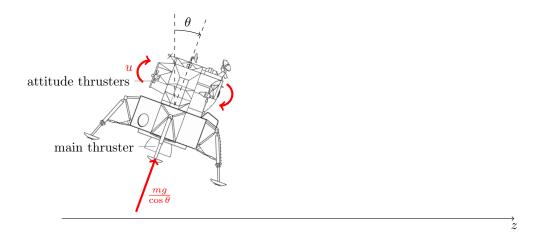
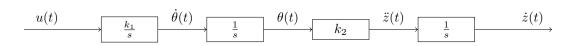
State feedback

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Example - The Apollo lunar module





Example - The Apollo lunar module

State variables:
$$x = \begin{bmatrix} x_1 & x_2 & x_3 \end{bmatrix}^T = \begin{bmatrix} \dot{\theta} & \theta & \dot{z} \end{bmatrix}^T$$
. With dynamics

$$\begin{cases} \dot{x}_1 = \ddot{\theta} = k_1 u \\ \dot{x}_2 = \dot{\theta} = x_1 \\ \dot{x}_3 = \ddot{z} = k_2 \theta = k_2 x_2 \end{cases}$$

$$\dot{x} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \underbrace{\begin{bmatrix} 0 & 0 & 0 \\ 1 & 0 & 0 \\ 0 & k_2 & 0 \end{bmatrix}}_{A} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \underbrace{\begin{bmatrix} k_1 \\ 0 \\ 0 \end{bmatrix}}_{B} u$$

Example - The Apollo lunar module

$$x(kh+h) = e^{Ah}x(kh) + \int_0^h e^{As}Bu(kh+h-s)ds$$

$$= \underbrace{e^{Ah}}_{\Phi(h)}x(kh) + \underbrace{\left(\int_0^h e^{As}Bds\right)}_{\Gamma(h)}u(kh)$$

$$= \begin{bmatrix} 1 & 0 & 0\\ h & 1 & 0\\ \frac{h^2k_2}{2} & hk_2 & 1 \end{bmatrix}x(kh) + k_1\begin{bmatrix} h\\ \frac{h^2}{2}\\ \frac{k_2h^3}{6} \end{bmatrix}u(kh)$$

Stability

Eigenvalues and eigenvectors

Definition The eigenvalues $\lambda_i \in \mathbb{R}$ and eigenvectors $v_i \in \mathbb{R}^n$ of a matrix $\Phi \in \mathbb{R}^{n \times n}$ are the *n* pairs $(\lambda_i, v_i \neq 0), i = 1, 2, ..., n$ that satisfy

$$\Phi v_i = \lambda_i v_i$$

Stability

The system

$$x(k+1) = \Phi x(k), \quad x(0) = x_0$$

is stable if $\lim_{t\to\infty} x(kh) = 0$, $\forall x_0 \in \mathbb{R}^n$.

A necessary and sufficient requirement for stability is that all the eigenvalues of Φ are inside the unit circle.

The eigenvalues of Φ are the poles of the system.

Eigenvalues and eigenvectors - exercise

Activity Verify that the vector

$$v = \begin{bmatrix} 1 \\ 0 \end{bmatrix}$$

is an eigenvector of

$$\Phi = \begin{bmatrix} 2 & 0 \\ 0 & \frac{1}{2} \end{bmatrix}.$$

What is the corresponding eigenvalue?

Controllability

Controllability is the answer to the question Can we drive the state of the system to any location in the state space by a suitable input sequence u(k), k = 0, 1, 2, ..., n - 1?

Consider

$$x(k+1) = \Phi x(k) + \Gamma u(k), \quad x(0) = x_0$$

with solution

$$x(n) = \Phi^{n} x(0) + \Phi^{n-1} \Gamma u(0) + \Phi^{n-2} \Gamma u(1) + \dots + \Gamma u(n-1)$$

= $\Phi^{n} x(0) + W_{c} U$, (1)

where

$$W_c = \begin{bmatrix} \Gamma & \Phi \Gamma & \cdots & \Phi^{n-1} \Gamma \end{bmatrix}$$

$$U = \begin{bmatrix} u(n-1) & u(n-2) & \cdots & u(0) \end{bmatrix}^{\mathrm{T}}$$



Controllability

To find the input sequence u(k) that takes the state from $x(0) = x_0$ to $x(n) = x_d$ we may solve for U in the equation

$$x_d = \Phi^n x_0 + W_c U.$$

$$U = W_c^{-1} (x_d - \Phi^n x(0))$$

This is possible when the matrix W_x is invertible:

The state-space system above is controllable if and only if the *Controllability matrix* W_c has rank n, i.e.

$$\det W_c \neq 0$$
.

Observability

Observability is the answer to the question "Can we determine the initial state x(0) if we only know y(k), k = 0, 1, 2, ..., n - 1?"

The first n values of the output sequence are given by

$$y(0) = Cx(0)$$

$$y(1) = Cx(1) = C (\Phi x(0) + \Gamma u(0))$$

$$\vdots$$

$$y(n-1) = Cx(n-1) = C (\Phi^{n-1}x(0) + W_c U).$$

This gives the equation

$$\begin{bmatrix} C \\ C\Phi \\ \vdots \\ C\Phi^{n-1} \end{bmatrix} \times (0) = \begin{bmatrix} y(0) \\ y(1) - C\Gamma u(0) \\ \vdots \\ y(n-1) - CW_c U \end{bmatrix}$$

which can be solved for x(0) if and only if the matrix

$$W_o = \begin{bmatrix} C \\ C\Phi \\ \vdots \\ C\Phi^{n-1} \end{bmatrix}$$

Observability, contd

The equation

$$\begin{bmatrix} C \\ C\Phi \\ \vdots \\ C\Phi^{n-1} \end{bmatrix} \times (0) = \begin{bmatrix} y(0) \\ y(1) - C\Gamma u(0) \\ \vdots \\ y(n-1) - CW_c U \end{bmatrix}$$

can be solved for x(0) if and only if the matrix

$$W_o = \begin{bmatrix} C \\ C\Phi \\ \vdots \\ C\Phi^{n-1} \end{bmatrix}$$

has full rank. If this is the case, the system is said to be observable.

State feedback control

State feedback control

Given

$$x(k+1) = \Phi x(k) + \Gamma u(k)$$

$$y(k) = Cx(k)$$
 (2)

and measurements (or an estimate) of the state vector x(k).

Linear state feedback is the control law

$$u(k) = f((x(k), u_c(k))) = -l_1x_1(k) - l_2x_2(k) - \dots - l_nx_n(k) + l_0u_c(k)$$

= $-Lx(k) + l_0u_c(k)$,

where

$$L = \begin{bmatrix} I_1 & I_2 & \cdots & I_n \end{bmatrix}.$$

Substituting this in the state-space model (2) gives

$$x(k+1) = (\Phi - \Gamma L)x(k) + m\Gamma u_c(k)$$

$$y(k) = Cx(k)$$
(3)

Pole placement by state feedback

Given a desired placement of the closed-loop poles p_1, p_2, \ldots, p_n , being roots of the desired characteristic polynomial

$$a_c(z) = (z - p_1)(z - p_2) \cdots (z - p_n) = z^n + \alpha_1 z^{n-1} + \cdots + \alpha_n.$$
 (4)

Linear state feedback gives the system

$$x(k+1) = (\Phi - \Gamma L)x(k) + l_0 \Gamma u_c(k)$$

$$y(k) = Cx(k)$$
(5)

with characteristic polynomial

$$\det(zI - (\Phi - \Gamma L)) = z^n + \beta_1(I_1, \dots, I_n)z^{n-1} + \dots + \beta_n(I_1, \dots, I_n).$$
 (6)

Set the coefficients of the desired characteristic polynomial (8) equal to the coefficients of (6) to obtain the system of equations

$$\beta_1(I_1, \dots, I_n) = \alpha_1$$
$$\beta_2(I_1, \dots, I_n) = \alpha_2$$
$$\vdots$$
$$\beta_n(I_1, \dots, I_n) = \alpha_n$$



Pole placement by state feedback

The system of equations

$$\beta_1(I_1, \dots, I_n) = \alpha_1$$

$$\beta_2(I_1, \dots, I_n) = \alpha_2$$

$$\vdots$$

$$\beta_n(I_1, \dots, I_n) = \alpha_n$$

is always linear in the parameters of the controller, henc

$$ML^{\mathrm{T}} = \alpha,$$

where
$$\alpha^{\mathrm{T}} = \begin{bmatrix} \alpha_1 & \alpha_2 & \cdots & \alpha_n \end{bmatrix}$$
.

Pole placement and controllability

It can be shown that the controllability matrix

$$W_c = \begin{bmatrix} \Gamma & \Phi \Gamma & \cdots & \Phi^{n-1} \Gamma \end{bmatrix}$$

is a factor of the matrix M

$$M = \bar{M}W_c$$
.

Hence, in general, the equations

$$\bar{M}W_cL^{\mathrm{T}} = \alpha \qquad \Rightarrow \qquad L^{\mathrm{T}} = W_c^{-1}\bar{M}^{-1}\alpha$$
 (7)

only has a solution if W_c is invertible, that is when the system is *controllable*.

Note that the equations (7) may also have solutions when the system is not controllable, if α is in the column space of M. That is

$$\alpha = b_1 M_{:,1} + b_2 M_{:,2} + \cdots + b_{M:,m}, \ m < n$$



Pole placement by state feedback

Given a desired placement of the closed-loop poles p_1, p_2, \ldots, p_n , being roots of the desired characteristic polynomial

$$a_c(z) = (z - p_1)(z - p_2) \cdots (z - p_n) = z^n + \alpha_1 z^{n-1} + \cdots + \alpha_n.$$
 (8)

and closed-loop system

$$x(k+1) = (\Phi - \Gamma L)x(k) + l_0 \Gamma u_c(k)$$

$$y(k) = Cx(k)$$
(9)

The Matlab (control systems toolbox) has methods for computing the gain vector L

1. Ackerman's method

2. Numerically more stable method

The reference input gain I_0

The closed-loop state space system

$$x(k+1) = \underbrace{(\Phi - \Gamma L)}_{\Phi_c} x(k) + l_0 \Gamma u_c(k)$$
$$y(k) = Cx(k)$$

has the steady-state solution (x(k+1) = x(k)) for constant reference signal $u_c(k) = u_{c,f}$

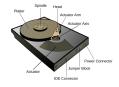
$$y_f = I_0 C (I - \Phi_c)^{-1} \Gamma u_{c,f}.$$

We want $y_f = u_{c,f}$,

$$\Rightarrow I_0 = \frac{1}{C(I - \Phi_c)^{-1} \Gamma}$$

Exercise - The harddisk drive arm

The model of the arm of the harddisk drive



can, with suitable choice of sampling period, be written

$$x(k+1) = \Phi x(k) + \Gamma u(k) = \begin{bmatrix} 1 & 0.4 \\ 0 & 1 \end{bmatrix} x + \begin{bmatrix} 0.16 \\ 0.8 \end{bmatrix} u.$$

With linear state feedback $u(k) = -Lx(k) + l_0u_c(k)$ the closed-loop system is

$$\begin{aligned} x(k+1) &= (\Phi - \Gamma L) x(k) + l_0 \Gamma u_c(k) \\ &= \begin{bmatrix} 1 - 0.16l_1 & 0.4 - 0.16l_2 \\ -0.8l_1 & 1 - 0.8l_2 \end{bmatrix} x(k) + l_0 \Gamma u_c(k). \end{aligned}$$

Determine the characteristic polynomial of the closed-loop system $\det \left(zI - (\Phi - \Gamma L)\right)$