

Attitude Estimation Examples from Chapter 3

Setup

```
In[214]:= << RigidBodyKinematics`  
  
In[215]:= tilde[x_] := {{0, -x[[3]], x[[2]]},  
                      {x[[3]], 0, -x[[1]]},  
                      {-x[[2]], x[[1]], 0}}
```

Example 3.14 (Triad)

Setup the true attitude states:

```
In[216]:= theta = {30., 20., -10.} Degree;  
BNtrue = Euler3212C[theta]  
  
Out[217]:= {{0.813798, 0.469846, -0.34202},  
            {-0.543838, 0.823173, -0.163176}, {0.204874, 0.318796, 0.925417}}
```

```
In[218]:= v1N = {1, 0, 0};  
v2N = {0, 0, 1};  
  
In[220]:= MatrixForm[v1Btrue = BNtrue.v1N]  
Out[220]/MatrixForm=  

$$\begin{pmatrix} 0.813798 \\ -0.543838 \\ 0.204874 \end{pmatrix}$$
  
  
In[221]:= MatrixForm[v2Btrue = BNtrue.v2N]  
Out[221]/MatrixForm=  

$$\begin{pmatrix} -0.34202 \\ -0.163176 \\ 0.925417 \end{pmatrix}$$

```

Setup the measured attitude states:

```
In[222]:= v1B = {0.8190, -0.5282, 0.2242};
          v2B = {-0.3138, -0.1584, 0.9362};
          v1B = v1B / Norm[v1B];
          v2B = v2B / Norm[v2B];
```

Develop Triad Frame

From measured states

```
In[226]:= MatrixForm[t1B = v1B]
          t2B = Cross[v1B, v2B];
          MatrixForm[t2B = t2B / Norm[t2B]]
          MatrixForm[t3B = Cross[t1B, t2B]]
```

Out[226]/MatrixForm=

$$\begin{pmatrix} 0.818991 \\ -0.528194 \\ 0.224198 \end{pmatrix}$$

Out[228]/MatrixForm=

$$\begin{pmatrix} -0.459282 \\ -0.837639 \\ -0.295669 \end{pmatrix}$$

Out[229]/MatrixForm=

$$\begin{pmatrix} 0.343967 \\ 0.13918 \\ -0.928609 \end{pmatrix}$$

From inertial states of the measurements

```
In[230]:= MatrixForm[t1N = v1N]
          t2N = Cross[t1N, v2N];
          MatrixForm[t2N = t2N / Norm[t2N]]
          MatrixForm[t3N = Cross[t1N, t2N]]
```

Out[230]/MatrixForm=

$$\begin{pmatrix} 1 \\ 0 \\ 0 \end{pmatrix}$$

Out[232]/MatrixForm=

$$\begin{pmatrix} 0 \\ -1 \\ 0 \end{pmatrix}$$

Out[233]/MatrixForm=

$$\begin{pmatrix} 0 \\ 0 \\ -1 \end{pmatrix}$$

Find Estimated Attitude

```
In[234]:= MatrixForm[BbarT = Transpose[{t1B, t2B, t3B}]]
```

```
Out[234]/MatrixForm=
```

$$\begin{pmatrix} 0.818991 & -0.459282 & 0.343967 \\ -0.528194 & -0.837639 & 0.13918 \\ 0.224198 & -0.295669 & -0.928609 \end{pmatrix}$$

```
In[235]:= NT = Transpose[{t1N, t2N, t3N}]
```

```
Out[235]= {{1, 0, 0}, {0, -1, 0}, {0, 0, -1}}
```

```
In[236]:= MatrixForm[BbarN = BbarT.Transpose[NT]]
```

```
Out[236]/MatrixForm=
```

$$\begin{pmatrix} 0.818991 & 0.459282 & -0.343967 \\ -0.528194 & 0.837639 & -0.13918 \\ 0.224198 & 0.295669 & 0.928609 \end{pmatrix}$$

Check accuracy of estimate

```
In[237]:= MatrixForm[BbarB = BbarN.Transpose[BNtrue]]
```

```
Out[237]/MatrixForm=
```

$$\begin{pmatrix} 0.999929 & -0.0112026 & -0.00410552 \\ 0.0113209 & 0.999485 & 0.0300232 \\ 0.00376707 & -0.0300675 & 0.999541 \end{pmatrix}$$

```
In[238]:= p = C2PRV[BbarB]
```

```
Out[238]= {0.0300506, 0.00393698, -0.0112637}
```

```
In[239]:= Norm[p] / Degree
```

```
Out[239]= 1.85253
```

Example 3.15 (Devenport's q-Method)

Setup the true attitude states:

```
In[240]:= thetaTrue = {30., 20., -10.} Degree;
```

```
BNtrue = Euler3212C[thetaTrue]
```

```
Out[241]= {{0.813798, 0.469846, -0.34202},
           {-0.543838, 0.823173, -0.163176}, {0.204874, 0.318796, 0.925417}}
```

```
In[242]:= v1N = {1, 0, 0};
```

```
v2N = {0, 0, 1};
```

```
In[244]:= MatrixForm[v1Btrue = BNtrue.v1N]
```

```
Out[244]//MatrixForm=
```

$$\begin{pmatrix} 0.813798 \\ -0.543838 \\ 0.204874 \end{pmatrix}$$

```
In[245]:= MatrixForm[v2Btrue = BNtrue.v2N]
```

```
Out[245]//MatrixForm=
```

$$\begin{pmatrix} -0.34202 \\ -0.163176 \\ 0.925417 \end{pmatrix}$$

Setup the measured attitude states:

```
In[246]:= v1B = {0.8190, -0.5282, 0.2242};
v2B = {-0.3138, -0.1584, 0.9362};
v1B = v1B / Norm[v1B];
v2B = v2B / Norm[v2B];
```

Setup q-method parameters

```
In[250]:= w1 = 1;
w2 = 1;
```

```
In[252]:= B = w1 Outer[Times, v1B, v1N] + w2 Outer[Times, v2B, v2N]
```

```
Out[252]= {{0.818991, 0., -0.313795}, {-0.528194, 0., -0.158398}, {0.224198, 0., 0.936185}}
```

```
In[253]:= S = B + Transpose[B]
```

```
Out[253]= {{1.63798, -0.528194, -0.0895975},
{-0.528194, 0., -0.158398}, {-0.0895975, -0.158398, 1.87237}}
```

```
In[254]:= σ = B[[1, 1]] + B[[2, 2]] + B[[3, 3]]
```

```
Out[254]= 1.75518
```

```
In[255]:= Z = {B[[2, 3]] - B[[3, 2]],
B[[3, 1]] - B[[1, 3]],
B[[1, 2]] - B[[2, 1]]}
```

```
Out[255]= {-0.158398, 0.537993, 0.528194}
```

```
In[256]:= MatrixForm[
  K = {{σ, Z[[1]], Z[[2]], Z[[3]]},
        {Z[[1]], S[[1, 1]] - σ, S[[1, 2]], S[[1, 3]]},
        {Z[[2]], S[[2, 1]], S[[2, 2]] - σ, S[[2, 3]]},
        {Z[[3]], S[[3, 1]], S[[3, 2]], S[[3, 3]] - σ}}
]
```

```
Out[256]/MatrixForm=

$$\begin{pmatrix} 1.75518 & -0.158398 & 0.537993 & 0.528194 \\ -0.158398 & -0.117194 & -0.528194 & -0.0895975 \\ 0.537993 & -0.528194 & -1.75518 & -0.158398 \\ 0.528194 & -0.0895975 & -0.158398 & 0.117194 \end{pmatrix}$$

```

Solve for optimal attitude

```
In[257]:= {λ, βvectors} = Eigensystem[K];
```

```
In[258]:= λ
```

```
Out[258]= {1.99967, -1.99967, 0.0365659, -0.0365659}
```

```
In[259]:= β = βvectors[[1]]
```

```
Out[259]= {0.948069, -0.117207, 0.141371, 0.259697}
```

```
In[260]:= MatrixForm[BbarN = EP2C[β]]
```

```
Out[260]/MatrixForm=

$$\begin{pmatrix} 0.825143 & 0.459282 & -0.328936 \\ -0.525561 & 0.837639 & -0.148814 \\ 0.207182 & 0.295669 & 0.932553 \end{pmatrix}$$

```

Check accuracy of estimate

```
In[261]:= MatrixForm[BbarB = BbarN.Transpose[BNtrue]]
```

```
Out[261]/MatrixForm=

$$\begin{pmatrix} 0.999794 & -0.0170009 & 0.0110648 \\ 0.0167585 & 0.999625 & 0.0216474 \\ -0.0114287 & -0.0214576 & 0.999704 \end{pmatrix}$$

```

```
In[262]:= q = C2PRV[BbarB]
```

```
Out[262]= {0.0215556, -0.0112484, -0.0168821}
```

```
In[263]:= ErrorQmethod = Norm[q] / Degree
```

```
Out[263]= 1.69597
```

Example 3.16 (QUEST)

Setup the true attitude states:

```
In[264]:= θtrue = {30., 20., -10.} Degree;
          BNtrue = Euler3212C[θtrue]

Out[265]:= {{0.813798, 0.469846, -0.34202},
           {-0.543838, 0.823173, -0.163176}, {0.204874, 0.318796, 0.925417}}
```

```
In[266]:= v1N = {1, 0, 0};
          v2N = {0, 0, 1};

In[268]:= MatrixForm[v1Btrue = BNtrue.v1N]
Out[268]//MatrixForm=

$$\begin{pmatrix} 0.813798 \\ -0.543838 \\ 0.204874 \end{pmatrix}$$

In[269]:= MatrixForm[v2Btrue = BNtrue.v2N]
Out[269]//MatrixForm=

$$\begin{pmatrix} -0.34202 \\ -0.163176 \\ 0.925417 \end{pmatrix}$$

```

Setup the measured attitude states:

```
In[270]:= v1B = {0.8190, -0.5282, 0.2242};
          v2B = {-0.3138, -0.1584, 0.9362};
          v1B = v1B / Norm[v1B];
          v2B = v2B / Norm[v2B];
```

Setup q-method parameters

```
In[274]:= w1 = 1;
          w2 = 1;

In[276]:= B = w1 Outer[Times, v1B, v1N] + w2 Outer[Times, v2B, v2N]
Out[276]:= {{0.818991, 0., -0.313795}, {-0.528194, 0., -0.158398}, {0.224198, 0., 0.936185}}
```

```
In[277]:= S = B + Transpose[B]
Out[277]:= {{1.63798, -0.528194, -0.0895975},
           {-0.528194, 0., -0.158398}, {-0.0895975, -0.158398, 1.87237}}
```

```
In[278]:= σ = B[[1, 1]] + B[[2, 2]] + B[[3, 3]]
Out[278]= 1.75518
```

```
In[279]:= Z = {B[[2, 3]] - B[[3, 2]],
               B[[3, 1]] - B[[1, 3]],
               B[[1, 2]] - B[[2, 1]]}
```

```
Out[279]= {-0.158398, 0.537993, 0.528194}
```

```
In[280]:= MatrixForm[
  K = {{σ, Z[[1]], Z[[2]], Z[[3]]},
        {Z[[1]], S[[1, 1]] - σ, S[[1, 2]], S[[1, 3]]},
        {Z[[2]], S[[2, 1]], S[[2, 2]] - σ, S[[2, 3]]},
        {Z[[3]], S[[3, 1]], S[[3, 2]], S[[3, 3]] - σ}}
]
```

```
Out[280]/MatrixForm=

$$\begin{pmatrix} 1.75518 & -0.158398 & 0.537993 & 0.528194 \\ -0.158398 & -0.117194 & -0.528194 & -0.0895975 \\ 0.537993 & -0.528194 & -1.75518 & -0.158398 \\ 0.528194 & -0.0895975 & -0.158398 & 0.117194 \end{pmatrix}$$

```

Setup QUEST parameters

```
In[281]:= λ = w1 + w2
```

```
Out[281]= 2
```

```
In[282]:= q = Inverse[(λ + σ) IdentityMatrix[3] - S].Z
```

```
Out[282]= {-0.123602, 0.1491, 0.273874}
```

```
In[283]:= MatrixForm[BbarN = Gibbs2C[q]]
```

```
Out[283]/MatrixForm=

$$\begin{pmatrix} 0.825193 & 0.45922 & -0.328897 \\ -0.525482 & 0.837693 & -0.148793 \\ 0.207186 & 0.295613 & 0.93257 \end{pmatrix}$$

```

```
In[284]:= MatrixForm[BbarB = BbarN.Transpose[BNtrue]]
```

```
Out[284]/MatrixForm=

$$\begin{pmatrix} 0.999793 & -0.0170853 & 0.0110912 \\ 0.0168417 & 0.999623 & 0.0216995 \\ -0.0114577 & -0.0215082 & 0.999703 \end{pmatrix}$$

```

```
In[285]:= q = C2PRV[BbarB]
```

```
Out[285]= {0.021607, -0.0112761, -0.016966}
```

```
In[286]:= Norm[q] / Degree
```

```
Out[286]= 1.70146
```

Iterate for Optimal Attitude

```

In[287]:= Det[K - s * IdentityMatrix[4]]
Out[287]= 0.00534646 + 7.35523 × 10-16 s - 4. s2 - 2.22045 × 10-16 s3 + s4

In[288]:= f[s_] := Det[K - s * IdentityMatrix[4]]

In[289]:= λ0 = λ
Out[289]= 2

In[290]:= f[λ0]
Out[290]= 0.00534646

In[291]:= λ1 = λ0 - f[λ0] / f'[λ0]
Out[291]= 1.99967

In[292]:= f[λ1]
Out[292]= 2.23288 × 10-6

In[293]:= λ2 = λ1 - f[λ1] / f'[λ1]
Out[293]= 1.99967

In[294]:= f[λ2]
Out[294]= 3.90176 × 10-13

In[295]:= λ3 = λ2 - f[λ2] / f'[λ2]
Out[295]= 1.99967

In[296]:= f[λ3]
Out[296]= -3.45856 × 10-16

```

Iteration Accuracy Study

```

In[297]:= Norm[C2PRV[Gibbs2C[Inverse[(λ0 + σ) IdentityMatrix[3] - S].Z].Transpose[BNtrue]]] /
Degree - ErrorQmethod
Out[297]= 0.00549088

In[298]:= Norm[C2PRV[Gibbs2C[Inverse[(λ1 + σ) IdentityMatrix[3] - S].Z].Transpose[BNtrue]]] /
Degree - ErrorQmethod
Out[298]= 2.29401 × 10-6

In[299]:= Norm[C2PRV[Gibbs2C[Inverse[(λ2 + σ) IdentityMatrix[3] - S].Z].Transpose[BNtrue]]] /
Degree - ErrorQmethod
Out[299]= 3.76588 × 10-13

```



```
In[300]:= Norm[C2PRV[Gibbs2C[Inverse[(λ3 + σ) IdentityMatrix[3] - S].Z].Transpose[BNtrue]]] /
Degree - ErrorQmethod
Out[300]:= -2.90878 × 10-14
```

Example 3.17 (OLAE)

Setup the true attitude states:

```
In[301]:= θtrue = {30., 20., -10.} Degree;
BNtrue = Euler3212C[θtrue]
Out[302]:= {{0.813798, 0.469846, -0.34202},
{-0.543838, 0.823173, -0.163176}, {0.204874, 0.318796, 0.925417}}

In[303]:= v1N = {1, 0, 0};
v2N = {0, 0, 1};

In[305]:= MatrixForm[v1Btrue = BNtrue.v1N]
Out[305]//MatrixForm=

$$\begin{pmatrix} 0.813798 \\ -0.543838 \\ 0.204874 \end{pmatrix}$$


In[306]:= MatrixForm[v2Btrue = BNtrue.v2N]
Out[306]//MatrixForm=

$$\begin{pmatrix} -0.34202 \\ -0.163176 \\ 0.925417 \end{pmatrix}$$

```

Setup the measured attitude states:

```
In[307]:= v1B = {0.8190, -0.5282, 0.2242};
v2B = {-0.3138, -0.1584, 0.9362};
v1B = v1B / Norm[v1B];
v2B = v2B / Norm[v2B];
```

Evaluate OLAE states

```
In[311]:= W = IdentityMatrix[6];
In[312]:= d = Join[v1B - v1N, v2B - v2N]
Out[312]:= {-0.181009, -0.528194, 0.224198, -0.313795, -0.158398, -0.0638147}

In[313]:= S = Join[tilde[v1B + v1N], tilde[v2B + v2N]]
Out[313]:= {{0, -0.224198, -0.528194}, {0.224198, 0, -1.81899}, {0.528194, 1.81899, 0},
{0, -1.93619, -0.158398}, {1.93619, 0, 0.313795}, {0.158398, -0.313795, 0}}
```

```

In[314]:= qBar = Inverse[Transpose[S].W.S].Transpose[S].W.d
Out[314]= {-0.12359, 0.148759, 0.274255}

In[315]:= BbarN = Gibbs2C[qBar]
Out[315]= {{0.825016, 0.459942, -0.328332},
           {-0.526039, 0.837338, -0.148823}, {0.206474, 0.295497, 0.932765}}

```

Check accuracy of estimate

```

In[316]:= BbarB = BbarN.Transpose[BNtrue]
Out[316]= {{0.999794, -0.0164875, 0.0118086},
           {0.0162316, 0.999638, 0.0214443}, {-0.0121579, -0.0212482, 0.9997}}

In[317]:= q = C2PRV[BbarB]
Out[317]= {0.0213494, -0.011985, -0.0163619}

In[318]:= ErrorQmethod = Norm[q] / Degree
Out[318]= 1.68721

```