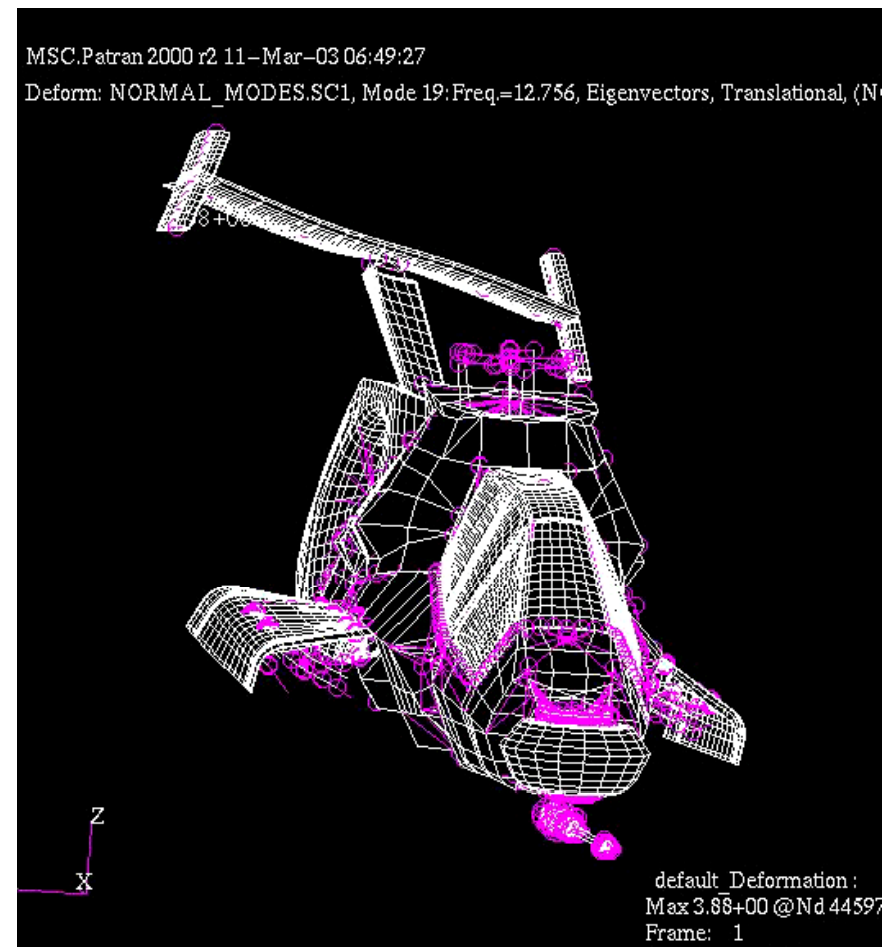


Frequency Response Design Example

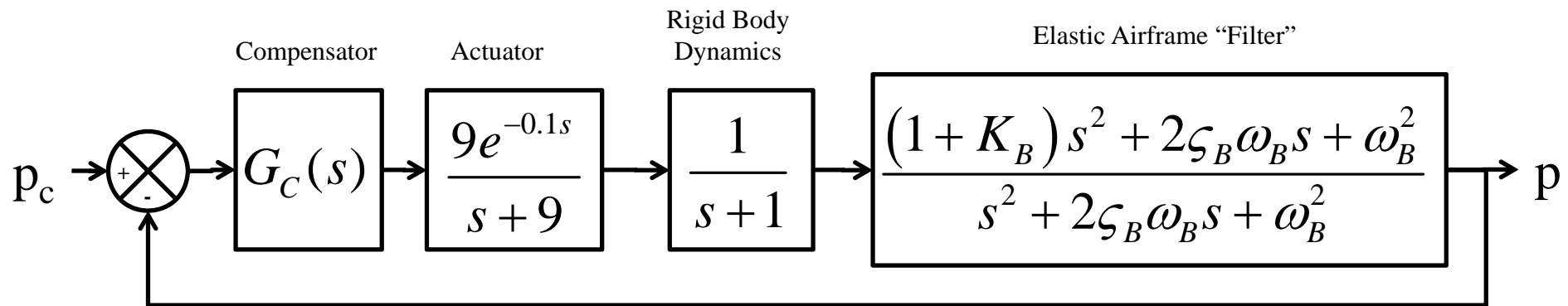
ESE 505 & MEAM 513

Bruce D. Kothmann

2014-04-02



Plant Description & Design Goals



- Overall Design Objective = Good Disturbance Rejection = High Velocity Error Constant = High Integral Gain in Compensator
- Some Design Constraints
 - Rigid Body Damping Ratio $> \sim 0.3$
 - Rigid Body Setting Time $< \sim 2.5$ sec
 - Phase Margin $> \sim 45$ Deg
 - Gain Margin $> \sim 6$ dB

Elastic Modes Very Common Challenge

VOL. 5, NO. 4, JULY-AUGUST 1982

J. GUIDANCE

403

AIAA 80-176R

Space Telescope Pointing Control System

H. Dougherty*

Lockheed Missiles & Space Company, Inc., Sunnyvale, Calif.

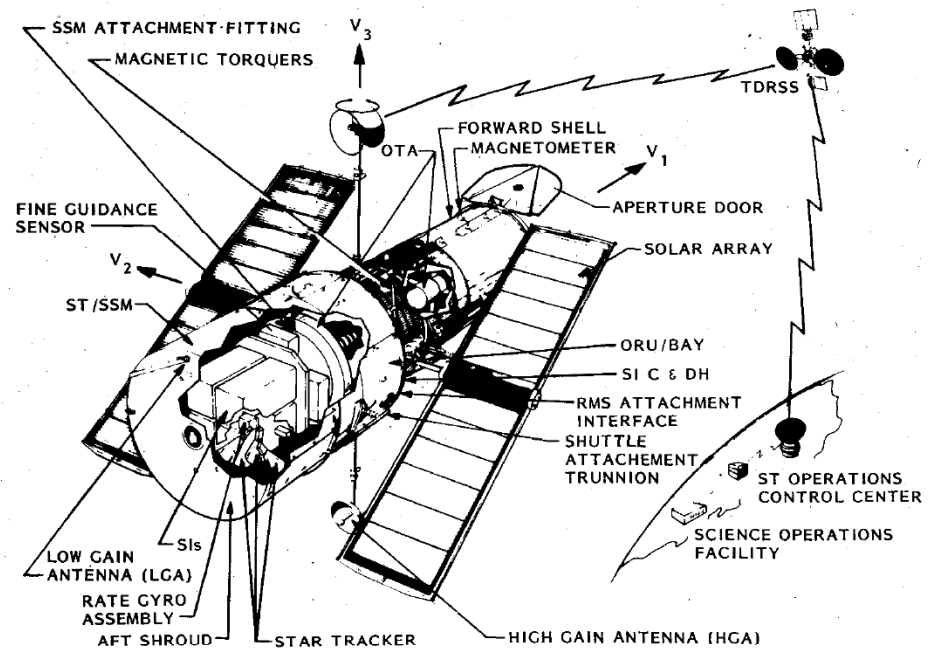
K. Tompetrini† and J. Levinthal‡

Bendix Guidance Systems Division, Teterboro, N.J.

and

G. Nure§

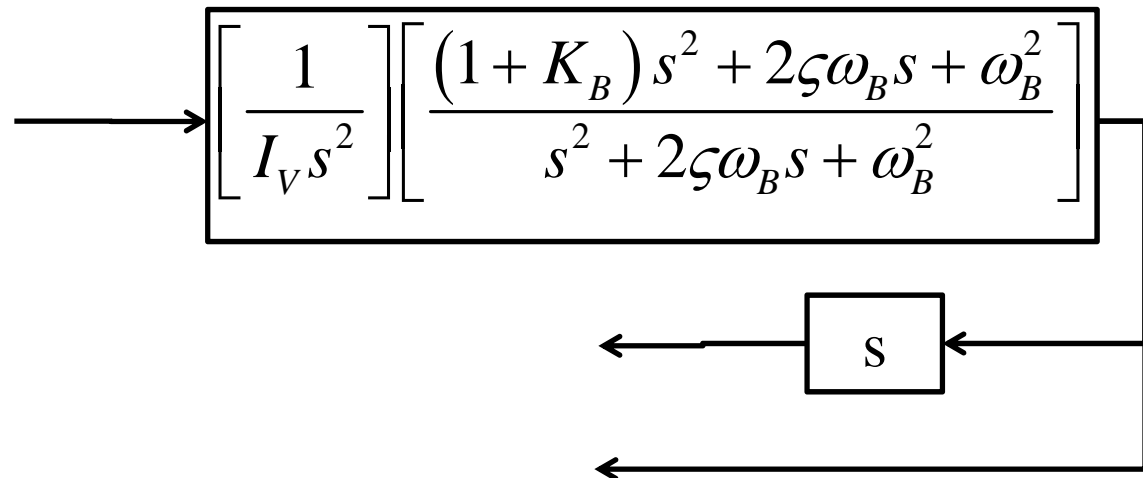
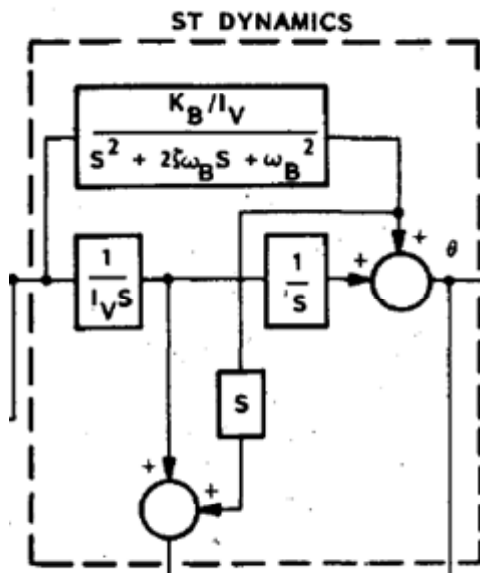
NASA Marshall Space Flight Center, Huntsville, Ala.



Space Telescope Dynamics

1) *Structural modes*: The solar array and optical telescope assembly modes have large modal coefficients. For example, the value of the solar array inertia about the Space Telescope center of mass is almost one-half that of the Space Telescope centerbody, which comprises the support systems module and optical telescope assembly. The control system sample rate and compensation are chosen to stabilize the modes. The command generator shapes maneuvers to limit structural excitations during maneuvers.

Modal Coefficient = K_B



Another Example : Space Station

Pointing, Control, and Stabilization of Solar Dynamic Systems on a Space Station

Eric T. Falangas and Henry H. Woo

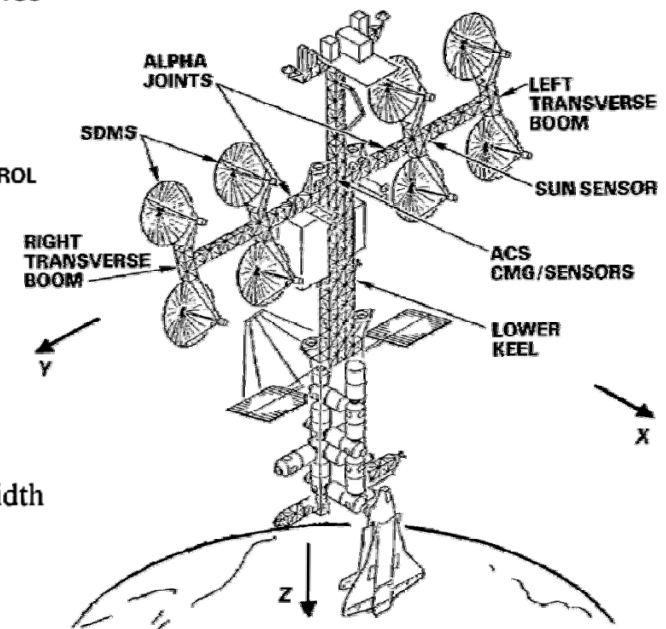
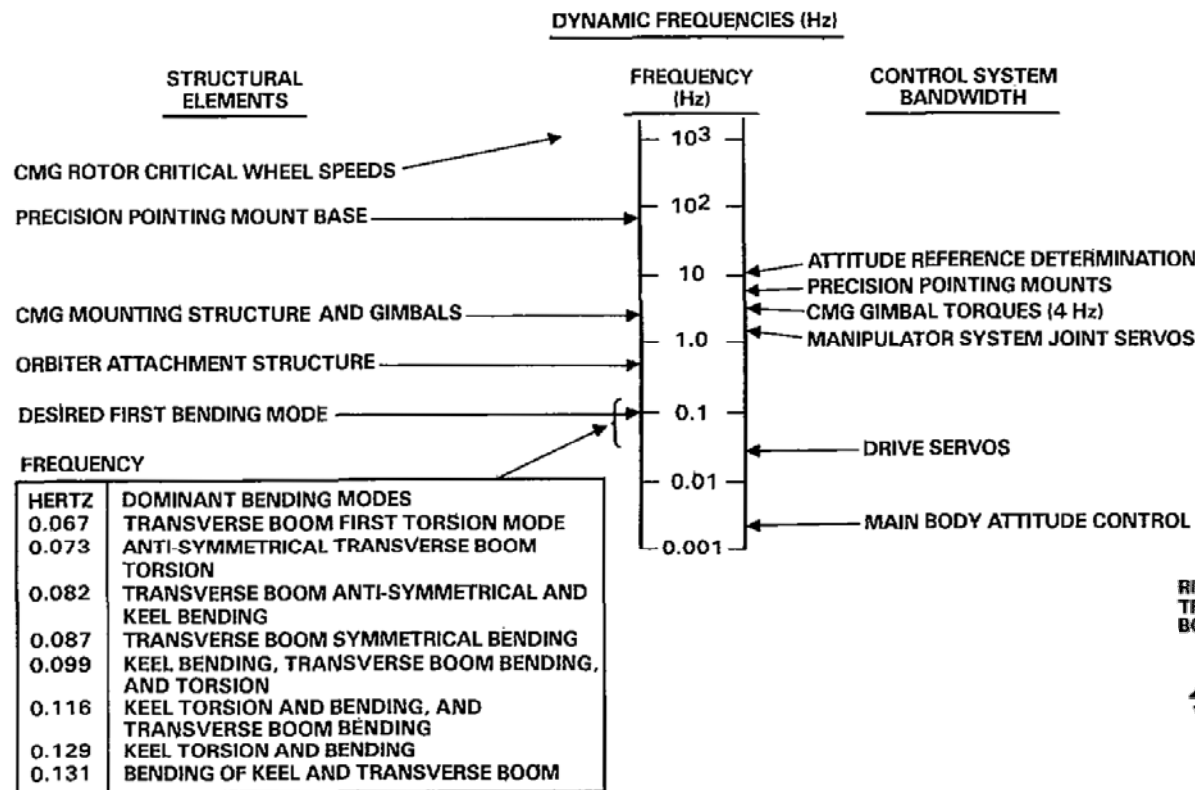
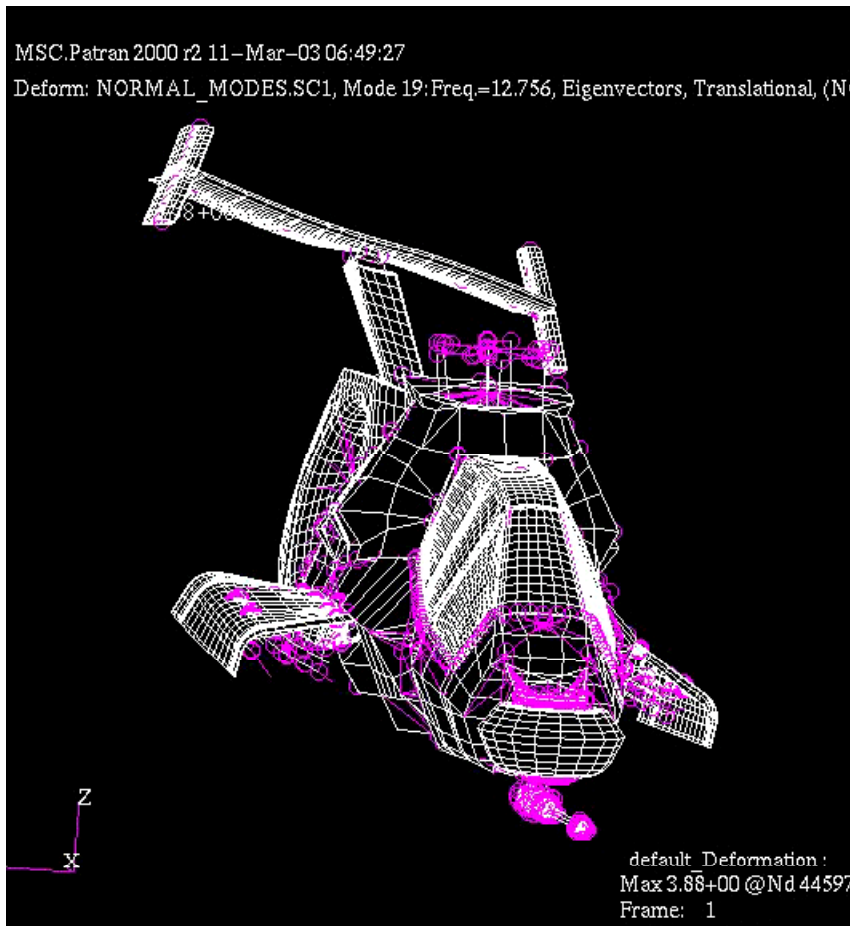


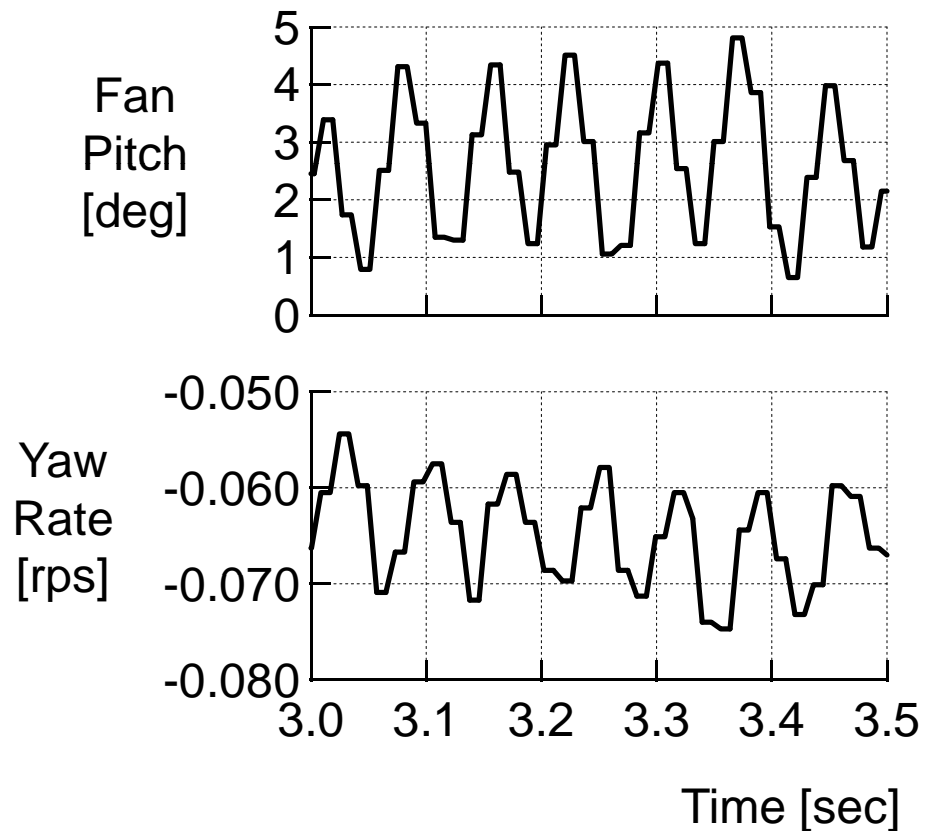
Fig. 1. Single-keel space station with solar dynamic system.

RAH-66 Comanche Aeroelastic Instability

Linear Instability With Actuator
Rate Limit \rightarrow Limit Cycle

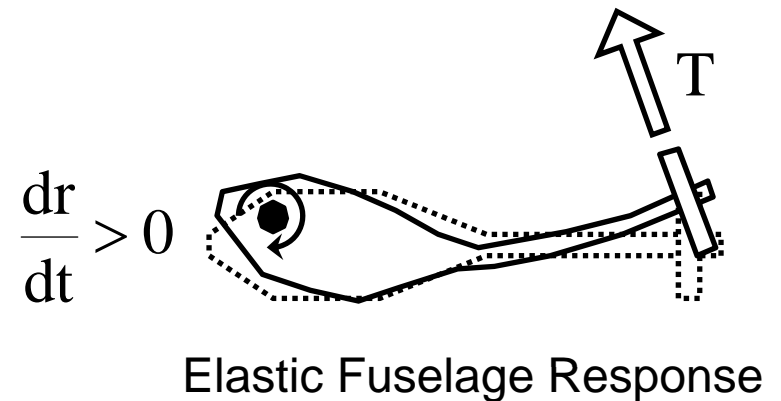
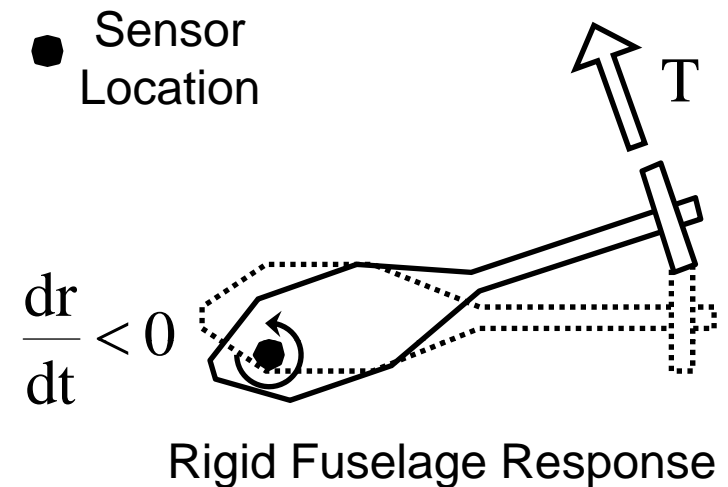


14-Hz Sustained Oscillation
90 Knots Descending Shallow Turn
MPFCS Mode - Yaw Rate Feedback Only

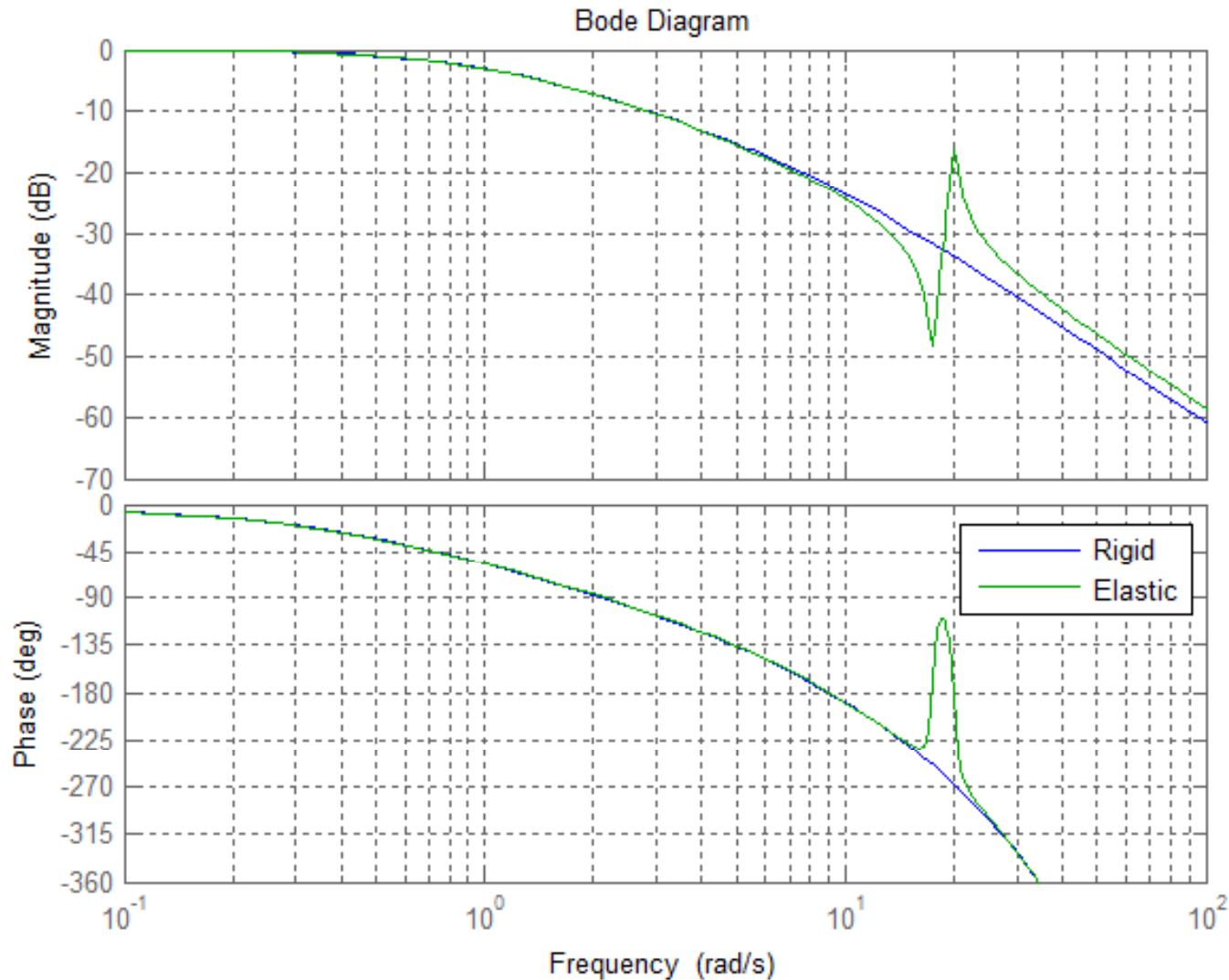


Simple Explanation of Feedback Instability

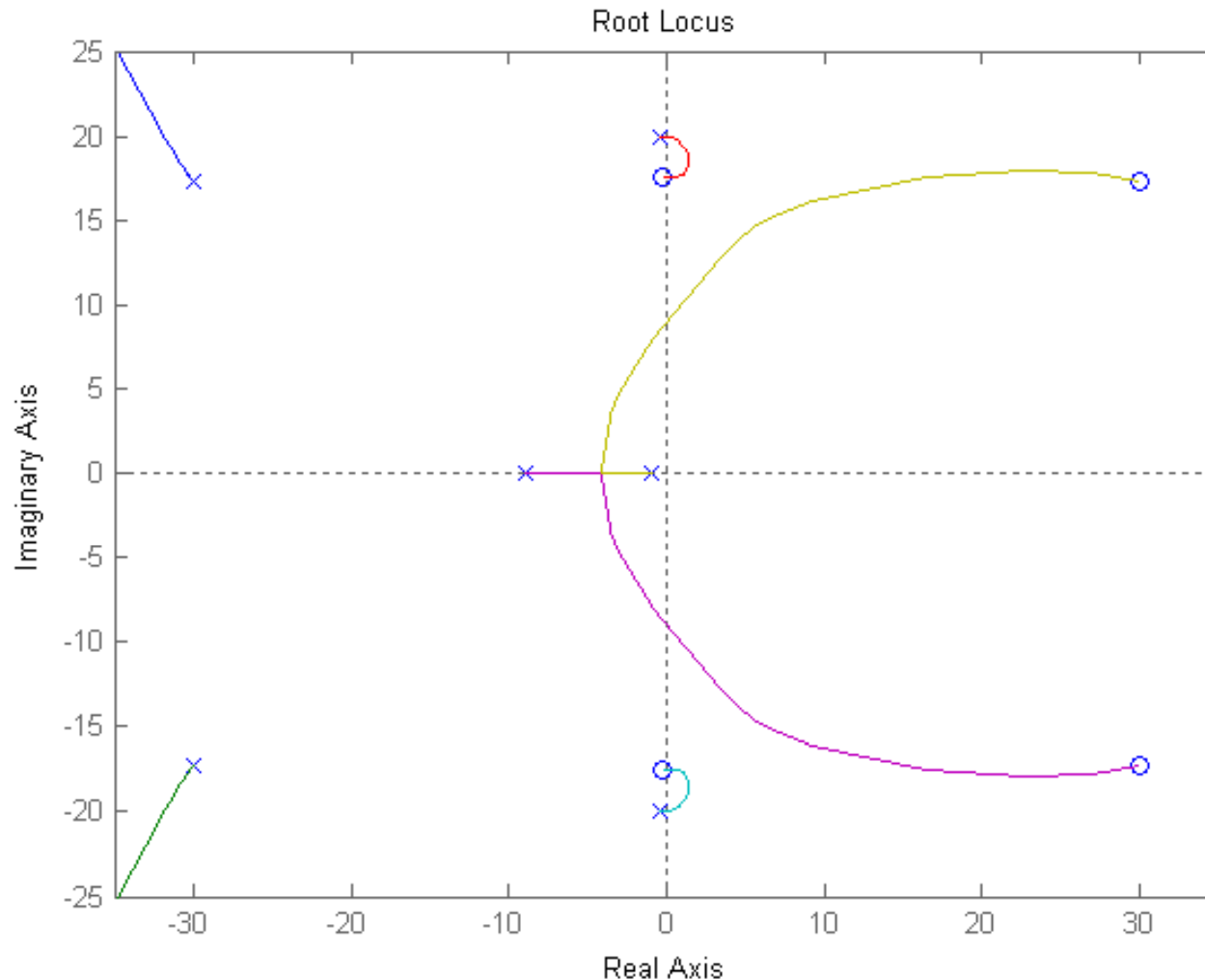
- Yaw Rate (r) Feedback to FANTAIL™ Thrust Required for Adequate Damping of Rigid-Body Modes
- Elastic Fuselage Response Can Change Sign of Response at High Frequency
- Effectively Positive Feedback Destabilizes Fuselage Elastic Mode
- Design: Linear Analysis Predicted Weaker Coupling



“Elastic”=Actuator + Rigid + Elastic Body Modes



Root Locus with Proportional Feedback

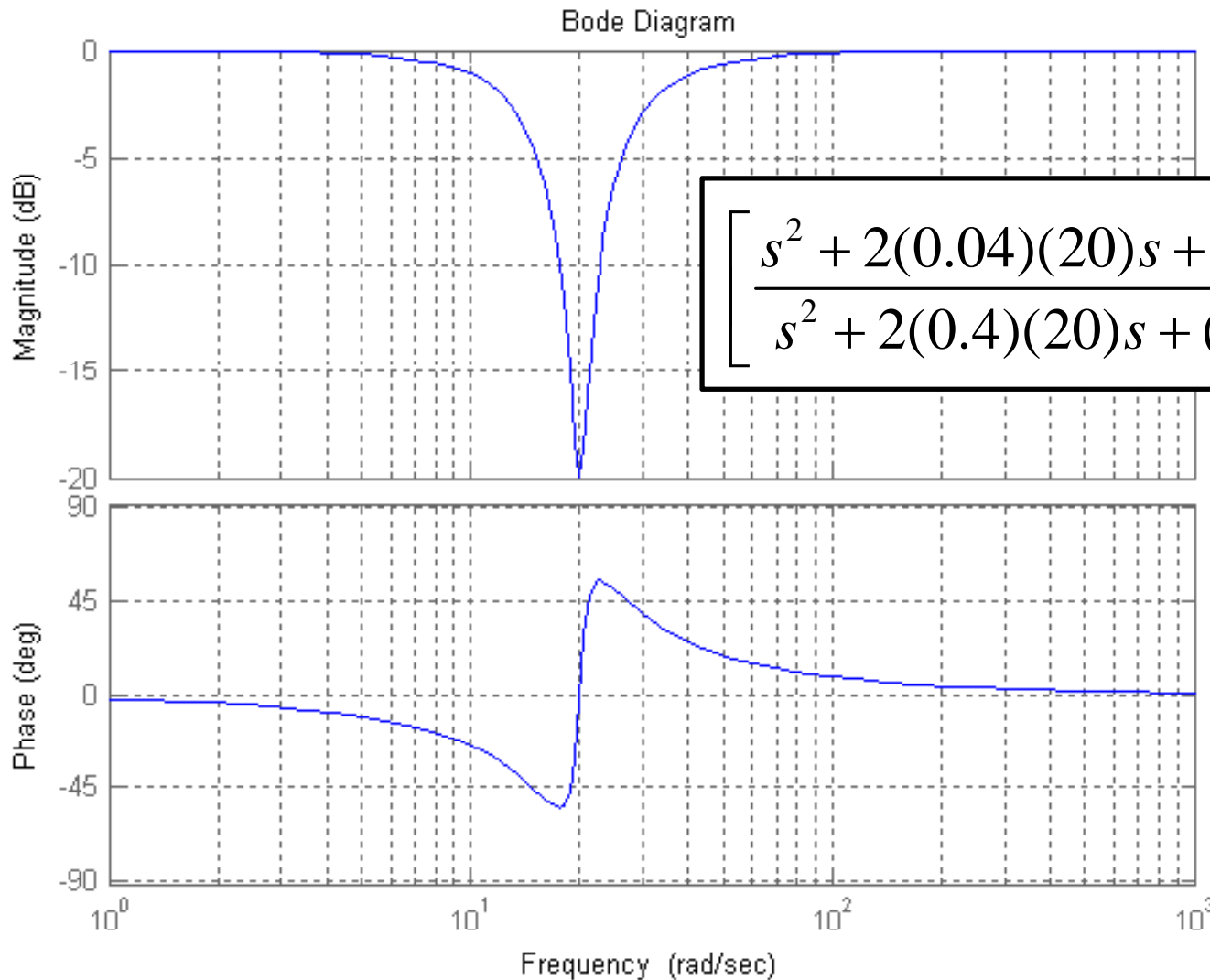


Elastic Modes
Unstable for
 $K > \sim 6$

Rigid Body Modes
Unstable for
 $K > \sim 14$

We Need to Do
Something to
Manage Elastic
Modes!

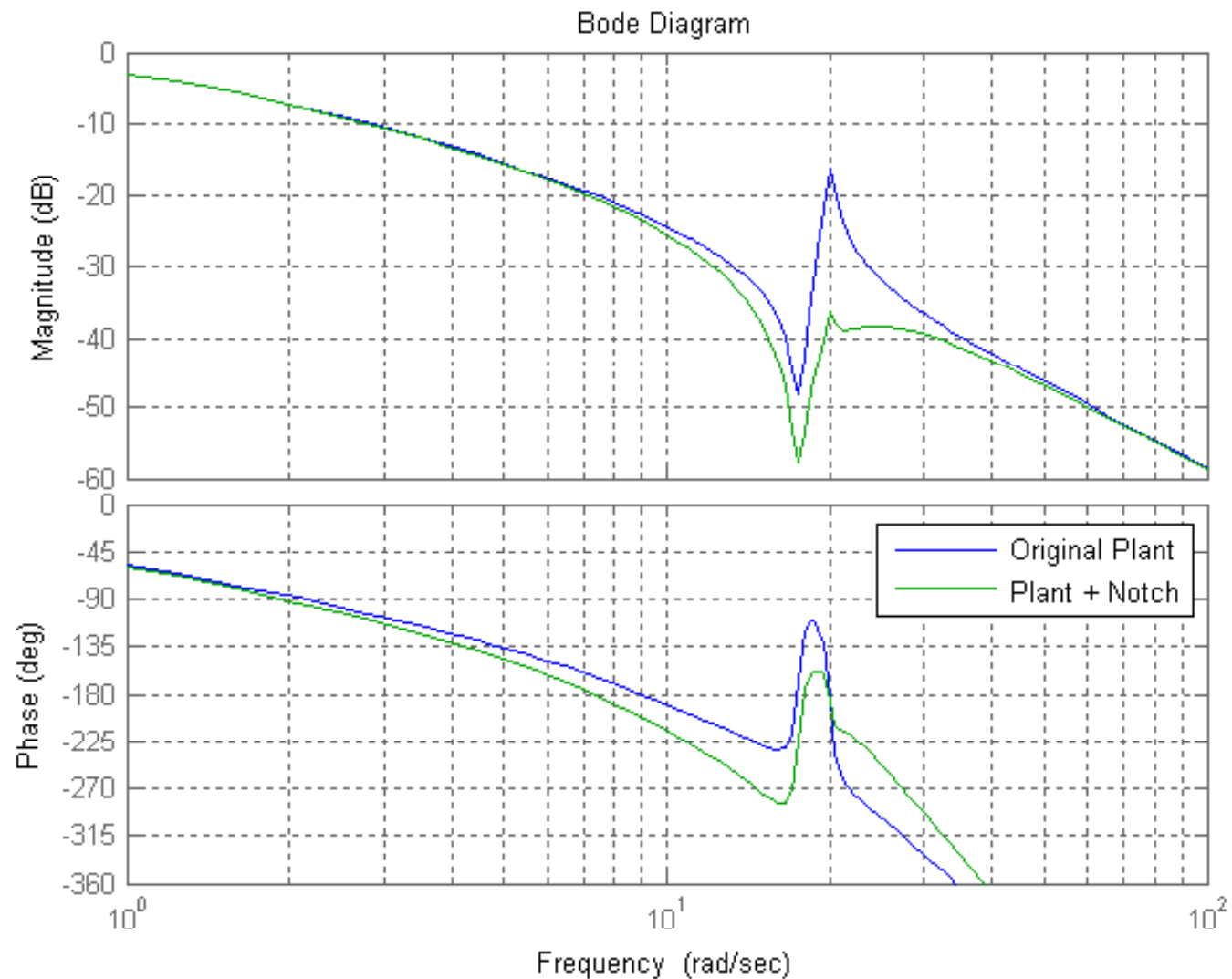
Notch Filter



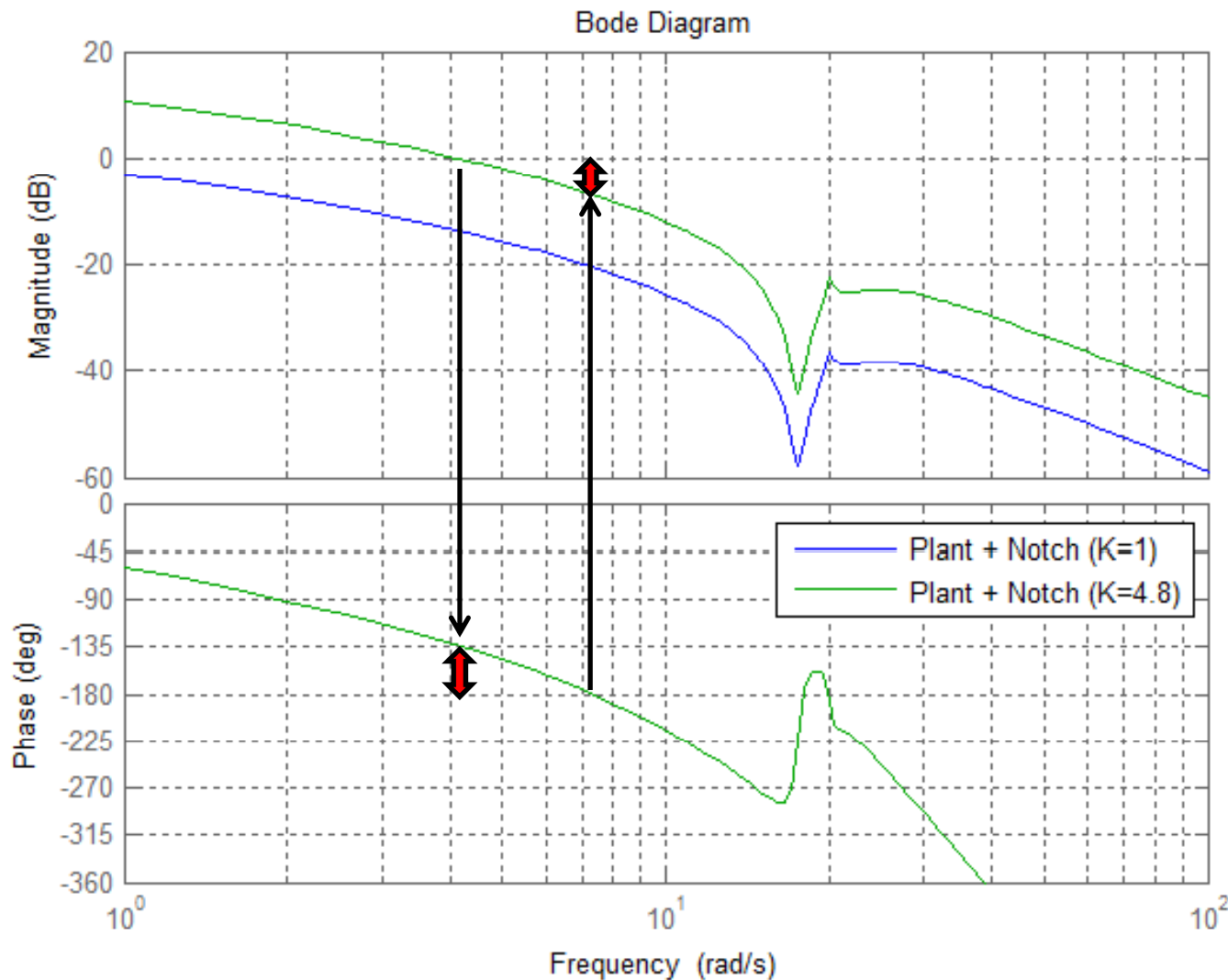
Start with a Deep
& Moderately
Wide Notch

(Conservative
Design)

Effect of Notch on Loop Bode Plot (K=1)



Loop Bode Plot with Notch (K=1 vs. K=4.8)

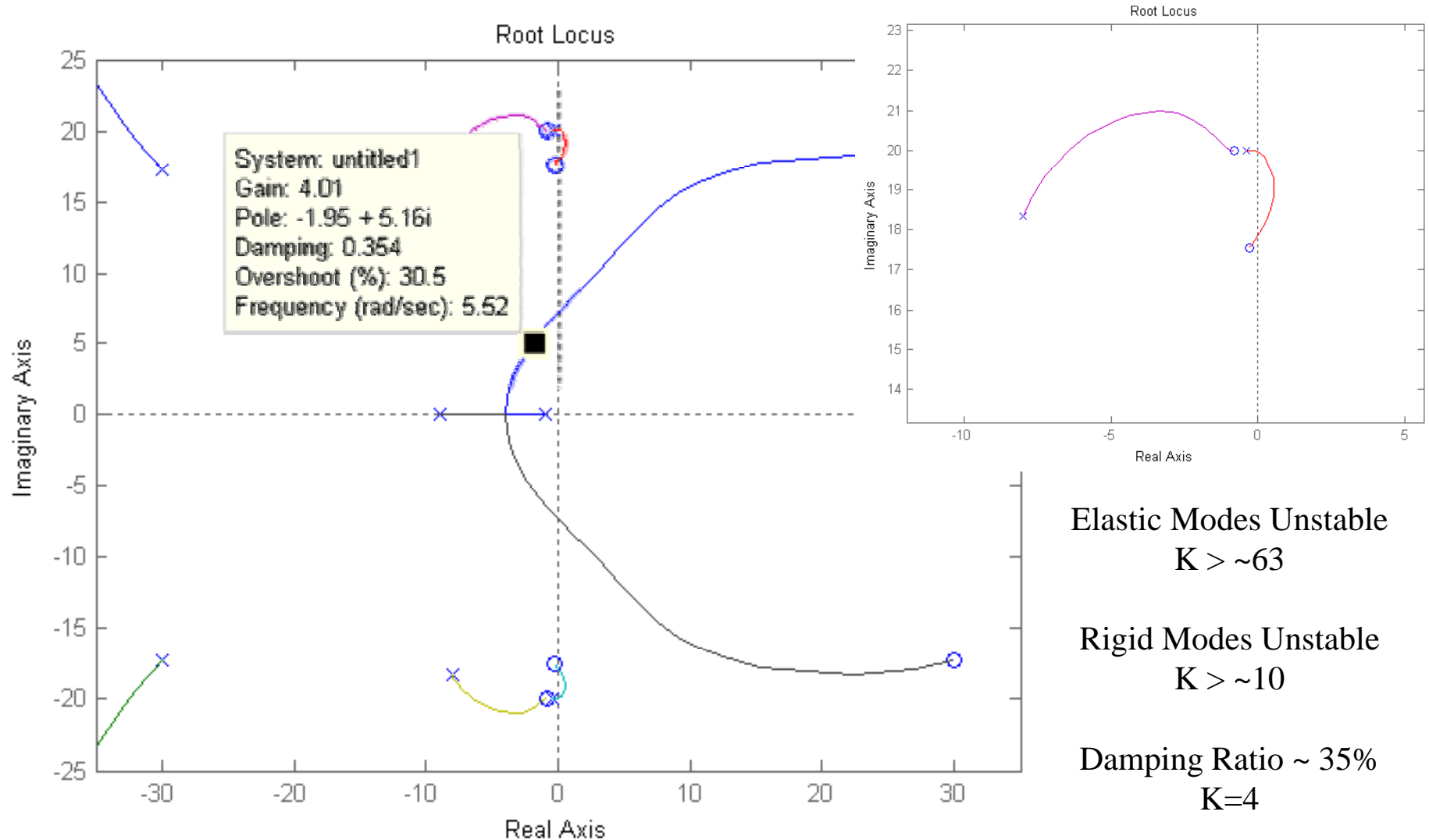


K=4.8

PM ~ 45 deg
@ 4.2 rps

GM ~ 6 dB
@ 7.3 rps

Effect of Notch on Root Locus



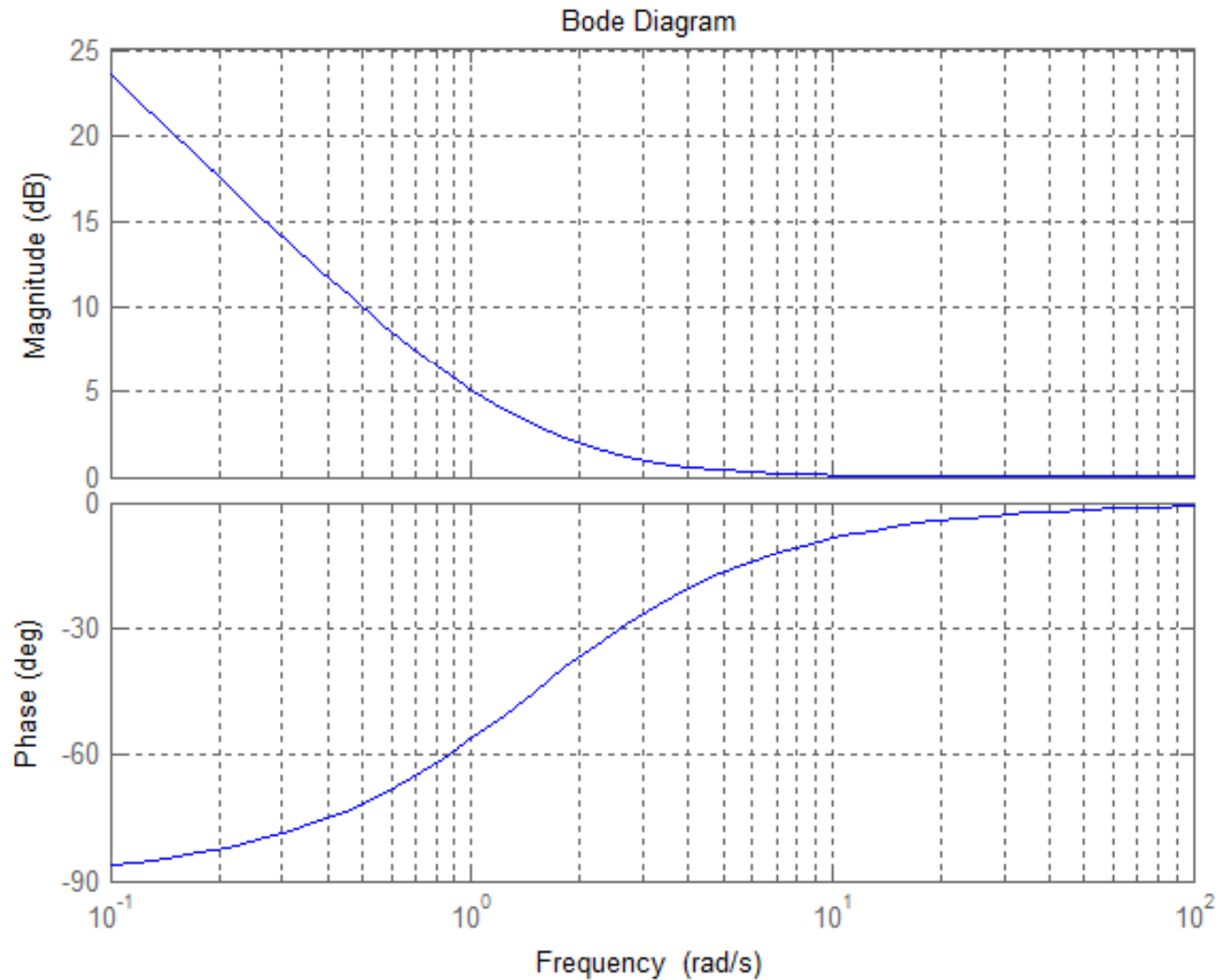
P+I = Special Form of “Lag Compensator”

- General Form of Lag Compensator ($z > p$)

$$G_c = K \frac{s + z}{s + p}$$

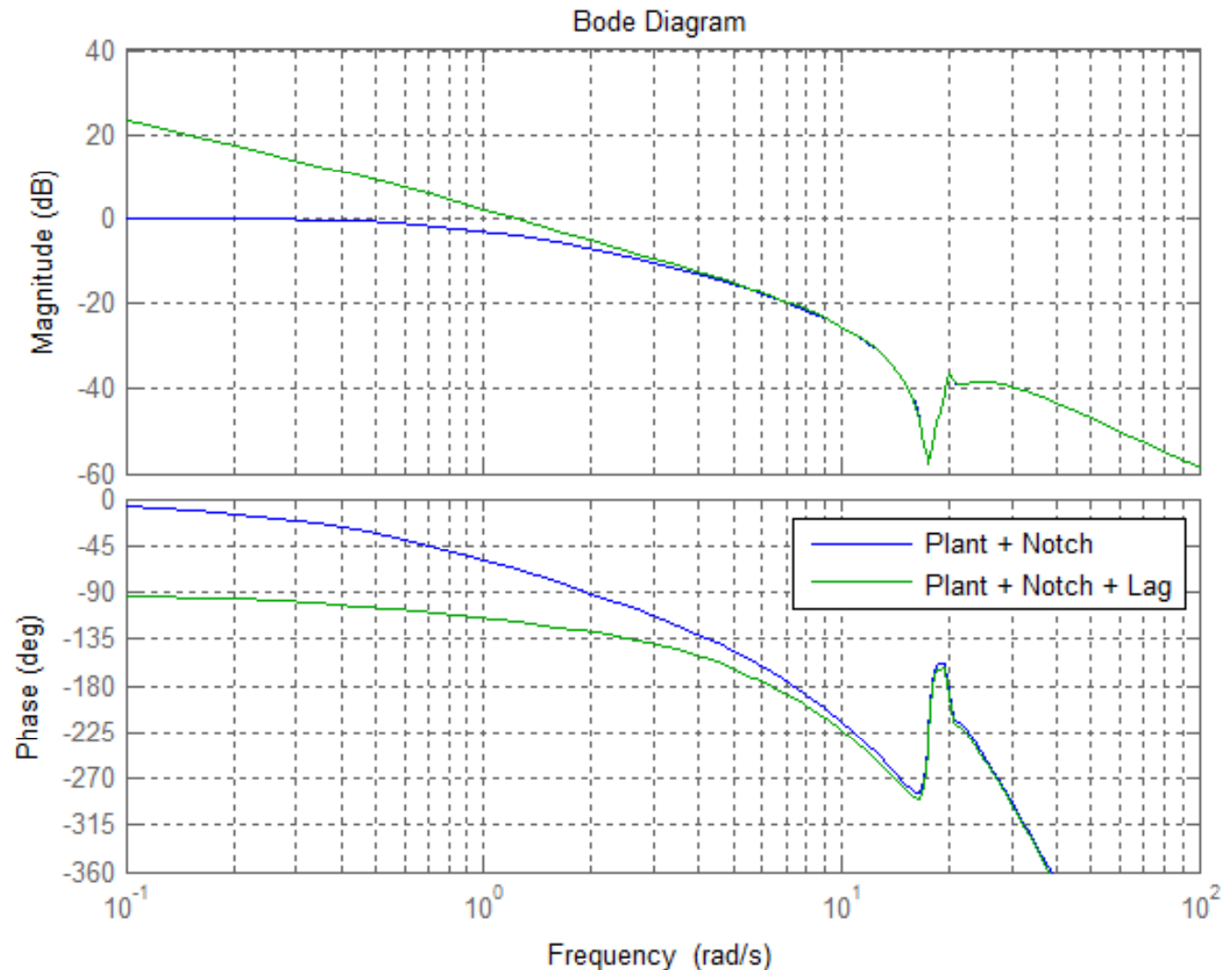
- P+I Compensator $\rightarrow p = 0$
- Note: Gain $\rightarrow K$ @ Frequencies $\gg z$
 - Design Crossover Region Without Lag Compensator
 - Add Lag Compensator with “ z ” Small Enough Not to “Mess Up” Bode Plot Near Crossover (Typically Want “ z ” Large for Tracking & Error Rejection Though)
 - Also Get Insight on Value of “ z ” From Root Locus

Bode Plot of Lag Filter ($z=1.5$, $p=0$)



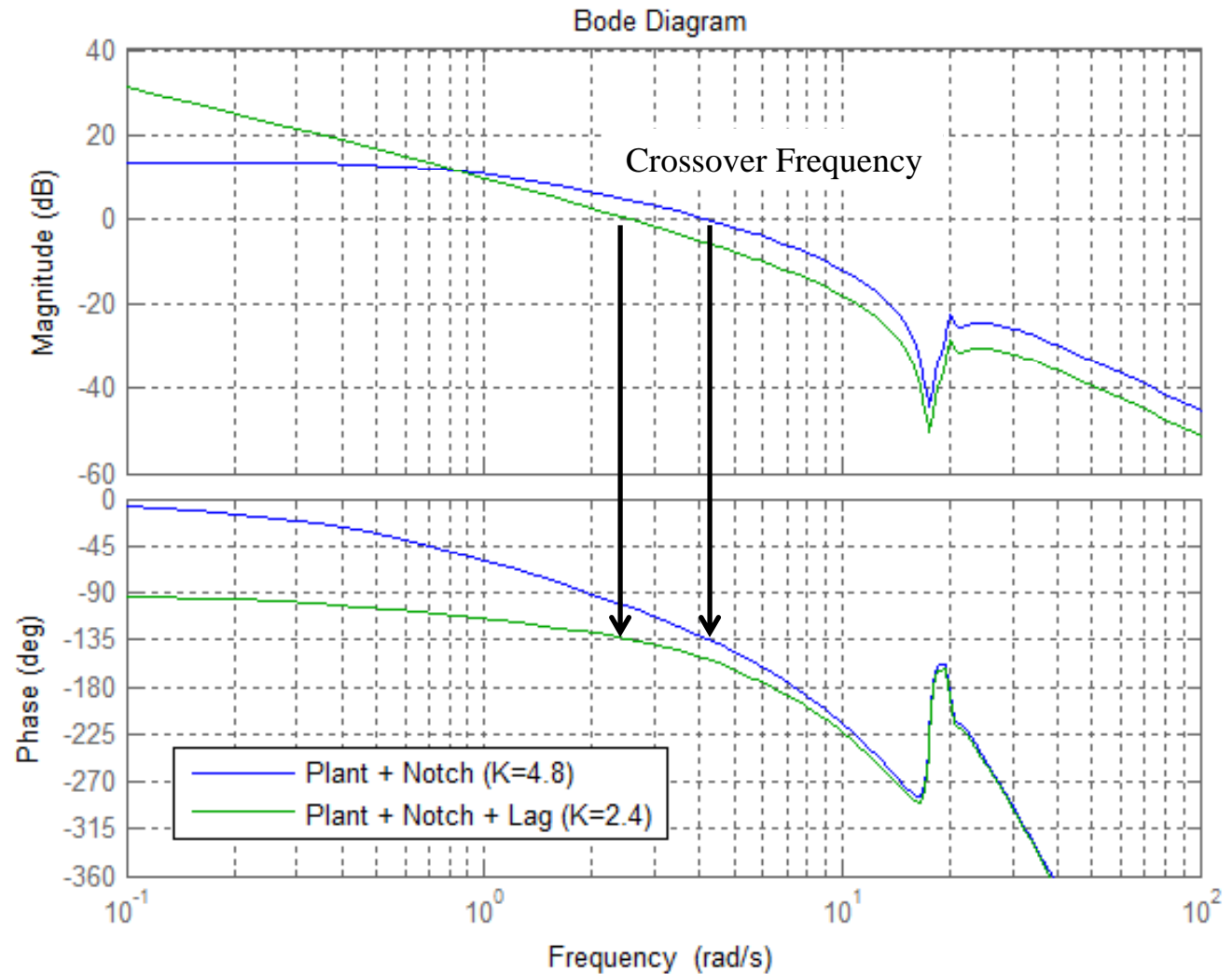
Notch vs. (Notch + Lag) (Both with K=1)

We “Pay For”
the Higher Gain
at Low
Frequency
Frequency
(Better Tracking
& Disturbance
Rejection) with
Lower Phase at
Moderate
Frequencies
(Limit on K)

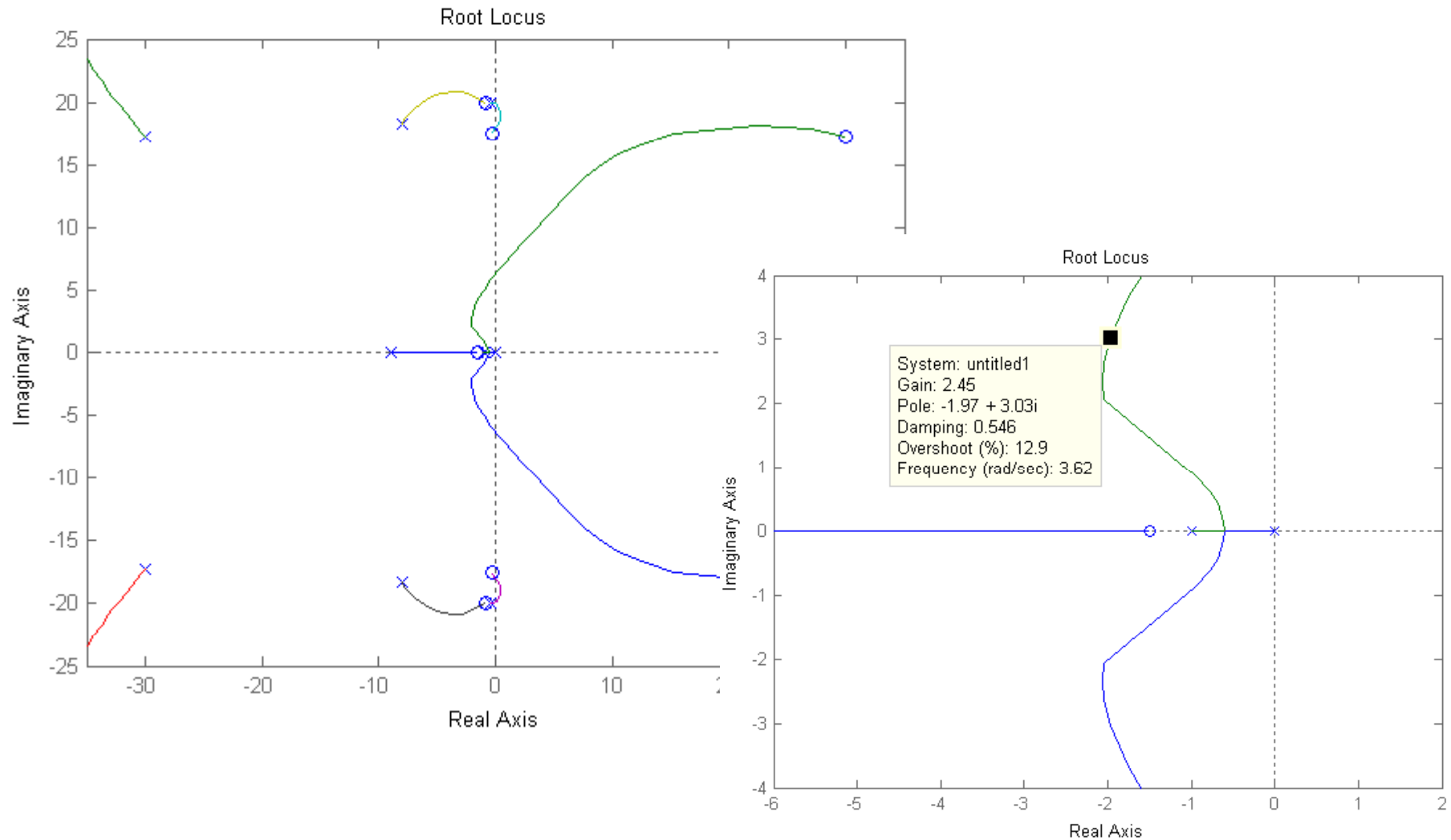


Notch vs. (Notch + Lag) (Both with PM=45 Deg)

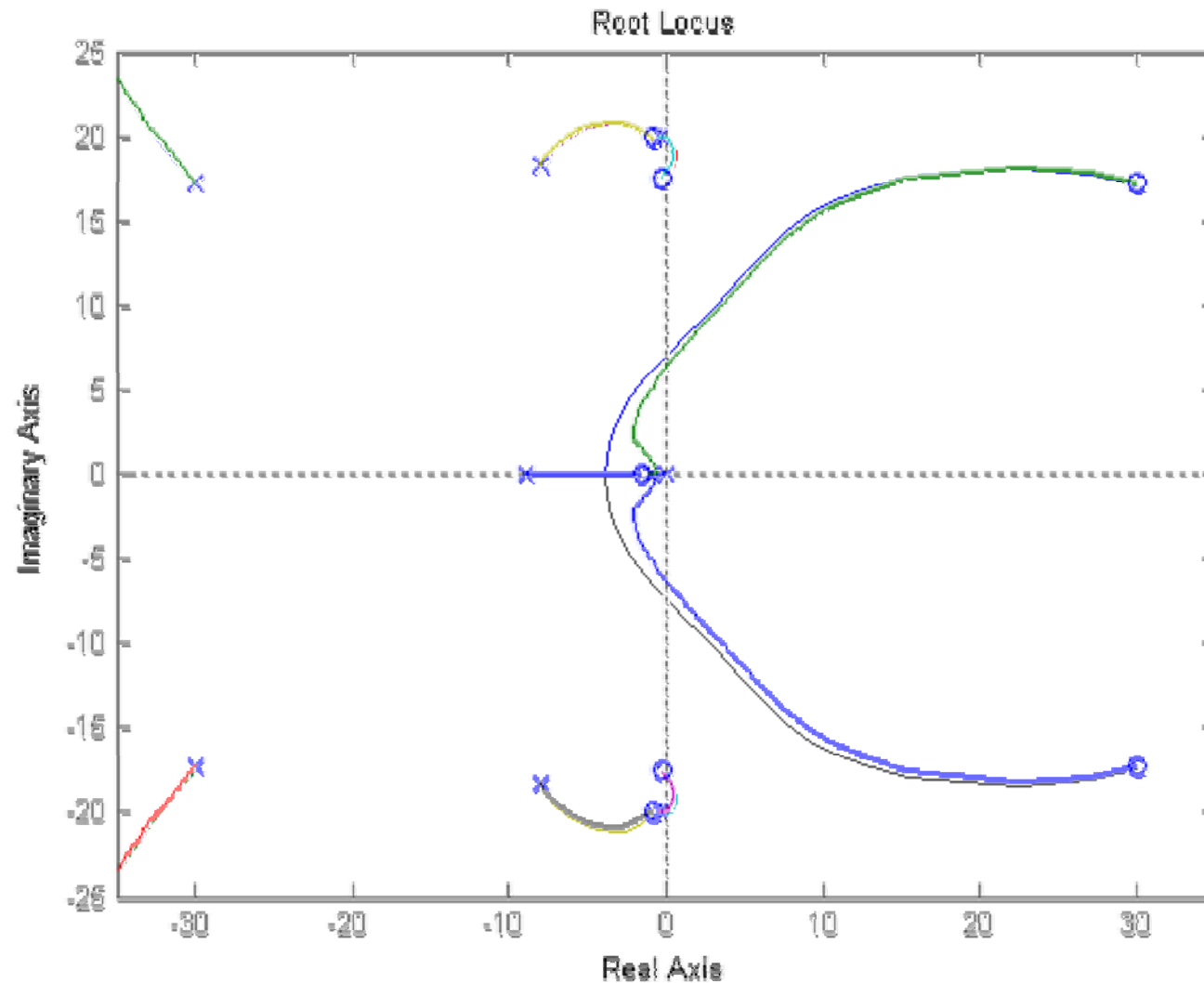
We “Pay For” the Higher Gain at Low Frequency (Better Tracking & Disturbance Rejection) with Lower Phase at Moderate Frequencies → Lower Crossover Frequency (~Bandwidth)



Root Locus with Lag Compensator ($z=1.5$, $p=0$)



Lag Compensator Design Difficult on Root Locus



Root Locus
with Lag
Compensator
Very Similar to
Root Locus for
Proportional
Feedback
(Overlay Here)

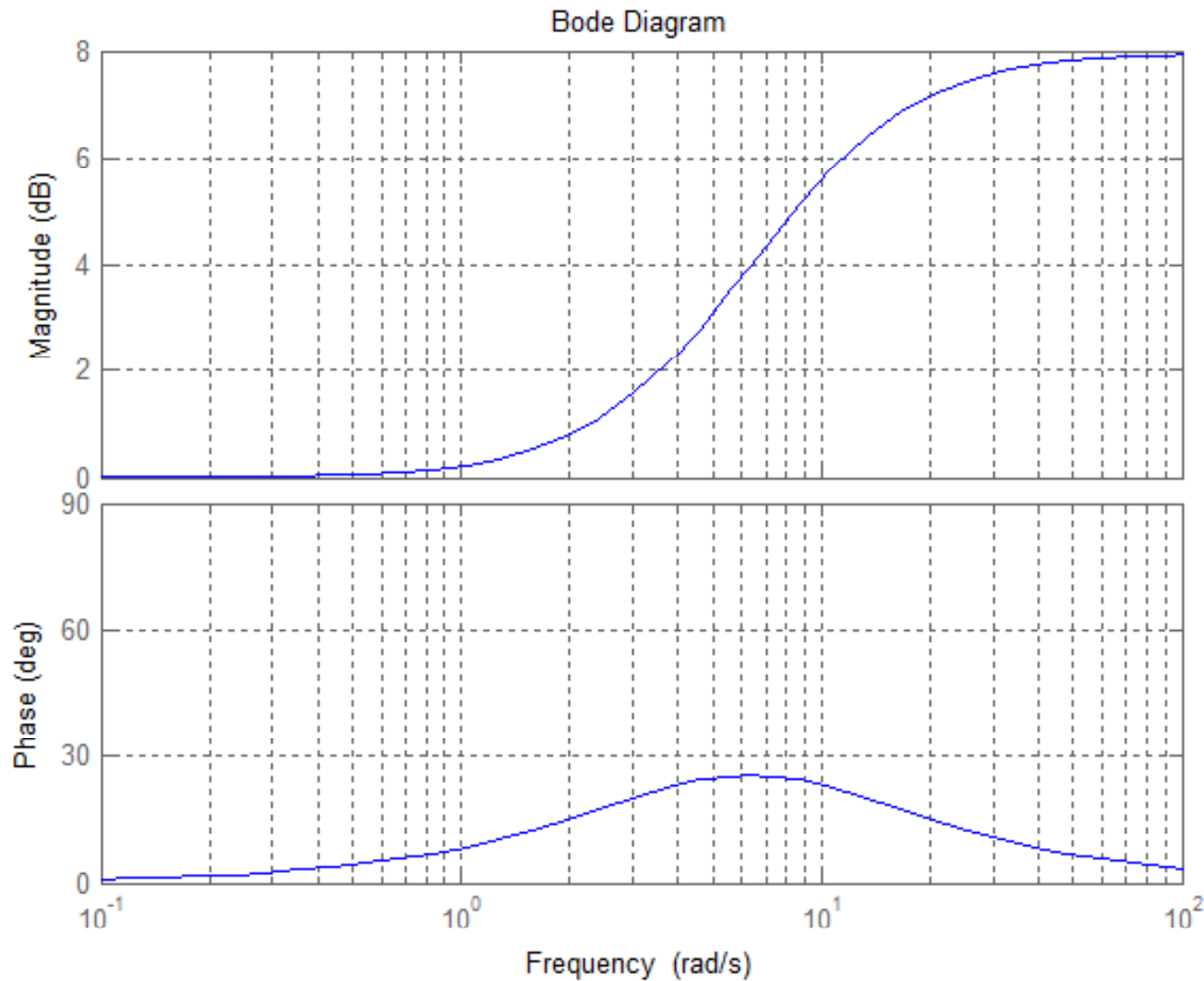
P+D = Special Form of Lead Compensator

- General Form of Lead Compensator ($z < p$)

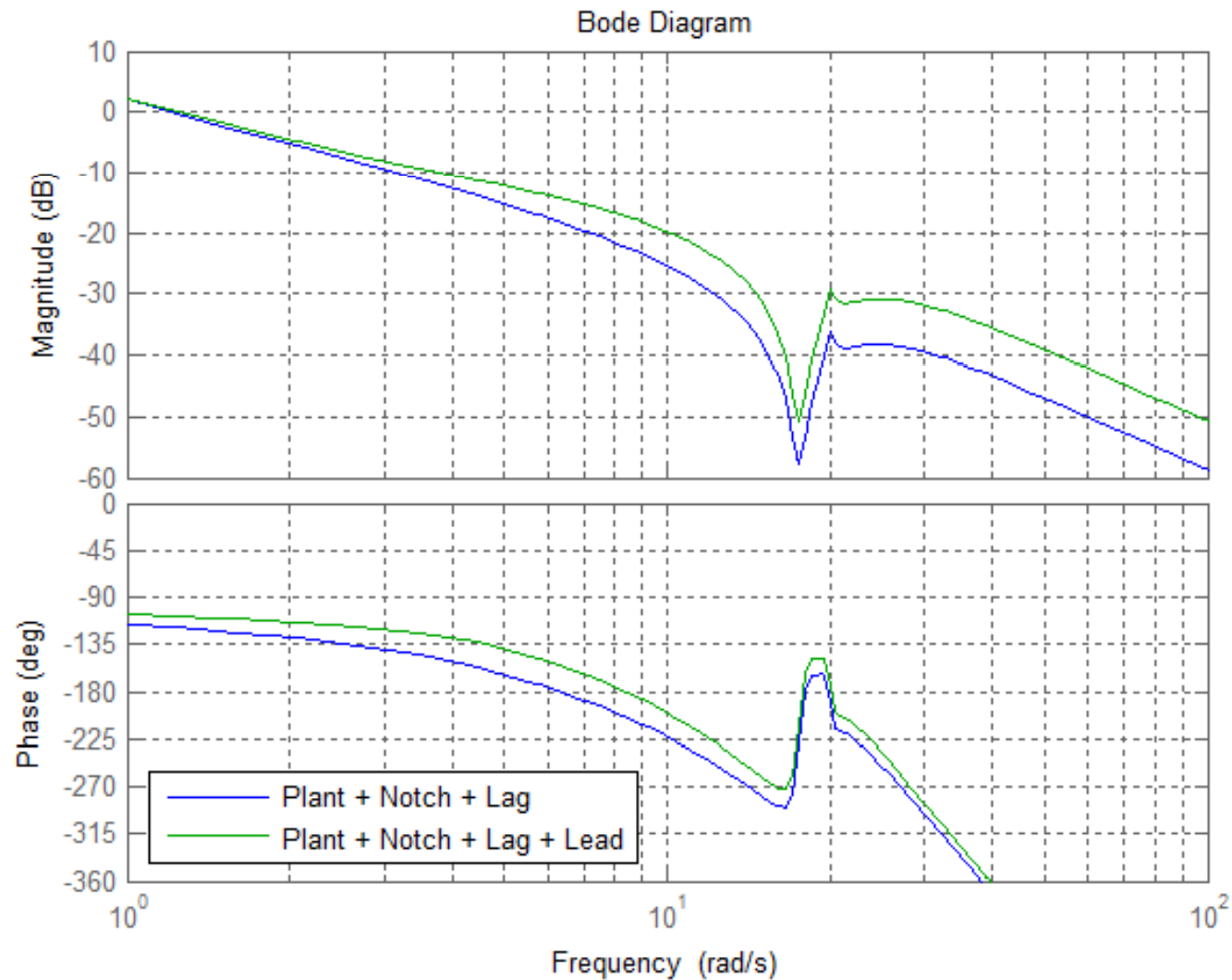
$$G_C = K \frac{\left(\frac{s}{z} + 1 \right)}{\left(\frac{s}{p} + 1 \right)}$$

- P+D Compensator $\rightarrow p \rightarrow \text{Infinity}$ (Usually Unrealistic)
- Note: Gain $\rightarrow K$ @ Frequencies $\ll z$
 - Design Low-Frequency (Tracking) Region Without Lead Compensator
 - Use Lead Compensator to Improve Stability Margins
 - Also Get Insight on Value of “ z ” From Root Locus

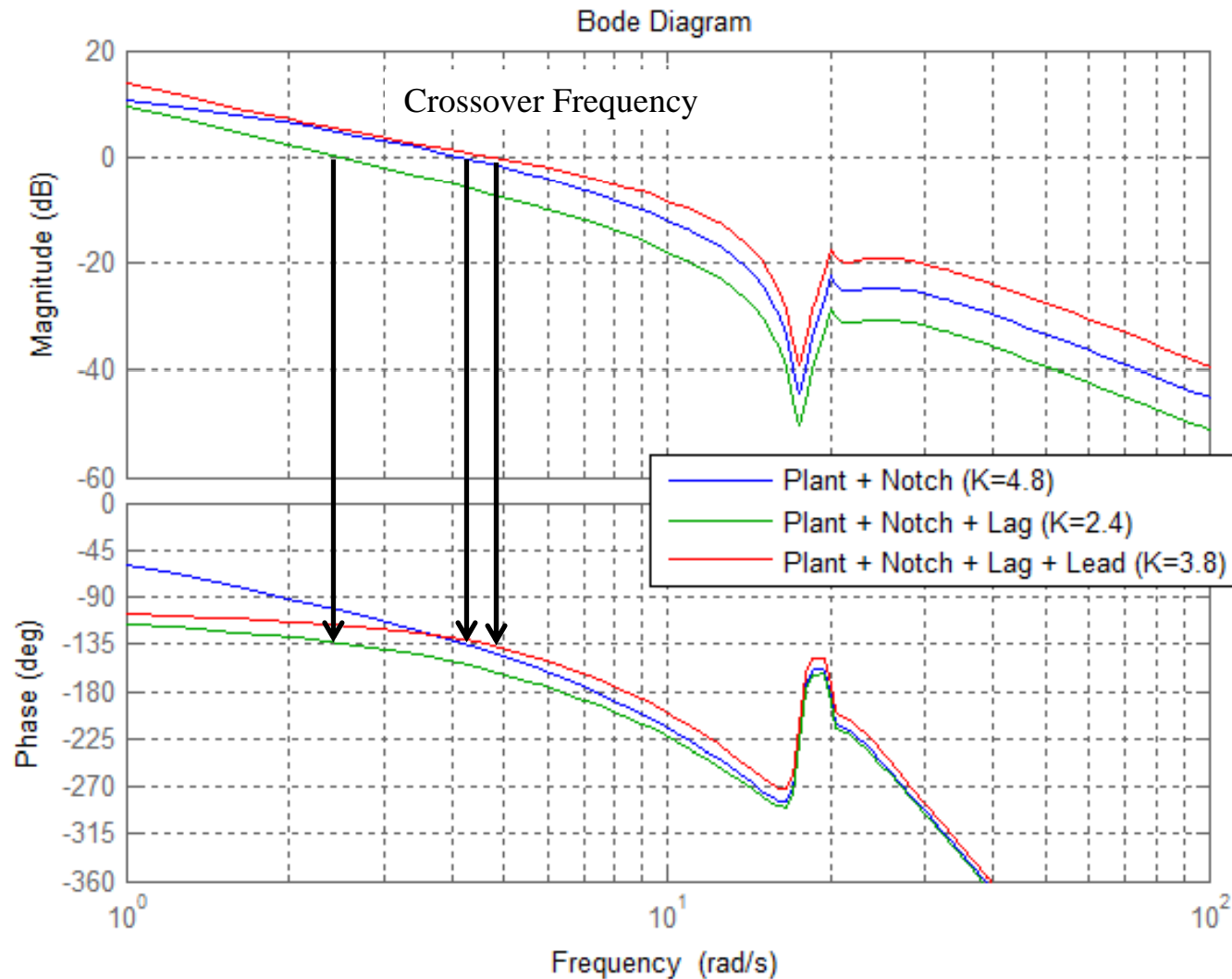
Bode Plot of Lead Filter ($z=4$, $p=10$)



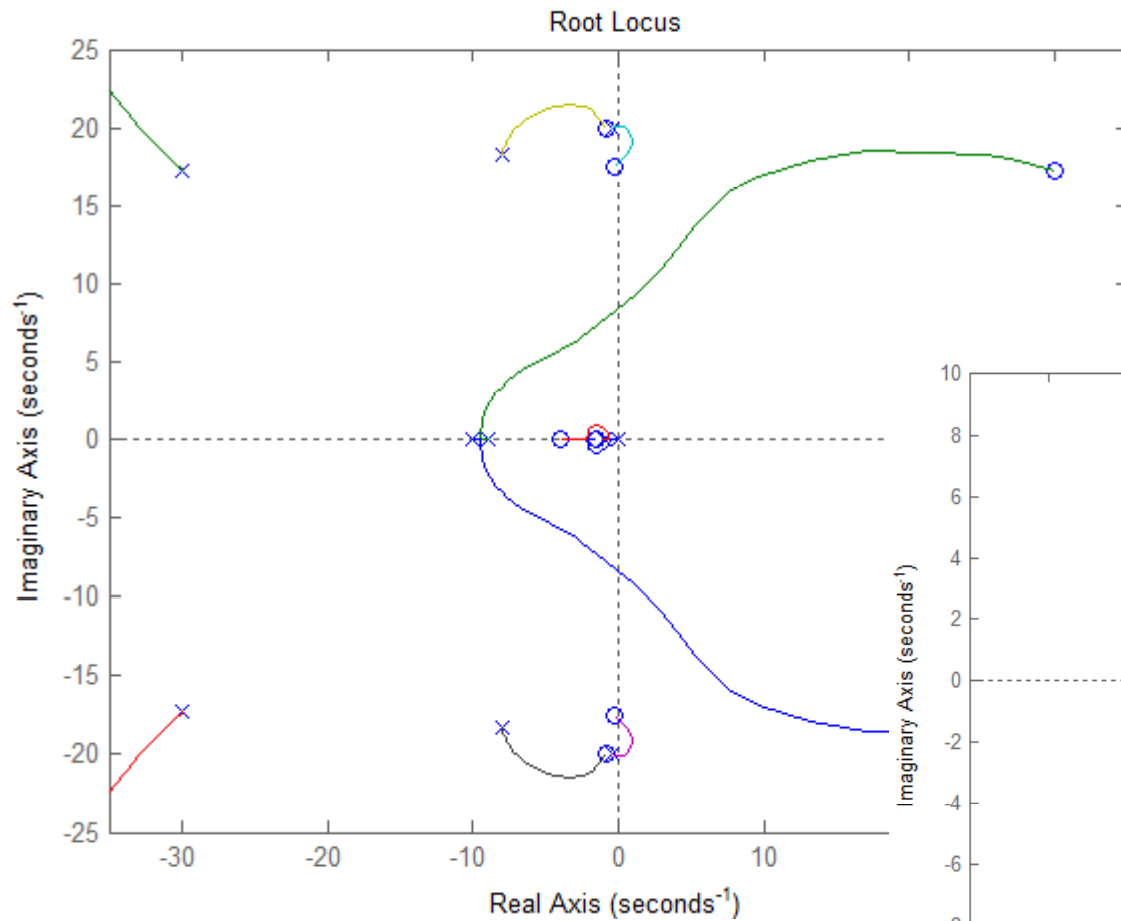
(Notch + Lag) vs. (Notch + Lag + Lead) ($K=1$)



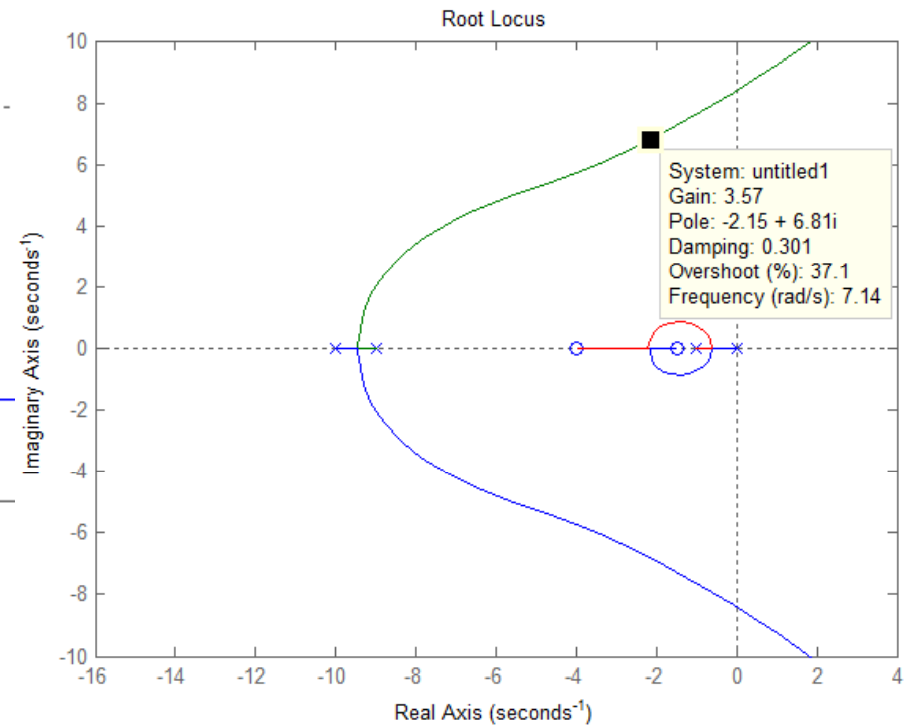
Compensators Compared (All with PM=45)



New Root Locus (Notch + Lag + Lead)



Gain Limited by Closed-
Loop Damping Ratio
 $K \sim 3.6$

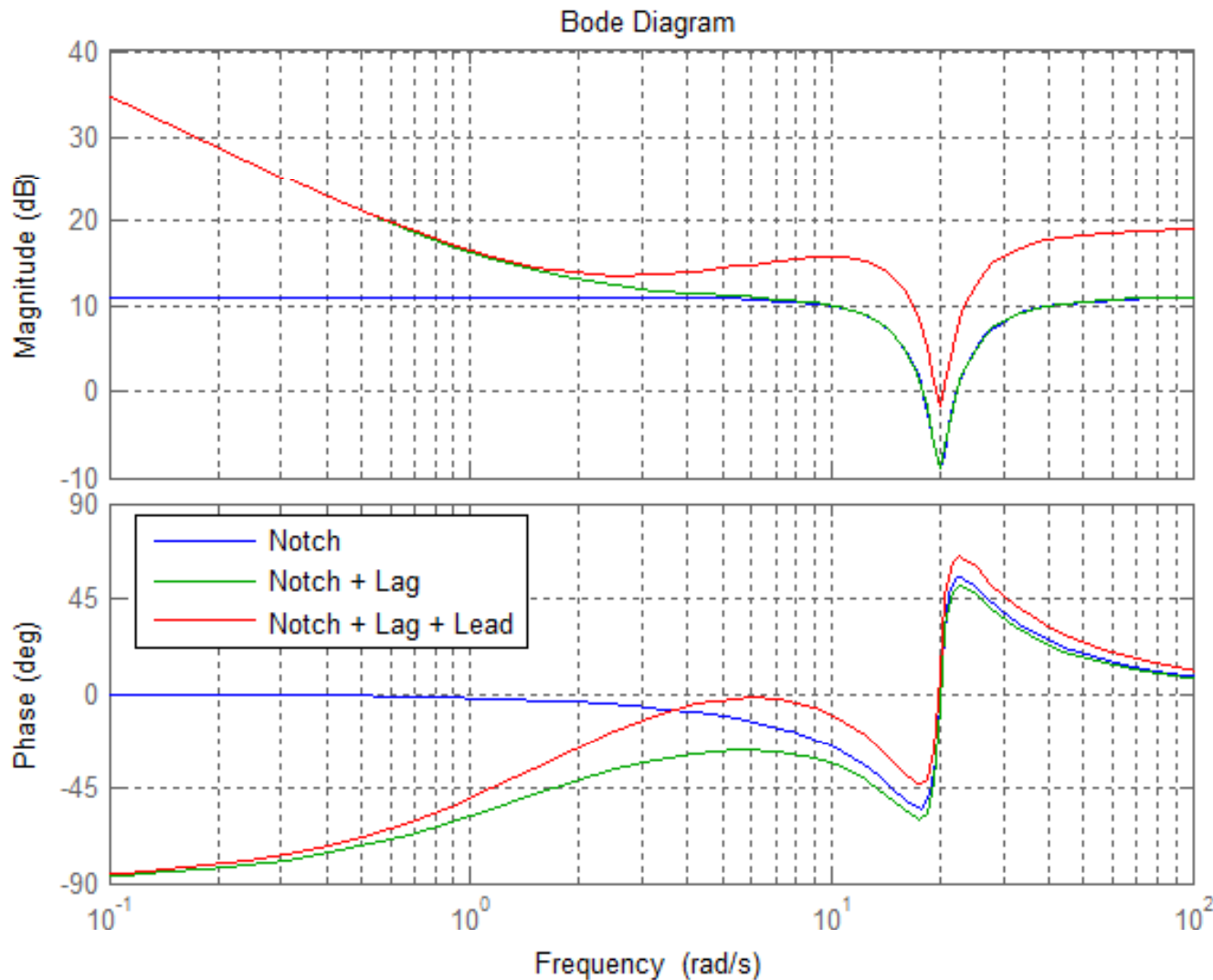


Final Form of Compensator

$$G_C = K \frac{\left(\frac{s}{z_{LEAD}} + 1 \right)}{\left(\frac{s}{p_{LEAD}} + 1 \right)} \frac{(s + z_{LAG})}{(s + p_{LAG})} \left(\frac{s^2 + 2\zeta_n \omega_n s + \omega_n^2}{s^2 + 2\zeta_d \omega_n s + \omega_n^2} \right)$$

- Notch Filter to Prevent Destabilization of Elastic Modes
- Lag Compensator (~PI) for Good Tracking & Disturbance Rejection @ Low Frequency
- Lead Compensator for Quick Response & Good Stability Robustness (Allows Higher KI Gain)

Recap of Compensator Design (All with $K=3.6$)



- Notch to Eliminate Elastic Modes = High Frequency Stability Robustness
- Lag Compensation (P+I) for Good Tracking & Disturbance Rejection @ Low Frequency
- Lead Compensation for Better Stability → Higher Bandwidth
- These Elements are Quite Typical of Many Control System Designs