

RECAP $AQ = Q\Lambda$

$$Q = \begin{bmatrix} q_1 & q_2 & \dots & q_n \end{bmatrix} \quad \Lambda = \begin{bmatrix} \lambda_1 & & & \\ & \lambda_2 & & \\ & & \ddots & \\ & & & \lambda_n \end{bmatrix}$$

$$Aq_k = \lambda_k q_k \quad \text{RIGHT EIGENVECTORS}$$

$$Q^{-1}A = \Lambda Q^{-1}$$

$$Q^{-1} = \begin{bmatrix} p_1^T \\ p_2^T \\ \vdots \\ p_n^T \end{bmatrix}$$

$$p_k^T A = \lambda_k p_k^T \quad \text{LEFT EIGENVECTORS}$$

ODE $\dot{x} = Ax, \quad x(0) = x_0$

$$x(t) = Qz(t) \Rightarrow \dot{z} = \Lambda z, \quad z(0) = z_0 = Q^{-1}x_0$$

$$\dot{z}_k = \lambda_k z_k \quad z_k(0) = p_k^T x_0$$

$$z_k(t) = e^{\lambda_k t} z_k(0)$$

$$\Rightarrow x(t) = Q e^{\Lambda t} \underbrace{Q^{-1}x_0}_{z_0} = e^{At} x_0$$

$z(t)$

$$e^{At} = Q e^{\Lambda t} Q^{-1}$$

IN GENERAL, $f(t) \rightarrow f(At)$

$$f(At) = Q f(\Lambda t) Q^{-1}$$

$$= Q \begin{bmatrix} f(\lambda_1 t) & & & \\ & f(\lambda_2 t) & & \\ & & \ddots & \\ & & & f(\lambda_n t) \end{bmatrix} Q^{-1}$$

CAUTION: DOESN'T ALWAYS WORK TO FIND

$$\Lambda! \quad \text{EX/ } A = \begin{bmatrix} 2 & 1 \\ 0 & 2 \end{bmatrix} \text{ HAS}$$

ONLY $\lambda=2$, $q = \begin{pmatrix} 1 \\ 0 \end{pmatrix}$. JORDAN FORM
 e^{At} HAS $te^{\lambda t}$!

EXAMPLE $A = \begin{pmatrix} 2 & -2 & 1 \\ 0 & 1 & 8 \\ 0 & 0 & 3 \end{pmatrix}$ $Q = \begin{pmatrix} 2 & 1 & 3 \\ 1 & 0 & 4 \\ 0 & 0 & 1 \end{pmatrix}$ $Q^{-1} = \begin{pmatrix} 0 & 1 & -4 \\ 1 & -2 & 5 \\ 0 & 0 & 1 \end{pmatrix}$

SOLVE $\dot{x} = Ax$ with $x(0) = \begin{pmatrix} 1 \\ 4 \\ 1 \end{pmatrix}$ $z(0) = \begin{pmatrix} 6 \\ -2 \\ 1 \end{pmatrix}$

$$\Rightarrow x(t) = 0 \cdot e^t \begin{pmatrix} 2 \\ 1 \\ 0 \end{pmatrix} + (-2)e^{2t} \begin{pmatrix} 1 \\ 0 \\ 0 \end{pmatrix} + 1 \cdot e^{3t} \begin{pmatrix} 3 \\ 4 \\ 1 \end{pmatrix}$$

USE MATLAB'S "expm" FUNCTION TO CONFIRM $x(t) = e^{At} x_0$

A COUPLE MORE USEFUL THINGS

$$\Delta(\lambda) = |\lambda I - A| = \lambda^n + a_{n-1}\lambda^{n-1} + \dots + a_1\lambda + a_0$$

"CHARACTERISTIC POLYNOMIAL" $\Delta(\lambda_k) = 0$

$$\Delta(A) = A^n + a_{n-1}A^{n-1} + \dots + a_1A + a_0I$$

$$A^n = (Q\Delta Q^{-1})(Q\Delta Q^{-1}) \dots (Q\Delta Q^{-1}) = Q\Delta^n Q^{-1}$$

$$\Rightarrow \Delta(A) = Q\Delta(A)Q^{-1} = Q \begin{bmatrix} \Delta(\lambda_1) & & \\ & \Delta(\lambda_2) & \\ & & \ddots \\ & & & \Delta(\lambda_n) \end{bmatrix} Q^{-1} = 0!$$

ALSO TRUE FOR
JORDAN FORM
 $\neq \Delta$

CAYLEY-HAMILTON THEOREM: $\Delta(A) = 0$

TAYLOR SERIES (USE e^{tA} TO AVOID CONFUSION)

$$e^{tA} = I + tA + \frac{t^2}{2}A^2 + \dots$$

$$\text{BUT } A^n = -a_{n-1}A^{n-1} - a_{n-2}A^{n-2} - \dots - a_0I$$

CAN ELIMINATE ALL POWERS OF A BIGGER THAN $n-1$

LET'S THINK ABOUT CONTROL...

$$\dot{\underline{x}} = A\underline{x} + B\underline{u} \Rightarrow \underline{x}(t) = e^{tA} \underline{x}_0 + \int_0^t e^{(t-\tau)A} B\underline{u}(\tau) d\tau$$

$$\underline{x}(0) = \underline{x}_0$$

τ = "DUMMY VARIABLE"

"PROOF" OF $\underline{x}(t)$ AS SOLUTION.

$$\underline{x}(0) = \underline{x}_0 \checkmark$$

$$\dot{\underline{x}} = A e^{tA} \underline{x}_0 + e^{(t-\tau)A} B\underline{u}(\tau) \Big|_{\tau=t} + \int_0^t \frac{\partial}{\partial t} [e^{(t-\tau)A} B\underline{u}(\tau)] d\tau$$

$\frac{d}{dt} \rightarrow \frac{\partial}{\partial t}$
BECAUSE τ IS FIXED

$$= A e^{tA} \underline{x}_0 + B\underline{u}(t) + \int_0^t A e^{(t-\tau)A} B\underline{u}(\tau) d\tau$$

$$= A\underline{x}(t) + B\underline{u}(t) \checkmark$$

"DEFINITION OF CONTROLLABILITY"

WHAT CAN I ACHIEVE WITH CONTROL?CAN I MAKE $\underline{x}(t_f) = \underline{x}_f$ ANYTHING I WANT FOR $t_f > 0$?ROUGH PROOF

$$\left[\underline{x}(t_f) - e^{t_f A} \underline{x}_0 \right] = e^{t_f A} \int_0^{t_f} \underbrace{e^{-\tau A} B\underline{u}(\tau) d\tau}_{\text{THIS IS A VECTOR } \underline{v}(\tau) \text{ MADE BY COMBINING COLUMNS OF } e^{-\tau A} B}$$

ARBITRARY VECTOR

ALWAYS NONSINGULAR

THIS IS A

VECTOR $\underline{v}(\tau)$ MADE BY COMBINING COLUMNS OF $e^{-\tau A} B$

CAREFUL!

$$e^{tA} e^{tY} \neq e^{t(X+Y)}$$

UNLESS $XY = YX$!BUT $e^{-\tau A}$ CONTAINS ONLY $I, A, A^2, \dots, A^{n-1}$ SO WE CAN ONLY GET MULTIPLES OF COLUMNS OF $B, AB, A^2B, \dots, A^{n-1}B$

$$\Rightarrow C \triangleq \begin{bmatrix} B & AB & A^2B & \dots & A^{n-1}B \end{bmatrix}$$

"CONTROLLABILITY MATRIX"

MUST HAVE FULL RANK! THIS TURNS OUT TO BE NECESSARY & SUFFICIENT CONDITION FOR CONTROLLABILITY

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ESE 505

MORE MODERN CONTROL

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RECAP $\dot{\underline{x}} = A\underline{x} + B\underline{u}$ $\underline{x}(0) = \underline{x}_0$

$$\underline{y} = C\underline{x} + D\underline{u}$$

SHOWED $\underline{x}(t) = e^{tA} \underline{x}_0 + \int_0^t e^{(t-\tau)A} B \underline{u}(\tau) d\tau$

$\text{RANK } C = \text{RANK} [B \ AB \ A^2B \ \dots \ A^{n-1}B] = n \iff \text{CONTROLLABLE}$

 $\Rightarrow \exists \underline{u}(t)$ s.t. $\underline{x}(t_f) = \underline{x}_f$ SKIP AHEAD

WHAT ABOUT OBSERVATIONS?

FORMAL QUESTION $\dot{\underline{x}} = A\underline{x}$, $\underline{x}(0) = \underline{x}_0$
 $\underline{u} = 0$ $\underline{y} = C\underline{x}$

GIVEN $\underline{y}(t)$, $0 \leq t \leq t_f$, CAN WE DETERMINE $\underline{x}(0)$?

KNOWING $\underline{y}(t)$ IS LIKE KNOWING $\underline{y}(0), \dot{\underline{y}}(0), \dots, \underline{y}^{(n)}(0), \dots$
 BECAUSE $\underline{y}(t)$ COULD BE WRITTEN WITH TAYLOR SERIES.

KTH DERIVATIVE $\underline{y}^{(k)}(0) = CA^k \underline{x}_0$ $\underline{y}^{(n)}(0) = CA^n \underline{x}_0 = \underbrace{[C \ A \ A^2 \ \dots \ A^{n-1}]}_{\text{LINEAR COMBO OF } \underline{y}(0) \text{ UP TO } \underline{y}^{(n-1)}(0)} \underline{x}_0$

$$\Rightarrow \underline{y} = \begin{pmatrix} \underline{y}(0) \\ \dot{\underline{y}}(0) \\ \vdots \\ \underline{y}^{(n-1)}(0) \end{pmatrix} = \begin{bmatrix} C \\ CA \\ \vdots \\ CA^{n-1} \end{bmatrix} \underline{x}_0 = \mathcal{O} \underline{x}_0$$

$\underline{x}_0 = (\mathcal{O}^T \mathcal{O})^{-1} \mathcal{O}^T \underline{y}$ UNIQUE IF $(\mathcal{O}^T \mathcal{O})^{-1}$ EXISTS
 $\Rightarrow \text{RANK } \mathcal{O} = n$

DUALITY

$$A = \begin{bmatrix} 1 & & \\ & 2 & \\ & & 3 \end{bmatrix} \quad B = \begin{bmatrix} 4 \\ 5 \\ 0 \end{bmatrix}$$

$$C = \begin{bmatrix} 4 & 4 & 4 \\ 5 & 10 & 20 \\ 0 & 0 & 0 \end{bmatrix}$$

$\xrightarrow{+3x}$
 $\xrightarrow{-2x}$

RANK $C = 2$

x_3 UNCONTROLLABLE!

IN GENERAL, CAN
IDENTIFY UNCONTROLLABLE
MODE(S)

WHAT IF $A = \begin{bmatrix} 1 & 0 & x \\ 0 & 2 & 0 \\ 0 & 0 & 3 \end{bmatrix} \quad B = \begin{bmatrix} 4 \\ 5 \\ 0 \end{bmatrix}$

C UNCHANGED!

WHAT IF $A = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 2 & 0 \\ x & 0 & 3 \end{bmatrix} \quad B = \begin{bmatrix} 4 \\ 5 \\ 0 \end{bmatrix}$

NOW $C = \begin{bmatrix} 4 & 4 & 4 \\ 5 & 10 & 20 \\ 0 & 4x & 10x \end{bmatrix}$

ANY NON-ZERO $x \Rightarrow$ CONTROLLABLE.

⊕ USE CONTROL TO INFLUENCE

x_1 & x_2 TO INFLUENCE x_3 .

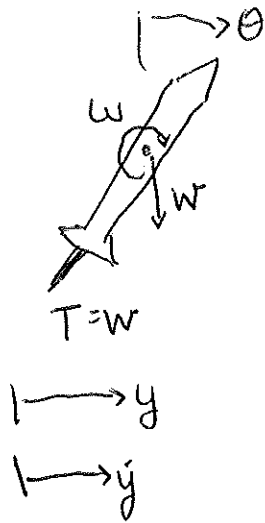
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STATE FEEDBACK EXAMPLE

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SPACE-X GRASSHOPPER ROCKET

$$d^3 = gC_1$$



$$\ddot{y} = \dot{y}$$

$$\ddot{y} = g\theta$$

$$\ddot{\theta} = w$$

$$\dot{w} = u + C_1 \dot{y}$$

ROCKET TIPS INTO WIND DUE TO FINs

$$\text{LET } x_4 = \frac{1}{g} \dot{y}$$

$$x_3 = \frac{1}{g} \ddot{y} \Rightarrow$$

$$x_2 = \theta$$

$$x_1 = w$$

$$\dot{x} = \begin{bmatrix} 0 & 0 & d^3 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix} x + \begin{bmatrix} 1 \\ 0 \\ 0 \\ 0 \end{bmatrix} u$$

SPECIAL FORM CALLED

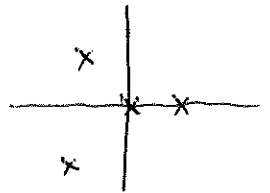
"CONTROL CANONICAL FORM" (7.86 IN FRANKLIN)

$$\text{TOP ROW IS } [-a_1 \quad -a_2 \quad \dots \quad -a_n]$$

$$\text{WHERE } \Delta(s) = s^n + a_1 s^{n-1} + \dots + a_n$$

HERE

$$\Delta(s) = s(s^3 - d^3)$$



- ⊗ FOR CONTROLLABLE SYSTEM, WE CAN ALWAYS FIND A STATE VARIABLE CHANGE TO GET A & B IN THIS FORM! ⊗
STATE VARIABLE CHANGE RELIES ON RANK $C = n$! $\{C^{-1}$ FOR SCALAR $u\}$

$$\text{STATE FEEDBACK: } \underline{u} = -K\underline{x} = [-k_1 \quad -k_2 \quad -k_3 \quad -k_4] \underline{x}$$

$$\dot{\underline{x}} = [A - BK] \underline{x} = \begin{bmatrix} -a_1 - k_1 & -a_2 - k_2 & \dots & -a_n - k_n \\ 1 & & & \\ & 1 & & \\ & & \ddots & \\ & & & 1 & 0 \end{bmatrix}$$

NEW CHAR POLY IS

$$\Delta_{CL}(s) = s^n + (a_1 + k_1)s^{n-1} + \dots + (a_n + k_n)$$

PICK ARBITRARY $\Delta_{CL}(s)$ USING STATE FEEDBACK!