

# Nonlinear Spacecraft Control

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ASEN 5010

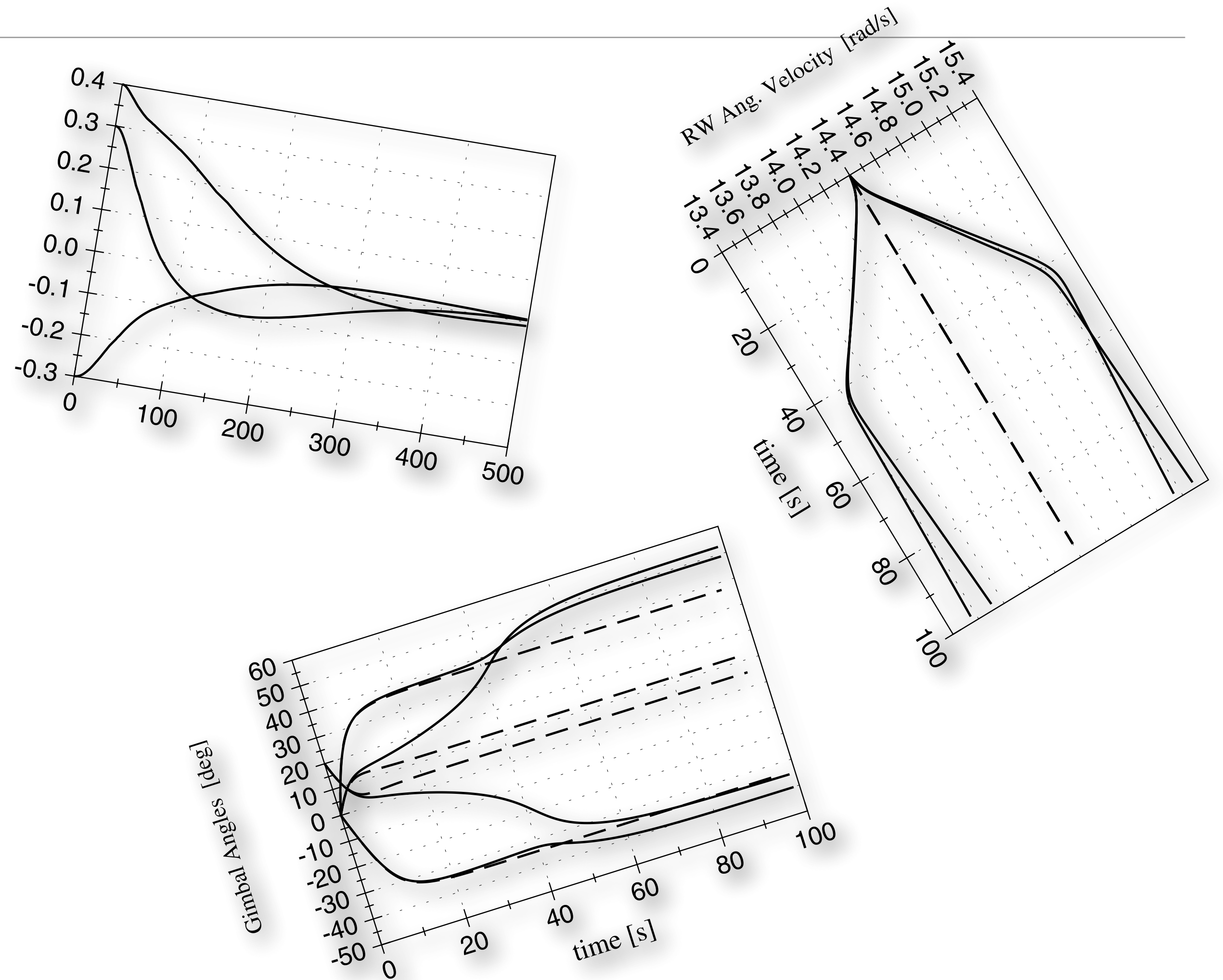
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# Outline

- Stability Definitions
- Lyapunov Functions
  - Velocity-based feedback
  - Position-based feedback
  - Lyapunov's Direct Method
- Nonlinear Feedback of Spacecraft Attitude
  - Full-state feedback for regulator and tracking problems
  - Feedback Gain Selection
- Lyapunov Optimal Feedback
- Linear Closed-Loop Dynamics



# Stability Definitions

Why isn't stable just stable?

# Definitions

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State Vector:  $\mathbf{x} = (x_1 \cdots x_N)^T$

EOM:  $\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}, t)$  — Non-Autonomous System

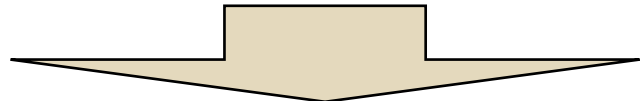
$\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x})$  — Autonomous System

Control Vector:  $\mathbf{u} = \mathbf{g}(\mathbf{x})$

Closed-Loop System:  $\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}, \mathbf{u}, t)$

**Equilibrium State:** A state vector point  $\mathbf{x}_e$  is said to be an equilibrium state (or equilibrium point) of a dynamical system described by  $\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}, t)$  at time  $t_0$  if

$$\mathbf{f}(\mathbf{x}_e, t) = 0 \quad \forall t > t_0$$

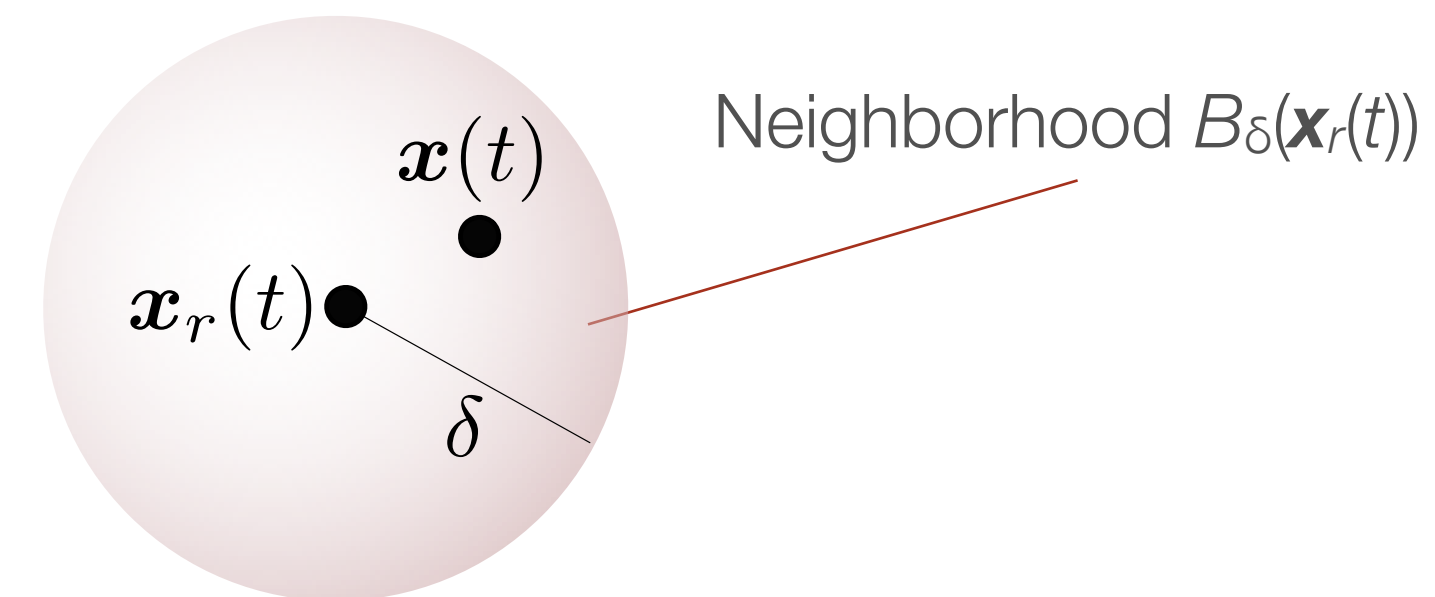

$$\dot{\mathbf{x}}_e = 0 \quad \mathbf{x}_e = \text{constant}$$

**Neighborhood:** Given  $\delta > 0$ , a state vector  $\mathbf{x}(t)$  is said to be in the neighborhood  $B_\delta(\mathbf{x}_r(t))$  of the state  $\mathbf{x}_r(t)$  if

$$\|\mathbf{x}(t) - \mathbf{x}_r(t)\| < \delta$$

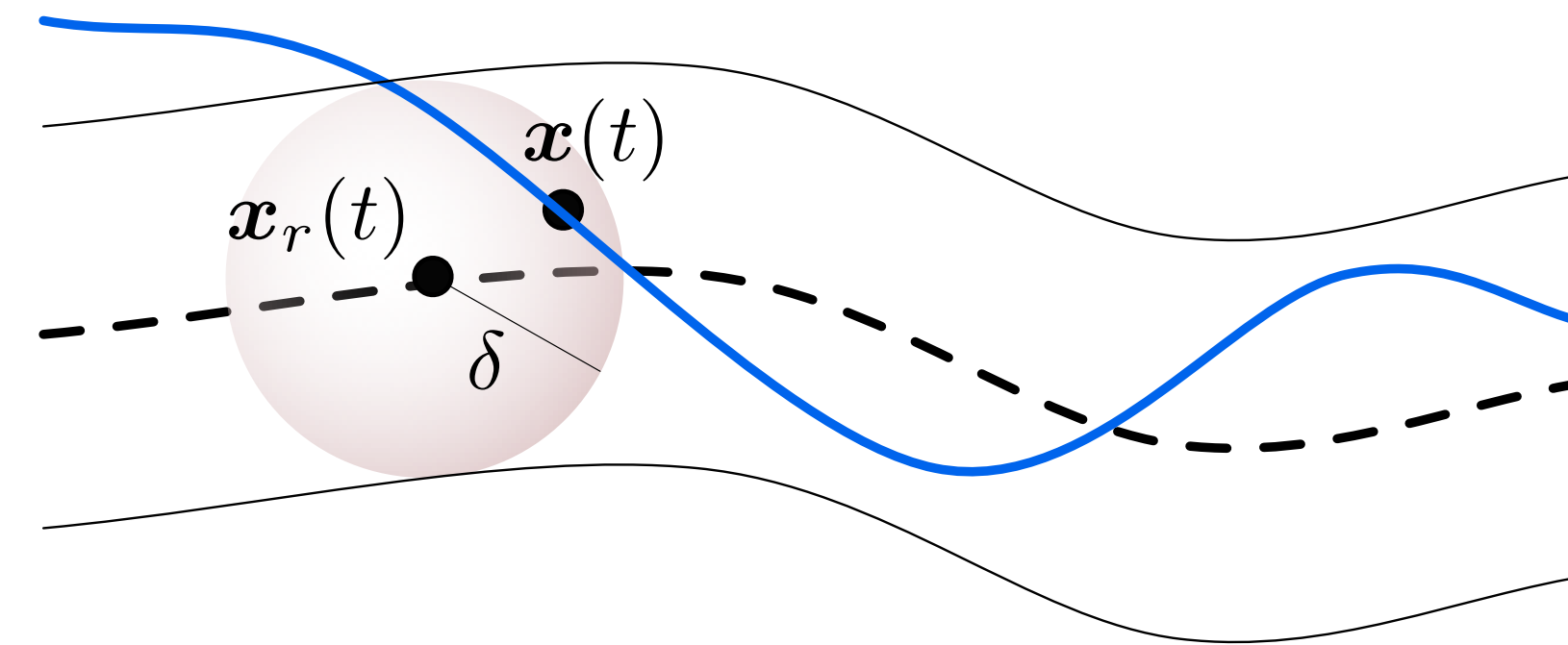
then

$$\mathbf{x}(t) \in B_\delta(\mathbf{x}_r(t))$$



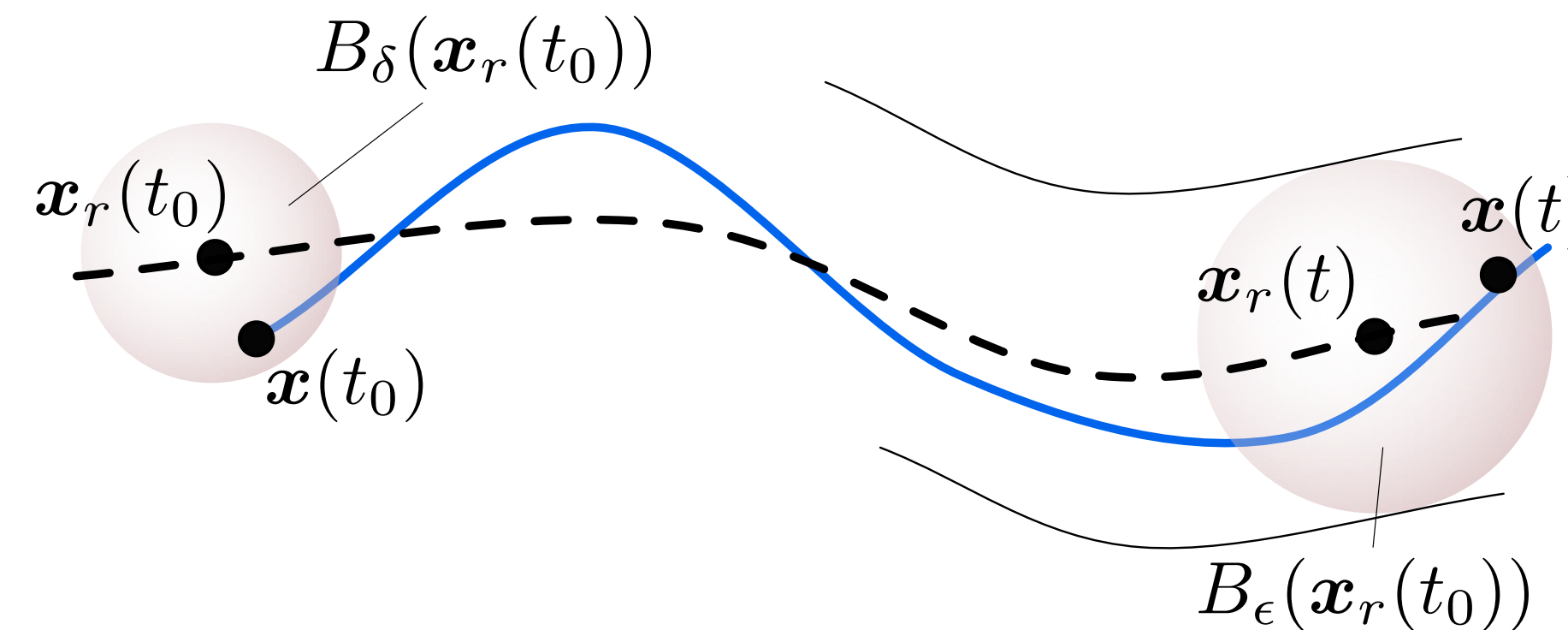
**Lagrange Stability:** The motion  $\mathbf{x}(t)$  is said to be Lagrange stable (or bounded) relative to  $\mathbf{x}_r(t)$  if there exists a  $\delta > 0$  such that

$$\mathbf{x}(t) \in B_\delta(\mathbf{x}_r(t)) \quad \forall t > t_0$$



**Lyapunov Stability:** The motion  $\mathbf{x}(t)$  is said to be Lyapunov stable (or stable) relative to  $\mathbf{x}_r(t)$  if for each  $\epsilon > 0$  there exists a  $\delta(\epsilon) > 0$  such that

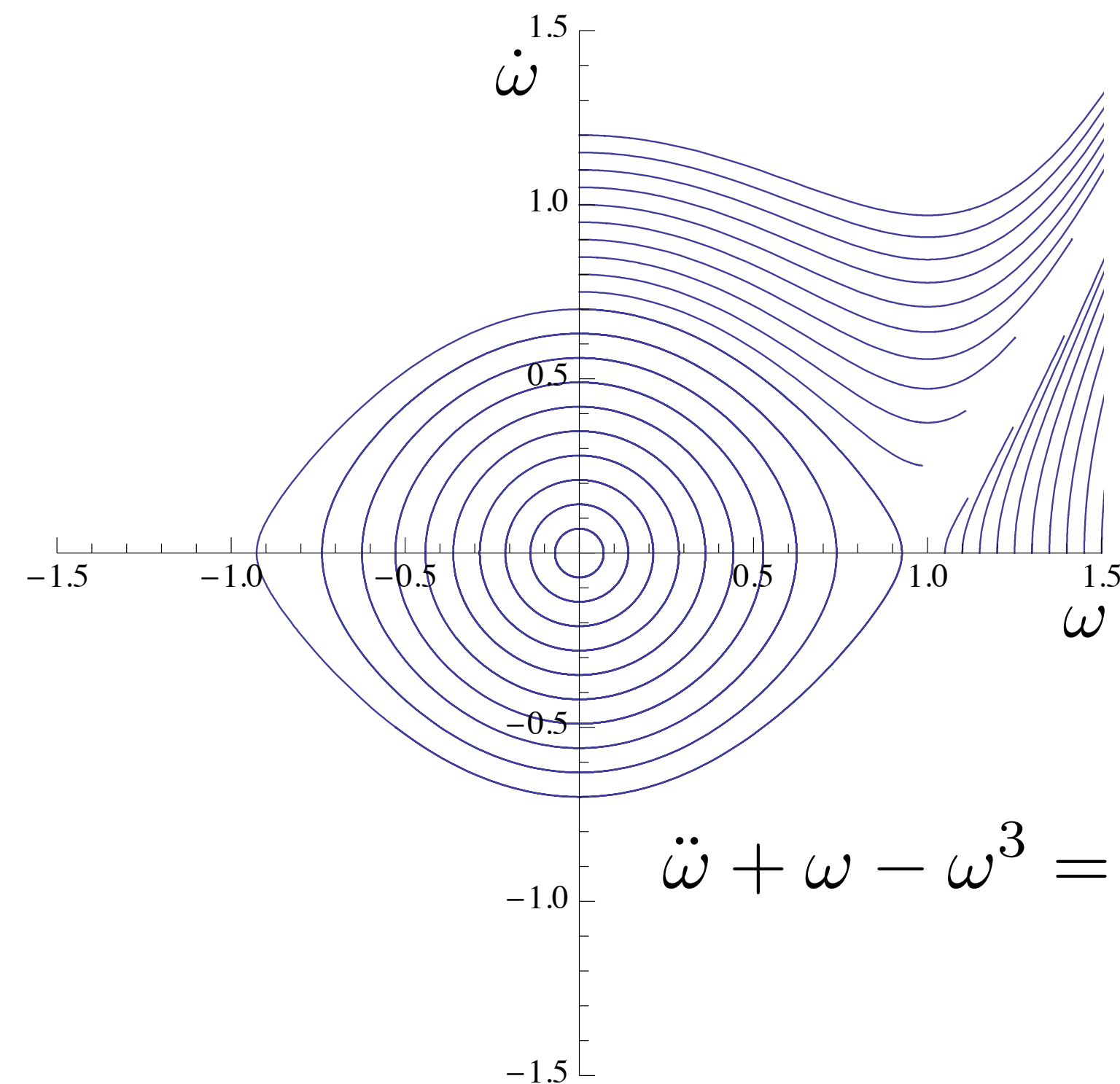
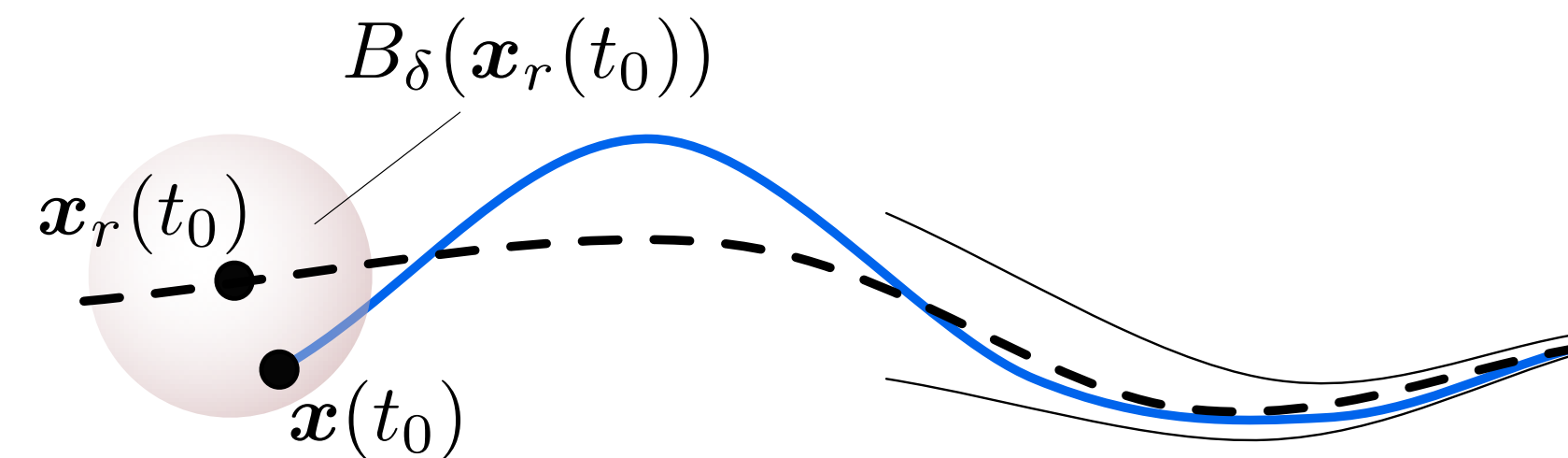
$$\mathbf{x}(t_0) \in B_\delta(\mathbf{x}_r(t_0)) \implies \mathbf{x}(t) \in B_\epsilon(\mathbf{x}_r(t)) \quad \forall t > t_0$$



**Asymptotic Stability:** The motion  $\mathbf{x}(t)$  is asymptotically stable relative to  $\mathbf{x}_r(t)$  if  $\mathbf{x}(t)$  is Lyapunov stable and there exists a  $\delta > 0$  such that

$$\mathbf{x}(t_0) \in B_\delta(\mathbf{x}_r(t_0)) \implies \lim_{t \rightarrow \infty} \mathbf{x}(t) = \mathbf{x}_r(t)$$

**Global Stability:** The motion  $\mathbf{x}(t)$  is globally stable relative to  $\mathbf{x}_r(t)$  if  $\mathbf{x}(t)$  is stable for any initial state vector  $\mathbf{x}(t_0)$ .



(Show Mathematica Example)



# Linearization of Dynamical System

Reference motion  
given by:

$$\dot{\mathbf{x}}_r = \mathbf{f}(\mathbf{x}_r, \overset{\text{Feedforward control}}{\mathbf{u}_r})$$

Nonlinear EOM:

$$\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}, \mathbf{u})$$


Feedback control:

$$\delta \mathbf{u} = \mathbf{u} - \mathbf{u}_r$$

Departure motion:

$$\delta \mathbf{x} = \mathbf{x} - \mathbf{x}_r$$

Performing a Taylor Series expansion of  $\mathbf{x}$  about  $(\mathbf{x}_r, \mathbf{u}_r)$  we obtain

$$\begin{aligned} \delta \dot{\mathbf{x}} &= \mathbf{f}(\mathbf{x}_r, \mathbf{u}_r) + \frac{\partial \mathbf{f}(\mathbf{x}_r, \mathbf{u}_r)}{\partial \mathbf{x}} \delta \mathbf{x} \\ &\quad + \frac{\partial \mathbf{f}(\mathbf{x}_r, \mathbf{u}_r)}{\partial \mathbf{u}} \delta \mathbf{u} + H.O.T - \mathbf{f}(\mathbf{x}_r, \mathbf{u}_r) \end{aligned}$$


$$\delta \dot{\mathbf{x}} \simeq \frac{\partial \mathbf{f}(\mathbf{x}_r, \mathbf{u}_r)}{\partial \mathbf{x}} \delta \mathbf{x} + \frac{\partial \mathbf{f}(\mathbf{x}_r, \mathbf{u}_r)}{\partial \mathbf{u}} \delta \mathbf{u}$$

Let us define:

$$[A] = \frac{\partial \mathbf{f}(\mathbf{x}_r, \mathbf{u}_r)}{\partial \mathbf{x}}$$

$$[B] = \frac{\partial \mathbf{f}(\mathbf{x}_r, \mathbf{u}_r)}{\partial \mathbf{u}}$$

The linearized system is then written in standard form as

$$\delta \dot{\mathbf{x}} \simeq [A] \delta \mathbf{x} + [B] \delta \mathbf{u}$$

If the nominal reference motion is an equilibrium state  $\mathbf{x}_e$ , then the linearized EOM simplify to:

$$\dot{\mathbf{x}} \simeq [A] \mathbf{x} + [B] \mathbf{u}$$