components of the velocity must be zero on both boundaries, i.e.

$$\mathbf{v}(r = R_1) = \mathbf{v}(r = R_2) = \mathbf{0}$$

Due to the separation of variables, and since $\xi(\theta) = \cos(k\theta)$ we have

$$u(r, \theta) = \frac{1}{r} \frac{\partial \Psi}{\partial \theta} = \frac{1}{r} \phi \xi' = -\frac{1}{r} \phi(r) k \sin(k\theta) \quad v(r, \theta) = -\frac{\partial \Psi}{\partial r} = -\phi'(r) \xi = -\phi'(r) \cos(k\theta)$$

It is obvious that the only way to insure no-slip boundary conditions is to have

$$\phi(R_1) = \phi(R_2) = \phi'(R_1) = \phi'(R_2) = 0$$

We could then choose

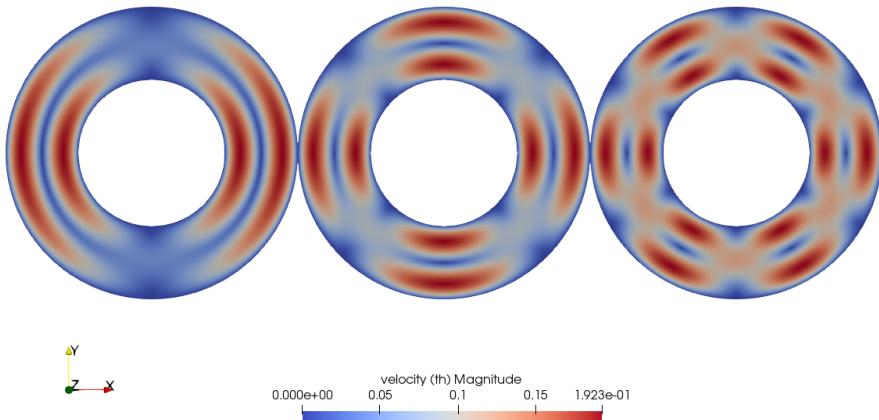
$$\phi(r) = (r - R_1)^2 (r - R_2)^2 \mathcal{F}(r) \quad (763)$$

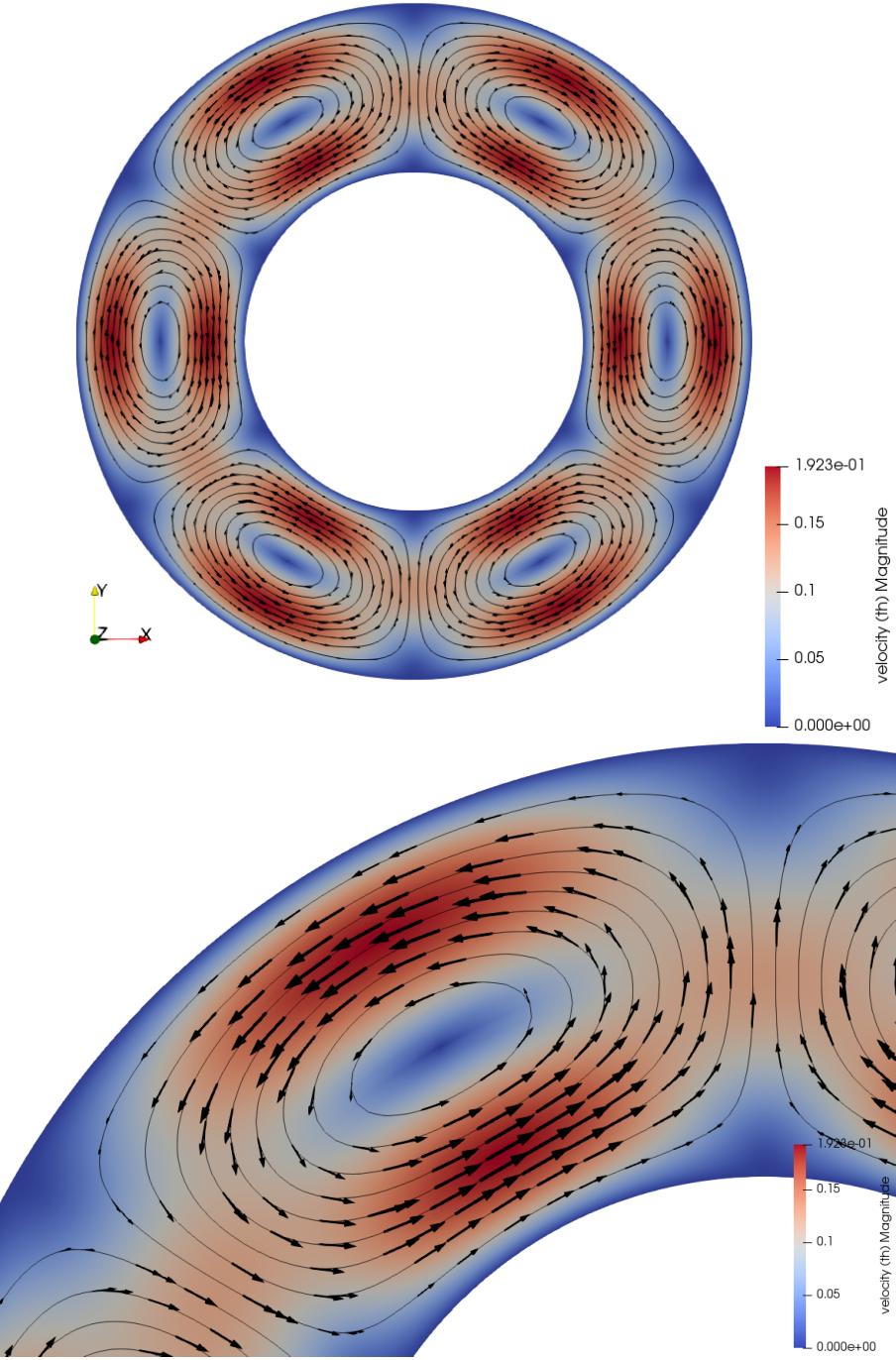
$$\phi'(r) = 2(r - R_1)(r - R_2)^2 \mathcal{F}(r) + 2(r - R_1)^2(r - R_2) \mathcal{F}(r) + (r - R_1)^2(r - R_2)^2 \mathcal{F}'(r) \quad (764)$$

which are indeed identically zero on both boundaries. Here $\mathcal{F}(r)$ is any (smooth enough) function of r . We would then have

$$\boxed{\Psi(r, \theta) = (r - R_1)^2 (r - R_2)^2 \mathcal{F}(r) \cos(k\theta)}$$

In what follows we will take $\mathcal{F}(r) = 1$ for simplicity.





COMPUTE f from ϕ and then the pressure.

Free slip boundary conditions

Before postulating the form of $\phi(r)$, let us now turn to the boundary conditions that the flow must fulfill, i.e. free-slip on both boundaries. Two conditions must be met:

- $\mathbf{v} \cdot \mathbf{n} = 0$ (no flow through the boundaries) which yields $u(r = R_1) = 0$ and $u(r = R_2) = 0$:

$$\frac{1}{r} \frac{\partial \Psi}{\partial \theta}(r = R_1, R_2) = 0 \quad \forall \theta$$

which gives us the first constraint since $\Psi(r, \theta) = \phi(r)\xi(\theta)$:

$$\phi(r = R_1) = \phi(r = R_2) = 0$$

- $(\sigma \cdot n) \times n = \mathbf{0}$ (the tangential stress at the boundary is zero) which imposes: $\sigma_{\theta r} = 0$, with

$$\sigma_{\theta r} = 2\eta \cdot \frac{1}{2} \left(\frac{\partial v}{\partial r} - \frac{v}{r} + \frac{1}{r} \frac{\partial u}{\partial \theta} \right) = \eta \left(\frac{\partial}{\partial r} \left(-\frac{\partial \Psi}{\partial r} \right) - \frac{1}{r} \left(-\frac{\partial \Psi}{\partial r} \right) + \frac{1}{r} \frac{\partial}{\partial \theta} \left(\frac{1}{r} \frac{\partial \Psi}{\partial \theta} \right) \right)$$

Finally Ψ must fulfill (on the boundaries!):

$$-\frac{\partial^2 \Psi}{\partial r^2} + \frac{1}{r} \frac{\partial \Psi}{\partial r} + \frac{1}{r^2} \frac{\partial^2 \Psi}{\partial \theta^2} = 0$$

$$-\phi''\xi + \frac{1}{r}\phi'\xi + \frac{1}{r^2}\phi\xi'' = 0$$

or,

$$-\phi'' + \frac{1}{r}\phi' - k^2 \frac{1}{r^2}\phi = 0$$

Note that this equation is a so-called Euler Differential Equation⁶⁴. Since we are looking for a solution ϕ such that $\phi(R_1) = \phi(R_2) = 0$ then the 3rd term of the equation above is by definition zero on the boundaries. We have to ensure the following equality on the boundary:

$$-\phi'' + \frac{1}{r}\phi' = 0 \quad \text{for } r = R_1, R_2$$

The solution of this ODE is of the form $\phi(r) = ar^2 + b$ and it becomes evident that it cannot satisfy $\phi(r = R_1) = \phi(r = R_2) = 0$.

Separation of variables leads to solutions which cannot fulfill the free slip boundary conditions

⁶⁴<http://mathworld.wolfram.com/EulerDifferentialEquation.html>

Stone 36: the annulus geometry elastic aquarium

This fieldstone was developed in collaboration with Lukas van de Wiel.

The domain is an annulus with inner radius R_1 and outer radius R_2 . It is filled with a single elastic material characterised by a Young's modulus E and a Poisson ratio ν , a density ρ_0 . The gravity $\mathbf{g} = -g_0 \mathbf{e}_r$ is pointing towards the center of the domain.

The problem at hand is axisymmetric so that the tangential component of the displacement vector v_θ is assumed to be zero as well as all terms containing ∂_θ . The components of the strain tensor are

$$\varepsilon_{rr} = \frac{\partial v_r}{\partial r} \quad (765)$$

$$\varepsilon_{\theta\theta} = \frac{v_r}{r} + \frac{1}{r} \frac{\partial v_\theta}{\partial \theta} = \frac{v_r}{r} \quad (766)$$

$$\varepsilon_{r\theta} = \frac{1}{2} \left(\frac{\partial v_\theta}{\partial r} - \frac{v_\theta}{r} + \frac{1}{r} \frac{\partial v_r}{\partial \theta} \right) = 0 \quad (767)$$

so that the tensor simply is

$$\boldsymbol{\varepsilon} = \begin{pmatrix} \varepsilon_{rr} & \varepsilon_{r\theta} \\ \varepsilon_{r\theta} & \varepsilon_{\theta\theta} \end{pmatrix} = \begin{pmatrix} \frac{\partial v_r}{\partial r} & 0 \\ 0 & \frac{v_r}{r} \end{pmatrix} \quad (768)$$

The pressure is

$$p = -\lambda \nabla \cdot \mathbf{v} = -\lambda \left(\frac{1}{r} \frac{\partial(rv_r)}{\partial r} \right) \quad (769)$$

and finally the stress tensor:

$$\boldsymbol{\sigma} = -p \mathbf{1} + 2\mu \boldsymbol{\varepsilon} = \begin{pmatrix} \lambda \frac{1}{r} \frac{\partial(rv_r)}{\partial r} + 2\mu \frac{\partial v_r}{\partial r} & 0 \\ 0 & \lambda \frac{1}{r} \frac{\partial(rv_r)}{\partial r} + 2\mu \frac{v_r}{r} \end{pmatrix} \quad (770)$$

The divergence of the stress tensor is given by [1647]:

$$\nabla \cdot \boldsymbol{\sigma} = \begin{pmatrix} \frac{\partial \sigma_{rr}}{\partial r} + \frac{\sigma_{rr} - \sigma_{\theta\theta}}{r} + \frac{1}{r} \frac{\partial \sigma_{\theta r}}{\partial \theta} \\ \frac{\partial \sigma_{r\theta}}{\partial r} + \frac{1}{r} \frac{\sigma_{\theta\theta}}{\partial \theta} + \frac{\sigma_{r\theta} + \sigma_{\theta r}}{r} \end{pmatrix} \quad (771)$$

Given the diagonal nature of the stress tensor this simplifies to (also remember that $\partial_\theta = 0$):

$$\nabla \cdot \boldsymbol{\sigma} = \begin{pmatrix} \frac{\partial \sigma_{rr}}{\partial r} + \frac{\sigma_{rr} - \sigma_{\theta\theta}}{r} \\ 0 \end{pmatrix} \quad (772)$$

Focusing on the r -component of the stress divergence:

$$(\nabla \cdot \boldsymbol{\sigma})_r = \frac{\partial \sigma_{rr}}{\partial r} + \frac{\sigma_{rr} - \sigma_{\theta\theta}}{r} \quad (773)$$

$$= \frac{\partial}{\partial r} \left[\lambda \frac{1}{r} \frac{\partial(rv_r)}{\partial r} + 2\mu \frac{\partial v_r}{\partial r} \right] + \frac{1}{r} \left[\lambda \frac{1}{r} \frac{\partial(rv_r)}{\partial r} + 2\mu \frac{\partial v_r}{\partial r} - \lambda \frac{1}{r} \frac{\partial(rv_r)}{\partial r} - 2\mu \frac{v_r}{r} \right] \quad (774)$$

$$= \lambda \frac{\partial}{\partial r} \frac{1}{r} \frac{\partial(rv_r)}{\partial r} + 2\mu \frac{\partial^2 v_r}{\partial r^2} + \lambda \frac{1}{r^2} \frac{\partial(rv_r)}{\partial r} + \frac{2\mu}{r} \frac{\partial v_r}{\partial r} - \lambda \frac{1}{r^2} \frac{\partial(rv_r)}{\partial r} - \frac{2\mu v_r}{r^2} \quad (775)$$

$$= \lambda \left(-\frac{v_r}{r^2} + \frac{1}{r} \frac{\partial v_r}{\partial r} + \frac{\partial^2 v_r}{\partial r^2} \right) + 2\mu \frac{\partial^2 v_r}{\partial r^2} + \frac{2\mu}{r} \frac{\partial v_r}{\partial r} - \frac{2\mu v_r}{r^2} \quad (776)$$

$$= -(2\mu + \lambda) \frac{v_r}{r^2} + (2\mu + \lambda) \frac{1}{r} \frac{\partial v_r}{\partial r} + (2\mu + \lambda) \frac{\partial^2 v_r}{\partial r^2} \quad (777)$$

So the momentum conservation in the r direction is

$$(\nabla \cdot \boldsymbol{\sigma} + \rho_0 \mathbf{g})_r = -(2\mu + \lambda) \frac{v_r}{r^2} + (2\mu + \lambda) \frac{1}{r} \frac{\partial v_r}{\partial r} + (2\mu + \lambda) \frac{\partial^2 v_r}{\partial r^2} - \rho_0 g_0 = 0 \quad (778)$$

or,

$$\frac{\partial^2 v_r}{\partial r^2} + \frac{1}{r} \frac{\partial v_r}{\partial r} - \frac{v_r}{r^2} = \frac{\rho_0 g_0}{\lambda + 2\mu} \quad (779)$$

We now look at the boundary conditions. On the inner boundary we prescribe $v_r(r = R_1) = 0$ while free surface boundary conditions are prescribed on the outer boundary, i.e. $\boldsymbol{\sigma} \cdot \mathbf{n} = 0$ (i.e. there is no force applied on the surface).

The general form of the solution can then be obtained:

$$v_r(r) = C_1 r^2 + C_2 r + \frac{C_3}{r} \quad (780)$$

with

$$C_1 = \frac{\rho_0 g_0}{3(\lambda + 2\mu)} \quad C_2 = -C_1 R_1 - \frac{C_3}{R_1^2} \quad C_3 = \frac{k_1 + k_2}{(R_1^2 + R_2^2)(2\mu + \lambda) + (R_2^2 - R_1^2)\lambda} \quad (781)$$

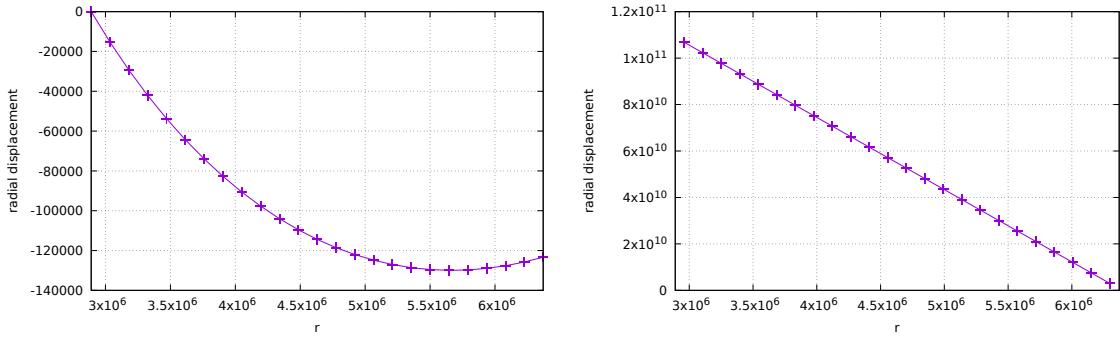
and

$$k_1 = (2\mu + \lambda)C_1(2R_1^2 R_2^3 - R_1^3 R_2^2) \quad k_2 = \lambda C_1(R_1^2 R_2^3 - R_1^3 R_2^2) \quad (782)$$

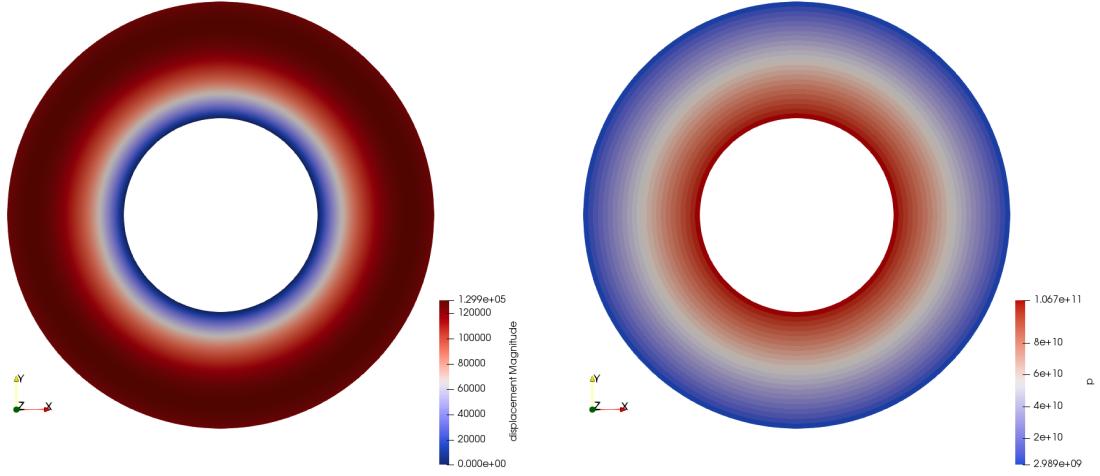
Pressure can then be computed as follows:

$$p = -\lambda \nabla \cdot \mathbf{v} = -\lambda \left(\frac{1}{r} \frac{\partial(r v_r)}{\partial r} \right) = -\lambda \left(\frac{1}{r} (3C_1 r^2 + 2C_2 r) \right) = -\lambda (3C_1 r + 2C_2) \quad (783)$$

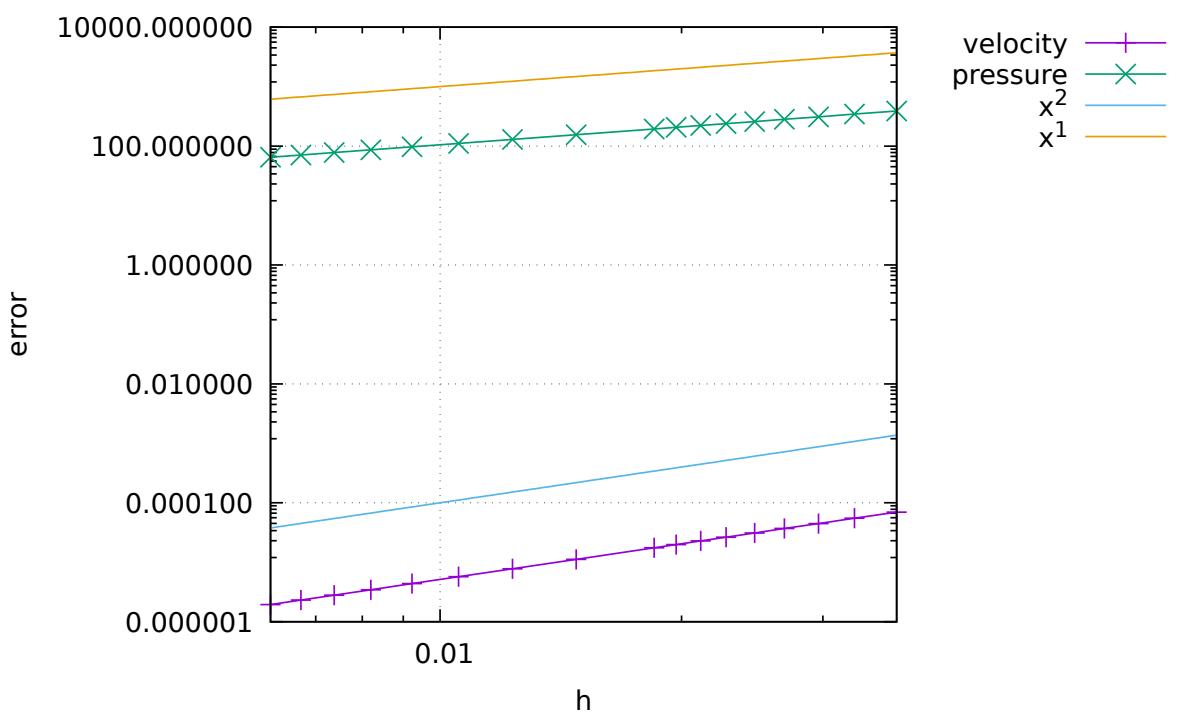
We choose $R_1 = 2890\text{km}$, $R_2 = 6371\text{km}$, $g_0 = 9.81\text{ms}^{-2}$, $\rho_0 = 3300$, $E = 6 \cdot 10^{10}$, $\nu = 0.49$.



radial profiles of the displacement and pressure fields



displacement and pressure fields in the domain



Stone 37: marker advection and population control



The domain is a unit square. The Stokes equations are not solved, the velocity is prescribed everywhere in the domain as follows:

$$u = -(z - 0.5) \quad (784)$$

$$v = 0 \quad (785)$$

$$w = (x - 0.5) \quad (786)$$

At the moment, velocity is computed on the marker itself (rk0 algorithm). When markers are advected outside, they are arbitrarily placed at location (-0.0123,-0.0123).

in construction.

Stone 38: Critical Rayleigh number

This fieldstone was developed in collaboration with Arie van den Berg.

The system is a layer of fluid between $y = 0$ and $y = 1$, with boundary conditions $T(x, y = 0) = 1$ and $T(x, y = 1) = 0$, characterized by ρ , C_p , k , η_0 . The Rayleigh number of the system is

$$\text{Ra} = \frac{\rho_0 g_0 \alpha \Delta T h^3}{\eta_0 \kappa}$$

We have $\Delta T = 1$, $h = 1$ and choose $\kappa = 1$ so that the Rayleigh number simplifies to $\text{Ra} = \rho_0 g_0 \alpha / \eta_0$.

The Stokes equation is $\vec{\nabla} \cdot \boldsymbol{\sigma} + \vec{b} = \vec{0}$ with $\vec{b} = \rho \vec{g}$. Then the components of this equation on the x - and y -axis are:

$$(\vec{\nabla} \cdot \boldsymbol{\sigma})_x = -\rho \vec{g} \cdot \vec{e}_x = 0 \quad (787)$$

$$(\vec{\nabla} \cdot \boldsymbol{\sigma})_y = -\rho \vec{g} \cdot \vec{e}_y = \rho g_0 \quad (788)$$

since \vec{g} and \vec{e}_y are in opposite directions ($\vec{g} = -g_0 \vec{e}_y$, with $g_0 > 0$). The stream function formulation of the incompressible isoviscous Stokes equation is then

$$\nabla^4 \Psi = \frac{g_0}{\eta_0} \frac{\partial \rho}{\partial x}$$

Assuming a linearised density field with regards to temperature $\rho(T) = \rho_0(1 - \alpha T)$ we have

$$\frac{\partial \rho}{\partial x} = -\rho_0 \alpha \frac{\partial T}{\partial x}$$

and then

$$\boxed{\nabla^4 \Psi = -\frac{\rho_0 g_0 \alpha}{\eta_0} g \frac{\partial T}{\partial x} = -Ra \frac{\partial T}{\partial x}} \quad (789)$$

For small perturbations of the conductive state $T_0(y) = 1 - y$ we define the temperature perturbation $T_1(x, y)$ such that

$$T(x, y) = T_0(y) + T_1(x, y)$$

The temperature perturbation T_1 satisfies the homogeneous boundary conditions $T_1(x, y = 0) = 0$ and $T_1(x, y = 1) = 0$. The temperature equation is

$$\rho C_p \frac{DT}{Dt} = \rho C_p \left(\frac{\partial T}{\partial t} + \vec{v} \cdot \vec{\nabla} T \right) = \rho C_p \left(\frac{\partial T_0 + T_1}{\partial t} + \vec{v} \cdot \vec{\nabla}(T_0 + T_1) \right) = k \Delta (T_0 + T_1)$$

and can be simplified as follows:

$$\rho C_p \left(\frac{\partial T_1}{\partial t} + \vec{v} \cdot \vec{\nabla} T_0 \right) = k \Delta T_1$$

since T_0 does not depend on time, $\Delta T_0 = 0$ and we assume the nonlinear term $\vec{v} \cdot \vec{\nabla} T_1$ to be second order (temperature perturbations and coupled velocity changes are assumed to be small). Using the relationship between velocity and stream function $v_y = -\partial_x \Psi$ we have $\vec{v} \cdot \vec{\nabla} T_0 = -v_y = \partial_x \Psi$ and since $\kappa = k/\rho C_p = 1$ we get

$$\boxed{\frac{\partial T_1}{\partial t} - \kappa \Delta T_1 = -\frac{\partial \Psi}{\partial x}} \quad (790)$$

Looking at these equations, we immediately think about a separation of variables approach to solve these equations. Both equations showcase the Laplace operator Δ , and the eigenfunctions of the biharmonic operator and the Laplace operator are the same. We then pose that Ψ and T_1 can be written:

$$\Psi(x, y, t) = A_\Psi \exp(pt) \exp(\pm i k_x x) \exp(\pm i k_y y) = A_\Psi E_\psi(x, y, t) \quad (791)$$

$$T_1(x, y, t) = A_T \exp(pt) \exp(\pm i k_x x) \exp(\pm i k_y y) = A_T E_T(x, y, t) \quad (792)$$

where k_x and k_y are the horizontal and vertical wave number respectively. Note that we then have

$$\nabla^2 \Psi = -(k_x^2 + k_y^2) \Psi \quad \nabla^2 T_1 = -(k_x^2 + k_y^2) T_1$$

The boundary conditions on T_1 , coupled with a choice of a real function for the x dependence yields:

$$E_T(x, y, t) = \exp(pt) \cos(k_x x) \sin(n\pi y).$$

from here onwards check for minus signs!

The velocity boundary conditions are $v_y(x, y=0) = 0$ and $v_y(x, y=1) = 0$ which imposes conditions on $\partial\Psi/\partial x$ and we find that we can use the same y dependence as for T_1 . Choosing again for a real function for the x dependence yields:

$$E_\Psi(x, y, t) = \exp(pt) \sin(k_x x) \sin(n\pi z)$$

We then have

$$\Psi(x, y, t) = A_\Psi \exp(pt) \sin(k_x x) \sin(n\pi z) = A_\Psi E_\Psi(x, y, t) \quad (793)$$

$$T_1(x, y, t) = A_T \exp(pt) \cos(k_x x) \sin(n\pi z) = A_T E_T(x, y, t) \quad (794)$$

In what follows we simplify notations: $k = k_x$. Then the two PDEs become:

$$pT_1 + \kappa(k^2 + n^2\pi^2) - kA_\Psi \exp(pt) \cos(k_x x) \sin(n\pi z) = kA_\Psi E_\theta \quad (795)$$

$$-RaA_T \cos(kx) \sin(n\pi z) + \kappa(k^2 + n^2\pi^2)^2 A_\Psi = -RaA_T E_\Psi + \kappa(k^2 + n^2\pi^2)^2 A_\Psi = 0 \quad (796)$$

These equations must then be verified for all ... which leads to write:

$$\begin{pmatrix} p + (k^2 + n^2\pi^2) & -k \\ -Ra k & (k^2 + n^2\pi^2)^2 \end{pmatrix} \begin{pmatrix} A_\theta \\ A_\Psi \end{pmatrix} = \begin{pmatrix} 0 \\ 0 \end{pmatrix}$$

The determinant of such system must be nul otherwise there is only a trivial solution to the problem, i.e. $A_\theta = 0$ and $A_\Psi = 0$ which is not helpful. CHECK/REPHRASE

$$D = [p + (k^2 + n^2\pi^2)](k^2 + n^2\pi^2)^2 - Ra k^2 = 0$$

or,

$$p = \frac{Ra k^2 - (k^2 + n^2\pi^2)^3}{(k^2 + n^2\pi^2)^2}$$

The coefficient p determines the stability of the system: if it is negative, the system is stable and both Ψ and T_1 will decay to zero (return to conductive state). If $p = 0$, then the system is meta-stable, and if $p > 0$ then the system is unstable and the perturbations will grow. The threshold is then $p = 0$ and the solution of the above system is

Stone 39: chpe15

The Drucker-Prager yield function is given by the function f :

$$f = p \sin \phi + c \cos \phi - \tau$$

where τ is the square root of the second invariant of the deviatoric stress. We have

$$p = \frac{1}{2}(\sigma_1 + \sigma_3)$$

and

$$\tau = \frac{1}{2}(\sigma_1 - \sigma_3)$$

Inserting these into f yields:

$$f = \frac{1}{2}(\sigma_1 + \sigma_3) \sin \phi + c \cos \phi - \frac{1}{2}(\sigma_1 - \sigma_3)$$

The yield condition $f = 0$ can be reworked as follows:

$$\sigma_1 - \frac{1 + \sin \phi}{1 - \sin \phi} \sigma_3 - 2 \frac{\cos \phi}{1 - \sin \phi} c = 0$$

The third term can further be modified as follows:

$$\frac{\cos \phi}{1 - \sin \phi} = \frac{\sqrt{1 - \sin^2 \phi}}{\sqrt{(1 - \sin \phi)^2}} = \frac{\sqrt{(1 - \sin \phi)(1 + \sin \phi)}}{\sqrt{(1 - \sin \phi)^2}} = \sqrt{\frac{1 + \sin \phi}{1 - \sin \phi}}$$

Finally, we define N_ϕ as follows

$$N_\phi = \frac{1 + \sin \phi}{1 - \sin \phi}$$

so that the yield condition becomes:

$$\sigma_1 - N_\phi \sigma_3 - 2\sqrt{N_\phi} c = 0$$

which is Eq. 3 of the article by Choi & Petersen [365].

This paper offers a solution to the problem of the angle of shear bands in geodynamic models. The underlying idea is based on simple modifications brought to existing incompressible flow codes. Note that the codes featured in that paper also implemented elastic behaviour but this can be easily switched off by setting $Z = 1$ in their equations.

Their plasticity implementation starts with a modification of the continuity equation:

$$\vec{\nabla} \cdot \vec{\nabla} = R = 2 \sin \psi \dot{\varepsilon}_p$$

where R is the dilation rate, Ψ is the dilation angle and $\dot{\varepsilon}_p$ is the square root of the second invariant of the plastic strain rate.

Under this assumption, the deviatoric strain rate tensor is given by

$$\dot{\varepsilon}^d(\vec{\nabla}) = \dot{\varepsilon}(\vec{\nabla}) - \frac{1}{3} \text{Tr}[\dot{\varepsilon}(\vec{\nabla})] \mathbf{1} = \dot{\varepsilon}(\vec{\nabla}) - \frac{1}{3} \vec{\nabla} \cdot \vec{\nabla} \mathbf{1} = \dot{\varepsilon}(\vec{\nabla}) - \frac{1}{3} R \mathbf{1}$$

Turning now to the momentum conservation equation:

$$\begin{aligned} -\vec{\nabla} p + \vec{\nabla} \cdot \boldsymbol{\tau} &= -\vec{\nabla} p + \vec{\nabla} \cdot (2\eta \dot{\varepsilon}^d(\vec{\nabla})) \\ &= -\vec{\nabla} p + \vec{\nabla} \cdot \left[2\eta \left(\dot{\varepsilon}(\vec{\nabla}) - \frac{1}{3} R \mathbf{1} \right) \right] \\ &= -\vec{\nabla} p + \vec{\nabla} \cdot (2\eta \dot{\varepsilon}(\vec{\nabla})) - \frac{2}{3} \vec{\nabla} (\eta R) \end{aligned} \tag{797}$$

The last term is then an addition to the right hand side of the momentum equation and its weak form is as follows:

$$\vec{f}' = \int_{\Omega} N_v \frac{2}{3} \vec{\nabla}(\eta R) dV = \frac{4}{3} \sin \Psi \int_{\Omega} N_v \vec{\nabla}(\eta \dot{\epsilon}_p) dV$$

This formulation proves to be problematic since in order to compute the gradient, we would need the viscosity and the plastic strain rate on the mesh nodes and both these quantities are effectively computed on the quadrature points. One option could be to project those quadrature values onto the nodes, which may introduce interpolation errors/artefacts and/or smoothing. Another option is to resort to integration by parts:

$$\int_{\Omega} N_v \vec{\nabla}(\eta \dot{\epsilon}_p) dV = [N_v \eta \dot{\epsilon}_p]_{\Gamma} - \int_{\Omega} \vec{\nabla} N_v (\eta \dot{\epsilon}_p) dV$$

The last term is now trivial to compute since the shape function derivatives, the viscosity and the plastic strain rate are known at the quadrature points. Remains the surface term. We will neglect it for now to simplify our implementation and note that a) it will not directly affect what happens inside the domain, b) it could be somewhat important when shear bands intersect with the free surface.

$$\vec{f}' = -\frac{4}{3} \sin \psi \int_{\Omega} \vec{\nabla} N_v (\eta \dot{\epsilon}_p) dV = -\frac{2}{3} \int_{\Omega} \vec{\nabla} N_v (\eta R) dV$$

Although the authors do indicate that they add a term in each rhs, it is not very clear how they deal with the implementation issue above. We then propose an alternative: instead of explicitly removing the deviatoric part of the strain rate as in Eq. 797 and replace the trace of the tensor by R , one could leave the term inside the matrix, thereby using a compressible form of the viscous block of the Stokes matrix. We will recover the same converged solution as before, but the path to convergence will be different than the first approach. In what follows, we denote the original approach by Choi & Petersen 'method 1' and the latter 'method 2'.

Finally, we need to define what the plastic strain rate tensor is. When using a rigid plastic rheology, the only deformation mechanism *is* plasticity so that the plastic strain rate *is* the strain rate. When using a visco-plastic rheology, the plastic strain rate is the strain rate of the zones above/at yield (the shear bands, where the vrm is active).

The 2010 brick

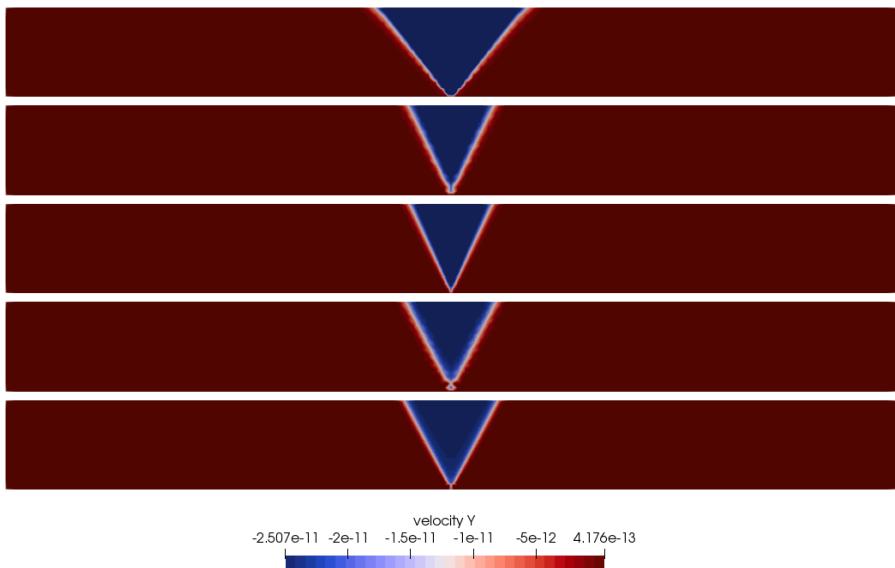
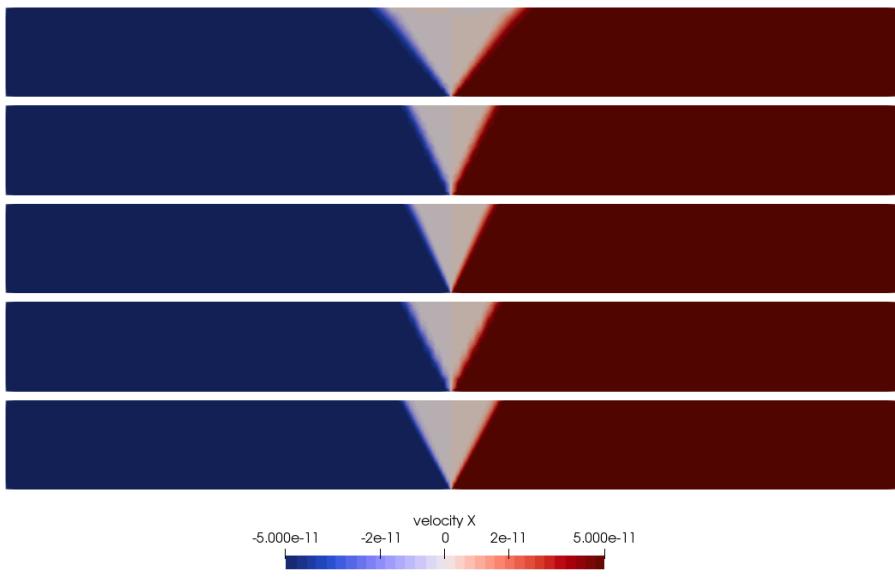
The setup is similar to the one in [1030]. It is a 2D Cartesian domain filled with a single rigid-plastic material characterised by a cohesion $c = 10\text{MPa}$, an angle of friction ϕ , a dilation angle ψ and a density $\rho = 2800\text{kg/m}^3$. Extensional boundary conditions are as follows:

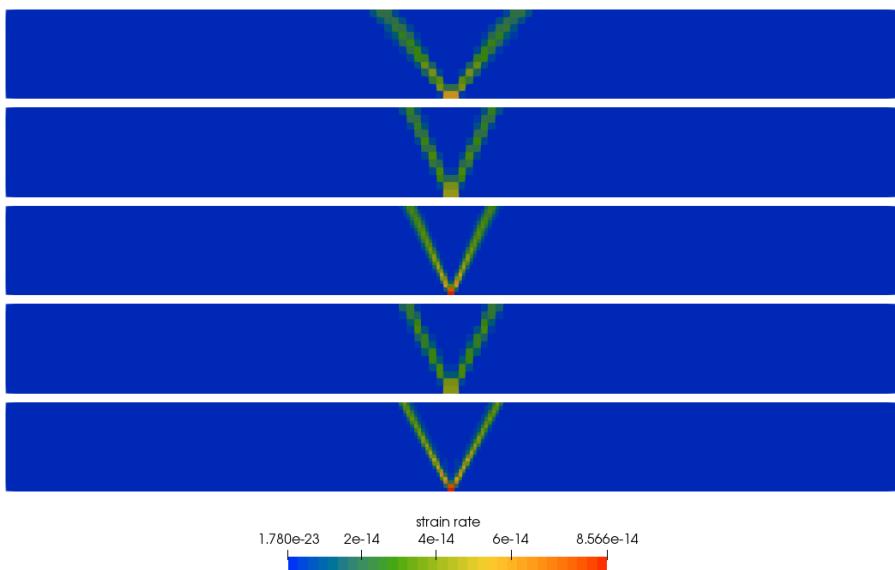
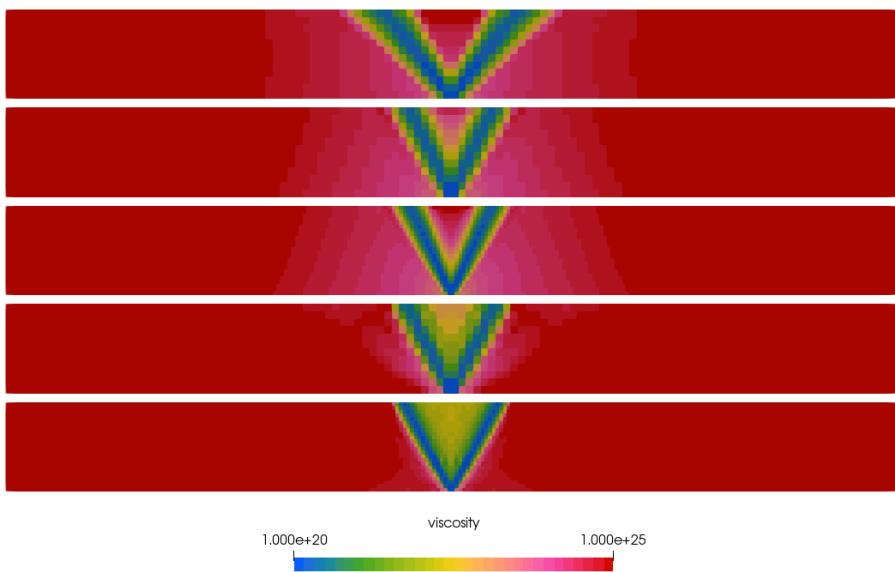
- left boundary: $u = -v_{bc}$;
- right boundary: $u = +v_{bc}$;
- bottom boundary: $v = 0$, $u = -v_{bc}$ for $x < L_x/2$, $u = +v_{bc}$ for $x > L_x/2$, and $u = 0$ if $x = L_x/2$;
- top boundary: zero traction.

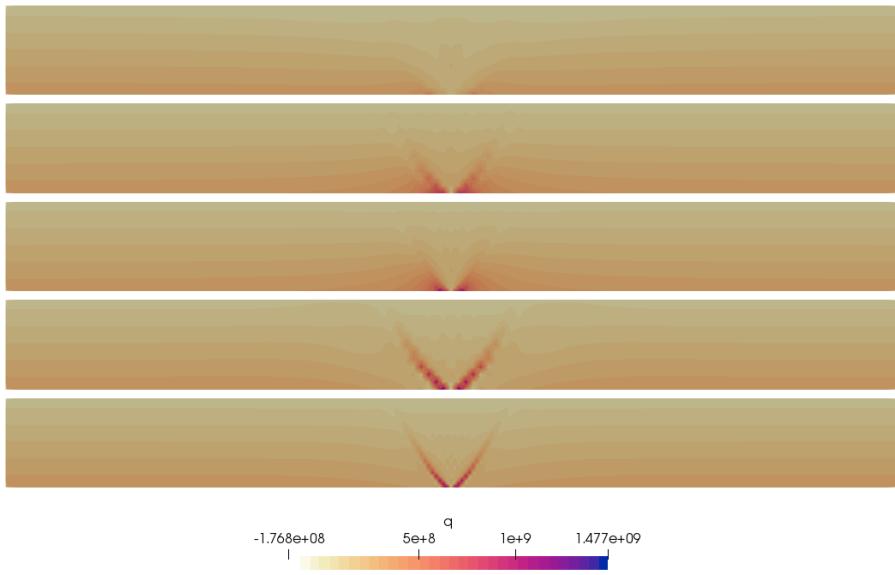
For compressional boundary conditions the signs of all horizontal velocities should be reversed. The nonlinear tolerance is set to $\text{tol} = 10^{-6}$. Nonlinear iterations stop when maximum of the normalised nonlinear residual reaches the desired tolerance.

Following Choi & Petersen [365], we run the experiment with an associative ($\phi = \psi$) plasticity and a non associative one ($\psi = 0$, i.e. $R = 0$). This second approach is essentially what many codes do.

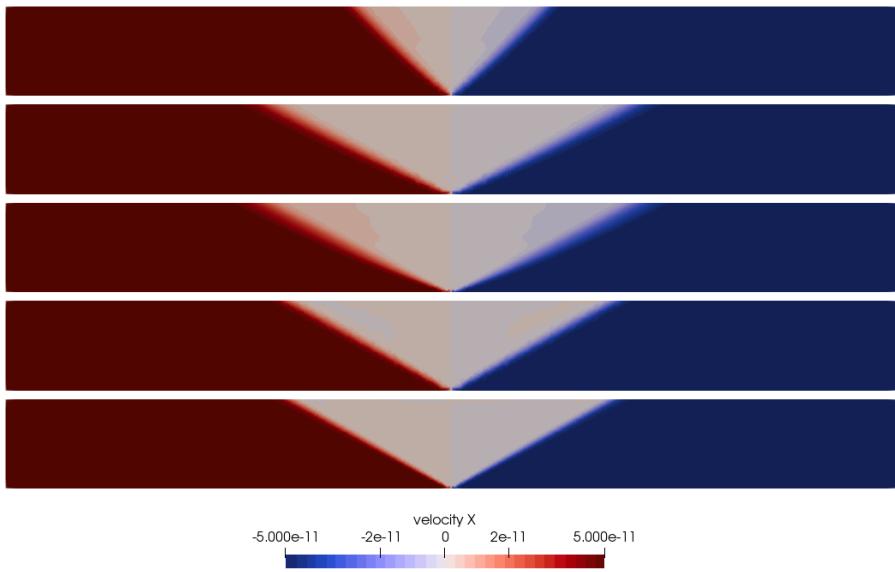
The velocity, pressure, strain rate, dilation rate, and velocity divergence are shown hereunder both in extension and compression.

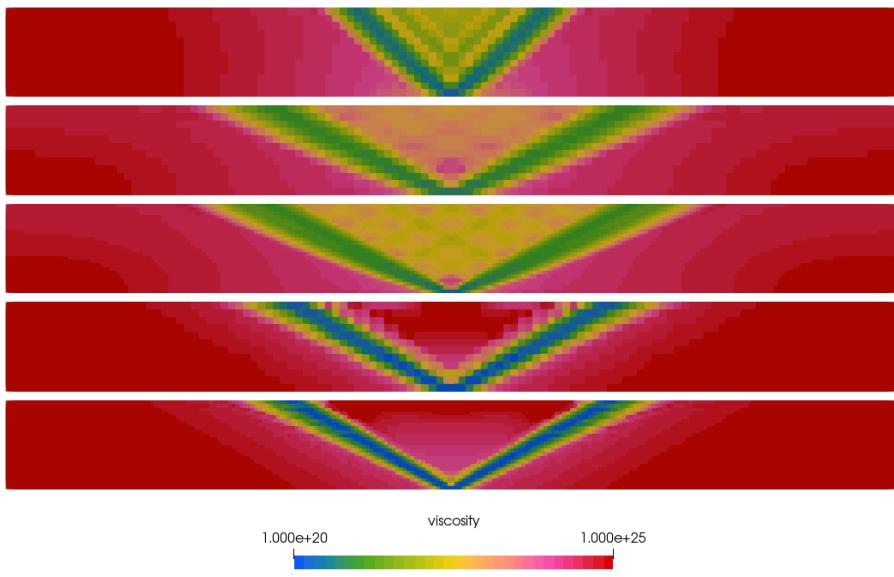
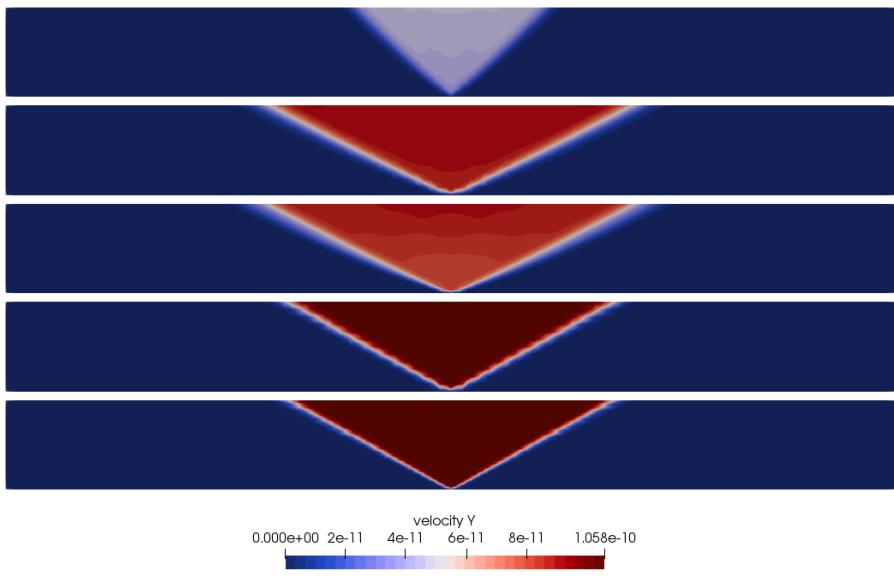


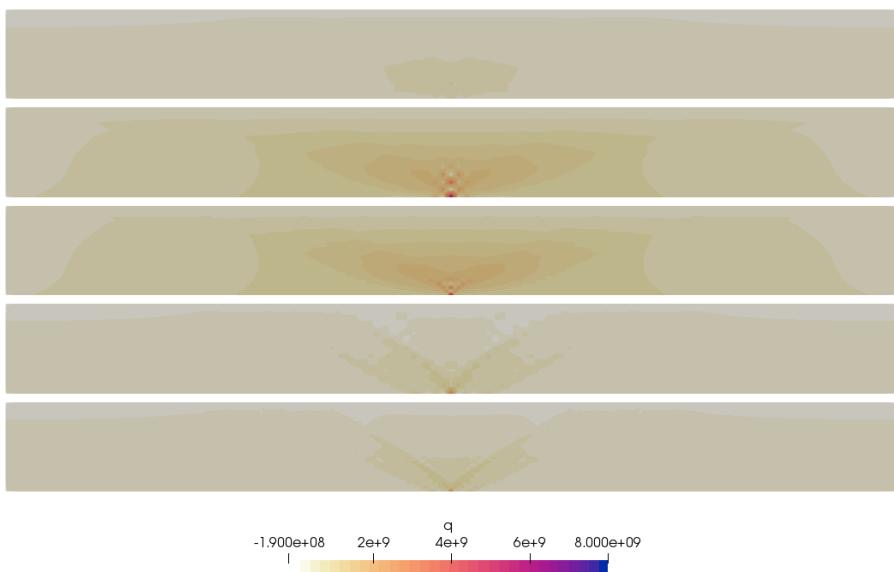
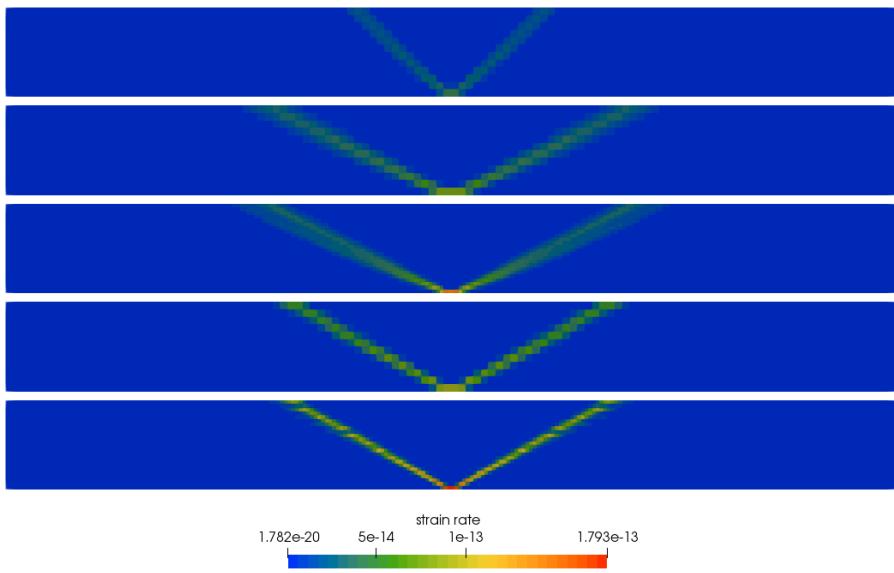




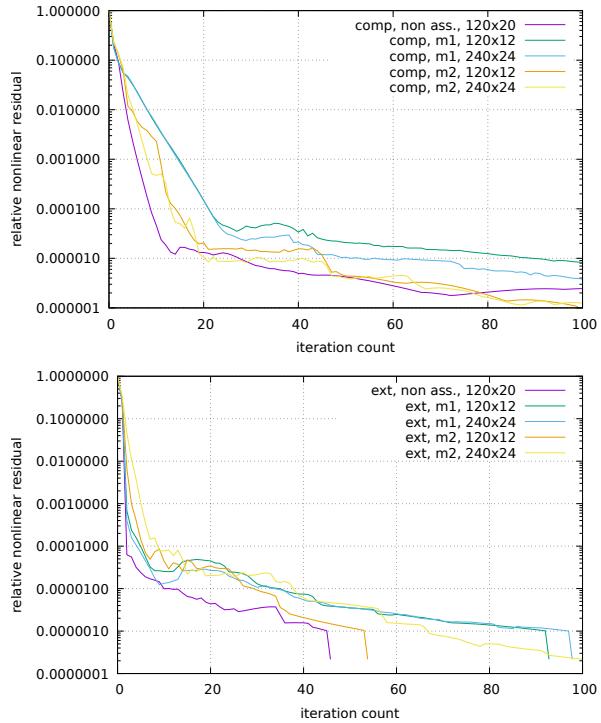
Extension. 1st row: Non-associative plasticity; 2nd and 3rd row: associative plasticity ($\psi = \phi$) with method 1 for two resolutions 120x12 and 240; 4th and 5th row: associative plasticity ($\psi = \phi$) with method 2 for two resolutions 120x12 and 240



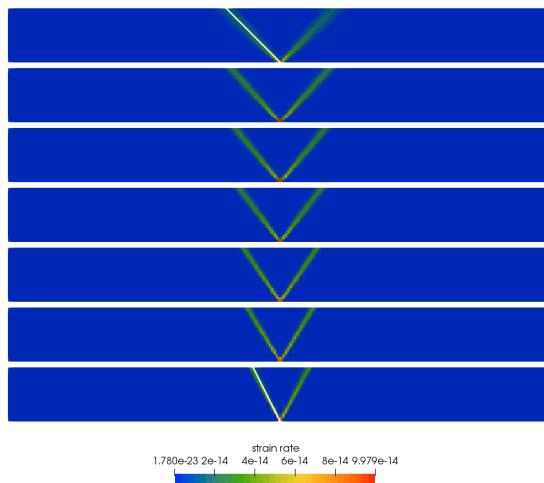


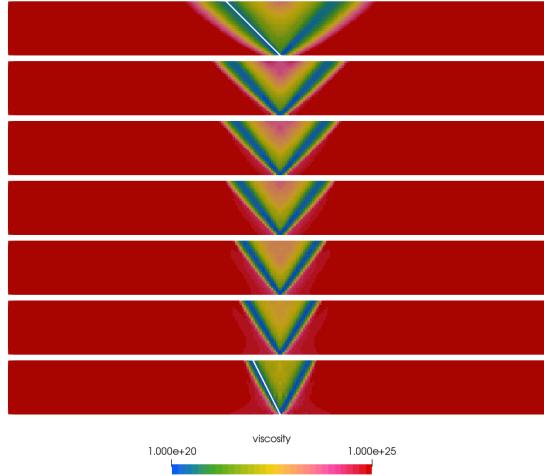


Compression. 1st row: Non-associative plasticity; 2nd and 3rd row: associative plasticity ($\psi = \phi$) with method 1 for two resolutions 120x12 and 240; 4th and 5th row: associative plasticity ($\psi = \phi$) with method 2 for two resolutions 120x12 and 240



One can also run the extension model for $\phi = \psi = 0, 5, 10, 15, 20, 25, 30^\circ$





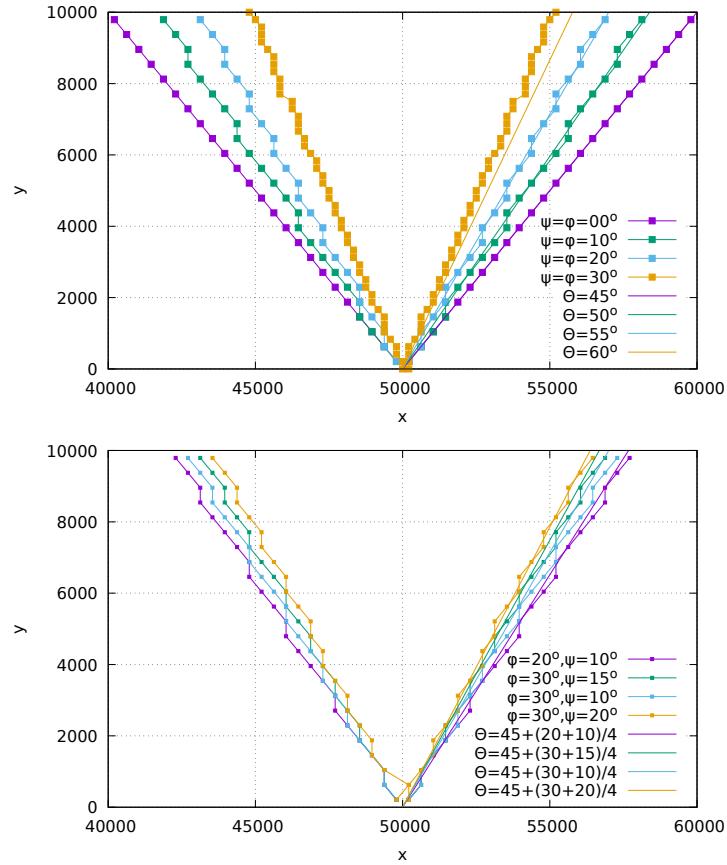
Three angles are mechanically stable (e.g. [1030]):

$$\theta = \frac{\pi}{4} \pm \frac{\psi}{2} \quad \text{Roscoe angle}$$

$$\theta = \frac{\pi}{4} \pm \frac{\phi}{2} \quad \text{Coulomb angle}$$

$$\theta = \frac{\pi}{4} \pm \frac{\phi + \psi}{4} \quad \text{Arthur angle}$$

In the case of associative plasticity, $\phi = \psi$, so that all three angles are the same. Per row of elements, and per half of the domain (left and right) we find the element with the highest strain-rate and record their center coordinates on the figure hereunder. These elements are shown for $\phi = \psi = \{0, 10, 20, 30\}^\circ$ alongside a line corresponding to the expected analytical shear band angle value.



Results obtained on a 240x24 grid, max 50 nl iterations.

Note that benchmarking this is not easy. One solution Timo and I found was to add a velocity field $\vec{v} = (x, y, z)$ (with $\nabla \cdot \vec{v} = 3$) to an existing analytical problem, e.g. the Burstedde benchamrk.

The 2016 brick

The setup is similar to the one in [1685]. It is a 2D Cartesian domain filled with an isoviscous layer at the bottom and a visco-plastic material on top, as shown here:

In what follows the nonlinear tolerance is set to 10^{-6} . Due to a lack of resolution, I do not implement the rounded edges of the seed. U_0 is set to 25mm/yr and the background viscosity of the brittle layer is set to $\eta_0 = 10^{24}$ Pa.s. Note that the effective viscosity of the brittle layer is computed as follows:

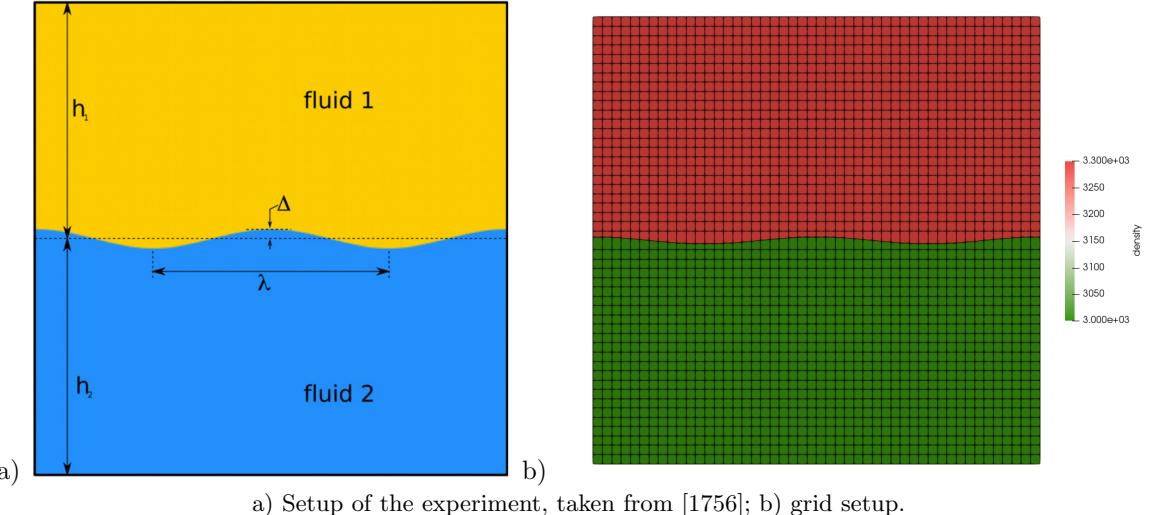
$$\eta_{eff} = \left(\frac{1}{\eta_0} + \frac{1}{\eta_p} \right)^{-1} = \left(\frac{1}{\eta_0} + \frac{2\dot{\epsilon}_{ii}}{Y} \right)^{-1}$$

with $Y = p \sin \phi + c \cos \phi$. Note that the pressure colour bars in Fig(6) of [1685] are most likely not correct at all, and in order for my results to look like theirs I had to change it for VM, DDM and DP.

Stone 40: Rayleigh-Taylor instability

This benchmark is carried out in [524, 722, 1756] and is based on the analytical solution by Ramberg (1968). It consists of a two-layer system driven by gravity. Free slip are imposed on the sides while no-slip boundary conditions are imposed on the top and the bottom of the box.

Fluid 1 (ρ_1, η_1) of thickness h_1 overlays fluid 2 (ρ_2, η_2) of thickness h_2 (with $h_1 + h_2 = L_y$). An initial sinusoidal disturbance of the interface between these layers is introduced and is characterised by an amplitude Δ and a wavelength $\lambda = L_x/2$ as shown in Figure ??.



a) Setup of the experiment, taken from [1756]; b) grid setup.

Under this condition, the velocity of the diapiric growth v_y is given by the relation

$$\frac{v_y}{\Delta} = -K \frac{\rho_1 - \rho_2}{2\eta_2} h_2 g$$

with the dimensionless growth factor K being

$$K = \frac{-d_{12}}{c_{11}j_{22} - d_{12}i_{21}}$$

and

$$c_{11} = \frac{\eta_1 2\phi_1^2}{\eta_2(\cosh 2\phi_1 - 1 - 2\phi_1^2)} - \frac{2\phi_2^2}{\cosh 2\phi_2 - 1 - 2\phi_2^2} \quad (798)$$

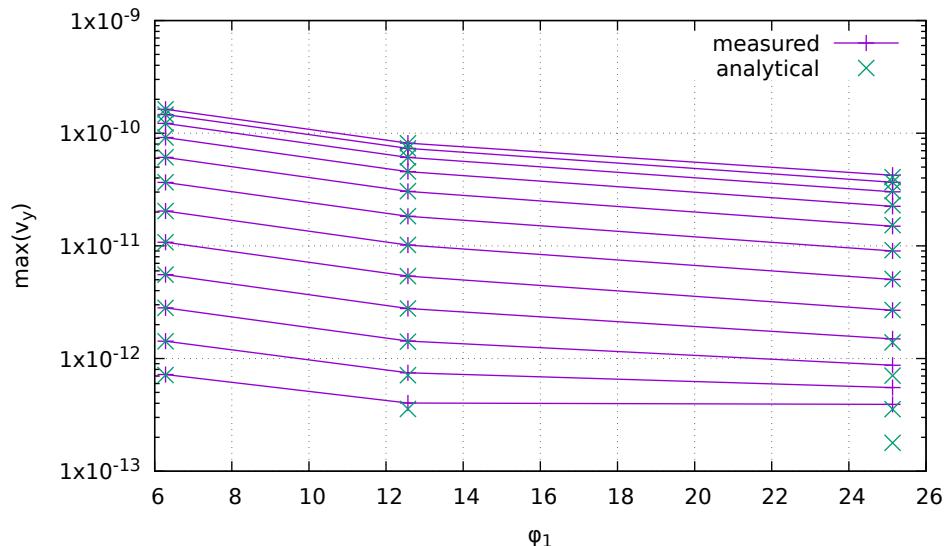
$$d_{12} = \frac{\eta_1(\sinh 2\phi_1 - 2\phi_1)}{\eta_2(\cosh 2\phi_1 - 1 - 2\phi_1^2)} + \frac{\sinh 2\phi_2 - 2\phi_2}{\cosh 2\phi_2 - 1 - 2\phi_2^2} \quad (799)$$

$$i_{21} = \frac{\eta_1 \phi_2 (\sinh 2\phi_1 + 2\phi_1)}{\eta_2(\cosh 2\phi_1 - 1 - 2\phi_1^2)} + \frac{\phi_2 (\sinh 2\phi_2 + 2\phi_2)}{\cosh 2\phi_2 - 1 - 2\phi_2^2} \quad (800)$$

$$j_{22} = \frac{\eta_1 2\phi_1^2 \phi_2}{\eta_2(\cosh 2\phi_1 - 1 - 2\phi_1^2)} - \frac{2\phi_2^3}{\cosh 2\phi_2 - 1 - 2\phi_2^2} \quad (801)$$

$$\phi_1 = \frac{2\pi h_1}{\lambda} \quad (802)$$

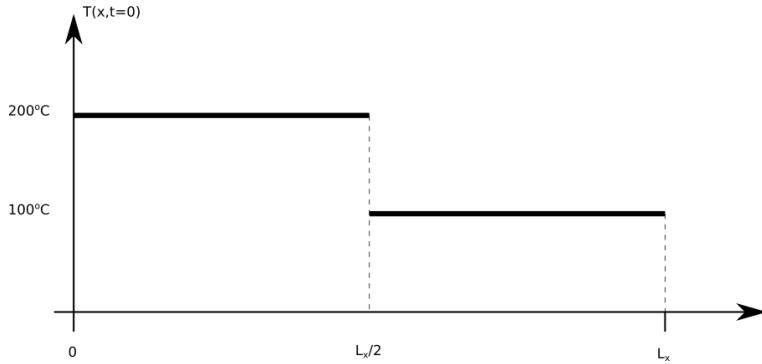
$$\phi_2 = \frac{2\pi h_2}{\lambda} \quad (803)$$



Note that in [1756] I fixed $\lambda = L_x/2$ and varied L_x . Here I keep L_x fixed and vary $\lambda = L_x/2, L_x, 4, L_x/8$. Each line corresponds to a different value of the viscosity η_2 .

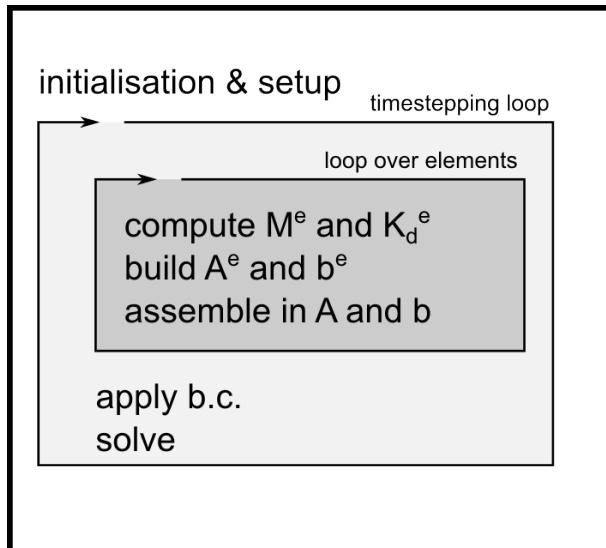
Stone 42: 1D diffusion

This is the simplest case for a FE code: a 1D (temperature) diffusion problem. It puts into practice what is presented in section 5.1. The initial temperature profile is as follows:



$$T(x, t = 0) = 200 \quad x < L_x/2 \quad T(x, t = 0) = 100 \quad x \geq L_x/2$$

The properties of the material are as follows: $\rho = 3000$, $k = 3$, $C_p = 1000$ and the domain size is $L_x = 100\text{km}$. Boundary conditions are $T(t, x = 0) = 200^\circ\text{C}$ and $T(t, x = L_x) = 100^\circ\text{C}$. There are `nelt` elements and `nnx` nodes. All elements are `hx` long. The code will carry out `nstep` timesteps of length `dt` or will stop before that when steady state is reached. The code structure is summarised hereunder:



Stone 43: the rotating cone

This benchmark originates in [533]. It is also carried out in [151]. It considers the advection of a product-cosine hill in a prescribed velocity field. The initial temperature is:

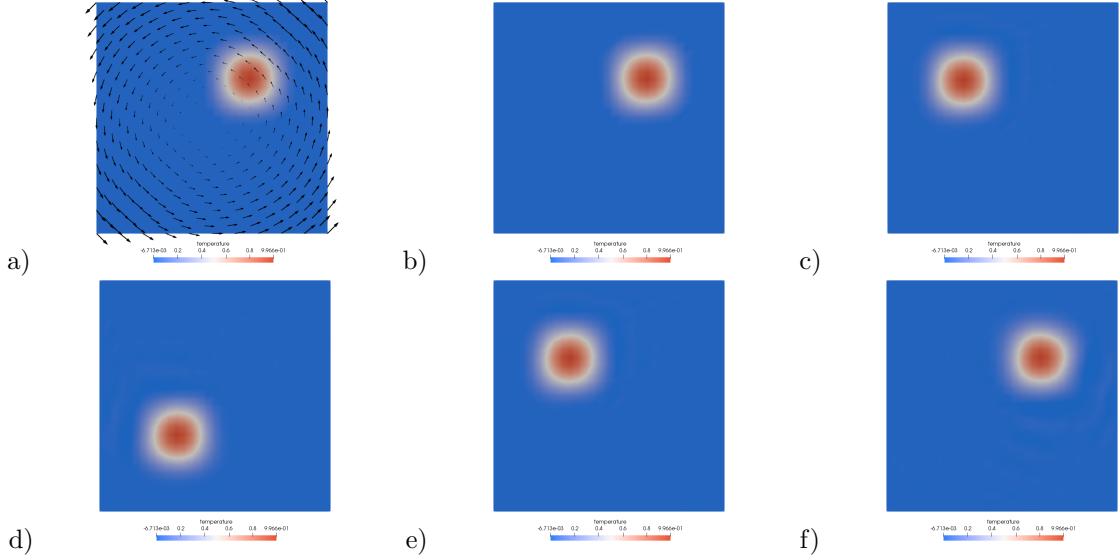
$$T_0(x, y) = \begin{cases} \frac{1}{4} \left(1 + \cos \pi \frac{x-x_c}{\sigma}\right) \left(1 + \cos \pi \frac{y-y_c}{\sigma}\right) & \text{if } (x - x_c)^2 + (y - y_c)^2 \leq \sigma^2 \\ 0 & \text{otherwise} \end{cases} \quad (804)$$

The boundary conditions are $T(x, y) = 0$ on all four sides of the unit square domain. In what follows we set $x_c = y_c = 1/6$ and $\sigma = 0.2$. The velocity field is analytically prescribed: $\vec{v} = (-(y - y_c), +(x - x_c))$.

In what follows we test the time integration scheme by setting $\alpha_T = 1$ (fully implicit formulation), $\alpha = 0$ (fully explicit formulation) and $\alpha_T = 1/2$ (Crank-Nicolson). The timestep is set to $\delta t = 2\pi/200$. The density and heat capacity values are set to 1. We monitor the minimum and maximum value of the temperature field, as well as the total thermal energy E_T in the system during the 200 time steps (2π rotation of the cone):

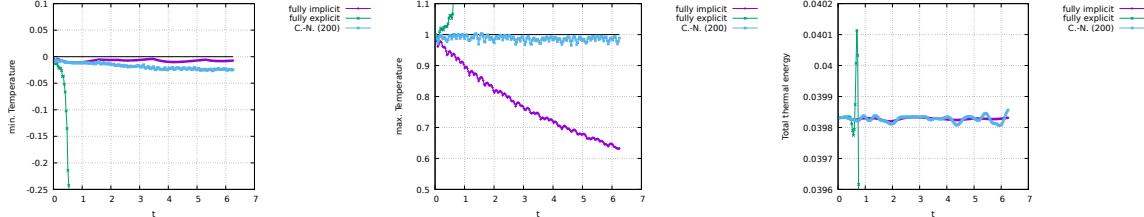
$$E_T = \int_{\Omega} \rho_0 C_p T dV = \int_{\Omega} T dV = |\Omega| \langle T \rangle \quad \text{where} \quad \langle T \rangle = \frac{1}{|\Omega|} \int_{\Omega} T dV$$

The time evolution of the temperature with the Crank-Nicolson algorithm is shown hereunder:



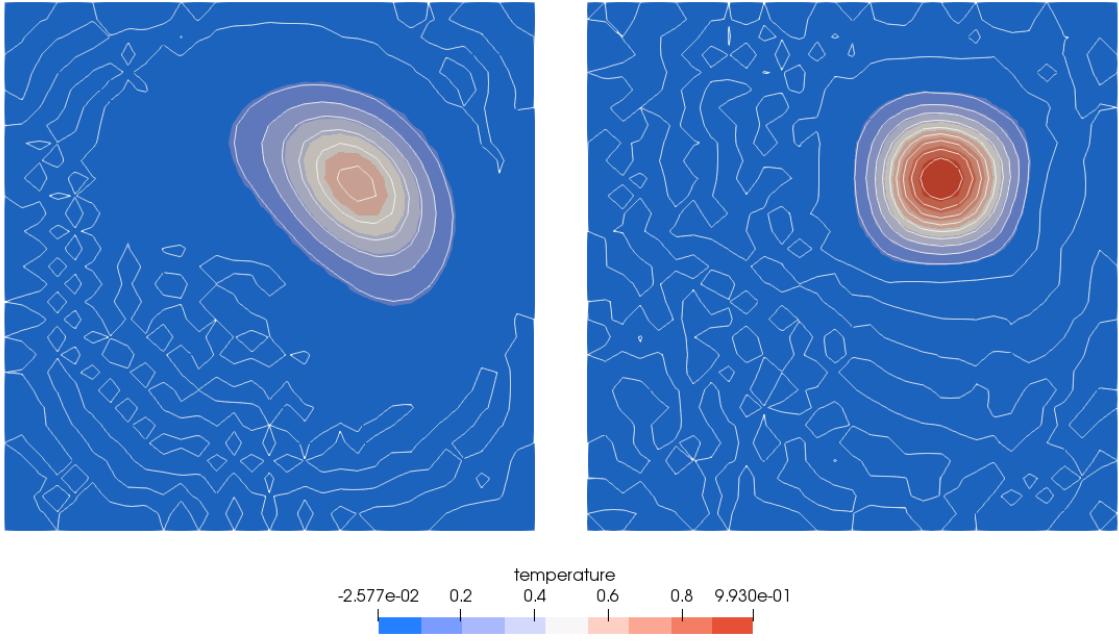
a) Velocity field and initial temperature; b,c,d,e,f) Temperature field at timesteps 0,50,100,150,199.

Turning now to the statistics, we plot $\min(T)$, $\max(T)$ and E_T as a function of time:



Time evolution of the min and max temperature and the total energy

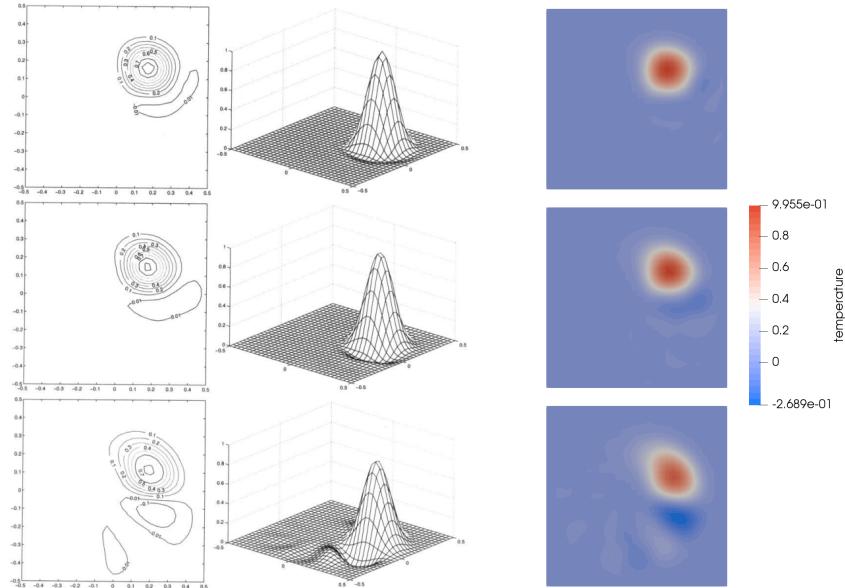
The conclusions are clear: the explicit method diverges quickly and is unusable. The fully implicit and Crank-Nicolson method yield similar energy conservation but the fully-implicit showcases a clear loss in maximum temperature as shown in the following figure:



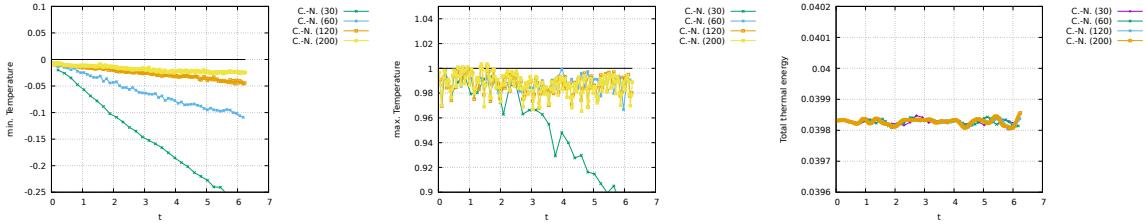
Temperature field after a full rotation with isocontours every 0.1 value.

Left: Fully-implicit; Right: Crank-Nicolson

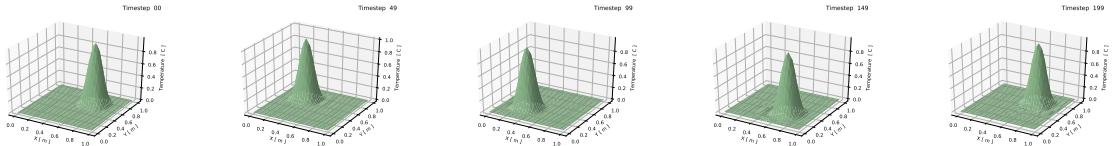
Finally we can run the experiment (still a 2π rotation) with three different time steps ($\delta t = 2\pi/30, 2\pi/60, 2\pi/120$) and we recover very similar results to those presented in [533]:



From top to bottom: $\delta t = 2\pi/120, 2\pi/60, 2\pi/30$ with Crank-Nicolson. Left panel is taken from donea & Huerta [533]



Time evolution of the min and max temperature and the total energy obtained with the Crank-Nicolson algorithm for 4 values of the timestep as indicated by the number between parenthesis.

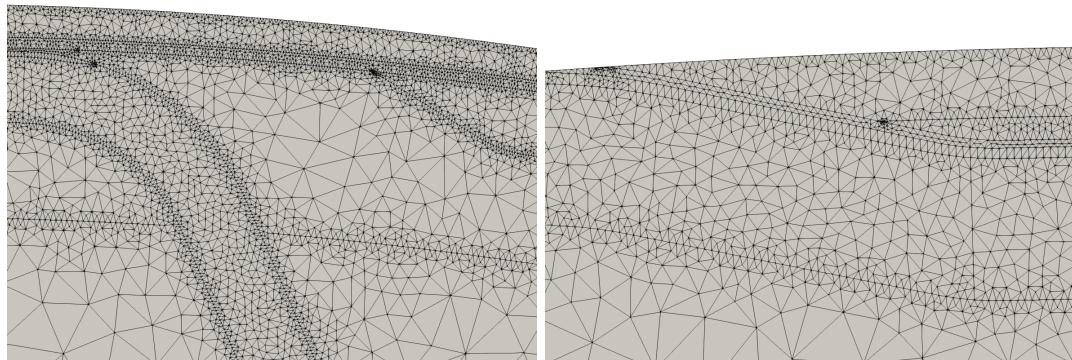
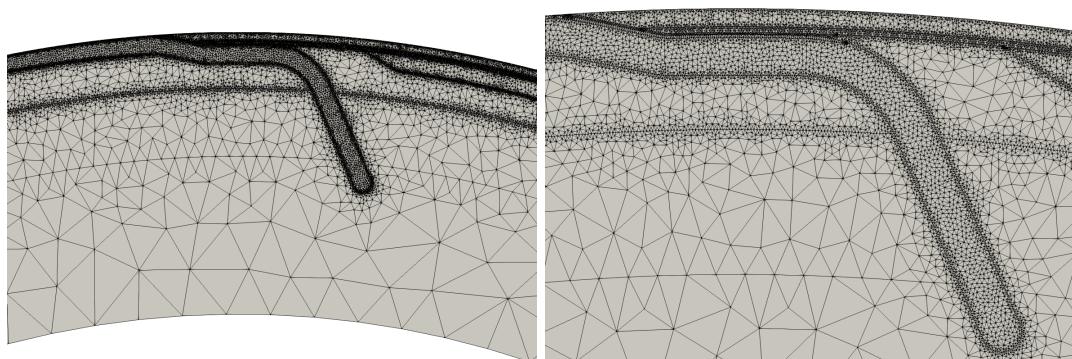


Time evolution of the temperature field for $\delta t = 2\pi/200$ with Crank-Nicolson.

Stone 44: the flat slab

WORK in PROGRESS

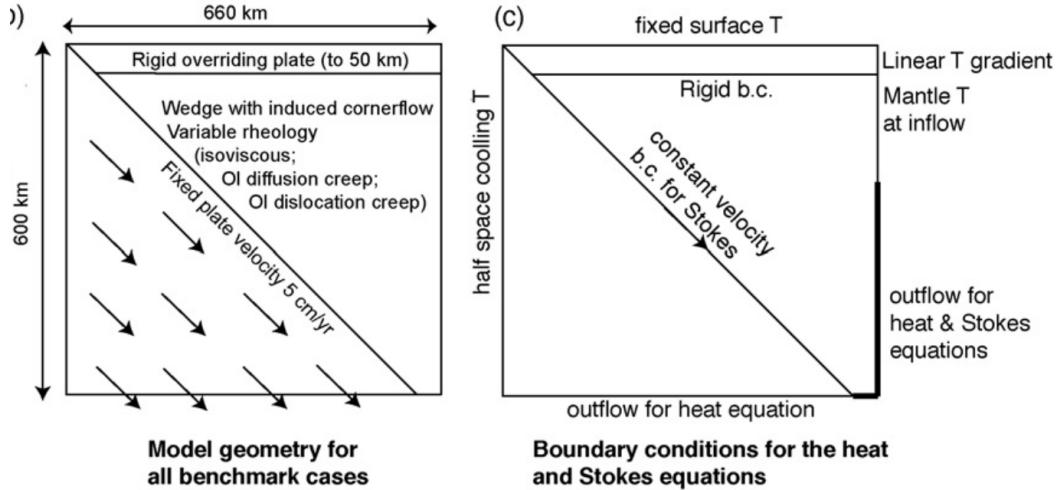
I need a list of nodes on the boundary I need a GCOORD.txt file with more decimals I need an even lower resolution grid I need the scaling factors for rho,eta, ...



Stone 45: the corner flow

This experiment is based on the benchmark paper by van Keken et al, 2008 [1845]. It shares similarities with the time dehydration processes in subduction zones work by Magni et al., 2014 [1245] and the 3D corner flow study of Plunder et al, 2018[1488]. See also Cerpa et al, 2017 [346] for a study of fluid migration in the mantle wedge.

The domain is 660km × 600km. Note that in the original paper the origin of the coordinate system is at the top left while it is at the lower left corner in our code.



As shown in the figure above, the inflow boundaries (at both wedge and trench sides) and top of the model have prescribed temperature. The wedge is assumed to be an incompressible fluid that is driven only by the kinematic forcing of the slab. The wedge is confined by the top of the slab and the base of the rigid overriding plate (located at a depth of 50km). The boundary conditions for the wedge are no-slip below the overriding plate and constant velocity along the top of the slab. The velocity boundary conditions for the boundaries of the wedge are either provided by the Batchelor cornerflow solution (cases 1a and 1b) or based on free inflow/outflow boundaries. The velocity field is discontinuous between the slab and the overriding plate. The velocity in the slab is constant (5cm/yr) and it dips at a 45° angle. There is no radiogenic or shear heating.

The flow is assumed to be incompressible and buoyancy effects are neglected. All the experiments shown in the paper are at steady state, i.e. the temperature field satisfies:

$$\rho C_p \vec{v} \cdot \vec{\nabla} T = \vec{\nabla} \cdot (k \vec{\nabla} T) \quad (805)$$

In the paper a simplified diffusion creep formulation is adopted and the effective diffusion creep viscosity is computed as follows:

$$\eta_{\text{diff}} = A_{\text{diff}} \exp \frac{Q_{\text{diff}}}{RT}$$

The dislocation creep effective viscosity is given by

$$\eta_{\text{disl}} = A_{\text{disl}} \dot{\varepsilon}^{(1-n)/n} \exp \frac{Q_{\text{disl}}}{nRT}$$

Note that in both the activation volume has been set to zero, which decouples pressure from the effective viscosities. Both effective viscosities are limited with a maximum viscosity as follows:

$$\eta_{\text{diff}}^* = \left(\frac{1}{\eta_{\text{diff}}} + \frac{1}{\eta_{\text{max}}} \right)^{-1} \quad \eta_{\text{disl}}^* = \left(\frac{1}{\eta_{\text{disl}}} + \frac{1}{\eta_{\text{max}}} \right)^{-1}$$

The top boundary condition is $T_{\text{top}} = T(y = L_y) = 273K$. At the inflow boundary of the wedge (i.e. where $u < 0$)⁶⁵ temperature is fixed at $T_0 = 1573K$ and a linear geotherm is used at the left hand

⁶⁵Think about it: it makes little sense to prescribe a temperature where the fluid is leaving the domain

boundary of the overriding plate from 0 to 50 km depth. The temperature at the slab inflow boundary is described by an error-function solution for half-space cooling for 50 Myr:

$$T(x = 0, y) = T_{top} + (T_0 - T_{top}) \operatorname{erf} \frac{L_y - y}{2\sqrt{\kappa t_{50}}}$$

where t_{50} is the age of the slab.

At the slab and wedge outflow boundaries we prescribe the natural boundary condition (zero curvature) for the heat equation.

the original paper considers multiple cases:

- Case 1a: analytical cornerflow model. The wedge flow is prescribed by the analytical expression for cornerflow [82], so that we do not need to solve for the Stokes equations, only the energy equation.
- Case 1b: dynamical flow in isoviscous wedge I This case is the same as 1a, except that the solution for the wedge flow is determined by solving the Stokes equations while the Batchelor solution is imposed on the inflow and outflow boundaries. This case tests the ability of the numerical method to accurately reproduce the corner flow solution.
- Case 1c: dynamical flow in isoviscous wedge II. Same as case 1b, but with stress-free boundary conditions on the mantle wedge.
- Case 2a: dynamical flow with diffusion creep
- Case 2b: dynamical flow with dislocation creep

The temperature field as discreted values T_{ij} on an equidistant grid with 6km spacing, which is a 111×101 matrix stored row-wise starting in the top left corner. From this grid the following measurements are extracted for direct comparison:

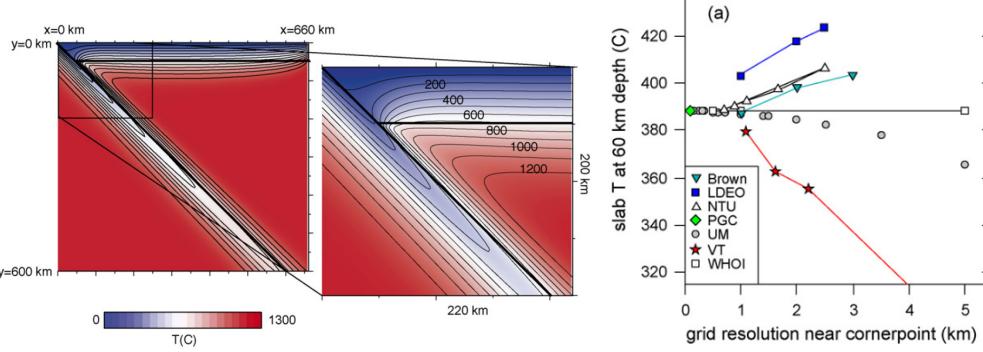
1. the temperature $T_{11,11}$ which is at coordinates (60, 60km) and just down-stream from the corner point. This provides therefore one of the most critical tests of accuracy of the numerical codes;
2. the L2 norm of the slab-wedge interface temperature between 0 and 210 km depth defined by

$$T_{\text{slab}} = \sqrt{\frac{1}{36} \sum_{i=1}^{36} T_{ii}^2}$$

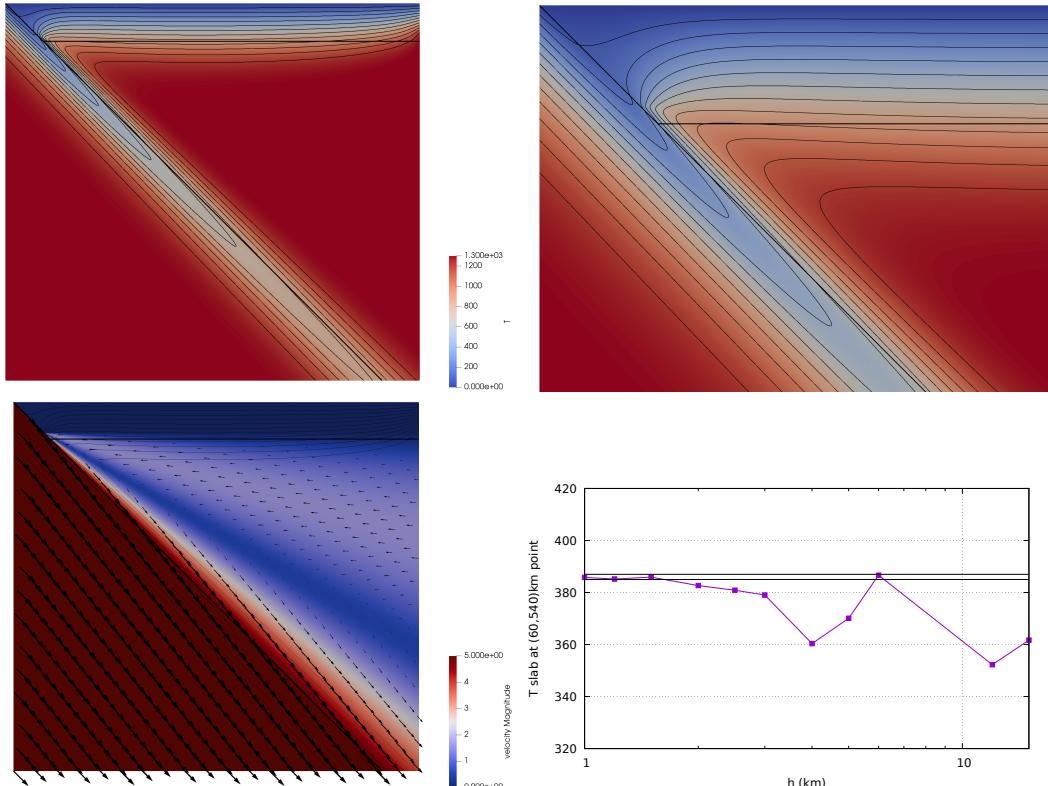
3. the L2 norm of the temperature in the triangular part of the tip of the wedge, between 54 and 120 km depth:

$$T_{\text{wedge}} = \sqrt{\frac{1}{78} \sum_{i=10}^{21} \sum_{j=10}^i T_{ij}^2}$$

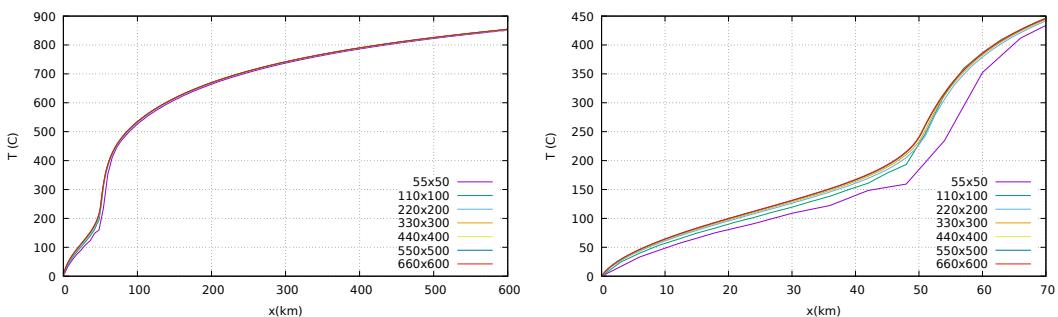
Results for case 1a



(a) Temperature prediction for case 1a. The bold lines indicate the top of the slab and base of the overriding plate. (b) Close up of the top left part of the model. Figures taken from [1845].



Results obtained on mesh 660x600(x2) elements. Black lines on lower right figure correspond to a visible range of values as shown in Fig.3a of [1845].

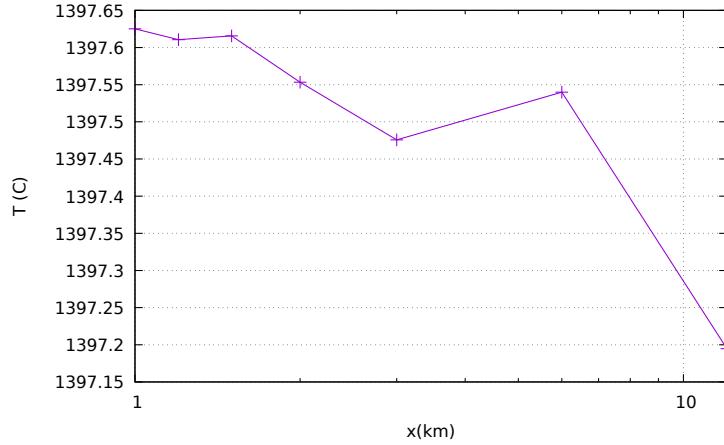


Temperature values on the slab top surface ($y = L_y - x$) plotted as a function of the x coordinate. Left is entire slab, right is zoom on the first 70km

The average temperature

$$\langle T \rangle = \frac{1}{|\Omega|} \int_{\Omega} T \, dV$$

is plotted in the following figure:



Average temperature in the domain.

We see that this measurement is not appropriate to assert whether the resolution is sufficient so that results converge to a single value, as opposed to the point wise temperature measurement presented above. The average temperature changes by about 0.4 for an average value of about 1397, which is not much for a factor 12 increase in resolution (from 55x50 to 660x600).

carry out other two measurements (need interpolation onto other grid for this

Results for case 1b

TODO

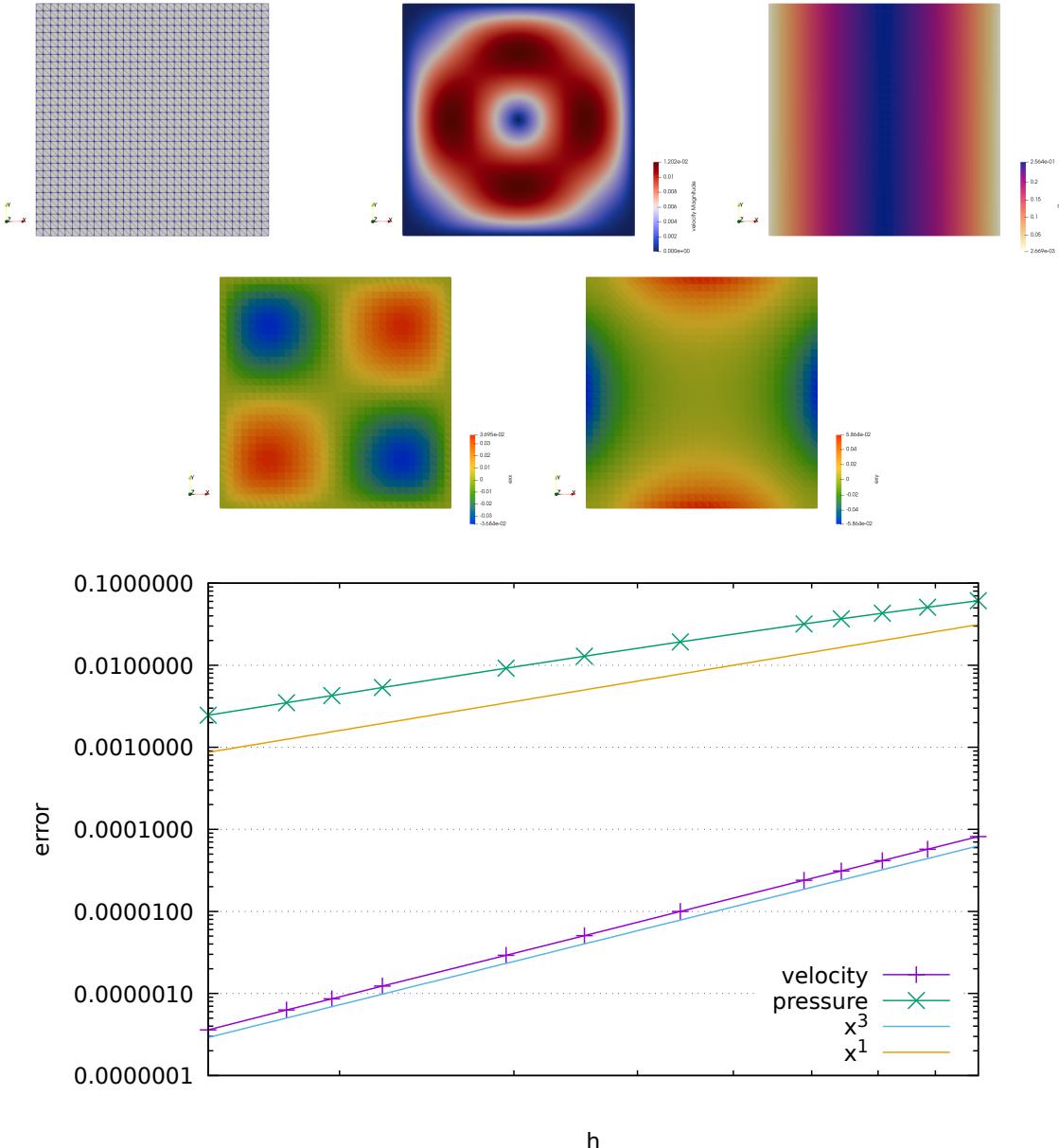
Results for case 1c

TODO

Stone 46: MMS1 with Crouzeix-Raviart ($P_2^+ \times P_{-1}$) elements

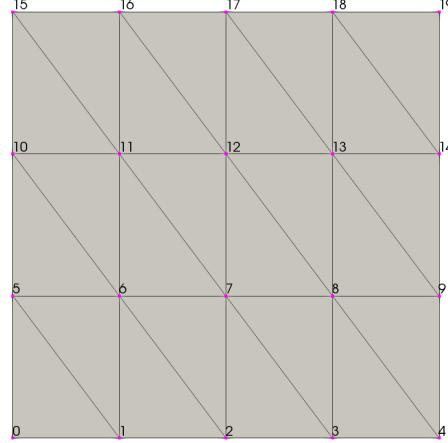
This stone showcases the Crouzeix-Raviart element (see Section 6.2.10) used to solve the analytical problem "Donea & Huerta" (see Section 8.6.1).

Out of convenience the pressure is set to zero at location $(x, y) = (1, 1)$, so that the analytical solution is now $p(x, y) = x(1 - x)$.



Stone 47: MMS1 with MINI ($P_1^+ \times P_1$) elements

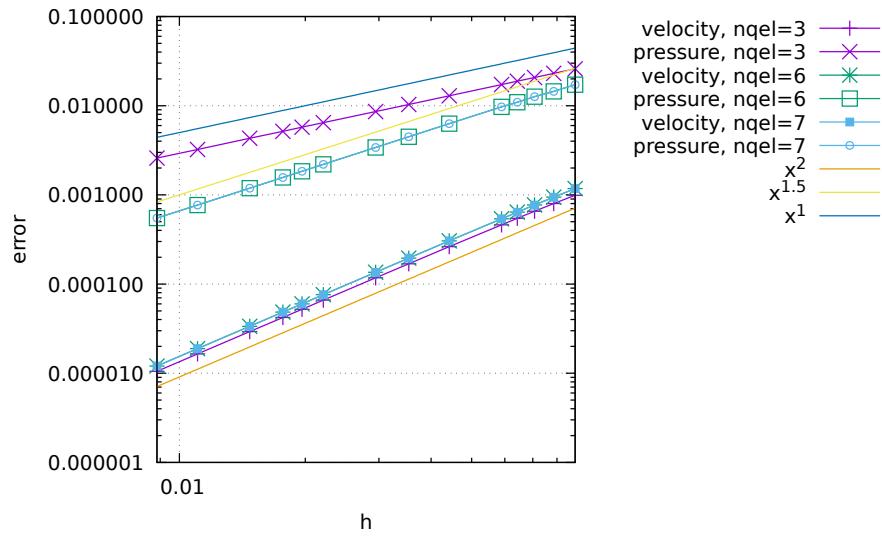
The grid is composed of triangles but for simplicity these are obtained by splitting rectangles in two, as shown hereunder:



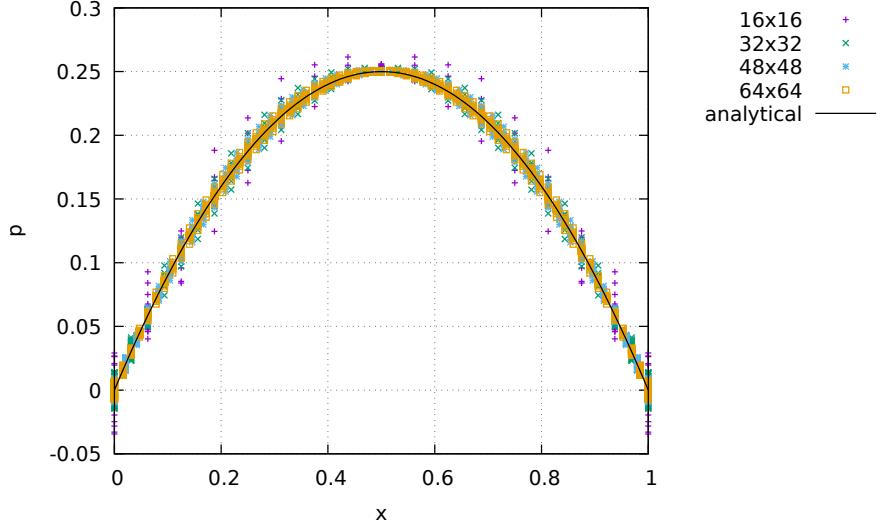
Not shown are the nodes for the bubbles in the middle of each triangle.

This stone showcases the MINI element (see Section 6.2.8) used to solve the analytical problem "Donea & Huerta" (see Section 8.6.1). Out of convenience the pressure is set to zero at location $(x, y) = (1, 1)$, so that the analytical solution is now $p(x, y) = x(1 - x)$.

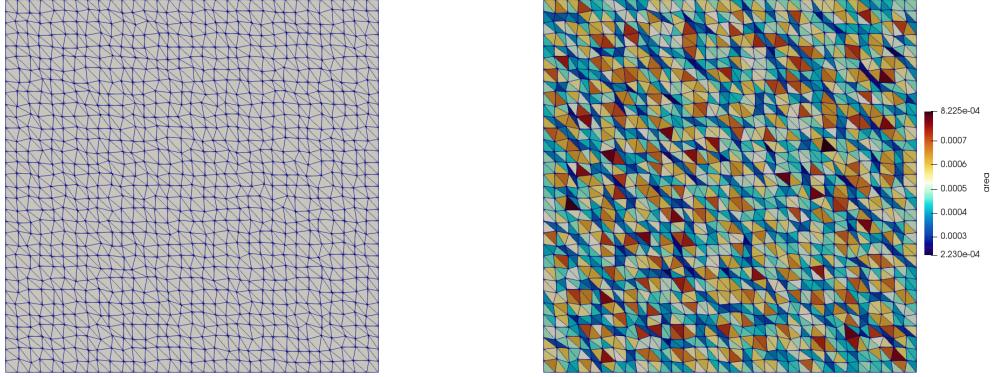
As an experiment I have run convergence tests for two cases: using $nqel=3$ quadrature points and using $nqel=6$ quadrature points. We find that the velocity and pressure errors converge depend on this crucial parameter. For $nqel=3$ the velocity and pressure errors converge quadratically and linearly respectively but for $nqel=6$ they converge as h^2 and $h^{1.5}$ respectively:



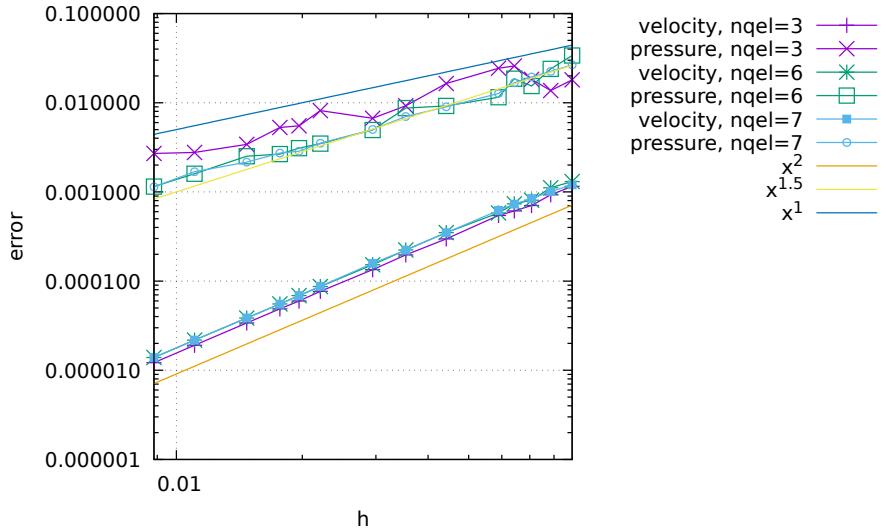
It is worth noticing that although the element is stable, and the error converges at a respectable rate, the pressure solution is not 'clean': as shown on the following figure, there is still some under/overshoot with respect to the analytical solution.



Let us now explore the case where the nodes inside the domain are randomly perturbed, i.e. a random value $(\delta_x, \delta_y) \in [-h_x/5, h_x/5] \times [-h_y/5, h_y/5]$ is added to their position (while preserving the position of the bubble as the barycenter of each triangle), as shown hereunder:



Looking again at the convergence rates of the errors, we see that the velocity errors are virtually unchanged but we observe that the pressure errors no more align on a single line and that the rates are only maintained on average.



Stone 48: D&H with $Q_1 \times P_0$, $Q_2 \times Q_1$, $Q_3 \times Q_2$ and $Q_4 \times Q_3$ elements

In this experiment we consider 4 different finite elements. The idea behind this stone is to build a code which code(the FEM build and assembly) is common to all. The setup is the Donea & HUerta benchmark (Section 8.6.1), which has been modified so that the pressure is zero at the top right corner.

Q4xQ3	Q3xQ2	Q2xQ1	Q1Q0
20====21====22====23====24	12====13====14====15	06=====07=====08	02=====03
20====21====22====23====24			
	08====09====10====11		
20====21====22====23====24			
	03=====04=====05		
	04====05====06====07		
20====21====22====23====24			
20====21====22====23====24	00=====01=====02=====03	00=====01=====02	00=====01
12====13====14====15	06=====07=====08	02=====03	.
08====09====10====11			
	03=====04=====05		00
04====05====06====07			
00=====01=====02=====03	00=====01=====02	00=====01	.
mV=25, mP=16	mV=16, mP=9	mV=9, mP=4	mV=4, mP=1

In the code the 'order' parameter can take values 1,2,3 and 4 which correspond to the polynomial order of the velocity approximation (Q_1 , Q_2 , Q_3 and Q_4).

When both nelx and nely values have been chosen, the total number of element for a regular 2D grid is simply:

```
nel=nelx*nely
```

The number of nodes in each direction is then easily computed:

```
nmx=order*nelx+1
nny=order*nely+1
```

and so is then the total number of velocity nodes:

```
NV=nmx*nny
```

The total number of pressure nodes is as follows:

```

if order==1:
    NP=nelx*nely
if order==2:
    NP=(nelx+1)*(nely+1)
if order==3:
    NP=(2*nelx+1)*(2*nely+1)
if order==4:
    NP=(3*nelx+1)*(3*nely+1)

```

Each velocity node has 2 dofs (ndofV=2) while pressure nodes have one dof (ndofP=1) so that the size of the blocks and the assembled FE matrix are given by:

```

NfemV=NV*ndofV
NfemP=NP*ndofP
Nfem=NfemV+NfemP

```

For the linear element, 2 quadrature points per dimension are enough (nqperdim=2), while 3 are necessary for the quadratic element (nqperdim=3) and 4 are used for the cubic element (nqperdim=4), and 5 for the quartic element, which can be conveniently implemented as follows:

```
nqperdim=order+1
```

The quadrature points location and weight is document in Section 4.1.

Because we wish to use a regular grid, the layout of the points for all three elements can be implemented easily:

```

counter=0
for j in range(0,nny):
    for i in range(0,nnx):
        xV[counter]=i*hx/order
        yV[counter]=j*hy/order
        counter+=1

```

The position of the pressure nodes follows a similar logic.

When it comes to the connectivity array, I first started by building it for each element as follows:

```

if order==1:
    counter=0
    for j in range(0,nely):
        for i in range(0,nelx):
            iconV[0,counter]=(i)*1+0+(j)*1*nnx+nnx*0
            iconV[1,counter]=(i)*1+1+(j)*1*nnx+nnx*0
            iconV[2,counter]=(i)*1+0+(j)*1*nnx+nnx*1
            iconV[3,counter]=(i)*1+1+(j)*1*nnx+nnx*1
            counter += 1

if order==2:
    counter = 0
    for j in range(0,nely):
        for i in range(0,nelx):
            iconV[0,counter]=(i)*2+0+(j)*2*nnx+nnx*0
            iconV[1,counter]=(i)*2+1+(j)*2*nnx+nnx*0
            iconV[2,counter]=(i)*2+2+(j)*2*nnx+nnx*0
            iconV[3,counter]=(i)*2+0+(j)*2*nnx+nnx*1
            iconV[4,counter]=(i)*2+1+(j)*2*nnx+nnx*1
            iconV[5,counter]=(i)*2+2+(j)*2*nnx+nnx*1
            iconV[6,counter]=(i)*2+0+(j)*2*nnx+nnx*2
            iconV[7,counter]=(i)*2+1+(j)*2*nnx+nnx*2
            iconV[8,counter]=(i)*2+2+(j)*2*nnx+nnx*2
            counter += 1

if order==3:
    counter = 0
    for j in range(0,nely):
        for i in range(0,nelx):
            iconV[0,counter]=(i)*3+0+(j)*3*nnx+0*nnx
            iconV[1,counter]=(i)*3+1+(j)*3*nnx+0*nnx
            iconV[2,counter]=(i)*3+2+(j)*3*nnx+0*nnx

```

```

...
iconV[13, counter]=(i)*3+1+(j)*3*nnx+3*nnx
iconV[14, counter]=(i)*3+2+(j)*3*nnx+3*nnx
iconV[15, counter]=(i)*3+3+(j)*3*nnx+3*nnx
counter += 1

```

Having done so, it becomes quickly apparent that the connectivity array can be computed for all elements as follows:

```

counter=0
for j in range(0, nely):
    for i in range(0, nelx):
        counter2=0
        for k in range(0, order+1):
            for l in range(0, order+1):
                iconV[counter2, counter]=i*order+l+j*order*nnx+nnx*k
                counter2+=1
        counter += 1

```

The same approach is taken to build the pressure connectivity array, although the $Q_1 \times P_0$ element requires special attention since the pressure is elemental and attributed to a single node inside the element.

For the other elements I started from:

```

if order==2:
    counter=0
    for j in range(0, nely):
        for i in range(0, nelx):
            iconP[0, counter]=(i)*1+0+(j)*1*(nelx+1)+(nelx+1)*0
            iconP[1, counter]=(i)*1+1+(j)*1*(nelx+1)+(nelx+1)*0
            iconP[2, counter]=(i)*1+0+(j)*1*(nelx+1)+(nelx+1)*1
            iconP[3, counter]=(i)*1+1+(j)*1*(nelx+1)+(nelx+1)*1
            counter += 1

if order==3:
    counter=0
    for j in range(0, nely):
        for i in range(0, nelx):
            iconP[0, counter]=(i)*2+0+(j)*2*(2*nelx+1)+(2*nelx+1)*0
            iconP[1, counter]=(i)*2+1+(j)*2*(2*nelx+1)+(2*nelx+1)*0
            iconP[2, counter]=(i)*2+2+(j)*2*(2*nelx+1)+(2*nelx+1)*0
            iconP[3, counter]=(i)*2+0+(j)*2*(2*nelx+1)+(2*nelx+1)*1
            iconP[4, counter]=(i)*2+1+(j)*2*(2*nelx+1)+(2*nelx+1)*1
            iconP[5, counter]=(i)*2+2+(j)*2*(2*nelx+1)+(2*nelx+1)*1
            iconP[6, counter]=(i)*2+0+(j)*2*(2*nelx+1)+(2*nelx+1)*2
            iconP[7, counter]=(i)*2+1+(j)*2*(2*nelx+1)+(2*nelx+1)*2
            iconP[8, counter]=(i)*2+2+(j)*2*(2*nelx+1)+(2*nelx+1)*2
            counter += 1

if order==4:
    etc ...

```

and quickly arrived at the following compact form:

```

if order >1:
    om1=order-1
    counter=0
    for j in range(0, nely):
        for i in range(0, nelx):
            counter2=0
            for k in range(0, order):
                for l in range(0, order):
                    iconP[counter2, counter]=i*om1+l+j*om1*(om1*nelx+1)+(om1*nelx+1)*k
                    counter2+=1
            counter += 1

```

The core of the code is rather similar if not identical to other stones (i.e. the loop over elements, the calculation of the elemental matrices, their assembly, and the solve).

What is here somewhat elegant is the projection of the pressure field onto the velocity grid nodes (mostly for plotting purposes). For each element I loop over each velocity node, and evaluate the pressure

shape function at this location, compute the pressure with it and add it in the array q while keeping count of how many contributions there are in total per velocity node.

```

for iel in range(0,nel):
    for i in range(0,mV):
        NNNP[0:mP]=NNP(rVnodes[i],sVnodes[i],order)
        q[iconV[i,iel]]+=np.dot(p[iconP[0:mP,iel]],NNNP[0:mP])
        c[iconV[i,iel]]+=1.

q=q/c

```

Finally, since the vtu format does not support higher order elements, I here chose to only extract the corner values for each element, which translates as follows:

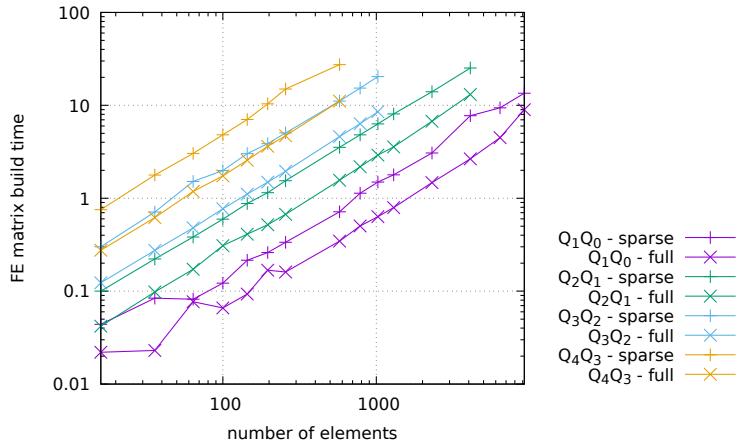
```

vtufile.write("<DataArray _type='Int32' _Name='connectivity' _Format='ascii'>\n")
if order==1:
    for iel in range (0,nel):
        vtufile.write("%d%d%d\n" %(iconV[0 ,iel] ,iconV[1 ,iel] ,iconV[3 ,iel] ,iconV[2 ,iel]))
if order==2:
    for iel in range (0,nel):
        vtufile.write("%d%d%d%d\n" %(iconV[0 ,iel] ,iconV[2 ,iel] ,iconV[8 ,iel] ,iconV[6 ,iel]))
if order==3:
    for iel in range (0,nel):
        vtufile.write("%d%d%d%d\n" %(iconV[0 ,iel] ,iconV[3 ,iel] ,iconV[15 ,iel] ,iconV[12 ,iel]))
if order==4:
    for iel in range (0,nel):
        vtufile.write("%d%d%d%d\n" %(iconV[0 ,iel] ,iconV[4 ,iel] ,iconV[24 ,iel] ,iconV[20 ,iel]))
vtufile.write("</DataArray>\n")

```

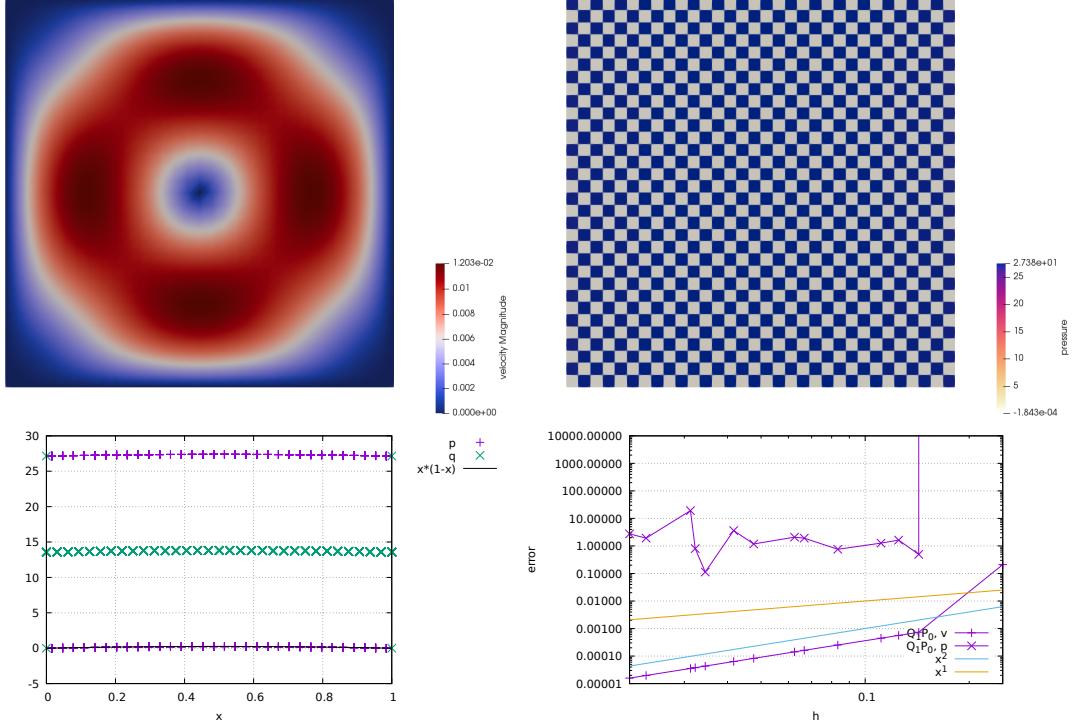
The following results are obtained by running one of the four scripts *script_errorsX* where X=1,2,3,4. The gnuplot script is to be found in the *images* folder.

The stone implements two ways of building the FE matrix. When the flag *sparse* is false, the \mathbb{K} and \mathbb{G} matrices are built as full arrays, later assembled in a larger full array, and then only converted to Compressed Sparse Row before it is passed to the solver. When the flag is true, the global FE matrix is defined as a *lil_matrix* (a List of Lists) and it grows in size/memory every time a new term is added to it. As shown on the following plot, it is about twice as slow compared to the first option, but it uses only a fraction of the memory that the first one does.



Also not very surprising: the cost of building the FE matrix increases with the order of the used elements. A matrix corresponding to 100 $Q_1 \times P_0$ elements can be built in about 1s, while it will take 7s when $Q_4 \times Q_3$ elements are used.

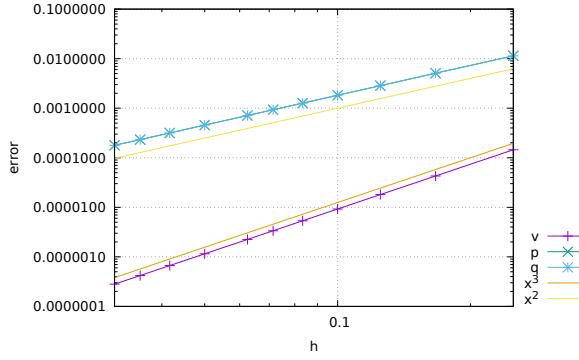
Results with $Q_1 \times P_0$ element



We see that we recover a second order convergence rate for velocity (as expected) but because of the checkerboard pattern the pressure convergence is simply random. The smoothed pressure q shows virtually no checkerboard pattern, except on the boundaries. This is a perfect example for the use of more accurate/clever smoothing procedure, see Section 8.12.

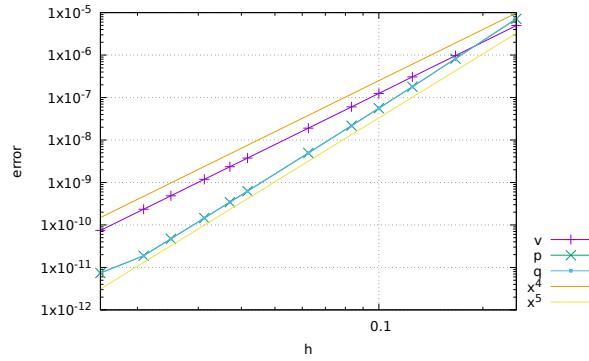
Results with $Q_2 \times Q_1$ element

We recover the cubic convergence for the velocity error and the quadratic convergence for the pressure:



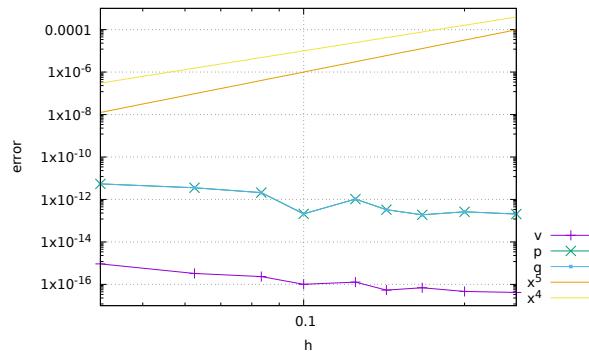
Results with $Q_3 \times Q_2$ element

The analytical solution is a second order polynomial, which means that the pressure shape functions can adequately represent the solution. We recover a fourth-order convergence for the velocity error and a superconvergent (fifth order) pressure error (but why is it 5th order ?).



Results with $Q_4 \times Q_3$ element

Rather interestingly, now both velocity and pressure analytical solutions can be represented exactly by their respective polynomial spaces, so that the errors are at machine precision.

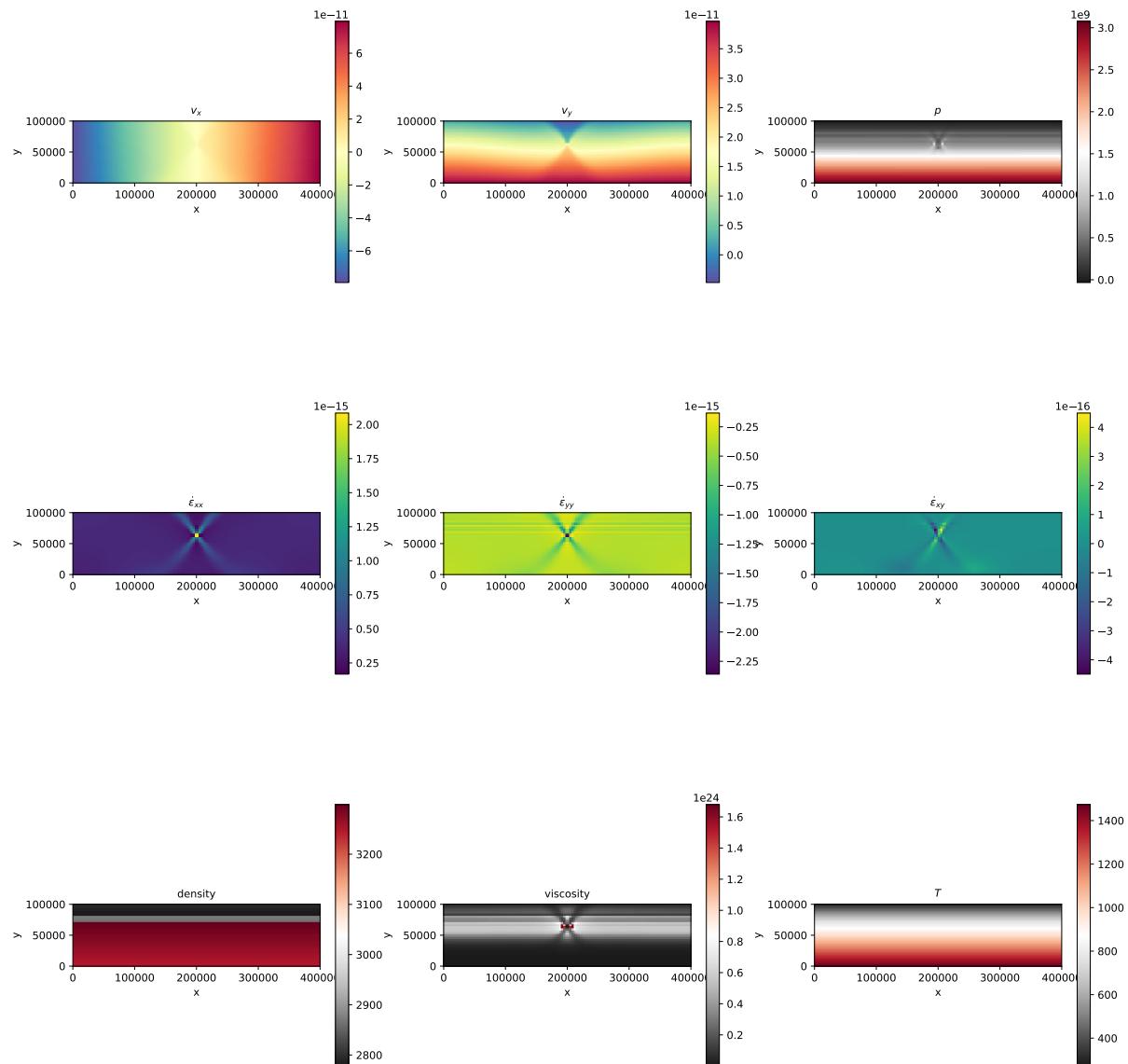


Stone 49: Consistent Boundary Flux method on D&H benchmark with 4 elements

Looking at the four different elements

Looking at the influence of the mas matrix lumping

Stone 50: Lithosphere extension



Stone 51: Triangular domain benchmark with MINI element

This fieldstone was developed in collaboration with L. van de Wiel.

The following problem is studied in [1000]. The equations that they solve are the thermo-mechanically coupled steady state equations:

$$-\vec{\nabla}p + \Delta\vec{v} + RaTe_y = 0 \quad (806)$$

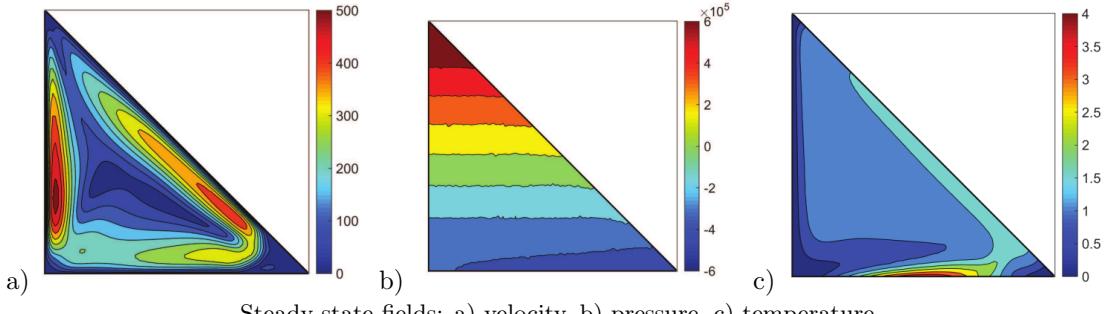
$$\vec{\nabla} \cdot \vec{v} = 0 \quad (807)$$

$$-\Delta T + \vec{v} \cdot \nabla T = 0 \quad (808)$$

In our case the code is based on the MINI element (a.k.a. $P_1^+ \times P_1$), see Section 6.2.8 and we set $Ra = 10^6$.

The domain is chosen to be the right triangle with vertices $(0, 0)$, $(1, 0)$, and $(0, 1)$. The boundary is considered to be solid walls (no-slip). For the temperature, a sinusoidal heat source is enforced on the bottom boundary with a Dirichlet condition ($T(x) = 2(1 - \cos(2\pi x))$), the left wall is set to a constant temperature of zero, and the hypotenuse wall is perfectly insulated so that a Neumann boundary condition is appropriate.

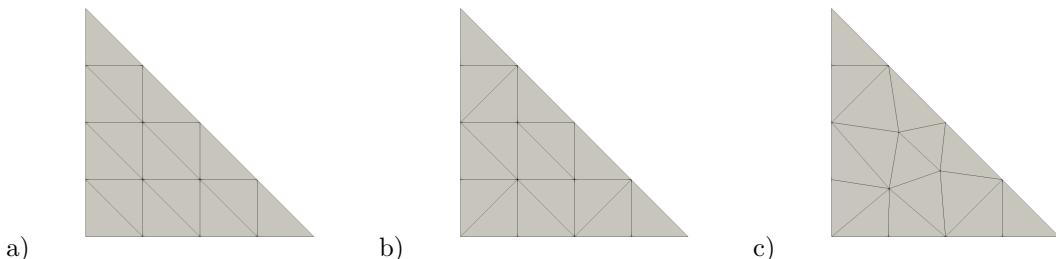
The steady state velocity pressure and temperature fields as shown in [1000] are as follows:



Steady state fields: a) velocity, b) pressure, c) temperature.

Although it is not mentioned in the original article it appears that the pressure field has been normalised so that $\langle p \rangle = \int_{\Omega} pdV = 0$.

As opposed to the mesh presented in [1000] I build a regular mesh. An example of such a mesh is shown hereunder (a) for $n = 5$ (the number of nodes per side of the triangular domain). In order to generate a mesh which is more isotropic some edges between triangles can be flipped (b). Note that this mesh can also be modified in such a way that the position of nodes inside the domain is perturbed by a small random value (c). In what follows I denote by h the distance between nodes on the horizontal (or vertical) boundaries, i.e. $h = L_x/(n - 1) = L_y/(n - 1)$.



a) regular mesh for $n=5$. b) flipped edges mesh. c) randomized+flipped edges mesh.

Since I am solving for the steady-state solution I set the mass matrix in the heat transport part of the code to zero. However, since I am solving the Stokes equations and the heat transport equation alternatively until convergence is reached, it is well known that this approach does not converge fast (if at all). I then implement a simple relaxation scheme [1892]. After I have solved for velocity (using the most recent temperature field in the rhs), I do:

$$\vec{v}^k = \gamma \vec{v}^k + (1 - \gamma) \vec{v}^{k-1}$$

and after having solved for temperature having used the most recent velocity field, I do the same for temperature:

$$T^k = \gamma T^k + (1 - \gamma) T^{k-1}$$

where the relaxation parameter γ is between 0 and 1.

Additionally I measure:

- the Nusselt number defined by

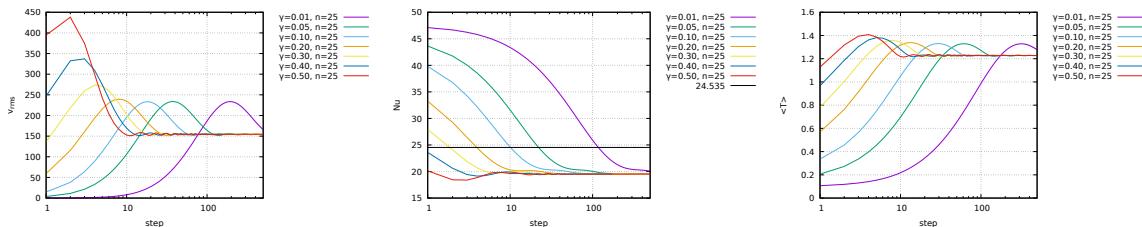
$$Nu = \int_{y=0} \vec{\nabla} T \cdot \vec{n} dS = - \int_{x=0}^{x=1} \frac{\partial T}{\partial y} dx$$

It is reported to be 24.535 in [1000].

- the temperature on the hypotenuse.
- the root mean square velocity
- the mean temperature $\langle T \rangle = |\Omega|^{-1} \int_{\Omega} T dV$

On the importance of the relaxation parameter γ

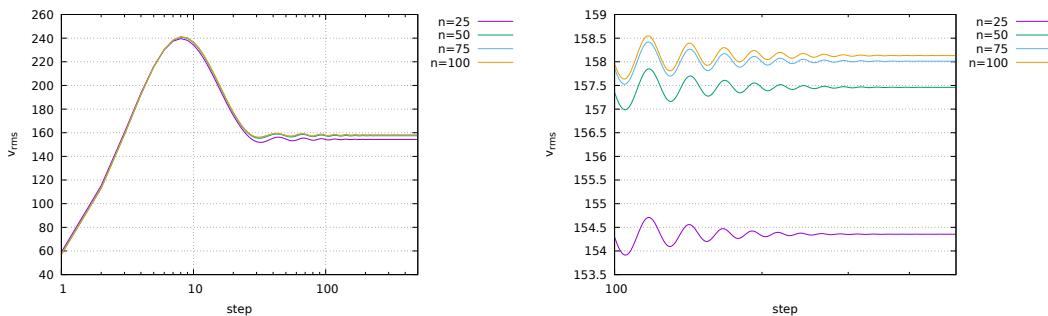
In what follows the internal node coordinate randomness is switched off. I have run the model for various values of γ for $n = 25$, and results are shown on the following figures. We see that all simulations seem to converge to the same steady state, which is very reassuring. However, it looks like too small a value of γ delays greatly the convergence and too large a value also seems detrimental. In light of this, I have chosen $\gamma = 0.2$ for all what follows.

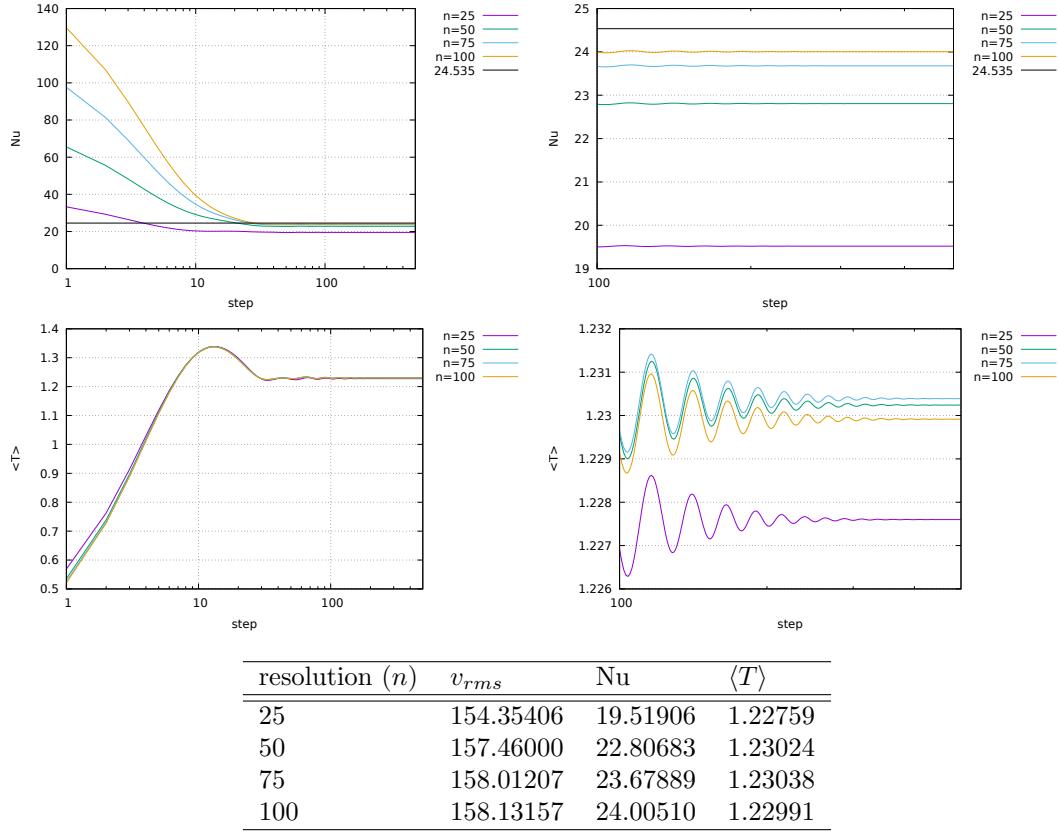


Left to right: Root mean square velocity, Nusselt number and average temperature as a function of the iteration counter.

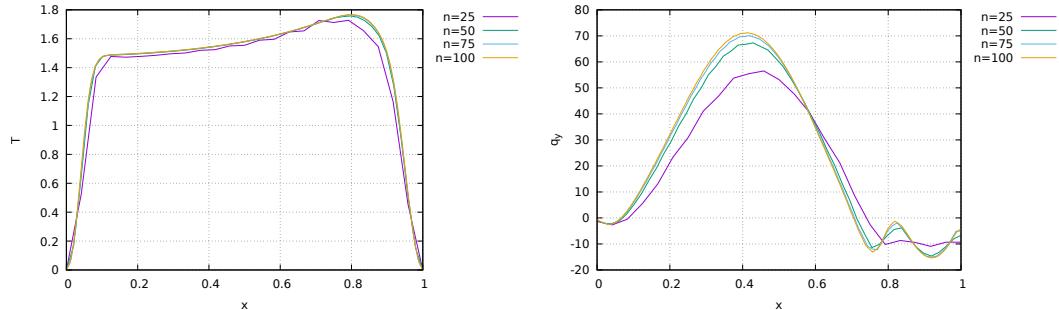
On the influence of mesh resolution

I now explore the influence of the mesh resolution on the results and run steady state calculations for $n = 25, 50, 75, 100$.

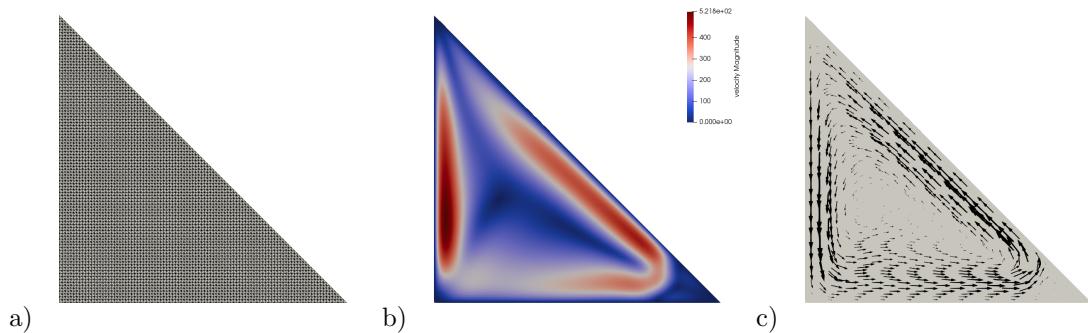


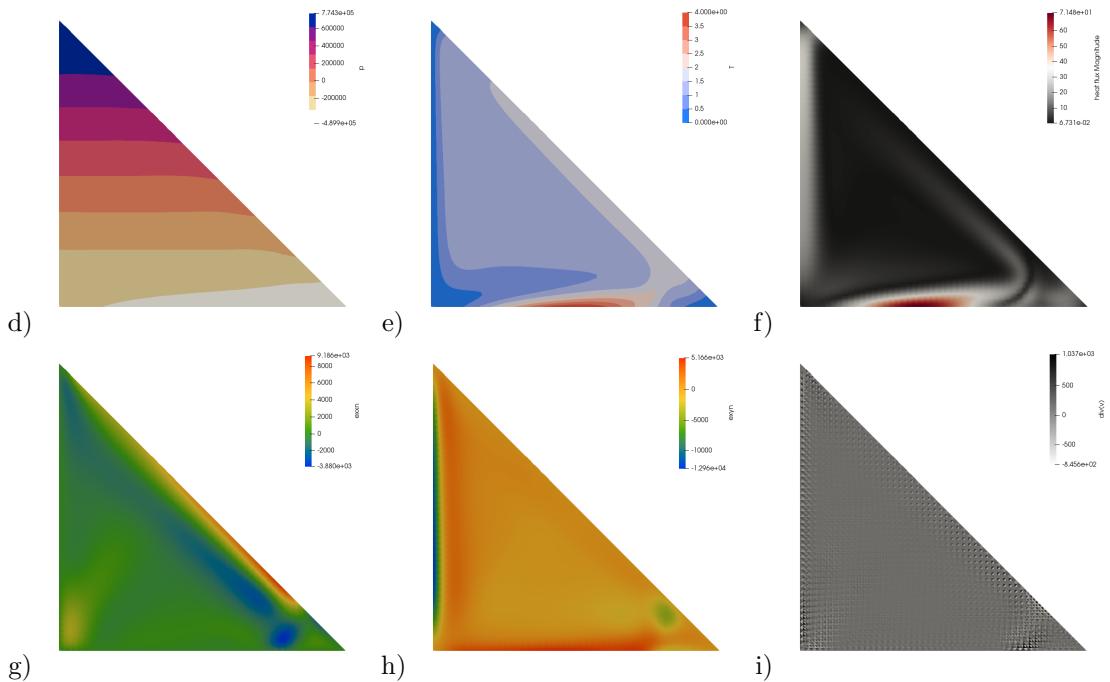


The following two plots show the temperature along the hypotenuse (as a function of x for simplicity) at steady-state and the heat flux measured at the bottom.



Finally, here are the fields for $n = 100$ at steady state:





Steady state fields: a) grid ($n=100$), b,c) velocity, d) pressure, e) temperature, f) heat flux, g) $\dot{\varepsilon}_{xx}$, h) $\dot{\varepsilon}_{xy}$, i) velocity divergence measured in the middle of the element.

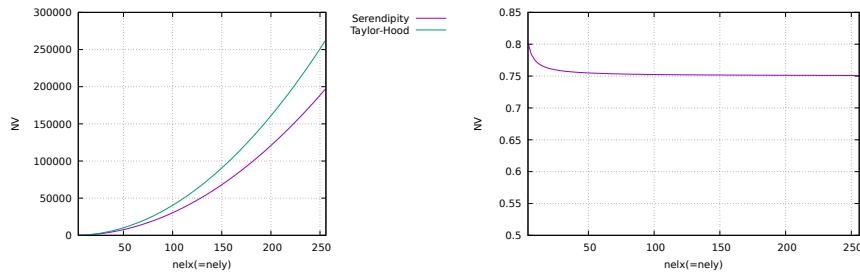
Stone 52: Serendipity element in 2D

In order to test the grid point and connectivity algorithms, we use this simple 4×3 element mesh:

Q_2 X Q1 (serendipity)	Q_2 X Q1 (regular)
15--47--16--48--17--49--18--50--19	54--55--56--57--58--59--60--61--62
	: : : :
42 43 44 45 46	45..46..47..48..49..50..51..52..53
	: : : :
10--38--11--39--12--40--13--41--14	36--37--38--39--40--41--42--43--44
	: : : :
33 34 35 36 37	27..28..29..30..31..32..33..34..35
	: : : :
05--29--06--30--07--31--08--32--09	18--19--20--21--22--23--24--25--26
	: : : :
24 25 26 27 28	09..10..11..12..13..14..15..16..17
	: : : :
00--20--01--21--02--22--03--23--04	00--01--02--03--04--05--06--07--08
iel= 0:	iel= 0
node 0 : 0 at pos. 0.0 0.0	node 0 : 0 at pos. 0.0 0.0
node 1 : 1 at pos. 1.0 0.0	node 1 : 2 at pos. 1.0 0.0
node 2 : 6 at pos. 1.0 1.0	node 2 : 20 at pos. 1.0 1.0
node 3 : 5 at pos. 0.0 1.0	node 3 : 18 at pos. 0.0 1.0
node 4 : 20 at pos. 0.5 0.0	node 4 : 1 at pos. 0.5 0.0
node 5 : 25 at pos. 1.0 0.5	node 5 : 11 at pos. 1.0 0.5
node 6 : 29 at pos. 0.5 1.0	node 6 : 19 at pos. 0.5 1.0
node 7 : 24 at pos. 0.0 0.5	node 7 : 9 at pos. 0.0 0.5
	node 8 : 10 at pos. 0.5 0.5

We see that the serendipity element-based mesh counts only 51 nodes, as opposed to 63 for its counterpart.

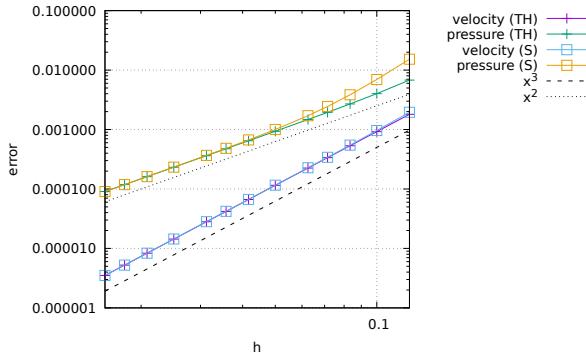
Setting $nelx = nely$, we can look at the number of velocity nodes for each as a function of $nelx$, as shown hereunder:



Looking at the ratio between both, we see that ultimately at high resolution, a mesh composed of serendipity elements will count 25% less nodes than a mesh with Taylor-Hood elements. Since there is not free lunch, what is the price paid in terms of accuracy when using the cheaper serendipity?

The shape functions and their derivatives are in Section 4.5.3.

Although the vtk format does not understand the Q_2 element in 2D or 3D, it surprisingly does understand the serendipity element in 2D (type=23) and 3D (type=25).

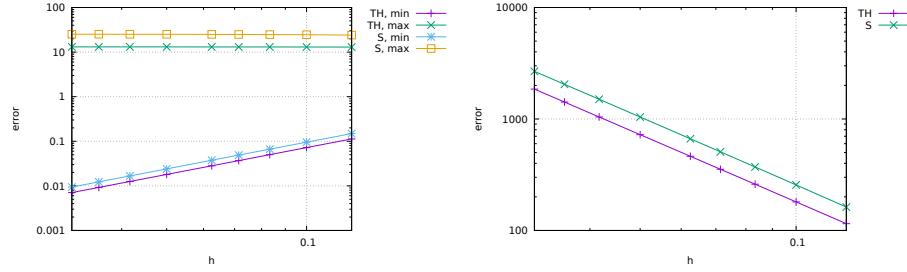


It looks like the serendipity element yields the same errors and error convergence rates as its Taylor-Hood counterpart. Since it is cheaper in terms of dofs, one could think that it should be preferred. However, most modern codes use an iterative solver approach to solve the discretised Stokes problem, and often the \mathbb{K} matrix (which is SPD) is 'solved' with a conjugate gradient solver. The convergence of this type of solver depends on the condition number of the matrix itself, i.e. the ratio of the largest and smallest eigenvalues. Note that this is rather trivial with Python:

```
print('condition_number:', nel, linalg.cond(K_mat))
```

However, since I was also curious about the values of the eigenvalues, I implemented it as follows:

```
eigvals, eigvecs = linalg.eig(K_mat)
print('eigenvalues:', nel, eigvals.min(), eigvals.max())
```



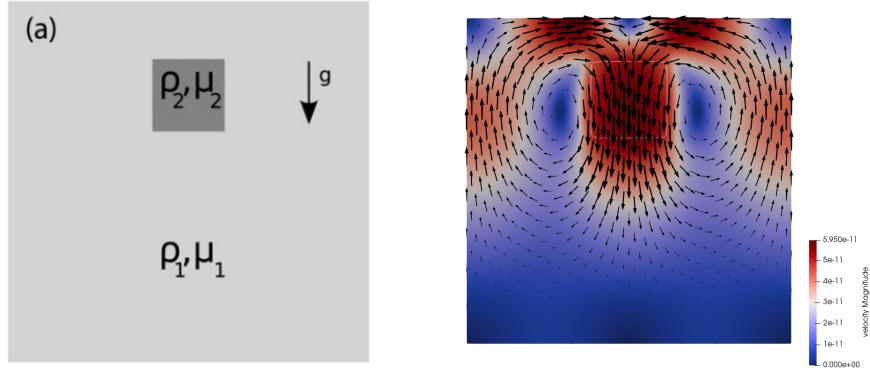
Left: min and max eigenvalues for both types of elements as a function of h ; Right: condition number

As it turns out, the condition number is twice as high for the serendipity element, which means that the CG would have to iterate more to arrive at the solution, thereby offsetting the benefit of less dofs.

Stone 53: the sinking block benchmark

https://github.com/cedrict/fieldstone/tree/master/python_codes/fieldstone_53

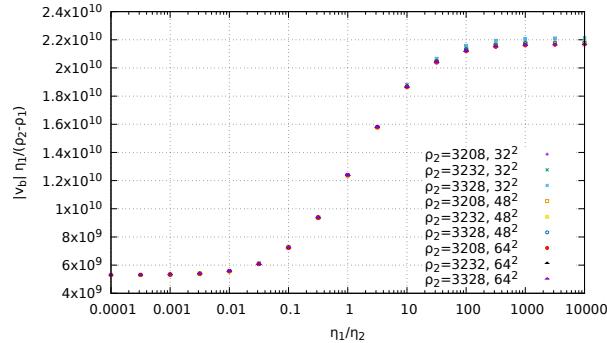
This simple benchmark provides challenging numerical experiments dealing with large viscosity variations within the simulation domain. It appears in [722] and consists of a bulk of fluid 1 (ρ_1, η_1) in which a block of fluid 2 (ρ_2, η_2) falls under its own weight. The domain is a square of size $L_x = L_y = 512\text{km}$ and the block is initially centred at point ($x = 256\text{ km}, y = 384\text{ km}$) with size $128 \times 128\text{ km}$:



Left: setup. Right: velocity field for $\rho_2 = 3208$, $\eta_1 = 10^{21}$ and $\eta_2 = 10^{22}$.

The simulation is carried out on 32×32 , 48×48 and 64×64 grids. Free slip boundary conditions are imposed on all sides of the domain. In all experiments the density of the surrounding fluid is $\rho_1 = 3200\text{kg/m}^3$. The velocity v_b of the falling block is measured in its centre (note that due to symmetry the horizontal component should be zero).

As explained in [1756], following physical intuition, one expects the velocity v_b of the block to (a) decrease when the viscosity of the surrounding medium η_1 increases; (b) increase with the density contrast $\rho_2 - \rho_1$. The quantity $v_b \eta_1 / (\rho_2 - \rho_1)$ is therefore monitored and shown hereunder as a function of the viscosity ratio.



Series of experiments have been conducted with $\rho_2 = 3208, 3232, 3328$, $\log_{10}(\eta_1) = 20, 21, 22$ and $\log_{10}(\eta_2) = 18, 18.5, 19, \dots, 22.5, 23, 23.5, 24$, all with 3 mesh resolutions.

All experimental points line up on a single curve which further indicates that the code can deal with gravity driven simulations in the presence of large viscosity contrasts. These results have been successfully compared with those obtained with ASPECT with the same setup.

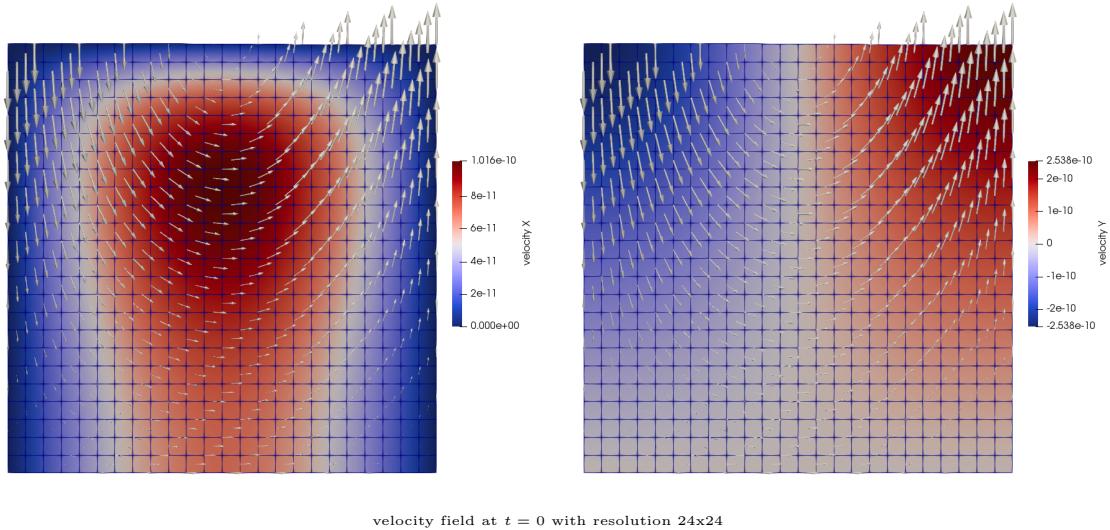
Stone 54: ALEs

https://github.com/cedrikt/fieldstone/tree/master/python_codes/fieldstone_54

This stone implements three different free surface/mesh deformation algorithms. The first one has all the nodes move with the computed velocity (Lagrangian method) and is coined 'method 1'. The second one ('method 2') only has the top row of nodes moving with the computed velocity, while all the nodes underneath are static (this is obviously not a viable method for large deformations). The third one ('method 3') is the method used in ASPECT and described in Rose et al (2017) [1599]. Its implementation is described in Section 8.1.

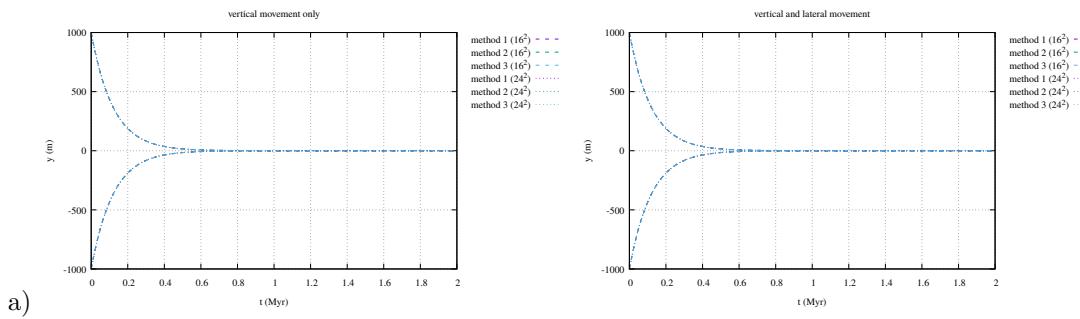
Experiment 1 - relaxation of topography

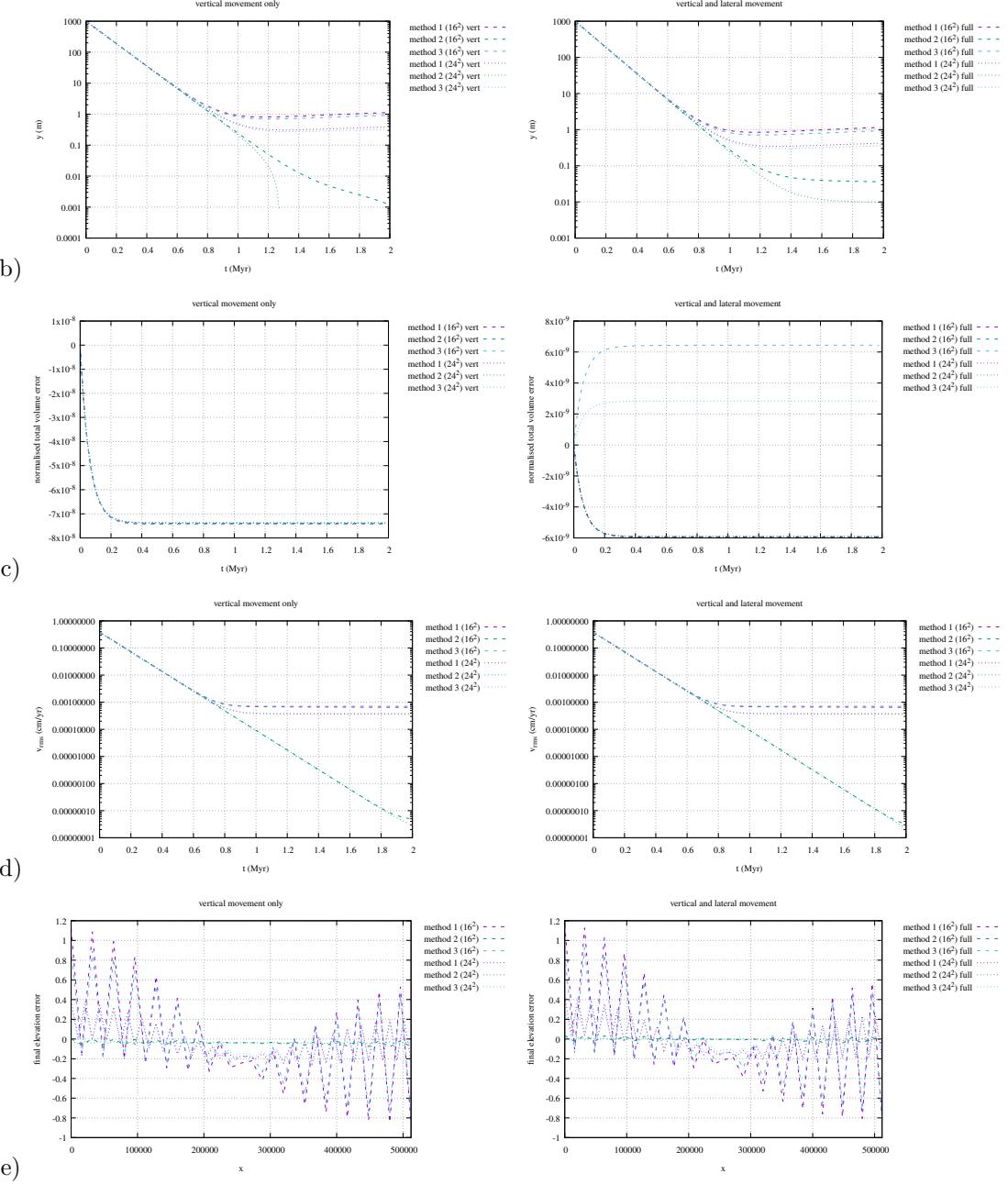
The domain is a 2D Cartesian box of size 512×512 km, with free slip on left, bottom and right sides, free surface at the top. Mantle material characterised by $\rho_m = 3200$ and $\eta_m = 10^{22}$. Gravity is vertical and Earth like. The surface is perturbed at startup by $\delta y = A \cos(\pi x / L_x)$ with $A=1$ km. 200 time steps with $\delta t = 10$ kyr are carried out. The root mean square velocity, the total volume of the domain, the min/max elevation values of the surface are recorded over time.



velocity field at $t = 0$ with resolution 24×24

The results hereunder are obtained for all three methods at two different resolutions (16×16 and 24×24 elements). The plots on the left column are obtained with the movement of the top nodes being constrained in the vertical direction, while the plots on the right column are obtained with nodes being allowed to move in both x and y directions (note that for method 3 the normals are not -yet- computed with the method of Eq. 49 in [1599] but instead by a simple geometric rule).

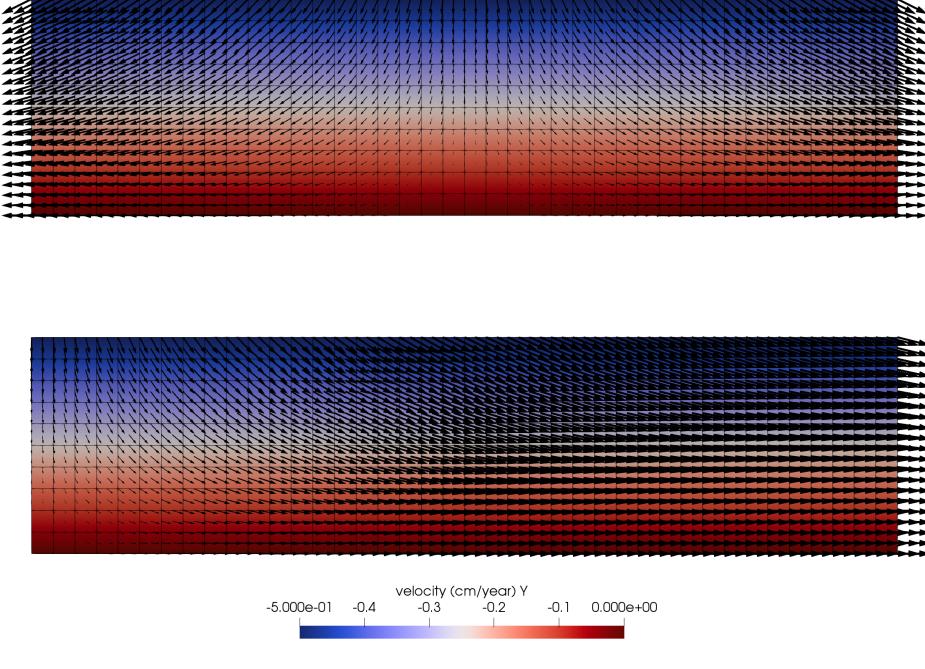




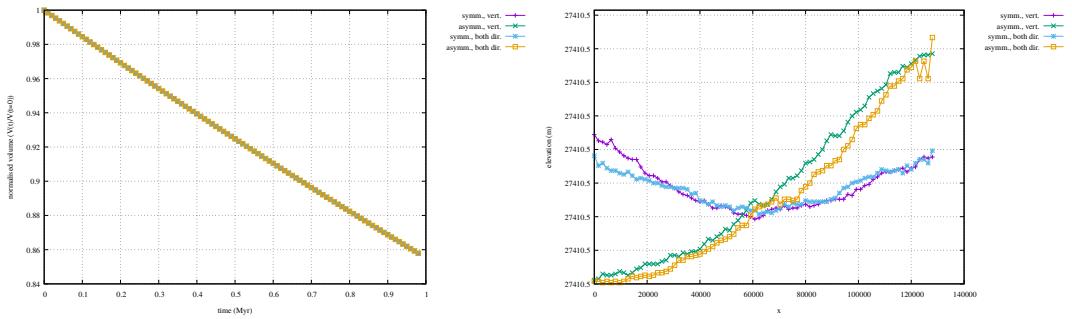
a) min/max of free surface topo as a function of time; b) free surface topo maximum (in log scale) as a function of time; c) measured volume of the domain with numerical quadrature normalised by the expected volume $L_x L_y$; d) root mean square velocity as a function of time; e) final elevation at the 200th time step.

Experiment 2,3 - extension

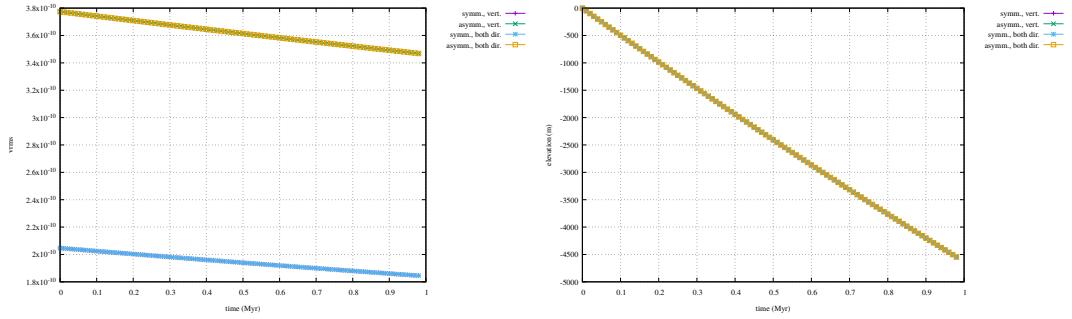
We now consider a rectangular domain (crust sized) of $128 \times 32\text{km}$, discretised by means of 40×10 elements. The fluid is identical to the one above and so is gravity. Extensional boundary conditions are applied. In the first case, a 1cm/yr horizontal velocity is applied on both sides, while in the second case a 2cm/yr velocity is applied on the right while 0 is prescribed to the left. Free slip conditions are otherwise prescribed on the sides and bottom. 100 time steps are carried out with $\delta t = 10^4\text{yr}$. Only method 3 is used here.



Despite the asymmetry in the boundary conditions, we expect the same evolution of the domain geometry. This is indeed what we recover with surprising accuracy. 'vert' stands for only vertical movement allowed (i.e. $n_x = 0$, $n_y = 1$) while 'both dir' stands for the use of dynamically computed normal vectors (based on geometrical consideration).



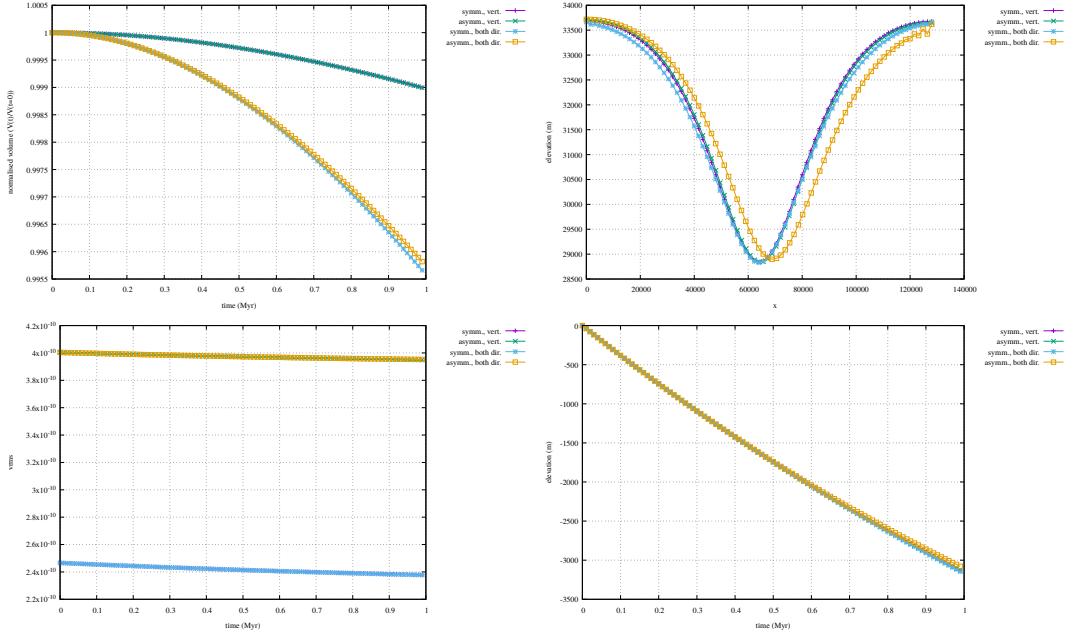
Left: time evolution of the normalised volume for both boundary condition types; Right: surface at the end of the simulation.



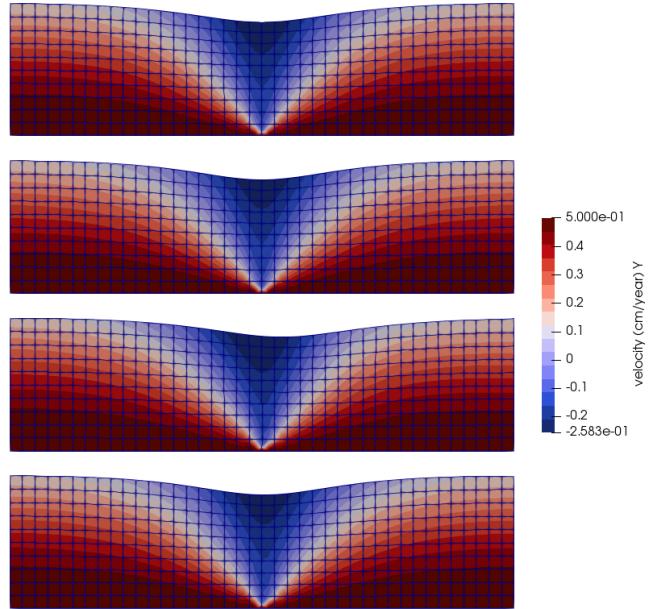
Left: Root mean square velocity as a function of time; Right: time evolution of the elevation (min and max virtually indistinguishable).

Experiment 4,5 - extension with bottom inflow

This is the same setup as above, but we now impose an influx boundary condition at the bottom: $v = 0.5\text{cm}$ (this balances the outflux exactly) and the horizontal component at the bottom is set to the left value (-1 or 0 cm/yr) for $x \leq L_x/2$ and to the right value +1 or +2 cm/yr for $x \geq L_x/2$.

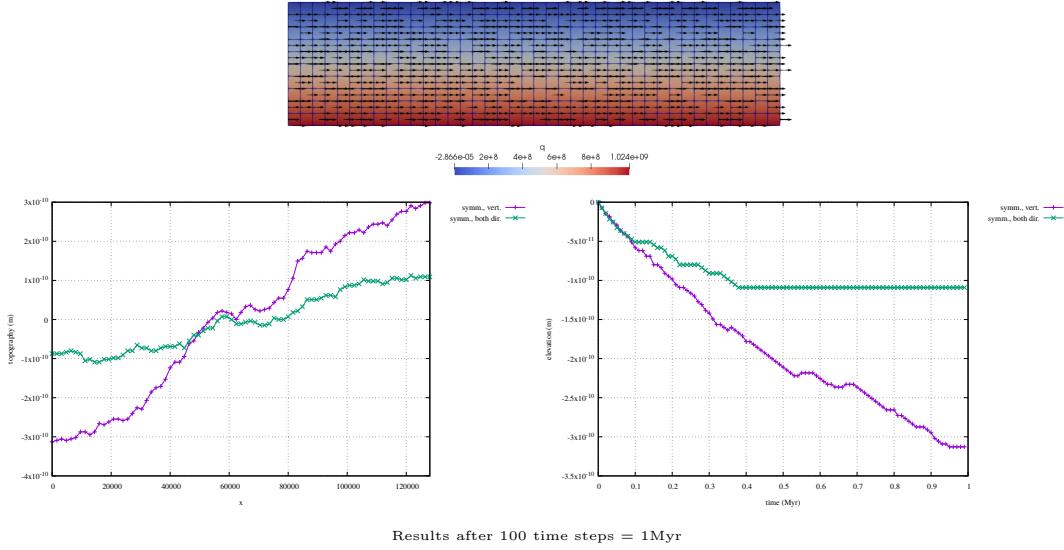


We can conclude that the 'vertical movement only' conserves volume/mass better and the free surface remains symmetrical unless the full normal is used which is simply explained by having $\vec{v} \cdot \vec{n}$ as a boundary condition: in the asymmetric extension case, the velocity is always to the right while the normal has an x component which is negative and positive, therefore introducing an asymmetry in the surface boundary conditions.



Top to bottom: symmetric extension, full normal vector; symmetric extension, vertical normal vector; asymmetric extension, full normal vector; asymmetric extension, vertical normal vector

Experiment 6 - pure advection test



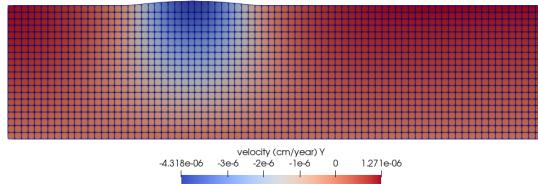
Experiment 7 - pure advection test of a cosine bump

The initial topography bump is given by

$$y(x) = A \left[1 + \cos \left(\frac{x - x_0}{w} \pi \right) \right]$$

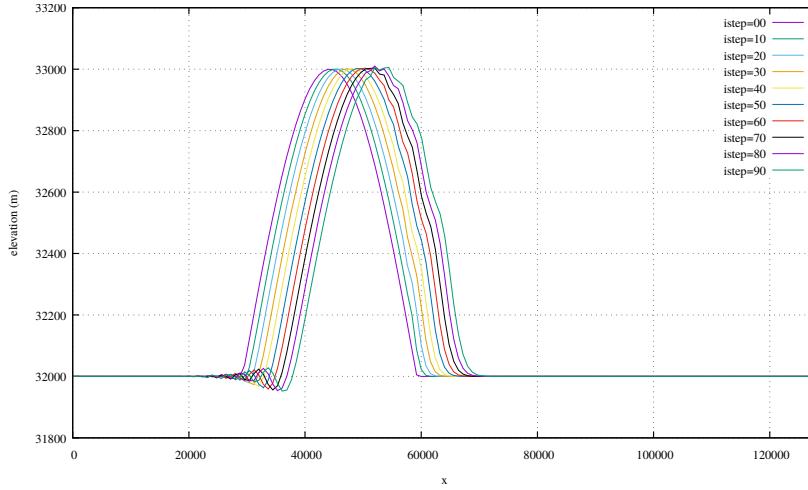
with $A = 1000\text{m}$, $x_0 = 0.345678L_x$ and $w = 15\text{km}$. This is a somewhat ideal case since the transition from flat to bump is very smooth.

The viscosity is set to $10^{26}\text{Pa}\cdot\text{s}$ so that the velocity of the viscous relaxation of the topography is negligible with regards to the advection velocity (+1cm/yr on left and right boundaries). I also choose a resolution of 80x20 elements.

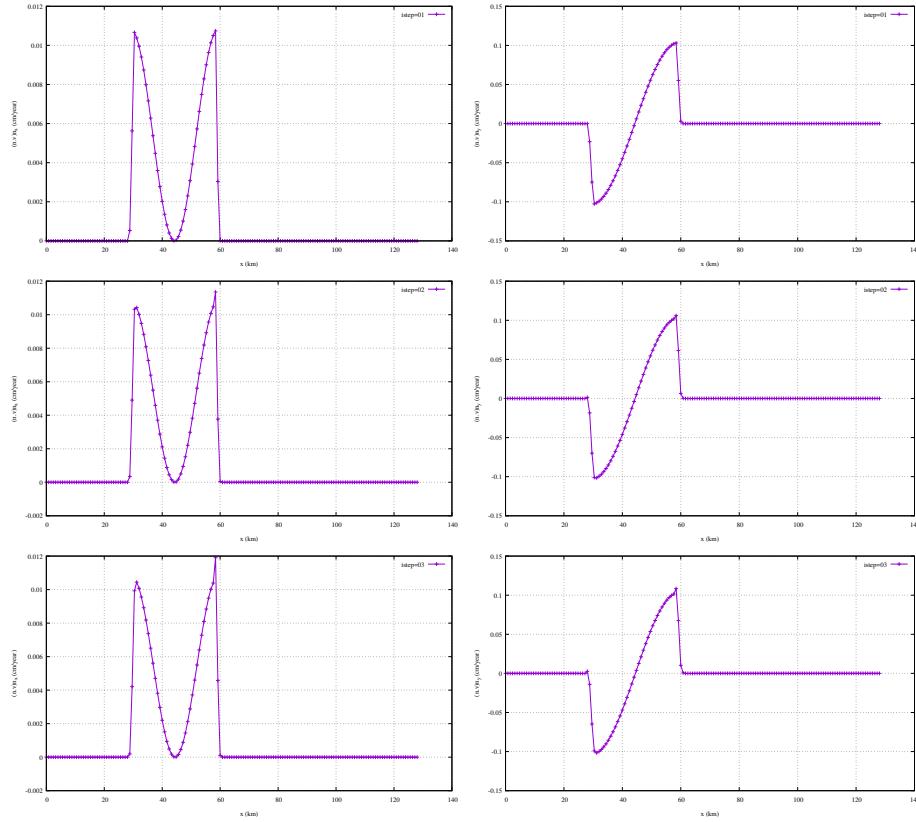


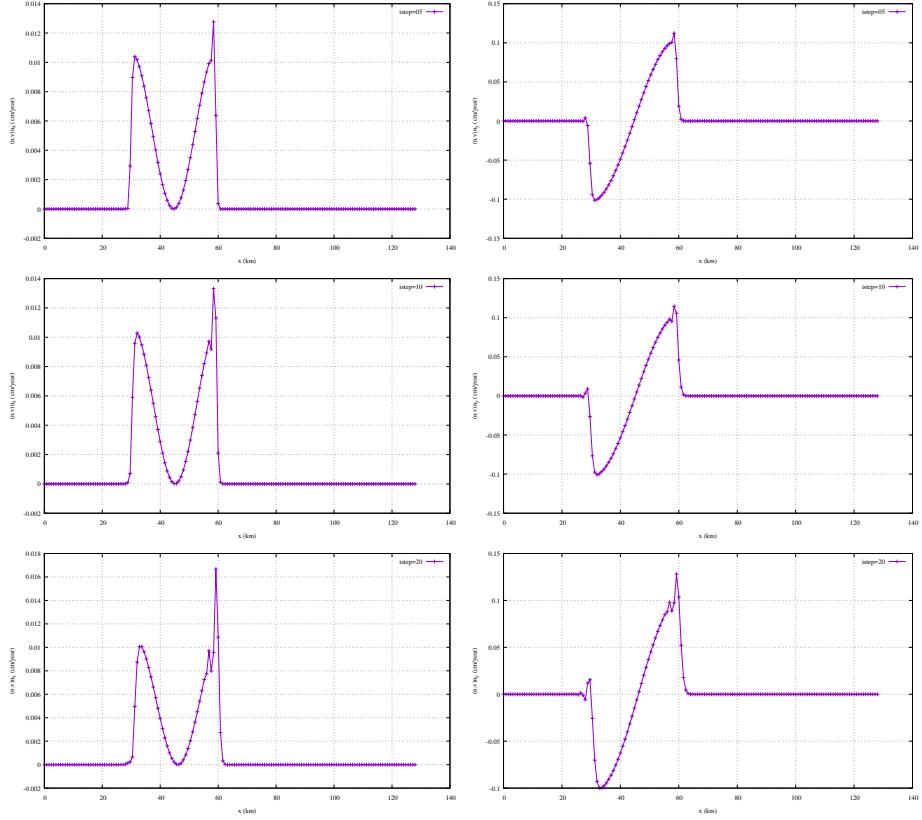
If only the vertical normal is used then *nothing* moves since the velocity is always perpendicular to the normal.

What follows is obtained when the full normal is used. After 20 timesteps the topography has been advected and we already observe some visible asymmetry (and oscillations) on the surface:



On the following plots the left column shows $(\vec{v} \cdot \vec{n})n_x$ as a function of the x -coordinate and the right column shows $(\vec{v} \cdot \vec{n})n_y$, both for timesteps 1,2,3,5,10,20. Both quantities form the boundary conditions for the mesh deformation. Since I have chosen x_0 such that it does not fall on a node, the initial topography is *not* symmetric and therefore the normal vectors at the nodes on the left and right of the peak are not identical. From the second timestep (the first one for which the free surface algo uses a non zero Stokes velocity as boundary condition) we see that both components $(\vec{v} \cdot \vec{n})n_x$ and $(\vec{v} \cdot \vec{n})n_y$ are asymmetric!

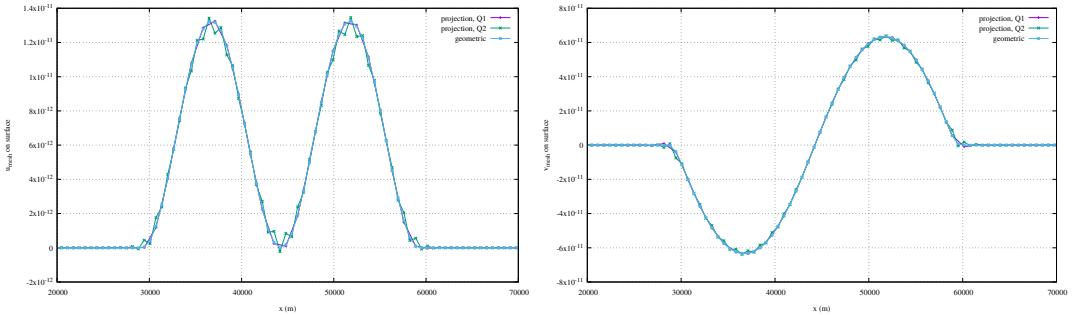


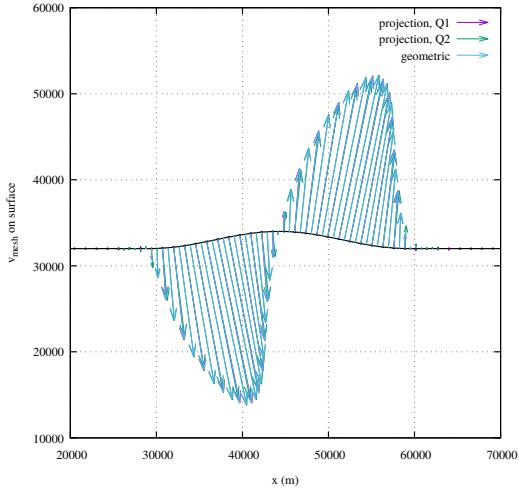


The inescapable conclusion is that the algorithm (as it is now implemented with the normal vector) is incapable of advecting a bump.

In what follows, the Stokes system is solved once, and the obtained velocity is used to compute the mesh velocity boundary condition for the Laplace system. The resolution is set to 100x25. I have implemented the L_2 projection approach of [1599] for both Q1 and Q2 elements. For such a smooth topography the Q1 and geometrical approach (i.e. using geometrically computed normal vectors) are very similar although Q1 produces tiny undershoots. Q2 however generates oscillations which makes the computed velocity not suited as a boundary condition to move surface nodes.

From left to right: horizontal component, vertical component, and vector form of the computed mesh velocity bc.

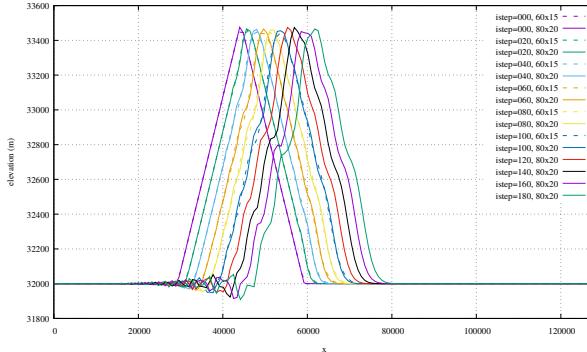
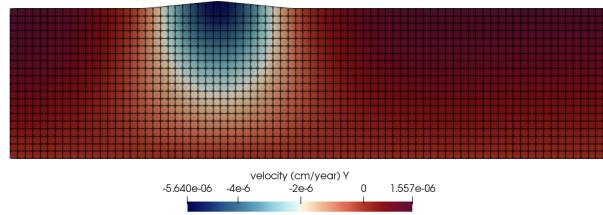


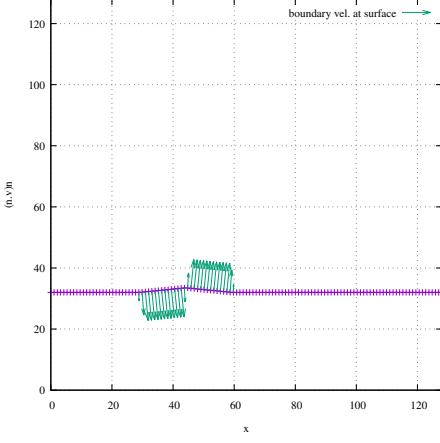


I haven't used any of the projections to carry out timestepping.

Experiment 8 - pure advection test of a pyramidal bump

This is a nearly identical experiment as the previous one, but now the bump is composed of two straight lines, of slopes ± 0.1 . Pyramid centered at $x = 0.345678L_x$, of half width 15km. $\rho = 3000$, $\eta = 10^{26}$. Resolution 60x15. +1cm/yr prescribed on left and right. $dt=10\text{kyr}$.

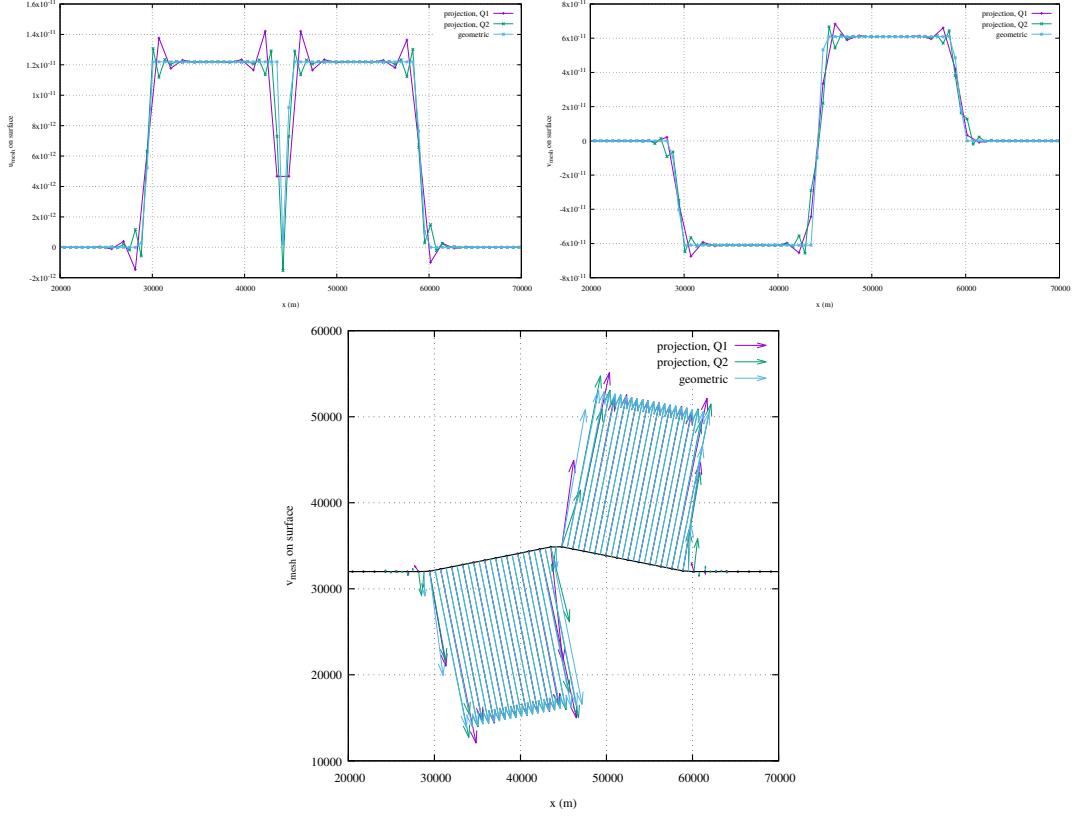




Results are not crazy bad, but we should be able to do better ... the pyramid becomes more and more deformed after only 20km of advection (but surprisingly retains its height).

In what follows, the Stokes system is solved once, and the obtained velocity is used to compute the mesh velocity boundary condition for the Laplace system. The resolution is set to 100x25. I have implemented the L_2 projection approach of [1599] for both Q1 and Q2 elements. For such a broken line topography the geometrical approach (i.e. using geometrically computed normal vectors) is better. Q1 and Q2 both produce under/overshoots/oscillations with no clear winner.

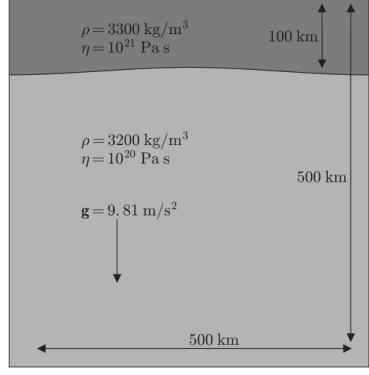
From left to right: horizontal component, vertical component, and vector form of the computed mesh velocity bc.



I haven't used any of the projections to carry out timestepping.

Experiment 9 - Rayleigh-Taylor instability

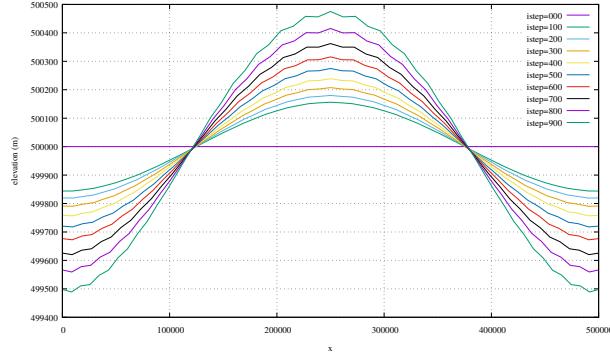
The setup is as follows:



The sinusoidal perturbation is given by

$$y(x) = 400e3 - 500 \cos(2\pi x/L_x)$$

Because I do not have compositional fields or markers implemented in this code I have to align the mesh with the initial sinusoidal perturbation and run the model in Lagrangian mode. Also, as shown in [1032], this experiment is prone to drunken sailor instabilities and I have therefore fixed $dt=1000\text{yr}$. As a consequence, the model will run until elements become too distorted.



Stone 55: Subduction as a thin-sheet problem

This stone was partially contributed by E. van der Wiel.

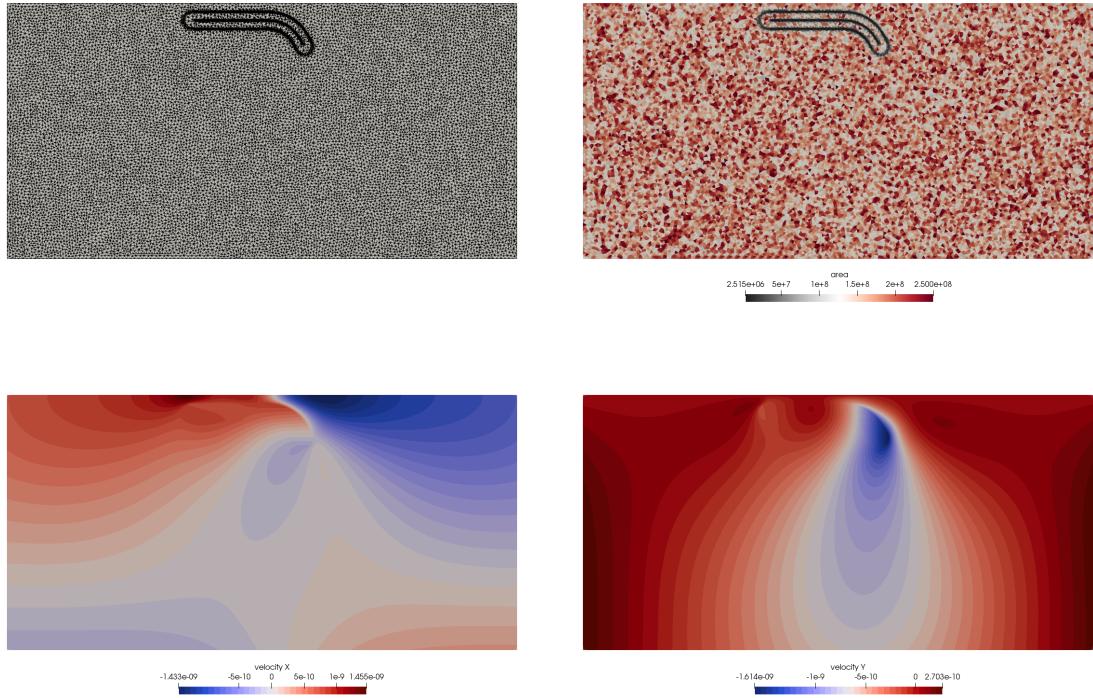
https://github.com/cedrict/fieldstone/tree/master/python_codes/fieldstone_55

Parameters for the setup are defined in *parameters.py*. This file is used in *generate_nodes.py* which produces the *subd.node* file which contains the coordinates of all key points on the boundary of the domain and along the material interfaces. This file is then further processed by the triangle program⁶⁶ as follows:

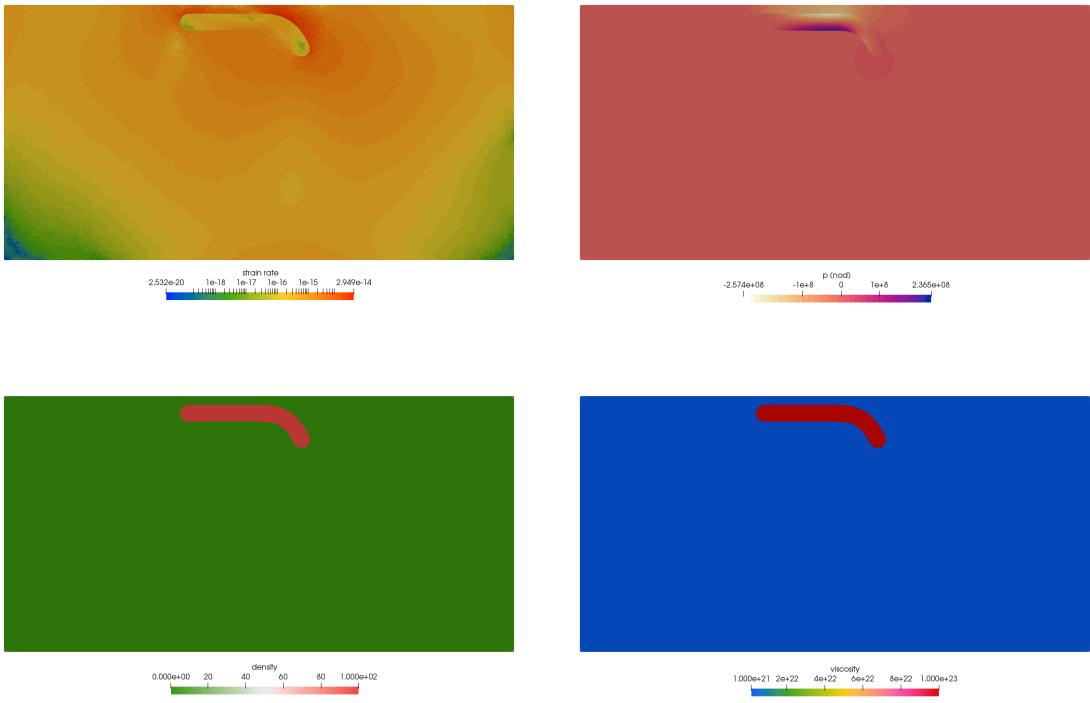
```
./triangle -q -a200000000 -o2 subd.node
```

The '*-q*' option adds vertices to the mesh to ensure that all angles are between 20 and 140 degrees. The '*-a*' makes sure that no triangle has an area larger than the supplied number. The '*-o2*' generates a mesh composed of second order triangles (six nodes per element, rather than three) and the three extra nodes of an element fall at the midpoints of the three edges. This generates two files: '*subd.1.ele*' which contains the connectivity of all generated triangles and '*subd.1.node*' which contains the coordinates of all nodal points. These two files are then read in *fieldstone.py* and stored in the *xV*, *yV* and *iconV* arrays.

Gravity is vertical and Earth-like. Free-slip boundary conditions are imposed on the top while the other boundaries are free (in/outflow determined freely based on the internal dynamics). In order to remove the horizontal null space the average horizontal velocity is set to zero. Crouzeix-Raviart elements are used, see Section 6.2.10. The density of the mantle is set to zero while the subducting plate has a density $\delta\rho$.

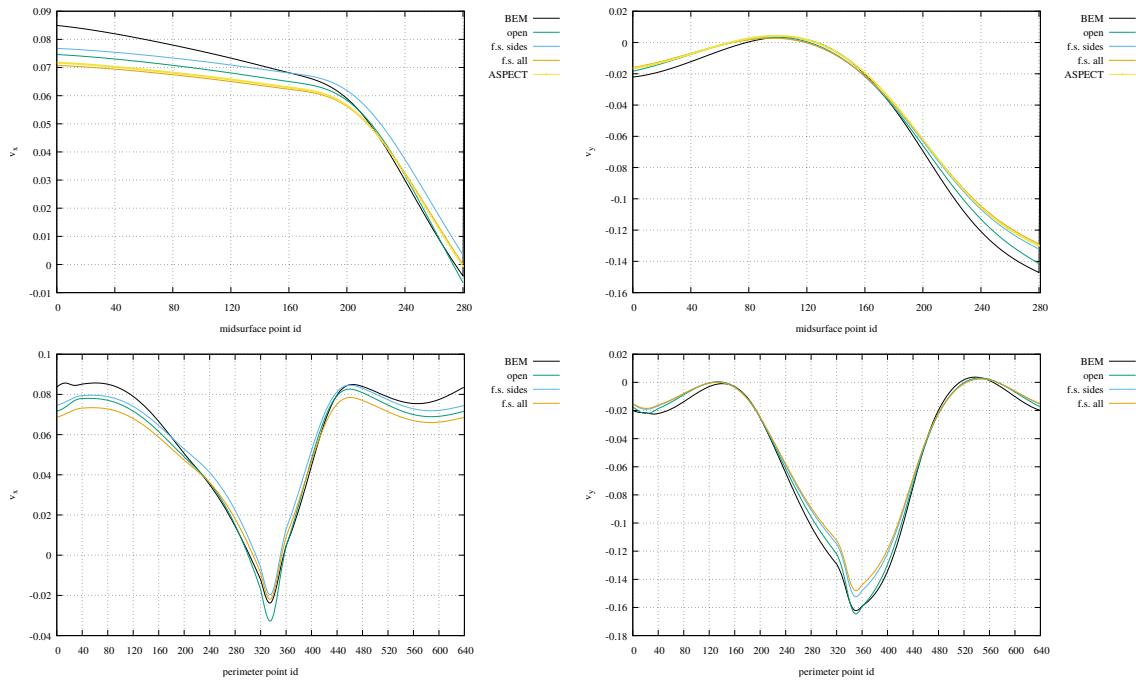


⁶⁶<https://www.cs.cmu.edu/~quake/triangle.html>



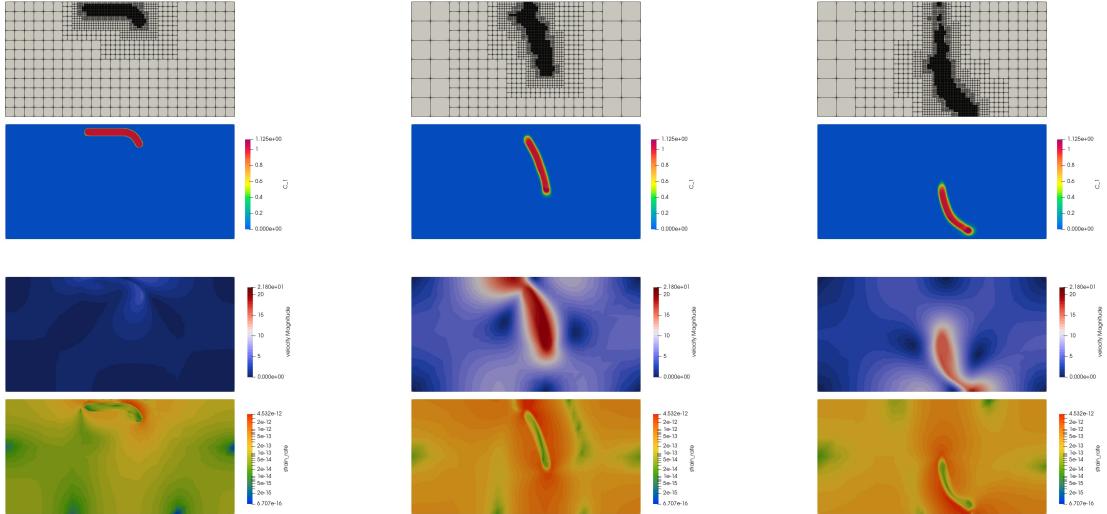
Mesh composed of 31,765 triangles. Other parameters: $\theta_0 = 60^\circ$, $\eta_1 = 10^{21}$, $\gamma = 100$, $L_x = 3000\text{km}$, $L_y = 1500\text{km}$, $\delta\rho = 100$, $L = 400\text{km}$, $h = 100\text{km}$, $d = 50\text{km}$. Note that the bottom boundary is open.

We have also run this model with ASPECT for the case where free slip boundary conditions are prescribed on all sides. Velocities on the midsurface are reported hereunder alongside those obtained with fieldstone and the BEM method.

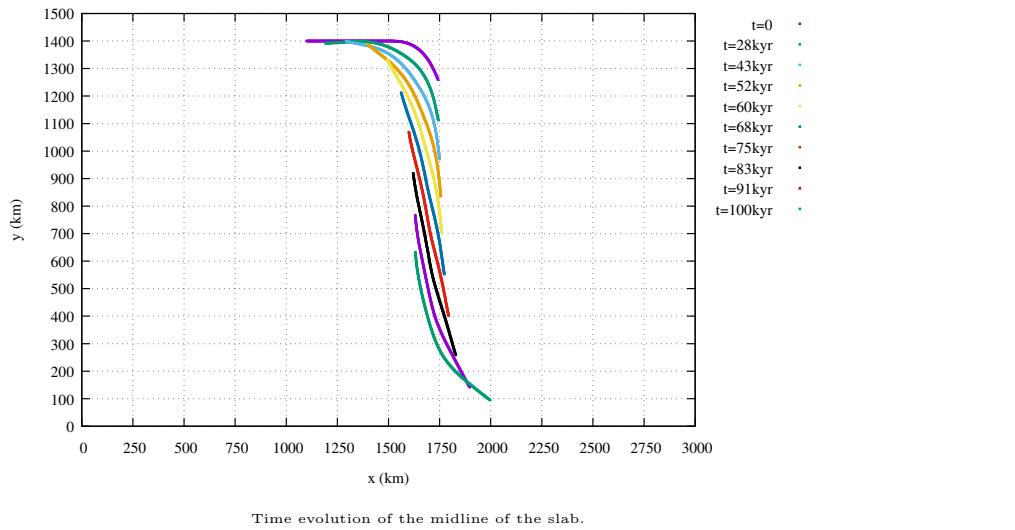


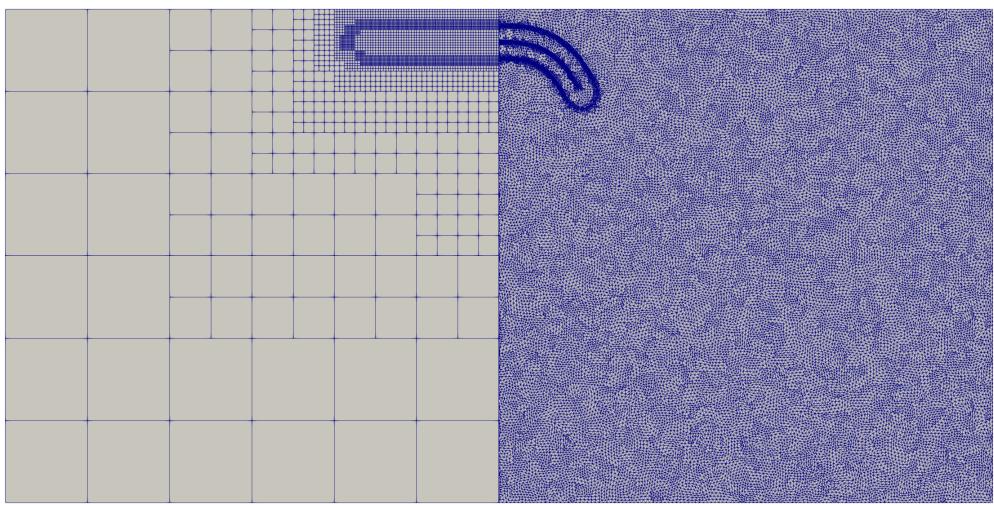
Top row: midsurface velocity measurements. Bottom row: slab/plate perimeter velocity measurements. 'open' means no b.c. on sides and bottom; 'f.s. sides' means free slip b.c. on left and right sides, open at the bottom; 'f.s. all' means free slip b.c. on sides and bottom.

Finally, we have run this experiment over time with ASPECT (unfortunately viscosities were 100 times too small so that the times should be multiplied by 100):



Left: $t=0$, middle $t=64\text{kyr}$, right: $t=100\text{kyr}$.





Literature: [641]

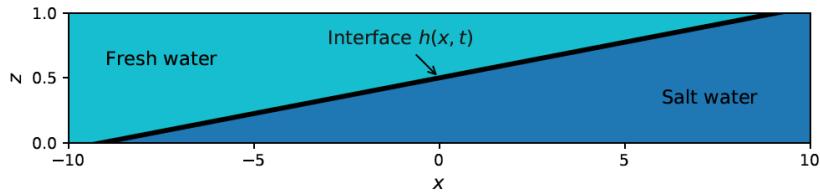
Stone 56: Dynamics of the Salt Water - Fresh Water Interface

Saltwater intrusion in coastal aquifers is a world wide observable phenomena. The mixing of salt and fresh water reduces groundwater quality. It potentially threatens the usability as drinking water. Extensive pumping causing a decrease of groundwater levels amplifies this effect.

Gravity drives salty sea water land inward due to its higher density if an aquifer's water table is lower than sea level. An interface develops between salt and fresh water, which is stable due to the density difference. Along the interface slight mixing occurs due to diffusion. As for most processes in nature, coupled partial differential equations describe the movement of the interface between salt and fresh water.

The target of this exercise is to model a simplified form of the process. Basic assumption hereby are: (i) groundwater flow is mostly horizontal (called *Deput assumption*); (ii) the salt and fresh water interface is sharp, meaning we neglect mixing due to dispersion and diffusion.

We investigate a portion of an aquifer of length L and depth H . Within the aquifer, an interface separates salt water and fresh water as illustrated here under:



The interface $h(x, t)$ is a function of the horizontal location x and time t . For simplicity, we assume an aquifer thickness of $H = 1$. Thus, the values for the position of the interface as a function in the z direction range between zero and one: $h(x, t) \in [0, 1]$.

The differential equation describing the movement of the interface $h(x, t)$ is given by

$$\frac{\partial h(x, t)}{\partial t} = \Gamma \frac{\partial}{\partial x} \left(\frac{h(1-h)\partial h/\partial x}{1 + (\partial h/\partial x)^2} \right) \quad (809)$$

The constant $\Gamma = \frac{\kappa}{\mu g(\rho_{\text{salt}} - \rho_{\text{fresh}})}$ summarizes the physical properties: of the aquifer (permeability κ), of the fluids (viscosity μ , density of salt water ρ_{salt} and density of fresh water ρ_{fresh}) and the gravity constant g .

The differential equation has the character of a non-linear diffusion equation with a non-constant diffusion coefficient $D(h)$:

$$\frac{\partial h}{\partial t} = \Gamma \frac{\partial}{\partial x} \left(D(h) \frac{\partial h}{\partial x} \right) \quad D(h(x, t)) = \frac{h(1-h)}{1 + (\partial h/\partial x)^2} \quad (810)$$

According to the value range of $h \in [0, 1]$, D has the property of $D(0) = D(1) = 0$ (this differential equation is therefore called *degenerate*).

Despite the complexity of the initial differential equation, an analytical solution can be postulated in the form of:

$$h(x, t) = a(t) \cdot x + 0.5 \quad (811)$$

The interface h is a linear function in x with a time-dependent slope $a(t)$. It rotates with time around the fix point $(0; 0.5)$. The slope $a(t)$ can be determined by the ordinary differential equation (REF??):

$$\frac{da(t)}{dt} = -\frac{2a^3}{1 + a^2} \quad (812)$$

The value range of a is limited to $0 < a(t) < 1$ corresponding to a maximal slope angle of 45° . The differential equation in $a(t)$ has an implicit solution:

$$t = \frac{1}{2} \ln \left(\frac{a(0)}{a(t)} \right) + \frac{1}{4} \left(\frac{1}{a(t)^2} - \frac{1}{a(0)^2} \right) \quad (813)$$

We aim to implement a numerical solution of the differential equation 1 which allows us to calculate the location of the interface $h(x, t)$ at specified locations x and times t .

To Do in 2D only

Stone 57: 1D diffusion with DG-FEM

opla

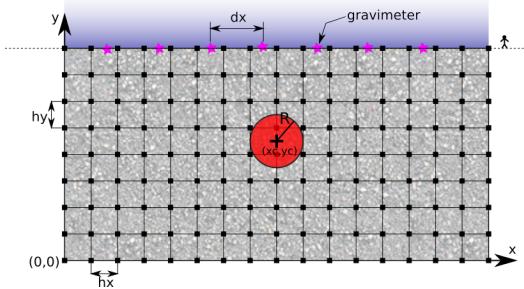
10 fieldstone: Gravity: buried sphere

Before you proceed further, please read :

http://en.wikipedia.org/wiki/Gravity_anomaly

<http://en.wikipedia.org/wiki/Gravimeter>

Let us consider a vertical domain $Lx \times Ly$ where $L_x = 1000\text{km}$ and $L_y = 500\text{km}$. This domain is discretised by means of a grid which counts $nnp = nnx \times nny$ nodes. This grid then counts $nel = nelx \times nely = (nnx - 1) \times (nny - 1)$ cells. The horizontal spacing between nodes is hx and the vertical spacing is hy .



Assume that this domain is filled with a rock type which mass density is given by $\rho_{medium} = 3000\text{kg/m}^3$, and that there is a circular inclusion of another rock type ($\rho_{sphere} = 3200\text{kg/m}^3$) at location $(xsphere, ysphere)$ of radius $rsphere$. The density in the system is then given by

$$\rho(x, y) = \begin{cases} \rho_{sphere} & \text{inside the circle} \\ \rho_{medium} & \text{outside the circle} \end{cases}$$

Let us now assume that we place $nsurf$ gravimeters at the surface of the model. These are placed equidistantly between coordinates $x = 0$ and coordinates $x = Lx$. We will use the arrays $xsurf$ and $ysurf$ to store the coordinates of these locations. The spacing between the gravimeters is $\delta_x = Lx/(nsurf - 1)$.

At any given point (x_i, y_i) in a 2D space, one can show that the gravity anomaly due to the presence of a circular inclusion can be computed as follows:

$$g(x_i, y_i) = 2\pi G(\rho_{sphere} - \rho_0)R^2 \frac{y_i - ysphere}{(x_i - xsphere)^2 + (y_i - ysphere)^2} \quad (814)$$

where r_{sphere} is the radius of the inclusion, $(xsphere, ysphere)$ are the coordinates of the center of the inclusion, and ρ_0 is a reference density.

However, the general formula to compute the gravity anomaly at a given point (x_i, y_i) in space due to a density anomaly of any shape is given by:

$$g(x_i, y_i) = 2G \int \int_{\Omega} \frac{\Delta\rho(x, y)(y - y_i)}{(x - x_i)^2 + (y - y_i)^2} dxdy \quad (815)$$

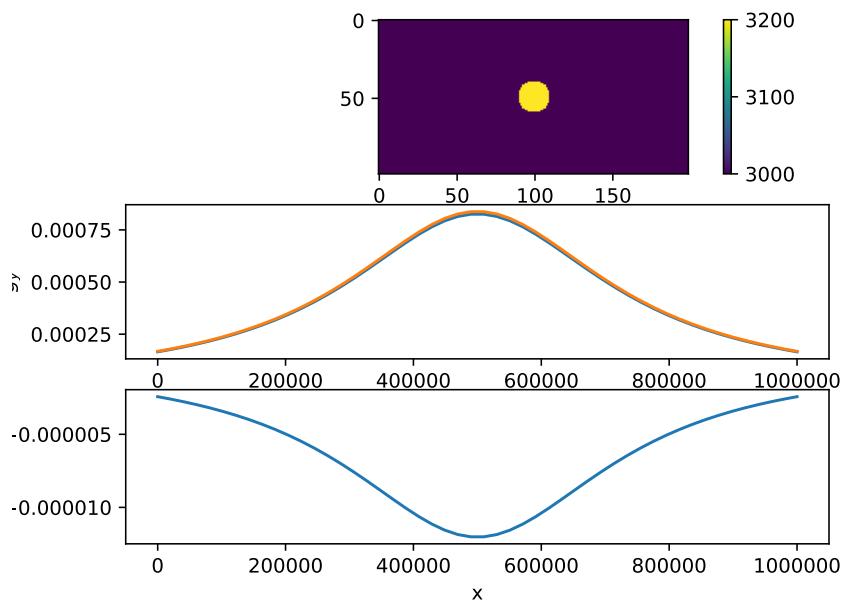
where Ω is the area of the domain on which the integration is to be carried out. Furthermore the density anomaly can be written : $\Delta\rho(x, y) = \rho(x, y) - \rho_0$. We can then carry out the integration for each cell and sum their contributions:

$$g(x_i, y_i) = 2G \sum_{ic=1}^{nel} \int \int_{\Omega_e} \frac{(\rho(x, y) - \rho_0)(y - y_i)}{(x - x_i)^2 + (y - y_i)^2} dxdy \quad (816)$$

where Ω_e is now the area of a single cell. Finally, one can assume the density to be constant within each cell so that $\rho(x, y) \rightarrow \rho(ic)$ and $\int \int_{\Omega_e} dxdy \rightarrow hx \times hy$ and then

$$g(x_i, y_i) = 2G \sum_{ic=1}^{nel} \frac{(\rho(ic) - \rho_0)(y(ic) - y_i)}{(x(ic) - x_i)^2 + (y(ic) - y_i)^2} s_x s_y \quad (817)$$

We will then use the array $gsurf$ to store the value of the gravity anomaly measured at each gravimeter at the surface.



To go further

- explore the effect of the size of the inclusion on the gravity profile.
- explore the effect of the ρ_0 value.
- explore the effect of the grid resolution.
- measure the time that is required to compute the gravity. How does this time vary with nsurf ? how does it vary when the grid resolution is doubled ?
- Assume now that $\rho_2 < \rho_1$. What does the gravity profile look like ?
- what happens when the gravimeters are no more at the surface of the Earth but in a satellite ?
- if you feel brave, redo the whole exercise in 3D...

11 Problems, to do list and projects for students

- Bsc thesis
 - Darcy flow. redo WAFLE (see <http://cedrichieulot.net/wafle.html>)
 - chunk grid
- MSc guided research/thesis
 - surface tension see [1551]p28-29 - see ibuprofem
 - elasticity with markers
 - navier-stokes ? (LUKAS) use dohu matlab code
 - pure shear deformation of inclusions [1792]
 - redo Buck and Sokoutis benchmark for continental convergence [254]
 - redo topo and geoid calculations a la [1060]
 - redo Travis study [1791] which is close to Blankenbach [175]. Note that [1248] looks at kinetic energy for [1791]
 - propagator matrix ? what is it ? [1570] (**MSc Thesis**)  Literature[846, 847, 1574, 2017, 423, 1335, 2013, 1167, 1174, 1374, 1955, 1321, 632, 1219, 258]
 - redo early compressional orogen study by Beaumont [103]
 - redo erosion solutions by [461] (**MSc Thesis**)
 - redo extension 3D Allken papers + [1496, 1027]
- Miscellaneous /to do
 - write about impose bc on el matrix
 - free-slip bc on annulus and sphere . See for example p540 Gresho and Sani book. find book [516]. also check [574] !!
 - constraints [2]
 - formatting of code style
 - Finish nonlinear cavity case5.
 - write about stream functions
 - create stone for layeredflow (see folder one up)
 - in the context of mesh generation on sphere cite [1360]
 - illustrate early boundary fitted static meshes with [1751]
 - [151] spell out the derivation of Jaumann derivative in appendix
 - look at strain-rate softening in [115]
 - write section in features about thermo mechanical simulations and how/why we solve vp, then T.
- carry out critical Rayleigh experiments for various geometries/aspect ratios. Use Arie's notes.
- Indentor/punch with stress b.c. ?
- read in crust 1.0 in 2D on chunk
- compute gravity based on tetrahedra
- NS a la http://ww2.lacan.upc.edu/huerta/exercises/Incompressible/Incompressible_Ex2.htm
- write Scott about matching compressible2 setup with his paper
- including phase changes (w. R. Myhill)

- GEO1442 indenter setup in plane ?
- redo puth17 2 layer experiment
- SIMPLE a la p667 [998]
- implement/monitor div v
- shape fct, trial fct, basis fct vs test fct doc
- write/draw the whole FEM process for a 4x3 grid for compgeo
- lukas' 2D and 3D benchmark
- ROTATING disc
- cylindrical footing on (elasto)-viscous medium - analytical solution, Haskell, etc ...
- check the BASIL code by Houseman et al <http://homepages.see.leeds.ac.uk/~eargah/basil/> on which the ELLE code is based <http://elle.ws/>
- try Anderson acceleration for Uzawa [909] with m=1
- $Q_1^+ \times P_0$ Look at fort81 , rota87b and vadv03
- check [309] for RT0 element use
- deformation around rigid particles [959]
- write about flexural isostasy [1248], bottom Sopale
- mesh containing both quadrilaterals and triangles [27]
- redo/adapt bsc thesis with inversion on stokes sphere
- implementation of fault in FEM codes [2016, 2006]
- https://en.wikipedia.org/wiki/Bernstein_polynomial
- cvi pukp16
- van keken instantane MINI elt
- check $Q_2 \times Q_{-1}$ element, [813] p 697. pressure basis function based at the four 2x2 gauss points.
- look at condition number of \mathbb{K} block for Q1P0 and Q2Q1 as a function of resolution. Insert results in section about why q1p0 should not be used.
- write about splines as shape functions [372]. second or third order shape functions , using extra nodes instead of using more nodes per element. Smaller matrix than Q2 or Q3 but: spline coeffs on nodes are no more unknowns. Plus bc are complicated. Does it work well with visc contrasts ?

`}; iDataArray type=Int32 Name=connectivity .../}; iDataArray type=Int32 Name=offsets .../}; iDataArray type=UInt8 Name=types .../}; i/Cells};`

open questions: what does it mean to have a negative pressure ? should we threshold it when computing yield strength ?

A Three-dimensional applications

In the following table I list many papers which showcase high-resolution models of geodynamical processes (subduction, rifting, mantle flow, plume transport, ...). Given the yearly output of our community and the long list of journal in which research can be disseminated, this list can not be exhaustive.

Ref.	topic	resolution
[187]	Effect of margin curvature on plate deformation in a 3D subduction zones	
[70]	Small-scale sublithospheric convection in the Pacific	$448 \times 56 \times 64$
[1693]	Migration and morphology of subducted slabs in the upper mantle	$50 \times 50 \times 25$
[1516]	Subduction scissor across the South Island of New Zealand	$17 \times 9 \times 9$
[1281]	Influence of a buoyant oceanic plateau on subduction zones	$80 \times 40 \times 80$
[328]	Subduction dynamics, origin of Andean orogeny and the Bolivian orocline	$96 \times 96 \times 64$
[570]	Feedback between rifting and diapirism, ultrahigh-pressure rocks exhumation	$100 \times 64 \times 20$
[19]	Numerical modeling of upper crustal extensional systems	$160 \times 160 \times 12$
[20]	Rift interaction in brittle-ductile coupled systems	$160 \times 160 \times 23$
[1140]	Kinematic interpretation of the 3D shapes of metamorphic core complexes	$67 \times 67 \times 33$
[976]	Role of rheology and slab shape on rapid mantle flow: the Alaska slab edge	$960 \times 648 \times 160$
[327]	Complex mantle flow around heterogeneous subducting oceanic plates	$96 \times 96 \times 64$
[244]	Oblique rifting and continental break-up	$150 \times 50 \times 30$
[150]	Influence of mantle plume head on dynamics of a retreating subduction zone	$80 \times 40 \times 80$
[243]	Rift to break-up evolution of the Gulf of Aden	$83 \times 83 \times 40$
[245]	Thermo-mechanical impact of plume arrival on continental break-up	$100 \times 70 \times 20$
[333]	Subduction and slab breakoff controls on Asian indentation tectonics	$96 \times 96 \times 64$
[590]	Modeling of upper mantle deformation and SKS splitting calculations	$96 \times 64 \times 96$
[1630]	Backarc extension/shortening, slab induced toroidal/poloidal mantle flow	$352 \times 80 \times 64$
[1250]	Sediment transports in the context of oblique subduction modelling	$500 \times 164 \times 100$
[2023]	Crustal growth at active continental margins	$404 \times 164 \times 100$
[1143]	Dynamics of India-Asia collision	$257 \times 257 \times 33$
[1926]	Strain-partitioning in the Himalaya	$256 \times 256 \times 40$
[1186]	Collision of continental corner from 3-D numerical modeling	$500 \times 340 \times 164$
[1509]	Dependence of mid-ocean ridge morphology on spreading rate	$196 \times 196 \times 100$
[589]	Mid mantle seismic anisotropy around subduction zones	$197 \times 197 \times 53$
[881]	Oblique rifting of the Equatorial Atlantic	$120 \times 80 \times 20$
[1347]	Dynamics of continental accretion	$256 \times 96 \times 96$
[1606]	Thrust wedges: infl. of decollement strength on transfer zones	$309 \times 85 \times 149$
[296]	Asymmetric three-dimensional topography over mantle plumes	$500 \times 500 \times 217$
[1762]	modelled crustal systems undergoing orogeny and subjected to surface processes	$96 \times 32 \times 14$
[1194]	Thermo-mechanical modeling of continental rifting and seafloor spreading	$197 \times 197 \times 197$
[149]	Geodynamics of oceanic plateau and plume head accretion	$256 \times 96 \times 96$

B Codes in geodynamics

In what follows I make a quick inventory of the main codes of computational geodynamics, for crust, lithosphere and/or mantle modelling.

in order to find all CIG-codes citations go to: <https://geodynamics.org/cig/news/publications-refbase/>

- ABAQUS

[214] [712] [680] [329] [1124] [1262] [330] [1401] [1462] [1391] [1390]

- ADELI

1997: [876]
2004: [768]
2006: [1875]
2008: [186][187]
2012: [711][757]
2013: [1901]
2014: [345]
2015: [343]
2018: [344][713]

- ASPECT

This code is hosted by CIG at <https://geodynamics.org/cig/software/aspect/>. It is an open source community code based on the finite element library deal.II [40]. It is massively parallel, relies on the p4est library for adaptive mesh refinement, uses the Trilinos solver library, and can deal with 2D and 3D geometries.

2012: [1119]
2015: [50][1784]
2016: [472][697][1997]
2017: [880][470][882][1599][1600][49][1758][233] [1433][1730][1993]
2018: [471][1435][766][893][698][1461][1506][232]
2019: [93][1695][434][1218][699][474][1420]

- BASIL/ELLE <http://elle.ws/>

[190] [1221]

- BEM

[455] [1027] [1365] [1366] [1369] [1525] [1524] [707]

- CHIC

[1421]

- CitComS and CitComCU

These codes are hosted by CIG at <https://geodynamics.org/cig/software/citcomcu/> and <https://geodynamics.org/cig/software/citcoms/>.

1995: [1357]
1996: [1680][1350]
1997: [1356][285]
1998: [1354][2009][839]
1999: [1359][286][1843][1157]
2000: [2014][834][1156][838]
2001: [160][1162]
2002: [1735]
2003: [1839][425][161][1164][1154][159]
2004: [1677][656][1165][428]
2005: [162][1304][1303][1163]

2006: [110][1478][1733][114][426][657][429] [2005]
 2007: [163][2012][1259][71][1581][1374][424] [1522][933]
 2008: [762][694][2010][913][1160][1214][354] [108][111][1670]
 2009: [1193][39][1995][33][630][289][106] [72][1168][203]
 2010: [273][1862][70][290][1994][156] [975][2026][192][915][869][596]
 2011: [112][1155][1836][1170][1213][68][204] [1798][1425][153]
 2012: [44][976][164][197][954][2018][1678] [903][978][1261][1678][1797]
 2013: [66][206][207][977][1526][1428] [45][427][291][626]
 2014: [625][272][1010][46][1897][1655] [10][1245][2027]
 2015: [67][205][198][1659][473][1837][1521] [1904][1896][1895][874][1743][1171][1743] [69][1373]
 2016: [1925][1924][974][666][1589][178][1287]
 2017: [9][1244][655][879][1239][756][1739]
 2018: [898][1061][1011][1898][931][1180]
 2019: [1286][668][1243][1238][1265][177][1158] [1632]

cross check with CIG database

- CONMAN This code is hosted by CIG at <https://geodynamics.org/cig/software/conman/> [1066][1059] [968] [1063] [1064] [969] [113][1157] [1161] [606] [1083] [1400] [487] [1065]
- CONVRS
[1979] [1977]
- DOUAR
[221] [1761] [1964] [223] [1226] [1388] [1926] [1405] [1104]
- DYNEARTHSOL
[366] [1224]
- ELMER Elmer is an open source multiphysical simulation software mainly developed by CSC - IT Center for Science (CSC). Elmer development was started 1995 in collaboration with Finnish Universities, research institutes and industry. Elmer includes physical models of fluid dynamics, structural mechanics, electromagnetics, heat transfer and acoustics, for example. These are described by partial differential equations which Elmer solves by the Finite Element Method (FEM). <https://www.csc.fi/web/elmer>
[1249]
- LaCoDe
[508]
- M-DOODS, Duretz code
[1965] [1963] [359][554]
- FENICS
[16]
- GAIA
[1785] [957] [1407]
- GALE
This code is hosted by CIG at <https://geodynamics.org/cig/software/gale/> [611] [798] [152] [459] [1140][1185] [45]
- (G)TECTON
1980: [1309]
1981: [1308]

1993: [792]
 1995: [793]
 1996: [837][761]
 1999: [794][682]
 2001: [263][797]
 2002: [264]
 2005: [795][1861][1863][795]
 2006: [645][1220][1637]
 2007: [1864]
 2008: [506][505]
 2009: [1127][1485]
 2010: [1815][1484]
 2011: [58][59]
 2013: [1486][1907]
 2014: [1816]
 2015: [1268][1411]
 2016: [704][1269]
 2017: [1443]
 2018: [796][1412][884]

- ELEFANT

[1784] [1242] [268] [1135] [1758] [1488] [1947] [650]

- ELLIPSIS Section 3 of [1522] presents the evolutionary path which lead to this code.

[1349][1931][1378] [1930] [1434] [1353][746][557][1437] [1436] [1516] [1170] [1166] [1568]

- FANTOM

[1756] [19] [20] [21] [581] [1762] [579] [1617] [580]

- FDCON

[578] [442] [670] [669]

- FLUIDITY [488] [693]

- geoFLAC (based on PARAVOZ) [981]

- IFISS Incompressible Flow Iterative Solution Solver is a MATLAB package that is a very useful tool for people interested in learning about solving PDEs. IFIGS includes built-in software for 2D versions of: the Poisson equation, the convection-diffusion equation, the Stokes equations and the Navier-Stokes equations.

<https://personalpages.manchester.ac.uk/staff/david.silvester/ifiss/>

- the I2(3)E(L)VIS code

2003: [742][743][741]
 2004: [744][745][739][728]
 2005: [282][1267][1703]
 2006: [260][720][781][734]
 2007: [721][780][732][783][785]
 2008: [1638][733][1810][591][2022][283][341][1324][1414][738]
 2009: [735][281][1199][593][1192]
 2010: [718][1413][88][1187]
 2011: [548][550][1188][719][737][174] [1415][776][1200][1459][2025][729]
 2012: [442][547][1988][592][784][1811]
 2013: [1186][1397][1250][1831][1830][2023][558][736] [1274][352][353][546][342][1196][1885] [730][726][1319]
 2014: [549][1509][1606][1884][63][1195][1701][1260] [296][782][1832][1275][64][775][723][731][759]
 2015: [545][1812][1607][740][1607][1102][894]
 2016: [1101][1251][624][1263]

2017: [1264][1103][1887]

2018: [2020]

2019: [1100][1190]

- I3MG

[589]

- LaMEM

[1638] [1032] [1142] [1288] [1143] [412] [89] [615] [614] [1507] [616] [411] [1036] [?]

- LAPEX2D,LAPEX3D (LAgrangian Particle EXplicit, based on the prototype code PAROVOZ)
[1672][55][1671] [260][56] [1468][1467] [54] [1638] [1673]

- LITMOD [6] [5] [675] [676]

- MARC [1404] [1403]

- MILAMIN

MILAMIN is a finite element method implementation in native MATLAB that is capable of doing one million degrees of freedom per minute on a modern desktop computer. This includes pre-processing, solving, and post-processing. The MILAMIN strategies and package are applicable to a broad class of problems in Earth science. <http://milamin.org/>

2008: [466]

2009: [774]

2010: [1121][1030]

2011: [1966]

2012: [708]

2014: [1001]

2015: [1234][709][1753][1386]

2016: [984][1277]

2018: [553][985][983][438][996][1538][358]

2019: [32][1669][93]

- PARAVOZ/FLAMAR/FLAC

1989: [462][911]

1993: [1490][2000]

1994: [1942]

1996: [868][2001]

1998: [715]

2000: [1137]

2001: [298][292]

2002: [57][395]

2003: [853][714][1814]

2004: [833][716][1788][1772][398][1789]

2005: [297]

2006: [301]

2007: [1961][299]

2008: [1962][1773]

2009: [717][1964][305][1776]

2012: [35][711][828][710]

2013: [1913][647][1774]

2014: [646][690][295]

2015: [1956][1279][703][525]

- PINK3D [1890]

- PLASTI [677]
- pTatin3D: A nice succinct description of the code is given in Appendix B of [1141].
 - 2013: [1470]
 - 2014: [1292]
 - 2015: [1289]
 - 2017: [1141]
 - 2018: [1006]
 - 2019: [1005]
- PyLith
 - [1]
- RHEA
 - [306] [1688] [15] [308]
- SAMOVAR [568]
- SEPRAN
 - 1993: [1819][1824]
 - 1994: [1882][1852]
 - 1995: [1826]
 - 1996: [1825]
 - 2002: [387][1838][1851]
 - 2003: [1855]
 - 2004: [1856][1857][1858]
 - 2005: [1859][1654]
 - 2006: [1203][1204]
 - 2007: [1854][389][213][212][1074]
 - 2008: [1483][211][1076]
 - 2009: [1869]
 - 2010: [1820][1708]
 - 2011: [1846]
 - 2012: [1821][129][360]
 - 2013: [34]
 - 2014: [361][1363]
 - 2015: [1822]
 - 2019: [2004][1823]
- SISTER
 - [1427]
- SLIM3D
 - [1498] [1530] [244] [245] [243] [242] [881] [1099] [394] [247] [60]
- SLOMO
 - [1029] [1037]
- SNAC SNAC (StGermaiN Analysis of Continua) is an updated Lagrangian explicit finite difference code for modelinga finitely deforming elasto-visco-plastic solid in 3D. The code is hosted at <https://geodynamics.org/cig/software/snac/>.
 - [368] [367] [1520] [369]
- SPECFEM3D
 - [1088]

- MICROFEM/SOPALE

1993: [1933]
1994: [1938][97][103]
1995: [678][569]
1996: [101]
1999: [1936][1934]
2000: [1513][102]
2001: [99]
2002: [950][1514]
2003: [947][1866][1940][1518][266][1945]
2004: [100][1515][1512][572][702]
2005: [701][953]
2006: [1511][1657][1449]
2007: [951][464][1346][1658]
2008: [1656][1910][1911][772][265][952]
2009: [1046][98][267][801][1667]
2010: [14][13][806][1517]
2011: [463][317][946]
2012: [807][808][1090][800][979]
2013: [315][357][620][621][787][809] [1075][1417][980][1122]
2014: [771]
2015: [17][316][892]
2016: [1217]
2017: [314]

- STAG3D/STAGYY

1994: [1723]
1995: [1646]
1996: [1724]
1998: [1343][1768]
2002: [605][1715]
2003: [1729]
2004: [1959][1958]
2005: [818][604]
2006: [1328]
2007: [820][819][886]
2008: [520][888][887]
2009: [521][1393][1045]
2010: [522][1392][1362]
2011: [1594][776]
2012: [1593][445][418][523][517][38]
2013: [1605][1716]

2014: [1974][443][1595][414][117][1184]
2015: [118][519]
2016: [1666][444]
2017: [413][1465]
2018: [419][180][416][42][447][449]
2019: [825][43][518][1454][415][448]

- SUBMAR

[1272] [1271] [1585]

- SULEC SULEC is a finite element code that solves the incompressible Navier-Stokes equations for slow creeping flows. The code is developed by Susan Ellis (GNS Sciences, NZ) and Susanne Buitter (NGU). <http://www.geodynamics.no/buitter/sulec.html>

2011: [1527][570]
2012: [262][1746][442][821]
2013: [747]
2014: [748][1528]
2015: [1396]
2016: [2042]
2017: [1398]
2018: [1748]
2019: [571]

- TERRA: The computational grid is based on a projection of the regular icosahedron onto a sphere and successive dyadic refinements [92]. Concentric copies of such spherical layers of nodes build the domain in radial direction.

[90] [763] [1720] [1721] [278] [277][1972] [275] [1718] [1472][1949][788] [1967] [1948] [485] [486] [1833]

- TERRA-NEO:

[1923]

- TerraFERMA

[1944] [1943] [1685] [346] [347]

- YACC

[1620] [1786] [1784] [1782]

- UNDERWORLD 1&2 Section 3 of [1522] presents the evolutionary path which lead to this code.

2006: [1692][1351]
2007: [1353][1629][1522]
2008: [1150][1444][798][1690]
2010: [334][1281][1693][1691][608][335]
2011: [1312][328]
2012: [327]
2013: [150][1630][590][333][432]
2014: [607]
2015: [1521][149][1631][1661]
2016: [1662][1445][1067]
2017: [94][1068][1624]
2018: [1311][1971][95]
2019: [1621][1969][332]

- VEMAN [151]

C Matrix properties

C.1 Symmetric matrices

Any symmetric matrix has only real eigenvalues, is always diagonalizable, and has orthogonal eigenvectors. A symmetric $N \times N$ real matrix \mathbf{M} is said to be

- **positive definite** if $\vec{x} \cdot \mathbf{M} \cdot \vec{x} > 0$ for every non-zero vector \vec{x} of n real numbers. All the eigenvalues of a Symmetric Positive Definite (SPD) matrix are positive. If A and B are positive definite, then so is $A+B$. The matrix inverse of a positive definite matrix is also positive definite. An SPD matrix has a unique Cholesky decomposition. In other words the matrix \mathbf{M} is positive definite if and only if there exists a unique lower triangular matrix \mathbf{L} , with real and strictly positive diagonal elements, such that $\mathbf{M} = \mathbf{LL}^T$ (the product of a lower triangular matrix and its conjugate transpose). This factorization is called the Cholesky decomposition of \mathbf{M} .
- **positive semi-definite** if $\vec{x} \cdot \mathbf{M} \cdot \vec{x} \geq 0$
- **negative definite** if $\vec{x} \cdot \mathbf{M} \cdot \vec{x} < 0$
- **negative semi-definite** if $\vec{x} \cdot \mathbf{M} \cdot \vec{x} \leq 0$

The Stokes linear system

$$\begin{pmatrix} \mathbb{K} & \mathbb{G} \\ \mathbb{G}^T & 0 \end{pmatrix} \cdot \begin{pmatrix} \mathbf{v} \\ \mathbf{p} \end{pmatrix} = \begin{pmatrix} \mathbf{f} \\ \mathbf{g} \end{pmatrix}$$

is **indefinite** (i.e. it has positive as well as negative eigenvalues).

A square matrix that is not invertible is called **singular** or degenerate. A square matrix is singular if and only if its determinant is 0. Singular matrices are rare in the sense that if you pick a random square matrix, it will almost surely not be singular.

C.2 Schur complement

From wiki. In linear algebra and the theory of matrices, the Schur complement of a matrix block (i.e., a submatrix within a larger matrix) is defined as follows. Suppose $\mathbf{A}, \mathbf{B}, \mathbf{C}, \mathbf{D}$ are respectively $p \times p$, $p \times q$, $q \times p$ and $q \times q$ matrices, and \mathbf{D} is invertible. Let

$$\mathbf{M} = \begin{pmatrix} \mathbf{A} & \mathbf{B} \\ \mathbf{C} & \mathbf{D} \end{pmatrix}$$

so that \mathbf{M} is a $(p+q) \times (p+q)$ matrix. Then the Schur complement of the block \mathbf{D} of the matrix \mathbf{M} is the $p \times p$ matrix

$$\mathbf{S} = \mathbf{A} - \mathbf{B} \cdot \mathbf{D}^{-1} \cdot \mathbf{C}$$

Application to solving linear equations: The Schur complement arises naturally in solving a system of linear equations such as

$$\begin{aligned} \mathbf{A} \cdot \vec{x} + \mathbf{B} \cdot \vec{y} &= \vec{f} \\ \mathbf{C} \cdot \vec{x} + \mathbf{D} \cdot \vec{y} &= \vec{g} \end{aligned}$$

where \vec{x}, \vec{f} are p -dimensional vectors, \vec{y}, \vec{g} are q -dimensional vectors, and $\mathbf{A}, \mathbf{B}, \mathbf{C}, \mathbf{D}$ are as above. Multiplying the bottom equation by $\mathbf{B} \cdot \mathbf{D}^{-1}$ and then subtracting from the top equation one obtains

$$(\mathbf{A} - \mathbf{B} \cdot \mathbf{D}^{-1} \cdot \mathbf{C}) \cdot \vec{x} = \vec{f} - \mathbf{B} \cdot \mathbf{D}^{-1} \cdot \vec{g}$$

Thus if one can invert \mathbf{D} as well as the Schur complement of \mathbf{D} , one can solve for \vec{x} , and then by using the equation $\mathbf{C} \cdot \vec{x} + \mathbf{D} \cdot \vec{y} = \vec{g}$ one can solve for \vec{y} . This reduces the problem of inverting a $(p+q) \times (p+q)$ matrix to that of inverting a $p \times p$ matrix and a $q \times q$ matrix. In practice one needs \mathbf{D} to be well-conditioned in order for this algorithm to be numerically accurate.

Considering now the Stokes system:

$$\begin{pmatrix} \mathbb{K} & \mathbb{G} \\ \mathbb{G}^T & -\mathbb{C} \end{pmatrix} \cdot \begin{pmatrix} \vec{v} \\ \vec{p} \end{pmatrix} = \begin{pmatrix} \vec{f} \\ \vec{g} \end{pmatrix}$$

Factorising for \vec{p} we end up with a **velocity-Schur complement**. Solving for \vec{p} in the second equation and inserting the expression for \vec{p} into the first equation we have

$$\mathbb{S}_v \cdot \vec{v} = \vec{f} \quad \text{with} \quad \mathbb{S}_v = \mathbb{K} + \mathbb{G} \cdot \mathbb{C}^{-1} \cdot \mathbb{G}^T$$

Factorising for \vec{v} we get a **pressure-Schur complement**.

$$\mathbb{S}_p \cdot \vec{p} = \mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot \vec{f} \quad \text{with} \quad \mathbb{S}_p = \mathbb{G}^T \cdot \mathbb{K}^{-1} \cdot \mathbb{G} + \mathbb{C}$$

D Dont be a hero - unless you have to

What follows was published online on July 17th, 2017 at <https://blogs.egu.eu/divisions/gd/2017/07/19/dont-be-a-hero-unless-you-have-to/> It was written by me and edited by Iris van Zelst, at the time PhD student at ETH Zürich.

In December 2013, I was invited to give a talk about the ASPECT code [1] at the American Geological Union conference in San Francisco. Right after my talk, Prof. Louis Moresi took the stage and gave a talk entitled: *Underworld: What we set out to do, How far did we get, What did we Learn?*

The abstract went as follows:

"Underworld was conceived as a tool for modelling 3D lithospheric deformation coupled with the underlying / surrounding mantle flow. The challenges involved were to find a method capable of representing the complicated, non-linear, history dependent rheology of the near surface as well as being able to model mantle convection, and, simultaneously, to be able to solve the numerical system efficiently. [] The elegance of the method is that it can be completely described in a couple of sentences. However, there are some limitations: it is not obvious how to retain this elegance for unstructured or adaptive meshes, arbitrary element types are not sufficiently well integrated by the simple quadrature approach, and swarms of particles representing volumes are usually an inefficient representation of surfaces."

Aside from the standard numerical modelling jargon, Louis used a term during his talk which I thought at the time had a nice ring to it: hero codes. In short, I believe he meant the codes written essentially by one or two people who at some point in time spent great effort into writing a code (usually choosing a range of applications, a geometry, a number of dimensions, a particular numerical method to solve the relevant PDEs(1), and a tracking method for the various fields of interest).

In the long list of Hero codes, one could cite (in alphabetical order) CITCOM [1], DOUAR [8], FANTOM [2], IELVIS [5], LaMEM [3], pTatin [4], SLIM3D [10], SOPALE [7], StaggYY [6], SULEC [11], Underworld [9], and I apologise to all other heroes out there whom I may have overlooked. And who does not want to be a hero? The Spiderman of geodynamics, the Superwoman of modelling?

Louis' talk echoed my thoughts on two key choices we (computational geodynamicists) are facing: Hero or not, and if yes, what type?

Hero or not?

Speaking from experience, it is an intense source of satisfaction when peer-reviewed published results are obtained with the very code one has painstakingly put together over months, if not years. But is it worth it?

On the one hand, writing one owns code is a source of deep learning, a way to ensure that one understands the tool and knows its limitations, and a way to ensure that the code has the appropriate combination of features which are necessary to answer the research question at hand. On the other hand, it is akin to a journey; a rather long term commitment; a sometimes frustrating endeavour, with no guarantee of success. Let us not deny it many a student has started with one code only to switch to plan B sooner or later. Ultimately, this yields a satisfactory tool with often little to no perennial survival over the 5 year mark, a scarce if at all existent documentation, and almost always not compliant with the growing trend of long term repeatability. Furthermore, the resulting code will probably bear the marks of its not-all-knowing creator in its DNA and is likely not to be optimal nor efficient by modern computational standards.

This brings me to the second choice: elegance & modularity or taylored code & raw performance? Should one develop a code in a very broad framework using as much external libraries as possible or is there still space for true heroism?

It is my opinion that the answer to this question is: both. The current form of heroism no more lies in writing ones own FEM(2)/FDM(3) packages, meshers, or solvers from scratch, but in cleverly taking advantage of state-of-the-art packages such as for example p4est [15] for Adaptive Mesh Refinement, PetSc [13] or Trilinos [14] for solvers, Saint Germain [17] for particle tracking, deal.ii [12] or Fenics [16] for FEM, and sharing their codes through platforms such as svn, bitbucket or github.

In reality, the many different ways of approaching the development or usage of a (new) code is linked to the diversity of individual projects, but ultimately anyone who dares to touch a code (let alone write one) is a hero in his/her own right: although (super-)heroes can be awesome on their own, they often

complete each other, team up and join forces for maximum efficiency. Let us all be heroes, then, and join efforts to serve Science to the best of our abilities.

Abbreviations

- (1) PDE: Partial Differential Equation
- (2) FEM: Finite Element Method
- (3) FDM: Finite Difference Method

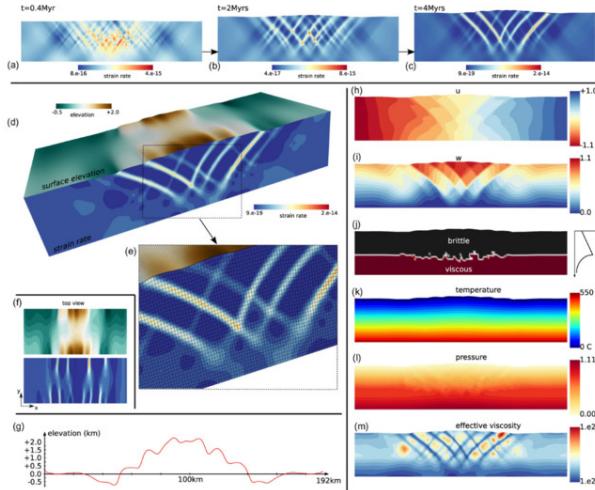
References

- [1] Zhong et al., JGR 105, 2000;
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- [3] Kaus et al., NIC Symposium proceedings, 2016;
- [4] May et al, CMAME 290, 2015
- [5] Gerya and Yuen, PEPI 163, 2007
- [6] Tackley, PEPI 171, 2008
- [7] Fullsack, GJI 120, 1995
- [8] Braun et al., PEPI 171, 2008
- [9] <http://www.underworldcode.org/>
- [10] Popov and Sobolev, PEPI 171, 2008
- [11] <http://www.geodynamics.no/buiter/sulec.html>
- [12] Bangerth et al., J. Numer. Math., 2016; <http://www.dealii.org/>
- [13] <http://www.mcs.anl.gov/petsc/>
- [14] <https://trilinos.org/>
- [15] Burstedde et al., SIAM journal on Scientific Computing, 2011; <http://www.p4est.org/>
- [16] <https://fenicsproject.org/>
- [17] Quenette et al., Proceedings 19th IEEE, 2007

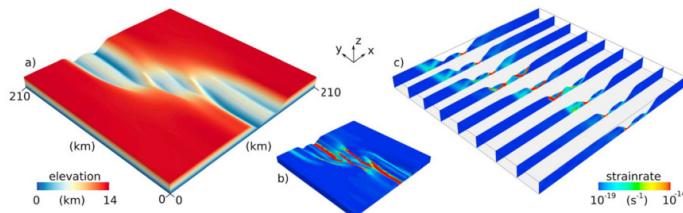
E A FANTOM, an ELEFANT and a GHOST

While a post-doctoral researcher at Bergen University I developed the FANTOM code. Here is what other people and I have published with it:

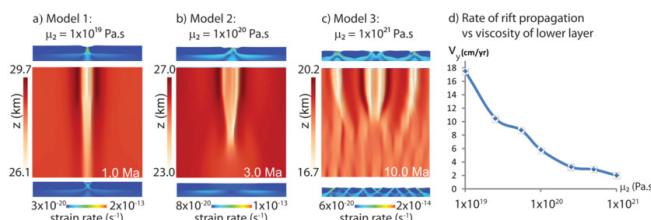
- *FANTOM : two- and three-dimensional numerical modelling of creeping flows for the solution of geological problems*, C. Thieulot, Physics of the Earth and Planetary Interiors, 188, 2011.



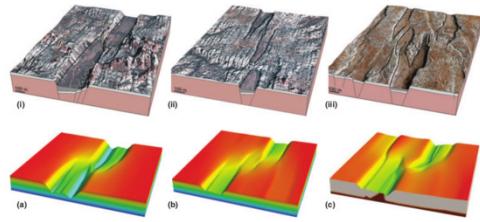
- *Three-dimensional numerical modeling of upper crustal extensional systems*, V. Allken, R.S. Huismans and C. Thieulot, JGR 116, 2011. <https://doi.org/10.1029/2011JB008319>



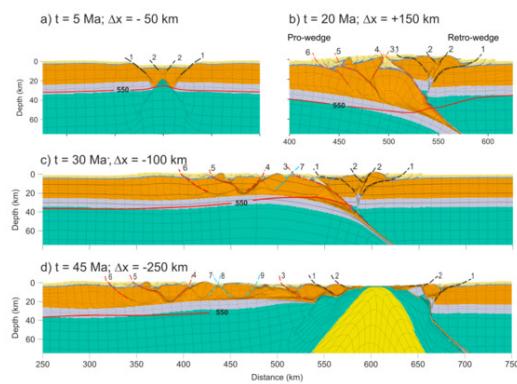
- *Factors controlling the mode of rift interaction in brittle-ductile coupled systems: A 3D numerical study*, V. Allken, R.S. Huismans and C. Thieulot, Geochem. Geophys. Geosyst. 13(5), 2012. <https://doi.org/10.1029/2012GC004077>



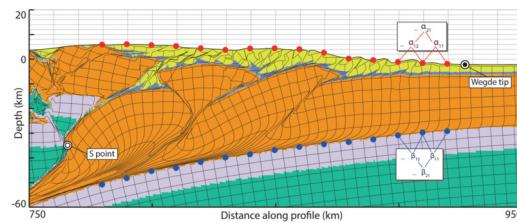
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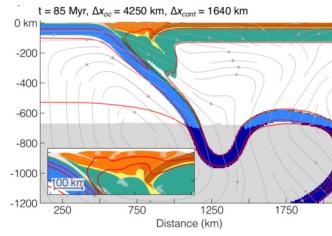
- Three-dimensional numerical simulations of crustal systems undergoing orogeny and subjected to surface processes, C. Thieulot, P. Steer and R.S. Huismans, *Geochem. Geophys. Geosyst.*, 15, 2014. doi:10.1002/2014GC005490
- Extensional inheritance and surface processes as controlling factors of mountain belt structure, Z. Erdős, R.S. Huismans, P. van der Beek, and C. Thieulot, *J. Geophys. Res. Solid Earth*, 119, 2014. doi:10.1002/2014JB011408
- First-order control of syntectonic sedimentation on crustal-scale structure of mountain belts, Z. Erdős, R.S. Huismans, P. van der Beek, *J. Geophys. Res. Solid Earth*, 120, 5362-5377, 2015. doi:10.1002/2014JB011785
- The Wilson Cycle and Effects of Tectonic Structural Inheritance on Rifted Passive Margin Formation, C.A. Salazar-Mora, R.S. Huismans, H. Fossen and M. Egydio-Silva, *Tectonics*, 37, 3085-3101, 2017. doi:10.1029/2018TC004962



- Control of increased sedimentation on orogenic fold-and-thrust belt structure - insights into the evolution of the Western Alps, Z. Erdős, R.S. Huismans and P. van der Beek, *Solid Earth*, 10, 391-404, 2019. <https://doi.org/10.5194/se-10-391-2019>

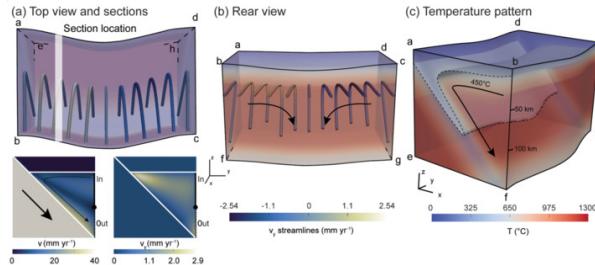


- Mountain building or backarc extension in ocean-continent subduction systems - a function of backarc lithospheric strength and absolute plate velocities, S.G. Wolf and R.S. Huismans, *JGR*, 2019. <https://doi.org/10.1029/2018JB017171>

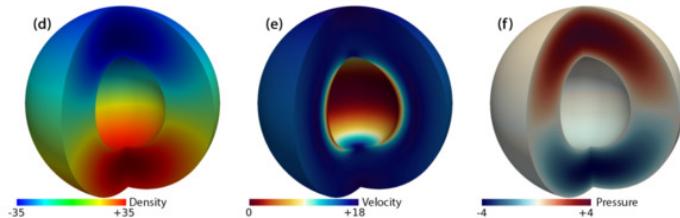


Upon my arrival at Utrecht University in 2012 I started working on a more flexible code, called ELEFANT, which has since very much diverged from FANTOM.

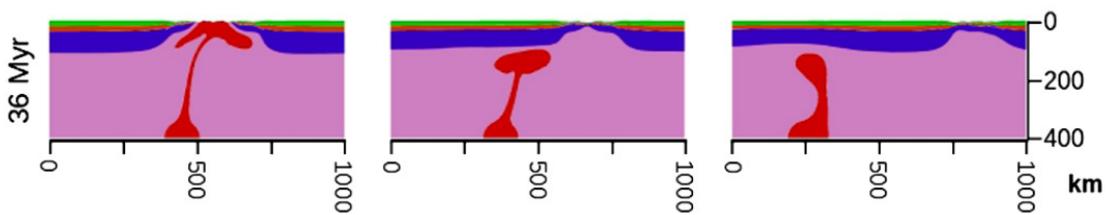
- *The effect of obliquity on temperature in subduction zones: insights from 3-D numerical modeling*, A. Plunder, C. Thieulot and D.J.J. van Hinsbergen, Solid Earth 9, 759-776, 2018. <https://doi.org/10.5194/se-9-759-2018>



- *Analytical solution for viscous incompressible Stokes flow in a spherical shell*, C. Thieulot, Solid Earth 8, 1181-1191, 2017. <https://doi.org/10.5194/se-8-1181-2017>



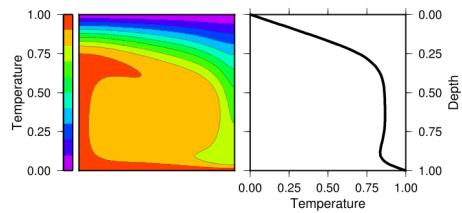
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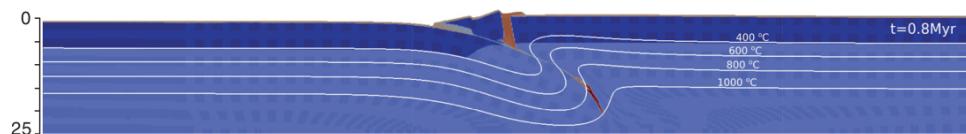
- *Benchmarking numerical models of brittle thrust wedges*, Susanne J.H. Buiter, Guido Schreurs, Markus Albertz, Taras V. Gerya, Boris Kaus, Walter Landry, Laetitia le Pourhiet, Yury Mishin, David L. Egholm, Michele Cooke, Bertrand Maillot, Cedric Thieulot, Tony Crook, Dave May, Pauline Souloumiac, Christopher Beaumont Journal of Structural Geology 92, p140-177, 2016. <https://doi:10.1016/j.jsg.2016.03.003>



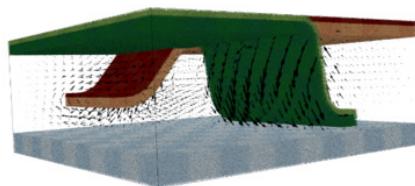
- *A community benchmark for viscoplastic thermal convection in a 2-D square box*, N. Tosi, C. Stein, L. Noack, C. Huettig, P. Maierova, H. Samuel, D.R. Davies, C.R. Wilson, S.C. Kramer, C. Thieulot, A. Glerum, M. Fraters, W. Spakman, A. Rozel, P.J. Tackley, *Geochem. Geophys. Geosyst.* 16, doi:10.1002/2015GC005807, 2015.



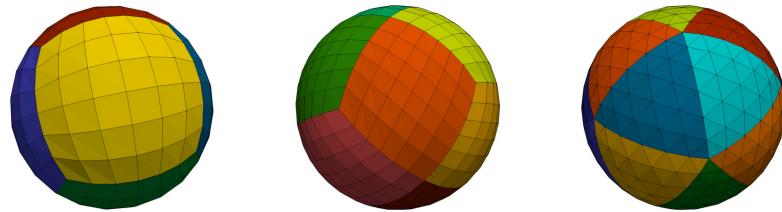
- *Dynamics of intraoceanic subduction initiation: 1. Oceanic detachment fault inversion and the formation of supra-subduction zone ophiolites*, M. Maffione, C. Thieulot, D.J.J. van Hinsbergen, A. Morris, O. Pluempfer and W. Spakman, *Geochem. Geophys. Geosyst.* 16, p1753-1770, 2015.



- *The Geodynamic World Builder: a solution for complex initial conditions in numerical modelling*, M. Fraters, C. Thieulot, A. van den Berg and W. Spakman, *Solid Earth*, <https://doi.org/10.5194/se-2019-24>, 2019.



- *GHOST: Geoscientific Hollow Sphere Tessellation*, C. Thieulot, *Solid Earth*, 9, 11691177, 2018. <https://doi.org/10.5194/se-9-1169-2018>



F Some useful Python commands

F.1 Sparse matrices

So far, the best way I have found to deal with sparse matrices is to declare the matrix as a 'lil_matrix' (linked list).

```
from scipy.sparse import csr_matrix, lil_matrix  
A_mat = lil_matrix((Nfem,Nfem), dtype=np.float64)
```

One then adds terms to it as if it was a full/dense matrix. Once the assembly is done, the conversion to CSR format is trivial:

```
A_mat=A_mat.tocsr()
```

Finally the solver can be called:

```
sol=sps.linalg.spsolve(A_mat,rhs)
```

F.2 condition number

if the matrix has been declared as `lil_matrix`, first convert it to a dense matrix:

```
A_mat=A_mat.dense()
```

The condition number of the matrix is simply obtained as follows:

```
from numpy import linalg as LA  
print(LA.cond(A_mat))
```

G Some useful maths

G.1 Inverse of a 3x3 matrix

Let us assume we wish to solve the system $\mathbf{A} \cdot \vec{X} = \vec{b}$, with $\vec{X} = (x, y)$. Then the solution is given by
The solution is given by

$$x = \frac{1}{\det(\mathbf{A})} \begin{vmatrix} b_1 & a_{21} \\ b_2 & a_{22} \end{vmatrix} \quad y = \frac{1}{\det(\mathbf{A})} \begin{vmatrix} a_{11} & b_1 \\ a_{21} & b_2 \end{vmatrix}$$

G.2 Inverse of a 3x3 matrix

Let us consider the 3x3 matrix \mathbf{M}

$$\mathbf{M} = \begin{pmatrix} M_{xx} & M_{xy} & M_{xz} \\ M_{yx} & M_{yy} & M_{yz} \\ M_{zx} & M_{zy} & M_{zz} \end{pmatrix}$$

1. Find $\det(\mathbf{M})$, the determinant of the Matrix \mathbf{M} . The determinant will usually show up in the denominator of the inverse. If the determinant is zero, the matrix won't have an inverse.
2. Find \mathbf{M}^T , the transpose of the matrix. Transposing means reflecting the matrix about the main diagonal.

$$\mathbf{M}^T = \begin{pmatrix} M_{xx} & M_{yx} & M_{zx} \\ M_{xy} & M_{yy} & M_{zy} \\ M_{xz} & M_{yz} & M_{zz} \end{pmatrix}$$

3. Find the determinant of each of the 2x2 minor matrices. For instance $\tilde{M}_{xx} = M_{yy}M_{zz} - M_{yz}M_{zy}$, or $\tilde{M}_{xz} = M_{xy}M_{yz} - M_{xz}M_{yy}$.
4. assemble the $\tilde{\mathbf{M}}$ matrix:

$$\tilde{\mathbf{M}} = \begin{pmatrix} +\tilde{M}_{xx} & -\tilde{M}_{xy} & +\tilde{M}_{xz} \\ -\tilde{M}_{yx} & +\tilde{M}_{yy} & -\tilde{M}_{yz} \\ +\tilde{M}_{zx} & -\tilde{M}_{zy} & +\tilde{M}_{zz} \end{pmatrix}$$

5. the inverse of \mathbf{M} is then given by

$$\mathbf{M}^{-1} = \frac{1}{\det(\mathbf{M})} \tilde{\mathbf{M}}$$

Another approach which of course is equivalent to the above is Cramer's rule. Let us assume we wish to solve the system $\mathbf{A} \cdot \vec{X} = \vec{b}$, with $\vec{X} = (x, y, z)$. Then the solution is given by

$$x = \frac{1}{\det(\mathbf{M})} \begin{vmatrix} b_1 & a_{12} & a_{13} \\ b_2 & a_{22} & a_{23} \\ b_3 & a_{32} & a_{33} \end{vmatrix} \quad y = \frac{1}{\det(\mathbf{M})} \begin{vmatrix} a_{11} & b_1 & a_{13} \\ a_{21} & b_2 & a_{23} \\ a_{31} & b_3 & a_{33} \end{vmatrix} \quad z = \frac{1}{\det(\mathbf{M})} \begin{vmatrix} a_{11} & a_{12} & b_1 \\ a_{21} & a_{22} & b_2 \\ a_{31} & a_{32} & b_3 \end{vmatrix}$$

H Topics in (computational) geodynamics

This is a *very* rough attempt at classifying my somewhat extensive bibliography per theme/topic. It goes without saying that this cannot be extensive and that since I started computational geodynamics myself around 2006 these lists are biased towards the last 2 decades or so. In retrospect, the categories I have chosen could have been subdivided into narrower fields. I understand that having 100+ references for 'subduction' or 'mantle convection' is not particularly useful, but it means that all these papers show up in the bibliography section of this book, and the titles of said papers are then searchable per keyword.

Big review papers - very good for students

- Advances and challenges in geotectonic modelling [302]
- Slab breakoff: A critical appraisal of a geological theory as applied in space and time [696]
- Dynamics and evolution of the deep mantle resulting from thermal, chemical, phase and melting effects [1727]
- Subduction
 - From the trench to the core-mantle boundary [1052]
 - Modeling the subduction dynamics [157]
 - Slab dynamics in the transition zone [156]
 - Future directions in subduction modeling [719]
 - Controls of subduction geometry, location of magmatic arcs, and tectonics of arc and back-arc regions [454]
 - Stagnant slabs in the upper and lower mantle transition region [672]
 - A Review of the Role of Subduction Dynamics for Regional and Global Plate Motions [106]
- Mantle & plates:
 - Mantle convection models featuring plate tectonic behavior [1231]
 - A mantle convection perspective on global tectonics [413]
 - review of Wilson Cycle plate margins [271]
 - Interior dynamics and long term evolution of habitable planets [1717]
 - Supercontinentsuperplume coupling, true polar wander and plume mobility [1193]
 - Mantle dynamics in the Mediterranean [598]
- Mantle structure
 - Thermo-chemical structure of the lower mantle [?]
- Computational geodynamics
 - Modelling plate tectonics and convection in the mantle [1358]
 - Attributes of a community computer code [431]
 - Attributes of a community lithospheric modeling computer code [431]
 - Overview of adaptive finite element analysis in computational geodynamics [1291]
- Extensional systems
 - Fault linkage and relay structures in extensional settings [638]
 - Rifted Margins: State of the Art and Future Challenges [1460]
 - Rifted margin architecture and crustal rheology: Reviewing Iberia-Newfoundland, Central South Atlantic, and South China Sea [249]

- Rheology
 - Rheology and strength of the lithosphere [1072, 303]
 - Rheology of deep upper mantle [1021]
- Miscellaneous
 - Vening Meinesz [1881]
 -
- Analogue modelling: historical outline [1113]; Approaches, scaling, materials and quantification, with an application to subduction experiments [1625]
- Exhumation of (ultra-)high-pressure terranes: concepts and mechanisms [1909]
- Paradigms, new and old, for ultra-high-pressure tectonism [844]
- The role of solid-solid phase transitions in mantle convection [594]
- Verification, validation and confirmation of numerical models [1438]
- Experimental modelling of orogenic wedges [805]
- Structure and dynamics of the mantle wedge [1850]
- Mountain building, observations and models of dynamic topography [626, 599]
- Reconciling laboratory and observational models of mantle rheology in geodynamic modelling [1056]
- Controlling parameters, surface expressions and the future directions in delamination modeling [773]
- The structural evolution of the deep continental lithosphere [430]
- What makes computational open source software libraries successful? [76]
- Lithosphere tectonics and thermo-mechanical properties: An integrated modelling approach for Enhanced Geothermal Systems exploration in Europe [399]
- Salt tectonics at passive margins: Geology versus models [241]
- Crustal versus mantle core complexes [240]
- Precambrian geodynamics: concepts and models [723]
- A review of brittle compressional wedge models [262]
- accreted terranes: a compilation of island arcs, oceanic plateaus, submarine ridges, seamounts, and continental fragments [1747]
- Hotspot swells [1062]

Analogue modelling

[528] [1740] [1458][458] [499] [1402] [482][113][600] [1626] [856] [483] [1007][1627] [1645][1775] [400][679] [1475] [484] [1236] [325][1598][1069] [324] [1481] [1333][1628][1387]

Archean tectonics, early Earth

[332][724]

Anisotropy, Lattice/Crystal preferred orientation, SKS splitting

[2000] [1377] [1378] [1379] [114] [424] [108][111] [1425] [590] [589] [1239]

Benchmark, analytical solutions, code comparisons, methodology, numerical methods, theory

1974: [904]
1975: [1893][1894]
1984: [1987]
1989: [175]
1990: [1791]
1993: [1153]
1994: [229]
1995: [230][1357][678]
1996: [2017][1355]
1997: [1580]
1999: [1206][169]
2002: [1377]
2003: [1728][1349][742][743][1729]
2004: [1035][1009][1013][1379]
2005: [1383]
2006: [1034][1367][1434][1382]
2007: [1783][364][1031][1026][1353][721]
2008: [2010][524][1799][1120][1290][715] [1845][887][221][466][354]
2009: [1060][700][1871][1529]
2010: [1030][1032][568][1054]
2011: [550][1813][896][1381][488][1151]
2012: [442][360][1115][1288][710][48]
2013: [366][1043][736]
2014: [1754][1292][1222]
2015: [1149][1604][365][1289]
2016: [555][176]
2017: [1599][1943][1254]
2018: [1311][447]
2019: [393][1218][508][699][650]

Core dynamics, CMB temperature

[866] [1138][777] [1392]

Dataset (gravity, goce, grace, tomography, gps, heat flow) and/or their calculations/use

1977: [1596]
1981: [559]
1990: [1202]
1998: [155]
2003: [1117][1682][1480]
2006: [1272]
2007: [1316][1233]
2008: [2019][2019]
2010: [492] 2012: [878][1562][878][617]
2013: [1563][561][494]
2014: [1452][560][1116]
2015: [199][237]
2016: [1082][1375]
2018: [1448]
2019: [1675]

Computational Structural geology

[623] [3] [1792] [1258] [1257] [661] [1462] [1391] [1390] [1221]

Core complexes

[746] [1773] [1776] [1140] [1463] [240]

Discontinuous Galerkin (DG)

2000: [408]
2002: [340][407]
2003: [410]
2005: [338][405]
2008: [1016][1341][531]
2009: [406][404]
2010: [1409][409][1342]
2011: [706][1410]
2012: [1028][1408][1225, chapt. 31]
2013: [1892]
2015: [1149]
2017: [613][958][880]
2018: [1506]

(role of) elasticity in geodynamics modelling

1984:[1987]
1995:[304]
1996:[868]
1998:[420]
2001:[1868]
2002:[1380][1348]
2003:[935][1914]
2005:[1383]
2006:[1034][1382]
2007:[1031]
2008:[54][684]
2009:[1529]
2010:[151]
2012:[710][1012]
2013:[1901]
2014:[607][641]
2015:[1753]
2016:[65][984][1427][65]
2018:[553]
2019:[1454]

(Geodynamics+) surface processes, erosion, sedimentation, topography evolution

1960: [461]
1992: [96]
1993: [1494][1933]
1994: [928][1095][1092]
1995: [348][1096]
1996: [52][101][1093]
1997: [220][691][85]

1998: [514][1817]
1999: [1936][280][86][1779]
2001: [1991][1805][231][292][439][450]
2002: [1939][1340][692][1928]
2003: [216]
2004: [622][768]
2005: [1133][1941]
2006: [1591][217][188]
2007: [299][1658][1778]
2008: [18][1592]
2009: [1927][1124][1262][1475][500]
2010: [1937][1803][227][226][223][459]
2011: [1590][804]
2012: [1070]
2013: [1876][222][620][621][219][352][1804]
2014: [451][412][581][579]
2015: [1812] 2019: [32]

Geotechnics

[1702] [790] [2021] [789] [791] [302]

Glacier dynamics, ice sheets, ice flow, ice rheology

[1942] [814] [656] [47][657] [1992] [1495] [1172] [961][662] [1114] [1224]

(use of) inverse methods, inversion, adjoint methods, assimilation

[321] [191][1033] [279] [965] [963] [1899] [1954][921] [1903] [749][179] [1177] [180]

Large scale mantle-plate interaction, whole Earth models

[1985] [1232] [426] [1908][769][1688] [15][1593][445] [751] [1255] [931]

Crust/Lithosphere modelling, plate motion, plate stress

1977: [456]
1981: [235]
1984: [1125]
1988: [498][497]
1990: [355]
1991: [362][499]
1992: [1334][300][1057]
1993: [1406][224][802]
1994: [254][97]
1995: [126][228][1063][304][569]
1996: [101][140]
1997: [1766][85][293][1356]
1998: [166][1173][1064][1270][1280][839] [141]
1999: [1934][169][396][682][1359][1157]
2000: [862][1137][1156][838][1161][1466]
2001: [926][109][1033]
2002: [1136][422][57][712]
2003: [1940][1914][741][1814][1866][266][1164]
2004: [1772][716][428]
2005: [854][1930][1297]

2006: [1220][261][56][1038][677][429][1449]
2007: [6][1105][746]
2008: [5][1773][873][284][400][368] [1037][611][367]
2009: [267][1485][1582][281]
2010: [877][608][806][1815][1484]
2011: [1566][369]
2012: [1906][1853][262][1090][784][1797]
2013: [1915][1122][1388][647][1907]
2014: [1010][552][1606][1897][1926][1635] [1249][827][688]
2015: [1896][1753][1268]
2017: [1608][1443][1887][1739]
2018: [1898][1412][95]
2019: [1104][1073][448]

Delamination, edge driven convection

[1063] [1064] [1059] [1345] [772] [1862] [1860] [1122] [78] [94]

Dynamic topography

[1567] [1049] [286] [425] [192] [227] [219][626] [50][1069][495] [1970][49] [1809] [518]

lithospheric stress, intraplate stress

1975: [631]
1976: [1576]
1979: [1578]
1989: [196]
1991: [1951]
1992: [1577][1957][2039][401]
2001: [1696]
2004: [1207]
2005: [1771]
2008: [170][753]
2009: [755][1399]
2010: [151]
2012: [1401][754]
2013: [752]
2014: [1816]

Folding, buckling

[1535] [1536] [1537] [628] [2000] [629] [2001] [663] [1636] [1549] [659] [1547][803] [1548] [658][660] [664]

Geoid

[489][848] [490] [2015][1049] [1996][1572] [1055] [1350] [321] [1060] [903] [799] [1061]

Geothermal Energy

[1564]

Grain size evolution & influence on geodynamics

[218] [503] [1679] [855] [1603] [143] [138] [1755][1808][1462] [346][470][1384] [133][116][1385]

LLSVP, ULVZ

1986: [491]
2007: [886]
2008: [694]
2010: [1697]
2011: [204]
2012: [1698][485][517]
2013: [1181][206]
2014: [272][1184][1781]
2015: [1386][874][1304][519]
2016: [536][1780]
2018: [471][1180]
2019: [890]

Magma transport / melting / two phase flow/ (intra-plate) volcanism / lava flow

1984: [1651][1296]
1986: [1652]
1988: [1650]
1993: [1686]
1995: [172]
1998: [1532]
2001: [144]
2003: [142]
2006: [1434]
2007: [1687][1346]
2008: [888][885]
2009: [72]
2010: [70][871]
2011: [68][2024][2025][153]
2012: [1965]
2013: [1043][1331]
2015: [1808][1373]
2016: [1044]
2017: [526]
2018: [1230]
2019: [474]

Mantle convection/dynamics, whole Earth models, plate interaction

1975: [897]
1976: [1299]
1978: [1278][929]
1979: [1237][311]
1980: [1429][988][1649]
1981: [312][973]
1982: [989][925]
1983: [927][924][373][1298]
1984: [1430][990][863][872][489][193][374] [1371][627]
1985: [991][91][378][460]
1986: [490][836][1519]
1987: [1986]
1989: [137][1640]
1990: [1791]

1991: [375][1240][687]
1992: [136][1252][1842]
1993: [1996][992][1713][313][322][1824][1720]
1994: [865][380][1721][1959][1958]
1995: [2006][1826][278][1544][139]
1996: [1999][916][1611][1612][1724][1800] [171][276][1015][837][1825][1546][1545]
1997: [917][970][277][1356]
1998: [28][971][511][1725][1719][1793] [1794][275]
1999: [542][1843][482][1718]
2000: [12][864][509][1714][2014][1152]
2001: [1841][1575][1162]
2002: [1711][1348][1715]
2003: [855][1154][1378][606][84]
2004: [1857]
2005: [1722][318][818][1163][1083]
2006: [110]
2007: [1400][1284][213][212][820][819][933]
2008: [1726][363][211][520][1483][913]
2009: [1949][630][788][521]
2010: [273][522][1967][1392][915]
2011: [1231][1594][1948][1155][112]
2012: [154][417][523][1678]
2013: [914][486][1786][206][308][1329] [599][418][432][427]
2014: [46][891][443][625][1595][414][117]
2015: [1753][1923][118]
2016: [666][1666][178]
2017: [756]
2018: [416][42][419]
2019: [825][1265][415][1158]

Mantle rheology, phase transitions, stratification, (temperature) profile

1982: [1984][376]
1985: [381]
1992: [2003]
1993: [1720][145]
1994: [1852]
1995: [1998][377][1646]
1996: [1457]
1997: [1325]
1998: [321]
2005: [889]
2007: [1455][1374]
2009: [1393]
2010: [1025]
2011: [972][609]
2012: [1727][1620]
2013: [603][1716]
2015: [69]
2016: [1777][128]

Mantle wedge

1978: [1790]
2001: [160][856]
2003: [1850]

2006: [781][734]
2007: [780][1074]
2008: [1076][341]
2009: [1145]
2010: [1585][919]
2011: [2025]
2014: [1883][1245]
2018: [1488]

Obduction, ophiolites

[842] [843] [7] [545]

Planets - Mars

[1648] [1418] [1165] [1859] [1045] [776] [1605] [1655] [1997]

Planets - Moon

[512] [2004]

Planets - Venus

[1648] [1049] [1680] [1241][1354][1473] [1859] [38] [759] [731] [446][481] [1061]

Plate motion and mantle, plate tectonic reconstruction

[1301] [2029] [2009] [588][1565] [1193][1827] [955][840][1525] [1368] [1980] [1745]

Plume dynamics

1994: [1395][867]
1997: [1844]
1998: [1768]
1999: [1131]
2002: [605][483]
2006: [966][1203][1204][2005][1328][1422][1523]
2009: [1869][203]
2011: [1785][1736][288][1312][484][1798]
2013: [480][1283][29]
2015: [473][874]
2017: [1993]
2018: [479]
2019: [43]

Plume-Lithosphere interaction

1994: [1571]
1996: [2008]
1998: [155][1343]
1999: [1344]
2001: [1868]
2002: [640]
2003: [1839]
2005: [297][604]
2008: [1810][1670]
2009: [305][2022][70]

2010: [596]
2011: [1673]
2013: [150][245]
2014: [296][731][271][270]
2015: [149][740][1102][1313]
2016: [624][697][1101]
2017: [80][233][80][130][1103]
2018: [471]
2019: [1100]

Regenauer-Lieb

2000: [1555]
2003: [1556]
2004: [1561]
2006: [1558][1554]
2008: [1560]
2009: [1559]
2013: [1553]

Rheology, material parameters, rock mechanics

1951: [538][845]
1952: [539]
1968: [320]
1969: [858]
1972: [337]
1974: [1087]
1979: [770][587]
1980: [208]
1981: [513]
1984: [1543][371][1874]
1986: [1022]
1987: [1071]
1990: [1932]
1992: [61][370][1019][1084]
1993: [1020]
1994: [649]
1995: [1086][764]
1996: [1900][906]
1997: [584, 583] 1998: [420][1241]
1999: [1014]
2000: [1609][1541][1305, 1306]
2001: [1223]
2002: [908]
2003: [907][1018][1339]
2005: [527][540]
2006: [1610][301][1351]
2007: [900][1085][610]
2008: [1150][287][1106][760]
2009: [1024][1040]
2011: [1144][1041]
2012: [1569]
2013: [1139][1318]
2015: [365]
2017: [184]

Rifting, seafloor spreading, ridges, pull-apart basins, extension

1973: [667]
1980: [234]
1984: [1496]
1985: [194]
1986: [923][2041][2040][1361]
1988: [253]
1989: [1310]
1990: [601][1201]
1991: [1795][255]
1992: [2028][563]
1993: [792]
1994: [1802]
1995: [793][1027]
1996: [544][127][1285]
1998: [1533]
1999: [239][256][794]
2001: [949][948][652][654][653]
2002: [950][860][478][1865][115]
2003: [947][861][437][1931]
2004: [899]
2005: [953][441][1861][1863]
2006: [1775][436][452][1468]
2007: [938][1276][1864][557][951]
2008: [435][826][265][952][1467]
2009: [8][1048]
2010: [51][722][639][718]
2011: [19][570][946]
2012: [20][244]
2013: [21][243][357][1075][1047][1322][1913] [726][1196][730]
2014: [881][1195][242][1099][562][1509][1194]
2015: [1396][394]
2016: [1427][994][2042]
2017: [1141][247][130][1398]
2018: [358][250][246][1748][995][1617][1464]
2019: [1178][2043][32][554][1333]

Salt tectonics

1978: [1946]
1991: [1731]
1992: [1990]
1993: [1402][1848][1918]
1996: [1293]
1998: [758]
2004: [964][702]
2006: [1282]
2007: [936][1282]
2009: [801]
2011: [241]
2012: [619][1185]
2013: [787]
2014: [89][615][614][748][1416]
2015: [616]
2016: [1269]

H.1 Seismo-tectonics, subduction earthquakes

[185] [1831, 1830][1319] [1832] [894] [796] [884]

Subduction

1980: [1309]
1982: [454]
1985: [1751]
1986: [986]
1989: [196]
1990: [930][1058]
1992: [2015][1929][835]
1993: [1003][556][1933]
1994: [2016][1938][1916][1917][103]
1995: [1273]
1996: [379][761]
1997: [876][1053]
1998: [969][269][252]
1999: [857][113][286]
2000: [1749][215][102]
2001: [298][263][349]
2002: [387][395][727]
2003: [1552][680][714][159]
2004: [1788][202][100][1789]
2005: [987][1128][795][578][36][795] [1267][1703][1627]
2006: [645][1602][1272][720][681][1478]
2007: [37][1961][464][389][1271][329] [783][785][1259][1365][1437]
2008: [1962][1444][762][1910][1911][187] [1324][186][54][591][733][679] [506][505][738][1414][1656]
2009: [1964][158][595][98][1046][717][735]
[593][1192][1366][1227][106]
2010: [852][1226][1281][330][334][1187][1691] [1369][2026][1530][1362]
2011: [1188][1213][317][59][328][737][1527]
[174][729]
2012: [35][979][976][978][1988][807][808]
[1587][1746][1752][197][388][331][327]
[708][1191][129][1811][164]
2013: [1397][844][34][1394][1977][2023][1186] [977][962][1236][541][1774][315][1630] [673][1250][1389][1317][809][1831][1830]
[342][1885][1526][291]
2014: [1586][918][1588][1347][607][641][646]
[693][1886][1884][1189][202][63][1701]
[64][1832]
2015: [149][198][205][343][1065][1837][495]
2016: [1782][841][1589][1287]
2017: [1068][1624][1465][1887][1239]
2018: [1971][449]
2019: [1243][1286][1632]

Subduction - slab detachment, break-off

1985: [671]
1992: [1952]
1995: [1981][1888][493]
1997: [1950]
1998: [502][336]
2000: [1953]
2002: [264]

2004: [744]
2006: [597]
2008: [2038]
2009: [33][289]
2010: [290][88]
2011: [548][1836][1633]
2012: [547][551]
2013: [333][1246][747][546]
2014: [549][138][1891]
2015: [1890]
2017: [655][1244]
2018: [696]
2019: [134]

Subduction + water (fluids)

[1851] [737] [592] [1528][1245] [346] [347]

Subduction/plate tectonics initiation

1978: [168]
1982: [402]
1984: [339]
1989: [397]
1991: [1376]
1998: [1787]
1999: [600]
2000: [1513]
2001: [530][1557][214]
2004: [1700][833]
2003: [853]
2005: [162]
2007: [1105]
2008: [1810]
2010: [1413][294]
2011: [58][1415]
2013: [558][1274][1108]
2014: [1568][1275]
2015: [1242][1463][1835]
2016: [444]
2018: [2020][60]
2019: [104][829]

Subduction - flat slabs - low angle subduction

2011: [463]
2012: [1261]
2015: [709][1743]

Teaching

[1042]

Tectonics, small deformation, rock mechanics

[959] [895] [586] [1166]

Wilson cycle, supercontinent cycles

[1801] [2012] [1995] [104]

Meshless methods (SPH, RKPM, DEM, FPM, ...)

[1235] [121] [1370] [125] [1183] [1215] [1902] [618][1216] [477] [1502][1123] [1089][1879] [476] [1419] [1118]

Element Free Galerkin Method

[119] [123] [124] [120][433] [122] [1497]

Planetary accretion, exoplanets, planet formation, segregation

[1160] [1199][774] [1200] [775][1974] [1407]

accretionary wedges, nappes

[1704][496] [467][469] [468] [1095] [1945] [1606] [1263] [1264][1608] [571]

Thrust-wrench fault

[1598]

Transpressional systems

[1769] [1766] [1097][1814] [1148] [1391][1608] [1390]

H.2 Urey ratio

[1107]

intrusions, diapirism, Rayleigh-Taylor instability

1972: [148]

1975: [528]

1981: [235]

1988: [1641]

1992: [1842][1990][1919][1922][1872]

1993: [1849][1402][1489][1494][1848][1490][1493][1920]

1995: [1921][172][1921]

2001: [1033]

2003: [741][1855]

2004: [739][964][728]

2007: [732]

2008: [283][2038][1335]

2011: [570][1459]

2013: [670]

2014: [614]

2015: [616][669]

Visualization, rendering

[602] [354] [1690] [447]

Locations

- South America, Andes [1917] [55][1671] [56] [1589]
- North America [285]
- Alps [1469] [1480] [282] [1877] [1236] [1634]
- Mediterranean region [1479] [1480] [192]
- New Zealand [1094] [228] [101] [86] [1220] [712][1514] [714][1097][1814] [1516] [821]
- Zagros [1875] [877] [1966] [1417] [646] [748] [1608]
- Himalayan region, Tibetan plateau, India collision [168] [1740] [1878] [1458] [1337] [1338] [1938] [348] [1601] [1829] [356][391] [99][1134][1991][1742] [1098] [1565] [100] [390][1583][1584] [392] [284] [877] [112] [333][1388][352][353] [1926] [314] [1481] [1628][1632]
- Pyrenees [758] [102] [1294] [982] [1880] [621] [554]
- Caribbean
- East mediterranean - Aegean region, Turkey [335]
- Ethiopian and Afar rift [1316] [435] [1048] [152] [1471] [247] [248] [434]
- Alaskan region [976] [977] [879] [1320]
- Farallon plate [1214]
- Japan [1051]
- Tonga-Kermadec subduction zone [159][161]
- Western United States [114]
- Australian plate [902] [869]

I Elemental mass matrices for simple geometries

In what follows I compute the mass matrix for a variety of reference elements. If you wish to use these in a code, do not forget to take the jacobian of the transformation/mapping into account.

I.1 1D segments

I.1.1 Linear basis functions

Let us start with the mass matrix (which we encountered in Section 5.1 – although we leave the ρC_p term out):

$$\mathbf{M}_e = \int_{\Omega_e} \vec{N}^T \vec{N} dV = \int_{-1}^{+1} \vec{N}^T \vec{N} dr \quad (818)$$

on the reference element, with

$$\vec{N}^T = \begin{pmatrix} N_1(r) \\ N_2(r) \end{pmatrix} = \frac{1}{2} \begin{pmatrix} 1-r \\ 1+r \end{pmatrix}$$

We have

$$\int_{-1}^{+1} N_1(r) N_1(r) dr = 2/3 \quad (819)$$

$$\int_{-1}^{+1} N_1(r) N_2(r) dr = 1/3 \quad (820)$$

$$\int_{-1}^{+1} N_2(r) N_2(r) dr = 2/3 \quad (821)$$

Following the procedure in Section 5.1 we arrive at

$$\mathbf{M}^e = \frac{1}{3} \begin{pmatrix} 2 & 1 \\ 1 & 2 \end{pmatrix}$$

The lumped mass matrix is then

$$\bar{\mathbf{M}}^e = \frac{1}{3} \begin{pmatrix} 2+1 & 0 \\ 0 & 1+2 \end{pmatrix} = \begin{pmatrix} 1 & 0 \\ 0 & 1 \end{pmatrix} \quad (822)$$

Remark. The sum of all the terms in the mass matrix must be equal to 2. Indeed:

$$\begin{aligned} \sum_{ij} M_{ij} &= \sum_{ij} \int_{-1}^{+1} N_i N_j dr \\ &= \int_{-1}^{+1} (N_1 N_1 + N_1 N_2 + N_2 N_1 + N_2 N_2) dr \\ &= \int_{-1}^{+1} [N_1(N_1 + N_2) + N_2(N_1 + N_2)] dr \\ &= \int_{-1}^{+1} (N_1 + N_2) dr \\ &= 2 \end{aligned}$$

I.1.2 Quadratic basis functions

There are now three nodes in the segment so that the mass matrix is now a 3×3 matrix. We have (see Section 4.4.1)

$$\vec{N}^T(r) = \begin{pmatrix} N_1(r) \\ N_2(r) \\ N_3(r) \end{pmatrix} = \begin{pmatrix} \frac{1}{2}r(r-1) \\ 1-r^2 \\ \frac{1}{2}r(r+1) \end{pmatrix} \quad (823)$$

We then have to compute

$$\begin{aligned}
\int_{-1}^{+1} N_1(r)N_1(r)dr &= \frac{8}{30} = 0.26666 \\
\int_{-1}^{+1} N_1(r)N_2(r)dr &= \frac{4}{30} = 0.13333 \\
\int_{-1}^{+1} N_1(r)N_3(r)dr &= -\frac{2}{30} = -0.06666... \\
\int_{-1}^{+1} N_2(r)N_2(r)dr &= \frac{16}{15} = 1.06666 \\
\int_{-1}^{+1} N_2(r)N_3(r)dr &= \frac{4}{30} = 0.13333 \\
\int_{-1}^{+1} N_3(r)N_3(r)dr &= \frac{8}{30} = 0.26666
\end{aligned}$$

and finally

$$\mathbf{M}^e = \frac{1}{30} \begin{pmatrix} 8 & 4 & -2 \\ 4 & 32 & 4 \\ -2 & 4 & 8 \end{pmatrix} \quad (824)$$

The lumped mass matrix is then

$$\begin{aligned}
\bar{\mathbf{M}}^e &= \frac{1}{30} \begin{pmatrix} 8+4-2 & 0 & 0 \\ 0 & 4+32+4 & -2+4+8 \\ 0 & 0 & -2+4+8 \end{pmatrix} \\
&= \frac{1}{30} \begin{pmatrix} 10 & 0 & 0 \\ 0 & 40 & 0 \\ 0 & 0 & 10 \end{pmatrix} \\
&= \frac{1}{3} \begin{pmatrix} 1 & 0 & 0 \\ 0 & 4 & 0 \\ 0 & 0 & 1 \end{pmatrix}
\end{aligned} \quad (825)$$

We can easily verify that

$$\sum_{ij} M_{ij} = 2 \quad \sum_{ij} \bar{M}_{ij} = 2$$

I.1.3 Cubic basis functions

There are now four nodes in the segment so that the mass matrix is now a 4×4 matrix. We have (see Section 4.4.3)

$$\vec{N}^T(r) = \begin{pmatrix} N_1(r) \\ N_2(r) \\ N_3(r) \\ N_4(r) \end{pmatrix} = \frac{1}{16} \begin{pmatrix} -1+r+9r^2-9r^3 \\ 9-27r-9r^2+27r^3 \\ 9+27r-9r^2-27r^3 \\ -1-r+9r^2+9r^3 \end{pmatrix} \quad (826)$$

$$\begin{aligned}
\int_{-1}^{+1} N_1(r)N_1(r)dr &= \frac{1}{256} \frac{4096}{105} \\
\int_{-1}^{+1} N_1(r)N_2(r)dr &= \frac{1}{256} \frac{1056}{35} \\
\int_{-1}^{+1} N_1(r)N_3(r)dr &= -\frac{1}{256} \frac{384}{35} \\
\int_{-1}^{+1} N_1(r)N_4(r)dr &= \frac{1}{256} \frac{608}{105} \\
\int_{-1}^{+1} N_2(r)N_2(r)dr &= \frac{1}{256} \frac{6912}{35} \\
\int_{-1}^{+1} N_2(r)N_3(r)dr &= -\frac{1}{256} \frac{864}{35} \\
\int_{-1}^{+1} N_2(r)N_4(r)dr &= -\frac{1}{256} \frac{384}{35} \\
\int_{-1}^{+1} N_3(r)N_3(r)dr &= \frac{1}{256} \frac{6912}{35} \\
\int_{-1}^{+1} N_3(r)N_4(r)dr &= \frac{1}{256} \frac{1056}{35} \\
\int_{-1}^{+1} N_4(r)N_4(r)dr &= \frac{1}{256} \frac{4096}{105}
\end{aligned}$$

and finally

$$\mathbf{M}^e = \frac{1}{16} \frac{1}{105} \begin{pmatrix} 256 & 198 & -72 & 38 \\ 198 & 1296 & -162 & -72 \\ -72 & -162 & 1296 & 198 \\ 38 & -72 & 198 & 256 \end{pmatrix} \quad (827)$$

The lumped mass matrix is then

$$\begin{aligned}
\bar{\mathbf{M}}^e &= \frac{1}{16} \frac{1}{105} \begin{pmatrix} 256 + 198 - 72 + 38 & 0 & 0 & 0 \\ 0 & 198 + 1296 - 162 - 72 & 0 & 0 \\ 0 & 0 & -72 - 162 + 1296 + 198 & 0 \\ 0 & 0 & 0 & 38 - 72 + 198 + 256 \end{pmatrix} \\
&= \frac{1}{16} \frac{1}{105} \begin{pmatrix} 420 & 0 & 0 & 0 \\ 0 & 1260 & 0 & 0 \\ 0 & 0 & 1260 & 0 \\ 0 & 0 & 0 & 420 \end{pmatrix} \\
&= \frac{1}{4} \begin{pmatrix} 1 & 0 & 0 & 0 \\ 0 & 3 & 0 & 0 \\ 0 & 0 & 3 & 0 \\ 0 & 0 & 0 & 1 \end{pmatrix}
\end{aligned}$$

We can easily verify that

$$\sum_{ij} M_{ij} = 2 \quad \sum_{ij} \bar{M}_{ij} = 2$$

I.1.4 Quartic basis functions

There are now five nodes in the segment so that the mass matrix is now a 5×5 matrix. We have (see Section 4.4.4)

$$\vec{N}^T(r) = \begin{pmatrix} N_1(r) \\ N_2(r) \\ N_3(r) \\ N_4(r) \\ N_5(r) \end{pmatrix} = \frac{1}{6} \begin{pmatrix} r - r^2 - 4r^3 + 4r^4 \\ -8r + 16r^2 + 8r^3 - 16r^4 \\ 6 - 30r^2 + 24r^4 \\ 8r + 16r^2 - 8r^3 - 16r^4 \\ -r - r^2 + 4r^3 + 4r^4 \end{pmatrix} \quad (828)$$

$$\begin{aligned} \int_{-1}^{+1} N_1(r)N_1(r)dr &= \frac{1}{36} \frac{1168}{315} \\ \int_{-1}^{+1} N_1(r)N_2(r)dr &= \frac{1}{36} \frac{1184}{315} \\ \int_{-1}^{+1} N_1(r)N_3(r)dr &= -\frac{1}{36} \frac{232}{105} \\ \int_{-1}^{+1} N_1(r)N_4(r)dr &= \frac{1}{36} \frac{32}{45} \\ \int_{-1}^{+1} N_1(r)N_5(r)dr &= -\frac{1}{36} \frac{116}{315} \\ \int_{-1}^{+1} N_2(r)N_2(r)dr &= \frac{1}{36} \frac{1024}{45} \\ \int_{-1}^{+1} N_2(r)N_3(r)dr &= -\frac{1}{36} \frac{512}{105} \\ \int_{-1}^{+1} N_2(r)N_4(r)dr &= \frac{1}{36} \frac{1024}{315} \\ \int_{-1}^{+1} N_2(r)N_5(r)dr &= \frac{1}{36} \frac{32}{45} \\ \int_{-1}^{+1} N_3(r)N_3(r)dr &= \frac{1}{36} \frac{832}{35} \\ \int_{-1}^{+1} N_3(r)N_4(r)dr &= -\frac{1}{36} \frac{512}{105} \\ \int_{-1}^{+1} N_3(r)N_5(r)dr &= -\frac{1}{36} \frac{232}{105} \\ \int_{-1}^{+1} N_4(r)N_4(r)dr &= \frac{1}{36} \frac{1024}{45} \\ \int_{-1}^{+1} N_4(r)N_5(r)dr &= \frac{1}{36} \frac{1184}{315} \\ \int_{-1}^{+1} N_5(r)N_5(r)dr &= \frac{1}{36} \frac{1168}{315} \end{aligned} \quad (829)$$

$$M^e = \frac{1}{36} \frac{1}{315} \begin{pmatrix} 1168 & 1184 & -696 & 224 & -116 \\ 1184 & 7168 & -1536 & 1024 & 224 \\ -696 & -1536 & 7488 & -1536 & -696 \\ 224 & 1024 & -1536 & 7168 & 1184 \\ -116 & 224 & -696 & 1184 & 1168 \end{pmatrix} \quad (830)$$

The lumped mass matrix is then

$$\bar{\mathbf{M}}^e = \frac{1}{36} \frac{1}{315} \begin{pmatrix} 1764 & 0 & 0 & 0 & 0 \\ 0 & 8064 & 0 & 0 & 0 \\ 0 & 0 & 3024 & 0 & 0 \\ 0 & 0 & 0 & 8064 & 0 \\ 0 & 0 & 0 & 0 & 1764 \end{pmatrix} = \frac{1}{45} \begin{pmatrix} 7 & 0 & 0 & 0 & 0 \\ 0 & 32 & 0 & 0 & 0 \\ 0 & 0 & 12 & 0 & 0 \\ 0 & 0 & 0 & 32 & 0 \\ 0 & 0 & 0 & 0 & 7 \end{pmatrix} \quad (831)$$

We can once again easily verify that

$$\sum_{ij} M_{ij} = 2 \quad \sum_{ij} \bar{M}_{ij} = 2$$

Note that all the integrals above were done very conveniently with the WolframAlpha software/web-site⁶⁷. Example:

The screenshot shows the WolframAlpha interface. At the top, there is a search bar containing the input: `int_{-1}^1 (r - r^2 - 4r^3 + 4r^4)(r - r^2 - 4r^3 + 4r^4) dr`. Below the search bar are several icons: a star, a square, a grid, and a camera. To the right of the search bar are two buttons: "Browse Examples" and "Surprise Me". In the main result area, it says "Definite integral:" followed by the integral expression. To its right are buttons for "More digits" and "Step-by-step solution" (which is checked). The result of the integral is given as $\int_{-1}^{+1} (r - r^2 - 4r^3 + 4r^4)(r - r^2 - 4r^3 + 4r^4) dr = \frac{1168}{315} \approx 3.7079$. At the bottom right of this area is a link "Open code" with a small icon.

I.2 Quadrilaterals: rectangular elements

We here assume that each element is a rectangle of size $h_x \times h_y$.

I.3 Hexahedra: cuboid elements

We here assume that each element is a cuboid⁶⁸ of size $h_x \times h_y \times h_z$.

⁶⁷<https://www.wolframalpha.com/>

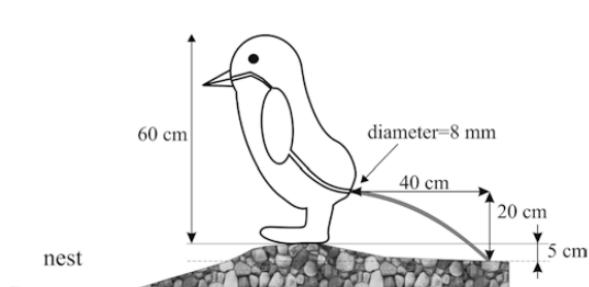
⁶⁸<https://en.wikipedia.org/wiki/Cuboid>

J Finite element terminology in various languages

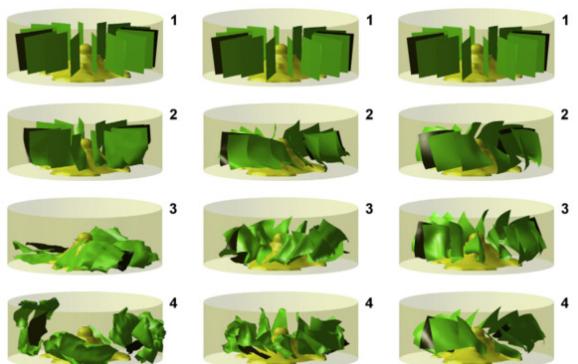
English	French	Dutch
Finite Element Method	Méthode des éléments finis	Eindige-elementenmethode
Finite Difference Method	Méthode des différences finies	Eindige-differentiemethode
Finite Volume Method	Méthode des volumes finis	
Matrix	Matrice	
Heat transport eq.	Equation de transport de la chaleur	Warmtetransport vergelijking
Momentum conservation eq.	équation de conservation du moment	Wet van behoud van impuls
Mass conservation / continuity eq.		Continuïteitsvergelijking
Iterative solver	solveur itératif	
Elemental matrix		
Boundary conditions	conditions aux limites	randvoorwaarden
(In)compressible	(in)compressible	
Surface processes	processus de surface	
an element	un élément	
Computational geodynamics	géodynamique numérique	
Assembly	assemblage	
Strong form		
Weak form	formulation variationnelle / formulation faible	
Basis function		
Shape function		
Partial differential eq. (PDE)	équation aux dérivées partielles (EDP)	partiële differentiaalvergelijking
Node	noeud	knooppunt
Grid, mesh	(la) maille / (le) maillage	rooster
Stiffness matrix	matrice de raideur	stijfheidsmatrix
Displacement vector	vecteur déplacement	verplaatsingsvector
Tessellation	pavage	Betegeling
Mass matrix	matrice de masse	
Classical mechanics	mécanique Newtonienne	(de) klassieke mechanica
Momentum	(le) moment	(de) impuls

K Fun modelling

Because sometimes numerical modelling is fun ...



Pressures produced when penguins pooh - calculations on avian defaecation [1314]



Clothes washing simulations [11]

Computers and Fluids Vol. 9, pp. 223-231
Pergamon Press Ltd., 1981. Printed in Great Britain

0045-7990(81)0001-022\$02.00/0

INTERNATIONAL JOURNAL FOR NUMERICAL METHODS IN FLUIDS, VOL. 9, 99-112 (1989)

DON'T SUPPRESS THE WIGGLES—THEY'RE TELLING YOU SOMETHING!†

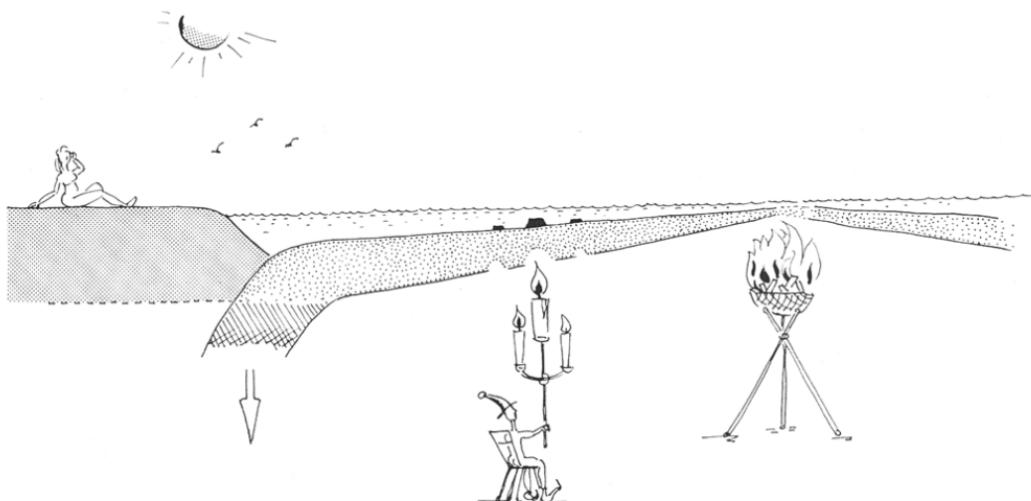
PHILIP M. GRESHO AND ROBERT L. LEE

[811]

ARE FEM SOLUTIONS OF INCOMPRESSIBLE FLOWS REALLY INCOMPRESSIBLE? (OR HOW SIMPLE FLOWS CAN CAUSE HEADACHES!)

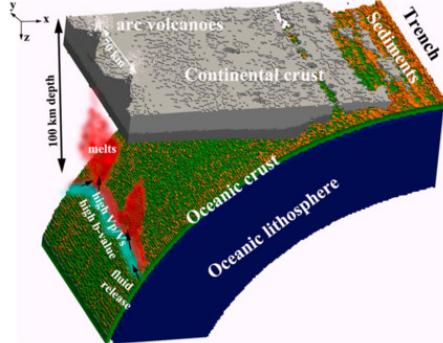
DOMINIQUE PELLETIER, ANDRE FORTIN AND RICARDO CAMARERO

[1456]

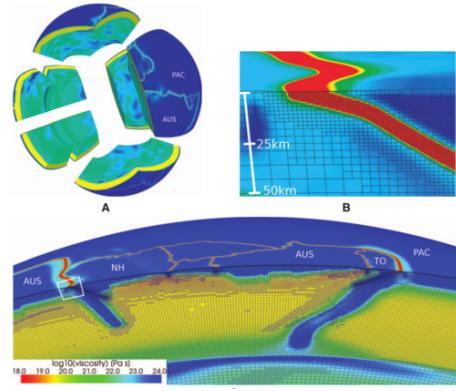


Lithospheric thickness anomaly near the trench and possible driving force of subduction [671]

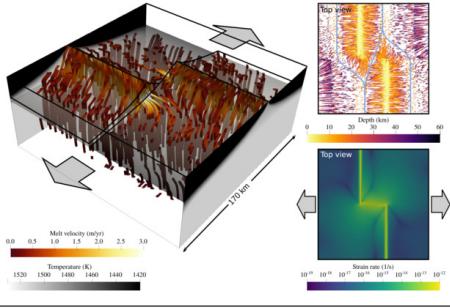
L Beautiful/interesting images from computational geodynamics



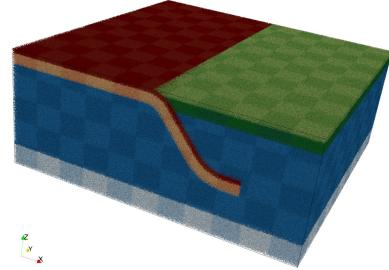
[2023]



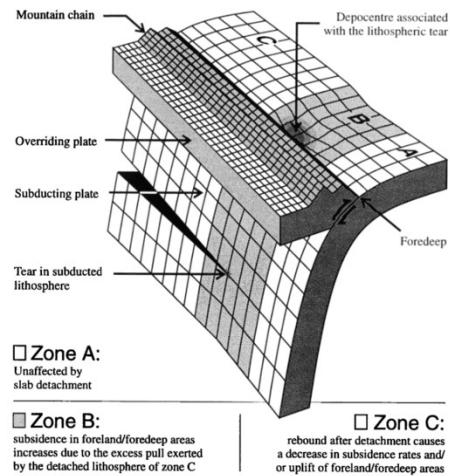
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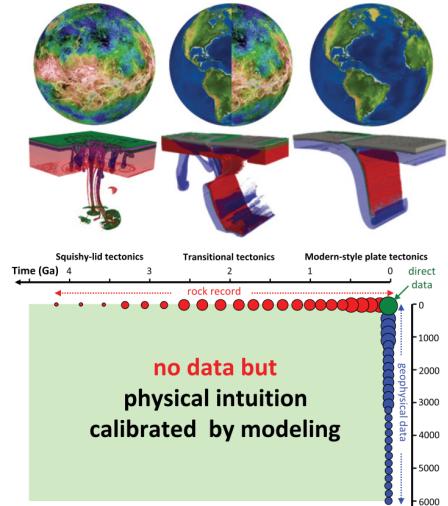
[474]



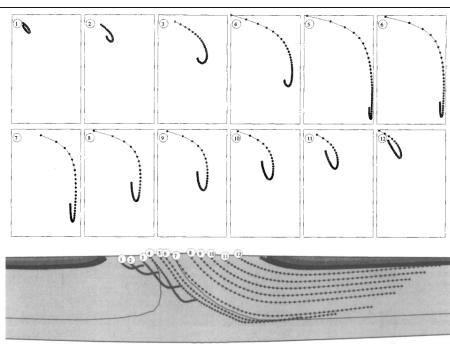
[650]



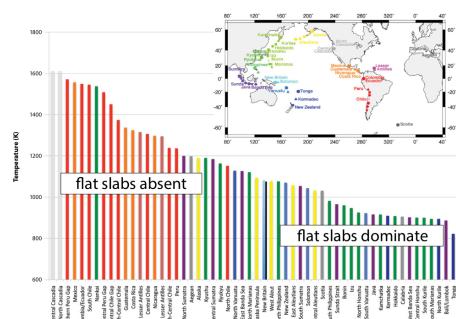
[1828]



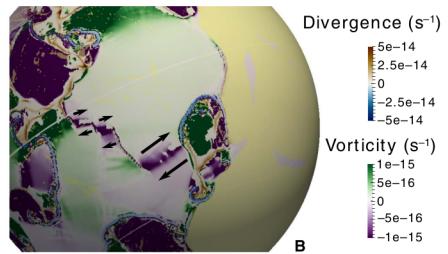
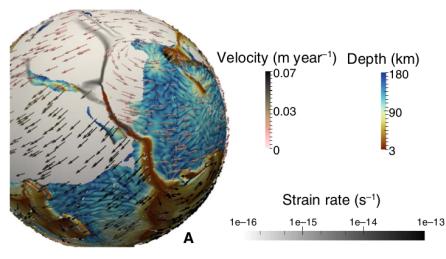
[724]



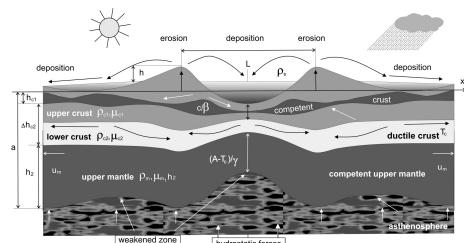
[85]



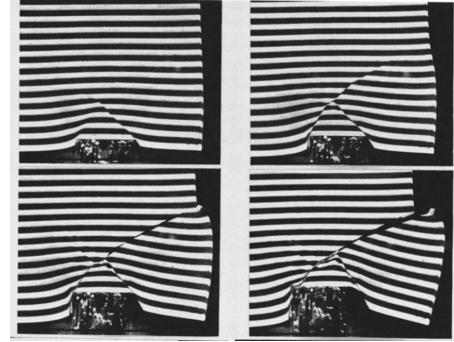
[1065]



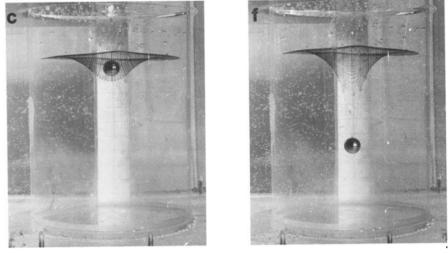
[415]



[293]



[1458]



[458]

M People

This is relevant to me only. Bibliography work.

- Thorsten Becker
- Sascha Brune
- Susanne Buiter. done
- Fabio Crameri
- Fabio Capitanio
- Anne Davaille
- Huw Davies
- Taras Gerya. done
- Gregor Golabek
- Mike Gurnis
- Laurent Husson
- Boris Kaus
- Louis Moresi. done
- Anne Replumaz
- Neil Ribe
- Tobias Rolf
- Paul Tackley. done
- Arie van den Berg
- Peter van Keken
- Ylona van Dinther
- Jeroen van Hunen
- Iris van Zelst

N Working with Git

This appendix was contributed by E. van der Wiel.

- make sure that you have an account on GitHub (say, <https://github.com/myusername>) and carry out a proper setup on your local computer as follows:

```
$ git config --global user.name "firstname lastname"
$ git config --global user.email your@email
```

- On github.com, fork the official fieldstone repository to your repository:
go to <https://github.com/cedrikt/fieldstone> and click on the 'fork' button on the upper right corner of the screen.

- On your machine, in a terminal, clone your repository with

```
$ git clone git@github.com:myusername/fieldstone.git
```

This is now your master branch.

- On your machine, find your security key ⁶⁹ with `$ less ~/.ssh/id_dsa.pub` and copy this into github.com so you can push to your repository. See <https://help.github.com/en/articles/connecting-to-github-with-ssh> on how to configure github with ssh support (no more login/-password to type – if you have cloned the repository wish ssh, not html). Please also check the instructions at <https://help.github.com/en/articles/connecting-to-github-with-ssh>.

- create a remote of your (online) repository

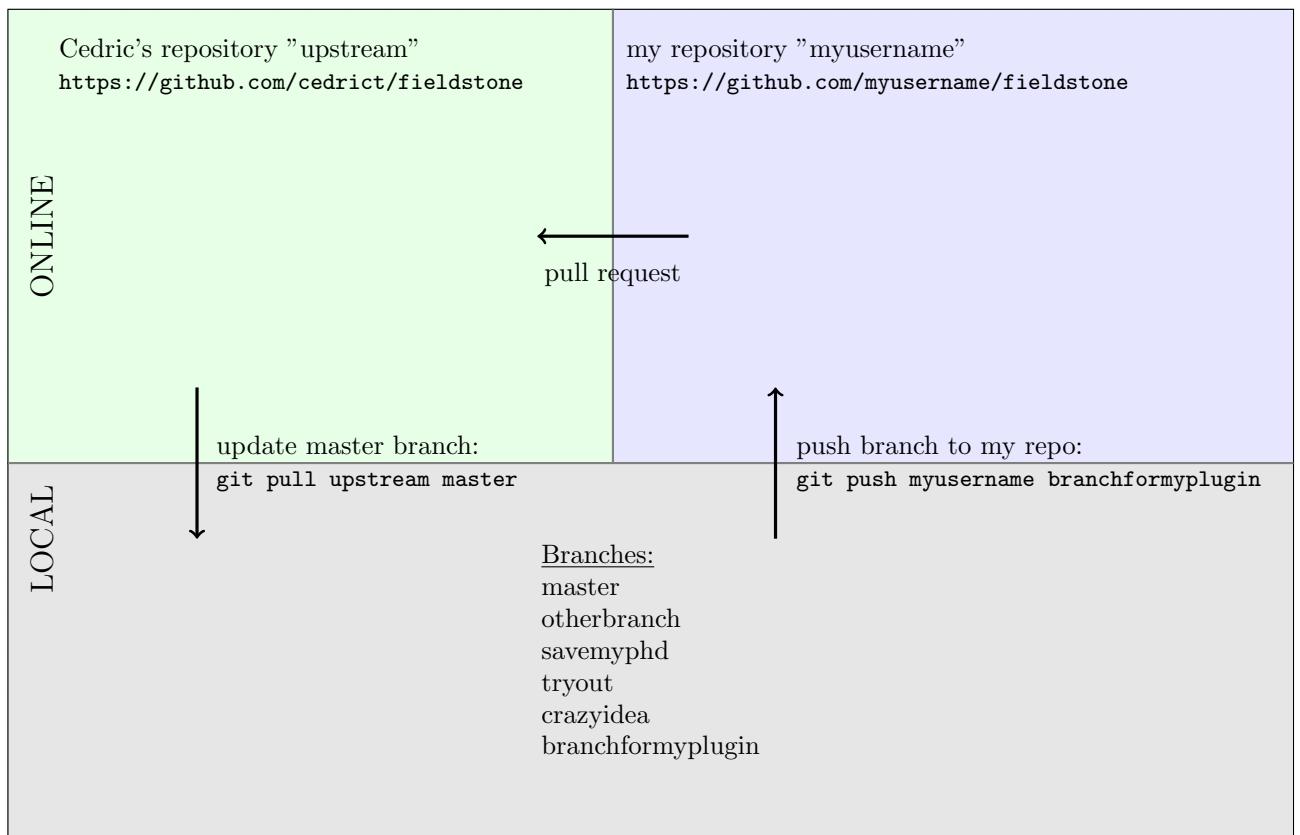
```
git remote add origin git@github.com:myusername/fieldstone.git
```

and in order to avoid potential confusion later on, we shall rename our github repo as follows:

```
$ git remote rename origin myusername
```

- Also create a remote of the official version with

```
$ git remote add upstream https://github.com/cedrikt/fieldstone
```



⁶⁹ Note that you can create a public key as follows: <https://help.github.com/articles/generating-ssh-keys/>

Very concretely, if you wish to contribute: Let us assume that you have found a typo in the file "physics.tex" and you wish to report the problem to me. The easiest way is to write me an email with the nature of the problem and the proposed fix. Although I will be grateful for your contribution, this approach can be improved upon by using git's "pull request" command.

1. If this is the very first time you use git, the following instruction is not needed. If not, make sure that your terminal points at your master branch: `$ git checkout master`
2. Then, make sure you update your local version: `$ git pull upstream master` and also update your own repo online: `$ git push`
3. Create a new branch with a self explanatory name: `$ git branch fix_typo` Then do `$ git branch` to see all your branches. The one that is coloured is the active branch. In order to switch to this new branch, do `$ git checkout fix_typo`. Redo `$ git branch` to verify that the branch 'fix_typo' is highlighted.
4. Edit the file and correct the typo, save and exit. Do `$ git status` and you should see the filename highlighted next to 'modified'.
5. If this is the only modification you wish to communicate, you then need to add and commit it as follows:

```
git add physics.tex
```

If you do git status again, the file should have changed colour (?). Then do

```
git commit -m "message"
```

where 'message' is a very short description of the modification (e.g. 'I fixed a typo').

6. We now need to bring this modification online

```
git push myusername fix_typo
```

7. You are nearly done. The last step takes place on github.com: when you log onto your own github fieldstone page, you should see a large green rectangle "Compare and pull request". Click on this button.
8. a new page opens, entitled "Open a pull request". If necessary, add a detailed description of the pull request (this makes sense when you contribute a piece of code, or a whole new section, etc ...). You can review the requested changes at the bottom. In the end, simply click "Create pull request".
9. Once you have done so, I will receive an email which notifies me of the pull request. I will then review it. If I have no comment, I will accept the PR and your modification will then be incorporated in the master repo. If I have comments, you will be notified via github and a back-and-forth discussion will ensue until I accept the PR.
10. After the PR has been accepted, the branch is no longer needed. Switch back to your local master branch: `$ git checkout master` . Update your local master and online repo (see step # 2) and then delete the no-longer-needed branch as follows:

```
git branch -d fix_typo
```

```

(base) UU062931:fieldstone vanderWiel$ git branch
  master
* my_work
(base) UU062931:fieldstone vanderWiel$ git checkout master
Switched to branch 'master'
Your branch is up to date with 'origin/master'.
(base) UU062931:fieldstone vanderWiel$ git pull upstream master
From https://github.com/cedrict/fieldstone
 * branch           master      -> FETCH_HEAD
 Already up to date.
(base) UU062931:fieldstone vanderWiel$ git push
Everything up-to-date
(base) UU062931:fieldstone vanderWiel$ █

(base) UU062931:fieldstone vanderWiel$ git branch fix_typos
(base) UU062931:fieldstone vanderWiel$ git branch
fix_typos
* master
my_work
(base) UU062931:fieldstone vanderWiel$ git checkout fix_typos
Switched to branch 'fix_typos'
(base) UU062931:fieldstone vanderWiel$ git branch
* fix_typos
master
my_work
(base) UU062931:fieldstone vanderWiel$ █

(base) UU062931:fieldstone vanderWiel$ vi physics.tex
(base) UU062931:fieldstone vanderWiel$ git status
On branch fix_typos
Changes not staged for commit:
  (use "git add <file>..." to update what will be committed)
    (use "git restore <file>..." to discard changes in working directory)
       modified:   physics.tex

no changes added to commit (use "git add" and/or "git commit -a")
(base) UU062931:fieldstone vanderWiel$ git add physics.tex
(base) UU062931:fieldstone vanderWiel$ git status
On branch fix_typos
Changes to be committed:
  (use "git restore --staged <file>..." to unstage)
       modified:   physics.tex

(base) UU062931:fieldstone vanderWiel$ git commit -m "I fixed a typo"
[fix_typos 85beb3e] I fixed a typo
  1 file changed, 1 insertion(+), 1 deletion(-)
(base) UU062931:fieldstone vanderWiel$ git push origin fix_typos
Enumerating objects: 5, done.
Counting objects: 100% (5/5), done.
Delta compression using up to 4 threads
Compressing objects: 100% (3/3), done.
Writing objects: 100% (3/3), 294 bytes | 294.00 KiB/s, done.
Total 3 (delta 2), reused 0 (delta 0)
remote: Resolving deltas: 100% (2/2), completed with 2 local objects.
remote:
remote: Create a pull request for 'fix_typos' on GitHub by visiting:
remote:   https://github.com/e-wiel/fieldstone/pull/new/fix_typos
remote:
To https://github.com/e-wiel/fieldstone.git
 * [new branch]      fix_typos -> fix_typos
(base) UU062931:fieldstone vanderWiel$ █



```

Screen captures of the procedure described above as carried out by E. van der Wiel on his Apple laptop.

In what follows we summarize the most important commands one should and remember while working with github. After creating an account one can 'fork' a repository (repo) in the online environment. This repository is a copy from the master directory of the developer and should not be used to adapt or change, as changes from the developer (updates) should be obtained in this 'fork', or as it could also be called; your master branch.

In order to be able to work within a repository, for instance, to run and compile different programs, you should have your own branch of the repository in which YOU CAN make changes. The following commands should be used to make, copy and publish your own version of the repo to your local device and the online github environment.

command	what it does
git branch	shows all branches of your repository and highlights the one you're in.
git checkout -b <my_own_branch>	This makes your own branch called "my_own_branch".
git push origin <my_own_branch>	This pushes your own, local, branch to as a second branch in the online repo of github.
git checkout <name>	changing the branch you're working in (e.g. master or my_own_branch). Or replace the name with a hyphen to switch to the last branch.
git branch -d <my_own_branch>	Delete your local branch.

The following commands should be used in order to update your own local branches from updates made by somewhere else (upstream/master is the main repository). One should do this for the local master branch and, where possible as well for the different local branches you have committed changes to already.

command	what it does
git checkout master	To make sure you are in the right branch
git fetch upstream	to fetch updates from upstream repositories to your own local branch (e.g. to update your master branch).
git merge upstream/master	Command to update the branch with the fetched repo from 'upstream'.
git push origin master	To level your own online repository again with the one on your local drive (and thus the one upstream).
git checkout <my_own_branch>	To switch to your own adapted branch of the repo.
git merge master	Used from another branch working directory to combine the new released version of the master repo with the one where all your own changes are put. -> Then git finds all conflicts in different files which you need to resolve.
git add .	This adds the resolved issues in your own local branch (not master). After which you are able to commit and push your changes back to the online repository.

While you are working in your own branch you can change, add or delete files in any amount you want. However, always check whether your changes do not inflict the outcome of for instance your code. And when uploading from your terminal: if you commit and then push from master branch your changes will automatically be inserted in the online version of your master branch, when done from another branch it will be shown as a pull request towards your master branch. This request can then, for instance be forwarded to the main repo.

command	what it does
git commit -a	This will send your changes/updates from your branch as a commit to your own local branch.
git push origin <changes>	To update the remote repository (on Github) from you local repository (in this case the 'changes' branch). (Actually upload the new version). Online one can then judge what to do with it. !! this is a pull request towards your own fork/local_branch
git status	Showing the status of your current branch; it shows which files are different between the master file and your adapted branch.
git diff <changes>	This shows the exact differences between the different branches; one can simply ask for the difference between two branches when pwd in one branch ask for the other branch.
git merge <my_own_branch>	When used from the master branch (or any other???) this accepts the changes made in your branch and puts them in your local(!) master branch.
git pull origin master	if the main repository changes, one can pull the newest version towards it's own master file. While keeping your own branches alive with you own changes and vica versa: by running this command the origin/master (remote file) will be cloned and updated to the working branch you are in.
git stash (apply)	?? While updating your local branch, sometimes git wants to overrule your own changes, with this command you can 'stash' them to look at the differences later. ??

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