## NAME \_\_\_\_\_\_ ESE406 - SPRING 2015 - Final EXAM CLOSED NOTES & CLOSED BOOK : NO CALCULATORS

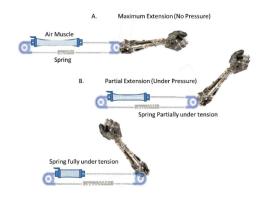
- Choose the one best answer for each question and enter it on your answer sheet.
- A correct answer is worth 3 points.
- No answer is worth 1 points.
- An incorrect answer is worth 0 points. Random guessing will lower your grade, on average.
- You must work completely independently.

## SUBMIT YOUR ANSWER SHEET BEFORE YOU LEAVE THE ROOM TAKE THIS EXAM WITH YOU AND SUBMIT "REVISED / COMPLETED" ANSWERS ONLINE FOR 1-POINT EACH

- 1.  $\frac{\left(1+j\right)}{e^{j\pi/4}}$  is equal to...
  - A.  $\sqrt{2}$
  - B.  $\sqrt{2}j$
  - C.  $\sqrt{2}e^{j3\pi/4}$
  - D.  $-\frac{\sqrt{2}}{2}$



- 2. The Parallax PING device, shown above, is an example of which type of system element?
  - A. Sensor
  - B. Actuator
  - C. Plant
  - D. Compensator
- 3. Lyapunov's Second Method...
  - A. ...is used to establish stability of equilibria of nonlinear dynamic systems.
  - B. ...is nice because it does not require an explicit solution of the ODE.
  - C. ...can be challenging because it requires the analyst to divine a suitable Lyapunov function (often some generalization of a total energy function).
  - D. All of the above.



- 4. The pneumatic device that moves the robotic arm in the figure above<sup>1</sup> is an example of what type of system element?
  - A. Sensor
  - B. Actuator
  - C. Plant
  - D. Compensator
- 5. An engineer uses a PD compensator,

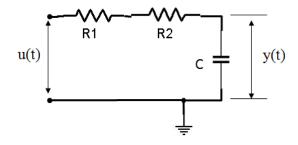
 $G_C(s) = K_P + K_D s$  to improve the stability of a closed-loop system. But she finds that derivative feedback creates undesired amplification of high-frequency noise in the system. So, she decides to low-pass filter the derivative feedback, using

$$G_{C}(s) = K_{P} + \frac{K_{D}s}{\tau s + 1}$$
 . This filtered PD

compensation is equivalent to...

- A. Lead Compensation
- B. Lag Compensation
- C. Notch Filter
- D. Bang-Bang Control

<sup>1</sup> http://sreal.eecs.ucf.edu/people/phd/rpillatproject8.php



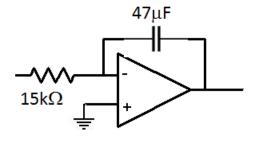
6. The ODE describing the circuit shown above is...

$$A. \quad C\frac{d^2y}{dt^2} + R_2y = R_1u$$

B. 
$$R_1 C \frac{d^2 y}{dt^2} + R_2 y = R_2 u$$

C. 
$$C \frac{dy}{dt} + (R_1 + R_2) y = (R_1 + R_2) u$$

D. 
$$(R_1 + R_2)C\frac{dy}{dt} + y = u$$



7. The op-amp circuit shown above has a transfer function given by...

A. 
$$\frac{1.4}{s}$$

B. 
$$\frac{1}{1.4s+1}$$

C. 
$$\frac{s}{1.4}$$

D. 
$$\frac{s-1.4}{s+1.4}$$

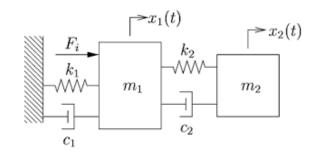
8. The Laplace Transform of  $[y(t)]^2$  ...

A. ...is equal to 
$$[Y(s)]^2$$

B. ...is equal to 
$$s^2Y(s)$$

C. ...is equal to 
$$\sqrt{Y(s)}$$

D. ...cannot generally be expressed simply in terms of Y(s).



9. For the system shown above, how many poles are in the transfer function  $\frac{X_2(s)}{F_1(s)}$ ?

A. 1

B. 2

C. 4

D. 8

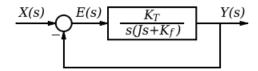
10. What transfer function corresponds to the differential equation  $RC \frac{dy}{dt} + y = RC \frac{du}{dt}$ ?

A. 
$$\frac{RCs}{RCs+1}$$

B. 
$$\frac{RC}{RC-s}$$

C. 
$$\frac{1}{s+RC}$$

D. We need initial conditions to determine a transfer function.



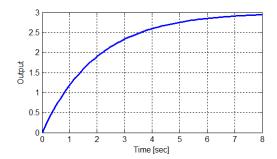
11. The figure above represents a position-control system for a DC motor. The closed-loop transfer function,  $\frac{Y(s)}{X(s)}$ , is...

A. 
$$\frac{1 - K_T}{Js^2 + K_f s}$$

B. 
$$\frac{K_T}{Js^2 + K_f s + K_T}$$

$$C. \quad \frac{K_T + 1}{Js^2 + K_f s + 1}$$

D. None of the Above



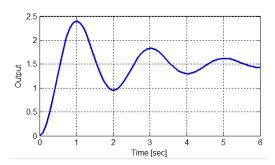
12. Which of the following transfer functions could MOST REASONABLY be thought to have the unit step response shown above?

A. 
$$\frac{2}{3s+1}$$

B. 
$$\frac{1}{s+3}$$

$$C. \quad \frac{3}{2s+1}$$

$$D. \quad \frac{3}{s+2}$$



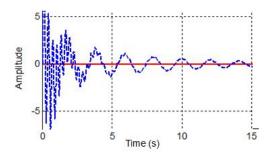
13. Which of the following transfer functions could MOST REASONABLY be thought to have the unit step response shown above?

A. 
$$\frac{3}{s+2}$$

$$B. \quad \frac{3}{s^2 + 2}$$

B. 
$$\frac{3}{s^2 + 2}$$
C.  $\frac{15}{s^2 + s + 10}$ 

$$D. \quad \frac{6}{s^2 + s + 4}$$



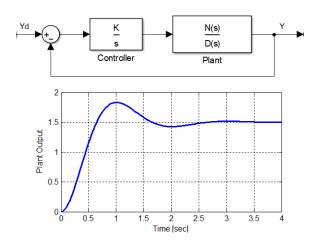
14. The figure above<sup>2</sup> shows the response of an electromechanical system to a momentary disturbance input at t=0. There are clearly two modes in the response. The mode with lower frequency corresponds MOST NEARLY to a pole located in the complex plane at...

A. 
$$1\pm 3j$$

B. 
$$-0.1 \pm 3j$$

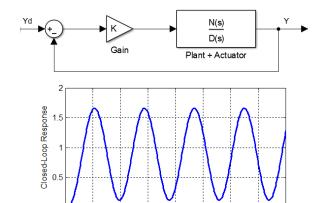
C. 
$$-3 \pm j$$

D. 
$$-3 \pm 0.1 j$$



- 15. The figure above shows the open-loop unit step response for the *plant only*. Which of the following is MOST ACCURATE about the use of Integral Feedback, as shown in the figure above, for this plant?
  - A. There will be a value of  $K_u > 0$  such that the closed-loop system will be unstable for  $K > K_{u}$ .
  - B. If the closed-loop system is stable, the steady-state error will be zero.
  - Both A and B.
  - D. Neither A nor B.

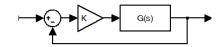
<sup>&</sup>lt;sup>2</sup> Hierarchical Fuzzy Control, By Carlos André Guerra Fonseca, Fábio Meneghetti Ugulino de Araújo and Marconi Câmara Rodrigues

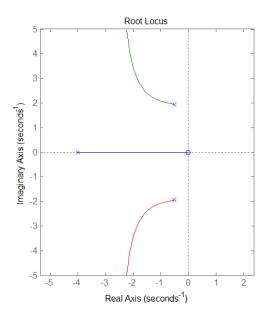


- 16. The figure above shows the closed-loop unit step response for the system shown in the block diagram, for some value of the proportional gain, *K*. The "Plant + Actuator" is stable open-loop. Which of the following is MOST ACCURATE about this system?
  - A. The chosen value of K results in neutral stability of the closed-loop system.
  - B. The system would be closed-loop stable for some smaller value of K.
  - C. Adding derivative feedback would likely be helpful in achieving closed-loop stability.
  - D. All of the above.

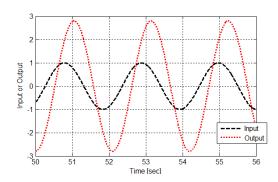
Ziegler–Nichols method <sup>[1]</sup>			
Control Type	$K_p$	$K_i$	$K_d$
Р	$0.5K_u$	-	-
PI	$0.45K_u$	$1.2K_p/T_u$	-
PD	$0.8K_u$	-	$K_pT_u/8$
classic PID <sup>[2]</sup>	$0.60K_{u}$	$2K_p/T_u$	$K_pT_u/8$

- 17. Given that K = 10 was the gain used in the previous problem, if we replaced proportional control with a PD controller, using the Ziegler-Nichols rules shown above, what would be the transfer function of the new controller?
  - A. 1.8s + 8
  - B. 8s + 1.8
  - C.  $1.8 + \frac{8}{s}$
  - D.  $\frac{s+1.8}{s+8}$

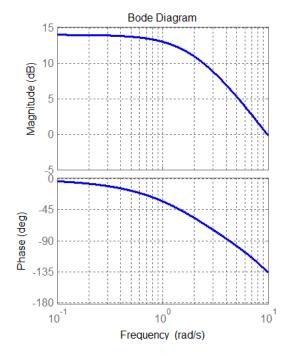




- 18. Which of the following is the MOST ACCURATE inference about the damping ratio of the oscillatory poles in the root locus shown above?
  - A. The maximum closed-loop damping ratio is about 0.15.
  - B. The maximum closed-loop damping ratio is about 0.55.
  - C. The maximum closed-loop damping ratio is about 0.95.
  - D. There is no maximum damping ratio of the closed-loop poles, because they go to infinity as the gain goes to infinity.
- 19. The MOST ACCURATE inference from the root locus of the previous problem about the effect of large gain is...
  - A. The closed-loop system will have zero steady-state error when the gain is large.
  - B. The closed-loop system will be stable regardless of how much the gain is increased.
  - C. The closed-loop system is unstable for gain larger than about 2.2.
  - D. None of the above is reasonable.



- 20. The figure above shows the response of a system to a sinusoidal input after the transient response has decayed to very nearly zero. The frequency of the input is...
  - A. About 0.5 rad/sec
  - B. About 1.5 rad/sec
  - C. About 3.0 rad/sec
  - D. About 56 rad/sec
- 21. Continuing with the frequency response of the previous problem, the magnitude at this frequency is...
  - A. About -3 dB
  - B. About +3 dB
  - C. About +6 dB
  - D. About +9 dB
- 22. Continuing with the frequency response of the previous problem, phase at this frequency is...
  - A. About +45 degrees
  - B. About -45 degrees
  - C. About -90 degrees
  - D. About -180 degrees

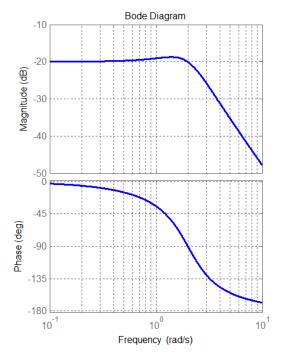


23. The bode plot shown above corresponds to a first-order system with time delay--that is, a

transfer function of the form  $\frac{Ae^{-Ts}}{\tau s+1}$  . The

MOST REASONABLE estimate of A is...

- A.  $A \approx 14$
- B.  $A \approx 9$
- C.  $A \approx 5$
- D.  $A \approx 1$
- 24. For the system in the previous problem, the MOST REASONABLE estimate of the time delay, T, is...
  - A.  $T \approx 0.1 \text{sec}$
  - B.  $T \approx 0.25 \text{ sec}$
  - C.  $T \approx 0.5 \text{sec}$
  - D. Since the bode plot does not cross -180 in the frequency range shown, the time delay is sufficiently small that it can be approximated as zero.



25. The bode plot shown above corresponds to a second-order system whose transfer function is

$$\frac{A}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$
 . The MOST

REASONABLE estimate of  $\omega_n$  is...

A. 
$$\omega_n = 1rps$$

B. 
$$\omega_n = 2rps$$

C. 
$$\omega_n = 4rps$$

D. 
$$\omega_n = 8rps$$

26. The MOST REASONABLE estimate of the damping ratio in the previous problem is...

A. 
$$\zeta = -0.2$$

B. 
$$\zeta = 0.0$$

C. 
$$\zeta = 0.2$$

D. 
$$\zeta = 0.5$$

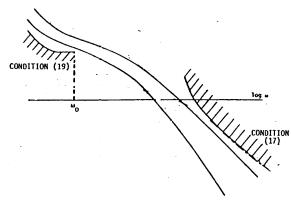
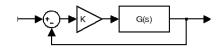


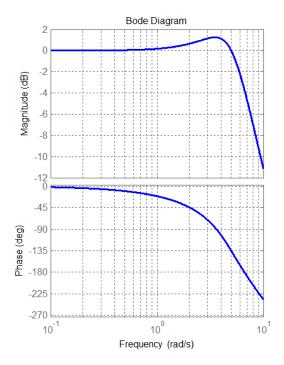
Fig. 3. The design tradeoff for GK.

- 27. The figure above<sup>3</sup> shows the design requirements for the loop transfer function frequency response magnitude as a function of frequency. The two curves in the center represent a "corridor" of acceptable designs. Based on ideas we discussed in class, the boundary denoted by "Condition (19)" at low frequency is MOST LIKELY...
  - A. ...to ensure that sensor noise is easily visible in the closed-loop response.
  - B. ...to ensure that the closed-loop system is stable.
  - C. ...to ensure good tracking and disturbance rejection.
  - D. All of the above.
- 28. Continuing with the figure in the previous question, based on ideas we discussed in class, the boundary denoted by "Condition (17)" at high frequency is MOST LIKELY...
  - A. ...to ensure adequate noise rejection.
  - B. ...to ensure zero steady-state error.
  - C. ...to ensure closed-loop damping ratio greater than 0.5.
  - D. All of the above.

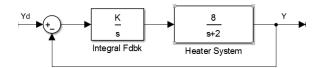
widely read. The application here is simplified and not totally accurate, but still reasonable.

<sup>&</sup>lt;sup>3</sup> John C. Doyle & Gunter Stein, "Multivariable Feedback Design: Concepts for a Classical/Modern Synthesis", 1981. This paper very important and widely read. The application here is simplified and





- 29. The bode plot shows KG(s) for the block diagram above with K = 5. Given that G(s) is open-loop stable, what is the gain margin of the system when K = 5?
  - A. There is no gain margin for this system, because the phase is always negative.
  - B. About 4 dB
  - C. About 12 dB
  - D. The system is closed-loop unstable with K = 5.
- 30. For the previous problem, what value of K results in neutral stability of the closed-loop system?
  - A.  $K \approx 0.6$
  - B.  $K \approx 3$
  - C.  $K \approx 8$
  - D.  $K \approx 20$
- 31. For the system of the previous 2 problems, what is the phase margin when K = 5?
  - A. About 45 degrees
  - B. About 90 degrees
  - C. About 180 degrees
  - D. This system has no phase margin, because the gain is 0dB at low frequency.

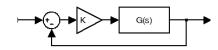


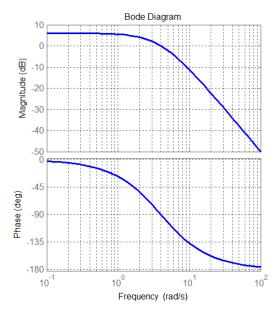
- 32. For the system shown above, approximately what value of K results in a 45-deg phase margin?
  - A.  $K \approx 0.7$
  - B.  $K \approx 2.8$
  - C.  $K \approx 12.6$
  - D. The phase margin is greater than 45 degrees for any K > 0.
- 33. In previous problem, suppose that gain chosen to provide 45 degrees of phase margin does not result in sufficiently large  $K_V$ . Which design change could be used to increase  $K_V$  while maintaining the phase margin?
  - A. Add lag compensation with a small reduction in K.
  - B. Replace pure integral feedback with P+I feedback,  $\left(G_C(s) = K_P + \frac{K_I}{s}\right)$ , and then use  $K_I > K$ , with whatever value of  $K_P$  is required to achieve the required phase margin.
  - C. Neither of the above.
  - D. Both of the above.
- 34. This problem is not related to the previous two problems. Using a lag compensator of the form

$$G_C(s) = K \frac{s+z}{s+p}$$
, an engineer chooses an

initial set of values for K, z > p, and p > 0. With this system, he finds that the steady-state error for a unit step input is 10 percent. What design change would reduce the steady error to zero, assuming that the closed-loop system remained stable?

- A. Increase K by a factor of 10.
- B. Set z = 0.
- C. Set p = 0.
- Any of the above would reduce the steady error to zero.





- 35. The graph shows the bode plot of G(s) for the block diagram shown above. For what value of K will the gain margin be approximately 6dB?
  - A.  $K \approx 0.5$
  - B.  $K \approx 2$
  - C.  $K \approx 8$
  - D. None of the other answers.
- 36. If the proportional feedback, K, of the previous problem is replaced with integral feedback,  $\frac{K}{s}$ , for what value of K will the gain margin be approximately 6dB?
  - A.  $K \approx 0.5$
  - B.  $K \approx 2$
  - C.  $K \approx 8$
  - D. None of the other answers.

37. Which of the following could be the *A* matrix in a state-space representation of a system that is governed by the equation

$$\frac{d^2y}{dt^2} + 4\frac{dy}{dt} - 5y = 6u$$
?

A. 
$$A = \begin{bmatrix} 0 & 1 \\ 5 & -4 \end{bmatrix}$$

B. 
$$A = \begin{bmatrix} 0 & 6 \\ 4 & -5 \end{bmatrix}$$

$$C. \quad A = \begin{bmatrix} 0 & -5 \\ 4 & 0 \end{bmatrix}$$

- D. This system has second-order derivatives and cannot be represented with the first-order state-space form.
- 38. Why did the previous question say "*could be* the *A* matrix" instead of "*is* the *A* matrix"?
  - A. The A matrix is not unique, because we can use a "change of basis" for the state vector, x = Tz, to find a new A matrix (which will have the same eigenvalues).
  - B. The professor was getting tired while writing exam questions and his use of the English language was a little sloppy.
  - C. We can't know if any given matrix is *the A* matrix until it is certified by the United Nations.
  - D. All of the above.
- 39. Which of the following is an eigenvalueeigenvector pair of the matrix  $A = \begin{bmatrix} -1 & 0 \\ 1 & -3 \end{bmatrix}$ ?

A. 
$$\lambda = 1$$
  $q = \begin{bmatrix} 1 \\ 2 \end{bmatrix}$ 

B. 
$$\lambda = 0$$
  $q = \begin{bmatrix} -1 \\ -3 \end{bmatrix}$ 

C. 
$$\lambda = -1$$
  $q = \begin{bmatrix} 2 \\ 1 \end{bmatrix}$ 

D. 
$$\lambda = -3$$
  $q = \begin{bmatrix} -1 \\ 2 \end{bmatrix}$ 

40. Suppose a system has the A matrix of the previous problem and has  $B = \begin{bmatrix} 2 \\ 1 \end{bmatrix}$ . Which of

the following is MOST ACCURATE about what can be achieved with control?

- A. Finite control can be used to move the state from an arbitrary initial value to an arbitrary final value in finite time.
- B. Proportional state feedback can be used to achieve an arbitrary closed-loop characteristic equation (pole placement).
- C. Both of the above
- D. Neither of the above
- 41. Consider a state-space system with the following matrices:

$$A = \begin{bmatrix} -2 & 0 \\ 0 & -4 \end{bmatrix}, B = \begin{bmatrix} 4 \\ 2 \end{bmatrix}, C = \begin{bmatrix} 1 & 1 \end{bmatrix}$$

What is the corresponding transfer function?

A. 
$$\frac{6s+18}{s^2+6s+9}$$

B. 
$$\frac{6s+20}{s^2+6s+8}$$

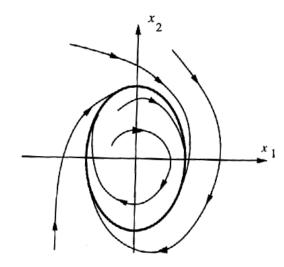
C. 
$$\frac{6s-18}{s^2-6s+9}$$

D. 
$$\frac{20}{s^2 + 6}$$

42. Which of the following is MOST ACCURATE about a linear state observer of the form

$$\frac{d\hat{x}}{dt} = A\hat{x} + Bu + L(y - \hat{y}), \text{ where } \hat{y} = C\hat{x} ?$$

- A. If the matrix pair (A, C) is observable, then the characteristic equation of the observer, whose roots are the eigenvalues of A-LC, can be chosen arbitrarily (observer pole placement).
- B. We typically implement the observer so that we can use the measurements (y) to obtain an estimate of the state, for use in feedback control, such as proportional state feedback,  $u = -K\hat{x}$ .
- C. Both of the above.
- D. Neither of the above.



- 43. Which of the following is MOST ACCURATE about the phase portrait, shown above, of a second-order dynamic system?
  - A. The system has 3 stable equilibrium points on the positive  $x_1$  axis.
  - B. The system has 4 unstable equilibrium points on the negative  $x_2$  axis.
  - C. The system has a limit cycle to which all trajectories converge.
  - D. All of the above.
- 44. If the dynamics of the system in the previous question are governed by  $\underline{\dot{x}} = f(\underline{x})$ , then...
  - A. ...the system is linear,  $\underline{f}(\underline{x}) = A\underline{x}$ , with A having real eigenvalues.
  - B. ...the system is linear,  $\underline{f}(\underline{x}) = A\underline{x}$ , with A having imaginary eigenvalues.
  - C. ...the system is nonlinear.
  - D. ...the system has a singularity at the origin, with  $\lim_{\underline{x} \to 0} \left| \underline{f}(\underline{x}) \right| \to \infty$