## One of the advantages of state space models is that it is possible to apply state feedback to place the closed loop poles into any desired positions. **State Space Design Methodology**

1. Design control law to place closed loop poles where desired

State-Feedback Control

1. Combine Observer and Controller -- this takes the place of the Classical Compensator

1. If full state not available for feedback, then design an *Observer* to compute the states from the system output

1. Introduce the *Reference Input* -- affects the closed loop zeros but not the poles making it possible to improve the transient

response and tracking accuracy

**State Feedback Compensator** 

output plant i/p Plant

```
ref. i/p
                                           X
                           -\mathbf{K}
                                                         Observer
                                   measured
                                   states
This Section

    Finding the control law

    State feedback for controller canonical form
```

**Additional Materials** 

Not examined

 Effect of state feedback on closed-loop zeros Effect of plant zeros on the feedback gains

**Finding the Control Law** 

Let the input to the plant, u be derived from the reference input r, and the states, x, as follows:

We shall only consider SISO systems here.

The state matrix for the closed loop system with input, r, is: A - BK.

Thus:

Therefore

Suppose the desired closed loop poles are to be at  $p_1$ ,  $p_2$ ,  $\cdots$ ,  $p_n$ , then the desired CE is:  $\alpha_c(s) = (s - p_1)(s - p_2) \cdots (s - p_n) = 0$ 

Equation (2) is multiplied out to a polynomial in s.

This leads to a set of linear equations in the k's which can always be solved to give the required feedback control law, for whatever closed loop pole locations are given.

Problem: Given,  $\dot{\mathbf{x}} = \begin{bmatrix} -4 & 0 \\ 0 & -11 \end{bmatrix} \mathbf{x} + \begin{bmatrix} 1 \\ -1 \end{bmatrix} \mathbf{u}$ find the feedback control law which places the closed-loop poles at:  $-10 \pm j10$ .

 $0 = \det \begin{bmatrix} s+4+k_1 & k_2 \\ -k_1 & s+11-k_2 \end{bmatrix}$ 

 $0 = (s + 4 + k_1)(s + 11 - k_2) + k_1k_2$ 

 $0 = (s + 4 + k_1)(s + 11 - k_2) - (k_2)(-k_1)$ 

The closed loop poles are the roots of s in the Characteristic Equation (CE):

Now the desired CE is:  $\alpha_c(s) = (s + 10 - j10)(s + 10 + j10) = 0$  $s^2 + 20s + 200 = 0$ 

 $0 = \det [s\mathbf{I} - \mathbf{A} + \mathbf{B}\mathbf{K}] = \det \left\{ \begin{bmatrix} s+4 & 0 \\ 0 & s+11 \end{bmatrix} + \begin{bmatrix} 1 \\ -1 \end{bmatrix} \begin{bmatrix} k_1 & k_2 \end{bmatrix} \right\}$ 

 $s^2 + (15 + k_1 - k_2)s + (44 + 11k_1 - 4k_2) = 0$ 

 $\begin{bmatrix} k_1 \\ k_2 \end{bmatrix} = \frac{1}{-4+11} \begin{bmatrix} -4 & 1 \\ -11 & 1 \end{bmatrix} \begin{bmatrix} 5 \\ 156 \end{bmatrix} = \frac{1}{7} \begin{bmatrix} 136 \\ 101 \end{bmatrix} = \begin{bmatrix} 19.429 \\ 14.429 \end{bmatrix}$ 

 $u = r - [19.429 \quad 14.429] \mathbf{x}$ 

 $\mathbf{A} = \begin{bmatrix} a_1 & a_2 & \cdots & a_n \\ 1 & 0 & \cdots & 0 \\ 0 & \ddots & \cdots & \vdots \end{bmatrix}; \quad \mathbf{B} = \begin{bmatrix} 1 \\ 0 \\ \vdots \end{bmatrix}$ 

State Feedback in the Case of the Controller Canonical Form

Therefore the required feedback control law is: **COMMENT** This matching of coefficients can always be done, though it is tedious for n > 3, **EXCEPT** in the case of the

Control Canonical Form.

with open loop CE:

where:

Example 2

Solved in class

**SOLUTION:** 

**Problem**: Given the system TF:

The control canonical form has matrices:

**NB**: **C** is obtained from the TF numerator (0s + 7). so:

In the control canonical form we have matrices:

SOLUTION:

with open loop CE: 
$$\det(s\mathbf{I}-\mathbf{A})=s^n+a_1s^{n-1}+\cdots+a_n=0.$$
 Feedback results in the closed loop CE: 
$$\det(s\mathbf{I}-\mathbf{A}+\mathbf{B}\mathbf{K})=0$$
 where: 
$$\mathbf{A}-\mathbf{B}\mathbf{K}=\mathbf{A}-\begin{bmatrix}1\\0\\\vdots\\0\end{bmatrix}\begin{bmatrix}k_1&k_2&\cdots&k_n\end{bmatrix}$$

 $\begin{bmatrix} (-a_1-k_1) & (-a_2-k_2) & \cdots & (-a_n-k_n) \end{bmatrix}$ 

 $s^{n-2}: a_2 + k_2 = \alpha_2 \rightarrow k_2 = \alpha_2 - a_2$  $s^0: a_n + k_n = \alpha_n \rightarrow k_n = \alpha_n - a_n$ 

 $G(s) = \frac{7}{(s+4)(s+11)}$ 

 $G(s) = \frac{7}{(s+4)(s+11)} = \frac{7}{(s^2+15s+44)}$ 

 $\mathbf{A} = \begin{bmatrix} -15 & -44 \\ 1 & 0 \end{bmatrix}; \quad \mathbf{B} = \begin{bmatrix} 1 \\ 0 \end{bmatrix}; \quad \mathbf{C} = \begin{bmatrix} 0 & 7 \end{bmatrix}; \quad \mathbf{D} = 0$ 

 $\mathbf{A} - \mathbf{B}\mathbf{K} = \begin{bmatrix} -15 - k_1 & -44 - k_2 \\ 1 & 0 \end{bmatrix}$ 

 $s^2 + (15 + k_1)s + (44 + k_2) = 0$ 

find the control law for the control canonical form which places the closed loop poles at  $s=-10\pm j10$ .

 $s^{n-1}: a_1 + k_1 = \alpha_1 \to k_1 = \alpha_1 - a_1$ 

The desired CE is: Comparing Eqs. (7) and (8) gives:

giving the control law as:

the feedback law:

so, taking Laplace transforms:

R(s)

**Ackermann's Formula** 

The formula

and

**Caveats** 

**MATLAB** Function

**Solution in MATLAB** 

In [1]:  $A = [1 \ 2; -1 \ 1]; B = [1; -2];$ 

-2

Using the function acker

Ka = acker(A, B, P)

-2

alpha c = A \* A + 2 \* A + eye(2);

 $K = [0 \ 1] * inv(ctrb(A, B)) * alpha c$ 

In [2]: P = [-1, -1]; % vector of desired pole locations

Using the formula

0

K =

Ka =

Comments

system

**Explanation of the terms** 

 ${\cal C}$  is the controllability matrix (see Section 8.1):

• The system must be *controllable* for  $C^{-1}$  to exist.

• C becomes numerically inaccurate for large n.

• Ackermann's formula is useful for SISO systems of order  $n \leq 10$ .

**Transfer Function Model of State Feedback** 

and

and the closed loop CE is:

A Transfer Function Model of State Feedback The last example had a system TF with no zeros. In this case it is easy to construct the equivalent classical controller. We had the feedback law: 
$$u=r-5x_1-156x_2$$
 so, taking Laplace transforms: 
$$U(s)=R(s)-5X_1(s)-156X_2(s)$$
 Now  $y=7x_2$  and  $\dot{x}_2=x_1$  therefore  $X_2(s)=Y(s)/7$  and  $X_1(s)=sX_2(s)=sY(s)/7$ . Therefore

 $U(s) = R(s) - \frac{1}{7}(5s + 156)Y(s)$ 

 $u = r - \begin{bmatrix} 5 & 156 \end{bmatrix} \mathbf{x}$ 

State Feedback Design for any Form of State Space Model • As stated previously, the derivation of the feedback law is tedious for systems of order n > 3 except in the case of the controller canonical form. • One approach to the problem is to transform the given model to controller canonical form, derive the control law in terms of these states and then transform back to the original system.

State feedback:  $u = r - \mathbf{K}\mathbf{x}$  will place the closed loop poles at the roots of the desired CE:

Ackermann derived the following formula by this method.

where the row vector  $\mathbf{K}$  is given by Ackermann's formula:

Problem: Given:  $\mathbf{A} = \begin{bmatrix} 1 & 2 \\ -1 & 1 \end{bmatrix} \quad \text{and} \quad \mathbf{B} = \begin{bmatrix} 1 \\ -2 \end{bmatrix}$ find the feedback vector **K** to place the closed loop poles at s = -1, -1 using Ackermann's formula. SOLUTION:  $\alpha_c(s) = (s+1)(s+1) = s^2 + 2s + 1$ therefore  $\alpha_c(s) = \mathbf{A}^2 + 2\mathbf{A}s + \mathbf{I}$   $\alpha_c(\mathbf{A}) = \begin{bmatrix} -1 & 4 \\ -2 & -1 \end{bmatrix} + 2 \begin{bmatrix} 1 & 2 \\ -1 & 1 \end{bmatrix} + \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} = \begin{bmatrix} 2 & 8 \\ -4 & 2 \end{bmatrix}$   $\mathbf{A}\mathbf{B} = \begin{bmatrix} -3 \\ -3 \end{bmatrix}; \quad C = \begin{bmatrix} \mathbf{B} : \mathbf{A}\mathbf{B} \end{bmatrix} = \begin{bmatrix} 1 & -3 \\ -2 & -3 \end{bmatrix}$  $\mathbf{K} = \begin{bmatrix} 0 & \cdots & 0 & 1 \end{bmatrix} C^{-1} \alpha_c(\mathbf{A})$  $= \begin{bmatrix} 0 & 1 \end{bmatrix} \begin{bmatrix} 1 & -3 \\ -2 & -3 \end{bmatrix}^{-1} \begin{bmatrix} 2 & 8 \\ -4 & 2 \end{bmatrix}$ 

 $= \begin{bmatrix} 0 & 1 \end{bmatrix} \frac{\begin{bmatrix} -3 & 3 \\ 2 & 1 \end{bmatrix}}{\begin{bmatrix} -3 - (+6) \end{bmatrix}} \begin{bmatrix} 2 & 8 \\ -4 & 2 \end{bmatrix}$ 

 $= \begin{bmatrix} 0 & 1 \end{bmatrix} \frac{1}{-9} \begin{bmatrix} -18 & -18 \\ 0 & 18 \end{bmatrix}$ 

 $= \begin{bmatrix} 0 & 1 \end{bmatrix} \begin{bmatrix} 2 & 2 \\ 0 & -2 \end{bmatrix}$ 

 $= \begin{bmatrix} 0 & -2 \end{bmatrix}$ 

Order/Type  $s^2 + \sqrt{2}s + 1$   $s^2 + \sqrt{3}s + 1$  $3 s^3 + 1.75s^2 + 2.15s + 1 s^3 + 2.43s^2 + 2.47s + 1$ 

example.

 Transfer function model Ackermann's formula Pole-selection for good design

 $u = r - \mathbf{K}\mathbf{x} = r - (k_1x_1 + k_2x_2 + \cdots + k_nx_n)$ 

= Ax + B(r - Kx)

= (A - BK)x + Br

 $s\mathbf{X}(s) = (\mathbf{A} - \mathbf{B}\mathbf{K})\mathbf{X}(s) + \mathbf{B}R(s)$ 

 $(s\mathbf{I} - \mathbf{A} + \mathbf{B}\mathbf{K})\mathbf{X}(s) = \mathbf{B}R(s).$ 

 $\det [s\mathbf{I} - \mathbf{A} + \mathbf{B}\mathbf{K}] = 0$ 

Finally, we need to find the k coefficients in  $\mathbf{K}$  such that the polynomials in equations (1) and (2) above have matching

 $\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}u$ 

Taking Laplace Transforms (ignoring initial conditions) gives:

coefficients in each power of s. **Example 1** Will be done in class.

Therefore matching coefficients in Eqs. (3) and (4):  $s^2: 1 = 1 \to OK$  $s^1: 15 + k_1 - k_2 = 20 \rightarrow k_1 - k_2 = 5$  $s^0: 44 + 11k_1 - 4k_2 = 200 \rightarrow 11k_1 - 4k_2 = 156$ Solving for the k's:  $\begin{bmatrix} 1 & -1 \\ 11 & -4 \end{bmatrix} \begin{bmatrix} k_1 \\ k_2 \end{bmatrix} = \begin{bmatrix} 5 \\ 156 \end{bmatrix}$ 

 $\mathbf{A} - \mathbf{B}\mathbf{K} = \begin{bmatrix} 1 & 0 & \cdots & \vdots \\ \vdots & \ddots & \cdots & \vdots \\ 0 & 0 & \cdots & 0 \end{bmatrix}$ therefore  $\det(s\mathbf{I} - \mathbf{A} + \mathbf{B}\mathbf{K}) = s^n + (-a_1 - k_1)s^{n-1} + (-a_2 - k_2)s^{n-2} + \dots + (-a_n - k_n) = 0.$ Suppose the desired CE is:  $\alpha_c(s) = (s - p_1)(s - p_2) \cdots (s - p_n) = 0$  $\alpha_c(s) = s^n + \alpha_1 s^{n-1} + \alpha_2 s^{n-2} + \dots + \alpha_n = 0$ Matching coefficients in Eqs. (5) and (6) is now simple:

> $\alpha_c(s) = (s + 10 - j10)(s + 10 + j10) = 0$  $s^2 + 20s + 200 = 0$  $15 + k_1 = 20 \rightarrow k_1 = 5$  $44 + k_2 = 200 \rightarrow k_2 = 156$

**Note**: the DC gain is affected -- this could be compensated for by introducing a gain term in series with input R.

 $\alpha_c(s) = s^n + \alpha_1 s^{n-1} + \dots + \alpha_n = 0$ 

 $\mathbf{K} = \begin{bmatrix} 0 & \cdots & 0 & 1 \end{bmatrix} C^{-1} \alpha_c(\mathbf{A})$ 

 $C = [\mathbf{B} : \mathbf{AB} : \cdots : \mathbf{A}^{n-1}\mathbf{B}]$ 

 $\alpha_c(\mathbf{A}) = \mathbf{A}^n + \alpha_1 \mathbf{A}^{n-1} + \dots + \alpha_n \mathbf{I}$ 

 $G(s) = \frac{7}{(s+4)(s+11)}$ 

Y(s)

 $\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}u$ with a feedback law of  $u = -\mathbf{K}\mathbf{x}$  has closed loop poles at the values specified in vector  $\mathbf{P}$ , i.e.,  $\mathbf{P} = \text{eig}(\mathbf{A} - \mathbf{B} * \mathbf{K})$ . Note: This algorithm uses Ackermann's formula. This method is NOT numerically reliable and starts to break down rapidly for problems of order greater than 10, or for weakly controllable systems. A warning message is printed if the nonzero closed-loop poles are greater than 10% from the desired locations specified in **P**. **Example 3** For class

From MATLAB CST, help acker: K = ACKER(A,B,P) calculates the feedback gain matrix K such that the single input

**Pole Selection for Good Design** Large control gains (large values in  $\mathbf{K}$ ) are to be avoided since they result in high energy costs and require high power/bandwidth actuators. A compromise must be achieved between good system response and control effort. Sensible choices of poles may be obtained from standard tables which optimise the step response in some way. e.g. The ITAE (Integral Time and Absolute Error) poles are designed to minimise,  $I = \int_{0}^{\infty} t |\text{error}| dt$ 

etc

where the choice of the parameter k determines the required compromise between,

Effect of State Feedback on the Closed Loop Zeros

• High Accuracy for High Control Effort (use a large value for k)

Lower Accuracy for Reduced Control Effort (use a smaller value for k)

E.g. if  $\omega_n = 5$  rad/s, the  $2^{\rm nd}$  order ITAE polynomial is:  $s^2 + 5\sqrt{2}s + 25$ .

Butterworth) poles if some overshoot is acceptable.

reduced order system as above.

the feedback vector  $\mathbf{K}$  such as to minimise,

**End of Pre-Class Presentation** 

**Non Examined Content** 

The new form for the TF is:

feedbck control.

Since,

These have overshoot. If this is really unacceptable (e.g. in machine tools) then Bessel polynomials can be used.

s+1

The above are normalised to give  $\omega_n=1$  rad/s. To obtain polynomials for  $\omega_n\neq 1$ , replace s in the above with  $s/\omega_n$ .

In general the Bessel polynomials have too much damping for normal applications --- it is preferable to use ITAE (or

If zeros are present they tend to "sharpen up" the transient response (faster rise times but consequently with more

shifted to a better damped location, if necessary, at a similar frequency (distance from the origin).

overshoot). In such cases it may be desirable to place a closed loop pole on top of a troublesome zero and work with the

The poles closest to the origin matter most; other poles give rise to shorter-term transients only and may need only to be

A most effective technique is to use optimal control to achieve a compromise between control effort, u, and error, e. i.e. Find

 $J = \int_0^\infty \left( e^2 + \frac{u^2}{k} \right) dt$ 

The remainder of this section will not be examined. It is nonetheless important for those of you wishing to implement state-

 $u = r - \mathbf{K}\mathbf{x}$ 

 $\frac{Y(s)}{R(s)} = \frac{\det \begin{bmatrix} (s\mathbf{I} - \mathbf{A}) & -\mathbf{B} \\ \mathbf{C} & \mathbf{D} \end{bmatrix}}{\det(s\mathbf{I} - \mathbf{A} + \mathbf{R}\mathbf{K})}$ 

Notice now that numerator is identical to that of the open loop TF. This implies that the state feedback control has left the open

loop zeros unchanged. The different denominator is due to the feedback action which alters the pole positions as required.

ITAE

etc

Bessel

etc

s+1

then the closed-loop system is:  $\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}\mathbf{u}; \ \mathbf{y} = \mathbf{C}\mathbf{x} + \mathbf{D}\mathbf{u}$  $\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}(r - \mathbf{K}\mathbf{x}); \ y = \mathbf{C}\mathbf{x} + \mathbf{D}(r - \mathbf{K}\mathbf{x})$  $\dot{\mathbf{x}} = (\mathbf{A} - \mathbf{B}\mathbf{K})\mathbf{x} + \mathbf{B}r; \ \ y = (\mathbf{C} - \mathbf{D}\mathbf{K})\mathbf{x} + \mathbf{D}r$ By analogy with previous work (see <u>Some Important Properties</u>, in <u>Section 7.2</u>), the TF from reference input r to output y is:  $\frac{Y(s)}{R(s)} = \frac{\det \begin{bmatrix} (s\mathbf{I} - \mathbf{A} + \mathbf{B}\mathbf{K}) & -\mathbf{B} \\ (\mathbf{C} - \mathbf{D}\mathbf{K}) & \mathbf{D} \end{bmatrix}}{\det(s\mathbf{I} - \mathbf{A} + \mathbf{B}\mathbf{K})}$ The closed loop TF zeros are determined by the numerator determinant.

Adding  ${\bf K}$  times the  $2^{\rm nd}$  column to the first cancels terms whilst leaving the determinant unchanged.

p-z

State feedback for controller canonical form

Transfer function model

Pole-selection for good design

Effect of state feedback on closed-loop zeros

Effect of plant zeros on the feedback gains

Ackermann's formula

Effect of Zero Locations on the Feedback Gains When a zero is close to a pole in the TF there is a marked increase in the feedback gains. This effect is best illustrated with an Given a system with a TF,  $\frac{Y(s)}{U(s)} = \frac{s-z}{s-p} = 1 + \frac{p-z}{s-p}$ find the control law to move the pole to  $p_c$ . Using the observer canonical form,  $\dot{x} = px + (p-z)u, \quad y = x + u.$ Design the feedback to move the closed-loop pole to  $p_c$ . Now, A = p; B = p - z;  $K = k_1$ . Desired CE polynomial:  $\alpha_c(s) = s - p_c$ . Actual CE polynomial:  $\det(s\mathbf{I} - \mathbf{A} - \mathbf{B}\mathbf{K}) = s - p + (p - z)k_1$ . Comparing the constant:  $-p_c = -p + (p-z)k_1$  $k_1 = \frac{p - p_c}{p - z}$ Notice that the feedback gain  $k_1$  is large if: • the zero is close to the pole:  $z \approx p$  the closed loop pole p<sub>c</sub> is far from p Effect of Zero Locations on the Feedback Gains Large feedback control gains are required if: 1. There exist almost cancelling pole-zero pairs in the open loop TF, making the system almost uncontrollable.

(Notice that in this condition the input u is almost disconnected from the integrator for the state.) 1. One tries to move the poles a long way, ( $|p - p_c|$  large). This imposes a practical limit on how arbitrarily the poles can be placed. You cannot make a slow system fast without using large gains requiring powerful, expensive actuators to force the plant response. Indeed, excessively large forces may destroy the plant. **Summary** Finding the control law