

and by making  $\theta$  equal to a right angle, which will render the axis of  $z'$  perpendicular to the axis of  $z''$ , we shall have  $C=A$ ; therefore the moments of inertia relative to all axes situated in the plane perpendicular to the axis of  $z''$  are then equal to each other. But it is easy to be assured that we have in this case for the system of the axis of  $z'$ , and of any two axes perpendicular to each other, and to this axis,

$$S. x'y'.dm = 0; \quad S. x'z''.dm = 0; \quad S. y'z''.dm = 0;$$

for if we denote by  $x''$  and  $y''$  the coordinates of a molecule  $dm$  referred to the principal axes, taken in the plane perpendicular to the axis of  $z''$ , and with respect to which the moments of inertia are supposed equal, we shall have

$$S.(x''^2 + y''^2).dm = S.(y''^2 + z''^2).dm;$$

or simply  $S.x''^2 dm = S.y''^2 dm$ ; but by naming  $\epsilon$  the angle which the axis of  $x'$  makes with the axis of  $x''$ , we have

$$x' = x'' \cdot \cos. \epsilon + y'' \cdot \sin. \epsilon;$$

$$y' = y'' \cdot \cos. \epsilon - x'' \cdot \sin. \epsilon;$$

consequently we have

$$\begin{aligned} S. x'y'.dm &= S. x''y''.dm (\cos. \epsilon \cdot \sin. \epsilon) \\ &\quad + S. (y''^2 - x''^2).dm \cdot \sin. \epsilon \cdot \cos. \epsilon = 0 \end{aligned}$$

we shall find in like manner  $S. x'z''.dm = 0$ ;  $S. y'z''.dm = 0$ ; therefore all axes perpendicular to the axis of  $z''$ , are in this case principal axes; and in this case the solid has an infinite number of similar axes.

follows from the equations  $x' = x'' \cdot \cos. \epsilon + y'' \cdot \sin. \epsilon$ ,  $y' = y'' \cdot \cos. \epsilon - x'' \cdot \sin. \epsilon$  for  $S. x'^2 dm = S.(x''^2 \cos. \epsilon^2 + S y''^2 \sin. \epsilon^2) = S. x''^2 dm = S. y''^2 dm$ , since  $Sx''y''.dm = 0$ , in the case of an ellipsoid generated by the revolution of an ellipse above its minor axis, we have always two of the principal moments of inertia equal, the moment which is the greatest is referred to the minor axis.

If we have at the same time  $A=B=C$ ; we shall have generally  $C=A$ ; \* that is to say, all the moments of inertia of the solid are equal, but then we have generally,

$$\Sigma.x'y'.dm=0; \Sigma.x'z'.dm=0; \Sigma.y'z'.dm=0;$$

whatever may be the position of the plane of  $x'$  and of  $y'$ ; so that all the axes are principal axes. This is the case of the sphere, and we shall see in the sequel that this property belongs to an infinite number of other solids of which the equation will be given.†

\* Since by hypothesis  $A=B=C$ ,  $\Sigma x''^2 dm = \Sigma y''^2 dm = \Sigma z''^2 dm$ ,  $\therefore$  if in the expression for  $z''^2$  in terms of  $x'', y'', z''$ , and of the angle  $\theta, \psi, \phi$ , we take this into account and also observe that  $\Sigma x''y''dm$ ,  $\Sigma x''z''dm$ ,  $\Sigma y''z''dm$ , are equal to nothing, we shall find  $\Sigma z''^2 dm = \Sigma z''^2 dm$  for  $z''=z''\cos\theta-y''\sin\theta\cos\phi-x''\sin\theta\sin\phi$ .  $\therefore z''^2=z''^2\cos^2\theta+y''^2\sin^2\theta\cos^2\phi+x''^2\sin^2\theta\sin^2\phi$  (when  $x''^2=y''^2=z''^2$ ) the same is true respecting  $y''^2$  and  $x''^2$  on the other hand if we equate  $z''^2$  and its value in a function of  $x', y', z'$  and the angles  $\theta, \psi, \phi$ , and also satisfy the equations  $\Sigma x'^2 dm = \Sigma y'^2 dm = \Sigma z'^2 dm$ , we must equate  $\Sigma x'y'dm$ ,  $\Sigma x'z'dm$ ,  $\Sigma y'z'dm$  to nothing. (See Book V. Chap. I. No. 2.)

†  $x'', y'', z''$  being the coordinates with respect to the principal axes of any point of the solid, if we transfer the origin to a point of which the coordinates are  $a, b, c$ , then the coordinates relative to the new origin will be  $x''-a, y''-b, z''-c$ , now if we suppose that the three principal moments of inertia with respect to this new origin are equal, then all rectangular axes, and  $\therefore$  the axes of  $x''-a, y''-b, z''-c$ , will be principal axes, consequently we shall have

$$\Sigma.(x''-a)(y''-b).dm = \Sigma.x''y''.dm - a \Sigma.y''dm - b \Sigma.x''dm + ab \Sigma dm = 0$$

$$\Sigma.(x''-a)(z''-c).dm = \Sigma.x''z''.dm - a \Sigma.z''dm - c \Sigma.x''dm + ac \Sigma dm = 0$$

$$\Sigma.(y''-b)(z''-c).dm = \Sigma.y''z''.dm - b \Sigma.z''dm - c \Sigma.y''dm + bc \Sigma dm = 0$$

now if we suppose the origin of the coordinates  $x'', y'', z''$ , to be at the centre of gravity the preceding equations will be reduced to  $ab \Sigma dm = 0$ ,  $ac \Sigma dm = 0$ ,  $bc \Sigma dm = 0$   $\therefore$  two of the preceding quantities must vanish, let  $b, c$ , be equal to nothing and  $a$  will be undetermined,  $\therefore$  the point required will be at a distance equal to  $a$  from the origin by a foregoing note the moments of inertia with respect to this point will be  $A, B+ma^2, C+ma^2$  and by the conditions of the problem they are supposed to be equal  $\therefore$  we have  $a = \pm$

$$\sqrt{\frac{A-C}{m}}, \therefore A$$
 being greater than  $C$  we have two values of  $a$  equally distant on

28. The quantities  $p, q, r$ , which we have introduced in the equations (C) of No. 26. have this remarkable property, that they determine the position of the real and instantaneous axis of rotation with respect to the principal axes. In fact, we have relatively to all points situated in the axis of rotation,  $dx'=0; dy'=0; dz'=0$ ; if we difference the values of  $x', y', z'$ , of No. 26, and then make  $\sin. \psi=0$  after the dif-

B B

opposite sides from the centre of gravity, but  $a$  is also equal to  $\sqrt{\frac{A-B}{\sqrt{m}}}$  ∵ in order that these

two values of  $a$  should be possible, it is requisite that  $B$  should be equal to  $C$ , ∵ when  $A \neq C$  are unequal there is no point which satisfies the required conditions and when two of the moments are equal, the third must be greater than either of them, and in this case the point required is situated on the axis relative to which the principal moment of inertia is the greatest, when the three moments of inertia are equal the two points are concentrated in the common centre of gravity. When  $B=C$  we have  $S.y'^2 dm = S.z'^2 dm$ .

In an ellipsoid generated by the revolution of an ellipse of an ellipse round its minor axis two of the three principal moments relative to the principal diameters are equal, and the greatest moment is relative to the minor axis, see note page 181, ∵ we shall have two points existing on this axis relatively to which all the moments of inertia are equal, it is easy to shew that the distance of those points from the centre of the ellipsoid is  $=$  to the square root of the fifth part of the difference between the squares of the semi-axes, and ∵ they may be within the ellipsoid, at its surface, or finally without this surface.

We might have inferred a priori that there is an axis with respect to which the moment of inertia is a maximum and a minimum, for from their nature all moments of inertia are positive and have a finite magnitude, and most authors deduce the properties of principal axes from the moments of inertia which are the greatest and least, the general expression for  $S.(x'^2 + y'^2).dm$  in terms of  $x', y'$  and  $z'$  is equal to

$$\begin{aligned} & S.x'^2 dm \cos. \theta \sin. \psi + S.x'^2 dm \cos. \theta \sin. \psi + S.y'^2 dm \cos. \theta \cos. \psi + S.y'^2 dm \sin. \theta \cos. \psi \\ & + S.z'^2 dm \sin. \theta + 2 Sx'y' dm \cos. \theta \sin. \psi \cos. \psi \\ & - 2 Sx'y' dm \sin. \psi \cos. \psi - 2 S.z'x' dm \sin. \theta \cos. \theta \sin. \psi - 2 S.z'y' dm \sin. \theta \cos. \theta \cos. \psi. \end{aligned}$$

When the law of the variation of the density and the equation of the generating curve of a solid of revolution are given, the value of  $S.(x'^2 + y'^2).dm$  may be computed by a method similar to that by which the centre of gravity of a body is determined; the value of  $S(x^2 + y^2).dm$  is computed for the earth in Book V. Chapter 1. No. 2.

ferentiations which we are permitted to do, since the position of the axis of  $x'$  on the plane of  $x', y'$ , is indeterminate, we shall have

$$\begin{aligned} dx' = & x'' \cdot \{d\psi \cdot \cos. \theta. \sin. \phi - d\phi \cdot \sin. \phi\} + y'' \cdot \{d\psi \cdot \cos. \theta. \cos. \phi \\ & - d\phi. \cos. \phi\} + z'' \cdot d\psi. \sin. \theta = 0; \end{aligned}$$

$$\begin{aligned} dy' = & x'' \cdot \{d\phi. \cos. \theta. \cos. \phi - d\theta. \sin. \theta. \sin. \phi - d\psi. \cos. \phi\} \\ & + y'' \cdot \{d\psi. \sin. \phi - d\phi. \cos. \theta. \sin. \phi - d\theta. \sin. \theta. \cos. \phi\} \\ & + z'' \cdot d\theta. \cos. \theta = 0; \end{aligned}$$

$$\begin{aligned} dz' = & -x'' \cdot (d\theta. \cos. \theta. \sin. \phi + d\phi. \sin. \theta. \cos. \phi) \\ & - y'' \cdot (d\theta. \cos. \theta. \cos. \phi - d\phi. \sin. \theta. \sin. \phi) - z'' \cdot d\theta. \sin. \theta = 0. \end{aligned}$$

If we multiply the first of these equations by  $-\sin. \phi$ ; the second by  $\cos. \theta. \cos. \phi$ , and the third by  $-\sin. \theta. \cos. \phi$ ; we shall have by adding them together,

$$0 = px'' - qz''.$$

Multiplying the first of the same equations by  $\cos. \phi$ ; the second by  $\cos. \theta. \sin. \phi$ , and the third by  $-\sin. \theta. \sin. \phi$ , and then adding them together we shall obtain

$$0 = py'' - rz''.$$

Finally, if we multiply the second of those equations by  $\sin. \theta$ , and the third by  $\cos. \theta$ , and then add them together, their addition will give \*

$$0 = qy'' - rx''$$

\* In taking the differentials of  $dx'$ ,  $dy'$ ,  $dz'$ , we may omit those quantities in which  $\sin. \psi$  occurs after the differentiations, and where  $\cos. \psi$  occurs, we may substitute unity; multiplying the value of  $dx'$  which results by  $-\sin. \phi$ , it becomes

$$\begin{aligned} -dx' \cdot \sin. \phi = & -x'' \cdot (d\psi \cdot \cos. \theta. \sin. {}^2 \phi - d\phi. \sin. {}^2 \phi) - y'' \cdot (d\psi \cdot \cos. \theta. \sin. \phi \cos. \phi \\ & - d\phi. \sin. \phi. \cos. \phi) - z'' \cdot d\psi. \sin. \theta. \sin. \phi; \end{aligned}$$

This last equation evidently results from the two preceding; thus the three equations  $dx' = 0$ ,  $dy' = 0$ ,  $dz' = 0$  reduce themselves to these two equations which belong to a right line, forming with the axes of  $x'$ ,

$$\text{B B } 2$$

and in like manner multiplying  $dy'$  and its value by  $\cos \theta \cdot \cos \varphi$ , we have

$$\begin{aligned} dy' \cdot \cos \theta \cdot \cos \varphi &= x''(d\phi \cdot \cos^2 \theta \cdot \cos^2 \varphi - d\theta \cdot \sin \theta \cdot \cos \theta \cdot \sin \varphi \cdot \cos \varphi - d\psi \cdot \cos \theta \cdot \cos^2 \varphi) \\ &+ y''(d\psi \cdot \sin \varphi \cdot \cos \varphi \cdot \cos \theta - d\phi \cdot \cos^2 \theta \cdot \sin \varphi \cdot \cos \varphi - d\theta \cdot \sin \theta \cdot \cos \theta \cdot \cos^2 \varphi) \\ &+ z''(d\theta \cdot \cos^2 \theta \cdot \cos \varphi) \end{aligned}$$

and the multiplication of  $dz'$ , and its value by  $-\sin \theta \cdot \cos \varphi$ , gives

$$\begin{aligned} -dz' \cdot \sin \theta \cdot \cos \varphi &= x''(d\theta \cdot \sin \theta \cdot \cos \theta \cdot \sin \varphi \cdot \cos \varphi + d\phi \cdot \sin^2 \theta \cdot \cos^2 \varphi) \\ &+ y''(d\theta \cdot \sin \theta \cdot \cos \theta \cdot \cos^2 \varphi - d\phi \cdot \sin^2 \theta \cdot \sin \varphi \cdot \cos \varphi) + z''(d\theta \cdot \sin^2 \theta \cdot \cos^2 \varphi) \end{aligned}$$

adding these quantities together and making the factors of the differentials of  $\theta, \psi, \varphi$ , which belong to the same coordinates coalesce we obtain

$$-x''(d\psi \cdot \cos \theta + d\varphi) + z''(d\theta \cdot \cos \varphi - d\psi \cdot \sin \theta \cdot \sin \varphi) = 0 =$$

(by substituting  $p$  and  $q$  instead of their values)  $x' \cdot p - z' \cdot q$ ; multiplying the first equation by  $\cos \varphi$ , the second by  $\cos \theta \cdot \sin \varphi$ , and the third by  $-\sin \theta \cdot \sin \varphi$ , we obtain

$$\begin{aligned} dx' \cdot \cos \varphi &= x''(d\psi \cdot \cos \theta \cdot \sin \varphi \cdot \cos \varphi - d\phi \cdot \sin \varphi \cdot \cos \varphi) \\ &+ y''(d\psi \cdot \cos \theta \cdot \cos^2 \varphi - d\phi \cdot \cos^2 \varphi) + z''(d\psi \cdot \sin \theta \cdot \cos \varphi) \\ dy' \cdot \cos \theta \cdot \sin \varphi &= x''(d\phi \cdot \cos^2 \theta \cdot \sin \varphi \cdot \cos \varphi - d\theta \cdot \sin \theta \cdot \cos \theta \cdot \sin^2 \varphi \\ &- d\psi \cdot \sin \varphi \cdot \cos \varphi \cdot \cos \theta) \\ &+ y''(d\psi \cdot \sin^2 \varphi \cdot \cos \theta - d\phi \cdot \cos^2 \theta \cdot \sin^2 \varphi) + z''(d\theta \cdot \sin \theta \cdot \cos \theta \cdot \sin \varphi \cdot \cos \varphi) \\ dz' \cdot \sin \theta \cdot \sin \varphi &= x''(d\theta \cdot \sin \theta \cdot \cos \theta \cdot \sin^2 \varphi + d\phi \cdot \sin^2 \theta \cdot \sin \varphi \cdot \cos \varphi) \\ &+ y''(d\theta \cdot \sin \theta \cdot \cos \theta \cdot \sin \varphi \cdot \cos \varphi - d\phi \cdot \sin^2 \theta \cdot \sin^2 \varphi) + z''(d\theta \cdot \sin^2 \theta \cdot \sin \varphi) \end{aligned}$$

adding and concinnating as before we obtain

$$y''(d\psi \cdot \cos \theta - d\varphi) + z''(d\psi \cdot \sin \theta \cdot \cos \varphi + d\theta \cdot \sin \varphi) = 0 = -y''p + z''r$$

of  $y''$  and of  $z''$ , angles of which the cosines are

$$\frac{q}{\sqrt{p^2+q^2+r^2}}, \quad \frac{r}{\sqrt{p^2+q^2+r^2}}, \quad \frac{p}{\sqrt{p^2+q^2+r^2}} *$$

multiplying the second equation by  $\sin. \theta$ , and the third by  $\cos. \theta$ , we obtain

$$\begin{aligned} dy'. \sin. \theta &= x''.(d\phi. \cos. \varphi. \sin. \theta. \cos. \theta - dt. \sin. {}^2 \theta. \sin. \varphi - d\psi. \sin. \theta. \cos. \varphi) \\ &+ y''.(d\psi. \sin. \theta. \sin. \varphi - d\phi. \sin. \theta. \cos. \theta. \sin. \varphi - dt. \sin. {}^2 \theta. \cos. \varphi) + z''.dt. \sin. \theta. \cos. \theta. \\ dz'. \cos. \theta &= -x''.(dt. \cos. {}^2 \theta. \sin. \varphi + d\phi. \sin. \theta. \cos. \theta. \cos. \varphi) \\ &- y''.(dt. \cos. {}^2 \theta. \cos. \varphi - d\phi. \sin. \theta. \cos. \theta. \sin. \varphi) - z''.dt. \sin. \theta. \cos. \theta \end{aligned}$$

∴ adding and concinnating we have

$$-x''.(dt. \sin. \varphi + d\psi. \sin. \theta. \cos. \varphi) - y''.(dt. \cos. \varphi - d\psi. \sin. \theta. \sin. \varphi) = -x''r + y''q.$$

\* The equations  $px'' - qz'' = 0$  &c. are the equations of the projections of the line, relatively to which  $dx' dy'$  are equal to nothing at any instant, on the planes  $x''z'', y''z'$ , &c. ∴ the cosines of the angles which this line makes with the axes are respectively

$$\frac{q}{\sqrt{p^2+q^2+r^2}}, \quad \frac{r}{\sqrt{p^2+q^2+r^2}}, \quad \frac{p}{\sqrt{p^2+q^2+r^2}}.$$

For these cosines are equal to

$$\frac{u'}{\sqrt{x''^2+y''^2+z''^2}} = \frac{\frac{qz''}{p}}{\sqrt{\frac{q^2z''^2}{p^2} + \frac{r^2z''^2}{p^2} + z''^2}} = \frac{q}{\sqrt{p^2+r^2+q^2}}$$

and the same is true of the other cosines.

From the preceding analysis it follows, that the locus of all the points whose velocity is nothing at any given moment is a right line, whose position with respect to the principal axes is determined by  $p, q, r$ , ∴ the preceding equations both evince the existence of such a line and indicate its position, and a body revolving about a fixed point may be considered as revolving about an axis determined in this manner, but as in general  $p, q, r$ , vary from one instant to another, being functions of the time, the position of this axis will also vary, and hence it is that this axis has been termed by some authors the axis of instantaneous rotation ; when  $p, q, r$ , are constant, the axis of rotation will remain immovable during the motion of the system.

Therefore this right line quiesces, and constitutes the real axis of rotation of the body. \*

\* The values which have been given for  $px'' - qz''$ ,  $py'' - rz''$ ,  $qy'' - rx''$ , enables us to determine the linear velocity of each point resolved parallel to the axes of  $x'$   $y'$  and  $z'$  for if we multiply the first of the preceding equations by  $\cos. \theta. \cos. \phi$ . the second by  $\cos. \theta. \sin. \phi$ . and the third by  $\sin. \theta$ . we shall obtain by adding them together

$$\begin{aligned} & -dx'. \cos. \theta. \sin. \phi. \cos. \phi + dy'. \cos. {}^2\theta. \cos. {}^2\phi - dz. \sin. \theta. \cos. \theta. \cos. {}^2\phi \\ & + dx'. \cos. \theta. \sin. \phi. \cos. \phi + dy'. \cos. {}^2\theta. \sin. {}^2\phi - dz. \sin. \theta. \cos. \theta. \sin. {}^2\phi + dy'. \sin. {}^2\phi \\ & + dz'. \sin. \theta. \cos. \theta = dy' = (px'' - qz''). \cos. \theta. \cos. \phi + (py'' - rz''). \cos. \theta. \sin. \phi \\ & + (qy'' - rx''). \sin. \theta; \text{ if we multiply } px'' - qz'' \text{ by } -\sin. \phi \text{ and } py'' - rz'' \\ & \text{by } \cos. \phi \text{ we shall obtain} \\ & dt'. \sin. {}^2\phi - dy'. \cos. \theta. \sin. \phi. \cos. \phi + dz'. \sin. \theta. \sin. \phi. \cos. \phi + dx'. \cos. {}^2\phi \\ & + dy'. \cos. \theta. \sin. \phi. \cos. \phi - dz. \sin. \theta. \sin. \phi. \cos. \phi = dx' \\ & = -(px'' - qz''). \sin. \phi + (py'' - rz''). \cos. \phi; \text{ multiplying } px'' - qz'' \text{ by } -\sin. \theta. \cos. \phi, \\ & \quad py'' - rz'' \text{ by } -\sin. \theta. \sin. \phi. \text{ and } qy'' - rx'' \text{ by } \cos. \theta; \end{aligned}$$

we shall obtain

$$\begin{aligned} & dx'. \sin. \theta. \sin. \phi. \cos. \phi - dy'. \sin. \theta. \cos. \theta. \cos. {}^2\phi + dz'. \sin. {}^2\theta. \cos. {}^2\phi \\ & - dx'. \sin. \theta. \sin. \phi. \cos. \phi - dy'. \sin. \theta. \cos. \theta. \sin. {}^2\phi + dz'. \sin. {}^2\theta. \sin. {}^2\phi + dy'. \sin. \theta. \cos. \theta. \\ & + dz'. \cos. {}^2\theta = dz' = -(px'' - qz''). \sin. \theta. \cos. \phi - (py'' - rz''). \sin. \theta. \sin. \phi + (qy'' - rx''). \cos. \theta; \end{aligned}$$

we might in like manner obtain the value of the accelerating forces resolved parallel to the axes of  $x'$   $y'$  and  $z'$ , by taking the differentials of  $dz'$ ,  $dy'$ ,  $dz'$ , and of their respective values.

Since as has been observed, in note, page 166, the coordinates of  $x''$ ,  $y''$ ,  $z''$ , do not vary with the time, and as the angles  $\theta$ ,  $\psi$ ,  $\phi$ , are functions of the time, it follows that when we take the differential of  $x''$   $y''$  and  $z''$  respectively in terms of the coordinates  $x'$ ,  $y'$ ,  $z'$ , and of the angles  $\theta$ ,  $\psi$ ,  $\phi$ , the sines and cosines of these angles must be considered as constant, ∴ keeping this in view and also that  $\sin. \psi = 0$  after the differentiations we shall obtain

$$\begin{aligned} dx'' &= dx'. \cos. \phi + dy'. \cos. \theta. \sin. \phi - dz'. \sin. \theta. \sin. \phi; dy'' = -dx'. \sin. \phi + dy'. \cos. \theta. \cos. \phi \\ & - dz'. \sin. \theta. \cos. \phi; dz'' = dy'. \sin. \theta + dz'. \cos. \theta; \end{aligned}$$

In order to determine the velocity of rotation of the body, let us consider that point of the axis of  $z''$ , of which the distance from the origin of the coordinates is represented by a quantity equal to unity. We shall have the velocities parallel to the axes of  $x'$  of  $y'$  and of  $z'$ , by making  $x''=0$ ,  $y''=0$ ,  $z''=1$ , in the preceding expressions of  $dx'$ ,  $dy'$ ,  $dz'$ , and then dividing them by  $dt$ , which gives for these partial velocities

$$\frac{d\psi}{dt} \cdot \sin. \theta; \frac{d\theta}{dt} \cdot \cos. \theta; -\frac{d\theta}{dt} \cdot \sin. \theta;$$

therefore the entire velocity of the point in question, is  $\sqrt{\frac{d\theta^2 + d\psi^2 \cdot \sin. ^2 \theta}{dt}}$

or  $\sqrt{q^2 + r^2}$ , and dividing this expression by the distance of the point from the instantaneous axis of rotation, we shall have the angular velocity of rotation of the body; but this distance is evidently equal to the sine of the angle, which the real axis of rotation makes with the axis of  $z''$ , and the cosine of this angle is equal to  $\frac{p}{\sqrt{p^2 + q^2 + r^2}}$ ;

but it is evident from what precedes that the second members of these equations are equal respectively to

$$(py'' - rz''), dt, (px'' - qz''), dt, (qy'' - rx''), dt$$

∴ we have

$$\frac{dx''}{dt} = py'' - rz'', \frac{dy''}{dt} = px'' - qz'', \frac{dz''}{dt} = qy'' - rx'',$$

consequently the quantities  $p, q, r$ , which determine the position of the axes of rotation, give also for any other point the linear and angular velocities of the different points of the body resolved parallel to the coordinates  $x'', y'',$  and  $z''$ .

therefore  $\sqrt{p^2+q^2+r^2}$  will be equal to the angular velocity of rotation.\*

It appears from what precedes that whatever may be the rotatory motion of a body, about a fixed point, or a point considered as fixed; this motion must be considered as a motion of rotation about a fixed axis *during* an instant, but which may vary from one moment to another. The position of this axis with respect to the principal axes, and the angular velocity of rotation depend on the variables  $p, q, r$ , the determination of which is most important in these investigations, and as they express quantities independent of the situation of the plane of  $x'$  and  $y'$ , are themselves independent of this situation.

29. Let us proceed to determine these variables in functions of the time, in that case in which the body is not solicited by any accelerating forces. For this purpose, let us resume the equations (*D*) of No. 26, existing between the variables  $p', q', r'$ , which are in a given ratio to

$$* \frac{p}{\sqrt{p^2+q^2+r^2}}$$

= the cosine of the angle which the axis of  $z'$  makes with the instantaneous axis of rotation :.

$$1 - \frac{p^2}{p^2+q^2+r^2} = \frac{q^2+r^2}{p^2+q^2+r^2}$$

is equal to the square of the sine of the same angle, and since  $z''$  is by hypothesis equal to unity we have the perpendicular distance of the point in question from the axis of rotation equal to this sine, ∴ dividing  $\sqrt{q^2+r^2}$  by this distance the quote will be equal to  $\sqrt{p^2+q^2+r^2}$ , and as the axis during an instant may be considered as fixed the angular velocity of all points during this instant will be the same, the selection of the point so circumstanced that  $x''=0, y''=0, z''=1$  is made in order to simplify the calculus, ∴ it appears from an inspection of the value of the angular velocity, that it is constant when  $p, q$  and  $r$  are constant, i.e., when the axis of rotation is immovable, but the converse of this proposition is not true for it is possible that the function  $\sqrt{p^2+q^2+r^2}$  should be constant, while at the same time its component parts may vary, see page 197.

the variables  $p, q, r, \dots$ \* In this case, the differentials  $dN, dN', dN''$  vanish, and these equations being multiplied by  $p', q',$  and  $r'$  respectively and then added together give

$$0 = p'.dp' + q'.dq' + r'.dr';$$

and integrating them we shall obtain

$$p'^2 + q'^2 + r'^2 = k^2;$$

$k$  being a constant arbitrary quantity.

If we multiply the equations (D) by  $AB.p', BC.q',$  and  $AC.r',$  and then add them together, we shall obtain by integrating their sum,

$$AB.p'^2 + BC.q'^2 + AC.r'^2 = H^2;$$

$H$  being a constant arbitrary quantity; this equation involves the principle of the conservation of living forces. † By means of the two pre-

\*  $p:p':C::1:1:S(x'^2+y'^2).dm$ , but this is a constant ratio, because the position of the principal axes being given, the quantity  $S(x'^2+y'^2).dm$  is constant, and when no exterior forces act on the body, the quantities  $N, N', N'',$  are constant and  $\therefore dN dN' dN''$  vanish.

† For substituting for  $p', q', r',$  their values, we obtain

$$\begin{aligned} A.B.C(Cp^2+Aq^2+Br^2)=H^2, \therefore S(x'^2+y'^2)dm \cdot p^2 + S(y'^2+z'^2)dm \cdot q^2 \\ + S(x'^2+z'^2)dm \cdot r^2 = \end{aligned}$$

a constant quantity, now we have seen in a preceding note, that the velocity of any point resolved parallel to the axes of  $x''$  of  $y''$  and of  $z''$  is equal to  $rx''-qz'', py''-rz'', qy''-rx''$

and the sum of the squares of these quantities

$$= p^2x''^2 + q^2z''^2 + r^2y''^2 + r^2z''^2 + q^2y''^2 + r^2x''^2 - 2pq \cdot x''z'' - 2pr \cdot y''z'' - 2qr \cdot y''x'' =$$

the square of the velocity of the point whose coordinates are  $x'', y'', z'', \therefore$  this expression multiplied by  $dm$  equals the living force of this molecule, now as the quantities  $p, q, r,$  are the same for all molecules at the same instant, the sum of the living forces of all the

ceding integrals we shall obtain

$$q'^2 = \frac{AC.k^2 - H^2 + A.(B-C).p'^2}{C.(A-B)} *$$

$$r'^2 = \frac{H^2 - BC.k^2 - B.(A-C).p'^2}{C(A-B)}$$

thus, we shall have  $q'$  and  $r'$  in functions of the time, when  $p'$  will be determined, but from the first of the equations (*D*) we have

$$dt = \frac{AB.dp'}{(A-B).q'r'} ;$$

consequently

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molecules will be equal to

$$\begin{aligned} p^2 \int (x'^2 + y'^2).dm + q^2 \int (y'^2 + z'^2).dm + r^2 \int (x'^2 + z'^2).dm \\ - 2pq \int x'z'.dm - 2pr \int y'z'.dm - 2qr \int y'x'.dm, \end{aligned}$$

but these latter quantities vanish,  $x'', y'', z''$ , belonging to the principal axes consequently

$$p^2 \int (x'^2 + y'^2).dm + q^2 \int (y'^2 + z'^2).dm + r^2 \int (x'^2 + z'^2).dm$$

is equal to the sum of the living forces, and being constant as has been just shewn, it follows that the expression

$$AB.p'^2 + BC.q'^2 + AC.r'^2 = H^2,$$

involves the principle of the conservation of living forces.

$$* r'^2 = k^2 - p'^2 - q'^2 = \frac{H^2 - AB.p'^2 - BC.q'^2}{AC}$$

$$\therefore AC.k^2 - AC.p'^2 - AC.q'^2 = H^2 - AB.p'^2 - BC.q'^2 \text{ therefore } q'^2 =$$

$$\frac{AC.k^2 - H^2 + A.(B-C)p'^2}{C.(A-B)},$$

the value of  $r'^2$  is derived in a similar manner.

$$dt = \frac{ABC.dp'}{\sqrt{\{AC.k^2 - H^2 + A.(B-C).p'^2\} \cdot \{H^2 - BC.k^2 - B.(A-C).p'^2\}}} *$$

$$\begin{aligned} * \text{ When } A=B, dt &= \frac{ABC.dp'}{\sqrt{(AC.k^2 - H^2 + A.(A-C).p'^2) \cdot (H^2 - AC.k^2 - A(A-C).p'^2)}} \\ &= \frac{ABC.dp'}{\sqrt{- (AC.k^2 - H^2 + A(A-C).p'^2)^2}} = \end{aligned}$$

$$\frac{ABC.dp'}{AC.k^2 - H^2 + A(A-C).p'^2}$$

and it may be made to assume the form

$$C_r \frac{a^2 \cdot dp'}{a^2 + p'^2} \left( C_r \text{ being equal to } \frac{ABC}{AC.k^2 - H^2} \right)$$

$$\text{and } a^2 \text{ being equal to } \frac{AC.k^2 - H^2}{A(A-C)}$$

and the integral of this expression  $= t = C_r$  (arc tangent  $= p'$  to radius  $= a$ .  
the constant quantity is equal to nothing because  $t=0$  at the same time with  $p'$ .

When  $A=C$  the expression for  $dt$  becomes

$$\frac{ABC.dp'}{\sqrt{(A^2.k^2 - H^2 + A.(B-A).p'^2) \cdot (H^2 - BA.k^2)}}$$

this expression may be reduced to the form

$$C_r \frac{dp'}{\sqrt{a^2 + p'^2}} \left( \text{in which } C_r \text{ is equal to } \frac{A^2.B}{A(B-A).(H^2 - BA.k^2)} \right)$$

$$\text{and } a^2 = \frac{Ak^2 - H^2}{A(B-A)}$$

the integral  $= C_r \log (p' + \sqrt{a^2 + p'^2})$

$$\text{If } B=C \text{ then } dt \frac{AB^2 dp'}{\sqrt{(AC.k^2 - H^2)(H^2 - B^2.k^2 - B)(A-B)}.p'^2} = -C_r \frac{a \cdot dp'}{\sqrt{a^2 - p'^2}}$$

and the integral will be arc sine  $= p'$  rad  $= a$

$$C_r \text{ being equal to } \frac{AB^2}{\sqrt{AC.k^2 - H^2 \cdot (H^2 - B^2.k^2)}} .$$

$$\text{and } a^2 = \frac{(AC.k^2 - H^2) \cdot (H^2 - B.k^2)}{-B.(A-B)}$$

if  $AC.k^2 = H^2$  then

this equation is only integrable in one of the three following cases,  $B=A$ ,  $B=C$ ,  $A=C$ .

The determination of the three quantities  $p'$ ,  $q'$ ,  $r'$ , involves three arbitrary quantities,  $H^2$ ,  $k^2$  and that which the integration of the preceding differential equation introduces. But these quantities only give the position of the instantaneous axis of rotation of the body, on the surface,  $i$ ,  $e$ , with respect to the three principal axes, and its angular velocity of rotation. In order to have the real motion of the body, about the fixed point, we must also know the position of the principal axes in space; \* this should introduce three new arbitrary quantities

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$$q'^2 = \frac{A.(B-C).p'^2}{C(A-B)} \text{ and } dt = \frac{ABC dp'}{\sqrt{A(B-C)p'^2(H^2-BCk^2-B.(A-C)p'^2)}} \\ = C_r \frac{2adp}{p' \sqrt{a^2-p'^2}}$$

in which  $2C_r = \frac{ABC}{\sqrt{(A(B-C)H^2-BC.k^2)}}$

and  $a^2 = \frac{A(B-C).(H^2-BCk^2)}{-B.(A-C)}$

its integral will be equal to  $C_r \log \frac{a-\sqrt{a^2-p'^2}}{a+\sqrt{a^2-p'^2}}$

See Lacroix, page 256, No. 174. and if  $H^2 = BC.k^2$  then

$$dt = \frac{ABC dp'}{\sqrt{ACK^2-H^2+A(B-C)p'^2(-B(A-C)p'^2)}}$$

$$= C_r \frac{2adp'}{p' \sqrt{a+p'^2}} \text{ and } t = C_r \log \frac{\sqrt{a^2+p'^2}-a}{\sqrt{a^2+p'^2}+a},$$

the constant quantities vanish for these integrals, because as has been already mentioned  $p'=0$  when  $t$  vanishes. The value of  $dt$  cannot be exhibited in a finite form except in the cases already specified, and when all the moments of inertia are equal, in every other case, the value of the integral of  $dt$  must be obtained by the method of quadratures.

\* From the quantities  $p'$ ,  $q'$ ,  $r'$ , we can collect the values of  $p$ ,  $q$ ,  $r$ , which are in a given ratio to them, and from these last quantities we obtain the cosines of the angles which the axis of instantaneous rotation makes with the principal axes, but as these axes though fixed in the body are moveable in space, we must know the position of these axes at the com-

which depend on the initial position of these axes, and which require three new integrals, which being joined to the preceding quantities will commence of the motion, in order to have the real motion of the body, which gives three constant quantities.

Substituting in the values of  $-N, -N', -N'', p'$  for  $Cp, q'$  for  $Aq, r'$  for  $Br$  we shall have

$$\begin{aligned} & q' \cdot \sin. \theta \cdot \sin. \phi + r' \cdot \sin. \theta \cdot \cos. \phi - p' \cdot \cos. \theta = -N \\ & q' \cdot \cos. \theta \cdot \sin. \phi \cdot \cos. \psi + r' \cdot \cos. \theta \cdot \cos. \phi \cdot \cos. \psi + p' \cdot \sin. \theta \cdot \cos. \psi \end{aligned}$$

$$+ r' \cdot \sin. \phi \cdot \sin. \psi - q' \cdot \cos. \phi \cdot \sin. \psi = -N'$$

$$- q' \cdot \cos. \theta \cdot \sin. \phi \cdot \sin. \psi - r' \cdot \cos. \theta \cdot \sin. \psi \cdot \cos. \phi - p' \cdot \sin. \theta \cdot \sin. \psi$$

$$+ r' \cdot \sin. \phi \cdot \cos. \psi - q' \cdot \cos. \phi \cdot \cos. \psi = -N''$$

squaring these quantities we obtain

$$q'^2 \sin. {}^2 \theta \cdot \sin. {}^2 \phi + r'^2 \sin. {}^2 \theta \cdot \cos. {}^2 \phi + p'^2 \cos. {}^2 \theta + 2q'r' \cdot \sin. {}^2 \theta \cdot \sin. \phi \cdot \cos. \phi$$

$$- 2p'q' \cdot \sin. \theta \cdot \cos. \theta \cdot \sin. \phi - 2p'r' \cdot \sin. \theta \cdot \cos. \theta \cdot \cos. \phi = N^2$$

$$q'^2 \cos. {}^2 \theta \cdot \sin. {}^2 \phi \cos. {}^2 \psi + r'^2 \cos. {}^2 \theta \cdot \cos. {}^2 \phi \cos. {}^2 \psi + p'^2 \cos. {}^2 \theta \cdot \sin. {}^2 \psi$$

$$+ 2q'r' \cdot \cos. {}^2 \theta \cdot \sin. \phi \cdot \cos. \phi \cdot \cos. \psi + 2p'r' \cdot \sin. \theta \cdot \cos. \theta \cdot \cos. \phi \cdot \cos. \psi$$

$$+ 2p'q' \cdot \sin. \theta \cdot \cos. \theta \cdot \sin. \phi \cdot \cos. \psi)$$

$$+ r'^2 \sin. {}^2 \phi \cdot \sin. {}^2 \psi + q'^2 \cos. {}^2 \phi \cdot \sin. {}^2 \psi - 2q'r' \cdot \sin. \phi \cdot \cos. \phi \cdot \sin. \psi = N'^2$$

$$q'^2 \cos. {}^2 \theta \cdot \sin. {}^2 \phi \cdot \sin. {}^2 \psi + r'^2 \cos. {}^2 \theta \cdot \sin. {}^2 \psi \cos. {}^2 \phi + p'^2 \cos. {}^2 \theta \cdot \sin. {}^2 \psi$$

$$+ 2q'r' \cdot \cos. {}^2 \theta \cdot \sin. {}^2 \psi \sin. \phi \cdot \cos. \phi + 2p'r' \cdot \sin. \theta \cdot \cos. \theta \cdot \sin. {}^2 \psi \cos. \phi$$

$$+ 2p'q' \sin. \theta \cdot \cos. \theta \cdot \sin. {}^2 \psi \cdot \sin. \phi)$$

$$+ r'^2 \sin. {}^2 \phi \cdot \cos. {}^2 \psi + q'^2 \cos. {}^2 \phi \cdot \cos. {}^2 \psi - 2q'r' \cdot \sin. \phi \cdot \cos. \phi \cos. {}^2 \psi = N''^2$$

$\therefore$  adding the first members of these equations together we obtain

$$q'^2 \sin. {}^2 \theta \cdot \sin. {}^2 \phi + q'^2 \cos. {}^2 \theta \cdot \sin. {}^2 \phi + q'^2 \cos. {}^2 \phi = (q'^2) + r'^2 \sin. {}^2 \theta \cdot \cos. {}^2 \phi$$

$$+ r'^2 \cos. {}^2 \theta \cdot \cos. {}^2 \phi + r'^2 \sin. {}^2 \phi = (r'^2) + p'^2 \cos. {}^2 \theta + p'^2 \sin. {}^2 \theta \cdot \cos. {}^2 \psi$$

$$+ p'^2 \sin. {}^2 \theta \cdot \sin. {}^2 \psi = p'^2$$

the parts of these squares which are the products of two different quantities vanish when added together and in the expressions for  $N'^2, N''^2$  we omit the product  $(q' \cdot \cos. \theta \cdot \sin. \phi \cdot \cos. \psi + r' \cdot \cos. \theta \cdot \cos. \phi \cdot \cos. \psi + p' \cdot \sin. \theta \cdot \cos. \phi \cdot \cos. \psi) \cdot (r' \cdot \sin. \phi \cdot \sin. \psi - q' \cdot \cos. \phi \cdot \sin. \psi)$  for this product occurs in  $N'^2$  and  $N''^2$  affected with contrary signs,  $\therefore$  it must vanish from  $N'^2 + N''^2$   $\therefore$  we shall have

$$p'^2 + q'^2 + r'^2 = N^2 + N'^2 + N''^2.$$

completely solve the problem. The equations (*C*) of No. 26, involve three arbitrary quantities  $N, N', N''$ ; but they are not entirely distinct from the arbitrary quantities  $H$  and  $k$ . In fact, if we add together the squares of the first members of the equations (*C*), we shall have

$$p'^2 + q'^2 + r'^2 = N^2 + N'^2 + N''^2;$$

and consequently

$$k^2 = N^2 + N'^2 + N''^2.$$

The constant quantities  $N, N', N''$ , correspond to the constant quantities  $c, c', c''$ , of No. 21, and the function  $\frac{1}{2} \cdot t \cdot \sqrt{p'^2 + q'^2 + r'^2}$  expresses the sum of the areas described in the time  $t$ , by the projection of each molecule of the body on the plane relatively to which this sum is a *maximum*.  $N', N''$ , vanish with respect to this plane,  $\therefore$  if we put their values, which have been found in No. 26, equal to nothing we shall have

$$0 = Br \cdot \sin. \phi - Aq \cdot \cos. \phi;$$

$$0 = Aq \cdot \cos. \theta \cdot \sin. \phi + Br \cdot \cos. \theta \cdot \cos. \phi + Cp \cdot \sin. \theta;^*$$

\* From the equation  $Br \cdot \sin. \phi - Aq \cdot \cos. \phi = 0$  we obtain by substitution

$$\tan. \phi = \frac{q'}{r'} \therefore \cos. \phi = \frac{r'}{\sqrt{q'^2 + r'^2}} \text{ and } \sin. \phi = \frac{q'}{\sqrt{q'^2 + r'^2}}$$

consequently we have  $\frac{q'^2 + r'^2}{\sqrt{q'^2 + r'^2}} \cdot \cos. \theta$ .

$$= \sqrt{(q'^2 + r'^2)} \cdot \cos. \theta = p' \cdot \sin. \theta \therefore (q'^2 + r'^2) \cdot \cos. \theta = p'^2 - p'^2 \cos. \theta \therefore$$

$$\cos. \theta = \frac{p'}{\sqrt{p'^2 + q'^2 + r'^2}},$$

if we multiply the first of the preceding equations by  $\cos. \theta \cdot \sin. \phi$ , and the second by  $\cos. \phi$ , we shall obtain by adding them together  $r' \cdot \cos. \theta + p' \cdot \sin. \theta \cdot \cos. \phi = 0 \therefore$  substituting for  $\cos. \theta$  its value we obtain

from which we deduce

$$\cos. \theta = \frac{p'}{\sqrt{p'^2 + q'^2 + r'^2}} ;$$

$$\sin. \theta. \sin. \varphi = \frac{-q'}{\sqrt{p'^2 + q'^2 + r'^2}} ;$$

$$\sin. \theta. \cos. \varphi = \frac{-r'}{\sqrt{p'^2 + q'^2 + r'^2}} .$$

By means of these equations, we can determine the values of  $\theta$  and  $\varphi$  in functions of the time with respect to the fixed plane which we have considered. We have only now to determine the angle  $\psi$ , which the intersection of this plane, and that of the two first principal axes, constitutes with the axis of  $x'$ ; but this requires a new integration.

From the values of  $q$  and of  $r$  which have been given in No. 26 we derive

$$d\psi. \sin. \theta = q. dt. \sin. \theta. \sin. \varphi + r. dt. \sin. \theta. \cos. \varphi ;$$

from which we deduce

$$\sin. \theta. \cos. \varphi = \frac{-r'}{\sqrt{p'^2 + q'^2 + r'^2}} ,$$

and if we multiply the first of the preceding equations by  $\cos. \theta. \cos. \varphi$ , and the second by  $\sin. \varphi$ , and then subtract the first from the second we shall obtain

$$q'. \cos. \theta + p'. \sin. \theta. \sin. \varphi = 0$$

$\therefore$  substituting for  $\cos. \theta$  its value, we obtain

$$\sin. \theta. \sin. \varphi = \frac{-q'}{\sqrt{p'^2 + q'^2 + r'^2}} .$$

$$d\psi = \frac{-k.dt.(Bq'^2 + Ar'^2)}{AB.(q'^2 + r'^2)} *$$

but from what precedes, we have

$$q'^2 + r'^2 = k^2 - p'^2; \quad Bq'^2 + Ar'^2 = \frac{H^2 - AB \cdot p'^2}{C};$$

therefore we shall have

$$d\psi = \frac{-k.dt(H^2 - AB \cdot p'^2)}{ABC(k^2 - p'^2)}$$

By substituting in place of  $dt$ , its value which has been given above; we shall have the value of  $\psi$  in a function of  $p'$ ; thus the three angles  $\theta$ ,  $\phi$ , and  $\psi$  will be determined in functions of the variables  $p'$ ,  $q'$ ,  $r'$ , which will be themselves determined in functions of the time  $t$ .†

Consequently we can have at any instant the values of these angles with respect to the plane of  $x'$ , and  $y'$ , which we have considered, and it will be easy by means of the formulæ of spherical trigonometry, to

\* If we multiply the values of  $qdt$ ,  $rdt$ , given in page 166, respectively by  $\sin. \theta \cdot \sin. \phi$ ,  $\sin. \theta \cdot \cos. \phi$ , and then add them together we shall have

$$d\psi \cdot \sin. 2\theta = q.dt \cdot \sin. \theta \cdot \sin. \phi + rdt \cdot \sin. \theta \cdot \cos. \phi =$$

$$\left( -\frac{q'^2 dt}{A.k} - \frac{r'^2 dt}{B.k} \right) = -\frac{(Bq'^2 + Ar'^2)}{AB.k} dt \text{ and as } \sin. 2\theta = \frac{q'^2 + r'^2}{k^2},$$

the value of  $d\psi$  will be

$$-\frac{k.dt.(Bq'^2 + Ar'^2)}{A.B(q'^2 + r'^2)}.$$

†  $\cos. \theta - \sin. \theta \cdot \sin. \phi, -\sin. \theta \cdot \cos. \phi$ , are the cosines of the angles which a perpendicular to the fixed plane or the axis of  $z'$  makes the principal axes, see page 180, and

$$\frac{p'}{\sqrt{p'^2 + q'^2 + r'^2}} \quad \frac{q'}{\sqrt{p'^2 + q'^2 + r'^2}} \quad \frac{r'}{\sqrt{p'^2 + q'^2 + r'^2}}$$

are the cosines of the angles which the principal axes,  $z''$ ,  $y''$ ,  $x''$ , make with the axis of the plane, on which the projection of the area is a maximum, consequently the cosine of the angle which the axis of the plane on which the projection of the area is a maximum makes

determine the values of the same angles with respect to any other plane ; \* this will introduce two new arbitrary quantities, which combined with the four preceding quantities will constitute the six arbitrary quantities, which ought to give the complete solution of the problem which we have discussed. But it is evident that the consideration of the above mentioned plane simplifies considerably this problem.

The position of the three principal axes on the surface, being supposed to be known ; if at any instant, the position of the real axis of rotation on this surface, is given and also the angular velocity of rotation, we

with the axis of the fixed plane, (see note to page 7).

$$= \frac{p' \cdot \cos. \theta - q' \cdot \sin. \theta \cdot \sin. \phi - r' \cdot \sin. \theta \cdot \cos. \phi}{\sqrt{p'^2 + q'^2 + r'^2}} = \frac{N}{k}$$

we might by a similar process shew that the cosine of the angle which the axis of the plane of greatest projection makes with  $y'$ , and  $x'$ , are respectively proportional to  $N'$  and  $N''$ , consequently the position of this plane with respect to the fixed axes of  $x' y'$ , and  $z'$  is given, therefore this plane remains fixed during the motion, and the values of  $N$ ,  $N'$ ,  $N''$ , are the three quantities which determine the position of the fixed axes, with respect to the plane of greatest projection.

\* The determination of  $p', q', r'$ , which give the position of the instantaneous axis of rotation requires three arbitrary quantities and the determination of  $\ell, \psi, \phi$ , which give the position of the principal axes with respect to the fixed axes requires three more arbitrary quantities, these are,  $H$ ,  $k$ , and the constant quantities which are introduced by the integration of  $dt$  and  $d\psi$ , the two remaining quantities are determined by the values of  $\cos. \ell, \sin. \ell, \sin. \phi, \sin. \ell. \cos. \phi$ , for any other fixed plane beside the invariable plane, ∵ by making the plane of greatest projection, to coincide with the fixed plane ; these new arbitrary quantities vanish, and the number of constant arbitrary quantities will be reduced to four.

The values of  $\ell, \phi, \psi$ , with respect to the plane on which the projection of the area is a maximum being given, and also the value of the angle which this plane makes with any other plane, it will be easy to deduce the cosine of the angle which each of the principal axes makes with the assumed plane, in fact by means of the values of  $N N' N''$  we can determine the angles  $\ell, \psi, \phi$ , where we have the values of the same angles for the plane on which the projection of the area is a maximum, i.e, where we have  $p', q', r'$ , and substituting  $p, q, r$ , in place of  $p', q', r'$ , in these expressions we obtain the cosine of the angle which the axis of instantaneous rotation makes with the axis of the fixed plane, the three quantities  $N, N', N''$ , are not undetermined, for if  $N'$  and  $N''$  have definite values the value of  $N$  is determined by means of the equation  $N^2 + N'^2 + N''^2 = k^2$ .

shall have the values of  $p$ ,  $q$ ,  $r$ , at this instant because these values divided by the angular velocity of rotation express the cosines of the angles, which the real axis of rotation constitutes with the three principal axes ; ∴ we shall have the values of  $p'$ ,  $q'$ ,  $r'$ , but these last values are proportional to the sines of the angles which the three principal axes constitute with the plane  $x'$  and  $y'$ , relatively to which the sum of the areas of the projections of the molecules of the body, multiplied respectively by these molecules, is a *maximum* ; therefore we can determine at all instants, the intersection of the surface of the body with the invariable plane ; and consequently find the position of this plane, by the actual conditions of the motion of the body.

Let us suppose, that the motion of rotation of the body arises from a primitive impulse, of which the direction does not pass through its centre of gravity. It follows from what has been demonstrated in Nos. 20 and 22, that the centre of gravity will acquire the same motion, as if this impulse was immediately applied to it, and that the body will move round this centre with the same rotatory motion as if this centre quiesced. The sum of the areas described about this point, by the radius vector of each molecule projected on a fixed plane, and multiplied respectively by these molecules will be proportional to the moment of the principal force projected on the same plane ; but this moment is evidently the greatest possible for the plane which passes through its direction and through the centre of gravity ; consequently this plane is the invariable plane. If the distance of the primitive impulse from the centre of gravity be  $f$  and if  $v$  be the velocity which is impressed on this point,  $m$  representing the mass of the body,  $mfv$  \* will be the moment of this im-

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\*  $v$  being the velocity of the centre of gravity, and  $m$  being the mass of the body, the measure of the force will be equal to  $mv$ , and its moment with respect to the centre of gravity will be equal to  $mf.v$ , see No. 3, and the motion of all the molecules of the body arising solely from this impulse it is evident from the principle of D'Alembert, which has been established in No. 18, that the quantities of motion which these molecules have at the commencement of the motion, estimated in a direction contrary to their true direction must

pulse and being multiplied by  $\frac{1}{2}t$ , the product will be equal to the sum of the areas described in the time  $t$ , but by what precedes this sum is equal to  $\frac{t}{2} \cdot \sqrt{p'^2 + q'^2 + r'^2}$ ; consequently we have .

$$\sqrt{p'^2 + q'^2 + r'^2} = m.fv.$$

If at the commencement of the motion we know the position of the principal axes with respect to the invariable plane,  $i, e$ , \* the angles  $\theta$  and  $\phi$ ; we shall have at this commencement the values of  $p' q'$  and  $r'$  and consequently those of  $p, q, r$ ; therefore at *any* instant we shall have the values of the same quantities. †

constitute an equilibrium with the force  $mv$  consequently the principal plane  $i, e$ , the plane with respect to which the moment is a maximum is the plane passing through the centre of gravity, and the direction of the primitive impulsion ∵ the sum of the areas described in the time  $t = \frac{1}{2}t.mfv$ .

\* The constant quantity  $k = m.fv$ ; in order to determine  $H$ , it may be remarked that the position of the principal axes at the commencement of the motion, with respect to the plane passing through the fixed point and the direction of the impulse being given, we have the the values of  $p' q', r'$ , being proportional to the cosines of the angles which the principal axes make with the axis to the invariable plane. Consequently we have the *constant* quantity

$$H = \sqrt{AB.p'^2 + BC.q'^2 + AC.r'^2};$$

the third constant quantity will be determined by integrating the value of  $dt$ , which will be equal to a function of  $p' +$  a constant arbitrary quantity;  $p'$  which is proportional to the cosine of the angle which the axis of  $z'$  makes with the axis to the plane of greatest moment has a determined value when  $t=0$  ∵ by means of this value we are enabled to find the value of the third constant quantity; with respect to the fourth constant quantity which arises from the integration of the value of  $d\psi$ , this gives  $\psi =$  to a function of  $p'$  plus a constant quantity,  $p'$  being proportional to  $\cos. \theta$ , we shall obtain the fourth constant quantity which is necessary to complete the solution of the problem, if we know what value of  $\psi$  corresponds to a given value of  $\theta$ .

† When a solid body is not solicited by any accelerating forces and can revolve freely about a point we shall have

$$dx = yd\pi - zd\phi, dy = zd\psi - xd\pi, dz = xd\phi - yd\psi, \&c.$$

By means of this theory, we are enabled to explain the double motion of rotation and of revolution, of the planets, by one initial impulse. In fact, let us suppose that a planet is an homogenous sphere whose radius is

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See page 89, if we multiply the equations (Z) of No. 21, by

$$\frac{d\varpi}{dt}, \frac{d\phi}{dt}, \frac{d\psi}{dt},$$

respectively we shall obtain

$$\begin{aligned} c \cdot \frac{d\varpi}{dt} + c' \cdot \frac{d\phi}{dt} + c'' \cdot \frac{d\psi}{dt} &= \Sigma m. \left\{ \frac{xd\varpi \cdot dy - y \cdot d\varpi \cdot dx}{dt^2} \right\} + \Sigma. \left\{ \frac{zd\phi \cdot dx - xd\phi \cdot dz}{dt^2} \right\} \\ + \Sigma m. \left\{ \frac{y \cdot d\psi \cdot dz - z \cdot d\psi \cdot dy}{dt^2} \right\} &= \Sigma m. \left\{ \frac{xd\varpi - z \cdot d\psi}{dt^2} \right\} \cdot dy + \Sigma m. \left\{ \frac{z \cdot d\phi - y \cdot d\varpi}{dt^2} \right\} \cdot dx \\ + \Sigma m. \frac{(y \cdot d\psi - x \cdot d\phi)}{dt^2} \cdot dz &= \Sigma m. \frac{(dx^2 + dy^2 + dz^2)}{dt^2} = \end{aligned}$$

const. (sec No. 19) now if we substitute for  $c, c', c''$ ,

$$\sqrt{c^2 + c'^2 + c''^2} \cos. \theta, \sqrt{c^2 + c'^2 + c''^2} \sin. \theta. \sin. \psi, -\sqrt{c^2 + c'^2 + c''^2} \sin. \theta. \cos. \psi,$$

to which they are respectively equal, and also for

$d\varpi, d\phi, d\psi, d\theta. \cos. l, d\theta. \cos. n, d\theta. \cos. m$ , see page 90, we shall obtain

$$\Sigma m. \left\{ \frac{dx^2 + dy^2 + dz^2}{dt^2} \right\} = \sqrt{c^2 + c'^2 + c''^2} \cdot \frac{d\theta}{dt}.$$

$$(\cos. \theta. \cos. l + \sin. \theta. \sin. \psi. \cos. n - \sin. \theta. \cos. \psi. \cos. m)$$

$$= \text{const. as } \cos. \theta. \sin. \theta. \sin. \psi. \sin. \theta. \cos. \psi,$$

are equal to the cosines of the angles which the axis of the plane of greatest projection, makes with three fixed axes, and as  $\cos. l, \cos. n, \cos. m$ , are the cosines of the angles which the axis of instantaneous rotation makes with the same axes, the last factor of the second member of the equation is equal to the cosine of the angle, which the axis of rotation makes with the axis of the plane on which projection of the areas is the greatest possible, . . . as  $\frac{d\theta}{dt}$  is the exponent of the velocity of rotation for any instant, this expression multiplied

equal to  $R$ , and that it revolves about the sun with an angular velocity equal to  $U$ ;  $r$  being supposed to express its distance from the sun, we shall have  $v=r U$ ; moreover if we conceive that the planet is put in motion by a primitive impulse, of which the direction is distant from its centre by a quantity equal to  $f$ , it is evident that it will revolve about an axis perpendicular to the invariable plane; therefore if we suppose that this axis coincides with the third principal axis \* we shall have  $\theta=0$ , and consequently  $q'=0$ ,  $r'=0$ ; therefore  $p'=mfv$  i, e,  $Cp=mfrU$ .

But in the sphere, we have  $C = \frac{2}{5} mR^2$ ; consequently,

$$f = \frac{2}{5} \cdot \frac{R^2}{r} \cdot \frac{p}{U};$$

which gives the distance of the direction of the primitive impulsion from the centre of the planet, and satisfies the ratio which is observed to obtain between  $p$  the angular velocity of rotation, and  $U$  the angular velocity of the revolution of the planet round the sun. With respect to the earth, we have  $\frac{p}{U} = 366,25638$ ; the parallax of the sun gives  $\frac{R}{r} = 0.000042665$ , and consequently  $f = \frac{1}{160} R$  very nearly.

into the cosine of the angle, which the axis of instantaneous rotation makes with the axis of the plane, on which the projection is a maximum, is a constant quantity. When the plane of  $xy$  coincides with the plane passing through the direction of the impulse, and the point about which the rotation is performed  $\cos. \theta=1$  and  $\sin. \theta=0$  ∴ we shall have  $c' \frac{d\phi}{dt}, c'' \frac{d\psi}{dt} = 0$ ; constant quantity= $c \cdot \frac{d\theta}{dt}$ .  $\cos. l$  consequently the velocity of rotation, i. e, parallel to the axis of  $z$ ,  $= \frac{d\theta}{dt} \cdot \cos. l$  is constant.

\* All the diameters of a sphere being principal axes, if we suppose that the axis of revolution which is evidently the axis of the invariable plane coincides with the axis of  $z'$ ,  $\theta=0$  ∴  $\cos. \theta=1$  ∴  $q'$  and  $r'$  respectively to  $\sin. \theta. \sin. \varphi$ ,  $\sin. \theta. \cos. \varphi$  vanish and this considerably simplifies the calculus.

The planets are not homogenous; but we may suppose them to be composed of concentrical spherical strata of unequal density. Let  $\rho$  denote the density of one of those stratas of which the radius is equal to  $R$ , we shall have

$$C = \frac{2m}{3} \cdot \frac{\int_{\rho} R^4 dR}{\int_{\rho} R^2 dR} . *$$

\* The moment of inertia for a sphere is calculated in Book V. No. II. in a general manner, but as it involves some steps which are demonstrated in the second and third books, it will be necessary to give here a special demonstration, let there be two concentrical circles, whose radii are  $q$ ,  $q+dq$ , the circumference of the interior is equal to  $2\pi q$ , and the area of the annulus contained between the peripheries of those circles is equal  $2\pi q dq$  ::  $2\pi q^3 dq$  is equal to the moment of inertia of this annulus and  $\frac{1}{2}\pi q^4$  is the moment of inertia of a concentrical annulus of a finite breadth, ∵ when the preceding integral is taken between the limits  $q=0$ ,  $q=R$  the expression becomes  $\frac{1}{2}\pi R^4$ , which is the moment of inertia for the entire circle, now in order to obtain the moment of inertia for the entire sphere, let us conceive a plane parallel to the axis of rotation cutting the sphere at a distance from the axis equal to  $x$ , its intersection with the surface of the sphere will a lesser circle of the sphere, let  $y$ =the radius of this circle, the moment of inertia of this circle with respect to its centre is equal by what precedes to  $\frac{1}{2}\pi y^4$  ∵ the moment of inertia of an indefinitely small slice is equal to  $\frac{1}{2}\pi y^4 dx = \frac{1}{2}\pi (2Rx-x^2)^2 dx$ , for  $y^2=2Rx-x^2$   $R$  being the radius of the sphere, ∴ integrating we have

$$\pi x^3 \left\{ \frac{2}{3}R^2 - \frac{1}{2}Rx + \frac{x^2}{10} \right\}$$

= the moment of inertia of a spherical segment and this integral being taken between the limits  $x=0$ , and  $x=R$  gives

$$\pi R_3 \cdot \left\{ \frac{2}{3}R^2 - \frac{1}{2}R^2 + \frac{R^2}{10} \right\} = \frac{8}{15} \cdot \pi R^6$$

= the moment of inertia of the entire sphere with respect to a diameter, and it is very easy by means of the expression which has been given in page 180, to obtain the moment of inertia for any axis parallel to the diameter, if  $R$  is supposed to be variable in the last expression, and if  $\rho$  the density varies from the centre to the circumference, the moment of inertia of any spherical stratum whose radius =  $R$  is

$$\frac{8.5}{15} \cdot \pi \rho R^4 dR$$

( $\rho$  being a function of  $R$ ).

If, as is very probable, the denser strata are nearer to the centre ; the function  $\frac{\int \rho \cdot R^4 dR}{\int \rho \cdot R^2 dR}$  will be less than  $\frac{3R^2}{5}$ , consequently the value of  $f$  will be less than in the case of homogeneity.

30. Let us now determine the oscillations of a body when it turns very nearly about the third principal axis. We might deduce them from the integrals which we obtained in the preceding number ; but it is

$\therefore$  the moment of inertia of a sphere composed of concentrical strata is equal to

$$\frac{8}{3} \pi f \rho \cdot R^4 dR, \text{ in like manner } m = \text{the mass of the sphere} = 4\pi \int \rho R^2 dR$$

$$\therefore \pi = \frac{m}{4 \int \rho R^2 dR} \text{ and } f = \frac{C_p}{m \cdot r U} = \frac{8m \int \rho \cdot f_\rho R^4 dR}{3.4m \cdot r \int \rho \cdot R^2 dR} = \frac{2p \cdot f_\rho R^4 dR}{3r \int \rho \cdot R^2 dR}$$

we obtain the ratio of  $U$  to  $p$  from knowing the period of the earth and the time of its rotation, for the angular velocities are inversely as the angles described in the same time,  $\rho$  being by hypothesis a function of  $R$  where the density increases towards the centre

$\epsilon - \frac{1}{\phi(R)}$   $\therefore$  the fraction in the text becomes

$$\begin{aligned} f &= \frac{R^4 \cdot dR}{\phi(R)} \\ &= \frac{R^2 dR}{\phi(R)} \end{aligned}$$

by partial integration

$$\frac{R^5}{5 \cdot \phi R} - \int \frac{R^5 \cdot d\phi(R)}{5(\phi(R))^2} + \frac{R^3}{3\phi(R)} - \int \frac{R^3 \cdot d\phi(R)}{3 \cdot (\phi R)^2}$$

and as the numerator is more diminished than the denominator the value of the fraction which in the case of homogeneity was  $\frac{3R^2}{5}$  will be diminished when the density increases towards the centre.

simpler to deduce them directly from the differential equations (*D*) of No. 26. The body not being actuated by any forces; these equations will become by substituting  $Cp$ ,  $Aq$ , and  $Br$ , in place of their respective values  $p'$ ,  $q'$ ,  $r'$ ,

$$dp + \frac{(B-A)}{C} \cdot qr \cdot dt = 0;$$

$$dq + \frac{(C-B)}{A} \cdot rp \cdot dt = 0;$$

$$dr + \frac{(A-C)}{B} \cdot pq \cdot dt = 0.$$

The solid being supposed to revolve very nearly about the third principal axis,  $q$  and  $r$ \* are very small quantities, therefore we may reject their squares and products; consequently we shall have  $dp=0$  and  $p$  will be constant. If in the other two equations we suppose

$$q = M \cdot \sin.(nt+\gamma); \quad r = M' \cdot \cos.(nt+\gamma);$$

we shall have

\* The solid being supposed to revolve very nearly about the principal axis,

$$\frac{p}{\sqrt{p^2+q^2+r^2}} \approx$$

the cosine of the angle which the instantaneous axis of rotation, make with the principal will be  $q.p$ . equal to unity consequently,  $q$  and  $r$  will be very small because the sine of the above mentioned angle which is equal to

$$\frac{\sqrt{q^2+r^2}}{\sqrt{p^2+q^2+r^2}}$$

very nearly vanishes.

$$n=p. \sqrt{\frac{(C-A).(C-B)}{AB}}; M=-M. \sqrt{\frac{A.(C-A)}{B.(C-B)}}, M \text{ and } \gamma^*$$

being two constant quantities, the velocity of rotation will be  $\sqrt{p^2+q^2+r^2}$  or simply  $p$ , the squares of  $q$  and  $r$  being neglected; therefore this velocity will be very nearly constant, finally the sine of the angle formed by the real axis of rotation, and the third principal axis will be  $\frac{\sqrt{q^2+r^2}}{p}$ .

If at the commencement of the motion we have  $q=0$  and  $r=0$ ,  $i, e$ , if at this instant the real axis of rotation coincides with the third principal axis; we shall have  $M=0$   $M'=0$ ; consequently  $q$  and  $r$  will be always equal to nothing, and the axis of rotation will always coincide with the third principal axis; from which it follows that if the body commences to revolve round one of its principal axes, it will continue to revolve uniformly about the same axis. It is from this remarkable pro-

\*  $q=M. \sin. (nt+\gamma)$   $r=M'. \cos. (nt+\gamma)$  satisfy the preceding differential equations, for by substituting these values we obtain

$$\begin{aligned} M.n.dt. (\cos. (nt+\gamma) + \frac{(C-B)}{A}.p M'.dt. \cos. (nt+\gamma)) &= 0 \\ -M'.n.dt. \sin. (nt+\gamma) + \frac{A-C}{B}.p M.dt. \sin. (nt+\gamma) &= 0 \\ \therefore M.n + \frac{(C-B)}{A}.p M' &= 0 \quad M'.n + \frac{A-C}{B}.p M = 0 \quad \therefore M' = \\ -\frac{M.n.A}{p.(C-B)} &= \frac{(A-C).p.M}{B.n} \quad \therefore n = \\ p^2 \cdot \frac{(C-A)(C-B)}{AB} \quad \therefore M' &= -M. \sqrt{\frac{A.(C-A)}{B.(C-B)}} \end{aligned}$$

the quantities  $M$  and  $\gamma$  are arbitrary consequently these values are perfect integrals of the two preceding differential equations which they satisfy. (See Lacroix traité élémentaire de Calcul intégral, No. 297).

perty, that these axes have been termed principal axes of rotation, it appertains to them exclusively; for if the real axis of rotation is invariable on the surface of the body, we have  $dp=0$ ,  $dq=0$ ,  $dr=0$ , therefore from the preceding values of those quantities we obtain

$$\frac{(B-A)}{C} \cdot rq = 0; \quad \frac{(C-B)}{A} \cdot rp = 0; \quad \frac{(A-C)}{B} \cdot pq = 0.$$

In the general case where  $A$ ,  $B$ ,  $C$ , are unequal, two of the three quantities  $p$ ,  $q$ ,  $r$ , vanish in consequence of these equations, which implies that the real axis of rotation coincides with one of the principal axes.\*

If two of the three quantities  $A$ ,  $B$ ,  $C$ , are equal, for example, if we have  $A=B$ ; the three preceding equations will be reduced to the following,  $rp=0$ ,  $pq=0$ ; and they may be satisfied by supposing  $p=0$ . The axis of rotation in this case exists in a plane perpendicular to the third principal axis; but we have seen in No. 27, that all axes existing in this plane, are in this case principal axes.

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\* The value of the quantities  $M$ ,  $M'$ , may be determined by knowing the position of the instantaneous axis of rotation at the commencement of the motion, whatever be their values at that instant they remain unaltered during the motion of the body ∵ if at the commencement of the motion, the real axis of rotation coincided with the principal axis

$$\frac{\sqrt{r^2+q^2}}{\sqrt{p_2+r^2+q^2}} = 0$$

∴  $q$  and  $r$  are respectively equal to nothing, and therefore  $M$  and  $M'$  will vanish, consequently the values of  $q$  and  $r$  will always be equal to nothing, and as  $p$  is constant and equal to the angular velocity, the body will revolve uniformly about the principal axis. If the position of the real axis of rotation is invariable on the surface of the body,  $p$ ,  $q$ ,  $r$ , must be constant, see No. 29, page 201, ∴  $dp$ ,  $dq$ ,  $dr$ , are respectively equal to nothing ∴ their values

$$\frac{B-A}{C} \cdot rq \frac{C-B}{A} \cdot rp \frac{A-C}{A} \cdot pq$$

respectively vanish, ∴ in order to satisfy these equations two of the three variable quantities  $p$ ,  $q$ ,  $r$ , must vanish.

Finally, if the three quantities  $A, B, C$ , are equal, the preceding equations will be satisfied, whatever may be the values of  $p, q, r$ ; but in this case, all the axes of the body are principal axes.\*

It follows from what precedes, that to the principal axes only belongs the property of being permanent axes of rotation; † but they do not

\* When  $A=B$ , the first of these three equations vanishes of itself, whatever may be the values of  $r$  and  $q$ , and we shall satisfy the two last equations by supposing  $p=0$ , ∵ the real axis of rotation is perpendicular to the third principal axis, see No. 29, notes, but as in this case all lines drawn in a plane perpendicular to the third principal axis, are principal axes, it follows that the axis of rotation is in this case a principal axis; if  $A=B=C$  the three preceding equations will be identical, and the values of  $p, q$ , and  $r$ , may be assumed at pleasure, but in this case all axes are principal axes, ∵ it follows universally, that if the axis of rotation remain permanently the same, it must be a principal axis.

In the general case when  $A, B$  and  $C$ , are unequal, we shall be always certain that  $p, q$ , and  $r$ , and  $M, M'$ , vanish at the commencement of the motion, when the impulse is made in a plane which coincides with the plane of two of the principal axes, for in this case the invariable plane to which we adverted in Note to page 184, coincides with the plane passing through two of the principal axes, and the axis of rotation or of this invariable plane will necessarily coincide with the third principal axis. See Notes, to page 188.

† It might be proved directly from the property of principal axes scilicet  $S.xz.dm = 0$ ,  $Syz.dm \neq 0$ , that the pressure on the axis of rotation which is produced by the centrifugal force must vanish, when this axis is a principal axis, and that consequently, when there is a fixed point given in a body, there exists always three axes passing through this point, about which the body may revolve uniformly without a displacement of the axis, and as if these lines were entirely free; for if the body is acted upon by an initial impulse,  $\omega$  denoting the angular velocity and  $r$  the distance of a molecule  $dm$  from the axis of rotation which we suppose to coincide with the axis  $z$ ,  $x$  and  $y$  being the coordinates with respect to the axes of  $x$  and  $y$ , we have the centrifugal force  $= \omega^2 r dm$ , this force resolved parallel to  $x$  and  $y = \frac{\omega^2 r x}{r} dm, \frac{\omega^2 r y}{r} dm$ , because  $\frac{x}{r}, \frac{y}{r}$  are equal to the cosines of the angles which the axes of  $x$  and  $y$  make with  $r$ , ∵ the sum of the forces for all the molecules of the body  $= \omega^2 Sx dm, \omega^2 Sy dm$ , and the respective sums of their moments for the axes of  $y$  and of  $x$  are  $\omega^2 Sx.z dm, \omega^2 Sy.z dm$ , and  $m$  being the mass of the body and  $x, y$ , being the coordinates of the centre of gravity, we have  $\omega^2.mx = \omega^2 Sx dm, \omega^2.my = \omega^2 Sy dm$ , and if  $z, z_{\perp}$ , represent the distances of the resultants  $\omega^2.mx, \omega^2.my$ , from the plane of the axis of  $x y$  we have by note to No. 3,  $\omega^2.mx.z = \omega^2 Sxy dm, \omega^2.my.z = \omega^2.Syz dm$ , when  $z, z_{\perp}$ , are equal, the resultants

possess this property in the same manner. The motion of rotation about the axis, of which the moment of inertia is intermediate between the moments of inertia of the two other axes, may be disturbed in a sensible degree by the slightest cause; so that in this motion, there is no stability.

The state of a system of bodies is termed *stable*, when the system being very slightly deranged, it deviates from the state by an indefinitely small degree, by making continual oscillations about this state. This being understood, let us suppose that the real axis of rotation deviates from the third principal axis by an indefinitely small quantity; in this case, the quantities  $M$  and  $M'$  are indefinitely small; and if  $n$  is a real quantity, the values of  $q$  and  $r$  will always remain indefinitely small, and the real axis of rotation will only make excursions of the

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$\omega^2 \cdot mx$ ,  $\omega^2 \cdot my$ , are applied to the same point,  $\therefore$  these two forces will compose one sole force  $= \omega^2 \cdot m \sqrt{x^2 + y^2}$ , now if the fixed axes pass through the centre of gravity we have  $x=0, y=0 \therefore \omega^2 Sx dm, \omega^2 Sy dm$  respectively vanish, and if the axis of rotation is a principal axes we have  $\omega^2 Sxz dm = 0, \omega^2 Syz dm = 0$ , from the first equation it follows that the axes does not experience any tendency to a progressive motion, and the second equations indicate that the sum  $\omega^2 \cdot Sxz dm$  of the moments of the forces vanish, from these two conditions it follows that the forces constitute an equilibrium independently of the axis. If the fixed axis of rotation and origin of the coordinates was transferred to a different point of the body, being still a principal axis, we should have as before  $Sxz dm = 0, Syz dm = 0 \therefore$  the sum of the moments of the forces with respect to the axes of  $y$  and of  $x$  vanish as before  $\therefore$  as  $x$ , and  $y$ , have in this case a finite value  $z$ , and  $z_{\perp}$ , must vanish, for  $\omega^2 \cdot mx, z, \omega^2 \cdot my, z_{\perp}$  vanish being equal to  $\omega^2 \cdot Sxz dm, \omega^2 \cdot Syz dm$ ,  $\therefore$  the pressure  $= \omega^2 \cdot m \sqrt{x'^2 + y'^2}$ , which as  $z, z_{\perp}$  vanish, must exist in the plane of  $x, y$ , and must pass through the origin of the coordinates,  $\therefore$  if this point is fixed the pressure will be destroyed, and the motion will be performed about the axis as if it was fixed, for the only pressure which could displace it is destroyed, by the resistance of the fixed point.

From what precedes it appears, that when the principal axis passes through the centre of gravity, it is not necessary that any point should be fixed, in order that the motion may be perpetuated uniformly about the fixed axes, in any other case it is necessary that the origin of the coordinates be fixed.

same order \* about the third principal axis. But if  $n$  was imaginary,  $\sin.(nt+\gamma)$ ,  $\cos.(nt+\gamma)$  will become exponential, and the expressions for  $q$  and  $r$  might then increase indefinitely, and at length cease to be very small; consequently there would be no stability in the motion of rotation of the body about the third principal axis. The value of  $n$  is real, if  $C$  is the greatest or the least of the three quantities  $A$ ,  $B$ ,  $C$ ; for the product  $(C-A)$ .  $(C-B)$  is positive; but this product is negative if  $C$  is intermediate between  $A$  and  $B$ , and in this case  $n$  is imaginary; thus, the motion of rotation is stable about the two principal

\* When  $n$  is a real quantity,  $p$  and  $q$  can be expressed by sines and cosines of  $nt$ , but these values are not susceptible of indefinite increase with the time, for they are periodic functions of  $t$ , and the limit of the values of  $\sin.(nt+\gamma)$ ,  $\cos.(nt+\gamma)$  is unity, if they are very small at the commencement of the motion,  $M$  and  $M'$  must be very small, and as these quantities are invariable, the expressions for  $q$  and  $r$  will always remain indefinitely small. If  $n$  is imaginary,  $\sin.(nt+\gamma)$ ,  $\cos.(nt+\gamma)$  are imaginary, and as

$$\cos.(nt+\gamma) + \sqrt{-1} \cdot \sin.(nt+\gamma) = c^{+(nt+\gamma)\sqrt{-1}}$$

and

$$\cos.(nt+\gamma) - \sin.(nt+\gamma) = c^{-(nt+\gamma)\sqrt{-1}}$$

we obtain by adding and subtracting

$$\cos.(nt+\gamma) = \frac{c^{(nt+\gamma)\sqrt{-1}} + c^{-(nt+\gamma)\sqrt{-1}}}{2}$$

$$\sin.(nt+\gamma) = \frac{c^{(nt+\gamma)\sqrt{-1}} - c^{-(nt+\gamma)\sqrt{-1}}}{2\sqrt{-1}}$$

if  $n$  is imaginary the preceding exponential expressions will become

$$\frac{c^{-nt+\gamma\sqrt{-1}} + c^{nt-\gamma\sqrt{-1}}}{2}, \frac{c^{-nt+\gamma\sqrt{-1}} - c^{nt-\gamma\sqrt{-1}}}{2\sqrt{-1}}$$

in these exponential expressions, the part which is not affected with the radical sign, is

axes of which the moments of inertia are the greatest, and the least; but not so about the other principal axis.\*

proportional to the time, and therefore the values of  $q$  and  $r$ , will increase indefinitely with the time,  $\therefore$  though they may have been indefinitely small at the commencement of the motion, still as there is no limit to the increase of the exponential expressions, they will at length exceed any assigned magnitude.

\* It might be shewn directly by means of the equations  $C^2p^2 + A^2q^2 + B^2r^2 = k^2$ ;  $ABC^2.p^2 + A^2BCq^2 + AB^2Cr^2 = H^2$ , that there is a limit to the increase of  $q$  and  $r$  when  $C$  is the greatest or least of the three quantities  $A, B, C$ , for if we multiply the first equation by  $AB$ , and then subduct it from the second we obtain  $A^2.B(C-A)q^2 + AB^2.(C-B).r^2 = H^2 - AB.k^2$ , if at any instant the quantities  $q, r$ , are very small  $H^2 - Ak^2$  which is constant will be very small, consequently in all the changes which  $r$  and  $q$  undergo they are subjected to the same condition, and this condition requires that  $r$  and  $q$  should be always very small when  $C-A$  and  $C-B$  are of the same sign, because then both the terms of the first member of the preceding differential equation will be either positive or negative, and the expressions

$$\frac{H^2 - AB.k^2}{A^2.B.(C-A)}, \quad \frac{H^2 - AB.k^2}{AB^2(C-B)},$$

are the limits to which the respective values of  $q$  and  $r$  can never attain. If  $C-B$  and  $C-A$  are of different signs, then the terms of the first member of the equation will be of different signs, and it is only the difference of the quantities  $A^2B(C-A).q^2 + AB^2.(C-B)r^2$ , that is indefinitely small  $\therefore$  since this difference depends on the relative values of these quantities,  $q$  and  $r$  may be very great, though the preceding residual is a quantity indefinitely small.

Philosophers have distinguished the equilibrium of stability into two species absolute and relative, in the first case the stability obtains whatever may be the oscillations of the system, in the second case it is necessary that the oscillations should be of a certain description, in order to insure the stability of the equilibrium. If a body revolving about a fixed axis passes through several positions of equilibrium, these will be alternately stable and unstable. For if a system deviates from a position of stable equilibrium, from the nature of this equilibrium it tends to revert, but according as the system deviates more and more from its first position, this tendency will diminish, and at length it will tend to deviate from the original position, but previous to this change of tendency there must have been a position in which the system neither tended to revert, or to deviate from its original position, consequently this is a position of equilibrium, but this equilibrium is evidently one of instability, for previous to the arrival of the system at this position it tended to revert to its primary position, and when it passed this position, it tends to deviate from the primary and consequently from this second position of

Now, in order to determine the position of the principal axes in space, we shall suppose the third principal axis to coincide very nearly with the plane of  $x'$  and of  $y'$ , so that  $\theta$  will be a very small quantity of which we may neglect the square.

By No. 26, we shall have

$$d\phi - d\psi = pdt *$$

and by integrating we obtain

$$\psi = \phi - pt - \varepsilon$$

$\varepsilon$  being a constant arbitrary quantity. If we afterwards make  
 $\sin. \theta. \sin. \varphi = s$ ;  $\sin. \theta. \cos. \varphi = u$ ;

from the values of  $q$  and of  $r$  which have been given in No. 26, we shall obtain, by the elimination of  $d\psi$

$$\frac{ds}{dt} - pu = r; \quad \frac{du}{dt} + ps = -q; \quad +$$

equilibrium, this tendency of the body to deviate from the second position of equilibrium gradually diminishes, and at length vanishes, afterward the system tends to revert to the second position of equilibrium, and where the tendency to deviate from the second position of equilibrium vanishes, is also a position of equilibrium, which is evidently an equilibrium of stability, for previous to the arrival of the system at this position it tends towards it, inasmuch as it tends to deviate from the second position, and after passing this third position of equilibrium it tends to revert to the second, and consequently to the third position of equilibrium, thus it appears that when a system has returned to its primary position, it has passed through an even number of positions of equilibrium, alternately stable and unstable.

\*  $d\phi - d\psi. \cos. \theta = p.dt$ , but  $\cos. \theta = 1 - \frac{\theta^2}{2} + \frac{\theta^4}{1.2.3.4} - \&c$  when  $\theta$  is very small,  
 $\therefore d\phi - d\psi = pdt$ .

+  $d\psi. \sin. \theta. \sin. \varphi - d\theta. \cos. \varphi = q.dt$ ;  $d\psi. \sin. \theta. \cos. \varphi + d\theta. \sin. \varphi = r.dt$ ,  
 substituting in place of  $d\psi$  its value  $d\phi - pdt$ , we shall have

and by integrating

$$s = \epsilon \cdot \sin(pt + \lambda) - \frac{A \cdot M}{C \cdot p} \cdot (\sin nt + \gamma);$$

$$u = \epsilon \cdot \cos(pt + \lambda) - \frac{B \cdot M'}{C \cdot p} \cdot \cos(nt + \gamma);^*$$

$$\begin{aligned} d\varphi \cdot \sin \theta \sin \varphi - p \cdot \sin \theta \cdot \sin \varphi \cdot dt - d\theta \cdot \cos \varphi = q \cdot dt; \\ d\varphi \cdot \sin \theta \cdot \cos \varphi - p \cdot \sin \theta \cos \varphi \cdot dt \\ + d\theta \cdot \sin \varphi = r \cdot dt; \end{aligned}$$

substituting  $\theta$  in place of  $\sin \theta$ , to which it is very nearly equal since the higher powers of  $\theta$  may be neglected, we obtain

$$\begin{aligned} -d\varphi \cdot \theta \cdot \sin \varphi + d\theta \cdot \cos \varphi + p \cdot \sin \theta \cdot \sin \varphi \cdot dt = -q \cdot dt, i.e., \\ d(\cos \varphi \cdot \sin \theta) + p \cdot \sin \theta \cdot \sin \varphi \cdot dt = -q \cdot dt, \end{aligned}$$

and by substituting for  $\sin \theta \cdot \sin \varphi \cdot \sin \theta \cdot \cos \varphi$ , their values which have been given in the text, we obtain

$$\frac{du}{dt} + ps = -q;$$

in like manner the second differential equation becomes,

$$\begin{aligned} d\varphi \cdot \theta \cdot \cos \varphi + d\theta \cdot \sin \varphi - p \cdot dt \cdot \sin \theta \cdot \cos \varphi = r \cdot dt, i.e., d(\sin \theta \cdot \sin \varphi) \\ - p \cdot dt \cdot \sin \theta \cdot \cos \varphi = r \cdot dt, \end{aligned}$$

and by substitution,  $\frac{ds}{dt} - pu = r$ .

\* The integrals assigned in the text are the complete values of  $s$  and  $u$  for

$$\frac{ds}{dt} = \epsilon \cdot p \cdot \cos(pt + \lambda) - \frac{M}{C} \cdot \sqrt{\frac{A}{B} \cdot (C-A) \cdot (C-B)} \cdot \cos(nt + \gamma),$$

this expression is equal to  $pu + r$ , for substituting in place of  $u$  and  $r$ , we shall have

$$\epsilon \cdot p \cdot \cos(pt + \lambda) - \frac{B \cdot M' p}{C p} \cos(nt + \gamma) + M' \cdot \cos(nt + \gamma) =$$

(by substituting for  $M'$  its value,)  $\epsilon \cdot p \cdot \cos(pt + \lambda)$

$$- \frac{M}{C} \cdot \sqrt{\frac{A}{B} \cdot (C-A) \cdot (C-B)} \cdot \cos(nt + \gamma),$$

$\epsilon$  and  $\lambda$  being two new arbitrary quantities : therefore the problem is completely resolved, since the values of  $s$  and of  $u$  give the angles  $\theta$  and  $\phi$  in functions of the time, and  $\psi$  is determined in a function of  $\phi$  and  $t$ . If  $\epsilon$  vanishes, the plane of  $x'$  and of  $y'$  becomes the invariable plane, to which we have referred in the preceding number, the angles  $\theta$ ,  $\phi$  and  $\psi$ . \*

$\therefore$  since the integrals given in the text satisfy the differential equations

$$\frac{ds}{dt} - pu = r, \quad \frac{du}{dt} + ps = -q;$$

and since there are two constant quantities introduced, these values of  $u$  and  $s$  are their complete integrals.

\* When  $\epsilon$  vanishes  $s = \sin. \theta. \sin. \phi = -\frac{A.q}{C.p}$ ;  $u = \sin. \theta. \cos. \phi = -\frac{B.r}{C.p}$ ,  $i, e,$

$$\sin. \theta. \sin. \phi = -\frac{q'}{p'}, \quad \sin. \theta. \cos. \phi = -\frac{r'}{p'},$$

and those are values of the cosines of the angles which the principal axes of  $x''$  and  $y''$  make with the axis of the invariable plane, see notes to page 198. *In this case*

$$\frac{s}{u} = \tan. \phi = \frac{AM}{BM} \cdot \tan. (nt + \gamma) \therefore \phi = \frac{AM}{BM} \cdot (nt + \gamma),$$

as  $\phi$  is equal to the angle formed by the intersection of the invariable plane, and of the plane of  $x'', y''$ , with the axis of  $x''$ , if we know this angle at the commencement of the motion, or at any given epoch, we shall have the value of  $\gamma$ ; we might in like manner find  $M$ , for

$$u^2 + s^2 = \sin. {}^2 \theta. (\sin. {}^2 \phi + \cos. {}^2 \phi) = \frac{A^2 M^2}{C^2 p^2} + \frac{B^2 M^2}{C^2 p^2} \left( \frac{A(C-A)}{B(C-B)} \right) = \sin. {}^2 \theta,$$

by substituting for  $(\sin. {}^2 \phi + \cos. {}^2 \phi)$  unity, and for  $M^2$  its value.

$$\psi = \frac{AM}{BM} \cdot (nt + \gamma) - pt - \epsilon$$

$\therefore$  as we have already determined the values of  $M$ ,  $M'$ , and  $\gamma$ , we can determine the value of  $\epsilon$ , when the value of  $\psi$  is given at the commencement of the motion; from the preceding value of  $\psi$  it appears that this angle increases proportionably to the time,  $\therefore$  the intersection of the invariable plane and the plane of  $x'' y''$  revolves about the axis of the invariable plane with an uniform angular velocity.

31. If the solid is free ; the analysis of the preceding numbers will determine its motion about its centre of gravity : if the solid is constrained to move about a fixed point, it will make known the motion of rotation about this point. It now remains for us to consider the motion of a solid constrained to revolve about a fixed axis.

Let us suppose this axis to be that of  $x'$ , which we will make horizontal : in this case, the last of the equations (*B*) of No. 25, will be sufficient to determine the motion. Moreover let us conceive that the axis of  $y'$  is horizontal, and thus that the axis of  $z'$  is vertical, and directed towards the centre of the earth, lastly let the plane which passes through the axis of  $y'$  and of  $z'$ ,\* pass through the centre of gravity of the body, and let us conceive an axis always passing through this centre and through the origin of the coordinates. If  $\theta$  represents the angle which this new axis constitutes with the axis of  $z$ ; and  $y''$ , and  $z''$ , the coordinates referred to this new axis, we shall have

$$y' = y'' \cdot \cos. \theta + z'' \cdot \sin. \theta; \quad z' = z'' \cdot \cos. \theta - y'' \cdot \sin. \theta;$$

from which we may obtain

$$S. \left\{ \frac{y' dx' - z' dy'}{dt} \right\} dm = - \frac{d\theta}{dt} S. dm. (y''^2 + z''^2). \dagger$$

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\* Since the plane passing through the axis of  $z'$ , and of  $y'$ , of which the former is vertical, and the latter horizontal, passes constantly through the centre of gravity, this centre must move in a vertical plane.

† As the coordinates  $x'', y'', z''$ , do not vary with the time, being always the same for the same molecule, in taking the differentials of  $y', z'$ , and their respective values, with respect to the time they become

$$dy = d\theta. (z'' \cdot \cos. \theta - y'' \cdot \sin. \theta); \quad dz = - d\theta. (z'' \cdot \sin. \theta + y'' \cdot \cos. \theta)$$

$$\therefore y' dz' - z' dy' = (y'' \cdot \cos. \theta + z'' \cdot \sin. \theta). (-d\theta. (z'' \cdot \sin. \theta + y'' \cdot \cos. \theta)) - (z'' \cdot \cos. \theta - y'' \cdot \sin. \theta). \\ (d\theta. (z'' \cdot \cos. \theta - y'' \cdot \sin. \theta))$$

$S.dm.(y'^2+z'^2)$  is the moment of inertia of the body with respect to the axis of  $x'$ : \* Let this moment be equal to  $C$ . The last of the equations ( $B$ ) of No. 25, will give

$$-C \cdot \frac{d^2\theta}{dt^2} = \frac{dN''}{dt}$$

Let us suppose that the body is only solicited by the force of gravity; the values of  $P$  and of  $Q$  of No. 25, will vanish, and  $R$  will be constant, which gives

$$\frac{dN''}{dt} = S.Ry'.dm = R \cdot \cos \theta \cdot S.y''.dm + R \cdot \sin \theta \cdot S.z''.dm.$$

The axis of  $z''$  passing through the centre of gravity of the body, we have  $S.y''.dm=0$ ; moreover, if we name  $h$  the distance of the centre of gravity of the body, from the axis of  $x'$ , we shall have  $S.z''.dm=mh$ ,  $m$  being the entire mass of the body; therefore we shall have

$$= -d\theta \cdot (y''.\cos \theta + z''.\sin \theta)^2$$

$$-d\theta(z''.\cos \theta - y''.\sin \theta)^2 = -d\theta(y''^2 + z''^2)$$

∴ multiplying by  $dm$ , and extending the expression to all the molecules we obtain,

$$S \cdot \frac{y'.dz' - z'.dy'}{dt} \cdot dm = -C \cdot \frac{d\theta}{dt}.$$

and since  $C$  is constant, we shall have

$$-C \cdot \frac{d^2\theta}{dt^2} = \frac{dN''}{dt}.$$

$$* y'^2 + z'^2 = y'^2 \cdot \cos^2 \theta + z'^2 \cdot \sin^2 \theta + 2y''z''.\sin \theta \cdot \cos \theta + y''^2 \cdot \sin^2 \theta + z''^2 \cdot \cos^2 \theta$$

$$-2y''z''.\sin \theta \cdot \cos \theta = y''^2 + z''^2 ∴ S(y'^2 + z'^2).dm,$$

the moment of inertia of the body relative to the axis of

$$x' = S(z'^2 + y'^2).dm = C.$$

$$\frac{dN''}{dt} = mh \cdot R \cdot \sin. \theta *$$

and consequently

$$\frac{d^2\theta}{dt^2} = \frac{-m.h.R \cdot \sin. \theta}{C}$$

Let us now consider a second body, all whose parts are concentrated in one point, of which the distance from the axis of  $x'$ , is equal to  $l$ ; we shall have for this body,  $C = m'l^2, m'$  expressing its mass; moreover  $h$  will be equal to  $l$ ; and therefore

$$\frac{d^2\theta}{dt^2} = \frac{-R}{l} \cdot \sin. \theta \dagger$$

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\*  $N''$  is always equal to  $S.y' - Q.z'$ .  $dt.dm \therefore Q$  vanishing we shall have

$$\frac{dN''}{dt} = S.Ry'.dm$$

and by substituting for  $y'$  we obtain the expression given in the text. In fact, since the axis of  $z''$  passes through the centre of gravity, we have  $S.y'.dm = 0$ , and  $S.z'.dm = mh$ , See No. 15, page 91, it also appears from note to same number, page 88, that when a body is constrained to move about an axis, one of the equations (B) of No. 25, is sufficient to determine the motion of the body;  $\therefore$  by substituting  $mh \cdot \sin. \theta$  for  $\sin. \theta \cdot Sz''.dm$  we shall have

$$\frac{dN''}{dt} = mh \cdot R \cdot \sin. \theta.$$

† For any body  $m'$  of which all the molecules are concentrated into a point at the distance equal to  $l$  from the axis of  $x'$  we have

$$\frac{d^2\theta}{dt^2} = -\frac{m'l \cdot R}{m'l^2} \sin. \theta = -\frac{R}{l} \sin. \theta,$$

for in this case the centre of gravity, is in this point, and the moment of its inertia, is equal to  $m'l^2$ ;

if this body has the same motion of oscillation with the body we have first considered, the values of  $\frac{d^2\theta}{dt^2}$  must be the same, i.e.,

$$-\frac{mh \cdot R \cdot \sin. \theta.}{C} = -\frac{R}{l} \sin. \theta \therefore l = \frac{C}{mh}.$$

Consequently these two bodies will have the same motion of oscillation, if their initial angular velocities, when their centres of gravity, exist in the vertical, are the same, and if we have also  $l = \frac{C}{mh}$  \* The second body which we have considered is the simple pendulum, the oscillations of which are determined in No. 11, and by means of this formula we are always enabled to assign the length  $l$  of the simple pendulum of which the oscillations are isochronous, with those of the solid which we have considered in this number, and which constitutes the compound pendulum. It is thus, that the length of the simple pendulum, which vibrates in a second, is determined by observations made on compound pendulums.†

Multiplying both sides of the equation  $\frac{d^2\theta}{dt^2} = -\frac{R \cdot \sin \theta}{l}$  by  $2d\theta$ , and integrating we obtain

$$\frac{d\theta^2}{dt^2} = \frac{2R}{l} \cdot \cos \theta + C,$$

the constant quantity  $C$ , depends on the angular velocity, and on the value of  $\theta$ , at the commencement of the motion.

\* From the expression  $l = \frac{C}{mh}$ , it appears that when the axis of rotation passes through the centre of gravity,  $l$  is infinite, and consequently the time of oscillation is infinite in this case, in fact the action of gravity being destroyed, the primitive impulse will communicate a rotatory motion which will be perpetuated for ever, if the resistance of the air be removed.

† The point which is distant from the axis of rotation by a quantity equal to  $l$  is termed the centre of oscillation of the body, and if the axis of rotation passed through this point, the centre of oscillation with respect to this new axis, will be in the former axis of rotation, for the moment of inertia with respect to the centre of gravity being equal to  $C - mh^2$ , the moment with respect to the new axis will be  $C + ml^2 - 2mlh$ . See note, page 182, ∵ the value of  $l$  for the new axis  $= \frac{C + ml^2 - 2mlh}{ml - mh}$  but  $C = mlh$  ∴ the value of  $l$  for the new axis  $= \frac{ml^2 - mlh}{ml - mh} = l$ .

$C' = A \sin. {}^2\theta \cdot \sin. {}^2\phi + B \cdot \sin. {}^2\theta \cdot \cos. {}^2\phi + C \cdot \cos. {}^2\theta + mh^2$ , see page 180, where  $A, B, C$ ,

are the moments of inertia, relative to the principal axis, passing through the centre of gravity, we shall have

$$l = \frac{mh^2 + A \cdot \sin. 2\theta. \sin. {}^2\phi + B \cdot \sin. 2\theta. \cos. {}^2\phi + C \cdot \cos. 2\theta}{mh}$$

$\therefore l$  will be a minimum when the quantity represented by  $C$  in the text is the least of the three principal moments of inertia, for in that case the other two moments vanish, let  $A$  be the least of the three moments then we shall have

$$l = \frac{mh^2 + A}{mh}, \text{ for } \sin. \theta. \cos. \phi = 0, \cos. \theta = 0, \therefore \text{when } l \text{ is a minimum}$$

$$dl = \frac{2m^2h^2 - m^2h^2 - mA}{m^2h^2} \cdot dh = 0 \therefore h = \sqrt{\frac{A}{m}}$$

$\therefore l$  and consequently the time of oscillation will be a minimum when the axis of rotation is that principal axis, relatively to which, the moment of inertia is a minimum, and at a distance from the centre of gravity by a quantity equal to  $\sqrt{\frac{A}{m}}$ . The product of  $lh$ , is constant

and = to  $\frac{C}{m}$ , this fraction is equal to the square of the distance of the centre of gyration from the axis of rotation, therefore this distance is a mean proportional, between the distances of the centres, of gravity and oscillation, from the axis of rotation, and it readily appears from what precedes, that when the time of vibration is a minimum, the distance of the centre of gyration from the axis of rotation is equal to the distance of the centre of gravity from this axis, and the distance of the centre of oscillation from the same axis =  $2\sqrt{\frac{A}{m}}$ . In this case, the centre of gyration, is termed the principal centre of gyration.

## CHAPTER VIII.

*Of the motion of fluids.*

32. We may make the laws of the motion of fluids, depend on those of their equilibrium ; in the same manner, as in the fifth chapter we have deduced the laws of the motion of a system of bodies, from those of the equilibrium of the system. For this purpose, let us resume the general equation of the equilibrium of fluids, which has been given in No. 17,

$$\delta p = \rho \{ P \cdot \delta x + Q \cdot \delta y + R \cdot \delta z \} ;$$

in which, the characteristic  $\delta$  refers only to the coordinates of the molecule  $x, y, z$ , being independent of the time. When the fluid is in motion, the forces which would retain the molecules in equilibrium are by No. 18,

$$P - \left\{ \frac{d^2x}{dt^2} \right\}; Q - \left\{ \frac{d^2y}{dt^2} \right\}; R - \left\{ \frac{d^2z}{dt^2} \right\} ;$$

( $dt$  being supposed constant) ; therefore it is necessary to substitute in the preceding equation of equilibrium, these forces in place of  $P, Q, R$ . If we suppose that  $P \delta x + Q \cdot \delta y + R \delta z$  is an exact variation, represented by  $\delta V$ ; we shall have

$$\delta V - \frac{\delta p}{\rho} = \delta x \cdot \left( \frac{d^2 x}{dt^2} \right) + \delta y \cdot \left( \frac{d^2 y}{dt^2} \right) + \delta z \cdot \left( \frac{d^2 z}{dt^2} \right); * \text{ (F)}$$

this equation is equivalent to three distinct equations; because the variations  $\delta x, \delta y, \delta z$ , being independent, we are permitted to make their coefficients, separately equal to nothing.

The coordinates  $x, y, z$ , are functions of the primitive coordinates, and of the time  $t$ ; † let  $a b c$  be the primitive coordinates, we shall have

$$\begin{aligned} * \delta p &= \epsilon \left\{ P - \frac{d^2 x}{dt^2} \right\} \delta x + \left\{ Q - \frac{d^2 y}{dt^2} \right\} \delta y + \left\{ R - \frac{d^2 z}{dt^2} \right\} \delta z \\ &= \epsilon \cdot (P \delta x + Q \delta y + R \delta z) - \epsilon \left\{ \frac{d^2 x}{dt^2} \cdot \delta x + \frac{d^2 y}{dt^2} \cdot \delta y + \frac{d^2 z}{dt^2} \cdot \delta z \right\}; \end{aligned}$$

we are permitted to consider  $P \delta x + Q \delta y + R \delta z$ , an exact variation where the forces which solicit the molecules, are those of attraction directed towards fixed or moveable points, or such as arise from the mutual attraction of the fluid molecules. We have seen in No. 17, that this is the condition which must be satisfied, when the molecules of the fluid, are in equilibrio by the action of the same forces.

† The position of a molecule at any instant, is known when we know the coordinates  $a, b, c$ , which determine its position at the commencement of the motion, or at any determined epoch, ∵  $x, y, z$ , are respectively functions of  $a, b, c$ , and  $t$ , consequently we have  $x=f(a, b, c, t), y=F.(a, b, c, t); z=\phi.(a, b, c, t)$ . and as the differences indicated by the characteristic δ refer solely to the variations of the coordinates  $a, b, c$ , being independent of the time, the expressions for  $\delta x, \delta y, \delta z$ , should be such as are given in the text, ∵ if it was proposed to compare the respective positions of two molecules at any given moment, the time should be considered as constant, and the expressions for  $\delta x \delta y \delta z$  should be those which are given in page 224, on the other hand, if we consider the motion of the *same* molecule for the time  $dt$ , the values of  $dx, dy, dz$ , deduced from the preceding expressions for  $x, y, z$ , must be taken on the hypothesis that  $t$  only varies and ∵ when  $t=0, x=a, y=b, z=c$ . If the form of the preceding functions was given, by eliminating the time from the equations which determine values of  $x, y, z$ , the two equations which result will be the equations of the curve described by the molecule, however as  $a b c$  are different for each molecule, the nature of this curve and its position will be different for each molecule, see Note page 31.

$$\delta x = \left\{ \frac{dx}{da} \right\} \cdot \delta a + \left\{ \frac{dx}{db} \right\} \cdot \delta b + \left\{ \frac{dx}{dc} \right\} \cdot \delta c ;$$

$$\delta y = \left\{ \frac{dy}{da} \right\} \cdot \delta a + \left\{ \frac{dy}{db} \right\} \cdot \delta b + \left\{ \frac{dy}{dc} \right\} \cdot \delta c ;$$

$$\delta z = \left\{ \frac{dz}{da} \right\} \cdot \delta a + \left\{ \frac{dz}{db} \right\} \cdot \delta b + \left\{ \frac{dz}{dc} \right\} \cdot \delta c .$$

By substituting these values in the equation ( $F$ ), we may put the coefficients of  $\delta a$ ,  $\delta b$ ,  $\delta c$ , separately equal to nothing ; which will give three equations of partial differences between the three coordinates of the molecule  $x, y, z$ , its primitive coordinates  $a, b, c$ , and the time  $t$ .

It remains to satisfy the condition of the continuity of the fluid.\* For this purpose, let us consider at the commencement of the motion, a rectangular fluid parallelepiped, of which the three dimensions are  $da, db, dc$ . If we denote its primitive density by  $(\rho)$ , its mass will be equal to  $(\rho) \cdot da \cdot db \cdot dc$ . Let this parallelepiped be represented by ( $A$ ), it is easy to see, that after the time  $t$ ,† it will be changed into an oblique angled parallelepiped ; for all the molecules which in the primitive situation existed on any face of the

\* In order to determine the condition of a fluid mass at each instant, we must know the direction of the motion of a molecule, its velocity, the pressure  $p$ , and the density  $\xi$ , but if we know the three partial velocities parallel to the coordinates, we shall have the entire velocity, and also the direction, for the partial velocities divided by the entire velocity, are proportional to the cosines of the angles which the coordinates make with the direction, see Note page 26, and page 227.

Three of the equations which are required for the determination of those sought quantities, are furnished by the equation ( $F$ ) ; another equation from the continuity of the fluid, for though each indefinitely small portion of the fluid changes its form, and if it is compressible, its volume during the motion, still the mass must be constant, consequently the product of the volume into the density must be the same as at the commencement, ∴ by equating those two values of the mass, we obtain the equation relative to the continuity of the fluid.

† After the time  $t$ , the coordinates of the summit of the parallelogram, which were  $a, b, c$ , at the commencement of the motion, will be  $x, y, z$ , or  $f(a b c t), F(a b c t), \phi(a b c t)$ , the coordinates of that point of which the initial coordinates were  $a, b, c+dc$ , will be

parallelepiped ( $A$ ) will still be in the same plane, at least if we neglect quantities indefinitely small of the second order; all the molecules situated on the parallel edges of ( $A$ ) will be found on small right lines, equal and parallel to each other. Denoting this new parallelepiped by ( $B$ ), and conceiving that through the extremities of the slice constituted, of those molecules which in the parallelepiped ( $A$ ) compose the side  $dc$ , we draw two planes parallel to the plane of  $x$  and  $y$ . Then producing the edges of the second parallelepiped to meet these two planes, we shall have a new parallelepiped ( $C$ ) contained between

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$f(a, b, c+dc, t), F(a, b, c+dc, t), \phi(a, b, c+dc, t) =$   
respectively to

$$x + \frac{dx}{dc} \cdot dc + \frac{d^2x}{2 \cdot dc^2} \cdot dc^2 + &c.y + \frac{dy}{dc} \cdot dc + \frac{d^2y}{2 \cdot dc^2} \cdot dc^2 + &c.z + \frac{dz}{dc} \cdot dc + \frac{d^2z}{2 \cdot dc^2} \cdot dc^2 + &c.$$

the difference between these coordinates and  $x, y, z$ , are

$$\frac{dx}{dc} \cdot dc + \frac{d^2x}{2dc^3} \cdot dc^2, \frac{dy}{dc} \cdot dc + \frac{d^2y}{2dc^3} \cdot dc^2, \frac{dz}{dc} \cdot dc + \frac{d^2z}{2dc^3} \cdot dc^2$$

and the square root of the sum of the squares of these three quantities, is the value of the side of the parallelepiped which answers to the side  $dc$  of the primitive parallelepiped; extracting the square root, and neglecting the third, and higher powers of  $dc$ , this side becomes equal to

$$\frac{dz}{dc} \cdot dc + \frac{d^2z}{2dc} \cdot dc^2,$$

in like manner it may be shewn that the quantities which in the original parallelepiped are equal to  $da, db$ , become

$$\frac{dx}{da} \cdot da + \frac{d^2x}{da^2} \cdot da^2, \frac{dy}{db} \cdot db + \frac{d^2y}{db^2} \cdot db^2,$$

the opposite sides of the figure are equal to these; for the value of  $x, y, z$ , which corresponds to the primitive coordinates  $a+da, b, c$ ,

$$are f(a+da, b+ct)F(a+da+b, ct)\phi(a+da, b, ct) =$$

$$x + \frac{dx}{da} \cdot da + \frac{d^2x}{2da^2} \cdot da^2, y + \frac{dy}{da} \cdot da + \frac{d^2y}{2da^2} \cdot da^2, z + \frac{dz}{da} \cdot da + \frac{d^2z}{2da^2} \cdot da^2$$

those planes, and equal to ( $B$ ) ; for it is manifest that what one of these planes takes from the parallelepiped ( $B$ ), is added by the other plane. The two bases of the parallelepiped ( $C$ ) will be parallel to the plane  $x, y$  : its altitude contained between its bases will be equal to the difference of  $z$ , taken on the hypothesis that  $c$  \* only varies ; consequently this altitude will be equal to  $\left(\frac{dz}{dc}\right) \cdot dc$ .

the values of  $x, y, z$ , which answers to the primitive coordinates  $a+da, b, c+dc$ , will be

$$\begin{aligned} & f(a+da, b, c+dc, t) F(a+da, b, c+dc, t) \varphi(a+da, b, c+dc, t) = \\ & z + \frac{dx}{da} \cdot da + \frac{d^2x}{2 \cdot da^2} \cdot da^2 + \frac{dx}{dc} \cdot dc + \frac{d^2x}{2 \cdot dc^2} \cdot dc^2 + \&c. y + \frac{dy}{da} \cdot da + \frac{d^2y}{2 \cdot da^2} \cdot da^2, \frac{dy}{dc} \cdot dc \frac{d^2y}{2 \cdot dc^2} \cdot 2dc^2 + \&c. \\ & z + \frac{dz}{dc} \cdot dc + \frac{d^2z}{2 \cdot dc^2} \cdot dc^2 + \frac{dz}{da} \cdot da + \frac{d^2z}{2 \cdot da^2} \cdot da^2 \end{aligned}$$

$\therefore$  the difference of the coordinates of these points

$$= \frac{dx}{dc} \cdot dc + \frac{d^2x}{2 \cdot dc^2} \cdot dc^2, \frac{dy}{dc} \cdot dc + \frac{d^2y}{2 \cdot dc^2} \cdot dc^2, \frac{dz}{dc} \cdot dc + \frac{d^2z}{2 \cdot dc^2} \cdot dc^2,$$

and as these differences are equal to the corresponding differences of the opposite side of the figure, it follows that these sides must be equal, being equal to the square root of the sum of the squares of these differences, in like manner it may be proved, that the other sides are respectively equal to those to which they are opposed ; and the parallelism of these sides is a necessary consequence of their equality, from which we infer that the figure which the molecules assume is a parallelepiped. The equation of the line connecting the points whose respective coordinates are

$$f(a, b, c, t), F(a, b, c, t), \varphi(a, b, c, t), f(a+da, b, c, t), F(a+da, b, c, t), \varphi(a+da, b, c, t),$$

will be that of a right line, if we neglect the indefinitely small quantities of the second order, and the same is true for all lines parallel to this line, of the sum of which the face may be conceived to made up,  $\therefore$  this face may be considered as a plane.

\* The difference between the values of  $z$  corresponding to the expressions

$$z = \varphi(a, b, c, t), z' = \varphi(a, b, c+dc, t) = \frac{dz}{dc} \cdot dc + \left\{ \frac{d^2z}{dc^2} \right\} \cdot \frac{dc^2}{1.2} = \left\{ \frac{dz}{dc} \right\} \cdot dc$$

We shall obtain its base, by remarking that it is equal to a section of ( $B$ ) made by a plane parallel to the plane of  $x, y,$ ; let us designate this section by ( $\varepsilon$ ). The value of  $z$  will be the same for all the molecules of which this base is constituted, therefore we shall have

$$o = \left\{ \frac{dz}{da} \right\} \cdot da + \left\{ \frac{dz}{db} \right\} \cdot db + \left\{ \frac{dz}{dt} \right\} \cdot dc.$$

Let  $\delta p, \delta q$ , be two contiguous sides of the section ( $\varepsilon$ ), of which the first is made up of molecules which existed on the face  $db \cdot dc$  of the parallelepiped ( $A$ ), and of which the second is composed of molecules which existed on the face  $da \cdot dc$ . If we conceive two lines to be drawn through the extremities of the side  $\delta p$ , parallel to the axis of  $x$ , by producing them to meet that side of the parallelogram ( $\varepsilon$ ), which is parallel to  $\delta p$ , they will intercept a new parallelogram ( $\lambda$ ) equal to ( $\varepsilon$ ), of which the base will be parallel to the axis of  $x$ . The side  $\delta p$  being composed of molecules which existed on the face  $db \cdot dc$ , and relatively to which the value of  $z$  is constant; it is easy to perceive that the altitude of the parallelogram ( $\lambda$ ) is the difference of  $y$ , on the supposition that  $a, z$ , and  $t$  are constant, consequently we have

$$dy = \left\{ \frac{dy}{db} \right\} \cdot db + \left\{ \frac{dy}{dc} \right\} \cdot dc; ^*$$

$$o = \left\{ \frac{dz}{db} \right\} \cdot db + \left\{ \frac{dz}{dc} \right\} \cdot dc;$$

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by neglecting quantities indefinitely small of the second order. For all the molecules situated on the edge, which corresponds to  $dc$  in the original parallelepiped, projected on the axis of  $z$ , the values  $a$  and  $b$  remain the same, nor do any molecules which occur in the face  $da \cdot db$  enter in the constitution of this perpendicular, therefore it is equal to  $dz$  on the hypothesis that  $c$  only varies.

\* If we conceive the molecules of the face  $db \cdot dc$  relatively to which  $dz$  is constant, to be projected on the axis of  $y$ , it is evident that the projected line is equal to the difference

from which may be obtained

$$\frac{dy = \left\{ \left( \frac{dy}{db} \right) \cdot \left( \frac{dz}{dc} \right) - \left( \frac{dy}{dc} \right) \cdot \left( \frac{dz}{db} \right) \right\} \cdot db}{\left( \frac{dz}{dc} \right)}$$

this is the expression for the altitude of the parallelogram ( $\lambda$ ). Its base is equal to a section of this parallelogram by a plane parallel to the axis of  $x$ ; this section is composed of those molecules of the parallelepiped ( $A$ ), with respect to which  $z$  and  $y$  are constant; its length will be equal to the differential of  $x$  taken on the hypothesis that  $z$ ,  $y$ , and  $t$  are constant, which gives the three following equations

$$dx = \left\{ \frac{dx}{da} \right\} \cdot da + \left\{ \frac{dx}{db} \right\} \cdot db + \left\{ \frac{dx}{dc} \right\} \cdot dc ; *$$

$$0 = \left\{ \frac{dy}{da} \right\} \cdot da + \left\{ \frac{dy}{db} \right\} \cdot db + \left\{ \frac{dy}{dc} \right\} \cdot dc ;$$

$$0 = \left\{ \frac{dz}{da} \right\} \cdot da + \left\{ \frac{dz}{db} \right\} \cdot db + \left\{ \frac{dz}{dc} \right\} \cdot dc.$$

of  $y$ , on the hypothesis that  $a$  is constant, for this projection is the same for every series of molecules, which exist on the face which corresponds to the primitive face  $db \cdot dc$ , and relatively to which  $z$  is the same. We obtain the expression which is given in the text for  $dy$  by eliminating  $dc$  between the two preceding equations.

\* Since the parallelogram ( $\lambda$ ) exists in the plane parallel to the axes of  $x$ ,  $y$ , the value of  $z$  will be constant for this parallelogram, and since the base of ( $\lambda$ ) is a line parallel to the axis of  $x$  the value of  $y$  will be the same for all molecules situated in this base, but since in this base molecules occur which belong to the faces  $da \cdot db$ ,  $da \cdot dc$ ,  $db \cdot dc$ ,  $a \cdot b \cdot c$ , will vary for these molecules.

In order to abridge, let us make

$$\epsilon = \left\{ \frac{dx}{da} \right\} \cdot \left\{ \frac{dy}{db} \right\} \cdot \left\{ \frac{dx}{dc} \right\} - \left\{ \frac{dx}{da} \right\} \cdot \left\{ \frac{dy}{dc} \right\} \cdot \left\{ \frac{dz}{db} \right\}$$

$$+ \left\{ \frac{dx}{db} \right\} \cdot \left\{ \frac{dy}{dc} \right\} \cdot \left\{ \frac{dz}{da} \right\} *$$

\* Multiplying the second equation by  $\left\{ \frac{dz}{dc} \right\}$ , and the third by  $\left\{ \frac{dy}{dc} \right\}$ , and then subtracting we shall eliminate dc

$$\left\{ \frac{dz}{dc} \right\} \cdot \left\{ \frac{dy}{da} \right\} \cdot da + \left\{ \frac{dz}{dc} \right\} \cdot \left\{ \frac{dy}{db} \right\} \cdot db + \left\{ \frac{dz}{dc} \right\} \cdot \left\{ \frac{dy}{dc} \right\} \cdot dc = 0$$

$$\left\{ \frac{dy}{dc} \right\} \cdot \left\{ \frac{dz}{da} \right\} \cdot da + \left\{ \frac{dy}{dc} \right\} \cdot \left\{ \frac{dz}{db} \right\} \cdot db + \left\{ \frac{dy}{dc} \right\} \cdot \left\{ \frac{dz}{dc} \right\} \cdot dc = 0$$

$$\therefore \left\{ \left( \frac{dz}{dc} \right) \cdot \left( \frac{dy}{da} \right) - \left( \frac{dy}{dc} \right) \cdot \left( \frac{dz}{da} \right) \right\} \cdot da + \left\{ \left( \frac{dz}{dc} \right) \cdot \left( \frac{dy}{db} \right) - \left( \frac{dy}{dc} \right) \cdot \left( \frac{dz}{db} \right) \right\} \cdot db = 0$$

$$\therefore db = \frac{\left\{ \left( \frac{dy}{dc} \right) \cdot \left( \frac{dz}{da} \right) - \left( \frac{dz}{dc} \right) \cdot \left( \frac{dy}{da} \right) \right\}}{\left( \frac{dz}{dc} \right) \cdot \left( \frac{dy}{db} \right) - \left( \frac{dy}{dc} \right) \cdot \left( \frac{dz}{db} \right)} \cdot da$$

in like manner we can obtain

$$\left\{ \left\{ \frac{dz}{db} \right\} \cdot \left\{ \frac{dy}{da} \right\} - \left\{ \frac{dy}{db} \right\} \cdot \left\{ \frac{dz}{da} \right\} \right\} \cdot da + \left\{ \left\{ \frac{dz}{db} \right\} \cdot \left\{ \frac{dy}{dc} \right\} - \left\{ \frac{dy}{db} \right\} \cdot \left\{ \frac{dz}{dc} \right\} \right\} \cdot dc = 0$$

$$\therefore dc = \frac{\left\{ \frac{dy}{db} \right\} \cdot \left\{ \frac{dz}{da} \right\} - \left\{ \frac{dz}{db} \right\} \cdot \left\{ \frac{dy}{da} \right\}}{\left\{ \frac{dz}{db} \right\} \cdot \left\{ \frac{dy}{dc} \right\} - \left\{ \frac{dy}{db} \right\} \cdot \left\{ \frac{dz}{dc} \right\}} \cdot da$$

$$dx = \left\{ \left( \frac{dx}{da} \right) + \frac{\left\{ \frac{dy}{dc} \right\} \cdot \left\{ \frac{dz}{da} \right\} - \left\{ \frac{dz}{dc} \right\} \cdot \left\{ \frac{dy}{da} \right\}}{\left\{ \frac{dz}{dc} \right\} \cdot \left\{ \frac{dy}{db} \right\} - \left\{ \frac{dy}{dc} \right\} \cdot \left\{ \frac{dz}{db} \right\}} \cdot \left\{ \frac{dx}{db} \right\} \right\}$$

$$-\left\{\frac{dx}{db}\right\} \cdot \left\{\frac{dy}{da}\right\} \cdot \left\{\frac{dz}{dc}\right\} + \left\{\frac{dx}{dc}\right\} \cdot \left\{\frac{dy}{da}\right\} \cdot \left\{\frac{dz}{db}\right\} - \left\{\frac{dx}{dc}\right\} \cdot \left\{\frac{dy}{db}\right\} \cdot \left\{\frac{dz}{da}\right\}$$

we shall have

$$dx = \frac{\epsilon \cdot da}{\left\{\frac{dy}{db}\right\} \cdot \left\{\frac{dz}{dc}\right\} - \left\{\frac{dy}{dc}\right\} \cdot \left\{\frac{dz}{db}\right\}}$$

this is the value of the base of the parallelogram ( $\lambda$ ); therefore the surface of this parallelogram will be equal to

$$\frac{\epsilon \cdot da \cdot db}{\left(\frac{dz}{dc}\right)}$$

This quantity also expresses the surface of the parallelogram ( $\epsilon$ ), if we multiply it by  $\left(\frac{dz}{dc}\right) \cdot dc$  we shall have  $\epsilon \cdot da \cdot db \cdot dc$  for the volume of the

$$\begin{aligned} &+ \frac{\left\{\frac{dy}{db}\right\} \cdot \left\{\frac{dz}{da}\right\} - \left\{\frac{dz}{db}\right\} \cdot \left\{\frac{dy}{da}\right\}}{\left\{\frac{dz}{dc}\right\} \cdot \left\{\frac{dy}{db}\right\} - \left\{\frac{dy}{dc}\right\} \cdot \left\{\frac{dz}{db}\right\}} \cdot \left\{\frac{dx}{dc}\right\} \cdot da = \\ &\left\{ \left\{\frac{dx}{da}\right\} \cdot \left\{\frac{dz}{dc}\right\} \cdot \left\{\frac{dy}{db}\right\} - \left\{\frac{dx}{da}\right\} \cdot \left\{\frac{dy}{dc}\right\} \cdot \left\{\frac{dz}{db}\right\} + \left\{\frac{dx}{db}\right\} \cdot \left\{\frac{dy}{dc}\right\} \cdot \left\{\frac{dz}{da}\right\} \right. \\ &- \left. \left\{\frac{dx}{db}\right\} \cdot \left\{\frac{dz}{dc}\right\} \cdot \left\{\frac{dy}{da}\right\} + \left\{\frac{dx}{dc}\right\} \cdot \left\{\frac{dy}{db}\right\} \cdot \left\{\frac{dz}{da}\right\} - \left\{\frac{dx}{dc}\right\} \cdot \left\{\frac{dz}{da}\right\} \cdot \left\{\frac{dy}{db}\right\} \right\} \cdot da \\ &\div \left\{\frac{dz}{dc}\right\} \cdot \left\{\frac{dy}{db}\right\} - \left\{\frac{dy}{dc}\right\} \cdot \left\{\frac{dz}{db}\right\} \\ &= \frac{\epsilon \cdot da}{\left\{\frac{dz}{dc}\right\} \cdot \left\{\frac{dy}{db}\right\} - \left\{\frac{dy}{dc}\right\} \cdot \left\{\frac{dz}{db}\right\}} \end{aligned}$$

= the base of the parallelogram ( $\lambda$ ), this expression being multiplied into the value of  $dy$  gives the area of ( $\lambda$ ), and this area being multiplied by the altitude gives the volume of ( $C$ )

parallelepipeds (*C*), and (*B*). Let  $\rho$  represent the density of the parallelepiped (*A*), after the time  $t$ ; we shall have its mass equal to  $\rho \epsilon.da.db.dc$ ; and by equating this to its primitive mass  $(\rho).da.db.dc$  we shall have

$$\rho\epsilon = (\rho); \quad (G)$$

for the equation relative to the continuity of the fluid.

33. The equations (*F*) and (*G*) may be made to assume another form, which is in certain circumstances of more convenient application. Let  $u$ ,  $v$ , and  $V$  be the velocities of a molecule of the fluid, parallel to the axes of  $x$ , of  $y$ , and of  $z$ ; we shall have

$$\left\{ \frac{dx}{dt} \right\} = u; \quad \left\{ \frac{dy}{dt} \right\} = v; \quad \left\{ \frac{dz}{dt} \right\} = V.$$

By differentiating these equations,  $u$ ,  $v$ ,  $V$  being considered as functions of the coordinates  $x$ ,  $y$ ,  $z$ , of the molecule, and of the time  $t$ , we shall have

$$\left\{ \frac{d^2x}{dt^2} \right\} = \left\{ \frac{du}{dt} \right\} + u. \left\{ \frac{du}{dx} \right\} + v. \left\{ \frac{du}{dy} \right\} + V. \left\{ \frac{du}{dz} \right\}; \quad *$$

\* $u$ ,  $v$ ,  $V$ , are respectively unknown functions of  $x$ ,  $y$ ,  $z$ , and  $t$ , they depend on the coordinates  $x$ ,  $y$ ,  $z$ , because for a given value of  $t$ , the velocity is different in different molecules, they depend on  $t$ , because for the same values of  $x$ ,  $y$ ,  $z$ , the velocity varies every instant,

$$\therefore du = \left\{ \frac{du}{dt} \right\} \cdot dt + \left\{ \frac{du}{dx} \right\} \cdot dx + \left\{ \frac{du}{dy} \right\} \cdot dy + \left\{ \frac{du}{dz} \right\} \cdot dz,$$

and since  $dx = u dt$ ,  $dy = v dt$ ,  $dz = V dt$ ,  
substituting and dividing by  $dt$ , we obtain

$$\frac{du}{dt} = \left\{ \frac{du}{dt} \right\} + \left\{ \frac{du}{dz} \right\} \cdot V + \left\{ \frac{du}{dy} \right\} \cdot V + \left\{ \frac{du}{dx} \right\} \cdot V,$$

$$\text{but } u = \frac{dx}{dt} \therefore \frac{du}{dt} = \frac{d^2x}{dt^2}.$$

From the values of  $\frac{du}{dt}$ ,  $\frac{dv}{dt}$ ,  $\frac{dV}{dt}$ , given in the text, it appears how the increment of each of the three velocities depends on the two other velocities. If we were able to determine the

$$\left\{ \frac{d^2y}{dt^2} \right\} = \left\{ \frac{dv}{dt} \right\} + u \cdot \left\{ \frac{dv}{dx} \right\} + v \cdot \left\{ \frac{dv}{dy} \right\} + V \cdot \left\{ \frac{dv}{dz} \right\};$$

$$\left\{ \frac{d^2z}{dt^2} \right\} = \left\{ \frac{dV}{dt} \right\} + u \cdot \left\{ \frac{dV}{dx} \right\} + v \cdot \left\{ \frac{dV}{dy} \right\} + V \cdot \left\{ \frac{dV}{dz} \right\}.$$

consequently the equation (*F*) of the preceding number will become,

$$\begin{aligned} \delta V - \frac{\delta p}{\rho} &= \delta x \cdot \left\{ \left\{ \frac{du}{dt} \right\} + u \cdot \left\{ \frac{du}{dx} \right\} + v \cdot \left\{ \frac{du}{dy} \right\} + V \cdot \left\{ \frac{du}{dz} \right\} \right\} \\ &\quad + \delta y \cdot \left\{ \left\{ \frac{dv}{dt} \right\} + u \cdot \left\{ \frac{dv}{dx} \right\} + v \cdot \left\{ \frac{dv}{dy} \right\} + V \cdot \left\{ \frac{dv}{dz} \right\} \right\}; (H) \\ &\quad + \delta z \cdot \left\{ \left\{ \frac{dV}{dt} \right\} + u \cdot \left\{ \frac{dV}{dx} \right\} + v \cdot \left\{ \frac{dV}{dy} \right\} + V \cdot \left\{ \frac{dV}{dz} \right\} \right\} \end{aligned}$$

In order to have the equation relative to the continuity of the fluid ; let us conceive that in the value of  $\epsilon$ , of the preceding number,  $a, b, c$ , were equal to  $x, y, z$ , and that  $x, y, z$ , were equal to  $x + u dt, y + v dt, z + V dt$ , which is equivalent to assuming the primitive coordinates  $a, b, c$ , indefinitely near to  $x, y, z$ ; we shall have

value of  $u$  in a function of  $x, y, z, t$ , we could by means of the equations  $\frac{dx}{dt} = u, \frac{dy}{dt} = v,$

$\frac{dz}{dt} = V$  determine the position of a molecule at any instant, provided we know the initial position of this molecule, and also what function of  $x, y, z, t, u, v, V$  are, for substituting in the equations  $\frac{dx}{dt} = u, \frac{dy}{dt} = v, \frac{dz}{dt} = V$  the values of  $u, v, V$ , in functions of  $x, y, z, t$ , and integrating, we would obtain the values of  $x, y, z$ , respectively in a function of  $t$ , the constant arbitrary quantities which are introduced are the values of  $x, y, z$ , at the commencement of the motion which by hypothesis are given, consequently the values of  $x, y, z$  will be completely determined for any instant. Eliminating  $t$  between values of  $x, y, z$ , to which we have arrived, we would obtain the two equations of the curve described by the molecule, but since the initial position of each molecule is different, the form of this curve will also be different, as will be in like manner, the position.

$$\epsilon = 1 + dt \cdot \left\{ \left\{ \frac{du}{dx} \right\} + \left\{ \frac{dv}{dy} \right\} + \left\{ \frac{dV}{dz} \right\} \right\};^*$$

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\* The first coordinates being assumed indefinitely near to  $x, y, z$ , we shall have  $da = dx$ , and the quantity which corresponds to  $dx$  to  $dx + du.dt$ , in like manner we shall have

$$\frac{dx + du.dt}{dz}, \frac{dx + du.dt}{dy}, \frac{dy + dv.dt}{dz}, \frac{dy + dv.dt}{dx}, \frac{dz + dV.dt}{dy}, \frac{dz + dV.dt}{dx}$$

respectively indefinitely small, because when  $t=0$  these quantities vanish,  $\therefore$  the product of any two of these quantities may be neglected, making these substitutions the expression for  $\epsilon$  becomes equal to

$$\begin{aligned} & \left\{ \frac{dx + du.dt}{dx} \right\} \cdot \left\{ \frac{dy + dv.dt}{dy} \right\} \cdot \left\{ \frac{dz + dV.dt}{dz} \right\} \\ & - \left\{ \frac{dx + du.dt}{dx} \right\} \cdot \left\{ \frac{dy + dv.dt}{dz} \right\} \cdot \left\{ \frac{dz + dV.dt}{dy} \right\} \\ & + \left\{ \frac{dx + du.dt}{dy} \right\} \cdot \left\{ \frac{dy + dv.dt}{dz} \right\} \cdot \left\{ \frac{dz + dV.dt}{dx} \right\} \\ & - \left\{ \frac{dx + du.dt}{dy} \right\} \cdot \left\{ \frac{dy + dv.dt}{dx} \right\} \cdot \left\{ \frac{dz + dV.dt}{dz} \right\} \\ & + \left\{ \frac{dx + du.dt}{dz} \right\} \cdot \left\{ \frac{dy + dv.dt}{dx} \right\} \cdot \left\{ \frac{dz + dV.dt}{dy} \right\} \\ & - \left\{ \frac{dx + du.dt}{dz} \right\} \cdot \left\{ \frac{dy + dv.dt}{dy} \right\} \cdot \left\{ \frac{dz + dV.dt}{dx} \right\} \end{aligned}$$

the first term of this expression

$$= \left\{ 1 + \frac{du.dt}{dx} \right\} \cdot \left\{ 1 + \frac{dv.dt}{dy} \right\} \cdot \left\{ 1 + \frac{dV.dt}{dz} \right\}$$

= by neglecting quantities indefinitely small

$$1 + \left\{ \frac{du}{dx} + \frac{dv}{dy} + \frac{dV}{dz} \right\} dt,$$

the other terms of this expression vanish. It appears from what precedes that  $\epsilon$  is a constant quantity independent of the time, when the fluid is incompressible  $\epsilon=1$ .

the equation (*G*) becomes,

$$\rho \cdot \left\{ \left\{ \frac{du}{dx} \right\} + \left\{ \frac{dv}{dy} \right\} + \left\{ \frac{dV}{dz} \right\} \right\} + p - (\rho) = 0.$$

If we consider  $\rho$  as a function of  $x, y, z$ , and  $t$ , we shall have

$$(\rho) = \rho - dt \cdot \left\{ \frac{d\rho}{dt} \right\} - u dt \cdot \left\{ \frac{d\rho}{dx} \right\} - v dt \cdot \left\{ \frac{d\rho}{dy} \right\} - V dt \cdot \left\{ \frac{d\rho}{dz} \right\};$$

therefore the preceding equation will become

$$0 = \left\{ \frac{d\rho}{dt} \right\} + \left\{ \frac{d \cdot \rho u}{dx} \right\} + \left\{ \frac{d \cdot \rho v}{dy} \right\} + \left\{ \frac{d \cdot \rho V}{dz} \right\}; (K) *$$

\* The density  $\rho$ , the pressure  $p$ , may be shewn to be functions of  $x, y, z, t$ , by reasoning analogous to that, by which  $u, v, V$ , were proved to be functions of these quantities;

$$\rho \cdot dt \cdot \left\{ \left\{ \frac{du}{dx} \right\} + \left\{ \frac{dv}{dy} \right\} + \left\{ \frac{dV}{dz} \right\} \right\}$$

is the increment of  $\rho$  on the supposition that  $t$  is constant,

$$dt \cdot \left\{ \frac{d\rho}{dt} \right\} - u \cdot dt \cdot \left\{ \frac{d\rho}{dx} \right\} - v \cdot dt \cdot \left\{ \frac{d\rho}{dy} \right\} - V \cdot dt \cdot \left\{ \frac{d\rho}{dz} \right\}$$

is the variation of  $\rho$  on the hypothesis that  $x, y, z, t$ , vary  $\therefore$  their difference

$$\begin{aligned} &= \rho \cdot \left\{ \frac{du}{dx} \right\} + u \cdot \left\{ \frac{d\rho}{dx} \right\} + \rho \cdot \left\{ \frac{dv}{dy} \right\} + v \cdot \left\{ \frac{d\rho}{dy} \right\} \\ &\quad + \rho \cdot \left\{ \frac{dV}{dz} \right\} + V \cdot \left\{ \frac{d\rho}{dz} \right\} + \left\{ \frac{d\rho}{dt} \right\} \\ &= \left\{ \frac{d \cdot \rho u}{dx} \right\} + \left\{ \frac{d \cdot \rho v}{dy} \right\} + \left\{ \frac{d \cdot \rho V}{dz} \right\} + \left\{ \frac{d\rho}{dt} \right\} \end{aligned}$$

is the differential of the equation (*G*) taken with respect to the time;

this is the equation relative to the continuity of the fluid, and it is easy to perceive that it is the differential of the equation (*G*) of the preceding number, taken with respect to the time *t*.

The equation (*H*) is susceptible of integration in a very extensive case that is, when  $u\partial x + v\partial y + V\partial z$  is an exact variation of *x*, *y*, *z*,  $\rho$  being any function whatever of the pressure *p*. Therefore if we re-

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when the fluid is incompressible, we have

$$\left\{ \frac{d\rho}{dx} \right\} \cdot u + \left\{ \frac{d\rho}{dy} \right\} \cdot v + \left\{ \frac{d\rho}{dz} \right\} \cdot V + \left\{ \frac{d\rho}{dt} \right\}$$

$$= 0; \text{ & } \left\{ \left\{ \frac{du}{dx} \right\} + \left\{ \frac{dv}{dy} \right\} + \left\{ \frac{dV}{dz} \right\} \right\} = 0;$$

for in this case both the magnitude, and density are constant,  $\therefore d\rho$  and  $d\rho_t$  are respectively equal to nothing, these two equations combined with the three, which may be derived from the equations (*H*), or (*F*), are sufficient to determine *p*,  $\rho$ , and the three partial velocities, *u*, *v*, *V*, in functions of *x*, *y*, *z*, *t*. When the differential coefficients  $\frac{d\rho}{dt}$ ,  $\frac{d\rho}{dx}$ ,  $\frac{d\rho}{dy}$ ,  $\frac{d\rho}{dz}$ , vanish of themselves,  $\rho$  must be a constant quantity, and the incompressible fluid will be also homogenous,  $\therefore$  in this case the number of unknown quantities is reduced to four, which is also the number of differential equations. When the fluid is elastic the number of unknown quantities will be ultimately reducible to four, for when the temperature is given  $p=k.\rho$ ,  $\therefore$  the equation (*K*) and the three equations (*H*) are sufficient to determine the unknown quantities, in this case

$$\frac{\partial p}{\rho} = \frac{1}{k} \cdot \frac{\partial \rho}{\rho} = \frac{1}{k} \cdot \partial \log \rho.$$

*k* will not be constant when the temperature varies, but if the law of its variation is known, since for each different instant, and point of space the temperature is a given function of *x*, *y*, *z*, *t*, *k* will be so likewise, so that even in this case the equations (*K*) and *H* are sufficient to determine  $\rho$ , *u*, *v*, *V*. It appears from what precedes, that we have *always* as many equations of partial differences as sought quantities, however the general integration of these equations has baffled the ingenuity of Philosophers and even granting that it is possible to effect this integration, still the determination of the arbitrary functions introduced by these integrations, is extremely difficult, these functions depend partly, on the primitive state of the fluid, and partly on the equation of the exterior surface.

present this variation by  $\delta\phi$ , the equation (*H*) will give

$$\delta V - \frac{\delta p}{\rho} = \delta \cdot \left\{ \frac{d\phi}{dt} \right\} + \frac{1}{2} \cdot \delta \cdot \left\{ \left\{ \frac{d\phi}{dx} \right\}^2 + \left\{ \frac{d\phi}{dy} \right\}^2 + \left\{ \frac{d\phi}{dz} \right\}^2 \right\};$$

from which may be obtained by integrating with respect to  $\delta$ ,

$$V - \int \frac{\delta p}{\rho} = \left\{ \frac{d\phi}{dt} \right\} + \frac{1}{2} \cdot \left\{ \left\{ \frac{d\phi}{dx} \right\}^2 + \left\{ \frac{d\phi}{dy} \right\}^2 + \left\{ \frac{d\phi}{dz} \right\}^2 \right\}. *$$

\* If we take the differential of the equation  $u\delta x + v\delta y + V\delta z$  with respect to  $t, x, y, z$ , we shall obtain

$$\frac{du}{dt} \cdot dt \cdot \delta x + \frac{dv}{dt} \cdot dt \cdot \delta y + \frac{dV}{dt} \cdot dt \cdot \delta z = \frac{d\delta\phi}{dt} \cdot dt = \frac{du}{dt} \cdot dt \cdot \delta x + \frac{dv}{dt} \cdot dt \cdot \delta y + \frac{dV}{dt} \cdot dt \cdot \delta z$$

$$\frac{du}{dx} \cdot dx \cdot \delta y + \frac{dv}{dx} \cdot dx \cdot \delta y + \frac{dV}{dx} \cdot dx \cdot \delta z = \frac{d\delta\phi}{dx} \cdot dx = \frac{du}{dx} \cdot u dt \cdot \delta x + \frac{dv}{dx} \cdot u dt \cdot \delta y + \frac{dV}{dx} \cdot u dt \cdot \delta z$$

$$\frac{du}{dy} \cdot dy \cdot \delta x + \frac{dv}{dy} \cdot dy \cdot \delta y + \frac{dV}{dy} \cdot dy \cdot \delta z = \frac{d\delta\phi}{dy} \cdot dy = \frac{du}{dy} \cdot v dt \cdot \delta x + \frac{dv}{dy} \cdot v dt \cdot \delta y + \frac{dV}{dy} \cdot v dt \cdot \delta z$$

$$\frac{du}{dz} \cdot dz \cdot \delta x + \frac{dv}{dz} \cdot dz \cdot \delta y + \frac{dV}{dz} \cdot dz \cdot \delta z = \frac{d\delta\phi}{dz} \cdot dz = \frac{du}{dz} \cdot V dt \cdot \delta z + \frac{dv}{dz} \cdot V dt \cdot \delta y + \frac{dV}{dz} \cdot V dt \cdot \delta z$$

now substituting  $udt, vdt, Vdt$ , in place of  $dx, dy, dz$ , and remarking that,  $u = \frac{d\phi}{dz}, v = \frac{d\phi}{dy}$ , &c. and also that  $\delta \cdot \frac{d\phi}{dt} = \frac{d\delta\phi}{dt}$  we shall have

$$\begin{aligned} & \delta \cdot \left\{ \frac{d\phi}{dt} \right\} + \delta \cdot \left\{ \frac{d\phi}{dx} \right\} \cdot \left\{ \frac{d\phi}{dx} \right\} + \delta \cdot \left\{ \frac{d\phi}{dy} \right\} \cdot \left\{ \frac{d\phi}{dy} \right\} + \delta \cdot \left\{ \frac{d\phi}{dz} \right\} \cdot \left\{ \frac{d\phi}{dz} \right\} \\ &= \delta \cdot \left\{ \frac{d\phi}{dt} + \frac{1}{2} \cdot \delta \cdot \left\{ \left\{ \frac{d\phi}{dx} \right\}^2 + \left\{ \frac{d\phi}{dy} \right\}^2 + \left\{ \frac{d\phi}{dz} \right\}^2 \right\} \right\} \end{aligned}$$

= the sum of the last members of the preceding equations, but these by concinnating, and dividing by  $dt$  are evidently equal to the second member of the equation (*H*). Since the integration is only made relative to the characteristic  $\delta$ , it is evident that the time is not involved in this expression. When the fluid is homogenous  $\frac{\delta r}{dx} &c. = 0 \therefore$  the equation of continuity is reduced to the second term, by means of this equation, and the equations  $u = \frac{d\phi}{dx}$ ,

It is necessary to add to this integral, a constant quantity, which is a function of  $t$ ; but we may suppose that this function is contained in the function  $\phi$ . This last function gives the velocity of the molecules of the fluid parallel to the axes of  $x$ , of  $y$ , and of  $z$ ; for we have

$$u = \left\{ \frac{d\phi}{dx} \right\}; v = \left\{ \frac{d\phi}{dy} \right\}; V = \left\{ \frac{d\phi}{dz} \right\}.$$

The equation ( $K$ ) relative to the continuity of the fluid, becomes

$$\begin{aligned} 0 = & \left\{ \frac{dp}{dt} \right\} + \left\{ \frac{dp}{dx} \right\} \cdot \left\{ \frac{d\phi}{dx} \right\} + \left\{ \frac{dp}{dy} \right\} \cdot \left\{ \frac{d\phi}{dy} \right\} + \left\{ \frac{dp}{dz} \right\} \cdot \left\{ \frac{d\phi}{dz} \right\} \\ & + p \cdot \left\{ \left\{ \frac{d^2\phi}{dx^2} \right\} + \left\{ \frac{d^2\phi}{dy^2} \right\} + \left\{ \frac{d^2\phi}{dz^2} \right\} \right\}; \end{aligned}$$

consequently, we shall have in the case of homogenous fluids,

$$0 = \left\{ \frac{d^2\phi}{dx^2} \right\} + \left\{ \frac{d^2\phi}{dy^2} \right\} + \left\{ \frac{d^2\phi}{dz^2} \right\}.$$

It may be observed, that if the function  $u\delta x + v\delta y + V\delta z$  is an exact variation of  $x, y, z$ , at any one instant, it will always remain so. In fact, let us suppose that at any instant whatever, it is equal to  $\delta\phi$ , in the subsequent instant it will be equal to

$$\delta\phi + dt \cdot \left\{ \left\{ \frac{du}{dt} \right\} \cdot \delta x + \left\{ \frac{dv}{dt} \right\} \cdot \delta y + \left\{ \frac{dV}{dt} \right\} \cdot \delta z \right\}; *$$

$v = \frac{d\phi}{dy}$ ,  $V = \frac{d\phi}{dz}$ , and the value for  $\int \frac{\delta p}{\epsilon}$ , in this case  $\frac{p}{\epsilon}$ , we can determine  $\phi$  and  $p$  and consequently  $u, v, V$ , in functions of  $x, y, z$ .

\* From the value of  $V - f \cdot \frac{\delta p}{\epsilon}$  it appears that the pressure of a molecule, of which the density is constant, diminishes when the velocity which is equal to

therefore it will be an exact variation at this new instant, if

$$\left\{ \frac{du}{dt} \right\} \cdot \delta x + \left\{ \frac{dv}{dt} \right\} \cdot \delta y + \left\{ \frac{dV}{dt} \right\} \cdot \delta z$$

$$\sqrt{\left\{ \frac{d\phi}{dx} \right\}^2 + \left\{ \frac{d\phi}{dy} \right\}^2 + \left\{ \frac{d\phi}{dz} \right\}^2}$$

is increased.

$$d \cdot \frac{d\phi}{dt} \cdot dt = \frac{d \cdot d\phi}{dt} \cdot dt = d \cdot \left\{ \frac{d\phi}{dt} \right\} \cdot dt = \left\{ \left\{ \frac{du}{dt} \right\} \delta x + \left\{ \frac{dv}{dt} \right\} \delta y + \left\{ \frac{dV}{dt} \right\} \delta z \right\} \cdot dt$$

substituting this value of  $d \cdot \left\{ \frac{d\phi}{dt} \right\}$  in the expression for  $\delta V - \frac{\delta p}{\epsilon}$  we obtain

$$\begin{aligned} & \left\{ \frac{du}{dt} \right\} \delta x + \left\{ \frac{dv}{dt} \right\} \delta y + \left\{ \frac{dV}{dt} \right\} \delta z = \delta V \\ & - \frac{1}{2} \cdot d \cdot \left\{ \left\{ \frac{d\phi}{dx} \right\}^2 + \left\{ \frac{d\phi}{dy} \right\}^2 + \left\{ \frac{d\phi}{dz} \right\}^2 \right\} - \frac{\delta p}{\epsilon} \end{aligned}$$

and since each of the terms, of the second member of this equation, are exact variations of  $x, y, z$ , the first member will also be an exact variation, we suppose  $\epsilon$  to be a function of  $p$ .

$$\left\{ \left\{ \frac{du}{dt} \right\} \cdot \delta x + \left\{ \frac{dv}{dt} \right\} \cdot \delta y + \left\{ \frac{dV}{dt} \right\} \cdot \delta z \right\} \cdot dt$$

is the differential of  $d\phi$ , on the supposition that the time only varies. Consequently, we are not obliged to determine  $\phi$  in  $x, y, z$ , in order to know whether it is an exact differential or not.  $\therefore$  It appears that if  $u \delta x + v \delta y + V \delta z$  be an exact variation, at the subsequent instant its increment will be an exact variation,  $\therefore \delta \phi +$  this increment will be an exact variation. As in general we know the condition of the fluid at the commencement of the motion, if at this moment  $u \delta x + v \delta y + V \delta z$  is an exact variation, it will be an exact variation when  $t = \pm dt$ ,  $t = \pm 2dt$ , &c. and in general whatever may the value of  $t$ .  $u \delta x + v \delta y + V \delta z$  will be an exact variation, if when  $t = 0$ , the fluid either has no velocity or a constant one, for in first case  $u = 0, v = 0, V = 0$  when  $t$  vanishes,  $\therefore u \delta x + v \delta y + V \delta z$  will be integrable for this moment, the second case will obtain when the motion is produced by an impulse on the surface of the fluid, such as that which arises from the action of a piston. For the velocities  $u, v, V$  which are communicated to each of the molecules, must be such, that if they are

is an exact variation at the first moment, but the equation (*H*) gives at this moment

$$\left\{ \frac{du}{dt} \right\} \cdot \delta x + \left\{ \frac{dv}{dt} \right\} \cdot \delta y + \left\{ \frac{dV}{dt} \right\} \cdot \delta z \\ = \delta V - \frac{1}{2} \delta \cdot \left\{ \left\{ \frac{d\varphi}{db} \right\}^2 + \left\{ \frac{d\varphi}{dy} \right\}^2 + \left\{ \frac{d\varphi}{dz} \right\}^2 \right\};$$

consequently the first member of this equation is an exact variation of  $x, y, z$ ; therefore if the function  $u\delta x + v\delta y + V\delta z$  be an exact variation at any one instant, it will be one in the next, therefore it will be an exact variation at all times.

When the motions are very small; the squares and products of  $u, v, V$ , may be neglected; and the equation (*H*) will then become

$$\delta V - \frac{\delta p}{\rho} = \left\{ \frac{du}{dt} \right\} \delta x + \left\{ \frac{dv}{dt} \right\} \delta y + \left\{ \frac{dV}{dt} \right\} \delta z; *$$

therefore in this case  $u\delta x + v\delta y + V\delta z$ , is an exact variation, provided that, as we have supposed,  $p$  is a function of  $\rho$ ; therefore if we designate

destroyed by impressing on each molecule, equal velocities in an opposite direction the *entire* fluid would quiesce; ∴ in consequence of the primitive impulsion, and the velocities  $u, v, V$ , applied in an opposite direction, there must be an equilibrium, ∴  $u v V$  must be such that  $u\delta x + v\delta y + V\delta z$  may be an exact variation, see No. 17; it appears from what precedes, that the integrability of the equation (*H*), and the consequent determination of  $p, \epsilon, u, v, V$ , depends on the nature of the velocities, communicated to the molecules at the commencement of the motion.

\* In the equation (*H*)  $u, v, V$ , are very small quantities, and in like manner

$$\left\{ \frac{du}{dx} \right\} \left\{ \frac{du}{dy} \right\} \left\{ \frac{dv}{dy} \right\} \left\{ \frac{dV}{dz} \right\} \text{ &c.}$$

∴ their product may be rejected ∴ naming this variation  $\delta\varphi$  we have as before,

$$\frac{d\delta\varphi}{dt} = \delta \frac{d\varphi}{dt} = \left\{ \left( \frac{du}{dt} \right) \cdot \delta x + \left( \frac{dv}{dt} \right) \cdot \delta y + \left( \frac{dV}{dt} \right) \cdot \delta z \right\}; \\ = \delta V - \frac{\delta p}{\epsilon}, \therefore \frac{d\varphi}{dt} = V - \int \frac{\delta p}{\epsilon}$$

this variation by  $\delta\phi$ , we shall have

$$V - \int \frac{\delta p}{\rho} = \left\{ \frac{d\phi}{dt} \right\};$$

and if the fluid be homogenous, the equation of continuity will become

$$0 = \left\{ \frac{d^2\phi}{dx^2} \right\} + \left\{ \frac{d^2\phi}{dy^2} \right\} + \left\{ \frac{d^2\phi}{dz^2} \right\};$$

$$= \frac{d\phi}{dt} + \frac{u^2 + v^2 + V^2}{2};$$

the expression

$$\left( \frac{d\phi}{dt} \right) + \frac{1}{2} \cdot \left\{ \left( \frac{d\phi}{dx} \right)^2 + \left( \frac{d\phi}{dy} \right)^2 + \left( \frac{d\phi}{dz} \right)^2 \right\}$$

is the value of  $V - \int \frac{\delta p}{\rho}$ , when  $u\delta x + v\delta y + V\delta z$  is an exact variation, it is reduced to its first term when  $u, v, V$ , are very small quantities.

However though the form of these equations is comparatively so much simpler, than the *general equations* which have been given in page 232, still the determination of the *laws* of the small oscillations of the waves of the sea, is yet a desideratum in Physics. Philosophers have been much more successful in investigating the oscillations of the pulses of the air, and in the determination of the velocity of the propagation of sound.

The integration of

$$\left\{ \frac{d^2\phi}{dx^2} \right\} + \left\{ \frac{d^2\phi}{dy^2} \right\} + \left\{ \frac{d^2\phi}{dz^2} \right\}$$

which is the equation relative to the continuity of the fluid, when  $u\delta x + v\delta y + V\delta z$  is an exact variation, and when the fluid is homogenous, which is consequently the simplest possible form, is extremely difficult, however it has been completed effected by Antonie Parseval.

these two equations contain the entire theory, of the very small undulations of homogeneous fluids.\*

## II

\* If the fluid which makes small oscillations be water, by making the axis of  $z$  vertical,  $R\partial z = g\cdot \partial z$ ,  $g$  representing the force of gravity,  $Pdx$ ,  $Qdy$  are respectively to nothing, in like manner we may conceive it to be homogeneous and incompressible, consequently we shall have

$$\int \frac{\partial p}{\epsilon} = \frac{p}{\epsilon}, \therefore \partial V - \frac{\partial p}{\epsilon} = \partial \left( \frac{d\phi}{dt} \right) = g \cdot \partial z - \frac{\partial p}{\epsilon}, \text{ & } gz - \frac{p}{\epsilon} = \left( \frac{d\phi}{dt} \right);$$

at the surface  $p$  vanishes,  $\therefore z = \frac{1}{g} \left\{ \frac{d\phi}{dt} \right\}$ , consequently when the form of  $\phi$  is determined, we can derive the equation of the part of the fluid in which  $p=0$ , i.e., the equation of the surface of the fluid.

We determine  $\phi$  as was already observed by means of the equation

$$\left\{ \frac{d^2\phi}{dx^2} \right\} + \left\{ \frac{d^2\phi}{dy^2} \right\} + \left\{ \frac{d^2\phi}{dz^2} \right\} = 0,$$

For, elastic fluids or those whose density varies,  $p=\epsilon \rho$ , and if  $(\epsilon)$  the density of the fluid in a state of rest, becomes in a state of motion equal to  $(\epsilon) + (\epsilon) \cdot q$ ,  $q$  being a very small quantity,  $\rho$  will be equal to  $(\epsilon) + (\epsilon) \cdot q$ , the oscillations being supposed very small,  $\partial V - \frac{\partial p}{\epsilon} = \partial \left( \frac{d\phi}{dt} \right)$  will become  $\partial V - \epsilon \cdot \frac{\partial \rho}{\epsilon} = \partial \left( \frac{d\phi}{dt} \right)$ , the only force acting being that of gravity, and the motion being supposed parallel to the horizon,  $\partial V$  will vanish and the equation will become  $- \frac{\epsilon \cdot \partial \rho}{\epsilon} = \partial \left( \frac{d\phi}{dt} \right)$  = by substituting for  $\rho$  its value,  $(\epsilon)$  being supposed constant,  $- \frac{\epsilon(\epsilon) \cdot \partial q}{(\epsilon) \cdot q}$ ;  $\therefore -\epsilon \cdot \log. q = \left\{ \frac{d\phi}{dt} \right\}$ , the equation relative to the continuity of the fluid will become

$$0 = \left\{ \frac{d(\epsilon)q}{dt} \right\} + \left\{ \frac{d(\epsilon)q}{dx} \right\} \cdot \left\{ \frac{d\phi}{dx} \right\} + \epsilon \left\{ \frac{d^2\phi}{dx^2} \right\}; \text{ for } \left\{ \frac{d\phi}{dy} \right\} \left\{ \frac{d\phi}{dz} \right\},$$

vanish, the motion being supposed to be performed in a direction parallel to the axis of  $x$ , and

34. Let us consider an homogeneous fluid mass which revolves uniformly about the axis of  $x$ .  $n$  representing the angular velocity of rotation, at a distance from the axis equal to unity, we shall have  $v = -nz$ ;  $V = ny$ ; \* consequently the equation ( $H$ ) of the preceding number, will become

$$\frac{\partial p}{\rho} = \delta V + n^2 \cdot \{ y \delta y + z \delta z \}; \dagger$$

consequently the velocities  $v, V$ , = respectively

$$\left( \frac{d\phi}{dz} \right), \left( \frac{d\phi}{dx} \right), \text{vanish}; \left( \frac{d(\epsilon)q}{dx} \right) \cdot \frac{d\phi}{dx} = (\epsilon) \cdot \left( \frac{dq}{dx} \right) u,$$

which is a quantity indefinitely small of the second order,  $\therefore$  it may be neglected, consequently the preceding equation becomes

$$(\epsilon) \left\{ \left( \frac{dq}{dt} \right) + (1+q) \left( \frac{d^2\phi}{dx^2} \right) \right\} = 0, \text{ but } \frac{d\phi}{dt} = -\epsilon, \log. q \therefore \frac{d^2\phi}{dt^2} = -\epsilon \left\{ \frac{dq}{q \cdot dt} \right\}$$

$$\text{consequently } \frac{q}{\epsilon} \cdot \left\{ \frac{d^2\phi}{dt^2} \right\} + (1+q) \cdot \left\{ \frac{d^2\phi}{dx^2} \right\} = 0;$$

this equation is of great celebrity in the history of the integral calculus, it was first integrated by D'Alembert, in an analysis of the problem of the vibrating chord, which leads to an equation of precisely the same form.

\* The linear velocity is equal to the angular velocity multiplied into the distance,  $\therefore$  at a distance represented by unity, the linear velocity =  $n$ , and since the angular velocity at all distances from the axis is the same, at a distance =  $\sqrt{z^2 + y^2}$  the linear velocity =  $n \cdot \sqrt{z^2 + y^2}$ , the direction of the motion being perpendicular to the radius in order to obtain the velocity parallel to the coordinates  $z, y$ , we should multiply  $n \cdot \sqrt{z^2 + y^2}$  into the cosines of the angles which  $z$  and  $y$  make with the tangent, but these cosines are respectively  $\frac{y}{\sqrt{z^2 + y^2}}$ ,  $\frac{-z}{\sqrt{z^2 + y^2}}$ , for the motion being circular, if one of the coordinates be increased, the other will be diminished  $\therefore v = -nz$ ,  $V = ny$ .

† The terms corresponding to  $\left\{ \frac{du}{dt} \right\}$ ,  $\left\{ \frac{dv}{dt} \right\}$ ,  $\left\{ \frac{dV}{dt} \right\}$ , in the equation ( $H$ ) vanish, because the time does not enter into the values of  $u, v, V$ , in like manner  $u$  and its differential coefficients vanish, and from the values of  $v, V$ , given above, it is manifest that

which equation is possible, because its two members are exact variations. The equation (*K*) of the same number will become

$$0 = dt \cdot \left\{ \frac{dp}{dt} \right\} + u \cdot dt \cdot \left\{ \frac{dp}{dx} \right\} + v \cdot dt \cdot \left\{ \frac{dp}{dy} \right\} + V \cdot dt \cdot \left\{ \frac{dp}{dz} \right\}; *$$

and it is manifest that this equation will be satisfied, if the fluid mass be homogeneous. The equations of the motion of fluids will therefore be satisfied, and consequently, the motion is possible.

The centrifugal force at the distance  $\sqrt{y^2+z^2}$  from the axis of rotation, is equal to the square  $n^2(z^2+y^2)$  of the velocity, divided by this distance; therefore the function  $n^2(y\delta y+z\delta z)$  † is the product of the

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$\left( \frac{dv}{dy} \right), \left( \frac{dV}{dz} \right)$ , are equal respectively to nothing, consequently the only terms which have a finite value are  $V \cdot \left( \frac{dv}{dz} \right), v \cdot \left( \frac{dV}{dy} \right)$ , which are respectively equal to  $= -n^2 y, -n^2 z$ , ∵ the equation (*H*) will become  $\frac{\delta p}{\epsilon} = \delta V + n^2(y\delta z + z\delta y)$ , this equation determines the pressure when  $\epsilon$  is constant, or when it is a function of  $p$ .

\* The equation (*K*) is resolvable into two parts as before,

$$\left( \frac{de}{dt} \right) + u \cdot \left( \frac{de}{dx} \right) + v \cdot \left( \frac{de}{dy} \right) + V \cdot \left( \frac{de}{dz} \right) + \epsilon \cdot \left\{ \left( \frac{du}{dx} \right) + \left( \frac{dv}{dy} \right) + \left( \frac{dV}{dz} \right) \right\},$$

the velocity being uniform, its increment resolved parallel to the axes of  $x, y, z, i, e$

$$\left( \frac{du}{dx} \right), \left( \frac{dv}{dy} \right), \left( \frac{dV}{dz} \right),$$

must be severally equal to nothing, this is evident for  $v, V$ , from their values which have been given above, with respect to the velocity  $u$ , it must be produced by the part of the velocity which is parallel to  $x$ , and if it was not uniform, the fluid would not have a uniform motion of rotation about the axis of  $x$ .

† The centrifugal force  $= n^2 \cdot \sqrt{z^2+y^2}$ , the variation of the distance  $= \frac{z\delta z+y\delta y}{\sqrt{z^2+y^2}}$   
 $\therefore n^2(z\delta z+y\delta y)$  is = to the centrifugal force multiplied into the element of the distance.

centrifugal force, by the element of its direction ; thus, if we compare the preceding equation of the motion of a fluid, with the general equation of the equilibrium of fluids, which has been given in No. 17, we may perceive that the conditions of the motion are reduced, to those of the equilibrium of the fluid mass, solicited by the same forces, and by the centrifugal force which arises from the motion of rotation ; which is sufficiently evident from the nature of the case.

If the exterior surface of the fluid mass be free, we shall have  $\delta p=0$ , at this surface, and consequently

$$0 = \delta V + n^2 \cdot \{ y\delta y + z\delta z \} ; *$$

Substituting for  $\delta V$  we obtain  $\frac{\delta p}{\epsilon} = P.\delta x + Q.\delta y + R.\delta z + n^2.y\delta y + n^2.z\delta z$ , the quantity added i, e, the centrifugal force multiplied into the element of distance, being an exact variation, it follows that the expression for  $\frac{\delta p}{\epsilon}$  will in this case be an exact variation,  $n$  is some function of the distance of the molecules from the axis of rotation, as the *time* is not involved in the preceding equation, it follows that the conditions of the motion of a fluid mass, about an axis, with a given velocity, are the same as the conditions of equilibrium of a fluid mass, the same forces as before soliciting the molecules, combined with the centrifugal force, arising from the uniform revolution about the axis. The molecules of the fluid, though they have a motion about an axis, are relatively at rest.

\* At the exterior free surface  $\delta p=0$ ,  $\therefore \delta V + n^2(y\delta y + z\delta z) = 0$ ,  $\therefore$  in order that the form of the fluid, may remain the same, during the entire motion,  $n$  must be constant. If the fluid was water contained in a vessel open at its upper surface,  $\epsilon$  is constant, and  $\delta V = g.\delta x$ , the axis of rotation being supposed vertical,  $\therefore Q.\delta y, R\delta z$  vanish, and  $P=g$ , consequently, we shall have  $\frac{P}{\epsilon} = -gx + n^2 \cdot \left( \frac{z^2 + y^2}{2} \right) + h$  and at the free surface, we have  $x = n^2 \cdot \left( \frac{z^2 + y^2}{2g} \right) + \frac{h}{g}$  for the equation of this surface; if  $n^2 \cdot \sqrt{z^2 + y^2}$  which expresses the centrifugal force varied at the  $2r-1$  power of the of the distance from the axis of rotation *i, e*, as

$$\begin{aligned} & (z^2 + y^2); n^2 = a^2 \cdot (z^2 + y^2)^{\frac{r-1}{2}}, \text{ and } n^2 \cdot (y\delta y + z\delta z) \\ & = a^2 \left( \frac{z^2 + y^2}{4r} \right)^r, \therefore x = a^2 \left( \frac{z^2 + y^2}{2r.g} \right)^r + \frac{h}{g} \end{aligned}$$

from which it follows that the resultant of all the forces which actuate each molecule, must be perpendicular to this surface, moreover it must be directed towards the interior of the fluid mass. If these conditions be satisfied, an homogeneous fluid mass will be in equilibrio, whatever may be the figure of the solid, which it covers.

The case which we have discussed, is one of those in which the variation  $u\delta x + v\delta y + V\delta z$  \* is not exact; for then this variation becomes

$\therefore$  if  $r$  is positive,  $x$  is least, when  $(z^2+y^2)=0$ , when  $r=1$  all the molecules revolve in the same time, and  $x=a^2 \cdot \left(\frac{z^2+y^2}{2g}\right) + \frac{h}{g}$  which is the equation of the concave surface of the paraboloid, of which the parameter  $= \frac{2g}{a^2}$ , the periodic time being equal to the force divided by the distance  $= \frac{\pi}{a}$ .  $\therefore$  if the time of revolution, be called  $T$ , we shall have the parameter of the generating curve  $= \frac{2g}{\pi^2} T^2 & \frac{p}{\xi} = \frac{\pi^2(z^2+y^2)}{2T^2} + h - gx$   $\therefore x$  being the same, the pressure is greater at a greater distance from the axis of rotation.

When  $r$  is negative, at the point where  $z^2+y^2=0$ ,  $x$  is infinite, and when  $=-\frac{1}{2}$  the surface of the fluid will be such, as would be generated by the revolution of a conical hyperbola, about its asymptote, the axis of  $x$  is in this case the asymptote. The constant quantity  $h$  denotes the distance of the origin of the coordinates from the other asymptote,  $\therefore$  both in this case and where the surface of the fluid is paraboloidal, the constant quantity depends on the quantity of water in the vessel. If the vessel was cylindrical, we could determine the area of the paraboloid, provided that we knew the area of the base of the cylinder, and also the points of greatest elevation and depression, for the paraboloid is half the circumscribing cylinder.

This paraboloidal figure is that which is assumed by the molecules of the fluid, in the experiment which Newton adduces, in order to shew that the effects by which *absolute* and *relative* motions are distinguished from each other, are the forces of receding from the axis of circular motion. See Princip. Math. page 10.

\*  $u\delta x + v\delta y + V\delta z$  is not an exact variation in the preceding investigation, for substituting for  $v$ , and  $V$ , we obtain  $v=-nz, V=ny, \therefore u\delta x + v\delta y + V\delta z = n(y\delta z - z\delta y)$ , consequently it appears, that though the circumstance of the preceding expression being an exact variation, would facilitate very much, our investigations, still it is not *essentially* necessary, that this should be the case, in order that the motion should be *possible*.  $\therefore$  Since in the case of the sea, revolving round with the earth round its axis, and relatively quiescing with respect to the

$-n\{zdy - ydz\}$ ; therefore in the theory of the flux and reflux of the sea, we are not permitted to assume, that the variation concerned is exact; since it is not so in the very simple case, in which the sea has no other motion, but that of rotation, which is common to it, and the earth.

35. Let us now determine the oscillations of a fluid mass which covers a spheroid revolving about the axis of  $x$ ; and let us suppose that it is deranged from the position of equilibrium, by the action of very small forces.

At the commencement of the motion, let  $r$  represent the distance of a molecule of the fluid, \* from the centre of gravity of the spheroid over which it is spread, and which we shall suppose immovable; let  $\theta$  be the angle which the radius  $r$  makes with the axis of  $x$ , and  $\varpi$  the angle which the plane passing through the axis of  $x$  and the radius  $r$ , constitutes with the plane of  $x$  and of  $y$ . Let us suppose that after the time  $t$ , the radius  $r$  is changed into  $r + \alpha s$ , that the angle  $\theta$  is changed into  $\theta + \alpha u$ , and finally, that the angle  $\varpi$  is changed into  $nt + \varpi + \alpha v$ ;  $\alpha s$ ,  $\alpha u$ , and  $\alpha v$ , being very small quantities, of which the squares and products may be neglected, we shall have

$$x = (r + \alpha s) \cdot \cos. (\theta + \alpha u);$$

$$y = (r + \alpha s) \cdot \sin. (\theta + \alpha u) \cdot \cos. (nt + \varpi + \alpha v);$$

$$z = (r + \alpha s) \cdot \sin. (\theta + \alpha u) \cdot \sin. (nt + \varpi + \alpha v).$$

earth,  $u\partial x + v\partial y + V\partial z$  is not an exact variation, we may conclude a *a fortiori*, that it is not one, where the oscillations arise from the attractions of the sun and moon, which produce the flux and reflux of the sea.

In order to ascertain whether an incompressible fluid solicited by accelerating forces, and also by a centrifugal force, may be at the surface of a given figure of revolution, we substitute in the equation  $0 = \partial V + n^2(y\partial y + z\partial z)$  the forces parallel to  $x$ ,  $y$ ,  $z$ , which would result from this hypothesis, the resulting expression should be the differential equation of the given surface, if it is not, then we may be certain that the given curve does not satisfy the equilibrium of the fluid. See Book 3. Chap III. No. 18.

\* If a perpendicular is let fall from the extremity of  $r$  on the axis of  $x$ , it will be equal to  $r \cdot \sin. \theta$ , and the projection of this perpendicular on the plane of  $y, x$ , is equal to the coordinate  $y$  and its value will be  $r \cdot \sin. \theta \cdot \cos. \varpi$ , and this perpendicular projected on the plane  $z, x$  will be the coordinate  $z$ , and it will be equal to  $r \cdot \sin. \theta \cdot \sin. \varpi$ .

Substituting these values in the equation (*F*) of No. 32, we shall obtain, the square of  $\alpha$  being neglected, \*

\* Since  $\alpha u$ ,  $\alpha \theta$ ,  $\alpha \omega$ , are very small quantities, of which the squares and products may be neglected, the time  $t$  will of the same order as  $\alpha$ , so that  $\alpha t$  is of the order  $\alpha^2$ , consequently

$$\sin. \alpha u = \alpha u - \frac{\alpha^3 u^3}{1.2.3} \&c. = \alpha u, \cos. \alpha u = 1 - \frac{\alpha^2 u^2}{2} = 1 \therefore x = (r + \alpha s) \cdot \cos. (\theta + \alpha u)$$

$$= r \cdot \cos. \theta \cdot \cos. \alpha u - r \cdot \sin. \theta \cdot \sin. \alpha u + \alpha s \cdot \cos. \theta \cdot \cos. \alpha u - \alpha s \cdot \sin. \theta \cdot \sin. \alpha u \\ = \text{by neglecting quantities of the order } \alpha^2, r \cdot \cos. \theta - r \cdot \sin. \theta \cdot \alpha u + \alpha s \cdot \cos. \theta,$$

$r$  and  $\theta$  are independent of  $t$ ,

$$\therefore \frac{dx}{dt} = -\frac{du}{dt} \cdot \alpha r \cdot \sin. \theta + \frac{ds}{dt} \alpha \cdot \cos. \theta; \frac{d^2 x}{dt^2} = -\frac{d^2 u}{dt^2} \alpha r \cdot \sin. \theta + \frac{d^2 s}{dt^2} \alpha \cos. \theta,$$

$$dx = dr \cdot \cos. \theta - d\theta \cdot r \cdot \sin. \theta - dr \cdot \sin. \theta \cdot \alpha u - d\theta \cdot \cos. \theta \cdot r \alpha u - d\theta \cdot \sin. \theta \cdot \alpha s; \quad dx \left( \frac{d^2 x}{dt^2} \right)$$

$$= -dr \cdot r \alpha \cdot \sin. \theta \cdot \cos. \theta \cdot \frac{d^2 u}{dt^2} + dr \cdot \alpha \cdot \cos. \theta \cdot \frac{d^2 s}{dt^2} + d\theta \cdot r^2 \alpha \cdot \sin. \theta \cdot \frac{d^2 u}{dt^2} - d\theta \cdot r \alpha \cdot \sin. \theta \cdot \cos. \theta \cdot \frac{d^2 s}{dt^2},$$

rejecting quantities involving  $\alpha^2$  &c.:

$$y = (r + \alpha s) \cdot \sin. (\theta + \alpha u) \cdot \cos. (nt + \omega + \alpha v) = r \cdot \sin. (\theta + \alpha u) \cdot \cos. (nt + \omega + \alpha v) \\ + \alpha s \cdot \sin. (\theta + \alpha u) \cdot \cos. (nt + \omega + \alpha v) = r \cdot \sin. \theta \cdot \cos. (nt + \omega) - r \cdot \sin. \theta \cdot \sin. (\omega + nt) \alpha v \\ + \alpha u \cdot r \cdot \cos. \theta \cdot \cos. (\omega + nt) + \alpha s \cdot \sin. \theta \cdot \cos. (\omega + nt)$$

rejecting as before quantities of the order  $\alpha^2$ , substituting  $\alpha u$ ,  $\alpha v$ , for  $\sin. \alpha u$ ,  $\sin. \alpha v$ , and observing that  $\alpha t$  is of the order  $\alpha^2$ ,  $\therefore y =$

$$r \cdot \sin. \theta \cdot \cos. \omega - nr \cdot \sin. \theta \cdot \sin. \omega - r \cdot \sin. \theta \cdot \sin. \omega \alpha v - nr \cdot \alpha v \cdot \sin. \theta \cdot \cos. \omega + \alpha u r \cdot \cos. \theta \cdot \cos. \omega \\ - \alpha u r n \cdot \cos. \theta \cdot \sin. \omega + \alpha s \cdot \sin. \theta \cdot \cos. \omega - \alpha s n \cdot \sin. \theta \cdot \sin. \omega;$$

$$\frac{dy}{dt} = -nr \cdot \sin. \theta \cdot \sin. \omega - r \cdot \sin. \theta \cdot \sin. \omega \alpha v - nr \cdot \alpha v \cdot \sin. \theta \cdot \cos. \omega - nr \alpha \cdot \frac{dv}{dt} \cdot \sin. \theta \cdot \cos. \omega \\ - \alpha r \cdot \cos. \theta \cdot \cos. \omega \cdot \frac{du}{dt} - \alpha u r n \cdot \cos. \theta \cdot \sin. \omega - \alpha s n \cdot \frac{du}{dt} \cdot \cos. \theta \cdot \sin. \omega$$

$$\begin{aligned}
& \alpha r^2 \delta \theta \cdot \left\{ \left( \frac{d^2 u}{dt^2} \right) - 2n \cdot \sin \theta \cdot \cos \theta \cdot \left( \frac{dv}{dt} \right) \right\} \\
& + \alpha r^2 \delta \varpi \cdot \left\{ \sin^2 \theta \cdot \left( \frac{d^2 v}{dt^2} \right) + 2n \cdot \sin \theta \cdot \cos \theta \cdot \left( \frac{du}{dt} \right) + \frac{2n \cdot \sin^2 \theta}{r} \cdot \left( \frac{ds}{dt} \right) \right\}; (L) \\
& + \alpha \cdot \delta r \cdot \left\{ \left( \frac{d^2 s}{dt^2} \right) - 2nr \cdot \sin^2 \theta \cdot \left( \frac{dv}{dt} \right) \right\} \\
& = \frac{n^2}{2} \cdot \delta \cdot \left\{ (r + \alpha s) \cdot \sin (\theta + \alpha u) \right\}^2 + \delta V - \frac{\delta p}{\epsilon}
\end{aligned}$$

$$+ \alpha \cdot \frac{ds}{dt} \cdot \sin \theta \cdot \cos \varpi - \alpha s \cdot \sin \theta \cdot \sin \varpi - \alpha n \cdot \sin \theta \cdot \sin \varpi \cdot \frac{ds}{dt}$$

$$\begin{aligned}
\frac{d^2 y}{dt^2} &= -\alpha r \cdot \sin \theta \cdot \sin \varpi \cdot \frac{d^2 v}{dt^2} - 2nr \cdot \alpha \cdot \sin \theta \cdot \cos \varpi \cdot \frac{dv}{dt} + \alpha r \cdot \cos \theta \cdot \cos \varpi \cdot \frac{d^2 u}{dt^2} \\
&- 2\alpha n \cdot \cos \theta \cdot \sin \varpi \cdot \frac{du}{dt} + \alpha \cdot \sin \theta \cdot \cos \varpi \cdot \frac{d^2 s}{dt^2} - 2\alpha n \cdot \sin \theta \cdot \sin \varpi \cdot \frac{ds}{dt}
\end{aligned}$$

$$\delta y = \delta r \cdot \sin \theta \cdot \cos \varpi + \delta \theta \cdot r \cdot \cos \theta \cdot \cos \varpi - \delta \varpi \cdot r \cdot \sin \theta \cdot \sin \varpi,$$

rejecting those quantities in the value of  $\delta y$ , where  $\alpha$  occurs, for in the product of the expression for  $\frac{d^2 y}{dt^2}$  into the value of  $\delta y$ , these would be of the order  $\alpha^2$ ,  $\therefore$  they ought to be neglected;

$$\begin{aligned}
\therefore \delta y &= \delta r \cdot (-\alpha r \cdot \sin^2 \theta \cdot \sin \varpi \cdot \cos \varpi) \frac{d^2 v}{dt^2} - 2nr \alpha \cdot \sin^2 \theta \cdot \cos^2 \varpi \cdot \frac{dv}{dt} \\
&+ \alpha r \cdot \sin \theta \cdot \cos \theta \cdot \cos^2 \varpi \cdot \frac{d^2 u}{dt^2} - 2\alpha n \cdot \sin \theta \cdot \cos \theta \cdot \sin \varpi \cdot \cos \varpi \cdot \frac{du}{dt} \\
&+ \alpha \cdot \sin^2 \theta \cdot \cos^2 \varpi \cdot \frac{d^2 s}{dt^2} - 2\alpha n \cdot (\sin^2 \theta \cdot \sin \varpi \cdot \cos \varpi) \cdot \frac{ds}{dt} \\
&\delta \theta (-\alpha r^2 \sin \theta \cdot \cos \theta \cdot \sin \varpi \cdot \cos \varpi) \frac{d^2 v}{dt^2} - 2nr^2 \alpha \cdot \sin \theta \cdot \cos \theta \cdot \cos^2 \varpi \cdot \frac{dv}{dt} \\
&+ \alpha r^2 \cdot \cos^2 \theta \cdot \cos^2 \varpi \cdot \frac{d^2 u}{dt^2} - 2\alpha n r^2 \cdot \cos^2 \theta \cdot \sin \varpi \cdot \cos \varpi \cdot \frac{du}{dt} + \alpha r \cdot \sin \theta \cdot \cos \theta \cdot \cos^2 \varpi \cdot \frac{d^2 s}{dt^2}
\end{aligned}$$

At the exterior surface of the fluid we have  $\delta p=0$ ; moreover in the state of equilibrium,

$$0 = \frac{n^2}{2} \delta \cdot \{(r + \alpha s) \cdot \sin(\theta + \alpha u)\}^2 + (\delta V);$$

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$$\begin{aligned}
& -2\alpha nr \cdot \sin \theta \cdot \cos \theta \cdot \sin \varpi \cdot \cos \varpi \cdot \frac{ds}{dt} \Big) + \delta \varpi (\alpha r^2 \cdot \sin^2 \theta \sin^2 \varpi \cdot \frac{d^2 v}{dt^2} \\
& + 2nr^2 \alpha \cdot \sin^2 \theta \sin \varpi \cdot \cos \varpi \cdot \frac{dv}{dt} - \alpha r^2 \cdot \sin \theta \cdot \cos \theta \cdot \sin \varpi \cdot \cos \varpi \cdot \frac{d^2 u}{dt^2} \\
& + 2\alpha r^2 n \cdot \sin \theta \cdot \cos \theta \cdot \sin^2 \varpi \cdot \frac{du}{dt} - \alpha r \cdot \sin^2 \theta \cdot \sin \varpi \cdot \cos \varpi \cdot \frac{d^2 s}{dt^2} + 2\alpha nr \cdot \sin^2 \theta \cdot \sin^2 \varpi \cdot \frac{ds}{dt}) \\
& (r + \alpha s) \sin(\theta + \alpha u) \cdot \sin(nt + \varpi + \alpha v) = (r + \alpha s) \sin \theta \cdot \sin(nt + \varpi + \alpha v) + \alpha ur \cdot \cos \theta \cdot \sin(nt + \varpi \\
& + \alpha v) = r \cdot \sin \theta \cdot \sin(nt + \varpi) + r \cdot \sin \theta \cdot \cos(nt + \varpi) \alpha v + \alpha ur \cdot \cos \theta \cdot \sin \theta \cdot \sin(nt + \varpi) \\
& \quad \alpha s \cdot \sin(\theta + \alpha u) \cdot \sin(nt + \varpi + \alpha v) (= \alpha s \cdot \sin \theta \cdot \sin(nt + \varpi)) \\
& = r \cdot \sin \theta \cdot \sin \varpi + ntr \cdot \sin \theta \cdot \cos \varpi + r \cdot \sin \theta \cdot \cos \varpi \cdot \alpha v - r \cdot \sin \theta \cdot \sin \varpi \cdot nt \cdot \alpha v \\
& + \alpha ur \cdot \cos \sin \theta \cdot \sin \varpi + \alpha ur \cdot \cos \theta \cdot \cos \varpi \cdot nt + \alpha s \cdot \sin \theta \cdot \sin \varpi + \alpha s \cdot \sin \theta \cdot \cos \varpi \cdot nt \therefore \frac{dz}{dt} = \\
& nr \cdot \sin \theta \cdot \cos \varpi + r \cdot \sin \theta \cdot \cos \varpi \cdot \alpha \frac{dv}{dt} - r \cdot \sin \theta \cdot \sin \varpi \cdot n \alpha v - r \cdot \sin \theta \cdot \sin \varpi \cdot nt \alpha \cdot \frac{dv}{dt} \\
& + \alpha r \cdot \cos \theta \cdot \sin \varpi \cdot \frac{du}{dt} + \alpha ur \cdot \cos \theta \cdot \cos \varpi \cdot n + \alpha r \cdot nt \cdot \cos \theta \cdot \cos \varpi \cdot \frac{du}{dt} + \alpha \cdot \sin \theta \cdot \sin \varpi \cdot \frac{ds}{dt} \\
& + \alpha \cdot \sin \theta \cdot \cos \varpi \cdot ns + \alpha \cdot \sin \theta \cdot \cos \varpi \cdot nt \cdot \frac{ds}{dt}; \frac{d^2 z}{dt^2} = r \alpha \cdot \sin \theta \cdot \cos \varpi \cdot \frac{d^2 v}{dt^2} \\
& - nr \alpha \cdot \sin \theta \cdot \sin \varpi \cdot \frac{dv}{dt} - nr \alpha \cdot \sin \theta \cdot \sin \varpi \cdot \frac{dv}{dt} + \alpha r \cdot \cos \theta \cdot \sin \varpi \cdot \frac{d^2 u}{dt^2} + \alpha nr \cdot \cos \theta \cdot \cos \varpi \cdot \frac{du}{dt} \\
& + \alpha nr \cdot \cos \theta \cdot \cos \varpi \cdot \frac{du}{dt} + \alpha \cdot \sin \theta \cdot \sin \varpi \cdot \frac{d^2 s}{dt^2} + \alpha n \cdot \sin \theta \cdot \cos \varpi \cdot \frac{ds}{dt} + \alpha n \cdot \sin \theta \cdot \cos \varpi \cdot \frac{ds}{dt}; \\
& \delta z = \delta r \cdot \sin \theta \cdot \sin \varpi + \delta \theta \cdot r \cdot \cos \theta \cdot \sin \varpi + \delta \varpi \cdot r \cdot \sin \theta \cdot \cos \varpi,
\end{aligned}$$

neglecting those terms which involve  $\alpha$ , for as was before mentioned, in the product

$(\delta V)$  being the value of  $\delta V$  which corresponds to this state. Let us suppose that the fluid in question, is the sea ; the variation  $(\delta V)$  will be the product of the gravity multiplied, into the element of its direction. Let  $g$  represent

$\delta z \cdot \frac{d^2z}{dt^2}$ , these quantities would produce terms of the order  $\alpha^2$  and would consequently be neglected.  $\therefore \delta z \cdot \frac{d^2z}{dt^2} =$

$$\delta r \cdot \left\{ r\alpha \cdot \sin^2\theta \cdot \sin \varpi \cdot \cos \varpi \cdot \frac{d^2v}{dt^2} - 2nr\alpha \cdot \sin^2\theta \cdot \sin^2\varpi \cdot \frac{dv}{dt} + \alpha r \cdot \sin \theta \cdot \cos \theta \cdot \sin^2\varpi \cdot \frac{d^2u}{dt^2} \right.$$

$$\left. + 2\alpha nr \cdot \sin \theta \cdot \cos \theta \cdot \sin \varpi \cdot \cos \varpi \cdot \frac{du}{dt} + \alpha \cdot \sin^2\theta \cdot \sin^2\varpi \cdot \frac{d^2s}{dt^2} + 2\alpha n \cdot \sin^2\theta \cdot \sin \varpi \cdot \cos \varpi \cdot \frac{ds}{dt} \right\}$$

$$+ \delta \theta \cdot \left( r^2 \alpha \sin \theta \cdot \cos \theta \cdot \sin \varpi \cdot \cos \varpi \cdot \frac{d^2v}{dt^2} - 2nr^2\alpha \cdot \sin \theta \cdot \cos \theta \cdot \sin^2\varpi \cdot \frac{dv}{dt} + \alpha r^2 \cdot \cos^2\theta \cdot \sin^2\varpi \cdot \frac{d^2u}{dt^2} \right.$$

$$\left. + 2\alpha nr^2 \cdot \cos^2\theta \cdot \sin \varpi \cdot \cos \varpi \cdot \frac{du}{dt} + \alpha r \cdot \sin \theta \cdot \cos \theta \cdot \sin^2\varpi \cdot \frac{d^2s}{dt^2} + 2\alpha nr \cdot \sin \theta \cdot \cos \theta \cdot \sin \varpi \cdot \cos \varpi \cdot \frac{ds}{dt} \right\}$$

$$\delta \varpi \cdot \left\{ r^2 \alpha \cdot \sin^2\theta \cdot \cos^2\varpi \cdot \frac{d^2v}{dt^2} - 2nr^2\alpha \cdot \sin^2\theta \cdot \sin \varpi \cdot \cos \varpi \cdot \frac{dv}{dt} + \alpha r^2 \cdot \sin \theta \cdot \cos \theta \cdot \sin \varpi \cdot \cos \varpi \cdot \frac{d^2u}{dt^2} \right.$$

$$\left. + 2\alpha nr^2 \cdot \sin \theta \cdot \cos \theta \cdot \cos^2\varpi \cdot \frac{du}{dt} + \alpha r \cdot \sin^2\theta \cdot \sin \varpi \cdot \cos \varpi \cdot \frac{d^2s}{dt^2} + 2\alpha nr \cdot \sin^2\theta \cdot \cos^2\varpi \cdot \frac{ds}{dt} \right\}$$

$$\therefore \delta x \cdot \frac{d^2x}{dt^2} + \delta y \cdot \frac{d^2y}{dt^2} + \delta z \cdot \frac{d^2z}{dt^2} =$$

$$\delta r \cdot \left\{ -r\alpha \cdot \sin \theta \cdot \cos \theta \cdot \frac{d^2u}{dt^2} + \alpha \cdot \cos^2\theta \cdot \frac{d^2s}{dt^2} - \alpha r \cdot \sin^2\theta \cdot \sin \varpi \cdot \cos \varpi \cdot \frac{d^2v}{dt^2} \right.$$

$$\left. - 2nr\alpha \cdot \sin^2\theta \cdot \cos^2\varpi \cdot \frac{dv}{dt} + \alpha r \cdot \sin \theta \cdot \cos \theta \cdot \cos^2\varpi \cdot \frac{d^2u}{dt^2} - 2\alpha nr \cdot \sin \theta \cdot \cos \theta \cdot \sin \varpi \cdot \cos \varpi \cdot \frac{du}{dt} \right.$$

$$\left. + \alpha \cdot \sin^2\theta \cdot \cos^2\varpi \cdot \frac{d^2s}{dt^2} - 2\alpha n \cdot \sin^2\theta \cdot \sin \varpi \cdot \cos \varpi \cdot \frac{ds}{dt} \right\}$$

$$+ r\alpha \cdot \sin^2\theta \cdot \sin \varpi \cdot \cos \varpi \cdot \frac{d^2v}{dt^2} - 2nr\alpha \cdot \sin^2\theta \cdot \sin^2\varpi \cdot \frac{dv}{dt} + \alpha r \cdot \sin \theta \cdot \cos \theta \cdot \sin^2\varpi \cdot \frac{d^2u}{dt^2}$$

$$+ 2\alpha nr \cdot \sin \theta \cdot \cos \theta \cdot \sin \varpi \cdot \cos \varpi \cdot \frac{du}{dt} + \alpha \cdot \sin^2\theta \cdot \sin^2\varpi \cdot \frac{d^2s}{dt^2} + 2\alpha n \cdot \sin^2\theta \cdot \sin \varpi \cdot \cos \varpi \cdot \frac{ds}{dt} \right\}$$

the force of gravity, and  $\alpha y$  the elevation of a molecule of water at its surface, above the surface of equilibrium, which surface we shall consider as the true level of the sea. The variation ( $\delta V$ ) in the state of motion, will in consequence of this elevation, be increased by the quan-

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(= by concinnating

$$\begin{aligned}
 & \alpha \delta r \cdot \left( \frac{d^2 s}{dt^2} - 2nr \cdot \sin^2 \theta \cdot \frac{dv}{dt} \right) + \delta \theta \cdot \left\{ r^2 \alpha \cdot \sin^2 \theta \cdot \frac{d^2 u}{dt^2} - r \alpha \cdot \sin \theta \cdot \cos \theta \cdot \frac{ds}{dt^2} \right. \\
 & \quad - \alpha r^2 \cdot \sin \theta \cdot \cos \theta \cdot \sin \pi \cdot \cos \pi \cdot \frac{d^2 v}{dt^2} - 2nr^2 \alpha \cdot \sin \theta \cdot \cos \theta \cdot \cos^2 \pi \cdot \frac{dv}{dt} \\
 & \quad + \alpha r^2 \cdot \cos^2 \theta \cdot \cos^2 \pi \cdot \frac{d^2 u}{dt^2} - 2\alpha r^2 n \cdot \cos^2 \theta \cdot \sin \pi \cdot \cos \pi \cdot \frac{du}{dt} + \alpha r \cdot \sin \theta \cdot \cos \theta \cdot \cos^2 \pi \cdot \frac{d^2 s}{dt^2} \\
 & \quad - 2\alpha nr \cdot \sin \theta \cdot \cos \theta \cdot \sin \pi \cdot \cos \pi \cdot \frac{ds}{dt} + r^2 \alpha \cdot \sin \theta \cdot \cos \theta \cdot \sin \pi \cdot \cos \pi \cdot \frac{d^2 v}{dt^2} \\
 & \quad - 2nr^2 \alpha \cdot \sin \theta \cdot \cos \theta \cdot \sin^2 \pi \cdot \frac{dv}{dt} + \alpha r^2 \cdot \cos^2 \theta \cdot \sin^2 \pi \cdot \frac{d^2 u}{dt^2} + 2\alpha nr^2 \cdot \cos^2 \theta \cdot \sin \pi \cdot \cos \pi \cdot \frac{du}{dt} \\
 & \quad \left. + \alpha r \cdot \sin \theta \cdot \cos \theta \cdot \sin^2 \pi \cdot \frac{d^2 s}{dt^2} + 2\alpha nr \cdot \sin \theta \cdot \cos \theta \cdot \sin \pi \cdot \cos \pi \cdot \frac{ds}{dt} \right\}
 \end{aligned}$$

(and by concinnating we obtain the coefficient of  $\delta \theta = 0$

$$\begin{aligned}
 & \left( r^2 \alpha \cdot \frac{d^2 u}{dt^2} - 2nr^2 \alpha \cdot \sin \theta \cdot \cos \theta \cdot \frac{dv}{dt} \right); \\
 & \delta \pi \cdot \left\{ \alpha r^2 \cdot \sin^2 \theta \cdot \sin^2 \pi \cdot \frac{d^2 v}{dt^2} + 2nr^2 \alpha \cdot \sin^2 \theta \cdot \sin \pi \cdot \cos \pi \cdot \frac{dv}{dt} - \alpha r^2 \cdot \sin \theta \cdot \cos \theta \cdot \sin \pi \cdot \cos \pi \cdot \frac{d^2 u}{dt^2} \right. \\
 & \quad + 2\alpha r^2 n \cdot \sin \theta \cdot \cos \theta \cdot \sin^2 \pi \cdot \frac{du}{dt} - \alpha r \cdot \sin^2 \theta \cdot \sin \pi \cdot \cos \pi \cdot \frac{d^2 s}{dt^2} + 2\alpha nr \cdot \sin^2 \theta \cdot \sin^2 \pi \cdot \frac{ds}{dt} \\
 & \quad + r\alpha^2 \cdot \sin^2 \theta \cdot \cos^2 \pi \cdot \frac{d^2 v}{dt^2} - 2nr^2 \alpha \cdot \sin^2 \theta \cdot \sin \pi \cdot \cos \pi \cdot \frac{dv}{dt} + \alpha r^2 \cdot \sin \theta \cdot \cos \theta \cdot \sin \pi \cdot \cos \pi \cdot \frac{d^2 u}{dt^2} \\
 & \quad \left. + 2\alpha nr^2 \cdot \sin \theta \cdot \cos \theta \cdot \cos^2 \pi \cdot \frac{du}{dt} + \alpha r \cdot \sin^2 \theta \cdot \sin \pi \cdot \cos \pi \cdot \frac{d^2 s}{dt^2} + 2\alpha nr \cdot \sin^2 \theta \cdot \cos^2 \pi \cdot \frac{ds}{dt} \right\}
 \end{aligned}$$

tity —  $\alpha g \cdot \delta y$ ; because the gravity is very nearly in the direction of  $\alpha y$ , and tends *towards* its origin; \* consequently, if we denote by  $\alpha \delta V$ , the part of  $\delta V$  relative to the new forces, which in the state of motion

concerning as before we obtain

$$\alpha \omega \left( \alpha r^2 \cdot \sin^2 \theta \cdot \frac{d^2 v}{dt^2} + 2\alpha r^2 n \cdot \sin \theta \cdot \cos \theta \cdot \frac{du}{dt} + 2\alpha nr \cdot \sin^2 \theta \cdot \frac{ds}{dt} \right);$$

the body having a rotatory motion about an axis, the part of the equation (H) which corresponds to the centrifugal force arising from the rotation is by the preceding number equal to  $n^2 (y \delta y + z \delta z) = \frac{n^2}{2} \cdot \delta (y^2 + z^2) = \frac{n^2}{2} \cdot \delta \left\{ (r + \alpha s) \cdot \sin (\theta + \alpha u) \right\}^2 \therefore$  the second numbers of the preceding equations, when concinnated, give the equation (L) of the text.

\* At the surface of the spheroid  $r = 1 + q l$ , in which  $l$  is for simplicity, considered as a function of  $\theta$  only, and the semi-axis minor = 1,  $\therefore \delta r = q \cdot \left( \frac{\partial l}{\partial \theta} \right) \alpha u$ ,  $q$  depends on the eccentricity,  $r$  receiving at the surface of the solid the increment  $\alpha s$ , the corresponding increment of  $\theta = \alpha u$ , therefore the expression for  $r$  will become  $1 + q l + \alpha u q \cdot \left( \frac{\partial l}{\partial \theta} \right)$ .  $\therefore \alpha s = \alpha u q \cdot \left( \frac{dl}{du} \right)$  and  $q$  being very small,  $s$  may neglected in comparison of  $u$ , and it is evidently of the order  $uq$ , i.e. of  $u$  multiplied into the eccentricity, and if  $l$  be considered as a function of  $\omega$  only, we might shew that  $\omega$  receiving an increment  $\alpha v$ , the corresponding increment of  $r$ , is to  $\alpha v$ , as the eccentricity multiplied into  $\left( \frac{dl}{d\omega} \right)$  is to unity. If we produce the radius  $r$  to the surface of the fluid in equilibrio; it will be represented by  $1 + q l + \gamma$ ,  $\gamma$  being the depth of the fluid, and a function of  $\theta$  and  $\omega$ ,  $\therefore \theta$  receiving the increment  $\alpha u$ , the corresponding increase of the radius, drawn to the surface of the fluid supposed in equilibrio, will be  $q \cdot \left( \frac{dl}{d\theta} \right) \cdot \alpha u + \left( \frac{d\gamma}{d\theta} \right) \cdot \alpha u$ ; when the fluid is in motion, the distance of the exterior surface from the centre, =  $r' + \alpha s'$ , is greater than the distance of the surface of equilibrium, from the centre of the spheroid, measured on the same radius, this last distance

$$= 1 + q l + \gamma + \alpha u \cdot \left( q \cdot \left( \frac{dl}{d\theta} \right) + \left( \frac{d\gamma}{d\theta} \right) \right) + \alpha v \cdot \left( q \cdot \left( \frac{dl}{d\omega} \right) + \left( \frac{d\gamma}{d\omega} \right) \right), r' + \alpha s' = 1 + q l + \gamma + \alpha s' \therefore \alpha \cdot \left( s' - u \cdot \left( q \cdot \left( \frac{dl}{d\theta} \right) + \left( \frac{d\gamma}{d\theta} \right) \right) + v \cdot q \cdot \left( \frac{dl}{d\omega} \right) + \left( \frac{d\gamma}{d\omega} \right) \right) = \alpha y$$

= the elevation of a molecule of water in the state of motion, above the surface of

agitate the molecule, and which arise either from the changes, which in the state of motion the attractions of the fluid and spheroid experience, or from the attractions of extraneous bodies ; we shall have at the surface,

$$\delta V = (\delta V) - \alpha g \cdot \delta y + \alpha \cdot \delta V'$$

The variation  $\frac{n^2}{2} \cdot \delta \{ (r + \alpha s) \cdot \sin(\theta + \alpha u) \}^2$  is increased by the quantity  $\alpha n^2 \cdot \delta y \cdot r \cdot \sin^2 \theta$ , \* in consequence of the elevation of the molecule of the water, above the level of the sea ; but this quantity may be neglected in comparison of the term  $- \alpha g \cdot \delta y$ , because the ratio  $\frac{n^2 \cdot r}{g}$  of the centrifugal force at the equator, to the gravity, is a very small fraction equal to  $\frac{1}{289}$ . Finally, the radius  $r$  is very nearly constant at the surface of the sea, because it differs very little from a spherical surface ; therefore we may make  $\delta r = 0$ . The equation ( $L$ ) will thus, become, at the surface of the sea,

$$r^2 \cdot \delta \theta \left\{ \left\{ \frac{d^2 u}{dt^2} \right\} - 2n \cdot \sin \theta \cdot \cos \theta \cdot \left\{ \frac{dv}{dt} \right\} \right\} \\ + r^2 \cdot \delta \pi \cdot \left\{ \sin^2 \theta \cdot \left\{ \frac{d^2 v}{dt^2} \right\} + 2n \cdot \sin \theta \cdot \cos \theta \cdot \left\{ \frac{du}{dt} \right\} + \frac{2n}{r} \sin^2 \theta \cdot \left\{ \frac{ds}{dt} \right\} \right\}$$

equilibrium ; it is evidently a function of  $\theta$  and  $\pi$ .  $q$  being the eccentricity, it is evident that the differential of the normal according to which the gravity acts, in case of equilibrium, differs from the differential of the radius, by a quantity which  $=$  the product of the eccentricity into the differential of  $N$ , a function of  $\theta$ . ∵ at the surface of the fluid in equilibrio,  $(\delta V) = g \cdot \delta \cdot (r' + q \cdot N)$ , at the surface of the fluid in motion, the normal corresponding to  $r' + \alpha y$ , has not the same direction as when in equilibrio, its variation  $= \delta \cdot (r' + qN + \alpha y + \alpha q \cdot qN)$  ; the attraction of the spheroid in motion differs from the attraction of the spheroid in equilibrio by quantities of the order  $\alpha y$  ∵ let it be equal to  $\alpha yg'$ , then  $(g + \alpha yg') \cdot \delta(r' + qN + \alpha y + \alpha q \cdot qN) = (g + \alpha yg') \cdot \delta(r + qN) + g \cdot \delta \alpha y$ , rejecting quantities of the order  $\alpha^2$ , and remarking that  $\delta r$  is of the order  $q \cdot \delta \theta$ , the first term of the second member of the preceding equation  $= (\delta V)$  ∵ the second term is the quantity by which in the state of motion  $(\delta V)$  is increased, as has been stated in the text.

$$= -g.\delta y + \delta V';$$

the variations  $\delta y$ , and  $\delta V'$ , being taken relatively to the two variables  $\theta$ , and  $\varpi$ .

Let us now, consider, the equation relative to the continuity of the fluid. For this purpose, let us conceive at the origin of the motion, a rectangular parallelepiped, of which the altitude is  $dr$ , the breadth  $r. d\varpi. \sin. \theta$ . and the length  $r.d\theta$ . \* Let  $r', \theta', \varpi'$ , represent what  $r, \theta, \varpi$ , become after the time  $t$ . By following the reasoning of No. 32, we shall find that after this interval, the volume of the molecule of the fluid, is equal to a rectangular parallelepiped, of which the height is  $\left\{ \frac{dr'}{dr} \right\}. dr$ ; of which the breadth is

$$r'. \sin'. \theta'. \left\{ \left\{ \frac{d\varpi'}{d\varpi} \right\}. d\varpi + \left\{ \frac{d\varpi'}{dr} \right\}. dr \right\},$$

$dr$  being eliminated, by means of the equation

$$0 = \left\{ \frac{dr'}{d\varpi} \right\}. d\varpi + \left\{ \frac{dr'}{dr} \right\}. dr;$$

Finally, its length is

$$r. \left\{ \left\{ \frac{d\theta'}{dr} \right\}. dr + \left\{ \frac{d\theta'}{d\theta} \right\}. d\theta + \left\{ \frac{d\theta'}{d\varpi} \right\}. d\varpi \right\}$$

\*  $r \sin. \theta$  = radius of a small circle, whose plane is parallel to the equator, and as the plane of the axes of  $x$ , and  $y$ , is fixed,  $r. \sin. \theta. d\varpi$  = the differential of the arc of this circle, to which  $dr$  is evidently perpendicular, also, the differential of the meridian =  $r.d\theta$ , is perpendicular both to  $r. \sin. \theta. d\varpi$  and to  $dr$ , ∴ these three differentials, constitute the parallelepiped mentioned in the text.

\* When the fluid is in motion, this expression becomes,  $\frac{n^2}{2} \delta(r + \alpha s + \alpha y) \sin.(\theta + \alpha u)^2$   
 $\therefore$  the part which corresponds to  $\alpha y$ , is  $n^2. \alpha \delta y. (r + \alpha s + \alpha y). \sin.(\theta + \alpha u)^2$  = by neglecting quantities of the order  $\alpha^2, n^2. \alpha \delta y. r. \sin. \theta^2$ .

$dr$ , and  $d\varpi$ , being eliminated by means of the equations

$$0 = \left\{ \frac{dr'}{dr} \right\} \cdot dr + \left\{ \frac{dr'}{d\theta} \right\} \cdot d\theta + \left\{ \frac{dr'}{d\varpi} \right\} \cdot d\varpi;$$

$$0 = \left\{ \frac{d\varpi'}{dr} \right\} \cdot dr + \left\{ \frac{d\varpi'}{d\theta} \right\} \cdot d\theta + \left\{ \frac{d\varpi'}{d\varpi} \right\} \cdot d\varpi.$$

Consequently, if we make

$$\begin{aligned} \epsilon' &= \left\{ \frac{dr'}{dr} \right\} \cdot \left\{ \frac{d\theta'}{d\theta} \right\} \cdot \left\{ \frac{d\varpi'}{d\varpi} \right\} - \left\{ \frac{dr'}{dr} \right\} \cdot \left\{ \frac{d\theta'}{d\varpi} \right\} \cdot \left\{ \frac{d\varpi'}{d\theta} \right\} \\ &\quad + \left\{ \frac{dr'}{d\theta} \right\} \cdot \left\{ \frac{d\theta'}{d\varpi} \right\} \cdot \left\{ \frac{d\varpi'}{dr} \right\} \\ &- \left\{ \frac{dr'}{d\theta} \right\} \cdot \left\{ \frac{d\theta'}{dr} \right\} \cdot \left\{ \frac{d\varpi'}{d\varpi} \right\} + \left\{ \frac{dr'}{d\varpi} \right\} \cdot \left\{ \frac{d\theta'}{dr} \right\} \cdot \left\{ \frac{d\varpi'}{d\theta} \right\} \\ &- \left\{ \frac{dr'}{d\varpi} \right\} \cdot \left\{ \frac{d\theta'}{d\theta} \right\} \cdot \left\{ \frac{d\varpi'}{dr} \right\}; \end{aligned}$$

after the time  $t$ , the volume of the parallelepiped will be equal to  $\epsilon' \cdot r^2 \cdot \sin \theta \cdot dr \cdot d\theta \cdot d\varpi$ ; \* therefore if  $(\rho)$  represent the primitive density of the molecule, and  $\rho$  its density, corresponding to the time  $t$ , we shall obtain, by putting the primitive value of its mass, equal to its value after the time  $t$ ,

$$\rho \cdot \epsilon' r^2 \cdot \sin \theta = (\rho) \cdot r^2 \cdot \sin \theta;$$

this is the equation relative to the continuity of the fluid. In the case we are at present considering,

$$r' = r + \alpha s; \quad \theta' = \theta + \alpha u; \quad \varpi' = nt + \varpi + \alpha v;$$

\*  $r'$ ,  $\theta'$ ,  $\varpi'$  are generally functions of  $r$ ,  $\theta$ ,  $\varpi$ , and  $t$ , see page 217, notes; the reasoning is precisely the same as in page 218, substituting the coordinates  $r$ ,  $\theta$ ,  $\varpi$ , in place of  $x$ ,  $y$ ,  $z$ .

consequently, we shall have by neglecting quantities of the order  $\alpha^2$

$$\epsilon' = 1 + \alpha \cdot \left\{ \frac{ds}{dr} \right\} + \alpha \cdot \left\{ \frac{du}{d\theta} \right\} + \alpha \cdot \left\{ \frac{dv}{d\varpi} \right\}. *$$

Let us suppose that after the time  $t$ , the primitive density  $(\rho)$  is changed into  $(\rho) + \alpha\rho'$ ; the preceding equation relative to the continuity of the fluid, will give

$$0 = r^2 \cdot \left\{ \rho' + (\rho) \cdot \left\{ \frac{du}{d\theta} \right\} + \left\{ \frac{dv}{d\varpi} \right\} + \frac{u \cdot \cos \theta}{\sin \theta} \right\} + (\rho) \cdot \left\{ \frac{d.r^2 s}{dr} \right\}.$$

36. Let us apply these results, to the oscillations of the sea. Its mass being homogeneous,  $\xi'$  vanishes, consequently,

$$0 = \left\{ \frac{d.r^2 s}{dr} \right\} + r^2 \cdot \left\{ \frac{du}{d\theta} \right\} + \left\{ \frac{dv}{d\varpi} \right\} + \frac{u \cos \theta}{\sin \theta} + \dots$$

$$* dr' = dr + \alpha ds, \quad d\theta' = d\theta + \alpha du, \quad d\varpi' = d\varpi + \alpha dv \quad \therefore \left( \frac{dr'}{dr} \right) \cdot \left( \frac{d\theta'}{d\theta} \right) \cdot \left( \frac{d\varpi'}{d\varpi} \right) =$$

$$\left( \frac{dr + \alpha ds}{dr} \right) \times \left( \frac{d\theta + \alpha du}{d\theta} \right) \times \left( \frac{d\varpi + \alpha dv}{d\varpi} \right) = 1 + \alpha \left( \frac{ds}{dr} \right) + \alpha \left( \frac{du}{d\theta} \right) + \alpha \left( \frac{dv}{d\varpi} \right),$$

it is plain, that if there was no motion, the differential of any coordinate  $\theta$ , with respect to another coordinate, would vanish, after the time  $t$ , this differential is of the order  $t^2$ .  $\left( \frac{d\theta + \alpha du}{d\varpi} \right) \times \left( \frac{d\varpi + \alpha dv}{d\theta} \right)$  is of the order  $t^2$  or  $\alpha^2$ , consequently it may be neglected, from which it appears, that all the terms in expression for  $\epsilon'$  after the first may be neglected.

$$+ \left\{ (\rho) + \alpha \rho' \right\} \cdot \left\{ 1 + \alpha \cdot \left\{ \frac{ds}{dr} \right\} + \alpha \cdot \left\{ \frac{du}{d\theta} \right\} + \alpha \cdot \left\{ \frac{dv}{d\varpi} \right\} \right\}.$$

$$r^2 (\sin \theta + \alpha u \cos \theta) \quad \left\{ = (\rho) \cdot r^2 \cdot \sin \theta \right\}$$

$$\left\{ (\rho) + \alpha \rho' \right\} \cdot 1 + \alpha \cdot \left\{ \frac{ds}{dr} \right\} + \alpha \cdot \left\{ \frac{du}{d\theta} \right\} + \alpha \cdot \left\{ \frac{dv}{d\varpi} \right\}.$$

$$(r^2 + 2\alpha s) \cdot \sin \theta + \alpha u \cos \theta \quad \left\{ = (\rho) \cdot r^2 \cdot \sin \theta, i, e, \right\}$$

Let us suppose, conformably to what appears to be the case of nature, that the depth of the sea is very small in comparison with the radius  $r$  of the terrestrial spheroid; let this depth be represented by  $\gamma$ ,  $\gamma$  being a very small function of  $\theta$  and  $\varpi$ , which depends on the law of this depth. If we integrate the preceding differential equation, with respect to  $r$ , from the surface of the solid which the sea covers, to the surface of the sea, \* it is obvious that the value of  $s$  will be equal to a function of  $\theta$ ,  $\varpi$ , and  $t$ , independent of

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$$\begin{aligned}
 & (\epsilon) \cdot (r^2 + 2\alpha rs) \cdot (\sin. \theta + \alpha u \cos. \theta) + (\epsilon) \cdot r^2 \cdot \sin. \theta \cdot \left\{ \alpha \cdot \left\{ \frac{ds}{dr} \right\} + \left\{ \frac{du}{d\theta} \right\} + \left\{ \frac{dv}{d\varpi} \right\} \right\} \\
 & + \alpha \epsilon' \cdot r^2 \cdot \sin. \theta = (\epsilon) r^2 \cdot \sin. \theta. \\
 \therefore & (\epsilon) \cdot r^2 \cdot \alpha u \cos. \theta + (\epsilon) 2\alpha rs \cdot \sin. \theta + (\epsilon) r^2 \cdot \sin. \theta \cdot \left\{ \alpha \left\{ \frac{ds}{dr} \right\} + \left\{ \frac{du}{d\theta} \right\} + \left\{ \frac{dv}{d\varpi} \right\} \right\} \\
 & + \alpha \epsilon' \cdot r^2 \cdot \sin. \theta = 0
 \end{aligned}$$

∴ dividing by  $\sin. \theta$  and  $\alpha$ , we obtain

$$\begin{aligned}
 & r^2 \cdot \left\{ (\epsilon) \cdot \left\{ \frac{du}{d\theta} \right\} + \left\{ \frac{dv}{d\varpi} \right\} + \frac{u \cdot \cos. \theta}{\sin. \theta} \right\} + \epsilon' \cdot \left\{ (\epsilon) 2rs + (\epsilon) r^2 \cdot \frac{ds}{dr} \right\} \\
 & \left\{ = \left\{ (\epsilon) \cdot \frac{2r \cdot dr \cdot s + r^2 ds}{dr} \right\} = (\epsilon) \left\{ \frac{d \cdot r^2 s}{dr} \right\} \right.
 \end{aligned}$$

\* The depth of the sea being inconsiderable, in comparison of the terrestrial radius, we may suppose, that for this depth  $r^2$ , and the factor of  $r^2$  in the second term, of the second member of this equation, are constant ∴ integrating we obtain

$$r^2 s' - r^2 s = \gamma \cdot r^2 \cdot \left( \left( \frac{du}{d\theta} \right) + \left( \frac{dv}{d\varpi} \right) + \frac{u \cos. \theta}{\sin. \theta} \right),$$

as the increment of the radius at the surface of the spheroid =  $\alpha u q \cdot \left( \frac{dl}{d\theta} \right) + \alpha v q \cdot \left( \frac{dl}{d\varpi} \right)$

see notes to page 252, ∴  $s'$  at the surface of the sea

$$= \gamma \cdot \left\{ \left\{ \frac{du}{d\theta} \right\} + \left\{ \frac{dv}{d\varpi} \right\} + \frac{u \cos. \theta}{\sin. \theta} \right\} + uq \cdot \left\{ \frac{dl}{d\theta} \right\} + vq \cdot \left( \frac{dl}{d\varpi} \right)$$

$r$ , together with a very small function which will be to  $u$  and to  $v$ , of the same order of smallness as the function  $\frac{\gamma}{r}$ ; but at the surface of the solid which the sea covers, when the angles  $\theta$ , and  $\varpi$ , are respectively changed into  $\theta + \alpha u$ ,  $\varpi + nt + \alpha v$ , it is easy to perceive that the distance of a molecule of water, contiguous to this surface, from the centre of gravity of the earth, only varies by a quantity very small with respect to  $\alpha u$  and  $\alpha v$ , and of the same order, as the products of these quantities, into the eccentricity of the spheroid covered by the sea: therefore, the function, which occurs in the expression for  $s$ , independent of the value of  $r$ , is a very small quantity of the same order; thus we can generally neglect  $s$ , as inconsiderable, in comparison of  $u$  and  $v$ . Consequently, the equation of the motion of the sea, which has been given in No. 35, becomes,

$$\begin{aligned} & r^2 \delta \theta \cdot \left\{ \left\{ \frac{d^2 u}{dt^2} \right\} - 2n \cdot \sin \theta \cdot \cos \theta \cdot \left\{ \frac{dv}{dt} \right\} \right\} \\ & + r^2 \delta \varpi \cdot \left\{ \sin^2 \theta \cdot \left\{ \frac{d^2 v}{dt^2} \right\} + 2n \cdot \sin \theta \cdot \cos \theta \cdot \left\{ \frac{du}{dt} \right\} \right\} = -g \cdot \delta y + \delta V; \quad (M) \end{aligned}$$

the equation (L) of the same number relative to any point of the interior of the fluid, gives in the state of equilibrium,

$$0 = \frac{n^2}{2} \cdot \delta \cdot \left( (r + \alpha s) \cdot \sin(\theta + \alpha u) \right)^2 + (\delta V) - \frac{(\delta p)}{\rho}$$

( $\delta V$ ) and ( $\delta p$ ) being the values of  $\delta V$  and  $\delta p$ , which in the state of equi-

these two last terms are to  $u$ , or  $v$ , as the product of these quantities into the eccentricity. With respect to the first term, it may be remarked that we can derive another expression for it, in terms of the difference of the eccentricities of the interior and exterior spheroids, divided by  $r$ , but this difference is evidently proportional to  $\gamma$ , in fact this term will be to  $ur$  as  $\gamma$  to  $r$ . The integral involves  $t$  because it was taken with respect to the characteristic  $d$  and not  $\delta$ .

The last member of the equation (L), becomes in a state of motion, in consequence of this substitution,

librium, answer to the quantities  $r + \alpha s$ ,  $\theta + \alpha u$ ,  $w + \alpha v$ . Suppose that when the fluid is in motion, we have

$$\delta V = (\delta V) + \alpha \delta V'; \quad \delta p = (\delta p) + \alpha \delta p';$$

the equation (L) will give

$$\left\{ d \cdot \left( V' - \frac{p'}{r} \right) \right\} = \left( \frac{d^2 s}{dt^2} \right) - 2nr \cdot \sin. 2\theta \cdot \left( \frac{dv}{dt} \right).$$

From a consideration of the equation (M), it appears that  $n \cdot \left( \frac{dv}{dt} \right)$  is of the same order as  $y$  or  $s$ , and consequently of the order  $\frac{\gamma u}{r}$ ; the value of the first member of this equation is therefore of the same order; \* thus, multiplying this value, by  $dr$ , and then integrating from the surface of the spheroid, to the surface of the sea; we shall have  $V' - \frac{p'}{r}$  equal to a very small function, of the order  $\frac{\gamma s}{r}$ , plus a function of  $\theta$ ,  $w$ , and  $t$ , independent of  $r$ , which we will denote by  $\lambda$ ; therefore, if in the

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\*  $\frac{n}{2} \cdot \partial \left\{ (r + \alpha s) \cdot \sin. (\theta + \alpha u) \right\}^2 + (\partial V) - \left\{ \frac{\partial p}{\epsilon} \right\} + \alpha \cdot \partial V' - \alpha \frac{\partial p'}{\epsilon}$ , the three first terms destroy each other.  $\therefore \alpha \partial V' - \alpha \frac{\partial p'}{\epsilon} + \lambda$  is equal to the first member of the equation (L), and since it is an exact variation, the first member of the equation (L) will be so also,  $\therefore V' - \frac{p'}{\epsilon}$  differenced with respect to  $r$ , is equal to the term of the first member of the equation (L), which is multiplied by  $\partial r$ .

$\therefore \sin. \theta. \cos. \theta = \left\{ \frac{d \cdot \cos. \theta^2}{d\theta} \right\} \therefore$  in order that  $-2n \cdot \left\{ \frac{dv}{dt} \right\} \cdot \sin. \theta. \cos. \theta$  may be of the same order as  $\left\{ \frac{d^2 u}{dt^2} \right\}$  it is necessary that  $n \left\{ \frac{dv}{dt} \right\}$  should be of the order  $y$  or  $s$ , which is of the order  $\frac{\gamma u}{r}$

equation (L) of No. 35, we only consider the two variables  $\theta$  and  $\varpi$ , it will be changed into the equation (M), with this sole difference, that the second member will be changed into  $\delta\lambda$ . But  $\lambda$  being independent of the depth of the molecule, which we consider; if we suppose this molecule very near the surface; the equation (L) must evidently coincide with the equation (M); therefore we have  $\delta\lambda = \delta V' - g\delta y$ , and consequently,

$$\delta \left\{ V' - \frac{p'}{\rho} \right\} = \delta V' - g\delta y;$$

the value of  $\delta V'$  in the second member of this equation, being relative to the surface of the sea.\* We shall find in the theory of the flux and reflux of the sea, that this value is very nearly the same for all molecules situated on the same terrestrial radius, from the surface of the solid which the sea covers, to the surface of the sea; therefore with respect to all these molecules  $\frac{\delta p'}{\rho} = g\delta y$ ; which gives  $p' = \rho gy$ , together with a function independent of  $\theta$ ,  $\varpi$ , and  $r$ ; but at the surface of the level of the sea, the value of  $\alpha p'$ , is equal to the pressure of a small column of water  $\alpha y$ , which is elevated

\*  $\int \frac{d^2 s}{dt^2} dr - 2 \int nr dr \sin^2 \theta \left\{ \frac{dv}{dt} \right\}$  integrated between the the surface of the spheroid, and the surface of sea, gives the integral of the text, the first term is  $\pm$  to  $\left\{ \frac{d^2 s}{dt^2} \right\} y$ , which is a function of  $\theta, \varpi$ , and  $t$ ,  $= \lambda$ , the other term being of the order  $\frac{y^2}{r}$  may be rejected. If we only consider the terms, which refer to  $\theta$  and  $\varpi$ , the first member of the equation (L) is the same as the first member of the equation (M), near the surfacee, the last term of the first member of the equation (L') vanishes  $\therefore$  the equation (L) must in this case coincide with the equation (M), but  $\lambda$  the member of the equation (L) does not vary  $\therefore$  we have the second member of the equation (L)  $=$  the second member of the equation(M) i. e.  $\delta\lambda = \delta V' - g\delta y$ ; but  $\delta\lambda = \delta \left\{ V' - \frac{p'}{\rho} \right\} \dots \delta \left\{ V' - \frac{p'}{\rho} \right\} = \delta V' - g\delta y$ , from the theory of the tides it appears that the  $\delta V'$  in these two members are the same,  $\therefore g\delta y = \frac{\delta p'}{\rho}$  and  $p' = \rho gy +$  a constant arbitrary quantity; when the integral is taken between the surface of spheroid, and surfacee of the sea, this constant arbitrary quantity may be rejected.

above this surface, and this pressure is equal to  $\alpha\rho gy$ ; therefore we have, in the entire of the interior of the fluid, from the surface of the spheroid covered by the sea, to the surface of the level of the sea,  $p' = \rho gy$ ; consequently, any point of the surface of the spheroid, which is covered by the sea, is more pressed than in the state of equilibrium, by the entire weight of a column of water, contained between the surface of the sea, and the surface of level. This excess of pressure becomes negative, for those points, where the surface of the sea is depressed beneath the surface of level.

It follows from which has been stated above, that if we only consider the variations of  $\theta$  and  $\varpi$ ; the equation (L) will be changed into the equation (M), for all the interior molecules of the fluid. Consequently, the values of  $u$ , and  $v$ , relative to all molecules, \* situated on the same terrestrial radius, are determined by the same differential equations; thus, supposing, as we shall do in the theory of the flux and reflux of the sea, that at the commencement of the motion, the values of  $u$ ,  $(\frac{du}{dt})$ ,  $v$ ,  $(\frac{dv}{dt})$ , were the same for all the molecules of the fluid, situated on the same radius, these molecules will exist the same radius, during the oscillations of the fluid. Therefore the values of  $r$ ,  $u$ , and  $v$ , may be supposed very nearly the same, on the small part of the radius, comprised between the solid, which the sea covers, and the surface of the sea; thus, if we integrate with respect to  $r$ , the equation

$$0 = \left\{ \frac{d.r^2 s}{dr} \right\} + r^2 \cdot \left\{ \left\{ \frac{du}{d\theta} \right\} + \left\{ \frac{dv}{d\varpi} \right\} + \frac{u \cos \theta}{\sin \theta} \right\}; +$$

\* At the commencement of the motion  $u$ , and  $v$ ,  $\left\{ \frac{du}{dt} \right\}$ ,  $\left\{ \frac{dv}{dt} \right\}$ , are the same, for all molecules situated on the same radius,  $\therefore$  after the interval  $dt$ , the corresponding values of  $u$  and  $v$ , will be the same for all molecules situated on the same radius.

$\dagger r^2 s - (r^2 s) = r^2 s - r^2 \cdot (s) + 2r\gamma \cdot (s) + \gamma^2 (s)$  for  $(r^2) = (r - \gamma)^2$   
 $\gamma$  being a function of  $\theta$ , and  $\varpi$ , when these angles are increased by the quantity  $\alpha u$ ,  $\alpha v$ , becomes  $\gamma + \alpha u \cdot \left\{ \frac{dy}{d\theta} \right\} + \alpha v \cdot \left\{ \frac{dy}{d\varpi} \right\}$  this is the value of  $\gamma$  corresponding to the angle  $\theta + \alpha u$ ,  $\varpi + nt + \alpha v$  for the surface of equilibrium,  $\therefore$  where the fluid is in motion, we must add  $\alpha y$  to this expression.

we shall have

$$0 = r^2 s - (r^2 s) + r^2 \gamma \left\{ \left\{ \frac{du}{d\theta} \right\} + \left\{ \frac{dv}{d\omega} \right\} + \frac{u \cdot \cos. \theta}{\sin. \theta} \right\};$$

$(r^2 s)$  being the value of  $r^2 s$ , at the surface of the spheroid covered by the sea. The function  $r^2 s - (r^2 s)$  is very nearly equal to  $r^2 \{s - (s)\} + 2r\gamma(s)$ ,  $(s)$  being what  $s$  becomes at the surface of the spheroid ; considering, the smallness of  $\gamma$ , and  $(s)$ , in comparison of  $r$ , we may neglect the term  $2r\gamma(s)$  ; therefore, we shall have

$$r^2 s - (r^2 s) = r^2 \{s - (s)\}.$$

Now, the depth of the sea, corresponding to the angles  $\theta + \alpha u$ ,  $\omega + nt + \alpha v$ , is  $\gamma + \alpha \{s - (s)\}$ . If the origin of the angles  $\theta$ , and  $nt + \omega$ , be referred to a point, and a meridian, which are fixed on the surface of the earth, which we are permitted to do, as we shall see very soon ; this same depth will be  $\gamma + \alpha u$ .

$\left\{ \frac{dy}{d\theta} \right\} + \alpha v \cdot \left\{ \frac{dy}{d\omega} \right\}$ , plus the elevation  $\alpha y$  of the molecule of the fluid at the surface of the sea, above the surface of level ; therefore, we shall have

If we make  $\cos. \theta = \mu$ , then

$$dt = \frac{-d\mu}{\sin. \theta}, \quad d. \sqrt{1-\mu^2} = \frac{-\mu \cdot d\mu}{\sqrt{1-\mu^2}} = \frac{-d\mu \cdot \cos. \theta}{\sin. \theta}$$

consequently the equation of continuity, on the supposition that the sea is homogeneous becomes,

$$\begin{aligned} & \left( \frac{d.r^2.s}{dr} \right) + r^2 \cdot \left( \frac{dv}{d\omega} \right) - \frac{r^2 \cdot du}{d\mu} \cdot \sqrt{1-\mu^2} + u \frac{d.\sqrt{1-\mu^2}}{d\mu} \\ & - \left( \frac{d.r^2.s}{dr} \right) + r^2 \cdot \left( \frac{dv}{d\omega} \right) - r^2 \cdot \left( \frac{d.(u \cdot \sqrt{1-\mu^2})}{d\mu} \right) \text{ see Book IV. Chap. 2.} \end{aligned}$$

$$\begin{aligned} & \text{In like manner, if } \gamma \text{ be constant, } y = -\gamma \cdot \left\{ \left\{ \frac{du}{d\theta} \right\} + \left\{ \frac{dv}{d\omega} \right\} + \frac{u \cos. \theta}{\sin. \theta} \right\} = \\ & -\gamma \cdot \left\{ \left\{ \frac{dv}{d\omega} \right\} + \frac{d.u \cdot \sqrt{1-\mu^2}}{d\mu} + \frac{u.d.\sqrt{1-\mu^2}}{d\mu} \right\} = -\gamma \cdot \left\{ \left\{ \frac{dv}{d\omega} \right\} + \frac{d.(u \cdot \sqrt{1-\mu^2})}{d\mu} \right\} \end{aligned}$$

See Book IV. Chap. 1, No. 2.

$$s-(s) = y + u \cdot \left\{ \frac{d\gamma}{d\theta} \right\} + v \cdot \left\{ \frac{d\gamma}{d\pi} \right\}.$$

Consequently the equation relative to the continuity of the fluid will become \*

$$y = - \left\{ \frac{d \cdot \gamma u}{d\theta} \right\} - \left\{ \frac{d \cdot \gamma v}{d\pi} \right\} - \frac{\gamma u \cdot \cos \theta}{\sin \theta}. \quad (\text{N})$$

It may be remarked, that in this equation, the angles  $\theta$  and  $nt+\pi$  are reckoned from a point, and a meridian, which are respectively fixed on the surface of the earth, and in the equation (M), these angles are reckoned from the axis of  $x$ , and from a plane, which passing through this axis, revolves about it with a rotatory motion, expressed by  $n$ ; but this axis, and this plane are not fixed on the surface of the earth, since the attraction and pressure of the fluid which covers it, as well as the rotatory motion of the spheroid, disturb a little their position. However it is easy to perceive that these perturbations † are to the values of  $\alpha u$ , and  $\alpha v$ , in the ratio of the mass

\* Substituting for  $s-(s)$ , its value

$$-\gamma \cdot \left\{ \frac{du}{d\theta} \right\} - \gamma \cdot \frac{dv}{d\pi} - \gamma \cdot \frac{u \cdot \cos \theta}{\sin \theta}$$

and observing that

$$\left\{ \frac{d \cdot \gamma \cdot u}{d\theta} \right\} = \gamma \cdot \left\{ \frac{du}{d\theta} \right\} + u \cdot \left\{ \frac{d\gamma}{d\theta} \right\},$$

we will arrive at the value of  $y$ , which is given in the text.

† In the state of equilibrium, neither the pressure or attraction of the ocean, can produce any motion in the spheroid covered by the sea, and it is only the stratum of water which in consequence of the attractions of the exterior bodies, and of the centrifugal force; is elevated above the surface, which can produce any effect. The effects of the pressure and attraction may be considered separately, with respect to the first, if the mean radius of the earth be supposed equal to unity,  $\alpha y$  being the elevation, the action of the aqueous stratum is equal to the difference of the attractions of two spheroids, of which the radius of the interior = 1, of the exterior =  $1 + \alpha y$ , naming this difference  $\alpha y h$ . and  $\tau$  its direction,  $\alpha y h d\tau$  will be the expression for this attraction; multiplied into the element of its direction,  $\tau$  being a function of  $\theta$ , and  $\pi$ ,  $d\tau$

of the sea, to the mass of the spheroid ; therefore, in order to refer the angles  $\theta$ , and  $nt + \varpi$ , to a point and meridian, which are invariable on the surface of the spheroid, in the two equations (M) and (N) ; we should alter  $u$ , and  $v$ , by quantities of the order  $\frac{\gamma u}{r}$  and  $\frac{\gamma v}{r}$ , which quantities we are permitted to neglect ; therefore we may suppose in these equations, that  $\alpha u$  and  $\alpha v$  are the motions of the fluid, in latitude and longitude.\*

It may also be observed, that the centre of gravity of the spheroid being supposed immovable, we should transfer in an opposite direction to the molecules, the forces by which it is actuated, in consequence of the reaction of the sea ; but the common centre of gravity of the sea and spheroid being invariable in consequence of this reaction ; it is manifest that the ratio of these forces, to those by which the molecules are solicited by the action of the spheroid, is of the same order, as the ratio of the mass of the fluid to that of the spheroid, and consequently of the order  $\frac{\gamma}{r}$ , therefore they may be omitted in the calculation of  $\delta V'$ .

$$= \left\{ \frac{d\tau}{d\theta} \right\} \alpha u + \left\{ \frac{d\tau}{d\varpi} \right\} \alpha v \therefore \alpha y \cdot h d\tau = \alpha y h \cdot \left\{ \frac{d\tau}{d\theta} \right\} \alpha u + \alpha y h \cdot \left\{ \frac{d\tau}{d\varpi} \right\} \alpha v$$

The attractions are of the order  $\alpha y$  ; for if  $y$  vanished there would be no pressure or action, but  $y$  is of the order  $\frac{\gamma u}{r}$ . The exact effect which the attractions, and pressures of the aqueous stratum produce are calculated in Book V. Nos. 10 and 11.

\* The centre of gravity of the spheroid is considered immovable, because we do not consider the *absolute* oscillations of the molecules in space, but only their oscillations relative to the mass of the fluid. The common centre of gravity of the fluid and spheroid covered by the fluid is not affected by the mutual action of these molecules, see No. 20. With respect to the action of foreign bodies, their effect is not to be neglected, as in case of the action of the sea, if we consider the centre of gravity of the spheroid immovable, we must transfer in a contrary direction to the molecule, the attraction which such bodies exert on the centre of gravity of the spheroid, the oscillations  $\alpha y$  and the force which actuates the particles are of the order  $\alpha \frac{d^2 y}{dt^2}$ , or  $\alpha \cdot q \cdot \left\{ \frac{d^2 s}{dt^2} \right\}$ , see preceding note.

37. Let us consider in the same manner, the motions of the atmosphere. In this investigation, we shall omit the consideration of the variation of heat in different latitudes, and different elevations, as well as all anomalous causes of perturbation, and consider only the regular causes which act upon it, as upon the ocean. Consequently, we may consider the sea as surrounded by an elastic fluid of an uniform temperature; we shall also suppose, that the density of this fluid is proportional to its pressure, which is conformable to experience. This supposition implies, \* that the atmosphere has an infinite height; but it is easy to be assured, that at a very small height, its density is so small, that it may be regarded as evanescent.

This being premised, let  $s'$ ,  $u'$ , and  $v'$ , denote for the molecules of the atmosphere, what  $s$ ,  $u$ ,  $v$ , designated, for the molecules of the sea; the equation (L) of No. 35, will then become

$$\begin{aligned} & \alpha r^2 \delta \theta \cdot \left\{ \left\{ \frac{d^2 u'}{dt^2} \right\} - 2n \cdot \sin. \theta \cdot \cos. \theta \cdot \left\{ \frac{dv'}{dt} \right\} \right\} \\ & + \alpha r^2 \delta \omega \cdot \left\{ \sin. \theta \cdot \left( \frac{d^2 v'}{dt^2} \right) + 2n \cdot \sin. \theta \cdot \cos. \theta \cdot \left( \frac{du'}{dt} \right) + \frac{2n \cdot \sin. \theta}{r} \cdot \left( \frac{ds'}{dt} \right) \right\} \\ & + \alpha \delta r \cdot \left\{ \left( \frac{d^2 s'}{dt^2} \right) - 2nr \cdot \sin. \theta \cdot \left( \frac{dv'}{dt} \right) \right\} = \frac{n}{2} \cdot \delta \cdot (r + \alpha s') \cdot \sin. \theta + \alpha u' \cdot \left\{ \right\}^2 \\ & + \delta V - \frac{\delta p}{\rho}. \end{aligned}$$

M M

\* According as the fluid is elevated above the surface of the earth, it becomes rarer, in consequence of its elasticity which dilates it more and more, as it is less compressed, and it would extend indefinitely, and eventually dissipate itself in space, if the molecules of its surface were elastic; consequently, if there is a state of rarity, in which the molecules are devoid of elasticity, the elasticity of the atmosphere must diminish in a greater ratio than the compressing force.

At first let us consider the atmosphere in a state of equilibrium, in which case  $s$ ,  $u'$  and  $v'$  vanish. Then, the preceding equation, being integrated becomes,

$$\frac{n^2}{2} \cdot r^2 \cdot \sin. {}^2 \theta + V - \int \frac{\delta p}{\rho} = \text{constant.}$$

The pressure  $p$  being by hypothesis proportional to the density; we shall make  $p = l \cdot g \cdot \rho$ ,  $g$  represents the gravity at a determined place, \* which we will suppose to be the equator, and  $l$  is a constant quantity which expresses the height of the atmosphere, of which the density is throughout the same as at the surface of the sea: this height is very small relative to the radius of the terrestrial spheroid, of which it is less than the 720th part.

The integral  $\int \frac{\delta p}{\rho}$  is equal to  $lg. \log. \rho$ ; consequently the preceding equation relative to the equilibrium of the atmosphere becomes,

$$lg. \log. \rho = \text{constant} + V + \frac{n^2}{2} \cdot r^2 \cdot \sin. {}^2 \theta.$$

At the surface of the sea, the value of  $V$  is the same for a molecule of air, as for a molecule of water contiguous to it, because the forces which solicit each molecule, are the same; but the condition of the equilibrium of the sea requires, that we should have

$$V + \frac{n^2}{2} \cdot r^2 \cdot \sin. {}^2 \theta = \text{constant};$$

\* An homogeneous atmosphere is an atmosphere, supposed to be of the same weight as that which actually surrounds the earth; its density being uniform, and every where equal to the density of the air at the surface of the earth. Let  $h$  be the height of the mercury in the barometer at the equator, and  $d$  its density, we shall have  $l \epsilon = h \cdot d$ :  $l = \frac{hd}{\epsilon}$  and by substituting for  $h$  and  $d$  and  $\epsilon$  their numerical values,  $l$  comes out equal to  $5\frac{1}{4}$  miles very nearly, which is somewhat less than the 720th part of the radius of the equator. When the temperature is given, this height is a constant quantity, whatever be the changes which the pressure undergoes.

therefore  $\rho$  is constant at this surface, *i.e.*, the density of the stratum of air contiguous to the sea, is every where the same, in the state of equilibrium.

Let  $R$  represent, the part of the radius  $r$ , comprehended between the centre of the spheroid and the surface of the sea, and  $r'$  the part comprised between this surface and a molecule of air elevated above it;  $r'$  will differ only by quantities nearly of the order  $\left(\frac{n^2 \cdot r'}{R}\right)^2$ , \* from the height of this molecule above the surface of

the sea; we may without sensible error neglect quantities of this order. The equation between  $\rho$  and  $r$  will give

$$\begin{aligned} \lg. \log. \rho = & \text{constant} + V + \frac{r'}{1} \left\{ \frac{dV}{dr} \right\} + \frac{r'^2}{1.2} \left\{ \frac{d^2V}{dr^2} \right\} \\ & + \frac{n^2}{2} \cdot R^2 \cdot \sin.^2 \theta + n^2 R r' \cdot \sin.^2 \theta; \end{aligned}$$

the values of  $V$ ,  $(\frac{dV}{dr})$  and  $(\frac{d^2V}{dr^2})$  being relative to the surface of the sea, where we have,

$$\text{constant} = V + \frac{n^2}{2} \cdot R^2 \cdot \sin.^2 \theta;$$

the quantity  $= (\frac{dV}{dr}) - n^2 R \cdot \sin.^2 \theta$ , expresses the gravity at the same

\*  $V$  being a function of  $R$ ,  $\theta$ , and  $\omega$ , if  $R$  receive the increment  $r'$ ,  $V$  becomes  $= V + \frac{r'}{1} \cdot \left\{ \frac{dV}{dr} \right\} + \frac{r'^2}{1.2} \cdot \left\{ \frac{d^2V}{dr^2} \right\} + \&c.$  and the expression  $\frac{n^2}{2} R^2 \cdot \sin.^2 \theta$  will be increased by the quantity  $n^2 R r' \cdot \sin.^2 \theta + \frac{n^2}{2} r'^2 \cdot \sin.^2 \theta$ , but this last term being indefinitely small, may be rejected.

*surface*; which we will represent by  $g'$ . The function  $\left\{ \frac{d^2V}{dr} \right\}$  \* being multiplied by a very small quantity  $r'$ ,<sup>2</sup> we may determine it on the hypothesis that the earth is spherical, and we may neglect the density of the atmosphere relatively to that of the earth; therefore, we shall have very nearly,

$$-\left\{ \frac{dV}{dr} \right\} = g = \frac{m}{R^2};$$

$m$  expressing the mass of the earth; consequently  $\left\{ \frac{d^2V}{dr^2} \right\} =$

$-\frac{2m}{R^2} = -\frac{2g'}{R}$ ; therefore we shall have  $lg. \log. \rho = \text{constant}$

$-r'g' - \frac{r'^2}{R}g'$ ; from which may be obtained

$$-\frac{r'g'}{lg} \cdot \left\{ 1 + \frac{r'}{R} \right\},$$

$\rho = \Pi.c \dagger$

\* If the earth was a sphere then  $r'$ , would be equal to the height of the molecule of the atmosphere above the surface of the sea, and as in the case of a spheroid the height is determined by a normal drawn to the surface from the molecule, the difference between  $r'$  and the part of this normal which is exterior to the surface, depends on the ellipticity of the spheroid, which is nearly of the order  $\left( \frac{n^2 r'}{g} \right)^2$ , for he afterwards supposes that the earth is at the surface of the sea very nearly  $\frac{n^2 r'}{g}$  spherical, ∴ the only aberration from sphericity can arise from the greater centrifugal force of the molecule of the air, the ratio of this excess of centrifugal force to gravity, for a molecule elevated at the equator, above the surface of the earth  $= \frac{n^2 r'}{g}$ , and the intercept at the surface between the direction of  $r$ , and the direction of a normal drawn from the molecule of the air must be evidently of the order of the ellipticity  $i$ ,  $e$ , of the order  $\frac{n^2 r'}{g}$ , and the difference between  $r'$  and this height is equal to the square of this quantity divided by  $R$  very nearly.

†  $dV = P \partial x + Q \partial y + R \partial z$ , and if we refer the molecules to the polar coordinates  $r, \theta, \varpi$ ,

$$dV = \left\{ \frac{dV}{dr} \right\} \cdot \partial r + \left\{ \frac{dV}{d\theta} \right\} \cdot \partial \theta + \left\{ \frac{dV}{d\varpi} \right\} \cdot \partial \varpi, \therefore \left\{ \frac{dV}{dr} \right\}$$

$c$  being the number of which the hyperbolical logarithm is equal to unity, and  $\Pi$  being a constant quantity evidently equal to the density of the air at the surface of the sea. Let  $h$  and  $h'$  represent the lengths of a pendulum, which vibrates seconds at the surface of sea, under the equator, and at the latitude of the molecule of the atmosphere, which has been

is that part of the force  $\delta V$ , which is resolved in the direction of the radius of the earth,  $\theta =$  the complement of latitude  $\therefore n^2 R \sin. ^2 \theta$  is the part of the centrifugal force, which acts in the direction of the terrestrial radius. The force varying inversely as the square of the distance,  $V \propto \frac{1}{R}$ , and  $\frac{dV}{dr} = \frac{m}{R^2}$  see Book II. No. 12.

The earth being supposed spherical  $\left\{ \frac{dV}{dr} \right\}$  is nearly the same in every parallel, and  $\therefore$  equal to its value at the equator, where it is equal to  $g$  very nearly; in the value of  $\left\{ \frac{d^2 V}{dr^2} \right\}$  we substitute  $g'$  in place of  $\frac{m}{R^2}$ , for thus the error of the supposition that  $g = \frac{m}{R^2}$  is somewhat corrected; substituting for

$$\left( \frac{dV}{dr} \right) + n^2 R \cdot \sin. ^2 \theta, \left( \frac{d^2 V}{dr^2} \right) + \frac{n^2}{2} R^2 \cdot \sin. ^2 \theta$$

their values and of remarking that  $V + \frac{n^2}{2} R \cdot \sin. ^2 \theta$  is constant, we obtain the value of  $\lg. \log. \epsilon$  which is given in the text.

The density of the atmosphere being inconsiderable with respect to that of the earth, we may without sensible error, neglect the attraction of its molecules.

The variable part of the value of  $\epsilon$  is necessarily negative, for the density decreases, according as we ascend in the atmosphere;

$$\log. \epsilon = \frac{\text{const}}{l.g} - \frac{r' g'}{l.g} \left( 1 + \frac{r'}{R} \right)$$

$$\text{and at the surface of the sea } r' = 0 \therefore \epsilon = c = \Pi \text{ which is consequently the value of } \epsilon$$

at the surface of the sea; when the times of vibration are given, the lengths of the isochronous pendulums are proportional to the forces of gravity,  $\therefore \frac{g'}{g} = \frac{h'}{h}$ .

considered : we shall have  $\frac{g'}{g} = \frac{h'}{h}$ , and consequently,

$$-\frac{r'h'}{lh} \cdot \left\{ 1 + \frac{r'}{R} \right\}.$$

$r \equiv \text{II. } c^*$

From this expression of the density of the air, it appears that strata of the same density, are throughout equally elevated about the surface of the sea, with the exception of the quantity  $\frac{r'(h'-h)}{h}$ ; however, in the exact determination of the heights of mountains by observations of the barometer, this quantity ought not to be neglected.

Let us now consider the atmosphere in a state of motion, and let the oscillations of a stratum of level, or of the same density in the state of equilibrium, be determined. Let  $\alpha\phi$  represent the elevation of a molecule of the fluid, above the surface of level, to which it appertains in the

\* If we expand the value of  $\epsilon$  into a series it becomes equal to

$$1 - \frac{r'h'}{lh} \cdot \left( 1 + \frac{r'}{R} \right) + \frac{1}{2} \cdot \frac{r'h'}{lh} \cdot \left( 1 + \frac{r'}{R} \right)^2 + \text{&c.}$$

and neglecting higher powers of  $r'$ ,  $= 1 - \frac{r'h'}{lh}$ . ∴ in strata of *equal elevation* above the level of the sea, the difference of density is equal to  $r' \left( \frac{h'-h}{lh} \right)$ ; in like manner, if the density of two strata, in latitudes of which the forces are respectively equal to  $g$  and  $g'$ ; be the same, we shall have

$$\frac{r'h'}{lh} \cdot \left( 1 + \frac{r'}{R} \right) = \frac{r''h}{lh} \cdot \left( 1 + \frac{r''}{R} \right)$$

$r'$  and  $r''$  being the heights which correspond to the respective latitudes, ∴ neglecting quantities of the second order we shall have, when the density is given,  $r'h' = r''h$  ∴  $r'' = \frac{r'h'}{h}$  consequently the difference between  $r'$  and  $r'' \left( = \frac{r'h'}{h} \right) = r' \cdot \left( \frac{h'-h}{h} \right)$ .

state of equilibrium; it is manifest that, in consequence of this elevation, the value of  $\delta V$  will be increased by the differential variation  $-\alpha g \cdot \delta \varphi$ ; thus we shall have,  $\delta V = (\delta V) - \alpha g \cdot \delta \varphi + \alpha \delta V'$ ; ( $\delta V$ ) being the value of  $\delta V$ , which, in the state of equilibrium, corresponds to the stratum of level, and to the angles  $\theta + \alpha u$ , and  $nt + \omega + \alpha v$ ;  $\delta V'$  being the part of  $\delta V$ , which is produced by the new forces, which in the state of motion, agitate the atmosphere.

Let  $\rho = (\rho) + \alpha \rho'$ ,  $\rho$  being the density of the stratum of level, in the state of equilibrium. By making  $\frac{l\rho'}{(\rho)} = y'$ , we shall have

$$\frac{\delta p}{\rho} = \frac{lg \cdot \delta(\rho)}{(\rho)} + \alpha g \cdot \delta y'; *$$

but in the state of equilibrium we have,

$$0 = \frac{n^2}{2} \cdot \delta \cdot \{(r + \alpha s) \cdot \sin. (\theta + \alpha u)\}^2 + (\delta V) - \frac{lg \cdot \delta(\rho)}{(\rho)};$$

therefore, the general equation relative to the motion of the atmosphere will become, relatively to the strata of level, with respect to which  $\delta r$  very nearly vanishes,

$$\begin{aligned} & r \cdot \delta \theta \cdot \left\{ \left\{ \frac{d^2 u'}{dt^2} \right\} - 2n \cdot \sin. \theta \cdot \cos. \theta \cdot \left\{ \frac{dv'}{dt} \right\} \right\} \\ & + r^2 \cdot \delta \omega \cdot \left\{ \sin. 2\theta \cdot \left\{ \frac{d^2 v'}{dt^2} \right\} + 2n \cdot \sin. \theta \cdot \cos. \theta \cdot \left\{ \frac{du'}{dt} \right\} + \frac{2n \cdot \sin. 2\theta}{r} \cdot \left\{ \frac{ds'}{dt} \right\} \right\} \\ & p * = l \cdot g \cdot \rho = lg \cdot (\rho) + \alpha lg \cdot \rho' \therefore \frac{\delta p}{\rho} = \frac{lg \cdot \delta(\rho)}{(\rho) + \alpha \rho'} + \frac{\alpha lg \cdot \delta \rho'}{(\rho) + \alpha \rho'} \\ & = \text{neglecting quantities of the order } \alpha^2, lg \cdot \delta(\rho) \cdot (\cdot(\rho)^{-1} \cdot (\rho) - \alpha \cdot \rho') \\ & + \alpha lg \cdot \delta \rho' \cdot (\cdot(\rho)^{-1} - \rho^{-2} \alpha \rho') = lg \cdot \frac{\delta(\rho)}{(\rho)} - \frac{lg \cdot \alpha \rho' \cdot \delta(\rho)}{(\rho^2)} + \frac{\alpha lg \cdot \delta \rho'}{(\rho)} = \frac{lg \cdot \delta(\rho)}{(\rho)} \\ & + \frac{lg \cdot \alpha \delta \rho' \cdot (\rho) - lg \cdot \alpha \rho' \cdot d(\rho)}{(\rho^2)} = \frac{lg \cdot \delta(\rho)}{(\rho)} + \alpha g \cdot \delta \cdot \frac{l \cdot \rho'}{(\rho)} (= \alpha g \cdot \delta y'). \end{aligned}$$

$$= \delta V' - g \cdot \delta\varphi - g \delta y' + n^2 r \cdot \sin. {}^2 \theta \cdot \delta. (s' - (s')), ^*$$

$\alpha (s')$  being the variation of  $r$ , which in the state of equilibrium corresponds to the variations  $\alpha u'$ ,  $\alpha v'$ , of the angles  $\theta$ , and  $\omega$ .

Let us suppose that all the molecules, which at the commencement of the motion existed on the same radius vector, remained constantly on the same radius in a state of motion, which, as appears from what precedes, obtains in the oscillations of the sea; and let us examine whether this supposition is consistent with the equations of the motion and continuity of the atmosphere. For this purpose, it is necessary that the values of  $u'$  and of  $v'$ , should be the same for all these molecules, as we shall see in the sequel, when the forces which cause this variation are determined; consequently, it is necessary that the variations  $\delta\varphi$  and  $\delta y$ , should be the same for these molecules, and moreover that the quantities

$$2nr. \delta\omega. \sin. {}^2 \theta. \left\{ \frac{ds'}{d\omega} \right\}, \text{ and } n^2 r. \sin. {}^2 \theta. \delta. \left\{ s' - (s') \right\},$$

may be neglected in the preceding equation.

At the surface of the sea, we have  $\varphi = y$ ,  $\alpha y$  being the elevation of the surface of the sea above the surface of level. Let us examine whether the supposition of  $\varphi$  equal to  $y$ , and of  $y$  constant for all molecules of the atmosphere, existing on the same radius vector, is compatible with the equation of the continuity of the fluid. This equation is by No. 35,

\*  $\alpha s'$  and  $\alpha(s')$  being the variations of  $r$ , corresponding respectively, in the states of motion and equilibrium, to the variations  $\alpha u'$  and  $\alpha v'$ , the expression

$$\frac{n^2}{2} \cdot \delta. \left\{ (r + \alpha s'). \sin. (\theta + \alpha u') \right\}^2 = \frac{n^2}{2} \cdot \delta. \left\{ (r + \alpha(s')) + \alpha(s' - (s')). \sin. (\theta + \alpha u') \right\}^2$$

and when we neglect quantities of the order  $\alpha^2$ , the part of this expression, which does not occur in the equation

$$0 = \frac{n^2}{2} \cdot \delta. \left\{ (r + \alpha s). \sin. (\theta + \alpha u) \right\}^2, \text{ is, } n^2 r \cdot \alpha \cdot \delta. \left\{ (s' - (s')) \right\} \cdot \sin. {}^2 \theta.$$

$$0 = r^2 \cdot \left\{ \rho' + (\rho) \cdot \left\{ \left\{ \frac{du'}{d\theta} \right\} + \left\{ \frac{dv'}{d\varpi} \right\} + \frac{u' \cos. \theta}{\sin. \theta} \right\} \right\} + (\rho) \cdot \left\{ \frac{d.r^2 s'}{dr} \right\}; *$$

from which we obtain

$$y = -l \cdot \left\{ \left\{ \frac{d.r^2 s'}{r^2 dr} \right\} + \left\{ \frac{du'}{d\theta} \right\} + \left\{ \frac{dv'}{d\varpi} \right\} + \frac{u' \cos. \theta}{\sin. \theta} \right\}.$$

$r + \alpha s'$  is equal to the value of  $r$  at the surface of level, which corresponds to the angles  $\theta + \alpha u$ , and  $\varpi + \alpha v$ , together with the elevation of a molecule of air above this surface; the part of  $\alpha s'$  which depends on the variation of the angles  $\theta$  and  $\varpi$ , † being of the order  $\frac{\alpha n^2 \cdot u}{g}$ , may be neglected in

N N

\* Dividing this equation by  $r^2$  ( $\rho$ ) we shall obtain

$$\frac{\rho'}{(\rho)} \left( = \frac{y}{l} \right) = - \left( \frac{du'}{d\theta} \right) - \left( \frac{dv'}{d\varpi} \right) - \frac{u' \cos. \theta}{\sin. \theta} - \left( \frac{d.r^2 s'}{r^2 dr} \right).$$

† The part of  $\alpha s'$  which corresponds to the variations  $\alpha u'$ ,  $\alpha v'$ , is of the same order as the products of these quantities by the eccentricity of the spheroid, see page 258, and the eccentricity in this case is proportional to the fraction  $\frac{n^2}{g}$ , consequently the variation of  $\alpha s'$

which corresponds to the variation of the angles  $\theta$  and  $\varpi$ ,  $= \frac{\alpha n^2 u}{g}$ ; the entire variation of  $\alpha s'$  is made up of two parts, of which one is equal to the elevation of the molecule above the surface of equilibrium, on the supposition that the angles  $\theta$  and  $\varpi$  are not varied, and this part of the variation of  $\alpha s' = \alpha \phi$ , the other part of the variation is the part which corresponds to the variations  $\alpha u'$  and  $\alpha v'$  of the angles  $\theta$  and  $\varpi$ , and from what precedes it appears that this part may be neglected, consequently we have

$$\alpha s' = \alpha \phi; \left( \frac{d.r^2 s'}{r^2 dr} \right) = \frac{2s'}{r} + \frac{ds'}{dr}$$

the second term  $= \left( \frac{d\phi}{dr} \right)$  by substituting  $\phi$  in place of  $y'$ , to which it is equal, and when  $\phi$  is supposed to be equal to  $y$ ; its derivative function with respect to  $r$  must vanish,  $\phi$  being the same for all the molecules, situated on the same radius,  $y$  is the same order as  $s'$ , or the eccen-

the preceding expression for  $y'$ , consequently it may be supposed in this expression that  $s' = \varphi$ ; by making  $\varphi = y$ , we shall have  $\left(\frac{d\varphi}{dr}\right) = 0$ , since the value of  $\varphi$  is then the same for all molecules situated on the same radius. Moreover, by what precedes,  $y$  is of the order  $l$  or  $\frac{n^2}{g}$ ; therefore the expression for  $y'$  will become,

$$y' = -l \cdot \left\{ \left\{ \frac{du'}{d\theta} \right\} + \left\{ \frac{dv'}{d\omega} \right\} + \frac{u' \cdot \cos. \theta}{\sin. \theta} \right\};$$

thus,  $u'$  and  $v'$  being the same for all molecules situated primitively on the same radius, the value of  $y'$  will be the same for all these molecules. Moreover, it is manifest from what has been stated that the quantities

$$2nr \cdot \delta\omega \cdot \sin. {}^2 \theta \cdot \left\{ \frac{ds'}{dt} \right\}, \text{ and } n^2 r \cdot \sin. {}^2 \theta \cdot \delta(s' - (s')),$$

may be neglected in the preceding equations of the motion of the atmosphere, which can then be satisfied, by supposing that  $u'$  and  $v'$  are the same for all the molecules of the atmosphere, which at the commencement of the motion existed on the same radius; therefore the supposition that all those molecules remain constantly on the same radius during the oscillations, is compatible with the equations of the motion and of the continuity of the atmospheric fluid. In this case, the oscillations of the different strata of level are the same, and may be determined by means of the equations,

tricity which is proportional to  $\frac{n^2}{g}$ , and this last quantity is proportional to  $l$ , see page 258 and 266, ∴ we may neglect both  $\frac{2s'}{r}$  and  $\frac{ds'}{dr}$  consequently we will obtain for  $y'$  the expression given in the text. It is manifest from what has been stated in notes to page 253, that  $2nr \cdot \delta\omega \cdot \sin. {}^2 \theta \cdot \left( \frac{ds'}{dt} \right); n^2 \cdot r \sin. {}^2 \theta \cdot \delta(s' - (s))$  may be neglected when the earth is nearly spherical.

$$\begin{aligned}
 & r^2 \delta \theta \left\{ \left\{ \frac{d^2 u'}{dt^2} \right\} - 2n \sin \theta \cos \theta \left\{ \frac{dv}{dt} \right\} \right\} \\
 & + r^2 \delta \varpi \left\{ \sin^2 \theta \left\{ \frac{d^2 v'}{dt^2} \right\} - 2n \sin \theta \cos \theta \left\{ \frac{du}{dt} \right\} \right\} = \delta V - g \cdot \delta y' - g \cdot \delta y; \\
 y' & = -l \left\{ \left\{ \frac{du}{d\theta} \right\} + \left\{ \frac{dv}{d\varpi} \right\} + \frac{u \cos \theta}{\sin \theta} \right\}.
 \end{aligned}$$

These oscillations of the atmosphere ought to produce corresponding oscillations, in the heights of the barometer. In order to determine these last by means of the first, we should suppose a barometer fixed at any elevation above the level of the sea. The altitude of the mercury is proportional to the pressure which the surface exposed to the action of the air experiences ; therefore it may be represented by  $\lg. \rho$  ; but this surface is successively exposed to the action of different strata of level, which are alternately elevated and depressed like the surface of the sea ; thus the value of  $\rho$  at the surface of the mercury varies, 1st, \* because it appertains to a stratum of level, which in the state of equilibrium was less elevated by the quantity  $\alpha y$  ; 2dly, because the density of a stratum increases in the state of motion, by  $\alpha \rho'$  or by  $\frac{\alpha(\rho) \cdot y'}{l}$ . In consequence of the first cause, the variation of  $\rho$  is augmented by the quantity  $-\alpha y \cdot \left( \frac{d\rho}{dr} \right)$  or  $\frac{\alpha(\rho) \cdot y}{l}$  therefore the entire variation of the density  $\rho$  at the surface of the mercury, is  $\alpha(\rho) \cdot \frac{(y+y')}{l}$ . It follows from this, that if we represent the height of the mercury, in

\*  $\left( \frac{dp}{dr} \right) = \lg. \left( \frac{d\rho}{dr} \right)$ , in the state of equilibrium  $\left( \frac{dp}{dr} \right) = g \cdot (\rho)$  see (page 223)  $\therefore \left( \frac{d\rho}{dr} \right) = \frac{g \cdot (\rho)}{g \cdot l}$  consequently  $-\alpha y \cdot \left( \frac{d\rho}{dr} \right) = \frac{\alpha y \cdot (\rho)}{l}, \left( \frac{d\rho}{dr} \right)$  is negative because the density increases as we ascend in the atmosphere.

The temperature of the air being supposed to remain unvaried, its specific gravity will vary as  $(\rho)$  its density, and this quantity varies as  $k$ .

the barometer, in the state of equilibrium by  $k$ ; its oscillations, in the state of motion will be represented by the function  $\frac{\alpha k.(y+y')}{l}$ ; consequently at all heights above the level of the sea, these oscillations are similar, and proportional to the altitudes of the barometer.

It only now remains, in order to *determine* the oscillations of the sea, and of the atmosphere, to know the forces which act on these respective fluids, and to integrate the preceding differential equations; which will be done in the sequel of this work.

END OF THE FIRST BOOK.

A

**TREATISE**  
OF  
**CELESTIAL MECHANICS,**  
BY P. S. LAPLACE,  
MEMBER OF THE NATIONAL INSTITUTE, &c.

---

PART THE FIRST—BOOK THE SECOND.

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TRANSLATED FROM THE FRENCH, AND ELUCIDATED WITH  
*EXPLANATORY NOTES.*

---

BY THE REV. HENRY H. HARTE, F.T.C.D. M.R.I.A.

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*The mean longitude of the first satellite, minus three times that of the second, plus twice that of the third, is constantly equal to two right angles.*

These theorems subsist notwithstanding any change which the mean motions of the satellites may undergo, either from a cause similar to what alters the mean motion of the moon, or from the resistance of a very rare medium. These theorems give rise to an arbitrary inequality, which only differs for each of the three satellites by the magnitude of its coefficient, and which according to observations is insensible. - - No. 66

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No. 73



## ERRATA.

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Page Line

- 20, 3, for This read The.  
28, 7, for  $(z''+z')^2$ , read  $(z''-z')^2$ .  
34, 6, for  $mm$ , read  $mm'$ .  
50, 19, for from the  $M$ , read from  $M$ .  
51, 12, for their, read its.  
52, 16, for  $z-z$ , read  $z-z'$ .  
62, 11, for its, read these.  
68, 5, for  $r \left( \frac{d^2 U}{du} \right)$ , read  $r \left( \frac{d^2 U}{du^2} \right)$ .  
81, last line, for the second  $\frac{\delta}{2}$ , read  $\frac{\delta}{2}$ .  
96, 8, for supply, read solely.  
96, 19, for  $\frac{dy}{dx}$ , read  $\frac{dy_i}{dx}$ .  
103, 1, for  $e$ , read  $c$ .  
143, 17, for cos.  $\epsilon n$ , read cos.  $\epsilon nt$ .  
152, 19, for  $u_i$ , read  $v$ .  
163, 3, for tan.  $\frac{1}{2}v$ , read tan.  $\frac{1}{2}v$ .  
166, 17, for value, read ratio.  
174, 10, for cos.  $\epsilon$ . cos.  $\epsilon$ , read cos.  $\epsilon$ . cos.  $\epsilon'$ .  
174, 11, for  $e$ , read  $c$ .  
174, 20, for sin.  $u$ . sin.  $u'^2$ , read sin.  $u$ . sin.  $u'$ .  
216, 1, for  $\delta\epsilon''$ , read  $\delta\epsilon'''$ .  
219, 1, for  $\frac{2}{r}$ , read  $-\frac{2}{r}$ .  
224, 20, for  $U-V'$ , read  $U'-V'$ .  
240, 2, for the second  $aQ$ , read  $aQ'$ .  
244, 11, for these, read the.  
256, 4, for  $-0$ , read  $=0$ .  
266, 1, for  $dR$ , read  $dR$ .  
271, 4, for  $dt'$ , read  $dt^*$ .  
284, 3, for  $a^2$ , read  $a^2$ .

ERRATA.

Page Line

285, 11, for  $a = \frac{a}{a'}$ , read  $a = \frac{a}{a'}$ .

287, 13, for  $\frac{1}{a'} - \frac{d\zeta^{(i)}}{da} \left( \frac{d\alpha}{da} \right)$ , read  $\frac{a}{a'^2} - \frac{1}{a'} \cdot \zeta^{(1)}$ .

287, 16, dele — before  $\frac{1}{a'^2}$ .

299, 2, for  $n+i$ , read  $nt+i$ .

300, 7, for  $e \cdot \cos. \varpi'$ , read  $e' \cos. \varpi'$ .

306, 5, for  $m'$ , read  $m$ .

315, 1,  $\int \chi'' dr$ , read  $\int \chi'' dR$ .

318, 5, for motion, read motions.

323, 14, for  $m' \sqrt{a}$ , read  $m' \sqrt{a'}$ .

328, 20, for the second  $\zeta_2 - \zeta$ , read  $\zeta_2 - \zeta_1$ .

380, 2, for in  $u$ , read  $(u)$ .

A  
TREATISE  
ON  
*CELESTIAL MECHANICS,*  
*&c. &c.*

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PART I.—BOOK II.

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OF THE LAW OF UNIVERSAL GRAVITATION, AND OF THE MOTIONS  
OF THE CENTRES OF GRAVITY OF THE HEAVENLY BODIES.

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CHAPTER I.

*Of the law of universal gravitation, deduced from the phenomena.*

1. AFTER having developed the laws of motion, we proceed to deduce from these laws, and from those of the celestial motions, which have been given in detail in the work entitled the *Exposition of the System of the World*, the general law of these motions. Of all the phenomena, that which seems most proper, to discover it, is the elliptic motion of the planets and of the comets round the sun, let us therefore consider what this law furnishes us with on the subject. For this purpose, let

PART. I.—BOOK II.

\* B

## CELESTIAL MECHANICS,

$x$  and  $y$  represent the rectangular coordinates of a planet, in the plane of its orbit, their origin being at the centre of the sun; moreover, let  $P$  and  $Q$  represent the forces with which the planet is actuated in its relative motion round the sun, parallel to the axes of  $x$  and of  $y$ , these forces being supposed to tend towards the origin of the coordinates; finally, let  $dt$  represent the element of the time which is supposed to be constant; by the second chapter of the first book,\* we shall have

$$0 = \frac{d^2x}{dt^2} + P; \quad (1)$$

$$0 = \frac{d^2y}{dt^2} + Q. \quad (2)$$

If we add the first of these equations multiplied by  $-y$ , to the second multiplied by  $x$ , the following equation will be obtained:

$$0 = \frac{d(xdy - ydx)}{dt^2} + xQ - yP.$$

It is evident that  $x dy - y dx$  is equal to twice the area which the radius vector of the planet describes about the sun during the instant  $dt$ ; by the first law of Kepler this area is proportional to the time, consequently we have

$$xdy - ydx = cdt,$$

$c$  being a constant quantity; hence it appears, that the differential of the first member of this equation is equal to cypher, which gives

$$xQ - yP = 0,$$

\* These laws refer strictly to the motion of the centre of gravity of each planet; it is therefore the motion of this point which is determined, and by the position and velocity

it follows from this, that the forces  $P$  and  $Q$  are to each other in the ratio of  $x$  to  $y$ ; and consequently their resultant must pass through the origin of the coordinates, that is, through the centre of the sun, and as the curve which the planet describes is\* concave towards the sun, it is evident that the force which acts on it, must tend towards this star.

The law of the areas, proportional to the times employed in their description, leads us therefore to this first remarkable result, namely, that the force which solicits the planets and comets, is directed towards the centre of the sun.

2. Let us in the next place, determine the law according to which this force acts at different distances from this star. It is evident that as the planets and the comets alternately approach to and recede from the sun, during each revolution, the nature of the elliptic motion ought to conduct us to this law. For this purpose, let the differential equations (1) and (2) of the preceding number be resumed. If we add the first, multiplied by  $dx$ , to the second, multiplied by  $dy$ , we shall obtain

$$0 = \frac{dx \cdot d^2x + dy \cdot d^2y}{dt^2} + Pdx + Qdy;$$

which gives by integrating

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of a planet, we always understand, unless the contrary be specified, the position and velocity of its centre of gravity; hence it is evident, that the equations of the motion of a *material point*, which have been given in the second chapter, are applicable in the present case.

\* The areas being proportional to the times, the curve described is one of single curvature, (see Book I. page 28, Notes), therefore two coordinates ( $x, y$ ) are sufficient to determine the circumstances of the planet's motion. As the curve described by the planet is *concave* to the sun, it is plain that in the equation  $\frac{d^2x}{dt^2} = P$ ;  $\frac{d^2y}{dt^2}$  must be taken negatively, because the force tends to diminish the coordinates. See Book I. Chapter II. page 31.

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$$0 = \frac{dx^2 + dy^2}{dt^2} + 2f(Pdx + Qdy), *$$

the arbitrary constant being indicated by the sign of integration. Substituting instead of  $dt$ , its value  $\frac{xdy - ydx}{c}$ , which is given by the law of the proportionality of the areas to the time, we shall have

$$0 = \frac{c^2 \cdot (dx^2 + dy^2)}{(xdy - ydx)^2} + 2f(Pdx + Qdy).$$

For greater simplicity, let us transform the coordinates  $x$  and  $y$ , into a radius vector, and a traversed angle, conformably to the practice of astronomers. Let  $r$  represent a radius drawn from the centre of the sun to that of the planet, or its radius vector; and let  $v$  be the angle which it makes with the axis of  $x$ , we shall have then,

$$x = r \cdot \cos. v; \quad y = r \cdot \sin. v; \quad r = \sqrt{x^2 + y^2}; \dagger$$

from which may be obtained,

$$dx^2 + dy^2 = r^2 \cdot dv^2 + dr^2; \quad xdy - ydx = r^2 dv.$$

If the *principal* force which acts on the planet be denoted by  $\varphi$ , we shall have by means of the preceding number,

$$P = \varphi \cdot \cos. v; \quad Q = \varphi \cdot \sin. v; \quad \varphi = \sqrt{P^2 + Q^2};$$

which gives

$$Pdx + Qdy = \varphi dr;$$

\* The equation  $0 = \frac{dx^2 + dy^2}{dt^2} + 2f(Pdx + Qdy)$ , has been already deduced in No 8; by substituting for  $dx^2$  and  $dy^2$  their values in terms of the polar coordinates, we obtain  $\frac{dr^2}{dt^2} + \frac{r^2 dv^2}{dt^2} + 2f\varphi dr = 0$ ; hence if  $\varphi$  be given in terms of  $r$  we shall immediately obtain the velocity at any distance from the centre of force.

† The most obvious way of determining the position of any body, is by means of rectangular coordinates, in which case the differential equations of motion are symmetrical; however, as the polar coordinates involve directly the quantities which are required to be known in astronomical investigations, namely, the distance, longitude and latitude of a planet, astronomers make use of these coordinates in determining the circumstances of its motion, &c.

and by substitution we shall have

$$0 = \frac{c^2 \cdot (r^2 \cdot dv^2 + dr^2)}{r^4 dv^2} + 2 \int \phi dr ; *$$

$\bullet dx = dr \cdot \cos v - dv \cdot \sin v \cdot r, dy = dr \cdot \sin v + dv \cdot \cos v \cdot r, \therefore dx^2 + dy^2 = dr^2 \cdot (\cos^2 v + \sin^2 v) - 2dr \cdot dvr \cdot \sin v \cdot \cos v + 2dr \cdot dvr \cdot \sin v \cdot \cos v + dv^2 \cdot r^2 \cdot (\sin^2 v + \cos^2 v) = dr^2 + dv^2 \cdot r^2; xdy = r \cdot \cos v \cdot (dr \cdot \sin v + rdv \cdot \cos v) = rdr \cdot \sin v \cdot \cos v + dv \cdot r^2 \cdot \cos v, ydx = r \cdot \sin v \cdot (dr \cdot \cos v - r \cdot dv \cdot \sin v) = rdr \cdot \sin v \cdot \cos v - dvr^2 \cdot \sin^2 v, \therefore xdy - ydx = r^2 \cdot dv; Pdx = \phi \cdot \cos v \cdot (dr \cdot \cos v - rdv \cdot \sin v); Qdy = \phi \cdot \sin v \cdot (dr \cdot \sin v + rdv \cdot \cos v), \therefore Pdx + Qdy = \phi dr \cdot (\cos^2 v + \sin^2 v) + \phi dv \cdot (r \cdot \cos v \cdot \sin v - r \cdot \cos v \cdot \sin v) = \phi dr; \text{ therefore by substituting in the equation } \frac{c^2(dx^2 + dy^2)}{(xdy - ydx)^2} + 2 \int (Pdx + Qdy) = 0,$   
 we obtain  $\frac{c^2(r^2 dv^2 + dr^2)}{r^4 dv^2} + 2 \int \phi dr = 0$ ; and  $\therefore (-c^2 r^2 - r^4 \cdot 2 \int \phi dr) \cdot dv^2 = c^2 dr^2$ ; as the

variables  $dv$  and  $dr$  are separated in the equation  $dv = \frac{cdr}{r \sqrt{-c^2 - 2r^2 \int \phi dr}}$ , it can be integrated and constructed, the radical ought to be affected with the sign  $\pm$ , when  $v$  and  $r$  increase the same time, the sign is  $+$ , and in the contrary case the sign is  $-$ ; these circumstances depend on the initial impulse of the planet. The determination of  $v$ , or of the orbit described by a body, when the law of the force  $\phi$  is given, is called the inverse problem of central forces, the expression for  $dv$  coincides with that given by Newton in Prop. 41, Lib. 1st. Princip. for it is there demonstrated that  $XY \cdot XC =$

$$\frac{Q \cdot IN \cdot CX^2}{A^2 \sqrt{ABTD - Z^2}}, \text{ from the construction it is evident that } \frac{XY}{XC} = dv, \text{ that } IN = dr,$$

that  $Q = c$ , and finally that  $A = r$ , and as  $Z^2 \propto \frac{Q^2}{A^2}$ , and  $ABTD =$  the square of

$$\text{the velocity}, \sqrt{ABTD - Z^2} = \sqrt{-\int \phi dr - \frac{c^2}{r^2}} \therefore \frac{XY}{XC} = dv = \frac{Q \cdot IN}{A^2 \sqrt{ABTD - Z^2}}$$

$$= \frac{cdr}{r^2 \sqrt{-c^2 - 2r^2 \int \phi dr}} \div \text{by } r.$$

If the force  $\phi$  be as any power  $n$  of the distance, then  $2 \int \phi dr = 2 \int r^n dr$  ( $=$  the square of the velocity)  $= b^2 + \frac{2}{n+1} \cdot r^{n+1} - \frac{2}{n+1} \cdot a^{n+1}$  ( $a$  being the initial distance), hence

$$dv = \frac{cdr}{r \sqrt{-c^2 - b^2 r^2 + \frac{2}{n+1} \cdot r^{n+3} + \frac{2}{n+1} \cdot a^n r_2}}, \text{ as } b \text{ is the velocity of projec-}$$

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from which we may obtain,

$$dv = \frac{cdr}{r \cdot \sqrt{-c^2 - 2r^2 \int \varphi dr}}. \quad (3)$$

This equation will give by the method of quadratures, the value of  $v$  in terms of  $r$ , when  $\varphi$  is a known function of  $r$ , but if, this force being unknown, the nature of the curve which it makes the planet describe, be given, then, by differentiating the preceding expression of  $2 \int \varphi dr$ , we shall have, to determine  $\varphi$ , the equation

tion, if  $p$  be the perpendicular on the tangent at this point,  $c \propto p b$ , and  $b^2 = m^2$

$$a^{n+1}, \because dv = \frac{pbdr}{r \cdot \sqrt{-(p^2 + r^2)m^2 a^{n+1} - \frac{2}{n+1} \cdot r^{n+3} + \frac{2}{n+1} \cdot a^{n+2} r^2}}, \text{ at the apsides}$$

$$p=a, dr=0, \text{ and } \because r = \frac{pb}{\text{vel.}} = \frac{pb}{\sqrt{b^2 + \frac{2}{n+1} \cdot (a^{n+1} - r^{n+1})}}, \text{ hence}$$

$$r \cdot \sqrt{b^2 + \frac{2}{n+1} \cdot (a^{n+1} - r^{n+1})} - pb = 0, \text{ by squaring this equation, we get } b^2 r^2 + \frac{2}{n+1} \cdot a^{n+1} r^2 - \frac{2}{n+1} \cdot r^{n+3} - p^2 b^2 = 0.$$

When  $n$  is even, this equation may have four possible roots, when it is odd, it can only have three; but as this equation is the square of the given equation, some of the roots are introduced by the operation, so that the equation to the apsides can never have more than two possible roots, consequently no orbit can have more than two apsides, *i. e.* there are only two different distances of the apsides, but there is no limit to the number of repetitions of these, without again falling on the *same* points, if  $n = -3$  or a greater negative number, the equation can have only one possible root, and the orbit but one apsid.

If in the equation  $\frac{c^2}{z^2} + \frac{dr^2}{r^4 \cdot dv^2} + 2 \int \varphi dr \cdot \frac{1}{z}$  be substituted in place of  $r$ , it becomes  $c^2 \cdot \left( \frac{dz^2}{dv^2} + z^2 \right) - 2 \int \varphi \cdot \frac{dz}{z^2}$ , which is a much more convenient form, particularly when the

$$\varphi = \frac{c^2}{r^3} - \frac{c^2}{2} \cdot d \cdot \underbrace{\left\{ \frac{dr^2}{r^4 dv^2} \right\}}_{dr} \quad (4)*$$

The orbits of the planets are ellipses, having the centre of the sun in one of the foci; if, in the ellipse,  $\omega$  represents the angle which the axis major makes with the axis of  $x$ , moreover if  $a$  represents the semiaxis major, and  $e$  the ratio of the excentricity to the semiaxis major, we shall have, the origin of the coordinates being in the focus,

$$r = \frac{a(1-e^2)}{1+e \cos(v-\omega)}, \dagger$$

which equation becomes that of a parabola, when  $e=1$ , and  $a$  is infinite, it appertains to an hyperbola, when  $e$  is greater than unity.

law of the force being given, the nature of the orbit is required; for instance the equation in page 5 becomes, when  $\frac{1}{z}$  is substituted for  $r$  then differentiated, and the result divided by  $2dz$ ,

$$c^2 \left( \frac{d^2 z}{dv^2} + z^2 \right) - \frac{\varphi}{z^2} = 0, \because \varphi = c^2 z^2 \left( \frac{d^2 z}{dv^2} + z^2 \right), \text{ and as } z = \frac{1}{r} = \frac{1+e \cos(v-\omega)}{a(1-e^2)}, \therefore$$

$$\text{differentiating twice } \frac{d^2 z}{dv^2} = -\frac{e \cos(v-\omega)}{a(1-e^2)}, \therefore \frac{d^2 z}{dv^2} + z^2 = \frac{1}{a(1-e^2)}, \therefore \varphi = \frac{c^2 z^2}{a(1-e^2)}.$$

$$* \quad \frac{c^2 \cdot (r^2 \cdot dv^2 + dr^2)}{r^4 \cdot dv} = \frac{c^2}{r^2} + \frac{c^2 \cdot dr^2}{r^4 \cdot dv^2} = -2f \varphi dr, \because \text{by differentiating and dividing by}$$

$$\text{by } dr \text{ we obtain } \frac{c^2}{r^3} - d \cdot \left( \frac{c^2 dr^2}{r^4 dv^2} \right) = \varphi.$$

$\dagger$  The greatest and least values of  $r$  correspond to  $v-\omega=\pi$ ,  $v-\omega'=0$ ,  $\therefore$  they are respectively  $a(1+e)$ ,  $a(1-e)$ , consequently they lie in directum; hence it is easy to perceive, that when  $\varphi$  varies as  $\frac{1}{r^2}$ , the apsides are  $180^\circ$  distant, and *vice versa*.

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This equation gives

$$\frac{dr^2}{r^4 dv^2} = \frac{2}{ar.(1-e^2)} - \frac{1}{r^2} - \frac{1}{a^2.(1-e^2)},$$

and consequently

$$\phi = \frac{c^2}{a.(1-e^2)} \cdot \frac{1}{r^2};$$

therefore, the orbits of the planets and comets being conic sections, the force  $\phi$  is reciprocally proportional to the square of the distance of the centres of these stars from that of the sun.

Moreover we may perceive, that, if the force  $\phi$  be inversely as the square of the distance, or expressed by  $\frac{h}{r^2}$ ,  $h$  being a constant coefficient, the preceding equation of conic sections, will satisfy the differential equation (4) between  $r$  and  $v$ , which gives the expression of  $\phi$ , when  $\phi$  is changed into  $\frac{h}{r^2}$ . We have then  $h = \frac{c^2}{a.(1-e^2)}$ , which

$$\frac{1}{r} = \frac{1+e. \cos.(v-\omega)}{a.(1-e^2)}, \therefore \frac{dr^2}{r^4 dv^2} = \left( \frac{e. \sin.(v-\omega)}{a.(1-e^2)} \right)^2, \text{ and } \frac{a.(1-e^2)}{r} - 1 = e.$$

$$\cos.(v-\omega) \therefore \left( \frac{a.(1-e^2)}{r} \right)^2 - \frac{2a.(1-e^2)}{r} + 1 = e^2. \cos.(v-\omega)^2 = e^2 - e^2. \sin.$$

$$(v-\omega), \therefore \frac{dr^2}{r^4 dv^2} = -\frac{1}{r^2} + \frac{2}{a.(1-e^2)} \times \frac{1}{r} - \frac{1}{a^2.(1-e^2)}, \text{ and the differential of the second member divided by } dr \text{ will be equal to } \frac{2}{r^3} - \frac{2}{a.(1-e^2)} \cdot \frac{1}{r^2}, \text{ consequently we have}$$

the value of

$$\phi = \frac{c^2}{r^3} - \frac{c^2}{2} \cdot d \cdot \left( \frac{dr^2}{r^4 dv^2} \right) = \frac{c^2}{r^3} - \frac{c^2}{r^3} + \frac{c^2}{a.(1-e^2)} \cdot \frac{1}{r^2}.$$

forms an equation of condition between the two arbitrary quantities  $a$  and  $e$ , of the equation of a conic section; therefore the three arbitrary quantities  $a$ ,  $c$ , and  $\varpi$ , of this equation, are reduced two distinct quantities, and as the differential equation between  $r$  and  $v$ , is only of the second order, the finite equation of conic sections is its complete integral.\*

From what precedes, it follows, that, if the curve described is a conic section, the force is in the inverse ratio of the square of the distance, and conversely, if the force be inversely as the square of the distance, the curve described is a conic section.

3. The intensity of the† force  $\phi$ , with respect to *each* planet and comet depends on the coefficient  $\frac{c^3}{a(1-e^2)}$ ; the laws of Kepler furnish us with the means of determining it. In fact, if we denote the time of the revolution of a planet by  $T$ ; the area, which its radius vector describes during this time, being the surface of the planetary ellipse, it

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\* Conversely, when  $\phi = \frac{h}{r^2}$ , the preceding equation of conic sections will satisfy the differential equation (4) between  $r$  and  $v$ , and  $h$  becomes  $= \frac{c^2}{a.(1-e^2)}$ , ∵ the three arbitrary quantities are reduced to two distinct ones, and this is the required number of arbitrary quantities, for the differential equation between  $r$  and  $v$  being of the second order, the number of arbitrary quantities introduced by the double integration is two, so that the equation of conic sections is the complete integral of this differential equation.

† The two first laws of Kepler, are sufficient to determine the ratio which exists between the intensities of the action of the sun on each planet, at different distances of the planet from the sun; by means of the third law we are enabled to find the relations which exist between the respective actions of the sun on *different* planets. As  $\frac{c^2}{a.(1-e^2)}$ , which expresses the intensity of the force for each planet, at the unity of its distance from the sun, depends on the three quantities  $a$ ,  $e$ ,  $c$ , which have particular values for each planet, we cannot determine without the third law, whether it changes, or remains the same, in passing from one planet to another.

will be  $\pi \cdot a^2 \cdot \sqrt{1-e^2}$ ,\*  $\pi$  being the ratio of the semicircumference to the radius; but, by what precedes, the area described during the instant  $dt$ , is equal to  $\frac{1}{2}cdt$ ; therefore the law of the proportionality of the areas to the times of describing them, will give the following proportion:

$$\frac{1}{2}cdt : \pi a^2 \cdot \sqrt{1-e^2} :: dt : T:$$

consequently

$$c = \frac{2\pi \cdot a^2 \cdot \sqrt{1-e^2}}{T}.$$

With respect to the planets, the law of Kepler, according to which the squares of the times of their revolutions, are as the cubes of the greater axes of their ellipses, gives  $T^2 = k^2 \cdot a^3$ ,  $k$  being the same for all the planets; therefore, we have

$$c = \frac{2\pi \cdot \sqrt{a \cdot (1-e^2)}}{k}.$$

$2a \cdot (1-e^2)$  is the parameter of the orbit, and in different orbits, the values of  $c$  are proportional to the areas, described by the radii vectores in equal times; therefore these areas are as the square roots of the parameters of the orbits.

This proportion obtains also, for the orbits described by the comets, compared either among themselves, or with the orbits of the planets; this is one of the fundamental points of their theory, which corresponds so exactly to all their observed motions. The greater axes of their orbits, and the times of their revolutions, being unknown, we compute the motion of these stars, on the hypothesis that it is performed in a

\* The area of the ellipse being equal to that of a circle, whose radius is a mean proportional between the semiaxes  $a$  and  $a\sqrt{1-e^2}$ ; it must be equal to  $\pi a^2 \cdot \sqrt{1-e^2}$ .

parabolic orbit, and expressing their perihelion distance by  $D$ ,<sup>\*</sup> we suppose  $c = \frac{2\pi\sqrt{2D}}{k}$ , which is equivalent to making  $e$  equal to unity, and  $a$  infinite, in the preceding expression of  $c$ ; consequently, we have relatively to the comets,  $T^2 = k^2 \cdot a^3$ , so that we can determine the greater axes of their orbits, when the periods of their revolution are known.

The expression for  $c$  gives,

$$\frac{c^2}{a \cdot (1-e^2)} = \frac{4\pi^2}{k^2};$$

therefore we have

$$c = \frac{4\pi^2}{k^2} \cdot \frac{1}{r^2}.$$

c 2

\* The polar equation of the parabola is  $r = \frac{p}{1 + \cos(v - \omega)}$ ; ∵ when  $v - \omega = 0$ , i. e. at the perihelion,  $r = \frac{a(1-e^2)}{2} = \frac{p}{2} = D$ , ∵  $a(1-e^2) = 2D$ . Now this is the same thing, as if  $a$  was made infinite, and  $e =$  to unity, in the equation,  $r = a \cdot \frac{(1-e^2)}{2}$ , which expresses the distance of the nearest apsis from the focus of the ellipse, for substituting for the eccentricity its value  $\sqrt{a^2 - b^2}$ ,  $r$  becomes equal to  $a \left( \frac{1}{2} - \frac{\sqrt{a^2 - b^2}}{2a} \right)$  = as ( $b^2 = ap$ )  $\frac{a(a - \sqrt{a^2 - ap})}{2a}$ , and as  $\sqrt{a^2 - ap} = a - \frac{p}{2} + (\dots)$ .  $\frac{1}{a} =$  when  $a$  is infinite  $a - \frac{p}{2}$ ,  $r = \frac{a(a - a + \frac{p}{2})}{2a} = \frac{p}{4}$ , and it is evident that  $e$  is equal in this case to unity. ∵ If we suppose that the synchronous areas are as the square roots of the parameters, or  $c = \frac{2\pi\sqrt{2D}}{k}$ , we will have  $\frac{2\pi\sqrt{2D}}{2k} \cdot dt : \pi a^{\frac{3}{2}} \sqrt{2D} :: dt : T$ ; ∵  $T^2 = k^2 \cdot a^3$ .

† The constant ratio which  $c$  bears to the square root of  $2D$ , is that of  $2\pi : k$ , which is the same for all the planets;  $\frac{4\pi^2}{k^2}$ , or  $\frac{c^2}{a(1-e^2)}$  is the value of the

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The coefficient  $\frac{4\pi^2}{k^2}$ , being the same for all the planets and comets, it

force  $\phi$  at the unity of the distance of a planet from the sun. The *accelerating* force of the planets being the same at equal distances from the sun, it follows that the *moving* force will be proportional to the mass; and if all the planets descended at the same instant, and without any initial velocities from different points of the same spheric surface, of which the centre coincided with that of the sun, they would arrive at the surface of the sun, being supposed spheric, in the same time; here, we may perceive, a remarkable analogy between this force and the terrestrial gravity, which also impresses the same motion, on all bodies situated at equal distances from its centre.

If the apparent diameter of the sun be observed accurately with a micrometer, it will be found to vary in the subduplicate ratio of his angular velocity; from this phenomenon the equable description of areas may be inferred; for as the apparent diameters of the sun are inversely as the distance of the sun from the earth, the angular velocity of the sun must be inversely as the square of the distance of the sun from the earth, therefore the product of the diurnal motion into the square of the distance, *i. e.* the small area must be constant. If the sun's mean apparent diameter be called  $m$ , and his least apparent diameter  $m-n$ , his apparent diameter at any other time, will be  $m-n \cos. z$ ,  $z$  being the angular distance of the sun from the point where his diameter is least, hence it may be inferred, that the orbit is elliptic;

for as the distance is inversely as the apparent diameter,  $r = \frac{B}{m-n \cos. (v-\omega)}$ , when  $r$  is greatest,  $v-\omega=0$ , when least  $v-\omega=\pi$ ,  $\therefore mr=nr \cos. (v-\omega) = x(m-n)$ ,  $x$  being the greatest distance, and  $mr = x(m-n) + nr. (\cos. v-\omega)$ , let  $(m-n) \cdot x = nx'$ , and then  $mr = n(r. \cos. (v-\omega) + x')$ ,  $\therefore m:n :: r. \cos. (v-\omega) + x' : r$ ; now  $r. (\cos. (v-\omega))$  is equal to a part of the axis intercepted between a perpendicular let fall from the sun's place on this axis, and the place the earth is supposed to occupy, and  $x'$  is a constant quantity,  $\therefore$  producing the axis in an opposite direction from the sun, till the distance from the earth is equal to  $x'$ , and erecting a perpendicular to the produced axis at the extremity of its production,  $x' + r \cos. (v-\omega)$  is equal to the distance of the sun from this perpendicular, and as it is to  $r$  the distance of the sun from the earth, in a given ratio of major inequality, namely  $m:n$ , it follows that the curve is an ellipse of which the directrix is a perpendicular, erected at the extremity of  $x'$ . This conclusion might also have been inferred from the

$$\text{polar equation to the ellipse } r = \frac{a(1-e^2)}{1+e \cos. (v-\omega)} = a(1-e^2) \cdot (1+e \cos. (v-\omega))^{-1}.$$

Kepler directed his observations to the planet of Mars, of which the motion appeared to be more irregular, than the motion of the other planets, and by determining several distances of the planet from the sun, and tracing the orbit which passes through them all, it will appear that this orbit must be an ellipse, of which the sun occupies one of the foci, it

follows that for each of these bodies, the force  $\varphi$ , is inversely as the square of the distance from the centre of the sun, and that it only varies from one planet to another, in consequence of the change of distance; from which it follows that it is the same for all these bodies supposed at equal distances from the sun.

We are thus conducted, by the beautiful laws of Kepler, to consider the centre of the sun as the focus of an attractive force, which, decreasing in the ratio of the square of the distance, extends indefinitely in every direction. The law of the proportionality of the areas to the times of their description, indicates that the principal force which solicits the planets and comets, is constantly directed towards the centre of the sun; the ellipticity of the planetary orbits, and the motions of the comets which are performed in orbits, which are very nearly parabolic, prove, that for each planet and for each comet, this force is in the inverse ratio of the square of the distance of these stars from the sun; finally, from the law of the squares of the periodic times proportional, to the cubes of the greater axes of their orbits, i. e. from the proportionality of the areas traced in equal times by the radii vectores in different orbits, to the square roots of the parameters of these orbits, which law involves the preceding, and is applicable to comets; it follows, that this force is the same for all the planets and comets, placed at equal distances from the sun, so that in *this* case, these bodies would fall towards the sun, with equal velocities.

4. If from the planets we pass to the consideration of the satellites, we find that the laws of Kepler being very nearly observed in their motions about their respective primary planets, they must gravitate towards the centres of these planets, in the inverse ratio of the squares of their distances from these centres; they must in like manner gravitate very nearly as their primaries towards the sun, in order that their relative motions about their respective primary planets, may be very nearly the same

can also be shewn that the angular velocities are inversely as the squares of the distances from the sun, from which it follows that the areas are proportional to the times.

as if these planets were at rest. Therefore the satellites are solicited towards their primaries and towards the sun, by forces which are inversely as the squares of the distances. The ellipticity of the orbits of the three\* first satellites of Jupiter is inconsiderable ; but the ellipticity of the fourth satellite is very perceptible. From the great distance of Saturn we have not been able hitherto to recognise the ellipticity of the orbits of his satellites, with the exception of the sixth, of which the orbit appears to be sensibly elliptic. But the law of the gravitation of the satellites of Jupiter, Saturn, and Uranus is principally conspicuous in the relation which exists between their mean motions, and their mean distances from the centre of these planets. This relation consists in this, that for each system of satellites, the squares of the times of their revolutions are as the cubes of their mean distances from the centre of the planet. Therefore let us suppose that a satellite describes a circular orbit, of which the radius  $a$  is equal to its mean distance from the centre of the primary,  $T$  expressing the number of seconds contained in the duration of a sidereal revolution, and  $\pi$  expressing as before the ratio of the semiperiphery to the radius,  $\frac{2.a\pi}{T}$  will be the small arc described by the satellite in a second of time. If, the attractive force of the pla-

\* The frequent recurrence of the eclipses of the satellites, enables us to determine the synodic revolution with great accuracy : and by means of this revolution, and of the motion of Jupiter, we can obtain the periodic time. The hypothesis of the orbits being very nearly circular, in the case of the first and second satellites, is confirmed by the phenomena, for the greatest elongations are always very nearly the same ; besides the supposition of the uniformity of the motions, satisfies very nearly the computations of the eclipses. The distances of the satellites from the centre of Jupiter, may be found, by measuring with a micrometer, their distances from this centre, at the time of their greatest elongation, and also the diameter of Jupiter at this time, by means of which, these distances may be obtained in terms of the diameter ; however they cannot be determined with the same precision as the periods of the satellites. As it is necessary in a comparison of a great number of observations, to modify the laws of circular motion, in the case of the third and fourth satellites, but especially in the case of the fourth, we conclude that the orbits of these satellites are elliptical.

net ceasing, the satellite was no longer retained in its orbit, it would recede from the centre of the planet along the tangent, by a quantity equal to the versed sine of the arc  $\frac{2a\pi}{T}$ , that is by the quantity\*  $\frac{2a\pi^2}{T^2}$ ; therefore this attractive force makes it to descend by this quantity, towards the primary. Relatively to another satellite, of which the mean distance from the centre of the primary is represented by  $a'$ ,  $T'$  being equal to the duration of a sidereal revolution, reduced into seconds, the descent in a second will be equal to  $\frac{2a'\pi^2}{T'^2}$ ; but if we name  $\varphi$ ,  $\varphi'$ , the attractive forces of the planet at the distances  $a$  and  $a'$ , it is manifest, that they are proportional to the quantities by which they make the two satellites to descend towards their primary in a second; therefore we have  $\varphi : \varphi' :: \frac{2a\pi^2}{T^2} : \frac{2a'\pi^2}{T'^2}$ .

The law of the squares of the times of the revolutions, proportional to the cubes of the mean distances of the satellites from the centre of their primary, gives

$$T^2 : T'^2 :: a^3 : a'^3 :$$

From these two proportions, it is easy to infer

$$\varphi : \varphi' :: \frac{1}{a^2} : \frac{1}{a'^2} ;$$

consequently, the forces  $\varphi$  and  $\varphi'$  are inversely as the squares of the distances  $a$  and  $a'$ .

\*  $T : 1'' :: 2a\pi : \text{arc described in a second}$ , on the hypothesis that the motion is uniform, the versed sine of this arc  $= \frac{4a^2\pi^2}{2aT^2}$ . As the orbits of all the satellites are not elliptic, we cannot determine from the nature of the orbits, whether the force for each satellite in particular, varies inversely as the square of the distance or not.

5. The earth having but one satellite, the ellipticity of the lunar orbit is the only phenomenon, which can indicate to us the law of its attractive force; but the elliptic motion of the moon, being very sensibly deranged by the\* perturbating forces, some doubts may exist, whether the law of the diminution of the attractive force of the earth, is in the inverse ratio of the square of the distance from its centre. Indeed, the analogy which exists between this force, and the attractive forces of the sun, of Jupiter, of Saturn, and of Uranus, leads us to think that it follows the same law† of diminution; but the experiments which have been instituted on terrestrial gravity, offer a direct means of verifying this law.

For‡ this purpose, we proceed to determine the lunar parallax, by

\* The orbit of the moon differs sensibly from the *elliptic* form, in consequence of the action of the disturbing forces, and the variation of its apparent diameter shews, that it deviates more from the *circular* form, than the orbit of the sun. The first law of Kepler may be proved to be true, in the case of the moon, in the same manner as for the sun, namely, by a comparison of her apparent motion, with her apparent diameter. Indeed, if great accuracy is required, the observations ought to be made in the syzygies and in the quadratures; for in the other points of the orbit, the disturbing force of the sun deranges the proportionality of the areas to the times employed in their description. See Princip. Math. Lib. 1. Prop. 66. and Lib. 3, Prop. 3 and 29.

† Newton demonstrates that the force which retains the moon in her orbit, is inversely as the square of the distance, in the following manner: if the distance between the apsides was  $180^\circ$ , the force would be inversely as the square of the distance, as has been already pointed out. See Note to page 7.

Now the apsides are observed to advance three degrees and three minutes every month, and the law of the force which would produce such an advance of the apsides, varies inversely as some power of the distance, intermediate between the square and the cube, but which is nearly sixty times nearer to the square; ∵ on the hypothesis, that the progression of the apsides, is produced by a deviation from the law of elliptical motion, the force must vary *very nearly* in the inverse ratio of the square of the distance; but if, as Newton demonstrates, the motion of the apsides arises from the disturbing force of the sun, it follows, *a fortiori*, that the force must be inversely as the square of the distance.

‡ The value of the constant part of the parallax is deduced on the hypothesis, that the force soliciting the moon, is the terrestrial gravity, diminished in the ratio of the square of

means of experiments on the length of the pendulum which vibrates seconds, and to compare it with observations made in the heavens. On the parallel of which the square\* of the sine of the latitude is  $\frac{1}{3}$ , the space through which bodies fall by the action of gravity in a second, is, from observations on the length of the pendulum, equal to 3<sup>metres</sup>, 65548,

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D

the distance ; and if this parallax agrees with the observed parallax corrected for the lunar inequalities, we are justified in inferring, that the diminished terrestrial gravity and the force solliciting the moon are identically the same.

Let unity represent the radius of a sphere equicapacious with a spheroid, its density being supposed to be the same with the mean density of this spheroid; if the greater semiaxis of the spheroid be  $= 1+\epsilon$ , and the lesser  $= 1-s$ , we shall have for the oblong spheroid the following equation,  $\frac{4\pi}{3} \cdot 1^3 = \frac{4\pi}{3} \cdot (1+\epsilon)(1-s)^2$ ,  $\therefore 1^3 = 1+\epsilon - 2s$  neglecting the squares and products of  $s$  and  $\epsilon$ , which is permitted as the ellipticity of the spheroid is supposed to be inconsiderable, consequently we have  $\epsilon=2s$ ,  $\therefore$  in an oblong spheroid, such as would be generated by a revolution about the greater axis, the elevation of the spheroid above the equicapacious sphere is double of the depression below this sphere ; and if  $r$  be the radius of the equicapacious sphere,  $a$  the greater, and  $b$  the lesser axis of the spheroid, we have  $a-r=2r-2b$ ,  $\therefore r = \frac{a+2b}{3}$  ; if the spheroid be oblate, i. e. such as would be generated by a revolution about the lesser axis,  $\frac{4\pi}{3} \cdot 1^3 = \frac{4\pi}{3} \cdot (1-s)(1+\epsilon)^2$ , hence  $s=2\epsilon$ , i.e. the depression in this case is equal to twice the elevation,  $\therefore 2a-2r=r-b$ , and  $r = \frac{2a+b}{3}$ .

If a sphere be inscribed in a spheroid, the elevation of any point of the spheroid above the inscribed sphere, is to the greatest elevation of a spheroid above the inscribed sphere, i. e. to the difference between the radius of the equator and semiaxis, as the square of the cosine of the angular distance  $\lambda$  from the axis major, to the square of radius,  $\therefore$  the elevation  $= (a-b) \cos.^2 \lambda$ , and as the equicapacious sphere is elevated above the lesser axis, and  $\therefore$  above the inscribed sphere by a quantity equal to  $r-b$ , the elevation of the spheroid above the equicapacious sphere  $= (a-b) \cos.^2 \lambda - r + b = (a-b)$ .

$\cos.^2 \lambda = \frac{2a+b}{3} + b$ , ( $= \frac{-2a+2b}{3}$ ), consequently when the elevation is 0, we have  $\cos.^2 \lambda = \frac{2}{3}$ ,  $\therefore \sin.^2 \lambda = \frac{1}{3}$ , and  $\lambda = 35^\circ 16'$ . This situation is also remarkable for being the distance from the quadrature at which the addititious force of the sun, is equal to that part of its ablatitious force, which acts in direction of the radius of the moon's orbit.

as we shall see in the third book : we select this parallel, because the attraction of the earth on the corresponding points of its surface, is very nearly, as at the distance of the moon, equal to the mass of the earth, divided by the square of its distance from its centre of gravity. Under this parallel, the gravity is less than the attraction of the earth, by  $\frac{2}{3}*$  of the centrifugal force which arises from the motion of rotation

at the equator ; this force is the  $\frac{1}{288}$ th part of the force of gravity ;

consequently we must augment the preceding space by its 432d part, in order to obtain the entire space which is due to the action of the earth, which on this parallel, is equal to its mass divided by the square of the terrestrial radius ; therefore this space will be equal to  $3^{me},66394$ . At the distance of the moon, it must be diminished in the ratio of the square of the radius of the spheroid of the earth, to the square of the distance of this star, to effect this, it is sufficient to multiply it by the square of the sine of the lunar parallax ; therefore  $x$  representing this sine under the parallel above mentioned, we shall have  $x^2.3^{me},66394$ , for the height through which the moon ought to fall in a second, by the attraction of the earth. But we shall see in the theory of the moon, that the action of the sun diminishes its gravity towards the earth by a quantity, of which the constant part is†

\* The centrifugal force at the equator is to the efficient part of the centrifugal force at any parallel, as the square of radius to the square of the cosine of latitude, i. e. in this case, as 1 to  $\frac{2}{3}$ , ∴ as the centrifugal force at the equator is the  $\frac{1}{288}$ th part of the gravity, the force at the parallel in question, will be  $= \frac{2}{3} \cdot \frac{1}{288} = \frac{1}{432}$ .

†  $m$  being the mass of the sun, and  $d$  its distance from the moon,  $a$  the radius of the moon's orbit, the addititious force  $= \frac{ma}{d^3}$ , and the part of the ablatitious force, which acts in the direction of the radius vector  $= \frac{ma}{d^3} \cdot 3 \sin. {}^2 \pi$ ,  $\pi$  being the angular distance from quadrature, see Newton, Princip. Prop. 66 ; ∴  $\frac{ma}{d^3} (1 - 3 \sin. {}^2 \pi)$  is the part of the sun's

equal to the  $\frac{1}{358}$ th part of this gravity; moreover, the moon, in its relative motion about the earth, is sollicited by a force equal to the sum of the masses\* of the earth and moon, divided by the square of their mutual distance; it is therefore necessary to diminish the preceding space by its 358th part, and to increase it in the ratio of the sum the masses of the earth and moon, to the mass of the earth; but we shall see in the fourth book, that the mass of the moon deduced from the phenomena of the tides, is a  $\frac{1}{58,7}$ th part of the mass of the earth; therefore the space through which the moon descends towards the earth, in the interval of a second, is equal to  $\frac{357}{358} \cdot \frac{59,7}{58,7} \cdot x^2 \cdot 3^{me} \cdot 66394$ .

Now  $a$  representing the mean radius of the lunar orbit, and  $T$ , the duration of a sidereal revolution of the moon, expressed in seconds;

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disturbing force acting in the direction of the radius, which is efficient at any point; (hence it appears that it vanishes when  $\sin. 2\omega = \frac{1}{3}$ , see Note, page 17); in order to obtain its mean quantity, multiply this expression by  $d\omega$  and it becomes  $\frac{ma}{d^3}$ .  
 $(d\omega - 3d\omega \cdot \sin. 2\omega) = \frac{ma}{d^3} (d\omega - \frac{3}{2} d\omega + \frac{3}{2} d\omega \cos. 2\omega)$ , and its integral  $= \frac{ma}{d^3} (\omega - \frac{3}{2}\omega + \frac{3}{4} \cdot \sin. 2\omega)$  = for the entire circumference, i.e. when  $\omega = \pi$ ,  $- \frac{ma}{d^3} \cdot \frac{\pi}{2}$ ; ∵ the mean disturbing force  $= - \frac{ma}{2d^3}$ , but  $\frac{m}{d^2} : F$  the force retaining the moon in its orbit ::  $\frac{d}{T^2} : \frac{a}{T'^2}$  ( $T, T'$  are the periods of the sun and moon) ∵  $\frac{ma}{d^3} = \frac{T'^2 F}{T^2} = \frac{F}{179}$ , for  $\frac{T'}{T} = \frac{1}{179}$ , and  $- \frac{ma}{2d^3} = - \frac{F}{358}$ , ∵ in consequence of the diminution of her gravity by the action of the disturbing force, the moon is sustained at a greater distance from the earth, than it would be if the action of the sun was removed, and as the mean area described in a given time in the primitive and disturbed orbits is the same, the radius vector is increased by a 358th part, and the angular velocity is diminished by a 179th part.

\* The moon being considered as a point, if it revolved about the centre of the earth, in

$\frac{2a\pi^2}{T^2}$  will be, as has been already observed, the versed sine of the arc which it describes during a second, and it expresses the quantity, by which the moon has descended towards the earth, in this interval. This value of  $a$  is equal to the radius of the earth, under the above mentioned parallel, divided by the sine of  $x$ ; this radius is equal to  $6369514^{me}$ ; therefore we have

$$a = \frac{6369514^{metre}}{x};$$

but in order to obtain a value of  $a$ , independent of the inequalities of the moon, it is necessary to assume for its mean parallax of which the sine is  $x$ , the part of this parallax, which is independent of these inequalities, and which has been therefore termed the *constant part of the parallax*. Thus,  $\pi$  representing the ratio of 355 to 113, and  $T'$  being  $= 2732166''$ ; the mean space through which the moon descends towards the earth, will be

$$\frac{2.(355)^2.6369514^{me}}{(113)^2.x.(2732166)^2}.$$

the same time in which it revolves about the common centre of gravity of the earth and moon, the central force which should exist in the centre of the earth capable of effecting this, should be  $\pm$  to the sum of the masses of the earth and moon; for  $a$  being the distance of the earth from the moon, and  $m, m'$  their respective masses, the distance  $y$  at which the moon would revolve round the earth by itself, considered as quiescent, is

$$\pm \frac{am^{\frac{1}{3}}}{(m+m')^{\frac{1}{3}}}, \text{ see Prin. Math. Prop. 59, Book I. and } T' \pm \frac{y^3}{m} = \frac{a^3}{m+m'}, \text{ hence if } a$$

be the distance, the central force  $= m+m'$ ,  $\therefore$  as the versed sine of the arc described in a second is the space through which the moon descends in consequence of the combined actions of the earth and moon, this must be diminished in the ratio of  $m:m+m'$  to obtain the space described in consequence of the sole action of  $m$ . The two corrections, which are here applied to the *space* through which a heavy body would descend at the latitude  $35^{\circ}16'$ , diminished in the ratio of the square of the distance, are in the *Système du Monde*, applied to the *versed sine* of the arc described in a second, hence it appears that they must be affected with contrary signs.

By equalling the two expressions, which we have found for this space, we shall have

$$x^3 = \frac{2.(355)^2.358.58,7.6369514}{(113)^2.357.59,7.3,66394(2732166)^2};$$

from which we obtain  $10536'',2$  for the constant part\* of the lunar parallax, under the parallel in question. This value differs very little from the constant quantity  $10540,7$  which Triesnecker collected from a great number of observations of eclipses, and of † occultations of the stars by the moon ; it is therefore certain that the *principal* force which retains the moon in its orbit, is the terrestrial gravity diminished in the ratio of the square of the distance ; thus, the law of the diminution of gravity, which in the planets attended by several satellites, is proved by a comparison of the times of their revolutions, and of their distances, is

\* In order to find the constant part of the parallax, we apply to the observed parallax, all the corrections which theory makes known, and we may perceive from this how the theory of gravity, by indicating the forces which act on the moon, furnishes us with the means of determining the mean motion, and the nature of the inequalities which act on it.

† If in a partial eclipse of the moon, the time be noted in which the two horns of the part which is not eclipsed, are observed to be in the same vertical line, it would be easy to shew that the height of the centre of the moon at this instant, will be the same as the height of the centre of the shadow ; ∵ if at this instant the height of each of the horns be observed, the mean height, which will be the height of the centre of the shadow, will be the apparent height affected by the parallax ; but as the centre of the shadow is diametrically opposite to the centre of the sun, the true height will be equal to the depression of the sun, which is known from the time of observation ; ∵ the difference of these heights will be the parallax of the moon for the observed altitude, by means of which we can easily determine the greatest parallax ; and if in a total and central eclipse, the height of the moon be observed *at the instant* that it is entirely immersed, and also when it *first* begins to emerge, the mean height will be the height of the centre of the shadow as it is affected by parallax.

In an occultation of a fixed star, the star's parallax vanishes, and the difference of apparent altitudes is = to the difference of the true altitudes + parallax in altitude of the moon ; hence by the known formulæ we can obtain the true parallax. A constant ratio exists between the horizontal parallax, and the moon's apparent diameter at the same terrestrial latitude.

demonstrated for the moon, by comparing its motion with that of projectiles near the surface of the earth. It follows from this, that the origin of the distances of the sun, and of the planets, ought in the computation of their attractive forces, on bodies placed at their surface, or beyond it, to be fixed in the centre of gravity of these bodies; since this has been demonstrated to be the case for the earth, of which the attractive force is, as has been remarked, of the same nature with that of these stars.

6. The sun and the planets which are accompanied with satellites, are consequently endowed with an attractive force, which decreasing indefinitely, in the inverse ratio of the squares of the distances, comprehends all bodies in the sphere of its activity. Analogy would induce us to think, that a like force inheres generally in all the planets and in the comets; but we may be assured of it directly in the following manner. It is a constant law of nature, that one body cannot act on another, without experiencing an equal and contrary reaction; therefore the planets and comets being attracted towards the sun, they ought to attract this star according to the same law. For the same reason, the satellites attract their respective primary planets; consequently this attractive force is common to the planets, to the comets, and to the satellites, and therefore we may consider the gravitation of the heavenly bodies, towards\* each other, as a general property which belongs to all the bodies of the universe.

We have seen, that it varies inversely as the square of the distance; indeed, this ratio is given by the laws of elliptic motion, which do not rigorously obtain in the celestial motions; but we should consider, that the simplest laws ought always to be preferred, unless observations compel us to abandon them; it is natural for us to suppose, in the first instance, that the law of gravitation is inversely as some power of the dis-

\* Besides, it follows from the sphericity of these bodies that their molecules are united about their centres of gravity, by a force which at equal distances solicits them equally towards these points; the existence of this force is also indicated by the perturbations which the planetary motions experience.

tancee, and by computation it has been found, that the slightest difference between this\* power and the square, would be very perceptible in the position of the perihelia of the orbits of the planets, in which observation has indicated motions hardly perceptible, and of which we shall hereafter develope the cause. In general, we shall see throughout this treatise, that the law of gravitation inversely as the square of the distance, represents with the greatest precision all the observed inequalities of the motions of the heavenly bodies; this agreement, combined with the simplicity of this law, justifies us in assuming that it is rigorously the law of nature.

The gravitation is proportional to the masses; for it follows from No. 3, that the planets and comets being supposed at equal distances from the sun, and then remitted to their gravity towards this star, would fall through equal spaces, in the same time; consequently their gravity will be proportional to their mass. The motions almost circular of the satellites about their primaries, demonstrate that they gravitate as their primaries towards the sun, in the ratio of their masses; the slightest difference in this respect, would be perceptible in the motions of the satellites, and observations have not indicated any inequality depending

\* See No. 58 of this book; this also follows from Prop. 45, Book 1st, Prin. For if the force which is added to the force varying in the inverse ratio of the square of the distance be called

$X$ , the angular distance between the apsides  $= 180 \cdot \frac{\sqrt{1+X}}{\sqrt{1+3X}} = 180.(1-X)$ , the square of

$X$  being neglected, and conversely if the distance between the apsides be given, we can determine  $X$ . The force  $X$  is supposed to vary as the distance.

† See Newton Princip. Prop. 6, Book 3, where it is shewn, that if the satellite gravitated more towards the sun than the primary at equal distances from the sun, in the ratio of  $d:e$ , the distance of the centre of the sun from the centre of the orbit of the satellite, would be greater than the distance of the centre of the sun from the centre of the primary, in the ratio of  $\sqrt{d}:\sqrt{e}$ ,  $\therefore$  if the difference between  $d$  and  $e$ , was the thousandth part of the entire gravity, the distance of the centre of the orbit from the centre of the sun, would be greater than the distance of the centre of Jupiter from that of the sun, by a  $\frac{1}{2000}$ th part of the entire distance.

on this cause. Therefore it appears that if the comets, the planets and satellites, were placed at equal distances from the sun, they would gravitate towards this star, in the ratio of their masses ; from which it follows, in consequence of the equality between action and reaction, that these stars must attract the sun, in the same ratio, and consequently their action on this star, is proportional to their\* masses divided by the square of their distance from its centre.

The same law obtains on the earth ; for from very exact experiments instituted by means of the pendulum, it has been ascertained, that if the resistance of the air was removed, all bodies would descend towards *its centre* with equal velocities ; therefore bodies near the earth gravitate towards its centre, in the ratio of their masses, in the same manner as the planets gravitate towards the sun, and the satellites towards their primaries. This conformity of nature with itself on the earth, and in the immensity of the heavens, evinces in the most striking manner, that the

\* The mutual attraction does not affect the elliptic motion of any *two* bodies when their mutual action is considered, for the relative motion is not affected when a common velocity is impressed on the bodies,  $\therefore$  if the motion which the sun has, and the action which it experiences on the part of the planet, be impressed in a contrary direction, on both the sun and the planet ; the sun may be regarded as immovable, and the planet will be solicited by a force  $\therefore^2$  to the sum of the masses of the sun and planet, divided by the square of their mutual distance ;  $\therefore$  the motion will be elliptic ; but the periodic time will be less than if the planet did not act on the sun, for the ratio of the cube of the greater axis of the orbit to the square of the periodic time, is proportional to the sum of the masses of the sun and planet ; however as this ratio of the square of the time to the cube of the distance, is very nearly the same for all the planets, it follows that the masses of the planets must be comparatively much smaller than the mass of the sun, which is confirmed by an estimation of their volumes. See No. 25, and Prop. 8, Lib. 3. Princip. Math. The comparative smallness of the masses is also confirmed by the laws which Kepler was enabled to announce, for these laws were deduced from observation, notwithstanding the various causes which disturb the elliptic motion ; hence appears the reason why, in the commencement of this chapter, the sun was supposed to be immoveable, and to exert its action on the planets as on so many points, which do not react on the sun, neither was the mutual action of the planets on each other taken into account ; the same simplifications were employed, when the motion of a satellite about its primary was considered.

gravity observed here on earth, is only a particular case of a general law, which obtains throughout the universe.

The attractive property of the heavenly bodies does not appertain to them solely in a mass, but is peculiar to each of their molecules. If the sun only acted on the centre of the earth, without attracting in particular each of its parts, there would be produced in the sea, oscillations much greater, and very different from those which we observe ; therefore the gravity of the earth to the sun, is the result of the gravitations of all its molecules, which consequently attract the sun, in the ratio of their respective masses. Besides, each body on the earth gravitates towards its centre, proportionally to its mass ; it reacts therefore on the earth, and attracts it in the same ratio. If this was not the case, and if any part of the earth, however small, did not attract the other part, as it is attracted by this other part, the centre of gravity of the earth would have a motion in space, in consequence of the force of gravity, which is impossible.

The celestial phenomena, compared with the laws of motion, conduct us therefore to this great principle of nature, namely, that all the molecules of matter mutually attract each other in the proportion of their masses, divided by the square of their distances. We may perceive already, in this *universal* gravitation, the cause of the perturbations, which the heavenly bodies experience ; for the planets and comets being subject to their reciprocal action, ought to deviate a little from the laws of elliptic motion, which they would accurately follow, if they only obeyed the action of the sun. The satellites in like manner deranged in their motions about their primaries, by their mutual attraction, and by that of the sun, deviate from these laws. We may perceive also, that the molecules of each of the heavenly bodies, united by their attraction, should constitute a mass nearly spherical, and that the result of their action at the surface of the body, should produce all the phenomena of gravitation. We see moreover, that the motion of rotation of the heavenly bodies, should slightly alter the sphericity of their figure, and flatten them at the poles, and that then, the resultant of their mutual action, not pass-

ing accurately through their centres of gravity, ought to produce in their axes of rotation, motions similar to those, which are indicated by observation. Finally, we may perceive why the molecules of the ocean, unequally acted on by the sun and moon, ought to have an oscillatory motion, similar to the ebbing and flowing of the sea. But the development of these different effects of universal gravitation, requires a profound analysis. In order to embrace them in all their generality, we proceed to give the differential equations of the motion of a system of bodies, subjected to their mutual attraction, and to investigate the exact integrals which may be derived from them. We will then take advantage of the facilities which the relations of the masses and distances of the heavenly bodies furnish us with, in order to obtain integrals more and more accurate, and thus to determine the celestial phenomena, with all the precision which the observations admit of.

## CHAPTER II.

*Of the differential equations of the motion of a system of bodies, subjected to their mutual attraction.*

7. LET  $m, m', m'', \&c.$  represent the masses of the different bodies of the system, considered as so many points; let  $x, y, z,$  be the rectangular coordinates of the body  $m;$   $x', y', z',$  those of the body  $m',$  and corresponding expressions for the coordinates of the other bodies. The distance of  $m'$  from  $m$  being equal to

$$\sqrt{(x'-x)^2 + (y'-y)^2 + (z'-z)^2},$$

its action on  $m,$  will be, by the law of universal gravitation, equal to

$$\frac{m'}{(x'-x)^2 + (y'-y)^2 + (z'-z)^2}.$$

If we resolve this action, parallel to the axes of  $x,$  of  $y,$  and of  $z,$  the force parallel to the axis of  $x,$  and directed from the origin, will be

$$\frac{m'(x'-x)}{\{(x'-x)^2 + (y'-y)^2 + (z'-z)^2\}^{\frac{3}{2}}}.$$

E 2

\* The force parallel to the axis of  $x:$   $\frac{m'}{(x'-x)^2 + (y'-y)^2 + (z'-z)^2} :: x' - x :$   
 $\sqrt{(x'-x)^2 + (y'-y)^2 + (z'-z)^2};$  and if  $\sqrt{x'-x)^2 + (y'-y)^2 + (z'-z)^2}$  be differenced with respect to  $x,$  and then divided by  $m.dx,$  it will become  
 $= \frac{1}{m.dx} \cdot \frac{mm'.(x'-x).dx}{((x'-x)^2 + (y'-y)^2 + (z'-z)^2)^{\frac{3}{2}}}.$

or

$$\frac{1}{m} \cdot \left\{ d \cdot \frac{mm'}{\sqrt{(x'-x)^2 + (y'-y)^2 + (z'-z)^2}} \right\}.$$

We shall have also,

$$\frac{1}{m} \cdot \left\{ d \cdot \frac{mm''}{\sqrt{(x''-x)^2 + (y''-y)^2 + (z''-z)^2}} \right\}$$

for the action of  $m''$  on  $m$ , resolved parallel to the axis of  $x$ , and corresponding expressions for the other bodies of the system. Consequently if

$$\lambda = \frac{mm'}{\sqrt{(x'-x)^2 + (y'-y)^2 + (z'-z)^2}} + \frac{mm''}{\sqrt{(x''-x)^2 + (y''-y)^2 + (z''-z)^2}}$$

$$+ \frac{m'm''}{\sqrt{(x''-x')^2 + (y''-y')^2 + (z''-z')^2}} + \text{&c.};$$

$\lambda$  representing the sum of the products of the masses  $m, m', m'', \text{ &c.}$ , taken two by two, and divided by their respective distances;  $\frac{1}{m}$ .

$\left\{ \frac{d\lambda}{dx} \right\}^*$  will express the sum of the actions of the bodies  $m', m'', \text{ &c.}$  on  $m$ , resolved parallel to the axis of  $x$ , and directed from the origin of

$$* \frac{1}{m} \cdot \left( \frac{d\lambda}{dx} \right) = \frac{1}{m} \left\{ d \cdot \frac{mm'}{\sqrt{(x'-x)^2 + (y'-y)^2 + (z'-z)^2}} + \right.$$

$$d \cdot \frac{mm''}{\sqrt{(x''-x)^2 + (y''-y)^2 + (z''-z)^2}} + \text{&c.} \left. \right\} = \frac{m' \cdot (x'-x)}{\left( (x'-x)^2 + (y'^2-y)^2 + (z'-z)^2 \right)^{\frac{3}{2}}} +$$

$$+ \frac{m'' \cdot (x''-x)}{\left( (x''-x)^2 + (y''-y)^2 + (z''-z)^2 \right)^{\frac{3}{2}}} + \text{&c.} = \text{the sum of the actions of the bodies } m', m'', m''', \text{ &c. on } m, \text{ resolved parallel to the axis of } x.$$

the coordinates. Therefore  $dt$  representing the element of the time, supposed constant; we shall have by the principles of dynamics, explained in the preceding book,

$$0 = m \cdot \frac{d^2x}{dt^2} - \left\{ \frac{d\lambda}{dx} \right\}.$$

In like manner we shall have

$$0 = m \cdot \frac{d^2y}{dt^2} - \left\{ \frac{d\lambda}{dy} \right\};$$

$$0 = m \cdot \frac{d^2z}{dt^2} - \left\{ \frac{d\lambda}{dz} \right\}.$$

If we consider, in the same manner, the action of the bodies  $m$ ,  $m''$ , &c. on  $m'$ ; that of the bodies  $m$ ,  $m'$ , on  $m''$ , and so of the rest, we shall have the following equations, namely,

$$0 = m' \cdot \frac{d^2x'}{dt^2} - \left\{ \frac{d\lambda}{dx'} \right\}; \quad 0 = m' \cdot \frac{d^2y'}{dt^2} - \left\{ \frac{d\lambda}{dy'} \right\};$$

$$0 = m' \cdot \frac{d^2z'}{dt^2} - \left\{ \frac{d\lambda}{dz'} \right\};$$

$$0 = m'' \cdot \frac{d^2x''}{dt^2} - \left\{ \frac{d\lambda}{dx''} \right\}; \quad 0 = m'' \cdot \frac{d^2y''}{dt^2} - \left\{ \frac{d\lambda}{dy''} \right\};$$

$$0 = m'' \cdot \frac{d^2z''}{dt^2} - \left\{ \frac{d\lambda}{dz''} \right\}. \text{ &c.}$$

The determination of the motions of  $m$ ,  $m'$ ,  $m''$ , &c., depends on the integration of these differential equations; but as yet they have not been completely integrated, except in the case in which the system is composed of only two bodies. In other cases, we have not been able to

obtain but a small number of perfect integrals, which we proceed to develope.

8. For this purpose, let us first consider the differential equations in  $x, x', x'', \&c.$ ; if we add them together, observing at the same time, that by the nature of the function  $\lambda$ , we have

$$0 = \left\{ \frac{d\lambda}{dx} \right\} + \left\{ \frac{d\lambda}{dx'} \right\} + \left\{ \frac{d\lambda}{dx''} \right\} + \&c.;^*$$

we shall obtain,  $0 = \Sigma.m. \frac{d^2x}{dt^2}$ . We shall have also,  $0 = \Sigma.m. \frac{d^2y}{dt^2}$ ;

$0 = \Sigma.m. \frac{d^2z}{dt^2}$ . Let  $X, Y, Z$  represent the three coordinates of the centre of gravity of the system; we shall have by the nature of this centre

$$X = \frac{\Sigma.mx}{\Sigma.m}; \quad Y = \frac{\Sigma.my}{\Sigma.m}; \quad Z = \frac{\Sigma.mz}{\Sigma.m};$$

therefore we shall have

$$0 = \frac{d^2X}{dt^2}; \quad 0 = \frac{d^2Y}{dt^2}; \quad 0 = \frac{d^2Z}{dt^2};$$

and by integrating, we shall obtain

$$X = a + bt; \quad Y = a' + b't; \quad Z = a'' + b''t;^{\dagger}$$

\* Suppose that there are only three bodies, then  $\Sigma.m. \frac{d^2x}{dt^2} = \left( \frac{d\lambda}{dx} \right) + \left( \frac{d\lambda}{dx'} \right) + \left( \frac{d\lambda}{dx''} \right)$   
 $= \frac{m'm.((x'-x)-(x''-x))}{((x'-x)^2+(y'-y)^2+(z'-z)^2)^{\frac{3}{2}}} + \frac{mm''((x''-x)-(x''-x))}{((x''-x)^2+(y''-y)^2+(z''-z)^2)^{\frac{3}{2}}}$   
 $+ \frac{m'm'((x''-x')-(x''-x'))}{((x''-x')^2+(y''-y')^2+(z''-z')^2)^{\frac{3}{2}}} = 0$ ; the same proof may be extended to any number of bodies.

†  $X = \frac{\Sigma.mx}{\Sigma.m}$ ,  $\therefore \frac{dX}{dt} = \Sigma.m. \frac{dx}{dt} \div \Sigma.m. \frac{d^2X}{dt^2} = \Sigma.m. \frac{d^2x}{dt^2} \div \Sigma.m. = 0$ , and by inte-

$a, a', a'', b, b', b''$ , being constant arbitrary quantities. We may perceive by this, that the motion of the centre of gravity of the system is rectilinear and uniform, and that consequently, it is not deranged by the reciprocal action of the bodies composing the system; which agrees with what has been demonstrated in the fifth chapter of the first book. Resuming the differential equations of the motion of these bodies, and multiplying the differential equations in  $y, y', y'', \&c.$ , respectively by  $x, x', x'', \&c.$ , and then adding them to the differential equations in  $x, x', x'', \&c.$  multiplied respectively by  $-y, -y', -y'', \&c.$ ; we shall obtain

$$\begin{aligned} 0 = & m \cdot \left\{ \frac{x d^2 y - y d^2 x}{dt^2} \right\} + m' \cdot \left\{ \frac{x' d^2 y' - y' d^2 x'}{dt^2} \right\} + \\ & m'' \cdot \left\{ \frac{x'' d^2 y'' - y'' d^2 x''}{dt^2} \right\} + \&c. \\ & + y \cdot \left\{ \frac{d\lambda}{dx} \right\} + y' \cdot \left\{ \frac{d\lambda}{dx'} \right\} + y'' \cdot \left\{ \frac{d\lambda}{dx''} \right\} + \&c. \\ & - x \cdot \left\{ \frac{d\lambda}{dy} \right\} - x' \cdot \left\{ \frac{d\lambda}{dy'} \right\} - x'' \cdot \left\{ \frac{d\lambda}{dy''} \right\} - \&c.; \end{aligned}$$

but from the nature of the function  $\lambda$ , it is evident that

$$0 = y \cdot \left\{ \frac{d\lambda}{dx} \right\} + y' \cdot \left\{ \frac{d\lambda}{dx'} \right\} + \&c.$$

$$- x \cdot \left\{ \frac{d\lambda}{dy} \right\} - x' \cdot \left\{ \frac{d\lambda}{dy'} \right\} - \&c.;$$

grating,  $\frac{dX}{dt} = a$ , and  $X = at + b$ , the constant quantity  $a$  depends on the velocity of the centre of gravity at the commencement of the motion, and  $b$  depends on the position of this centre, at the same instant.

consequently,\* by integrating the preceding equation, we shall obtain

$$c = \Sigma.m. \left\{ \frac{xdy - ydx}{dt} \right\}.$$

In like manner we shall have,

$$c' = \Sigma.m. \left\{ \frac{xdz - zdx}{dt} \right\};$$

$$c'' = \Sigma.m. \left\{ \frac{ydz - zdy}{dt} \right\};$$

$c, c', c'', \&c.$  being constant arbitrary quantities. These three integrals involve the principle of the conservation of areas, which has been explained in the fifth chapter of the first book.

Finally, if we multiply the differential equations in  $x, x', x'', \&c.$ , respectively by  $dx, dx', dx'', \&c.$ ; and those in  $y, y', y'', \&c.$  respectively by  $dy, dy', dy'', \&c.$ ; those in  $z, z', z'', \&c.$ , respectively by  $dz, dz', dz'', \&c.$ ; and then add them together, we shall obtain

$$0 = \Sigma.m. \frac{(dx.d^2x + dy.d^2y + dz.d^2z)}{dt^2} - d\lambda, \dagger$$

\* Suppose that there are only three bodies, then  $y\left(\frac{d\lambda}{dx}\right) + y'\left(\frac{d\lambda}{dx'}\right) + y''\left(\frac{d\lambda}{dx''}\right) - x\left(\frac{d\lambda}{dy}\right) - x'\left(\frac{d\lambda}{dy'}\right) - x''\left(\frac{d\lambda}{dy''}\right) = m. \left( \frac{xd^2y - yd^2x}{dt^2} \right) + m'. \left( \frac{x'd^2y' - y'd^2x'}{dt^2} \right) + m''. \left( \frac{x''d^2y'' - y''d^2x''}{dt^2} \right) = \frac{mm'.(y(x'-x) - y'(x'-x))}{((x'-x)^2 + (y'-y)^2 + (z'-z)^2)^{\frac{3}{2}}} + \frac{mm''(y(x''-x) - y''(x''-x))}{((x''-x)^2 + (y''-y)^2 + (z''-z)^2)^{\frac{3}{2}}} + \frac{m'm'.(y'(x''-x') - y''(x''-x'))}{((x''-x')^2 + (y''-y')^2 + (z''-z')^2)^{\frac{3}{2}}} - \frac{mm'(x(y'-y) - x'(y'-y))}{((x'-x)^2 + (y'-y)^2 + (z'-z)^2)^{\frac{3}{2}}} - \frac{mm''(x(y''-y) - x''(y''-y))}{((x''-x)^2 + (y''-y)^2 + (z''-z)^2)^{\frac{3}{2}}} - \frac{m''m'(x'(y''-y') - x''(y''-y'))}{((x''-x')^2 + (y''-y')^2 + (z''-z')^2)^{\frac{3}{2}}} = 0.$

† By multiplying  $\left\{ \frac{d\lambda}{dx} \right\} \left\{ \frac{d\lambda}{dx'} \right\} + \&c. by dx, dx', dx'', \&c.; \left\{ \frac{d\lambda}{dy} \right\}, \left\{ \frac{d\lambda}{dy'} \right\}, +$

and by integrating,

$$h = \Sigma.m.\left(\frac{dx^2 + dy^2 + dz^2}{dt^2}\right) - 2\lambda;$$

$h$  being a new arbitrary quantity. This integral contains the principle of the conservation of living forces, which has been treated of in the fifth chapter of the first book.

The\* seven preceding integrals are the only exact integrals, which we have hitherto been able to obtain; when the system is composed of only two bodies, the determination of their motions is reduced to differential equations of the first order, which can be integrated, as we will see in the sequel; but when the system is composed of three or a greater number of bodies, we are then obliged to recur to the methods of approximation.

9. As we can only observe the relative motions of bodies; we refer the motions of the planets and of the comets, to the centre of the sun, and the motions of the satellites, to the centre of their primaries. Therefore in order to compare the theory with observations, it is necessary to determine the relative motions of a system of bodies, about a body which is considered as the centre of their motions.

Let  $M$  represent this last body,  $m, m', m'', \&c.$ , being the other bodies, the relative motion of which about  $M$ , is required; Let  $\zeta, \Pi$  and  $\gamma$  be the rectangular coordinates of  $M$ ,  $\zeta+x, \Pi+y, \gamma+z$ , those of  $m$ ;  $\zeta+x', \Pi+y', \gamma+z'$ , those of  $m', \&c.$ ; it is manifest that  $x, y, z$ , will be the coordinates of  $m$ , with respect to  $M$ ; that  $x', y', z'$ , will be those

$\&c.$ , by  $dy, dy', dy'', \&c.$  and then adding these quantities together, their aggregate is equal to the differential of  $\lambda$  considered as a function of  $x, x', \&c. y, y', \&c. z, z', \&c.$ , and  $\therefore$  it is equal to  $d\lambda$ .

\* Three of these integrals are furnished by the principle of the conservation of areas, three by the principle of the conservation of the motion of the centre of gravity, and one by the conservation of living forces.

of  $m'$  referred to the same body, and so of the rest. Let  $r, r', \&c.$ , represent the distances of  $m, m', \&c.$  from the body  $M$ , so that

$$r = \sqrt{x^2 + y^2 + z^2}; \quad r' = \sqrt{x'^2 + y'^2 + z'^2}; \quad \&c.$$

and let us also suppose

$$\begin{aligned} \lambda &= \frac{m'm}{\sqrt{(x'-x)^2 + (y'-y)^2 + (z'-z)^2}} + \\ &\quad \frac{mm''}{\sqrt{(x''-x)^2 + (y''-y)^2 + (z''-z)^2}} \\ &+ \frac{m'm''}{\sqrt{(x''-x')^2 + (y''-y')^2 + (z''-z')^2}} + \&c. \end{aligned}$$

This being premised, the action of  $m$  on  $M$ , resolved parallel to the axis of  $x$ , and tending from the origin, will be  $\frac{mx}{r^3}$ ; that of  $m'$  on  $M$  resolved in the same direction, will be  $\frac{m'x'}{r'^3}$ , and so of the other bodies of the system. Therefore, to determine  $\zeta$ , we will have the following differential equation :

$$0 = \frac{d^2\zeta}{dt^2} - \Sigma \cdot \frac{mx}{r^3};$$

and in like manner,

$$0 = \frac{d^2\Pi}{dt^2} - \Sigma \cdot \frac{my}{r^3},$$

$$0 = \frac{d^2\gamma}{dt^2} - \Sigma \cdot \frac{mz}{r^3}.$$

The action of  $M$  on  $m$ , resolved parallel to the axis of  $x$ , and directed from the origin, will be  $-\frac{Mx}{r^3}$ , and the sum of the actions of the bodies  $m'$ ,  $m''$ , &c. on  $m$ , resolved in the same direction, will be  $\frac{1}{m} \left( \frac{d\lambda}{dx} \right)$ ; consequently, we will have

$$0 = \frac{d^2(\xi+x)}{dt^2} + \frac{Mx}{r^3} - \frac{1}{m} \cdot \left\{ \frac{d\lambda}{dx} \right\}; *$$

and substituting in place of  $\frac{d^2\xi}{dt^2}$ , its value  $\Sigma \cdot \frac{mx}{r^5}$ , we will obtain

$$0 = \frac{d^2x}{dt^2} + \frac{Mx}{r^3} + \Sigma \cdot \frac{mx}{r^5} - \frac{1}{m} \cdot \left\{ \frac{d\lambda}{dx} \right\}; \quad (1)$$

in like manner, we will have

$$0 = \frac{d^2y}{dt^2} + \frac{My}{r^3} + \Sigma \cdot \frac{my}{r^5} - \frac{1}{m} \cdot \left\{ \frac{d\lambda}{dy} \right\}; \quad (2)$$

$$0 = \frac{d^2z}{dt^2} + \frac{Mz}{r^3} + \Sigma \cdot \frac{mz}{r^5} - \frac{1}{m} \cdot \left\{ \frac{d\lambda}{dz} \right\}; \quad (3)$$

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\*  $\frac{1}{m} \cdot \left\{ \frac{d\lambda}{dx} \right\}$  is equal to the sum of the actions of the bodies  $m'$ ,  $m''$ , &c. on  $m$ , resolved parallel to the axis of  $x$ ,  $\therefore$  if we add to this expression the action of  $M$  on  $m$ , which is equal to  $-\frac{Mx}{r^3}$ , we will have the actions of all bodies of the system on  $m$ , and  $\therefore$  by the principles of dynamics established in the first book,  $\frac{d^2(\xi+x)}{dt^2} + \frac{Mx}{r^3} - \frac{1}{m} \cdot \left\{ \frac{d\lambda}{dx} \right\} = 0$ .

If in the equations (1), (2), (3), we change successively the quantities  $m, x, y, z$ , into  $m', x', y', z'$ ;  $m'', x'', y'', z''$ , &c.; and reciprocally, we will obtain the equations of the motion of the bodies  $m', m'',$  &c. about  $M$ .

If we multiply the differential equation in  $\zeta$ , by  $M + \Sigma.m.$ ; that in  $x$ , by  $m$ ; that in  $x'$ , by  $m'$ , and performing similar operations on the other differential equations; by adding them together, and observing that by the nature of the function  $\lambda$ , we have

$$0 = \left\{ \frac{d\lambda}{dx} \right\} + \left\{ \frac{d\lambda}{dx'} \right\} + \text{&c.};$$

we will obtain

$$0 = (M + \Sigma.m.) \cdot \frac{d^2\zeta}{dt^2} + \Sigma.m. \cdot \frac{d^2x}{dt^2}; *$$

from which we obtain by integrating

\* The differential equation in  $\zeta$ , becomes by this multiplication,  $(M + \Sigma.m.) \frac{d^2\zeta}{dt^2}$   
 $- M.\Sigma. \frac{mx}{r^3} - \Sigma.m.\Sigma. \frac{mx}{r^3} = 0$ ; and if the differential equations in  $x, x', x'',$  &c. be multiplied by  $m, m', m'',$  &c., respectively, and then added together, their sum will be  $\Sigma.m. \left\{ \frac{d^2x}{dt^2} \right\} + M.\Sigma. \frac{mx}{r^3} + \Sigma.m.\Sigma. \frac{mx}{r^3} - \left\{ \frac{d\lambda}{dx} \right\} - \left\{ \frac{d\lambda}{dx'} \right\} - \left\{ \frac{d\lambda}{dx''} \right\} - \text{&c.} = 0$ , if this expression be added to the preceding, we will have, observing the quantities which destroy each other, and likewise those which are equal to cypher,  $(M + \Sigma.m.) \left\{ \frac{d^2\zeta}{dt^2} \right\} + \Sigma.m. \left\{ \frac{d^2x}{dt^2} \right\} = 0$ , and by integrating we have  $(M + \Sigma.m.). \left\{ \frac{d\zeta}{dt} \right\} + \Sigma.m. \left\{ \frac{dx}{dt} \right\} = d$ ,  $\therefore (M + \Sigma.m).\zeta + \Sigma.m.x = c + dt$ , and  $\therefore$  if  $\frac{c}{M + \Sigma.m.} = a \frac{d}{M + \Sigma.m.} = b$ , we shall have  $\zeta =$  the expression given in the text.

$$\zeta = a + bt - \frac{\Sigma.m.x}{M+\Sigma.m};$$

$a$  and  $b$  being two constant arbitrary quantities. We will obtain also

$$\Pi = a' + b't - \frac{\Sigma.my}{M+\Sigma.m};$$

$$\gamma = a'' + b''t - \frac{\Sigma.m.z}{M+\Sigma.m};$$

$a'$ ,  $b'$ ,  $a''$ ,  $b''$ , being constant arbitrary quantities: we shall thus obtain the absolute motion of  $M$  in space, when the relative motions of  $m$ ,  $m'$ , &c., about it, are known.

If we multiply the differential equation in  $x$ , by

$$-my+m.\frac{\Sigma.my}{M+\Sigma.m};$$

and the differential equation in  $y$ , by

$$mx-m.\frac{\Sigma.mx}{M+\Sigma.m};$$

and in like manner, the differential equation in  $x'$ , by

$$-m'y'+m'.\frac{\Sigma.my}{M+\Sigma.m};$$

and the differential equation in  $y'$ , by

$$m'x'-m'.\frac{\Sigma.m.x}{M+\Sigma.m};$$

and if the same operations be performed on the coordinates of the other bodies of the system, by adding all these equations together, and observing that by the nature of the function  $\lambda$ ,

$$0 = \Sigma.x. \left\{ \frac{d\lambda}{dy} \right\} - \Sigma.y. \left\{ \frac{d\lambda}{dx} \right\};$$

$$0 = \Sigma. \left\{ \frac{d\lambda}{dx} \right\}; \quad 0 = \Sigma. \left\{ \frac{d\lambda}{dy} \right\};$$

we will obtain

$$0 = \Sigma.m. \frac{(xd^2y - yd^2x)}{dt^2} - \frac{\Sigma.mx}{M + \Sigma.m}. \Sigma.m. \frac{d^2y}{dt^2} + \frac{\Sigma.my}{M + \Sigma.m}. \Sigma.m. \frac{d^2x}{dt^2}; *$$

\* Performing these operations, the differential equation in  $x$  becomes  $= - my. \frac{d^2x}{dt^2} - M.m. \frac{yx}{r^3} - my \Sigma \frac{mx}{r^3} + y. \left\{ \frac{d\lambda}{dx} \right\} + \frac{m}{M + \Sigma.m}. \frac{d^2x}{dt^2}. \Sigma.my + \frac{Mmx}{r^3}. \frac{\Sigma.my}{M + \Sigma.m} + \frac{m}{M + \Sigma.m}. \Sigma. \frac{mx}{r^3}. \Sigma.my - \frac{\Sigma.my}{M + \Sigma.m}. \left\{ \frac{d\lambda}{dx} \right\}$ ; and corresponding operations being performed on the differential equations in  $x'$ ,  $x''$ , &c. we obtain, by adding them all together,  $- \Sigma.my. \frac{d^2x}{dt^2} - M.\Sigma.m. \frac{yx}{r^3} - \Sigma.my \Sigma. \frac{mx}{r^3} + \Sigma.y. \left\{ \frac{d\lambda}{dx} \right\} + \frac{\Sigma.my}{M + \Sigma.m}. \Sigma.m. \frac{d^2x}{dt^2} + \frac{\Sigma.my.M}{M + \Sigma.m}. \Sigma. \frac{mx}{r^3} + \frac{\Sigma.m.\Sigma.my}{M + \Sigma.m}. \Sigma. \frac{mx}{r^3} - \frac{-\Sigma.my}{M + \Sigma.m}. \Sigma. \left\{ \frac{d\lambda}{dx} \right\}$ ; multiplying the differential equations in  $y$ ,  $y'$ ,  $y''$ , &c. by  $mx - m. \frac{\Sigma.mx}{M + \Sigma.m}$ ,  $m'x' - m'. \frac{\Sigma.mx}{M + \Sigma.m}$ , &c., we obtain for the equation in  $y$ ,  $mx. \frac{d^2y}{dt^2} + M.m. \frac{xy}{r^3} + mx.\Sigma. \frac{my}{r^3} - x. \left\{ \frac{d\lambda}{dy} \right\} - \frac{m}{M + \Sigma.m}. \frac{d^2y}{dt^2}. \Sigma.mx - \frac{mM}{M + \Sigma.m}. \frac{y}{r^3}. \Sigma.mx - m. \frac{\Sigma.mx}{M + \Sigma.m}. \Sigma. \frac{my}{r^3} + \frac{\Sigma.mx}{M + \Sigma.m}. \left\{ \frac{d\lambda}{dy} \right\}$ ; if the same operation be performed for the equations in  $y'$  and  $y''$ , &c. we obtain, by adding these equations, and concinnating

$$\Sigma.mx.\Sigma. \frac{d^2y}{dt^2} + M.\Sigma. \frac{m.xy}{r^3} + \Sigma.mx.\Sigma. \frac{my}{r^3} - \Sigma.x. \left\{ \frac{d\lambda}{dy} \right\} - \frac{\Sigma.m. \frac{d^2y}{dt^2}. \Sigma.mx.}{M + \Sigma.m}.$$

of which equation the integral is

$$\begin{aligned} \text{Const}^{\text{nt}}. = \Sigma.m. \frac{(xdy - ydx)}{dt} - \frac{\Sigma.m.x}{M + \Sigma.m.} \cdot \Sigma.m. \frac{dy}{dt} \\ + \frac{\Sigma.my}{M + \Sigma.m.} \cdot \Sigma.m. \frac{dx}{dt}; \end{aligned}$$

or  $c =$

$$\begin{aligned} M.\Sigma.m. \left\{ \frac{xdy - ydx}{dt} \right\} + \\ \Sigma.mm'. \left\{ \frac{(x' - x).(dy' - dy) - (y' - y.(dx' - dx))}{dt} \right\}; \quad (4) \\ \rightarrow \frac{M}{M + \Sigma.m.} \cdot \Sigma. \frac{my}{r^3}, \Sigma.mx. - \frac{\Sigma.m.\Sigma.mx.}{M + \Sigma.m.} \cdot \Sigma. \frac{my}{r^3} + \Sigma.mx.\Sigma. \left\{ \frac{d\lambda}{dy} \right\}; \end{aligned}$$

this equation being added to the equation obtained, by taking the sum of the equations in  $x, x', \&c.$  gives

$$\begin{aligned} 0 = \Sigma.m. \left\{ \frac{xd^2y - yd^2x}{dt^2} \right\} + \Sigma. \left\{ y. \left\{ \frac{d\lambda}{dx} \right\} - x. \left\{ \frac{d\lambda}{dy} \right\} \right\} + \\ \left\{ \frac{\Sigma.my.}{M + \Sigma.m.} \Sigma.m. \frac{d^2x}{dt^2} - \frac{\Sigma.m.x}{M + \Sigma.m.} \Sigma.m. \frac{d^2y}{dt^2} \right\} + \frac{\Sigma.mx.\Sigma.}{M + \Sigma.m.} \left\{ \frac{d\lambda}{dy} \right\} - \frac{\Sigma.my.\Sigma.}{M + \Sigma.m.} \left\{ \frac{d\lambda}{dx} \right\}, \end{aligned}$$

the quantities which destroy each other, by the opposition of signs are omitted.

$$\Sigma. \left\{ y. \left\{ \frac{d\lambda}{dx} \right\} - x. \left\{ \frac{d\lambda}{dy} \right\} \right\} = 0, \text{ and } \Sigma. \left\{ \frac{d\lambda}{dx} \right\} = 0, \quad \Sigma. \left\{ \frac{d\lambda}{dy} \right\} = 0, \text{ see}$$

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The first term of the second member of this equation is evidently an exact differential, see page 2, and the integral of the remaining terms which do not vanish  $\rightarrow \frac{\Sigma.my.}{M + \Sigma.m.} \cdot \Sigma.m. \frac{dx}{dt} -$

$$\int \frac{\Sigma.m.dy}{M + \Sigma.m.} \cdot \Sigma.m. \frac{dx}{dt} - \frac{\Sigma.m.x}{M + \Sigma.m.} \cdot \Sigma.m. \frac{dy}{dt} + \int \frac{\Sigma.mdx}{M + \Sigma.m.} \cdot \Sigma.m. \frac{dy}{dt}.$$

$\epsilon$  being\* a constant arbitrary quantity. By a similar process we may obtain the two following integrals :

- \* If there are but three bodies

$$\begin{aligned} \text{const.} = & m. \left\{ \frac{xdy - ydx}{dt} \right\} + m'. \left\{ \frac{x'dy' - y'dx'}{dt} \right\} + m''. \left\{ \frac{x''dy'' - y''dx''}{dt} \right\} - \\ & \frac{mx}{M+m+m'+m''} \cdot \frac{mdy}{dt} - \frac{m'x'.m'dy'}{(M+m+m'+m'')dt} - \frac{m''x''.m''dy''}{(M+m+m'+m'')dt} - \frac{mx.m'dy'}{(M+m+m'+m'')dt} - \\ & \frac{mx.m''dy''}{(M+m+m'+m'')dt} - \frac{m'x'.m'dy}{(M+m+m'+m'')dt} - \frac{m'x'.m''dy''}{(M+m+m'+m'')dt} - \frac{m''x''.m'dy}{(M+m+m'+m'')dt} - \\ & \frac{m''x''.m'dy'}{(M+m+m'+m'')dt} + \frac{my.mdx}{(M+m+m'+m'')dt} + \frac{m'y'.m'dx'}{(M+m+m'+m'')dt} + \frac{m''y''.m'dx''}{(M+m+m'+m'')dt} \\ & + \frac{my.m'dx'}{(M+m+m'+m'')dt} + \frac{my.m''dx''}{(M+m+m'+m'')dt} + \frac{m'y'.mdx}{(M+m+m'+m'')dt} + \frac{m'y'.m''dx''}{(M+m+m'+m'')dt} \\ & + \frac{m''y''.mdx}{(M+m+m'+m'')dt} + \frac{m''y''.m'dx'}{(M+m+m'+m'')dt}; \text{ multiplying both sides of this equation by } \\ & M+\Sigma.m. \text{ we have} \end{aligned}$$

$$\begin{aligned} M+\Sigma.m. \text{ Const.} = & M. \left\{ m. \frac{(xdy - ydx)}{dt} + m'. \frac{(x'dy' - y'dx')}{dt} + m''. \frac{(x''dy'' - y''dx'')}{dt} \right\} + \\ & mm'. \frac{(xdy - ydx + x'dy' - y'dx')}{dt} + mm''. \frac{(xdy - ydx + x'dy'' - y''dx')}{dt}, \\ & + m'm''. \frac{(x'dy' - y'dx' + x''dy'' - y''dx'')}{dt} + m^2. \frac{(xdy - ydx)}{dt} + m'^2. \frac{(x'dy' - y'dx')}{dt} \\ & + m''^2. \frac{(x''dy'' - y''dx'')}{dt} - m^2. \frac{(xdy - ydx)}{dt} - m'^2. \frac{(x'dy' - y'dx')}{dt} - m''^2. \frac{(x''dy'' - y''dx'')}{dt} \\ & + mm'. \frac{(ydx' - xdy')}{dt} + mm''. \frac{(ydx'' - xdy'')}{dt} + mm'. \frac{(y'dx - x'dy)}{dt} \\ & + m'm''. \frac{(y'dx'' - x'dy'')}{dt} + mm''. \frac{(y''dx - x''dy)}{dt} + m''m'. \frac{(y''dx' - x''dy')}{dt}; \text{ But} \\ & mm'. \frac{(xdy - ydx + x'dy' - y'dx')}{dt} + mm'. \frac{(ydx' - xdy')}{dt} + mm'. \frac{(y'dx - x'dy)}{dt} + \&c.= \\ & mm'. \frac{((x'-x).(dy' - dy) - (y' - y).(dx' - dx))}{dt}, \because \text{making the factors of } nm'', m'm'', \&c. \end{aligned}$$

$$c = M \cdot \Sigma.m. \frac{(xdz - zdx)}{dt} +$$

$$\Sigma.mm'. \left\{ \frac{(x' - x).(dz' - dz) - (z' - z).(dx' - dx)}{dt} \right\}; \quad (5)$$

$$c'' = M \cdot \Sigma.m. \frac{(ydz - zdy)}{dt} +$$

$$\Sigma.mm'. \left\{ \frac{(y' - y).(dz' - dz) - (z' - z).(dy' - dy)}{dt} \right\}; \quad (6)$$

$c'$  and  $c''$  being two new arbitrary quantities.

If we multiply the differential equation in  $x$ , by

$$2mdx - 2m \cdot \frac{\Sigma.m.dx}{M + \Sigma.m};$$

the differential equation in  $y$ , by

$$2mdy - 2m \cdot \frac{\Sigma.m.dy}{M + \Sigma.m};$$

the differential equation in  $z$ , by

$$2mdz - 2m \cdot \frac{\Sigma.m.dz}{M + \Sigma.m};$$

and if, in like manner, we multiply the differential equation in  $x'$ , by

also to coalesce, and obliterating the quantities which destroy each other, we have  $(M + \Sigma.m). \text{Const.} = c$  = the second member of the equation in the text, it is evident that the same proof is applicable to any number of bodies.

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$$2m'.dx' = -2m' \cdot \frac{\Sigma.m.dx}{M+\Sigma.m};$$

the differential equation in  $y'$ , by

$$2m'.dy' = -2m' \cdot \frac{\Sigma.m.dy}{M+\Sigma.m};$$

the differential equation in  $z'$ , by

$$2m'.dz' = -2m' \cdot \frac{\Sigma.m.dz}{M+\Sigma.m};$$

and so of the other bodies; if we then add together these different equations, observing that

$$0 = \Sigma. \left\{ \frac{d\lambda}{dx} \right\}; \quad 0 = \Sigma. \left\{ \frac{d\lambda}{dy} \right\}; \quad 0 = \Sigma. \left\{ \frac{d\lambda}{dz} \right\};$$

we will obtain

$$0 = 2\Sigma.m. \frac{(dx.d^2x + dy.d^2y + dz.d^2z)}{dt^4} - \frac{2\Sigma.mdx}{M+\Sigma.m} \cdot \Sigma.m. \frac{d^2x}{dt^2} - \frac{2\Sigma.mdy}{M+\Sigma.m} \cdot \Sigma.m. \frac{d^2y}{dt^2} - \frac{2\Sigma.mdz}{M+\Sigma.m} \cdot \Sigma.m. \frac{d^2z}{dt^2} + 2M.\Sigma. \frac{mdr}{r^2} - 2d\lambda;$$

\* The differential equation in  $x$ , being multiplied by this quantity becomes =  
 $2m. \frac{dx.a^2x}{dt^2} + M. \frac{2mx}{r^2} dx + 2mdx. \Sigma. \frac{mx}{r^3} - 2. \left( \frac{d\lambda}{dx} \right) dx - \frac{2\Sigma.mdx}{M+\Sigma.m} \cdot m. \frac{d^2x}{dt^2} - \frac{2M}{M+\Sigma.m}$   
 $\frac{mx}{r^3} \Sigma.mdx - \frac{2m}{M+\Sigma.m} \Sigma.mdx. \Sigma. \frac{mx}{r^3} + \frac{2}{M+\Sigma.m} \Sigma.mdx. \left( \frac{d\lambda}{dx} \right)$ , if corresponding operations be performed on the differential equations in  $x'$ ,  $x''$ , &c. we will obtain by adding them together,

$$2\Sigma.m. \frac{dx.d^2x}{dt^2} + M.2\Sigma. \frac{mx}{r^3} dx + 2\Sigma.mdx. \Sigma. \frac{mx}{r^3} - 2\Sigma. \left\{ \frac{d\lambda}{dx} \right\} dx -$$

which gives by integrating

$$\text{const.} = \Sigma.m. \frac{(dx^2 + dy^2 + dz^2)}{dt^2} - \frac{-(\Sigma.mdx)^2 - (\Sigma.mdy)^2 - (\Sigma.mdz)^2}{(M + \Sigma.m)dt^2} - 2M.\Sigma.\frac{m}{r} - 2\lambda,$$

or

$$h = M.\Sigma.m. \frac{(dx^2 + dy^2 + dz^2)}{dt^2} +$$

$$\Sigma.mm'. \left\{ \frac{(dx' - dx)^2 + (dy' - dy)^2 + (dz' - dz)^2}{dt^2} \right\}^* \\ G 2$$

$$\frac{2\Sigma.mdx.\Sigma.m. \frac{d^2x}{dt^2}}{M + \Sigma.m} - \frac{2M}{M + \Sigma.m} \cdot \Sigma.mdx.\Sigma. \frac{mx}{r^3} - \frac{2.\Sigma.m}{M + \Sigma.m} \cdot \Sigma.mdx.\Sigma. \frac{mx}{r^3} + \frac{2}{M + \Sigma.m} \cdot \Sigma.mdx.\Sigma.$$

$$\left\{ \frac{d\lambda}{dx} \right\}, \text{ this equation by reducing, and observing that } -\frac{2M}{M + \Sigma.m} \cdot \Sigma.mdx.\Sigma. \frac{mx}{r^3} - \frac{2\Sigma.m}{M + \Sigma.m} \cdot \Sigma.mdx.\Sigma. \frac{mx}{r^3} = -2\Sigma.mdx.\Sigma. \frac{mx}{r^3}, \text{ and also that } \frac{2}{M + \Sigma.m} \cdot \Sigma.mdx.\Sigma. \frac{d\lambda}{dx} = 0,$$

becomes

$$\frac{2\Sigma.mdx. \frac{d^2x}{dt^2} + M.2\Sigma. \frac{mx dx}{r^3} - 2\Sigma. \left\{ \frac{d\lambda}{dx} \right\}. dx - 2\Sigma. m dx. \Sigma.m. \frac{d^2x}{dt^2}}{M + \Sigma.m};$$

if this equation be added to the differential equations, which result by performing corresponding operations on the equations in  $y, y', y'', \&c. z, z', z'', \&c.$ , observing also that  $2xdx + 2ydy + 2zdz = 2rdr$ , we shall obtain the differential equation of the text.

$$\Sigma. \left\{ \frac{d\lambda}{dx} \right\}. dx + \Sigma. \left\{ \frac{d\lambda}{dy} \right\}. dy + \Sigma. \left\{ \frac{d\lambda}{dz} \right\}. dz = d\lambda \text{ see page 28.}$$

\* If there are but three bodies, we have by multiplying by  $(M + m + m' + m'')$ :  
 $\text{Const. } (M + m + m' + m'') = h$ ; and if we only consider the coordinates parallel to the axis of  $x$ , we will have  $M(mdx^2 + m'dx'^2 + m''dx''^2) + (m + m' + m'')$ .  
 $(mdx^2 + m'dx'^2 + m''dx''^2) - (m + m' + m'')dx + dx' + dx'' = M.\Sigma.mdx^2 + m^2dx^2 + m'^2dx'^2 + m''^2dx''^2 + mm'dx^2 + mm'dx'^2 + mm''dx^2 + mm''dx'^2 + m'm''dx'^2 + m'm''dx''^2 - m^2dx^2 - m'^2dx'^2 - m''^2dx''^2 - 2mm'dxdx' - 2mm''dxdx'' - 2m'm''dx'dx'' = M.\Sigma.mdx^2 + mm'(dx - dx')^2 + (mm''.(dx - dx'))^2 + m'm''.(dx' - dx'')^2 = M\Sigma.mdx^2 + \Sigma.mmn'(dx' - dx)^2$ ; similar expressions may be obtained for the differentials of the coordinates parallel to the axes of  $z$  and  $y$ , and if to these be added  $-(2M.\Sigma.m + 2\lambda)$  multiplied by  $M + \Sigma.m$ , we will have the expression in the text.

$$-\left\{2M.\Sigma.\frac{m}{r} + 2\lambda\right\}.(M+\Sigma.m); \quad (7)$$

$h$  being a constant arbitrary quantity. These different integrals were already obtained in the fifth chapter of the first book, relatively to a system of bodies which react on each other in any manner; but considering their utility in the theory of the system of the world, we thought it necessary to demonstrate them here again.

10. The preceding being the only integrals which have been obtained in the actual state of analysis; we are compelled to recur to the methods of approximation, and to avail ourselves of the facilities which the constitution of the system of the world furnishes us with for this object. One of the greatest arises from the circumstance of the solar system being distributed into partial systems, composed of the planets and their respective satellites; these systems are so constituted that the distances of the satellites from their primaries, are considerably less than the distance of the primary from the sun; it follows from this, that the action of the sun, being very nearly the same on the primary and on the satellites, they move very nearly in the same manner, as if they were only subject to the action of the primary. The following remarkable property also follows, from this arrangement of the planets and satellites, namely, that the motion of the centre of gravity of a planet, and of its satellites, is very nearly the same,\* as if all these bodies were concentrated in this centre.

In order to demonstrate this, let us suppose that the mutual distances of the bodies  $m$ ,  $m'$ , &c. are very small, compared with the distance of their centre of gravity, from the body  $M$ . Let

$$x = X+x; \quad y = Y+y; \quad z = Z+z;$$

$$x' = X+x'; \quad y' = Y+y'; \quad z = Z+z'; \\ \text{&c.};$$

\* See Princip. Math. Lib. 1st Prop. 65.

$X, Y, Z$ , being the coordinates of the centre of gravity of the system of bodies  $m, m', m'', \&c.$ ; the origin of these coordinates, as also that of the coordinates,  $x, y, z, x', y', z', \&c.$ , being at the centre of  $M$ . It is manifest that  $x, y, z, x', \&c.$  will be the coordinates of  $m, m', \&c.$  relatively to their common centre of gravity; we shall suppose these to be very small quantities of the first order, in relation to  $X, Y, Z$ . This being premised, we will obtain, as we have seen in the first book, the force which solicits the centre of gravity of the system parallel to any right line, by taking the sum of the forces, which solicit the bodies parallel to this line, multiplied respectively by their masses, and then dividing this sum by the sum of the masses. Moreover, we have seen in the same book, that the mutual action of bodies connected together in any manner, does not derange the motion of the centre of gravity of the system; and by No. 8, the mutual attraction of those bodies, does not alter this motion, consequently, in the investigation of the forces, which actuate the centre of gravity of the system, it is sufficient to consider the action of the body  $M$ , which does not belong to this system.

The action of the body  $M$  on  $m$ , resolved parallel to the axis of  $x$ , and in a direction tending from the origin is  $-\frac{Mx}{r^3}$ , therefore the entire force which solicits the centre of gravity of the system of bodies  $m, m', \&c.$  parallel to this line, is\*

$$-\frac{M \cdot \Sigma \cdot \frac{mx}{r^3}}{\Sigma \cdot m},$$

and by substituting in place of  $x$  and of  $r$ , their values, we have

\* By what has been stated in No. 20 of the first book, it appears that  $\frac{d^2 X}{dt^2} = \frac{\Sigma \cdot m \cdot P}{\Sigma \cdot m}$ ; now in the present case  $\Sigma \cdot m \cdot P = -M \Sigma \cdot \frac{mx}{r^3}$ , for  $P = -\frac{Mx}{r^3}$ .

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$$\frac{x}{r^3} = \frac{X+x_i}{((X+x_i)^2 + (Y+y_i)^2 + (Z+z_i)^2)^{\frac{3}{2}}}$$

If we neglect very small quantities of the second order, namely the squares, and the products of the variables  $x, y, z, x', &c.$ ; and if we denote by  $R$ , the distance  $\sqrt{X^2+Y^2+Z^2}$ , of the centre of gravity of the system, from the body  $M$ ; we shall obtain

$$\frac{x}{r^3} = \frac{X}{R^3} + \frac{x_i}{R^3} - 3X \cdot \frac{(Xx_i + Yy_i + Zz_i)}{R^5}, *$$

we shall have the values of  $\frac{x'}{r'^3}, \frac{x''}{r''^3}, &c.$  by distinguishing the letters  $x, y, z, &c.$  by one, two accents, &c.; but by the nature of the centre of gravity,

$$0 = \Sigma mx_i; \quad 0 = \Sigma my_i; \quad 0 = \Sigma mz_i;$$

therefore we will have, neglecting quantities of the second order,

$$\frac{M \cdot \Sigma \frac{mx}{r^3}}{\Sigma m} = -\frac{MX}{R^3};$$

$$* \frac{X+x_i}{((X+x_i)^2 + (Y+y_i)^2 + (Z+z_i)^2)^{\frac{3}{2}}} = (X+x_i)((X+x)^2 + (Y+y)^2 + (Z+z)^2)^{-\frac{3}{2}} =$$

$$\text{by neglecting quantities very small of the second order, } X(X^2 + 2Xx_i + Y^2 + 2Yy_i + Z^2 + 2Zz_i)^{-\frac{3}{2}} + x_i(X^2 + Y^2 + Z^2)^{-\frac{3}{2}} = X(X^2 + Y^2 + Z^2)^{-\frac{3}{2}} - \frac{3}{2}X(2Xx_i + 2Yy_i + 2Zz_i)R^6 + x_i(X^2 + Y^2 + Z^2)^{-\frac{3}{2}} = (\text{by substituting } R^2 \text{ for } X^2 + Y^2 + Z^2) \frac{X}{R^3} + \frac{x_i}{R^3} -$$

$$3X \cdot \frac{(Xx_i + Yy_i + Zz_i)}{R^5}; \quad \because \text{as } \Sigma mx_i = 0, \Sigma my_i = 0, \Sigma mz_i = 0, -M \cdot \frac{mx}{r^3} = \left( -\frac{MX \cdot \Sigma m}{R^3 \Sigma m} - \frac{\Sigma mx_i}{R^3 \Sigma m} + 3X \cdot \frac{(X\Sigma mx_i + Y\Sigma my_i + Z\Sigma mz_i)}{R^5 \Sigma m} \right)$$

$= -\frac{MX}{R^3}$ , for the two last terms of the second member of this equation vanish.

consequently, the centre of gravity of the system is sollicited by the action of the body  $M$  parallel to the axis of  $x$ , in very nearly the same manner as if all bodies of the system were concentrated in this centre. The same conclusion evidently obtains for the axes of  $y$  and of  $z$ , so that the forces by which the centre of gravity of the system is actuated parallel to these axes, by the action of  $M$ , are  $-\frac{MY}{R^3}$ ,  $-\frac{MZ}{R^3}$ .

When we consider the relative motion of the centre of gravity of the system about  $M$ , we should transfer in an opposite direction, the force which sollicits this body. This force resulting from the action of  $m$ ,  $m'$ ,  $m''$ , &c. on  $M$ , resolved parallel to  $x$ , and acting in a direction tending from their origin, is  $\Sigma \frac{mx}{r^3}$ ; if quantities of the second order are neglected, this function is by what precedes, equal to

$$\frac{X \cdot \Sigma m}{R^3}.$$

In like manner, the forces by which  $M$  is sollicited, in consequence of the action of the system, parallel to the axes of  $y$  and of  $z$ , in a direction tending from the origin, are

$$\frac{Y \cdot \Sigma m}{R^3}, \text{ and } \frac{Z \cdot \Sigma m}{R^3}.$$

It appears from this, that the action of the system on the body  $M$ , is very nearly the same, as if all the bodies were condensed in their common centre of gravity. By transferring to this centre, and with a contrary sign, the three preceding forces; this point will be sollicited parallel to the axes of  $x$ , of  $y$ , and of  $z$ , in its relative motion round  $M$ , by the three following forces:

$$-(M + \Sigma m) \cdot \frac{X}{R^3}; \quad -(M + \Sigma m) \cdot \frac{Y}{R^3}; \quad -(M + \Sigma m) \cdot \frac{Z}{R^3}.$$

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These forces are the same as if all the bodies  $m, m', m'', \&c.$  were united in their common centre of gravity;\* consequently neglecting very small quantities of the second order, this centre moves as if all the bodies were concentrated in this point.

\* The action of  $m$  on  $M$  resolved parallel to the axis of  $x = \frac{mx}{r^3}$ ,  $\therefore$  the sum of the actions of all the bodies  $m, m', m'', \&c.$  on  $M = \Sigma \frac{mx}{r^3}$ ,  $=$  by what precedes  $\frac{X\Sigma.m}{R^3}$ ;  $\therefore$  if this action be transferred to the centre of gravity, with a contrary sign, this centre in its relative motion about  $M$ , will be sollicited parallel to the axis of  $x$ , by the force  $-(M+\Sigma.m).\frac{X}{R^3}$ ; now if all the bodies  $m, m', m'', \&c.$  were concentrated in their common centre of gravity, this centre would be acted on parallel to axis of  $x$ , by the force  $-(M+\Sigma.m)X$ ,  $\therefore$  this centre moves as if all the bodies were concentrated in it, consequently it describes very nearly an ellipse about  $M$ , the quantities which are neglected are of the order of the square and higher powers of  $x$ , and it is easy to shew, that the aberration of the force, by which the common centre of gravity is sollicited, from the inverse ratio of the square of the distance, is much less than the aberration of the forces solliciting any of the bodies composing the system, from the inverse square of the distance. For if there are but three bodies, and if the distance of the greatest  $M$  from the remaining  $m$  and  $m'$ , be much greater than the distance of  $m$  from  $m'$ , then if  $R$  be the distance of  $M$  from the common centre of gravity of  $m$  and  $m'$ ,  $p$  and  $q$  the distances of this centre from  $m$  and  $m'$ , respectively, and  $\varpi$  the angle which  $r=p+q$ , makes with  $R$ , the distance of  $M$  from  $m, = R-p \cos. \varpi$ , the distance of  $M$  from  $m' = R+q \cos. \varpi$ ,  $\therefore$  the attraction of  $M$  on  $m$ , resolved parallel to

$$R = \frac{MR}{(R-p \cos. \varpi)^3} = MR(R^{-3} + 3R^{-4}p \cos. \varpi + 6R^{-5}p^2 \cos^2 \varpi + \&c.) = \frac{M}{R^2} + \frac{3Mp \cos. \varpi}{R^3} + \frac{6Mp^2 \cos^2 \varpi}{R^4} + \&c.;$$

in like manner, the action of  $M$  on  $m'$ , resolved parallel to  $R = \frac{MR}{(R+q \cos. \varpi)^3} = \frac{M}{R^2} - \frac{3Mq \cos. \varpi}{R^3} + \frac{6Mq^2 \cos^2 \varpi}{R^4}, \&c.$

now we know from what has been already established in the first book, that the accelerating force by which the centre of gravity of  $m$  and  $m'$ , is sollicited in the direction of  $R$ , is obtained by dividing the sum of the motive forces, by which  $m$  and  $m'$  are sollicited in this direction, by  $m+m'$ ,  $\therefore$  this force is  $=$  to

$$\left\{ \frac{MmR}{(R-p \cos. \varpi)^3} + \frac{Mm'R}{(R+q \cos. \varpi)^3} \right\} \cdot \frac{1}{m+m'} = \text{by substitution}$$

It follows from what precedes, that if there are several systems, of which the centres of gravity are at considerable distances from each other, compared with the respective distances of the bodies of each

## PART I.—BOOK II.

## H

$$\left\{ \frac{Mm}{R^2} + \frac{3Mmp \cdot \cos. \omega}{R^3} + \frac{6Mmp^2 \cdot \cos.^2 \omega}{R^4} + \text{etc.} + \frac{Mm'}{R^2} - \frac{3Mm'q \cdot \cos. \omega}{R^3} \right. \\ \left. + \frac{6Mm'q^2 \cdot \cos.^2 \omega}{R^4} + \text{etc.} \right\} \cdot \frac{1}{m+m'} = \frac{M}{R^2} \cdot \frac{m+m'}{m+m'} + \frac{3M \cdot \cos. \omega}{R^3 \cdot (m+m')} \cdot (mp - m'q)$$

$\frac{6M \cos. \omega}{R^4 \cdot (m+m')} \cdot (mp^2 + m'q^2) + \text{etc.}$ , the first term gives the law of elliptic motion; the second term vanishes by the nature of the centre of gravity,  $\therefore$  the third and following terms are those which cause an aberration from the law of elliptic motion in the centre of gravity. The actions of  $m$  and  $m'$  on  $M$ , resolved parallel to  $R$ , are respectively

$\frac{mR}{(R-p \cdot \cos. \omega)^3}, \frac{m'R}{(R+q \cdot \cos. \omega)^3}$ , which become by reducing,  $\frac{m}{R^2}, \frac{m'}{R^2}$ , and if these be transferred to  $M$  with a contrary sign, the entire force by which the centre is urged, is  $\frac{M+m+m'}{R^2}$ . It appears from this discussion that the centre of gravity of the earth and

moon describes very nearly an ellipse about the sun; now a comparison of this expression, with that which gives the action of  $M$  on  $m$ , disturbed by the action of  $m'$  on  $M$  and on  $m$ , shews that the curve described by the centre of gravity, approaches much nearer to an ellipse than the curve described by  $m$ , for the force on  $m$ , acting in the direction of  $R-p \cdot \cos. \omega$

$$= \frac{M+m}{(R-p \cdot \cos. \omega)^2} + \frac{m' \cdot (R-p \cdot \cos. \omega)}{r^3} + m' \cdot \left\{ \frac{1}{(R+q \cdot \cos. \omega)^2} - \frac{R+q \cdot \cos. \omega}{r^3} \right\}.$$

$\cos. \theta, \theta$  being the angle at which  $r$  is inclined to a radius drawn from  $M$  to  $m$ , this expression becomes by rejecting very small quantities of the second and higher orders,

$$\frac{M+m+m'}{(R-p \cdot \cos. \omega)^2} + \frac{m' \cdot \cos. \theta}{(R+q \cdot \cos. \omega)^2}, \text{ and the last term is evidently greater than}$$

$$\frac{6M \cdot \cos. \omega}{R^4} \cdot \frac{mp^2 + m'q^2}{m+m'}. \text{ The force which is perpendicular to } R-p \cdot \cos. \omega \text{ is equal to}$$

$$m' \cdot \left\{ \frac{R+q \cdot \cos. \omega}{r^3} - \frac{1}{(R+q \cdot \cos. \omega)^2} \right\} \cdot \sin. \theta = \text{by reducing } \frac{m' \cdot \sin. \theta}{(R+q \cdot \cos. \omega)^2}; \text{ but}$$

if the force of  $M$  on  $m$ , be resolved parallel to  $r$  it will be  $= \frac{Mp}{(R-p \cdot \cos. \omega)^2}$ , and the

force of  $M$  on  $m'$  parallel to  $r = \frac{Mq}{(R+q \cdot \cos. \omega)^2}$ ,  $\therefore$  the accelerating force on the centre of

$$\text{gravity parallel to } r = \left\{ \frac{Mmp}{(R-p \cdot \cos. \omega)^3} - \frac{Mm'q}{(R+q \cdot \cos. \omega)^3} \right\} \frac{1}{m+m'} = \left\{ \frac{Mmp}{R^3} \right.$$

$$\left. + \frac{3Mmp^2 \cdot \cos. \omega}{R^4} - \frac{Mm'q}{R^3} + \frac{3Mm'q^2 \cdot \cos. \omega}{R^4} \right\} \frac{1}{m+m'} = \text{because } mp - m'q = 0,$$

$\frac{3M \cos. \omega}{R^4 \cdot (m+m')} (mp^2 + m'q^2)$ ; the part of this force which is perpendicular to  $R$  disturbs the

system ; these centres will move very nearly in the same manner, as if the bodies of the respective systems were concentrated in them ; for the action of the first system on each body of the second system, is, by what precedes, very nearly the same, as if all the bodies of the first system were united in their common centre of gravity ; the action of the first system on the centre of gravity of the second, will, therefore, by what has been just established, be the same as in this hypothesis, from which we may conclude generally, that the reciprocal action of different systems, on their respective centres of gravity, is the same as if the bodies of each system

proportionality of the areas described by the centre of gravity to the times, and it is evidently less than  $\frac{m'. \sin. \theta}{(R+q. \cos. \varpi)^2}$ , See Princip. Math. Lib. 1. Prop. 66. Cor. 3, 4, &c.

The distance of the centre of gravity from  $M$  differs from the distance of  $m$  from  $M$  resolved parallel to  $R$ , by  $p. \cos. \varpi = \frac{m'}{m+m'} \cdot r. \cos. \varpi$ . (by the nature of the centre of gravity). In like manner the aberration in longitude  $= p. \sin. \varpi = \frac{m'}{m+m'} \cdot r. \sin. \varpi$ ,  $\because$  it varies as the sine of the angle of elongation of  $M$  from  $m$  ; if  $s$  be the tangent of the latitude of the earth, the distance of the earth from the plane passing through  $M$  and the centre of gravity of  $m$  and  $m' = sp = rs \cdot \frac{m'}{m+m'}$ , now  $s = \tan. \phi. \sin. (v-\theta)$ ,  $\phi$  being the inclination of the orbit of the moon to the above mentioned plane, and  $v-\theta$  being  $=$  to the distance of the moon from her node. The distance from this plane, as seen from the  $M = \frac{m'}{m+m'} \cdot \frac{rs}{R}$ . See Book 7, and Newton Princip. Math. Prop. 65, 66, 67, 68. What has been stated at the commencement of this note, shews the truth of Newton's 65 and 67 Prop. Lib. 1. And it would be easy to demonstrate, as Newton states in Prop. 64, that when the force varies as the distance, the centre of gravity describes an *accurate* ellipse about  $M$ , for the force solliciting  $m$  parallel to the axis of  $x$ ,  $= -Mx$ ,  $\therefore$  the force which solicits the centre of gravity parallel to this axis,  $- \frac{M \cdot \Sigma mx}{\Sigma m} = -MX - \frac{M \cdot \Sigma mx}{\Sigma m}$ , now this last term vanishes, if we add to this force, the force  $\Sigma mx = X\Sigma m + \Sigma m \cdot x$ , by which  $M$  is sollicitated in a contrary direction, the entire force on the centre of gravity parallel to this axis  $= -(M + \Sigma m)X$ ,  $\therefore$  the centre of gravity describes an accurate ellipse, and  $m$  describes an ellipse about the common centre of gravity of  $M$  and  $m'$  ; the periodic time in this ellipse depends on the number of bodies composing the system, and it varies inversly as the square root of the sum of the masses.

were concentrated in them, and that consequently those centres move, as they would do, in the case of this concentration. It is manifest, that this conclusion equally obtains, whether the bodies of each system are free, or connected together in any manner whatever, because their mutual action does not affect the motion of their common centre of gravity.

Therefore, the system of a planet and its satellites acts very nearly in the same manner on the other bodies of the solar system, as if the planet and its satellites were united in their common centre of gravity ; and this centre is attracted by the several bodies of the solar system, as in this hypothesis.

Each of the heavenly bodies, being composed of an infinite number of molecules, endowed with an attractive power, and their dimensions being very small compared with its distance from the other bodies of the system of the world ; its centre of gravity is attracted very nearly in the same manner, as if the entire mass was concentrated in it, and it acts itself on the several bodies of the system, as on this hypothesis ; therefore in the investigation of the motion of the centre of gravity of the heavenly bodies, we may consider these bodies as so many massive points, placed in their centres of gravity. But the sphericity of the planets, and of their satellites, render this hypothesis, already very near to the truth, still more exact. In fact, these several bodies may be conceived to be made up of strata very nearly spherical, and of a density which varies according to any given law ; and we now proceed to show that the action of a spherical stratum on a body, which is exterior to it, is the same as if its mass was united in its centre. For this purpose, we will establish some general propositions, relative to the attractions of spheroids, which will be very useful in the sequel.

11. Let  $x, y, z$ , represent the three coordinates of the attracted point, which we will denote by  $m$  ; let  $dM$  represent a molecule of the spheroid, and  $x', y', z'$ , the coordinates of this molecule,  $\rho$  denoting the density, which is a function of  $x', y', z'$ , independent of  $x, y, z$  ; we will have

$$dM = \rho \cdot dx' \cdot dy' \cdot dz'.$$

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The action of  $dM$  on  $m$ , resolved parallel to the axis of  $x$ , and tending towards the origin, will be

$$\frac{\rho \cdot dx' \cdot dy' \cdot dz' \cdot (x - x')}{((x - x')^2 + (y - y')^2 + (z - z')^2)^{\frac{3}{2}}};$$

and it will consequently be equal to

$$-\left\{ d \cdot \frac{\rho \cdot dx' \cdot dy' \cdot dz'}{\sqrt{(x - x')^2 + (y - y')^2 + (z - z')^2}} \right\};$$

therefore if  $V$  denote the integral

$$\int \frac{\rho \cdot dx' \cdot dy' \cdot dz'}{\sqrt{(x - x')^2 + (y - y')^2 + (z - z')^2}}; *$$

\* The action of  $dM$  on  $m$ , is expressed by  $\frac{\rho \cdot dx' \cdot dy' \cdot dz'}{(x - x')^2 + (y - y')^2 + (z - z')^2}$ ,  $\therefore$  the force parallel to the axis of  $x$ :  $\frac{\rho \cdot dx' \cdot dy' \cdot dz'}{(x - x')^2 + (y - y')^2 + (z - z')^2} :: (x - x')$ :

$\sqrt{(x - x')^2 + (y - y')^2 + (z - z')^2}$ , consequently it is  $= \frac{\rho \cdot dx' \cdot dy' \cdot dz'}{((x - x')^2 + (y - y')^2 + (z - z')^2)^{\frac{3}{2}}}$ , the expression  $\frac{\rho \cdot dx' \cdot dy' \cdot dz'}{\sqrt{(x - x')^2 + (y - y')^2 + (z - z')^2}}$ , differenced with respect to  $x$ , and divided by  $dx$ , becomes  $= \frac{\rho \cdot dx' \cdot dy' \cdot dz' \cdot (x - x')}{((x - x')^2 + (y - y')^2 + (z - z')^2)^{\frac{3}{2}}}$ ;  $\therefore$  this expression or

$-\left\{ d \cdot \frac{\rho \cdot dx' \cdot dy' \cdot dz'}{\sqrt{(x - x')^2 + (y - y')^2 + (z - z')^2}} \right\}$ , expresses the action of a molecule of the spheroid, on a point without the surface of the spheroid, consequently, if we take the sum of the corresponding expressions for all the molecules of the spheroid, i. e. if we take

$$-\left\{ d \cdot \int \frac{\rho \cdot dx' \cdot dy' \cdot dz'}{\sqrt{(x - x')^2 + (y - y')^2 + (z - z')^2}} \right\} = -\left\{ \frac{dV}{dx} \right\}, \text{ this quantity expresses the}$$

action of the spheroid, on the point  $m$ , resolved parallel to the axis of  $x$ ; the characteristic,  $d$  refers solely to the coordinates  $x, y, z$ , it does not denote an operation the reverse of that indicated by the characteristic  $\int$ .

extended to the entire mass of the spheroid;  $-\left\{ \frac{dV}{dx} \right\}$  will represent the entire action of the spheroid on the point  $m$ , resolved parallel to the axis of  $x$ , and directed towards their origin.  $V$  is the sum of the molecules of the spheroid, divided by their respective distances from the point attracted; in order to obtain the attraction of the spheroid on this point, we should consider  $V$  as a function of three rectangular coordinates, of which one may be parallel to this line, and then take the differential of the function, with respect to this coordinate; the coefficient of this differential, affected with a contrary sign, will express the attraction of the spheroid parallel to the given line, and directed towards the origin of the coordinate to which it is parallel.

Denoting the function  $((x-x')^2 + (y-y')^2 + (z-z')^2)^{-\frac{1}{2}}$ , by  $\epsilon$ , we will have

$$V = \int \epsilon \cdot dx' \cdot dy' \cdot dz'.$$

As the integration only respects the variables  $x'$ ,  $y'$ ,  $z'$ , it is manifest that we will have

$$\begin{aligned} \left\{ \frac{d^2 V}{dx^2} \right\} + \left\{ \frac{d^2 V}{dy^2} \right\} + \left\{ \frac{d^2 V}{dz^2} \right\} &= \int \epsilon \cdot dx' \cdot dy' \cdot dz' \\ \left\{ \left\{ \frac{d^2 \epsilon}{dx^2} \right\} + \left\{ \frac{d^2 \epsilon}{dy^2} \right\} + \left\{ \frac{d^2 \epsilon}{dz^2} \right\} \right\}; \end{aligned}$$

but we have

$$\begin{aligned} 0 &= \left\{ \frac{d^2 \epsilon}{dx^2} \right\} + \left\{ \frac{d^2 \epsilon}{dy^2} \right\} + \left\{ \frac{d^2 \epsilon}{dz^2} \right\}; \\ \bullet \quad \frac{d\epsilon}{dx} &= \frac{-(x-x')}{((x-x')^2 + (y-y')^2 + (z-z')^2)^{\frac{3}{2}}} \therefore \frac{d^2 \epsilon}{dx^2} = \frac{-1}{((x-x')^2 + (y-y')^2 + (z-z')^2)^{\frac{5}{2}}}, \\ + \frac{3(x-x')^2}{((x-x')^2 + (y-y')^2 + (z-z')^2)^{\frac{5}{2}}} &= \frac{-(x-x')^2 - (y-y')^2 - (z-z')^2 + 3(x-x')^2}{((x-x')^2 + (y-y')^2 + (z-z')^2)^{\frac{5}{2}}}, \end{aligned}$$

consequently we will have also

$$0 = \left\{ \frac{d^2 V}{dx^2} \right\} + \left\{ \frac{d^2 V}{dy^2} \right\} + \left\{ \frac{d^2 V}{dz^2} \right\}; \quad (\text{A})$$

This remarkable equation will be extremely useful in the theory of the figure of the heavenly bodies ; we may make it to assume other forms, which will in different circumstances be more convenient ; for instance, let a radius be drawn from the origin of the coordinates to the point attracted, which radius we will represent by  $r$ , let  $\theta$  be equal to the angle, which this radius makes with the axis of  $x$ , and  $w$  the angle which the plane passing through  $r$  and this axis, makes with the plane of the co-ordinates  $x$  and  $y$  ; we will have

$$x = r \cdot \cos. \theta; \quad y = r \cdot \sin. \theta \cdot \cos. w; \quad z = r \cdot \sin. \theta \cdot \sin. w;$$

consequently we shall obtain

$$r = \sqrt{x^2 + y^2 + z^2}; \quad \cos. \theta = \frac{x}{\sqrt{x^2 + y^2 + z^2}}; \quad \tan. w = \frac{z}{y};$$

by means of these expressions, we can obtain the partial differences of

in like manner,  $\frac{d^2 \xi}{dy^2}$ ,  $\frac{d^2 \xi}{dz^2}$ , are respectively equal to

$$\frac{-(x-x')^2 - (y-y')^2 - (z-z')^2 + 3(y-y')^2}{((x-x')^2 + (y-y')^2 + (z-z')^2)^{\frac{5}{2}}}, \quad \frac{-(x-x')^2 - (y-y')^2 - (z-z')^2 + 3(z-z')^2}{((x-x')^2 + (y-y')^2 + (z-z')^2)^{\frac{5}{2}}}$$

$$\therefore \frac{d^2 \xi}{dx^2} + \frac{d^2 \xi}{dy^2} + \frac{d^2 \xi}{dz^2} =$$

$$\frac{-3(x-x')^2 - 3(y-y')^2 - 3(z-z')^2 + 3(x-x')^2 + 3(y-y')^2 + 3(z-z')^2}{((x-x')^2 + (y-y')^2 + (z-z')^2)^{\frac{5}{2}}} = 0.$$

$r$ ,  $\theta$ , and  $\varpi$ , with respect to the variables  $x$ ,  $y$ ,  $z$ ; from which we can deduce the values of  $\left\{ \frac{d^2 V}{dx^2} \right\}$ ,  $\left\{ \frac{d^2 V}{dy^2} \right\}$ ,  $\left\{ \frac{d^2 V}{dz^2} \right\}$ , in partial differences of  $V$ , with respect to the variables  $r$ ,  $\theta$ , and  $\varpi$ . As we shall have occasion frequently to consider these transformations of partial differences; it will be useful here to trace the principle of them.  $V$  being considered first as a function of the variables  $x$ ,  $y$ ,  $z$ , and then, of the variables  $r$ ,  $\theta$ , and  $\varpi$ , we have

$$\left\{ \frac{dV}{dx} \right\} = \left\{ \frac{dV}{dr} \right\} \cdot \left\{ \frac{dr}{dx} \right\} + \left\{ \frac{dV}{d\theta} \right\} \cdot \left\{ \frac{d\theta}{dx} \right\} + \left\{ \frac{dV}{d\varpi} \right\} \cdot \left\{ \frac{d\varpi}{dx} \right\}. *$$

In order to obtain the partial differences,  $\left\{ \frac{dr}{dx} \right\}$ ,  $\left\{ \frac{d\theta}{dx} \right\}$ ,  $\left\{ \frac{d\varpi}{dx} \right\}$ , it is only necessary to make  $x$  the sole variable in the preceding expressions for  $r$ , cos.  $\theta$ , and tan.  $\varpi$ , consequently, if we difference these expressions, we will have

$$|\left\{ \frac{dr}{dx} \right\} = \cos. \theta; \left\{ \frac{d\theta}{dx} \right\} = -\frac{\sin. \theta}{r}; \frac{d\varpi}{dx} = 0;$$

$$\begin{aligned} * \left\{ \frac{dr}{dx} \right\} &= \frac{x}{\sqrt{x^2+y^2+z^2}} = \frac{x}{r} = \cos. \theta; -\left\{ \frac{d\theta}{dx} \right\} \cdot \sin. \theta = \frac{1}{\sqrt{x^2+y^2+z^2}} \\ &- \frac{x^2}{(x^2+y^2+z^2)^{\frac{3}{2}}} = \frac{y^2+z^2}{(x^2+y^2+z^2)^{\frac{3}{2}}}, \text{ and as } \frac{y^2+z^2}{x^2+y^2+z^2} = \sin. ^2 \theta; \left\{ \frac{d\theta}{dx} \right\} = -\frac{\sin. \theta}{r}, \\ \text{by substituting for } \left\{ \frac{dr}{dx} \right\}, \left\{ \frac{d\theta}{dx} \right\}, \text{ we obtain the value of } \left\{ \frac{dV}{dx} \right\}, \text{ which has been given} \\ \text{in the text.} \end{aligned}$$

which gives

$$\left\{ \frac{dV}{dx} \right\} = \cos. \theta. \left\{ \frac{dV}{dr} \right\} - \frac{\sin. \theta}{r}. \left\{ \frac{dV}{d\theta} \right\}.$$

By this means we can obtain the partial difference  $\left\{ \frac{dV}{dx} \right\}$ , in partial differences of the function  $V$ , taken with respect to the variables  $r$ ,  $\theta$ , and  $w$ . By differencing this value of  $\left\{ \frac{dV}{dx} \right\}$  a second time, we shall obtain the difference  $\left\{ \frac{d^2 V}{dx^2} \right\}$  in terms of the partial differences of  $V$ , taken relatively to the variables  $r$ ,  $\theta$ , and  $w$ . We can obtain, by a similar process, the values of  $\left\{ \frac{d^2 V}{dy^2} \right\}$ , and  $\left\{ \frac{d^2 V}{dz^2} \right\}$ .

By the preceding operations, we can transform the equation (A) into the following :

$$0 = \left\{ \frac{d^2 V}{d\theta^2} \right\} + \frac{\cos. \theta}{\sin. \theta} \cdot \left\{ \frac{dV}{d\theta} \right\} + \left\{ \frac{d^2 V}{d\omega^2} \right\} + r. \left\{ \frac{d^2 rV}{dr^2} \right\}; \quad (B)^*$$

$$\begin{aligned} \left\{ \frac{dr}{dx} \right\} &= \frac{x}{r}; \quad \therefore \left\{ \frac{d^2 r}{dx^2} \right\} = \frac{1}{r} - \frac{x^2}{r^3} = \frac{y^2+z^2}{r^3} = \frac{\sin. ^2 \theta}{r}; \quad \left\{ \frac{d\theta}{dx} \right\} = -\frac{\sin. \theta}{r} \\ &= -\frac{\sqrt{y^2+z^2}}{r^2}, \quad \therefore \left\{ \frac{d^2 \theta}{dx^2} \right\} = \frac{2x\sqrt{y^2+z^2}}{r^4} = \frac{2\sin. \theta. \cos. \theta}{r^2}; \quad \left\{ \frac{d\omega}{dx} \right\} \text{ and} \\ \left\{ \frac{d^2 \omega}{dx^2} \right\} &= 0; \quad \therefore \left\{ \frac{d^2 V}{dx^2} \right\} = \frac{d^2 V}{dr^2} \cdot \frac{dr^2}{dx^2} + \frac{dV}{dr} \cdot \frac{d^2 r}{dx^2} + \frac{d^2 V}{d\theta^2} \cdot \frac{d\theta^2}{dx^2} + \frac{dV}{d\theta} \cdot \frac{d^2 \theta}{dx^2} = \\ &\quad \frac{d^2 V}{dr^2} \cdot \cos. ^2 \theta + \frac{dV}{dr} \cdot \frac{\sin. ^2 \theta}{r} + \frac{d^2 V}{d\theta^2} \cdot \frac{\sin. ^2 \theta}{r^2} + \frac{dV}{d\theta} \cdot \frac{2\sin. \theta. \cos. \theta}{r^2}; \\ \left\{ \frac{dr}{dy} \right\} &= \frac{y}{r} = \sin. \theta. \cos. w; \quad \left\{ \frac{d^2 r}{dy^2} \right\} = \frac{1}{r} - \frac{y^2}{r^3} = \frac{x^2+z^2}{r^3} = \frac{\cos. ^2 \theta + \sin. ^2 \theta. \sin. ^2 w}{r}; \\ -\left\{ \frac{d\theta}{dy} \right\} \cdot \sin. \theta. &= -\frac{xy}{r^2}, \quad \therefore \left\{ \frac{d\theta}{dy} \right\} = \frac{xy}{\sqrt{y^2+z^2)r^2}}, \text{ by substituting for } -\sin. \theta \text{ its} \end{aligned}$$

value  $\frac{\sqrt{y^2+z^2}}{r}$ ; and by substituting  $r \cos. \theta$  for  $x$ , and  $r \sin. \theta \cos. \varpi$  for  $y$ , we obtain,

$$\frac{xy}{\sqrt{y^2+z^2} \cdot r^2} = \frac{\cos. \theta \cos. \varpi}{r} \quad \therefore$$

$$\begin{aligned} \left\{ \frac{d^2 \theta}{dy^2} \right\} &= \frac{x}{\sqrt{y^2+z^2} \cdot r^2} - \frac{xy^2}{(y^2+z^2)^{\frac{3}{2}} \cdot r^2} - \frac{2xy^2}{\sqrt{y^2+z^2} \cdot r^4} = \frac{\cos. \theta}{\sin. \theta \cdot r^2} - \frac{\cos. \theta \cos. \varpi}{\sin. \theta \cdot r^2} \\ &- \frac{2 \cos. \theta \sin. \theta \cos. \varpi}{r^2}; \left\{ \frac{d\varpi}{dy} \right\} = d \cdot \frac{\tan. \varpi}{dy} = -\frac{z}{y^2} \div \left\{ \left( 1 + \frac{z^2}{y^2} \right) \right\} = -\frac{z}{y^2+z^2} = \\ &- \frac{\sin. \varpi}{\sin. \theta \cdot r}; \left\{ \frac{d^2 \varpi}{dy^2} \right\} = \frac{2yz}{(y^2+z^2)^2} = \frac{2 \cdot \sin. \varpi \cos. \varpi}{\sin. \theta \cdot r^2}; \therefore \left\{ \frac{dV}{dy} \right\} = \left\{ \frac{dV}{dr} \right\} \cdot \left\{ \frac{dr}{dy} \right\} \\ &+ \left\{ \frac{dV}{d\theta} \right\} \cdot \left\{ \frac{d\theta}{dy} \right\} + \left\{ \frac{dV}{d\varpi} \right\} \cdot \left\{ \frac{d\varpi}{dy} \right\} = \left\{ \frac{dV}{dr} \right\} \cdot \sin. \theta \cos. \varpi + \left\{ \frac{dV}{d\theta} \right\} \cdot \\ &\frac{\cos. \theta \cos. \varpi}{r} - \left\{ \frac{dV}{d\varpi} \right\} \cdot \frac{\sin. \varpi}{r \sin. \theta}; \therefore \left\{ \frac{d^2 V}{dy^2} \right\} \\ &= \left\{ \frac{d^2 V}{dr^2} \right\} \cdot \left\{ \frac{dr^2}{dy^2} \right\} + \left\{ \frac{dV}{dr} \right\} \cdot \left\{ \frac{d^2 r}{dy^2} \right\} + \left\{ \frac{d^2 V}{d\theta^2} \right\} \cdot \left\{ \frac{d\theta^2}{dy^2} \right\} + \left\{ \frac{dV}{d\theta} \right\} \cdot \\ &\left\{ \frac{d^2 \theta}{dy^2} \right\} + \left\{ \frac{d^2 V}{d\varpi^2} \right\} \cdot \left\{ \frac{d\varpi^2}{dy^2} \right\} + \left\{ \frac{dV}{d\varpi} \right\} \cdot \left\{ \frac{d^2 \varpi}{dy^2} \right\} = \frac{d^2 V}{dr^2} \sin. \theta \cos. \varpi + \frac{dV}{dr} \\ &\frac{\cos. \theta + \sin. \theta \sin. \varpi}{r} + \left\{ \frac{d^2 V}{d\theta^2} \right\} \cdot \frac{\cos. \theta \cos. \varpi}{r^2} + \left\{ \frac{dV}{d\theta} \right\} \cdot \left\{ \frac{\cos. \theta - \cos. \theta \cos. \varpi}{r^2 \sin. \theta} \right\} \\ &- \frac{2 \sin. \theta \cos. \theta \cos. \varpi}{r^2} + \frac{d^2 V}{d\varpi^2} \cdot \frac{\sin. \varpi}{r^2 \sin. \theta} + \frac{dV}{d\varpi} \cdot \frac{2 \sin. \varpi \cos. \varpi}{r^2 \sin. \theta}; \frac{dr}{dz} = \frac{z}{r} = \\ &\sin. \theta \sin. \varpi; \frac{d^2 r}{dz^2} = \frac{1}{r} - \frac{z^2}{r^3} = \frac{x^2+y^2}{r^3} = \frac{\cos. \theta + \sin. \theta \cos. \varpi}{r}, - \left\{ \frac{d\theta}{dz} \right\} \cdot \sin. \theta \\ &= -\frac{zx}{r^3} = -\frac{\sin. \theta \cos. \theta \sin. \varpi}{r}; \therefore \left\{ \frac{d\theta}{dz} \right\} = \frac{\cos. \theta \sin. \varpi}{r} = \frac{zx}{\sqrt{y^2+z^2} \cdot r^2} \therefore \frac{d^2 \theta}{dz^2} \\ &= \frac{x}{\sqrt{y^2+z^2} \cdot r^2} - \frac{z^2 x}{(y^2+z^2)^{\frac{3}{2}} r^2} - \frac{2z^2 x}{\sqrt{y^2+z^2} \cdot r^4} = \frac{\cos. \theta}{r^2 \sin. \theta} - \frac{\cos. \theta \sin. \varpi}{\sin. \theta \cdot r^2} \\ &- \frac{2 \sin. \theta \cos. \theta \sin. \varpi}{r^2}; \left\{ \frac{d\varpi}{dz} \right\} = \frac{y}{y^2+z^2} = \frac{\cos. \varpi}{r \sin. \theta}; \frac{d^2 \varpi}{dz^2} = -\frac{2zy}{(y^2+z^2)^2} = \\ &- \frac{2 \sin. \varpi \cos. \varpi}{\sin. \theta \cdot r^2}; \left\{ \frac{dV}{dz} \right\} = \left\{ \frac{dV}{dr} \right\} \cdot \left\{ \frac{dr}{dz} \right\} + \left\{ \frac{dV}{d\theta} \right\} \cdot \left\{ \frac{d\theta}{dz} \right\} + \left\{ \frac{dV}{d\varpi} \right\} \cdot \left\{ \frac{d\varpi}{dz} \right\} \end{aligned}$$

if  $\cos. \theta$  be put equal to  $\mu$ , this last equation will become

$$\begin{aligned}
 &= \left\{ \frac{dV}{dr} \right\} \sin. \theta. \sin. \varpi + \left\{ \frac{dV}{d\theta} \right\} \frac{\cos. \theta. \sin. \varpi}{r} + \left\{ \frac{dV}{d\varpi} \right\} \cdot \frac{\cos. \varpi}{r. \sin. \theta} ; \left\{ \frac{d^2 V}{dz^2} \right\} \\
 &= \left\{ \frac{d^2 V}{dr^2} \right\} \cdot \sin. ^2 \theta. \sin. ^2 \varpi + \left\{ \frac{dV}{dr} \right\} \cdot \left\{ \frac{\cos. ^2 \theta + \sin. ^2 \theta. \cos. ^2 \varpi}{r} \right\} + \left\{ \frac{d^2 V}{d\theta^2} \right\}. \\
 &\quad \frac{\cos. ^2 \theta. \sin. ^2 \varpi}{r^2} + \left\{ \frac{dV}{d\theta} \right\} \frac{\cos. \theta - \cos. \theta. \sin. ^2 \varpi}{r^2 \sin. \theta} - \frac{2 \sin. \theta}{r^2} \cos. \theta. \sin. ^2 \varpi . \\
 &\quad + \left\{ \frac{d^2 V}{d\varpi^2} \right\} \cdot \frac{\cos. ^2 \varpi}{r^2. \sin. ^2 \theta} - \left\{ \frac{dV}{d\varpi} \right\} \cdot \frac{2 \sin. \varpi. \cos. \varpi}{r^2. \sin. ^2 \theta} ;
 \end{aligned}$$

if the corresponding terms are made to coalesce in the values of  $\left\{ \frac{d^2 V}{dx^2} \right\} + \left\{ \frac{d^2 V}{dy^2} \right\}$

+  $\left\{ \frac{d^2 V}{dz^2} \right\}$ , we will obtain the following expression

$$\begin{aligned}
 &\left\{ \frac{d^2 V}{dr^2} \right\} \cdot (\cos. ^2 \theta + \sin. ^2 \theta. \cos. ^2 \varpi + \sin. ^2 \theta. \sin. ^2 \varpi) + \frac{dV}{dr} \cdot \left\{ \frac{\sin. ^2 \theta}{r} + \right. \\
 &\quad \frac{\cos. ^2 \theta + \sin. ^2 \theta. \sin. ^2 \varpi}{r} + \frac{\cos. ^2 \theta + \sin. ^2 \theta. \cos. ^2 \varpi}{r} \left. + \left\{ \frac{d^2 V}{d\theta^2} \right\} \right. \\
 &\quad \left. \frac{\sin. ^2 \theta + \cos. ^2 \theta. \cos. ^2 \varpi + \cos. ^2 \theta. \sin. ^2 \varpi}{r^2} \right\} + \left\{ \frac{dV}{d\theta} \right\} \cdot \left\{ \frac{2 \sin. \theta. \cos. \theta}{r^2} + \frac{\cos. \theta}{r^2 \sin. \theta} \right. \\
 &\quad \left. - \frac{\cos. \theta. \cos. ^2 \varpi}{r^2. \sin. \theta} - \frac{2 \sin. \theta. \cos. \theta. \cos. ^2 \varpi}{r^2} + \frac{\cos. \theta}{r^2. \sin. \theta} - \frac{\cos. \theta. \sin. ^2 \varpi}{r^2. \sin. \theta} \right. \\
 &\quad \left. - \frac{2 \sin. \theta. \cos. \theta. \sin. ^2 \varpi}{r^2} \right\} + \left\{ \frac{d^2 V}{d\varpi^2} \right\} \frac{\sin. ^2 \varpi}{r^2. \sin. ^2 \theta} + \frac{\cos. ^2 \varpi}{r^2. \sin. ^2 \theta} \left\{ \frac{dV}{d\varpi} \right\} + \\
 &\quad \left( \frac{2 \sin. \varpi. \cos. \varpi - 2 \sin. \varpi. \cos. \varpi}{r^2. \sin. ^2 \theta} \right) = \left\{ \frac{d^2 V}{dr^2} \right\} + 2 \left\{ \frac{dV}{dr} \right\} \cdot \frac{1}{r} + \left\{ \frac{d^2 V}{d\theta^2} \right\} \cdot \frac{1}{r^2} \\
 &+ \left\{ \frac{dV}{d\theta} \right\} \cdot \frac{\cos. \theta}{r^2. \sin. \theta} + \left\{ \frac{d^2 V}{d\varpi^2} \right\} \cdot \frac{1}{r^2. \sin. ^2 \theta} = 0, \because \text{hence multiplying by } r^2, \text{ we obtain} \\
 &\left\{ \frac{d^2 V}{dr^2} \right\} \cdot r^2 + 2r \cdot \left\{ \frac{dV}{dr} \right\} + \left\{ \frac{d^2 V}{d\theta^2} \right\} + \left\{ \frac{dV}{d\theta} \right\} \cdot \frac{\cos. \theta}{\sin. \theta} + \left\{ \frac{d^2 V}{d\varpi^2} \right\} \cdot \frac{1}{\sin. ^2 \theta} = 0; \text{but as}
 \end{aligned}$$

$$0 = \left\{ d. (1 - \mu^2) \cdot \frac{dV}{d\mu} \right\} + \left\{ \frac{d^2 V}{dr^2} \right\} + r \cdot \left\{ \frac{d^2 \cdot rV}{dr^2} \right\}. \quad (C)^*$$

12. Let us now suppose, that the spheroid is a spherical stratum, the origin of the coordinates being at the centre; it is obvious that  $V$  will only depend on  $r$ , and that it will not contain  $\mu$  or  $w$ ; the equation (C) will therefore be reduced to

$$0 = \left\{ \frac{d^2 \cdot rV}{dr^2} \right\};$$

from which we obtain by integrating,

$$V = A + \frac{B}{r}; \dagger$$

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$$\left\{ \frac{d^2 \cdot rV}{dr^2} \right\} = r \cdot \left\{ \frac{d^2 V}{dr^2} \right\} + 2 \cdot \left\{ \frac{dV}{dr} \right\}, \text{ } dr \text{ being considered as constant, } \therefore r \cdot \left\{ \frac{d^2 \cdot rV}{dr^2} \right\}$$

$$\text{may be substituted in place of } r^2 \cdot \left\{ \frac{d^2 V}{dr^2} \right\} + 2r \cdot \left\{ \frac{dV}{dr} \right\}.$$

$$\begin{aligned} * \text{ If we make } \cos. \theta = \mu, \text{ then } \frac{dV}{d\theta} &= \left( \frac{dV}{d\mu} \right) \cdot \left( \frac{d\mu}{d\theta} \right), \text{ and } \frac{d^2 V}{d\theta^2} = \left( \frac{d^2 V}{d\mu^2} \right) \cdot \frac{d\mu^2}{d\theta^2} \\ &+ \left( \frac{dV}{d\mu} \right) \cdot \left( \frac{d^2 \mu}{d\theta^2} \right), \text{ and as } d\theta \text{ is constant, and } d\mu = -d\theta \cdot \sin. \theta, \text{ } d^2 \mu = -d\theta^2 \cdot \cos. \theta; \\ \frac{d^2 V}{d\theta^2} &= \left( \frac{d^2 V}{d\mu^2} \right) \cdot (1 - \mu^2) - \left( \frac{dV}{d\mu} \right) \cdot \mu; \left( \frac{dV}{d\theta} \right) = \left( \frac{dV}{d\mu} \right) \cdot \frac{d\mu}{d\theta} = -\frac{dV}{d\mu} \cdot \sqrt{1 - \mu^2}, \text{ and } \left( \frac{dV}{d\theta} \right) \\ \frac{\cos. \theta}{\sin. \theta} &= -\frac{dV}{d\mu} \cdot \frac{\sqrt{1 - \mu^2} \cdot \mu}{\sqrt{1 - \mu^2}}. \quad \therefore \left( \frac{d^2 V}{d\theta^2} \right) + \left( \frac{dV}{d\theta} \right) \cdot \frac{\cos. \theta}{\sin. \theta} = \frac{d^2 V}{d\mu^2} \cdot (1 - \mu^2) \end{aligned}$$

$$- 2 \left( \frac{dV}{d\mu} \right) \cdot \mu = d \left( (1 - \mu^2) \frac{dV}{d\mu} \right); \text{ hence it appears how the equation (B) may be reduced to the equation (C).}$$

† If the attracting body be spherical, the quantity  $V$  will be always the same, when  $r$  is

$A$  and  $B$  being two constant arbitrary quantities. Consequently we have

$$-\left\{\frac{dV}{dr}\right\} = \frac{B}{r^2}.$$

from what precedes, it is manifest, that  $-\left\{\frac{dV}{dr}\right\}$  expresses the action of the spherical stratum on the point  $m$ , resolved in the direction of the radius  $r$ , and directed towards the centre of the stratum ; but it is evident, that the entire action of the stratum must be in the direction of the radius ; therefore  $-\left\{\frac{dV}{dr}\right\}$  expresses the total action of the spherical stratum on the point  $m$ .\*

First, let us suppose this point to be placed within the stratum. If it was at the centre itself, the action of the stratum would vanish ; therefore when  $r=0$ , we have  $-\left\{\frac{dV}{dr}\right\}=0$ , i. e.  $\frac{B}{r^2}=0$ , from

the same, and it only varies when  $r$  is increased or diminished. For suppose the attracted point to move on the surface of a sphere, concentrical with the attracting body, it is evident that the value of  $V$  remains the same when the attracting body is spherical, but when this body is any other figure,  $V$  will vary from one position to another of the point moving on the spheric surface.

$$\left(\frac{d^2 r V}{dr^2}\right)=0, \therefore \frac{d r V}{dr}=A, \text{ and } r V=A r + B,$$

it appears from this equation, that if  $r=0$ ,  $B=0$ .

\* From what has been stated in page 42, relative to the action of a spheroid, it appears that  $-\left(\frac{dV}{dr}\right)$  expresses the action of the stratum parallel to  $r$ , but it is evident that the entire action of the stratum is equivalent to this expression, for if equal elements be assumed at each side, equally distant from the direction of  $r$ , their action perpendicular to  $r$  will be destroyed, and the remaining action will be in the direction of  $r$ , and this being the case for every two corresponding elements, it is true for the entire spherical stratum.

which it follows that  $B=0$ , and consequently whatever may be the value of  $r$ ,  $-\left\{ \frac{dV}{dr} \right\} = 0$ ; from this it appears, that a point situated within a spherical stratum does not experience any action, or, which is the same thing, it is equally attracted in every direction.

If the point  $m$  exists without the spherical stratum; it is manifest that if we suppose it at an infinite distance from its centre, the action of the stratum on this point, will be the same, as if the entire mass was collected in this centre; therefore if  $M$  represent the mass of this stratum;  $-\left\{ \frac{dV}{dr} \right\}$  or  $\frac{B}{r^2}$  will become in this case, equal to  $\frac{M}{r^2}$ , from which we obtain  $B=M$ , therefore we have universally,\*

\* When the point is at the centre  $\frac{B}{r^2} = 0$ , when  $r=0$ , as has been already remarked, see preceding page; this is also evident from other considerations, and as  $B$  must be the same, wherever the point is assumed within the surface,  $B$  in all such cases  $= 0$ ;  $\therefore V=A$ , the value of  $A$  may be easily determined.

When the point is infinitely distant, the action is the same as if all the molecules were united in the centre of gravity of the sphere, see page 47, and in this case the action is equal to  $\frac{M}{r^2}$ ,  $\because -\left\{ \frac{dV}{dr} \right\}$  or  $\frac{B}{r^2} = \frac{M}{r^2}$ ,  $\therefore B=M$ ;  $V=A + \frac{M}{r}$ , hence when the attracted point is infinitely distant,  $A=0$ ,  $\therefore$  it is always  $=0$ ; and  $V=\frac{B}{r} = \frac{M}{r}$ .

If the attracted point be without the sphere, the attraction towards the convex part is equal to the attraction to the concave part of the surface: and when the point is on the surface, the attraction to the spherical stratum is only half of what it is, when the point is at a distance from the surface. This is immediately evident from the expression  $\frac{u^2.du.d\pi.d\theta. \sin.\theta}{f} \cdot \frac{1}{(r-u \cos.\theta.\theta.(f))}$ , which, when  $\phi.(f) \propto \frac{1}{f^2}$  becomes  $\frac{u^2.du.d\pi.d\theta. \sin.\theta.}{f^3} \cdot \frac{1}{r-u \cos.\theta.}$ , and it is easy to shew that this expression is the same for two elements situated on the convex and concave sides of the spherical stratum, and which lie on two lines drawn from the attracted point, and making an indefinitely small angle with each other, for  $u \sin.\theta =$  a perpendicular let fall on  $r$  from the attracting element,  $r-u \cos.\theta =$

with respect to exterior points,

$$-\left\{ \frac{dV}{dr} \right\} = \frac{M}{r^2};$$

that is to say, they are attracted by the spherical stratum, in the same manner, as if the entire mass was united in its centre.

A sphere being a spherical stratum, of which the radius of the interior surface vanishes ; it is obvious, that its attraction on a point situated on its surface, or beyond it, is the same as if its mass was united in its centre.\*

This conclusion is equally true, for globes composed of concentrical strata, of which the density varies from the centre to the surface according to any given law ; for this is true for each of its strata ; thus, as the sun, the planets, and the satellites may be considered, very nearly, as globes of this nature ; they attract exterior bodies almost, as if their masses were concentrated in their centres of gravity, which is conformable to the result of observation, as we have seen in No. 5. Indeed, the figure of the heavenly bodies deviates a little from the spherical form ; however, the difference is very small, and the error which results

part of  $r$  intercepted between attracted point and this perpendicular, and it is manifest from similar triangles that the perpendicular let fall on  $r$ , and also the intercepts between these perpendiculars and attracted point are respectively as the distances of the attracting elements from the attracted point, and  $ud\theta$  is also in the same ratio in both cases, see Princip. Math. Book I. Prop. 72, ∵ for the two elements at above mentioned,

$\frac{u.d\theta.u.\sin.\theta.(r-u.\cos.\theta)}{f^3}$  is the same for both, consequently the attractions which vary as

these expressions will be equal, and this being true for every two corresponding elements existing on the same right lines, it is true for the entire stratum. Hence if the attracted point is indefinitely near to the spherical surface, its attraction to the molecule contiguous to it, is equal to its attraction to the rest of the spherical stratum ; if the attracted point approaches still nearer, so as to become identified with this molecule, it will then be a part of the stratum, and its attraction will now be only half what it was previous to its contact with the stratum.

\* For  $u$  being the radius of the homogeneous sphere  $M = \frac{4\pi}{3} \cdot u^3$ , ∵  $-\left\{ \frac{dV}{dr} \right\} =$

from the preceding supposition, is of the same order as this difference, relative to points contiguous to this surface ; and with respect to those points which are at a considerable distance,\* the error is of the same order as the product of this difference, by the square of the ratio of the radii of the attracting bodies to their distances from the points attracted, because we have seen, in No. 10, that the sole consideration of the great distance of the attracted points, renders the error of the preceding supposition, of the same order as the square of this ratio ; the heavenly bodies, therefore attract one another very nearly as if their masses were concentrated in their centres of gravity, not only because they are at considerable distances from each other, relatively to their respective dimensions ; but also because their figures differ little from the spherical form.

The property which spheres possess in the law of nature, of attracting, as if their masses were united in their centres, is very remarkable, and it is interesting to know whether it obtains in other laws of attraction. For this purpose, it may be observed, that if the law of gravity is such, that a homogeneous sphere attracts a point placed without it, as if the entire mass was united in its centre ; the same result will have place for a spherical stratum of a uniform thickness ; for if we take away from a sphere, a spherical stratum of a uniform thickness, we will obtain a new sphere of a smaller radius, which will possess the property equally with the first sphere, of attracting as if the entire mass

$\frac{M}{r^2}$  = when  $r=a$ ,  $\frac{4\pi}{3} \cdot a$ ; for a point which is situated within the sphere, it is evident the action of the strata between the point and exterior surface vanishes, consequently this case is reduced to the former.

\* This ratio may be deduced from what has been established in No. 46, page 10; see also Systeme du Monde, page 255, and Book 3, No. 9. If the force varied as the distance, a homogeneous body of any figure will attract a particle of matter placed anywhere, with the same force and in the same direction, as if all the matter of the body was collected in the centre of gravity. See notes to page 50. This will appear immediately if the force of each element be resolved into other forces parallel to three rectangular co-ordinates.

was united in its centre ; but it is evident, that if this property belongs to these two spheres, it must also belong to the spherical stratum which constitutes their difference. Consequently the problem reduces itself to determine the laws of attraction, according to which a spherical stratum, of an uniform and indefinitely small thickness, attracts an exterior point, as if the entire mass was collected in its centre.

Let  $r$  represent the distance of the attracted point from the centre of the spherical stratum ;  $u$  the radius of this stratum, and  $du$  its thickness. Let  $\theta$  be the angle, which the radius  $u$ , makes with the right line  $r$ ,  $\varpi$  the angle made by the plane which passes through the two lines  $r$  and  $u$ , with a fixed plane, passing through the right line  $r$ ;  $u^2 du.d\varpi.d\theta.\sin.\theta$ ,\* will represent the element of the spherical stratum. If then  $f$  denote the distance of this element, from the point attracted, we will have

$$f^2 = r^2 - 2ru \cdot \cos.\theta + u^2.$$

Let us represent the law of the attraction, at the distance  $f$  by  $\phi(f)$ , the action of the element of the stratum, resolved parallel to  $r$ , and directed towards the centre of the stratum, will be

$$u^2 du.d\varpi.d\theta.\sin.\theta \cdot \frac{(r-u \cdot \cos.\theta)}{f} \cdot \phi(f);$$

but we have

$$\frac{r-u \cdot \cos.\theta}{f} = \left\{ \frac{df}{dr} \right\};$$

in consequence of which, the preceding expression assumes this form

\* The three sides of the element, are  $du$  in the direction of the radius,  $ud\theta$  the element of the curve in the plane passing through the radius  $u$  and  $r$ , and  $u \sin.\theta d\varpi$  the element perpendicular to this plane ; see Book 3, No. 1.

$$u^2.du.d\varpi.d\theta. \sin. \theta. \left\{ \frac{df}{dr} \right\}. \phi(f);^*$$

therefore if we denote  $\int df. \phi(f)$ , by  $\phi(f)$ ; we shall obtain the entire action of the spherical stratum on the point attracted, by means of the integral  $u^2.du. \int d\varpi.d\theta. \sin. \theta. \phi(f)$ , differenced with respect to  $r$ , and divided by  $dr$ .

This integral relatively to  $\varpi$ , should be taken from  $\varpi=0$ , to  $\varpi$  equal to the circumference, and after this integration, it becomes

$$2\pi.u^2.du. \int d\theta. \sin. \theta. \phi(f);$$

$\pi$  expressing the ratio of the semi-circumference to the radius. The value of  $f$  differenced with respect to  $\theta$ , will give

$$d\theta. \sin. \theta = \frac{fdf}{ru};$$

and consequently,

$$2\pi.u^2du. \int d\theta. \sin. \theta. \phi(f) = 2\pi. \frac{udu}{r} \cdot \int fd\theta. \phi(f).$$

\* The attraction in the direction of  $f = u^2du.d\varpi.d\theta. \sin. \theta. \phi(f)$ , and as  $r=u$ .  $\cos. \theta$  = the distance of the attracted point from a perpendicular demitted from the attracting element on the direction of  $r$ , it is evident that  $u^2du.d\varpi.d\theta. \sin. \theta. \phi(f) \frac{(r-u.\cos. \theta)}{f}$  is equal to the action of the attracting element in the direction of  $r$ ,

$$f = \sqrt{r^2 - 2ru.\cos. \theta + u^2}, \therefore df = \frac{dr(r-u.\cos. \theta) + ru.d\theta. \sin. \theta + (u-r.\cos. \theta) du}{f};$$

$$\therefore \left\{ \frac{df}{dr} \right\} = \frac{r-u.\cos. \theta}{f}; \frac{df}{ru} = \frac{d\theta. \sin. \theta}{f}.$$

The integral relative to  $\theta$ , must be taken from  $\theta=0$ , to  $\theta=\pi$ , and at these two limits, we have  $f=r-u$ , and  $f=r+u$ ; consequently the integral relative to  $f$ , must be taken from  $f=r-u$ , to  $f=r+u$ ; therefore let  $\int f df \cdot \varphi(f) = \psi(f)$ ; we shall have\*

$$\frac{2\pi \cdot u du}{r} \cdot \int f df \cdot \varphi(f) = \frac{2\pi \cdot u du}{r} \cdot \left\{ \psi(r+u) - \psi(r-u) \right\}.$$

The coefficient of  $dr$ , in the differential of the second member of this equation, taken with respect to  $r$ , will give the attraction of the spherical stratum, on the point attracted, and it is easy to infer from thence, that in the case of nature in which  $\varphi(f) = \frac{1}{f^2}$ ,† this attrac-

\* The action of the entire stratum, in the direction of  $r=u^2 du \cdot f d\pi \cdot d\theta \cdot \sin \theta \cdot \left\{ \frac{df}{dr} \right\}$ .

$\varphi(f) = u^2 du \cdot f d\pi \cdot d\theta \cdot \sin \theta \cdot \frac{d\varphi(f)}{dr} = u^2 du \cdot f d\pi \cdot d\theta \cdot \sin \theta \cdot \varphi(f)$  differenced with respect to  $r$ , and divided by  $dr$ ,  $dr$  and  $d\theta$  being independent variables. The attracting force for each molecule  $= u^2 \cdot du \cdot d\pi \cdot d\theta \cdot \sin \theta \cdot \left\{ \frac{df}{dr} \right\}$ .  $\varphi(f)$ , ∵ in order to obtain the entire force a triple integration is requisite, with respect to  $f$ , to  $\theta$ , and to  $\pi$ .

In order to integrate with respect to  $d\theta \cdot \sin \theta \cdot \varphi(f)$ , this expression is reduced to a function of  $f$  only, and as  $f$  is here considered as a function of  $\theta$  only,  $r$  comes from under the sign of integration; by substituting  $\frac{f df}{ru}$  for  $d\theta \sin \theta$ , we get  $2\pi u^2 du \cdot f df \cdot \sin \theta$ .

$\varphi(f) = \frac{2\pi u}{r} \cdot du \int f df \cdot \varphi(f)$ , and as  $df$  is only concerned as far as  $f$  is a function of  $\theta$ , and as the limits between which the integral of the first member of this equation ought to be taken, are  $\theta=0$ ,  $\theta=\pi$ , to which limits the corresponding values of  $f$  are  $r-u$ ,  $r+u$ , i. e. the least and greatest values of  $f$ , it is evident that by making  $\int f df \cdot \varphi(f) = \psi(f) = -f$ , the integral of the second member will assume the form in the text.

†  $\varphi(f) = \frac{1}{f^2}$ , ∵  $\int f df \cdot \varphi(f) = \varphi(f) = -\frac{1}{f}$ , and  $\int f df \cdot \varphi(f) = \psi(f) = -f$  at the limits,  $-r-u$ ,  $+r-u$ ; ∵  $\psi(r+u) - \psi(r-u) = -2u$ , consequently, the differential

tion is equal to  $\frac{4\pi.u^2du}{r^2}$ , that is to say, it is the same, as if the entire mass of the spherical stratum was united in its centre; which furnishes a new demonstration of the property, which we have already established, on the attraction of spheres.

Let us now determine  $\phi(f)$ , from the condition that the attraction of the stratum is the same as if its mass was united in its centre. This mass is equal to  $4\pi.u^2du$ , and if it was collected in its centre, its action on the attracted point, will be  $4\pi.u^2du.\phi(r)$ ; therefore we shall have

$$2\pi.udu \left\{ \frac{d \cdot \left\{ \frac{1}{r} \cdot [(\psi(r+u) - \psi(r-u))] \right\}}{dr} \right\} = 4\pi.u^2du \cdot \phi(r); \quad (\text{D})$$

and by integrating with respect to  $r$ , we shall have

$$\psi(r+u) - \psi(r-u) = 2ru \int dr \cdot \phi(r) + rU,$$

$U$  being a function of  $u$ , and of constant quantities, added to the integral\*  $2u \int dr \cdot \phi(r)$ . If we represent  $\psi(r+u) - \psi(r-u)$ , by  $R$ , we shall obtain by differentiating the preceding equation,

$$\left\{ \frac{d^2R}{dr^2} \right\} = 4u \cdot \phi(r) + 2ru \cdot \frac{d \cdot \phi(r)}{dr};$$

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coefficient of the second member of this equation, with respect to  $r = -\frac{2\pi.udu}{r^2} \cdot (-2u) = \frac{4\pi u^2 \cdot du}{r^2}$ ;  $4\pi u^2 du$  = the mass of the spherical stratum, for  $\pi u^2$  = the area of a circle whose radius =  $u$ ,  $\therefore 4\pi u^2$  = the surface of the spherical stratum, and  $4\pi u \cdot u^2 du$  = the mass of the stratum, of which the thickness =  $du$ .

\* Multiplying both sides by  $dr$ , and dividing by  $2\pi.udu$  we obtain by integrating  
 $\frac{\psi(r+u) - \psi(r-u)}{r} = 2ud \int dr \cdot \phi(r) + U.$

$$\left\{ \frac{d^2 R}{du^2} \right\} = r \cdot \left\{ \frac{d^2 U}{du^2} \right\}; *$$

but by the nature of the function  $R$ , we have

$$\left\{ \frac{d^2 R}{dr^2} \right\} = \left\{ \frac{d^2 R}{du^2} \right\}; †$$

consequently,

$$2u \cdot \left\{ 2\phi(r) + \frac{r \cdot d\phi(r)}{dr} \right\} = r \cdot \left\{ \frac{d^2 U}{du^2} \right\};$$

or

$$\frac{2\phi(r)}{r} + \frac{d\phi(r)}{dr} = \frac{1}{2u} \cdot \left\{ \frac{d^2 U}{du^2} \right\}.$$

Thus, the first member of this equation being independent of  $u$ , and the second member being independent of  $r$ , each of these members

$$* \quad \left\{ \frac{dR}{dr} \right\} = 2u \int dr \cdot \phi(r) + 2ru \cdot \phi(r) + U; \quad \frac{d^2 R}{dr^2} = 2u \cdot \phi(r) + 2u \cdot \phi(r) + \frac{2ru d\phi(r)}{dr};$$

$$\frac{dR}{du} = 2r \int dr \cdot \phi(r) + r \cdot \frac{dU}{du}; \quad \frac{d^2 R}{du^2} = r \cdot \left\{ \frac{d^2 U}{du^2} \right\}.$$

† For  $\int f df \cdot \phi(f) = \psi(f)$ ,  $\therefore f df \cdot \phi(f) = d \cdot \psi(f)$ , and  $d f^2 \phi(f) + d f^2 f \cdot \phi(f) = d^2 \cdot \psi(f)$ ,  $\therefore (dr+du)^2 \cdot (\phi(r+u)+r+u) \cdot \phi(r+u) = d^2 \cdot \psi(r+u)$ ,  $(dr-du)^2 \cdot (\phi(r-u)+(r-u)) \cdot \phi(r-u) = d^2 \cdot \psi(r-u)$ ;

$$\therefore \frac{d^2 \cdot \psi(r+u) - d^2 \cdot \psi(r-u)}{dr^2} = \frac{d^2 R}{dr^2} = \frac{d^2 \cdot \psi(r+u) - d^2 \cdot \psi(r-u)}{du^2} = \frac{d^2 R}{du^2};$$

In order to obtain the attraction to a *sphere*, we should integrate the expression  $\frac{2\pi \cdot u du}{r} (\psi(r+u) - \psi(r-u))$  from  $u=0$  to  $u=L$ ,  $L$  being the radius of the sphere, and then the differential of this function taken with respect to  $r$ , and divided by  $dr$ , will give the attraction of the sphere.—See Book 12, No. 2.

must be equal to a constant arbitrary quantity, which we will denote by  $3A$ ; therefore, we have

$$\frac{2\cdot\phi(r)}{r} + \frac{d\cdot\phi(r)}{dr} = 3A;^*$$

from which we obtain by integrating,

$$\phi(r) = Ar + \frac{B}{r^2};$$

$B$  being a new arbitrary quantity. Consequently, all the laws of attraction, in which a sphere acts on an exterior point, placed at the distance  $r$  from its centre, as if the entire mass was collected in this centre, are comprised in the general formula

$$Ar + \frac{B}{r^2}.$$

In fact, it is evident, that this value satisfies the equation ( $D$ ),† whatever may be the values of  $A$  and  $B$ .

If we suppose  $A = 0$ , we shall have the law of nature, and it is evident that in the infinite number of laws which render the attraction very small at great distances, that of nature is the only one, in which

\* Since  $u$  does not occur in the first member, nor  $r$  in the second member of this equation, the equality of these members can only arise from their being respectively equal to a constant quantity, independent of both  $u$  and  $r$ .

Multiplying both sides by  $r^2 dr$ , we shall have

$$2r\cdot dr\cdot\phi(r) + r^2\cdot d\cdot\phi(r) = 3Ar^2\cdot dr. \therefore r^2\cdot\phi r = Ar^3 + B.$$

† In this hypothesis  $\int df\phi(f) = A\cdot f df\cdot f + B\int \frac{df}{f^2} = \frac{Af^2}{2} - \frac{B}{f} = \phi(f)$ , and

spheres are endowed with the power of attracting, as if their masses were united in their centres.

And if a body be situated within a spherical stratum of a uniform thickness throughout, it is in this law only that the body will be equally attracted in every direction. From the foregoing analysis, it appears that the attraction of a spherical stratum, of which the thickness is expressed by  $du$ , on a point placed in its interior, is equal to

$$2\pi.udu.\left\{ \frac{d.\frac{1}{r}\left\{ \psi(u+r)-\psi(u-r) \right\}}{dr} \right\}.$$

In order that this function should vanish, we should have

$$\psi(u+r)-\psi(u-r) = r.U,$$

$U^*$  being a function of  $u$ , independent of  $r$ , and it is easy to perceive

$$\int dff. \varphi(f) = \psi(f) = -\frac{A}{2}. \int dff^3 - B \int df = \frac{A}{8}. f^4 - Bf; \therefore \psi(r+u) = \frac{A}{8}. (r^4 + 4r^3u + 6r^2u^2 + 4ru^3 + u^4) - B(r+u), \text{ and } \psi(r-u) = \frac{A}{8}. (r^4 - 4r^3u + 6r^2u^2 - 4ru^3 + u^4) - B(r-u), \therefore \psi(r+u) - \psi(r-u) = A.(r^3u + ru^3) - 2Bu; \text{ and}$$

$$\begin{aligned} d.\left\{ \frac{1}{r} \cdot (\psi(r+u) - \psi(r-u)) \right\} &= d.\left\{ \frac{1}{r} \cdot A(r^3u + ru^3) - 2Bu \right\} \\ &= \frac{3Ar^2u + Au^3r - Ar^3u - Au^3r + 2Bu}{r^2} = 2Aru + \frac{2Bu}{r^2}; \end{aligned}$$

and if we substitute for  $\varphi(r)$  its value  $Ar + \frac{B}{r^2}$ , in the second member of the equation (D), it comes out equal to  $2Aru + \frac{2Bu}{r^2}$ .

\*  $U$  being the constant arbitrary quantity which is introduced by the integration of

that this is the case in the law of nature, in which  $\phi(f) = \frac{B}{f^2}$ .

But in order to demonstrate that it only obtains in this law, we shall represent by  $\psi'(f)$ , the difference of  $\psi(f)$ , divided by  $df$ ; we shall likewise denote by  $\psi''(f)$ , the difference of  $\psi'(f)$  divided by  $df$ , and so on; we shall thus obtain by two successive differentiations of the preceding equation, with respect to  $r$ ,

$$\psi''(u+r) - \psi''(u-r) = 0.*$$

As this equation obtains, whatever may be the values of  $u$  and  $r$ , it follows that  $\psi''(f)$  must be equal to a constant quantity, whatever may be the value of  $f$ ; and that therefore  $\psi''(f) = 0$ ; but, we have by what precedes,

$$\psi'(f) = f \cdot \phi(f),$$

from which we deduce

$$\psi''(f) = 2 \cdot \phi(f) + f \cdot \phi'(f);$$

*d.*  $\frac{1}{r} (\psi(u+r) - \psi(u-r))$ , differenced with respect to  $r$ , if  $\frac{\psi(u+r) - \psi(u-r)}{r}$  is only equal to  $U$ , its differential with respect to  $r$  must vanish, for then the quantity to which this differential is equal vanishes: *i.e.*  $4\pi u^2 du \phi r = 0$ . If  $\phi(f) = \frac{B}{f^2}$ ,  $f df \cdot \phi(f) = \phi(f) = -\frac{B}{f}$ , and  $f df f \phi(f) = -f B df = -B(f)$ ,  $\therefore \psi(u+r) - \psi(u-r) = B(-r-u) - B(-u+r) = -2Br$ ;  $\therefore d \cdot \left\{ \frac{1}{r} \cdot \psi(u+r) - \psi(u-r) \right\} = -d \cdot \frac{2Br}{r} = 0$ ;  $r$  is less

than  $u$  when the point is assumed within the sphere,  $\therefore$  the limits of  $f$  must be taken  $u+r$ ,  $u-r$ .

$$* \frac{d \cdot \psi(u+r) - d \cdot \psi(u-r)}{dr} = U = \psi'(u+r) - \psi'(u-r); \text{ and } \psi''(u+r) - \psi''(u-r) =$$

and therefore

$$0 = 2 \cdot \phi(f) + f \cdot \phi'(f);$$

which gives by integrating,  $\phi(f) = \frac{B}{f^2}$ ,\* and consequently the law of nature.

13. Let us resume the equation (*C*) of No. 11. If this equation could be generally integrated in every case, we would obtain an expression for  $V$ , involving two arbitrary functions, which could be determined by seeking the attraction of the spheroid on a point situated in a position which facilitates this investigation, and then comparing this attraction with its general expression. But the integration of the equation (*C*) can only be effected in some particular cases, such as when the attracting spheroid becomes a sphere, in which case the equation is reduced to one of ordinary differences; it is also possible, in the case in which the spheroid is a cylinder, of which the base is a curve returning into itself, and of which the length is infinite: we shall see in the third book, that this particular case involves the theory of the rings of Saturn.

Let us fix the origin of the distances  $r$ , on the axis itself of the cylinder, which we shall suppose to be indefinitely extended on each side of the origin. Denoting the distance of the point attracted, from the axis by  $r'$ , we shall have

$$r' = r \cdot \sqrt{1 - \mu^2}.$$

$$\frac{d^2 \psi(u+r) - d^2 \psi(u-r)}{dr^2} = \frac{d_u U}{dr} = 0.$$

$\psi'(u+r)$  is always equal to  $\psi'(u-r)$ , now this could not always be the case unless each of them was constant.

\*  $\psi(f) = f f d f \cdot \phi(f)$ ;  $\therefore \psi'(f) = f \cdot \phi(f)$ , and  $\psi''(f) = \phi(f) + f \cdot \phi'(f)$ , and  $\psi'''(f) = \phi(f) + \phi(f) + f \phi''(f) = 0$ , multiplying by  $f d f$  we obtain  $2f\phi(f)df + f^2\phi'(f).df = 0$ ,  $\therefore f^2 \cdot \phi(f) = B$ , and  $\phi(f) = \frac{B}{f^2}$ .

It is obvious that  $V$  depends solely on  $r'$  and  $\varpi$ , because it is the same for all points, relatively to which, these two variables are the same; consequently it only involves  $\mu$ , inasmuch as  $r'$  is a function of this variable; which gives

$$\left\{ \frac{dV}{d\mu} \right\} = \left\{ \frac{dV}{dr'} \right\} \cdot \left\{ \frac{dr'}{d\mu} \right\} = -\frac{r\mu}{\sqrt{1-\mu^2}} \cdot \left\{ \frac{dV}{dr'} \right\};$$

$$\left\{ \frac{d^2V}{d\mu^2} \right\} = \frac{r^2\mu^2}{1-\mu^2} \cdot \left\{ \frac{d^2V}{dr'^2} \right\} - \frac{r}{(1-\mu^2)^{\frac{3}{2}}} \cdot \left\{ \frac{dV}{dr'} \right\};$$

thus, the equation (C) becomes,

$$0 = r^2 \cdot \left\{ \frac{d^2V}{dr'^2} \right\} + \left\{ \frac{d^2V}{d\varpi^2} \right\} + r' \cdot \left\{ \frac{dV}{dr'} \right\}; *$$

\*  $r'$  = a perpendicular let fall from the attracted point, on the axis of the cylinder,  $\theta$  = the angle which  $r$  makes with the axis,  $\therefore r' = r \cdot \sin \theta = r \cdot \sqrt{1-\mu^2}$ ; if the base of the cylinder was circular,  $V$  would be always the same, when  $r'$  was the same, i. e. it would be a function of  $r'$  only, but as this curve may be an ellipse, or any other curve which returns into itself,  $V$  must depend also on the angle which the plane of  $x, y$  makes with the plane passing through  $r$ , and the axis of  $x$ , i. e., on  $\varpi$ .

$$\begin{aligned} * dr' &= dr \cdot \sqrt{1-\mu^2} - \frac{r\mu d\mu}{\sqrt{1-\mu^2}}, \quad d^2r' = -\frac{dr d\mu \mu}{\sqrt{1-\mu^2}} - \frac{dr d\mu \mu}{\sqrt{1-\mu^2}} - \frac{r d\mu^2}{\sqrt{1-\mu^2}} \\ &- \frac{r\mu^2 d\mu^2}{(1-\mu^2)^{\frac{3}{2}}}, \quad \therefore \frac{d^2r'}{d\mu^2} = \frac{r\mu^2 - r\mu^2 - r}{(1-\mu^2)^{\frac{3}{2}}}; \quad -\frac{d^2V}{d\mu^2} = \left\{ \frac{d^2V}{dr'^2} \right\} \cdot \left\{ \frac{dr'^2}{d\mu^2} \right\} + \left\{ \frac{dV}{dr'} \right\}. \end{aligned}$$

$$\left\{ \frac{d^2r'}{d\mu^2} \right\}; \left\{ \frac{d^2V}{d\mu^2} \right\} \cdot (1-\mu^2) - 2\mu \cdot \left\{ \frac{dV}{d\mu} \right\} + \frac{d^2V}{d\varpi^2} + r \cdot \left\{ \frac{d^2r' V}{dr'^2} \right\} = \text{by (substituting } \frac{1}{1-\mu} \text{)}$$

$$\text{for } \left\{ \frac{d^2V}{d\mu^2} \right\}, \left\{ \frac{dV}{d\mu} \right\}; \frac{d^2V}{dr'^2} \cdot \frac{r^2\mu^2(1-\mu^2)}{1-\mu^2} - \left\{ \frac{dV}{dr'} \right\} \cdot \frac{r \cdot (1-\mu^2)}{(1-\mu^2)^{\frac{3}{2}}} + \left\{ \frac{dV}{dr'} \right\}.$$

from which we obtain by integrating,

$$V = \varphi(r'. \cos. \varpi + r'. \sqrt{-1}. \sin. \varpi) + \psi(r'. \cos. \varpi - r'. \sqrt{-1}. \sin. \varpi); *$$

$$\frac{2\mu^2 r}{\sqrt{1-\mu^2}} + \left\{ \frac{d^2 V}{d\varpi^2} \right\} + r. \left\{ \frac{d^2 r V}{dr^2} \right\}; \text{ (but } d^2 r V = 2dr.dV + rd^2 V; \text{ and } dr = \frac{dr'}{\sqrt{1-\mu^2}};$$

$$\therefore d^2 r V = \frac{2dr'}{\sqrt{1-\mu^2}}. dV + r.d^2 V; \text{ hence } r. \left\{ \frac{d^2 r V}{dr^2} \right\} = \frac{2dV}{dr'} \cdot r. \sqrt{1-\mu^2} + r^2.$$

$(1-\mu^2). \left\{ \frac{d^2 V}{dr'^2} \right\}$ . By substituting these values, the equation (C) becomes

$$\left\{ \frac{d^2 V}{dr'^2} \right\} \cdot r^2 \mu^2 - \left\{ \frac{dV}{dr'} \right\} \cdot \frac{r}{\sqrt{1-\mu^2}} + \left\{ \frac{dV}{dr'} \right\} \cdot \frac{2\mu^2 r}{\sqrt{1-\mu^2}}$$

$$+ \left\{ \frac{d^2 V}{d\varpi^2} \right\} + \left\{ \frac{d^2 V}{dr'^2} \right\} \cdot r^2.(1-\mu^2) + 2 \left\{ \frac{dV}{dr'} \right\} \cdot r. \sqrt{1-\mu^2}.$$

$$= \left\{ \frac{d^2 V}{dr'^2} \right\} \cdot r^2(\mu^2 + 1 - \mu^2) - \left\{ \frac{dV}{dr'} \right\} \cdot \frac{r}{\sqrt{1-\mu^2}} + \left\{ \frac{dV}{dr'} \right\} \frac{2r\mu^2}{\sqrt{1-\mu^2}} + \frac{d^2 V}{d\varpi^2}$$

+  $\frac{dV}{dr'} \cdot \frac{2r(1-\mu^2)}{\sqrt{1-\mu^2}} = 0$ , and if both sides of this equation be multiplied by  $1-\mu^2$ , we will obtain the expression given in the text, by substituting  $r'$  for  $r.\sqrt{1-\mu^2}$ .

\* This integral may be deduced a priori in the following manner: let  $\frac{d^2 V}{d\varpi^2} = r$ ,  $\frac{d^2 V}{dr'^2} = t$ ,  $\frac{dV}{dr'} = q$ , then we will have  $r + r'^2 \cdot t + r'.q = 0$ ; the general expression  $Rk^2 + Sk + T = 0$ , Lacroix, tom. 2. No. 752, 753, &c. becomes  $k^2 + r'^2 = 0$ ,  $\therefore k = \pm r'.\sqrt{-1}$ , and  $du = \frac{du}{dr'} \cdot (dr' + k.d\varpi)$ ,  $dv = \frac{dv}{dr'} \cdot (dr' + k'd\varpi)$  become by making  $\frac{du}{dr'} = \frac{du}{dr'}$ ,  $\frac{dv}{dr'} = \frac{dv}{dr'}$  respectively  $\frac{1}{r'}$ , and substituting  $\pm \sqrt{-1}.r'$ ,  $-\sqrt{-1}.r'$ , for  $k$  and  $k'$ ;  $du = \frac{dr'}{r'} + \sqrt{-1}.d\varpi$ ,  $dv = \frac{dr'}{r'} - \sqrt{-1}.d\varpi$ , consequently  $u = \log. r' + \sqrt{-1}.\varpi$ ,  $v = \log. r' - \sqrt{-1}.$

$\varphi(r')$  and  $\psi(r')$  being arbitrary functions of  $r'$ , which may be deter-

## I. 2

w, are particular integrals of the preceding differential equations; let  $\frac{du}{dr'} = \frac{1}{r'} = n$ ;  $\frac{dv}{dr'} =$

$$= \frac{1}{r'} = n'; \frac{du}{d\omega} = \sqrt{-1} = m; \frac{dv}{d\omega} = -\sqrt{-1} = m'; q = np' + nq',$$

(see Collection of

Examples of differential and integral calculus, page 466,)  $= \frac{p'}{r'} + \frac{q'}{r'}; r = -r' + 2s' - t; t = \frac{r'}{r'^2} + \frac{2s'}{r'^2} - \frac{t'}{r'^2} - \frac{p'}{r'^2} - \frac{q'}{r'^2}, \because r + r'^2 t + r' q = -r' + 2s' - t' + r' + t' + 2s' - p' - q' + q' + q' = 0, \therefore 4s' = 0, i.e. \frac{d^2V}{dv.du} = 0,$  and  $V = \varphi'(u) + \psi(v) = \varphi'(\log. r' + \sqrt{-1}. \omega) + \psi'(\log. r' - \sqrt{-1}. \omega)$  respectively,  $(\varphi' \log. r' + \log. e^{\omega} \sqrt{-1}) + \psi'(\log. r' - \log e^{-\omega} \sqrt{-1}) = \varphi'(\log. (r'. (\cos. \omega + \sqrt{-1}. \sin. \omega)) + \psi'(\log. r'. (\cos. \omega - \sqrt{-1}. \sin. \omega)) = \varphi(r'. \cos. \omega + r'. \sqrt{-1}. \sin. \omega) + \psi(r'. \cos. \omega - r'. \sqrt{-1}. \sin. \omega),$  by substituting  $\cos. \omega \pm \sqrt{-1}. \sin. \omega$  for  $e^{\pm \omega} \sqrt{-1}$ , and assuming the arbitrary function  $\varphi$  = the function  $\varphi'$ . log. This integral evidently satisfies the preceding equation, for

$$\begin{aligned} \left( \frac{dV}{dr} \right) &= \frac{d(\varphi(r'. \cos. \omega + r'. \sqrt{-1}. \sin. \omega))}{d(r'. \cos. \omega + r'. \sqrt{-1}. \sin. \omega)} \cdot \frac{d(r'. \cos. \omega + r'. \sqrt{-1}. \sin. \omega)}{dr'} \\ &\quad + d(\psi(r'. \cos. \omega - r'. \sqrt{-1}. \sin. \omega)) \cdot \frac{d(r'. \cos. \omega - r'. \sqrt{-1}. \sin. \omega)}{dr'} \\ \left( \frac{d^2V}{dr'^2} \right) &= \frac{d^2(\varphi(r'. \cos. \omega + r'. \sqrt{-1}. \sin. \omega))}{d(r'. \cos. \omega + r'. \sqrt{-1}. \sin. \omega)^2} \cdot \frac{d(r'. \cos. \omega + r'. \sqrt{-1}. \sin. \omega)^2}{dr'^2} \\ &\quad + \frac{d(\varphi(r'. \cos. \omega + r'. \sqrt{-1}. \sin. \omega))}{d(r'. \cos. \omega + r'. \sqrt{-1}. \sin. \omega)} \cdot \frac{d^2(r'. \cos. \omega + r'. \sqrt{-1}. \sin. \omega)}{dr'^2} \\ &\quad + \frac{d^2(\psi(r'. \cos. \omega - r'. \sqrt{-1}. \sin. \omega))}{d(r'. \cos. \omega - r'. \sqrt{-1}. \sin. \omega)^2} \cdot \frac{d(r'. \cos. \omega - r'. \sqrt{-1}. \sin. \omega)^2}{dr'^2} \\ &\quad + \frac{d(\psi(r'. \cos. \omega - r'. \sqrt{-1}. \sin. \omega))}{d(r'. \cos. \omega - r'. \sqrt{-1}. \sin. \omega)} \cdot \frac{d^2(r'. \cos. \omega - r'. \sqrt{-1}. \sin. \omega)}{dr'^2} \end{aligned}$$

mined, by investigating the attraction of the cylinder, when  $\varpi$  is equal to cipher, and when it becomes equal to a right angle.

$$\text{but } \frac{d.(r'. \cos. \varpi \pm r'. \sqrt{-1}. \sin. \varpi)}{dr'} = \cos. \varpi \pm \sqrt{-1}. \sin. \varpi, \therefore$$

$$\frac{d^2.(r'. \cos. \varpi \pm r'. \sqrt{-1}. \sin. \varpi)}{dr'^2} = 0;$$

$$r'. \left( \frac{dV}{dr'} \right) = \frac{d.\phi(r'. \cos. \varpi + r'. \sqrt{-1}. \sin. \varpi)}{d.(r'. (\cos. \varpi + r'. \sqrt{-1}. \sin. \varpi))} \cdot (r'. \cos. \varpi + r'. \sqrt{-1}. \sin. \varpi)$$

$$+ \frac{d.\psi(r'. \cos. \varpi - r'. \sqrt{-1}. \sin. \varpi)}{d.(r'. \cos. \varpi - r'. \sqrt{-1}. \sin. \varpi)} \cdot (r'. \cos. \varpi - r'. \sqrt{-1}. \sin. \varpi)$$

$$r'^2. \left( \frac{d^2 V}{dr'^2} \right) = \frac{d^2.\phi(r'. \cos. \varpi + r'. \sqrt{-1}. \sin. \varpi)}{d.(r'. \cos. \varpi + r'. \sqrt{-1}. \sin. \varpi)^2} \cdot r'^2. (\cos. \varpi + 2\sqrt{-1}. \sin. \varpi. \cos. \varpi - \sin. \varpi)$$

$$+ \frac{d^2.\psi(r'. \cos. \varpi - r'. \sqrt{-1}. \sin. \varpi)}{d.(r'. \cos. \varpi - r'. \sqrt{-1}. \sin. \varpi)^2} \cdot r'^2. (\cos. \varpi - 2\sqrt{-1}. \sin. \varpi. \cos. \varpi - \sin. \varpi)$$

$$\left( \frac{dV}{d\varpi} \right) = \frac{d.\phi(r'. \cos. \varpi + r'. \sqrt{-1}. \sin. \varpi)}{d.(r'. \cos. \varpi + r'. \sqrt{-1}. \sin. \varpi)} \cdot \frac{d.(r'. \cos. \varpi + r'. \sqrt{-1}. \sin. \varpi)}{d\varpi}$$

$$+ \frac{d.\psi(r'. \cos. \varpi - r'. \sqrt{-1}. \sin. \varpi)}{d.(r'. \cos. \varpi - r'. \sqrt{-1}. \sin. \varpi)} \cdot \frac{d.(r'. \cos. \varpi - r'. \sqrt{-1}. \sin. \varpi)}{d\varpi},$$

$$\therefore \frac{d^2 V}{d\varpi^2} = \frac{d^2.\phi(r'. \cos. \varpi + r'. \sqrt{-1}. \sin. \varpi)}{d.(r'. \cos. \varpi + r'. \sqrt{-1}. \sin. \varpi)^2} \cdot \frac{d.(r'. \cos. \varpi + r'. \sqrt{-1}. \sin. \varpi)^2}{d\varpi^2},$$

$$+ \frac{d.\phi(r'. \cos. \varpi + r'. \sqrt{-1}. \sin. \varpi)}{d.(r'. \cos. \varpi + r'. \sqrt{-1}. \sin. \varpi)} \cdot \frac{d^2.(r'. \cos. \varpi + r'. \sqrt{-1}. \sin. \varpi)}{d\varpi^2},$$

$$+ \frac{d^2.\psi(r'. \cos. \varpi - r'. \sqrt{-1}. \sin. \varpi)}{d.(r'. \cos. \varpi - r'. \sqrt{-1}. \sin. \varpi)^2} \cdot \frac{d.(r'. \cos. \varpi - r'. \sqrt{-1}. \sin. \varpi)^2}{d\varpi^2}$$

$$+ \frac{d.\psi(r'. \cos. \varpi - r'. \sqrt{-1}. \sin. \varpi)}{d.(r'. \cos. \varpi - r'. \sqrt{-1}. \sin. \varpi)} \cdot \frac{d^2.(r'. \cos. \varpi - r'. \sqrt{-1}. \sin. \varpi)}{d\varpi^2}$$

If the base of the cylinder is a circle,  $V$  will be evidently a function of  $r'$ , independent of  $\varpi$ ; the preceding equation of partial differences will consequently become,

$$0 = r'^2 \cdot \left\{ \frac{d^2 V}{dr'^2} \right\} + r' \cdot \left\{ \frac{dV}{dr'} \right\},$$

which gives, by integrating,

$$-\left\{ \frac{dV}{dr'} \right\} = \frac{H}{r'},$$

$$\begin{aligned} \frac{d(r' \cdot \cos. \varpi \pm r' \cdot \sqrt{-1} \cdot \sin. \varpi)}{d\varpi} &= -r' \cdot \sin. \varpi \pm r' \cdot \sqrt{-1} \cdot \cos. \varpi. \\ \frac{d^2(r' \cdot \cos. \varpi \pm r' \cdot \sqrt{-1} \cdot \sin. \varpi)}{d\varpi^2} &= -r' \cdot \cos. \varpi \mp r' \cdot \sqrt{-1} \cdot \sin. \varpi; \because \left( \frac{d^2 V}{d\varpi^2} \right) \\ &= \frac{d^2 \phi(r' \cdot \cos. \varpi + r' \cdot \sqrt{-1} \cdot \sin. \varpi)}{d(r' \cdot \cos. \varpi + r' \cdot \sqrt{-1} \cdot \sin. \varpi)^2} \cdot r'^2 \cdot (\sin. {}^2 \varpi - 2\sqrt{-1} \cdot \sin. \varpi \cdot \cos. \varpi - \cos. {}^2 \varpi) \\ &\quad + \frac{d\phi(r' \cdot \cos. \varpi + r' \cdot \sqrt{-1} \cdot \sin. \varpi)}{d(r' \cdot \cos. \varpi + r' \cdot \sqrt{-1} \cdot \sin. \varpi)} \cdot (-r' \cdot (\cos. \varpi + \sqrt{-1} \cdot \sin. \varpi)) \\ &\quad + \frac{d\psi(r' \cdot \cos. \varpi - r' \cdot \sqrt{-1} \cdot \sin. \varpi)}{d(r' \cdot \cos. \varpi - r' \cdot \sqrt{-1} \cdot \sin. \varpi)} \cdot r'^2 \cdot (\sin. {}^2 \varpi) + 2\sqrt{-1} \cdot \sin. \varpi \cdot \cos. \varpi - \cos. {}^2 \varpi). \\ &\quad + \frac{d\psi(r' \cdot \cos. \varpi - r' \cdot \sqrt{-1} \cdot \sin. \varpi)}{d(r' \cdot \cos. \varpi - r' \cdot \sqrt{-1} \cdot \sin. \varpi)} \cdot (-r' \cdot (\cos. \varpi - \sqrt{-1} \cdot \sin. \varpi)). \end{aligned}$$

$$\therefore r'^2 \left\{ \frac{d^2 V}{dr'^2} \right\} + r' \cdot \left\{ \frac{dV}{dr'} \right\} + \left\{ \frac{d^2 V}{d\varpi^2} \right\} = 0, \text{ when the values of } \frac{d^2 V}{dr'^2}, \frac{dV}{dr'}, \frac{d^2 V}{d\varpi^2} \text{ are}$$

substituted; consequently this integral satisfies the given differential equation.

When  $\varpi$  vanishes  $V = \phi(r') + \psi(r')$ , and when  $\varpi = 90^\circ$ ,  $V = \phi(r' \cdot \sqrt{-1}) + \psi(-r' \cdot \sqrt{-1})$ , and as the attraction in the direction of  $r' = \left\{ \frac{dV}{dr'} \right\}$ ,  $\phi(r')$ , and  $\psi(r')$  may be determined.

$H$  being a constant quantity. In order to determine it, we will suppose  $r'$  very great with respect to the radius of the base of the cylinder, which consideration permits us to regard the cylinder as an infinite right line. Let  $A$  represent this base, and  $z$  the distance of any point of the axis of the cylinder, from the point where  $r'$  meet this axis, the action of the cylinder supposed to be concentrated in its axis, and resolved parallel to  $r'$ , will be equal to

$$\int \frac{Ar' \cdot dz}{(r'^2 + z^2)^{\frac{3}{2}}},$$

the integral being taken from  $z = -\infty$ , to  $z = \infty$ ; which reduces this integral to  $\frac{2A}{r'}$ ; this is the value of  $-\left\{ \frac{dV}{dr'} \right\}$ , when  $r'$  is very considerable. By comparing it with the preceding expression, we obtain  $H = 2A$ , and it is evident that whatever may be the value of  $r'$ , the action of the cylinder on an exterior point, is  $\frac{2A}{r'}.$ \*

\* If the base of the cylinder be circular,  $V$  will be always the same, when  $r'$  is given,  $\therefore V$  will be a function of  $r'$ , independent of  $w$ ; dividing by  $r'$ , and multiplying both sides by  $dr'$ , we obtain

$$0 = r dr' \cdot \left\{ \frac{d^2 V}{dr'^2} \right\} + \left\{ \frac{dV}{dr'} \right\} \cdot dr' = d.r' \cdot \left\{ \frac{dV}{dr'} \right\}; \therefore r' \cdot \left\{ \frac{dV}{dr'} \right\} = -H.$$

$r = \sqrt{r'^2 + z^2}$ ,  $\therefore$  the attraction in a direction perpendicular to the base, : to the attraction towards the assumed point  $= \frac{1}{r'^2 + z^2}$   $\therefore r' : \sqrt{r'^2 + z^2}$ , hence as  $Adz$  is the differential of the area of the base;  $\frac{Ar' dz}{(r'^2 + z^2)^{\frac{3}{2}}}$  is the differential of the entire force and its integral  $= \frac{Az}{r' \sqrt{r'^2 + z^2}}$ , (see Lacroix, No. 192), when  $z = \infty$  this integral becomes  $\frac{A}{r'}$ , and when  $z = -\infty$ , it becomes  $-\frac{A}{r'}$ ; and as we want the attraction of the point to the cylinder between these two values of  $z$ , the difference of the expressions in these

If the attracted point lies within a circular cylindrical stratum, of an uniform thickness, and of an infinite length; we have also —  $\left\{ \frac{dV}{dr'} \right\}$

two cases,  $= \frac{2A}{r'}$ , must give the attraction required.

When  $r'$  is very considerable with respect to the radius of the cylinder, it is the same thing as if the mass of the cylinder was concentrated in its axis. When the point is situated within the cylinder,  $V$  is of a different form from what it is, when the point is situated without the cylinder; and as it is of the *same* form wherever the point is assumed within the cylinder, whatever it is in one case, it will be the same in all. The length of the cylinder must be infinite, otherwise the point, even when situated in the axis, would not be equally attracted in the direction of the axis.

When the base is circular, —  $\left\{ \frac{dV}{dr'} \right\} = \frac{H}{r'} \therefore - \left\{ \frac{dV}{dr'} \right\} \cdot dr' = H \cdot \frac{dr'}{r'}, \therefore - V = H \log r' + C$ . The cylinder being of an infinite length, the attraction perpendicular to the axis is the only attraction which it is necessary to estimate.

Therefore the force varying inversely as the square of the distance, there are two cases in which a point is equally attracted in every direction; the first is when the point is situated in the interior of a spherical stratum, (it will be proved in the third book, that this conclusion may be extended to the case of elliptic strata, the interior and exterior surfaces being similar, and similarly situated;) the second is that in which the point is situated in the interior of a hollow cylinder, whose base is circular and length infinite.

If the cylinder was concentrated into a right line of a finite length, the attraction in a direction perpendicular to this line  $= \frac{r' \cdot dz}{(r'^2 + z^2)^{\frac{3}{2}}}$ . of which the integral is  $\frac{z}{\sqrt{(r'^2 + z^2)} r'}$ .

And if  $a$  is  $\equiv$  the length of this line, the entire attraction in a direction perpendicular to it  $= \frac{a}{\sqrt{a^2 + r^2} \cdot r'}$ ; hence if  $a$  be infinite, the attraction is as  $\frac{1}{r'}$ ; the attraction in the direction of  $a$ , is as  $\frac{z}{(r'^2 + z^2)^{\frac{3}{2}}}$ ;  $\therefore$  the differential of the force  $= \frac{z dz}{(r'^2 + z^2)^{\frac{3}{2}}}$ , the integral of which is  $\frac{-1}{\sqrt{r'^2 + z^2}} + C$ , when  $z = 0$ ,  $C = \frac{1}{r'}$ ,  $\therefore$  the entire attraction  $= \frac{1}{r'} - \frac{1}{\sqrt{r'^2 + z^2}} = \frac{\sqrt{r'^2 + z^2} - r'}{r' \cdot \sqrt{r'^2 + z^2}}$  when  $z = a$ ;  $\frac{\sqrt{r'^2 + a^2} - r'}{r' \cdot \sqrt{r'^2 + a^2}}$ ;  $\therefore$  the attraction in the direction of  $a$  is to the attraction in the direction of  $r'$  ::  $\sqrt{r'^2 + a^2} - r' : a$ ; hence it is easy to determine the direction in which the point would commence to move; it may be easily

$= \frac{H}{r'}$ ; and as the attraction vanishes, when the attracted point is on the axis itself of the stratum, we have  $H = 0$ , and consequently

shewn that a point placed in the vertex of a triangle is attracted towards the segments made by the perpendicular with a force reciprocally proportional to the secants of the angles which the base makes with the sides. For if  $r'$  be the altitude, and  $a, a'$ , the segments of the base, it is evident from the expression  $\frac{a}{r' \cdot \sqrt{a^2 + r'^2}}$  that the attractions to the segments

$a, a'$  are as  $\frac{a}{\sqrt{a^2 + r'^2}}$  to  $\frac{a'}{\sqrt{a'^2 + r'^2}}$ , but these expressions will be evidently proportional to the reciprocals of the secants of the angles at the base of the triangle.

If the attracted point exist in a perpendicular to the plane of a circle which passes through the centre,  $x$  being the distance of the attracted point from the circumference of a circle, concentrical with the given circle, the distance of the centre from this point being  $= r'$ , then  $\pi(x^2 - r'^2)$  = the area of this circle, and  $2\pi x dx$  is the differential of the area, and as the attraction in the direction of  $r'$  is as  $\frac{r'}{x^3}$ ; the differential of the attraction of the point towards the circle

$= \frac{2\pi r' \cdot dx}{x^2}$ , of which the integral is  $- \frac{2\pi r'}{x} + C$ , and when  $x = r'$  the attraction va-

nishes,  $\therefore C = 2\pi$ , and the corrected integral  $= 2\pi(1 - \frac{r}{x})$ , hence the attraction of a point situated in the vertex of a cone to all circular sections of the cone is the same, and for similar cones the attraction varies as the side of the cone. If the attracted point exist in the produced axis of a finite cylinder with a circular base, of which the radius  $= a$ ,  $r'$  being as before the distance of the attracted point from any point in the axis,  $\sqrt{a^2 + r'^2}$

will be the distance of the circumference of the cylinder from this point, the attraction towards this circumference is as  $1 - \frac{r'}{\sqrt{a^2 + r'^2}}$ , and the differential of this attraction is as

$dr' - \frac{r' dr'}{\sqrt{a^2 + r'^2}}$  of which the integral  $= r' - \sqrt{a^2 + r'^2}$ ,  $r$ , and  $r_{\infty}$  being the greatest and least values of  $r'$ , the attraction to the entire cylinder  $= -r + r_{\infty} - \sqrt{a^2 + r_{\infty}^2} + \sqrt{a^2 + r^2}$ ;  $r - r_{\infty}$  = the length of the cylinder. If the length be infinite  $r_{\infty} = \sqrt{a^2 + r'^2}$ ,  $\therefore$  the attraction is as  $r - \sqrt{a^2 + r^2}$ , and if  $a$  be infinite the attraction is as  $r - r_{\infty}$ , the length of the cylinder.

a point situated in the interior of the stratum is equally attracted in every direction.

14. We may apply to the motion of a body, the equations  $A$ ,  $B$ , and  $C$ , of No. 11, and then elicit from them, an equation of condition, which will be found very useful, in verifying as well the computations of the theory, as also the theory itself of universal gravitation. The differential equations (1), (2), (3) of No. 9, which determine the relative motion of  $m$  about  $M$ , may be made to assume the following form :

$$\frac{d^2x}{dt^2} = \left\{ \frac{dQ}{dx} \right\}; \quad \frac{d^2y}{dt^2} = \left\{ \frac{dQ}{dy} \right\}; \quad \frac{d^2z}{dt^2} = \left\{ \frac{dQ}{dz} \right\}; \quad (i)$$

$Q$  being equal to  $\frac{M+m}{r} - \Sigma \frac{m' \cdot (xx' + yy' + zz')}{r'^3} + \frac{\lambda}{m}$ ; and it is easy to perceive that we have

$$0 = \left\{ \frac{d^2Q}{dx^2} \right\} + \left\{ \frac{d^2Q}{dy^2} \right\} + \left\{ \frac{d^2Q}{dz^2} \right\}; \quad (E)^*$$

provided that the variables  $x'$ ,  $y'$ ,  $z'$ ,  $x''$ , &c., which  $Q$  contains, are independent of  $x$ ,  $y$  and  $z$ .

$$\begin{aligned} * \left\{ \frac{dQ}{dx} \right\} &= \frac{-(M+m)x}{(x^2+y^2+z^2)^{\frac{3}{2}}} - \Sigma \frac{m'x'}{r'^3} + \frac{1}{m} \cdot \left\{ \frac{d\lambda}{dx} \right\}; \quad \left\{ \frac{dQ}{dy} \right\} = - \frac{(M+m)y}{(x^2+y^2+z^2)^{\frac{3}{2}}} \\ &- \Sigma \frac{m'y'}{r'^3} + \frac{1}{m} \cdot \left\{ \frac{d\lambda}{dy} \right\}; \quad \left\{ \frac{dQ}{dz} \right\} = - \frac{(M+m)z}{(x^2+y^2+z^2)^{\frac{3}{2}}} - \Sigma \frac{mz'}{r'^3} + \frac{1}{m} \cdot \left\{ \frac{d\lambda}{dz} \right\}; \text{ but} \\ &- \frac{mx}{(x^2+y^2+z^2)^{\frac{3}{2}}} - \Sigma \frac{m'x'}{r'^3} = - \Sigma \frac{mx}{r^3}, \quad \therefore \frac{d^2x}{dt^2} = - \frac{Mx}{r^3} - \Sigma \frac{mx}{r^3} + \frac{1}{m} \cdot \\ \left\{ \frac{d\lambda}{dx} \right\} &= \left\{ \frac{dQ}{dx} \right\}; \text{ see page 35.} \\ \frac{d^2Q}{dx^2} &= - \frac{(M+m)}{(x^2+y^2+z^2)^{\frac{3}{2}}} + \frac{3(M+m)x^2}{(x^2+y^2+z^2)^{\frac{5}{2}}} + \frac{1}{m} \cdot \left\{ \frac{d^2\lambda}{dx^2} \right\} = \\ &\left( \frac{-m'}{(x'-x)^2 + (y'-y)^2 + (z'-z)^2} + \frac{3m' \cdot (x'-x)^2}{((x'-x)^2 + (y'-y)^2 + (z'-z)^2)^{\frac{5}{2}}} \right) + \&c. \end{aligned}$$

The variables  $x, y, z$ , may be transformed into others, which are more convenient for astronomical purposes.  $r$  being the radius drawn from the centre of  $M$  to that of  $m$ , let  $v$  represent the angle which the projection of this radius on the plane of  $x$ , and of  $y$ , makes with the axis of  $x$ ; and  $\theta$ , the inclination of  $r$  on the same plane; we shall have,

$$x = r \cdot \cos. \theta. \cos. v;$$

$$y = r \cdot \cos. \theta. \sin. v;$$

$$z = r \cdot \sin. \theta.$$

By referring the equation ( $E$ ) to these new variables, we shall have by No. 11,

$$0 = r^2 \cdot \left\{ \frac{d^2 Q}{dr^2} \right\} + 2r \cdot \left\{ \frac{dQ}{dr} \right\} + \left\{ \frac{d^2 Q}{dv^2} \right\} + \left\{ \frac{d^2 Q}{d\theta^2} \right\} - \frac{\sin. \theta.}{\cos. \theta.} \cdot \left\{ \frac{dQ}{d\theta} \right\}; \quad (F)$$

Multiplying the first of the equations ( $i$ ) by  $\cos. \theta. \cos. v$ ; the second, by  $\cos. \theta. \sin. v$ ; the third, by  $\sin. \theta$ ; and then, in order to abridge, making

$$M' = \frac{d^2 r}{dt^2} - \frac{r \cdot dv^2}{dt^2} \cdot \cos. \theta. \cos. v - \frac{r \cdot d\theta^2}{dt^2};$$

$$\frac{d^2 Q}{dy^2} = - \frac{(M+m)}{(x^2+y^2+z^2)^{\frac{3}{2}}} + \frac{3(M+m) \cdot y^2}{(x^2+y^2+z^2)^{\frac{5}{2}}} + \frac{1}{m} \cdot \left\{ \frac{d^2 \lambda}{dy^2} \right\} = \\ \left( \frac{-m}{(x'-x)^2+(y'-y)^2+(z'-z)^2)^{\frac{3}{2}} + \frac{3m'(y-y)^2}{((x'-x)^2+(y'-y)^2+(z'-z)^2)^{\frac{5}{2}}} \right) + \text{&c.}$$

$$\frac{d^2 Q}{dz^2} = - \frac{(M+m)}{(x^2+y^2+z^2)^{\frac{3}{2}}} + \frac{3(M+m) \cdot z^2}{(x^2+y^2+z^2)^{\frac{5}{2}}} + \frac{1}{m} \cdot \left\{ \frac{d^2 \lambda}{dz^2} \right\} =$$

$$\left( \frac{-m'}{(x'-x)^2+(y'-y)^2+(z'-z)^2)^{\frac{3}{2}} + \frac{3m'(z'-z)^2}{((x'-x)^2+(y'-y)^2+(z'-z)^2)^{\frac{5}{2}}} \right) + \text{&c.}$$

$$\therefore \frac{d^2 Q}{dx^2} + \frac{d^2 Q}{dy^2} + \frac{d^2 Q}{dz^2} = \frac{-3(M+m) \cdot r^2 + 3(M+m) \cdot r^2}{r^5}$$

$$\frac{-3m' \cdot (x'-x)^2 + (y'-y)^2 + (z'-z)^2}{((x'-x)^2 + (y'-y)^2 + (z'-z)^2)^{\frac{5}{2}}}$$

$$+ 3m' \cdot \frac{(x'-x)^2 + (y'-y)^2 + (z'-z)^2}{((x'-x)^2 + (y'-y)^2 + (z'-z)^2)^{\frac{5}{2}}} = 0. \quad \text{In the expression for } \frac{1}{m} \left\{ \frac{d^2 \lambda}{dx^2} + \frac{d^2 \lambda}{dy^2} + \frac{d^2 \lambda}{dz^2} \right\}$$

we shall obtain, by adding them together,

$$M' = \left\{ \frac{dQ}{dr} \right\}.$$

In like manner, if we multiply the first of the equations (*i*), by  $-r \cdot \cos. \theta \cdot \sin. v$ ; the second, by  $r \cdot \cos. \theta \cdot \cos. v$ , we shall obtain by their addition

$$N' = \left\{ \frac{dQ}{dv} \right\};$$

$$N' \text{ being supposed equal to } \frac{d \left\{ r^2 \cdot \frac{dv}{dt} \cdot \cos. {}^2 \theta \right\}}{dt}.$$

Finally, if we multiply the first of the equations (*i*), by  $-r \cdot \sin. \theta \cdot \cos. v$ ; the second by  $-r \cdot \sin. \theta \cdot \sin. v$ ; and if then we add them to the third, multiplied by  $\cos. \theta$ , we shall obtain, by making  $P'$  equal to

$$r^2 \cdot \frac{d^2 \theta}{dt^2} + r^2 \cdot \frac{dv^2}{dt^4} \cdot \sin. \theta \cdot \cos. \theta + \frac{2r \cdot dr \cdot d\theta}{dt^2};$$

$$P' = \left\{ \frac{dQ}{d\theta} \right\}.*$$

M 2

are only considered the first terms in each, but as the other terms are precisely of the same form, it is evident, that the sum of the three differential coefficients, for each of the other terms respectively constitute a result equal to cipher.

$$\begin{aligned} * dx &= dr \cdot \cos. \theta \cdot \cos. v - d\theta \cdot r \cdot \sin. \theta \cdot \cos. v - dv \cdot r \cdot \cos. \theta \cdot \sin. v; \therefore d^2x = d^2r \cdot \cos. \theta \cdot \cos. v - dr \cdot d\theta \cdot \sin. \theta \cdot \cos. v - dr \cdot dv \cdot \cos. \theta \cdot \sin. v - d^2\theta \cdot r \cdot \sin. \theta \cdot \cos. v - d\theta \cdot dr \cdot \sin. \theta \cdot \cos. v - \\ &d\theta^2 \cdot r \cdot \cos. \theta \cdot \cos. v + d\theta \cdot dv \cdot r \cdot \sin. \theta \cdot \sin. v - d^2v \cdot r \cdot \cos. \theta \cdot \sin. v - dv \cdot dr \cdot \cos. \theta \cdot \sin. v + \\ &dv \cdot d\theta \cdot r \cdot \sin. \theta \cdot \sin. v - dv^2 \cdot r \cdot \cos. \theta \cdot \cos. v, \therefore d^2x \cdot \cos. \theta \cdot \cos. v = d^2r \cdot \cos. {}^2 \theta \cdot \cos. {}^2 v - 2dr \cdot \\ &d\theta \cdot \sin. \theta \cdot \cos. \theta \cdot \cos. {}^2 v - 2dr \cdot dv \cdot \cos. {}^2 \theta \cdot \sin. v - d^2\theta \cdot r \cdot \sin. \theta \cdot \cos. \theta \cdot \cos. {}^2 v - d\theta^2 \cdot r \cdot \\ &\cos. {}^2 \theta \cdot \cos. {}^2 v + 2dv \cdot d\theta \cdot r \cdot \sin. \theta \cdot \cos. \theta \cdot \sin. v - d^2v \cdot r \cdot \cos. {}^2 \theta \cdot \sin. v - dv^2 \cdot r \cdot \\ &\cos. {}^2 \theta \cdot \cos. {}^2 v; dy = dr \cdot \cos. \theta \cdot \sin. v - r d\theta \cdot \sin. \theta \cdot \sin. v + rdv \cdot \cos. \theta \cdot \cos. v; \therefore d^2y = d^2r \cdot \cos. \theta \cdot \sin. v - \\ &dr \cdot d\theta \cdot \sin. \theta \cdot \sin. v + dr \cdot dv \cdot \cos. \theta \cdot \cos. v - dr \cdot d\theta \cdot \sin. \theta \cdot \sin. v - r d\theta^2 \cdot \cos. \theta \cdot \sin. v - \\ &rdv \cdot d\theta \cdot \sin. \theta \cdot \cos. v - rd^2v \cdot \cos. \theta \cdot \sin. v + rd^2v \cdot \cos. \theta \cdot \cos. v; \therefore d^2y \cdot \cos. \theta \cdot \sin. v = d^2r \cdot \cos. {}^2 \theta \cdot \sin. {}^2 v. \end{aligned}$$

The values of  $r$ ,  $v$ , and  $\theta$ , involve six arbitrary quantities, which are introduced by the integration of the preceding differential equa-

$$\begin{aligned}
 & -2dr.d\theta \sin. \theta. \cos. \theta. \sin. {}^2 v + 2dr.dv. \cos. {}^2 \theta. \sin. v. \cos. v - rd\theta^2. \cos. {}^2 \theta. \sin. {}^2 v - rd^2 \theta. \\
 & \sin. \theta. \cos. \theta. \sin. {}^2 v - rdv^2. \cos. {}^2 \theta. \sin. {}^2 v + rd^2 v. \cos. {}^2 \theta. \sin. v. \cos. v - 2rd\theta.dv. \sin. \theta. \cos. \theta. \\
 & \sin. v. \cos. v; dz = dr. \sin. \theta + r d\theta. \cos. \theta; \therefore d^2 z = d^2 r. \sin. \theta + 2dr.d\theta. \cos. \theta + rd^2 \theta. \cos. \theta \\
 & - rd\theta^2. \sin. \theta; \therefore d^2 z. \sin. \theta = d^2 r. \sin. {}^2 \theta + 2dr.d\theta. \sin. \theta. \cos. \theta + rd^2 \theta. \sin. \theta. \cos. \theta - rd\theta^2. \\
 & \sin. {}^2 \theta, \text{ consequently, } \frac{d^2 x}{dt^2} \cos. \theta. \cos. v + \frac{d^2 y}{dt^2} \cos. \theta. \sin. v + \frac{d^2 z}{dt^2} \sin. \theta = \frac{d^2 r}{dt^2} - \frac{rd\theta^2}{dt^2} \\
 & - \frac{rdv^2}{dt^2}. \cos. {}^2 \theta, \text{ but } \frac{dx}{dr} = \cos. \theta. \cos. v; \frac{dy}{dr} = \cos. \theta. \sin. v; \frac{dz}{dr} = \sin. \theta. \therefore \frac{d^2 x}{dt^2} = \\
 & \cos. \theta. \cos. v + \frac{d^2 y}{dt^2}. \cos. \theta. \sin. v + \frac{d^2 z}{dt^2}. \sin. \theta = \\
 & \left\{ \frac{dQ}{dx} \right\} \cdot \left\{ \frac{dx}{dr} \right\} + \left\{ \frac{dQ}{dy} \right\} \cdot \left\{ \frac{dy}{dr} \right\} + \left\{ \frac{dQ}{dz} \right\} \cdot \left\{ \frac{dz}{dr} \right\} = \left\{ \frac{dQ}{dr} \right\} = M'.
 \end{aligned}$$

In like manner, if  $d^2 x$  and its value be respectively multiplied by the differential of  $x$ , on the hypothesis that  $v$  is the only variable quantity, we shall obtain;  $-r.d^2 x. \cos. \theta. \sin. v = -rd^2 r. \cos. {}^2 \theta. \sin. v + 2dr.d\theta.r. \sin. \theta. \cos. \theta. \sin. v. \cos. v + 2dr. dv.r. \cos. {}^2 \theta. \sin. {}^2 v + d^2 \theta. r^2. \sin. v. \cos. v. \sin. \theta. \cos. \theta + d\theta^2. r^2. \cos. {}^2 \theta. \sin. v. \cos. v + d^2 v. r^2. \cos. {}^2 \theta. \sin. {}^2 v + dv^2. r^2. \cos. {}^2 \theta. \sin. v. \cos. v - 2dv.d\theta. r^2. \sin. \theta. \cos. \theta. \sin. {}^2 v$ ; and multiplying  $d^2 y$  and its value by the differential of  $y$ , taken on the same hypothesis, we obtain  $r.d^2 y. \cos. \theta. \cos. v = r.d^2 r. \cos. {}^2 \theta. \sin. v. \cos. v - 2dr. d\theta. r. \sin. \theta. \cos. \theta. \sin. v. \cos. v + 2dr. dv. r. \cos. {}^2 \theta. \cos. {}^2 v - r^2 d^2 \theta. \sin. \theta. \cos. \theta. \sin. v. \cos. v - r^2. dv^2. \cos. {}^2 \theta. \sin. v + r^2 d^2 v. \cos. {}^2 \theta. \cos. {}^2 v - 2r^2 d\theta. dv. \sin. \theta. \cos. \theta. \cos. {}^2 v; \therefore -r.d^2 y. \cos. \theta. \sin. v + rd^2 y. \cos. \theta. \cos. v = 2rdr.dv. \cos. {}^2 \theta + r^2 d^2 v. \cos. {}^2 \theta - 2r^2 dv. d\theta. \sin. \theta. \cos. \theta.$

$$= d(r^2.dv. \cos. {}^2 \theta); \frac{dx}{dv} = -r. \cos. \theta. \sin. v; \frac{dy}{dv} = r. \cos. \theta. \cos. v; \therefore - \left\{ \frac{d^2 x}{dt^2} \right\}.$$

$$r. \cos. \theta. \sin. v + \left\{ \frac{d^2 y}{dt^2} \right\} \cdot r. \cos. \theta. \cos. v = \left\{ \frac{dQ}{dx} \right\} \cdot \left\{ \frac{dx}{dv} \right\} + \left\{ \frac{dQ}{dy} \right\} \cdot \left\{ \frac{dy}{dv} \right\}$$

$$= \left\{ \frac{dQ}{dv} \right\} = N'. \text{ Multiplying } d^2 x \text{ and its value, by the differential of } x, \text{ taken on the supposition that } \theta \text{ is the variable quantity; } -rd^2 x. \sin. \theta. \cos. v = -rd^2 r. \sin. \theta. \cos. \theta. \cos. {}^2 v + 2rdr.d\theta. \sin. {}^2 \theta. \cos. {}^2 v + 2rdr.dv. \sin. \theta. \cos. \theta. \sin. v. \cos. v + r^2 d\theta^2. \sin. \theta. \cos. \theta. \cos. {}^2 v - 2r^2 d\theta. dv. \sin. {}^2 \theta. \sin. v. \cos. v + r^2 d^2 v. \sin. \theta. \cos. \theta. \sin. v. \cos. v; \text{ performing a similar operation on } d^2 y \text{ and its value, we obtain } -d^2 y. r. \sin. \theta. \sin. v = -rd^2 r. \sin. \theta. \cos. \theta. \sin. {}^2 v + 2rdr.d\theta. \sin. {}^2 \theta. \sin. {}^2 v - 2rdr.dv. \sin. \theta. \cos. \theta. \sin. v. \cos. v + r^2 d\theta^2. \sin. \theta. \cos. \theta. \sin. {}^2 v + 2r^2 d\theta. dv. \sin. {}^2 \theta.$$

tions.\* Let us consider any three of these which we will denote by  $a, b, c$ ; the equations  $M' = \left\{ \frac{dQ}{dr} \right\}$  will furnish us with the three following equations:

$$\left\{ \frac{d^2Q}{dr^2} \right\} \cdot \left\{ \frac{dr}{da} \right\} + \left\{ \frac{d^2Q}{dr \cdot dv} \right\} \cdot \left\{ \frac{dv}{da} \right\} + \left\{ \frac{d^2Q}{dr \cdot d\theta} \right\} \cdot \left\{ \frac{d\theta}{da} \right\} = \left\{ \frac{dM'}{da} \right\};$$

$$\left\{ \frac{d^2Q}{dr^2} \right\} \cdot \left\{ \frac{dr}{db} \right\} + \left\{ \frac{d^2Q}{dr \cdot dv} \right\} \cdot \left\{ \frac{dv}{db} \right\} + \left\{ \frac{d^2Q}{dr \cdot d\theta} \right\} \cdot \left\{ \frac{d\theta}{db} \right\} = \left\{ \frac{dM'}{db} \right\};$$

$$\left\{ \frac{d^2Q}{dr^2} \right\} \cdot \left\{ \frac{dr}{dc} \right\} + \left\{ \frac{d^2Q}{dr \cdot dv} \right\} \cdot \left\{ \frac{dv}{dc} \right\} + \left\{ \frac{d^2Q}{dr \cdot d\theta} \right\} \cdot \left\{ \frac{d\theta}{dc} \right\} = \left\{ \frac{dM'}{dc} \right\};$$

We can obtain by means of those equations, the value of  $\left\{ \frac{d^2Q}{dr^2} \right\}$ ,

and if we make

$$m = \left\{ \frac{dv}{db} \right\} \cdot \left\{ \frac{d\theta}{dc} \right\} - \left\{ \frac{dv}{dc} \right\} \cdot \left\{ \frac{d\theta}{db} \right\};$$

$$n = \left\{ \frac{dv}{dc} \right\} \cdot \left\{ \frac{d\theta}{da} \right\} - \left\{ \frac{dv}{da} \right\} \cdot \left\{ \frac{d\theta}{dc} \right\};$$

$$\begin{aligned} & \sin. v. \cos. v + r^2 dv^2. \sin. \theta. \cos. \theta. \sin. ^2 v + r^2 d^2 \theta. \sin. ^2 \theta. \sin. ^2 v - r^2 d^2 v. \sin. \theta. \cos. \theta. \\ & \sin. v. \cos. v; \text{ and in like manner } d^2 z. r \cos. \theta = rd^2 r. \sin. \theta. \cos. \theta - 2rdr. d\theta. \cos. ^2 \theta - r^2 d\theta^2. \\ & \sin. \theta. \cos. \theta + r^2 d^2 \theta. \cos. ^2 \theta, \therefore - \frac{d^2 x. r}{dt^2} \sin. \theta. \cos. v - \frac{d^2 y. r}{dt^2} \sin. \theta. \sin. v + \frac{d^2 z. r}{dt^2}. \\ & \cos. \theta = \frac{2rdr. d\theta}{dt^2} + \frac{r^2. dv^2}{dt^2} \sin. \theta. \cos. \theta + r^2. \frac{d^2 \theta}{dt^2}, \text{ but } \frac{dx}{d\theta} = r. \sin. \theta. \cos. v; \frac{dy}{d\theta} = - r. \\ & \sin. \theta. \sin. v. \end{aligned}$$

$$\frac{dz}{d\theta} = \cos. \theta; \text{ and } - \left\{ \frac{d^2 x}{dt^2} \right\} \cdot r. \sin. \theta. \cos. v - \left\{ \frac{d^2 y}{dt^2} \right\} \cdot r. \sin. \theta. \sin. v + \left\{ \frac{d^2 z}{dt^2} \right\} r. \cos. \theta.$$

$$= \left\{ \frac{dQ}{dx} \right\} \cdot \left\{ \frac{dx}{d\theta} \right\} + \left\{ \frac{dQ}{dy} \right\} \cdot \left\{ \frac{dy}{d\theta} \right\} + \left\{ \frac{dQ}{dz} \right\} \cdot \left\{ \frac{dz}{d\theta} \right\} = \left\{ \frac{dQ}{d\theta} \right\}.$$

$$= \frac{2r. dr. d\theta}{dt^2} + r^2. \frac{d^2 \theta}{dt^2} \sin. \theta. \cos. \theta + r^2. \left\{ \frac{dv^2}{dt^2} \right\} = P'.$$

\* The values of  $r, v$  and  $\theta$  are determined by the integration of equations of the second order,  $\because$  two arbitrary quantities are involved in the determination of each variable.

## CELESTIAL MECHANICS,

$$\begin{aligned}
 p &= \left\{ \frac{dv}{da} \right\} \cdot \left\{ \frac{d\theta}{db} \right\} - \left\{ \frac{dv}{db} \right\} \cdot \left\{ \frac{d\theta}{da} \right\}; \\
 \epsilon &= \left\{ \frac{dr}{dc} \right\} \cdot \left\{ \frac{dv}{db} \right\} \cdot \left\{ \frac{d\theta}{dc} \right\} - \left\{ \frac{dr}{da} \right\} \cdot \left\{ \frac{dv}{dc} \right\} \cdot \left\{ \frac{d\theta}{db} \right\}; \\
 &\quad + \left\{ \frac{dr}{db} \right\} \cdot \left\{ \frac{dv}{dc} \right\} \cdot \left\{ \frac{d\theta}{da} \right\} - \left\{ \frac{dr}{db} \right\} \cdot \left\{ \frac{dv}{da} \right\} \cdot \left\{ \frac{d\theta}{dc} \right\}; \\
 &\quad + \left\{ \frac{dr}{dc} \right\} \cdot \left\{ \frac{dv}{da} \right\} \cdot \left\{ \frac{d\theta}{db} \right\} - \left\{ \frac{dr}{dc} \right\} \cdot \left\{ \frac{dv}{db} \right\} \cdot \left\{ \frac{d\theta}{da} \right\};
 \end{aligned}$$

we shall have

$$\epsilon \cdot \left\{ \frac{d^2 Q}{dr^2} \right\} = m \cdot \left\{ \frac{dM'}{da} \right\} + n \cdot \left\{ \frac{dM'}{db} \right\} + p \cdot \left\{ \frac{dM'}{dc} \right\}.*$$

\* From the value of  $\left\{ \frac{dQ}{dr} \right\} = M'$ ; it is evident that  $M'$  is a function of  $r, v$  and  $\theta$ ; and as these coordinates are functions of  $a, b, c$ , and conversely, it follows that

$$\begin{aligned}
 \left\{ \frac{dM'}{da} \right\} &= \left\{ \frac{dM'}{dr} \right\} \cdot \left\{ \frac{dr}{da} \right\} + \left\{ \frac{dM'}{dv} \right\} \cdot \left\{ \frac{dv}{da} \right\} + \left\{ \frac{dM'}{d\theta} \right\} \cdot \left\{ \frac{d\theta}{da} \right\} = \\
 &\quad \left( \text{by substituting for } M' \text{ its value } \left\{ \frac{dQ}{dr} \right\} \right) \\
 &\quad \left\{ \frac{d^2 Q}{dr^2} \right\} \cdot \left\{ \frac{dr}{da} \right\} + \left\{ \frac{d^2 Q}{dr.dv} \right\} \cdot \left\{ \frac{dv}{da} \right\} + \left\{ \frac{d^2 Q}{dr.d\theta} \right\} \cdot \left\{ \frac{d\theta}{da} \right\},
 \end{aligned}$$

by similar operations we obtain the values of  $\left\{ \frac{dM'}{db} \right\} \cdot \left\{ \frac{dM'}{dc} \right\}$ , &c. Multiplying  $\left\{ \frac{dM'}{da} \right\}$  and its value, by  $m$  and its value,  $\left\{ \frac{dM'}{db} \right\}$  and its value, by  $n$  and its value,  $\left\{ \frac{dM'}{dc} \right\}$  and its value, by  $p$  and its value, we obtain

$$\begin{aligned}
 &\left\{ \frac{d^2 Q}{dr^2} \right\} \cdot \left\{ \frac{dr}{da} \right\} \left( \left\{ \frac{dv}{db} \right\} \cdot \left\{ \frac{d\theta}{dc} \right\} - \left\{ \frac{dv}{dc} \right\} \cdot \left\{ \frac{d\theta}{db} \right\} \right) + \left\{ \frac{d^2 Q}{dr.dv} \right\} \cdot \\
 &\left\{ \frac{dv}{da} \right\} \cdot \left( \left\{ \frac{dv}{db} \right\} \cdot \left\{ \frac{d\theta}{dc} \right\} - \left\{ \frac{dv}{dc} \right\} \cdot \left\{ \frac{d\theta}{db} \right\} \right) + \left\{ \frac{d^2 Q}{dr.d\theta} \right\} \cdot \left\{ \frac{d\theta}{da} \right\} \cdot \left( \left\{ \frac{dv}{db} \right\} \cdot
 \end{aligned}$$

In like manner if we make

$$m' = \left\{ \frac{dr}{dc} \right\} \cdot \left\{ \frac{d\theta}{db} \right\} - \left\{ \frac{dr}{db} \right\} \cdot \left\{ \frac{d\theta}{dc} \right\};$$

$$n' = \left\{ \frac{dr}{da} \right\} \cdot \left\{ \frac{d\theta}{dc} \right\} - \left\{ \frac{dr}{dc} \right\} \cdot \left\{ \frac{d\theta}{da} \right\};$$

$$p' = \left\{ \frac{dr}{ab} \right\} \cdot \left\{ \frac{d\theta}{da} \right\} - \left\{ \frac{dr}{da} \right\} \cdot \left\{ \frac{d\theta}{db} \right\};$$

$$\left( \left\{ \frac{d\theta}{dc} \right\} - \left\{ \frac{dv}{dc} \right\} \cdot \left\{ \frac{d\theta}{db} \right\} \right) = m \cdot \left\{ \frac{dM'}{da} \right\} \cdot$$

$$\left\{ \frac{d^2 Q}{dr^2} \right\} \cdot \left\{ \frac{dr}{db} \right\} \cdot \left( \left\{ \frac{dv}{dc} \right\} \cdot \left\{ \frac{d\theta}{da} \right\} - \left\{ \frac{dv}{da} \right\} \cdot \left\{ \frac{d\theta}{dc} \right\} \right) + \left\{ \frac{d^2 Q}{dr \cdot dv} \right\} \cdot$$

$$\left\{ \frac{dv}{db} \right\} \cdot \left( \left\{ \frac{dv}{dc} \right\} \cdot \left\{ \frac{d\theta}{da} \right\} - \left\{ \frac{dv}{da} \right\} \cdot \left\{ \frac{d\theta}{dc} \right\} \right) + \left\{ \frac{d^2 Q}{ar \cdot d\theta} \right\} \cdot \left\{ \frac{d\theta}{db} \right\} \cdot \left( \left\{ \frac{dv}{dc} \right\} \cdot \right.$$

$$\left. \left\{ \frac{d\theta}{da} \right\} - \left\{ \frac{dv}{da} \right\} \cdot \left\{ \frac{d\theta}{dc} \right\} \right) = n \cdot \left\{ \frac{dM'}{db} \right\}$$

$$\left\{ \frac{d^2 Q}{dr^2} \right\} \cdot \left\{ \frac{dr}{dc} \right\} \left( \left\{ \frac{dv}{da} \right\} \cdot \left\{ \frac{d\theta}{db} \right\} - \left\{ \frac{dv}{db} \right\} \cdot \left\{ \frac{d\theta}{da} \right\} \right) + \left\{ \frac{d^2 Q}{dr \cdot dv} \right\} \cdot$$

$$\left\{ \frac{dv}{dc} \right\} \left( \left\{ \frac{dv}{da} \right\} \cdot \left\{ \frac{d\theta}{db} \right\} - \left\{ \frac{dv}{db} \right\} \cdot \left\{ \frac{d\theta}{da} \right\} \right) + \left\{ \frac{d^2 Q}{dr \cdot dv} \right\} \cdot \left\{ \frac{d\theta}{dc} \right\} \cdot \left( \left\{ \frac{dv}{da} \right\} \cdot \right.$$

$$\left. \left\{ \frac{d\theta}{db} \right\} - \left\{ \frac{dv}{db} \right\} \cdot \left\{ \frac{d\theta}{da} \right\} \right) = p \cdot \left\{ \frac{dM'}{dc} \right\}.$$

Adding these three expressions together, and observing that the coefficients of  $\left( \frac{d^2 Q}{dr \cdot dv} \right)$ ,

$\left( \frac{d^2 Q}{dr \cdot d\theta} \right)$  are respectively equal to cipher, and that the coefficient of  $\left( \frac{d^2 Q'}{dr^2} \right) = \epsilon$ , we will obtain the expression given in the text. We can by a similar process obtain the values of  $\left( \frac{d^2 Q}{dv^2} \right)$ ,  $\left( \frac{d^2 Q}{d\theta^2} \right)$ , now if we substitute these values in the equation (F), and also  $M'$  and  $P'$ , for  $\left\{ \frac{dQ}{dr} \right\}$ ,  $\left\{ \frac{dQ}{d\theta} \right\}$ , and multiply by  $\epsilon$  and  $\cos^2 \theta$ , we will arrive at the equation (G).

the equation  $N' = \left( \frac{dQ}{dv} \right)$  will give

$$\epsilon \cdot \left\{ \frac{d^2 Q}{d^2 v} \right\} = m' \cdot \left\{ \frac{dN'}{da} \right\} + n' \cdot \left\{ \frac{dN'}{db} \right\} + p' \cdot \left\{ \frac{dN'}{dc} \right\}.$$

Finally, if we make

$$m'' = \left\{ \frac{dr}{db} \right\} \cdot \left\{ \frac{dv}{dc} \right\} - \left\{ \frac{dr}{dc} \right\} \cdot \left\{ \frac{dv}{db} \right\};$$

$$n'' = \left\{ \frac{dr}{dc} \right\} \cdot \left\{ \frac{dv}{da} \right\} - \left\{ \frac{dr}{da} \right\} \cdot \left\{ \frac{dv}{dc} \right\};$$

$$p'' = \left\{ \frac{dr}{da} \right\} \cdot \left\{ \frac{dv}{db} \right\} - \left\{ \frac{dr}{db} \right\} \cdot \left\{ \frac{dv}{da} \right\};$$

The equation  $P = \left\{ \frac{dQ}{d\theta} \right\}$  will give

$$\epsilon \cdot \left\{ \frac{d^2 Q}{d\theta^2} \right\} = m'' \cdot \left\{ \frac{dP'}{da} \right\} + n'' \cdot \left\{ \frac{dP'}{db} \right\} + p'' \cdot \left\{ \frac{dP'}{dc} \right\}.$$

Consequently, the equation (F) will become,

$$\begin{aligned} 0 = & m \cdot r^2 \cos^2 \theta \cdot \left\{ \frac{dM'}{da} \right\} + n \cdot r^2 \cos^2 \theta \cdot \left\{ \frac{dM'}{db} \right\} + p \cdot r^2 \cdot \cos^2 \theta \cdot \left\{ \frac{dM'}{dc} \right\} \\ & + m' \cdot \left\{ \frac{dN'}{da} \right\} + n' \cdot \left\{ \frac{dN'}{db} \right\} + p' \cdot \left\{ \frac{dN'}{dc} \right\}. \quad (G) \\ & + m'' \cdot \cos^2 \theta \cdot \left\{ \frac{dP'}{da} \right\} + n'' \cdot \cos^2 \theta \cdot \left\{ \frac{dP'}{db} \right\} + p'' \cdot \cos^2 \theta \cdot \left\{ \frac{dP'}{dc} \right\}. \\ & + \epsilon (2rM' \cdot \cos^2 \theta - P' \cdot \sin \theta \cdot \cos \theta). \end{aligned}$$

In the theory of the moon, we neglect the perturbations, that its action produces in the relative motion of the sun about the earth, which implies that its mass is indefinitely small. Then the variables  $x', y', z'$ , which are relative to the sun, are independent of  $x, y, z$ , and the

equation (G) obtains in this theory ; it is therefore necessary that the values found for  $r$ ,  $v$  and  $\theta$ , should satisfy this equation, which furnishes us with a means of verifying these values. If the inequalities which are observed in the motion of the moon, are the result of a mutual attraction between these three bodies, namely, the sun, the earth, and the moon, the observed values of  $r$ ,  $v$  and  $\theta$ , deduced from observation, should satisfy the equation (G), which furnishes us with a means of verifying the theory of universal gravitation ; for the mean longitudes of the moon, of its perigee, and of its ascending node, occur in these values, and  $a$ ,  $b$ ,  $c$ , may be assumed equal to these longitudes.

In like manner, if in the theory of the planets, we neglect the square of the disturbing forces, which we are almost always permitted to do ; then, in the theory of the planet, of which the coordinates are  $x$ ,  $y$ ,  $z$ , we can suppose that the coordinates  $x'$ ,  $y'$ ,  $z'$ ,  $x''$ , &c. of the other planets, are relative to their elliptic motion, and consequently, independent of  $x$ ,  $y$ ,  $z$  ; therefore the equation (G) obtains in this theory.\*

15. The differential equations of the preceding No.

$$\left. \begin{aligned} \frac{d^2r}{dt^2} - \frac{rdv^2}{dt^2} \cdot \cos.^2\theta - r \cdot \frac{d\theta^2}{dt^2} &= \left\{ \frac{dQ}{dr} \right\}; \\ \frac{d(r^2 \cdot \frac{dv}{dt} \cdot \cos.^2\theta)}{dt} &= \left\{ \frac{dQ}{dv} \right\}; \\ r^2 \cdot \frac{d^2\theta}{dt^2} + r^2 \cdot \frac{dv^2}{dt^2} \cdot \sin.\theta \cdot \cos.\theta + \frac{2rdr.d\theta}{dt^2} &= \left\{ \frac{dQ}{d\theta} \right\} \end{aligned} \right\}; \quad (\text{H})$$

\* We arrived at the equation (G) on the supposition that  $x$ ,  $y$ ,  $z$  were independent of  $x'$ ,  $y'$ ,  $z'$ , &c. In the case of elliptic motion  $x$ ,  $y$ ,  $z$ , are independent of  $x'$ ,  $y'$ ,  $z'$ , and conversely, and as when the square of the perturbing force is neglected, the motion is q.p. elliptic, it follows that  $x$ ,  $y$ ,  $z$ , are in this case independent of  $x'$ ,  $y'$ ,  $z'$ . See page 49, of the text.

are only a combination of the differential equations (*i*) of the same No. ; but they are more convenient, and better adapted to astronomical computations. We can assign other forms to them, which may be useful in different circumstances.

Instead of the variables  $r$  and  $\theta$ , let us consider  $u$  and  $s$ ,  $u$  being equal to  $\frac{1}{r \cdot \cos \theta}$ , that is to unity divided by the projection of the radius vector, on the plane of  $x$  and of  $y$ ; and  $s$  being equal to the tangent of  $\theta$ , or to the tangent of latitude of  $m$  above the same plane, by multiplying the second of the equations (H) by  $r^2 dv \cdot \cos^2 \theta$ , and then integrating, we shall obtain

$$\left\{ \frac{dv}{u^2 \cdot dt} \right\}^2 = h^2 + 2 \int \left\{ \frac{dQ}{dv} \right\} \cdot \frac{dv}{u^2}; ^*$$

$h$  being a constant arbitrary quantity; consequently we have

$$dt = \frac{dv}{u^2 \cdot \sqrt{h^2 + 2 \int \left\{ \frac{dQ}{dv} \right\} \cdot \frac{dv}{u^2}}}.$$

If the first of the equations (H) multiplied by  $-\cos \theta$ , be added to the third multiplied by  $\frac{\sin \theta}{r}$ , we shall obtain

$$-\frac{d^2 \cdot \frac{1}{u}}{dt^2} + \frac{1}{u} \cdot \frac{dv^2}{dt^2} = u^2 \cdot \left\{ \frac{dQ}{du} \right\} + us \cdot \left\{ \frac{dQ}{ds} \right\};$$

from which we deduce

There are two distinct objects, one to verify the values of  $r$ ,  $v$ ,  $\theta$ , and the other to verify the theory of universal gravitation.

$$\begin{aligned} & r^2 \cdot dv \cdot \cos^2 \theta \cdot d(r^2 \cdot \frac{dv}{dt} \cdot \cos^2 \theta) = \frac{dv}{u^2} \cdot d \left( \frac{dv}{u^2 \cdot dt} \right) = \frac{dv}{u^2} \cdot \left( \frac{dQ}{dv} \right) \because \left( \frac{dv}{u^2 \cdot dt} \right)^2 \\ & = h^2 + 2 \int \frac{dv}{u^2} \cdot \left( \frac{dQ}{dv} \right). \end{aligned}$$

$$d \cdot \left\{ \frac{du}{u^2 \cdot dt} \right\} + \frac{dv^2}{u \cdot dt} = u^2 \cdot dt \cdot \left\{ \left\{ \frac{dQ}{du} \right\} + \frac{s}{u} \cdot \left\{ \frac{dQ}{ds} \right\} \right\}.$$

If we consider  $dv$  as constant, we shall obtain by substituting for  $dt$  its value, which has been already given

$$0 = \frac{d^2 u}{dv^2} + u + \frac{\left\{ \frac{dQ}{dv} \right\} \cdot \frac{du}{u^2 dv} - \frac{dQ}{du} - \frac{s}{u} \cdot \left\{ \frac{dQ}{ds} \right\}}{h^2 + 2 \cdot \int \left\{ \frac{dQ}{dv} \right\} \cdot \frac{dv}{u^2}} *$$

N 2

$$* = \frac{d^2 r}{dt^2} \cdot \cos. \theta + r \cdot \frac{dv^2}{dt^2} \cdot \cos. {}^3 \theta + r \cdot \frac{d\theta^2}{dt^2} \cdot \cos. \theta + r \cdot \frac{d^2 \theta}{dt^2} \cdot \sin. \theta + r \cdot \frac{dv^2}{dt^2} \cdot \sin. {}^2 \theta.$$

$$\cos. \theta + \frac{2dr \cdot d\theta}{dt^2} \cdot \sin. \theta = - \left( \frac{dQ}{dr} \right) \cdot \cos. \theta + \left( \frac{dQ}{d\theta} \right) \cdot \frac{\sin. \theta}{r}; \text{ but } \frac{1}{u} = r \cdot \cos. \theta; \text{ and } d \cdot \frac{1}{u} = dr \cdot \cos. \theta - rd\theta \cdot \sin. \theta; \therefore d^2 \cdot \frac{1}{u} = d^2 r \cdot \cos. \theta - 2dr \cdot d\theta \cdot \sin. \theta - d^2 \theta \cdot r \cdot \sin. \theta - rd\theta^2 \cdot \cos. \theta;$$

∴ by concinnating and substituting  $-d^2 \cdot \frac{1}{u}$ , for its value, and noting that  $r \cdot \left( \frac{dv^2}{dt^2} \right)$ .

$$\cos. {}^3 \theta = r \cdot \left( \frac{dv^2}{dt^2} \right) \cdot \cos. \theta - r \cdot \left( \frac{dv^2}{dt^2} \right) \cdot \cos. \theta \cdot \sin. {}^2 \theta, \text{ we obtain } -d^2 \cdot \frac{1}{u} + r \cdot \frac{dv^2}{dt^2}.$$

$$\cos. \theta = - \left( \frac{dQ}{dr} \right) \cdot \cos. \theta + \left( \frac{dQ}{d\theta} \right) \cdot \frac{\sin. \theta}{r}; - \frac{du}{u^2} = dr \cdot \cos. \theta - rd\theta \cdot \sin. \theta; \left( \frac{dQ}{dr} \right) = \left( \frac{dQ}{du} \right).$$

$$\left( \frac{du}{dr} \right), \text{ but } \frac{du}{dr} = -u^2 \cdot \cos. \theta. \therefore - \left( \frac{dQ}{dr} \right) \cdot \cos. \theta = \left( \frac{dQ}{du} \right) \cdot u^2 \cdot \cos. {}^2 \theta; \left( \frac{dQ}{d\theta} \right) = \left( \frac{dQ}{du} \right).$$

$$\left( \frac{du}{d\theta} \right) + \left( \frac{dQ}{ds} \right) \cdot \left( \frac{ds}{d\theta} \right), \text{ and as } s = \tan. \theta; \frac{ds}{d\theta} = (1+s^2); \left( \frac{du}{d\theta} \right) = r \cdot \sin. \theta \cdot u^2 \cdot \left( \frac{dQ}{d\theta} \right) \cdot \frac{\sin. \theta}{r}$$

$$= \left( \frac{dQ}{du} \right) \cdot \left( \frac{du}{d\theta} \right) \cdot \frac{\sin. \theta}{r} + \left( \frac{dQ}{ds} \right) \cdot \left( \frac{ds}{d\theta} \right) \cdot \frac{\sin. \theta}{r} = \left( \frac{dQ}{du} \right) \cdot \sin. {}^2 \theta \cdot u^2 + \left( \frac{dQ}{ds} \right) \cdot (1+s^2).$$

$$\frac{\sin. \theta}{r}; \sin. \theta = \frac{s}{\sqrt{1+s^2}}; \text{ and } \frac{1}{r} = \frac{u}{\sqrt{1+s^2}}. \therefore \left( \frac{dQ}{ds} \right) \cdot (1+s^2) \frac{\sin. \theta}{r} = \left( \frac{dQ}{ds} \right) \cdot us, \text{ and making}$$

the two coefficients of  $\left( \frac{dQ}{du} \right)$  to coalesce, we obtain  $d^2 \frac{1}{u} + \frac{dv^2}{u \cdot dt^2} = u^2 \cdot (\sin. {}^2 \theta + \cos. {}^2 \theta)$ .

In the same manner, by treating  $dv$  as if it was constant, the third of the equations (H), will become

$$0 = \frac{d^2s}{dv^2} + s + \frac{ds}{dv} \cdot \left\{ \frac{dQ}{dv^2} \right\} - (1+s^2) \left\{ \frac{dQ}{ds} \right\} - us \cdot \left\{ \frac{dQ}{du} \right\}^*$$

$$u^2 \cdot (h^2 + 2f \left\{ \frac{dQ}{dv} \right\} \cdot \frac{dv}{u^2})$$

$$\left( \frac{dQ}{du} \right) + us \cdot \left( \frac{dQ}{ds} \right). \text{ Substituting for } dt \text{ we obtain}$$

$$d \cdot \left( \frac{du}{dv} \cdot \sqrt{h^2 + 2f \left( \frac{dQ}{dv} \right) \cdot \frac{dv}{u^2}} \right) + dv \cdot u \cdot \sqrt{h^2 + 2f \left( \frac{dQ}{dv} \right) \cdot \frac{dv}{u^2}}$$

$$= \frac{dv}{\sqrt{h^2 + 2f \left( \frac{dQ}{dv} \right) \cdot \frac{dv}{u^2}}} \left( \frac{dQ}{du} + \frac{s}{u} \cdot \left( \frac{dQ}{ds} \right) \right) = \frac{d^2u}{dv} \cdot \sqrt{h^2 + 2f \left( \frac{dQ}{dv} \right) \cdot \frac{dv}{u^2}}$$

$$+ du \cdot dv \cdot \left( \frac{dQ}{dv} \right) \cdot \frac{1}{u^2} + dv \cdot u \cdot \sqrt{h^2 + 2f \left( \frac{dQ}{dv} \right) \cdot \frac{dv}{u^2}} =$$

$$\frac{dv}{dv \cdot \sqrt{h^2 + 2f \left( \frac{dQ}{dv} \right) \cdot \frac{dv}{u^2}}} \cdot \left\{ \frac{dQ}{du} + \frac{s}{u} \left\{ \frac{dQ}{ds} \right\} \right\}, \because \text{ dividing by } dv, \text{ and the radical}$$

$$\sqrt{h^2 + 2f \left\{ \frac{dQ}{dv} \right\} \cdot \frac{dv}{u^2}}$$

quantity we obtain the expression which is given in the text.

$$* d\theta = \frac{ds}{1+s^2}, \because d^2\theta = \frac{d^2s}{1+s^2} - \frac{2sds^2}{(1+s^2)^2}; r^2 = \frac{1+s^2}{u^2}, \therefore r^2 \cdot \frac{d^2\theta}{dt^2} = \frac{d^2s}{u^2 dt^2} - \frac{2s}{(1+s^2)}$$

$$\frac{ds^2}{u^2 dt^2}, (\sin. \theta. \cos. \theta = \frac{s}{1+s^2}, \therefore r^2 \cdot \frac{dv^2}{dt^2} \cdot \sin. \theta. \cos. \theta = \frac{s}{u^2} \cdot \frac{dv^2}{dt^2}; 2rdr = \frac{2uds}{u^2}$$

$$- \frac{2du(1+s^2)}{u^3}, \therefore 2rdr \cdot d\theta = \frac{2s \cdot ds^2}{(1+s^2)u^2} - \frac{2udu \cdot ds}{u^3}; \text{ but } r^2 \cdot \frac{d^2\theta}{dt^2} = \frac{r^2}{dt} \cdot d \cdot \left\{ \frac{d\theta}{dt} \right\} = \frac{r^2}{dt}.$$

$$\left\{ \frac{d^2\theta}{dt^2} - \frac{d\theta \cdot d^2t}{dt^2} \right\}; \therefore \text{ by substituting for } d^2\theta, d\theta \text{ and } r^2 \text{ their values already given, and}$$

for  $-\frac{d^2t}{dt^2}$  its value

$$\frac{d(u^2 \cdot \sqrt{h^2 + 2f \left\{ \frac{dQ}{dv} \right\} \cdot \frac{dv}{u^2}})}{dv} = \frac{2udu}{dv} \cdot \sqrt{h^2 + 2f \left\{ \frac{dQ}{dv} \right\} \cdot \frac{dv}{u^2}}$$

Therefore in place of the three differential equations (H), we shall have the following:

$$\begin{aligned}
 dt &= \frac{dv}{u \cdot \sqrt{h^2 + 2f \left\{ \frac{dQ}{dv} \right\} \cdot \frac{dv}{u^2}}}, \\
 0 &= \frac{\frac{d^2u}{dv^2} + u + \left\{ \frac{dQ}{dv} \right\} \cdot \frac{du}{u^2 dv} - \left\{ \frac{dQ}{du} \right\} - \frac{s}{u} \cdot \left\{ \frac{dQ}{ds} \right\}}{\frac{h^2 + 2f \left\{ \frac{dQ}{dv} \right\} \cdot \frac{dv}{u^2}}{u^2}}, \quad (\text{K}) \\
 0 &= \frac{\frac{d^2s}{dv^2} + s + \frac{ds}{dv} \cdot \left\{ \frac{dQ}{dv} \right\} - us \cdot \left\{ \frac{dQ}{dv} \right\} - (1+s^2) \cdot \left\{ \frac{dQ}{ds} \right\}}{u^2 \cdot \left\{ h^2 + 2f \left\{ \frac{dQ}{dv} \right\} \cdot \frac{dv}{u^2} \right\}}.
 \end{aligned}$$

By making these equations to assume the following form, we avoid fractions and radicals,

$$\begin{aligned}
 &\frac{+u^2 \cdot \frac{dQ}{dv} \cdot \frac{dv}{u^2}}{dv \cdot \sqrt{h^2 + 2f \left\{ \frac{dQ}{dv} \right\} \cdot \frac{dv}{u^2}}}, \text{ the third equation (H) becomes =} \\
 &\left\{ \frac{d^2s}{u^2 \cdot dt^2} - \frac{2s \cdot ds^2}{(1+s^2) \cdot u^2 \cdot dt^2} + \left\{ \frac{ds}{u^2} \cdot \frac{2u \cdot du}{dv \cdot dt} \right\} \right. \\
 &\left. \left\{ \sqrt{h^2 + 2f \left\{ \frac{dQ}{dv} \right\} \cdot \frac{dv}{u^2}} \right\} + \frac{ds}{u^2} \cdot u^2 \cdot \frac{dQ}{dv} \cdot \frac{dv}{u^2} + \frac{s}{u^2} \cdot \frac{dv^2}{dt^2} \right. \\
 &\left. + \frac{2sds^2}{(1+s^2) \cdot u^2 \cdot dt^2} - \frac{2du \cdot ds}{u^3 \cdot dt^2} = \left\{ \frac{dQ}{du} \right\} \cdot \left\{ \frac{du}{d\theta} \right\} + \left\{ \frac{dQ}{ds} \right\} \cdot \left\{ \frac{ds}{d\theta} \right\} \right\} = \text{by substituting for } dt \text{ its value, and calling } \sqrt{h^2 + 2f \left\{ \frac{dQ}{dv} \right\} \cdot \left\{ \frac{dv}{u^2} \right\}} \cdot p \cdot \frac{d^2s}{dv^2} \cdot u^2 \cdot p^2 - \frac{2s}{(1+s^2)} \cdot \frac{ds^2 \cdot u^2}{dv^2} \cdot p^2 + \frac{ds}{dv^2} \cdot 2udu \cdot p^2 + \frac{ds}{dv^2} \cdot \frac{dQ}{dv} \cdot dv + s \cdot u^2 \cdot p^2 + \frac{2s \cdot ds^2}{1+s^2} \cdot \frac{u^2 \cdot p^2}{dv^2} \right. \\
 &\left. - \frac{-2du \cdot ds \cdot p^2}{dv^2} = + \left\{ \frac{dQ}{du} \right\} \cdot su + \left\{ \frac{dQ}{ds} \right\} \cdot (1+s^2); \text{ equal evidently to the third equation (K).} \right.
 \end{aligned}$$

## CELESTIAL MECHANICS,

$$\begin{aligned}
 0 &= \frac{d^2t}{dv^2} + \frac{2du.dt}{u.dv^2} + u^2 \cdot \left\{ \frac{dQ}{dv} \right\} \cdot \frac{dt^3}{dv^3}; * \\
 0 &= \left\{ \frac{d^2u}{dv^2} + u \cdot \right\} \cdot \left\{ 1 + \frac{2}{h} \cdot J \left\{ \frac{dQ}{dv} \right\} \cdot \frac{dv}{u^2} \right\} \\
 &\quad + \frac{1}{h^2} \left\{ \left\{ \frac{dQ}{dv} \right\} \cdot \frac{du}{u^2.dv} - \left\{ \frac{dQ}{du} \right\} - \frac{s}{u} \cdot \left\{ \frac{dQ}{ds} \right\} \right\}; \quad (L) \\
 0 &= \left\{ \frac{d^2s}{dv^2} + s \right\} \cdot \left\{ 1 + \frac{2}{h^2} \cdot J \left\{ \frac{dQ}{dv} \right\} \cdot \frac{dv}{u^2} \right\} \\
 &\quad + \frac{1}{h^2u^2} \cdot \left\{ \frac{ds}{dv} \cdot \left\{ \frac{dQ}{dv} \right\} - us \cdot \left\{ \frac{dQ}{du} \right\} - (1+s^2) \cdot \left\{ \frac{dQ}{ds} \right\} \right\}.
 \end{aligned}$$

By making use of other coordinates, we might form new systems of differential equations ; suppose, for example, that the coordinates  $x$  and  $y$ , of the equations (*i*) of No. 14, are transformed into others, relative to two moveable axes situated in the plane of these coordinates, and of which the first indicates the mean longitude of the body  $m$ , the second lying perpendicular to it. Let  $x$ , and  $y$ , represent the coordinates of  $m$ , relatively to these axes, and let  $nt + \varepsilon$  denote the mean longitude of  $m$ , or

\* By differentiating the first of the equations (K), we obtain  $d^2t$

$$= \frac{-2du.dv}{u^3 \cdot \sqrt{(h^2 + 2J \left\{ \frac{dQ}{dv} \right\} \cdot \frac{dv}{u^2})}} - \frac{dv^2 \cdot \frac{dQ}{dv}}{u^4(h^2 + 2J \left\{ \frac{dQ}{dv} \right\} \cdot \frac{dv}{u^2})^{\frac{3}{2}}}; \text{ and by substituting } \frac{dt}{dv}, \frac{dt^3}{dv^3},$$

$$\text{for their values, } d^2t = -\frac{2du}{u} \cdot \frac{dt}{dv} - \frac{dt^2 \cdot \left\{ \frac{dQ}{dv} \right\}}{\sqrt{h^2 + 2J \left\{ \frac{dQ}{dv} \right\} \cdot \frac{dv}{u^2}}} = dt^3 \cdot \left\{ \frac{dQ}{dv} \right\} \cdot u^2; \quad \therefore$$

dividing by  $dv^2$ ; we obtain  $\frac{d^2t}{dv^2} = \frac{2du.dt}{udv^2} - \frac{dt^3}{dv^3} \cdot u^2 \cdot \left\{ \frac{dQ}{dv} \right\}$ , in the second and third equations, the second should be multiplied by the denominator, and then divided by  $h^2$ , the third should be multiplied by the denominator, and afterwards divided by  $h^2 \cdot u^2$ .

the angle which the moveable axis of  $x$ , makes with the axis of  $x$ ; we shall have

$$\begin{aligned}x &= x, \cos. (nt+\epsilon) - y, \sin. (nt+\epsilon); \\y &= x, \sin. (nt+\epsilon) + y, \cos. (nt+\epsilon);\end{aligned}$$

from which we collect, on the supposition that  $dt$  is constant,

$$\begin{aligned}d^2x, \cos. (nt+\epsilon) + d^2y, \sin. (nt+\epsilon) &= d^2x, -n^2x, dt^2 - 2ndy, dt; \\d^2y, \cos. (nt+\epsilon) - d^2x, \sin. (nt+\epsilon) &= d^2y, -n^2y, dt^2 + 2ndx, dt.\end{aligned}$$

By substituting in  $Q$ , in place of  $x$  and of  $y$ , their preceding values, we will obtain

$$\begin{aligned}\left\{\frac{dQ}{dx}\right\} &= \left\{\frac{dQ}{dx}\right\} \cdot \cos. (nt+\epsilon) - \left\{\frac{dQ}{dy}\right\} \cdot \sin. (nt+\epsilon); \\ \left\{\frac{dQ}{dy}\right\} &= \left\{\frac{dQ}{dx}\right\} \cdot \sin. (nt+\epsilon) + \left\{\frac{dQ}{dy}\right\} \cdot \cos. (nt+\epsilon);\end{aligned}$$

This being premised, the differential equations (*i*) will give the three following;

$$\begin{aligned}0 &= \frac{d^2x}{dt^2} - n^2x - 2n \cdot \frac{dy}{dt} - \left\{\frac{dQ}{dx}\right\};^* \\0 &= \frac{d^2y}{dt^2} - n^2y + 2n \cdot \frac{dx}{dt} - \left\{\frac{dQ}{dy}\right\}; \\0 &= \frac{d^2z}{dt^2} - \left\{\frac{dQ}{dz}\right\}.\end{aligned}$$

\*  $dx = dx, \cos. (nt+\epsilon) - dy, \sin. (nt+\epsilon) - nx, dt. \sin. (nt+\epsilon) - ny, dt. \cos. (nt+\epsilon)$ .

$dy = dx, \sin. (nt+\epsilon) + dy, \cos. (nt+\epsilon) + nx, dt. \cos. (nt+\epsilon) - ny, dt. \sin. (nt+\epsilon)$ .

$d^2x = d^2x, \cos. (nt+\epsilon) - d^2y, \sin. (nt+\epsilon) - 2ndx, dt. \sin. (nt+\epsilon) - 2ndy, dt. \cos. (nt+\epsilon) - n^2x, dt^2. \cos. (nt+\epsilon) + n^2y, dt^2. \sin. (nt+\epsilon)$ .

$d^2y = d^2x, \sin. (nt+\epsilon) + d^2y, \cos. (nt+\epsilon) + 2ndx, dt. \cos. (nt+\epsilon) - 2ndy, dt. \sin. (nt+\epsilon) - n^2x, dt^2. \sin. (nt+\epsilon) - n^2y, dt^2. \cos. (nt+\epsilon)$ .

$\therefore d^2x, \cos. (nt+\epsilon) + d^2y, \sin. (nt+\epsilon) = d^2x, -2ndy, dt - n^2x, dt^2.$

After having deduced the differential equations of a system of bodies subject to their mutual attraction, and also the only exact integrals, which we have hitherto been able to obtain, being determined ; it remains for us to integrate these equations by successive approximations. In the solar system, the heavenly bodies move very nearly as if they were only subject to the principal force which actuates them, and the disturbing forces are inconsiderable ; we are therefore permitted in a first approximation, solely to consider the mutual action of two bodies, namely, that of a planet or of a comet, and of the Sun, in the theory of the planets, and of the comets ; and the mutual action of a planet and its satellite, in the theory of the satellites. We will, therefore, commence with determining rigorously the motion of two bodies which attract each other ; this first approximation will conduct us to a second, in which we will consider the first power of the disturbing forces ; afterwards we will take into account, the squares and products of these forces ; and proceeding in this manner, we will determine the celestial motions with all the precision which the observations admit of.

$$d^2y \cdot \cos. (nt+\epsilon) - d^2x \cdot \sin. (nt+\epsilon) = d^2y + 2ndx, dt - n^2y, dt^2.$$

$$\begin{aligned} \frac{dQ}{dx} &= \left\{ \frac{dQ}{dx} \right\} \cdot \left\{ \frac{dx}{dx} \right\} + \left\{ \frac{dQ}{dy} \right\} \cdot \left\{ \frac{dy}{dx} \right\}; \text{ but } \left\{ \frac{dx}{dx} \right\} = \cos. (nt+\epsilon); \quad \left\{ \frac{dy}{dx} \right\} \\ &= -\sin. (nt+\epsilon) \therefore \left\{ \frac{dQ}{dx} \right\} = \left\{ \frac{dQ}{dx} \right\} \cdot \cos. (nt+\epsilon) - \left\{ \frac{dQ}{dy} \right\} \cdot \sin. (nt+\epsilon). \end{aligned}$$

$x = x \cdot \cos. (nt+\epsilon) + y \cdot \sin. (nt+\epsilon)$ ;  $y = y \cdot \cos. (nt+\epsilon) - x \cdot \sin. (nt+\epsilon)$ ; hence may be inferred the values of  $\frac{dx}{dx}$ ,  $\frac{dy}{dx}$ , &c. &c.

$$\begin{aligned} \frac{d^2x}{dt^2} \cdot \cos. (nt+\epsilon) &= \left\{ \frac{dQ}{dx} \right\} \cdot \cos. (nt+\epsilon) = \left\{ \frac{dQ}{dx} \right\} \cdot \cos. ^2(nt+\epsilon) - \left\{ \frac{dQ}{dy} \right\} \cdot \sin. (nt+\epsilon) \cdot \\ \cos. (nt+\epsilon); \quad \frac{d^2y}{dt^2} \cdot \sin. (nt+\epsilon) &= \left\{ \frac{dQ}{dy} \right\} \cdot \sin. (nt+\epsilon) = \left\{ \frac{dQ}{dx} \right\} \cdot \sin. ^2(nt+\epsilon) + \left\{ \frac{dQ}{dy} \right\} \cdot \\ \sin. (nt+\epsilon) \cdot \cos. (nt+\epsilon). \quad \therefore \frac{d^2x}{dt^2} \cdot \cos. (nt+\epsilon) + \frac{d^2y}{dt^2} \cdot \sin. (nt+\epsilon) = \frac{d^2x}{dt^2} - n^2x, dt - \\ 2ndy, dt &= \left\{ \frac{dQ}{dx} \right\}. \end{aligned}$$

## CHAPTER III.

*First approximation of the celestial motions, or the theory of elliptic motion.*

16. It has been already demonstrated in the first Chapter, that a body attracted to a fixed point, by a force which is inversely as the square of the distance, describes a conic section; but in the relative motion of the body  $m$  about  $M$ , if this last body be considered at rest, we should transfer to  $m$  in an opposite direction, the action which  $m$  exercises on  $M$ ; therefore, in this relative motion,  $m$  is sollicited towards  $M$  by a force which is equal to the sum of the masses divided by the square of their distance, consequently the body  $m$  describes a conic section about  $M$ . But the importance of this subject in the theory of the system of the world, requires that it should be resumed under new points of view.

For this purpose, let us consider the equations (K) of No. 15. If  $M+m$  be made  $= \mu$ , it is evident from No. 14, that if we only consider the reciprocal action of  $M$  on  $m$ ,  $Q$  is equal to  $\frac{\mu}{r}$  or to

$\frac{\mu u}{\sqrt{1+s^2}}$ , the equations (K) will consequently become,

$$dt = \frac{dv}{h.u^2};$$

$$0 = \frac{d^2u}{dv^2} + u - \frac{\mu}{h^2(1+s^2)^{\frac{3}{2}}};^*$$

$$0 = \frac{d^2s}{dv^2} + s.$$

The area described by the projection of the radius vector, during the element of time  $dt$ , being equal to  $\frac{1}{2} \cdot \frac{dv}{u^2}$ ;† the first of these equations indicates that this area is proportional to this element, and that consequently in a finite time, it is proportional to the time. By integrating the last equation we obtain

$$s = \gamma \cdot \sin. (v - \theta), \ddagger$$

\*  $\left\{ \frac{dQ}{du} \right\} = \frac{\mu}{\sqrt{1+s^2}}$ ,  $\left\{ \frac{dQ}{ds} \right\} = \frac{-\mu us}{(1+s^2)^{\frac{3}{2}}}$ ,  $\left\{ \frac{dQ}{dv} \right\} = 0$ ; therefore if these values of  $\left\{ \frac{dQ}{dv} \right\}$ ,  $\left\{ \frac{dQ}{ds} \right\}$ ,  $\left\{ \frac{dQ}{du} \right\}$  be substituted in the equations (K); the second of these equations becomes

$$\frac{d^2u}{dv^2} + u - \frac{dQ}{du} - \frac{s}{u} \cdot \frac{dQ}{ds} = \frac{d^2u}{dv^2} + u - \frac{\mu}{h^2\sqrt{1+s^2}} + \frac{us^2}{h^2(1+s^2)^{\frac{3}{2}}} = \frac{d^2u}{dv^2} +$$

$$+ u - \frac{\mu}{h^2(1+s^2)^{\frac{3}{2}}}, \text{ and the third equation becomes } \frac{d^2s}{dv^2} + s -$$

$$\frac{\mu us}{\sqrt{1+s^2}} + \frac{\mu us}{\sqrt{1+s^2}} = \frac{d^2s}{dv^2} + s.$$

†  $\frac{1}{2} \cdot \frac{dv}{u^2} = \frac{1}{2} \cdot dv \cdot r^2 \cdot \cos^2 \theta$  = the element of the area described in a given time by the projection of the radius vector; see page 4.

‡  $\frac{d^2s}{dv^2} + s = 0$ ; ∵  $\frac{d^2s \cdot ds}{dv^2} + sds = 0$ , therefore by integrating  $\frac{ds^2}{dv^2} + s^2 = c$ , it is evident that  $s = \sin. v$ . or  $s = \cos. v$ , and that ∵  $s = a \sin. v$ , or  $s = b \cdot \cos. v$ , and consequently  $s = a \cdot \sin. v + b \cdot \cos. v$ . will satisfy the given equation, and be its complete integral; as it contains two independent arbitrary quantities. Now,  $a \sin. v + b \cdot \cos. v$ . may be reduced to the form  $\gamma \sin. (v - \theta)$ , by assuming  $a = \gamma \cdot \cos. \theta$ ,  $b = -\gamma \cdot \sin. \theta$ , which gives  $a \cdot \sin. v + b \cdot \cos. v = \gamma \cdot (\sin. v \cdot \cos. \theta - \cos. v \cdot \sin. \theta) = \gamma \cdot \sin. (v - \theta)$ , and it may be shewn that  $\gamma \cdot \sin. (v - \theta)$ , likewise satisfies this equation. It is also

$\gamma$  and  $\theta$  being two arbitrary quantities. Finally, the second equation gives by its integration

$$u = \frac{\mu}{h^2(1+\gamma^2)} \left\{ \sqrt{1+s^2} + e. \cos.(v-\varpi) \right\} = \frac{\sqrt{1+s^2}}{r}; *$$

$e$  and  $\varpi$  being two new arbitrary quantities. By substituting in this

o 2

evident, that  $s = a. \sin.(v-\theta) + b. \cos.(v-\theta)$  will satisfy the equation  $\frac{d^2s}{dv^2} + s = 0$ , and may be used when convenient, but in this case  $a$ ,  $b$  and  $\theta$ , must be selected in such a manner, that they may be reduced to two *independent* quantities.

\* In the equation  $\frac{d^2u}{dv^2} + u - \frac{\mu}{h^2(1+s^2)^{\frac{3}{2}}} = 0$ , let  $P = \frac{\mu}{h^2(1+s^2)^{\frac{3}{2}}}$ , and  $u = a. \sin.(v-\theta) + b. \cos.(v-\theta)$  will be the complete integral of the equation  $\frac{d^2u}{dv^2} + u = 0$ ; and  $a. \sin.(v-\theta)$  and  $b. \cos.(v-\theta)$  will respectively satisfy the equation  $\frac{d^2u}{dv^2} + u = 0$ ; now if the expression  $a. \sin.(v-\theta) + b. \cos.(v-\theta)$  be regarded as the integral of the differential equation  $\frac{d^2u}{dv^2} + v - P = 0$ ;  $a$  and  $b$  must in this case be functions of the variables  $v$ , and as there is only one equation to verify by means of  $a$  and  $b$ , we can impose certain conditions on them which will facilitate their determination; supposing them to be functions of  $v$  in the equation  $u = a. \sin.(v-\theta) + b. \cos.(v-\theta)$ , we shall have

$$du = adv. \cos.(v-\theta) - b. dv. \sin.(v-\theta) + da. \sin.(v-\theta) + db. \cos.(v-\theta);$$

but as there are two quantities to be determined, and as the proposed question furnishes us with but one condition, we are at liberty to select the other condition; for this purpose let

$$da. \sin.(v-\theta) + db. \cos.(v-\theta) = 0;$$

then  $du = dv. (a. \cos.(v-\theta) - b. \sin.(v-\theta))$ ; and consequently,

$$d^2u = -dv^2. (a. \sin.(v-\theta) + b. \cos.(v-\theta)) + dv. da. \cos.(v-\theta) - dv. db. \sin.(v-\theta);$$

and this value of  $d^2u$  being substituted in the equation  $\frac{d^2u}{dv^2} + u - \frac{\mu}{h^2(1+s^2)^{\frac{3}{2}}} = 0$  gives,  
 $adv^2. (\sin.(v-\theta) - \sin.(v-\theta)) + bdv^2. (\cos.(v-\theta) - \cos.(v-\theta)) + da. dv. \cos.(v-\theta) - db. dv. \sin.(v-\theta) - Pdv^2 = 0$ ;  $\therefore da. dv. \cos^2.(v-\theta) - db. dv. \sin.(v-\theta). \cos.(v-\theta)$

expression for  $u$ , in place of  $s$ , its value in terms of  $v$ , and then substituting this expression, in the equation  $dt = \frac{dv}{h \cdot u^2}$ ; the integral of the resulting equation will give  $t$  in a function of  $v$ ; therefore we shall have  $v$ ,  $u$  and  $s$ , in functions of the time.

$-P \cdot \cos.(v-\theta) \cdot dv^2 = 0$ ; and if this equation be divided by  $dv$ , and then added to the equation  $da \cdot \sin^2.(v-\theta) + db \cdot \sin.(v-\theta) \cdot \cos.(v-\theta) = 0$ , we shall have  $da = P \cdot \cos.(v-\theta) \cdot dv$ , of which the integral is  $a = a' + \int P \cdot \cos.(v-\theta) \cdot dv$ ; in like manner if the same equations be respectively multiplied by  $\cos.(v-\theta)$ ,  $\sin.(v-\theta)$ , we obtain by subtracting the second, divided by  $dv$ , from the first;  $db = -P \cdot \sin.(v-\theta) \cdot dv$ ; and  $b = b' - \int P \cdot \sin.(v-\theta) \cdot dv$ . Therefore  $u = a \cdot \sin.(v-\theta) + b \cdot \cos.(v-\theta) = a' \cdot \sin.(v-\theta) + \sin.(v-\theta) \int P \cdot \cos.(v-\theta) \cdot dv + b' \cdot \cos.(v-\theta) - \cos.(v-\theta) \int P \cdot \sin.(v-\theta) \cdot dv$ ;  $a'$  and  $b'$  are the values of  $a$  and  $b$  when  $P = 0$ ;

$$P = \frac{\mu}{h^2(1+s^2)^{\frac{3}{2}}} = (\text{by substituting for } s^2 \text{ its value}) \frac{\mu}{h^2(1+\gamma^2 \cdot \sin^2(v-\theta))^{\frac{3}{2}}}, \text{ therefore}$$

$$\sin.(v-\theta) \int P \cdot \cos.(v-\theta) \cdot dv = \frac{\mu \cdot \sin.(v-\theta)}{h^2} \cdot \int \frac{\cos.(v-\theta) \cdot dv}{(1+\gamma^2 \cdot \sin^2(v-\theta))^{\frac{3}{2}}}, \text{ but}$$

$$\int \frac{\cos.(v-\theta) \cdot dv}{h^2(1+\gamma^2 \cdot \sin^2(v-\theta))^{\frac{3}{2}}} = \frac{\sin.(v-\theta)}{h^2(1+\gamma^2 \cdot \sin^2(v-\theta))^{\frac{1}{2}}}, \text{ for } d \cdot \frac{\sin.(v-\theta)}{h^2(1+\gamma^2 \cdot \sin^2(v-\theta))^{\frac{1}{2}}}$$

$$= \frac{\cos.(v-\theta) \cdot dv}{h^2(1+\gamma^2 \cdot \sin^2(v-\theta))^{\frac{1}{2}}} - \frac{\gamma^2 \cdot \sin^2(v-\theta) \cdot \cos.(v-\theta) \cdot dv}{h^2(1+\gamma^2 \cdot \sin^2(v-\theta))^{\frac{3}{2}}} = \text{by reducing to a com-}$$

$$\text{mon denominator } \frac{\cos.(v-\theta) \cdot dv}{h^2(1+\gamma^2 \cdot \sin^2(v-\theta))^{\frac{3}{2}}}; \text{ consequently } \frac{\mu \cdot \sin.(v-\theta)}{h^2}.$$

$$\int \frac{\cos.(v-\theta) \cdot dv}{(1+\gamma^2 \cdot \sin^2(v-\theta))^{\frac{3}{2}}} = \frac{\mu}{h^2} \cdot \frac{\sin^2(v-\theta)}{(1+\gamma^2 \cdot \sin^2(v-\theta))^{\frac{1}{2}}}.$$

$$- \cos.(v-\theta) \cdot \int P \cdot \sin.(v-\theta) \cdot dv =$$

$$- \mu \cdot (\cos.(v-\theta) \cdot \int \frac{\sin.(v-\theta) \cdot dv}{h^2(1+\gamma^2 \cdot \sin^2(v-\theta))^{\frac{3}{2}}}, \text{ and } \int \frac{\sin.(v-\theta) \cdot dv}{h^2(1+\gamma^2 \cdot \sin^2(v-\theta))^{\frac{3}{2}}}$$

$$= \frac{-1}{(1+\gamma^2)} \cdot \frac{\cos.(v-\theta)}{h^2(1+\gamma^2 \cdot \sin^2(v-\theta))^{\frac{1}{2}}}, \text{ for } \frac{1}{1+\gamma^2} \cdot d \cdot \frac{\cos.(v-\theta)}{h^2(1+\gamma^2 \cdot \sin^2(v-\theta))^{\frac{1}{2}}}$$

$$= -\frac{1}{1+\gamma^2} \cdot \frac{\sin.(v-\theta) \cdot dv}{h^2(1+\gamma^2 \cdot \sin^2(v-\theta))^{\frac{1}{2}}} - \frac{1}{1+\gamma^2} \cdot \gamma^2 \cdot \frac{\sin.(v-\theta) \cdot \cos^2(v-\theta) \cdot dv}{h^2(1+\gamma^2 \cdot \sin^2(v-\theta))^{\frac{3}{2}}}$$

$$= \text{by reducing } - \frac{1}{1+\gamma^2} \cdot \frac{(\sin.(v-\theta) + \gamma^2 \cdot \sin.(v-\theta) (\sin^2(v-\theta) + \cos^2(v-\theta))) \cdot dv}{h^2(1+\gamma^2 \cdot \sin^2(v-\theta))^{\frac{3}{2}}}$$

The calculus may be considerably simplified, by observing that the value of  $s$  indicates that the orbit exists entirely\* in a plane of which  $\gamma$  is the tangent of the inclination to a fixed plane, and of which  $\theta$  represents the longitude of the node, reckoned from the origin of the angle  $v$ . Consequently, if we refer the motion of  $m$  to this plane, we shall have  $s=0$ , and  $\gamma=0$ , which gives

$$u = \frac{1}{r} = \frac{\mu}{h^2} \left\{ 1 + e \cdot \cos. (v - \omega) \right\}.$$

This is the equation of an ellipse, in which the origin of the radii is at the focus :  $\frac{h^2}{\mu \cdot (1 - e^2)}$ , is the semiaxis major, which we will represent by  $a$ ;  $e$  is the ratio of the excentricity to the semiaxis major;

$$\begin{aligned} &= -\frac{1}{1+\gamma^2} \cdot \frac{\sin. (v-\theta) \cdot (1+\gamma^2) \cdot dv}{h^2(1+\gamma^2 \cdot \sin. ^2(v-\theta))^{\frac{3}{2}}} = -\frac{\sin. (v-\theta) \cdot dv}{h^2(1+\gamma^2 \cdot \sin. ^2(v-\theta))^{\frac{3}{2}}}, \\ &\therefore \frac{\mu \cdot \sin. (v-\theta) \cdot \int \cos. (v-\theta) \cdot dv}{h^2(1+\gamma^2 \cdot \sin. ^2(v-\theta))^{\frac{3}{2}}} = \frac{\mu \cdot \cos. (v-\theta) \int \sin. (v-\theta) \cdot dv}{h^2(1+\gamma^2 \cdot \sin. ^2(v-\theta))^{\frac{3}{2}}} \\ &= \frac{\mu \sin. ^2(v-\theta)}{h^2(1+\gamma^2 \cdot \sin. ^2(v-\theta))^{\frac{1}{2}}} + \frac{1}{1+\gamma^2}, \frac{\mu \cos. ^2(v-\theta)}{h^2(1+\gamma^2 \cdot \sin. ^2(v-\theta))^{\frac{1}{2}}} = \\ &\mu \cdot \frac{(\sin. ^2(v-\theta) + \cos. ^2(v-\theta) + \gamma^2 \cdot \sin. ^2(v-\theta))}{(1+\gamma^2)^{\frac{1}{2}} \cdot h^2(1+\gamma^2 \cdot \sin. ^2(v-\theta))^{\frac{1}{2}}} = \mu \cdot \frac{(1+\gamma^2 \cdot \sin. ^2(v-\theta))^{\frac{1}{2}}}{(1+\gamma^2) \cdot h^2} \\ &= \mu \cdot \frac{(1+s^2)^{\frac{1}{2}}}{(1+\gamma^2) \cdot h^2}, \therefore u = a' \cdot \sin. (v-\theta) + b' \cdot \cos. (v-\theta) + \mu \cdot \frac{(1+s^2)^{\frac{1}{2}}}{(1+\gamma^2) \cdot h^2}, \text{ and as } e' \\ &\cos. (v-\omega) \text{ satisfies the equation } \frac{d^2u}{dv^2} + u = 0, \text{ we may write this function instead of } \\ &a' \cdot \sin. (v-\theta) + b' \cdot (\cos. (v-\theta)), \text{ and as } e' \text{ is arbitrary we can assume it equal to } \frac{\mu}{h^2(1+\gamma^2)}. \\ &\text{e, by means of which the expression for } u \text{ will assume the form given in the text.} \end{aligned}$$

\*  $\gamma$  is evidently equal to the tangent of latitude, when  $v-\theta=90$ , and consequently it is in this case equal to the inclination of the orbit; and as  $\sin. (v-\theta) = \frac{s}{\gamma} = s \cdot \cotangent$  of inclination; the orbit described must be a plane, for this equation expresses the relation between the two sides, and *invariable* angle of a spherical triangle.

finally,  $\varpi$  is the longitude of the perihelium. The equation  $dt = \frac{dv}{h.u^2}$  becomes, by substituting in place of  $u^2$ ,

$$dt = \frac{a^{\frac{3}{2}}(1-e^2)^{\frac{3}{2}}. dv}{\sqrt{\mu} \cdot (1+e \cdot \cos(v-\varpi))^2} .*$$

Let us expand the second member of this equation, into a series proceeding according to the cosines of the angle  $v-\varpi$ , and of its multiples.

For this purpose, we will commence by expanding  $\frac{1}{1+e \cdot \cos(v-\varpi)}$  into a similar series. By making

$$\lambda = \frac{e}{1+\sqrt{1-e^2}};$$

we shall have

$$\frac{1}{1+e \cdot \cos(v-\varpi)} = \frac{1}{\sqrt{1-e^2}} \left\{ \frac{1}{1 + \lambda \cdot c^{(v-\varpi)\sqrt{-1}}} - \frac{\lambda \cdot c^{-(v-\varpi)\sqrt{-1}}}{1 + \lambda \cdot c^{-(v-\varpi)\sqrt{-1}}} \right\}; \dagger$$

$$* \quad \frac{1}{u} = r = \frac{h^2}{\mu(1+e \cdot \cos(v-\varpi))} = \frac{a(1-e^2)}{1+e \cdot \cos(v-\varpi)}, \quad \because a = \frac{h^2}{\mu(1-e^2)}; \text{ hence } h =$$

$$\sqrt{a} \cdot \sqrt{\mu(1-e^2)}, \text{ and } dt = \frac{dv}{h.u^2} = \frac{dv.h^3}{\mu^2(1+e \cdot \cos(v-\varpi))^2} = \frac{dv.a^{\frac{3}{2}}.\mu^{\frac{3}{2}}.(1-e^2)^{\frac{3}{2}}}{\mu^2.(1+e \cdot \cos(v-\varpi))^2}.$$

$\ddagger$  By reducing the coefficient of  $\frac{1}{\sqrt{1-e^2}}$ , in the second member of this equation to the same denominator, it becomes equal to

$$\frac{1-\lambda^2}{\sqrt{1-e^2}.(1+\lambda^2+\lambda(c^{(v-\varpi)\sqrt{-1}}+\lambda(c^{-(v-\varpi)\sqrt{-1}})))}$$

$$\text{but } c^{(v-\varpi)\sqrt{-1}}+c^{-(v-\varpi)\sqrt{-1}}=2 \cos(v-\varpi), \quad \therefore \text{this second member}=$$

$e$  being the number of which the hyperbolical logarithm is equal to unity. By expanding the second member of this equation, into a series; namely, the first term relatively to the powers of  $c^{(v-\omega)\sqrt{-1}}$  and the second term relatively to the powers  $c^{-(v-\omega)\sqrt{-1}}$ , and then substituting in place of the imaginary exponentials their expressions in sines and cosines; we shall find

$$\frac{1}{1+e \cdot \cos. (v-\omega)} = \frac{1}{\sqrt{1-e^2}} \cdot *$$

$$(1-2\lambda \cdot \cos. (v-\omega) + 2\lambda^2 \cdot \cos. 2(v-\omega) - 2\lambda^3 \cdot \cos. 3.(v-\omega) + \&c.) ;$$

By representing the second member of this equation by  $\varphi$ , and making  $q = \frac{1}{e}$ , we shall have generally,

$$\frac{1-\lambda^2}{\sqrt{1-e^2}(1+\lambda^2 + \lambda \cdot \cos. (v-\omega))}; \text{ and from the equation } \lambda = \frac{e}{(1+\sqrt{1-e^2})}, \text{ we obtain}$$

$$1-\lambda^2 = \frac{2(1+\sqrt{1-e^2}-e^2)}{(1+\sqrt{1-e^2})^2}, \text{ and } 1+\lambda^2 = \frac{2(1+\sqrt{1-e^2})}{(1+\sqrt{1-e^2})^2}; \therefore \text{ by substituting for}$$

$$1-\lambda^2, \text{ and } 1+\lambda^2 \text{ we obtain } \frac{2(1-e^2+\sqrt{1-e^2})}{2\sqrt{1-e^2}(1+\sqrt{1-e^2})(1+e \cdot \cos. (v-\omega))} = \frac{1}{1+e \cdot \cos. (v-\omega)}.$$

\* The expression of the first term gives the following series :

$$1-\lambda \cdot c^{(v-\omega)\sqrt{-1}} + \lambda^2 \cdot c^{2(v-\omega)\sqrt{-1}} - \lambda^3 \cdot c^{3(v-\omega)\sqrt{-1}} + \lambda^4 \cdot c^{4(v-\omega)\sqrt{-1}} - \&c;$$

the expansion of the second term gives

$$-\lambda \cdot c^{-(v-\omega)\sqrt{-1}} \cdot (1-\lambda \cdot c^{-(v-\omega)\sqrt{-1}} + \lambda^2 \cdot c^{-2(v-\omega)\sqrt{-1}} - \lambda^3 \cdot c^{-3(v-\omega)\sqrt{-1}} + \lambda^4 \cdot c^{-4(v-\omega)\sqrt{-1}} + \&c.);$$

making the factors of the same powers of  $\lambda$  to coalesce in the two series, and observing that  $\lambda^i (c^{i(v-\omega)\sqrt{-1}} + c^{-i(v-\omega)\sqrt{-1}}) = \lambda^i \cdot \cos. i(v-\omega)$ , we will obtain the value of  $\frac{1}{1+e \cdot \cos. (v-\omega)}$ , which is given in the text.

$$\frac{1}{(1+e \cdot \cos(v-\omega))^{m+1}} = \pm \frac{e^{-m-1} \cdot d \cdot \left\{ \frac{\phi}{q} \right\}}{1 \cdot 2 \cdot 3 \cdots \cdots m \cdot dq^m}; *$$

in which  $dq$  is supposed to be constant, and the sign is + or -, according as  $m$  is even or odd. From this, it is easy to infer, that if we make

$$\frac{1}{(1+e \cdot \cos(v-\omega))^2} = (1-e^2)^{-\frac{3}{2}}$$

$$(1+E^{(1)} \cdot \cos(v-\omega) + E^{(2)} \cdot \cos 2(v-\omega) + E^{(3)} \cdot \cos 3(v-\omega) + \text{&c.});$$

we shall have, whatever may be the value of  $i$ ,

$$E^{(i)} = \pm \frac{2e^i (1+i \cdot \sqrt{1-e^2})}{(1+\sqrt{1-e^2})^i}; †$$

the sign being +, if  $i$  is even, and - if  $i$  is odd; therefore if  $n$  be

\* Substituting  $\frac{1}{q}$  for  $e$  we obtain  $\frac{1}{1+e \cdot \cos(v-\omega)} = \frac{q}{q+\cos(v-\omega)} = \phi$ , ∵

$$\frac{1}{(q+\cos(v-\omega))} = \frac{\phi}{q}, \text{ and } d \cdot \frac{1}{q+\cos(v-\omega)} \div dq = \frac{-1}{(q+\cos(v-\omega))^2} = d \cdot \left\{ \frac{\phi}{q} \right\} ;$$

$$\text{and } d^2 \cdot \left\{ \frac{\phi}{q} \right\} = d \cdot \frac{-1}{(q+\cos(v-\omega))^2} \div dq = \frac{2}{(q+\cos(v-\omega))^3}, \text{ and } d^3 \cdot \left\{ \frac{\phi}{q} \right\} = \\ = d \cdot \frac{2}{(q+\cos(v-\omega))^3} \div dq = \frac{-2.3}{(q+\cos(v-\omega))^4} : \text{ hence generally we obtain } d^m \left\{ \frac{\phi}{q} \right\} \\ \div dq^m$$

$$= \frac{\pm \cdot 1 \cdot 2 \cdot 3 \cdots \cdots m}{(q+\cos(v-\omega))^{m+1}} = \frac{\pm 1 \cdot 2 \cdot 3 \cdots \cdots m e^{m+1}}{(1+e \cdot \cos(v-\omega))^{m+1}}.$$

† Substituting  $\frac{1}{q}$  for  $e$ , in the value of  $\phi$ , we obtain  $\frac{\phi}{q} = \frac{1}{\sqrt{q^2-1}} \cdot (1-2\lambda \cdot \cos(v-\omega)+2\lambda^2 \cdot \cos 2(v-\omega)-2\lambda^3 \cdot \cos 3(v-\omega)+\text{&c.})$ .

$$\therefore \frac{1}{(1+e \cdot \cos(v-\omega))^2} = e^{-2} \cdot d \cdot \left\{ \frac{\phi}{q} \right\} = \text{the}$$

supposed equal to  $a^{-\frac{3}{2}} \cdot \sqrt{\mu}$ , we shall have

$$ndt = dv \cdot (1 + E^{(1)} \cdot \cos(v - \varpi) + E^{(2)} \cdot \cos. 2(v - \varpi) + E^{(3)} \cdot \cos. 3(v - \varpi) + \text{&c.}) ;$$

and by integrating

$$nt + \epsilon = v + E^{(1)} \cdot \sin. (v - \varpi) + \frac{1}{2} E^{(2)} \cdot \sin. 2(v - \varpi) + \frac{1}{3} E^{(3)} \cdot \sin. 3(v - \varpi) + \text{&c.}$$

$\epsilon$  being a constant arbitrary quantity. This expression for  $nt + \epsilon$  is very converging\* when the orbits have a very small excentricity, such as the orbits of the planets and of the satellites; and we can, by the

preceding series differenced with respect to  $q$ , and divided by  $e^2$ ; the differential of the  $i$  term  $= e^{-2} d \cdot \frac{1}{\sqrt{(q^2-1)}} 2\lambda^i = 2e^{-2} d \cdot \frac{1}{\sqrt{q^2-1}} \cdot \left\{ \frac{1}{(q+\sqrt{q^2-1})^i} \right\} \div dq = \pm 2e^{-2} \frac{q}{(q^2-1)^{\frac{3}{2}}} \cdot \frac{1}{(q+\sqrt{q^2-1})^i} \pm 2e^{-2} \frac{1}{\sqrt{q^2-1}} \cdot i \cdot \left\{ 1 + \frac{q}{\sqrt{q^2-1}} \right\} \frac{1}{(q+\sqrt{q^2-1})^{i+1}} =$

$$\pm 2e^{-2} \cdot \left\{ \frac{q}{(q^2-1)^{\frac{3}{2}}} \cdot \frac{1}{(q^2+\sqrt{q^2-1})^i} \right\} \pm \frac{e^{-2} \cdot i}{q^2-1} \cdot \frac{(q+\sqrt{q^2-1})}{(q+\sqrt{q^2-1})^{i+1}} ;$$

= by simplifying and reducing to a common denominator,

$$\pm 2e^{-2} \cdot \frac{(q+i\sqrt{q^2-1})}{(q^2-1)^{\frac{3}{2}} \cdot (q+\sqrt{q^2-1})^i} , \text{ which becomes, by substituting } \frac{1}{e} \text{ for } q,$$

$$\pm \frac{2e^i(1+i\sqrt{1-e^2})}{(1-e^2)^{\frac{3}{2}}(1+\sqrt{1-e^2})^i} , \text{ the expression given in the text.}$$

\*  $(1-e^2)^{\frac{3}{2}}$  occurs both in the numerator and also in the denominator of the value of  $n.dt$ , as is evident from the value of  $dt$  given in page 101, compared with the preceding expression; when the excentricity of the orbit is inconsiderable,  $e$  which expresses the ratio of the excentricity to the semiaxis major will be very small,  $\therefore$  the value of  $E^{(i)}$ , in which  $e^i$  occurs as a factor will be very small, and perpetually less and less.

reversion of series, conclude the value of  $v$  in terms of  $t$ ; we will effect this, in the subsequent Nos.

When\* the planet returns to the same point in its orbit,  $v$  is increased by the circumference which is always represented by  $2\pi$ ; naming  $T$  the periodic time, we shall have

$$T = \frac{2\pi}{n} = \frac{2\pi \cdot a^{\frac{3}{2}}}{\sqrt{\mu}},$$

This value of  $T$  may be easily deduced from the differential expression for  $dt$ , without recurring to series. In fact, let us resume the equation  $dt = \frac{dv}{h \cdot u^2}$ , or  $dt = \frac{r^2 \cdot dv}{h}$ . From it, we obtain  $T = \int \frac{r^2 \cdot dv}{h}$ ;  $\int r^2 \cdot dv$  is double the surface of the ellipse, and consequently it is equal to  $2\pi \cdot a^2 \cdot \sqrt{1-e^2}$ ; moreover,  $h^2$  is equal to  $\mu a \cdot (1-e^2)$ ; thus we shall obtain the same expression for  $T$ , as has been given above.

If the masses of the planets be neglected relatively to that of the sun, we have  $\sqrt{\mu} = \sqrt{M}$ ; the value of  $\mu$  is then the same for all the planets;  $T$  is therefore proportional to  $a^{\frac{3}{2}}$ , and consequently, the squares of the periodic times, are as the cubes of the greater axes of the orbits. It is evident, that the same law obtains in the motions of the satellites about their primary, their masses being neglected relatively to that of the primary.

### 17. The equations of the motion of two bodies, which attract each

\* When the [planet returns to the same point, the terms of this equation will become

$$n(t+T)+\varepsilon = v + 2\pi + E^{(1)} \cdot \sin.((v-\varpi)+2\pi) + E^{(2)} \cdot \sin.2(v-\varpi)+2\pi + \&c.$$

if this equation be taken from the equation  $nt+\varepsilon =$

$v+E^{(1)} \cdot \sin.(v-\varpi) + E^{(2)} \cdot \sin.2(v-\varpi) + E^{(3)} \cdot \sin.3(v-\varpi) + \&c.$  the difference will be  $nT=2\pi$ .

other in the inverse ratio of the squares of the distances, may be also integrated in the following manner: the equations (1), (2), (3), of No. 9, become, when we only consider the action of the two bodies  $M$  and  $m$ ,

$$\left. \begin{aligned} 0 &= \frac{d^2x}{dt^2} + \frac{\mu.x}{r^3} \\ 0 &= \frac{d^2y}{dt^2} + \frac{\mu.y}{r^3} \\ 0 &= \frac{d^2z}{dt^2} + \frac{\mu.z}{r^3} \end{aligned} \right\} \quad (O)$$

( $\mu$  being equal to  $M + m$ ).

The integrals of these equations will give the three coordinates  $x, y, z$ , of the body  $m$ , referred to the centre of  $M$ , in a function of the time, and then by No. 9, we can obtain the coordinates  $\zeta, \Pi$  and  $\gamma$  of the body  $M$ , referred to a fixed point, by means of the equations

$$\zeta = a + bt - \frac{mx}{M+m}; \quad \Pi = a' + b't - \frac{my}{M+m}; \quad \gamma = a'' + b''t - \frac{mz}{M+m}.$$

Finally, we shall have the coordinates of  $m$ , referred to the same fixed point, by adding  $\zeta$  to  $x$ ,  $\Pi$  to  $y$ , and  $\gamma$  to  $z$ ; by this means we shall obtain the relative motions of the bodies  $M$  and  $m$ , and also their absolute motion in space. Therefore every thing depends on the integration of the differential equations (O).

For this purpose, it may be observed, that if there is given between the  $n$  variables  $x^{(1)}, x^{(2)}, x^{(3)}, \dots, x^{(n)}$ , and the variable  $t$ , of which the difference is supposed to be constant, a number  $n$  of differential equations determined by the following :

$$0 = \frac{d^i x^{(s)}}{dt^i} + \frac{A.d^{i-1}x^{(s)}}{dt^{i-1}} + \frac{B.d^{i-2}x^{(s)}}{dt^{i-2}} + \dots + H.x^{(s)},$$

\* In every equation of the same form as that in the text, if the  $x^{(n-i+1)}, x^{(n-i+2)}$ ,

in which we suppose that  $s$  is successively equal to 1, 2, 3, ..... $n$ ;  $A, B, \dots H$  being functions of the variables  $x^{(1)}, x^{(2)}, x^{(3)}, \dots x^{(n)}$ , and of  $t$ , symmetrical with respect to the variables  $x^{(1)}, x^{(2)}, \dots x^{(n)}$ , that is such, that they remain the same when any one of these variables is changed into the other, and vice versa, we can suppose

$$\begin{aligned}x^{(1)} &= a^{(1)} x^{(n-i+1)} + b^{(1)} \cdot x^{(n-i+2)} \dots + h^{(1)} \cdot x^{(n)}; \\x^{(2)} &= a^{(2)} x^{(n-i+1)} + b^{(2)} \cdot x^{(n-i+2)} \dots + h^{(2)} \cdot x^{(n)}; \\&\dots \dots \dots \dots \dots \dots \dots \dots \\x^{(n-i)} &= a^{(n-i)} \cdot x^{(n-i+1)} + b^{(n-i)} \cdot x^{(n-i+2)} \dots + h^{(n-i)} \cdot x^{(n)},\end{aligned}$$

$a^{(1)}, b^{(1)}, \dots, h^{(1)}, a^{(2)}, b^{(2)}, \dots$ , &c. being arbitrary quantities of which the number is equal to  $i(n-i)$ . It is evident that these values satisfy the proposed system of differential equations: moreover, they reduce these equations, to  $i$  differential equations between the  $i$  variables  $x^{(n-i+1)}, x^{(n-i+2)}, \dots, x^{(n)}$ . Their integrals will introduce  $i^2$  new

$x^{(n-i+3)}, \dots, x^{(n)}$ , quantities satisfy this equation; then their sum will also satisfy the same equation, as will appear by substitution, and we are at liberty to assume  $x^{(1)} = a^{(1)} \cdot x^{(n-i+1)} + b^{(1)} \cdot x^{(n-i+2)} \dots + h^{(1)} \cdot x^{(n)}$ . In each of the values of  $x^{(1)}, x^{(2)}, \dots, x^{(n)}$ , there are  $i$  arbitrary quantities;  $\therefore$  in the sum of all the values of the  $n-i$  quantities these are  $i(n-i)$  arbitrary quantities. In the integration of a differential equation of the  $i$  order, there are  $i$  arbitrary quantities introduced.  $\therefore$  In the integration of  $i$  differential equations of the  $i$  order, there must be in all,  $i^2$  arbitrary quantities.

This theorem is evidently applicable to the differential equations (O); for these equations are symmetrical with respect to  $x, y, z$ , and remain the same, when any one of the variables is changed into another;  $\therefore$  as  $x, y, z$ , correspond to  $x^{(i)}, x^{(n-i+1)}, x^{(n-i+2)}$ , &c. in the theorem, we are at liberty to assume one of them  $z$  equal to the other two, multiplied respectively by arbitrary quantities.

arbitrary variables, which combined with the  $i(n-i)$  variables, already given, will constitute the arbitrary quantities, which would be produced by the integration of the proposed differential equations.

The application of this theorem, to the equations (O), gives  $z=ax+by$ ,  $a$  and  $b$  being two arbitrary quantities. This equation is that of a plane passing through the origin of the coordinates; consequently, the orbit of  $m$  exists entirely in the same plane.

The equations (O) give

$$\left. \begin{aligned} 0 &= d. \left\{ r^3. \frac{d^2x}{dt^2} \right\} + \mu.d_x \\ 0 &= d. \left\{ r^3. \frac{d^2y}{dt^2} \right\} + \mu.d_y \\ 0 &= d. \left\{ r^3. \frac{d^2z}{dt^2} \right\} + \mu.d_z \end{aligned} \right\}; \quad (O')$$

but by differentiating twice successively, the equation  $rdr = xdx+ydy+zdz$ , we obtain

$$\begin{aligned} r.d^3r + 3dr.d^2r &= x.d^3x + y.d^3y + z.d^3z \\ &\quad + 3.(dx.d^2x + dy.d^2y + dz.d^2z), \end{aligned}$$

and consequently,

$$\begin{aligned} d. \left\{ r^3. \frac{d^2r}{dt^2} \right\} &= r^3. \left\{ x. \frac{d^3x}{dt^3} + y. \frac{d^3y}{dt^3} + z. \frac{d^3z}{dt^3} \right\} * \\ &\quad + 3r^2. \left\{ dx. \frac{d^2x}{dt^2} + dy. \frac{d^2y}{dt^2} + dz. \frac{d^2z}{dt^2} \right\}. \end{aligned}$$

By substituting in the second member of this equation, in place of  $d^3x$ ,  $d^3y$ ,  $d^3z$ , their values determined by the equations (O'), and then,

\*  $rd^2r + dr^2 = xd^2x + yd^2y + zd^2z + dx^2 + dy^2 + dz^2$ ,  $\therefore rd^3r + 3drd^2r = xd^3x + yd^3y + zd^3z + 3dx.d^2x + 3dy.d^2y + 3dz.d^2z$ , and multiplying by  $r^2$  we obtain the expression in the text.

in place of  $d^2x, d^2y, d^2z$ , their values given by the equations (O); we shall find

$$0 = d \left\{ r^3 \cdot \frac{d^2r}{dt^2} \right\} + \mu \cdot dr.$$

The comparison of this equation with the equations (O'), will give, in consequence of the theorem which has been announced above,  $(\frac{dx}{dt}, \frac{dy}{dt}, \frac{dz}{dt}, \frac{dr}{dt})$ , being considered as corresponding to the particular variables  $x^{(1)}, x^{(2)}, x^{(3)}, x^{(4)}$ , and  $r$  being supposed a function of the time  $t$  ;)

$$dr = \lambda \cdot dx + \gamma \cdot dy;$$

$\lambda, \gamma$ , being constant arbitrary quantities; and by integrating,

$$r = \frac{h^2}{\mu} + \lambda x + \gamma y, \dagger$$

$\frac{h^2}{\mu}$  being a constant quantity. This equation combined with the following :

$$z = ax + by; \quad r^2 = x^2 + y^2 + z^2,$$

\* From the equation (O') we obtain  $r^3 \cdot x \cdot \frac{d^3x}{dt^2} = -3r^2 \cdot x \cdot \frac{d^2x}{dt^2} \cdot dr - \mu x dx$ , and by substituting for  $\frac{d^2x}{dt^2}$ , we have  $r^3 \cdot x \cdot \frac{d^3x}{dt^2} = 3 \frac{\mu \cdot x^2}{r} dr - \mu x dx$ ; ∴ the second member of the preceding equation =  $+3\mu \cdot \frac{(x^2 + y^2 + z^2)}{r^2} \cdot dr - \mu \cdot \frac{(xdx + ydy + zdz)}{r} - 3 \frac{\mu r^2}{r^3} (xdx + ydy + zdz)$ , hence the second member is reduced to  $-\mu \cdot dr$ , which combined with the member at the right hand side, gives the expression in the text.

† It is clear from an inspection of the equations (O') that the theorem already announced, is applicable to them, and to this last equation, since any one of these variables may be changed into the other without affecting the constant quantities, ∵  $\frac{dr}{dt} = \lambda \cdot \frac{dx}{dt}$

$$+ \gamma \cdot \frac{dy}{dt}.$$

gives an equation of the second degree, between either  $x$  and  $y$ ,  $x$  and  $z$ , or  $y$  and  $z$ , consequently the three projections of the curve described by  $m$ , about  $M$ , are lines of the second order, and therefore as all the points of this curve exist in the same plane, it is itself a line of the second order, or a conic section. It is easy to prove from the nature of this species of curves, that when the radius vector  $r$  is expressed by a linear function of the coordinates  $x, y$ ; the origin of the coordinates must be\* in the focus of the section. Now from the equation,  $r = \frac{h^2}{\mu} + \lambda \cdot x + \gamma \cdot y$ , we can obtain, in consequence of the equations (O),

$$0 = \frac{d^2r}{dt^2} + \mu \cdot \left\{ r - \frac{h^2}{\mu} \right\}$$

By multiplying this equation by  $dr$ , and then integrating, we shall obtain

$$r^2 \cdot \frac{dr^2}{dt^2} - 2\mu \cdot r + \frac{\mu r^2}{a'} + h^2 = 0, \dagger$$

$a'$  being a constant arbitrary quantity. From which may be obtained

$$dt = \frac{rdr}{\sqrt{\mu} \sqrt{2r - \frac{r^2}{a'} - \frac{h^2}{\mu}}};$$

this equation will give  $r$  in a function of  $t$ ; and as by what precedes,

\* It is a distinguishing property of the foci of conic sections, that if their equation be expressed by means of polar coordinates, these coordinates will be linear, when the origin is at the focus.

$$\dagger \frac{d^2r}{dt^2} = \lambda \cdot \frac{d^2x}{dt^2} + \gamma \cdot \frac{d^2y}{dt^2} = -\mu \cdot \left\{ \frac{\lambda x}{r^3} + \frac{\gamma y}{r^3} \right\} = -\mu \cdot \left\{ r - \frac{h^2}{\mu} \right\}. \text{ Multi-}$$

$$\text{plying by } dr, \text{ we obtain, } \frac{d^2rdr}{dt^2} = -\mu \cdot \frac{dr}{r^2} + h^2 \cdot \frac{dr}{r^3}; \text{ and by integrating } \frac{dr^2}{dt^2} \\ = \frac{2\mu}{r} - \frac{h^2}{r^2} + \frac{\mu}{a'}.$$

$x, y, z$ , are determined in functions of  $r$ ; we shall have the coordinates of  $m$ , in functions of the time.

18. We might arrive at these several equations, by the following method, which has this advantage, that it determines the arbitrary quantities in functions of the coordinates  $x, y, z$ , and of their first differences; which will be extremely useful in what follows.

Let us suppose that  $V = \text{constant}$ , is an integral of the first order of the equations (O),  $V$  being a function of  $x, y, z, \frac{dx}{dt}, \frac{dy}{dt}, \frac{dz}{dt}$ : Let  $x', y', z'$ , represent these three last quantities, and then the equation  $V = \text{constant}$ , will give by its differentiation,

$$0 = \left\{ \frac{dV}{dx} \right\} \cdot \frac{dx}{dt} + \left\{ \frac{dV}{dy} \right\} \cdot \frac{dy}{dt} + \left\{ \frac{dV}{dz} \right\} \cdot \frac{dz}{dt} \\ + \left\{ \frac{dV}{dx'} \right\} \cdot \frac{dx'}{dt} + \left\{ \frac{dV}{dy'} \right\} \cdot \frac{dy'}{dt} + \left\{ \frac{dV}{dz'} \right\} \cdot \frac{dz'}{dt}; *$$

but the equations (O) give

$$\frac{dx'}{dt} = -\frac{\mu x}{r^3}; \quad \frac{dy'}{dt} = -\frac{\mu y}{r^3}; \quad \frac{dz'}{dt} = -\frac{\mu z}{r^3};$$

consequently, we have the following identical equation, of partial differences,

$$0 = x' \cdot \left\{ \frac{dV}{dx} \right\} + y' \cdot \left\{ \frac{dV}{dy} \right\} + z' \cdot \left\{ \frac{dV}{dz} \right\} - \frac{\mu}{r}. \quad (I)$$

$$\left\{ x \cdot \left\{ \frac{dV}{dx'} \right\} + y \cdot \left\{ \frac{dV}{dy'} \right\} + z \cdot \left\{ \frac{dV}{dz'} \right\} \right\};$$

It is manifest, that every function of  $x, y, z, x', y', z'$ , which, substituted in place of ( $V$ ) in this equation, renders it identically nothing,

\* As  $V$  is in an immediate function of the six variables,  $x, y, z, x', y', z'$ , its differential coefficient with respect to another variable  $t$ , must be equal to the several differential coefficients of  $V$ , considered as a function of  $x, y, z, x', y', z'$ , multiplied respectively, into the differential coefficients of these variables, considered as a functions of  $t$ .

becomes, when it is put equal to a constant arbitrary quantity, an integral of the first order of the equations (O).

Let us suppose

$$V = U + U' + U'' + \text{ &c.}$$

$U$  being a function of the three variables  $x, y, z$ ;  $U'$  being a function of the six variables  $x, y, z, x', y', z'$ , but of the first order relatively to  $x', y', z'$ ;  $U''$  being a function of the same variables, and of the second order relatively to  $x', y', z'$ , and so of the rest. Substituting this value in the equation (I), and comparing separately, first, the terms in which  $x', y', z'$ , does not occur; secondly, those which involve the first power of these variables; thirdly, those which contain their squares, and their products, and so on of the rest; we shall have

$$0 = x \cdot \left\{ \frac{dU'}{dx'} \right\} + y \cdot \left\{ \frac{dU'}{dy'} \right\} + z \cdot \left\{ \frac{dU'}{dz'} \right\};$$

$$x' \cdot \left\{ \frac{dU}{dx} \right\} + y' \cdot \left\{ \frac{dU}{dy} \right\} + z' \cdot \left\{ \frac{dU}{dz} \right\} = \frac{\mu}{r^3}. \quad \left. \begin{array}{l} \\ \end{array} \right\}$$

$$\left( x \cdot \left\{ \frac{dU''}{dx'} \right\} + y \cdot \left\{ \frac{dU''}{dy'} \right\} + z \cdot \left\{ \frac{dU''}{dz'} \right\} \right);$$

$$x' \cdot \left\{ \frac{dU'}{dx} \right\} + y' \cdot \left\{ \frac{dU'}{dy} \right\} + z' \cdot \left\{ \frac{dU'}{dz} \right\} = \frac{\mu}{r^3}. \quad \left. \begin{array}{l} \\ \end{array} \right\}$$

$$\left( x \cdot \left\{ \frac{dU''}{dx'} \right\} + y \cdot \left\{ \frac{dU''}{dy'} \right\} + z \cdot \left\{ \frac{dU''}{dz'} \right\} \right);$$

$$x' \cdot \left\{ \frac{dU''}{dx} \right\} + y' \cdot \left\{ \frac{dU''}{dy} \right\} + z' \cdot \left\{ \frac{dU''}{dz} \right\} = \frac{\mu}{r^3}. \quad \left. \begin{array}{l} \\ \end{array} \right\}$$

$$\left( x \cdot \left\{ \frac{dU'''}{dx'} \right\} + y \cdot \left\{ \frac{dU'''}{dy'} \right\} + z \cdot \left\{ \frac{dU'''}{dz'} \right\} \right);$$

&c.

The integral of the first of these equations is, as we know by the theory of equations of partial differences,

$$U' = \text{func. } (xy' - yx', xz' - zx', yz' - zy', x, y, z).^*$$

As the value of  $U'$  must be linear with respect to  $x', y', z'$ , we shall suppose it of the following form :

$$U' = A.(xy' - yx') + B.(xz' - zx') + C.(yz' - zy');$$

$A, B, C$ , being constant arbitrary quantities. Let the value of  $V$  be continued as far as the term  $U''$ , so that  $U''', U''''$ , &c. may vanish ; the third of the equations (I') will become

$$0 = x'. \left\{ \frac{dU'}{dx} \right\} + y'. \left\{ \frac{dU'}{dy} \right\} + z'. \left\{ \frac{dU'}{dz} \right\}.$$

The preceding value of  $U'$  satisfies also this equation. The fourth of the equations (I') becomes

$$0 = x'. \left\{ \frac{dU''}{dx} \right\} + y'. \left\{ \frac{dU''}{dy} \right\} + z'. \left\{ \frac{dU''}{dz} \right\};$$

The integral of which equation, is

$$U'' = \text{funct. } (xy' - yx', xz' - zx', yz' - zy', x', y', z').†$$

This function ought to satisfy the second of the equations (I'), and

\* For the integration of this equation see Euler Integral Calculus, tome 3, chapter 3, No. , and Lacroix Traité Complete, Tom. 2, No. 634.

†  $F'$  being the derivative function of  $U'$ ,  $\frac{dU}{dx'} = -(y+z) \cdot F'$ ,  $\frac{dU'}{dy'} = (x-z) \cdot F'$ ;  $\frac{dU'}{dz'} = (x+y) \cdot F'$ ;  $\therefore x \cdot \frac{dU'}{dx'} + y \cdot \frac{dU'}{dy'} + z \cdot \frac{dU'}{dz'} = (-x(y+z) + y(x-z) + z(x+y)) \cdot F' = 0$ ;  $\frac{dU''}{dx} = (y'+z') \cdot F''$ ;  $\frac{dU''}{dy} = (z'-x') \cdot F''$ ;  $\frac{dU''}{dz} = -(x'+y') \cdot F''$ ;  $\therefore x \cdot \frac{dU''}{dx} + y \cdot \frac{dU''}{dy} + z \cdot \frac{dU''}{dz} = (x'(y'+z') + y'(z'-x') - z'(x'+y')) \cdot F'' = 0$ ; Multiplying the first member by  $dt$ , and substituting we obtain  $\left\{ \frac{dU}{dx} \cdot \frac{dx}{dt} + \frac{dU}{dy} \cdot \frac{dy}{dt} + \frac{dU}{dz} \cdot \frac{dz}{dt} \right\} \cdot d$   $= dU$ .

the first member of this equation multiplied by  $dt$ , is evidently equal to  $dU$ ; therefore the second member must be an exact differential of a function of  $x, y, z$ . But it is evident that we can satisfy at once this condition, the nature of the function  $U''$ , and the supposition that this function is of the second order in  $z', y', x'$ ; by making

$$U'' = (Dy' - Ex').(xy' - yx') + (Dz' - Fx').$$

$$(xz' - zx') + (Ez' - Fy').(yz' - zy') + G.(x'^2 + y'^2 + z'^2);$$

$D, E, F, G$ , being constant arbitrary quantities; and then  $r$  being equal to  $\sqrt{x^2 + y^2 + z^2}$ , we have

$$U = -\frac{\mu}{r} \cdot (Dx + Ey + Fz + 2G); *$$

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$$\begin{aligned} * \frac{dU''}{dx'} &= -D.(yy' + zz') + E.(2yx' - xy') + F.(2xz' - zx') + 2Gx', \\ \frac{dU''}{dy'} &= D.(2y'x - yx') - E.(xx' + zz') + F.(2zy' - yz') + 2Gy', \\ \frac{dU''}{dz'} &= D.(2z'x - zx') + E.(2yz' - zy') - F.(xx' + yy') + 2Gz'. \\ \therefore \frac{\mu}{r^3} \left\{ x \cdot \frac{dU''}{dx'} + y \cdot \frac{dU''}{dy'} + z \cdot \frac{dU''}{dz'} \right\} &= \\ &= -D.((xyy' + zxz') + E.(2yx.x' - x^2y') + F.(2zx.x' - x^2z' + 2Gxx')) \frac{\mu}{r^3} \\ &\quad + D.((2xyy' - y^2x') - E.(xyx' + zyz') + F.(2zy.y' - y^2z') + 2Gyy') \frac{\mu}{r^3} \\ &\quad + D.((2xzz' - z^2x') + E.(2yzz' - z^2y') - F.(xxx' + yzy') + 2Gzz') \frac{\mu}{r^3}, \end{aligned}$$

= by concinnating and omitting those terms which destroy each other, ( $-D.(y^2 + z^2)x' - E.(x^2 + z^2).y' - F.(x^2 + z^2)z' + D.(xy)y' + D.(xz)z' + E.(yx)x' + E.(yz)z' + F.(zx)x' + F.(zy)y'$   
 $y' + 2G.(xx' + yy' + zz')$ )  $\frac{\mu}{r^3}$  = (by observing that  $y^2 + z^2 = r^2 - x^2$ ;  $x^2 + z^2 = r^2 - y^2$   
&c.) the value of  $U$ , differenced with respect to  $x, y, z$ , successively, for

$$\frac{dU}{dx} = -\frac{\mu}{r} \cdot D + \frac{\mu}{r^3} \cdot Dx^2 + \frac{\mu}{r^3} (Eyx + Fzx + 2Gx) = -\frac{\mu}{r^3} \cdot (D.(y^2 + z^2) - Eyx -$$

consequently we can obtain, by this means, the values of  $U$ ,  $U'$ ,  $U''$ ; and the equation  $V = \text{constant}$ , will become

$$\begin{aligned} \text{const.} = & -\frac{\mu}{r} \cdot (Dx + Ey + Fz + 2G) + (A + Dy' - Ex') \cdot (xy' - yx')^* \\ & + (B + Dz' - Fx') \cdot (xz' - zx') + (C + Ez' - Fy') \cdot (yz' - zy') \\ & + G \cdot (x'^2 + y'^2 + z'^2). \end{aligned}$$

This equation satisfies the equation (I), and consequently the dif-

$$\begin{aligned} Fzx - 2Gx; \frac{dU}{dy} = & -\frac{\mu}{r} \cdot E + \frac{\mu}{r^3} \cdot Ey^2 + \frac{\mu}{r^3} \cdot (Dxy + Fyz + 2Gy) = -\frac{\mu}{r^3} \cdot (E(x^2 + z^2) - \\ & D \cdot xy - F \cdot yz - 2Gy), \quad \frac{dU}{dz} = -\frac{\mu}{r} \cdot F + \frac{\mu}{r^3} \cdot Fz^2 + \frac{\mu}{r^3} \cdot (Dxz + Eyz + 2Gz) = -\frac{\mu}{r^3} \cdot (F \\ & (x^2 + y^2) - Dxz - Eyz - 2Gz), \quad \therefore \text{if these equations be multiplied by } x', y', z', \text{ respectively, the sum of the terms at the left hand side will be equal to } dU, \text{ and the sum of those} \\ & \text{on the right hand, will coincide with those already given.} \end{aligned}$$

\* This equation evidently satisfies the equation (I), for

$$\begin{aligned} \frac{dV}{dx} = & -\frac{\mu}{r} \cdot D + \frac{\mu}{r^3} \cdot (Dx^2 + Exy + Fxz + 2Gx) + Ay' + Dy'^2 - Ex'y' + Bz' + Dz^2 - Fx'z' \\ \frac{dV}{dy} = & -\frac{\mu}{r} \cdot E + \frac{\mu}{r^3} \cdot (Ey^2 + Dxy + Fyz + 2Gy) - Ax' - Dy'x' + Ex'^2 + Cz' + Ez'^2 - Fy'z', \\ \frac{dV}{dz} = & -\frac{\mu}{r} \cdot F + \frac{\mu}{r^3} \cdot (Fz^2 + Dxz + Eyz + 2Gz) - Bx' - Dz'x' + Fx'^2 - Cy' - Ez'y' + Fy'^2. \\ \therefore x' \cdot \frac{dV}{dx} + y' \cdot \frac{dV}{dy} + z' \cdot \frac{dV}{dz} = & \\ & -\frac{\mu}{r^3} (D(y^2 + z^2) - Exy - Fxz - 2Gx)x' + Ay'x' + Dy'^2x' - Ex'^2y' + Bz'x + Dz'^2z' - Fx'^2z', \\ & -\frac{\mu}{r^3} (E(x^2 + z^2) - Dxy - Fyz - 2Gy)y' - Ay'x' - Dy'^2x' + Ex'^2y' + Cz'y' + Ez'^2y' - Fy'^2z', \\ & -\frac{\mu}{r^3} (F(x^2 + y^2) - Dxz - Eyz + 2Gz)z' - Bx'z' - Dz'^2z' + Fx'^2z' - Cy'z' - Ez'^2y' + Fy'^2z'. \\ = & \text{by obliterating the quantities which destroy each other} \\ & -\frac{\mu}{r^3} (D(y^2 + z^2) - Exy - Fxz - 2Gx)x' + E((x^2 + z^2) - Dxy - Fyz - 2Gy)y' + F(x^2 + y^2) - Dxz - Eyz \\ & - 2Gz)z'; \frac{dV}{dx'} = -E(xy' - yx') - y(A + Dy' - Ex') - F(xz' - zx') - z(B + Dz' - Fx') + 2Gx', \end{aligned}$$

ferential equations (O), whatever may be the arbitrary quantities  $A, B, C, D, E, F, G$ . Supposing them all to vanish first, with the exception of  $A$ ; 2dly, with the exception of  $B$ ; 3dly, with the exception of  $C$ , &c., and restoring  $\frac{dx}{dt}, \frac{dy}{dt}, \frac{dz}{dt}$ , in place of  $x', y', z'$ , we shall obtain the integrals

$$\left. \begin{aligned} c &= \frac{xdy - ydx}{dt}; \quad c' = \frac{xdz - zdx}{dt}; \quad c'' = \frac{ydz - zdy}{dt}; \\ 0 &= f + x \cdot \left\{ \frac{\mu}{r} - \frac{(dy^2 + dz^2)}{dt^2} \right\} + \frac{ydy \cdot dx}{dt^2} + \frac{zdz \cdot dx}{dt^2};^* \\ 0 &= f' + y \cdot \left\{ \frac{\mu}{r} - \frac{(dx^2 + dz^2)}{dt^2} \right\} + \frac{xdx \cdot dy}{dt^2} + \frac{zdz \cdot dy}{dt^2}; \\ 0 &= f'' + z \cdot \left\{ \frac{\mu}{r} - \frac{(dx^2 + dy^2)}{dt^2} \right\} + \frac{xdx \cdot dz}{dt^2} + \frac{ydy \cdot dz}{dt^2}; \\ 0 &= \frac{\mu}{a} - \frac{2\mu}{r} + \frac{dx^2 + dy^2 + dz^2}{dt^2}; \end{aligned} \right\} ; \quad (\text{P})$$

$c, c', c'', f, f', f''$ , and  $a$  being constant arbitrary quantities.

$$\frac{dV}{dy'} = D(xy' - yx') + x(A + Dy' - Ex') - F(yz' - zy') - z(C + Ez' - Fy') + 2Gy',$$

$$\frac{dV}{dz'} = D(xz' - zx') + x(B + Dz' - Fx') + E(yz' - zy') + y(C + Ez' - Fy') + 2Gz',$$

Multiplying these three equations by  $x, y, z$ , respectively, and observing that those terms, of which one factor is the product of two of the coordinates,  $x, y, z$ , destroy each other, we obtain, by concinnating  $\frac{dV}{dx'}x + \frac{dV}{dy'}y + \frac{dV}{dz'}z = -E(x^2 + z^2)y' - D(y^2 + z^2)x' - F(y^2 + x^2)z' + E(yx) + Fxz + 2Gx)x' + (Dxy + Fzy + 2Gy)y' + (Dxz + Eyz + 2Gz)z'$ , and this expression, when multiplied by  $\frac{\mu}{r^3}$  is identical with the preceding value of  $x' \frac{dV}{dx} + y' \frac{dV}{dy} + z' \frac{dV}{dz}$ .

\* Supposing all the constant quantities but  $A$  to vanish, the preceding equation becomes const.  $= A(xy' - yx')$ ; supposing them all except  $D$  to vanish, we shall have const.  $=$

The differential equations (O) can only have six\* distinct integrals of the first order, by means of which, if the differences  $dx, dy, dz$ , be eliminated, we shall obtain the three variables  $x, y, z$ , in functions of the time  $t$ ; therefore one at least of the seven preceding integrals should occur in the six others. We may perceive even, *a priori*, that two of these integrals must occur in the five remaining. In fact, as the sole element of the time, occurs in these integrals; they are not sufficient to determine the variables  $x, y, z$ , in functions of the time, and consequently they are inadequate to the complete determination of the motion of  $m$  about  $M$ . We proceed to examine how it happens that these integrals are only equivalent to five distinct integrals.

If we multiply the fourth of the equations (P) by  $\frac{zdy - ydz}{dt}$ , and

then add it to the fifth, multiplied by  $\frac{x dz - z dx}{dt}$ ; we shall obtain

$$0 = f \cdot \frac{(zdy - ydz)}{dt} + f' \cdot \frac{(xdz - zdx)}{dt} + z \cdot \frac{(xdy - ydx)}{dt}.$$

$$\left\{ \frac{\mu}{r} - \frac{(dx^2 + dy^2)}{dt^2} \right\} + \frac{(xdy - ydx)}{dt} \cdot \left\{ \frac{xdx.dz}{dt^2} + \frac{ydy.dz}{dt^2} \right\} +$$

$$-\frac{\mu}{r} \cdot Dx + Dy(xy' - yx') + Dz'(xz' - zx'); \because \frac{\text{const.}}{D} = f = -x \cdot \left( \frac{\mu}{r} + (y'^2 + z'^2) \right) -$$

$yy'.x' - zz'.x'$  which will be equal to the fourth of the equations (P), by substituting for  $x', y', z'$ , their values. Supposing  $G$  to be the only constant arbitrary quantity, we ob-

tain, const.  $= G \left( -\frac{2\mu}{r} + (x'^2 + y'^2 + z'^2) \right)$ ;  $\therefore$  making  $\frac{\text{const.}}{G} = \frac{\mu}{a}$ , and substituting

for  $x', y', z'$ , we obtain the expression given in the text.

\* As the differential equations (O) are of the second order, and since the complete integration of each equation furnishes two constant arbitrary quantities, the entire number cannot exceed six.

+ Performing this multiplication and addition, we obtain

$$-fc'' + \frac{\mu}{r} \frac{(xzdy - xydz) - xzdy^3 - xzdy.dz^2 + xydy^2.dz + xydz^3}{dt^3} + \frac{zydy^2.dx + z^2dx dy dz}{dt^3}$$

By substituting in place of  $\frac{xly-ydx}{dt}$ ,  $\frac{xdz-zdx}{dt}$ ,  $\frac{ydz-zdy}{dt}$  their values, which have been determined by the three first of the equations (P), we shall have

$$0 = \frac{f'c'-fc''}{c} + z \cdot \left\{ \frac{\mu}{r} - \left\{ \frac{dx^2+dy^2}{dt^2} \right\} \right\} + \frac{xdx.dz}{dt^2} + \frac{ydy.dz}{dt^2}.$$

This equation coincides with the sixth of the integrals (P), by making  $f'' = \frac{f'c'-fc''}{c}$ , or  $0 = fc'' - f'c' + f'c$ . Thus the sixth of the integrals (P), results from the five preceding, and the six arbitrary quantities  $c, c', c'', f, f', f''$ , are connected together by the preceding equation.

If we take the squares of the values of  $f, f', f''$ , which are determined by the equations (P), and then add them together, we shall obtain

$$\begin{aligned} & \frac{-y^2.dx dy dz - yz d^2 z dx}{dt^3} + f' c' + \frac{\mu}{r} \frac{(xy dz - yz dx)}{dt} - \frac{yx dx^2 dz - yx dz^3}{dt^3} \\ & + \frac{yz dx^3 + yz dz^2 dx}{dt^3} + \frac{x^2 dx dy dz + x z dz^2 dy}{dt^3} - \frac{xz dx^2 dy}{dt^3} - \frac{z^2 dx dy dz}{dt^3} = \text{by making} \\ & \text{ing factors to coalesce } -fc'' + f'c' + z \cdot \frac{\mu}{r} \frac{(xdy - ydx)}{dt} - z \cdot \frac{x dy}{dt} \frac{(dy^2 + dz^2)}{dt^2} + z \cdot \frac{xdy}{dt} \\ & \frac{(dz^2 - dx^2)}{dt^2} + \frac{xy dz}{dt} \frac{(dy^2 + dz^2)}{dt^2} - \frac{xy dz}{dt} \frac{(dx^2 + dz^2)}{dt^2} + z \cdot \frac{y dx}{dt} \frac{(dy^2 - dz^2)}{dt^2} + z \cdot \frac{y dx}{dt} \\ & \frac{(dz^2 + dx^2)}{dt} + (z^2 - y^2) \frac{dx dy dz}{dt^3} + (x^2 - z^2) \frac{dx dy dz}{dt^3} = \text{after all reductions, and obli-} \\ & \text{terating quantities which destroy each other, } f \frac{(zdy - ydz)}{dt} + f' \frac{(xdz - zdx)}{dt} + \frac{z\mu}{r}. \\ & \frac{(xdy - ydx)}{dt} - z \cdot \frac{(xdy - ydx)}{dt} \cdot \frac{(dx^2 + dy^2)}{dt^2} + \frac{xdy}{dt} \cdot \left\{ \frac{ydy dz}{dt^2} + \frac{xdx dz}{dt^2} \right\} - \frac{ydx}{dt}. \\ & \left\{ \frac{xdx dz}{dt^2} + \frac{ydy dz}{dt^2} \right\} \text{ which is the expression in the text.} \end{aligned}$$

$$l - \mu^2 = \left( r^2 \cdot \left\{ \frac{dx^2 + dy^2 + dz^2}{dt^2} \right\} - \left\{ \frac{r dr}{dt} \right\}^2 \right) \cdot \left\{ \frac{dx^2 + dy^2 + dz^2}{dt^2} - \frac{2\mu}{r} \right\}; ^*$$

in which  $l^2$  is, for the sake of abridging, put equal to  $f^2 + f'^2 + f''^2$ ; but if we take the square of the values of  $c, c', c''$ , which are given by the same equations, and then add them together, we shall have, by making  $c^2 + c'^2 + c''^2 = h^2$ ;

$$\begin{aligned} * f^2 &= \frac{\mu^2 x^2}{r^2} + x^2 \cdot \frac{(dy^2 + 2dy \cdot dz + dz^2)}{dt^2} - \frac{2\mu x^2}{r} \cdot \frac{(dy^2 + dz^2)}{dt^2} \\ &+ \frac{y^2 dy^2 \cdot dx^2 + z^2 dz^2 \cdot dx^2}{dt^4} + \frac{2yz \cdot dy \cdot dz \cdot dx^2}{dt^4} + \frac{2\mu x}{r} \cdot \frac{(ydy \cdot dx + zdz \cdot dx)}{dt^2} - 2x \cdot \frac{(dy^2 + dz^2)}{dt^2} \cdot \\ &\frac{(ydy \cdot dx + zdz \cdot dx)}{dt^4}; f'^2 = \frac{\mu^2 y^2}{r^2} + y^2 \cdot \frac{(dx^2 + 2dx \cdot dz + dz^2)}{dt^2} - \frac{2\mu y^2}{r} \cdot \frac{(dx^2 + dz^2)}{dt^2} \\ &+ \frac{x^2 dx^2 \cdot dy^2 + z^2 dz^2 \cdot dy^2}{dt^4} + \frac{2xz \cdot dx \cdot dz \cdot dy^2}{dt^4} + \frac{2\mu y}{r} \cdot \frac{(xdx \cdot dy + zdz \cdot dy)}{dt^2} - 2y \cdot \frac{(dx^2 + dz^2)}{dt^2} \cdot \\ &\frac{(xdx \cdot dy + zdz \cdot dy)}{dt^4}; f''^2 = \frac{\mu^2 z^2}{r^2} + z^2 \cdot \frac{(dx^2 + 2dx^2 \cdot dy^2 + dy^4)}{dt^4} - \frac{2\mu}{r} z^2 \cdot \frac{(dx^2 + dy^2)}{dt^2} \\ &+ \frac{x^2 dx^2 \cdot dz^2 + y^2 dy^2 \cdot dz^2}{dt^4} + \frac{2xy \cdot dx \cdot dy \cdot dz^2}{dt^4} + \frac{2\mu}{r} \cdot z \cdot \frac{(xdx \cdot dz + ydy \cdot dz)}{dt^2} - 2z \cdot \\ &\frac{(dx^2 + dy^2)}{dt^2} \cdot \frac{(xdx \cdot dz + ydy \cdot dz)}{dt^2}, \therefore \text{we obtain } f^2 + f'^2 + f''^2 - \mu^2 = \\ &x^2 \cdot \frac{(dy^2 + 2dy \cdot dz + dz^2)}{dt^4} + y^2 \cdot \frac{(dx^2 + 2dx \cdot dz + dz^2)}{dt^4} + z^2 \cdot \frac{(dx^2 + 2dx^2 \cdot dy^2 + dy^4)}{dt^4} \\ &- \frac{2\mu}{r} \cdot x^2 \cdot \frac{(dy^2 + dz^2)}{dt^2} - \frac{2my^2}{r} \cdot \frac{(dx^2 + dz^2)}{dt^2} - \frac{2\mu z^2}{r} \cdot \frac{(dx^2 + dy^2)}{dt^2} \\ &+ \frac{y^2 dy^2}{dt^4} \cdot \frac{(dx^2 + dz^2)}{dt^2} + \frac{x^2 dx^2}{dt^4} \cdot \frac{(dy^2 + dz^2)}{dt^2} + \frac{z^2 dz^2}{dt^4} \cdot \frac{(dy^2 + dx^2)}{dt^2} \\ &+ 2yz \cdot \frac{dy \cdot dz}{dt^2} \cdot \frac{dx^2}{dt^2} + \frac{2xz \cdot dx \cdot dz}{dt^2} \cdot \frac{dy^2}{dt^2} + \frac{2yz \cdot dy \cdot dz}{dt^2} \cdot \frac{dx^2}{dt^2} \cdot \\ &\frac{2\mu}{r} \left\{ xy \cdot \frac{dy \cdot dx}{dt^2} + xz \cdot \frac{dx \cdot dz}{dt^2} + xy \cdot \frac{dx \cdot dy}{dt^2} + yz \cdot \frac{dy \cdot dz}{dt^2} + xz \cdot \frac{dx \cdot dz}{dt^2} + yz \cdot \frac{dy \cdot dz}{dt^2} \right\} \end{aligned}$$

$$r^2 \cdot \left\{ \frac{dx^2 + dy^2 + dz^2}{dt^2} \right\} - \left\{ \frac{rdr}{dt} \right\}^2 = h^2, *$$

consequently, the preceding equation will become,

$$\begin{aligned} & \left\{ -2xy \cdot \frac{dx \cdot dy}{dt^2} - 2xz \cdot \frac{dx \cdot dz}{dt^2} \right\} \cdot \frac{dy^2 + dz^2}{dt^2} + \left\{ -2xy \cdot \frac{dy \cdot dx}{dt^2} - 2yz \cdot \frac{dy \cdot dz}{dt^2} \right\} \\ & \left\{ \frac{dx^2 + dz^2}{dt^2} \right\} + \left\{ -2xz \cdot \frac{dx \cdot dz}{dt^2} - 2yz \cdot \frac{dy \cdot dz}{dt^2} \right\} \cdot \left\{ \frac{dx^2 + dy^2}{dt^2} \right\} = \\ & (x^2 + y^2 + z^2) \cdot \frac{(dx^4 + dy^4 + dz^4 + 2dx^2 \cdot dy^2 + 2dx^2 \cdot dz^2 + 2dy^2 \cdot dz^2)}{dt^4} - x^2. \\ & \frac{(dx^4 + 2dx^2 \cdot dy^2 + 2dx^2 \cdot dz^2)}{dt^4} - y^2 \cdot \frac{(dy^4 + 2dx^2 \cdot dy^2 + 2dy^2 \cdot dz^2)}{dt^4} - z^2 \cdot \frac{(dz^4 + 2dz^2 \cdot dx^2 + 2dz^2 \cdot dy^2)}{dt^4} \\ & - (2xy \cdot \frac{dx \cdot dy}{dt^2} + 2z \cdot \frac{dx \cdot dz}{dt^2} + 2yz \cdot \frac{dy \cdot dz}{dt^2}) \cdot \frac{(dx^2 + dy^2 + dz^2)}{dt^2} - \frac{2\mu}{r} \cdot (x^2 + y^2 + z^2). \\ & \frac{(dx^2 + dy^2 + dz^2)}{dt^2} + \frac{2\mu}{r} \cdot \frac{(x^2 \cdot dt^2 + y^2 \cdot dy^2 + z^2 \cdot dz^2 + 2xy \cdot dx \cdot dy)}{dt^4} + \frac{(2xz \cdot dx \cdot dz + 2yz \cdot dy \cdot dz)}{dt^2} \\ & + y^2 \cdot \frac{(dy^2 \cdot dx^2 + dy^2 \cdot dz^2)}{dt^4} + x^2 \cdot \frac{(dx^2 \cdot dy^2 + dx^2 \cdot dz^2)}{dt^4} + z^2 \cdot \frac{(dz^2 \cdot dy^2 + dz^2 \cdot dx^2)}{dt^4} = (\text{by ob-}) \end{aligned}$$

literating the quantities which destroy each other, and observing that  $r^2 dr^2 = x^2 dx^2 + y^2 dy^2 + z^2 dz^2 + 2xy \cdot dx \cdot dy + 2xz \cdot dx \cdot dz + 2yz \cdot dy \cdot dz$

$$\begin{aligned} & r^2 \cdot \frac{(dx^4 + dy^4 + dz^4 + 2dx^2 \cdot dy^2 + 2dx^2 \cdot dz^2 + 2dy^2 \cdot dz^2)}{dt^4} (-x^2 \cdot dx^4 - y^2 \cdot dy^4 - z^2 \cdot dz^4 \\ & \left\{ -\frac{2xy \cdot dx \cdot dy}{dt^2} - \frac{2xz \cdot dz \cdot dy}{dt^2} - \frac{2yz \cdot dy \cdot dz}{dt^2} \right\} \cdot \left\{ \frac{dx^2 + dy^2 + dz^2}{dt^2} \right\} - \frac{2\mu}{r} \cdot r^2 + \frac{2\mu}{r}. \\ & \left\{ \frac{rdr}{dt^2} \right\}^2, \text{ which may be evidently reduced to the expression in the text.} \end{aligned}$$

\* Squaring these equations and then adding them together, gives

$$\begin{aligned} & x^2 \cdot \frac{(dy^2 + dz^2)}{dt^2} + y^2 \cdot \frac{(dx^2 + dz^2)}{dt^2} + z^2 \cdot \frac{(dx^2 + dy^2)}{dt^2} - \frac{2xy \cdot dy \cdot dx}{dt^2} - \frac{2xz \cdot dx \cdot dz}{dt^2} \\ & - \frac{2yz \cdot dy \cdot dz}{dt^2} = (x^2 + y^2 + z^2) \cdot \frac{(dx^2 + dy^2 + dz^2)}{dt^2} - \frac{x^2 \cdot dx^2}{dt^2} - \frac{y^2 \cdot dy^2}{dt^2} - \frac{z^2 \cdot dz^2}{dt^2} \\ & - \frac{2xy \cdot dx \cdot dy}{dt^2} - \frac{2xz \cdot dx \cdot dz}{dt^2} - \frac{2yz \cdot dy \cdot dz}{dt^2} = r^2 \cdot \frac{(dx^2 + dy^2 + dz^2)}{dt^4} - \left\{ \frac{rdr}{dt} \right\}^2. \end{aligned}$$

$$0 = \frac{dx^2 + dy^2 + dz^2}{dt^2} - \frac{2\mu}{r} + \frac{\mu^2 - l^2}{h^2}.$$

The comparison of this equation, with the last of the equations (P), will give the following equation of condition

$$\frac{\mu^2 - l^2}{h^2} = \frac{\mu}{a}.$$

Therefore it follows, that the last of the equations (P), occurs in the six first, which are themselves only equivalent to five distinct integrals, the seven arbitrary quantities  $c, c', c'', f, f', f''$ , and  $a$  being connected by the two preceding equations of condition. From hence it results, that we shall obtain the most general expression for  $V$ , which satisfies the equation (I), by assuming for this expression, an arbitrary function of the values of  $c, c', c'', f$ , and  $f'$ , which are determined by the five first of the equations (P).

19. Although these integrals are inadequate to the determination of  $x, y, z$ , in functions of the time, they nevertheless determine the species of the curve described by  $m$ , about  $M$ . In fact, if we multiply the first of the equations (P), by  $z$ , the second by  $-y$ , and the third by  $x$ , we shall obtain, by their addition,

$$0 = cz - c'y + c''x, *$$

which is the equation of a plane, of which the position depends on the constant quantities  $c, c', c''$ .

If we multiply the fourth of the equations (P) by  $x$ ; the fifth by  $y$ , and the sixth by  $z$ , we shall obtain

\* Performing this multiplication the members at the right hand side of the equation will disappear, for they become

$$cz - c'y + c''x = \frac{xz \cdot dy - yz \cdot dx}{dt} - \frac{xy \cdot dz + zy \cdot dx}{dt} + \frac{yx \cdot dz - zx \cdot dy}{dt} = 0.$$

$$0 = fx + f'y + f''z + \mu r - r^2 \cdot \frac{(dx^2 + dy^2 + dz^2)}{dt^2} + \frac{r^2 \cdot dr^2}{dt^2};$$

but by the preceding number we have,

$$r^2 \cdot \frac{(dx^2 + dy^2 + dz^2)}{dt^2} - \frac{r^2 \cdot dr^2}{dt^2} = h^2;$$

consequently,

$$0 = \mu r - h^2 + fx + f'y + f''z.$$

This equation, combined with the following, namely,

$$0 = c''x - c'y + cz; \quad r^2 = x^2 + y^2 + z^2;$$

gives the equation of conic sections, the origin of  $r$  being at the focus. From this it follows,\* that the planets and the comets describe very nearly conic sections about the sun, this star existing in one of the foci, and these stars move in such a manner, that the areas described by the radii vectores, increase proportionally to the time. In fact, if  $dv$  re-

R 2

\* Performing this multiplication, and then adding the products together, we obtain

$$\begin{aligned} fx + f'y + f''z &= \frac{\mu}{r} \cdot (x^2 + y^2 + z^2) - (x^2 + y^2 + z^2) \frac{(dx^2 + dy^2 + dz^2)}{dt^2} + x^2 \cdot \frac{dx^2}{dt^2} \\ &+ y^2 \cdot \frac{dy^2}{dt^2} + z^2 \cdot \frac{dz^2}{dt^2} + 2xy \cdot \frac{dx \cdot dy}{dt} + 2xz \cdot \frac{dx}{dt} \cdot \frac{dz}{dt} + 2yz \cdot \frac{dy}{dt} \cdot \frac{dz}{dt} = \mu r - r^2. \\ \frac{(dx^2 + dy^2 + dz^2)}{dt^2} + r^2 \cdot \frac{dr^2}{dt^2}. \end{aligned}$$

From the first of these equations we obtain

$\mu^2 \cdot r^2 = h^4 - 2h^2 \cdot (fx + f'y + f''z) + f^2 x^2 + f^2 y^2 + f''^2 z^2 + 2ff' \cdot xy + 2ff'' \cdot xz + 2f'f'' \cdot yz$ , and by means of the equation  $0 = c''x - c'y + cz$ , and  $r^2 = x^2 + y^2 + z^2$ , we can eliminate,  $z^2$  and  $z$ , and then substituting for  $r^2$  its value, we arrive at an equation of the second degree between  $y$  and  $x$ , by similar process we obtain equations of the second degree between  $x$  and  $z$ ,  $y$  and  $z$ , from which it follows that the curve described is a conic section; and as the value of  $r$  is given in a linear function of the coordinates  $x$ ,  $y$ ,  $z$ , the origin must be at the focus.

represents the indefinitely small angle, intercepted between the radii  $r$  and  $r+dr$ , we shall have

$$dx^2 + dy^2 + dz^2 = r^2 dv^2 + dr^2 ; *$$

the equation

$$r^2 \cdot \frac{(dx^2 + dy^2 + dz^2)}{dt^2} - \frac{r^2 dv^2}{dt^2} = h^2.$$

will consequently become,  $r^4 dv^2 = h^2 dt^2$ ; therefore

$$dv = \frac{h dt}{r^2}.$$

From this it appears that the elementary area  $\frac{1}{2}rdv$ , described by the radius vector  $r$ , is proportional to the element of time  $dt$ , consequently the area described in a finite time, is proportional to this time. It also appears, that the angular motion of  $m$  about  $M$ , is at each point of the orbit, inversely proportional to the square of the radius vector; and as we can, without sensible error, assume very short intervals of time, for the indefinitely small moments; by means of the preceding

\* The differential of the curve  $= ds = \sqrt{dx^2 + dy^2 + dz^2}$  = the hypotenuse of a right angle triangle, of which one side  $= dr$ , and the other side about the right angle  $= rdv$ ,  $\therefore dx^2 + dy^2 + dz^2 = ds^2 = dr^2 + r^2 dv^2$ .

As  $h$  varies as the square root of the parameter, it follows that the angular velocity  $\frac{dv}{dt}$  varies as the square root of the synchronous areas divided by the square of the distance, see page 10; hence the angular velocity in a conic section is to that in a circle at the same distance  $r$ , as  $h :: \sqrt{r}$ ;  $\therefore$  they are equal at the extremity of the focal ordinate; substituting for  $h$  its value  $\frac{2\pi a^2 \sqrt{1-e^2}}{T}$ ;  $\frac{dv}{dt}$  will be  $\frac{2\pi a^2 \sqrt{1-e^2}}{T r^2}$ ; if a body describes a circle at the unity of distance in a time equal to  $T$ , then the angular velocity in the circle  $= \frac{2\pi}{T}$  = the mean angular velocity in the ellipse, consequently, when the angular velocity in the ellipse is equal to the mean angular velocity, we have  $\frac{2\pi}{T} = \frac{2\pi a^2 \sqrt{1-e^2}}{T r^2}$ , and  $\therefore r = a(1-e^2)^{\frac{1}{2}}$ , = a mean proportional between the semiaxes; in this position the equation of the centre is a maximum.

equation, we can obtain the horary motions of the planets and comets in different parts of their orbits.

The elements of the conic section described by  $m$ , are the constant arbitrary quantities of its motion; they are consequently functions of the preceding arbitrary quantities  $c, c', c'', f, f', f''$ , and  $\frac{\mu}{a}$ ; we now proceed to determine these functions. Let  $\theta$  represent the angle which the intersection of the plane of the orbit with the plane of  $x$  and of  $y$ , constitutes with the axis of  $x$ , which intersection is termed the *line of the nodes*; let  $\phi$  be the mutual inclination of these two planes. If  $x'$  and  $y'$  represent the coordinates of  $m$ , referred to the line of the nodes, as axis of the abscissæ; we shall have

$$\begin{aligned}x' &= x \cdot \cos. \theta + y \cdot \sin. \theta; \\y' &= y \cdot \cos. \theta - x \cdot \sin. \theta.\end{aligned}$$

We have also

$$z = y' \cdot \tan. \phi;$$

consequently we shall have

$$z = y \cdot \cos. \theta \cdot \tan. \phi - x \cdot \sin. \theta \cdot \tan. \phi.$$

The comparison of this equation with the following,

$$0 = c''x - c'y + cz;$$

will give

$$\begin{aligned}c' &= c \cdot \cos. \theta \cdot \tan. \phi; \\c'' &= c \cdot \sin. \theta \cdot \tan. \phi;^*\end{aligned}$$

from which may be obtained

\* A comparison of these equations, gives  $y \cdot \cos. \theta \cdot \tan. \phi - x \cdot \sin. \theta \cdot \tan. \phi = \frac{c'}{c}$ .

$y - \frac{c''}{c} \cdot x \because \frac{c'}{c} = \cos. \theta \cdot \tan. \phi; \frac{c''}{c} = \sin. \theta \cdot \tan. \phi, \therefore \frac{c'^2 + c''^2}{c^2} = \tan. {}^2 \phi.$

See page 3, and page 34 of 1st Book.

$$\tan. \theta = \frac{c''}{c'} ;$$

$$\tan. \phi = \frac{\sqrt{c'^2 + c''^2}}{c}.$$

By means of the preceding equations, the positions of the nodes, and the inclination of the orbit are determined in functions of the constant arbitrary quantities,  $c, c', c''$ . At the perihelium, we have

$$rdr = 0; \text{ or } xdx + ydy + zdz = 0;$$

let therefore  $X, Y, Z$ , represent the coordinates of the planet at this point; and from the fourth and fifth of the equations (P), of the preceding No. may be obtained,

$$\frac{Y}{X} = \frac{f'}{f}. *$$

But if we name  $I$  the longitude of the projection of the perihelium, on the plane of  $x$  and of  $y$ , this longitude being reckoned from the axis of  $x$ , we have

$$\frac{Y}{X} = \tan. I;$$

consequently,

$$\tan. I = \frac{f'}{f};$$

this equation determines the position of the axis major of the conic section.

\* Substituting  $-xdx$  for  $ydy + zdz$ , and  $-ydy$  for  $xdx + zdz$  in the two last terms of the second member of this equation, and they will become

$$0 = f + X \left\{ \frac{\mu}{r} - \frac{(dX^2 + dY^2 + dZ^2)}{dt^2} \right\}; 0 = f + Y \left\{ \frac{\mu}{r} - \frac{dX^2 + dY^2 + dZ^2}{dt^2} \right\}$$

∴ multiplying the first by  $Y$ , and the second by  $X$ , and then subtracting, we obtain the expression given in the text.

If by means of the last of the equations (P),  $\frac{dx^2 + dy^2 + dz^2}{dt^2}$  be eliminated from the equation  $r^2 \cdot \frac{(dx^2 + dy^2 + dz^2)}{dt^2} - \frac{r^2 dr^2}{dt^2} = h^2$ , we shall obtain

$$2\mu r - \frac{\mu r^2}{a} - \frac{r^2 dr^2}{dt^2} = h^2;$$

but  $dr$  vanishes at the extremities of the greater axis; therefore at these points we have,

$$0 = r^2 - 2ar + \frac{ah^2}{\mu}.$$

The sum of the two values of  $r$  in this equation, is the axis major of the conic section, and their difference is equal to twice the eccentricity; thus,  $a$  is the semiaxis\* major of the orbit, or the mean distance of  $m$  from  $M$ ; and  $\sqrt{1 - \frac{h^2}{\mu a}}$  is the ratio of the eccentricity to the semi-axis major. Let  $e$  represent this ratio; and by the pre-

\* The coefficient of  $r$  with its sign changed is the sum of the two values of  $r$ , and their difference is equal to twice the radical, and ∵ = to  $2a\sqrt{1 - \frac{h^2}{\mu a}}$ , and

$\sqrt{1 - \frac{h^2}{\mu a}}$  is the ratio of the eccentricity to  $a$ ;  $\sqrt{a^2 - \frac{\mu h^2}{a}} = \mu e$   
 $\mu \cdot \left\{ \mu - \frac{h^2}{a} \right\} = \mu^2 e^2 = l^2$ ;  $dr = ae \cdot \sin. u du$ , ∵  $rdr = a^2 e \cdot \sin. u du \cdot (1 - e \cos. u)$ ,  
 $2r - \frac{r^2}{a} = a((2 - 2e \cos. u) - (1 + e^2 \cos. ^2 u - 2e \cos. u)) = a(1 - e^2 \cos. u)$ , and  
 $\therefore 2r - \frac{r^2}{a} - a(1 - e^2) = ae^2(1 - \cos. ^2 u) = ae^2 \cdot \sin. ^2 u$ , and therefore

$$\frac{rdr}{\sqrt{\mu} \cdot \sqrt{2r - \frac{r^2}{a} - a(1 - e^2)}} (=dt) = \frac{a^2 e \cdot \sin. u \cdot (1 - e \cos. u) du}{\sqrt{\mu} \cdot \sqrt{ae^2 \cdot \sin. ^2 u}} = \frac{a^{\frac{3}{2}}}{\sqrt{\mu}}.$$

ceding number, we have

$$\frac{\mu}{a} = \frac{\mu^2 - l^2}{h^2};$$

therefore  $\mu e = l$ . Thus, we can know all the elements which determine the nature of the conic section, and its position in space.

20. The three finite equations found in the preceding number, between  $x, y, z$ , and  $r$ , give  $x, y, z$ , in functions of  $r$ ; thus, in order to determine these coordinates in a function of the time, it is sufficient to have the radius vector  $r$ , in a similar function, which requires a new integration. For this purpose, let us resume the equation

$$2\mu r - \frac{\mu r^2}{a} - \frac{r^2 \cdot dr^2}{dt^2} = h^2;$$

by the preceding number, we have,

$$h^2 = \frac{a}{\mu} \cdot (\mu^2 - l^2) = a\mu \cdot (1 - e^2);$$

therefore we shall obtain

$$dt = \frac{rdr}{\sqrt{\mu} \cdot \sqrt{2r - \frac{r^2}{a} - a \cdot (1 - e^2)}}.$$

In order to integrate this equation, let  $r = a(1 - e \cos. u)$ , we shall have

$$dt = \frac{a^{\frac{3}{2}} \cdot du}{\sqrt{\mu}} \cdot (1 - e \cos. u).$$

from which may be obtained by integrating,

$$t + T = \frac{a^{\frac{3}{2}}}{\sqrt{\mu}} \cdot (u - e \sin. u); \quad (S)$$

$T$  being a constant arbitrary quantity. This equation determines  $u$ ,

and consequently  $r$  in a function of  $t$ ; and as  $x, y, z$  are determined in functions of  $r$ ; we shall obtain the values of these coordinates, for any instant whatever.

We have thus completely integrated the differential equations (O) of No. 17; this integration introduces the six arbitrary quantities  $a, e, I, \theta, \varphi$ , and  $T$ : the two first depend on the nature of the orbit; the three following depend on its position in space; and the last is relative to the position of the body  $m$ , at a determined period, or, what comes to the same thing, it depends on the instant of its transit through the perihelium.

Let us refer the coordinates of the body  $m$ , to other coordinates which are more convenient for the usages of astronomy, and for this purpose, let  $v$  represent the angle which the radius vector  $r$  makes with the greater axis, reckoning from the perihelium; the equation of the ellipse will be

$$r = \frac{a.(1-e^2)}{1+e.\cos.v}.$$

The equation  $r=a.(1-e\cos.u)$ , of the preceding number, indicates that  $u$  vanishes at the perihelium, so that this point is the origin of the two angles  $u$  and  $v$ ; it is easy to shew, that the angle  $u$  is formed by the greater axis of the orbit, and by the radius drawn from its centre, to the point where the circumference described on the greater axis as diameter, meets the ordinate drawn from the body  $m$ , perpendicular to the greater axis. This angle is termed the *excentric anomaly*, and the angle  $v$  is the *true anomaly*. A comparison of the two values of  $r$ , gives

$$1 - e.\cos.u = \frac{1-e^2}{1+e.\cos.v};$$

from which may be obtained

$$\tan \frac{1}{2}v = \sqrt{\frac{1+e}{1-e}} \cdot \tan \frac{1}{2}u. *$$

If the origin of the time  $t$  be fixed at the very moment of the passage, through the perihelium,  $T$  will vanish; and by making, in order to abridge,  $\frac{\sqrt{\mu}}{a^{\frac{3}{2}}} = n$ , we shall have,  $nt = u - e \cdot \sin u.$

By collecting together the equations of the motion of  $m$ , about  $M$ , we shall have

$$\left. \begin{array}{l} nt = u - e \cdot \sin u, \\ r = a(1 - e \cdot \cos u) \\ \tan \frac{1}{2}v = \sqrt{\frac{1+e}{1-e}} \cdot \tan \frac{1}{2}u. \end{array} \right\}; \quad (f)$$

the angle  $nt$  being what is termed the *mean anomaly*. The first of these equations determines  $u$  in a function of the time  $t$ , and the two remaining equations will give  $r$  and  $v$ , when  $u$  shall be determined. The equation between  $u$  and  $v$  is transcendental, and can only be resolved by approximation. Fortunately, from the circumstances of the celestial motions, the approximation is very rapid. In fact, the orbits of the celestial bodies are either almost circular, or extremely ex-

$$* \quad \frac{\sin a + \sin b}{\cos a + \cos b} = \tan \frac{(a+b)}{2}, \text{ let } b = 0, \text{ and } \frac{\sin a}{1 + \cos a} = \tan \frac{a}{2}, \text{ i.e.}$$

$$\frac{\sqrt{1 - \cos^2 a}}{1 + \cos a} = \frac{\sqrt{1 - \cos a}}{\sqrt{1 + \cos a}} = \tan \frac{a}{2}; \text{ now } e \cdot \cos u = e \frac{(e + \cos v)}{1 + e \cdot \cos v}, \text{ and } \cos v =$$

$$\frac{\cos u - e}{1 - e \cdot \cos u}; \because \tan \frac{v}{2} = \frac{\sqrt{1 - \cos v}}{\sqrt{1 + \cos v}} = \frac{\sqrt{1 + \frac{e - \cos u}{1 - e \cdot \cos u}}}{\sqrt{1 + \frac{-e + \cos u}{1 - e \cdot \cos u}}} =$$

$$\frac{\sqrt{1 + e - e \cdot \cos u - \cos u}}{\sqrt{1 - e - e \cdot \cos u + \cos u}} = \frac{\sqrt{(1+e)(1-\cos u)}}{\sqrt{(1-e)(1+\cos u)}} = \frac{\sqrt{1+e}}{\sqrt{1-e}} \tan \frac{u}{2}.$$

centric, and in these two cases, we can determine  $u$  in terms of  $t$ , by very convergent formulæ, which we proceed to develope. We shall give for this purpose, some general theorems on the reduction of functions into series, which will be extremely useful in the sequel.

21. Let  $u$  be any function of  $\alpha$ , which it is required to expand into a series proceeding according to the powers  $\alpha$ ; this series being supposed to be represented by

$$u = u + \alpha q_1 + \alpha^2 q_2 + \alpha^3 q_3, \dots + \alpha^n q_n + \alpha^{n+1} q_{n+1} + \text{ &c.}$$

$u$ ,  $q_1$ ,  $q_2$ , &c., being quantities independent of  $\alpha$ ; it is evident that  $u$  is what  $u$  becomes, when  $\alpha$  is supposed to be equal to cypher, and that, whatever be the value of  $n$ ,

$$\left\{ \frac{d^n u}{d\alpha^n} \right\} = 1.2.3\dots n.q_n + 2.3\dots (n+1).q_{n+1} + \text{ &c.}$$

the difference  $\left\{ \frac{d^n u}{d\alpha^n} \right\}$ , being taken on the hypothesis, that in  $u$  every thing is made to vary which ought to vary with  $\alpha$ . Consequently, if we suppose that after the differentiations,  $\alpha=0$ , in the expression of  $\left\{ \frac{d^n u}{d\alpha^n} \right\}$ ; we shall have

$$q_n = \frac{\left\{ \frac{d^n u}{d\alpha^n} \right\}}{1.2.3\dots n}$$

If  $u$  is a function of the two quantities  $\alpha$  and  $\alpha'$ , and it is proposed to expand it into a series, proceeding according to the powers and products of  $\alpha$  and  $\alpha'$ ; this series being represented by

$$\begin{aligned} u = & u + \alpha \cdot q_{1,0} + \alpha^2 \cdot q_{2,0} + \text{ &c.} \\ & + \alpha' \cdot q_{0,1} + \alpha \alpha' \cdot q_{1,1} + \text{ &c.} \\ & + \alpha'^2 \cdot q_{0,2} + \text{ &c.} \end{aligned}$$

the coefficient  $q_{n,n}$  of the product  $\alpha^n \cdot \alpha'^n$ , will be in like manner equal to

$$\frac{\left\{ \frac{d^{n+n'} u}{d\alpha^n \cdot d\alpha'^{n'}} \right\}}{1.2.3.\dots.n.1.2.3.\dots.n'};$$

$\alpha$  and  $\alpha'$  being supposed to vanish after the differentiations.

In general, if  $u$  is a function of  $\alpha, \alpha', \alpha'', \&c.$ , and if it is proposed to expand  $u$  into a series, ranged according to the powers and products of  $\alpha, \alpha', \alpha'', \&c.$ , the term of this series, of which the factor is the product  $\alpha^n \cdot \alpha'^{n'} \cdot \alpha''^{n''} \dots$  will be  $\alpha^n! \cdot \alpha'^{n'}! \cdot \alpha''^{n''}! \dots q_{n,n',n''}.$  we shall have

$$q_{n,n',n''} = \frac{\left\{ \frac{d^{n+n'+n''+\&c.} u}{d\alpha^n \cdot d\alpha'^{n'} \cdot d\alpha''^{n''} \&c.} \right\}}{1.2.3.\dots.n.1.2.3.\dots.n.1.2.3.\dots.n'. \&c.};$$

provided  $\alpha, \alpha', \alpha'', \&c.$ , are supposed to vanish after the differentiations.

Let us now suppose that  $u$  is a function of  $\alpha, \alpha', \alpha'', \&c.$ , and of the variables  $t, t', t'', \&c.$ ; if by the nature of this function, or by an equation of partial differences which represents it, we have obtained

$$\left\{ \frac{d^{n+n'+n''+\&c.} u}{d\alpha^n \cdot d\alpha'^{n'} \cdot d\alpha''^{n''} \&c.} \right\}$$

in a function of  $u$  and of its differences, taken with respect to  $t, t', \&c.$ ;  $F$  representing this function, when  $u$  is changed into  $u$ ,  $u$  being what  $u$  becomes when  $\alpha, \alpha', \&c.$  vanish, it is manifest that we shall obtain  $q_{n,n}, \&c.$  by dividing  $F$  by the product  $1.2.3\dots n.1.2.3\dots n', \&c.$ ; therefore we shall obtain the law of the series according to which  $u$  is expanded.

In the next place, let  $u$  be equal to any function of  $t+\alpha, t'+\alpha', t''+\alpha'', \&c.$ , which we will represent by  $\phi(t+\alpha, t'+\alpha', t''+\alpha'')$ , in this

case the  $m^{\text{th}}$  difference of  $u$ , taken with respect to  $\alpha$ , and divided by  $d\alpha^m$ , is evidently equal to this same difference, taken with respect to  $t$ , and divided by  $dt_m$ . The same equality obtains between the differences taken relatively to  $\alpha'$  and  $t'$ , or relatively to  $\alpha''$  and  $t''$ , &c.; hence it follows, that in general, we have

$$\left\{ \frac{d^{n+n'+n''+\&c.} u}{d\alpha^n \cdot d\alpha'^{n'} \cdot d\alpha''^{n''} \cdot \&c.} \right\} = \left\{ \frac{d^{n+n'+n''+\&c.} u}{dt^n \cdot dt'^{n'} \cdot dt''^{n''}} \right\}.$$

If in the second member of this equation,  $u$  be changed into  $u$ , that is, into  $\phi(t, t', t'', \&c.)$ ; we shall have, by what precedes,

$$q_{n, n', n''}, \&c. = \left\{ \frac{d^{n+n'+n''+\&c.} \cdot \phi(t, t', t'', \&c.)}{1.2.3...n. 1.2.3...n'. 1.2.3...n''. \&c.} \right\}$$

If  $u$  is a function of  $t$  and  $\alpha$ , only, we shall have

$$q_n = \frac{d^n \cdot \phi(t)}{1.2.3...n \cdot dt^n},$$

therefore

$$\phi(t+\alpha) = \phi(t) + \frac{\alpha \cdot d \cdot \phi(t)}{dt} + \frac{\alpha^2}{1.2} \cdot \frac{d \cdot \phi(t)}{dt^2} + \frac{\alpha^3}{1.2.3} \cdot \frac{d^3 \phi(t)}{dt^3} + \&c. \quad (i)$$

Let us in the next place suppose that  $u$ , instead of being given immediately in  $\alpha$  and  $t$ , as in the preceding case, is a function of  $x$ ,  $x$  being given by the equation of partial differences,  $\left\{ \frac{dx}{d\alpha} \right\} = z \cdot \left\{ \frac{dx}{dt} \right\}$ , in which  $z$  is any function whatever of  $x$ .

In order to reduce  $u$  into a series proceeding according to the powers of  $\alpha$ , the value of  $\left\{ \frac{d^n u}{d\alpha^n} \right\}$  must be determined in the case in which  $\alpha=0$ ; but in consequence of the proposed equation of partial differences, we have

$$\left\{ \frac{du}{d\alpha} \right\} = \left\{ \frac{du}{dx} \right\} \cdot \left\{ \frac{dx}{d\alpha} \right\} = z \cdot \left\{ \frac{du}{dx} \right\} \cdot \left\{ \frac{dx}{dt} \right\};$$

therefore, we shall have

$$\left\{ \frac{du}{d\alpha} \right\} = \frac{d \int z \cdot du}{dt}, * \quad (k)$$

This equation being differenced with respect to  $\alpha$ , gives

$$\left\{ \frac{d^2 u}{d\alpha^2} \right\} = \frac{d^2 \int z \cdot du}{d\alpha \cdot dt}, †$$

but the equation (k) gives, by changing  $u$  into  $\int z \cdot du$ ,

$$\left\{ \frac{d \int z \cdot du}{d\alpha} \right\} = \left\{ \frac{d \int z^2 \cdot du}{dt} \right\};$$

consequently

$$\left\{ \frac{d^2 u}{d\alpha^2} \right\} = \frac{d^2 \int z^2 \cdot du}{dt^2}.$$

This equation being differenced again with respect to  $\alpha$ , gives

$$\left\{ \frac{d^3 u}{d\alpha^3} \right\} = \frac{d^3 \int z^2 \cdot du}{d\alpha \cdot dt^2};$$

but the equation (k) gives, by changing  $u$  into  $\int z^2 \cdot du$

$$\left\{ \frac{d \int z \cdot du}{d\alpha} \right\} = \left\{ \frac{d \int z^3 \cdot du}{dt} \right\};$$

\* Let  $\int z \cdot du = u'$ , then  $\frac{du'}{d\alpha} = \frac{du'}{dx} \cdot \frac{dx}{d\alpha} = z \cdot \frac{du'}{dx} \cdot \frac{dx}{dt} = z \cdot \frac{du'}{dt} = d \cdot \frac{\int z \cdot du'}{dt}$  and by

substituting for  $du'$  its value, we obtain  $\frac{d \int z \cdot du}{d\alpha} = \frac{d \int z^2 \cdot du}{dt}$ .

† As the characteristic  $\int$  indicates an operation, the reverse of that denoted by  $d$ , we can remove the sign  $\int$ , by depressing the index of  $d$  by unity.

therefore

$$\left\{ \frac{d^3 u}{d\alpha^3} \right\} = \left\{ \frac{d^3 \cdot f z^3 \cdot du}{dt^3} \right\}.$$

By continuing this process, it is easy to infer generally

$$\left\{ \frac{d^n u}{d\alpha^n} \right\} = \left\{ \frac{d^n \cdot f z^n \cdot du}{dt^n} \right\} = \left\{ \frac{d^{n-1} \cdot z^n \cdot \left\{ \frac{du}{dt} \right\}}{dt^{n-1}} \right\}.$$

Let us now suppose that by making  $\alpha=0$ , we have  $x=T$ ,  $T$  being a function  $t$ ; we shall substitute this value of  $x$ , in  $z$ , and in  $u$ .

Let  $Z$  and  $u$  represent what these quantities then become; we shall have on the hypothesis that  $\alpha=0$ ,

$$\left\{ \frac{d^n u}{d\alpha^n} \right\} = \frac{d^{n-1} \cdot Z^n \cdot \frac{du}{dt}}{dt^{n-1}},$$

and consequently, by what precedes, we shall obtain,

$$q_n = \frac{d^{n-1} Z^n \cdot \frac{du}{dt}}{1 \cdot 2 \cdot 3 \cdots n \cdot dt^{n-1}};$$

which gives

$$u = u + \alpha \cdot Z \cdot \frac{du}{dt} + \frac{\alpha^2}{1 \cdot 2} \cdot d \cdot \left\{ Z^2 \cdot \frac{du}{dt} \right\} + \frac{\alpha^3}{1 \cdot 2 \cdot 3} \cdot d^2 \cdot \left\{ Z^3 \cdot \frac{du}{dt} \right\} + \&c; \quad (P)$$

It only now remains to determine what function of  $t$  and  $\alpha$ ,  $x$  represents; which will be effected by the integration of the equation of partial differences  $\left\{ \frac{dx}{d\alpha} \right\} = z \cdot \left\{ \frac{dx}{dt} \right\}$ . For this purpose, we shall observe, that

$$dx = \left\{ \frac{dx}{dt} \right\} \cdot dt + \left\{ \frac{dx}{d\alpha} \right\} \cdot d\alpha;$$

and by substituting in place of  $\left\{ \frac{dx}{d\alpha} \right\}$ , its value  $z \cdot \left\{ \frac{dx}{dt} \right\}$  we will obtain

$$dx = \left\{ \frac{dx}{dt} \right\} \cdot (dt + zd\alpha) = \frac{dx}{dt} \cdot (d(t + \alpha z) - \alpha \cdot \left\{ \frac{dz}{dx} \right\})$$

therefore, we shall have

$$dx = \frac{\frac{dx}{dt} \cdot d(t + \alpha z)}{1 + \alpha \cdot \left\{ \frac{dz}{dx} \right\} \cdot \left\{ \frac{dx}{dt} \right\}} ;^*$$

which gives by its integration,  $x = \phi(t + \alpha z)$ ,  $\phi(t + \alpha z)$  being an arbitrary function of  $t + \alpha z$ ; so that the quantity which we have termed  $T$ , is equal to  $\phi(t)$ . Consequently, as often as there exists between  $\alpha$  and  $x$ , an equation reducible to the form  $x = \phi(t + \alpha z)$ ; the value of  $u$  will be determined by the formula (P) in a series proceeding according to the powers of  $\alpha$ .

\*  $zd\alpha = d.\alpha z - \alpha \cdot \frac{dz}{dx} \cdot dx$ , therefore, by substituting this value of  $zd\alpha$ , we obtain the expression for  $dx$  given in the text; now as  $dx$  is an exact differential, the member, at the right hand side of the equation must be also an exact differential, consequently,  $\frac{dx}{dt} \div \left( 1 + \alpha \cdot \frac{dz}{dx} \cdot \frac{dx}{dt} \right)$ , must be equal to  $\phi'(t + \alpha z)$ ,  $\phi'$  denoting the derivative function of  $\phi$ .

$z$  being by hypothesis a function of  $x$ , let it equal  $F(x)$  and we shall have  $x = \phi(t + \alpha F(x))$ , and it is easy to obtain from this expression the proposed differential equation of partial differences, for

$$\frac{dx}{d\alpha} = \phi'(t + \alpha F(x)) \cdot \left\{ (F(x)) + \alpha F'(x) \cdot \frac{dx}{d\alpha} \right\}; \quad \frac{dx}{dt} = \phi'(t + \alpha F(x)) \left\{ 1 + \alpha F'(x) \cdot \frac{dx}{dt} \right\};$$

and by eliminating  $\phi'(t + \alpha F(x))$ , and reducing, we shall obtain

$$\frac{dx}{d\alpha} = F(x) \cdot \frac{dx}{dt};$$

Let us now suppose, that  $u$  is a function of the two variables  $x$  and  $x'$ , these variables being given by the equations of partial differences

$$\left\{ \frac{dx}{d\alpha} \right\} = z \cdot \left\{ \frac{dx}{dt} \right\}; \quad \left\{ \frac{dx'}{d\alpha} \right\} = z' \cdot \left\{ \frac{dx'}{dt} \right\};$$

in which  $z$  and  $z'$  are any functions whatever of  $x$  and  $x'$ . It is easy to be assured that the integrals of these equations are respectively

$$x = \varphi(t + \alpha z); \quad x' = \psi(t' + \alpha' z');$$

$\varphi(t + \alpha z)$ , and  $\psi(t' + \alpha' z')$  being arbitrary functions, the one of  $t + \alpha z$ ,

and as  $u$  is supposed to be equal to  $\varphi(x)$ ,

$$\frac{du}{d\alpha} = \varphi'(x) \cdot \frac{dx}{dx}; \quad \frac{du}{dt} = \varphi'(x) \cdot \frac{dx}{dt};$$

hence, by eliminating  $\varphi(x)$  we obtain  $\frac{du}{d\alpha} \cdot \frac{dx}{dt} = \frac{du}{dt} \cdot \frac{dx}{d\alpha}$ , and by substituting for  $\frac{dx}{d\alpha}$

its value  $F(x) \cdot \frac{dx}{dt}$ , and making  $F(x) = z$ , we obtain after all reductions  $\frac{du}{d\alpha} = z \cdot \frac{du}{dt}$ ;

when  $x = t + \alpha F(x)$ ;  $x = t$  when  $\alpha = 0$ ;  $\frac{dx}{dt} = 1$ ;  $u$ ,  $Z$ , and  $\frac{du}{dt}$  become respectively

$\psi(t)$ ,  $F(t)$ , and  $\psi'(t)$ , consequently, the equation (P) will become  $\psi(t) + \psi'(t) \cdot F(t)$

$\frac{\alpha}{1} + \frac{d(\psi(t) \cdot F(t)^2)}{dt^2} \frac{\alpha^2}{1.2} + \frac{d^2(\psi'(t) \cdot F(t)^3)}{dt^3} \cdot \frac{\alpha^3}{1.2.3} + \text{&c.}$ , if in the preceding equation,  $a=1$ , then we shall have  $x = t + F(x)$ , and the preceding series becomes  $\psi(x) =$

$\psi t + \psi'(t) \cdot F(t) + \frac{1}{1.2} \cdot \frac{d(\psi'(t) \cdot F(t)^2)}{da} + \text{&c.}$ , which Lagrange first announced in 1772,

an epoch deservedly celebrated in the history of science for the many beautiful applications of this series, if  $F(x) = 1$ , then  $x = F(t + \alpha)$ , and  $\therefore u = \psi(x)$ .

\* Let  $z = F(x x')$ ;  $z' = F(x x')$ ;  $\therefore x = \varphi(t + \alpha \cdot F(x x'))$ ,  $x' = \psi(t' + \alpha' \cdot F(x x'))$ , if the functions indicated by  $F$ ,  $F'$ , be defined, and if the form of the preceding equations permits us to eliminate, the values of  $z'$  and  $z$ , may be respectively obtained in terms of  $z$ ,  $t$ ,  $\alpha$ ,  $t'$ , we may  $\therefore$  regard  $x$ ,  $x'$ , as functions of those four quantities.

$$\frac{dx}{dt} = \varphi'(t + \alpha \cdot F(x x')) \cdot (1 + \alpha \cdot \frac{dF}{dt}); \quad \frac{dx}{d\alpha} = \varphi'(t + \alpha \cdot F(x x')) \cdot (F + \alpha \cdot \frac{dF}{d\alpha});$$

and the other of  $t' + \alpha' z'$ . Moreover, we have

$$\left\{ \frac{du}{d\alpha} \right\} = z \cdot \left\{ \frac{du}{dt} \right\}; \quad \left\{ \frac{du}{d\alpha'} \right\} = z' \cdot \left\{ \frac{du}{dt'} \right\}.$$

$$\frac{dx}{dt} = \phi'(t + \alpha \cdot F(x, x')).(1 + a \cdot \frac{dF}{dt'}) ; \quad \frac{dx}{da'} = \phi'(t + a \cdot F(x, x')).a \cdot \frac{dF}{da'} ;$$

$$\frac{dx'}{dt} = \psi'(t' + \alpha' \cdot F(x, x')) \cdot \alpha \cdot \frac{dF}{dt}; \quad \frac{dx'}{d\alpha} = \psi'(t' + \alpha' \cdot F(x, x')) \cdot \alpha' \cdot \frac{dF}{d\alpha};$$

$$\frac{dx'}{dt'} = \psi'(t' + \alpha' \cdot F(x, x')).(1 + \alpha' \cdot \frac{dF}{dt'}) ; \quad \frac{dx'}{d\alpha'} = \psi'(t' + \alpha' \cdot F(x, x')).\alpha' \cdot \frac{dF}{d\alpha'} ;$$

$$\frac{dx'}{dt'} = \psi'(t' + \alpha' \cdot F'(x, x')).(1 + \alpha' \cdot \frac{dF'}{dt'}); \quad \frac{dx'}{d\alpha'} = \psi'(t' + \alpha' \cdot F'(x, x')).\alpha' \cdot \frac{dF'}{d\alpha'};$$

when  $\alpha, \alpha'$  vanish, we have  $\frac{dx}{dt} = \phi'(t)$ ,  $\frac{dx}{d\alpha} = \phi'(t) \cdot F$ ;

$$\frac{dx}{dt'} = 0; \quad \frac{dx}{d\alpha'} = 0; \quad \frac{dx'}{d\alpha} = 0; \quad \frac{dx}{dt'} = \psi'(t'); \quad \frac{dx'}{d\alpha'} = \psi'(t') \cdot F;$$

in this case  $x = \phi(t)$ ;  $x' = \psi(t')$ ,  $\therefore u$  is a function of  $t, t'$ , only; as  $u$  is only an explicit function of  $x, x'$ , we shall have

$$\frac{du}{d\alpha} = \frac{du}{dx} \cdot \frac{dx}{d\alpha} + \frac{du}{dx'} \cdot \frac{dx'}{d\alpha}; \text{ and when } \alpha \text{ and } \alpha' \text{ vanish } \frac{dx}{d\alpha} = \phi'(t) \cdot F.$$

$$\frac{dx'}{d\alpha} = 0; \quad \therefore \frac{du}{d\alpha} = \frac{du}{dx} \cdot \phi'(t) \cdot F; \quad \frac{du}{d\alpha'} = \frac{du}{dx} \cdot \frac{dx}{d\alpha} + \frac{du}{dx'} \cdot \frac{dx'}{d\alpha}, \text{ and when } a, a' = 0,$$

$$\frac{dx}{d\alpha'} = 0; \quad \frac{dx'}{d\alpha'} = \psi'(t') \cdot F, \text{ and } \therefore \frac{du}{d\alpha} = \frac{du}{dx} \cdot \psi'(t') \cdot F; \quad \phi'(t) = \frac{dx}{dt}; \quad \psi'(t) = \frac{dx}{dt'};$$

$$\therefore \frac{du}{d\alpha} = \frac{du}{dx} \cdot \frac{dx}{dt} \cdot F = \frac{du}{dt} \cdot F; \quad \frac{du}{d\alpha'} = \frac{du}{dx'} \cdot \frac{dx'}{dt'} \cdot F = \frac{du}{dt'} \cdot F, \quad \therefore \text{ by substituting } z \text{ for}$$

$F$  we obtain  $\frac{du}{d\alpha} - z \cdot \frac{du}{dt} = 0$  when  $x = \phi(t + \alpha z)$ , conversely, when this differential equation obtains, we can determine the value of  $x = \phi(t + \alpha z)$ .

As  $u$  depends explicitly only on  $x, t', \alpha'$ , and as  $\alpha'$  is one of the independent variables in differencing  $u$  with respect to  $\alpha$ , it is only necessary to have respect to  $x$ ,  $\therefore$  the reasoning of the preceding page is applicable in this case.

When  $\alpha$  is equal to cipher  $\frac{du}{d\alpha} = z \cdot \frac{du}{dt}$ ,  $\therefore$  in this case we may substitute  $\frac{du}{d\alpha}$  for  $z \cdot \frac{du}{dt}$ .

This being premised, if we conceive that  $x'$  is eliminated from  $u$  and from  $z$ , by means of the equation  $x' = \psi(t' + \alpha' z')$ ;  $u$  and  $z$  will become functions of  $x$ ,  $\alpha'$  and  $t'$  without  $\alpha$  or  $t$ ; therefore we shall obtain, by what goes before,

$$\left\{ \frac{d^n u}{d\alpha^n} \right\} = \left\{ \frac{d^{n-1} \cdot z^n \cdot \left\{ \frac{du}{dt} \right\}}{dt^{n-1}} \right\}.$$

If we suppose  $\alpha=0$  after the differentiations, and if besides, we make  $x=\phi(t+\alpha z^n)$  in the second member of this equation  $x=\phi(t+\alpha z^n)$ , and consequently  $\left\{ \frac{du}{d\alpha} \right\} = z^n \cdot \left\{ \frac{du}{dt} \right\}$ , we shall have on these suppositions,

$$\left\{ \frac{d^n u}{d\alpha^n} \right\} = \left\{ \frac{d^{n-1} \cdot \left\{ \frac{du}{d\alpha} \right\}}{dt^{n-1}} \right\},$$

and consequently,

$$\left\{ \frac{d^{n+n'} u}{d\alpha^n \cdot d\alpha'^{n'}} \right\} = \left\{ \frac{d^{n-1} \cdot (d \cdot \left\{ \frac{d^{n'} u}{d\alpha'^{n'}} \right\})}{dt^{n-1}} \right\}.$$

We shall have in like manner,

$$\left\{ \frac{d^{n'} u}{d\alpha'^{n'}} \right\} = \left\{ \frac{d^{n'-1} \cdot \left\{ \frac{du}{d\alpha} \right\}}{dt^{n-1}} \right\}.$$

If we suppose  $a'$  to vanish after the differentiations, and if besides we suppose that in the second member of this equation,  $x' = \psi(t' + a' z'^{n'})$ ; we shall obtain

$$\left\{ \frac{d^{n+n'} u}{d\alpha^n \cdot d\alpha'^{n'}} \right\} = \left\{ \frac{d^{n+n'-2} \left\{ \frac{d^2 u}{d\alpha \cdot d\alpha'} \right\}}{dt^{n-1} \cdot dt^{n'-1}} \right\};$$

provided that we make  $\alpha$  and  $\alpha'$  to vanish after the differentiations, and also that we suppose in the second member of this equation

$$x = \varphi(t + \alpha z^n); \quad x' = \psi(t' + \alpha' z'^n);$$

which comes to supposing in the second member as well as in the first

$$x = \varphi(t + \alpha z); \quad x' = \psi(t + \alpha' z'),$$

and to change in the partial difference  $\left\{ \frac{d^2 u}{dx.d\alpha'} \right\}$ , of this second member  $z$  into  $z^n$ , and  $z'$  into  $z'^n$ . Thus, we shall have on those suppositions, and also by changing  $z$  into  $Z$ ,  $z'$  into  $Z'$ , and  $u$  into  $u$ ,

$$q_{n,n} = \left\{ \frac{d^{n+n'-2} \cdot \left\{ \frac{d u}{d\alpha.d\alpha'} \right\}}{1.2.3.....n. 1.2.3.....n'. dt^{n-1}.dt'^{n'-1}} \right\}.$$

By following on this reasoning, it is easy to infer, that if we have  $r$  equations,

$$\begin{aligned} x &= \varphi(t + \alpha z); \\ x' &= \psi(t' + \alpha' z'); \\ x'' &= \Pi(t'' + \alpha'' z''); \\ &\text{&c.} \end{aligned}$$

$z, z', z'', \text{ &c.}$ , being any functions whatever of  $x, x', x'', \text{ &c.}$ ;  $u$  being supposed to be a function of the same variables, we shall have generally

$$q_{n, n' n' \text{ &c.}} = \left\{ \frac{d^{n+n'+n''+\text{&c.}-r} \cdot \left\{ \frac{d^r u}{d\alpha.d\alpha'.d\alpha''. \text{ &c.}} \right\}}{1.2.3...n.1.2.3...n'.1.2.3...n''.\text{&c.} dt^{n-1}.dt'^{n'-1}.dt''^{n''-1}} \right\};$$

provided that in the partial difference  $\left\{ \frac{d^r u}{d\alpha.d\alpha'.d\alpha''. \text{ &c.}} \right\}$ , we change  $z$  into  $z^n$ ,  $z'$  into  $z'^n$ , &c., and that afterwards we change  $z$  into  $Z$ ,  $z'$  into  $Z'$ ,  $z''$  into  $Z''$ , &c., and  $u$  into  $u'$ .

If there is but one variable  $x$ , we shall have

$$\left\{ \frac{du}{d\alpha} \right\} = z \cdot \left\{ \frac{du}{dt} \right\};$$

therefore

$$q_n = \frac{d^{n-1} \cdot (z^n \cdot \left\{ \frac{du}{dt} \right\})}{1.2.3.\dots.n.dt^{n-1}}$$

If there are two variables  $x$  and  $x'$ ; we shall have

$$\left\{ \frac{du}{d\alpha} \right\} = z \cdot \left\{ \frac{du}{dt} \right\}; *$$

this equation differenced with respect to  $\alpha'$ , gives

$$\left\{ \frac{d^2u}{d\alpha.d\alpha'} \right\} = \left\{ \frac{dz}{d\alpha'} \right\} \cdot \left\{ \frac{du}{dt} \right\} + z \cdot \left\{ \frac{d^2u}{d\alpha'.dt} \right\};$$

but we have  $\left\{ \frac{du}{d\alpha'} \right\} = z' \cdot \left\{ \frac{du}{dt'} \right\}$ ; and if in this equation  $z$  is substituted in place of  $u$ , we have  $\left\{ \frac{dz}{d\alpha'} \right\} = z' \cdot \left\{ \frac{dz}{dt'} \right\}$ ; therefore

$$\left\{ \frac{d^2u}{d\alpha.d\alpha'} \right\} = z \cdot \left\{ \frac{d.z' \cdot \left\{ \frac{du}{dt'} \right\}}{dt} \right\} + z' \cdot \left\{ \frac{dz}{dt'} \right\} \cdot \left\{ \frac{du}{dt} \right\}.$$

\* By substituting  $z^n$  for  $z$ , &c. we have made the coefficient  $\frac{d^{n-1}u}{d\alpha^n.d\alpha'^n} = q_{n,n}$  to depend on a coefficient of the second order, and the differentiations relative to  $t$  and  $t'$  will not be difficult when  $\alpha$ ,  $\alpha'$  are  $=$  to cipher.

$$\frac{du}{d\alpha'} = z' \cdot \frac{du}{dt'}, \therefore z \cdot \frac{du^2}{d\alpha'.dt} = z.d(z' \cdot \left\{ \frac{du}{dt'} \right\}) = zz' \cdot \left\{ \frac{d^2u}{dt.dt'} \right\} + z \cdot \frac{dz'}{dt} \cdot \frac{du}{dt},$$

$\therefore$  by substituting  $z^n$ ,  $z'^n$ , for  $z$ ,  $z'$ , respectively, we obtain the expression which is given in the text.

If we suppose  $\alpha$  and  $\alpha'$  equal to nothing, in the second member of this equation, and if we change  $z$  into  $Z^n$ ,  $z'$  into  $Z^{n'}$ , and  $u$  into  $u$ ; we shall obtain the value of  $\left\{ \frac{d^2u}{d\alpha.d\alpha'} \right\}$ , on the same suppositions; hence we obtain

$$q_{n,n'} = d^{n+n'-2} \cdot \frac{\left\{ Z^n \cdot Z^{n'} \cdot \left\{ \frac{d^2u}{dt.dt'} \right\} + Z^{n'} \cdot \left\{ \frac{d.Z^n}{dt'} \right\} \cdot \left\{ \frac{du}{dt} \right\} \right\} + Z^n \cdot \left\{ \frac{d.Z^{n'}}{dt} \right\} \cdot \left\{ \frac{du}{dt} \right\}}{1.2.3.\dots.n.dt^{n-1}.1.2.3.\dots.n'.dt'^{n'-1}}$$

by proceeding in this manner the value of  $q_{n,n',nn'}$ , &c., for any number of variables whatever, may be obtained.

Although we have supposed that  $u, z, z', z'', \&c.$ , are functions of  $x, x', x'', \&c.$ , without  $t, t', t'', \&c.$ ; we can however suppose, that they contain these last variables: but then denoting these variables by  $t, t', t'', \&c.$ , it is necessary to suppose  $t, t', t'', \&c.$ , constant in the differentiations, and after these operations to restore  $t, t', \&c.$ , in place of  $t, t', \&c.$ .

22. Let us apply these results to the elliptic motion of the planets; and for this purpose, let the equations (*f*) of No. 20, be resumed. The equation  $nt = u - e. \sin. u$ , or  $u = nt + e. \sin. u$ , being compared with  $x = \varphi(t+az)$ ;  $x$  will be changed into  $u$ ,  $t$  into  $nt$ , and  $a$  into  $e$ ,  $z$  into  $\sin. u$ , and  $\varphi(t+az)$  into  $nt+e. \sin. u$ , consequently, the formula (*P*) of the preceding number will become

$$\begin{aligned} \psi(u) &= \psi(nt) + e. \psi'(nt). \sin. nt + \frac{e^2}{1.2} \cdot d. \frac{(\psi'(nt). \sin. {}^3nt)}{ndt} + \frac{e^3}{1.2.3} \cdot \\ &d^2. \frac{(\psi'(nt). \sin. {}^3nt)}{n^2 dt^2} + \&c.; (q) \end{aligned}$$

\* If in the equation  $u = nt + e. \sin. u$  it be required to develope  $\psi(u)$  into a series arranged according to the powers of  $e$ , then applying the preceding formula, besides the changes indicated in the text,  $u$  will be changed into  $\psi(u)$ .

$\psi'(nt)$  being equal to  $\frac{d\psi(nt)}{ndt}$ . In order to expand this formula, it is to be observed that  $c$  being the number of which the hyperbolical logarithm is unity, we have

$$\sin^i nt = \left\{ \frac{c^{nt\sqrt{-1}} - c^{-nt\sqrt{-1}}}{2\sqrt{-1}} \right\}^i; \quad \cos^i nt = \left\{ \frac{c^{nt\sqrt{-1}} + c^{-nt\sqrt{-1}}}{2} \right\}^{i^*};$$

$i$  being any number whatever. If we expand the second members of these equations, and then substitute, in place of  $c^{rnt\sqrt{-1}}$  and of  $c^{-rnt\sqrt{-1}}$ , their values  $\cos. rnt + \sqrt{-1} \sin. rnt\sqrt{-1}$ , and  $\cos. rnt - \sqrt{-1} \sin. rnt\sqrt{-1}$ ,  $r$  being any number whatever; we will obtain the  $i$  powers of  $\sin. nt$  and of  $\cos. nt$ , evolved according to the sines and cosines of the angles  $nt$  of its multiples; this being premised, we shall find

$$\begin{aligned} & \sin. nt + \frac{e}{1.2} \cdot \sin. {}^2 nt + \frac{e^2}{1.2.3} \cdot \sin. {}^3 nt + \frac{e^3}{1.2.3.4} \cdot \sin. {}^4 nt + \text{&c.} \\ &= \sin. nt - \frac{e}{1.2.2} (\cos. 2nt - 1) \\ & \quad - \frac{e^2}{1.2.3.2^2} (\sin. 3nt - 3 \sin. nt) \\ & \quad + \frac{e^5}{1.2.3.4.2^3} (\cos. 4nt - 4 \cos. 2nt + \frac{1}{2} \cdot \frac{4.3}{1.2}) \\ & \quad + \frac{e^4}{1.2.3.4.5.2^4} (\sin. 5nt - 5 \sin. 3nt + \frac{5.4}{1.2} \cdot \sin. nt) \\ & \quad - \frac{e^5}{1.2.3.4.5.6.2^5} (\cos. 6nt - 6 \cos. 4nt + \frac{6.5}{1.2} \cdot \cos. 2n - \frac{1}{2} \cdot \frac{6.5.4}{1.2.3}) \\ & \quad - \text{&c.} \end{aligned}$$

\* See Lacroix, Traité Complete, Tome 1, page 76, 95, of the Introduction.

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Let  $P^*$  represent this function ; if it be multiplied by  $\psi(nt)$  and then if each of its terms be differenced, with respect to  $t$ , as often as there are units in the power of  $e$ , by which it is multiplied,  $dt$  being supposed constant ; and if then these differentials be divided by the corresponding power of  $ndt$ , the formula ( $q$ ) will become

$$\psi u = \psi(nt) + eP'.$$

$P$  representing the sum of these differentials thus divided.

\* The series  $P$  is always the same where the equation  $u = nt + e \sin. nt$  obtains, whatever be the form of the function indicated by  $\psi$ ; therefore when the form of  $\psi$  is given, the expression for  $\psi(u)$  will be obtained by performing the operations indicated in the text.

When the value of  $P$ , is multiplied by  $e \cos. nt$ , the form of the terms multiplied into the even powers of  $e$ , will be  $\cos. i. nt. \sin. s. nt$ , and the expansion of this product is effected by the formula  $\sin. a. \cos. b = \sin. \frac{(a+b)+\sin.(a-b)}{2}$ , therefore the terms multiplied by the even powers will be the sines. The form of the terms multiplied into the odd powers of  $e$ , will be  $\cos. in. \cos. snt$  the developement of which is effected by the formula  $\cos. a. \cos. b = \frac{\cos.(a+b)+\cos.(a-b)}{2}$ , consequently the terms multiplied by the odd powers of  $e$  will be the cosines. If any term of the form  $Ke^{2r} \cdot \sin. snt$  be differenced as often as there are units in  $2r$ , it is evident that when this term is divided by  $ndt^{2r}$ , the resulting terms will be  $Ke^{2r} \cdot s^{2r} \cdot \sin. snt$ , for as the terms are alternately  $\cos. snt$ ,  $\sin. snt$  when the number of differentiations is even the last term must be  $\sin. snt$ , and as  $s$  is introduced as a factor at each successive differentiation when the number of differentiations is  $2r$ ,  $s^{2r}$  will be a factor of this last term, the first term is  $+$   $\cos. int$ , and the signs of the subsequent terms are minus and plus in pairs,  $\therefore$  the signs of the successive differential coefficients including the first, are plus minus, minus plus, plus minus ; i. e.  $+$   $-$ ,  $-$   $+$ ,  $+$   $-$ , &c.; hence it appears, that when  $r$  is an odd number, the sign of the last term will be  $-$ , and when  $r$  is an even number, the last term will be  $+$ . In a term of the form of the  $Ke^{2r+1} \cdot \cos. snt$  the number of differentiations being odd, the last term must be of the form  $Ke^{2r+1} \cdot s^{2r+1} \cdot \sin. snt$ , the signs of the terms in this case are alternately minus and plus in pairs, i. e.  $--$ ,  $+$   $+$ ,  $--$ ,  $+$   $+$ , and as the sign of  $\sin. snt$ , is the opposite of the sign of the penultimate term, when  $r$  is even this sign is evidently  $--$ , and when  $r$  is odd this sign is  $+$ .

It would be easy by this method to obtain the values of the angle  $u$ , and of the sines and cosines of this angle, and of its multiples. If for example, we suppose  $\psi(u) = \sin. iu$ ; we shall obtain  $\psi(nt) = i \cos. int$ . The preceding value of  $P$ , must be multiplied by  $i \cos. int$ , and the product should be expanded into sines and cosines of the angle  $nt$ , and of its multiples. The sines will be multiplied by the even powers of  $e$ , and the cosines will be multiplied by the odd powers of  $e$ . Then any term of the form  $Ke^{2r} \cdot \sin. snt$  will be changed into  $\pm Ke^{2r} \cdot s^{2r}$ ,  $\sin. snt$ , the sign + having place, if  $r$  is even, and the sign — obtaining, if  $r$  is odd. In like manner any term of the form  $Ke^{2r+1} \cdot \cos. snt$  will be changed into  $\mp Ke^{2r+1} \cdot s^{2r+1} \cdot \sin. snt$ , the sign — having place if  $r$  is even, and the sign + obtaining, if  $r$  be odd. The sum of all these terms will be the value of  $P'$ , and we shall obtain

$$\sin. iu = \sin. int + eP'.$$

If  $\psi(u)$  be supposed equal to  $u^*$ ,  $\psi(nt)$  will be equal to unity, and we will find

$$\begin{aligned} u = nt + e \cdot \sin. nt + & \frac{e^2}{1.2.2} \cdot 2 \sin. 2nt + \frac{e^3}{1.2.3.2^2} (3^2 \cdot \sin. 3nt - 3 \sin. nt) \\ & + \frac{e^4}{1.2.3.4.2^3} (4^3 \cdot \sin. 4nt - 4 \cdot 2^3 \cdot \sin. 2nt) \\ & + \frac{e^5}{1.2.3.4.5.2^4} (5^4 \cdot \sin. 5nt - 5 \cdot 3 \cdot \\ & \sin. 3nt + \frac{5 \cdot 4}{1.2} \cdot \sin. nt). \end{aligned}$$

\* If  $\psi(u) = u$ , then  $\psi(nt) = nt$ , and  $\psi'(nt) = \frac{ndt}{ndt} = 1$ , the series  $P'$  becomes  $\sin. nt$   
 $- \frac{e}{1.2.2} \cdot d \cdot \frac{\cos. (2nt-1)}{ndt} - \frac{e^2}{1.2.3.2^2} \cdot d^2 \cdot \frac{(\sin. 3nt - 3 \sin. nt)}{(ndt)^2} - \frac{e^3}{1.2.3.4.2^3} \cdot$   
 $d^3 \cdot \frac{(\cos. 4nt - 4 \cos. 2nt + \frac{4 \cdot 3}{1.2})}{(ndt)^3} + \&c.$  which will be reduced to the expression in the  
text, by performing the prescribed differentiations.

This series is very converging for the planets.  $u$  being thus determined for any instant; the corresponding values of  $r$  and  $v$ , will be given by means of the equations ( $f$ ) of N°. 20; but we can obtain these last quantities directly in converging series, in the following manner:

For this purpose, it may be remarked, that by No. 20, we have  $r = a(1 - e \cos. u)$ ; and if in the formula ( $g$ ), we suppose  $\psi(u) = 1 - e \cos. u$ , we shall have  $\psi'(nt) = e \sin. nt$ , and consequently

$$1 - e \cos. u = 1 - e \cos. nt + e^2 \cdot \sin. ^2 nt + \frac{e^3}{1.2} \cdot \frac{d \cdot \sin. ^3 nt}{ndt} + \frac{e^4}{1.2.3} \cdot \frac{d^2 \cdot \sin. ^4 nt}{n^2 dt^2} + \text{&c.}$$

Therefore by the preceding analysis, we shall obtain

$$\begin{aligned} \frac{r}{a} &= 1 + \frac{e^2}{2} - e \cos. nt - \frac{e^2}{2} \cos. 2nt^* \\ &\quad - \frac{e^3}{1.2.2^2} (3 \cdot \cos. 3nt - 3 \cdot \cos. nt) \\ &\quad - \frac{e^4}{1.2.3.2^3} (4^2 \cdot \cos. 4nt - 4.2^2 \cdot \cos. 2nt) \\ &\quad - \frac{e^5}{1.2.3.4.2^4} (5^3 \cdot \cos. 5nt - 5.3^3 \cdot \cos. 3nt + \frac{5.4}{1.2} \cdot \\ &\quad \cos. nt) \end{aligned}$$

\* Since  $\psi(u) = 1 - e \cos. u$ ,  $\psi(nt) = 1 - e \cos. nt$ ; by substituting for  $\sin. ^2 nt$ ,  $\sin. ^3 nt$  &c. their values, the expression for  $1 - e \cos. u$  becomes  $1 - e \cos. nt + \frac{e^2}{2} \cdot (1 - \cos. 2nt) + \frac{e^3}{1.2} \cdot d \cdot \frac{(-\sin. 3nt + 3 \sin. nt)}{4ndt} + \frac{e^4}{1.2.3} \cdot d^2 \cdot \frac{(\cos. 4nt - 4 \cos. 2nt + 3)}{8(ndt)^2} + \text{&c.}$

+&c., now when the differentiations indicated by the characteristics  $d$ ,  $d^2$ , &c. are performed, the resulting terms only contain  $\cos. nt$ , and its multiples, for those terms, in which the differentiation is performed an odd number of times, involve the sines of  $nt$  and of its multiples, therefore the resulting terms are cosines, and where the cosines of  $nt$ , and of its multiples are to be operated upon, the differentiation must be performed an even number of times, ∵ the resulting terms are in this case also cosines. The reason why in terms of the form  $Ke^{2r} \cdot \sin. snt$  the resulting quantity becomes  $Ke^{2r} \cdot s^2 r \cdot \sin. snt$ , is the same as that assigned for a similar expression in the preceding page.

$$-\frac{e^6}{1.2.3.4.5.2^5} \cdot (6^4 \cdot \cos. 6nt - 6.4^4 \cdot \cos. 4nt + \\ \frac{6.5}{1.2} \cdot 2^4 \cdot \cos. 2nt) \cdot$$

Let us now consider the third of the equations (*f*) of No. 20; by means of it we obtain

$$\frac{\sin. \frac{1}{2}v}{\cos. \frac{1}{2}v} = \sqrt{\frac{1+e}{1-e}} \cdot \frac{\sin. \frac{1}{2}u}{\cos. \frac{1}{2}u}.$$

By substituting in this equation, in place of the sines and of the cosines, their values expressed in imaginary exponentials, we shall have

$$\frac{e^{\frac{v\sqrt{-1}}{2}-1}-1}{e^{\frac{v\sqrt{-1}}{2}+1}} = \sqrt{\frac{1+e}{1-e}} \left\{ \frac{e^{\frac{u\sqrt{-1}}{2}-1}-1}{e^{\frac{u\sqrt{-1}}{2}+1}} \right\}; *$$

and by supposing

$$\lambda = \frac{e}{1 + \sqrt{1-e^2}};$$

we shall have

U 2

$$* \sin. \frac{1}{2}v = \frac{\frac{v}{2}\sqrt{-1} - c}{2\sqrt{-1}}, \cos. \frac{1}{2}v = \frac{\frac{v}{2}\sqrt{-1} + c}{2}, \therefore \text{substituting}$$

these expressions for  $\sin. \frac{v}{2} \cos. \frac{v}{2}$ , in the expression  $\sin. \frac{v}{2}$ , multiplying both numerator  
 $\cos. \frac{v}{2}$

and denominator by  $c^{\frac{v}{2}\sqrt{-1}}$ , and performing similar operations on  $\frac{\sin. \frac{u}{2}}{\cos. \frac{u}{2}}$ , we shall

have the expression in the text.

$$c^{v\sqrt{-1}} = c^{u\sqrt{-1}} \cdot \left\{ \frac{1 - \lambda \cdot c^{-u\sqrt{-1}}}{1 + \lambda \cdot c^{u\sqrt{-1}}} \right\};$$

and consequently,

$$v = u + \frac{\log. (1 - \lambda \cdot c^{-u\sqrt{-1}}) - \log. (1 - \lambda \cdot c^{u\sqrt{-1}})}{\sqrt{-1}};$$

from which may be obtained, by reducing the logarithms into series,†

$$\begin{aligned} * \because c^{v\sqrt{-1}} - 1 &= c^{u\sqrt{-1}} \left( \frac{1 - \lambda c^{-u\sqrt{-1}}}{1 + \lambda c^{u\sqrt{-1}}} - 1 \right) = \frac{(1 + \lambda) \cdot (c^{u\sqrt{-1}} - 1)}{1 - \lambda c^{u\sqrt{-1}}}; \text{ and} \\ c^{v\sqrt{-1}} + 1 &= c^{u\sqrt{-1}} \left( \frac{1 - \lambda c^{-u\sqrt{-1}}}{1 + \lambda c^{u\sqrt{-1}}} + 1 \right) = \frac{(1 - \lambda) \cdot (c^{u\sqrt{-1}} + 1)}{1 - \lambda c^{u\sqrt{-1}}}, \therefore \frac{c^{v\sqrt{-1}} - 1}{c^{v\sqrt{-1}} + 1} \\ &= \frac{(1 + \lambda) \cdot (c^{u\sqrt{-1}} - 1)}{(1 - \lambda) \cdot (c^{u\sqrt{-1}} + 1)}; 1 + \lambda = \frac{1 + e + \sqrt{1 - e^2}}{1 + \sqrt{1 - e^2}} = \sqrt{1 + e} \cdot \frac{(\sqrt{1 + e} + \sqrt{1 - e})}{1 + \sqrt{1 - e^2}}; (1 - \lambda) \\ &= \frac{1 - e + \sqrt{1 - e^2}}{1 + \sqrt{1 - e^2}} = \sqrt{1 - e} \cdot \frac{(\sqrt{1 - e} + \sqrt{1 + e})}{1 + \sqrt{1 - e^2}}, \therefore \text{by substituting these values of } 1 + \lambda \\ \text{and } 1 - \lambda, \text{ we obtain} \\ \frac{c^{v\sqrt{-1}} - 1}{c^{v\sqrt{-1}} + 1} &= \frac{\sqrt{1 + e} (\sqrt{1 - e} + \sqrt{1 + e})}{\sqrt{1 - e} (\sqrt{1 + e} + \sqrt{1 - e})} \cdot \frac{(c^{u\sqrt{-1}} - 1)}{(c^{u\sqrt{-1}} + 1)} = \frac{\sqrt{1 + e}}{\sqrt{1 - e}} \cdot \frac{c^{u\sqrt{-1}} - 1}{c^{u\sqrt{-1}} + 1}. \\ + \log. c^{v\sqrt{-1}} &\equiv v\sqrt{-1} = \log. c^{u\sqrt{-1}} + \log. (1 - \lambda \cdot c^{-u\sqrt{-1}}) - \log. (1 - \lambda \cdot c^{u\sqrt{-1}}) = u\sqrt{-1} + \log. (1 - \lambda \cdot c^{-u\sqrt{-1}}) - \log. (1 - \lambda \cdot c^{u\sqrt{-1}}); \log. (1 - \lambda \cdot c^{-u\sqrt{-1}}) \\ &= -\frac{\lambda}{1} c^{-\sqrt{-1}} - \frac{\lambda^2}{2} \cdot c^{-2u\sqrt{-1}} - \frac{\lambda^3}{3} \cdot c^{-3u\sqrt{-1}} - \&c. - \log. (1 - \lambda) \cdot c^{u\sqrt{-1}} \\ &= \frac{\lambda}{1} \cdot c^{u\sqrt{-1}} + \frac{\lambda^2}{2} \cdot c^{2u\sqrt{-1}} + \frac{\lambda^3}{3} \cdot c^{3u\sqrt{-1}} + \&c.; \therefore \log. (1 - \lambda) \cdot c^{-u\sqrt{-1}} - \\ \log. (1 - \lambda) \cdot c^{u\sqrt{-1}} &= \frac{\lambda}{1} \cdot (c^{u\sqrt{-1}} - c^{-u\sqrt{-1}}) + \frac{\lambda^2}{2} \cdot (c^{2u\sqrt{-1}} - c^{-2u\sqrt{-1}}) + \frac{\lambda^3}{3}. \end{aligned}$$

$$v = u + 2\lambda \cdot \sin. u + \frac{2\lambda^2}{2} \cdot \sin. 2u + \frac{2\lambda^3}{3} \cdot \sin. 3u + \frac{2\lambda^4}{4} \cdot \sin. 4u + \text{&c.}$$

by what goes before, we have  $u$ ,  $\sin. u$ ,  $\sin. 2u$ , &c. in a series arranged according to the powers of  $e$ , and expanded into sines and cosines of the angle  $nt$  and its multiples, therefore in order to obtain  $v$  expressed in a similar series, it is only necessary to expand the successive powers of  $t$  into a series ranged according to the powers of  $e$ .

The equation  $u = 2 - \frac{e^2}{u}$ , will give by the formula ( $p$ ) of the preceding number,

$$\frac{1}{u^i} = \frac{1}{2^i} + \frac{i \cdot e^2}{2^{i+2}} + \frac{i \cdot (i+3)}{1 \cdot 2} \cdot \frac{e^4}{2^{i+4}} + \frac{i \cdot (i+3) \cdot (i+5)}{1 \cdot 2 \cdot 3} \cdot \frac{e^6}{2^{i+6}} + \text{&c. ;}$$

and as we have,

$$u = 1 + \sqrt{1 - e^2}; \text{ we shall have}$$

$$\lambda^i = \frac{e^i}{2^i} \left\{ 1 + i \left( \frac{e}{2} \right)^2 + \frac{i \cdot (i+3)}{1 \cdot 2} \cdot \left( \frac{e}{2} \right)^4 + \frac{i \cdot (i+3) \cdot (i+5)}{1 \cdot 2 \cdot 3} \cdot \left( \frac{e}{2} \right)^6 + \text{&c.} \right\}$$

This being premised, we shall find by continuing the approximation to

$(c \frac{3u\sqrt{-1}}{c} - c \frac{-3u\sqrt{-1}}{c}) + \text{&c.}, \because$  dividing by  $\sqrt{-1}$ , and substituting  $2\sqrt{-1} \sin. su$ . for  $c \frac{su\sqrt{-1}}{c} - c \frac{-su\sqrt{-1}}{c}$ , we obtain the expression which is given in the text.

\* The equation  $u = 2 - \frac{e^2}{u}$ , being compared with the expression  $x = \varphi(t+az)$  gives

$$z = F(x) = \frac{1}{u}, \quad a = -e^2, \quad t = 2, \quad \text{and } \psi(x) = \frac{1}{u^i}, \quad \therefore \text{when } \psi(x) = \frac{1}{u^i}, \quad \psi(t) = \frac{1}{2^i},$$

$$\psi'(t) = \frac{-i}{2^{i+1}}, \quad F(t) = \frac{1}{2}, \quad \text{consequently } \frac{1}{u^i} = \frac{1}{2^i} + \frac{ie^2}{2^{i+2}} + \frac{i(i+3)}{1 \cdot 2} \cdot \frac{e^4}{2^{i+4}} + \frac{i(i+3)(i+5)}{1 \cdot 2 \cdot 3} \cdot \frac{e^6}{2^{i+6}} + \text{&c.}$$

From the equation  $u = 2 - \frac{e^2}{u}$  we obtain  $u^2 - 2u = -e^2, \therefore \frac{1}{u} = 1 + \sqrt{1 - e^2} = \frac{\lambda}{e}$   
hence  $\lambda^i = \frac{e^i}{u^i} =$  the expression given in the text. And if  $i = 1$ ,  $\lambda = \frac{e}{2} \left( 1 + \left( \frac{e}{2} \right)^2 \right)$

quantities of the order  $i^5$  inclusively,

$$v = nt + \left\{ 2e - \frac{1}{4} \cdot e^3 + \frac{5}{96} \cdot e^5 \right\} \sin(nt) + \left\{ \frac{5}{4} \cdot e^2 - \frac{11}{24} \cdot e^4 + \frac{17}{192} \cdot e^6 \right\} \sin(2nt)$$

$$+ \left\{ \frac{13}{12} \cdot e^3 - \frac{43}{64} \cdot e^5 \right\} \sin(3nt) + \left\{ \frac{103}{96} \cdot e^4 - \frac{451}{480} \cdot e^6 \right\} \sin(4nt)$$

$$+ \frac{1097}{960} \cdot e^5 \sin(5nt) + \frac{1223}{960} \cdot e^6 \sin(6nt).$$

$$+ \left\{ \frac{1+3}{1.2} \right\} \cdot \frac{e^2}{2^4} = \frac{e}{2} + \frac{e^3}{8} + \frac{e^5}{16}, \quad (\text{as the approximation is not carried beyond the fifth powers}),$$

$$\lambda^2 = \frac{e^2}{2^2} \left\{ 1 + 2 \left\{ \frac{e}{2} \right\}^2 + \frac{2(2+3)}{1.2} \cdot \left\{ \frac{e}{2} \right\}^4 \right\} = \frac{e^2}{4} + \frac{e^4}{8} + \frac{5}{4} \cdot \frac{e^6}{16};$$

$$\lambda^3 = \frac{e^3}{2^3} \cdot (1+3 \cdot \left\{ \frac{e}{2} \right\}^2) = \frac{e^3}{8} + \frac{3}{8} \cdot \frac{e^5}{4};$$

$$\lambda^4 = \frac{e^4}{2^4} \cdot (1+4 \cdot \left\{ \frac{e}{2} \right\}^2) = \frac{e^4}{16} + \frac{e^6}{32};$$

$$\lambda^5 = \frac{e^5}{2^5} \cdot (1+5 \cdot \left\{ \frac{e}{2} \right\}^2) = \frac{e^5}{32} + \frac{5}{32} \cdot \frac{e^7}{48};$$

$$\lambda^6 = \frac{e^6}{2^6} \cdot (1+6 \cdot \left\{ \frac{e}{2} \right\}^2) = \frac{e^6}{64} + \frac{6}{64} \cdot \frac{e^8}{192}.$$

\* When  $u$  is expressed in the manner prescribed in the text, the five first terms are those given in the preceding page; and as the approximation is carried to the sixth powers of  $e$ , we must add the additional term which is

$$\frac{e^5}{1.2.3.4.5.6.2^5} \cdot (6^5 \cdot \sin(6nt) - 6.4^5 \cdot \sin(4nt) + \frac{6.5}{1.2} \cdot 2^5 \cdot \sin(2nt));$$

If to these terms expressing the value of  $u$ , be added the values of  $2\lambda \cdot \sin(u)$ ,  $2\lambda^2 \cdot \sin(2u)$ , &c., reduced into a series ranged according to the powers of  $e$ , and developed into sines and cosines of the angle  $nt$  and its multiples, we shall have

$$2\lambda \cdot \sin(u) = \left\{ e + \frac{e^3}{4} + \frac{e^5}{8} \right\} \sin(nt) + \left\{ \frac{e^2}{2} + \frac{e^4}{8} + \frac{e^6}{16} \right\} \sin(2nt) + \left\{ \frac{e^3}{8} + \frac{e^5}{32} \right\} (3 \cdot \sin(3nt) - \sin(nt)) + \left\{ \frac{e^7}{48} + \frac{e^9}{192} \right\} (4^2 \cdot \sin(4nt) - 2^2 \cdot \sin(2nt)) + \frac{e^5}{384} \cdot (5^3 \cdot \sin(5nt) - 3.3^3 \cdot \sin(3nt) + 2 \cdot \sin(nt)) + \frac{e^6}{3840} \cdot (6^4 \cdot \sin(6nt) + 4.4^4 \cdot \sin(4nt) + 5.2^4 \cdot \sin(2nt));$$

$$\frac{2\lambda^2}{2} \cdot \sin(2u) = \left\{ \frac{e^2}{4} + \frac{e^4}{8} + \frac{5e^6}{64} \right\} \sin(2nt) + \left\{ \frac{e^3}{4} + \frac{e^5}{8} \right\} (\sin(3nt) - \sin(nt)) + \left\{ \frac{e^4}{14} + \frac{e^6}{32} \right\} (4 \cdot \sin(4nt) - 4 \cdot \sin(2nt)) + \frac{e^5}{96} \cdot (5^2 \cdot \sin(5nt) -$$

The angles  $v$  and  $nt$  are here reckoned from the perihelium, but if we wish to count them from aphelion, it is evident that to effect this, it is only necessary to make  $e$  negative in the preceding expressions of  $r$  and  $v$ . It will also be sufficient to augment in those expressions, the angle  $nt$ , by the semicircumference, which renders the sines and cosines of the odd multiples of  $nt$  negative, consequently, as the results of these two methods ought to be identical, it is necessary that in the expressions of  $r$  and of  $v$ , the sines and cosines of the odd multiples of  $nt$ , should be multiplied by the odd powers of  $e$ , and that the sines and cosines of the even multiples of the same angle, should be multiplied by the even powers of this quantity. This is, in fact, confirmed *a posteriori* by the calculus.

Let us suppose that in place of reckoning the angle  $v$ , from the perihelium, we fix its origin at any point whatever; it is evident that this angle will be increased by a constant quantity, which we will denote by  $w$ , and which will express the longitude of the perihelium.

$$\begin{aligned} & 3.3^2 \cdot \sin. 3nt + 4 \sin. nt) + \frac{e^6}{768} \cdot (6^3 \cdot \sin. 6nt - 4.4^3 \cdot \sin. 4nt + 7.2^3 \cdot \sin. 2nt); \\ & \frac{2\lambda^5}{3} \cdot \sin. 3u = \left\{ \frac{e^3}{12} + \frac{e^5}{16} \right\} \cdot \sin. 3nt + \left\{ \frac{e^6}{8} + \frac{3e^5}{32} \right\} \cdot (\sin. 4nt - \sin. 2nt) + \\ & \frac{e^5}{32} \cdot (5 \cdot \sin. 5nt - 2.3 \cdot \sin. 3nt + \sin. nt) + \frac{e^6}{192} \cdot (6^2 \cdot \sin. 6nt - 3.4^2 \cdot \sin. 4nt + 3.2^2 \cdot \\ & \sin. 2nt); \frac{2\lambda^4}{4} \cdot \sin. 4u = \left\{ \frac{e^4}{32} + \frac{e^6}{32} \right\} \cdot \sin. 4nt + \frac{e^5}{16} \cdot (\sin. 5nt - \sin. 3nt) + \frac{e^6}{64} \cdot \\ & (6 \cdot \sin. 6nt - 2.4 \cdot \sin. 4nt + 2 \cdot \sin. 2nt); \frac{2\lambda^5}{5} \cdot \sin. 5u = \frac{e^5}{80} \cdot \sin. 5nt + \frac{e^6}{16} \cdot (\sin. 6nt - \\ & \sin. 4nt). \end{aligned}$$

If the several factors of  $\sin. nt$ ,  $\sin. 2nt$ , &c., be collected and arranged, they will give the respective terms of the value of  $v$ , for instance, the factors which multiply  $\sin. nt$ , are, taking into account the value of  $u$  which is given in page 145).

$$\begin{aligned} & (2e - \frac{e^3}{8} + \frac{e^3}{4} - \frac{e^3}{8} + \frac{e^5}{8} - \frac{e^3}{4} - \frac{e^5}{32} - \frac{e^5}{192} - \frac{e^5}{8} + \frac{e^5}{24} + \frac{e^5}{32} + \frac{e^5}{192}) \cdot \sin. nt. \\ & = (2e - \frac{e^3}{4} + \frac{5e^5}{96}) \cdot \sin. nt; \text{ the factors of } \sin. 2nt \text{ are } \frac{e^2}{2} + \frac{e^2}{4} + \frac{e^2}{2} + \frac{e^4}{6} + \\ & \frac{e^4}{6} + \frac{e^4}{8} + \frac{e^4}{12} + \frac{e^4}{8} - \frac{e^4}{4} - \frac{e^4}{8}, \text{ &c. See page 145.} \end{aligned}$$

If instead of fixing the origin of  $t$ , at the moment of the passage through the perihelium, we fix it at any instant whatever ; the angle  $nt$  will be increased by a constant quantity, which we will denote by  $\epsilon - \omega$ ; and consequently the preceding expressions of  $\frac{r}{a}$ , and of  $v$  will become

$$\frac{r}{a} = 1 + \frac{1}{2}e^2 - \left(e - \frac{3}{8}e\right) \cdot \cos.(nt + \epsilon - \omega) - \left(\frac{1}{2}e^2 - \frac{1}{3}e^4\right).$$

$$\cos. 2(nt + \epsilon - \omega) - \&c.;$$

$$v = nt + \epsilon + \left(2e - \frac{1}{4}e^3\right) \cdot \sin.(nt + \epsilon - \omega) + \left(\frac{5}{4}e^2 - \frac{11}{24}e^4\right).$$

$$\sin. 2(nt + \epsilon - \omega) + \&c.;$$

$v$  is the true longitude of the planet, and  $nt + \epsilon$  is its mean longitude, these two longitudes being referred to the plane of the orbit.

Let us now refer the motion of the planet, to a fixed plane, a little inclined to that of the orbit. Let  $\phi$  represent the mutual inclination of these two planes, and  $\theta$  the longitude of the ascending node of the orbit, reckoned on the fixed plane ; let  $\epsilon$  be this longitude reckoned on the fixed plane of the orbit, so that  $\theta$  is the projection of  $\epsilon$  ; also let  $v$  be the projection of  $v$  on the fixed plane. We shall have

$$\tan.(v - \theta) = \cos.\phi. \tan.(v - \epsilon).$$

This equation gives  $v$ , in terms of  $u$ , and vice versa ; but we can have these two angles, each in terms of the other, in very converging series, by the following method. The series

$$\frac{1}{2}v = \frac{1}{2}u + \lambda. \sin.u + \frac{\lambda^2}{2}. \sin.2u + \frac{\lambda^3}{3}. \sin.3u + \&c.$$

has been already deduced from the equation

$$\tan. \frac{1}{2}v = \sqrt{\frac{1+e}{1-e}} \cdot \tan. \frac{1}{2}u.$$

by making

$$\lambda = \frac{\sqrt{\frac{1+e}{1-e}} - 1}{\sqrt{\frac{1+e}{1-e}} + 1}.$$

If  $\frac{1}{2}v$ , be changed into  $v - \theta$ ;  $\frac{1}{2}u$  into  $u - \epsilon$ ; and  $\sqrt{\frac{1+e}{1-e}}$  into  $\cos \varphi$ ;

we shall have

$$\lambda = \frac{\cos. \varphi - 1}{\cos. \varphi + 1} = -\tan. \frac{\frac{1}{2}v - u}{2} \varphi; \dagger$$

the equation between  $\frac{1}{2}v$  and  $\frac{1}{2}u$ , will be changed into an equation be-

\* By making  $e$  negative in the equation  $r = \frac{a(1-e^2)}{1+e \cdot \cos. v}$ ,  $v$  will be equal to cipher, when  $r = a(1+e)$ , i. e. at the aphelion,  $\therefore$  it is from this point that the angle  $v$  is reckoned.

Since the results must be identically the same, when  $v$  is reckoned from perihelium and aphelion, and since the signs of the odd multiples are necessarily changed, in order that these expressions may remain the same as before, the sign of the factors which multiply these odd multiples, must be changed at the same time, i. e. these factors must be odd powers of  $e$ .

$$\dagger 1 - 2 \sin. \frac{\frac{1}{2}v - u}{2} \varphi = \cos. \varphi; \quad 2 \cos. \frac{\frac{1}{2}v - u}{2} \varphi - 1 = \cos. \varphi, \quad \therefore \frac{\cos. \varphi - 1}{\cos. \varphi + 1} = -\tan. \varphi.$$

$$\frac{\frac{1}{2}v - u}{2} \varphi, \frac{\sqrt{\frac{1+e}{1-e}} - 1}{\sqrt{\frac{1+e}{1-e}} + 1} = \frac{\sqrt{1+e} - \sqrt{1-e}}{\sqrt{1+e} + \sqrt{1-e}}, \text{ multiplying both numerator and denominator by}$$

$$\sqrt{1+e} + \sqrt{1-e}, \text{ we obtain after all reductions } \frac{2e}{2+2\sqrt{1-e^2}} = \frac{e}{1+\sqrt{1-e^2}} = \lambda;$$

tween  $v - \theta$  and  $v - \epsilon$ , and the preceding series will give

$$\begin{aligned} v - \theta &= v - \epsilon - \tan^2 \frac{1}{2}\phi \cdot \sin. 2(v - \epsilon) + \frac{1}{2} \cdot \tan. \frac{1}{2}\phi \cdot \sin. 4(v - \epsilon) \\ &\quad - \frac{1}{3} \cdot \tan. \frac{1}{2}\phi \cdot \sin. 6(v - \epsilon) + \text{&c.} \end{aligned}$$

If in the equation between  $\frac{1}{2}v$  and  $\frac{1}{2}u$ , we change  $\frac{1}{2}v$  into  $v - \epsilon$ ,  $\frac{1}{2}u$  into  $v - \theta$ , and  $\sqrt{\frac{1+e}{1-e}}$  into  $\frac{1}{\cos. \phi}$ ; we will obtain

$$\lambda = \tan. \frac{1}{2}\phi, *$$

and

$$\begin{aligned} v - \epsilon &= v - \theta + \tan. \frac{1}{2}\phi \cdot \sin. 2(v - \theta) + \frac{1}{2} \cdot \tan. \frac{1}{2}\phi \cdot \sin. 4(v - \theta) \\ &\quad + \frac{1}{3} \cdot \tan. \frac{1}{2}\phi \cdot \sin. 6(v - \theta) + \text{&c.} \end{aligned}$$

It is evident from an inspection of the two preceding series, that they may be converted one into the other, by changing the sign of the  $\tan. \frac{1}{2}\phi$ , and by changing the angles,  $v - \theta$ , and  $v - \epsilon$ , the one into the other. We will obtain  $v - \theta$ , in a function of the sines and cosines of the angle  $nt$  and its multiples, by observing that by what goes before, we have,

$$v = nt + \epsilon + eQ,$$

( $Q$  being a function of the sine of the angle  $nt + \epsilon - \omega$ , and of its multiples); and that the formula (*i*) of No. 21, gives, whatever may be the value of  $i$ ,

$2(v - \epsilon)$  being substituted for  $u$ , and observing that when  $-\tan. \frac{1}{2}$ .  $v$  is substituted for  $\lambda$ , the even multiples of two are positive, and the odd multiples negative, we obtain the expression which is given in the text.

$$* \text{ In this case } \frac{\frac{1}{\cos. \phi} - 1}{\frac{1}{\cos. \phi} + 1} = \frac{1 - \cos. \phi}{1 + \cos. \phi} = \frac{\sin. \frac{1}{2}\phi}{\cos. \frac{1}{2}\phi} = \tan. \frac{1}{2}\phi.$$

$$\begin{aligned}\sin i.(v-\epsilon) = \sin i.(nt+\epsilon-\epsilon+eQ) &= \left\{ 1 - \frac{i^2 e^2 Q^2}{1.2} + \frac{i^4 e^4 Q^4}{1.2.3.4} - \text{&c.} \right\}^* \\ &\quad . \sin i.(nt+\epsilon-\epsilon) \\ &\quad + \left\{ ieQ - \frac{i^3 e^3 Q^3}{1.2.3} + \frac{i^5 e^5 Q^5}{1.2.3.4.5} - \text{&c.} \right\} \\ &\quad . \cos i.(nt+\epsilon-\epsilon).\end{aligned}$$

Finally,  $s$  being the tangent of the latitude of the planet, above the fixed plane, we have

$$s = \tan \phi. \sin(v-\theta);$$

and if  $r$ , represents the radius vector  $r$  projected on the fixed plane, we shall have

$$r = r.(1+s^2)^{-\frac{1}{2}} = r(1 - \frac{1}{2}s^2 + \frac{3}{8}s^4 - \text{&c.}); \dagger$$

by this means we are enabled to determine  $v$ ,  $r$ , and  $s$  in converging series of sines, and cosines of the angle  $nt$ , and of its multiples.

23. Let us now consider the orbits which are very eccentric, such as are those of the comets; and for this purpose let the equations of No. 20, be resumed, namely

$$r = \frac{a.(1-e^2)}{1+e. \cos v};$$

$$nt = u - e. \sin u;$$

$$\tan \frac{1}{2}v = \sqrt{\frac{1+e}{1-e}}. \tan \frac{1}{2}u.$$

x 2

\* By the formulæ of No. 21, if the function  $\sin i.(nt+\epsilon-\epsilon)$  receive the increment  $ieQ$ , the value of this function so increased, will be the successive differential coefficients of  $\sin i.(nt+\epsilon-\epsilon)$ , (which are ultimately its sines and cosines) multiplied into the successive powers of  $ieQ$ , and divided by the products  $1.2.3\dots r$ ; and these terms being concinnated, give the expression in the text.

†  $s. \cot \phi = \sin(v-\theta)$ ,  $\therefore s = \tan \phi. \sin(v-\theta)$ ;  $r = r \cos \text{lat.} = \frac{r}{\sqrt{1+s^2}}$ ,  $= r(1 - \frac{1}{2}s^2 + \frac{3}{8}s^4 - \text{&c.})$ .

In the case of very excentric orbits,  $e$  differs very little from unity; therefore let us suppose,  $1 - e = \alpha$ ,  $\alpha$  being very small. If we name  $D$ , the perihelium distance of the comet; we shall have  $D = a(1 - e) = \alpha a$ ; therefore the expression of  $r$  will become

$$r = \frac{(2-\alpha).D}{2 \cdot \cos. \frac{^2}{2}v - \alpha \cdot \cos. v} = \frac{D}{\cos. \frac{^2}{2}v \cdot \left\{ 1 + \frac{\alpha}{2-\alpha} \tan. \frac{^2}{2}v \right\}};$$

which gives, by reducing into a series,

$$r = \frac{D}{\cos. \frac{^2}{2}v} \left\{ 1 - \frac{\alpha}{2-\alpha} \cdot \tan. \frac{^2}{2}v + \left\{ \frac{\alpha}{2-\alpha} \right\}^2 \cdot \tan. \frac{^4}{2}v - \text{&c.} \right\}$$

In order to have the ratio of  $v$  to the time  $t$ , we will observe that the expression of the arc in terms of the tangent, gives

$$u = 2 \cdot \tan. \frac{1}{2}u \cdot (1 - \frac{1}{3} \cdot \tan. \frac{^2}{2}u + \frac{1}{5} \cdot \tan. \frac{^4}{2}u - \text{&c.}) ; \dagger$$

but we have

$$\tan. \frac{1}{2}u = \sqrt{\frac{\alpha}{2-\alpha}} \cdot \tan. \frac{1}{2}v;$$

therefore we shall have

$$u = 2 \cdot \sqrt{\frac{\alpha}{2-\alpha}} \cdot \tan. \frac{1}{2}v \cdot \left\{ 1 - \frac{1}{3} \left( \frac{\alpha}{2-\alpha} \right) \cdot \tan. \frac{^2}{2}v + \frac{1}{5} \left( \frac{\alpha}{2-\alpha} \right)^2 \cdot \tan. \frac{^4}{2}v - \text{&c.} \right\}$$

$$\ast \quad e^2 = \alpha^2 - 2\alpha + 1; \therefore r = \frac{a(1-\alpha^2+2\alpha-1)}{1+2 \cdot \cos. \frac{^2}{2}v - 1-2\alpha \cdot \cos. \frac{^2}{2}v+\alpha} =$$

$$\frac{\alpha \cdot a \cdot (2-\alpha)}{2 \cdot \cos. \frac{^2}{2}v + \alpha(1-2 \cos. \frac{^2}{2}v)} = \frac{D \cdot (2-\alpha)}{2 \cdot \cos. \frac{^2}{2}v - \alpha \cdot \cos. \frac{^2}{2}v + \alpha \cdot \sin. \frac{^2}{2}v}$$

$= \frac{D \cdot (2-\alpha)}{\cos. \frac{^2}{2}v(2-\alpha) + \alpha \cdot \sin. \frac{^2}{2}v}$ ; dividing the numerator and denominator by  $2-\alpha$ , we obtain the expression in the text.

$$\dagger \quad \tan. \frac{1}{2}u = \frac{\sqrt{1-e}}{\sqrt{1+e}} \cdot \tan. \frac{1}{2}v = \frac{\sqrt{\alpha}}{\sqrt{2-\alpha}} \cdot \tan. \frac{1}{2}v;$$

we have likewise

$$\sin. u = \frac{2 \tan. \frac{1}{2}u}{1 + \tan. \frac{1}{2}u} = 2 \cdot \tan. \frac{1}{2}u \cdot (1 - \tan. \frac{1}{2}u + \tan. \frac{1}{2}u + \text{&c.}); *$$

from which may be obtained

$$e. \sin. u = 2(1 - \alpha) \cdot \sqrt{\frac{\alpha}{2 - \alpha}} \cdot \tan. \frac{1}{2}v \cdot \left\{ 1 - \frac{\alpha}{2 - \alpha} \cdot \tan. \frac{1}{2}v + \left\{ \frac{\alpha}{2 - \alpha} \right\}^2 \cdot \tan. \frac{1}{2}v + \text{&c.} \right\}$$

These values of  $u$  and of  $e. \sin. u$ , being substituted in the equation  $nt = u - e. \sin. u$ , will determine, in a very converging series, the time  $t$ , in a function of the anomaly  $v$ ; but previous to making this substitution, it may be observed that by No. 20,  $n = a \cdot \sqrt[3]{\mu}$ , and as  $D = \alpha a$ , we shall have

$$\frac{1}{n} = \frac{D^{\frac{2}{3}}}{\alpha^{\frac{2}{3}} \cdot \sqrt{\mu}}.$$

This being premised, we shall find

$$t = \frac{2D^{\frac{2}{3}}}{\sqrt{(2 - \alpha)\mu}} \cdot \tan. \frac{1}{2}v \cdot \left\{ 1 + \frac{\left\{ \frac{2}{3} - \alpha \right\}}{2 - \alpha} \cdot \tan. \frac{1}{2}v - \frac{\left\{ \frac{4}{5} - \alpha \right\} \cdot \alpha \tan. \frac{1}{2}v + \text{&c.}}{(2 - \alpha)^2} \right\} +$$

$$* \frac{2 \tan. \frac{1}{2}u}{1 + \tan. \frac{1}{2}u} = \frac{2 \sin. \frac{1}{2}u}{\cos. \frac{1}{2}u} = \frac{2 \sin. \frac{1}{2}u \cdot \cos. \frac{1}{2}u}{\sin. \frac{1}{2}u + \cos. \frac{1}{2}u} = \sin. u.$$

$$+ n = \frac{\sqrt{\mu}}{\alpha^{\frac{2}{3}}}, \quad \frac{D}{\alpha} = a, \quad \therefore n = \frac{\alpha^{\frac{2}{3}} \sqrt{\mu}}{D^{\frac{2}{3}}}; \quad t = \frac{u - e. \sin. u}{n} = \frac{D^{\frac{2}{3}}}{\alpha^{\frac{2}{3}} \sqrt{\mu}} \left\{ \frac{2 \sqrt{\alpha}}{\sqrt{2 - \alpha}} \right\}$$

$$(\tan. \frac{1}{2}v(1 - \frac{1}{3} \left\{ \frac{\alpha}{2 - \alpha} \right\}) \cdot \tan. \frac{1}{2}v + \frac{1}{5} \left\{ \frac{\alpha}{2 - \alpha} \right\}^2 \cdot \tan. \frac{1}{2}v + \text{&c.}) - 2(1 - \alpha) \sqrt{\frac{\alpha}{2 - \alpha}}.$$

If the orbit be parabolical,  $\alpha = 0$ , and consequently

$$r = \frac{D}{\cos. \frac{v_1}{2} v};$$

$$t = \frac{D^{\frac{3}{2}}}{\sqrt{\mu}} \sqrt{2} \cdot (\tan. \frac{1}{2} v + \frac{1}{3} \cdot \tan. \frac{3}{2} v). *$$

The time  $t$ , the distance  $D$ , and the sum  $\mu$  of the masses of the sun and of the comet, are heterogeneous quantities, and in order to render them comparable, they should be divided by the respective units of their species. Let therefore the mean distance of the sun from the earth represent the unity of the distance, so that  $D$  may be expressed in parts of this distance. It may then be remarked, that if  $T$  be called the time of a sidereal revolution of the earth, which we will suppose to depart from the perihelium, we shall have in the equation  $nt = u - e \cdot \sin. u$ ,  $u = 0$ , at the commencement of the revolution, and  $u = 2\pi$ , at its completion,  $\pi$  being the semicircumference of which the radius is unity; therefore we shall have  $nT = 2\pi$ ; but we have  $n = a^{-\frac{3}{2}} \cdot \sqrt{\mu} = \sqrt{\mu}$ , because  $a = 1$ ; therefore

$$\sqrt{\mu} = \frac{2\pi}{T}.$$

The value of  $\mu$  is not exactly the same, in the case of the earth and of the comet; since, in the first case, it expresses the sum of the masses of the

$$\tan. \frac{1}{2} v \left( 1 - \frac{\alpha}{2-\alpha} \cdot \tan. \frac{v_1}{2} v + \left\{ \frac{\alpha}{2-\alpha} \right\}^2 \cdot \tan. \frac{3}{2} v - \text{etc.} \right),$$

if the parts which destroy each other in this expression be obliterated, and if  $\alpha^{\frac{3}{2}}$  which occurs both in the numerator and denominator, of the part which remains, be likewise obliterated, the resulting quantity will be value of  $t$  given in the text.

\* It appears from this value of  $t$ , that the times in which different comets moving in parabolick orbits, describe equal angles about the sun placed in the focus, are in the sesquicuple ratio of the perihelium distance. See Newton, Prop. 37, Book 3, and also No. 27.

sun and earth ; in place of which, in the second case, it expresses the sum of the masses of the sun and comet ; but the masses of the earth, and of the comet, being much less than that of the sun, they may be neglected, and we may suppose that  $\mu$  is the same for all these bodies, and that it expresses the mass of the sun. Therefore by substituting in place of  $\sqrt{\mu}$ , its value  $\frac{2\pi}{T}$  in the preceding expression of  $t$ ; we shall have

$$t = \frac{D^{\frac{3}{2}} \cdot T}{\pi \cdot \sqrt{\frac{2}{2}}} \cdot (\tan. \frac{1}{2}v + \frac{1}{3} \cdot \tan. \frac{3}{2}v).$$

This equation contains no quantities which are not comparable with each other, it will easily determine  $t$ , whenever  $\mu$  will be known ; but in order to determine  $v$ , by means of  $t$ , we must solve an equation of the third degree which admits of but one real root. We may dispense with the resolution, by making a table of the values of  $v$ , corresponding to those of  $t$ , in a parabola of which the perihelium distance is equal to unity, or equal to the mean distance of the earth from the sun. This table will give the time which corresponds to the anomaly  $v$ , in any parabola of which  $D$  represents the perihelium distance, by multiplying by  $D^{\frac{3}{2}}$ , the time which answers to the same anomaly, in the table. We shall obtain the anomaly  $v$ , which answers to the time, by dividing  $t$  by  $D^{\frac{3}{2}}$ , and then seeking in the table, the anomaly which answers to the quotient of this division.

Let us now suppose that the anomaly  $v$ , which corresponds to the time  $t$ , in a very eccentric ellipse, is required. If quantities of the order  $\alpha^2$  be neglected, and of  $1-e$  be substituted, instead of  $\alpha$ ; the preceding ex-

When this equation is reduced to an original form there will be only one mutation of sign ;  $\therefore$  there will be only one real and affirmative root ; when  $u$  and  $D$  are given,  $r$  and  $t$  may be obtained immediately by the solution of a simple equation.

pression of  $t$  in  $v$ , in the ellipse, will give

$$t = \frac{D^{\frac{3}{2}} \cdot \sqrt{2}}{\sqrt{\mu}} \left\{ \tan. \frac{1}{2}v + \frac{1}{3} \cdot \tan. \frac{3}{2}v + (1-e) \cdot \tan. \frac{1}{2}v \cdot (\frac{1}{4} - \frac{1}{4} \cdot \tan. \frac{3}{2}v) v - \frac{1}{5} \cdot \tan. \frac{4}{2}v \right\}^*$$

We should seek, in the table of the motion of comets, the anomaly which answers to the time  $t$ , in a parabola of which  $D$  represents the perhelium distance; let  $U$  represent this anomaly,  $U+x$  being the true anomaly in the ellipse, corresponding to the same time,  $x$  being a very small angle. If we substitute in the preceding equation  $U+x$  in place of  $v$ , and then reduce the second member of this equation into a series arranged according to the powers of  $x$ ; we shall obtain by neglecting the square of  $x$ , and the product of  $x$  into  $1-e$ ,

$$t = \frac{D^{\frac{3}{2}} \cdot \sqrt{2}}{\sqrt{\mu}} \left\{ (\tan. \frac{1}{2}U + \frac{1}{3} \cdot \tan. \frac{3}{2}U) + \frac{x}{2 \cdot \cos. \frac{4}{2}U} + \frac{1-e}{4} \cdot \tan. \frac{1}{2}U \cdot (1 - \tan. \frac{3}{2}U - \frac{1}{5} \cdot \tan. \frac{4}{2}U) \right\};$$

$$* \quad \frac{2D^{\frac{3}{2}}}{\sqrt{\mu} \cdot (2-\alpha)} = \frac{2D^{\frac{3}{2}}}{\sqrt{\mu}} \left\{ 2^{-\frac{1}{2}} + \frac{2^{-\frac{3}{2}} \cdot \alpha}{2} \right\} \quad (\text{neglecting the square and higher powers}$$

$$\text{of } \alpha) = \frac{\sqrt{2} \cdot D^{\frac{3}{2}}}{\sqrt{\mu}} \cdot \left\{ 1 + \frac{\alpha}{4} \right\}, \quad \therefore \text{the value of } t \text{ becomes} =$$

$$\begin{aligned} & \frac{\sqrt{2} \cdot D^{\frac{3}{2}}}{\sqrt{\mu}} \cdot \left( 1 + \frac{\alpha}{4} \right) \cdot \tan. \frac{1}{2}v \left\{ 1 + \left( \frac{1}{3} - \frac{\alpha}{3} \right) \cdot \tan. \frac{3}{2}v - (\frac{1}{4} - \alpha) \cdot \alpha \cdot 2^{-2} \tan. \frac{4}{2}v \right\} \\ &= \frac{\sqrt{2} \cdot D^{\frac{5}{2}}}{\sqrt{\mu}} \left\{ \tan. \frac{1}{2}v + \frac{\alpha}{4} \cdot \tan. \frac{1}{2}v + \frac{1}{3} \cdot \tan. \frac{3}{2}v + \frac{\alpha}{4 \cdot 3} \cdot \tan. \frac{3}{2}v - \frac{\alpha}{3} \cdot \tan. \frac{4}{2}v - \right. \\ & \left. \frac{4\alpha}{5} \cdot 2^{-2} \cdot \tan. \frac{4}{2}v \right\} = \frac{\sqrt{2} \cdot D^{\frac{5}{2}}}{\sqrt{\mu}} \cdot (\tan. \frac{1}{2}v + \frac{1}{3} \cdot \tan. \frac{3}{2}v + (1-e) \cdot \tan. \frac{1}{2}v \cdot (\frac{1}{4} + (\frac{1}{12} - \frac{1}{3}) \cdot \tan. \frac{3}{2}v) - \frac{1}{2} \cdot \tan. \frac{4}{2}v); \quad 1-e \text{ being substituted for } \alpha. \end{aligned}$$

\* Substituting  $U+x$  for  $v$ ; this equation becomes

$$t = \frac{D^{\frac{3}{2}}}{\sqrt{\mu}} \sqrt{2} \cdot (\tan. \frac{1}{2}(U+x) + \frac{1}{3} \cdot \tan. \frac{3}{2}(U+x) + (1-e) \tan. \frac{1}{2}(U+x) \cdot (\frac{1}{3} - \frac{1}{4} \cdot \tan. \frac{3}{2}$$

but by hypothesis, we have

$$t = \frac{D^{\frac{3}{2}}\sqrt{2}}{\sqrt{\mu}} \cdot (\tan. \frac{1}{2}U + \frac{1}{3} \cdot \tan. \frac{3}{2}U); *$$

$$(U+x) - \frac{1}{3} \cdot \tan. {}^4(U+v)) = \frac{D^{\frac{3}{2}}}{\sqrt{\mu}} \sqrt{2} \left\{ \frac{\tan. \frac{1}{2}U + \tan. \frac{1}{2}x}{1 - \tan. \frac{U}{2} \cdot \tan. \frac{x}{2}} + \frac{1}{3} \left\{ \frac{\tan. \frac{1}{2}U + \tan. \frac{1}{2}x}{1 - \tan. \frac{U}{2} \cdot \tan. \frac{x}{2}} \right\}^3 \right. \\ \left. + \frac{(1-e) \cdot \tan. \frac{U}{2} + \tan. \frac{x}{2}}{1 - \tan. \frac{U}{2} \cdot \tan. \frac{x}{2}} \left\{ \frac{\frac{1}{4} \cdot \tan. \frac{U}{2} + \tan. \frac{x}{2}}{1 - \tan. \frac{U}{2} \cdot \tan. \frac{x}{2}} \right\}^2 - \frac{1}{3} \right. \\ \left. \left\{ \frac{\tan. \frac{U}{2} + \tan. \frac{x}{2}}{1 - \tan. \frac{U}{2} \cdot \tan. \frac{x}{2}} \right\}^4 \right\} = \frac{D^{\frac{3}{2}}\sqrt{2}}{\sqrt{\mu}} \cdot \left( \tan. \frac{U}{2} + \tan. \frac{x}{2} + \tan. \frac{x}{2} \cdot \tan. \frac{2U}{2} + \frac{1}{3}(\tan. \frac{3U}{2} \right. \\ \left. + 3 \tan. \frac{2U}{2} \cdot \tan. \frac{x}{2} + 3 \tan. \frac{4U}{2} \cdot \tan. \frac{x}{2}) + (1-e) \cdot \tan. \frac{U}{2} \left( \frac{1}{4} - \frac{1}{4} \tan. \frac{2U}{2} - \frac{1}{2} \right. \\ \left. \tan. \frac{4U}{2} \right) = \frac{D^{\frac{3}{2}}\sqrt{2}}{\sqrt{\mu}} \left( \tan. \frac{U}{2} + \frac{1}{3} \tan. \frac{3U}{2} + \tan. \frac{x}{2} \left( 1 + 2 \cdot \tan. \frac{2U}{2} + \tan. \frac{4U}{2} \right) \right. \\ \left. + \frac{1-e}{4} \cdot \tan. \frac{U}{2} \cdot (1 - \tan. \frac{2U}{2} - \frac{1}{3} \tan. \frac{3U}{2}) \right), \text{ and since } \tan. \frac{x}{2} = \frac{x}{2}, \text{ when } x^2, x^3, \right. \\ \left. \text{ &c. are rejected, and } 1 + 2 \tan. \frac{2U}{2} + \tan. \frac{4U}{2} = \left( 1 + \tan. \frac{2U}{2} \right)^2 = \frac{1}{\cos. \frac{4U}{2}}, \text{ by sub-} \right. \\ \left. \text{ stituting } \frac{x}{\cos. \frac{4U}{2}} \text{ for } \tan. \frac{x}{2} \cdot (1 + 2 \tan. \frac{2U}{2} + \tan. \frac{4U}{2}), \text{ we shall have the expres-} \right. \\ \left. \text{ sion given in the text.} \right)$$

\* Therefore the two last terms of the second member of this equation are equal to cipher, consequently  $\frac{x}{2 \cos. \frac{4U}{2}} = \frac{1-e}{4} \cdot \tan. \frac{U}{2} \left( -1 + \tan. \frac{2U}{2} + \frac{1}{3} \tan. \frac{4U}{2} \right); \because x \text{ or } \sin. x = \frac{1-e}{4} \cdot \tan. \frac{U}{2} \left( -2 \cdot \cos. \frac{4U}{2} + 2 \sin. \frac{2U}{2} \cdot \cos. \frac{2U}{2} + \frac{8}{3} \sin. \frac{4U}{2} \right), \text{ (by substituting} \\ \text{ for } \tan. \frac{U}{2} \text{ its value); } = \frac{1-e}{4} \left( \tan. \frac{U}{2} \left( -2 \cdot \cos. \frac{4U}{2} + 2 \cos. \frac{2U}{2} - 2 \cos. \frac{4U}{2} + \frac{8}{3} \right) \right)$

therefore by substituting in place of the small arc  $x$ , its sine, we shall obtain

$$\sin. x = \frac{1}{10} \cdot (1-e) \cdot \tan. \frac{1}{2}U \cdot (4-3 \cdot \cos. \frac{1}{2}U - 6 \cdot \cos. \frac{1}{2}U).$$

Thus, by constructing a table of the logarithms of the expression,

$$\frac{1}{10} \cdot \tan. \frac{1}{2}U \cdot (4-3 \cdot \cos. \frac{1}{2}U - 6 \cdot \cos. \frac{1}{2}U);$$

it will be sufficient to add to them the logarithm of  $1-e$ , in order to obtain that of  $\sin. x$ ; consequently if this correction be made to the anomaly  $U$ , computed for the parabola, we will have the corresponding anomaly in a very eccentric ellipse.

24. It remains for us to consider the motion in an hyperbolic orbit. For this purpose, it may be observed that in the hyperbola, the semi-axis major  $a$  becomes negative, and the excentricity  $e$  surpasses unity. If therefore in the equation ( $f$ ) of No. 20, we make  $a=-a'$ , and  $u=\frac{u'}{\sqrt{-1}}$ , and then substitute in place of the sines and cosines, their values in imaginary exponentials; the first of these equations will give

$$\frac{t \cdot \sqrt{\mu}}{a^{\frac{3}{2}}} = \frac{e}{2} \cdot (e^{u'} - e^{-u'}) - u'.$$

$$\begin{aligned} \left(1 - 2 \cos. \frac{^2U}{2} + \cos. \frac{^4U}{2}\right) &= \frac{1-e}{4} \cdot \tan. \frac{U}{2} \left(-4 \cdot \cos. \frac{^4U}{2} + 2 \cos. \frac{^2U}{2} + \frac{e}{2}\right) \\ \left(1 - 2 \cos. \frac{^2U}{2} + \cos. \frac{^4U}{2}\right) &= \frac{1-e}{4} \cdot \tan. \frac{U}{2} \cdot \left(\frac{e}{2} - \frac{2}{3} \cos. \frac{^4U}{2} - \frac{e}{3} \cdot \cos. \frac{^2U}{2} + \right. \\ &\quad \left. \frac{e}{3} \cos. \frac{^4U}{2}\right) \end{aligned}$$

= evidently the expression given in the text.

\*  $nt=u-e \sin. u$ , ( $n$  in this case =  $\frac{\sqrt{\mu}}{\sqrt{-a^3}}$ ) =  $\frac{\sqrt{\mu} \cdot i}{a^{\frac{3}{2}} \sqrt{-1}}$ ; ∵  $nt=-\frac{u'}{\sqrt{-1}}+e \cdot \sin. \frac{u'}{\sqrt{-1}}$ ;

The second will become

$$r = a \cdot (\frac{1}{2} \cdot e \cdot (c^u + c^{-u}) - 1); *$$

finally, if we make a corresponding change in the sign of the radical of the third equation, in order that  $v$  may increase with  $t$ , and consequently with  $u'$ ; we shall have

$$\tan. \frac{1}{2}v = \sqrt{\frac{e+1}{e-1}} \cdot \left\{ \frac{c^{u'}-1}{c^{u'}+1} \right\}. †$$

Let us suppose that in these formulæ,  $\tilde{u} = \log. \tan. (\frac{1}{4}\pi + \frac{1}{2}\varpi)$ ,  $\pi$  being the semicircumference of which the radius is equal to unity, and the preceding logarithm being hyperbolic; we shall have

$$\frac{t \cdot \sqrt{\mu}}{a'^{\frac{3}{2}}} = e \cdot \text{tang. } \varpi - \log. \tan. (\frac{1}{4}\pi + \frac{1}{2}\varpi); ‡$$

## ¶ 2

$$\sin. \frac{u'}{\sqrt{-1}} = \frac{c^{u'} - c^{-u'}}{2\sqrt{-1}}; \therefore \frac{\sqrt{\mu}}{a'^{\frac{3}{2}}} \cdot t = -u' + e, \frac{c^{u'} - c^{-u'}}{2}.$$

$$* r = a(1 - e \cos. u), \text{ becomes } r' = -a'(1 - e) \left( \frac{c^{u'} + c^{-u'}}{2} \right).$$

$$† \text{Tang. } \frac{v}{2} = \frac{\sqrt{1+e}}{\sqrt{1-e}} \cdot \frac{\sin. \frac{u'}{2\sqrt{-1}}}{\cos. \frac{u'}{2\sqrt{-1}}} = \frac{\sqrt{1+e}}{\sqrt{1-e}} \cdot \frac{\frac{u'}{2} - \frac{-u'}{2}}{\frac{u'}{2} + \frac{-u'}{2}} = \frac{\sqrt{e+1}}{\sqrt{e-1}}.$$

$$\frac{c^{u'} - 1}{c^{u'} + 1}.$$

$$‡ \text{Tang. } \left( \frac{\pi}{4} + \frac{\varpi}{2} \right) = c^{u'} \text{ and } \frac{1}{\tan. \left( \frac{\pi}{4} + \frac{\varpi}{2} \right)} = \cot. \left( \frac{\pi}{4} + \frac{\varpi}{2} \right) = e^{-u'}; \therefore$$

$$c^{u'} - c^{-u'} = \tan. \left( \frac{\pi}{4} + \frac{\varpi}{2} \right) - \cot. \left( \frac{\pi}{4} + \frac{\varpi}{2} \right) = \frac{\sin. \left( \frac{\pi}{4} + \frac{\varpi}{2} \right)}{\cos. \left( \frac{\pi}{4} + \frac{\varpi}{2} \right)} - \frac{\cos. \left( \frac{\pi}{4} + \frac{\varpi}{2} \right)}{\sin. \left( \frac{\pi}{4} + \frac{\varpi}{2} \right)}$$

$$r = a' \cdot \left\{ \frac{e}{\cos. \varpi} - 1 \right\};$$

$$\tan. \frac{1}{2}v = \sqrt{\frac{e+1}{e-1}} \cdot \tan. \frac{1}{2}\varpi.$$

The arc.  $\frac{t \sqrt{\mu}}{a'^{\frac{3}{2}}}$  is the mean angular motion of the body  $m$ , during the time  $t$ , supposed to move in a circle about  $M$ , at a distance equal to  $a'$ . This arc may easily be determined by reducing it into parts of the radius; the first of the preceding equations will give by trials, the value of the angle  $\varpi$ , corresponding to the time  $t$ ; the two other equations will then give the corresponding values of  $r$  and of  $v$ .

25. Expressing the sidereal revolution of a planet of which  $a$  is the mean distance from the sun; the first of the equations ( $f$ ) of No. 20,

will give  $T = 2\pi$ ; but by the same number we have  $\frac{\sqrt{\mu}}{a^{\frac{3}{2}}} = n$ ; there-

$$= \frac{\sin.^2\left(\frac{\pi}{4} + \frac{\varpi}{2}\right) - \cos.^2\left(\frac{\pi}{4} + \frac{\varpi}{2}\right)}{\sin.\left(\frac{\pi}{4} + \frac{\varpi}{2}\right) \cdot \cos.\left(\frac{\pi}{4} + \frac{\varpi}{2}\right)} = -2 \frac{\cos.\left(\frac{\pi}{2} + \varpi\right)}{\sin.\left(\frac{\pi}{2} + \varpi\right)} = -2 \cot.\left(\frac{\pi}{2} + \varpi\right) = 2.$$

$\tan. \varpi$ ;  $\therefore$  by substituting this expression for  $\frac{c^{u'} - c^{-u'}}{2}$ ; we obtain the value of

$$\frac{t \sqrt{\mu}}{a'^{\frac{3}{2}}} \text{ given in the text. } c^{u'} + c^{-u'} = \tan.\left(\frac{\pi}{4} + \frac{\varpi}{2}\right) + \cot.\left(\frac{\pi}{4} + \frac{\varpi}{2}\right) = 2 \sec^{\text{nt.}} \varpi$$

$$= \frac{2}{\cos. \varpi}; \therefore r' = a' \left( \frac{1}{2} e(c^u + c^{-u}) - 1 \right) = a' \left( \frac{e}{\cos. \varpi} - 1 \right); \tan. \frac{v}{2} = \frac{\sqrt{e+1}}{\sqrt{e-1}}.$$

$$\frac{\tan. \left( \frac{\pi}{4} + \frac{\varpi}{2} \right) - 1}{\tan. \left( \frac{\pi}{4} + \frac{\varpi}{2} \right) + 1} = \frac{\sqrt{e+1}}{\sqrt{e-1}} \cdot \frac{\sin. \left( \frac{\pi}{4} + \frac{\varpi}{2} \right) - \cos. \left( \frac{\pi}{4} + \frac{\varpi}{2} \right)}{\sin. \left( \frac{\pi}{4} + \frac{\varpi}{2} \right) + \cos. \left( \frac{\pi}{4} + \frac{\varpi}{2} \right)} = \frac{\sqrt{e+1}}{\sqrt{e-1}}.$$

$$\frac{2 \cos. \frac{\pi}{4} \cdot \sin. \frac{\varpi}{2}}{2 \sin. \frac{\pi}{4} \cdot \cos. \frac{\varpi}{2}} = \frac{\sqrt{e+1}}{\sqrt{e-1}} \cdot \cot. \frac{\pi}{4} \cdot \tan. \frac{\varpi}{2} = , (\text{as } \cot. \frac{\pi}{4} = 1), \frac{\sqrt{e+1}}{\sqrt{e-1}} \cdot \tan. \frac{\varpi}{2}.$$

fore we shall have

$$T = \frac{2\pi \cdot a^{\frac{3}{2}}}{\sqrt{\mu}}.$$

If the masses of the planets, relatively to that of the sun, be neglected ;  $\mu$  will express the mass of this star, and this quantity will be the same for all the planets ; thus, for a second planet, of which  $a'$  and  $T'$  express the mean distances from the sun, and the time of the sidereal revolution ; we shall have in like manner

$$T' = \frac{2\pi \cdot a'^{\frac{3}{2}}}{\sqrt{\mu}};$$

consequently we shall have

$$T^2 : T'^2 :: a^3 : a'^3;$$

that is to say, the squares of the times of the revolutions of different planets, are to each other, as the cubes of the greater axes of their orbits ; this is one of the laws discovered by Kepler. It appears from the preceding analysis, that this law is not rigorously true, and that it only obtains when we neglect the action of the planets, on each other, and on the sun.

If we assume for the measure of the time, the mean motion of the earth, and for the unit of distance, its mean distance from the sun ;  $T$  will in this case be equal to  $2\pi$ , and we will have  $a = 1$  ; therefore the preceding expression for  $T'$  will give  $\mu = 1$  ; from which it follows that the mass of the sun ought then to be taken for the unity of mass. We can thus, in the theory of the planets and of the comets, suppose  $\mu = 1$ , and assume for the unity of distance, the mean distance of the earth from the sun ; but then, the time  $t$  is measured by corresponding arc of the mean sidereal motion of the earth.

The equation

$$T = \frac{2\pi \cdot a^{\frac{3}{2}}}{\sqrt{\mu}}$$

enables us to determine, in a very simple manner, the ratios of the masses of the planets which are accompanied by satellites, to the mass of the sun. In fact,  $M$  representing this mass, if we neglect the mass  $m$  of the planet relatively to that of  $M$ ; we shall have

$$T = \frac{2\pi \cdot a^{\frac{3}{2}}}{\sqrt{M}}.$$

If we afterwards consider a satellite of any planet  $m'$ ; and if  $p$  represent the mass of this satellite, and  $h$  its mean distance from the centre of  $m'$ , and  $T$ , the time of its sidereal revolution, we shall have

$$T = \frac{2\pi \cdot h^{\frac{3}{2}}}{\sqrt{m'+p}};$$

therefore,

$$\frac{m'+p}{M} = \frac{h^3}{a^3} \cdot \left( \frac{T}{T} \right)^2.$$

This equation gives the ratio of the sum of the masses of the planet  $m'$  and of its satellite, to the mass  $M$  of the sun; if therefore the mass of the satellite be neglected in comparison with that of its primary, or if we suppose that the ratio of these masses is known; we will obtain the value of the mass of the planet, to that of the sun. We will give, in the theory of the planets, the values of the masses of the planets about which satellites have been observed to revolve.

## CHAPTER IV.

*Determination of the elements of Elliptic Motion.*

26. After having treated of the general theory of elliptic motion, and of the mode of computing it by converging series, in the two cases of nature, namely, in that of orbits very nearly circular, and in the case of very eccentric orbits; it now remains for us to determine the elements of these orbits. If the circumstances of the primitive motions of the heavenly bodies were given, we could easily deduce the elements from them. In fact, if we name  $V$  the velocity of  $m$ , in its relative motion about  $M$ ; we shall have

$$V^2 = \frac{dx^2 + dy^2 + dz^2}{dt^2};$$

and the last of the equations ( $p$ ) of No. 18, will give

$$V^2 = \mu \cdot \left\{ \frac{2}{r} - \frac{1}{a} \right\}.$$

In order to make  $\mu$  to disappear from this expression; let  $U$  denote the velocity which  $m$  would have, if it described about  $M$ , a circle of which the radius is equal to the unity of distance. In this hypothesis, we have  $r=a=1$ , and consequently  $U^2=\mu$ ; therefore

$$V^2 = U^2 \cdot \left\{ \frac{2}{r} - \frac{1}{a} \right\}.$$

This equation will give the semiaxis major  $a$ , of the orbit, by means of the primitive velocity of  $m$ , and of its primitive distance from  $M$ .  $a$  is positive in the ellipse; it is infinite in the parabola, and negative in

the hyperbola ; therefore the orbit described by  $m$ , is an ellipse, a parabola, or an hyperbola, according as  $V$  is less, equal to or greater than  $U \cdot \sqrt{\frac{2}{r}}$ . It is remarkable that the direction of the primitive motion, does not at all influence the species of conic section.\*

In order to determine the excentricity of the orbit, it may be observed, that if  $\epsilon$  represent the angle which the direction of the relative motion of  $m$ , makes with the radius vector  $r$ ; we have  $\frac{dr^2}{dt^2} = V^2 \cdot \cos \epsilon$ .

\*. By substituting in place of  $V^2$ , its value  $\mu \cdot \left\{ \frac{2}{r} - \frac{1}{a} \right\}$ , we shall have

$$\frac{dr^2}{dt^2} = \mu \cdot \left\{ \frac{2}{r} - \frac{1}{a} \right\} \cdot \cos \epsilon; \dagger$$

\* From the equation  $\frac{1}{a} = \frac{2}{r} - \frac{V^2}{U^2}$ , it appears that when  $V$  and  $r$  are given, the axis major and therefore the periodic time are constantly the same. Hence since  $U \cdot \sqrt{\frac{1}{r}} =$  the velocity in a circle at the same distance, it follows that in the ellipse the velocity at any point is to that in a circle at the same distance in a less ratio than that of  $\sqrt{2} : 1$ ; in a parabola, it is in the ratio of  $\sqrt{2} : 1$ ; and in the hyperbola it is in a greater ratio than that of  $\sqrt{2} : 1$ . See Princip. Math. Prop. 16. In the ellipse when the velocity of projection diminishes, the distance increases, and when  $V$  vanishes,  $r$  becomes equal to  $2a$ , in this case the excentricity  $e$  becomes equal to unity. In the hyperbola, the limit of the velocity, when  $r$  is infinite, is  $U^2 \frac{1}{a} =$  the velocity in a circle, at the distance of a transverse semiaxis from focus.

It is also manifest that when the distance is equal to the semiaxis major, the velocity is equal to that in a circle at the same distance, and that in general the velocity in an ellipse, is to the velocity in a circle at the same distance in the subduplicate ratio of the distance from the other focus to the semiaxis ; for it is as  $\sqrt{2a-r} : \sqrt{a}$ .

+  $\frac{dr}{dt} =$  the velocity resolved in the direction of the radius,  $\therefore$  it is equal to  $V$ , multiplied into the cosine of the angle which the radius vector makes with the curve or tangent, i. e. it is equal to  $V \cdot \cos \epsilon$ .

but by No. 19, we have

$$2\mu r - \frac{\mu r^2}{a} - \frac{r^2 \cdot dr^2}{dt^2} = \mu a(1-e^2);$$

therefore we shall have

$$a(1-e^2) = r^2 \cdot \sin. {}^2 \varepsilon. \left\{ \frac{2}{r} - \frac{1}{a} \right\};$$

by means of this equation, we can determine  $ae$  the excentricity of the orbit.

From the polar equation of a conic section, namely

$$r = \frac{a(1-e^2)}{1+e \cdot \cos. v},$$

we obtain

$$\cos. v = \frac{a(1-e^2)-r}{er}.$$

Substituting for  $\frac{dr^2}{dt^2}$  its value, we shall have  $2\mu.r - \frac{\mu r^2}{a} - \mu \left( \frac{2}{r} - \frac{1}{a} \right) r^2 \cdot \cos. {}^2 \varepsilon = \mu a(1-e^2)$ ,  $\therefore (2r - \frac{r^2}{a}) \cdot (1 - \cos. {}^2 \varepsilon) = a(1-e^2) =$  the parameter; hence it appears that when the distance and axis major are given, the parameter varies as the square of the sine of projection. Since  $V^2 = \frac{dr^2 + r^2 \cdot dv^2}{dt^2}$ , see page 4,  $a(1-e^2) = r^2 \cdot \frac{r^2 \cdot dv^2}{dt^2}$ ,  $\therefore$ .

the parameter depends on that part of the velocity which acts perpendicularly to the radius vector, it is termed the paracentric velocity, and it is evidently a maximum at the extremity of the focal ordinate.

From the expression  $a(1-e^2) = r^2 \cdot \sin. {}^2 \varepsilon \left( \frac{2}{r} - \frac{1}{a} \right)$ , it follows that  $\sin. {}^2 \varepsilon$  varies inversely as  $r \left( \frac{2a-r}{a} \right)$ , but the sum of the two factors is given, being equal to  $2a$ ,  $\therefore$  the product is a maximum, and consequently the sine of projection is the least possible, when the distance from the focus is equal to the semiaxis major.

We shall thus obtain the angle  $v$ , which the radius vector  $r$  constitutes with the perihelion distance, consequently we have the position of the perihelion. The equations ( $f$ ) of No. 20, will make known the angle  $u$ , and by means of it, the instant of the passage through the perihelion.

In order to determine the position of the orbit, with respect to a fixed plane passing through the centre of  $M$ , supposed immovable; let  $\phi$  represent the inclination of the orbit on this plane, and  $\epsilon$  the angle which the radius  $r$  constitutes with the line of the nodes; moreover let  $z$  be the primitive elevation of  $m$ , above the fixed plane, which elevation we suppose to be known; we shall have

$$r \cdot \sin. \epsilon \cdot \sin. \phi = z;$$

so that the inclination  $\phi$  of the orbit will be known, when we shall have determined  $\epsilon$ . For this purpose, let  $\lambda$  represent the angle, which the primitive direction of the relative motion of  $m$ , makes with the fixed plane, which angle we suppose to be known; if we consider the triangle formed by this direction produced to meet the line of the nodes, by this last line, and by the radius  $r$ ;  $l$  representing the side of the triangle which is opposed to the angle  $\epsilon$ , we shall have

$$l = \frac{r \cdot \sin. \epsilon}{\sin. (\epsilon + \varepsilon)};$$

we have also  $\frac{z}{l} = \sin. \lambda$ ; therefore we shall have

$$\tan. \epsilon = \frac{z \cdot \sin. \varepsilon}{r \cdot \sin. \lambda - z \cdot \cos. \varepsilon}.*$$

\*  $r \cdot \sin. \epsilon = a$  perpendicular let fall from the extremity of  $r$ , on the line of the nodes, and  $z =$  this perpendicular multiplied into the sine of  $\phi$ . The supplement of the angle which the primitive direction makes with the line of the nodes  $= \epsilon + \varepsilon$ ,  $\therefore l : r :: \sin. \varepsilon : \sin. (\varepsilon + \varepsilon)$ ;

$$\therefore l = \frac{r \cdot \sin. \varepsilon}{\sin. \varepsilon \cdot \cos. \varepsilon + \sin. \varepsilon \cdot \cos. \varepsilon} = \frac{r \cdot \tan. \varepsilon}{\sin. \varepsilon + \tan. \varepsilon \cdot \cos. \varepsilon} = \frac{z}{\sin. \lambda}, \therefore (r \cdot \sin. \lambda - z \cdot \cos. \varepsilon).$$

$$\tan. \epsilon = z \cdot \sin. \varepsilon.$$

The elements of the orbit of the planet being determined by these formulæ, in functions of the coordinates  $r$  and  $z$ , of the velocity of the planet and of the direction of its motion ; the variations of these elements, corresponding to the variations which are supposed to take place in its velocity and in its direction may be obtained ; it will be easy, by the methods which will be given in the sequel, to infer the differential variations of these elements, arising from the action of disturbing forces.

Let us resume the equation

$$V^2 = U^2 \cdot \left\{ \frac{2}{r} - \frac{1}{a} \right\}.$$

In the circle  $a=r$ , and consequently  $V=U\sqrt{\frac{1}{r}}$ ; from which it appears, that the velocities of the planets in different circles are reciprocally as the square roots of their radii.

In the parabola,  $a=\infty$ ,  $\therefore V=U\sqrt{\frac{2}{r}}$ ; therefore the velocities in different points of the orbit, are in this case reciprocally as the square roots of the radii vectores, and the velocity in each point is to that which the planet would have, if it described a circle whose radius was equal to the radius vector  $r$ , as  $\sqrt{2} : 1$ .

An ellipse, of which the minor axis is indefinitely small, is changed into a right line ; and in this case,  $V$  expresses the velocity of  $m$ , if it descended in a right line towards  $M$ . Let us suppose that  $m$  sets out from a state of repose, and that its primitive distance from  $M$  is  $r$  ; let us moreover suppose, that having attained the distance  $r'$ , it has acquired the velocity  $V'$ ; the preceding expression for the velocity, will give the two following equations :

$$0 = \frac{2}{r} - \frac{1}{a}; \quad V'^2 = U^2 \cdot \left\{ \frac{2}{r'} - \frac{1}{a} \right\};$$

from which we obtain

$$V = U \cdot \sqrt{\frac{2 \cdot (r - r')}{rr'}};$$

this is the expression of the relative velocity acquired by  $m$ , in departing from the distance  $r$ , and in falling towards  $M$ , through the height  $r - r'$ . We can easily determine by means of this formula, from what height a body  $m$ , which moves in a conic section, ought to fall towards  $M$ , in order to acquire, in departing from the extremity of the radius vector  $r$ , a relative velocity equal to that which it has at this extremity ; for  $V$  expressing this last velocity, we have

$$V^2 = U^2 \cdot \left\{ \frac{2}{r} - \frac{1}{a} \right\};$$

but the square of the velocity acquired in falling through the height  $r - r'$ , is  $\frac{2U^2 \cdot (r - r')}{rr'}$ ; by equating these two expressions, we shall have

$$r - r' = \frac{r \cdot (2a - r)}{4a - r}.$$

\* By equating these expressions we have  $\frac{2a - r}{ar} = \frac{2r - 2r'}{rr'}$ ,  $\therefore$   
 $(2a - r)r' = 2a(r - r')$ ; and consequently  $(4a - r)r' = 2ar$ ;  $\therefore r' = \frac{2ar}{4a - r}$ , and  $r - r' = \frac{2ar - r^2}{4a - r}$ , in the ellipse  $4a - r$  is greater than twice  $2a - r$ ,  $\therefore r - r'$  is less than  $\frac{r}{2}$ ; in the parabola  $a$  being infinite,  $r - r' = \frac{r}{2}$ , in the hyperbola  $r - r' = \frac{r(2a + r)}{4a + r}$ , and as in this case  $4a + r$  is less than twice  $2a + r$ ,  $r - r'$  is greater than  $\frac{r}{2}$ .

In order to determine the space through which a body must fall externally, so that it may acquire the velocity which it has in a conic section,  $r' - r$  must be substituted for  $r - r'$ , and then we equate  $\frac{2(r' - r)}{rr'}$  to  $\frac{2a - r}{ar}$ , from which we obtain  $2ar' - 2ar =$

In the circle  $a=r$ , and then  $r-r'=\frac{1}{2}r$ ; in the ellipse, we have  $r-r' < \frac{1}{2}r$ ;  $a$  being infinite in the parabola, we have  $r-r'=\frac{1}{2}r$ ; and in the hyperbola, in which  $a$  is negative, we have  $r-r' > \frac{1}{2}r$ .

27. The equation

$$0 = \frac{dx^2 + dy^2 + dz^2}{dt^2} - \mu \cdot \left\{ \frac{2}{r} - \frac{1}{a} \right\}$$

is remarkable, in that it determines the velocity independently of the eccentricity of the orbit. It is contained in a more general equation, which exists between the axis major of the orbit, the chord of the elliptic arc, the sum of its extreme radii vectores, and the time employed to describe this arc. In order to arrive at this last equation, we will resume the equations of elliptic motion, which have been given in No. 20;  $\mu$  being supposed for the sake of simplicity equal to unity. These equations will consequently become

$$\begin{aligned} r &= \frac{a(1-e^2)}{1+e \cdot \cos v}; \\ r &= a(1-e \cdot \cos u); \\ t &= a^{\frac{3}{2}}(u-e \cdot \sin u). \end{aligned}$$

Let us suppose that  $r$ ,  $v$ , and  $t$  correspond to the first extremity of the elliptic arc, and that  $r'$ ,  $v'$ , and  $t'$  correspond to the other extremity; we will have

$$\begin{aligned} r' &= \frac{a(1-e^2)}{1+e \cdot \cos v'}; \\ r' &= a(1-e \cdot \cos u'); \\ t' &= a^{\frac{3}{2}}(u'-e \cdot \sin u'). \end{aligned}$$

$$\text{Let } t-t=T; \quad \frac{u'-u}{2}=\epsilon; \quad \frac{u'+u}{2}=\epsilon'; \quad r+r=R;$$

$2ar'-r$ ,  $\therefore$  in an ellipse  $r'$  is  $=$  to the axis major, in a circle it is  $=$  to the diameter, it is infinite in the parabola; and in the hyperbola  $r'$  becomes  $= -2a$ .

subtracting the expression of  $t$ , from that of  $t'$ , and observing at the same time that

$$\sin. u' - \sin. u = 2. \sin. \epsilon. \cos. \epsilon';$$

we shall have

$$T = 2a^{\frac{3}{2}}. (\epsilon - e. \sin. \epsilon. \cos. \epsilon').$$

If we add together the two expressions of  $r$  and of  $r'$  in terms of  $u$  and  $u'$ , and if we observe that

$$\cos. u' + \cos. u = 2 \cos. \epsilon. \cos. \epsilon';$$

we shall have

$$R = 2a.(1 - e. \cos. \epsilon. \cos. \epsilon).$$

Now, let  $e$  represent the chord of the elliptic arc, we have

$$c^2 = r^2 + r'^2 - 2rr'. \cos. (v - v');$$

but from the two equations

$$r = \frac{a.(1-e^2)}{1+e. \cos. v}; \quad r = a.(1-e. \cos. u),$$

we obtain

$$\cos. v = \frac{a.(\cos. u - e)}{r}; \quad \sin. v = \frac{a\sqrt{1-e^2}. \sin. u}{r}.$$

In like manner we have

$$\cos. v' = \frac{a.(\cos. u' - e)}{r'}; \quad \sin. v' = \frac{a\sqrt{1-e^2}. \sin. u'}{r'};$$

therefore we shall have

$$rr'. \cos. (v - v') = a^2. (e - \cos. u). (e - \cos. u') + a^2. (1 - e^2). \sin. u. \sin. u'^2;$$

and consequently

$$c^2 = 2a^2.(1 - e^2).(1 - \sin. u. \sin. u' - \cos. u. \cos. u') \\ + a^2e^2.(\cos. u - \cos. u')^2;$$

but we have

$$\begin{aligned}\sin. u \cdot \sin. u' + \cos. u' \cdot \cos. u &= 2 \cos. {}^2 \epsilon - 1; \\ \cos. u - \cos. u' &= 2 \cdot \sin. \epsilon. \sin. \epsilon';\end{aligned}$$

therefore

$$c^2 = 4a^2 \cdot \sin. {}^2 \epsilon. (1 - e^2 \cdot \cos. {}^2 \epsilon'); *$$

consequently we will by this means obtain the three following equations,

$$\begin{aligned}R &= 2a \cdot (1 - e \cdot \cos. \epsilon. \cos. \epsilon'); \\ T &= 2a^{\frac{3}{2}} \cdot (\epsilon - e \cdot \sin. \epsilon. \cos. \epsilon'); \\ c^2 &= 4a^2 \cdot \sin. {}^2 \epsilon. (1 - e^2 \cdot \cos. {}^2 \epsilon').\end{aligned}$$

\*  $u' = \epsilon + \epsilon$ ,  $u = \epsilon - \epsilon$ ;  $\therefore \sin. u' = \sin. \epsilon' \cdot \cos. \epsilon + \sin. \epsilon. \cos. \epsilon'$ ;  $\sin. u = \sin. \epsilon' \cdot \cos. \epsilon - \sin. \epsilon. \cos. \epsilon'$ ;  $\therefore \sin. u' - \sin. u = 2 \cdot \sin. \epsilon. \cos. \epsilon'$ , hence  $T = t' - t = a^{\frac{3}{2}} \cdot (u' - u - e \cdot (\sin. u' - \sin. u))$  = the expression in the text.  $\cos. u' = \cos. \epsilon' \cdot \cos. \epsilon - \sin. \epsilon. \sin. \epsilon'$ ;  $\cos. u = \cos. \epsilon. \cos. \epsilon' + \sin. \epsilon. \sin. \epsilon'$ ,  $\therefore \cos. u + \cos. u' = 2 \cdot \cos. \epsilon' \cdot \cos. \epsilon$ , and  $r' + r = R = a \cdot (2 - e \cdot (\cos. u' + \cos. u))$  = the expression given in the text.  $v - v'$  is evidently equal to the angle contained between  $r$  and  $r'$ .

$$\begin{aligned}1 + e \cdot \cos. v &= \frac{a \cdot (1 - e^2)}{r}; \quad \therefore e \cdot \cos. v = \frac{a \cdot (1 - e^2) - a \cdot (1 - e \cdot \cos. u)}{r}, \text{ (by substituting for } \\ &\text{r its value)} = \frac{ae \cdot (\cos. u - e)}{r}; \quad \therefore \cos. v = \frac{a \cdot (\cos. u - e)}{r}; \quad \sin. {}^2 v = \\ &\frac{a^2 \cdot (1 - 2e \cdot \cos. u + e^2 \cdot \cos. {}^2 u) - \cos. {}^2 u + 2e \cdot \cos. u - e^2}{r^2} = \frac{a^2 \cdot (1 - \cos. {}^2 u - e^2 \cdot (\sin. {}^2 u))}{r^2} \\ &= \frac{a^2 \cdot (1 - e^2) \cdot \sin. {}^2 u}{r^2}, \text{ consequently } \cos. |v - v'| = \cos. v' \cdot \cos. v + \sin. v \cdot \sin. v' = \\ &\frac{a^2 \cdot (\cos. u - e) \cdot (\cos. u' - e) + a^2 \cdot (1 - e^2) \cdot \sin. u \cdot \sin. u'}{r^2}; \quad r^2 + r'^2 = a^2 \cdot (2 - 2e \cdot (\cos. u + \cos. u') + \\ &e^2 \cdot (\cos. {}^2 u + \cos. {}^2 u')), \quad \therefore r^2 + r'^2 - 2rr' \cdot \cos. (v - v') = a^2 \cdot (2 - 2e \cdot (\cos. u + \cos. u') + e^2 \cdot (\cos. {}^2 u + \cos. {}^2 u')) - 2a^2(e^2 - e \cdot (\cos. u + \cos. u') + \cos. u \cdot \cos. u') - 2a^2 \cdot \sin. u \cdot \sin. u' + 2a^2 \cdot e^2 \cdot \sin. u \cdot \sin. u' = \text{by reduction } 2a^2(1 - e^2) \cdot (1 - \sin. u \cdot \sin. u') - \cos. u \cdot \cos. u' + a^2 \cdot e^2 \cdot (\cos. {}^2 u + \cos. {}^2 u') - 2a^2 \cdot e^2 \cdot \cos. u \cdot \cos. u'. \\ &\cos. u \cdot \cos. u' + \sin. u \cdot \sin. u' = \cos. 2\epsilon = \cos. \epsilon^2 - \sin. \epsilon^2 = 2 \cdot \cos. {}^2 \epsilon - 1. \\ &\therefore c^2 = 2a^2(1 - e^2) \cdot (1 + 1 - 2 \cdot \cos. {}^2 \epsilon) + a^2 \cdot e^2 \cdot (4 \cdot \sin. {}^2 \epsilon \cdot \sin. {}^2 \epsilon') = 4a^2 \cdot (1 - e^2) \cdot (1 - \cos. {}^2 \epsilon) + 4a^2 \cdot e^2 \cdot \sin. {}^2 \epsilon \cdot \sin. {}^2 \epsilon' = 4a^2 \cdot (1 - e^2) \cdot \sin. {}^2 \epsilon + 4a^2 \cdot e^2 \cdot \sin. {}^2 \epsilon - 4a^2 \cdot e^2 \cdot \sin. \epsilon \cdot \cos. \epsilon' = 4a^2 \cdot (1 - e^2) \cdot \cos. \epsilon \cdot \cos. \epsilon'.\end{aligned}$$

The first of these equations gives

$$e \cdot \cos. \epsilon' = \frac{2a - R}{2a \cdot \cos. \epsilon} ;$$

by substituting this value of  $e \cdot \cos. \epsilon'$ , in the two others, we shall have

$$\begin{aligned} T &= 2a^{\frac{2}{3}} \left\{ \epsilon + \left( \frac{R - 2a}{2a} \right) \cdot \tan. \epsilon \right\} ; \\ c^2 &= 4a^2 \cdot \tan. {}^2 \epsilon \cdot \left\{ \cos. {}^2 \epsilon - \left( \frac{2a - R}{2a} \right)^2 \right\}. \end{aligned}$$

These two equations do not involve the excentricity  $e$ ; and if in the first we substitute in place of  $\epsilon$ , its value given by the second, we shall obtain  $T$  in a function of  $c$ ,  $R$  and  $a$ . It appears from this, that the time  $T$  depends only on the axis major, the chord  $c$ , and the sum  $R$  of the extreme radii vectores.

If we make

$$z = \frac{2a - R + c}{2a} ; \quad z' = \frac{2a - R - c}{2a} ,$$

the last of the preceding equations will give

$$\cos. 2\epsilon = zz' + \sqrt{(1-z^2)(1-z'^2)} ;$$

from which may be obtained,

$$2\epsilon = \text{arc. cos. } z' - \text{arc. cos. } z ;^*$$

\*  $\frac{c^2}{4a^2} = \sin. {}^2 \epsilon - \left( \frac{2a - R}{2a} \right)^2 \cdot \frac{\sin. {}^2 \epsilon}{\cos. {}^2 \epsilon}$ , let  $\frac{c^2}{4a^2} = n$ ,  $\frac{2a - R}{2a} = m$ , and as  $\sin. \epsilon = 1 - \cos. {}^2 \epsilon$ ,  $\therefore n \cdot \cos. {}^2 \epsilon = \cos. {}^2 \epsilon - \cos. {}^4 \epsilon - m^2 + m^2 \cdot \cos. {}^2 \epsilon$ ,  $\therefore \cos. {}^4 \epsilon + (n - m^2 - 1) \cdot \cos. {}^2 \epsilon = -m^2$ , and solving this equation  $\cos. {}^2 \epsilon = \frac{-(n - m^2 - 1)}{2} \pm \frac{\sqrt{(n - m^2 - 1)^2 - 4m^2}}{2}$ , and as  $\cos. 2\epsilon = 2 \cos. {}^2 \epsilon - 1$ , we have  $\cos. 2\epsilon = -n + m^2 \pm \sqrt{(n - m^2 - 1)^2 - 4m^2}$ , and substituting for  $n$  and  $m$ , we obtain;

arc. cos.  $z$  denotes here the arc, of which the cosine is  $z$ ; consequently we have

$$\tan. \epsilon = \frac{\sin. (\text{arc. cos. } z') - \sin. (\text{arc. cos. } z)}{z + z'} ;$$

we have likewise  $z + z' = \frac{2a - R}{a}$ ; therefore the expression of  $T$  will become, (by observing, that if  $T$  is the duration of the sidereal revolution of the earth, of which the mean distance from the sun is taken for unity, we have by No. 16,  $T = 2\pi$ );

$$T = \frac{a^2 \cdot T}{2\pi} \cdot (\text{arc. cos. } z' - \text{arc. cos. } z - \sin. (\text{arc. cos. } z') + \sin. (\text{arc. cos. } z)). \quad (a)$$

As the same cosine may appertain to several arcs, this expression of

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$$\frac{\sqrt{2a - R)^2 - c^2}}{4a^2} \pm \left( \frac{(c^4 + (2a - R)^4 - 2c^2 \cdot (2a - R)^2)}{(2a)^4} - \frac{2c^2 - 2 \cdot (2a - R)^2}{4a^2} + 1 \right)^{\frac{1}{2}} = \cos. 2\epsilon;$$

the part of this radical of which the denominator  $(2a)^4 = z^2 \cdot z'^2$ ;  
 for  $z^2 = \frac{(2a - R)^2 + c^2 + 2 \cdot c \cdot (2a - R)}{(2a)^2}$  and  $z'^2 = \frac{(2a - R)^2 + c^2 - 2 \cdot c \cdot (2a - R)}{(2a)^2}$ ; and the part of this radical of which the denominator is  $4a^2 = z^2 - z'^2 = -(2a - R)^2 - c^2 - (2a - R)^2 - c^2 + 2c \cdot (2a - R) - 2c \cdot (2a - R)$ , the part without the radical is evidently equal to  $z'$ , ∴ by substituting we shall find the cosine of  $2\epsilon = zz' + \sqrt{z^2 z'^2 - z^2 - z'^2 + 1}$ , which is evidently equal to the expression given in the text.

Let  $z, z'$  represent the cosines of two arcs, and the cosine of the difference of these arcs will be  $= zz' + \sqrt{1 - z^2 \cdot (1 - z'^2)} = \cos. 2\epsilon$ ; ∴  $2\epsilon$  = the difference of two arcs of which the cosines are  $z$  and  $z'$ .

$$\sin. a - \sin. b = 2 \cdot \cos. \frac{(a+b)}{2} \cdot \sin. \frac{(a-b)}{2}, \cos. a + \cos. b = 2 \cdot \cos. \frac{(a+b)}{2} \cdot \cos. \frac{(a-b)}{2}$$

$$\therefore \frac{\sin. a - \sin. b}{\cos. a + \cos. b} = \frac{\sin. \frac{(a-b)}{2}}{\cos. \frac{(a-b)}{2}} = \tan. \frac{(a-b)}{2}, \text{ from this formula may be inferred the}$$

value of  $\tan. \epsilon$ , which is given in the text.

T is ambiguous, and it is necessary carefully to distinguish the arcs to which the cosines  $z$  and  $z'$  belong.

In the parabola, the semiaxis major  $a$  is infinite, and we have

$$\text{arc. cos. } z' - \sin. (\text{arc. cos. } z') = \frac{1}{6} \cdot \left( \frac{R+c}{a} \right)^{\frac{3}{2}} *$$

By making  $c$  negative, we obtain the value of the arc. cos.  $z - \sin. (\text{arc. cos. } z)$ ; the formula (a) will therefore give for the time T, employed to describe the arc subtended by the chord  $c$ ,

$$T = \frac{T}{12\pi} \cdot \left( (r + r' + c)^{\frac{3}{2}} - (r + r' - c)^{\frac{3}{2}} \right);$$

The sign — having place, when the two extremities of the parabolic

$$\begin{aligned} * \text{ Arc. cos. } z' - \sin. \text{arc. cos. } z' &= \text{arc. sin. } \sqrt{1-z'^2} - \sqrt{1-z^2} = \text{by expressing the} \\ \text{arc. in terms of the sine, } \sqrt{1-z'^2} &+ \frac{(1-z'^2)^{\frac{3}{2}}}{2.3} + \text{ &c. } - \sqrt{1-z^2} \\ &= \left( \frac{(4a^2 - (2a-R)^2 + 2c(2a-R)-c^2)}{(2.3.4a^2)} \right)^{\frac{3}{2}} = \frac{((4Ra-R^2+4ac-2cR-c^2))^{\frac{3}{2}}}{(2.3.4a^2)} \\ &= \text{when } a \text{ is } \infty, \frac{(4a(R+c))^{\frac{3}{2}}}{(2.3.4a^2)}. \end{aligned}$$

In the expression for arc. sin.  $\sqrt{1-z'^2}$ ; the approximation is not continued beyond the second term, because the subsequent terms disappear in the value of T, when  $a$  is supposed to be infinite. The second term of the value of T vanishes when  $c$  passes through the focus, and T is less when the angle formed by  $r, r'$  is turned towards the perhelion, than when the second term vanishes, it is manifest that the sign of the second term must be in this case negative, and positive in every other case.

The second term of the second member of this equation vanishes when the extremities of the arc described, are bounded by the focal ordinates,  $\therefore$  the time of describing the parabolic arc intercepted between vertex and focal ordinate varies in the sesquuplicate ratio of the parameter. See Newton, Princip. Vol. 3, Lem. 9, 10. Indeed it appears from the value of T, that the time of describing any parabolic arc, of which the chord passes through the focus, varies in the sesquuplicate ratio of the chord.

arc are situated on the same side of the axis of the parabola, or when one of them being situated below, the angle formed by the two radii vectores, is turned towards the perihelion, it is necessary to make use of the sign + in every other case.  $T$  being equal to 365<sup>days</sup>, 25638, we have

$$\frac{T}{2\pi} = 9^{\text{days}}, \text{ 688754.}$$

In the hyperbola,  $a$  is negative ;  $z$  and  $z'$  become greater than unity ; the arcs, arc. cos.  $z$ , and arc. cos.  $z'$  are imaginary, and their hyperbolic logarithms are,

$$\text{arc. cos. } z = \frac{1}{\sqrt{-1}} \cdot \log. (z + \sqrt{z^2 - 1}) ;$$

$$\text{arc. cos. } z' = \frac{1}{\sqrt{-1}} \cdot \log. (z' + \sqrt{z'^2 - 1}) ;$$

consequently the formula (a) becomes by changing  $a$  into  $-a$ ,

$$T = \frac{a^{\frac{3}{2}} T}{2\pi} \cdot (\sqrt{z'^2 - 1} \mp \sqrt{z^2 - 1} - \log. (z' + \sqrt{z'^2 - 1}) \pm \log. (z + \sqrt{z^2 - 1}).$$

The formula (a) determines the time, of rectilinear descent of a body towards the focus, when it departs with a given velocity, from a given distance ; it is sufficient for this purpose, to suppose that the ellipse which it then describes, is infinitely compressed. If, for example, we suppose that the body departs from a state of rest, at the distance  $2a$  from the focus, and that the time  $T$ , which it employs to describe the distance  $c$  is sought ; in this case  $R = 2a + r$ ;  $r = 2a - c$  ; which gives  $z' = -1$ ;  $z = \frac{c-a}{a}$ ; the formula (a) will consequently give

$$T = \frac{a^{\frac{3}{2}} T}{2\pi} \left\{ \pi - \text{arc. cos.} \left( \frac{c-a}{a} \right) + \sqrt{\frac{2ac-c^2}{a^2}} \right\}.$$

There\* is, however an essential difference between the elliptic motion towards the focus, and the motion in an ellipse infinitely compressed. In the first case, the body arrives at the focus, passes beyond it, and elongates itself to the distance from which it commenced to move; in the second case the body having attained the focus, returns to the point from which it set out. A tangential velocity at the aphelion, ever so small, suffices to produce this difference, which does not influence the time employed by the body in descending towards the focus.

28. As the circumstances of the primitive motions of the heavenly bodies are not known from observation, the elements of their orbit cannot be determined by the formulæ of No. 26. In order to effect this object, we should compare together their respective positions observed at different epochs; which presents considerable difficulties, as these bodies are not observed from the centre of their motions. Indeed, with respect to the planets, we can, by means of their oppositions and conjunctions, obtain their longitude such as it would be observed from the centre itself of the sun; and this consideration, combined with the small excentricity, and small inclination of their orbits to the ecliptic, simplifies very much the determination of their elements. Besides,

$$* R=2a+r=4a-c, \therefore z' = \frac{2a-4a+c-c}{2a} = -1, z = \frac{2a-4a+2c}{2a} = \frac{c-2a}{2a};$$

$$\text{If } a \text{ be infinite } T = T \cdot \frac{a^{\frac{3}{2}}}{2\pi} \cdot (\pi - \pi + \sqrt{\frac{2ac}{a^2}}) = \frac{a^{\frac{3}{2}} \sqrt{c}}{\pi \cdot \sqrt{2} \cdot a^{\frac{3}{2}}} = T \frac{\sqrt{c}}{\sqrt{2} \pi};$$

$$\text{arc. cos. } \frac{c-a}{a} = \text{arc. sin.} = \sqrt{\frac{2ac-c^2}{a^2}}, \therefore \text{as } \pi - \text{arc. sin.} = \sqrt{\frac{2ac-c^2}{a^2}}, \text{ and arc.}$$

$\sin. = \sqrt{\frac{2ac-c^2}{a^2}}$  have the same sine,  $T$  varies in an ellipse as the arc — sin.; which agrees with Newton's conclusion; Princip. Math. Lib. 1. Prop. 37. See Prony Mécanique Analytique, Tom. 2. No. 914, and Euler's Mechanics, No. 272, 672.

If  $c=2a$  the time of falling to the centre will be equal to  $\frac{a^{\frac{3}{2}}T}{2\pi} \cdot \pi$ .

in the actual state of astronomy, the elements\* of these orbits require only very slight corrections ; and as the variations of the distances of the planets from the earth are not at any time sufficiently great to render them invisible to us, we can observe them perpetually, and by a comparison of a great number of observations, correct the elements of their orbits, and also the errors themselves to which the observations are liable. This method cannot be applied in the case of the comets, as they are only visible near their perihelion ; and if the observations which are made on them during the time of their appearance, are inadequate to the determination of their elements, we have not then any means of following these stars in imagination, through the immensity of space ; so that when the lapse of ages brings them back towards the sun, it is impossible for us to recognise them ; it is therefore of the greatest consequence to be able to determine by observations made during the time of the appearance of a comet, the elements of its orbits ; but the rigorous solution of this problem surpasses the powers of analysis, and we are obliged to recur to methods of approximation, in order to obtain the first values of these elements, which we can afterwards correct with all the precision which the observations admit of.

If we employ observations which are at a considerable distance from each other, the eliminations would lead to impracticable computations ; it is therefore necessary to restrict ourselves to the consideration of near observations ; and even with this restriction, the problem presents considerable difficulties. It has appeared to me, after mature reflection, that instead of employing directly the observations themselves, it would be more advantageous to deduce from them data, which offer a simple and exact result ; and I am satisfied that the geocentric latitude and longitude of the comet, at a given moment, and their first and second

\* In the present state of Astronomy, the motions of the planets may be considered as very accurately known, and the object of these observations is to determine them with still greater accuracy. And when the elements have been determined under the most favourable circumstances, *i.e.* in those in which they have the greatest influence, they should be afterwards corrected simultaneously, by the method of *the equations of condition.*

differences divided by corresponding powers of the element of the time, are those which best satisfy this condition ; for by means of these data, we can determine rigorously, and with facility, the elements, without having recourse to any integration, and by the sole consideration of the differential equations of the orbit. This mode of considering the problem permits us also to employ a great number of neighbouring observations, and by this means, to embrace a considerable interval between the extreme observations, which is very useful in diminishing the influence of the errors, to which these observations are always liable, in consequence of the nebulosity which surrounds the comets. I proceed now to present the formulæ, by means of which the first differences of the longitude and latitude may be deduced from any number of neighbouring observations ; I will afterwards determine the elements of the orbit of the comet by means of these differences. Finally, I will point out the means which have appeared to me the simplest, for correcting these elements, by three observations, made at a considerable distance from each other.

29. Let at any given epoch,  $\alpha$  be the geocentric longitude of a comet, and  $\theta$  its northern geocentric latitude, the southern latitudes being supposed negative. If we denote by  $s$ , the number of days which have elapsed since this epoch ; the geocentric longitude and latitude of the comet, after this interval, will be expressed in consequence of the formula (i) of No. 21, by the two following series,

$$\alpha + s \cdot \left( \frac{d\alpha}{ds} \right) + \frac{s^2}{1.2} \cdot \left( \frac{d^2\alpha}{ds^2} \right) + \frac{s^3}{1.2.3} \cdot \left( \frac{d^3\alpha}{ds^3} \right) + \&c;$$

$$\theta + s \cdot \left( \frac{d\theta}{ds} \right) + \frac{s^2}{1.2} \cdot \left( \frac{d^2\theta}{ds^2} \right) + \frac{s^3}{1.2.3} \cdot \left( \frac{d^3\theta}{ds^3} \right) + \&c;$$

The values of  $\alpha$ ,  $\left( \frac{d\alpha}{ds} \right)$ ,  $\left( \frac{d^2\alpha}{ds^2} \right)$ , &c ;  $\theta$ ,  $\left( \frac{d\theta}{ds} \right)$ , &c. may be determined by means of several observed geocentric longitudes and latitudes.

In order to obtain them in the simplest manner, let us consider the infinite series which expresses the geocentric longitudes. The coefficients of the powers of  $s$ , in this series, may be determined by the condition that it ought to represent each observed longitude, when we substitute for  $s$ , the number of days which corresponds to it; we shall by this means obtain as many equations as there are observations; and if the number of these last be  $n$ , we cannot determine by their means, in the infinite series, but  $n$  quantities  $\alpha, \left(\frac{d\alpha}{dt}\right), \text{ &c.}$

However, it ought to be observed, that  $s$  being supposed very small, we can neglect the terms\* multiplied by  $s^n, s^{n+1}, \text{ &c.}$ , so that the infinite series is reduced to the  $n$  first terms, which we are able to determine by the  $n$  observations. These determinations are only approximative, and their accuracy will depend on the smallness of the terms which we have neglected; they will be always more exact, in proportion to the smallness of  $s$ , and to the greater number of observations employed. Therefore the question is reduced to a problem in the theory of interpolations, namely to find an entire and rational function of  $s$ , of such a nature, that when we substitute for  $s$ , the number of days which correspond to each observation, this function is changed into the observed longitude.

Let  $\ell, \ell', \ell'', \text{ represent the observed longitudes of the comet, and}$

\* As the values of the differential coefficients in the series expanded according to the formula of No. 21, are independent of the value of the increments, these values will remain, when the increment varies; and there are as many series of the form  $\alpha + \frac{s}{1!} \left( \frac{d\alpha}{ds} \right) + \frac{s^2}{2!} \cdot \left( \frac{d^2\alpha}{ds^2} \right) + \text{ &c.}$  as there are observations; if  $s$  be very small, it may be proved that the terms of the series after the  $n$  first diminish very rapidly, and consequently may be neglected; and as there will remain but  $n$  terms, if we have  $n$  observations we have as many observations as unknown quantities; if the number of observations be increased, a greater number of coefficients can be determined, and if  $s$  become less, the value of the terms which are rejected will be less.

$i, i', i'', \dots$ , the number of days which intervene between them and the given epoch; these numbers ought to be supposed negative, for the observations anterior to this epoch. By making

$$\begin{aligned}\frac{\epsilon' - \epsilon}{i' - i} &= \delta\epsilon; \quad \frac{\epsilon'' - \epsilon}{i'' - i'} = \delta\epsilon'; \quad \frac{\epsilon''' - \epsilon''}{i''' - i'} = \delta\epsilon''; \quad \text{&c.;} \\ \frac{\delta\epsilon' - \delta\epsilon}{i' - i} &= \delta^2\epsilon; \quad \frac{\delta\epsilon'' - \delta\epsilon'}{i''' - i'} = \delta^2\epsilon'; \quad \text{&c.} \\ \frac{\delta^2\epsilon' - \delta^2\epsilon}{i''' - i} &= \delta^3\epsilon \quad \text{&c.} \\ &\quad \text{&c.};\end{aligned}$$

the function sought will be

$$\epsilon + (s-i).\delta\epsilon + (s-i).(s-i').\delta^2\epsilon + (s-i).(s-i').(s-i').\delta^3\epsilon + \text{&c.};$$

for it is easy to be assured, that if we make successively,  $s=i$ ,  $s=i'$ ,  $s=i''$ , &c. this function will be converted into  $\epsilon$ ,  $\epsilon'$ ,  $\epsilon''$ , &c.

Now, the comparison of the preceding function, with the following:

$$\alpha + s. \left( \frac{d\alpha}{ds} \right) + \frac{s^2}{1.2} \left( \frac{d^2\alpha}{ds^2} \right) + \text{&c.};$$

will give, by putting the coefficients of similar powers of  $s$  equal to each other,

$$\alpha - \epsilon - i.\delta\epsilon + i.i'.\delta^2\epsilon - i.i'.i'.\delta^3\epsilon + \text{&c.};$$

$$\left( \frac{d\alpha}{ds} \right) = \delta\epsilon - (i+i').\delta^2\epsilon + (ii'+ii''+i'i'').\delta^3\epsilon - \text{&c.}$$

$$\frac{1}{2} \cdot \left( \frac{d^2\alpha}{ds^2} \right) = \delta^2\epsilon - (i+i'+i'').\delta^3\epsilon + \text{&c.}; *$$

the ulterior differences of  $\alpha$  will be useless to us. The coefficients of

\* These equations evidently obtain from the principle of indeterminate coefficients, and it is manifest that the greater the number of observations the more accurately will they be determined, and the less  $i'$ ,  $i''$ ,  $i'''$ , &c. are, the more rapid will be the convergence of the series.

these expressions are alternately positive and negative ; the coefficient of  $\delta^r \epsilon$  is, abstracting from the sign, the product of  $r$  into  $r$ , of the  $r$  quantities  $i, i', i'', i^{(r-1)}$ , in the value of  $\alpha$  ; it is the sum of the products of the same quantities,  $r-1$ , into  $r-1$ , in the value of  $\left(\frac{d\alpha}{ds}\right)$  ; finally it is the sum of the products of the quantities,  $r-2$ , into  $r-2$ , in the value of  $\frac{1}{2} \cdot \left(\frac{d^2\alpha}{ds^2}\right)$ .

If  $\gamma, \gamma', \gamma'', \&c.$  represent the observed geocentric latitudes of the comet, we shall obtain the values of  $\theta, \left(\frac{d\theta}{ds}\right), \left(\frac{d^2\theta}{ds^2}\right) \&c.$ ; by changing in the preceding expressions for  $\alpha, \left(\frac{d\alpha}{ds}\right), \left(\frac{d^2\alpha}{ds^2}\right), \&c.$ , the quantities  $\epsilon, \epsilon', \epsilon'', \&c.$  into  $\gamma, \gamma', \gamma'', \&c.$

These expressions will be more exact, according as the number of observations is increased, and as the intervals which separate them, are less ; we could therefore\* employ all the neighbouring observations of the selected epoch, provided that they were exact ; but the errors to which they are always liable, would lead to an erroneous result ; therefore in order to diminish the influence of these errors, the interval between the extreme observations should be increased, in proportion as a greater number of observations is employed. We are able in this manner, with five observations, to embrace an interval of thirty-five or forty degrees, which ought to lead to very approximate values of the geocentric longitudes and latitudes, and of their first and second differences. If the epoch which we select, is such that there exists an equal number of observations before and after it, so that each longitude which follows, has a corresponding longitude which precedes it by the same interval ; this condition will render the

\* The number of observations will of itself produce an increase in the error,  $\therefore$  in order that the error may be distributed over a greater number of degrees, we must increase the interval between the extreme observations.

values of  $\alpha$ ,  $\left(\frac{d\alpha}{ds}\right)$ ,  $\left(\frac{d^2\alpha}{ds^2}\right)$  more accurate,\* and it is easy to perceive

\* When the observations are assumed at different sides of the epoch which is selected  $i'$ ,  $i'''$ ,  $i''''$ , &c. are negative when  $i$ ,  $i''$ ,  $i'''$ , &c. are positive, and vice versa. In the values of  $\alpha$ , which are given above, the terms after the first, are negative and positive in pairs and in the values of  $\frac{d\alpha}{ds}$ ,  $\frac{d^2\alpha}{ds^2}$ . the coefficients of  $d\zeta$ ,  $d^2\zeta$ , &c. are less than when all the observations are made at the same side of the selected epoch, ∴ the convergence of the terms will be more rapid, and the terms which are omitted are of less consequence.

Let the number of observations be odd, and  $= 2r+1$ , and let  $i$  be the number of days between each observation, and let the epoch from which we count be the instant of the mean observation when  $\alpha = \zeta^{(r)}$ , then we have

$$\begin{aligned} \frac{d\alpha}{ds} &= \frac{1}{2i} \cdot \left\{ \begin{array}{l} \Delta \zeta^{(r)} + \Delta \zeta^{(r-1)} - \frac{1}{1.2.3} \cdot (\Delta^3 \zeta^{(r-1)} + \Delta^3 \zeta^{(r-2)}) \\ + \frac{2^2}{1.2.3.4.5} \cdot \left\{ \Delta^5 \zeta^{(r-2)} + \Delta^5 \zeta^{(r-3)} \right\} \\ - \frac{2^2 \cdot 3^2}{1.2.3.4.5.6.7} \left\{ \Delta^7 \zeta^{(r-3)} + \Delta^7 \zeta^{(r-4)} \right\} \\ + \text{ &c.} \end{array} \right\} \\ \frac{d^2\alpha}{ds^2} &= \frac{\Delta^2 \zeta^{(r-1)}}{2.i^2} - \frac{1}{2.3.4.i^2} \cdot \Delta^4 \zeta^{(r-4)} \\ &\quad + \frac{2^2}{1.2.3.4.5.6.i^2} \cdot \Delta^6 \zeta^{(r-3)} - \frac{2^2 \cdot 3^2}{2.3.4.5.6.7.8.i^2} \cdot \Delta^8 \zeta^{(r-4)} + \text{ &c.} \end{aligned}$$

$\Delta$  is the characteristic of finite differences, so that  $\Delta \zeta^{(r)} = \zeta^{(r+1)} - \zeta^{(r)}$ .

If the number of observations be even, and equal to  $2r$ , we should assume for the epoch, the mean time between the first and last observation, and then we shall have

$$\alpha = \frac{1}{2} \cdot \left\{ \begin{array}{l} \zeta^{(r)} + \zeta^{(r-1)} - \frac{1}{2.4} \cdot \Delta^2 (\zeta^{(r-1)} + \zeta^{(r-2)}) \\ + \frac{3^2}{2.4.6.8} \cdot \Delta^4 (\zeta^{(r-2)} + \zeta^{(r-3)}) \\ - \frac{3^2 \cdot 5^2}{2.4.6.8.10.12} \cdot \Delta^6 (\zeta^{(r-3)} + \zeta^{(r-4)}) \end{array} \right\}$$

$$\frac{d\alpha}{ds} = \frac{\Delta \zeta^{(r-1)}}{i^2} - \frac{1}{4.6.i} \cdot \Delta^3 \zeta^{(r-2)} + \frac{3^2}{4.6.8.10.i} \cdot \Delta^5 \zeta^{(r-3)} - \text{ &c.};$$

that new observations assumed at equal intervals, and at opposite sides of this epoch, will cause quantities to be added to those values, which will be, with respect to their last terms, of the same order, as the ratio of  $s^2 \cdot \left( \frac{d^2\alpha}{ds^2} \right)$  to  $\alpha$ . This symmetric disposition obtains when all the observations being equidistant, we fix the epoch in the middle of the interval contained between them ; it is therefore advantageous to employ corresponding observations.

In general it will be always useful to fix the epoch towards the middle of this interval ; because that the number of days which separates the extreme observations, being less considerable, the approximations are more convergent. The calculus will be likewise simplified by fixing the epoch at the very instant of one of the observations ; for then the values of  $\alpha$  and of  $\theta$  will be immediately given.

When, by the preceding process, we have determined,  $\left( \frac{d\alpha}{ds} \right)$ ,  $\left( \frac{d^2\alpha}{ds^2} \right)$ ,  $\left( \frac{d\theta}{ds} \right)$ ,  $\left( \frac{d^2\theta}{ds^2} \right)$ ; we can deduce in this manner the first and second differences of  $\alpha$  and  $\theta$ , divided by the corresponding powers of the element of the time. If the masses of the planets and of the comets, are neglected in comparison with that of the sun assumed to represent the unity of the mass ; if, moreover, we assume for the unity of distance, its mean distance from the earth ; the mean motion of the earth round the sun, will be, by No. 23, the measure of the time  $t$  ; let, therefore  $\lambda$  represent the number of seconds which the earth de-

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$$\begin{aligned} \frac{d^2\alpha}{1.2.ds^2} &= \frac{1}{4.i^2} \cdot \Delta^2(\xi^{(r-1)} + \xi^{(r-2)}) - \frac{3^2}{4.6.8.i^2} \cdot \Delta^4(\xi^{(r-2)} + \xi^{(r-3)}) \\ &\quad + \frac{259}{4.6.8.10.12.i^3} \cdot \Delta^6(\xi^{(r-3)} + \xi^{(r-4)}) - \&c.; \end{aligned}$$

It is easy to prove these theorems from the theory of finite differences. See Lacroix, Tom. 3.

scribes in a day, in consequence of its mean sidereal motion ; the time  $t$  corresponding to the number  $s$  of days, will be  $\lambda s$  ; therefore we shall have

$$\left( \frac{d\alpha}{dt} \right) = \frac{1}{\lambda} \cdot \left( \frac{d\alpha}{ds} \right), \quad \left( \frac{d^2\alpha}{dt^2} \right) = \frac{1}{\lambda^2} \cdot \left( \frac{d^2\alpha}{ds^2} \right).$$

Observations give in logarithms of the tables,  $\log. \lambda = 4,0394622$  ; moreover,  $\log. \lambda^2 = \log. \lambda + \log. \frac{\lambda}{R}$ ,  $R$  being the radius of the circle, reduced into seconds ; from this it appears that  $\log. \lambda^2 = 2,2750444$  ; therefore, if the values of  $\left( \frac{d\alpha}{ds} \right)$ , and of  $\left( \frac{d^2\alpha}{ds^2} \right)$ , be reduced into seconds ; the logarithms of  $\left( \frac{d\alpha}{ds} \right)$  and of  $\left( \frac{d^2\alpha}{ds^2} \right)$  will be obtained, by subducting from the logarithms of these values, the logarithms, 4,0394622, and 2,2750444. We shall obtain in like manner, the the logarithm of  $\left( \frac{d\theta}{dt} \right)$ , and of  $\left( \frac{d^2\theta}{dt^2} \right)$  ; by subtracting respectively the same logarithms, from the logarithms of their values reduced into seconds.

As the precision of the following results depends on the accuracy of the values of  $\alpha$ ,  $\left( \frac{d\alpha}{dt} \right)$ ,  $\left( \frac{d^2\alpha}{dt^2} \right)$ ,  $\theta$ ,  $\left( \frac{d\theta}{dt} \right)$ , and  $\left( \frac{d^2\theta}{dt^2} \right)$ , and as their formation is very simple, the observations ought to be selected and multiplied in such a manner, as to obtain them with the greatest possible precision. We now proceed to the determination of the elements of the orbit of the comet by means of these values, and in order to generalize these results, we will consider the motion of a system of bodies actuated by any forces whatever.

30. Let  $x, y, z$ , be the rectangular coordinates of the first body ;  $x', y', z'$ , those of the second body, and so on of the rest. Let us conceive that the first body is solicited parallel to the axis of  $x$ , of  $y$ , and of  $z$ , by the forces  $X, Y$ , and  $Z$ , which forces we will suppose

to tend to diminish these variables. Let us conceive, in like manner, that the second body is sollicited parallel to the same axes, by the forces  $X'$ ,  $Y'$ ,  $Z'$ , and so of the rest. The motions of all these bodies will be given by differential equations of the second order,

$$\begin{aligned} 0 &= \frac{d^2x}{dt^2} + X; \quad 0 = \frac{d^2y}{dt^2} + Y; \quad 0 = \frac{d^2z}{dt^2} + Z; \\ 0 &= \frac{d^2x'}{dt^2} + X'; \quad 0 = \frac{d^2y'}{dt^2} + Y'; \quad 0 = \left( \frac{d^2z'}{dt^2} \right) + Z'. \\ &\text{&c.} \end{aligned}$$

If the number of bodies is  $n$ , the number of these equations will be  $3n$ , and their finite integrals will involve  $6n$  arbitrary quantities, which will be the elements of the orbits of these different bodies.\*

In order to determine these elements by means of observations, we should transform the coordinates of each body into others, of which the origin will be at the observer. Therefore supposing a plane, of which the position may remain always parallel to itself, to pass through the eye of the observer, while the observer moves on a given curve, let  $\rho$ ,  $\rho'$ ,  $\rho''$ , represent the distances of the observer from the different bodies, projected on this plane; and  $\alpha$ ,  $\alpha'$ ,  $\alpha''$ , &c., the apparent longitudes of these bodies, referred to the same plane, and  $\theta$ ,  $\theta'$ ,  $\theta''$ , their apparent latitudes. The variables  $x$ ,  $y$ ,  $z$ , will be given in a function of  $\rho$ ,  $\alpha$ ,  $\theta$ , and of the coordinates of the observer. In like manner,  $x'$ ,  $y'$ ,  $z'$ , will be given in functions of  $\rho'$ ,  $\alpha'$ ,  $\theta'$ , and of the coordinates of the observer, and so of the rest. Moreover, if we suppose that the forces  $X$ ,  $Y$ ,  $Z$ ,  $X'$ ,  $Y'$ ,  $Z'$ , &c., arise from the reciprocal action of the bodies of the system, and from the action of foreign bodies; they will be given in functions of  $\rho$ ,  $\rho'$ ,  $\rho''$ , &c.;  $\alpha$ ,  $\alpha'$ ,

\* Each body furnishes three equations, and consequently the  $n$  bodies furnish  $3n$  equations, and as in the integration of each differential equation of the second order, two arbitrary quantities are introduced, the total number of arbitrary quantities must be  $6n$ .

$\alpha''$ , &c.;  $\theta, \theta', \theta''$ , &c.; and of known quantities; consequently the preceding differential equations will be between these new variables, and their first and second differences; now observations make known, for a given time instant, the values of  $\alpha, \left(\frac{d\alpha}{dt}\right), \left(\frac{d^2\alpha}{dt^2}\right), \theta, \left(\frac{d\theta}{dt}\right), \left(\frac{d^2\theta}{dt^2}\right); \alpha', \left(\frac{d\alpha'}{dt}\right), \left(\frac{d^2\alpha'}{dt^2}\right)$ , &c.; therefore, the quantities which remain unknown, are  $\varepsilon, \varepsilon', \varepsilon'',$  &c., their first and second differences. These unknown quantities are  $3n$  in number, and as we have  $3n$  differential equations, we can determine them. There is also this advantage connected with this method, that the first and second differences of  $\varepsilon, \varepsilon', \varepsilon'',$  &c. occur in these equations, in a linear form.

The quantities  $\alpha, \theta, \varepsilon, \alpha', \theta', \varepsilon',$  &c., and their first differentials divided by  $dt$ , being known; we shall have for any given instant, the values of  $x, y, z, x', y', z',$  &c., and of their first differentials divided by  $dt$ . These values being\* substituted in the  $3n$  integrals of the preceding differential equations, and in the first differences of these integrals will give  $6n$  equations, by means of which we can determine the  $6n$  arbitrary quantities of these integrals, or the elements of the orbits of these different bodies.

31. Let us apply this method to the motion of the comets. For this purpose it may be observed, that the principal force which actuates them, being the attraction of the sun, we may abstract from the consideration of every other force. However, if the comet passes sufficiently near to any large planet, to experience a sensible perturbation, the preceding method would still make known its velocity, and its distance from the earth; but this case being of rare occurrence, we shall only consider, in the subsequent researches, the action of the sun.

\* The number of unknown quantities for each body is three, namely  $\varepsilon, \frac{d\varepsilon}{dt}, \frac{d^2\varepsilon}{dt^2}$ , therefore there are  $3n$  unknown quantities in the system of  $n$  bodies.

Assuming the mass of the sun to represent the unity of mass, and its mean distance from the earth, the unity of distance, and moreover placing the origin of the coordinates  $x, y, z$ , of a comet of which the radius is  $r$ , at the sun ; the differential equations (O) of No. 17, will become, (the mass of the comet, in comparison with that of the sun being neglected)

$$\left. \begin{aligned} 0 &= \frac{d^2x}{dt^2} + \frac{x}{r^3}; \\ 0 &= \frac{d^2y}{dt^2} + \frac{y}{r^3}; \\ 0 &= \frac{d^2z}{dt^2} + \frac{z}{r^3}; \end{aligned} \right\}. \quad (k)$$

Let us now suppose that the plane of  $x$  and of  $y$ , is the plane of the ecliptic ; that the axis of  $x$  is the line drawn from the centre of the sun to the first point of Aries, at a given epoch ; that the axis of  $y$  is the line drawn from the centre of the sun to the first point of Cancer, at the same epoch ; that the positive  $z$ 's are on the same side with the north pole of the ecliptic ; and finally, that  $x'$  and  $y'$  are the coordinates of the earth, and  $R$  its radius vector ; this being premised,

Let the coordinates  $x, y, z$ , be transformed into others relative to the observer ; and for this purpose let  $\alpha$  represent the geocentric longitude of the comet,  $\theta$  its geocentric latitude, and  $\rho$  its distance from the earth projected on the ecliptic ; we shall have

$$x = x' + \rho \cdot \cos. \alpha; \quad y = y' + \rho \cdot \sin. \alpha; \quad z = \rho \cdot \tan. \theta.$$

If from the first of the equations (k), multiplied by  $\sin. \alpha$ , be subtracted the second multiplied by  $\cos. \alpha$ , we shall have

$$0 = \sin. \alpha \cdot \frac{d^2x}{dt^2} - \cos. \alpha \cdot \frac{d^2y}{dt^2} + \frac{x \cdot \sin. \alpha - y \cdot \cos. \alpha}{r^3} *$$

$$* \quad \frac{dx}{dt} = \frac{dx'}{dt} + \frac{d\rho}{dt} \cdot \cos. \alpha - \rho \cdot \sin. \alpha \cdot \frac{d\alpha}{dt}; \quad \therefore \frac{d^2x}{dt^2} \cdot \sin. \alpha = \frac{d^2x'}{dt^2} \cdot \sin. \alpha + \frac{d^2\rho}{dt^2} \cdot \sin. \alpha.$$

hence we deduce, by substituting for  $x$  and  $y$  their preceding values

$$\begin{aligned} \sin. \alpha. \frac{d^2x'}{dt^2} - \cos. \alpha. \frac{d^2y'}{dt^2} + \frac{x'. \sin. \alpha - y'. \cos. \alpha}{r^3} \\ - 2. \left\{ \frac{d\varrho}{dt} \right\} \cdot \left\{ \frac{d\alpha}{dt} \right\} - \varrho. \left\{ \frac{d^2\alpha}{dt^2} \right\}. \end{aligned}$$

The earth being retained in its orbit, as the comet, by the attraction of the sun, we have

$$0 = \frac{d^2x'}{dt^2} + \frac{x'}{R^3}; \quad 0 = \frac{d^2y'}{dt^2} + \frac{y'}{R^3};$$

consequently,

$$\sin. \alpha. \frac{d^2x'}{dt^2} - \cos. \alpha. \frac{d^2y'}{dt^2} = \frac{y'. \cos. \alpha - x'. \sin. \alpha}{R^3};$$

therefore, we shall have

$$0 = (y'. \cos. \alpha - x'. \sin. \alpha). \left\{ \frac{1}{R^3} - \frac{1}{r^3} \right\} - 2. \left\{ \frac{d\varrho}{dt} \right\} \cdot \left\{ \frac{d\alpha}{dt} \right\} - \varrho. \left\{ \frac{d^2\alpha}{dt^2} \right\}.$$

Let  $A$  be the longitude of the earth, as seen from the sun; we shall have

$$x' = R. \cos. A; \quad y' = R. \sin. A;$$

therefore

$$y'. \cos. \alpha - x'. \sin. \alpha = R. \sin. (A - \alpha);$$

the preceding equation will consequently become,

$$\begin{aligned} \sin. \alpha. \cos. \alpha - 2. \frac{d\varrho}{dt} \cdot \frac{d\alpha}{dt} \cdot \sin. {}^2 \alpha - \varrho. \sin. \alpha. \cos. \alpha. \frac{d\alpha^2}{dt^2} - \varrho. \sin. {}^2 \alpha. \frac{d^2\alpha}{dt^2} \cdot \frac{dy}{dt} = \frac{dy'}{dt} \\ + \frac{d\varrho}{dt} \cdot \sin. \alpha + \varrho. \cos. \alpha. \frac{d\alpha}{dt}; \quad \therefore \frac{d^2y}{dt^2} \cdot \cos. \alpha = \frac{d^2y'}{dt^2} \cdot \cos. \alpha + \frac{d^2\varrho}{dt^2} \cdot \sin. \alpha. \cos. \alpha \\ + 2. \frac{d\varrho}{dt} \cdot \frac{d\alpha}{dt} \cdot \cos. {}^2 \alpha - \varrho. \sin. \alpha. \cos. \alpha. \frac{d\alpha^2}{dt^2} + \varrho. \cos. {}^2 \alpha. \frac{d^2\alpha}{dt^2}, \quad \text{by subtracting this} \\ \text{equation from the value of } \frac{d^2x}{dt^2} \cdot \sin. \alpha, \quad \text{observing the quantities which destroy each} \\ \text{other, and also those which coalesce, we arrive at the expression given in the text.} \end{aligned}$$

$$\left\{ \frac{d\xi}{dt} \right\} = \frac{R \cdot \sin. (A - \alpha)}{2 \cdot \left\{ \frac{d\alpha}{dt} \right\}} \cdot \left\{ \frac{1}{R^3} - \frac{1}{r^3} \right\} - \xi \cdot \frac{\left\{ \frac{d^2\alpha}{dt^2} \right\}}{2 \cdot \left\{ \frac{d\alpha}{dt} \right\}}. \quad (1)$$

Let us now investigate a second expression for  $\left\{ \frac{d\xi}{dt} \right\}$ . For this purpose multiplying the first of the equations (*k*), by tan.  $\theta \cdot \cos. \alpha$ ; the second by tan.  $\theta \cdot \sin. \alpha$ ; and then subtracting the third equation, from the sum of these two products; we shall have

$$0 = \tan. \theta \cdot \left\{ \cos. \alpha \cdot \frac{d^2x}{dt^2} + \sin. \alpha \cdot \frac{d^2y}{dt^2} \right\} + \tan. \theta \cdot \left\{ \frac{x \cdot \cos. \alpha + y \cdot \sin. \alpha}{r^3} \right\} - \frac{d^2z}{dt^2} - \frac{z}{r^3}.$$

This equation will become, by substituting for  $x, y, z$ , their values,

$$0 = \tan. \theta \cdot \left\{ \left\{ \frac{d^2x'}{dt^2} + \frac{x'}{r^3} \right\} \cdot \cos. \alpha + \left\{ \frac{d^2y'}{dt^2} + \frac{y'}{r^3} \right\} \cdot \sin. \alpha \right\} - * \\ \frac{2 \cdot \left\{ \frac{d\theta}{dt} \right\} \cdot \left\{ \frac{d\xi}{dt} \right\} \cdot}{\cos. ^2 \theta}.$$

$$* \quad \frac{d^2x}{dt^2} = \frac{d^2x'}{dt^2} + \frac{d^2\xi}{dt^2} \cdot \cos. \alpha - 2 \cdot \frac{d\xi}{dt} \cdot \frac{d\alpha}{dt} \cdot \sin. \alpha - \xi \cdot \cos. \alpha \cdot \frac{d\alpha^2}{dt^2} - \xi \cdot \sin. \alpha \cdot \frac{d^2\alpha}{dt^2} \\ \frac{d^2y}{dt^2} = \frac{d^2y'}{dt^2} + \frac{d^2\xi}{dt^2} \cdot \sin. \alpha + 2 \cdot \frac{d\xi}{dt} \cdot \frac{d\alpha}{dt} \cdot \cos. \alpha - \xi \cdot \sin. \alpha \cdot \frac{d\alpha^2}{dt^2} + \xi \cdot \cos. \alpha \cdot \frac{d^2\alpha}{dt^2} \\ \therefore \frac{d^2x}{dt^2} \cdot \cos. \alpha + \frac{d^2y}{dt^2} \cdot \sin. \alpha = \frac{d^2x'}{dt^2} \cdot \cos. \alpha + \frac{d^2y'}{dt^2} \cdot \sin. \alpha + \frac{d^2\xi}{dt^2} - \xi \cdot \frac{d\alpha^2}{dt^2} \\ \frac{x \cdot \cos. \alpha + y \cdot \sin. \alpha}{r^3} = \frac{x' \cdot \cos. \alpha + y' \cdot \sin. \alpha}{r^3} + \frac{\xi}{r^3}, \\ z = \xi \cdot \tan. \theta \therefore \frac{dz}{dt} = \frac{d\xi}{dt} \cdot \tan. \theta + \xi \cdot \frac{d\theta}{dt} \cdot \frac{d^2z}{dt^2} = \frac{d^2\xi}{dt^2} \cdot \tan. \theta + \\ 2 \cdot \frac{d\xi}{dt} \cdot \frac{d\theta}{dt} \cdot \frac{1}{\cos. ^2 \theta} + \frac{d^2\theta}{dt^2} \cdot \frac{\xi}{\cos. ^3 \theta} + \frac{2\xi}{\cos. ^3 \theta} \cdot \frac{d\theta^2}{dt^2} \cdot \sin. \theta, \text{ this expression being subtracted from the preceding multiplied into } \tan. \theta, \text{ gives } 0 =$$

$$-\dot{\varrho} \cdot \left\{ \frac{\left\{ \frac{d^2\theta}{dt^2} \right\}}{\cos^2\theta} + 2 \cdot \frac{\left\{ \frac{d\theta}{dt} \right\}^2}{\cos^3\theta} \cdot \sin\theta + \frac{\left\{ \frac{d\alpha}{dt} \right\}^2}{\cos^3\theta} \cdot \tan\theta \right\};$$

but we have

$$\left\{ \frac{d^2x'}{dt^2} + \frac{x'}{r^3} \right\} \cdot \cos\alpha + \left\{ \frac{d^2y'}{dt^2} + \frac{y'}{r^3} \right\} \cdot \sin\alpha = (x' \cdot \cos\alpha + y' \cdot \sin\alpha).$$

$$\left\{ \frac{1}{r^3} - \frac{1}{R^3} \right\} = R \cdot \cos(A-\alpha) \cdot \left\{ \frac{1}{r^3} - \frac{1}{R^3} \right\};$$

therefore

$$\begin{aligned} \left\{ \frac{d\varrho}{dt} \right\} &= -\frac{1}{2}\dot{\varrho} \cdot \left\{ \frac{\left\{ \frac{d^2\theta}{dt^2} \right\}}{\left\{ \frac{d\theta}{dt} \right\}} + 2 \cdot \frac{\left\{ \frac{d\theta}{dt} \right\} \cdot \tan\theta}{\left\{ \frac{d\theta}{dt} \right\}} + \frac{\left\{ \frac{d\alpha}{dt} \right\}^2 \cdot \sin\theta \cdot \cos\theta}{\left\{ \frac{d\theta}{dt} \right\}} \right\} \\ &\quad + \frac{R \cdot \sin\theta \cdot \cos\theta \cdot \cos(A-\alpha)}{2 \cdot \left\{ \frac{d\theta}{dt} \right\}} \cdot \left\{ \frac{1}{r^3} - \frac{1}{R^3} \right\}; \end{aligned} \quad (2)$$

if this value of  $\frac{d\varrho}{dt}$  be subtracted from the first, and if we suppose

$$\mu' = \frac{\left\{ \frac{d\alpha}{dt} \right\} \cdot \left\{ \frac{d^2\theta}{dt^2} \right\} - \left\{ \frac{d\theta}{dt} \right\} \cdot \left\{ \frac{d^2\alpha}{dt^2} \right\} + 2 \cdot \left\{ \frac{d\alpha}{dt} \right\} \cdot \left\{ \frac{d\theta}{dt} \right\}^2 \cdot \tan\theta + \left\{ \frac{d\alpha}{dt} \right\}^3 \cdot \sin\theta \cdot \cos\theta}{\left\{ \frac{d\alpha}{dt} \right\} \cdot \sin\theta \cdot \cos\theta \cdot \cos(A-\alpha) + \left\{ \frac{d\theta}{dt} \right\} \cdot \sin(A-\alpha)} +$$

$$\begin{aligned} &\tan\theta \cdot \left( \frac{d^2x'}{dt^2} \cdot \cos\alpha + \frac{d^2y'}{dt^2} \cdot \sin\alpha + \frac{x'}{r^3} \cdot \cos\alpha + \frac{y'}{r^3} \cdot \sin\alpha \right) + \frac{\dot{\varrho} \cdot \tan\theta}{r^3} - \\ &\frac{d^2\dot{\varrho}}{dt^2} \cdot \tan\theta - 2 \cdot \frac{d\dot{\varrho}}{dt} \cdot \frac{d\theta}{dt} \cdot \frac{1}{\cos^2\theta} - \frac{d^2\theta}{dt^2} \cdot \frac{\dot{\varrho}}{\cos^2\theta} - \frac{2\dot{\varrho}}{\cos^3\theta} \cdot \frac{d\theta^2}{dt^2} \cdot \sin\theta - \frac{da^2}{dt^2} \cdot \tan\theta - \\ &\frac{\dot{\varrho} \cdot \tan\theta}{r^3}. \end{aligned}$$

\* This value of  $\frac{d\varrho}{dt}$  is derived immediately from the preceding equations, by multiplying the entire expression, by  $\cos^2\theta$ , and dividing by  $\frac{d\theta}{dt}$ , and observing that  $\tan\theta = \frac{\sin\theta}{\cos\theta}$ .

† If the two values of  $\frac{d\varrho}{dt}$ , be multiplied by  $\frac{d\alpha}{dt} \cdot \frac{d\theta}{dt}$ , and if the second be then sub-

we shall obtain

$$\xi = \frac{R}{\mu'} \left\{ \frac{1}{r^3} - \frac{1}{R^3} \right\}. \quad (3)$$

The projected distance  $\xi$ , of the comet from the earth, being always positive; this equation shews that the distance  $r$  of the comet from the sun is greater or less than the distance  $R$  of the earth from the sun, according as  $\mu'$  is positive or negative; these two distances are equal, if  $\mu' = 0$ .

We can, by the sole inspection of the celestial globe, determine the sign of  $\mu'$ ; and consequently, whether the comet is nearer or farther than the earth from the sun. For this purpose, let us conceive a great circle, which passes through two geocentric positions of the comet, indefinitely near to each other. Let  $\gamma$  represent the inclination of this circle to the ecliptic, and  $\lambda$ , the longitude of its ascending node; we shall have

$$\tan. \gamma. \sin. (\alpha - \lambda) = \tan. \theta;$$

from which may be obtained

$$d\theta. \sin. (\alpha - \lambda) = d\alpha. \sin. \theta. \cos. \theta. \cos. (\alpha - \lambda);$$

differentiating a second time, we shall have

$$0 = \left\{ \frac{d\alpha}{dt} \right\} \cdot \frac{d^2\theta}{dt^2} - \frac{d\theta}{dt} \cdot \frac{d^2\alpha}{dt^2} + 2 \cdot \left\{ \frac{d\alpha}{dt} \right\} \cdot \left\{ \frac{d\theta}{dt} \right\}^2 \cdot \tan. \theta$$

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tracted from the first, the quantity by which  $\xi$  is multiplied is the numerator of the value of  $\mu'$ , and the quantity independent of  $\xi$ , is its denominator.

If  $r$  be less than  $R$ ,  $\frac{1}{r^3} - \frac{1}{R^3}$  is positive,  $\therefore$  in this case  $\mu'$  must be positive; if  $r$  is greater than  $R$ , then  $\frac{1}{r^3} - \frac{1}{R^3}$  is negative,  $\therefore \mu'$  must in this case be negative; when  $r = R$ ,  $\mu' = \frac{R}{\xi} \cdot \left( \frac{1}{r^3} - \frac{1}{R^3} \right) = 0$ .

$$+ \left\{ \frac{d\alpha}{dt} \right\}^3 \cdot \sin. \theta. \cos. \theta; ^*$$

$d^2\theta$ , being the value of  $d^2\theta$ , which it would have, if the apparent motion of the comet continued in the great circle. Consequently the value of  $\mu'$  becomes, by substituting for  $d\theta$ , its value

$$\frac{d\alpha. \sin. \theta. \cos. \theta. \cos. (\alpha - \lambda)}{\sin. (\alpha - \lambda)};$$

$$\mu' = \frac{\left\{ \left\{ \frac{d^2\theta}{dt^2} \right\} - \left\{ \frac{d^2\theta}{dt^2} \right\} \right\} \cdot \sin. (\alpha - \lambda)}{\sin. \theta. \cos. \theta. \sin. (A - \lambda)} +$$

The function  $\frac{\sin. (\alpha - \lambda)}{\sin. \theta. \cos. \theta}$  is constantly positive ; therefore the value of  $\mu'$  is positive or negative, according as  $\left\{ \frac{d^2\theta}{dt^2} \right\} - \left\{ \frac{d^2\theta}{dt^2} \right\}$  has the same or a contrary sign, to  $\sin. (A - \lambda)$ ;  $A - \lambda$  is equal to two right angles, plus the distance of the sun from the ascending node of the

\*  $d\alpha. \cos. (\alpha - \lambda). \tan. \gamma = \frac{d\theta}{\cos. \theta^2}$ ,  $\therefore d\alpha. \frac{\cos. (\alpha - \lambda). \tan. \theta}{\sin. (\alpha - \lambda)} = \frac{d\theta}{\cos. \theta^2}$ ,  $\therefore d\alpha. \cos. (\alpha - \lambda). \sin. \theta. \cos. \theta = d\theta. \sin. (\alpha - \lambda)$ ; hence  $d^2\theta. \sin. (\alpha - \lambda) + d\alpha. d\theta. \cos. (\alpha - \lambda) = d^2\alpha. \sin. \theta. \cos. \theta. \cos. (\alpha - \lambda) + d\alpha. d\theta. \cos. \theta. \cos. (\alpha - \lambda) - d\alpha. d\theta. \sin. \theta. \cos. (\alpha - \lambda) - d\alpha^2. \sin. \theta. \cos. \theta. \sin. (\alpha - \lambda)$ , by substituting for  $\sin. (\alpha - \lambda)$  its value  $\frac{d\alpha}{d\theta}. \sin. \theta. \cos. \theta. \cos. (\alpha - \lambda)$ , we obtain  $\frac{d^2\theta. d\alpha}{d\theta^2}. \sin. \theta. \cos. \theta. \cos. (\alpha - \lambda) + d\alpha. d\theta. (\cos. (\alpha - \lambda)) = d^2\alpha. \sin. \theta. \cos. \theta. \cos. (\alpha - \lambda) + d\alpha. d\theta. \cos. \theta. \cos. (\alpha - \lambda) - d\alpha. d\theta. \sin. \theta. \cos. (\alpha - \lambda) - \frac{d\alpha^3}{d\theta}. \sin. \theta. \cos. \theta. \cos. (\alpha - \lambda)$ ; dividing both sides of this equation by  $\frac{\sin. \theta. \cos. \theta}{d\theta} \cdot \cos. (\alpha - \lambda)$ , we obtain the expression which is given in the text.

† By substituting for  $\frac{2d\alpha}{dt} \cdot \frac{d\theta^2}{dt^2} \cdot \tan. \theta + \left( \frac{d\alpha}{dt} \right)^3 \cdot \sin. \theta. \cos. \theta$ , its value given in the preceding equation, and for  $d\theta$  its value, we obtain  $\mu' =$

$$\left( \frac{d^2\theta}{dt^2} - \frac{d^2\theta}{dt^2} \right) \cdot \sin. (\alpha - \lambda) \text{ divided by } \sin. \theta. \cos. \theta. (\sin. (\alpha - \lambda). \cos. (A - \alpha) + \cos. (\alpha - \lambda),$$

$$\sin. (A - \alpha)) = (\sin. \theta. \cos. \theta. \sin. (\alpha - \lambda + A - \alpha) = \sin. \theta. \cos. \theta. \sin. (A - \lambda).$$

great circle ; hence it is easy to infer that  $\mu'$  will be positive or negative, according as in a third geocentric position of the comet, indefinitely near\* to the two first, the comet deviates from this great circle from the very side in which the sun exists, or from the opposite side. Let us conceive, therefore, that a great circle of the sphere passes through two geocentric positions of the comet, indefinitely near to each other ; if in a third consecutive geocentric position indefinitely near to the two first, the comet deviates from this great circle, from the same side as the sun, or from the opposite side, it will be nearer or farther than the earth from the sun, it will be equally distant, if it continues to appear in this great circle ; thus the different inflexions of its apparent route will throw some light on the variations of its distances from the sun.

In order to eliminate  $r$  from the equation (3), so that this equation may only involve the unknown  $\varrho$ , it is to be observed that we have  $r^2 = x^2 + y^2 + z^2$ , and by substituting in place of  $x, y, z$ , their values in terms of  $\varrho, \alpha$  and  $\theta$  ; we shall have

$$r^2 = x'^2 + y'^2 + 2\varrho(x' \cdot \cos. \alpha + y' \cdot \sin. \alpha) + \frac{\varrho^2}{\cos. \theta}; \dagger$$

\*  $A = 180 + a$  ( $=$  the sun's longitude, as seen from the earth) and  $\therefore A - \lambda = 180 + a - \lambda = 180 +$  the distance of the sun from the ascending node of the great circle,  $\therefore$  when  $a > \lambda$  the sign of  $\sin. (A - \lambda)$  is negative, and if  $\frac{d^2\theta}{dt^2} - \frac{d^2\theta'}{dt^2}$  be also negative, the comet in the third position must deviate from the great circle from the very direction in which the sun appears as seen from the earth, if  $a$  be  $< \lambda$ , then  $\sin. (A - \lambda)$  is positive,  $\therefore$  if  $\frac{d^2\theta}{dt^2} - \frac{d^2\theta'}{dt^2}$  be also positive, it is evident that the comet must be nearer than the earth to the sun, and  $\therefore$  that in the third position, the comet must deviate from the great circle, from the direction in which the sun appears from the earth ; on the contrary, if  $\sin. (A - \lambda)$  be negative, and  $\frac{d^2\theta}{dt^2} - \frac{d^2\theta'}{dt^2}$  positive, in order that this may obtain, in this situation of the bodies, it is necessary that in the third position the comet should deviate from the great circle, from the opposite side to that in which the sun appears, as seen from the earth. See Memoirs of the Academy of Berlin, for the years 1772, and 1778.

$\dagger x^2 = x'^2 + 2\varrho x' \cdot \cos. \alpha + \varrho^2 \cdot \cos. ^2 \alpha ; y^2 = y'^2 + 2ry' \cdot \sin. \alpha + \varrho^2 \cdot \sin. ^2 \alpha ; z^2 = \varrho^2$

but we have  $x' = R \cdot \cos. A$ ;  $y' = R \cdot \sin. A$ ; therefore

$$r^2 = \frac{\dot{\varrho}^2}{\cos. \dot{\theta}} + 2R\dot{\varrho} \cdot \cos. (A - \alpha) + R^2.$$

By squaring the members of the equation (3), when arranged under the following form,

$$r^3 \cdot (\mu' \cdot R^2 \dot{\varrho} + 1) = R^3;$$

we shall obtain, by substituting in place of  $r^2$ , its value,

$$\left( \frac{\dot{\varrho}^2}{\cos. \dot{\theta}} + 2R\dot{\varrho} \cdot \cos. (A - \alpha) + R^2 \right)^3 \cdot (\mu' \cdot R^2 \dot{\varrho} + 1)^2 = R^6; \quad (4)$$

In this\* equation,  $\dot{\varrho}$  is the only unknown quantity, and it ascends to the seventh degree, because the term which is entirely known in the first member being equal to  $R^6$ , the entire equation is divisible by  $\dot{\varrho}$ . Having by this means determined  $\dot{\varrho}$ , we will obtain  $\left\{ \frac{d\varrho}{dt} \right\}$  by means of the equations (1) and (2). By substituting, for example in the equation (1), instead of  $\frac{1}{r^3} - \frac{1}{R^3}$ , its value  $\frac{\mu' \rho}{R}$ , which is given by the equation (3); we shall have

$$\left\{ \frac{d\varrho}{dt} \right\} = \frac{-\dot{\varrho}}{2 \cdot \left\{ \frac{d\alpha}{dt} \right\}} \cdot \left\{ \left\{ \frac{d^2\alpha}{dt^2} \right\} + \mu' \cdot \sin. (A - \alpha) \right\}.$$

The equation (4) is frequently susceptible of several real and positive roots; by making its second member to coalesce with the first, and then dividing by  $\dot{\varrho}$ , its last term will be

$$\tan. \dot{\theta}. \therefore x^2 + y^2 + z^2 = x'^2 + y'^2 + 2\dot{\varrho} \cdot (x' \cdot \cos. \alpha + y' \cdot \sin. \alpha) + \dot{\varrho}^2 \cdot (1 + \tan. \dot{\theta})^2 = \left( \frac{\dot{\varrho}^2}{\cos. \dot{\theta}} + 2R\dot{\varrho} \cdot \cos. (A - \alpha) + R^2 \right)^2.$$

Multiplying both sides of equation (3) by  $\mu' \cdot R^3 \cdot r^3$ , and we obtain  $\mu' \cdot R^3 \cdot \dot{\varrho} \cdot r^3 = R^4 - R \cdot r^3$ ,  $\therefore r^3 \cdot (\mu' \cdot R^2 \dot{\varrho} + 1) = R^3$ ;  $\therefore$  substituting for  $r^3$  its value we obtain  $\left( \frac{\dot{\varrho}^2}{\cos. \dot{\theta}} + 2R\dot{\varrho} \cdot \cos. (A - \alpha) + R^2 \right)^{\frac{3}{2}} \cdot (\mu' R^2 \dot{\varrho} + 1) = R^3$ .

\*  $R^6$  occurs on both sides of this equation with the same sign, therefore it may be omitted, and as the remaining quantity is divisible by  $\dot{\varrho}$ , it may be depressed to the seventh degree.

$$2. R^5 \cdot \cos. {}^6\theta. (\mu' \cdot R^3 + 3. \cos. (A - \alpha)); ^*$$

Thus the equation in  $\xi$ , being of the seventh degree, it will have at least two roots which are real and positive, if  $\mu' \cdot R^3 + 3. \cos. (A - \alpha)$  is positive; † for by the nature of the problem, it must always have a positive root, and it is evident from the nature of equations that when this is the case the number of its positive roots cannot be odd. Each real and positive value of  $\xi$ , gives a different conick section for the orbit of the comet; therefore we will have as many curves which satisfy three neighbouring observations, as  $\xi$  will have real and positive values, and in order to determine the true orbit of the comet, we must have recourse to a new observation.

32. The value of  $\xi$ , deduced from the equation (4) would be rigorously exact, if  $\alpha$ ,  $\left(\frac{d\alpha}{dt}\right)$ ,  $\left(\frac{d^2\alpha}{dt^2}\right)$ ,  $\theta$ ,  $\left(\frac{d\theta}{dt}\right)$ ,  $\left(\frac{d^2\theta}{dt^2}\right)$ , were exactly known; but these are only approximate values. Indeed, we can by the method already laid down approach to them nearer and nearer, by employing a considerable number of observations, which has also the advantage, of enabling us to consider intervals sufficiently great, and thus to compensate by each other, the errors of observations. But this method is liable to the analytic inconvenience of employing more than three observations, in a problem in which three is sufficient. We can obviate this inconvenience in the following manner, which at the same

\* This equation when expanded becomes

$$\left(\frac{\xi^2}{\cos. {}^2\theta} + 2R\xi \cdot \cos. (A - \alpha)\right)^3 + 3. \left(\frac{\xi^2}{\cos. {}^2\theta} + 2R\xi \cdot \cos. (A - \alpha)\right)^2 \cdot R^2 + 3. \left(\frac{\xi^2}{\cos. {}^2\theta} +$$

$2R\xi \cdot \cos. (A - \alpha) \cdot R^4 + R^6\right) \cdot (\mu'^2 \cdot R^4 \xi^2 + 2\mu R^2 \xi + 1) = R^6$ , when  $R^6$  is obliterated, and this expression is multiplied by  $\cos. {}^6\theta$ , and divided by  $\xi$ , the absolute quantity is evidently equal to  $(2R \cdot \cos. (A - \alpha) \cdot 3R^4 + 2\mu R^8) \cdot \cos. {}^6\theta$ .

† This equation being of the seventh dimension, when the absolute quantity is positive it must have one real negative root, and from the nature of the problem it has one real affirmative root, ∴ as impossible roots enter questions by pairs, the number of those in the proposed equation cannot exceed four; consequently, in order that the sign of the absolute quantity may be positive, the remaining real root must be positive.

time that it only employs three observations, will render our solution as accurate as we please.

For this purpose let us suppose that  $\alpha$  and  $\theta$  represent the geocentric longitude and latitude of the intermediate observations; by substituting in place of  $x, y, z$ , their values  $x' + \rho \cdot \cos. \alpha$ ;  $y' + \rho \cdot \sin. \alpha$ ; and  $\rho \cdot \tan. \theta$ ; they will give  $\left\{ \frac{d^2 \rho}{dt^2} \right\}$ ,  $\left\{ \frac{d^2 \alpha}{dt^2} \right\}$  and  $\left\{ \frac{d^2 \theta}{dt^2} \right\}$ , in functions of  $\rho, \alpha$ , and  $\theta$ , of their first differences and of known quantities. By differentiating these functions, we will obtain,  $\left\{ \frac{d^3 \rho}{dt^3} \right\}$ ,  $\left\{ \frac{d^3 \alpha}{dt^3} \right\}$  and  $\left\{ \frac{d^3 \theta}{dt^3} \right\}$ , in functions of  $\rho, \alpha, \theta$ , and of their first and second differences. We can eliminate the second difference of  $\rho$ , by means of its value, and its first difference, by means of the equation (2) of the preceding number. By continuing to difference successively, the values of  $\left\{ \frac{d^3 \alpha}{dt^3} \right\}$ ,

$\left\{ \frac{d^3 \theta}{dt^3} \right\}$ , and then by eliminating the differences of  $\alpha$  and  $\theta$ , superior to the second, and all the differences of  $\rho$ , we will obtain the values of  $\left\{ \frac{d^3 \alpha}{dt^3} \right\}$ ,  $\left\{ \frac{d^4 \alpha}{dt^4} \right\}$ , &c.,  $\left\{ \frac{d^3 \theta}{dt^3} \right\}$ ,  $\left\{ \frac{d^4 \theta}{dt^4} \right\}$ , &c., in functions of  $\rho, \alpha$ ,  $\left\{ \frac{d \alpha}{dt} \right\}$ ,  $\theta$ ,  $\left\{ \frac{d \theta}{dt} \right\}$ ,  $\left\{ \frac{d^2 \theta}{dt^2} \right\}$ ; this being premised, let  $\alpha, \alpha', \alpha''$ , be the three observed geocentric longitudes of the comet;  $\theta, \theta', \theta''$ , its three corresponding geocentric latitudes; let  $i$  be the number of days which intervene between the first and second observation, and  $i'$ , the number which separates the second observation from the third; finally, let  $\lambda$  be the arc which the earth describes in a day by its mean sidereal motion; by No. 29, we shall have

$$\alpha = \alpha - i \cdot \lambda \cdot \left( \frac{d \alpha}{dt} \right) + \frac{i^2 \cdot \lambda^2}{1.2} \cdot \left( \frac{d^2 \alpha}{dt^2} \right) - \frac{i^3 \cdot \lambda^3}{1.2.3} \cdot \left( \frac{d^3 \alpha}{dt^3} \right) + \text{&c.};$$

$$\alpha' = \alpha + i' \cdot \lambda \cdot \left( \frac{d \alpha}{dt} \right) + \frac{i'^2 \cdot \lambda^2}{1.2} \cdot \left( \frac{d^2 \alpha}{dt^2} \right) + \frac{i'^3 \cdot \lambda^3}{1.2.3} \cdot \left( \frac{d^3 \alpha}{dt^3} \right) + \text{&c.};$$

$$\theta = -i\lambda \cdot \left( \frac{d\theta}{dt} \right) + \frac{i^2 \cdot \lambda^2}{1 \cdot 2} \cdot \left( \frac{d^2\theta}{dt^2} \right) - \frac{i^3 \cdot \lambda^3}{1 \cdot 2 \cdot 3} \cdot \left( \frac{d^3\theta}{dt^3} \right) + \text{&c.};$$

$$\vartheta = \theta + i' \cdot \lambda \cdot \left( \frac{d\theta}{dt} \right) + \frac{i'^2 \cdot \lambda^2}{1 \cdot 2} \cdot \left( \frac{d^2\theta}{dt^2} \right) + \frac{i'^3 \cdot \lambda^3}{1 \cdot 2 \cdot 3} \cdot \left( \frac{d^3\theta}{dt^3} \right) + \text{&c.}$$

By substituting in these series, for  $\left\{ \frac{d^3\alpha}{dt^3} \right\}$ ,  $\left\{ \frac{d^4\alpha}{dt^4} \right\}$ , &c.  $\left\{ \frac{d^3\theta}{dt^3} \right\}$ ,

$\left\{ \frac{d^4\theta}{dt^4} \right\}$ , &c., their values obtained by the preceding method; we shall have four equations between the five unknown quantities  $\varepsilon$ ,  $\left\{ \frac{d\alpha}{dt} \right\}$ ,  $\left\{ \frac{d^2\alpha}{dt^2} \right\}$ ,  $\left\{ \frac{d\theta}{dt} \right\}$ ,  $\left\{ \frac{d^2\theta}{dt^2} \right\}$ . These equations will be always more exact, according as we consider a greater number of terms in the preceding series. By this means we shall obtain,  $\left\{ \frac{d\alpha}{dt} \right\}$ ,  $\left\{ \frac{d^2\alpha}{dt^2} \right\}$ ,  $\left\{ \frac{d\theta}{dt} \right\}$  and  $\left\{ \frac{d^2\theta}{dt^2} \right\}$ , in functions of  $\varepsilon$  and of known quantities; and by substituting them in the equation (4) of the preceding number, it will only involve the unknown quantity  $\varepsilon$ . In fine, this method which has been detailed here, merely in order to shew how by means of three observations only we can obtain continually approaching values of  $\varepsilon$ , would require in practice, very troublesome computations, and it is at the same time more exact and more simple, to consider a greater number, by the method explained in No. 29.

33. When the values of  $\varepsilon$  and of  $\left\{ \frac{d\varepsilon}{dt} \right\}$  shall have been determined, we can obtain those of  $x$ ,  $y$ ,  $z$ ,  $\left( \frac{dx}{dt} \right)$ ,  $\left( \frac{dy}{dt} \right)$  and  $\left( \frac{dz}{dt} \right)$  by means of the equations  $x = R \cdot \cos. A + \varepsilon \cdot \cos. \alpha$ ;  $y = R \cdot \sin. A + \varepsilon \cdot \sin. \alpha$ ;  $z = \varepsilon \cdot \tan. \theta$ ; and of their differentials divided by  $dt$ ,

$$\begin{aligned} \left(\frac{dx}{dt}\right) &= \left(\frac{dR}{dt}\right) \cdot \cos. A - R \cdot \left(\frac{dA}{dt}\right) \cdot \sin. A + \left(\frac{d\varrho}{dt}\right) \cdot \cos. \alpha - \varrho \cdot \left(\frac{d\alpha}{dt}\right) \cdot \sin. \alpha ; \\ \left(\frac{dy}{dt}\right) &= \left(\frac{dR}{dt}\right) \cdot \sin. A + R \cdot \left(\frac{dA}{dt}\right) \cdot \cos. A + \left(\frac{d\varrho}{dt}\right) \cdot \sin. \alpha + \varrho \cdot \left(\frac{d\alpha}{dt}\right) \cdot \cos. \alpha ; \\ \left(\frac{dz}{dt}\right) &= \left(\frac{d\varrho}{dt}\right) \cdot \tan. \theta + \varrho \cdot \frac{\left(\frac{d\theta}{dt}\right)}{\cos. \theta} . \end{aligned}$$

The values of  $\left(\frac{dA}{dt}\right)$  and of  $\left(\frac{dR}{dt}\right)$  are given by the theory of the motion of the earth: in order to facilitate their computation, let  $E$  represent the eccentricity of the earth's orbit, and  $H$  the longitude of its perihelion; by the nature of the elliptic motion we have,

$$\left(\frac{dA}{dt}\right) = \sqrt{\frac{1-E^2}{R^2}} ; \quad R = \frac{1-E^2}{1+E \cdot \cos. (A-H)} .^*$$

These two equations give

$$\left(\frac{dR}{dt}\right) = \frac{E \cdot \sin. (A-H)}{\sqrt{1-E^2}} ;$$

let  $R'$  represent the radius vector of the earth corresponding to  $A$ , the longitude of this planet, increased by a right angle; we shall have

$$R' = \frac{1-E^2}{1-E \cdot \sin. (A-H)} ;$$

from which may be obtained

$$E \cdot \sin. (A-H) = \frac{R'-1+E^2}{R'} ;$$

\*  $\frac{dA}{dt}$  being equal to the angular velocity of the earth, it is equal to the square root of the parameter divide by the square of the distance,  $\therefore$  it is equal to  $\frac{\sqrt{1-E^2}}{R^2}$ .

consequently

$$\left(\frac{dR}{dt}\right) = \frac{R' + E^2 - 1}{R' \sqrt{1 - E^2}}.*$$

If we neglect the square of the excentricity of the terrestrial orbit, which is very small, we shall have

$$\left(\frac{dA}{dt}\right) = \frac{1}{R^2}; \quad \left(\frac{dR}{dt}\right) = R' - 1;$$

the preceding values of  $\left(\frac{dx}{dt}\right)$  and  $\left(\frac{dy}{dt}\right)$  will consequently become

$$\left(\frac{dx}{dt}\right) = (R' - 1) \cdot \cos. A - \frac{\sin. A}{R} + \left(\frac{d\varrho}{dt}\right) \cdot \cos. \alpha - \varrho \cdot \left(\frac{d\alpha}{dt}\right) \cdot \sin. \alpha.$$

$$\left(\frac{dy}{dt}\right) = (R' - 1) \cdot \sin. A + \frac{\cos. A}{R} + \left(\frac{d\varrho}{dt}\right) \cdot \sin. \alpha + \varrho \cdot \left(\frac{d\alpha}{dt}\right) \cdot \cos. \alpha;$$

$R$ ,  $R'$  and  $A$  being given immediately by the tables of the sun, the computation of the six quantities  $x$ ,  $y$ ,  $z$ ,  $\left(\frac{dx}{dt}\right)$ ,  $\left(\frac{dy}{dt}\right)$ , and  $\left(\frac{dz}{dt}\right)$  will be easy,

when  $\varrho$ , and  $\left(\frac{d\varrho}{dt}\right)$  will be known. The elements of the orbit of the comet can be deduced from them, in the following manner.

### D D 2

$$* \quad \frac{dR}{dt} = \frac{dA}{dt} \cdot \frac{(1 - E^2) \cdot E \cdot \sin. (A - H)}{(1 + E \cdot \cos. (A - H))^2} = \frac{\sqrt{1 - E^2}}{(1 - E^2)^2} \cdot (1 + E \cdot \cos. (A - H))^2.$$

$$\frac{(1 - E^2) \cdot E \cdot \sin. (A - H)}{(1 + E \cdot \cos. (A - H))^2} = \frac{E \cdot \sin. (A - H)}{\sqrt{1 - E^2}}; \quad R' = \frac{1 - E^2}{1 + E \cdot \cos. (A + 90 - H)}$$

$$= \frac{1 - E^2}{1 - E \cdot \sin. (A - H)}; \quad = (1 - E^2) \cdot (1 - E \cdot \sin. (A - H))^{-1} = (\text{when the square of } E \text{ is neglected}) \frac{1 - E^2}{1 + E \cdot \sin. (A - H)}, \therefore R' - 1 (= E \cdot \sin. (A - H)) \text{ is equal} \left(\frac{dR}{dt}\right), \text{ when } E^2 \text{ is neglected.}$$

The indefinitely small sector, which the projection of the radius vector of the comet on the plane of the ecliptic, describes during the element of time  $dt$ , is  $\frac{x dy - y dx}{2}$ ; and it is manifest that this sector is positive or negative according as the motion of the comet is direct or retrograde; thus, the sign of the quantity  $x \cdot \left(\frac{dy}{dt}\right) - y \cdot \left(\frac{dx}{dt}\right)$ , will indicate the direction of the motion of the comet.

In order to determine the position of the orbit, let us name  $\phi$ , its inclination to the ecliptic, and  $I$  the longitude of the node, which will be the ascending one, if the motion of the comet be direct; we shall have

$$z = y \cdot \cos. I \cdot \tan \phi - x \cdot \sin. I \cdot \tan \phi.$$

This equation, combined with its differential, gives

$$\tan. I = \frac{y \cdot \left(\frac{dz}{dt}\right) - z \cdot \left(\frac{dy}{dt}\right)}{x \cdot \left(\frac{dz}{dt}\right) - z \cdot \left(\frac{dx}{dt}\right)};^*$$

\*  $z = \tan. \phi$ . multiplied into the distance of  $z$  from the line of the nodes, and if the axis of  $x$  be a line drawn to the first point of Aries, this last distance  $= y \cdot \cos. I - x \cdot \sin. I$ .

$$\begin{aligned} dz &= dy \cdot \cos. I \cdot \tan \phi - dx \cdot \sin. I \cdot \tan \phi; \therefore \frac{y \cdot \frac{dz}{dt} - z \cdot \frac{dy}{dt}}{x \cdot \frac{dz}{dt} - z \cdot \frac{dx}{dt}} = \\ &\frac{\left( y \cdot \frac{dy}{dt} \cdot \cos. I - y \cdot \frac{dx}{dt} \cdot \sin. I - y \cdot \frac{dy}{dt} \cdot \cos. I + x \cdot \frac{dy}{dt} \cdot \sin. I \right) \cdot \tan. \phi}{\left( x \cdot \frac{dy}{dt} \cdot \cos. I - x \cdot \frac{dx}{dt} \cdot \sin. I - y \cdot \frac{dx}{dt} \cdot \cos. I + x \cdot \frac{dx}{dt} \cdot \sin. I \right) \cdot \tan. \phi}. \\ &= \frac{\left( x \cdot \frac{dy}{dt} - y \cdot \frac{dx}{dt} \right) \cdot \sin. I}{\left( x \cdot \frac{dy}{dt} - y \cdot \frac{dx}{dt} \right) \cdot \cos. I} = \tan. I. \\ &= \tan. I. \end{aligned}$$

$$\tan. \varphi = \frac{y. \left( \frac{dz}{dt} \right) - z. \left( \frac{dy}{dt} \right)}{\sin. I. \left\{ x. \left( \frac{dy}{dt} \right) - y. \left( \frac{dx}{dt} \right) \right\}}$$

$\varphi$  must be always positive and less than a right angle ; this condition determines the sign of  $\sin. I$  ; but the tangent of  $I$ , and the sign of its sine being determined, the angle  $I$  is entirely determined. This angle is equal to the longitude of the ascending node of the orbit, provided that the motion is direct, but if the motion is retrograde, we must add to it two right angles, in order to obtain the longitude of this node. It will be simpler to consider only the direct motions, by making  $\varphi$  the inclination of the orbits to vary, from zero to two right angles ; for it is manifest, that then the retrograde motions correspond to an inclination greater than a right angle. In this case,  $\tan. \varphi$  is of the same sign as  $x. \left( \frac{dy}{dt} \right) - y. \left( \frac{dx}{dt} \right)$ , which determines  $\sin. I$ , and consequently the angle  $I$ , which expresses always the longitude of the ascending node.

$a$  and  $ea$  representing the semiaxis major, and excentricity of the orbit, we have, by N°s. 18 and 19,  $\mu$  being supposed = 1,

$$\text{By substituting we obtain } \frac{\left( y. \frac{dz}{dt} - z. \frac{dy}{dt} \right)}{\sin. I. \left( x. \frac{dy}{dt} - y. \frac{dx}{dt} \right)} =$$

$$\frac{y. \frac{dy}{dt}. \cos. I - y. \frac{dx}{dt}. \sin. I - y. \frac{dy}{dt}. \cos. I + x. \frac{dy}{dt}. \sin. I. \tan. \varphi.}{\left( x. \frac{dy}{dt} - y. \frac{dx}{dt} \right). \sin. I.}$$

$$= \frac{\left( x. \frac{dy}{dt} - y. \frac{dx}{dt} \right). \sin. I. \tan. \varphi.}{\left( x. \frac{dy}{dt} - y. \frac{dx}{dt} \right). \sin. I.} = \tan. \varphi.$$

$$\frac{1}{a} = \frac{2}{r} - \left( \frac{dx}{dt} \right)^2 - \left( \frac{dy}{dt} \right)^2 - \left( \frac{dz}{dt} \right)^2;$$

$$a.(1-e^2) = 2r - \frac{r^2}{a} - \left\{ x. \left( \frac{dx}{dt} \right) + y. \left( \frac{dy}{dt} \right) + z. \left( \frac{dz}{dt} \right) \right\}^2.$$

The first of these equations determines the semiaxis major of the orbit, and the second determines its excentricity. The sign of the function  $x. \left( \frac{dx}{dt} \right) + y. \left( \frac{dy}{dt} \right) + z. \left( \frac{dz}{dt} \right)$  makes known whether the comet has already passed through its perihelion ; for if this function is negative, it approaches towards it ; in the contrary case, it has already passed this point.

Let  $T$  represent the interval of time comprised between the epoch which we have selected, and the passage of the comet through the perihelion ; the two first of the equations ( $f$ ) of No. 20, will give, by observing that  $\mu$  being supposed equal to unity, we have  $n=a^{-\frac{5}{2}}$ ,

$$r=a.(1-e. \cos. u); \quad T=a^{\frac{5}{2}}.(u-e. \cos. u).$$

The first of these equations gives the angle  $u$ , and the second makes known the time  $T$ . This time added or subtracted from the epoch, according as the comet approaches or departs from the perihelion, will give the instant of its passage through this point. The values of  $x$  and of  $y$ , determine the angle which the projection of the radius vector  $r$  makes with the axis of  $x$ , and as we know the angle  $I$  made by this axis, with the line of the nodes, we shall have the angle which this last line constitutes with the projection of  $r$  ; from which may be obtained, by means of the inclination  $\varphi$  of the orbit, the angle which the line of the nodes makes with the radius  $r$ . But the angle  $u$  being known, we shall have by means of the third of the equations ( $f$ ), of No. 20, the angle  $v$ , which this radius forms, with the line of the apsides ; therefore we will have the angle comprised between the two lines, of the apsides and the nodes, and, consequently, the position of the peri-

helion. All the elements of the orbit will be thus determined.

34. These elements are given, by what precedes, in functions of  $\xi$  ( $\frac{d\xi}{dt}$ ), and of known quantities; and as ( $\frac{d\xi}{dt}$ ) is given in  $\xi$ , by No. 31; the elements of the orbit will be functions of  $\xi$ , and of known quantities. If one of them was given, we would have a new equation, by means of which we could determine  $\xi$ ; this equation will have a common divisor with the equation (4) of No. 31; and seeking this divisor by the ordinary method we will arrive at an equation of the first degree in  $\xi$ , we shall have besides, an equation of condition between the data of the observations, and this equation will be that which should have place, in order that the given element might belong to the orbit of the comet.

Let us now apply this consideration to nature. For this purpose, we may observe that the orbits of the comets are very elongated ellipses, which are sensibly confounded with a parabola, in that part of their orbit in which these stars are visible; therefore we may suppose without sensible error, that  $a = \infty$ , and  $\frac{1}{a} = 0$ ; consequently the

expression for  $\frac{1}{a}$  of the preceding No. will give,

$$0 = \frac{2}{r} - \frac{(dx^2 + dy^2 + dz^2)}{dt^2}.$$

If we afterwards substitute, instead of ( $\frac{dx}{dt}$ ), ( $\frac{dy}{dt}$ ), ( $\frac{dz}{dt}$ ) their values, which are found in the same No.; we shall have, after all reductions, and by neglecting the square of  $R' - 1$ ,

$$0 = \left( \frac{d\xi}{dt} \right)^2 + \xi^2 \cdot \left( \frac{d\alpha}{dt} \right)^2 + \left\{ \left( \frac{d\xi}{dt} \right) \cdot \tan. \theta + \xi \cdot \frac{\left( \frac{d\theta}{dt} \right)}{\cos. \theta} \right\}^2$$

$$+2\cdot\left(\frac{d\varrho}{dt}\right)\left\{(R'-1)\cdot\cos.(A-\alpha)-\frac{\sin.(A-\alpha)}{R}\right\} \quad (5)$$

$$+2\varrho\cdot\left(\frac{d\alpha}{dt}\right)\cdot\left\{(R'-1)\cdot\sin.(A-\alpha)+\frac{\cos.(A-\alpha)}{R}\right\}+\frac{1}{R^2}-\frac{2}{r};$$

by substituting in this equation, instead of  $\left(\frac{d\varrho}{dt}\right)$  its value

$$\frac{-\xi}{2\cdot\left(\frac{d\alpha}{dt}\right)}\cdot\left\{\left(\frac{d^2\alpha}{dt^2}\right)+\mu'\cdot\sin.(A-\alpha)\right\}^*$$

which has been found in No. 31; and then by making

$$4\cdot\left(\frac{d\alpha}{dt}\right)^2\cdot B=4\cdot\left(\frac{d\alpha}{dt}\right)^4+\left\{\left(\frac{d^2\alpha}{dt^2}\right)+\mu'\cdot\sin.(A-\alpha)\right\}^2$$

$$+\left\{\tan.\theta\cdot\left(\frac{d^2\alpha}{dt^2}\right)+\mu'\cdot\tan.\theta\cdot\sin.(A-\alpha)-2\cdot\frac{\left(\frac{d\alpha}{dt}\right)\cdot\left(\frac{d\theta}{dt}\right)}{\cos.^2\theta}\right\}^2;$$

$$C=\frac{\left\{\left(\frac{d^2\alpha}{dt^2}\right)+\mu'\cdot\sin.(A-\alpha)\right\}}{\left(\frac{d\alpha}{dt}\right)}\cdot\left\{\frac{\sin.(A-\alpha)}{R}-(R'-1)\cdot\cos.(A-\alpha)\right\}$$

\* By making this substitution, the equation (5) becomes

$$4\cdot\frac{\xi^2}{dt^2}\left(\frac{d^2\alpha}{dt^2}+\mu'\cdot\sin.(A-\alpha)\right)^2+\xi^2\cdot\left(\frac{d\alpha}{dt}\right)^2+\xi^2\cdot\left\{\frac{\left(\frac{d^2\alpha}{dt^2}+\mu'\cdot\sin.(A-\alpha)\right)\cdot\tan.\theta}{-2\cdot\left(\frac{d\alpha}{dt}\right)}\right.$$

$$\left.+\frac{\left(\frac{d\theta}{dt}\right)}{\cos.^2\theta}\right\}^2-\frac{\xi}{dt}\left(\frac{d^2\alpha}{dt^2}+\mu'\cdot\sin.(A-\alpha)\right)\cdot(R'-1)\cdot\cos.(A-\alpha)-\sin.\frac{(A-\alpha)}{R})'$$

$$+2\xi\cdot\left(\frac{d\alpha}{dt}\right)\left\{(R'-1)\cdot\sin.(A-\alpha)+\frac{\cos.(A-\alpha)}{R}\right\}+\frac{1}{R^2}-\frac{2}{r}.$$

It is evident from an inspection of this expression, that  $B$  is equal to the quantity by which  $\xi^2$  is multiplied, and that  $C$  is equal to the corresponding factor of  $\xi$ .

$$+ 2 \cdot \left( \frac{d\alpha}{dt} \right) \cdot \left\{ (R' - 1) \cdot \sin. (A - \alpha) + \frac{\cos. (A - \alpha)}{R} \right\};$$

we shall have

$$0 = B \cdot \xi^2 + C \cdot \xi + \frac{1}{R^2} - \frac{2}{r};$$

and consequently

$$r^2 \cdot \left\{ B \cdot \xi^2 + C \cdot \xi + \frac{1}{R^2} \right\}^2 = 4;$$

this equation is only of the sixth degree, and in\* this respect it is simpler than the equation (4) of No. 31; but it is peculiar to the parabola, on the contrary, the equation (4) is applicable to every species of conic section.

35. We may perceive by the preceding analysis, that the determination of the parabolic orbits of comets, leads to more equations† than unknown quantities, we can, by different combinations of these equations, form as many different methods of calculating these orbits. Let us investigate those from which we ought to expect the greatest precision in the results, or which participate the least in the errors of observations.

It is principally on the values of the second differences  $\left( \frac{d^2 \alpha}{dt^2} \right)$  and  $\left( \frac{d^2 \theta}{dt^2} \right)$ , that these errors have a sensible influence; in fact, it is necessary, in order to determine them, to take the finite differences of the geocentric longitudes and latitudes of the comet, observed during a

\* This equation is of the sixth degree for  $\xi^4$  and  $r^2$  occurs in it, and if we substitute for  $r^2$  its value, in terms of  $\xi$ ;  $\xi^6$  will be the highest dimension of  $\xi$  which occurs in it.

† The reason why there are more equations than unknown quantities in this case, is because the axis major is supposed to be infinite.

very short interval of time ; but these differences being less than the first differences, the errors of observation are a greater aliquot part of them ; besides, the formulae of No. 29, which determine, by the comparison of observations, the values of  $\alpha$ ,  $\theta$ ,  $(\frac{d\alpha}{dt})$ ,  $(\frac{d\theta}{dt})$ ,  $(\frac{d^2\alpha}{dt^2})$ , and  $(\frac{d^2\theta}{dt^2})$ ,

determine with greater precision the four first of these quantities, than the two last ; it is therefore advantageous to rely as little as possible on the second differences of  $\alpha$  and of  $\theta$  ; and as we cannot reject them all at once, the method which only employs the greatest ought to lead to the most exact results ; this being premised,

Let the equations which have been found in the N°s. 31 and 34, be resumed

$$r^2 = \frac{\xi^2}{\cos. \xi \theta} + 2R\xi \cdot \cos. (A - \alpha) + R^2 ;$$

$$\left( \frac{d\xi}{dt} \right) = \frac{R \cdot \sin. (A - \alpha)}{2 \cdot \left( \frac{d\alpha}{dt} \right)} \cdot \left\{ \frac{1}{R^3} - \frac{1}{r^3} \right\} - \frac{\xi \cdot \left( \frac{d^2\alpha}{dt^2} \right)}{2 \cdot \left( \frac{d\alpha}{dt} \right)} ; \quad (\text{L})$$

$$\left( \frac{d\theta}{dt} \right) = -\frac{1}{2} \rho \cdot \left\{ \frac{\left( \frac{d^2\theta}{dt^2} \right)}{\left( \frac{d\theta}{dt} \right)} + 2 \cdot \left( \frac{d\theta}{dt} \right) \cdot \tan. \theta + \frac{\left( \frac{d\alpha}{dt} \right)^2 \cdot \sin. \theta \cdot \cos. \theta}{\left( \frac{d\theta}{dt} \right)} \right\}$$

$$+ R \cdot \frac{\sin. \theta \cdot \cos. \theta \cdot \cos. (A - \alpha)}{2 \cdot \left( \frac{d\theta}{dt} \right)} \cdot \left\{ \frac{1}{r^3} - \frac{1}{R^3} \right\} ;$$

$$0 = \left( \frac{d\xi}{dt} \right)^2 + \xi^2 \cdot \left( \frac{d\alpha}{dt} \right)^2 + \left\{ \left( \frac{d\xi}{dt} \right) \cdot \tan. \theta + \xi \cdot \left( \frac{d\theta}{dt} \right) \right\}^2$$

$$+ 2 \cdot \left( \frac{d\xi}{dt} \right) \cdot \left\{ (R' - 1) \cdot \cos. (A - \alpha) - \frac{\sin. (A - \alpha)}{R} \right\}$$

$$+2\epsilon \cdot \left(\frac{d\alpha}{dt}\right) \cdot \left\{ (R'-1) \cdot \sin. (A-\alpha) + \frac{\cos. (A-\alpha)}{R} \right\} + \frac{1}{R^2} - \frac{2}{r}.$$

If we wish to reject  $\left(\frac{d^2\theta}{dt^2}\right)$ , it is only necessary to consider the first, the second, and the fourth of these equations ; eliminating  $\left(\frac{d\epsilon}{dt}\right)$ , from the last, by means of the second, we will obtain an equation which freed from fractions will contain a term multiplied by  $r^6 \cdot \epsilon^2$ , and other terms affected with even and odd powers of  $\epsilon$  and of  $r$ . If all the terms affected with the even powers of  $r$ , be reduced into one member, and likewise all the terms affected with the odd powers of  $r$  ;\* the term multiplied by  $r^6 \cdot \epsilon^2$  will produce one multiplied by  $r^2 \cdot \epsilon^4$  ; therefore by substituting instead of  $r^2$ , its value given by the first of the equations (L), we shall have a final equation of the sixteenth degree in  $\epsilon$ . But instead of forming this equation, in order afterwards to resolve it, it will be simpler to satisfy by trials, the three preceding equations.

If we wish to reject  $\left(\frac{d^2\alpha}{dt^2}\right)$  ; we must consider the first, the third and the fourth of the equations (L). These three equations would also lead us to a final equation of the sixteenth degree in  $\epsilon$ , which can be easily satisfied by trials.

The two preceding methods appear to me the most exact which can be employed in the determination of the parabolic orbits of the comets ; it is even indispensably requisite to recur to them, if the motion of the comet in longitude or in latitude is insensible, or too small for the errors of the observations not to alter sensibly its second difference ; in this case we should reject that one of the equations (L), which contains this difference. But although in these methods, we

\* By squaring each member, we get rid of the odd powers of  $r$ , and the value of any even power will be obtained by means of the first of the equations (L).

only employ three of the preceding equations ; yet the fourth is useful, in order to determine amongst all the real and positive values of  $\epsilon$ , which satisfy the system of the three other equations, that which ought to be admitted.

36. The elements of the orbit of a comet, determined by what precedes, would be exact, if the values of  $\alpha$ ,  $\theta$ , and of their first and second differences, were rigorously correct ; because we have taken into account in a very simple manner, the excentricity of the earth's orbit, by means of the radius vector  $R'$  of the earth, corresponding to its true anomaly, increased by a right angle ; we are only permitted to neglect the square of this excentricity, as being too small a fraction for its omission to influence sensibly the results. But  $\theta$ ,  $\alpha$ , and their differences, are always liable to some inaccuracy, as well on account of the errors of observation, as because these differences are collected from the observations in an approximate manner. It is therefore necessary to correct these elements by means of three observations at considerable intervals from each other, which may be effected in an indefinite number of ways ; for if we know very nearly two quantities relative to the motion of a comet, such as the radii vectores corresponding to the two observations, or the position of the node, and the inclination of its orbit, by computing the observations, at first with these quantities, and then with other quantities which differ very little from them ; the law of the differences between these results, will easily make known the corrections which those quantities ought to undergo. But among the binary combinations of quantities relative to the motion of the comets, there is one of which the calculation is the simplest, and which on this account deserves to be preferred ; and in a problem so complicated, it is a matter of importance, to spare the computer every superfluous operation. The two elements which have appeared to me to afford this advantage, are the perihelion distance, and the instant of the passage of the comet through this point ; for they not only may be readily deduced from the values of  $\epsilon$  and of  $(\frac{d\epsilon}{dt})$  ; but it also

is very easy to correct them by observations, without being obliged, at each variation which they are made to undergo, to determine the other corresponding elements of the orbit.

Let us resume the equation which has been found in No. 19,

$$a.(1-e^2) = 2r - \frac{r^2}{a} - \frac{r^2 \cdot dr^2}{dt^2};$$

$a.(1-e^2)$  is the semiparameter of the conic sections of which  $a$  is the semiaxis major, and  $ea$  the excentricity; in the parabola, where  $a$  is infinite, and  $ea$  equal to unity,  $a.(1-e^2)$  is equal to twice the perihelion distance; let  $D$  equal this distance, the preceding equation becomes, relatively to this curve,

$$D = r - \frac{1}{2} \cdot \left\{ \frac{rdr}{dt} \right\}^2.$$

$\frac{rdr}{dt}$  is equal to  $\frac{\frac{1}{2}d.r^2}{dt^2}$ ; by substituting in place of  $r^2$ , its value  $\frac{e^2}{\cos^2 \theta} + 2R \cdot \cos(A-\alpha) + R^2$ , and instead of  $\left\{ \frac{dR}{dt} \right\}$  and of  $\left\{ \frac{dA}{dt} \right\}$ ,

their values found in No. 33, we shall have

$$\begin{aligned} \frac{rdr}{dt} &= \frac{e}{\cos^2 \theta} \cdot \left\{ \left\{ \frac{d\varphi}{dt} \right\} + e \cdot \left\{ \frac{d\theta}{dt} \right\} \cdot \tan \theta \right\} + R \cdot \left\{ \frac{d\varphi}{dt} \right\} \cdot \cos(A-\alpha)^* \\ &\quad + e \cdot \left\{ (R-1) \cdot \cos(A-\alpha) - \frac{\sin(A-\alpha)}{R} \right\} \\ &\quad + e \cdot R \cdot \left\{ \frac{d\alpha}{dt} \right\} \cdot \sin(A-\alpha) + R \cdot (R'-1). \end{aligned}$$

$$\begin{aligned} * \quad \frac{rdr}{dt} &= \frac{e \cdot d\varphi}{\cos^2 \theta \cdot dt} + e^2 \cdot \frac{\sin \theta}{\cos^3 \theta} \cdot \left( \frac{d\theta}{dt} \right) + \cos(A-\alpha) \cdot \left\{ e \cdot \left( \frac{dR}{dt} \right) + R \cdot \left( \frac{d\varphi}{dt} \right) \right\} \\ &\quad - R \cdot \sin(A-\alpha) \left\{ \left( \frac{dA}{dt} \right) - \left( \frac{d\alpha}{dt} \right) \right\} + R \cdot \left( \frac{dR}{dt} \right); \text{ and by substituting } R'-1 \text{ for } \left( \frac{dR}{dt} \right), \end{aligned}$$

Let  $P$  represent this quantity; if it is negative the radius vector  $r$  goes on diminishing, and consequently the comet\* tends towards its perihelion; but it moves from it, if  $P$  is positive. We have then

$$D = r - \frac{1}{2} \cdot P^2;$$

the angular distance  $v$  of the comet from the perihelion will be determined by the polar equation of the parabola

$$\cos. \frac{1}{2}v = \frac{D}{r};$$

finally, the time employed to describe the angle  $v$  will be obtained, by the table of the motion of comets. This time added or subtracted from that of the epoch, according as  $P$  is negative or positive, will give the moment of the passage through the perihelion.

37. These different results being collected together, will give the following method, for determining the parabolick orbits of comets.

*A general method for determining the Orbits of the Comets.*

This method will be divided into two parts; in the first, we will give the means of obtaining very nearly the perihelion distance of the comet, and the instant of its passage through the perihelion; in the second, we will determine exactly all the elements of the orbit, these quantities being supposed to be very nearly known.

and  $\frac{1}{R^2}$  for  $\left(\frac{dA}{dt}\right)$  we shall have  $\frac{rdr}{dt} = \frac{\xi}{\cos^2 \theta} \left\{ \left(\frac{d\xi}{dt}\right)^2 + \xi \cdot \frac{\sin. \theta}{\cos. \theta} \cdot \left(\frac{d\theta}{dt}\right) \right\} + \cos. (A-\alpha).$

$\xi \cdot (R'-1) + \cos. (A-\alpha) \cdot R \cdot \frac{d\xi}{dt} = \frac{R \cdot \xi \cdot \sin. (A-\alpha)}{R^2} + R \cdot \xi \cdot \sin. (A-\alpha) \cdot \left(\frac{d\alpha}{dt}\right) + R \cdot (R'-1).$

*An approximate determination of the perihelion distance of a Comet,  
and of the instant of its passage through perihelion.*

Let three, four, or five, &c. observations of the comet be selected as nearly as possible\* equi-distant from each other; with four observations we can embrace an interval of  $30^\circ$ ; with five observations, an interval of  $36^\circ$ , or  $40^\circ$ , and so on of the rest; but it is necessary always that the interval comprised between the observations should be more considerable, as they are more numerous, in order to diminish the influence of their errors; this being premised,

Let  $\epsilon, \epsilon', \epsilon'', \&c.$  be the successive geocentric longitudes of the comet;  $\gamma, \gamma', \gamma'',$  the corresponding latitudes, these latitudes being supposed positive or negative, according as they are north or south. Let the difference  $\epsilon' - \epsilon$  be divided by the number of days which separates the first from the second observation; in like manner, the difference  $\epsilon'' - \epsilon'$  be divided by the number of days which separates the second from the third observation; we will also divide the difference  $\epsilon''' - \epsilon''$ , by the number of days which separates the third from the fourth observation, and so of the rest. Let  $\delta\epsilon, \delta\epsilon', \delta\epsilon'',$  be these quotients; let the difference  $\delta\epsilon' - \delta\epsilon$ , be divided by the number of days which separates the first observation from the third; in like manner let the difference  $\delta\epsilon'' - \delta\epsilon'$ , be divided by the number of days which separates the second observation from the

\* The precision which might be expected from an increased number of observations would not (as M. Laplace has since ascertained) compensate for the errors to which the observations are liable, and also for the greater length of the calculus; he therefore proposes in the 15th Book, to employ only three observations, and by fixing the epoch at the intermediate observation, to render the extreme observations at such inconsiderable distances from each other, that for the interval which separates them, the preceding data may be supposed very nearly the same; an additional advantage in having the intervals short is, that the differences superior to the second are inconsiderable, and may therefore be neglected.

fourth ; and  $\delta^2\epsilon' - \delta^2\epsilon''$ , by the number of days which separates the third observation from the fifth ; and so of the rest. Let  $\delta^2\epsilon$ ,  $\delta^2\epsilon'$ ,  $\delta^2\epsilon''$ , represent these quotients.

Dividing the difference  $\delta^2\epsilon' - \delta^2\epsilon$ , by the number of days which separates the first observation from the fourth ; and in like manner  $\delta^2\epsilon'' - \delta^2\epsilon'$ , by the number of days which intervenes between the second and fifth observation, and so of the rest. Let  $\delta^3\epsilon$ ,  $\delta^3\epsilon'$ , &c. represent these quotients. Let these operations be continued till we arrive at  $\delta^{n-1}\epsilon$ ,  $n$  being the number of observations employed.

This being performed, let an observation which is a mean, or very nearly so between the instants of the extreme observations be selected, and let  $i$ ,  $i'$ ,  $i''$ ,  $i'''$ , &c. represent the number of days by which it precedes each observation,  $i$ ,  $i'$ ,  $i''$ , being supposed to be negative for the observations which are anterior to this epoch ; the longitude of the comet, after a small number  $z$  of days reckoned from the epoch, will be expressed by the following formula :

$$\begin{aligned} \epsilon &= i.\delta\epsilon + i.i'.\delta^2\epsilon - i.i'.i''.\delta^3\epsilon + \text{&c.} \\ &+ z.( \delta\epsilon - (i+i').\delta^2\epsilon + (i.i' + i.i'' + i'.i'').\delta^3\epsilon - (i.i'.i'' + i.i'.i''' + i.i''.i''') + \\ &\quad i'.i''.i''').\delta^4\epsilon + \text{&c.} ); \quad (p) \\ &+ z^2.( \delta^2\epsilon - (i+i'+i'').\delta^3\epsilon + i.i' + i.i'' + i.i''' + i'.i'' + i'.i''').\delta^4\epsilon - \text{&c.} ). \end{aligned}$$

The coefficients of  $-\delta\epsilon$ ,  $+\delta^2\epsilon$ ,  $-\delta^3\epsilon$ , &c. in the part which is independent of  $z$ , are, 1<sup>st</sup>. the number  $i$ ; 2<sup>nd</sup>. the product of the two numbers  $i$  and  $i'$ ; 3<sup>rd</sup>. the product of the three numbers  $i$ ,  $i'$ ,  $i''$ , &c.

The coefficients of  $-\delta^2\epsilon$ ,  $+\delta^3\epsilon$ ,  $-\delta^4\epsilon$ , &c. in the part multiplied by  $z$  are, 1<sup>st</sup>. the sum of the two numbers  $i$  and  $i'$ ; 2<sup>nd</sup>. the sum of the binary products of the three numbers  $i$ ,  $i'$ ,  $i''$ ; 3<sup>rd</sup>. the sum of the products of the four numbers  $i$ ,  $i'$ ,  $i''$ ,  $i'''$ , &c. taken three by three.

The coefficients of  $-\delta^3\epsilon$ ,  $+\delta^4\epsilon$ ,  $-\delta^5\epsilon$ , &c. in the part multiplied by  $z^2$ , are 1<sup>st</sup>. the sum of the three numbers  $i$ ,  $i'$ ,  $i''$ ; 2<sup>nd</sup>. the sum of the products of the four numbers  $i$ ,  $i'$ ,  $i''$ ,  $i'''$ , taken two by two; 3<sup>rd</sup>. the

sum of the products of the five numbers,  $i, i', i'', i''', \&c.$  taken three by three.

In place of forming these products, it would be as simple to develop the function

$$\epsilon + (z-i) \cdot \delta\epsilon + (z-i) \cdot (z-i') \cdot \delta^2\epsilon + (z-i') \cdot (z-i'') \cdot (z-i''') \cdot \delta^3\epsilon + \&c.$$

the powers of  $z$  superior to the second, which the preceding formulæ would give, being rejected.

If we perform similar operations on the observed geocentric latitudes of the comet; its geocentric latitude after the number  $z$  of days, reckoned from the epoch, will be expressed by the formula ( $p$ ), by changing  $\epsilon$  into  $\gamma$ ; and let ( $q$ ) represent what this formula becomes after this change; this being premised,  $\alpha$  will be the part independent of  $z$ , in the formula ( $p$ );  $\theta$  will be the part independent of  $z$ , in the formula ( $q$ ).

If the coefficient of  $z$  be reduced to seconds, in the formula ( $p$ ), and if the logarithm 4,0394622 be subducted from the tabular logarithm of this number of seconds; it will give the logarithm of a number which we will denote by  $a$ .

And if the coefficient of  $z^2$  in the same formula be reduced to seconds, and if the logarithm 1,9740144 be then subtracted from this number of seconds, it will give the logarithm of a number, which we will denote by  $b$ .

The coefficients of  $z$  and of  $z^2$  being in like manner reduced to seconds in the formula ( $q$ ), and then the logarithms 4,0394622, and 1,9740144 being subducted from the logarithms of these numbers respectively, will give the logarithms of two numbers, which we will denote by  $h$  and  $l$ .

The accuracy of this method depends on the precision of the values of  $a, b, h, l$ ; and as their formation is very simple, we should select and multiply the observations, so as to obtain them with all the precision which the observations admit of. It is easy to perceive that these

values are the quantities  $\left(\frac{d\alpha}{dt}\right)$ ,  $\left(\frac{d^2\alpha}{dt^2}\right)$ ,  $\left(\frac{d\theta}{dt}\right)$  and  $\left(\frac{d^2\theta}{dt^2}\right)$ , which for greater simplicity we have expressed by the preceding letters.

If the number of observations be odd, we can fix the epoch at the instant of the mean observation; this enables us to dispense with the computation of the parts independent of  $z$ , in the two preceding formulæ; for it is evident that these values are then respectively equal to the longitude and latitude of the mean observation.

The values of  $\alpha$ ,  $a$ ,  $b$ ,  $\theta$ ,  $h$  and  $l$ , being thus determined; the longitude of the sun at the instant which we select for the epoch, must next be determined; let this longitude be equal to  $E$ ,  $R$  being the corresponding distance of the sun from the earth, and  $R'$  the distance which answers to  $E$  increased by a right angle, the following equations will be obtained,

$$r^2 = \frac{x^2}{\cos. z\theta} - 2R.x. \cos. (E-\alpha) + R^2; \quad (1)$$

$$\dot{y} = \frac{R. \sin. (E-\alpha)}{2a} \cdot \left\{ \frac{1}{r^3} - \frac{1}{R^3} \right\} - \frac{bx}{2a}; \quad (2)$$

$$y = -x \cdot \left\{ h. \tan. \theta + \frac{l}{2h} + \frac{a^2. \sin. \theta. \cos. \theta}{2h} \right\} \\ + \frac{R. \sin. \theta. \cos. \theta}{2h} \cdot \cos. (E-\alpha) \cdot \left\{ \frac{1}{R^3} - \frac{1}{r^3} \right\}. \quad (3)$$

$$0 = y^2 + a^2 \cdot x^2 + \left\{ y. \tan. \theta + \frac{h \cdot x}{\cos. z\theta} \right\}^2 + 2y \cdot \left\{ \frac{\sin. (E-\alpha)}{R} - (R' - 1) \cdot \cos. (E-\alpha) \right\} \quad (4)$$

\* All the observations made in the interval between the extreme observations may be made use of in determining  $\alpha$ ,  $a$ ,  $b$ ,  $\theta$ ,  $h$ , and  $l$ ; for if each observation be expressed in a linear function of these data, there will be more equations than unknown quantities; the first final equation will be obtained if each equation be multiplied by the coefficient the first unknown quantity, the second final equation will be obtained by a similar process, and so on; and the data will be given by a resolution of these equations with a precision which will be greater, as more observations are made use of. This advantage is peculiar to this method. (See Connaissance des Temps, Année 1824.)

$$-2ax.((R'-1). \sin. (E-\alpha) + \frac{\cos. (E-\alpha)}{R}) + \frac{1}{R^2} + \frac{2}{r}.$$

In order to deduce from these equations, the values of the unknown quantities  $x$ ,  $y$ , and  $r$ ; we must consider, in the first place, whether, abstracting from the sign,  $b$  is greater or less than  $l$ . In the first case, we employ the equations (1), (2) and (4). We make a first supposition for  $x$ , by supposing it, for example, equal to unity; and from this we conclude, by means of the equations (1) and (2), the values of  $r$  and of  $y$ . We substitute then, these values in the equation (4), and if the remainder vanishes, it shews that the value of  $x$  has been rightly assumed; but if this remainder be negative, the value of  $x$  must be increased, and it must be diminished, if this remainder be positive. By this means, we shall obtain by a small number of trials, the values of  $x$ ,  $y$ , and  $r$ . But as these unknown quantities are susceptible of several real and positive values, it is necessary to select that value which satisfies exactly or very nearly the equation (3).

## F F 2

Since the publication of this book M. Laplace has ascertained that the best means of diminishing the influence which the errors of observation have on their results, consists in combining the equations (2) and (3), by multiplying the first by  $a^2$ , and the second by  $h^2$ , and then adding the products together, by means of which the following equation will be obtained,

$$\left. \begin{aligned} y &= \frac{a \sin. (E-a) - h. \sin. \theta. \cos. \theta. \cos. (E-a). R}{2.(a^2+h^2)} \cdot \left( \frac{1}{r^3} - \frac{1}{R^3} \right) \\ &- \frac{x. h^3. \tan. \theta + \frac{1}{2} al. + \frac{1}{2} h.l + \frac{1}{2} a^2 h. \sin. \theta. \cos. \theta}{a^2+h^2} \end{aligned} \right\} \quad (5)$$

This equation combined with the equations (1), (4), will give the values of  $x$ ,  $y$ ,  $r$ . By making a first hypothesis for  $x$ , the equations (a) will give the corresponding values of  $r$ , and then the equation (5) will give  $y$ . Now if the value of  $x$  has been properly assumed, these values, when substituted in the equation (4) ought to satisfy it; if this equation is not satisfied, a second value of  $x$  should be taken, and so on. Hence the perhelion distance  $D$ , and the instant of the passage through the perhelion, may be determined.

In the second case, *i.e.*, if we have  $l > b$ , we must employ the equations (1), (3), and (4), and then the equation (2) will serve to verify the values deduced from these equations.

Having by this means obtained the values of  $x$ ,  $y$ , and  $r$ ; let  $P$  be assumed

$$\begin{aligned} &= \frac{x}{\cos. {}^2 \theta} \cdot \left\{ y + h \cdot x \cdot \tan. \theta \right\} - R y \cdot \cos. (E - \alpha) \\ &+ x \cdot \left\{ \frac{\sin. (E - \alpha)}{R} - (R' - 1) \cdot \cos. (E - \alpha) \right\} - R \cdot a x \cdot \sin. (E - \alpha) \\ &+ R \cdot (R' - 1). \end{aligned}$$

The perihelion distance  $D$  of the comet will be determined by the equation

$$D = r - \frac{1}{2} \cdot P^2;$$

the cosine of its anomaly  $v$  will be given by the equation

$$\cos. {}^2 \frac{1}{2} \cdot v = \frac{D}{r};$$

and from this we infer, by the table of the motion of the comets, the time employed to describe the angle  $v$ . In order to obtain the instant of the passage through the perihelion, this time should be added to the epoch, if  $P$  is negative, and subtracted from it, if  $P$  is positive, because, in the first case, the comet approaches the perihelion, and in the second case, it moves from it.

Having thus determined very nearly the perihelion distance of the comet, and the instant of its passage through the perihelion, we can correct them by the following method, which has the advantage of being independent of an approximative knowledge of the other elements of the orbit.

*An exact determination of the elements of the orbit, when we know very nearly the perihelion distance of the Comet, and the instant of its passage through the perihelion.*

In the first place, three observations of the comet, at a considerable distance from each other, should be selected, and then from the perihelion distance of the comet, and from the instant of its passage through the perihelion, as data which are determined by what precedes, we compute three anomalies of the comet, and the radii vectores which correspond to the instants of the three observations. Let  $v, v', v''$ , represent these anomalies, (those which precede the passage through the perihelion being supposed negative); moreover, let  $r, r', r''$ , represent the corresponding radii vectores of the comet;  $v' - v, v'' - v$ , will be the angles contained between  $r$  and  $r'$ , and between  $r$ , and  $r''$ ; let  $U$  be the first of these angles, and  $U'$  the second.

Likewise let  $\alpha, \alpha', \alpha''$ , represent the three observed geocentric longitudes of the comet, referred to a fixed equinox;  $\theta, \theta', \theta''$ , its three geocentric latitudes, the southern latitudes being supposed to be negative; let  $\epsilon, \epsilon', \epsilon''$ , be its three corresponding heliocentric longitudes; and  $\varpi, \varpi', \varpi''$ , its three heliocentric latitudes, finally, let  $E, E', E'$ , be the three corresponding longitudes of the sun; and  $R, R', R'$ , its three distances from the centre of the earth.

Let us suppose that the letter  $S$  indicates the centre of the sun;

$T$  that of the earth ;  $C$  the centre of the comet, and  $C'$ , its projection on the plane of the ecliptic. The angle  $STC'$ , is the difference of the geocentric longitudes of the sun and of the comet ; by adding the logarithm of the cosine of this angle, to the logarithm of the cosine of the geocentric latitude of the comet,\* we will obtain the logarithm of the cosine of the angle  $STC$ ; therefore in the triangle  $STC$  there will be given the side  $ST$  or  $R$ ; the side  $SC$  or  $r$ , and the angle  $STC$ ; we can thus by trigonometry obtain the angle  $CST$ . The heliocentric latitude of the comet will then be obtained by means of the equation

$$\sin. \varpi = \frac{\sin. \theta. \sin. CST}{\sin. CTS} . \dagger$$

The angle  $TSC'$  is the side of a right angled spherical triangle, of which the hypotenuse is the angle  $TSC$ , and of which one of the sides is the angle  $\varpi$ ; from which we can easily obtain the angle  $TSC'$ , and consequently, the heliocentric longitude  $\epsilon$  of the comet.

In like manner,  $\varpi'$ ,  $\epsilon'$ ,  $\varpi''$ ,  $\epsilon''$ ; and the values of  $\epsilon$ ,  $\epsilon'$ ,  $\epsilon''$ , will determine whether the motion of the comet is direct or retrograde.

If we conceive the two arcs of latitude  $\varpi$ ,  $\varpi'$ , to meet in the pole of the ecliptic, they will make an angle equal to  $\epsilon' - \epsilon$ ; and in the spherical triangle formed by this angle, and by the sides  $\frac{\pi}{2} - \varpi$ , and  $\frac{\pi}{2} - \varpi'$ ,

\* If  $E$  be the longitude of the sun,  $STC' = \alpha - E$ , and in the right angled spherical triangle, of which one side is the measure of  $\alpha - E$ , and the other side about the right angle the measure of  $\theta$ , the hypotenuse will be equal to the measure of the angle at the earth between the sun and comet *i.e.* equal to  $STC$ ,  $\therefore$  by Napier's rules we have  $\cos. (\alpha - E). \cos. \theta = \cos. STC$ .

†  $\sin. CST : CTS : \text{distance of comet from earth} : r :: \sin. \varpi : \sin. \theta$ ,  $\therefore \sin. \varpi = \frac{\sin. \theta. \sin. CST}{\sin. CTS}$ .

$\pi$  being the semicircumference, the side opposite to the angle  $\epsilon' - \epsilon$ , will be the angle at the sun, contained between the radii vectores  $r$  and  $r'$ . It may be easily determined, by spherical trigonometry, or by the following formula :

$$\sin. {}^{\circ}\frac{1}{2}V = \cos. {}^{\circ}\frac{1}{2}. (\varpi + \varpi') - \cos. {}^{\circ}\frac{1}{2}. (\epsilon' - \epsilon). \cos. \varpi. \cos. \varpi', *$$

in which  $V$  represents this angle ; so that if we name  $A$ , the angle of which the square of the sine is  $\cos. {}^{\circ}\frac{1}{2}. (\epsilon' - \epsilon). \cos. \varpi. \cos. \varpi'$ , and which can be readily derived from the tables, we shall obtain

$$\sin. {}^{\circ}\frac{1}{2}V = \cos. (\frac{1}{2}\varpi + \frac{1}{2}\varpi' + A). \cos. (\frac{1}{2}\varpi + \frac{1}{2}\varpi' - A).$$

Naming in like manner  $V'$  the angle constituted by the two radii vectores  $r$  and  $r'$ , we will have

$$\sin. {}^{\circ}\frac{1}{2}V' = \cos. (\frac{1}{2}\varpi + \frac{1}{2}\varpi' + A'). \cos. (\frac{1}{2}\varpi + \frac{1}{2}\varpi' - A'),$$

$A'$  being what  $A$  becomes, when  $\varpi'$  and  $\epsilon'$  are changed into  $\varpi''$  and  $\epsilon''$ . Now, if the perihelion distance of the comet, and the moment of its passage through the perihelion were accurately determined, and if the

\* This expression may be easily derived from the known formulæ of spherical trigonometry, for if we assume  $B = (\epsilon' - \epsilon)$ ;  $C = \frac{\pi}{2} - \varpi$ ;  $C' = \frac{\pi}{2} - \varpi'$ ; we shall have,  $\cos. B$

$$= \frac{\cos. V - \cos. C. \cos. C'}{\sin. C. \sin. C'}; \therefore 1 - \cos. B = 2 \sin. {}^{\circ}\frac{1}{2}. B$$

$$= \frac{\sin. C. \sin. C' - \cos. V + \cos. C. \cos. C'}{\sin. C. \sin. C'} = \frac{\cos. (C - C') - \cos. V}{\sin. C. \sin. C'};$$

$\therefore 2 \sin. {}^{\circ}\frac{1}{2} B. \sin. C. \sin. C' = \cos. (C - C') - \cos. V = 2 \sin. {}^{\circ}\frac{1}{2}. V - 2 \sin. {}^{\circ}\frac{1}{2}. (C - C')$ , and since  $\sin. {}^{\circ}\frac{1}{2}B = 1 - \cos. {}^{\circ}\frac{1}{2}B$ ; and  $\sin. {}^{\circ}\frac{1}{2}. (C - C) = \sin. {}^{\circ}\frac{1}{2}. (C + C') - \sin. C. \sin. C'$ , we shall have  $(2 - 2 \cos. {}^{\circ}\frac{1}{2}. B). \sin. C. \sin. C' = 2 \sin. {}^{\circ}\frac{1}{2}V - 2 \sin. {}^{\circ}\frac{1}{2}. (C + C') + 2. \sin. C. \sin. C'$ .  $\therefore \sin. {}^{\circ}\frac{1}{2}V = \sin. \frac{1}{2}(C + C') - \cos. {}^{\circ}\frac{1}{2}B. \sin. C. \sin. C'$ ; which will give the expression in the text when their values are substituted for  $B$ ,  $C$ ,  $C'$ .

observations were rigorous, we would have

$$V = U; \quad V' = U;$$

but as this can never be the case, we will suppose

$$m = U - V; \quad m' = U' - V'.$$

(It is to be observed here, that the computation of the triangle  $STC$ , gives for the angle  $CST$  two different\* values. Most frequently, the nature of the cometary motion will make known which of them ought to be employed, especially if these two values are very different; for then one of them will place the comet farther than the other from the earth, and it will be easy to determine, by the apparent motion of the comet at the instant of observation, which ought to be selected. But any uncertainty which remains on this account may be removed, by taking care to select that value which renders  $V$  and  $V'$  very little different from  $U$  and  $U'$ .)

Then we will make a second hypothesis, in which the instant of the transit through the perihelion remaining the same as before the perihelion distance varies by a small quantity; *e.g.* by a five hundredth part of its value, and then we seek in this hypothesis the values of  $U - V$ , and of  $U' - V'$ ; let then

$$n = U - V; \quad n' = U - V'.$$

Finally, we make a third supposition in which, the distance of the perihelion remaining the same as in the first hypothesis, we make to vary by half of a day, more or less, the instant of the passage through the perihelion. And then let the values of  $U - V$ , and of  $U' - V'$  be investigated on this new hypothesis. Let in this case

$$p = U - V; \quad p' = U' - V'.$$

This being premised, if  $u$  represents the number by which the sup-

\* The values of  $CST$ , are  $CST$ , and  $180 - 2STC - CST$ .

posed variation in the perihelion distance should be multiplied, in order to obtain the true distance, and  $t$  the number by which the supposed variation in the instant of the passage through the perihelion should be multiplied, in order to obtain the true instant; we shall have the two following equations,

$$(m - n) \cdot u + (m - p) \cdot t = m;$$

$$(m' - n') \cdot u + (m' - p') \cdot t = m';$$

by means of which equations we obtain the values of  $u$  and of  $t$ , and consequently the corrected distance of the perihelion, and true instant of the passage of the comet through the perihelion.

The preceding corrections suppose that the elements determined by the first approximation, are sufficiently accurate to enable us to treat their errors as indefinitely small. Both if the second approximation does not appear to be sufficient, we must recur to a third, by operating on the elements already corrected, as we have done on their first values; it is solely necessary in addition to secure that they undergo small variations. It will also suffice to compute by these corrected elements the values of  $U - V$ , and of  $U' - V'$ ; by representing them by  $M$  and  $N$ , and substituting them in place of  $m$  and  $m'$ , in the second members of the two preceding equations; we shall have by this means two new equations which will give the values of  $u$  and of  $t$ , relative to the corrections of these new elements.\*

Having by this method obtained the accurate distance of the peri-

If in place of computing  $U$ ,  $U'$ ,  $V$ ,  $V'$ , on the three hypothesis mentioned in the text, they were computed on the five following hypotheses, 1st, with the elements found in the first approximation; 2dly, by making the perihelion distance to vary by a very small quantity; 3dly, by making it to vary by twice the same quantity; 4thly, the same perihelion distance as in the first hypothesis being preserved, by making the instant of the passage

helion, and the true instant of the passage of the comet through the perihelion ; the other elements of the orbit may be inferred in the following manner.

Let  $j$  be the longitude of the node which will be the ascending one, if the motion of the comet be direct, and  $\varphi$  the inclination of the orbit ; we shall obtain by a comparison of the first and last observation,

$$\tan. j = \frac{\tan. \varpi. \sin. \epsilon'' - \tan. \varpi''. \sin. \epsilon}{\tan. \varpi. \cos. \epsilon'' - \tan. \varpi''. \cos. \epsilon} ; *$$

$$\tan. \varphi = \frac{\tan. \varpi''}{\sin. (\epsilon'' - j)}.$$

As we can thus compare two by two, the three observations, it

through the perhelion to vary by a very small quantity ; 5thly, by making the same instant to vary by twice this quantity. Let  $m, m', m'', m''', m''''$ , be the values of  $U-V$ ;  $n, n', n'', n''', n''''$ , the values of  $U'-V'$ ; in order to determine in this case the value of  $n$ , and  $t$ , the two following equations should be formed

$$(4m' - 3m - m'')u + (m'' - 2m' + m)u^2 + (4m''' - 3m - m''')t \\ + (m'''' - 2m'''' - m)t^2 = 2m; (4n' - 3n - n'').u + n'' - 2n' + n).u^2 \\ + (4n''' - 3n - n''').t + (n'''' - 2n'''' + n).t^2 = 2n.$$

The values of  $u$  and of  $t$  which satisfy those equations, are much more precise than the preceding. Although this precision is for the most part unnecessary, it is however indispensably necessary to form these equations, when the terms depending on the second differences will be of the same order as those which depend on the first differences, as for instance, when the radius vector is very nearly at right angles to the visual ray from the earth to the comet ; in which case the angle  $SCT$  is very nearly equal to a right angle ; on the other hand, if  $SCT$  was  $= 45^\circ$ , the two values of  $SCT$  would be very nearly equal.

\* Let  $I$  be the inclination of the orbit to the plane of the ecliptic, and we shall have

$$\text{rad. sin. } (\varphi - j) = \cot. I. \tan. \varpi = \text{rad. sin. } (\epsilon'' - j) = \cot. I. \tan. \varpi''. \text{ therefore}$$

will be more exact to select those which give to the preceding fractions, the greatest numerators and the greatest denominators.

Tan.  $j$  may appertain to the two angles  $j$  and  $j + \pi$ ,  $j$  being the smallest of the positive angles to which its value belongs; in order to determine which of these two angles we ought to select, it may be observed that  $\varphi$  is positive and less than a right angle; and that thus  $\sin. (\epsilon'' - j)$  must have the same sign as  $\tan. \varpi''$ . This condition determines the angle  $j$ , and this angle will give the position of the ascending node, if the motion of the comet is direct; but if its motion be retrograde, we should add two right angles to the angle  $j$ , in order to have the position of this node.

The hypotenuse of the spherical triangle of which  $\epsilon'' - j$  and  $\varpi''$  are the sides, is the distance of the comet, from its ascending node in the third observation; and the difference between  $\tau''$  and this hypotenuse is the interval between the node and the perihelion, reckoned on the orbit.

If we wish to give to the cometary theory all the precision which the observations admit of, it ought to be established on a comparison of all the best observations, which can be effected in the following manner: denoting by one, two strokes, &c. the letters  $m, n, p$ , relative to the second observation, to the third, &c. compared all with the first observation, we shall form the following equations,

$$(m - n). u + (m - p). t = m;$$

$$(m' - n'). u + (m' - p'). t = m';$$

$$(m'' - n''). u + (m'' - p''). t = m'';$$

&c.

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$$\frac{\sin. \epsilon. \cos. j - \cos. \epsilon. \sin. j}{\tan. \varpi} = \frac{\sin. \epsilon''. \cos. j - \cos. \epsilon''. \sin. j}{\tan. \varpi''}; \therefore \text{dividing by } \cos. j.$$

we have  $\frac{\sin. \epsilon - \cos. \epsilon. \tan. j}{\tan. \varpi} = \frac{\sin. \epsilon'' - \cos. \epsilon''. \tan. j}{\tan. \varpi''}$ , hence we derive the expression for  $\tan. j$ , which is given in the text.

If then these equations be combined in the most advantageous manner, in order to determine  $u$  and  $t$ , we will have the corrections of the perihelion distance, and of the instant of the transit through the perihelion deduced from all the observations compared together. From these values, we can deduce the values of  $\epsilon$ ,  $\epsilon'$ ,  $\epsilon''$ , &c.  $\varpi$ ,  $\varpi'$ ,  $\varpi''$ , &c., and we shall have

$$\tan. j = \frac{\tan. \varpi. (\sin. \epsilon' + \sin. \epsilon'' + \text{&c.}) - \sin. \epsilon. (\tan. \varpi' + \tan. \varpi'' + \text{&c.})}{\tan. \varpi. (\cos. \epsilon' + \cos. \epsilon'' + \text{&c.}) - \cos. \epsilon. (\tan. \varpi' + \tan. \varpi'' + \text{&c.})};^*$$

$$\tan. \phi = \frac{\tan. \varpi' + \tan. \varpi'' + \text{&c.}}{\sin. (\epsilon' - j) + \sin. (\epsilon'' - j) + \text{&c.}}.$$

38. There is a case, of rare occurrence indeed, in which the orbit of a comet can be determined in a rigorous and simple manner; namely, when the comet has been observed in the two nodes. The right line which joins these two observed positions, passes then through the centre of the sun, and coincides with the line of the nodes. The length of this line can be determined by the time which intervenes between the two observations;  $T$  representing this time reduced to decimals of a day, and  $c$  denoting the right line in question, we shall have, by N°. 27,

$$c = \frac{1}{2} \sqrt{\frac{T^2}{(9^d,688724)^2}}.$$

Now let  $\epsilon$  be the heliocentric longitude of the comet, at the instant of the first observation; and  $r$  its radius vector,  $\rho$  its distance from the earth, and  $\alpha$  its geocentric longitude. Also let  $R$  be the radius of the orbit of the earth, and  $E$  the corresponding longitude of the sun at the same instant; we shall have

\* By composition of ratios we obtain these values of  $\tan. j$ ,  $\tan. \phi$ , which are more accurate than the preceding.

$$r \cdot \sin. \epsilon = \rho \cdot \sin. \alpha - R \cdot \sin. E;$$

$$r \cdot \cos. \epsilon = \rho \cdot \cos. \alpha - R \cdot \cos. E.$$

$\pi + \epsilon$  will be the heliocentric longitude of the comet, at the instant of the second observation ; and if we denote by one stroke the quantities  $r$ ,  $\alpha$ ,  $\rho$ ,  $R$  and  $E$ , relative to the same instant, we shall have

$$r' \cdot \sin. \epsilon = R' \cdot \sin. E' - \rho' \cdot \sin. \alpha';$$

$$r' \cdot \cos. \epsilon = R' \cdot \cos. E' - \rho' \cdot \cos. \alpha'.$$

These four equations give

$$\tan. \epsilon = \frac{\rho \cdot \sin. \alpha - R \cdot \sin. E}{\rho \cdot \cos. \alpha - R \cdot \cos. E} = \frac{\rho' \cdot \sin. \alpha' - R' \cdot \sin. E'}{\rho' \cdot \cos. \alpha' - R' \cdot \cos. E'},$$

hence we obtain

$$\rho' = \frac{RR' \cdot \sin. (E-E') - R \cdot \rho \cdot \sin. (\alpha-E')}{\rho \cdot \sin. (\alpha'-\alpha) - R \cdot \sin. (\alpha'-E')}.$$

We have afterwards

$$(r+r') \cdot \sin. \epsilon = \rho \cdot \sin. \alpha - \rho' \cdot \sin. \alpha' - R \cdot \sin. E + R' \cdot \sin. E'.$$

$$(r+r') \cdot \cos. \epsilon = \rho \cdot \cos. \alpha - \rho' \cdot \cos. \alpha' - R \cdot \cos. E + R' \cdot \cos. E'.$$

By squaring these two equations, and adding them together, we shall obtain, ( $c$  being substituted in place of  $r+r'$ )

$$\begin{aligned} c^2 &= R^2 - 2RR' \cdot \cos. (E-E) + R'^2 \\ &\quad + 2\rho(R' \cdot \cos. (\alpha-E') - R \cdot \cos. (\alpha-E)) \\ &\quad + 2\rho'(R \cdot \cos. (\alpha'-E) - R' \cdot \cos. (\alpha'-E')) \\ &\quad + \rho^2 - 2\rho\rho' \cdot \cos. (\alpha'-\alpha) + \rho'^2. \end{aligned}$$

If in this equation, we substitute instead of  $\rho'$  its preceding values given in terms of  $\rho$ , we shall have an equation in  $\rho$  of the fourth degree, which can be resolved by the known methods ; but it will be simpler to suppose  $\rho$  equal to some given value, to infer from it the value of  $\rho'$ , then to substitute these values in the preceding equation, and see

whether they satisfy it. A few trials will serve to determine with accuracy,  $\epsilon$  and  $\varphi$ .

By means of these quantities we can obtain  $\epsilon$ ,  $r$  and  $r'$ . And  $v$  representing the angle which the radius  $r$  makes with the perihelion distance denoted by  $D$ ;  $\pi - v$  will be the angle formed by this same distance, and by the radius  $r'$ , thus we will obtain by No. 23,

$$r = \frac{D}{\cos. {}^{\circ} \frac{1}{2} v}; \quad r' = \frac{D}{\sin. {}^{\circ} \frac{1}{2} v};$$

consequently\*

$$\tan. {}^{\circ} \frac{1}{2} v = \frac{r}{r'}; \quad D = \frac{rr'}{r+r'}.$$

Therefore we shall have  $v$  the anomaly of the comet at the instant of the first observation, and its perihelion distance  $D$ , hence it is easy to infer the position of the perihelion, and the instant of the passage of the comet through this point. Thus, of the five elements of the orbit of the comet, four are known, namely, the perihelion distance, the position of the perihelion, the instant of the transit of the comet through this point, and the position of the node. It only remains to find out the inclination of the orbit; but for this purpose it will be necessary to recur to a third observation, which will also be useful in indicating amongst the different real and positive roots of the equation in  $\varphi$ , that of which we ought to make use.

38. The hypothesis of the parabolick motion of the comets, is not rigorously true, it is even very improbable, considering the infinite number of cases which give an elliptic or a hyperbolic motion, relatively to those which determine a parabolic motion. Besides, a comet which moves

\* Dividing  $r$  and its value by  $r'$  and its value respectively, we have  $\frac{r}{r'} = \frac{\sin. {}^{\circ} \frac{1}{2} v}{\cos. {}^{\circ} \frac{1}{2} v}$ .

$\frac{D}{r}$ , and also we have  $\frac{1}{r} + \frac{1}{r'} = \frac{r'+r}{rr} = \frac{1}{D}$ .

in either a parabolic or an hyperbolick orbit, would be only visible once ; therefore we may with great appearance of probability suppose, that the comets which describe these curves, if any such ever existed, have long since disappeared, so that at the present day, we only observe those, which moving in orbits returning into themselves, are perpetually brought back, after greater or less intervals, into the regions of space, near to the sun. We can by the following method, determine nearly within an interval of some years, the duration of their revolutions, when we shall have made a great number of very accurate observations before and after the passage through the perihelion.

For this purpose, let us suppose that we had four or a greater number of accurate observations, which may embrace all the visible part of the orbit, and that we have determined by the preceding method, the parabola, which satisfies very nearly these observations. Let  $v, v', v'', v''', \&c.$ , be the corresponding anomalies,  $r, r', r'', r''', \&c.$ , the corresponding radii vectores. Let also

$$v - v = U; v' - v = U'; v'' - v = U''; \&c.;$$

this being agreed upon, we compute by the preceding method, with the parabola already found, the values of  $U, U', U'', \&c., V, V', V''', \&c.$ ; let

$$m = U - V; m' = U' - V'; m'' = U'' - V''; m''' = U''' - V'''; \&c.$$

Afterwards, suppose the perihelion distance in this parabola, to vary by a very small quantity ; and let in this hypothesis,

$$n = U - V; n' = U' - V'; n'' = U'' - V''; n''' = U''' - V'''; \&c.$$

We then make a third hypothesis, in which the same distance of the perihelion being preserved, as in the first, the instant of the passage through the perihelion is varied by a very small quantity ; let then

$$p = U - V; p' = U' - V'; p'' = U'' - V''; p''' = U''' - V'''; \&c.$$

Finally, we will compute with the perihelion distance, and the instant

of the passage of the comet through the perihelion of the first hypothesis, the angle  $v$ , and the radius vector  $r$ , on the hypothesis that the orbit is elliptic, and that the difference  $1 - e$  between its eccentricity, and unity, is equal to a very small quantity, for example, to a 50th part. In order to obtain the value of the angle  $v$  on this hypothesis, it will suffice, by No. 23, to add to the anomaly  $v$ , computed in the parabola of the first hypothesis, a small angle of which the sine is

$$\frac{1}{10} \cdot (1 - e) \cdot \tan. \frac{1}{2}v \cdot (4 - 3 \cdot \cos. \frac{3}{2}v - 6 \cdot \cos. \frac{1}{2}v).^*$$

By substituting then in the equation

$$r = \frac{D}{\cos. \frac{1}{2}v} \left\{ 1 - \frac{(1-e)}{2} \cdot \tan. \frac{1}{2}v \right\};$$

in place of  $v$ , this anomaly thus computed in the ellipse; we will obtain the corresponding radius vector. In a similar manner we can compute,  $v'$ ,  $r'$ ,  $v''$ ,  $r''$ ,  $v'''$ ,  $r'''$ , &c.; by means of which we can obtain the values of  $U$ ,  $U'$ ,  $U''$ ,  $U'''$ , &c., and by No. 37, those of  $V$ ,  $V'$ ,  $V''$ , &c. Let in this case

$$q = U - V; q' = U' - V'; q'' = U'' - V''; q''' = U''' - V''' ; \text{ &c.}$$

Lastly, let  $u$  denote the number by which we must multiply the supposed variation in the distance of the perihelion, in order to obtain the true distance; and  $t$  the number by which the supposed variation in the instant of the transit through the perihelion must be multiplied, in order to obtain the true instant; and  $s$  the number by which the

\* When the orbit is supposed elliptic, we must have at least four observations; and then if the arc observed be considerable, and particularly if it is greater than  $90^\circ$ , the ellipticity will be very sensible, and the periodic time may be determined with tolerable precision, if the four observations be made with all the precision of modern observations. If the square of  $\alpha$  be neglected, the expression for  $r$  will be  $\frac{D}{\cos. \frac{1}{2}v} \left( 1 - \frac{\alpha}{2-\alpha} \cdot \tan. \frac{1}{2}v \right)$  which becomes the expression in the text when  $1-e$  is substituted for  $\alpha$ .

supposed value of  $1-e$  must be multiplied, in order to obtain the accurate value, we will thus form the following equations,

$$\begin{aligned} (m-n). u + (m-p). t + (m-q). s &= m \\ (m'-n'). u + (m'-p'). t + (m'-q'). s &= m' \\ (m''-n''). u + (m''-p''). t + (m''-q''). s &= m'' \\ (m'''-n'''). u + (m'''-p'''). t + (m'''-q'''). s &= m''' ; \\ \text{&c.} \end{aligned}$$

The values of  $u$ ,  $t$ ,  $s$ , may be determined by means of these equations, from which we can infer the true distance of the perihelion, the true instant of the transit of the comet through the perihelion, and the true value of  $1-e$ . Let  $D$  be the perihelion distance, and  $a$  the semiaxis major of the orbit ; we shall have  $a = \frac{D}{1-e}$  ; the time of the comets sidereal revolution will be expressed by a number of sidereal years, equal to  $a^{\frac{2}{3}}$ , or to  $\left(\frac{D}{1-e}\right)^{\frac{2}{3}}$ , the mean distance of the sun from the earth being taken for unity. Afterwards by N°. 37, we shall get the inclination of the orbit, and the position of the node.

Whatever be the precision of the observations, they will always leave some uncertainty as to the duration of the comets revolution. The most exact method to determine it, consists in comparing the observations of a comet, in two consecutive revolutions ; but this means is not practicable, except when the lapse of time brings the comet back towards its perihelion.\*

## CHAPTER V.

*General methods for determining, by successive approximations, the motions of the heavenly bodies.*

40. In the first approximation of the motions of the heavenly bodies, we have only considered the principal forces which actuate them, and from thence the laws of the elliptic motion have been deduced. We will consider, in the following investigations, the forces which disturb this motion. In consequence of the action of these forces, it is only requisite to add small terms to the differential equations of the elliptic motion, of which we have previously determined the finite integrals: it is necessary now to determine, by successive approximations, the integrals of the same equations, increased by the terms which arise from the action of the disturbing forces. For this object, we here subjoin a general method, which is applicable whatever be the number and the degree of the differential equations, of which it is proposed to find the perpetually approaching integrals.

Let us suppose that we have between the  $n$  variables  $y, y', y'', \&c.$  and the variable  $t$ , of which the element  $dt$  may be considered as constant, the  $n$  differential equations

$$0 = \frac{dy}{dt^i} + P + \alpha Q;$$

$$0 = \frac{dy'}{dt^i} + P' + \alpha Q';$$

&c.

$P, Q, P', Q', \&c.$  being functions of  $t, y, y', \&c.$ ; and of their differences continued to the order  $i-1$  inclusively, and  $\alpha$  being a very

small constant coefficient, which, in the theory of the celestial motions, is of the order of the disturbing forces. Let us in the next place suppose that we have obtained the finite integrals of these equations, when  $Q, Q', \&c.$  vanish; by differencing each,  $i-1$  times in succession, they will constitute with their differentials, *in* equations by means of which we can determine by elimination, the arbitrary quantities  $c, c', c'', \&c.$  in functions of  $t, y, y', y'', \&c.$  and of their differentials to the order  $i-1.$  Therefore, if  $V, V', V'', \&c.$  represent these functions, we shall have\*

$$c = V; \quad c' = V'; \quad c'' = V''; \quad \&c.$$

These equations are the *in* integrals of the order  $i-1;$  which the differential equations ought to have, and which their finite integrals furnish by the elimination of the differences of these variables.

Now, by differentiating the preceding integrals of the order  $i-1,$  we shall have

$$0 = dV; \quad 0 = dV'; \quad 0 = dV''; \quad \&c.$$

but it is evident that these last equations being differentials of the order  $i,$  without involving arbitrary quantities; they can be no other than the sums of the equations

$$0 = \frac{d^i y}{dt^i} + P; \quad 0 = \frac{d^i y'}{dt^i} + P'; \quad \&c.$$

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\* In every differential equation of the order  $i,$  the number of *first* integrals is equal to  $i,$  these integrals are of the order  $i-1,$  and therefore they only contain the  $i-1$  differential coefficients  $\frac{dy}{dt}, \frac{d^2y}{dt^2}, \dots, \frac{d^{i-1}y}{dt^{i-1}},$  and if these could be eliminated we would have

the  $1^{th}$  integral, or the primitive equation, which corresponds to the proposed differential equation; consequently, if we have  $n$  differential equations, the number of first integrals, or of integrals of the order  $i-1,$  must be *in*, from which if the differential coefficients of the variables  $y, y', y'', \&c.$  could be eliminated we would obtain the  $n$  finite integrals of the proposed equations.

multiplied respectively by suitable factors, in order that these sums may be exact differences; therefore representing the factors which ought to multiply these equations respectively in order to form the equation  $dV = 0$ , by  $Fdt$ ,  $F'dt$ , &c., and in like manner, representing by  $Hdt$ ,  $H'dt$ , &c. the factors which ought respectively to multiply the same equations, in order to constitute the equation  $0 = dV'$ ; and so of the rest, we shall have

$$dV = F.dt. \left\{ \frac{d^i y}{dt^i} + P \right\} + F'.dt. \left\{ \frac{d^i y'}{dt^i} + P' \right\} + \text{&c.};$$

$$dV' = H.dt. \left\{ \frac{d^i y}{dt^i} + P \right\} + H'.dt. \left\{ \frac{d^i y'}{dt^i} + P' \right\} + \text{&c.};$$

$F$ ,  $F'$ , &c.  $H$ ,  $H'$ , &c. are functions of  $t$ ,  $y$ ,  $y'$ ,  $y''$ , &c. and of their differences to the order  $i-1$ : it is easy to determine them, when  $V$ ,  $V'$ , &c. are known; for  $F$  is evidently the coefficient of  $\frac{d^i y}{dt^i}$ , in the differential of  $V$ ;  $F'$  is the coefficient of  $\frac{d^i y'}{dt^i}$ , in the same differential, and so on of the rest. In like manner,  $H$ ,  $H'$ , &c. are the coefficients of  $\frac{d^i y}{dt^i}$ ,  $\frac{d^i y'}{dt^i}$ , &c. in the differential of  $V'$ ; consequently, as the functions of  $V$ ,  $V'$ , &c. are supposed to be known, by differencing them solely with respect to  $\frac{d^{i-1} y}{dt^{i-1}}$ ,  $\frac{d^{i-1} y'}{dt^{i-1}}$ , &c. we will obtain the factors by which the differential equations

$$0 = \frac{d^i y}{dt^i} + P; \quad 0 = \frac{d^i y'}{dt^i} + P'; \quad \text{&c.}$$

should be multiplied in order to obtain the exact differences; this being premised, let us resume the differential equations

$$0 = \frac{d^i y}{dt^i} + P + \alpha.Q; \quad 0 = \frac{d^i y'}{dt^i} + P' + \alpha.Q'; \quad \text{&c.}$$

The first being multiplied by  $Fdt$ , the second by  $F'dt$ , and so of the rest, and then added together, will give

$$0 = dV + \alpha \cdot dt \cdot (FQ + F'Q' + \&c.) ;$$

in like manner will have,

$$0 = dV' + \alpha \cdot dt \cdot (HQ + H'Q' + \&c.) ;$$

\&c.

hence we obtain by integrating,

$$\begin{aligned} c - \alpha \cdot \int dt \cdot (FQ + F'Q' + \&c.) &= V ; \\ c' - \alpha \cdot \int dt \cdot (HQ + H'Q' + \&c.) &= V' ; \\ \&c. ; \end{aligned}$$

we will have by this means *in* differential equations which will be of the same form as when  $Q$ ,  $Q'$ , &c. are equal to nothing, with this sole difference, that the arbitrary quantities,  $c$ ,  $c'$ ,  $c''$ , &c. must be changed into

$$c - \alpha \cdot \int dt \cdot (FQ + F'Q' + \&c.) ; \quad c' - \alpha \cdot \int dt \cdot (HQ + H'Q' + \&c.) ; \quad \&c.$$

Now if, on the hypothesis of  $Q$ ,  $Q'$ , &c. equal to zero, we eliminate from the *in* integrals of the order  $i-1$ , the differences of the variables  $y$ ,  $y'$ , &c. ; we shall have the  $n$  finite integrals of the proposed equations ; consequently, the integrals of the same equations, when  $Q$ ,  $Q'$ , &c. do not vanish, will be had, by changing in the first integrals  $c$ ,  $c'$ , &c. into

$$c - \alpha \cdot \int dt \cdot (FQ + F'Q' + \&c.) ; \quad c' - \alpha \cdot \int dt \cdot (HQ + H'Q' + \&c.)$$

41. If the differentials

$$dt \cdot (FQ + FQ' + \&c.), \quad dt \cdot (HQ + H'Q' + \&c.) \quad \&c.$$

were exact, we could obtain by the preceding method the finite integrals of the proposed differential equations ; but this does not obtain except in some particular cases, of which the most extensive and the most interesting, is that in which these equations are linear. Let us, there-

fore, suppose that  $P, P'$ , &c. are linear functions of  $y, y'$ , &c. and of their differences to the order  $i-1$ , without any term independent of these variables, and at first let us consider the case, in which  $Q, Q'$ , &c. vanish. The differential equations being linear, their successive integrals will be also linear, so that  $c = V, c' = V'$ , being the  $m$  integrals of the order  $i-1$ , of the differential linear equations

$$0 = \frac{dy}{dt^i} + P; \quad 0 = \frac{dy'}{dt^i} + P'; \quad \text{&c.}$$

$V, V'$ , &c. may be considered as linear functions of  $y, y'$ , &c. and of their differences, to the order  $i-1$ . In order to demonstrate this, let us suppose, that in the expressions of  $y, y'$ , &c. the constant arbitrary quantity  $c$  is equal to a determinate quantity, added to an indeterminate  $\delta c$ ; the constant quantity  $c'$ , is equal to a determinate quantity added to the indeterminate,  $\delta c'$ , &c.; these expressions being reduced into series, arranged with respect to the powers and products of  $\delta c, \delta c'$ , &c., we will have by the formula of No. 21,

$$\begin{aligned} y &= Y + \delta c \cdot \left( \frac{dY}{dc} \right) + \delta c' \cdot \left( \frac{dY}{dc'} \right) + \text{&c.} \\ &\quad + \frac{\delta c^2}{1.2} \cdot \left( \frac{d^2 Y}{dc^2} \right) + \text{&c.;} \end{aligned}$$

$$\begin{aligned} y' &= Y' + \delta c \cdot \left( \frac{dY'}{dc} \right) + \delta c' \cdot \left( \frac{dY'}{dc'} \right) + \text{&c.} \\ &\quad + \frac{\delta c^2}{1.2} \cdot \left( \frac{d^2 Y'}{dc^2} \right) + \text{&c.;} \end{aligned}$$

&c.

$Y, Y', \left( \frac{dY}{dc} \right)$ , &c. being functions of  $t$ , without arbitrary quantities. By substituting these values in the proposed differential equations, it is manifest that  $\delta c, \delta c'$ , &c. being indeterminate, the coefficients of the first powers of each of them, must vanish in those different equations; but these equations being linear, we shall have

evidently the terms affected with the first powers of  $\delta c$ ,  $\delta c'$ , &c., by substituting  $\left(\frac{dY}{dc}\right) \cdot \delta c + \left(\frac{dY'}{dc'}\right) \cdot \delta c' + \&c.$  in place of  $y$ , &c.  $\left(\frac{dY}{dc'}\right) \cdot \delta c + \left(\frac{dY'}{dc'}\right) \cdot \delta c' + \&c.$  in place of  $y'$ , &c. These expressions of  $y$ ,  $y'$ , &c. satisfy separately the proposed differential equations; and as they contain the *in* arbitrary quantities  $\delta c$ ,  $\delta c'$ , &c. they are their complete integrals. Therefore it follows that the arbitrary quantities exist in a linear form, in the expressions of  $y$ ,  $y'$ , &c. and consequently also, in their differentials; hence it is easy to infer that the variables  $y$ ,  $y'$ , &c. and their differences may be supposed to exist in a linear form, in the successive integrals of the proposed differential equations.

It follows from what has been stated that  $F$ ,  $F'$ , &c. being the coefficients of  $\frac{dy}{dt^i}$ ,  $\frac{dy'}{dt^i}$ , &c. in the differential of  $V$ ,  $H$ ,  $H'$  &c. being the coefficients of the same differences, in the differential of  $V'$ ; and so of the rest; these quantities are functions of the sole variable  $t$ . Therefore, if we suppose  $Q$ ,  $Q'$ , &c. to be functions of  $t$  only, the differences  $dt.(FQ+F'Q'+\&c.)$ ;  $dt.(HQ+H'Q'+\&c.)$ ; will be exact.

From the above results a simple means of obtaining the integrals of any number  $n$  of linear differential equations of the order  $i$ , and which involve any terms  $\alpha Q$ ,  $\alpha Q'$ , &c. which<sup>1</sup> are functions of the sole variable  $t$ ; when we know how to integrate the same equations, in the case in which these terms vanish; for then, if we difference their  $n$  finite integrals,  $i-1$  times in succession, we shall have *in* equations which will give by elimination, the values of the *in* arbitrary quantities  $c$ ,  $c'$ , &c., in functions of  $t$ ,  $y$ ,  $y'$ , &c., and of the differences of these variable quantities to the order  $i-1$ . We will thus form, the *in* equations,  $c = V$ ,  $c' = V'$ , &c.; this being premised,  $F$ ,  $F'$ , &c. will be the coefficients of  $\frac{d^{i-1}y}{dt^{i-1}}$ ,  $\frac{d^{i-1}y'}{dt^{i-1}}$ , &c., in  $V$ ;  $H$ ,  $H'$ , &c., will be the coefficients of the same differences in  $V'$ , and so of the rest; therefore,

we will obtain the finite integrals of the linear differential equations,

$$0 = \frac{d^i y}{dt^i} + P + \alpha \cdot Q; \quad 0 = \frac{d^i y'}{dt^i} + P' + \alpha \cdot Q; \quad \text{&c.,}$$

by changing in the finite integrals of these equations deprived of their last terms  $\alpha Q$ ,  $\alpha Q'$ , &c., the arbitrary quantities  $c$ ,  $c'$ , &c., into  $c - \alpha \int dt (FQ + F'Q' + \text{&c.})$ ,  $c' - \alpha \int dt (HQ + H'Q' + \text{&c.})$ ; &c.

Let us, for example, consider the linear differential equation

$$0 = \frac{d^2 y}{dt^2} + a^2 y + \alpha \cdot Q.$$

The finite integral of the equation  $0 = \frac{d^2 y}{dt^2} + a^2 y$  is

$$y = \frac{c}{a} \cdot \sin. at + \frac{c'}{a} \cdot \cos. at;$$

$c$  and  $c'$  being arbitrary quantities. By differentiating this integral we obtain

$$\frac{dy}{dt} = c \cdot \cos. at - c' \cdot \sin. at.$$

If this differential be combined with the integral itself, we can form two integrals of the first order,

$$c = ay \cdot \sin. at + \frac{dy}{dt} \cdot \cos. at;$$

$$c' = ay \cdot \cos. at - \frac{dy}{dt} \cdot \sin. at;$$

thus we shall have in this case,

$$F = \cos. at; \quad H = -\sin. at;$$

therefore, the complete integral of the proposed will be

$$y = \frac{c}{a} \cdot \sin. at + \frac{c'}{a} \cdot \cos. at - \frac{\alpha \cdot \sin. at}{a} \cdot \int Q. dt. \cos. at + \\ \frac{\alpha \cdot \cos. at}{a} \cdot \int Q. dt. \sin. at.$$

It is easy to perceive that if  $Q$  is composed of terms of the form  $K \cdot \frac{\sin. (mt+\epsilon)}{\cos. (mt+\epsilon)}$ , each of these terms will produce in the value of  $y$ , the corresponding term

$$\frac{\alpha K}{m^2 - a^2} \cdot \frac{\sin. (mt+\epsilon)^*}{\cos. (mt+\epsilon)^*}$$

\* If  $Q$  be of the form  $\sin. (mt+\epsilon)$  then we shall have  $-\int Q. dt. \cos. at = -\int \sin. (mt+\epsilon) dt. \cos. at$ , which by partial integration becomes

$$\begin{aligned} & \frac{1}{a} \cdot \sin. (mt+\epsilon) \cdot \sin. at - \frac{m}{a} \cdot \int \cos. (mt+\epsilon) \cdot \sin. at. dt. (= - \frac{m}{a^2} \cdot \cos. (mt+\epsilon) \cdot \cos. at \\ & - \frac{m^2}{a^2} \cdot \int \sin. (mt+\epsilon) \cdot \cos. at. dt. (= \frac{m^2}{a^3} \cdot \sin. (mt+\epsilon) \cdot \sin. at - \frac{m^3}{a^3} \cdot \int \cos. (mt+\epsilon) \sin. at. dt.) \\ & (= - \frac{m^3}{a^4} \cdot \cos. (mt+\epsilon) \cdot \cos. at - \frac{m^4}{a^4} \cdot \int \sin. (mt+\epsilon) \cdot \cos. at. dt.) = \\ & \quad \frac{m^4}{a^5} \cdot \sin. (mt+\epsilon) \cdot \sin. at - \frac{m^5}{a^5} \cdot \int \cos. (mt+\epsilon) \cdot \sin. at. dt; \end{aligned}$$

now if the factors of  $\sin. at$ , and of  $\cos. at$ , be collected respectively, we shall obtain

$$\begin{aligned} & - \frac{\alpha \cdot \sin. at}{a} \cdot \int \sin. (mt+\epsilon) \cdot \cos. at. dt = \sin. (mt+\epsilon) \cdot \sin. ^2 at. \alpha. (a^{-2} + a^{-4}. m^2 + a^{-6}. m^4 + \&) \\ & - \cos. (mt+\epsilon) \cdot \sin. at. \cos. at. \alpha. (a^{-3}. m + a^{-5}. m^3 + a^{-7}. m^5 + \&c.) \end{aligned}$$

and if the term  $\int \sin. (mt+\epsilon) \cdot \sin. at. dt$ , be expanded into a series by a similar process we shall have

$$\begin{aligned} & \frac{\alpha \cdot \cos. at}{a} \cdot \int \sin. (mt+\epsilon) \cdot \sin. at. dt = \sin. (mt+\epsilon) \cdot \cos. ^2 at. \alpha. \\ & (a^{-2} + a^{-4}. m^2 + a^{-6}. m^4 + \&c.) + \cos. (mt+\epsilon) \cdot \sin. at. \cos. at. a. (a^{-3}. m + a^{-5}. m^3 + a^{-7}. m^5 + \&c.) \end{aligned}$$

If  $m$  is equal to  $a$ , the term  $K \frac{\sin.}{\cos.} (mt + \epsilon)$  will produce in  $y$ , 1st, the term  $-\frac{\alpha K}{4a^2} \frac{\sin.}{\cos.} (at + \epsilon)$ , which being comprised in the two terms  $\frac{c}{a} \cdot \sin. at + \frac{c'}{a} \cdot \cos. at$ , may be neglected; 2dly, the term  $\pm \frac{\alpha Kt}{2a} \frac{\cos.}{\sin.} (at + \epsilon)$ , the sign + obtaining, if the term of the expression of  $Q$  is a sine, and the sign — having place,\* if this term is a cosine. It appears from what has been stated above, how the arc  $t$  is produced without the signs of sine or cosine, in the values of  $y, y'$ , &c. by the

$\therefore$  adding these two expressions, and observing that  $a^{-2} + a^{-4} \cdot m^2 + a^{-6} \cdot m^4 + \&c. = \frac{1}{a^2 - m^2}$ , we shall arrive at the expression given in the text.

\* The parts under the sign of integration in this case are respectively  $f. \sin. (at + \epsilon)$ .  $\cos. at. dt$ ,  $f. \sin. (at + \epsilon) \cdot \sin. at. dt = f. \sin. at. \cos. at. \cos. \epsilon. dt + f. \cos. ^2 at. dt. \sin. \epsilon$ ,  $f. \sin. ^2 at. \cos. \epsilon. dt + f. \sin. at. \cos. at. \sin. \epsilon. dt$ , and these expressions are  $= \frac{1}{2} f. \sin. 2at. \cos. \epsilon. dt + \frac{1}{2} f. \cos. 2at. \sin. \epsilon. dt + \frac{1}{2} f. \sin. \epsilon. dt$ , and  $- \frac{1}{2} f. \cos. 2at. \cos. \epsilon. dt + \frac{1}{2} f. \cos. \epsilon. dt$ , and by integrating these expressions become  $\frac{-1}{4a} \cdot \cos. 2at. \cos. \epsilon$   
 $+ \frac{1}{4a} \cdot \sin. 2at. \sin. \epsilon + \frac{1}{2} \cdot \sin. \epsilon. t - \frac{1}{4a} \cdot \sin. 2at. \cos. \epsilon + \frac{1}{2} \cos. \epsilon. t - \frac{1}{4a} \cdot \cos. 2at. \sin. \epsilon$ , and if the three first terms be multiplied by  $-\frac{\alpha. \sin. at}{a}$ , and the three last by  $\frac{\alpha. \cos. at}{a}$  they become respectively

$$+ \frac{\alpha}{4a^2} \cdot \sin. at. \cos. 2at. \cos. \epsilon - \frac{\alpha}{4a^2} \cdot \sin. 2at. \sin. at. \sin. \epsilon - \frac{\alpha}{2a} \cdot \sin. at. \sin. \epsilon. t - \frac{\alpha}{4a^2} \cdot \sin. 2at. \cos. at. \cos. \epsilon + \frac{\alpha}{2a} \cdot \cos. at. \cos. \epsilon. t - \frac{\alpha}{4a^2} \cdot \cos. at. \cos. 2at. \sin. \epsilon = \\ + \frac{\alpha}{4a^2} \cdot \sin. at. (\cos. 2at. \cos. \epsilon - \sin. 2at. \sin. \epsilon) - \frac{\alpha}{4a^2} \cdot \cos. at. (\sin. 2at. \cos. \epsilon + \cos. 2at. \sin. \epsilon) + \frac{\alpha}{2a} \cdot (\cos. at. \cos. \epsilon - \sin. at. \sin. \epsilon) = \frac{\alpha}{4a^2} \cdot \sin. at. \cos. (2at + \epsilon) - \cos. at. \sin. (2at + \epsilon)) = \frac{-\alpha}{4a^2} \cdot \sin. (at + \epsilon) + \frac{\alpha t}{2a} \cdot \cos. (at + \epsilon).$$

successive integrations, although the differential equations do not contain it under this form. It is evident that this will be the case as often as the functions  $FQ$ ,  $F'Q'$ , &c.  $HQ$ ,  $H'Q'$ , &c. contain constant terms.

42. If the differences  $dt.(FQ + \&c.)$ ,  $dt.(HQ + \&c.)$ , are not exact, the preceding analysis will not give their rigorous integrals; but it suggests a simple means of obtaining integrals more and more approaching, when  $\alpha$  is very small, and when the values of  $y$ ,  $y'$ , &c. on the hypothesis of  $\alpha$  being equal to cypher, are known. By differentiating these values,  $i-1$  times in succession, we will obtain the following differential equations of the order  $i-1$ ,

$$c = V; c' = V'; \&c.$$

The coefficients of  $\frac{d^i y}{dt^i}$ ,  $\frac{d^i y'}{dt^i}$ , in the differentials of  $V$ ,  $V'$ , &c. being the values of  $F$ ,  $F'$ , &c.  $H$ ,  $H'$ , &c.; we will substitute them in the differential functions

$$dt.(FQ + F'Q' + \&c.); dt.(HQ + H'Q' + \&c.)*$$

Afterwards, we must substitute, in place of  $y$ ,  $y'$ , &c., their first approximate values; which will give their differences in functions of  $t$ , and of the arbitrary quantities  $c$ ,  $c'$ , &c. Let  $T.dt$ ,  $T'.dt$ , &c., be

### I I 2

\* Let  $y = \phi(t, c, c', c'', \&c.)$  be the value of  $y$ , when  $\alpha=0$ , which being substituted in place of  $y$ , in the function  $dt.(FQ+F'Q')$ ,  $dt.(HQ+H'Q')+\&c.$  these functions will depend on  $t$ , and  $c$ ,  $c'$ ,  $c''$ , &c.  $\because y = \phi(t, c-\alpha \int T.dt, c'-\alpha \int T'.dt, \&c.)$ , and if this value of  $y$  be also substituted in  $dt.(FQ+F'Q'+\&c.)$ ,  $dt.(HQ+H'Q'+\&c.)$ , they will become  $= T.dt$ ,  $T'.dt$ , &c.; hence  $y = \phi(t, c-\alpha \int T.dt, c'-\alpha \int T'.dt, \&c.)$

The successive powers of  $\alpha$  must necessarily occur in these approaching values of  $y$ . This method corresponds to the method of continued substitutions adopted by Newton.

these functions. If in the first approximate values of  $y$ ,  $y'$ , &c., we change the arbitrary quantities  $c$ ,  $c'$ , &c. respectively into  $c - \alpha \int T dt$ ;  $c' - \alpha \int T' dt$ , &c. we will have the second approximate values of those variables.

These second values being substituted again, in the differential functions

$$dt.(FQ + \text{&c.}) ; dt.(HQ + \text{&c.}) ; \text{&c.}$$

it is manifest that these functions are then what  $Fdt$ ,  $F'dt$ , become, when the arbitrary quantities  $c$ ,  $c'$ ,  $c''$ , &c. are changed into  $c - \alpha \int T dt$ ;  $c' - \alpha \int T' dt$ , &c. Therefore, let  $F$ ,  $F'$ , &c. be what  $T$ ,  $T'$ . &c. become in consequence of these changes; we shall have the third approximate values of  $y$ ,  $y'$ , &c.; by changing in the first,  $c$ ,  $c'$ , &c. respectively into  $c - \alpha \int T dt$ ,  $c' - \alpha \int T' dt$ ; &c.

In like manner,  $T_{\prime\prime}$ ,  $T'_{\prime\prime}$ , &c. representing what  $T$ ,  $T'$ , &c. become when  $c$ ,  $c'$ , &c. are changed into  $c - \alpha \int T dt$ ,  $c' - \alpha \int T' dt$ , &c.: we shall have the fourth approximate values of  $y$ ,  $y'$ , &c. by changing in the first approaching values of these variables,  $c$ ,  $c'$ , into  $c - \alpha \int T_{\prime\prime} dt$ ,  $c' - \alpha \int T'_{\prime\prime} dt$ , &c.; and so on of the rest.

We shall see in the sequel, that the determination of the celestial motions depends almost always on differential equations of the form

$$0 = \frac{d^2y}{dt^2} + a^2y + \alpha Q, *$$

$Q$  being an entire and rational function of  $y$ , and of the sines and cosines of angles increasing proportionably to the time represented by  $t$ . The following is the easiest means of integrating this equa-

\* Let  $Q = y \cdot \cos. 2t$ , and we have  $0 = \frac{d^2y}{dt^2} + a^2y + \alpha y \cdot \cos. 2t$ ; let  $a=0$ , and we shall have  $0 = \frac{d^2y}{dt^2} + a^2y$ ; of which the integral is  $\frac{c}{a} \cdot \sin. at + \frac{c'}{a} \cdot \cos. at$ , which value

tion. We suppose, in the first place,  $\alpha$  equal to nothing, and by the preceding number we will obtain a first value of  $y$ .

This value being substituted in  $Q$ , it will by this means become an entire and rational function of the sines and cosines of angles proportional to the time  $t$ . Afterward by integrating the differential equation we will obtain a second value of  $y$ , approximate as far as quantities of the order  $\alpha$  inclusively.

This value being substituted in  $Q$ , will give, by integrating the differential equation, a third approximate value of  $y$ , and so on of the rest.

This manner of integrating by approximation, the differential equations of the celestial motions, although the simplest of all, is however liable to the inconvenience of giving in the expressions of the variables  $y, y', \&c.$ , the arcs of circles without the signs of the *sine* and *cosine*, even in the case in which these arcs do not exist in the accurate values of these variables; in fact, we may conceive, that if these values involve the sines and cosines of angles of the order  $\alpha t$ , these sines and cosines ought to be exhibited in the form of a series, in the approximate values which are found by the preceding method; because these last values are arranged according to the powers of  $\alpha$ . This expansion into a series of the sines and cosines of angles of the order  $\alpha t$ , ceases to be exact, when in the progress of time, the arc  $\alpha t$  becomes considerable; consequently the approximate values of  $y, y', \&c.$ , cannot be extended to an indefinite time. As it is\* of consequence to have values

being substituted for  $y$  in  $ay. \cos. 2t$ , the differential equation  $\frac{d^2y}{dt^2} + a^2y + ay. \cos. 2t$ , can be integrated by the method pointed out in the preceding page.

\* It would seem at first sight only necessary to substitute for the arc  $t$  and its powers their developements deduced from the series  $t = \sin. t + \frac{\sin. t^3}{1.2.3} + \frac{3. \sin. t^5}{2.3.4.5} + \&c.$  but it is to be considered, that when  $t$  exceeds a quadrant the series ceases to be exact, ∵ this series cannot be substituted for any arc *whatever*.

which embrace the past as well as future ages; the reversion of the arcs of a circle, which the approximate values contain, to the functions which would produce them by their expansion into a series, is a very delicate problem, and of great interest in analysis. The following is a very simple and general method of resolving it.

43. Let us consider the differential equation of the order  $i$ ,

$$0 = \frac{d^i y}{dt^i} + P + \alpha Q;$$

$\alpha$  being a very small quantity, and  $P$  and  $Q$  being algebraic functions of  $y$ ,  $\frac{dy}{dt}, \dots, \frac{d^{i-1}y}{dt^{i-1}}$ , and of the sines and cosines of angles increasing proportionably to  $t$ . Let us suppose that we have the complete integral of this differential equation, in the case of  $\alpha = 0$ , and that the value of  $y$ , determined by this integral, does not involve the arc  $t$ , without the signs *sine* and *cosine*; let us afterwards suppose, that this equation being integrated by the preceding method of approximation, when  $\alpha$  does not vanish, gives

$$y = X + t \cdot Y + t^2 \cdot Z + t^3 \cdot S + \&c.$$

$X, Y, Z, \&c.$ , being periodic functions of  $t$ , which involve the  $i$  arbitrary quantities  $c, c', c'', \&c.$ ; and the powers of  $t$ , in this expression of  $y$ , extending to infinity by the successive approximations. It is manifest that the coefficients of these powers will always decrease with greater rapidity, as  $\alpha$  is smaller. In the theory of the motions of the heavenly bodies,  $\alpha$  expresses the order of the disturbing forces relatively to the principal forces which actuate them.

If the preceding value of  $y$ , be substituted in the function  $\frac{d^i y}{dt^i} + P + \alpha Q$ ; it will assume the following form,  $k + k't + k''t^2 + \&c.$ ;  $k, k', k''$ , being periodic functions of  $t$ ; but by hypothesis the value of  $y$

satisfies the differential equation

$$0 = \frac{d^i y}{dt^i} + P + \alpha Q;$$

therefore we ought to have identically

$$0 = k + k't + k''t^2 + \&c.$$

If  $k, k', k'', \&c.$ , do not vanish, this equation will give by the reversion of series, the arc  $t$ , in a function of the sines and cosines of angles proportionable to  $t$ ; therefore  $\alpha$  being supposed to be indefinitely small, we would have  $t$  equal to a finite function of the sines and cosines of similar angles, which is impossible; consequently, the functions  $k, k', \&c.$ , are identically equal to cypher.

Now, if the arc  $t$  is only elevated to the first power under the sign *sine* and *cosine*, as is the case in the theory of the celestial motions,\* this arc will not be produced by the successive differences of  $y$ ; therefore by substituting the preceding value of  $y$ , in the function  $\frac{d^i y}{dt^i} + P + \alpha Q$  the function  $K + K't + \&c.$ , into which it is transformed, will not contain the arc  $t$ , without the sines *sin.*, and *cos.*, but as far as it is already contained in  $y$ ; thus, by changing in the expression of  $y$ , the arc  $t$ , without the periodic signs; into  $t - \theta$ ,  $\theta$  being a constant quantity, the function  $k + k't + \&c.$ , will be changed into  $k + k'.(t - \theta) + \&c.$ ; and because this last function is identically equal to nothing, in consequence of the identical equations  $k = 0, k' = 0, \&c.$ ; it follows that the expression†

$$y = X + (t - \theta). Y + (t - \theta)^2. Z + \&c.,$$

\* If a term of the form  $\sin.(at^n)$  occurred in the value of  $y$ , then in the successive differences of  $y$ , powers of the arc  $t$  will be produced.

† The values of  $Y, Z, \&c.$  in this second value of  $y$  are different from the quantities represented by  $Y, Z, \&c.$  in the first value of  $y$ .

satisfies also the differential equation

$$0 = \frac{d^i y}{dt^i} + P + \alpha Q.$$

Although this second value of  $y$  seems to involve  $i + 1$  arbitrary quantities, namely the  $i$  arbitrary quantities  $c, c', c'', \dots$ , and the arbitrary  $\theta$ ; however, it can only contain the number  $i$  of arbitrary quantities which are really distinct. It is therefore necessary, that by suitable transformation in the constant quantities  $c, c', c'', \dots$ , the arbitrary quantity  $\theta$  should disappear from this second expression of  $y$ , and that, consequently, it should coincide with the first. This consideration furnishes us with means of making the arcs of circle which exist without the periodic signs to disappear.

The second expression of  $y$ , may be made to assume the following form :

$$y = X + (t - \theta) \cdot R.$$

As we have supposed that  $\theta$  disappears from  $y$ , we will have  $\left(\frac{dy}{d\theta}\right) = 0$ , and consequently

$$R = \frac{dX}{d\theta} + (t - \theta) \cdot \left(\frac{dR}{d\theta}\right).$$

This equation being differenced successively, will give

If  $\varphi(t, \theta, t - \theta)$  be expanded by the formula of No. 21, it will become  $\varphi(t, \theta) + \frac{d\varphi(t, \theta)}{d\theta} \cdot (t - \theta) + \frac{d^2\varphi(t, \theta)}{d\theta^2} \cdot (t - \theta)^2 + \dots$  &c. = (as  $\varphi(t, \theta) = X$ ) the expression in the text.

$$\left(\frac{dy}{d\theta}\right) = \left(\frac{dX}{d\theta}\right) + (t - \theta) \left(\frac{dR}{d\theta}\right) - R.$$

See an example of this method in Chapter 7, Article 53.

$$2. \left( \frac{dR}{d\theta} \right) = \left( \frac{d^2X}{d\theta^2} \right) + (t-\theta) \cdot \left( \frac{d^2R}{d\theta^2} \right);$$

$$3. \left( \frac{d^2R}{d\theta^2} \right) = \left( \frac{d^3X}{d\theta^3} \right) + (t-\theta) \cdot \left( \frac{d^3R}{d\theta^3} \right);$$

&c.

hence it is easy to infer by eliminating  $R$ , and its differentials, from the preceding expression of  $y$ , that

$$y = X + \frac{(t-\theta)}{1} \cdot \left( \frac{dX}{d\theta} \right) + \frac{(t-\theta)^2}{1.2} \cdot \left( \frac{d^2X}{d\theta^2} \right) + \frac{(t-\theta)^3}{1.2.3} \cdot \left( \frac{d^3X}{d\theta^3} \right) + \text{&c.}$$

$X$  is a function of  $t$ , and of the constant quantities  $c, c', c'', \text{ &c.}$ ; and as these constants are functions of  $\theta$ ,  $X$  is a function of  $t$  and of  $\theta$ , which we can represent by  $\varphi.(t, \theta)$ . The expression of  $y$  is, by the formula (*i*) of N°. 21, the expansion of the function  $\varphi.(t, \theta + t - \theta)$ , according to the powers of  $t - \theta$ ; therefore we have  $y = \varphi.(t, t)$ ; it follows from this that the value of  $y$  will be had by changing  $\theta$  into  $t$ , in  $X$ . The proposed problem is by this means reduced to the determination of  $X$ , in a function of  $t$ , and of  $\theta$ , and consequently to the determination of  $c, c', c'', \text{ &c.}$ , in functions of  $\theta$ .

For this purpose, let the equation

$$y + X + (t-\theta) \cdot Y + (t-\theta)^2 \cdot Z + (t-\theta)^3 \cdot S + \text{&c.}$$

be resumed. The constant quantity  $\theta$  being supposed to disappear from this value of  $y$ , we have the identical equation

$$0 = \left( \frac{dX}{d\theta} \right) - Y + (t-\theta) \cdot \left\{ \left( \frac{dY}{d\theta} \right) - 2Z \right\} + (t-\theta)^2 \cdot \left\{ \left( \frac{dZ}{d\theta} \right) - 3S \right\} + \text{&c.} \quad (a)$$

By applying to this equation the same reasoning as in the case of the equation  $0 = k + k't + k''t^2 + \text{ &c.}$ , it will appear that the coefficients of the successive powers of  $(t-\theta)$ , must of themselves be equal to zero. The functions  $X, Y, Z, \text{ &c.}$ , do not involve  $\theta$ , except as far as it is

contained in  $c, c', \&c.$ ; so that in order to constitute the partial differences  $\left(\frac{dX}{d\theta}\right), \left(\frac{dY}{d\theta}\right), \left(\frac{dZ}{d\theta}\right), \&c.$ , it is sufficient merely to make  $c, c', \&c.$ , vary in these functions, which gives

$$\begin{aligned}\left(\frac{dX}{d\theta}\right) &= \left(\frac{dX}{dc}\right) \cdot \frac{dc}{d\theta} + \left(\frac{dX}{dc'}\right) \cdot \frac{dc'}{d\theta} + \left(\frac{dX}{dc''}\right) \cdot \frac{dc''}{d\theta} + \&c.; \\ \left(\frac{dY}{d\theta}\right) &= \left(\frac{dY}{dc}\right) \cdot \frac{dc}{d\theta} + \left(\frac{dY}{dc'}\right) \cdot \frac{dc'}{d\theta} + \left(\frac{dY}{dc''}\right) \cdot \frac{dc''}{d\theta} + \&c.; \\ &\&c.\end{aligned}$$

Now it may happen that some of the arbitrary quantities  $c, c', c'', \&c.$ , multiply the arc  $t$  in the periodic functions  $X, Y, Z, \&c.$ ; the differentiation of these functions relatively to  $\theta$ , or which comes to the same thing, relatively to these arbitrary quantities, will develope this arc, and make it issue from without the signs of the periodic functions; the differences  $\left(\frac{dX}{d\theta}\right), \left(\frac{dY}{d\theta}\right), \left(\frac{dZ}{d\theta}\right), \&c.$ , will then be of the following form :

$$\begin{aligned}\left(\frac{dX}{d\theta}\right) &= X' + t.X''; \\ \left(\frac{dY}{d\theta}\right) &= Y' + t.Y''; \\ \left(\frac{dZ}{d\theta}\right) &= Z' + t.Z''; \\ &\&c.,\end{aligned}$$

in which  $X', X'', Y', Y'', Z', Z'', \&c.$ , are periodic functions of  $t$ , and moreover involve the arbitrary quantities  $c, c', c'', \&c.$ , and their first differences divided by  $d\theta$ , which differences do not occur in these functions, except under a linear form; we shall therefore, have

$$\begin{aligned}\left(\frac{dX}{d\theta}\right) &= X' + \theta X'' + (t - \theta). X''; \\ \left(\frac{dY}{d\theta}\right) &= Y' + \theta Y'' + (t - \theta). Y'';\end{aligned}$$

$$\left( \frac{dZ}{d\theta} \right) = Z' + \theta Z'' + (t-\theta) \cdot Z'' ; \\ &c.$$

This value being substituted in the equation (*a*), will give

$$0 = X' + \theta X'' - Y \\ + (t-\theta) \cdot (Y' + \theta Y'' + X'' - 2Z) \\ + (t-\theta)^2 \cdot (Z' + \theta Z'' + Y'' - 3S) + &c.$$

hence we deduce, by putting the coefficients of the powers of  $t-\theta$ , separately equal to nothing,

$$0 = X' + \theta \cdot X'' - Y ; \\ 0 = Y' + \theta \cdot Y'' + X'' - 2Z ; \\ 0 = Z' + \theta \cdot Z'' + Y'' - 3S ; \\ &c.$$

The first of these equations, being differenced  $i-1$  times in succession, with respect to  $t$ , will give a corresponding number of equations between the arbitrary quantities  $c, c', c'', &c.$ , and their first differences divided by  $dt$ ; the resulting equations being afterwards integrated, with respect to  $\theta$ , will give these constant quantities in functions of  $\theta$ . The sole inspection of the first of the preceding equations will almost always suffice to determine the differential equations in  $c, c', c'', &c.$ , by comparing separately the coefficients of the sines and of the cosines which it contains; because it is manifest that the values of  $c, c', &c.$  being independent of  $t$ , the differential equations which determine them ought to be equally independent of this quantity. The simplicity which this consideration produces in the computation, is one of the principal advantages of this method. Most frequently these equations can only be integrated by successive approximation, which may introduce the arc  $\theta$ , without the periodic signs, in the values of  $c, c', &c.$ , even when this arc does not occur in the rigorous integrals; but we can make it to disappear by the method which we have laid down.

It may happen that the first of the preceding equations, and its  $i-1$  differentials in  $t$ , do not give a number  $i$  of distinct equations, between the quantities  $c, c', c'', \&c.$ , and their differences. In this case, we should recur to the second and subsequent equations.

When the values of  $c, c', c'', \&c.$ , shall have been determined in functions of  $\theta$ ; we can substitute them in  $X$ , and by changing afterwards  $\theta$  into  $t$ , we will have the value of  $y$ , in which no function of the arcs of a circle occur, which are not affected by periodic signs, when this is possible. If this value still preserves them, it will be a proof that they existed in the exact integrals.

44. Let us now consider any number  $n$  of differential equations

$$0 = \frac{d^i y}{dt^i} + P + \alpha.Q; \quad 0 = \frac{d^i y'}{dt^i} + P' + \alpha.Q'; \quad \&c.$$

$P, Q, P', Q', \&c.$ , being functions of  $y, y', \&c.$ , and of their differentials, continued to the order  $i-1$ , and of the sines and cosines of angles increasing proportionably to the variable  $t$ , of which the difference is supposed to be constant. Let us suppose that the approximate integrals of these equations are

$$\begin{aligned} y &= X + t.Y + t^2.Z + t^3.S + \&c.; \\ y' &= X' + t.Y' + t^2.Z' + t^3.S' + \&c.; \end{aligned}$$

$X, Y, Z, \&c.$ ,  $X', Y', Z', \&c.$ , being periodic functions of  $t$ , and containing the *in* arbitrary quantities  $c, c', c'', \&c.$ , we will have, as in the preceding number,

$$\begin{aligned} 0 &= X'' + \theta.X''' - Y; \\ 0 &= Y' + \theta.Y'' + X'' - 2Z; \\ 0 &= Z' + \theta.Z'' + Y'' - 3S; \\ &\&c. \end{aligned}$$

The value of  $y'$  will in like manner give equations of the following form :

$$\begin{aligned}0 &= X' + \theta. X'' + Y; \\0 &= Y' + \theta. Y'' + X'' - 2Z; \\&\text{&c.}\end{aligned}$$

The values of  $y'', y''', \dots$ , will furnish similar equations. By means of these different equations we can determine the values of  $c, c', c'', \dots$ , in functions of  $\theta$ , those equations being selected, which are the simplest and most approximative : by substituting these values in  $X, X', \dots$ , and afterwards changing  $\theta$  into  $t$ , we will have the values of  $y, y', \dots$ , not containing the arcs of a circle *without the periodic signs*, when this is possible.

45. Let us resume the method which has been explained in N°. 40. It follows from it, that if in place of supposing the parameters  $c, c', c'', \dots$ , constant, we make them to vary, so that we may have

$$\begin{aligned}dc &= -\alpha dt. (FQ + FQ' + \dots); \\dc' &= -\alpha dt. (HQ + HQ' + \dots), \\&\text{&c.}\end{aligned}$$

we will have always the *in* integrals of the order  $i-1$ , namely

$$c = V; c' = V'; c'' = V''; \dots$$

as in the case of  $\alpha$  equal to zero ; hence it follows, that not only the finite integrals, but also all the equations in which only differences of an order inferior to  $i$ , enter, preserve the same form in the case of  $\alpha$  equal to nothing, and of  $\alpha$  being any finite value whatever ; because these equations can result solely from a comparison of the preceding integrals of the order  $i-1$ . Consequently, we can equally, in these two cases, difference  $i-1$  times in succession the finite integrals, without making  $c, c', \dots$ , to vary ; and as we are at liberty to make all, vary at once, there results an equation of condition between the parameters  $c, c', \dots$ , and their differences.

In the two cases namely, of  $\alpha$  equal to nothing, and of  $\alpha$  being any quantity whatever, the values of  $y, y', y'', \dots$ , and of their differences to the

order  $i-1$  inclusively, are the same functions of  $t$ , and of the parameters  $c', c'', \&c.$ ; let, therefore,  $Y$  be any function of the variables,  $y, y', y'', \&c.$ , and of their differentials inferior to the order  $i-1$ , and let us name  $T$ , the function of  $t$ , into which it is changed, when we substitute in place of those variables and of their differences, their values in  $t$ . We can difference the equation  $Y=T$ , by considering the parameters  $c, c', c'', \&c.$ , as constant; we can even assume the partial difference of  $Y$ , relatively to one only, or to several of the variables  $y, y', \&c.$ , provided that we only make to vary that part of  $T$ , which varies with them. In all these differentiations the parameters  $c, c', c'', \&c.$ , may be always regarded as constant; because, by substituting for  $y, y', \&c.$ , and their differences, their values in  $t$ , we will have equations identically nothing, in the two cases of  $\alpha$  equal to nothing, of  $\alpha$  having any finite value whatever.

When the differential equations are of the order  $i-1$ , it is no longer permitted, in differentiating them, to treat the parameters  $c, c', c'', \&c.$ , as if they were constant. In order to difference those equations, let us consider the equation  $\phi=0$ ,  $\phi$  being a differential function of the order  $i-1$ , and which contains the parameters  $c, c', c'', \&c.$ : let  $\delta\phi$  be the difference of this function taken on the supposition that  $c, c', \&c.$ , as well as the differences  $dy^{i-1}, d^{i-1}y', \&c.$ , are constant. Let  $S$  be the coefficient of  $\frac{d^i y}{dt^{i-1}}$  in the entire difference of  $\phi$ : let  $S'$  be the coefficient of  $\frac{d^i y'}{dt^{i-1}}$ , in this same difference, and so of the rest. The equation  $\phi=0$ , being differenced, will give

$$0 = \delta\phi + \left( \frac{d\phi}{dc} \right) \cdot dc + \left( \frac{d\phi}{dc'} \right) \cdot dc' + \&c.$$

$$+ S \cdot \frac{d^i y}{dt^{i-1}} + \&c.;$$

by substituting in place of  $\frac{d^i y}{dt^{i-1}}$ , its value  $-dt(P+\alpha.Q)$ ; in place

of  $\frac{d^i y'}{dt^{i-1}}$ , its value  $-dt.(P + \alpha Q)$ , &c.; we shall have

$$0 = \delta\varphi + \left(\frac{d\varphi}{dc}\right) \cdot dc + \left(\frac{d\varphi}{dc'}\right) \cdot dc' + \text{&c.}$$

$$-dt.(SP + S'P' + \text{&c.}) - \alpha \cdot dt.(SQ + S'Q' + \text{&c.}); \quad (t)$$

The parameters  $c, c', c'', \text{ &c.}$ , are constant on the hypothesis that  $\alpha$  vanishes; thus we have

$$0 = \delta\varphi - dt.(SP + S'P' + \text{&c.})$$

If we substitute in this equation, in place of  $c, c', c'', \text{ &c.}$ , their values  $V, V', V'', \text{ &c.}$ , we will have a differential equation of the order  $i-1$ , without arbitrariness, which is impossible unless the terms of this equation are identically equal to cypher. Therefore the function

$$\delta\varphi - dt.(SP + S'P' + \text{&c.})$$

becomes identically equal to nothing, in consequence of the equations  $c = V, c' = V', \text{ &c.}$ ; and as the same equations also obtain when the parameters  $c, c', c'', \text{ &c.}$ , are variable, it is manifest, that in this case also, the preceding function is identically equal to nothing, the equation (t) will consequently become

$$0 = \left(\frac{d\varphi}{dc}\right) \cdot dc + \left(\frac{d\varphi}{dc'}\right) \cdot dc' + \text{&c.}$$

$$- \alpha \cdot dt.(SQ + S'Q' + \text{&c.}): \quad (x)$$

It appears from this, that in order to difference the equation  $\varphi = 0$ , it is sufficient to make the parameters  $c, c', c'', \text{ &c.}$ , and the differences  $d^{i-1}y, d^{i-1}y', \text{ &c.}$ , to vary in  $\varphi$ , and after the differentiations to substitute  $-\alpha Q, -\alpha Q', \text{ &c.}$ , in place of the quantities  $\frac{d^i y}{dt^i}, \frac{d^i y'}{dt^i}, \text{ &c.}$

Let  $\psi = 0$ , be a finite equation between  $y, y', \text{ &c.}$ , and the variable  $t$ ; if we denote by  $\delta\psi, \delta^2\psi, \text{ &c.}$ , the successive differences of  $\psi$ , taken

on the supposition that  $c, c', \&c.$ , are constant ; by what goes before we shall have, even in the case in which  $c, c', \&c.$ , are variable, the following equations :

$$\psi = 0; \delta\psi = 0; d^2\psi = 0. \dots \dots \dots d^{i-1}\psi = 0;$$

therefore, by changing successively in the equation ( $x$ ), the function  $\varphi$  into  $\psi, \delta\psi, d^2\psi, \&c.$ , we shall have

$$0 = \left( \frac{d\psi}{dc} \right) \cdot dc + \left( \frac{d\psi}{dc'} \right) \cdot dc' + \&c.;$$

$$0 = \left( \frac{d.\delta\psi}{dc} \right) \cdot dc + \left( \frac{d.\delta\psi}{dc'} \right) \cdot dc' + \&c.$$

... ... ... ... ... ... ... ... ...

$$0 = \left( \frac{d.\delta^{i-1}.\psi}{dc} \right) \cdot dc + \left( \frac{d.\delta^{i-1}.\psi}{dc'} \right) \cdot dc' + \&c.$$

$$- \alpha dt. \left\{ Q. \left( \frac{d\psi}{dy} \right) + Q'. \left( \frac{d\psi}{dy'} \right) + \&c. \right\}.$$

Thus the equations  $\psi = 0, \psi' = 0, \&c.$ , being supposed to be the  $n$  finite integrals of the differential equations,

$$0 = \frac{d^i y}{dt^i} + P; 0 = \frac{d^i y'}{dt^i} + P'; \&c.;$$

we will have the  $in$  equations by means of which the parameters  $c, c', c'', \&c.$ , may be determined without the necessity of forming for this purpose the equations  $c = V, c' = V', \&c.$ , but when the integrals will be under this last form, the determination of  $c, c', \&c.$ , will be more simple.

**45.** This method of making the parameters to vary is of the greatest use in analysis, and in its applications. In order to shew a new application of it, let us consider the differential equation

$$0 = \frac{d^i y}{dt^i} + P;$$

$P$  being a function of  $t$ ,  $y$ , of its differences to the order  $i-1$ , and of the quantities  $q$ ,  $q'$ , &c., which are functions of  $t$ . Let us suppose that we have the finite integral of this equation on the hypothesis that  $q$ ,  $q'$ , &c., are constant, and let  $\phi=0$  represent this integral, which will contain the  $i$  arbitrary quantities  $c$ ,  $c'$ , &c., let  $\delta\phi$ ,  $\delta^2\phi$ ,  $\delta^3\phi$ , &c., denote the successive differences of  $\phi$ , taken on the supposition that  $q$ ,  $q'$ , &c., and also the parameters  $c$ ,  $c'$ , &c., are constant. If all these quantities are supposed to vary, the difference of  $\phi$  will be

$$\delta\phi + \left\{ \frac{d\phi}{dc} \right\} \cdot dc + \left\{ \frac{d\phi}{dc'} \right\} \cdot dc' + \text{&c.} + \left\{ \frac{d\phi}{dq} \right\} \cdot dq + \left\{ \frac{d\phi}{dq'} \right\} \cdot dq' + \text{&c.}$$

therefore, by making

$$0 = \left\{ \frac{d\phi}{dc} \right\} \cdot dc + \left\{ \frac{d\phi}{dc'} \right\} \cdot dc' + \text{&c.} + \left\{ \frac{d\phi}{dq} \right\} \cdot dq + \left\{ \frac{d\phi}{dq'} \right\} \cdot dq' + \text{&c.};$$

$\delta\phi$  will be yet the first difference of  $\phi$  when  $c$ ,  $c'$  &c.,  $q$ ,  $q'$ , &c., are variable. If in like manner we make,

$$0 = \left\{ \frac{d.\delta\phi}{dc} \right\} \cdot dc + \left\{ \frac{d.\delta\phi}{dc'} \right\} \cdot dc' + \text{&c.} + \left\{ \frac{d.\delta\phi}{dq} \right\} \cdot dq + \left\{ \frac{d.\delta\phi}{dq'} \right\} \cdot dq' + \text{&c.};$$

...   ...   ...   ...   ...   ...   ...   ...   ...   ...   ...   ...

$$0 = \left\{ \frac{d.\delta^{i-1}\phi}{dc} \right\} \cdot dc + \left\{ \frac{d.\delta^{i-1}\phi}{dc'} \right\} \cdot dc' + \text{&c.}$$

$$+ \left\{ \frac{d.\delta^{i-1}\phi}{dq} \right\} dq + \left\{ \frac{d.\delta^{i-1}\phi}{dq'} \right\} dq' + \text{&c.};$$

$\delta^2\phi$ ,  $\delta^3\phi$ ..... $\delta^i\phi$ , will be also the second, third, ..... $i^{\text{th}}$  differences of  $\phi$ , when  $c$ ,  $c'$ , &c.,  $q$ ,  $q'$ , &c. are supposed to be variable.

Now, in the case of  $c$ ,  $c'$ , &c.,  $q$ ,  $q'$ , &c., being constant, the differential equation

$$0 = \frac{dy}{dt} + P,$$

is the result of the elimination of the parameters,  $c, c', \&c.$ , by means of the equations

$$\phi = 0; \delta\phi = 0; \delta^2\phi = 0; \dots\dots\dots\delta^i\phi = 0;$$

thus, as these last equations obtain even when  $q, q', \&c.$ , are supposed to be variable, the equation  $\phi=0$ , satisfies also, in this case the proposed differential equation, provided that the parameters  $c, c', \&c.$ , are determined by means of the preceding  $i$  differential equations; and as their integration gives  $i$  constant arbitrary quantities, the function  $\phi$  will contain these arbitrary quantities, and the equation  $\phi = 0$ , will be the complete integral of the proposed.

This method of making the constant arbitrary quantities to vary can be employed with advantage, when the quantities  $q, q', \&c.$  vary with great slowness, because this consideration generally renders the integration by approximation, of the differential equations which determine the parameters  $c, c', c'', \&c.$ , much easier.

## CHAPTER VI.

*The second approximation of the celestial motions, or the theory of their perturbations.*

46. Let us now apply the preceding methods to the perturbations of the motions of the heavenly bodies, in order to infer from them the simplest expressions of their periodic and secular inequalities. For this purpose, let the differential equations (1), (2) and (3), of No. 9, be resumed, which determine the relative motion of  $m$  about  $M$ . Let

$$R = m' \cdot \frac{(xx' + yy' + zz')}{(x'^2 + y'^2 + z'^2)^{\frac{3}{2}}} + m'' \cdot \frac{(xx'' + yy'' + zz'')}{(x''^2 + y''^2 + z''^2)^{\frac{3}{2}}} + \text{&c.} - \frac{\lambda}{m};$$

$\lambda$  being by the number cited, equal to

$$\begin{aligned} & \frac{mm'}{((x'-x)^2 + (y'-y)^2 + (z'-z)^2)^{\frac{1}{2}}} + \frac{mm''}{((x''-x)^2 + (y''-y)^2 + (z''-z)^2)^{\frac{1}{2}}} \\ & + \frac{m'm''}{((x''-x')^2 + (y''-y')^2 + (z''-z')^2)^{\frac{1}{2}}} + \text{&c.}; \end{aligned}$$

Moreover, if we suppose  $M+m=\mu$ ; and  $r = \sqrt{x^2 + y^2 + z^2}$ ;  $r' = \sqrt{x'^2 + y'^2 + z'^2}$ ; &c., we will have

$$\left. \begin{aligned} 0 &= \frac{d^2x}{dt^2} + \frac{\mu \cdot x}{r^3} + \left\{ \frac{dR}{dx} \right\} \\ 0 &= \frac{d^2y}{dt^2} + \frac{\mu \cdot y}{r^3} + \left\{ \frac{dR}{dy} \right\} \\ 0 &= \left\{ \frac{d^2z}{dt^2} \right\} + \frac{\mu \cdot z}{r^3} + \left\{ \frac{dR}{dz} \right\} \end{aligned} \right\}; \quad (P)$$

The sum of these three equations multiplied respectively by  $dx$ ,  $dy$ ,  $dz$ , gives by their integration,

$$0 = \frac{dx^2 + dy^2 + dz^2}{dt^2} - \frac{2\mu}{r} + \frac{\mu}{a} + 2 \int dR; \quad (Q)$$

the differential  $dR$  being solely relative to the coordinates  $x$ ,  $y$ ,  $z$ , of the body  $m$ , and  $a$  being a constant arbitrary quantity, which when  $R$  vanishes, becomes by Nos. 18 and 19, the semiaxis major of the ellipse described by  $m$  about  $M$ .

The equations  $(P)$  multiplied respectively by  $x$ ,  $y$ ,  $z$ , and added to the integral  $(Q)$ , will give\*

$$\begin{aligned} 0 = & \frac{1}{2} \cdot \frac{d^2 r^2}{dt^2} - \frac{\mu}{r} + \frac{\mu}{a} + 2 \int dR \\ & + x \cdot \left\{ \frac{dR}{dx} \right\} + y \cdot \left\{ \frac{dR}{dy} \right\} + z \cdot \left\{ \frac{dR}{dz} \right\}; \quad (R) \end{aligned}$$

Now, we may conceive that the disturbing masses  $m'$ ,  $m''$ , &c., are multiplied by a coefficient  $\alpha$ ; and then the value of  $r$  will be a function of the time  $t$  and of  $\alpha$ . If this function be expanded with respect to the powers of  $\alpha$ ; and if  $\alpha$  be made  $= 1$ , after this expansion, it will be ranged according to the powers and products of the disturbing masses. Let the characteristic  $\delta$ , placed before a quantity, denote the differential of this quantity, taken with respect to  $\alpha$ , and divided by  $d\alpha$ . When the value of  $\delta r$  shall have been determined in a series arranged according to the powers of  $\alpha$ , we will have the radius  $r$ , by multiplying this series by  $d\alpha$ , then integrating it with respect to  $\alpha$ , and adding to this integral a function of  $t$ , independent of  $\alpha$ , which function is evidently the value of  $r$  when the perturbing forces vanish, and when consequently the curve described is a conic section. The determination of  $r$  is therefore reduced to the forming and integrating the differential equation which determines  $\delta r$ .

\*  $\frac{1}{2} \frac{d^2 r^2}{dt^2} = \frac{1}{2} \cdot \frac{d^2(x^2 + y^2 + z^2)}{dt^2} = x \cdot \frac{d^2x}{dt^2} + y \cdot \frac{d^2y}{dt^2} + z \cdot \frac{d^2z}{dt^2} + \frac{dx^2 + dy^2 + dz^2}{dt^2}.$

For this purpose, let us resume the differential equation (R), and let us make for greater simplicity,

$$x \cdot \left\{ \frac{dR}{dx} \right\} + y \cdot \left\{ \frac{dR}{dy} \right\} + z \cdot \left\{ \frac{dR}{dz} \right\} = r \cdot R';$$

differentiating it with respect to  $\alpha$ , we will have

$$0 = \frac{d^2 \cdot r \delta r}{dt^2} + \frac{\mu \cdot r \delta r}{r^3} + 2 \int \delta \cdot dR + \delta \cdot r R'; \quad (\text{S})$$

naming  $dv$  the indefinitely small arc intercepted between the two radii vectores  $r$  and  $r+dr$ ; the element of the curve described by  $m$  about  $M$ , will be  $\sqrt{dr^2+r^2dv^2}$ ; therefore we will have  $dx^2+dy^2+dz^2=dr^2+r^2dv^2$ ; and the equation (Q) will become

$$0 = \frac{r^2 dv^2 + dr^2}{dt^2} - \frac{2\mu}{r} + \frac{\mu}{a} + 2 \int dR.$$

eliminating  $\frac{\mu}{a}$ , from this equation, by means of the equation (R), we shall have\*

$$\frac{r^2 dv^2}{dt^2} = \frac{r \cdot d^2 r}{dt^2} + \frac{\mu}{r} + r \cdot R';$$

hence we deduce by differentiating with respect to  $\alpha$ ,

$$\frac{2r^2 \cdot dv \cdot d\delta v}{dt^2} = \frac{rd^2 \cdot \delta r - \delta r d^2 r}{dt^2} - \frac{3\mu \cdot r \delta r}{r^3} + r \cdot \delta R' - R' \cdot \delta r.$$

By substituting in this equation, in place of  $\frac{\mu \cdot r \delta r}{r^3}$  its value deduced

\* Substituting for  $\frac{\mu}{a}$  its value given by the equation (R), we obtain  $\frac{r^2 \cdot dv^2}{dt^2} + \frac{dr^2}{dt^2} - \frac{dr^2}{dt^2} - \frac{r \cdot d^2 r}{dt^2} + \frac{\mu}{r} - 2 \int dR - r R' - \frac{2\mu}{r} + 2 \int dR = 0$ ; which by obliterating quantities which destroy each other becomes the expression in the text; and its differential with respect to  $\alpha$  is obtained by dividing by  $r^2$ , and then differentiating.

from the equation (S), we will have

$$d.\delta v = \frac{d.(dr.\delta r + 2r.d\delta r) + dt^2.(3.s.\delta dR + 2r.\delta R + R'.\delta r)}{r^2.dv}; * \quad (T)$$

We can obtain, by means of the equations (S) and (T), the values of  $\delta r$  and of  $\delta v$  as accurately as we please; but it ought to be observed, that  $dv$  being the angle intercepted between the radii  $r$  and  $r+dr$ , the integral  $v$  of these angles does not exist in one and the same plane. In order to deduce from it value of the angle described about  $M$ , by the projection of the radius vector  $R$  on the fixed plane, let  $v$ , represent this last angle, and let  $s$  denote the tangent of the latitude of  $m$  above this plane;  $r.(1+s^2)^{-\frac{1}{2}}$  will be the expression of the projected radius vector, and the square of the element of the curve described by  $m$ , will be

$$\frac{r^2 dv^2}{1+s^2} + dr^2 + \frac{r^2 ds^2}{(1+s^2)^2}; †$$

but the square of this element is  $r^2 dv^2 + dr^2$ ; therefore we will obtain, by putting these two expressions equal to each other,

$$dv = \frac{dv \sqrt{(1+s^2) - \frac{ds^2}{dv^2}}}{\sqrt{1+s^2}}.$$

Thus  $dv$ , can be determined by means of  $dv$  when  $s$  will be known.

$$\begin{aligned} * \quad \frac{d.r.\delta r}{dt} &= \frac{dr.\delta r}{dt} + \frac{r.d\delta r}{dt}; \quad \frac{d^2.r.\delta r}{dt^2} = \frac{d^2.r.\delta r}{dt^2} + \frac{2dr.d\delta r}{dt^2} + \frac{r.d^2\delta r}{dt^2}, \quad \therefore - \frac{3\mu.r\delta r}{r^3} \\ &= + \frac{3d^2r.\delta r}{dt^2} + \frac{6dr.d\delta r}{dt^2} + \frac{3r.d^2\delta r}{dt^2} + 6s\delta dR + 3\delta.rR', \text{ and } \frac{2r^2.dv.d\delta v}{dt^2} = \\ &\frac{r.d^2\delta r - \delta r.d^2r}{dt^2} + \frac{3d^2r.\delta r}{dt^2} + \frac{6.d.r.d\delta r}{dt^2} + \frac{3r.d^2\delta r}{dt^2} + 6s\delta dR + 3r.\delta R' + 3R'.\delta r + r.\delta R' \\ &- R'.\delta r; \quad \therefore \frac{r^2.dv.d\delta v}{dt^2} = \frac{2r.d^2.\delta r}{dt^2} + \frac{d^2.r.\delta r}{dt^2} + \frac{3d.r.d\delta r}{dt^2} \\ &= \frac{d.(dr.\delta r + 2r.d\delta r) + dt^2.(3.s.\delta dR + 2r.\delta R' + R'.\delta r)}{dt^2}. \end{aligned}$$

†  $s$  being equal to the tangent of latitude,  $\frac{ds}{1+s^2}$  is equal to the differential of the latitude, and  $\frac{r^2}{1+s^2} \cdot \frac{ds^2}{1+s^2} + \frac{r^2}{1+s^2} \cdot dv^2 + dr^2 = r^2.dv^2 + dr^2$ .

If the fixed plane is assumed to be the plane of the orbit of  $m$  at a given epoch,  $s$  and  $\frac{ds}{dv}$  will be manifestly of the order of the perturbing forces; therefore, by neglecting the square and the products of these forces, we shall have  $v = v_0$ . In the theory of the planets and comets, these squares and products may be neglected, with the exception of certain terms of this order, which particular circumstances render sensible, and which can be easily determined by means of the equations (S) and (T). These last equations assume a simpler form when we only take into account the first power of the perturbing forces. In fact, we can then consider  $\delta v$  and  $\delta r$  as the parts of  $r$  and  $v$  arising from these forces; and  $\delta R$ ,  $\delta \cdot rR'$ , are what  $R$  and  $rR'$ , become, when we substitute in place of the coordinates of these bodies their values relative to the elliptic motion: \* they can be denoted by these last quantities, subject to this condition. Consequently, the equation S becomes,

$$0 = \frac{d^2 \cdot r \delta r}{dt^2} + \frac{\mu \cdot r \delta r}{r^3} + 2 \int dR + rR'.$$

The fixed plane of  $x$  and of  $y$  being supposed to be that of the orbit of  $m$ , at a given epoch,  $z$  will be of the order of perturbing forces, and because the square of these forces is neglected, the quantity  $z \cdot \left\{ \frac{dR}{dz} \right\}$  may likewise be neglected. Moreover, the radius  $r$  only

\* In the equation (R) when coordinates relative to the elliptic motion are substituted in place of the coordinates of the body, the three first terms vanish; consequently if in place of  $r$  in the equation (R) be substituted, a radius  $r'$  (which is relative to the elliptic motion) plus an indefinitely small quantity  $\delta r'$  which is the effect of the disturbing forces, then the equation (R) becomes  $= \frac{1}{2} \cdot \frac{d^2 \cdot (r' + \delta r')^2}{dt^2} - \frac{\mu}{r' + \delta r'} + \frac{\mu}{a} + 2 \int dR + rR' = \frac{1}{2} \cdot \frac{d^2 r'}{dt^2} - \frac{\mu}{r'} + \frac{\mu}{a} + \frac{d^2 \cdot r' \delta r'}{dt^2} + \frac{\mu \cdot \delta r'}{r'^2} + 2 \int dR + rR'$ , in this case the three first terms are  $=$  to cypher, and the three last are what (S) is reduced to; when  $dR$  and  $rR'$  are substituted for  $\delta dR$  and  $\delta \cdot rR'$ .

differs from its projected value by quantities of the order  $z^2$ . The angle which this radius makes with the axis of  $x$ , differs only from its projection by quantities of the same order; therefore this angle may be supposed equal to  $v$ , and we have, excepting quantities of the same order,

$$x = r \cdot \cos. v; \quad y = r \cdot \sin. v;$$

hence we deduce\*

$$x \cdot \left\{ \frac{dR}{dx} \right\} + y \cdot \left\{ \frac{dR}{dy} \right\} = r \cdot \left\{ \frac{dR}{dr} \right\};$$

and consequently,  $r.R' = r \cdot \left\{ \frac{dR}{dr} \right\}$ . It is easy to be assured by differentiation, that if we neglect the square of the perturbing force, the preceding differential equation will give, in consequence of the two first equations (P)

$$r.\dot{r} = \frac{x \int y dt \cdot \left\{ 2 \int dR + r \cdot \left\{ \frac{dR}{dr} \right\} \right\} - y \int x dt \cdot \left\{ 2 \int dR + r \cdot \left\{ \frac{dR}{dr} \right\} \right\} \dagger}{\left\{ \frac{xdy - ydx}{dt} \right\}}.$$

\*  $\frac{dR}{dr} = \frac{dR}{dx} \cdot \frac{dx}{dr} + \frac{dR}{dy} \cdot \frac{dy}{dr} = \cos. v \cdot \frac{dR}{dx} + \sin. v \cdot \frac{dR}{dy}$  multiplying both sides by  $r$ , and substituting  $x$  for  $r \cdot \cos. v$ , and  $y$  for  $r \sin. v$ , we shall have the value of  $rR'$ .

† Add to the first of the equations (P) multiplied into  $r.\dot{r}$ , the equation (S) multiplied into  $-x$ , and we shall have

$$0 = r.\dot{r} \cdot \left( \frac{d^2x}{dt^2} + \frac{\mu x}{r^3} + \left( \frac{dR}{dx} \right) \right) - \frac{x \cdot d^2 \cdot r \dot{r}}{dt^2} - \frac{\mu x r \dot{r}}{r^3} - 2x \int dR - x \cdot r R';$$

in like manner if the second of the equations (P) be multiplied by  $r.\dot{r}$ , and then added to the equation (S) multiplied into  $-y$  we shall have

$$0 = r.\dot{r} \cdot \left( \frac{d^2y}{dt^2} + \frac{\mu y}{r^3} + \left( \frac{dR}{dy} \right) \right) + y \cdot \frac{d^2 \cdot r \dot{r}}{dt^2} + \frac{\mu y r \dot{r}}{r^3} + 2y \int dR + y \cdot r R';$$

now if these equations be respectively integrated, and then the integral of the first multi-

In the second member of this equation the coordinates may refer to the elliptic motion, which gives  $\left\{ \frac{xdy-ydx}{dt} \right\}$  constant, and equal by N°. 19, to  $\sqrt{\mu.a(1-e^2)}$ ,  $ae$  being the excentricity of the orbit of  $m$ . If we substitute in the expression of  $r\partial r$ , in place of  $x$  and of  $y$ , their values  $r \cdot \cos v$ , and  $r \cdot \sin v$ , and instead of  $\frac{xdy-ydx}{dt}$ , the quantity  $\sqrt{\mu a(1-e^2)}$ ; and if finally we observe that by N°. 20, we have  $\mu = n^2 \cdot a^3$ ; we will obtain

$$\delta r = \frac{\begin{cases} a \cdot \cos v \int ndt \cdot r \cdot \sin v \cdot \left\{ 2 \int dR + r \cdot \left\{ \frac{dR}{dr} \right\} \right\} \\ -a \cdot \sin v \int ndt \cdot r \cdot \cos v \cdot \left\{ 2 \int dR + r \cdot \left\{ \frac{dR}{dr} \right\} \right\} \end{cases}}{\mu \cdot \sqrt{1-e^2}}; \quad (\text{X})$$

the equation (T) gives by integrating and neglecting the square of the perturbating forces;

plied into  $y$ , be added to the second multiplied into  $-x$ , we shall obtain obliterating the quantities which destroy each other

$$\begin{aligned} &r\partial r \cdot y \cdot \frac{dx}{dt} + y \int r\partial r \cdot \left( \frac{dR}{dx} \right) - \frac{xy \cdot dr \cdot \partial r}{dt} - 2y \int dt \cdot x \cdot dR - y \int x \cdot r \cdot R \\ &- r\partial r \cdot x \cdot \frac{dy}{dt} - x \int r \cdot \partial r \cdot \left( \frac{dR}{dx} \right) + \frac{xy \cdot dr \cdot \partial r}{dt} + 2x \int dt \cdot y \cdot dR + x \int y \cdot r \cdot R, \end{aligned}$$

$\therefore$  neglecting quantities of the same order as the square of the disturbing force, we have  $r\partial r \cdot \frac{ydx-xdy}{dt} = x \cdot \int ydt \cdot (2dR + rR') - y \cdot \int x \cdot dt \cdot (2dR + rR')$ , which becomes the

expression in the text, when  $r \cdot \left( \frac{dR}{dr} \right)$  is substituted for  $r \cdot R'$ ; and by substituting for  $x$  and  $y$  their respective values  $r \cdot \cos v$ ,  $r \sin v$ , this equation is divisible by  $r$ ; now  $\frac{xdy-ydx}{dt} =$

$$= \sqrt{\mu \cdot a \cdot (1-e^2)}; \quad \text{and } \sqrt{\mu} = n \cdot a^{\frac{3}{2}}: \text{consequently } \frac{an}{\mu \cdot \sqrt{1-e^2}} = \frac{1}{na^{\frac{1}{2}} \cdot \sqrt{1-e^2}} = \frac{1}{\sqrt{\mu a \cdot (1-e^2)}}.$$

$$\delta v = \frac{\frac{2r.d.\delta r + dr.\delta r}{a^2.ndt} + \frac{3a}{\mu} \cdot \iint n dt. dR + \frac{2a}{\mu} \cdot \int n. dt. r. \left\{ \frac{dR}{dr} \right\}}{\sqrt{1-e^2}}; \quad (Y)^*$$

By means of this equation the perturbations of the motions of  $m$  in longitude can be easily determined, when those of the radius vector shall have been determined.

It now remains to determine the perturbations of the motion in latitude. For this purpose, we shall resume the third of the equations (P), and by integrating it as we have integrated the equation (S), and making  $z = r\delta s$ , we shall have

$$\delta s = \frac{a. \cos. v. \int n dt. r. \sin. v. \left\{ \frac{dR}{dz} \right\} - a. \sin. v. \int n dt. r. \cos. v. \left\{ \frac{dR}{dz} \right\}^+}{\mu. \sqrt{1-e^2}}; \quad (Z)$$

$\delta s$  is the latitude of  $m$  above the plane of the primitive orbit: if we

$$* \delta v = \frac{dr.\delta r + 2rd.\delta r}{\sqrt{\mu.a.(1-e^2)}} + \iint dt^2. 3.(dR + 2r. \left( \frac{dR}{dr} \right)) =$$

the expression in the text;  $R'\delta r$  is omitted as being of the order of the squares of the disturbing forces.

† Multiplying the third of the equations (P) by  $x$ , and subtracting it from the first multiplied by  $z$ , and then integrating, we shall obtain neglecting quantities of the order of the square of the disturbing forces

$$z. \frac{dx}{dt} - x. \frac{dz}{dt} = - \int x. dt. \left( \frac{dR}{dz} \right),$$

in like manner subtracting the second of the equations (P) multiplied into  $z$  from the third multiplied into  $y$ , we shall obtain by integrating,

$$y. \frac{dz}{dt} - z. \frac{dy}{dt} = \int y. dt. \left( \frac{dR}{dz} \right),$$

and multiplying the first of these equations by  $y$ , and the second by  $x$ , we obtain by adding them together

would wish to refer the motion of  $m$ , on a plane a little inclined to this orbit; by naming  $s$  its latitude, when it is supposed to exist on this plane,  $s + \delta s$  will be very nearly the latitude of  $m$  above the proposed plane.

47. The formulæ (X), (Y) and (Z), have the advantage of exhibiting the perturbations under a finite form; which is very useful in the theory of the comets, in which these perturbations can only be determined by quadratures. But in consequence of the little excentricity and inclination of the respective orbits of the planets, we are permitted to expand their perturbations, in converging series of the sines and cosines of angles increasing proportionably to the time, and to form tables of them which may serve for an indefinite time. Then, instead of the preceding expressions of  $\delta r$  and  $\delta s$ , it is more convenient to make use of differential equations which determine these variables. By arranging these equations with respect to the powers and to the products of the excentricities and inclinations of the orbits, we can always reduce the determination of the values of  $\delta r$  and  $\delta s$ , to the integration of equations of the form

$$0 = \frac{d^2y}{dt^2} + n^2y + Q;$$

the integration of this species of differential equation has been given in N°. 42. But we can immediately give this very simple form, to the preceding differential equations, by the following method.

Resuming the equation (R) of the preceding number, and in order to abridge making,

$$Q = 2. \int dR + r. \left\{ \frac{dR}{dr} \right\};$$

M M 2

$$z. \left( y. \frac{dx}{dt} - x. \frac{dy}{dt} \right) = x \int y. dt. \left( \frac{dR}{dz} \right) - y \int x. dt. \left( \frac{dR}{dz} \right),$$

and by substituting for  $\frac{ydx - xdy}{dt}$  its value  $\sqrt{\mu.a.(1-e^2)}$ , and for  $x$  and  $y$  their values  $r.\cos.v$ ,  $r.\sin.v$ , we obtain the expression which is given in the text.

it thus becomes,

$$0 = \frac{1}{2} \cdot \frac{d^2 r^2}{dt^2} - \frac{\mu}{r} + \frac{\mu}{a} + Q ; \quad (R')$$

In the case of elliptic motion, in which  $Q = 0$ ,  $r^2$  is by the N°. 22, a function of  $e \cdot \cos.(nt + \epsilon - \varpi)$ ,  $ae$  being the excentricity of the orbit, and  $nt + \epsilon - \varpi$  being the mean anomaly of the planet  $m$ . Let  $e \cdot \cos.(nt + \epsilon - \varpi) = u$ ; and let us suppose that  $r^2 = \phi(u)$ ; we shall have

$$0 = \frac{d^2 u}{dt^2} + n^2 u . *$$

In the case of the disturbed motion we can also suppose  $r^2 = \phi(u)$ ; but  $u$  will be no longer equal to  $e \cdot \cos.(nt + \epsilon - \varpi)$ ; it will therefore be given by the preceding differential equation increased by a term depending on the disturbing forces. In order to determine this term, it ought to be observed, that if we make  $u = \psi(r^2)$ , we shall have

$$\frac{d^2 u}{dt^2} + n^2 u = \frac{d^2 r^2}{dt^2} \cdot \psi.(r^2) + \frac{4r^2 dr^2}{dt^2} \cdot \psi'.(r^2) + n^2 \cdot \psi.(r^2), \dagger$$

$\psi'(r^2)$  being the differential of  $\psi.(r^2)$  divided by  $d.(r^2)$  and  $\psi''(r^2)$  being the differential of  $\psi'(r^2)$ , divided by  $d.r^2$ . The equation (R') gives  $\frac{d^2 r^2}{dt^2}$  equal to a function of  $r$ , plus a function depending on the disturbing force. If we multiply this equation by  $2rdr$ , and then in-

\*  $\frac{du}{dt} = -e \cdot n \cdot \sin.(nt + \epsilon - \varpi)$ ;  $\frac{d^2 u}{dt^2} = -en^2 \cdot \cos.(nt + \epsilon - \varpi)$ , therefore

$$\frac{d^2 u}{dt^2} + n^2 u = -en^2 \cdot \cos.(nt + \epsilon - \varpi) + e \cdot n^2 \cdot \cos.(nt + \epsilon - \varpi) = 0;$$

†  $\frac{du}{dt} = \frac{dr^2}{dt} \cdot \psi.(r^2)$ ,  $\therefore \frac{d^2 u}{dt^2} = \frac{d^2 r^2}{dt^2} \cdot \psi.(r^2) + \left(\frac{dr^2}{dt}\right)^2 \cdot \psi''(r^2)$ , and by substituting for  $dr^2$  its value  $2rdr$ , we obtain the expression for  $\frac{d^2 u}{dt^2} + n^2 u$ , which is given in the text.

tegrate it; we shall have  $\frac{r^2 dr^2}{dt^2}$  equal to a function of  $r$ , plus a function depending on the disturbing force. By substituting these values of  $\frac{d^2 r}{dt^2}$  and of  $\frac{r^2 dr^2}{dt^2}$ , in the preceding expression of  $\frac{d^2 u}{dt^2} + n^2 u$ ; the function of  $r$  independent of the disturbing force will disappear of itself, because the terms are identically equal to nothing, when this force vanishes, therefore we shall obtain the value of  $\frac{d^2 u}{dt^2} + n^2 u$ , by substituting in its expression, in place of  $\frac{d^2 r}{dt^2}$  and of  $\frac{r^2 dr^2}{dt^2}$ , the parts of their expressions which depend on the disturbing force. But, if we only consider these parts, the equation ( $R'$ ) and its integral, give

$$\frac{d^2 r}{dt^2} = -2Q;$$

$$\frac{4r^2 \cdot dr^2}{dt^2} = -8 \cdot \int Q r dr;$$

therefore,

$$\frac{d^2 u}{dt^2} + n^2 u = -2Q \psi'(r^2) - 8 \cdot \psi''(r^2) \cdot \int Q r dr.$$

Now, from the equation  $u = \psi(r^2)$  we deduce  $du = 2rdr \cdot \psi'(r^2)$ ; the equation  $r^2 = \varphi(u)$ , gives  $2rdr = du \cdot \varphi'(u)$ , and consequently

$$\psi'(r^2) = \frac{1}{\varphi'(u)}.$$

By differentiating this last equation, and by substituting  $\varphi'(u)$  instead of  $\frac{2rdr}{du}$ , we shall have

$$\psi''(r^2) = \frac{-\phi''(u)}{\phi'(u)^3},$$

$\phi''(u)$  being equal to  $\frac{d\cdot\phi'(u)}{du}$ , in like manner as  $\phi'(u)$  is equal to  $\frac{d\cdot\phi(u)}{du}$ . This being premised, if we make

$$u = e \cdot \cos. (nt + \epsilon - \omega) + \delta u,$$

the differential equation in  $u$  will become

$$0 = \frac{d^2 \cdot \delta u}{dt^2} + n^2 \cdot \delta u - \frac{4 \cdot \phi''(u)}{\phi'(u)^3} \cdot \int Q \cdot du \cdot \phi'(u) + \frac{Q}{\phi'(u)}; \dagger$$

and if we neglect the square of the disturbing force,  $u$  may be supposed to be equal to  $e \cdot \cos. (nt + \epsilon - \omega)$ , in the terms depending on  $Q$ .

The value of  $\frac{r}{a}$  found in N°. 22, gives, by carrying the precision to quantities of the order  $e^2$  inclusively,

$$r = a \cdot (1 + e^2 - u \cdot (1 - \frac{5}{2}e^2) - u^2 - \frac{5}{2}u^3);$$

hence we deduce

$$r^2 = a^2 \cdot (1 + 2e^2 - 2u \cdot (1 - \frac{1}{2}e^2) - u^2 - u^3) = \phi(u). \ddagger$$

If this value of  $\phi(u)$  be substituted in the differential equation in  $\delta u$ ,

† Substituting in place of  $u$  its value, the part which involves the cosine will be equal to nothing, as is evident from the preceding page; the other part is what is given in the text.

‡  $\frac{r}{a} =$  (as powers of  $e$  higher than the third are rejected)  $1 + \frac{1}{2}e^2 - (e - \frac{5}{8}e^3) \cdot \cos. (nt + \epsilon - \omega) - \frac{1}{2}e^2 \cdot \cos. 2(nt + \epsilon - \omega) - \frac{5}{8}e^3 \cdot \cos. 3(nt + \epsilon - \omega); \cos. 2(nt + \epsilon - \omega) = \frac{2u^2}{e^2} - 1, \cos. 3(nt + \epsilon - \omega) = \cos. 2(nt + \epsilon - \omega) \cdot \cos. (nt + \epsilon - \omega) - \sin. 2(nt + \epsilon - \omega) \cdot \sin. (nt + \epsilon - \omega) = \frac{2u^3}{e^3} - \frac{u}{e} + \frac{2u^3}{e^3} - \frac{2u}{e}; = \frac{4u^3}{e^3} - \frac{3u}{e}; \therefore \frac{r}{a} = 1 + \frac{e^2}{2} - (e - \frac{5}{8}e^3) \frac{u}{e} - \frac{1}{2}e^2.$

and if we then restore in place of  $Q$ , its value  $2.\int dR + r.\left\{ \frac{dR}{dr} \right\}$ , and  $e. \cos. (nt + \varepsilon - \varpi)$ , instead of  $u$ , we shall have, as far as quantities of the order  $e^3$ ,

$$0 = \frac{d^2 \delta u}{dt^2} + n^2 \delta u$$

$$- \frac{1}{a^2} \cdot (1 + \frac{1}{4} e^2 - e. \cos. (nt + \varepsilon - \varpi) - \frac{1}{4} e^2. \cos. (2nt + 2\varepsilon - 2\varpi)).$$

$$\left\{ (2.\int dR + r.\left\{ \frac{dR}{dr} \right\}) \right\};^*$$

$$- \frac{2e}{a^2} \cdot \int ndt. [\sin. (nt + \varepsilon - \varpi). [1 + e. \cos. (nt + \varepsilon - \varpi)].$$

$$\left\{ 2.\int dR + r. \left\{ \frac{dR}{dr} \right\} \right\}].$$

When  $\delta u$  shall have been determined, by means of this differential equation;  $\delta r$  will be obtained by differentiating the expression of  $r$ , with respect to the characteristic  $\delta$ , which gives

$$\delta r = -a\delta u.(1 + \frac{3}{4}e^2 + 2e. \cos. (nt + \varepsilon - \varpi) + \frac{9}{4}e^2. \cos. (2nt + 2\varepsilon - 2\varpi)).$$

This value of  $\delta r$  will give the value of  $\delta v$  by means of the formula (Y) of the preceding number.

$$\left( \frac{2u^2}{e^2} - 1 \right) - \frac{3}{8}e^3 \cdot \left( \frac{4u^3}{e^3} - \frac{3u}{e} \right) = 1 + \frac{1}{2}e^2 + \frac{1}{2}e^2 - \left( 1 - \frac{12}{8}e^2 \right) \cdot u - u^2 - \frac{12}{8}u^3;$$

$$\frac{r^2}{a^2} = (1 + e^2 - (1 - \frac{3}{2}e^2)u)^2 + 2(1 + e^2 - u(1 - \frac{3}{2}e^2)) \cdot (u^2 + \frac{3}{2}u^3) + (u^2 + \frac{3}{2}u^3)^2, \text{ as } u \text{ involves}$$

$$e, \text{ powers of } u \text{ higher than the third may be neglected, } \therefore \frac{r^2}{a^2} = 1 + 2e^2 - 2(u(1 - \frac{3}{2}e^2))$$

$$- 2ue^2 + u^2 - 2u^3 - \frac{3}{2}2u^3.$$

$$* \quad \phi'(u) = \frac{2rdr}{du} = -2a^2(1 - \frac{1}{2}e^2 + u + \frac{3}{2}u^2), \therefore \frac{1}{\phi'(u)} = -\frac{1}{2a^2} \cdot (1 - \frac{1}{2}e^2 + u + \frac{3}{2}u^2)^{-1}$$

$$= -\frac{1}{2a^2} \cdot (1 + \frac{1}{2}e^2 - u - \frac{3}{2}u^2 + u^2), \text{ because powers of } u \text{ higher than the second are rejected, inasmuch as they would involve powers of } e \text{ higher than the third, when their}$$

It now remains to determine  $\delta s$ ; but if the formulæ (X) and (Z) of the preceding number be compared together, it will appear that  $\delta r$  is changed into  $\delta s$ , by changing in its expression  $2 \int dR + r \cdot \left\{ \frac{dR}{dr} \right\}$  into  $\left\{ \frac{dR}{dz} \right\}$ ; hence it follows, that in order to obtain  $\delta s$ , it is sufficient to effect this change in the differential equation of  $\delta u$ , and afterwards to substitute the value of  $\delta u$  given by this equation, and which we will denote by  $\delta u'$ , in the expression of  $\delta r$ . Thus, we shall have,

$$0 = \frac{d^2 \cdot \delta u'}{dt^2} + n^2 \cdot \delta u'$$

$$- \frac{1}{a^2} \cdot (1 + \frac{1}{4}e^2 - e \cdot \cos(nt + \epsilon - \omega) - \frac{1}{4}e^2 \cdot \cos(2nt + 2\epsilon - 2\omega)) \cdot \left( \frac{dR}{dz} \right)$$

$$- \frac{2e}{a^2} \cdot \int n dt \cdot (\sin(nt + \epsilon - \omega) \cdot (1 + e \cdot \cos(nt + \epsilon - \omega)) \cdot \left( \frac{dR}{dz} \right)); \quad (Z)$$

values are substituted in place of  $u$ , hence substituting for  $u$  and  $u^2$  their values, namely,  $e \cdot \cos(nt + \epsilon - \omega)$ ,  $\frac{e^2}{2} \cdot \cos(2(nt + \epsilon - \omega)) + \frac{e^2}{2}$ ,  $\frac{2Q}{\phi'(u)} = \left( 2 \int dR + r \cdot \left( \frac{dR}{dr} \right) \right) \cdot \left( -\frac{2}{2a^2} \right)$ ,  $(1 + \frac{1}{2e^2} - \frac{1}{4e^2} - e \cdot \cos(nt + \epsilon - \omega) - \frac{e^2}{4} \cdot \cos(2nt + 2\epsilon - 2\omega))$ ;  $\phi''(u) = d \cdot \frac{\phi'(u)}{du} = -2a^2(1 + 3u)$ ; and  $\frac{1}{\phi'(u)^3} = -\frac{1}{8a^6} \cdot (1 + \frac{3}{2}e^2 - 3u)$ , the other terms are omitted because powers of  $e$  higher than the third would occur when we substitute for  $du$  and  $\phi'(u)$ .  
 $\therefore \frac{\phi'(u)}{\phi'(u)^3} = \frac{-2a^2}{-8a^6} \cdot (1 + 3u) \cdot (1 - 3u)$   
= (omitting terms which would by their multiplication produce powers of  $e$  higher than the third)  $\frac{1}{4a^4}$ ;  $du = -e \cdot ndt \cdot \sin(nt + \epsilon - \omega)$ ;  $\phi'(u) = -2a^2(1 + e \cdot \cos(nt + \epsilon - \omega))$ , hence substituting for  $\phi''(u)$ ,  $\frac{1}{\phi'(u)^3}$ , and  $du \cdot \phi'(u)$ , their values just given we obtain the last term of the equation (X').

$$\delta r = -a \cdot (\delta u \cdot (1 - \frac{3}{2}e^2 + 2u + \frac{9}{2} \cdot u^2), \frac{9}{2}u^2 = \frac{9}{4} \cdot \cos(2(nt + \epsilon - \omega) + \frac{9}{4}e^2), \therefore \delta r = -a \delta u \cdot (1 + \frac{3}{4}e^2 + 2e \cdot \cos(nt + \epsilon - \omega) + \frac{9}{4} \cdot \cos(2nt + 2\epsilon - 2\omega)).$$

$$ds = -a\delta u \cdot (1 + \frac{3}{4}e^2 + 2e \cdot \cos(nt + \epsilon - \omega) + \frac{9}{4}e^2 \cdot \cos(2nt + 2\epsilon - 2\omega)).$$

The system of equations (X'), (Y'), (Z'), will give, in a very simple manner, the troubled motion of  $m$ , if we only consider the first power of the perturbing force. The consideration of the terms due to this power being very nearly sufficient in the theory of the planet; we now proceed to deduce from them formulæ which may be conveniently applied in determining the motion of these bodies.

48. For this purpose, it is necessary to expand the function  $R$  into a series. If we only consider the action of  $m$  on  $m'$ , we have, by N°. 46,

$$R = \frac{m \cdot (xx' + yy' + zz')}{(x'^2 + y'^2 + z'^2)^{\frac{3}{2}}} - \frac{m'}{((x' - x)^2 + (y' - y)^2 + (z' - z)^2)^{\frac{1}{2}}}.$$

This function is entirely independent of the position of the plane of  $x$  and of  $y$ , for as the radical  $\sqrt{(x' - x)^2 + (y' - y)^2 + (z' - z)^2}$ , expresses the distance of  $m$  from  $m'$ , it is independent of it, consequently the function  $x^2 + y^2 + z^2 + x'^2 + y'^2 + z'^2 - 2xx' - 2yy' - 2zz'$  is equally independent of it; but the squares  $z^2 + y^2 + z^2$ , and  $x'^2 + y'^2 + z'^2$ , of the radii vectores do not all depend on this position, therefore the quantity  $xx' + yy' + zz'$ , does not depend on it, and consequently the function  $R$  is independent of it. Let us suppose that in this function

$$\begin{aligned} x &= r \cdot \cos v; & y &= r \cdot \sin v; \\ x' &= r' \cdot \cos v'; & y' &= r' \cdot \sin v'; \end{aligned}$$

we shall have

$$R = m' \cdot \frac{(rr' \cdot \cos(v' - v) + zz')}{(r'^2 + z'^2)^{\frac{3}{2}}} - \frac{m'}{(r^2 - 2rr' \cdot \cos(v' - v) + r'^2 + (z' - z)^2)^{\frac{1}{2}}}.$$

As the orbits of the planets are very nearly circular, and inclined at small angles to each other, the plane of  $x$  and of  $y$  may be so selected, that  $z$  and  $z'$  shall be very small. In this case,  $r$  and  $r'$  differ very little from the greater semiaxes  $a$  and  $a'$  of the elliptic orbits; there-

fore we can suppose that

$$r = a.(1+u); \quad r' = a'.(1+u'); \quad$$

$u$ , and  $u'$  being small quantities. As the angles  $v$  and  $v'$  differ little from the mean longitudes  $nt+\epsilon$ , and  $n't+\epsilon'$ ; we may suppose that

$$v=nt+\epsilon+v; \quad v'=n't+\epsilon'+v'; \quad$$

$v$ , and  $v'$  being very small quantities. Hence it appears that if  $R$  be arranged into a series proceeding according to the powers and products of  $u$ ,  $v$ ,  $z$ ,  $u'$ ,  $v'$ , and  $z'$ ; this series will be very converging.

Let

$$\begin{aligned} & \frac{a}{a'^2} \cdot \cos. (n't - nt + \epsilon' - \epsilon) - (a^2 - 2aa' \cdot \cos. (n't - nt + \epsilon' - \epsilon) + a'^2)^{-\frac{1}{2}} \\ &= \frac{1}{2} \cdot A^{(0)} + A^{(1)} \cdot \cos. (n't - nt + \epsilon' - \epsilon) + A^{(2)} \cdot \cos. 2.(n't - nt + \epsilon' - \epsilon) \\ & \quad + A^{(3)} \cdot \cos. 3.(n't - nt + \epsilon' - \epsilon) + \text{&c.}; \end{aligned}$$

this series may be made to assume the following form, namely,  $\frac{1}{2} \cdot \Sigma. A^{(i)}$ .  $\cos. i.(n't - nt + \epsilon' - \epsilon)$ . the characteristic  $\Sigma$  of finite integrals being relative to the number  $i$ , which ought to extend to all entire numbers from  $i = -\infty$  to  $i = \infty$ ; the value  $i = 0$ , being also comprised in this infinite number of values; but then it ought to be observed, that in this case  $A^{(-i)} = A^{(i)}$ . This form has not only the advantage of enabling us to express, in a very simple manner, the preceding series, but also the product of this series, by the sine or the cosine of any angle  $ft + w$ ; for it is easy to see that this product is equal to

$$\frac{1}{2} \cdot \Sigma. A^{(i)} \cdot \frac{\sin. (i.(n't - nt + \epsilon' - \epsilon) + ft + w)}{\cos.}$$

\* Let  $w = n't - nt + \epsilon' - \epsilon$ , and  $ft + w = p$ , we shall have  $\cos. i.(n't - nt + \epsilon' - \epsilon) \cdot \cos. (ft + w) = \cos. iw \cdot \cos. p$ , now  $\cos. (-iw) = \cos. iw$ ;  $\sin. -iw = -\sin. iw$ ;  $(-i \cdot \sin. (-iw)) = i \cdot \sin. iw$ ,  $\because$  if  $i$  denote the positive values of  $i$ , we shall have  $\cos. iw \cdot \cos. p = 2 \cos. iw \cdot \cos. p = \cos. (p + iw) + \cos. (p - iw)$ ;  $\cos. iw \cdot \sin. p = 2 \cos. iw \cdot \sin. p = \sin. (p + iw) + \sin. (p - iw)$ ;  $i \cdot \sin. iw \cdot \cos. p = 2i \cdot \sin. iw \cdot \cos. p = i \cdot \sin. (p + iw) - i \cdot \sin. (p - iw)$ ;  $i \cdot \sin. iw \cdot \sin. p = 2i \cdot \sin. iw \cdot \sin. p = -i \cdot \cos. (p + iw) + i \cdot \cos. (p - iw)$ ; in the second member of these equations, the first term is changed into the second when  $i$  has a negative value,  $\therefore$  if  $i$  is indifferently positive or negative, the second member is contained in the first; hence we have  $\cos. iw \cdot \cos. p = \frac{1}{2} \cdot \cos. (iw + p)$ . &c. See note page 290.

This property will also enable us to express in a very commodious manner the perturbations of the motions of the planets. Let, in like manner

$$(a^2 - 2aa' \cdot \cos.(n't - nt + \epsilon' - \epsilon) + a'^2)^{-\frac{3}{2}} \\ = \frac{1}{2} \cdot \Sigma. B^{(i)} \cdot \cos. i.(n't - nt + \epsilon' - \epsilon);$$

$B^{(-i)}$  being equal to  $B^{(i)}$ . This being premised, we shall have by the theorems of N°. 21,

$$R = \frac{m'}{2} \cdot \Sigma. A^{(i)} \cdot \cos. i.(n't - nt + \epsilon' - \epsilon)^* \\ + \frac{m'}{2} \cdot u_i \cdot \Sigma. a \cdot \left( \frac{dA^{(i)}}{da} \right) \cdot \cos. i.(n't - nt + \epsilon' - \epsilon) \\ + \frac{m'}{2} \cdot u'_i \cdot \Sigma. a' \cdot \left( \frac{dA^{(i)}}{da'} \right) \cdot \cos. i.(n't - nt + \epsilon' - \epsilon) \\ - \frac{m'}{2} \cdot (v'_i - v_i) \cdot \Sigma. i. A^{(i)} \cdot \sin. i.(n't - nt + \epsilon' - \epsilon) \\ + \frac{m'}{4} \cdot u_i^2 \cdot \Sigma. a^2 \cdot \left( \frac{d^2 A^{(i)}}{da^2} \right) \cdot \cos. i.(n't - nt + \epsilon' - \epsilon)$$

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\* Substituting for  $r, r', v, v'$ , their values, the constant part of the value of  $R$  will become  $\frac{m'.aa' \cdot \cos.(n't - nt + \epsilon' - \epsilon) + zz'}{(a^2 + z^2)^{\frac{3}{2}}} - \frac{m'}{(a^2 - 2aa' \cdot \cos.(n't - nt + \epsilon' - \epsilon) + a'^2 + (z - z')^2)^{\frac{3}{2}}}$ ,

which becomes (by reducing, and observing that terms higher than of the order of the square of the disturbing forces are neglected)  $= m'.(aa' \cdot \cos.((n't - nt + \epsilon' - \epsilon) + zz' \cdot (a'^{-3} - \frac{3}{2}a'^{-5}z^2))$

$$= \frac{m'}{(a^2 - 2aa' \cdot \cos.(n't - nt + \epsilon' - \epsilon) + a'^2)^{\frac{1}{2}} + \frac{\frac{m'}{2} \cdot (z - z')^2}{(a^2 - 2aa' \cdot \cos.(n't - nt + \epsilon' - \epsilon) + a'^2)^{\frac{3}{2}}}; \\ = \frac{m'}{2} \cdot \Sigma. A^{(i)} \cdot \cos. i.(n't - nt + \epsilon' - \epsilon) + \frac{m'.zz'}{a'^3} - \frac{\frac{3}{2}m'.az^2 \cdot \cos. (n't - nt + \epsilon' - \epsilon)}{a'^4} \\ + \frac{m'}{4} \cdot (z - z')^2 \cdot \Sigma. B^{(i)} \cdot \cos. i.(n't - nt + \epsilon' - \epsilon);$$

now if  $a, a', nt + \epsilon, n't + \epsilon'$ , be supposed to be increased by  $u', u'_i, v, v'$  respectively, the value of  $R$  will be given by the formula of N°. 21, in the manner expressed in the text.

$$\begin{aligned}
& + \frac{m'}{2} \cdot (u, u' \cdot \Sigma aa' \cdot \left( \frac{d^2 \cdot A^{(i)}}{da \cdot da'} \right) \cdot \cos. i \cdot (n't - nt + \epsilon' - \epsilon)) \\
& + \frac{m'}{4} \cdot u'^2 \cdot \Sigma a'^2 \cdot \left( \frac{d^2 \cdot A^{(i)}}{da'^2} \right) \cdot \cos. i \cdot (n't - nt + \epsilon' - \epsilon) \\
& - \frac{m'}{2} \cdot (v' - v) \cdot u \cdot \Sigma ia \cdot \left( \frac{dA^{(i)}}{da} \right) \cdot (\sin. i \cdot (n't - nt + \epsilon' - \epsilon)) \\
& - \frac{m'}{2} \cdot (v' - v) \cdot u' \cdot \Sigma ia' \cdot \left( \frac{dA^{(i)}}{da'} \right) \cdot \sin. i \cdot (n't - nt + \epsilon' - \epsilon) \\
& - \frac{m'}{4} \cdot (v' - v)^2 \cdot \Sigma i^2 \cdot A^{(i)} \cdot \cos. i \cdot (n't - nt + \epsilon' - \epsilon) \\
& + \frac{m' \cdot zz'}{a'^3} - \frac{3m' \cdot az'^2}{2a'^4} \cdot \cos. (n't - nt + \epsilon' - \epsilon) \\
& + m' \cdot \frac{(z' - z)^2}{4} \cdot \Sigma B^{(i)} \cdot \cos. i \cdot (n't - nt + \epsilon' - \epsilon) \\
& + \text{&c.}
\end{aligned}$$

If in this expression of  $R$ , the values relative to the elliptic motion, are substituted in place of  $u, u', v, v', z$  and  $z'$ , which values are functions of the sines and cosines of the angles  $nt + \epsilon, n't + \epsilon'$ , and of their multiples ;  $k$  will be expressed by an infinite series of cosines of the form  $m'k \cdot \cos. (i'n't - int + A)$ , \*  $i$  and  $i'$  being entire numbers.

It is evident that the action of the bodies  $m'', m''', \text{ &c.}$ , on  $m$ , will produce in  $R$ , terms analogous to those which result from the action of  $m'$ , and that we shall obtain them, by changing in the preceding expression of  $R$ , all that which is relative to  $m'$ , into the same quantities relative to  $m'', m''', \text{ &c.}$

Let any term  $m'k \cdot \cos. (i'n't - int + A)$  of the expression for  $R$ , be considered. If the orbits were circular, and existed in the same

\* The form of this function is always that of a cosine, for the values of  $u, u'$ , are expressed by series of the cosines of  $nt + \epsilon, n't + \epsilon'$ , and of their multiples, which are multiplied into a function of the form  $\Sigma. \cos. i \cdot (n't - nt + \epsilon' - \epsilon)$ , the value of  $v - v'$  is expressed by a series involving  $\sin. (nt + \epsilon), \sin. (n't + \epsilon')$ ; and their multiples, and this is multiplied into a function of the form  $\Sigma. \sin. i \cdot (n't - nt + \epsilon' - \epsilon)$ .

plane, we would have  $i = i$ , therefore  $i'$  cannot surpass  $i$ , or be surpassed by it, but by means of the sines and cosines of the expressions of  $u, v, z, u', v', z'$  which by combining with the sines and cosines of the angle  $n't - nt + \epsilon' - \epsilon$ , and of its multiples, would produce sines and cosines of angles in which  $i$  is different from  $i$ .

If we consider the excentricities and inclinations of the orbits, as very small quantities of the first order, it results from the formulæ of N° 22, that in the expressions of  $u, v, z$  or  $rs, s$  being the tangent of the latitude of  $m$ , the coefficient of the sine or of the cosine of an angle, such as  $f.(nt + \epsilon)$ , is expressed by a series, of which the first term is of the order  $f$ , the second term of the order  $f + 2$ , the third of the order\*  $f + 4$ ; and so of the rest. The same obtains for the coefficient of the sine and cosine of the angle  $f'(n't + \epsilon')$ , in the expressions of  $u', v', z'$ . It follows from this, that  $i$  and  $i'$  being supposed positive, and  $i'$  greater than  $i$ : the coefficient  $k$  in the term of  $m'k \cdot \cos(i'n't - int + A)$ , is of the order  $i' - i$ , and that in the series which expresses it, the first term is of the order  $i' - i$ , the second term is of the order  $i' - i + 2$ , and so of the rest, so that this series is very converging. If  $i$  be greater than  $i'$ , the terms of the series will be successively of the orders  $i - i', i - i' + 2$ , &c.

\* It is evident from inspection of the series in pages 150, 152, that when all the coefficients of the function  $\cos. f.(nt + \epsilon)$  are collected together, they will constitute a series of the form  $es \pm es^{+2} \pm es^{+4} \pm es^{+6}$ , &c., hence multiplying  $\cos. f.(nt + \epsilon)$  into  $\cos. i.(n't - nt + \epsilon' - \epsilon)$  the product will be of the form of

$$\frac{\cos. i.(n't - nt + \epsilon' - \epsilon) + f.(n't + \epsilon')}{2},$$

= by making  $f' + i = i'$

$$\frac{\cos. (i'n't - int + A)}{2};$$

which is to be multiplied into the series  $es, es^{+2}, es^{+4}, \text{&c.} = (\text{as } f = i' - i = ), e^{i - i}, e^{i - i + 2}, \text{ &c.}$

Let  $\varpi$  denote the longitude of the perihelion of the orbit of  $m$ , and  $\theta$  the longitude of its node ; and in like manner let  $\varpi'$  denote the longitude of the perihelion of the orbit of  $m'$ , and  $\theta'$  that of its node ; these longitudes being reckoned on a plane very little inclined to that of its orbit. It follows from the formulæ of N°. 22, that in the expressions of  $u$ ,  $v$ , and  $z$ , the angle  $nt+\epsilon$  is always accompanied by  $-\varpi$ , or by  $-\theta$  ; and that in the expressions of  $u'$ ,  $v'$ , and  $z'$ , the angle  $n't+\epsilon'$  is always accompanied by  $-\varpi'$ , or by  $-\theta'$ , hence it follows that the term  $m'k \cos. (i'n't - int + A)$  is of the following form

$$m'k. \cos. (i'n't - int + i'\epsilon' - i\epsilon - g\varpi - g'\varpi' - g''\theta - g'''\theta'),$$

$g$ ,  $g'$ ,  $g''$ ,  $g'''$ , being entire numbers, positive or negative, and such that we have

$$0 = i' - i - g - g' - g'' - g'''.$$

This also follows from considering, that the value of  $R$ , and its different terms are independent of the position of the right line, from which we reckon the longitudes. Moreover, in the formulæ of N°. 22, the coefficient of the sine and cosine of the angle  $\varpi$ , has always for factor the excentricity  $e$  of the orbit of  $m$ , the coefficient of the sine and cosine of the angle  $2\varpi$ , has for factor the square of this excentricity, and so of the rest. In like manner, the coefficient of the sine and cosine of the angle  $\theta$ , has for factor  $\tan. \frac{1}{2}\phi$ ,  $\phi$  being the inclination of the orbit of  $m$  on a fixed plane. The coefficient of the sine and cosine of the angle  $2\theta$ , has for factor  $\tan. \frac{1}{2}\phi$ , and so of the rest ; from this it follows, that the coefficient  $k$  has for factor,  $e^e. e'^{e'} \tan. \phi''(\frac{1}{2}\phi)$ .  $\tan. \phi'''(\frac{1}{2}\phi)$  the numbers  $g$ ,  $g'$ ,  $g''$ ,  $g'''$ , being taken positively in the exponents of these factors. If all these numbers are positive in themselves, this factor will be of the order  $i' - i$ , in consequence of the equation

$$0 = i' - i - g - g' - g'' - g''' ;$$

but if one of them, such as  $g$ , be negative and equal to  $-g$ , this factor will be of the order  $i - i + 2g$ .<sup>\*</sup> Therefore, if we only preserve, among the terms of  $k$ , those which depending on the angle  $i'nt - int$ , are of the order  $i - i$ , and neglect all those which depending on the same angle, are of the orders  $i - i + 2$ ,  $i - i + 4$ , &c.; the expression of  $k$  will be constituted in the following manner:

$$H.e^g.e'^{g'}.\tan\left(\frac{1}{2}\phi\right).\tan\left(\frac{g''}{2}\phi'\right).\cos(i'nt - int \\ + i'\varepsilon - i\varepsilon - g.\varpi - g'.\varpi' - g''.\theta - g'''.\theta')$$

$H$  being a coefficient independent of the excentricities and of the inclinations of the orbits, and the numbers  $g, g', g'', g'''$ , being all positive, and such that their sum is equal to  $i - i$ .

If we substitute in  $R$ ,  $a.(1+u)$ , in place of  $r$ , we shall have

$$r.\left(\frac{dR}{dr}\right) = a.\left(\frac{dR}{da}\right).$$

If in this same function, we substitute in place of  $u$ ,  $v$ , and  $z$ , their values given by the formulæ of N°. 22, we shall have

$$\left(\frac{dR}{dv}\right) = \left(\frac{dR}{dz}\right).$$

provided we suppose that  $\varepsilon - \varpi$ , and  $\varepsilon - \theta$  are constant, in the differential of  $R$ , taken relatively to  $\varepsilon$ ; for then  $u$ ,  $v$ , and  $z$  are constant in this differential, and as we have  $v = nt + \varepsilon + v_0$ , it is evident that the preceding equation has place. We can therefore easily obtain the values of  $r.\left(\frac{dR}{dr}\right)$  and of  $\left(\frac{dR}{dv}\right)$ , which occur in

\* For in this case  $i - i + 2g = g + g' + g'' + g'''$ .

$$r = a.(1+u); \frac{dr}{da} = 1+u, \frac{dR}{da} = \frac{dR}{dr} \cdot \frac{dr}{da}, = \frac{dr}{da} \cdot (1+u), \therefore a.\left(\frac{dR}{da}\right) = \\ r.\left(\frac{dR}{dr}\right).$$

the differential equations of the preceding numbers, when we shall have obtained the value of  $R$  expanded into a series of the cosines of angles increasing proportionally to the time. It will also be very easy to determine the differential  $dR$ , by taking care that the angle  $nt$ , solely varies in the expression of  $R$ , the angle  $n't$  being supposed to be constant; because  $dR$  is the difference of  $R$ , taking on the supposition, that the coordinates of  $m'$ , which are functions of  $n't$ , are constant.

49. The difficulty of the expansion of  $R$  into a series, is reduced to the determination of the quantities  $A^{(i)}$ ,  $B^{(i)}$ , and their differences, taken relatively to  $\alpha$  and  $\alpha'$ . For this purpose, let us consider generally the function  $(\alpha^2 - 2\alpha\alpha' \cdot \cos. \theta + \alpha'^2)^{-s}$ , and let us expand it according to the cosines of the angle  $\theta$ , and of its multiples. By making  $\frac{\alpha}{\alpha'} = \alpha$ , it will become  $\alpha'^{-2s} \cdot (1 - \alpha \cdot \cos. \theta + \alpha)^{-s}$ . Let

$$(1 - 2\alpha \cdot \cos. \theta + \alpha^2)^{-s} = \frac{1}{2} \cdot b_s^{(0)} + b_s^{(1)} \cdot \cos. \theta + b_s^{(2)} \cdot \cos. 2\theta + b_s^{(3)} \cdot \cos. 3\theta + \text{&c.}$$

$b_s^{(0)}$ ,  $b_s^{(1)}$ ,  $b_s^{(2)}$ , &c., being functions of  $\alpha$  and  $s$ . If we take the logarithmic differences of the two members of this equation, with respect to the variable  $\theta$ , we shall have

$$\frac{-2s \cdot \alpha \cdot \sin. \theta}{1 - 2\alpha \cdot \cos. \theta + \alpha^2} = \frac{-b_s^{(1)} \cdot \sin. \theta - 2b_s^{(2)} \cdot \sin. 2\theta - \text{&c.}}{\frac{1}{2} \cdot b_s^{(0)} + b_s^{(1)} \cdot \cos. \theta + b_s^{(2)} \cdot \cos. 2\theta + \text{&c.}},$$

by multiplying transversely, and comparing together like cosines, we find generally

$$b_s^{(i)} = \frac{(i-1) \cdot (1 + \alpha^2) \cdot b_s^{(i-1)} - (i+s-2) \cdot \alpha \cdot b_s^{(i-2)}}{(i-s) \cdot \alpha}; \quad (a)$$

by this means the values of  $b_s^{(2)}$ ,  $b_s^{(3)}$ , &c., will be given, when  $b_s^{(0)}$ ,  $b_s^{(1)}$ , are known.

$s$  being changed into  $s+1$  in the preceding expression of  $(1-2\alpha \cdot \cos. \theta + \alpha^2)^{s-1}$ , we shall have

$$(1-2\alpha \cdot \cos. \theta + \alpha^2)^{s-1} = \frac{1}{2} \cdot b_{s+1}^{(0)} + b_{s+1}^{(1)} \cos. \theta + b_{s+1}^{(2)} \cos. 2\theta + b_{s+1}^{(3)} \cos. 3\theta + \text{&c.}$$

By multiplying the two members of this equation, by  $1-2\alpha \cdot \cos. \theta + \alpha^2$ , and by substituting in place of  $(1-2\alpha \cdot \cos. \theta + \alpha^2)^s$ , its value in a series, we shall have

$$\begin{aligned} & \frac{1}{2} \cdot b_s^{(0)} + b_s^{(1)} \cos. \theta + b_s^{(2)} \cos. 2\theta + \text{&c.} \\ & = (1-2\alpha \cdot \cos. \theta + \alpha^2) \cdot (\frac{1}{2} \cdot b_{s+1}^{(0)} + b_{s+1}^{(1)} \cos. \theta + b_{s+1}^{(2)} \cos. 2\theta + b_{s+1}^{(3)} \cos. 3\theta + \text{&c.}) ; \end{aligned}$$

from which may be obtained, by a comparison of similar cosines

$$b_s^{(i)} = (1+\alpha^2) \cdot b_{s+1}^{(i)} - \alpha \cdot b_{s+1}^{(i-1)} - \alpha \cdot b_{s+1}^{(i+1)} . *$$

The formula (a) gives

$$b_{s+1}^{i+1} = \frac{i \cdot (1+\alpha^2) \cdot b_{s+1}^{(i)} - (i+s) \cdot \alpha \cdot b_{s+2}^{(i-1)}}{(i-s) \cdot \alpha} ;$$

the preceding expression of  $b_s^{(i)}$ , will therefore become

$$b_s^{(i)} = \frac{2s \cdot \alpha \cdot b_{s+1}^{(i-1)} - s \cdot (1+\alpha^2) \cdot b_{s+1}^{(i)}}{i-s}.$$

When this transverse multiplication is performed we must substitute for  $\cos. \theta \cdot \sin. i\theta$ ,  $\sin. \theta \cdot \cos. i\theta$ , their values in terms of  $\frac{\sin. (i+1)\theta}{2} + \frac{\sin. (i-1)\theta}{2}$ ; hence we obtain  
 $-s\alpha \xi_s^{(i-2)} \cdot \sin. (i-1)\theta - s\alpha \xi_s^{(i)} \cdot \sin. (i-1)\theta - (1+\alpha^2)(i-1)\xi_s^{i-1} \sin. (i-1)\theta + \alpha i \xi_s^{(i)} \cdot \sin. (i-1)\theta + \alpha(i-2)\xi_s^{(i-2)} \cdot \sin. (i-1)\theta = 0$ .  $\therefore \xi_s^i \cdot \alpha \cdot (i-s) = (1+\alpha^2)(i-1)\xi_s^{(i-1)} - \alpha(i-2+s)\xi_s^{i-2}$ .

\* To obtain this value of  $\xi_s^{(i)}$ , it is to be remarked that  $\cos. \theta \cdot \cos. i\theta = \frac{\cos. (i+1)\theta + \cos. (i-1)\theta}{2}$ , hence multiplying the two factors of the second member of this equation, the coefficient of  $\cos. i\theta$  is  $(1+\alpha^2) \xi_{s+1}^{(i)} - \alpha b_{s+1}^{(i+1)} - \alpha b_{s+1}^{(i-1)}$ .

By changing  $i$  into  $i+1$ , in this equation, we shall have

$$b_s^{(i+1)} = \frac{2s.\alpha.b_{s+1}^{(i)} - s.(1+\alpha^2).b_{s+1}^{(i+1)}}{i-s+1};$$

and if we substitute in place of  $b_{s+1}^{(i+1)}$ , its preceding value, we will have

$$b_s^{(i+1)} = \frac{s.(i+s).\alpha.(1+\alpha^2).b_{s+1}^{(i-1)} + s.(2(i-s).\alpha^2 - i.(1+\alpha^2)^2).b_{s+1}^{(i)}}{(i-s).(i-s+1).\alpha}.$$

These two expressions of  $b_s^{(i)}$ , and of  $b_s^{(i+1)}$ , give

$$b_{s+1}^{(i)} = \frac{\frac{(i+s)}{s}.(1+\alpha^2).b_s^{(i)} - 2.\frac{(i-s+1)}{s}.\alpha b_s^{(i+1)}}{(1-\alpha^2)^2}; \quad (b)$$

by substituting for  $b_s^{(i+1)}$ , its value deduced from the equation (a), we shall have

$$b_{s+1}^{(i)} = \frac{\frac{(s-i)}{s}.(1+\alpha^2).b_s^{(i)} + \frac{2(i+s-1)}{s}.\alpha b_s^{(i-1)}}{(1-\alpha^2)^2}; \quad (c)^*$$

which expression might have been inferred from the preceding by changing  $i$  into  $-i$ , and by remarking that  $b_s^{(i)} = b_s^{(-i)}$ . We shall consequently obtain by means of this formula, the values of  $b_{s+1}^{(0)}, b_{s+1}^{(1)}, b_{s+1}^{(2)},$  &c., when the values of  $b_s^{(0)}, b_s^{(1)}, b_s^{(2)},$  &c., will have been known.

In order to abridge, let  $\lambda$  denote the function  $1 - 2\alpha \cdot \cos \theta + \alpha^2$ , by differentiating with respect to  $\alpha$ , the equation

$$\lambda^{-s} = \frac{1}{2} \cdot b_s^{(0)} + b_s^{(1)} \cos \theta + b_s^{(2)} \cos 2\theta + \text{&c.};$$

we will obtain

$$-2s.(\alpha - \cos \theta) \cdot \lambda^{-s-1} = \frac{1}{2} \cdot \frac{db_s^{(0)}}{d\alpha} + \frac{db_s^{(1)}}{d\alpha} \cdot \cos \theta + \frac{db_s^{(2)}}{d\alpha} \cdot \cos 2\theta + \text{&c.};$$

\* Hence if we know the coefficients of the multiple cosines in the series which is equivalent to  $(1 - 2\alpha \cdot \cos \theta + \alpha^2)^{-s}$ , we know the coefficients of the multiple cosines in the series which is equivalent to  $(1 - 2\alpha \cdot \cos \theta + \alpha^2)^{-s-1}$ .

but we have

$$-\alpha + \cos. \theta = \frac{1 - \alpha^2 - \lambda}{2\alpha};$$

therefore we shall have

$$\frac{s.(1 - \alpha^2)}{\alpha} \cdot \lambda^{-s-1} - \frac{s \cdot \lambda^{-s}}{\alpha} = \frac{1}{2} \cdot \frac{db_s^{(0)}}{d\alpha} + \frac{db_s^{(1)}}{d\alpha} \cdot \cos. \theta + \text{ &c. ;}$$

hence we deduce generally

$$\frac{db_s^{(i)}}{d\alpha} = \frac{s.(1 - \alpha^2)}{\alpha} \cdot b_{s+1}^{(i)} - \frac{s \cdot b_s^{(i)}}{\alpha}.*$$

By substituting in place of  $b_{s+1}^{(i)}$  its value given by the formula (b), we will obtain

$$\frac{db_s^{(i)}}{d\alpha} = \left\{ \frac{i + (i+2s) \cdot \alpha^2}{\alpha \cdot (1 - \alpha^2)} \right\} \cdot b_s^{(i)} - \frac{2 \cdot (i-s+1)}{1 - \alpha^2} \cdot b_{s+1}^{(i)} +$$

This equation being differentiated, will give

$$\begin{aligned} \frac{d^2 b_s^{(i)}}{d\alpha^2} &= \left\{ \frac{i + (i+2s) \cdot \alpha^2}{\alpha \cdot (1 - \alpha^2)} \right\} \cdot \frac{db_s^{(i)}}{d\alpha} + \left\{ \frac{2 \cdot (i+s) \cdot (1 + \alpha^2)}{(1 - \alpha^2)^2} - \frac{i}{\alpha^2} \right\} \cdot b_s^{(i)}. \\ &\quad - 2 \cdot \frac{(i-s+1)}{1 - \alpha^2} \cdot \frac{db_{s+1}^{(i+1)}}{d\alpha} - 4 \cdot \frac{(i-s+1) \cdot \alpha}{(1 - \alpha^2)^2} \cdot b_{s+1}^{(i+1)}. \\ &\quad \text{on 2} \end{aligned}$$

\* Substituting for  $\lambda^{-s-1}$ ,  $\lambda^{-s}$ , their values given in the preceding page, the coefficient of  $\cos. i\theta$ , in the value of  $\lambda^{-s-1}$  is  $\xi_{s+1}^{(i)}$ , and the coefficient of the same quantity in the value of  $\lambda^{-s}$  is  $\xi_s^{(i)}$ .

† Differencing the coefficient of  $\xi_s^{(i)}$  with respect to  $\alpha$  it becomes

$$\begin{aligned} &\frac{-i \cdot (1 - \alpha^2) + 2\alpha^2 i}{\alpha^2 \cdot (1 - \alpha^2)^2} + \frac{(i+2s) \cdot (1 - \alpha^2) + 2\alpha^2 \cdot (i+2s)}{(1 - \alpha^2)^2} = \\ &\frac{i \cdot (1 - 3\alpha^2)}{\alpha^2 \cdot (1 - \alpha^2)^2} + \frac{(i+2s) \cdot \alpha^2 \cdot (1 - \alpha^2)}{\alpha^2 \cdot (1 - \alpha^2)^2} + \frac{2\alpha^2 \cdot (i+2s)}{(1 - \alpha^2)^2} = - \frac{i \cdot (1 - 2\alpha^2 + \alpha^4)}{\alpha^2 \cdot (1 - \alpha^2)^2} \\ &+ \frac{(1 + \alpha^2) \cdot 2(s+i)}{(1 - \alpha^2)^2}. \end{aligned}$$

By differentiating again, we will obtain

$$\begin{aligned} \frac{d^3 b_s^{(i)}}{dx^3} &= \left\{ \frac{i+(i+2s)\cdot\alpha^2}{\alpha\cdot(1-\alpha^2)} \right\} \cdot \frac{d^2 b_s^{(i)}}{d\alpha^2} + 2 \cdot \left\{ \frac{2\cdot(i+s)\cdot(1+\alpha^2)}{(1-\alpha^2)^2} \cdot \frac{i}{\alpha^2} \right\} \cdot \frac{db_s^{(i)}}{d\alpha} \\ &+ \left\{ \frac{4(i+s)\cdot\alpha(3+\alpha^2)}{(1-\alpha^2)^3} + \frac{2i}{\alpha^3} \right\} \cdot b_s^{(i)} - \frac{2\cdot(i-s+1)}{1-\alpha^2} \cdot \frac{d^2 \cdot b_s^{(i+1)}}{d\alpha^2} \\ &- \frac{8\cdot(i-s+1)\cdot\alpha}{(1-\alpha^2)^2} \cdot \frac{db_s^{(i+1)}}{d\alpha} - \frac{4\cdot(i-s+1)\cdot(1+3\alpha^2)}{(1-\alpha^2)^3} \cdot b_s^{(i+1)}. \end{aligned}$$

It appears from this that in order to determine the values of  $b_s^{(i)}$ , and of its successive differentials, it is sufficient to know those of  $b_s^{(0)}$ , and of  $b_s^{(1)}$ . These two quantities may be determined in the following manner :

Let  $c$  represent the number of which the hyperbolical logarithm is unity ; the expression of  $\lambda^{-s}$ , may be made to assume the following form :

$$\lambda^{-s} = (1 - \alpha \cdot c^{i\theta\sqrt{-1}})^{-s} \cdot (1 - \alpha \cdot c^{-i\theta\sqrt{-1}})^{-s}.$$

By expanding the second member of this equation, with respect to the powers of  $c^{i\theta\sqrt{-1}}$ , and of  $c^{-i\theta\sqrt{-1}}$ , it is evident that the two exponential quantities  $c^{i\theta\sqrt{-1}}$ , and  $c^{-i\theta\sqrt{-1}}$  will have the same coefficients which we will denote by  $k$ . The sum of the two terms  $k \cdot c^{i\theta\sqrt{-1}}$ , and  $k \cdot c^{-i\theta\sqrt{-1}}$  is  $2k \cdot \cos i\theta$  ; this will be the value of  $b_s^{(i)} \cdot \cos i\theta$  ; therefore we will obtain  $b_s^{(i)} = 2k$ . Now the expression of  $\lambda^{-s}$  is equal to the product of the two series

$$1 + s\alpha \cdot c^{i\theta\sqrt{-1}} + \frac{s \cdot (s+1)}{1 \cdot 2} \cdot \alpha^2 \cdot c^{2i\theta\sqrt{-1}} + \text{&c.};$$

$$1 + s\alpha \cdot c^{-i\theta\sqrt{-1}} + \frac{s \cdot (s+1)}{1 \cdot 2} \cdot \alpha^2 \cdot c^{-2i\theta\sqrt{-1}} + \text{&c.}$$

these two series being multiplied, the one by the other, will give, in

the case of  $i = 0$ ,\*

$$k = 1 + s^2 \alpha^2 + \left\{ \frac{s.(s+1)}{1.2} \right\}^2 \cdot \alpha^4 + \text{&c.};$$

and in the case of  $i = 1$ ,

$$k = \alpha \cdot \left\{ s+s \cdot \frac{s.(s+1)}{1.2} \cdot \alpha^2 + \frac{s.(s+1)}{1.2} \cdot \frac{s.(s+1).(s+2)}{1.2.3} \cdot \alpha^4 + \text{&c.} \right\},$$

consequently,

$$b_s^{(0)} = 2 \cdot \left\{ 1 + s^2 \cdot \alpha^2 + \left\{ \frac{s.(s+1)}{1.2} \right\}^2 \cdot \alpha^4 + \left\{ \frac{s.(s+1).(s+2)^2}{1.2.3} \right\} \cdot \alpha^6 + \text{&c.} \right\}$$

$$b_s^{(1)} = 2\alpha \cdot \left\{ s+s \cdot \frac{s.(s+1)}{1.2} \cdot \alpha^2 + \frac{s.(s+1)}{1.2} \cdot \frac{s.(s+1).(s+2)}{1.2.3} \cdot \alpha^4 + \text{&c.} \right\}$$

In order that this series may converge, it is necessary that  $\alpha$  should be less than unity; this may be always effected by assuming  $\alpha$  equal to the ratio of the smaller of the distances  $a$  and  $a'$  to the greater, and as we have already supposed  $a = \frac{a}{a'}$ , we will assume that  $a$  is less than  $a'$ .

In the theory of the motions of the bodies  $m, m', m'', \text{&c.}$ , it is necessary to know the values of  $b_s^{(0)}$ , and of  $b_s^{(1)}$ , when  $s = \frac{1}{2}$ , and  $s = \frac{5}{2}$ . In these two cases these values do not converge rapidly unless  $\alpha$  is a very small fraction. These series converge with greater rapidity when  $s = -\frac{1}{2}$ , and we have

$$\frac{1}{2} b_{-\frac{1}{2}}^{(0)} = 1 + (\frac{1}{2})^2 \cdot \alpha^2 + \left( \frac{1.1}{2.4} \right)^2 \cdot \alpha^4 + \left( \frac{1.1.3}{2.4.6} \right)^2 \cdot \alpha^6 + \left( \frac{1.1.3.5}{2.4.6.8} \right)^2 \cdot \alpha^8 + \text{&c.}$$

$$b_{-\frac{1}{2}}^{(1)} = -\alpha \cdot \left\{ 1 - \frac{1.1}{2.4} \cdot \alpha^2 - \frac{1}{4} \cdot \frac{1.1.3}{2.4.6} \cdot \alpha^4 - \frac{1.3}{4.6} \cdot \frac{1.1.3.5}{2.4.6.8} \cdot \alpha^6 - \frac{1.3.5}{4.6.8} \cdot \frac{1.1.3.5.7}{2.4.6.8.10} \cdot \alpha^8 - \text{&c.} \right\}.$$

\*  $i = 0$  when equal powers of  $b^{\theta\sqrt{-1}}$ , and  $c^{-\theta\sqrt{-1}}$ , are multiplied together and,  $i=1$ , when powers of  $b^{\theta\sqrt{-1}}$ , are multiplied into powers of  $c^{-\sqrt{-1}}$ , which are less by unit than these. This is evident from the value of  $k$ .

In the theory of the planets and of the satellites, it will be sufficient to assume the sum of the first eleven or twelve terms, the subsequent being neglected, or more accurately, by summing\* them as a geometric progression of which the ratio is  $1-\alpha^2$ . When  $b_{-\frac{1}{2}}^{(0)}$ ,  $b_{-\frac{1}{2}}^{(1)}$ , shall have been thus determined we will obtain  $b_{-\frac{1}{2}}^{(0)}$  by making  $i=0$ , and  $s=-\frac{1}{2}$ , in the formula (b), and we will find

$$b_{\frac{1}{2}}^{(0)} = \frac{(1+\alpha^2) \cdot b_{-\frac{1}{2}}^{(0)} + 6 \cdot \alpha \cdot b_{-\frac{1}{2}}^{(1)}}{(1-\alpha^2)^2}.$$

If in the formula (c), we suppose  $i=1$ , and  $s=-\frac{1}{2}$ , we will have

$$b_{\frac{1}{2}}^{(1)} = \frac{2 \cdot \alpha b_{-\frac{1}{2}}^{(0)} + 3 \cdot (1+\alpha^2) \cdot b_{-\frac{1}{2}}^{(1)}}{(1-\alpha^2)^2}.$$

By means of these values of  $b_{\frac{1}{2}}^{(0)}$ , and of  $b_{\frac{1}{2}}^{(1)}$ , we will obtain by the preceding formula, the values of  $b_{\frac{1}{2}}^{(i)}$ , and of its partial differences, whatever may be the number  $i$ , from which we may we may conclude the values of  $b_{\frac{3}{2}}^{(i)}$ , and of its differences. The values of  $b_{\frac{3}{2}}^{(0)}$ , and of  $b_{\frac{3}{2}}^{(1)}$  may be determined very simply, by the following formulæ;

$$b_{\frac{3}{2}}^{(0)} = \frac{b_{-\frac{1}{2}}^{(0)}}{(1-\alpha^2)^2}; \quad b_{\frac{3}{2}}^{(1)} = -3 \cdot \frac{b_{-\frac{1}{2}}^{(1)}}{(1-\alpha^2)^2}. \dagger$$

\* For if  $(1-\alpha^2)^{-1}$  be expanded to a series, the sum of the remaining terms will be very nearly equal to this series multiplied into the eleventh term.

† By formula (b)  $b_{\frac{3}{2}}^{(0)} = \frac{(1+\alpha^2) \cdot b_{-\frac{1}{2}}^{(0)} - 2\alpha b_{-\frac{1}{2}}^{(1)}}{(1-\alpha^2)^2}$ ; substituting for  $b_{-\frac{1}{2}}^{(0)}$ ,  $b_{-\frac{1}{2}}^{(1)}$ , their values we obtain  $b_{\frac{3}{2}}^{(0)} = \frac{(1+\alpha^2)^2 b_{-\frac{1}{2}}^{(0)} + 6\alpha \cdot (1+\alpha^2) \cdot b_{-\frac{1}{2}}^{(1)} - 4\alpha^2 \cdot b_{-\frac{1}{2}}^{(0)} - 6\alpha \cdot (1+\alpha^2) \cdot b_{-\frac{1}{2}}^{(1)}}{(1-\alpha^2)^4} = \frac{(1-\alpha^2)^2 b_{-\frac{1}{2}}^{(0)}}{(1-\alpha^2)^4}$   
 $\frac{b_{-\frac{1}{2}}^{(0)}}{(1-\alpha^2)^2}$ ; in a similar manner we obtain the value of  $b_{\frac{3}{2}}^{(1)}$ .

$$\frac{d^2 A^{(1)}}{da^2} = \frac{1}{a'^2} \cdot \frac{d^2 b_{\frac{1}{2}}^{(1)}}{da^2} \cdot \frac{da}{d\alpha} = -\frac{1}{a'^3} \cdot \frac{d^2 b_{\frac{1}{2}}^{(1)}}{da^2}$$

Now, in order to obtain the quantities  $A^{(0)}$ ,  $A^{(1)}$ , &c., and their differences, it may be remarked that by the preceding number, the series

$$\frac{1}{2} \cdot A^{(0)} + A^{(1)} \cdot \cos. \theta + A^{(2)} \cdot \cos. 2\theta + \text{&c.}$$

results from the expansion of the function

$$\frac{a \cdot \cos. \theta}{a'^2} - (a^2 - 2aa' \cdot \cos. \theta + a'^2)^{-\frac{1}{2}},$$

in a series ranged according to the cosines of the angle  $\theta$  and of its multiples; by making  $\frac{a}{a'} = \alpha$ , this same function is reduced to

$$-\frac{1}{2a'} \cdot b_{\frac{1}{2}}^{(0)} + \left( \frac{a}{a'^2} - \frac{1}{a'} \cdot b_{\frac{1}{2}}^{(1)} \right) \cdot \cos. \theta - \frac{1}{a'} \cdot b_{\frac{1}{2}}^{(2)} \cdot \cos. 2\theta - \text{&c.}$$

which gives generally

$$A^{(i)} = -\frac{1}{a'} \cdot b_{\frac{1}{2}}^{(i)};$$

when  $i$  is zero, or greater than unity, abstracting from the sign. In the case of  $i = 1$ , we have

$$A^{(1)} = -\frac{1}{a'} \cdot \frac{db_{\frac{1}{2}}^{(1)}}{d\alpha} \cdot \left( \frac{d\alpha}{da} \right);$$

we have then

$$\frac{dA^{(i)}}{d\alpha} = -\frac{1}{a'} \cdot \frac{db_{\frac{1}{2}}^{(i)}}{d\alpha} \cdot \left( \frac{d\alpha}{da} \right);$$

but we have  $\left( \frac{d\alpha}{da} \right) = \frac{1}{a'}$ ; therefore

$$\frac{dA^{(i)}}{d\alpha} = -\frac{1}{a'^2} \cdot \frac{db_{\frac{1}{2}}^{(i)}}{d\alpha};$$

and in the case of  $i = 1$ , we have

$$\left( \frac{dA^{(i)}}{da} \right) = -\frac{1}{a'^2} \cdot \left( 1 - \frac{db_{\frac{1}{2}}^{(i)}}{d\alpha} \right).$$

Finally, even in the case of  $i = 1$ , we have,

$$\left( \frac{d^2 A^{(i)}}{da^2} \right) = -\frac{1}{a'^3} \cdot \frac{d^2 b_{\frac{1}{2}}^{(i)}}{d\alpha^2};$$

$$\left( \frac{d^3 A^{(i)}}{da^3} \right) = -\frac{1}{a'^4} \cdot \frac{d^3 b_{\frac{1}{2}}^{(i)}}{d\alpha^3};$$

&c.

In order to obtain the differences of  $A^{(i)}$  relative to  $a'$  it may be observed, that  $A^{(i)}$  being an homogeneous function of  $a$  and  $a'$ , of the dimension  $-1$ , we have by the nature of this kind of functions,

$$a \cdot \left( \frac{dA^{(i)}}{da} \right) + a' \cdot \left( \frac{dA^{(i)}}{da'} \right) = -A^{(i)},$$

hence we deduce

$$a' \cdot \left( \frac{dA^{(i)}}{da'} \right) = -A^{(i)} - a \cdot \left( \frac{dA^{(i)}}{da} \right);$$

$$a' \cdot \left( \frac{d^2 A^{(i)}}{da \cdot da'} \right) = -2 \cdot \left( \frac{dA^{(i)}}{da} \right) - a \cdot \left( \frac{d^2 A^{(i)}}{da^2} \right);$$

$$a'^2 \cdot \left( \frac{d^2 A^{(i)}}{da'^2} \right) = 2 \cdot A^{(i)} + 4a \cdot \left( \frac{dA^{(i)}}{da} \right) + a^2 \cdot \left( \frac{d^2 A^{(i)}}{da^2} \right);$$

$$a'^3 \cdot \left( \frac{d^3 A^{(i)}}{da \cdot da^2} \right) = 6 \cdot \left( \frac{dA^{(i)}}{da} \right) + 6a \cdot \left( \frac{d^2 A^{(i)}}{da^2} \right) + a^2 \cdot \left( \frac{d^3 A^{(i)}}{da^3} \right);$$

$$a'^3 \cdot \left( \frac{d^3 A^{(i)}}{da^3} \right) = -6 \cdot A^{(i)} - 18a \cdot \left( \frac{dA^{(i)}}{da} \right) - 9a^2 \cdot \left( \frac{d^2 A^{(i)}}{da^2} \right) - a^3 \cdot \left( \frac{d^3 A^{(i)}}{da^3} \right);$$

&c.

$B^{(i)}$  and its differences will be obtained by observing that by the preceding number, the series

$$\frac{1}{2} \cdot B^{(0)} + B^{(1)} \cdot \cos \theta + B^{(2)} \cdot \cos^2 \theta + \text{&c.}$$

is the expansion of the function  $a^{-\frac{3}{2}} \cdot (1 - 2a \cdot \cos \theta + a^2)^{-\frac{3}{2}}$  according to

the cosines of the angle  $\theta$  and of its multiples ; but this function thus expanded, is equal to

$$a'^{-3} \cdot \left( \frac{1}{2} \cdot b_{\frac{3}{2}}^{(0)} + b_{\frac{3}{2}}^{(1)} \cdot \cos. \theta + b_{\frac{3}{2}}^{(2)} \cdot \cos. 2\theta + \text{&c.} \right);$$

therefore we have generally

$$B^{(i)} = \frac{1}{a'^3} \cdot b_{\frac{3}{2}}^{(i)};$$

hence we obtain

$$\left( \frac{dB^{(i)}}{da} \right) = \frac{1}{a'^4} \cdot \frac{db_{\frac{3}{2}}^{(i)}}{da}; \quad \left( \frac{d^2 B^{(i)}}{da^2} \right) = \frac{1}{a'^5} \cdot \frac{d^2 b_{\frac{3}{2}}^{(i)}}{da^2}; \quad \text{&c.}$$

Moreover,  $B^{(i)}$  being an homogeneous function of  $a$  and of  $a'$ , of the dimension  $-3$ , we have

$$a \cdot \left\{ \frac{dB^{(i)}}{da} \right\} + a' \cdot \left\{ \frac{dB^{(i)}}{da'} \right\} = -3B^{(i)};$$

from which it is easy to infer the partial differences of  $B^{(i)}$  taken relatively to  $a'$ , by means of its partial differences relatively to  $a$ .

In the theory of the perturbations of  $m'$  by the action of  $m$ , the values of  $A^{(i)}$  and of  $B^{(i)}$  are the same as above, with the exception of  $A^{(1)}$ , which in this theory becomes  $\frac{a'}{a^2} - \frac{1}{a'} \cdot b_{\frac{1}{2}}^{(1)}$ . Thus the computation of the values of  $A^{(i)}$ ,  $B^{(i)}$ , and of their differences, serves at once for the theories of the two bodies  $m$  and  $m'$ .

50. After this digression on the expansion of  $R$  into a series, let us resume the differential equations ( $X'$ ), ( $Y'$ ) and ( $Z'$ ) of Nos. 46 and 47 ; and let us determine by their means, the values of  $\delta r$ ,  $\delta v$ , and  $\delta s$ , the approximation being extended to quantities of the order of the excentricities and of the inclinations of the orbits.

If in the elliptic orbits, we suppose

$$\begin{aligned} r &= a \cdot (1 + u); \quad r' = a' \cdot (1 + u'); \\ v &= nt + \epsilon + v; \quad v' = n't + \epsilon' + v'; \end{aligned}$$

by N°. 22 we shall have

$$\begin{aligned} u &= -e \cdot \cos.(nt + \epsilon - \varpi); \quad u' = -e' \cdot \cos.(n't + \epsilon' - \varpi') \\ v &= 2e \cdot \sin.(nt + \epsilon - \varpi); \quad v' = 2e' \cdot \sin.(n't + \epsilon' - \varpi'); \end{aligned}$$

$nt + \epsilon$ ,  $n't + \epsilon'$  being the mean longitudes of  $m$  and  $m'$ ;  $a$  and  $a'$  being the greater semiaxes of their orbits;  $e$  and  $e'$  being the ratios of the excentricities to the greater semiaxes; finally,  $\varpi$  and  $\varpi'$  being the longitudes of their perihelions. All these longitudes, may be referred indifferently to the planes themselves of the orbits, or to a plane which is very little inclined to them; because quantities of the order of the squares and products of the excentricities and of the inclinations are neglected. The preceding values being substituted, in the expression of  $R$  of N°. 48, will give

$$R = \frac{m'}{2} \cdot \Sigma A^{(i)} \cdot (\cos. i \cdot (n't - nt + \epsilon' - \epsilon))^*$$

\* As the approximation is carried only as far as terms involving the first power of the excentricity, the only terms in the general expression for  $R$  which are to be considered, are the four first. Now as  $A^{(i)} = A^{(-i)}$  and  $\left(\frac{dA^{(i)}}{da}\right) = \left(\frac{dA^{(-i)}}{da}\right)$ , and  $\cos. i.w = \cos. (-i.w)$   $w$  representing  $(n't - nt + \epsilon' - \epsilon)$ , and  $\sin. (-i.w) = -\sin. iw$ , we shall have generally ( $i$  representing the positive values of  $i$ , and  $n$  representing  $(nt + \epsilon - \varpi)$ ),  $\cos. i.w \cdot \cos. n = 2 \cos. i.w \cdot \cos. n = \cos. (i.w + n) + \cos. (i.w - n)$ , and  $i \cdot \sin. i.w \cdot \sin. n = 2i \cdot \sin. i.w \cdot \sin. n = +i \cdot \cos. ((i.w + n) - i \cdot \cos. (n - i.w))$ . See page 274, Notes). Hence substituting for  $n$ , its value,  $(nt + \epsilon - \varpi)$ , and observing that  $\cos. i \cdot (n't - nt + \epsilon' - \epsilon) \cdot \cos. (nt + \epsilon - \varpi)$

$$= \cos. i \cdot (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \varpi + \cos. i \cdot (n't - nt + \epsilon' - \epsilon) - (nt + \epsilon - \varpi),$$

and also that when  $2e \cdot \sin. (nt + \epsilon - \varpi)$  is substituted for  $v'$ ,  $\sin. i \cdot (n't - nt + \epsilon' - \epsilon)$ .

$$\sin. (nt + \epsilon - \varpi) = \cos. i \cdot (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \varpi - \cos.$$

$i \cdot (n't - nt + \epsilon' - \epsilon) - nt - \epsilon - \varpi$ , we obtain the second term in the expression; in like manner the third term is obtained, by taking the index  $i-1$ ; in the third term the circular part is

$$\begin{aligned}
 & -\frac{m'}{2} \cdot \Sigma \left\{ a \cdot \left\{ \frac{dA^{(i)}}{da} \right\} + 2i \cdot A^{(i)} \right\} \cdot e \cdot \cos(i \cdot (n't - nt + \varepsilon' - \varepsilon) + nt + \varepsilon - \omega) \\
 & -\frac{m'}{2} \cdot \Sigma \left\{ a' \cdot \left\{ \frac{dA^{(i-1)}}{da'} \right\} - 2 \cdot (i-1) \cdot A^{(i-1)} \right\} e' \cos(i \cdot (n't - nt + \varepsilon' - \varepsilon) + nt + \varepsilon - \omega');
 \end{aligned}$$

the sign  $\Sigma$  of finite integrals, extending to all integral values positive and negative of  $i$ , the value  $i=0$  being comprehended among them. From which we obtain

$$\begin{aligned}
 & 2 \cdot \int dR + r \cdot \left( \frac{dR}{dr} \right) = \\
 & 2m' \cdot g + \frac{m'}{2} a \cdot \left\{ \frac{dA^{(0)}}{da} \right\} + \frac{m'}{2} \cdot \Sigma \left\{ a \cdot \left\{ \frac{dA^{(i)}}{da} \right\} + \frac{2n}{n-n'} \cdot A^{(i)} \right\} \\
 & \quad \cos(i \cdot (n't - nt + \varepsilon' - \varepsilon)) \\
 & - \frac{m'}{2} \cdot \left\{ a^2 \cdot \left\{ \frac{d^2 A^{(0)}}{da^2} \right\} + 3a \cdot \left\{ \frac{dA^{(0)}}{da} \right\} \right\} \cdot e \cdot (\cos nt + \varepsilon - \omega) \\
 & - \frac{m'}{2} \cdot \left\{ aa' \cdot \left\{ \frac{d^2 A^{(1)}}{da \cdot da'} \right\} + 2a \cdot \left\{ \frac{dA^{(1)}}{da} \right\} + 2a' \cdot \left\{ \frac{dA^{(1)}}{da'} \right\} + 4A^{(1)} \right\} \\
 & \quad e' \cdot \cos(nt + \varepsilon - \omega') \\
 & - \frac{m'}{2} \cdot \Sigma \cdot \left\{ a^2 \cdot \left\{ \frac{d^2 A^{(i)}}{da^2} \right\} + (2i+1) \cdot a \cdot \left\{ \frac{dA^{(i)}}{da} \right\} \right. \\
 & \quad \left. + \frac{2 \cdot (i-1) \cdot n}{i \cdot (n-n') - n} \cdot \left\{ a \cdot \left\{ \frac{dA^{(i)}}{da} \right\} + 2i \cdot A^{(i)} \right\} \right\} \\
 & e \cdot \cos(i \cdot (n't - nt + \varepsilon' - \varepsilon) + nt + \varepsilon - \omega);
 \end{aligned}$$

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made to assume a more symmetrical form, for it becomes by performing the prescribed operations,  $\cos(i-1) \cdot (n't - nt + \varepsilon' - \varepsilon) + n't + \varepsilon - \omega)$ , which is evidently identical with the expression  $\cos i \cdot (n't - nt + \varepsilon' - \varepsilon) + nt + \varepsilon - \omega')$ , besides the values when  $i=0$ , are comprised in this expression.

$$-\frac{m'}{2} \cdot \Sigma \left\{ aa' \cdot \left\{ \frac{d^2 A^{(i-1)}}{da da'} \right\} - 2 \cdot (i-1) \bar{a} \cdot \left\{ \frac{dA^{(i-1)}}{da} \right\} \right. \\ \left. \frac{2 \cdot (i-1) \cdot n}{i \cdot (n-n')-n} \cdot \left\{ a' \cdot \left\{ \frac{dA^{(i-1)}}{da'} \right\} - 2 \cdot (i-1) A^{(i-1)} \right\} \right\}^* \\ e' \cdot \cos. (i \cdot (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \omega');$$

The sign  $\Sigma$  extending in this and the following formulæ to all the integral values of  $i$ , positive and negative, the sole value  $i = 0$  being

\* When the value of  $i = 0$ , is excepted out of the positive and negative values of  $i$ , we shall have

$$\frac{dR}{dr} = \frac{m'}{2} \cdot \frac{dA^{(0)}}{da} + \frac{m'}{2} \cdot \frac{d\Sigma A^{(i)}}{da} \cdot \cos. i \cdot (n't - nt + \epsilon' - \epsilon) - \frac{m'}{2} \cdot a \cdot \frac{d^2 A^{(0)}}{da^2} \cdot e \cdot \cos. (nt + \epsilon - \omega) \\ - \frac{m'}{2} \cdot \Sigma \left( a \cdot \frac{d^2 A^{(i)}}{da^2} + 2i \cdot \frac{dA^{(i)}}{da} \right) e \cdot \cos. i \cdot (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \omega \\ - \frac{m'}{2} \cdot \left( a' \cdot \frac{d^2 A^{(1)}}{da' da} + \frac{2 \cdot dA^{(1)}}{da} \right) e' \cdot \cos. (nt + \epsilon - \omega) - \frac{m'}{2} \cdot \Sigma a' \cdot \frac{d^2 A^{i-1}}{da' da} - 2(i-1). \\ \frac{dA^{(i-1)}}{da} \right) e' \cdot \cos. i \cdot (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \omega), \therefore \text{substituting for } r \text{ its value, } a \cdot (1-e).$$

$$\cos. (nt + \epsilon - \omega), \text{ we shall have } r \cdot \left( \frac{dR}{dr} \right) = \frac{m'}{2} \cdot a \cdot \frac{dA^{(0)}}{da} - \frac{m'}{2} \cdot \frac{dA^{(0)}}{da} \cdot \\ e \cdot \cos. (nt + \epsilon - \omega) + \frac{m'}{2} \cdot a \cdot \frac{d\Sigma A^{(i)}}{da} \cdot \cos. i \cdot (n't - nt + \epsilon' - \epsilon) - \frac{m'}{2} \cdot a \cdot \frac{d\Sigma A^{(i)}}{da} \cdot e \cdot \cos. i \\ (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \omega \\ - \frac{m'}{2} \left( \Sigma a^2 \cdot \frac{d^2 A}{da^2} + 2ia \cdot \frac{dA}{da} \right) \cdot e \cdot \cos. i \cdot (n't - nt' + \epsilon' - \epsilon) + nt + \epsilon - \omega \\ - \frac{m'}{2} \cdot a^2 \cdot \frac{d^2 A^{(0)}}{da^2} \cdot e \cdot \cos. (nt + \epsilon - \omega) \\ - \frac{m'}{2} \cdot a^2 a' \cdot \left( \frac{d^2 A^{(1)}}{da' da} + a \cdot \frac{2dA^{(1)}}{da} \right) \cdot e' \cdot \cos. (nt + \epsilon - \omega) \\ - \frac{m'}{2} \cdot \Sigma \left( a^2 a' \cdot \frac{dA^{(i-1)}}{da'} - 2(i-1) d \cdot \frac{A^{(i-1)}}{da} \right) e' \cdot \cos. ((n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \omega)$$

(the remaining terms are omitted because  $e^2$  occurs)

excepted, because the terms in which  $i = 0$ , are extricated from this sign:  $mg$  is a constant quantity added to the integral  $\int dR$ . Therefore by making

$$C = \frac{1}{2}a^3 \cdot \left\{ \frac{d^2 A^{(0)}}{da^2} \right\} + 3a^2 \cdot \left\{ \frac{dA^{(0)}}{da} \right\} + 6ag;$$

$$D = \frac{1}{2}a^2 a' \cdot \left\{ \frac{d^2 A^{(1)}}{da da'} \right\} + a^2 \cdot \left\{ \frac{dA^{(1)}}{da} \right\} + aa' \cdot \left\{ \frac{dA^{(1)}}{da'} \right\} + 2aA^{(1)};$$

$$\begin{aligned} C^{(i)} &= \frac{1}{2}a^3 \cdot \left\{ \frac{d^2 A^{(i)}}{da^2} \right\} + \frac{(2i+1)}{2} \cdot a^2 \cdot \left\{ \frac{dA^{(i)}}{da} \right\} \\ &\quad + \frac{(i \cdot (n-n') - 3n)}{2 \cdot (i \cdot (n-n') - n)} \cdot a^2 \cdot \left\{ \frac{dA^{(i)}}{da} \right\} + \frac{2n}{n-n'} \cdot aA^{(i)} \\ &\quad + \frac{(i-1) \cdot n}{i \cdot (n-n') - n} \cdot \left\{ a^2 \cdot \left\{ \frac{dA^{(i)}}{da} \right\} + 2i \cdot aA^{(i)} \right\}; \end{aligned}$$

$$D^{(i)} = \frac{1}{2}a^2 a' \cdot \left\{ \frac{d^2 A^{(i-1)}}{da da'} \right\} - (i-1) \cdot a^2 \cdot \left\{ \frac{dA^{(i-1)}}{da} \right\}$$

$$\begin{aligned} dR &= + \frac{m'}{2} \cdot indt \cdot \Sigma A^{(i)} \cdot \sin. i \cdot (n't - nt + \epsilon' - \epsilon) + \frac{m'}{2} \cdot ndta \cdot \frac{dA^{(0)}}{da} \cdot e. \sin. (nt + \epsilon - \omega) \\ &\quad + \frac{m'}{2} (n-in) \cdot dt \cdot \Sigma \left( a \cdot \left( \frac{dA^{(i)}}{da} \right) + 2iA^{(i)} \right) e. \sin. i \cdot (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \omega \\ &\quad + \frac{m'}{2} \cdot ndt \cdot \left( \Sigma a' \cdot \frac{dA^{(1)}}{da'} + 2A^{(1)} \right) e' \cdot \sin. (nt + \epsilon - \omega') + \frac{m'}{2} \cdot (1-i) \cdot n \cdot dt \cdot (\Sigma a' \cdot \frac{dA^{(i-1)}}{da'}) \\ &\quad - 2 \cdot (i-1) \cdot A^{(i-1)} \cdot e' \cdot \sin. i \cdot (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \omega', \therefore 2 \int dR \\ &= 2m'g - \frac{m'}{2} \cdot \frac{2n}{n'-n} \cdot \Sigma A^{(i)} \cdot \cos. i \cdot (n't - nt + \epsilon' - \epsilon) - \frac{m'}{2} \cdot 2a \cdot \frac{dA^0}{da} \cdot e. \cos. (nt + \epsilon - \omega) - \frac{m'}{2} \cdot 2a' \\ &\quad \cdot \frac{2(n-in)}{in'-in+n} \cdot \left( \Sigma a \cdot \frac{dA^{(i)}}{da} + 2iA^{(i)} \right) e. \cos. i \cdot (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \omega) - \frac{m'}{2} \cdot 2a' \cdot \\ &\quad \left( \frac{dA^{(1)}}{da'} + 4A^{(1)} \right) \cdot e' \cdot \cos. (nt + \epsilon - \omega) - \frac{m'}{2} \cdot \frac{2(i-1)n}{i(n'-n)+n} \cdot \left( \Sigma a' \cdot \frac{dA^{(i-1)}}{da'} - 2(i-1) \cdot A^{(i-1)} \right) \\ &\quad e' \cdot \cos. i \cdot (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \omega), \therefore \text{by reducing we obtain } 2 \int dR + r \cdot \frac{dR}{dr} = \text{the expression which is given in the text.} \end{aligned}$$

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$$+ \frac{(i-1).n}{i.(n-n')-n} \cdot \left\{ aa' \cdot \left\{ \frac{dA^{(i-1)}}{da'} \right\} - 2.(i-1).aA^{(i-1)} \right\};$$

the sum of the masses  $M+m$  being assumed equal to unity, and  $\frac{M+m}{a^3}$  being supposed equal to  $n^2$ ; the equation (X') will become

$$0 = \frac{d^2.\delta u}{dt^2} + n^2.\delta u - 2n^2.m'ag - \frac{n^2m'}{2} \cdot a^2 \cdot \left\{ \frac{dA^{(0)}}{da} \right\} *$$

\* The equation (X') becomes by neglecting the square of the excentricity,  $\frac{d^2.\delta u}{dt^2} + n^2.\delta u - n^2a.(1-e \cdot \cos.(nt+\epsilon-\omega)) \cdot \left( 2f dR + r \cdot \frac{dR}{dr} \right) - 2ean^2 \int ndt \cdot (\sin.(nt+\epsilon-\omega)) \cdot \left( 2f dR + r \cdot \frac{dR}{dr} \right)$ ; ( $n^2a$  being substituted for  $\frac{1}{a^2}$  and  $M+m$  being by hypothesis = 1).

By substituting for  $2f dR + r \frac{dR}{dr}$ , its value, this equation becomes =

$$\begin{aligned} & \frac{d^2.\delta u}{dt^2} + n^2.\delta u - 2n^2.m'ag - \frac{n^2m'}{2} \cdot a^2 \cdot \left( \frac{dA^{(0)}}{da} \right) - \frac{n^2m'}{2} \cdot \left( \Sigma a^2 \cdot \frac{dA^{(i)}}{da} + \frac{2n}{n-n'} \cdot aA^{(i)} \right) \cos \\ & i.(n't-nt+\epsilon'-\epsilon) + \frac{n^2m'}{2} \cdot \left( a^2 \cdot \frac{d^2A^{(0)}}{da^2} + 3a^2 \cdot \frac{dA^{(0)}}{da} \right) \cdot e \cdot \cos.(nt+\epsilon-\omega) + \\ & \left( (n^2a) \cdot (2m'g + \frac{m'}{2}) \cdot a \cdot \frac{dA^{(0)}}{da} + 2an^2 \cdot (2m'g + \frac{m'}{2}) \cdot a \cdot \frac{dA^{(0)}}{da} \right) \cdot e \cdot \cos.(nt+\epsilon-\omega); (= n^2m'.Ce. \\ & \cos.(nt+\epsilon-\omega)) + n^2a \cdot \frac{m'}{2} \left( aa' \cdot \frac{d^2A^{(1)}}{da da'} + 2a \cdot \frac{dA^{(1)}}{da} + 2a' \cdot \frac{dA^{(1)}}{da'} + 4A^{(1)} \right) e' \cdot (\cos. nt + \\ & - \omega') = (n^2m'.De' \cdot \cos.(nt+\epsilon-\omega')). \end{aligned}$$

$$\begin{aligned} & + \frac{m'}{2} \cdot n^2a \cdot \Sigma \left\{ \begin{aligned} & a^2 \cdot \frac{d^2A^{(i)}}{da^2} + (2i+1)a \cdot \frac{dA^{(i)}}{da} \\ & \frac{2.(i-1).n}{i.(n-n')-n} \left( a \cdot \frac{dA^{(i)}}{da} + 2iA^{(i)} \right) \end{aligned} \right\} e \cdot \cos. (i.(n't-nt+\epsilon'-\epsilon) + nt+\epsilon-\omega) \\ & + \frac{m'}{2} \cdot n^2a \cdot \left( \Sigma a \cdot \frac{dA^{(i)}}{da} + \frac{2n}{n-n'} \cdot A^{(i)} \right) e \cdot \cos. i.(n't-nt+\epsilon'-\epsilon) + nt+\epsilon-\omega \\ & - 2an^2 \cdot e \cdot \int ndt \cdot (\sin. i.(n't-nt+\epsilon'-\epsilon) + nt+\epsilon-\omega) \left( \frac{m'}{2} \cdot \Sigma a \cdot \frac{dA^{(i)}}{da} + \frac{2n}{n-n'} A^{(i)} \right) = \left( \frac{m'}{2} \cdot \right. \end{aligned}$$

$$\begin{aligned}
& -\frac{n^2 m'}{2} \cdot \Sigma. \left\{ a^2 \cdot \left( \frac{dA^{(i)}}{da} \right) + \frac{2n}{n-n'} \cdot a A^{(i)} \right\} \cdot \cos. i. (n't - nt + \epsilon' - \epsilon) \\
& + n^2 m'. C.e. \cos. (nt + \epsilon - \omega) + n^2 m'. De' \cos. (nt + \epsilon - \omega') \\
& + n^2 m'. \Sigma. C^{(i)}. e. \cos. (i. (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \omega) \\
& + n^2 m'. \Sigma. D^{(i)}. e' \cos. (i. (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \omega');
\end{aligned}$$

and by integrating

$$\begin{aligned}
\delta u = & 2m'ag + \frac{m'^2}{2} \cdot a^2 \cdot \left( \frac{dA^{(0)}}{da} \right) \\
& - \frac{m'}{2} \cdot n^2 \cdot \Sigma. \frac{\left\{ a^2 \cdot \left( \frac{dA^{(i)}}{da} \right) + \frac{2n}{n-n'} \cdot a A^{(i)} \right\}}{i^2 \cdot (n-n')^2 - n^2} \cdot \cos. i. (n't - nt + \epsilon' - \epsilon) \\
& + m'f_e \cos. (nt + \epsilon - \omega) + m'f'_e \cdot e' \cos. (nt + \epsilon - \omega') \\
& - \frac{m'}{2} \cdot C.n.t. e. \sin. (nt + \epsilon - \omega) - \frac{m}{2} \cdot D.n.t. e' \sin. (nt + \epsilon - \omega') \\
& + m'. \Sigma. \frac{C^{(i)}. n^2}{(i. (n-n') - n)^2 - n^2} \cdot e. \cos. (i. (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \omega) \\
& + m'. \Sigma. \frac{D^{(i)}. n^2}{(i. (n-n') - n)^2 - n^2} \cdot e' \cos. (i. (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \omega),
\end{aligned}$$

$$2an^2 \cdot \left( \frac{n}{i.(n'-n)+n} \right) \cdot \Sigma a \cdot \frac{dA^{(i)}}{da} + \left( \frac{2n}{n-n'} \right) \cdot A^{(i)} \cdot e. \cos. i. (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \omega)$$

which added to the preceding term becomes, by changing the signs of the numerator and denominator of  $\frac{2n}{i.(n'-n)+n}$

$$\frac{m'}{2} \left( n^2 \cdot \frac{i.(n-n') - 3n}{i.(n-n') - n} \cdot \Sigma a^2 \cdot \frac{dA^{(i)}}{da} + \frac{4na}{n-n'} \cdot A^{(i)} \right) \cdot e. \cos. (i. (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \omega)$$

and by adding this quantity to  $\frac{m'}{2} \cdot n^2 a \cdot \left( \Sigma a^2 \cdot \frac{d^2 A^{(i)}}{da^2} + (2i+1) a \cdot \frac{dA^{(i)}}{da} \right)$ , it will appear that the coefficient of  $n^2 m'. e. \cos. (i. (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \omega)$  is equal to  $C^{(i)}$ ; it is evident from an inspection of the coefficients of  $e^2. \cos. (nt + \epsilon - \omega')$ ,  $e'. \cos. i. (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \omega'$  that they are respectively equal to  $D$ , and  $D^{(i)}$ .

$f$ , and  $f'$ , being two arbitrary quantities.

The expression for  $\delta r$  in  $\delta u$  which has been found in N°. 47, will give

$$\begin{aligned} \frac{\delta r}{a} = & -2m'.ag - \frac{m'}{2} \cdot a^2 \cdot \left( \frac{dA^{(0)}}{da} \right) \\ & + \frac{m'}{2} \cdot n^2 \Sigma \cdot \left\{ \frac{a^2 \cdot \left\{ \frac{dA^{(i)}}{da} \right\} + \frac{2n}{n-n'} \cdot aA^{(i)}}{i^2 \cdot (n-n')^2 - n^2} \right\} \cdot \cos. i. (n't - nt + \varepsilon' - \varepsilon) \\ & - m' \cdot fe \cdot \cos. (nt + \varepsilon - \omega) - m' f' e' \cdot \cos. (nt + \varepsilon - \omega') \\ & + \frac{1}{2} \cdot m' C \cdot nt \cdot e \cdot \sin. (nt + \varepsilon - \omega) + \frac{1}{2} \cdot m' D \cdot nt e' \cdot \sin. (nt + \varepsilon - \omega') \\ & + m' \cdot n^2 \Sigma \cdot \left\{ \left\{ \frac{a^2 \cdot \left\{ \frac{dA^{(i)}}{da} \right\} + \frac{2n}{n-n'} \cdot aA^{(i)}}{i^2 \cdot (n-n')^2 - n^2} - \frac{C^{(i)}}{(i \cdot (n-n') - n)^2 - n^2} \right\} \right\} \\ & (e \cdot \cos. (i \cdot (n't - nt + \varepsilon' - \varepsilon) + nt + \varepsilon - \omega)) \\ & - m' \cdot n^2 \cdot \Sigma \cdot \frac{D^{(i)}}{i \cdot (n-n') - n} \cdot e' \cdot \cos. (i \cdot (n't - nt + \varepsilon' - \varepsilon) + nt + \varepsilon - \omega); \end{aligned}$$

$f$  and  $f'$  being two arbitrary quantities depending on  $f$  and  $f'$ .

This value of  $\delta r$ , substituted in the formula (Y) of N°. 46, will give  $\delta v$ , or the perturbations of the motion of the planet in longitude; but it may be observed, that  $nt$  expressing the mean motion of  $m$ , the term proportional to the time must disappear from the expression of  $\delta v$ . This condition determines the constant quantity  $g$ , and we find

$$g = -\frac{1}{3}a \cdot \left\{ \frac{dA^{(0)}}{da} \right\}.*$$

The introduction of the arbitrary quantities  $f$  and  $f'$ , might have

\* As  $nt$  must vanish from this expression, by substituting for  $dR$  and  $\delta r \cdot \frac{dR}{dr}$  their values in the expression for  $\delta v$  given in page 263, and then integrating, the terms involving  $nt$  are  $3am'gnt$  and  $2 \frac{m'}{2} \cdot a^2 nt \cdot \left( \frac{dA^{(0)}}{da} \right)$ , hence we will have  $3m'g = -\frac{2m'}{2} \cdot a \cdot \frac{dA^{(0)}}{da}$ .

been dispensed with, by supposing them to be comprised in the elements  $e$  and  $\varpi$  of elliptic motion; but then the expression for  $\delta r$ , would have involved terms depending on the mean anomaly, which would not have been included in those which are given by the elliptic motion: now it is more convenient to make those terms to disappear from the expression for the longitude, in order to introduce them into the expression for the radius vector;  $f$ , and  $f'$ , will be so determined as to satisfy this condition. This being premised, by substituting in place of

$a' \cdot \left\{ \frac{dA^{(i-1)}}{da'} \right\}$  its value  $-A^{(i-1)} - a \cdot \left\{ \frac{dA^{(i-1)}}{da} \right\}$ , we shall obtain

$$C = a^2 \cdot \left( \frac{dA^{(0)}}{da} \right) + \frac{1}{2} \cdot a^3 \cdot \left( \frac{d^2 A^{(0)}}{da^2} \right);$$

$$D = aA^{(1)} - a^2 \cdot \left( \frac{dA^{(1)}}{da} \right) - \frac{1}{3} a^3 \cdot \left( \frac{d^2 A^{(1)}}{da^2} \right);$$

$$D^{(i)} = \frac{(i-1) \cdot (2i-1) \cdot n}{n-i \cdot (n-n')} \cdot aA^{(i-1)} + \frac{(i^2 \cdot (n-n') - n)}{n-i \cdot (n-n')} \cdot a^2 \cdot \left( \frac{dA^{(i-1)}}{da} \right) \\ - \frac{1}{2} a^3 \cdot \left( \frac{d^2 A^{(i-1)}}{da^2} \right);$$

$$f = \frac{2}{3} a^2 \cdot \left( \frac{dA^{(0)}}{da} \right) + \frac{1}{4} a^3 \cdot \left( \frac{d^2 A^{(0)}}{da^2} \right)$$

$$f' = \frac{1}{4} \cdot \left\{ a \cdot A^{(1)} - a^2 \cdot \left( \frac{dA^{(1)}}{da} \right) + a^3 \cdot \left( \frac{d^2 A^{(1)}}{da^2} \right) \right\};$$

moreover let  $E^{(i)} =$

$$= \frac{3n}{n-n'} \cdot aA^{(i)} + \frac{i^2 \cdot (n-n') \cdot (n+i \cdot (n-n')) - 3n^2}{i^2 \cdot (n-n')^2 - n^2}$$

$$\times \left\{ a^2 \cdot \left( \frac{dA^{(i)}}{da} \right) + \frac{2n}{n-n'} \cdot aA^{(i)} \right\} + \frac{1}{2} a^3 \cdot \left( \frac{d^2 A^{(i)}}{da^2} \right);$$

$$\begin{aligned}
 F^{(i)} &= \frac{(i-1).n}{n-n'} \cdot aA^{(i)} + \frac{in}{2} \cdot (n+i(n-n')-3n^2) \\
 &\quad - \frac{i^2 \cdot (n-n')^2 - n^2}{i^2 \cdot (n-n')^2 - n^2} \\
 &\times \left\{ a^2 \cdot \left( \frac{dA^{(i)}}{da} \right) + \frac{2n}{n-n'} \cdot aA^{(i)} \right\} - \frac{2n^2 E^{(i)}}{n^2 - (n-i(n-n'))^2}; \\
 G^{(i)} &= \frac{(i-1).(2i-1).na.A^{(i-1)} + (i-1).na^2 \cdot \left( \frac{dA^{(i-1)}}{da} \right)}{2.(n-i(n-n'))} \\
 &- \frac{2n^2.D^{(i)}}{n^2 - (n-i(n-n'))^2};
 \end{aligned}$$

we shall have

$$\begin{aligned}
 \frac{\delta r}{a} &= \frac{m'}{6} \cdot a^2 \cdot \left( \frac{dA^{(0)}}{da} \right) + \frac{m' \cdot n^2}{2} \cdot \Sigma \left\{ a \cdot \left( \frac{dA^{(i)}}{da} \right) + \frac{2n}{n-n'} \cdot \right. \\
 &\quad \left. aA^{(i)} \right\} \cdot \cos. i.(n't - nt + \epsilon' - \epsilon) \\
 &\quad - m'.fe \cdot \cos. (nt + \epsilon - \omega) - m'.f'e' \cdot \cos. (nt + \epsilon - \omega') \\
 &\quad + \frac{1}{2}m'.C.nt.e \cdot \sin. (nt + \epsilon - \omega) + \frac{1}{2}m'.D.nt.e' \cdot \sin. (nt + \epsilon - \omega') \\
 &\quad + n^2.m'.\Sigma \left\{ \begin{array}{l} \frac{*E^{(i)}}{n^2 - (n-i(n-n'))^2} \cdot e \cdot \cos. (i.(n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \omega) \\ + \frac{D^{(i)}}{n^2 - (n-i(n-n'))^2} \cdot e' \cdot \cos. (i.(n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \omega') \end{array} \right\}; \\
 \delta v &= \frac{m'}{2} \cdot \Sigma \left\{ \frac{n^2}{i.(n-n')^2} \cdot a.A^{(i)} + \frac{2n^3 \cdot \left\{ a^2 \cdot \left\{ \frac{dA^{(i)}}{da} \right\} + \frac{2n}{n-n'} \cdot aA^{(i)} \right\}}{i.(n-n').(i^2 \cdot (n-n')^2 - n^2)} \right\} \cdot \\
 &\quad \sin. i.(n't - nt + \epsilon' - \epsilon), \\
 &\quad + m'.C.nt.e \cdot \cos. (nt + \epsilon - \omega) + m'D.nt.e' \cdot \cos. (nt + \epsilon - \omega')
 \end{aligned}$$

\* If to the value of  $C^{(i)}$  be added the terms  $\pm \left( \frac{2i+1}{2} \right) \cdot \frac{2n}{n-n'} \cdot aA^{(i)} \pm$

$$+nm'.\Sigma \left\{ \begin{array}{l} \frac{E^{(i)}}{n-i.(n-n')} \cdot e. \sin. (i.(n't-nt+\varepsilon'-\varepsilon) + nt + \varepsilon - \varpi) \\ + \frac{G^{(i)}}{n-i.(n-n')} \cdot e'. \sin. (i.(n't-nt+\varepsilon'-\varepsilon) + n + \varepsilon - \varpi') \end{array} \right\};$$

in these expressions the integral sign  $\Sigma$  extends to the whole values of  $i$  both positive and negative, the sole value  $i = 0$  being excepted.

It may be observed here, that in the very ease in which the series represented by  $\Sigma.A^{(i)}.\cos. i.(n't-nt+\varepsilon'-\varepsilon)$  converges slowly, the expressions of  $\frac{\delta r}{a}$ , and of  $\delta v$ , may be rendered converging by means of the divisors which they acquire. This observation is extremely important,

## Q Q 2

$$\begin{aligned} \frac{(i-1).n}{i.(n-n')-n} \cdot \frac{2n}{n-n'} \cdot aA^{(i)} \text{ it will become } &= \frac{1}{2} \cdot \frac{d^2 A^{(i)}}{da^2} + \frac{2i+1}{2} \left( a^2 \cdot \frac{dA^{(i)}}{da} \right. \\ &\left. + \frac{2n}{n-n'} \cdot aA^{(i)} \right) - \left( \frac{2i+1}{2} \right) \frac{2n}{n-n'} \cdot aA^{(i)} + \frac{i.(n-n')-3n}{2.i.(n-n')-n}. \end{aligned}$$

$$\left( a^2 \cdot \left( \frac{dA^{(i)}}{da} \right) + \frac{2n}{n-n'} \cdot aA^{(i)} \right) + \frac{(i-1).n}{i.(n-n')-n} \cdot \left( a^2 \cdot \frac{dA^{(i)}}{da} + \frac{2n}{n-n'} \cdot aA^{(i)} \right) - \frac{2.(i-1)n}{n-n'}$$

$aA^{(i)}$ ; now by reducing the two terms which constitute the factor of  $e. \cos. i.(n't-nt+\varepsilon'-\varepsilon) + nt + \varepsilon - \varpi$  in page 296, to a common denominator, it will become  $=$  to

$$(2i^2.(n-n')^2 - 4in.(n-n') - 2i.(i^2.(n-n')^2 + 2i.n^2 - i^2.(n-n')^2 + n^2 - i^2.(n-n')^2 + 2in.(n-n')) + 3n^2 - 2in.(i.(n-n') + 2in.(n-n') - 2in^2 + 2n^2)). \left( a^2 \cdot \frac{dA^{(i)}}{da} + \frac{2n}{n-n'} \cdot aA^{(i)} \right) \text{ (divided by}$$

$$2i^2((n-n')^2 - n^2)) + \frac{4in + 2n - 4i.n + 4n}{2(n-n')} \cdot aA^{(i)} = -2i^2.n - n'.i.(n-n') + n + 6n^2$$

$\left( a^2 \cdot \frac{dA^{(i)}}{da} + \frac{2n}{n-n'} \cdot aA^{(i)} \right)$ . divided by  $2i^2.(n-n')^2 - n^2 + \frac{6n}{2.(n-n')} \cdot aA^{(i)}$ ; which is evidently equal to  $E^{(i)}$ .

because without it it would be impossible to express analytically the reciprocal perturbations of the planets, the ratio of whose distances from the sun, differ little from unity.

These expressions may be made to assume the following form, which will be extremely useful in the sequel ; let

$$h = e \cdot \sin. \varpi; \quad h' = e' \sin. \varpi';$$

$$l = e \cdot \cos. \varpi; \quad l' = e' \cos. \varpi';$$

we shall have

$$\begin{aligned} \frac{\delta r}{a} &= \frac{m'}{6} \cdot a^2 \cdot \left( \frac{dA^{(i)}}{da} \right) + \frac{m'n^2}{2} \cdot \Sigma. \left\{ \frac{a^2 \cdot \left( \frac{dA^{(i)}}{da} \right) + \frac{2n}{n-n'} \cdot aA^{(i)}}{i^2 \cdot (n-n')^2 - n^2} \right\} \cdot \\ &\quad \cos. i.(n't - nt + \varepsilon' - \varepsilon) \\ &- m' \cdot (hf + h'f') \cdot \cos. nt + \varepsilon - m' \cdot (lf + l'f') \cdot \sin. (nt + \varepsilon) \\ &+ \frac{m'}{2} \cdot (l.C + l'D) \cdot nt \cdot (\sin. (nt + \varepsilon) - \frac{m'}{2} \cdot (h.C + h'D) \cdot nt \cdot \cos. (nt + \varepsilon)) \\ &+ n^2 m' \cdot \Sigma. \left\{ \begin{array}{l} \frac{hE^{(i)} + h' \cdot D^{(i)}}{n^2 - (n - i \cdot (n - n'))^2} \cdot \sin. (i \cdot (n't - nt + \varepsilon' - \varepsilon) + nt + \varepsilon) \\ + \frac{(l.E^{(i)} + l' \cdot D^{(i)})}{n^2 - (n' - (n - n'))^2} \cdot \cos. (i \cdot (n't - nt + \varepsilon' - \varepsilon) + nt + \varepsilon) \end{array} \right\}; \\ \delta v &= \frac{m'}{2} \cdot \Sigma. \left\{ \frac{n^2}{i \cdot (n - n')^2} \cdot aA^{(i)} + 2n^3 \left\{ \frac{a^2 \cdot \left( \frac{dA^{(i)}}{da} \right) + \frac{2n}{n-n'} \cdot aA^{(i)}}{i \cdot (n - n') \cdot (i^2 \cdot (n - n')^2 - n^2)} \right\} \right\} \cdot \\ &\quad \sin. i.(n't - nt + \varepsilon' - \varepsilon). \\ &+ m' \cdot (h.C + h'.D) \cdot nt \cdot \sin. (nt + \varepsilon) + m' \cdot (l.C + l'.D) \cdot nt \cdot \cos. (nt + \varepsilon) \\ &+ n.m' \cdot \Sigma. \left\{ \begin{array}{l} \frac{l.F^{(i)} + l' \cdot G^{(i)}}{n - i \cdot (n - n')} \cdot \sin. (i \cdot (n't - nt + \varepsilon' - \varepsilon) + nt + \varepsilon) \\ - \frac{(h.F^{(i)} + h' \cdot G^{(i)})}{n - i \cdot (n - n')} \cdot \cos. (i \cdot (n't - nt + \varepsilon' - \varepsilon) + nt + \varepsilon) \end{array} \right\}; \end{aligned}$$

these expressions of  $\delta r$  and  $\delta v$  being added to the values of  $r$  and of  $v$ , relative to the elliptic motion, will give the entire values of the radius vector of  $m$ , and of its motion in longitude.

51. Let us at present, consider the motion of  $m$ , in latitude. For this purpose let the formula (Z') of N°. 47, be resumed; and if the product of the inclinations, by the excentricities of the orbits, be neglected it becomes

$$0 = \frac{d^2 \delta u'}{dt^2} + n^2 \cdot \delta u' - \frac{1}{a^2} \cdot \left( \frac{dR}{dz} \right); *$$

the expression for  $R$  of N°. 48, gives, by assuming for the fixed plane, the plane of the primitive orbit of  $m$ ,

$$\left( \frac{dR}{dz} \right) = \frac{m' z'}{a'^3} - \frac{m' z'}{2} \cdot \Sigma. B^{(i)} \cdot \cos. i. (n't - nt + \epsilon' - \epsilon);$$

the value of  $i$  comprehending all whole numbers both positive and negative, including  $i = 0$ . Let  $\gamma$  represent the tangent of the inclination of the orbit of  $m'$ , to the primitive orbit of  $m$ , and  $\Pi$  the longitude of the ascending node of the first of these orbits, on the second; we shall have very nearly,

$$z' = a' \cdot \gamma \cdot \sin. (n't + \epsilon' - \Pi); \dagger$$

which gives

$$\left( \frac{dR}{dz} \right) = \frac{m'}{a'^2} \cdot \gamma \cdot \sin. (n't + \epsilon' - \Pi) - \frac{m'}{2} \cdot a' \cdot B^{(i)} \cdot \gamma \cdot \sin. (nt + \epsilon - \Pi)$$

\* When the primitive orbit of  $m$  is assumed as the fixed plane, the differential of the two last terms in the value of  $R$  (which is given in page 276) with respect to  $z$ , becomes (when quantities of the order  $m'^2$  are neglected) the expression which is given in the text.

† When quantities of the higher orders of the inclinations are neglected, we may substitute for  $\sin. (n't + \epsilon' - \Pi)$ , the longitude on the fixed plane, and we can also assume the distance of the planet from the centre of its orbit, equal to the mean distance  $a'$ ; under these restrictions it will readily appear that the tangent of latitude of  $m'$  above the fixed plane  $= \gamma \cdot \sin. (n't + \epsilon' - \Pi)$ , and  $\because z' = a' \cdot \gamma \cdot \sin. (n't + \epsilon' - \Pi)$ .

$$-\frac{m'}{2} \cdot a' \cdot \Sigma B^{(i-1)} \cdot \gamma \cdot \sin. (i \cdot (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \Pi),$$

the value of  $i$ , in this and the following expressions extending to all whole numbers, as well positive as negative, the sole value  $i=0^*$  being excepted. The differential equation in  $\delta u'$ , will consequently become, by multiplying the value of  $\left(\frac{dR}{dz}\right)$ , by  $n^2 a^3$ . which is equal to unity,

$$\begin{aligned} 0 = & \frac{d^2 \cdot \delta u'}{dt^2} + n^2 \cdot \delta u' - m' \cdot n \cdot \frac{a}{a'^2} \cdot \gamma \cdot \sin. (n't + \epsilon' - \Pi) \\ & + \frac{m'n^2}{2} \cdot aa' \cdot B^{(1)} \cdot \gamma \cdot \sin. (nt + \epsilon - \Pi) \\ & + \frac{m'n^2}{2} \cdot aa' \cdot \Sigma B^{(i-1)} \cdot \gamma \cdot \sin. (i \cdot (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \Pi); \end{aligned}$$

from which we obtain, by integrating, and by remarking that by N°. 47,  $\delta s = -a \cdot \delta u'$ ,

$$\begin{aligned} \delta s = & -\frac{m' \cdot n^2}{n^2 - n'^2} \cdot \frac{a^2}{a'^2} \cdot \gamma \cdot \sin. (n't + \epsilon' - \Pi) \\ & - \frac{m' \cdot a^2 \cdot a'}{4} \cdot B^{(1)} \cdot nt \cdot \gamma \cdot \cos. (nt + \epsilon - \Pi) \dagger \end{aligned}$$

\* When this value of  $z'$  is multiplied into  $\Sigma B^{(i)} \cdot \cos. i \cdot (n't - nt + \epsilon' - \epsilon)$ , it becomes, when  $i=1$ , equal to  $B^{(1)} \cdot \sin. (n't - nt + \epsilon' - \epsilon) + n't + \epsilon - \Pi + B^{(1)} \cdot \sin. (nt + \epsilon - \Pi)$ , and when  $i=0$  it becomes  $= B^{(0)} \cdot \sin. (n't + \epsilon - \Pi)$ ; now had this product been expressed generally  $= \frac{m'}{2} \cdot a' \cdot \Sigma B^{(i)} \cdot \gamma \cdot \sin. (i \cdot (n't - nt + \epsilon' - \epsilon) + nt + \epsilon - \Pi)$ , it would not answer to the two cases in which  $i=1$ , and in which  $i=0$ ; hence we see the reason why this product is resolved into parts in the expression for  $\left(\frac{dR}{dz}\right)$ , and also why the value  $i=0$ , is excepted out of the values of  $i$ .

† This differential equation is integrated in the manner prescribed in N°. 41.

$$+ \frac{m' \cdot n^2 \cdot a^2 \cdot a'}{2} \cdot \Sigma. \frac{B^{(i-1)}}{n^2 - (n - i \cdot (n - n'))^2} \cdot \gamma \cdot \sin.(i \cdot (n't - nt + \varepsilon' - \varepsilon) + nt + \varepsilon - \Pi).$$

In order to obtain the latitude of  $m$ , above a fixed plane, a little inclined to the plane of its primitive orbit, naming  $\varphi$  the inclination of this orbit on the fixed plane, and  $\theta$  the longitude of its ascending node on the same plane, it will be sufficient to add to  $\delta s$ , the quantity  $\tan. \varphi. \sin. (\nu - \theta)$ , or  $\tan. \varphi. \sin. (nt + \varepsilon - \theta)$ , the eccentricity of the orbit being neglected.\* Let  $\varphi'$  and  $\theta'$  represent what  $\varphi$  and  $\theta$  become relatively to  $m'$ . If  $m$  moved in the primitive orbit of  $m'$ , the tangent of latitude will be  $\tan. \varphi'. \sin. (nt + \varepsilon - \theta')$ ; it will be  $\tan. \varphi. \sin. (nt + \varepsilon - \theta)$ , if  $m$  continued to move on its primitive orbit. The difference of these two tangents is very nearly the tangent of the latitude of  $m$ , above the plane of the primitive orbit, it being supposed to move on the plane of the primitive orbit of  $m'$ ; therefore we have

$$\tan. \varphi'. \sin. (nt + \varepsilon - \theta') - \tan. \varphi. \sin. (nt + \varepsilon - \theta) = \gamma. \sin. (nt + \varepsilon - \Pi).$$

Let

$$\begin{aligned} \tan. \varphi. \sin. \theta &= p; & \tan. \varphi'. \sin. \theta' &= p'; \\ \tan. \varphi. \cos. \theta &= q; & \tan. \varphi'. \cos. \theta' &= q'; \end{aligned}$$

we shall obtain

$$\gamma. \sin. \Pi = p' - p; \quad \gamma. \cos. \Pi = q' - q;$$

and consequently, if  $s$  denote the latitude of  $m$  above the fixed plane, we shall have very nearly,

$$\begin{aligned} s &= q. \sin. (nt + \varepsilon) - p. \cos. (nt + \varepsilon) \\ &- \frac{m' \cdot a^2 \cdot a'}{4} \cdot (p' - p) \cdot B^{(1)} \cdot nt. \sin. (nt + \varepsilon) \end{aligned}$$

\* This expression for the latitude of  $m$  above the fixed plane, which is a *little* inclined to the plane of its primitive orbit, is true when quantities of the higher orders are neglected.

$$\begin{aligned}
 & -\frac{m'.a^2 a'}{4} \cdot (q' - q) \cdot B^{(1)} \cdot n t \cdot \cos.(n t + \epsilon) \\
 & - \frac{m' n^2}{n^2 - n'^2} \cdot \frac{a^2}{a'^2} \cdot ((q' - q) \cdot \sin.(n' t + \epsilon') - (p' - p) \cdot \cos.(n' t + \epsilon')) \\
 & + \frac{m' \cdot n^2 \cdot a^2 \cdot a'}{2} \cdot \Sigma \cdot \left\{ \begin{array}{l} \frac{(q' - q) \cdot B^{(i-1)}}{n^2 - (n - i \cdot (n - n'))^2} \cdot \sin.(i \cdot (n' t - n t + \epsilon' - \epsilon) + n t + \epsilon) \\ \frac{-(p' - p) \cdot B^{(i-1)}}{n^2 - (n - i \cdot (n - n'))^2} \cdot \cos.(i \cdot (n' t - n t + \epsilon' - \epsilon) + n t + \epsilon) \end{array} \right\}.
 \end{aligned}$$

52. Let us now sum up the formulæ which we have investigated. If ( $r$ ) and ( $v$ ) represent the parts of the radius vector and of the longitude  $v$  of the orbit, which depend on the elliptic motion ; we will have

$$r = (r) + \delta r; v = (v) + \delta v;$$

The preceding value of  $s$  will be the latitude of  $m$  above the fixed plane ; but it will be more exact to employ instead of its two first terms which are independent of  $m'$ , the value of the latitude which would obtain in case that  $m$  did not depart from the plane of its primitive orbit. These expressions contain the entire theory of the planets, when the squares and products of the excentricities and of the inclinations of the orbits are neglected, which we are in most cases permitted to do. They have besides the advantage of appearing under a very simple form, which enables easily to perceive the law of their different terms.

Sometimes it will be necessary to recur to terms depending on the squares and the products of the excentricities and of the inclinations, and even on higher powers and products. These terms may be determined by means of the preceding analysis : the consideration which renders them necessary will always facilitate their determination. The approximations in which we will have occasion to take them into account, will introduce new terms depending on new arguments. These will again reproduce the arguments which the preceding approximations give, but with coefficients which are smaller and smaller

according to the following law, which it is easy to infer from the expansion of  $R$  into a series, and which has been given in N°. 48; *an argument which in the successive approximations is found for the first time among quantities of any order  $r$ , is only produced again by quantities of the orders  $r+2$ ,  $r+4$ , &c.*

It follows from this that the coefficients of the terms of the form  $t \cdot \frac{\sin.}{\cos.} (nt + \varepsilon)$ , which occur in the expressions of  $r$ ,  $v$ , and  $s$ ,

are approximate as far as quantities of the third order, that is to say, the approximation in which we only consider the squares and products of the excentricities and of the inclinations of their orbits, will add nothing to their values; therefore they have all the required accuracy; this observation is the more important, in as much as the secular variations of the orbits depend on these coefficients.

The different terms of the perturbations of  $r$ ,  $v$ ,  $s$ , are comprised in the form

$$k \cdot \frac{\sin.}{\cos.} \left\{ i \cdot (n't - nt + \varepsilon' - \varepsilon) + rnt + r\varepsilon \right\},$$

$r$  being either a positive integral number, or equal to cypher, and  $k$  being a function of the excentricities and of the inclinations of the orbits, of the order  $r$ , or of a superior order: we are enabled by means of this, to determine of what order a term depending on a given angle is.

It is manifest that the action of the bodies  $m''$ ,  $m'''$ , &c., only cause to be added to the preceding values of  $r$ ,  $v$  and  $s$ , terms analogous to those which result from the action of  $m'$ , and that if we neglect the square of the perturbating force, the sum of all these terms will give the complete values of  $r$ ,  $v$  and  $s$ . This fol-

lows from the nature of the formulæ  $(X')$ ,  $(Y')$  and  $(Z')$ ,\* which are linear with respect to quantities which depend on the perturbing force.

Finally, we shall obtain the perturbation of  $m'$ , produced by the action of  $m'$ , by changing in the preceding formulæ,  $a$ ,  $n$ ,  $h$ ,  $l$ ,  $\epsilon$ ,  $\varpi$ ,  $p$ ,  $q$ , and  $m'$ , into  $a'$ ,  $n'$ ,  $h'$ ,  $l'$ ,  $\epsilon'$ ,  $\varpi'$ ,  $p'$ ,  $q'$ , and  $m$ , and *vice versa*.

\* When quantities of the order of the square of the perturbing forces are neglected, the formulæ  $X'$ ,  $Y'$ ,  $Z'$ , are linear with respect to the perturbing force, from which it follows, that the variation of the sum is equal to the sum of the variations.

## CHAPTER VII.

*Of the secular inequalities of the celestial motions.*

53. The perturbating forces which disturb the elliptic motion introduce into the expressions of  $r$ ,  $\frac{dv}{dt}$  and of  $s$ , which are given in the preceding chapter, the time without the signs of the *sine* and *cosine*, or under the form of arcs of circles, which increasing indefinitely, must at length render these expressions erroneous ; it is therefore essentially necessary to make these arcs to disappear, and to obtain the functions which produce them by their expansion into a series. There has been given for this object, in the fifth chapter, a general method, from which it follows, that these arcs arise from the variations of the elliptic motion, which are then functions of the time. These variations being performed with extreme slowness, have been termed *secular inequalities*. Their theory is one of the most interesting points in the system of the world : we proceed to present it here, in all the detail which its importance requires.

By the preceding chapter we have

$$r = a \cdot \left\{ \begin{array}{l} 1 - h \cdot \sin. (nt + \varepsilon) - l \cdot \cos (nt + \varepsilon) - \&c. \\ + \frac{m'}{2} \cdot (l \cdot C + l' \cdot D) \cdot nt \cdot \sin. (nt + \varepsilon) \\ - \frac{m'}{2} \cdot (h \cdot C + h' \cdot D) \cdot nt \cdot \cos. (nt + \varepsilon) + m' S. \end{array} \right\};$$

$$\frac{dv}{dt} = n + 2nh \cdot \sin. (nt + \varepsilon) + 2nl \cdot \cos. (nt + \varepsilon) + \&c.$$

$$\begin{aligned}
 & - m' \cdot (l.C + l'D) \cdot n^2 t \cdot \sin. (nt + \epsilon) \\
 & + m' \cdot (h.C + h'D) \cdot n^2 t \cdot \cos. (nt + \epsilon) + m' \cdot T; \\
 s = & q \cdot \sin. (nt + \epsilon) - p \cdot \cos. (nt + \epsilon) + \&c. \\
 & - \frac{m'}{4} \cdot a^2 \cdot d \cdot (p' - p) \cdot B^{(1)} \cdot nt \cdot \sin. (nt + \epsilon) \\
 & - \frac{m'}{4} \cdot a^2 \cdot d' \cdot (q' - q) \cdot B^{(1)} \cdot nt \cdot \cos. (nt + \epsilon) + m' \cdot \chi;
 \end{aligned}$$

$s$ ,  $T$ ,  $\chi$ , being periodic functions of the time  $t$ . Let us at first consider the expression of  $\frac{dv}{dt}$ , and compare it with the expression of  $y$  of No. 43. As the arbitrary quantity  $n$  multiplies the arc  $t$ , under the periodic signs, in the expression for  $\frac{dv}{dt}$ ; we must employ the following equations, which have been found in N°. 43,

$$\begin{aligned}
 0 &= X' + \theta. X'' - Y; \\
 0 &= Y' + \theta. Y'' + X'' - 2Z;
 \end{aligned}$$

let us consider what  $X$ ,  $X'$ ,  $X''$ ,  $Y$ , &c. become in this case; the expression of  $\frac{dv}{dt}$ , being compared with that of  $y$  of the above cited N°. gives

$$\begin{aligned}
 X &= n + 2nh \cdot \sin. (nt + \epsilon) + 2nl \cdot \cos. (nt + \epsilon) + m' \cdot T; \\
 Y &= m' \cdot n^2 \cdot (h.C + h'D) \cdot \cos. (nt + \epsilon) - m'n^2 \cdot (l.C + l'D) \cdot \sin. (nt + \epsilon).
 \end{aligned}$$

The product of the partial differences of the constant quantities, into the disturbing masses being neglected,\* which we are permitted

\* Since the product of the partial differences of the constants into the disturbing masses are neglected, it will not be necessary to take into account the periodic function  $m' \cdot T$ ; the second and third terms of the value of  $X$  involve  $nt$  under the periodic signs, ∴ differencing the arbitraries contained under the signs with respect to  $n$ , we obtain the value of  $X''$ , which is given in the text.

to do, because these differences are of the order of the masses, we shall have by N°. 43,

$$\begin{aligned} X' &= \left( \frac{dn}{d\theta} \right) \cdot (1 + 2h \cdot \sin.(nt+\varepsilon) + 2l \cdot \cos.(nt+\varepsilon)) \\ &\quad + 2n \cdot \left( \frac{d\varepsilon}{d\theta} \right) \cdot (h \cdot \cos.(nt+\varepsilon) - l \cdot \sin.(nt+\varepsilon)) \\ &\quad + 2n \cdot \left( \frac{dh}{d\theta} \right) \cdot \sin.(nt+\varepsilon) + 2n \cdot \left( \frac{dl}{d\theta} \right) \cdot \cos.(nt+\varepsilon); \\ X'' &= 2n \cdot \left( \frac{dn}{d\theta} \right) \cdot (h \cdot \cos.(nt+\varepsilon) - l \cdot \sin.(nt+\varepsilon)). \end{aligned}$$

The equation  $0 = X' + \theta \cdot X'' - Y$ , will consequently become

$$\begin{aligned} 0 &= \left( \frac{dn}{d\theta} \right) \cdot (1 + 2h \cdot \sin.(nt+\varepsilon) + 2l \cdot \cos.(nt+\varepsilon)) \\ &\quad + 2n \cdot \left( \frac{d\varepsilon}{d\theta} \right) \cdot \sin.(nt+\varepsilon) + 2n \cdot \left( \frac{dl}{d\theta} \right) \cdot \cos.(nt+\varepsilon) \\ &\quad + 2n \cdot \left\{ \theta \cdot \left( \frac{dn}{d\theta} \right) + \left( \frac{d\varepsilon}{d\theta} \right) \right\} \cdot (h \cdot \cos.(nt+\varepsilon) - l \cdot \sin.(nt+\varepsilon)) \\ &\quad - m' \cdot n^2 \cdot (h \cdot C + h' \cdot D) \cdot \cos.(nt+\varepsilon) + m' \cdot n^2 \cdot (l \cdot C + l' \cdot D) \cdot \sin.(nt+\varepsilon). \end{aligned}$$

The coefficients of the corresponding sines and cosines, being put separately equal to nothing, we shall have

$$\begin{aligned} 0 &= \left( \frac{dn}{d\theta} \right) \\ 0 &= \left( \frac{dh}{d\theta} \right) - l \cdot \left( \frac{d\varepsilon}{d\theta} \right) + \frac{m' \cdot n}{2} \cdot (l \cdot C + l' \cdot D); \\ 0 &= \left( \frac{dl}{d\theta} \right) + h \cdot \left( \frac{d\varepsilon}{d\theta} \right) - \frac{m' \cdot n}{2} \cdot (h \cdot C + h' \cdot D). \end{aligned}$$

If these equations be integrated, and if in their integrals,  $\theta$  be changed into  $t$ , we shall have by N°. 43, the value of the arbitrary

quantities, in functions of  $t$ , and we can efface the arcs of the circle from the expressions for  $\frac{dv}{dt}$  and for  $r$ , but instead of this change we can all at once change  $\theta$  into  $t$ , in these differential equations. The first of these equations indicates that  $n$  is constant, and as the arbitrary quantity  $a$  of the expressions for  $r$  depends upon it, in consequence of the equations  $n^2 = \frac{1}{a^3}$ ;  $a$  is likewise constant. The two other equations are not sufficient to determine  $h$ ,  $l$ ,  $\epsilon$ . We shall have a new equation by observing, that the expression for  $\frac{dv}{dt}$ , gives by integrating,  $\int n dt$ , for the value of the mean longitude of  $m$ ; but we have supposed that this longitude is equal to  $nt + \epsilon$ ; therefore we have  $nt + \epsilon = \int n dt$ , which gives

$$t \cdot \frac{dn}{dt} + \frac{d\epsilon}{dt} = 0;$$

and as  $\frac{dn}{dt} = 0$ ; we shall have also  $\frac{d\epsilon}{dt} = 0$ . Thus the two arbitrary quantities  $n$  and  $\epsilon$  are constant; the arbitrary quantities  $h$  and  $l$  will be consequently determined by means of the differential equations,

$$\frac{dh}{dt} = -\frac{m' \cdot n}{2} \cdot (l \cdot C + l' \cdot D); \quad (1)$$

$$\frac{dl}{dt} = \frac{m' \cdot n}{2} \cdot (h \cdot C + h' \cdot D); \quad (2)$$

The consideration of the expression of  $\frac{dv}{dt}$  being sufficient to determine the values of  $n$ ,  $a$ ,  $h$ ,  $l$  and  $\epsilon$ ; we may perceive *a priori*, that the differential equation between the same quantities, which results from the expression for  $r$ , must coincide with the preceding. We may be easily assured of this *a posteriori*, by applying to this expression the method of N°. 43.

Let us now consider the expression of  $s$ . By comparing it with the expression of  $y$  in the N°. already cited ; we shall have

$$\begin{aligned} X &= q \cdot \sin. (nt + \varepsilon) - p \cdot \cos. (nt + \varepsilon) + m' \cdot \chi \\ Y &= \frac{m' \cdot n}{4} \cdot a^2 \cdot a' \cdot B^{(1)} \cdot (p - p') \cdot \sin. (nt + \varepsilon) \\ &\quad + \frac{m' \cdot n}{4} \cdot a^2 \cdot a' \cdot B^{(1)} \cdot (q - q') \cdot \cos. (nt + \varepsilon). \end{aligned}$$

$n$  and  $\varepsilon$  being constant, as is evident from what precedes ; by N°. 43, we have

$$\begin{aligned} X' &= \left( \frac{dq}{d\theta} \right) \cdot \sin. (nt + \varepsilon) - \left( \frac{dp}{d\theta} \right) \cdot \cos. (nt + \varepsilon) \\ X'' &= 0. \end{aligned}$$

The equation  $0 = X' + \theta \cdot X'' - Y$  consequently becomes,

$$\begin{aligned} 0 &= \left( \frac{dq}{d\theta} \right) \cdot \sin. (nt + \varepsilon) - \left( \frac{dp}{d\theta} \right) \cdot \cos. (nt + \varepsilon) \\ &\quad - \frac{m' \cdot n}{4} \cdot a^2 \cdot a' \cdot B^{(1)} \cdot (p - p') \cdot \sin. (nt + \varepsilon) \\ &\quad - \frac{m' \cdot n}{4} \cdot a^2 \cdot a' \cdot B^{(1)} \cdot (q - q') \cdot \cos. (nt + \varepsilon); \end{aligned}$$

from this we deduce, by comparing the coefficients of corresponding sines and cosines, and by changing  $\theta$  into  $t$ , in order to obtain  $p$  and  $q$  directly in functions of  $t$ ,

$$\frac{dp}{dt} = - \frac{m' \cdot n}{4} \cdot a^2 \cdot a' \cdot B^{(1)} \cdot (q - q'); \quad (3)$$

$$\frac{dq}{dt} = - \frac{m' \cdot n}{4} \cdot a^2 \cdot a' \cdot B^{(1)} \cdot (p - p'); \quad (4)$$

After that  $p$  and  $q$  shall have been determined by these equations, if we substitute them in the preceding expression of  $s$ , by obliterating the terms which contain the arcs of a circle, we will have

$$s = q \cdot \sin. (nt + \epsilon) - p \cdot \cos. (nt + \epsilon) + m' \cdot x.$$

54. The equation  $\frac{dn}{dt} = 0$ , to which we have arrived, is of great importance in the theory of the system of the world, in that it indicates that the mean motions of the heavenly bodies, and the greater axes of their orbits are invariable; but this equation is only accurate as far as quantities of the order  $m' \cdot h$ , inclusively. If quantities of the order  $m' \cdot h^2$ , or of the superior orders, would produce in  $\frac{dv}{dt}$ , a term of the form\*  $2kt$ ,  $k$  being a function of the elements of the orbits of  $m$  and of  $m'$ ; a term of the order  $kt^2$  would be produced in the expression of  $v$ , which by changing the longitudes of  $m$ , proportionably to the square of the time, would at length become extremely sensible. The equation  $\frac{dn}{dt} = 0$ , would no longer obtain, but in place of this equation there would be obtained by the preceding number  $\frac{dn}{dt} = 2k$ ; it is therefore of importance to ascertain whether there exists in the expressions for  $v$  terms of the form  $kt^2$ . We proceed to demonstrate that if we only consider the first power of the disturbing masses, however far we extend the approximations relative to the powers of the excentricities and the inclinations of the orbits; the expression of  $v$  will not involve terms of this kind.

For this purpose let the formula (X) of No. 46 be resumed,

$$\delta r = \frac{a \cdot \cos. v. \sqrt{n} dt. r. \sin. v. \left\{ 2 \int dR + r. \left\{ \frac{dR}{dr} \right\} \right\}}{\mu. \sqrt{1 - e^2}}$$

\* If the value of  $v$  contained a term of the order  $kt^2$ , there would exist in the expression of  $\frac{dv}{dt}$ , the term  $2kt$ , and consequently this term would exist in  $X$ , so that in comparing coefficients of corresponding terms, we would have  $\frac{dn}{dt} = 2k$ .

$$\frac{-a \cdot \sin v \int ndt \cdot r \cdot \cos v \left\{ 2 \int dR + r \left\{ \frac{dR}{dr} \right\} \right\}}{\mu \cdot \sqrt{1-e^2}}.$$

Let us consider the part of  $\partial r$  which involves terms multiplied by  $t^2$ , or for greater generality, let us consider the terms, which being multiplied by the sine or cosine\* of the angle  $\alpha t + \epsilon$ , in which  $\alpha$  is very small, have at the same time  $\alpha^2$  for a divisor. It is evident that  $\alpha$  being supposed = 0, there will result a term multiplied by  $t^2$ , so that the first case is contained in the second. The terms which have  $\alpha^2$  for a divisor can only be produced by a double integration; therefore they must be produced by the part of  $\partial r$ , which involves the double integral sign  $\int$ . Let us first examine the term

$$\frac{2a \cdot \cos v \int ndt \cdot (r \cdot \sin v \int dR)}{\mu \cdot \sqrt{1-e^2}}.$$

The origin of the angle  $i$  being fixed at the perihelion, we have in the elliptic orbit, by No. 20,

$$r = \frac{a(1-e^2)}{1+e \cdot \cos v},$$

and consequently

$$\cos v = \frac{a(1-e^2) - r}{er};$$

hence we deduce by differencing

$$r^2 \cdot dv \cdot \sin v = \frac{a(1-e^2)}{e} dr; \dagger$$

\*  $\alpha$  must be very small, because the sine is supposed to increase with great slowness; it is evident that if  $\alpha$  be supposed equal to nothing, the double integrations would produce a term proportional to the square of the time.

$$\dagger - dv \cdot \sin v = \frac{-er \cdot dr - e \cdot a \cdot ((1-e^2) + r) \cdot dr}{e^2 r^2} = \frac{a(1-e^2)}{er^2} \cdot dr.$$

but by No. 19, we have

$$r^2 \cdot dv = dt \cdot \sqrt{\mu a(1-e^2)} = a^2 \cdot ndt \cdot \sqrt{1-e^2};$$

consequently,

$$\frac{andt \cdot r \cdot \sin v}{\sqrt{1-e^2}} = \frac{r dr}{e}.$$

The term  $\frac{2a \cdot \cos v \cdot \int ndt \cdot (r \cdot \sin v \cdot \int dR)}{\mu \cdot \sqrt{1-e^2}}$ , will therefore become

$$\frac{2 \cdot \cos v}{\mu \cdot e} \cdot \int (r dr \cdot \int dR), \text{ or } \frac{\cos v}{\mu \cdot e} \cdot (r^2 \cdot \int dR - \int r^2 \cdot dR).$$

It is evident that as this last function does not contain any double integrals, there cannot arise any term which has  $e^2$  for a divisor.

Let us now consider the term

$$\frac{2a \cdot \sin v \cdot \int ndt \cdot (r \cdot \cos v \cdot \int dR)}{\mu \cdot \sqrt{1-e^2}},$$

of the expression of  $dr$ . By substituting for  $\cos v$ , its preceding value in  $r$ , this term becomes

$$\frac{2 \cdot \sin v \cdot \int ndt \cdot (r - a(1-e^2)) \cdot \int dR}{\mu e \cdot \sqrt{1-e^2}}.$$

By N°. 22, we have

$$r = a(1 + \frac{1}{2}e^2 + e\chi'),$$

$\chi'$  being an infinite series of the cosines of the angle  $nt + \iota$ , and its multiples; therefore we shall have

$$\frac{\int ndt}{e} \cdot (r - a(1-e^2)) \cdot \int dR = a \cdot \int ndt \cdot (\frac{3}{2}e + \chi') \cdot \int dR. *$$

$$* r - a(1 - e^2) = a(1 + \frac{1}{2}e^2 + e\chi') - a(1 - e^2) = a(\frac{3}{2}e^2 + e\chi').$$

Denoting by  $\chi''$  the integral  $\int \chi' ndt$ , we will have

$$a. \int ndt. (\frac{3}{2}e + \chi') \int dR = \frac{3}{2}ae. \int ndt. \int dR + a\chi''. \int dR - a. \int \chi''. dk.$$

As these two last terms do not involve the double sign of integration, no term which has  $\alpha^2$  for a denominator can arise from it; therefore if we only consider terms of this kind, we will have

$$\begin{aligned} - \frac{2a. \sin. v. \int ndt. (r. \cos. v. \int dR)}{\mu. \sqrt{1-e^2}} &= \frac{3a^2.e. \sin. v. \int ndt. \int dR}{\mu. \sqrt{1-e^2}} \\ &= \frac{dr}{ndt} \cdot \frac{3a}{\mu} \cdot \int ndt. \int dR; \end{aligned}$$

and the radius  $r$  will become

$$(r) + \left( \frac{dr}{ndt} \right) \cdot \frac{3a}{\mu} \cdot \int ndt. \int dR;$$

(r) and  $\left( \frac{dr}{ndt} \right)$  being the values of  $r$  and  $\frac{dr}{ndt}$  in the case of elliptic motion. Thus, in order to consider in the expression of the radius vector, the part of the perturbations, which is divided by  $\alpha^2$ , it will be sufficient to increase the mean longitude  $nt + \epsilon$ , by the quantity  $\frac{3a}{\mu} \cdot \int ndt. \int dR$ , in the expression for the mean longitude in the case of the elliptic motion.

Let us now examine whether this part of the perturbations should be taken into account in the expression for the longitude  $v$ . The formula (Y) of N°.46, gives by substituting  $\frac{3a}{\mu} \cdot \frac{dr}{ndt} \cdot \int ndt. \int dR$  in place of  $\delta r$ , when the terms divided by  $\alpha^2$  are only considered

$$\delta v = \frac{\left\{ \frac{2rd^2r + dr^2}{a^2n^2dt^2} + 1 \right\}}{\sqrt{1-e^2}} \cdot \frac{3a}{\mu} \cdot \int ndt. \int dR;$$

but by what goes before, we have

$$\delta r = \frac{ae.ndt. \sin. v}{\sqrt{1-e^2}} ; \quad r^2 dv = a^2 ndt. \sqrt{1-e^2} ;$$

hence it is easy to conclude, by substituting for  $\cos. v$ , its value, which has been already given\* in terms of  $r$

$$\frac{\left\{ \frac{2rd^2r+dr^2}{a^2n^2dt^2} + 1 \right\}}{\sqrt{1-e^2}} = \frac{dv}{ndt} ;$$

therefore if we only consider the part of the perturbations, of which the divisor is  $a^2$ , the longitude  $v$  will become

$$(v) + \left( \frac{dv}{ndt} \right) \cdot \frac{3a}{\mu} \cdot \int ndt. \int dR ;$$

(v) and  $\left( \frac{dv}{ndt} \right)$  being the parts of  $v$  and of  $\frac{dv}{ndt}$  which are relative to

\*  $r.d^2r = \frac{r.ae.n.dt. \cos. v.dv}{\sqrt{1-e^2}}$  equal by substituting for  $\cos. v$ ;  $\frac{(a.(a.(1-e^2)-r).ndt.dv)}{\sqrt{1-e^2}}$ ,

$$\therefore \frac{2rd^2r}{a^2n^2dt^2\sqrt{1-e^2}} = \frac{2a^2.ndt.\sqrt{1-e^2}.dv}{a^2n^2dt^2\sqrt{1-e^2}} - \frac{2ar.ndt.dv}{(1-e^2).a^2n^2dt^2} ; \quad dr^2 = \frac{a^2e^2n^2dt^2}{1-e^2}$$

$$-\frac{(a^2e^2n^2dt^2.(a^2.(1-e^2)^2-2ar.(1-e^2)+r^2)}{e^2.r^2.(1-e^2)}, \quad \therefore \frac{dr^2}{a^2n^2dt^2\sqrt{1-e^2}} = \frac{e^2}{(1-e^2)^2} - \frac{a^2.\sqrt{1-e^2}}{r^2}$$

$$+ \frac{2a}{r.\sqrt{1-e^2}} + \frac{1}{(1-e^2)^2}, \quad \therefore \frac{2rd^2r+dr^2}{a^2n^2dt^2} = \frac{2dv}{ndt} - \frac{2rdv}{(1-e^2).andt} + \frac{e^2}{(1-e^2)^2}$$

$$- \frac{a^2.\sqrt{1-e^2}}{r^2} + \frac{2a}{r.\sqrt{1-e^2}} - \frac{1}{(1-e^2)^2} + \frac{1}{\sqrt{1-e^2}} ; \text{ now } - \frac{2rdv}{(1-e^2).andt} + \frac{2a}{r.\sqrt{1-e^2}}$$

$$= - \frac{2r^2dv+2a^2ndt.\sqrt{1-e^2}}{(1-e^2).r.andt} = 0, \text{ and } \frac{e^2}{(1-e^2)^2} + \frac{1}{\sqrt{1-e^2}} - \frac{1}{(1-e^2)^2} = 0, \quad \because \text{since}$$

$$- \frac{a^2.\sqrt{1-e^2}}{r^2} = - \frac{dv}{ndt}, \text{ the preceding expression becomes equal to } \frac{dv}{ndt}.$$

the elliptic motion. Therefore in order to consider this part of the perturbations in the expression for the longitude of  $m$ , we should follow the same rule as we have given, when considering the expression of the radius vector, that is to say, it is necessary to increase in the elliptic expression of the true longitude, the mean longitude  $nt + \epsilon$  by the quantity  $\frac{3a}{\mu} \cdot \int ndt \cdot \int dR$ .

The constant part of the expression for  $\left( \frac{dv}{ndt} \right)$ , being expanded into a series of the cosines of the angle  $nt + \epsilon$  and of its multiples, is reduced to unity, as we have seen in N°. 22; hence arises the term  $\frac{3a}{\mu} \cdot \int ndt \cdot \int dR$  in the expression for the longitude. If  $dR$  contains the constant term  $km'ndt$ , this term would produce  $\frac{3}{2} \cdot \frac{am'}{\mu} \cdot k \cdot n^2 t^2$ , in the expression for the longitude  $v$ . Therefore in order to ascertain whether such terms exist in this expression, we must consider whether  $dR$  contains a constant term.

When the excentricities of the orbits and their mutual inclinations to each other are small,  $R$  can be reduced always into an infinite series of the sines and cosines of angles proportional to the time  $t$ . They can be generally represented by the term  $km' \cdot \cos(i'n't + int + A)$ ,  $i$  and  $i'$  being integral numbers, either positive or negative, or equal to cypher. The differential of this term taken solely with respect to the mean motion of  $m$ , is  $-ikm'ndt \cdot \sin(i'n't + int + A)$ ; this is the part of  $dR$ , which is relative to this term: it cannot be constant unless we have  $0 = i'n' + in$ ; but this supposes that the mean motions of the bodies  $m$  and  $m'$  are commensurable with each other; and as this is not the case in the solar system, we ought to infer from it, that the value of  $dR$  does not contain constant terms; and that consequently if we only consider the first power of the perturbing masses, the mean motions of the celestial bodies are uniform, or what comes to the

same thing,  $\frac{dn}{dt} = 0$ . The value of  $a$  being connected with that of  $n$ , by means of the equation  $n^2 = \frac{\mu}{a}$ ; it follows that if we do not take into account periodic quantities, the greater axes of the orbits are constant.

If the mean motions of the bodies  $m$  and  $m'$ , though not exactly commensurable are very nearly so; there will exist in the theory of their motions, inequalities of a very long period, and which may become very sensible, on account of the smallness of the divisor  $\alpha^2$ . We will see in the sequel that this obtains in the case of Jupiter and Saturn. The preceding analysis will give in a very simple manner, the part of the perturbations which depend on this divisor. It follows from it, that then it is sufficient to make the mean longitude  $nt + \varepsilon$  or  $\int ndt$  vary by the quantity  $\frac{3an}{\mu} \cdot \int ndt.dR$ ; which comes to make  $n$ , in the integral  $\int ndt$ , increase by the quantity  $\frac{3an}{\mu} \cdot \int dR$ ; the orbit of  $m$  being considered as a variable ellipse, we have  $n^2 = \frac{\mu}{a^3}$ ; therefore the preceding variation of  $n$  must introduce in the semiaxis major of the orbit, the variation\*  $= \frac{2a^2 \int dR}{\mu}$ .

If in the value of  $\frac{dv}{dt}$  we carry the approximation as far as quantities of the order of the squares of the perturbing masses, terms proportional to the times will arise; but by attentively considering the differential equations of the motion of the bodies  $m$ ,  $m'$ ,  $m''$ , &c.; it will readily appear that these terms are at the same time of the order of the squares and of the products of the excentricities and of

\* From the equation  $n^2 = \frac{\mu}{a^3}$  we have  $da = - \frac{2na^4 dn}{3\mu}$  substituting  $\frac{3an}{\mu} \cdot \int dR$  for  $dn$ , and we have  $da = - \frac{2n^2 a^5 \cdot \int dR}{\mu^2} = - \frac{2a^4}{\mu} \cdot \int dR$ .

the inclinations of the orbits. However, as every thing which affects the mean motion, may at length become very sensible, we will consider in the sequel those terms, and we shall see that they produce the secular equations which have been observed in the motion of the moon.

55. Let us now resume the equations (1) and (2) of No. 55, and let

$$(0, 1) = -\frac{m'nC}{2}; \quad [\underline{0}, \underline{1}] = \frac{m'.n.D}{2};$$

they will become

$$\frac{dh}{dt} = (0, 1). l - [\underline{0}, \underline{1}]. l';$$

$$\frac{dl}{dt} = -(0, 1). h + [\underline{0}, \underline{1}]. h'.$$

The expressions of  $(0, 1)$  and of  $[\underline{0}, \underline{1}]$  may be determined very simply in the following manner. By substituting in place of  $C$ , and of  $D$ , their values, which have been determined in N°. 50, there will be obtained

$$(0, 1) = -\frac{m'.n}{2} \cdot \left\{ a^2 \cdot \left( \frac{dA^{(0)}}{da} \right) + \frac{1}{2} a^3 \cdot \left( \frac{d^2 A^{(0)}}{da^2} \right) \right\};$$

$$[\underline{0}, \underline{1}] = \frac{m'.n}{2} \cdot \left\{ a \cdot A^{(1)} - a^2 \cdot \left( \frac{dA^{(1)}}{da} \right) - \frac{1}{2} a^3 \cdot \left( \frac{d^2 A^{(1)}}{da^2} \right) \right\}.$$

By N°. 49, we have

$$a^2 \cdot \left( \frac{dA^{(0)}}{da} \right) + \frac{1}{2} a^3 \cdot \left( \frac{d^2 A^{(0)}}{da^2} \right) = -\alpha^2 \cdot \frac{db_{\frac{1}{4}}^{(0)}}{d\alpha} - \frac{1}{2} \alpha^3 \cdot \frac{d^2 b_{\frac{1}{4}}^{(0)}}{d\alpha^2};$$

we will readily obtain by the same N°.  $\frac{db_{\frac{1}{4}}^{(0)}}{d\alpha}$ ,  $\frac{d^2 b_{\frac{1}{4}}^{(0)}}{d\alpha^2}$ , in functions of  $b_{\frac{1}{4}}^{(0)}$ , and of  $b_{\frac{1}{4}}^{(1)}$ ; and these quantities are given in linear functions of  $b_{\frac{1}{4}}^{(0)}$ , and of  $b_{\frac{1}{4}}^{(1)}$ ; this being premised we shall find

$$a^2 \cdot \left( \frac{dA^{(0)}}{da} \right) + \frac{1}{2} a^3 \cdot \left( \frac{d^2 A^{(0)}}{da^2} \right) = \frac{3\alpha^2 \cdot b_{\frac{1}{4}}^{(1)}}{2(1-\alpha^2)^2};$$

therefore

$$(0, 1) = -\frac{3m' \cdot n \cdot \alpha^2 \cdot b_{\frac{1}{2}}^{(1)}}{4 \cdot (1 - \alpha^2)^2},$$

let

$$(a^2 - 2aa' \cdot \cos \theta + a'^2)^{\frac{1}{2}} = (a, a') + (a, a)' \cdot \cos \theta + (a, a'')'' \cdot \cos 2\theta + \text{&c.}$$

by No. 49, we shall have

$$(aa') = \frac{1}{2}a' \cdot b_{\frac{1}{2}}^{(0)}; * \quad (a, a)' = (a' \cdot b_{\frac{1}{2}}^{(1)}), \text{ &c.}$$

therefore we shall have

$$(0, 1) = -\frac{3m' \cdot n \alpha^2 \cdot a' \cdot (a, a)'}{4 \cdot (a'^2 - a^2)^2}.$$

consequently by N°. 49, we obtain

$$aA^{(1)} - a^2 \cdot \left\{ \frac{dA^{(1)}}{da} \right\} - \frac{1}{2}a^3 \cdot \left\{ \frac{d^2 A^{(1)}}{da^2} \right\} = -\alpha \cdot \left\{ b_{\frac{1}{2}}^{(1)} - \alpha \cdot \frac{db_{\frac{1}{2}}^{(1)}}{d\alpha} - \frac{1}{2}\alpha^2 \cdot \frac{d^2 b_{\frac{1}{2}}^{(1)}}{d\alpha^2} \right\};$$

by substituting in place of  $b_{\frac{1}{2}}^{(1)}$  and of its differences, their values in  $b_{\frac{1}{2}}^{(0)}$ , and  $b_{\frac{1}{2}}^{(1)}$ , the preceding function will be found equal to

$$-\frac{3\alpha \cdot ((1 + \alpha^2) \cdot b_{\frac{1}{2}}^{(1)} + \frac{1}{2}\alpha \cdot b_{\frac{1}{2}}^{(0)})}{(1 - \alpha^2)^2};$$

therefore

$$[\underline{0, 1}] = -\frac{3\alpha \cdot m' \cdot n \cdot ((1 + \alpha^2) \cdot b_{\frac{1}{2}}^{(1)} + \frac{1}{2}\alpha \cdot b_{\frac{1}{2}}^{(0)})}{2 \cdot (1'^2 - \alpha^2)^2};$$

or

$$[\underline{0, 1}] = -\frac{3m' \cdot n \cdot ((a^2 + a'^2) \cdot (a, a') + a \cdot a' \cdot (a, a)')}{2 \cdot (a'^2 - a^2)^2},$$

we shall obtain by this means very simple expressions for  $(0, 1)$  and for  $[\underline{0, 1}]$ , and it is easy to conclude by the values in a series for  $b_{\frac{1}{2}}^{(0)}$ , and for  $b_{\frac{1}{2}}^{(1)}$ , which are given in N°. 49, that these expressions are positive, if  $n$  be positive, and negative, if  $n$  be negative.

Naming  $(0, 2)$  and  $[\underline{0, 2}]$  what  $(0, 1)$  and  $[\underline{0, 1}]$  become, when  $a'$  and  $m'$  are changed into  $a''$  and  $m''$ , and in like manner let  $(0, 3)$  and  $[\underline{0, 3}]$  represent what these same quantities become when  $a'$  and  $m'$  are changed into  $a'''$  and  $m'''$ ; and so on. Moreover let  $h'', l'', h''', l'''$ , denote what

$h$  and  $l$  become relative to the bodies  $m'$ ,  $m''$ , &c.; we shall obtain in consequence of the combined actions of the different bodies  $m'$ ,  $m''$ ,  $m'''$ , &c. on  $m$ ,

$$\frac{dh}{dt} = ((0, 1) + (0, 2) + (0, 3) + \&c.) \cdot l - [\underline{0}, \underline{1}] \cdot l' - [\underline{0}, \underline{2}] \cdot l'' - \&c.;$$

$$\frac{dl}{dt} = -((0, 1) + (0, 2) + (0, 3) + \&c.) \cdot h + [\underline{0}, \underline{1}] \cdot h' + [\underline{0}, \underline{2}] \cdot h'' + \&c..$$

It is manifest that  $\frac{dh'}{dt}$ ,  $\frac{dl'}{dt}$ ;  $\frac{dh''}{dt}$ ,  $\frac{dl''}{dt}$ , &c., will be determined by expressions similar to those of  $\frac{dh}{dt}$  and of  $\frac{dl}{dt}$ , and that it is easy to infer them from the preceding by changing successively, that which is relative to  $m$ , into that which refers to  $m'$ ,  $m''$ , &c., and *vice versa*. Let therefore

$$(1, 0), [\underline{1}, \underline{0}]; \quad (1, 2), [\underline{1}, \underline{2}]; \quad \&c.$$

be what

$$(0, 1), [\underline{0}, \underline{2}]; \quad (0, 2), [\underline{0}, \underline{2}]; \quad \&c.$$

become when we change in them that which is relative to  $m$ , into that which is relative to  $m'$ , and conversely; let also

$$(2, 0), [\underline{2}, \underline{0}]; \quad (2, 1), [\underline{2}, \underline{1}]; \quad \&c.$$

be what

$$(0, 2), [\underline{0}, \underline{1}]; \quad (0, 1), [\underline{0}, \underline{1}]$$

become when that which is relative to  $m$ , is changed into that which is relative to  $m'$ , and conversely, and so of the rest. The preceding

\* In this case  $(1 - 2a \cos. \theta + a^2)^{-s} = (1 - 2a \cos. \theta + a^2)^{\frac{1}{2}}$ ,  $\because s = -\frac{1}{2}$ ; see page 278;  $\therefore$  the first term in the expansion of  $a'^{-2s} \cdot (1 - 2a \cos. \theta + a^2)^{-s}$  becomes (when  $s = -\frac{1}{2}$ )  $a' \cdot b_{-\frac{1}{2}}^{(0)}$ , and the coefficient of  $\cos. \theta = a' \cdot b_{-\frac{1}{2}}^{(1)}$ .

differential equations referred successively to the bodies  $m$ ,  $m'$ ,  $m''$ , &c. will give for the determination of  $h$ ,  $l$ ,  $h'$ ,  $l'$ ,  $h''$ ,  $l''$ , &c. the following system of equations,

$$\left. \begin{aligned} \frac{dh}{dt} &= ((0, 1) + (0, 2) + (0, 3) + \&c.) \cdot l - [\underline{0}, \underline{1}] \cdot l' - [\underline{0}, \underline{2}] \cdot l'' - [\underline{0}, \underline{3}] \cdot l''' - \&c. \\ \frac{dl}{dt} &= -((0, 1) + (0, 2) + (0, 3) + \&c.) \cdot h + [\underline{0}, \underline{1}] \cdot h' + [\underline{0}, \underline{2}] \cdot h'' + [\underline{0}, \underline{3}] \cdot h''' + \&c. \\ \frac{dh'}{dt} &= ((1, 0) + (1, 2) + (1, 3) + \&c.) \cdot l' - [\underline{1}, \underline{0}] \cdot l - [\underline{1}, \underline{2}] \cdot l'' - [\underline{1}, \underline{3}] \cdot l''' - \&c. \\ \frac{dl'}{dt} &= -((1, 0) + (1, 2) + (1, 3) + \&c.) \cdot h' + [\underline{1}, \underline{0}] \cdot h + [\underline{1}, \underline{0}] \cdot h'' + [\underline{1}, \underline{5}] \cdot h''' + \&c. \\ \frac{dh''}{dt} &= ((2, 0) + (2, 1) + (2, 3) + \&c.) \cdot l'' - [\underline{2}, \underline{0}] \cdot l - [\underline{2}, \underline{1}] \cdot l' - [\underline{2}, \underline{3}] \cdot l''' - \&c. \\ \frac{dl''}{dt} &= -((2, 0) + (2, 1) + (2, 3) + \&c.) \cdot h'' + [\underline{2}, \underline{0}] \cdot h + [\underline{2}, \underline{1}] \cdot h' + [\underline{2}, \underline{5}] \cdot h''' + \&c. \end{aligned} \right\}; \quad (A)$$

The quantities  $(0, 1)$  and  $(1, 0)$ ,  $[\underline{0}, \underline{1}]$  and  $[\underline{1}, \underline{0}]$  have remarkable relations, which will very much facilitate the computation, and which will be useful in the sequel. By what precedes we have,

$$(0, 1) = -\frac{3m'na^2 \cdot a' \cdot (a, a')'}{4(a'^2 - a^2)^2}.$$

If in this expression for  $(0, 1)$ ,  $m'$  be changed into  $m$ ,  $n$  into  $n'$ ,  $a$  into  $a'$ , and *vice versa*; we shall have the expression of  $(1, 0)$  which will be consequently

$$(1, 0) = -\frac{3m.n'a'^2 \cdot a \cdot (a', a)'}{4(a'^2 - a^2)^2};$$

but we have  $(a', a)' = (aa')'$ , because each of these quantities results from the expansion of the function  $(a^2 - 2aa' \cdot \cos. \theta + a'^2)^\frac{1}{2}$  into a series arranged according to the cosines of the angle  $\theta$  and of its multiples; therefore we will have

$$(0, 1) m.n'a' = (1, 0) \cdot m'.n.a;$$

but, when the masses  $m$ , and  $m'$ , &c., are neglected with respect to  $M$ ,

$$n^2 = \frac{M}{a^3}; \quad n'^2 = \frac{M}{a'^3}; \quad \text{&c.}$$

therefore

$$(0, 1). m. \sqrt{a} = (1, 0). m'. \sqrt{a'};$$

by means of this equation we can easily obtain  $(1, 0)$  when  $(0, 1)$  will be determined. In like manner we have

$$[\underline{0}, \underline{1}] m. \sqrt{a} = [\underline{1}, \underline{0}] m'. \sqrt{a'}.$$

These two equations will also subsist when  $n$  and  $n'$  have contrary signs; that is to say, when the two bodies  $m$  and  $m'$  revolve in contrary directions; but then we must give the sign of  $n$  to the radical  $\sqrt{a}$ , and the sign of  $n'$  to the radical  $\sqrt{a'}$ .

The following equations result evidently from the two preceding:

$$(0, 2) m. \sqrt{a} = (2, 0) m''. \sqrt{a''}; \quad [\underline{0}, \underline{2}] m. \sqrt{a} = [\underline{2}, \underline{0}] m''. \sqrt{a''}; \quad \text{&c.}$$

$$(1, 2) m'. \sqrt{a} = (2, 1) m''. \sqrt{a''}; \quad [\underline{1}, \underline{2}] m'. \sqrt{a'} = [\underline{2}, \underline{1}] m''. \sqrt{a''}; \quad \text{&c.}$$

56. Now in order to integrate the equations (A) of the preceding number, let

$$h = N. \sin. (gt + \epsilon); \quad l = N. \cos. (gt + \epsilon);$$

$$h' = N'. \sin. (gt + \epsilon); \quad l' = N'. \cos. (gt + \epsilon);$$

these values being substituted in the equation (A), will give

$$\left. \begin{aligned} Ng &= ((0, 1) + (0, 2) + \text{&c.}). N - [\underline{0}, \underline{1}] N' - [\underline{0}, \underline{2}] N'' - \text{&c.} \\ N'g &= ((1, 0) + (1, 2) + \text{&c.}). N' - [\underline{1}, \underline{0}] N - [\underline{1}, \underline{2}] N'' - \text{&c.} \\ N''g &= ((2, 0) + (2, 1) + \text{&c.}). N'' - [\underline{2}, \underline{0}] N' - [\underline{2}, \underline{1}] N - \text{&c.} \end{aligned} \right\}; \quad (\text{B})^*$$

&c.

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\* In general, the number of these algebraic equations is equal to that of the coefficients.

The number of bodies  $m, m', m'', \&c.$ , being equal to  $i$ , the number of these equations will be also  $i$ , and by eliminating the constant quantities  $N, N' \&c.$ , we will have a final equation in  $g$ , of the degree  $i$ , which can easily be obtained in the following manner :

Naming  $\phi$  the function

$$\begin{aligned} & N^2 \cdot m \cdot \sqrt{a} \cdot (g - (0, 1) - (0, 2) - \&c.) \\ & + N'^2 m' \cdot \sqrt{a'} \cdot (g - (1, 0) - (1, 2) - \&c.) \\ & + \&c. \\ & + 2N \cdot m \cdot \sqrt{a} \cdot ([\underline{0}, \underline{1}] \cdot N' + [\underline{0}, \underline{1}] \cdot N'' + \&c.) \\ & + 2N' \cdot m' \cdot \sqrt{a'} \cdot ([\underline{1}, \underline{2}] \cdot N'' + [\underline{1}, \underline{5}] \cdot N''' + \&c.) \\ & + 2N'' \cdot m'' \cdot \sqrt{a''} \cdot ([\underline{2}, \underline{3}] \cdot N'' + \&c.) \\ & + \&c. \end{aligned}$$

In consequence of the relations which are given in the preceding number, the equations ( $B$ ) are reduced to the following  $\left( \frac{d\phi}{dN} \right) = 0$  ;  $\left( \frac{d\phi}{dN'} \right) = 0$  ;  $\left( \frac{d\phi}{dN''} \right) = 0$ , &c.; therefore  $N, N', N'', \&c.$  being considered as so many variables,  $\phi$  will be a *maximum*. Moreover,  $\phi$  being an homogeneous function of these variables of the second dimension ; we have

$$N \cdot \left( \frac{d\phi}{dN} \right) + N' \cdot \left( \frac{d\phi}{dN'} \right) + \&c. = 2\phi ;$$

therefore in consequence of the preceding equations,  $\phi = 0$ .

Now, we can determine in the following manner the *maximum* of the function  $\phi$ . First, let this function be differenced relatively to  $N$ , and then substitute in  $\phi$ , in place of  $N$  its value deduced from the equation

$N, N', \&c.$ ; by means of the operations performed on the function  $\phi$ , the ratio of these coefficients is obtained ; one of them remains undetermined.

$\left(\frac{d\phi}{dN}\right) = 0$ ; this value will be a linear function of the quantities  $N'$ ,  $N''$ , &c.; in this manner we shall obtain a rational function, which is both integral and homogeneous, of the second dimension in  $N'$ ,  $N''$ , &c., let  $\phi^{(1)}$  be this function. By differencing  $\phi^{(1)}$  relatively to  $N'$ , and by substituting in  $\phi^{(1)}$  in place of  $N^{(1)}$  its value deduced from the equation  $\left(\frac{d\phi^{(1)}}{dN'}\right) = 0$ ; we shall obtain an homogeneous function, which will be likewise of the second dimension in  $N''$ ,  $N'''$ , &c. let  $\phi^{(2)}$  be this function. By continuing this operation, we will arrive at a function  $\phi^{(i-1)}$  of the second dimension, in  $N^{(i-1)}$ , and which will consequently be of the form  $(N^{(i-1)})^2 \cdot k$ ;  $k$  being a function of  $g$ , and of constant quantities. If the differential of  $\phi^{(i-1)}$  taken with respect to  $N^{(i-1)}$ , be put equal to cypher, we shall have  $k = 0$ ; this will give an equation in  $g$  of the degree  $i$ , of which the different roots will give so many different systems for the indeterminate quantities  $N$ ,  $N'$ ,  $N''$ , &c.; the indeterminate  $N^{(i-1)}$ , will be the arbitrary quantity of each system, we shall obtain immediately, the ratio of the other indeterminate quantities  $N$ ,  $N'$ , &c. of the same system to this, by means of the preceding equations taken in an reverse order, namely

$$\left(\frac{d\phi^{(i-2)}}{dN^{i-2}}\right) = 0; \quad \left(\frac{d\phi^{(i-3)}}{dN^{i-3}}\right) = 0; \quad \text{&c.}$$

Let  $g$ ,  $g_1$ ,  $g_2$ , be the  $i$  roots of the equation in  $g$ ; let  $N$ ,  $N'$ ,  $N''$ , &c. be the system of indeterminate quantities relative to the root  $g$ ; let  $N$ ,  $N'$ ,  $N''$ , &c. be the system of indeterminate quantities relative to the root  $g_1$ , and so on of the rest: by the known theory of differential linear equations we will have

$$h = N \cdot \sin.(gt + \epsilon) + N_1 \cdot \sin.(g_1 t + \epsilon_1) + N_2 \cdot \sin.(g_2 t + \epsilon_2) + \text{&c.};$$

$$h' = N' \cdot \sin.(gt + \epsilon) + N'_1 \cdot \sin.(g_1 t + \epsilon_1) + N'_2 \cdot \sin.(g_2 t + \epsilon_2) + \text{&c.};$$

$$h'' = N'' \cdot \sin.(gt + \epsilon) + N''_1 \cdot \sin.(g_1 t + \epsilon_1) + N''_2 \cdot \sin.(g_2 t + \epsilon_2) + \text{&c.};$$

&c.

$\epsilon, \epsilon_1, \epsilon_2$ , being constant arbitrary quantities. The values of  $l, l', l'', \&c.$  will be obtained by changing in the expressions for  $h, h', h'', \&c.$  the sines into the cosines. These different values contain twice as many arbitrary quantities, as there are roots  $g, g_1, g_2, \&c.$ ; for each system of indeterminate quantities contains one arbitrary quantity, and besides, there are  $i$  arbitrary quantities  $\epsilon, \epsilon_1, \epsilon_2, \&c.$ ; these values are consequently the complete integrals of the equations (A) of the preceding number.

It is only now required to determine the constant quantities  $N, N, \&c. N', N'. \&c. \epsilon, \epsilon', \&c.$  These constant quantities are not given immediately by observation; but they make known at a given epoch, the excentricities  $e, e. \&c.$  of the orbits, and the longitudes  $\varpi, \varpi', \&c.$  of their perihelions, and consequently the values of  $h, h', \&c. l, l', \&c.$ ; thus we shall derive from them the values of the preceding constant quantities. For this purpose it may be observed, that if we multiply the first, third, and fifth, &c. of the differential equations (A) of the preceding number, by  $Nm\sqrt{a}, N'm'\sqrt{a'}, \&c.$  respectively, we will have in consequence of the equations (B), and of the relations found in the preceding number, between  $(0, 1)$  and  $(1, 0)$ ,  $(0, 2)$  and  $(2, 0)$ , &c.

$$\begin{aligned} & N \cdot \frac{dh}{dt} \cdot m \cdot \sqrt{a} + N' \cdot \frac{dh'}{dt} \cdot m' \cdot \sqrt{a'} + N'' \cdot \frac{dh''}{dt} \cdot m'' \cdot \sqrt{a''} + \&c.)^* \\ & = g \cdot (N \cdot l \cdot m \cdot \sqrt{a} + N' \cdot l' \cdot m' \cdot \sqrt{a'} + N'' \cdot l'' \cdot m'' \cdot \sqrt{a''} + \&c.) \end{aligned}$$

\* Multiplying the first of the equations (A) by  $N \cdot m \cdot \sqrt{a}$ , and the third by  $N' \cdot m' \cdot \sqrt{a'}$ , we shall obtain by adding them together,

$$\begin{aligned} & N \cdot \frac{dh}{dt} \cdot m \cdot \sqrt{a} + N' \cdot \frac{dh'}{dt} \cdot m' \cdot \sqrt{a'} = (0, 1) + (0, 2) + (0, 3) + \&c.) l \cdot N \cdot m \cdot \sqrt{a} - \\ & [\underline{0,1}] \cdot l \cdot N \cdot m \cdot \sqrt{a} - \&c. + ((1, 0) + (1, 2) + (1, 3) + \&c.) l \cdot N' \cdot m' \cdot \sqrt{a'} - [\underline{1,0}] \cdot l \cdot N' \cdot m' \cdot \sqrt{a'} - \&c. = (\text{as } [\underline{0,1}] \cdot m \cdot \sqrt{a} = [\underline{1,0}] \cdot m' \cdot \sqrt{a'},) l \cdot m \cdot \sqrt{a} \cdot ((0, 1) + (0, 2) + (0, 3) + \&c.) N - [\underline{0,1}] \cdot N' - \&c.) + l \cdot m \cdot \sqrt{a'} \cdot ((1, 0) + (1, 2) + (1, 3) + \&c.) N'. \end{aligned}$$

By substituting in this equation, in place of  $h, h', h'', \&c.$   $l, l', l'', \&c.$  their preceding values ; we will have by comparing the coefficients of the same cosines,

$$\begin{aligned} 0 &= N.N.m.\sqrt{a} + N'.N'.m'.\sqrt{a'} + N''.N''.m''.\sqrt{a''} + \&c.; \\ 0 &= N.N_2.m.\sqrt{a} + N'.N'_2.m'.\sqrt{a'} + N''.N''_2.m''.\sqrt{a''} + \&c. \\ &\&c. \end{aligned}$$

This being premised, if the preceding values of  $h, h', \&c.$ , be multiplied by  $N.m.\sqrt{a}, N'.m'.\sqrt{a'}, \&c.$ , respectively, we will have in consequence of these last equations,

$$\begin{aligned} &N.mh.\sqrt{a} + N'.m'h'.\sqrt{a'} + N''.m''h''.\sqrt{a''} + \&c. \\ &= (N^2.m.\sqrt{a} + N'^2.m'.\sqrt{a'} + N''^2.m''.\sqrt{a''} + \&c.). \sin.(gt+\epsilon). \end{aligned}$$

we shall have in like manner,

$$\begin{aligned} &N.ml.\sqrt{a} + N'm'l'.\sqrt{a'} + N''.m''l''.\sqrt{a''} + \&c. \\ &= (N^2.m.\sqrt{a} + N'^2.m'.\sqrt{a'} + N''^2.m''.\sqrt{a''} + \&c.) \cos.(gt+\epsilon). \end{aligned}$$

The commencement of the time being fixed at an epoch, for which the values of  $h, l, h', l', \&c.$  are supposed to be known ; the two preceding equations give

$\therefore [1.6]. N \dots \&c.) = (Nlm.\sqrt{a} + N'l'm'.\sqrt{a'} + \&c.) g$ ; now by substituting for  $\frac{dh}{dt} + \frac{dh'}{dt} + \&c. l, l', \&c.$  we obtain ;  $m.\sqrt{a}.(N^2.g. \cos(gt+\epsilon) + NN_2g_1. \cos(gt+\epsilon_1) + NN_2g_2. \cos(gt+\epsilon_2)) + \&c. + m'.\sqrt{a'}.(N'^2.g. \cos(gt+\epsilon) + N'N'_1.g. \cos(gt+\epsilon_1) + N'N'_2.g. \cos(gt+\epsilon_2) + \&c.) = g(N^2.m.\sqrt{a}. \cos(gt+\epsilon) + NN_2. \cos(gt+\epsilon) + N'N'_2. \cos(gt+\epsilon_2) + m'.\sqrt{a'}.N'^2. \cos(gt+\epsilon) + N'N'_1. \cos(gt+\epsilon_1) + N'N'_2. \cos(gt+\epsilon_2) + \&c.)$  From hence it follows, that in order for this equation always to obtain, we must have  $NN_2.m.\sqrt{a} + N'N'_1.m'.\sqrt{a'} + \&c. = 0$ .

$$\tan. \epsilon = \frac{N.h.m.\sqrt{\bar{a}} + N'.h'.m'.\sqrt{\bar{a}'} + N''.h''m''.\sqrt{\bar{a}''} + \&c. *}{N.l.m.\sqrt{\bar{a}} + N'.l'.m'.\sqrt{\bar{a}'} + N''.l''m''.\sqrt{\bar{a}''} + \&c.}$$

This expression of  $\tan. \epsilon$  does not contain any indeterminate quantity ; for although the constant quantities  $N, N', N'',$  depend on the indeterminate quantity  $N^{(i-1)}$  ; yet, as, their ratio to this indeterminate quantity is known by what precedes, it must disappear from the  $\tan. \epsilon$ .  $\epsilon$  being thus determined, we shall obtain  $N^{(i-1)}$ , by means of one of the two equations which determine  $\tan. \epsilon$ , and from it we infer the system of indeterminates  $N, N', N'', \&c.$ , relative to the root  $g$ . And if in the preceding expressions, this root be successively changed into  $g_1, g_2, g_3, \&c.$ , the values of the arbitrary quantities relative to each of these roots will be obtained.

These values being substituted in the expressions for  $h, l, h', l', \&c.$  the excentricities  $e, e', \&c.$  of the orbits may be deduced from them, as also the longitudes  $\varpi, \varpi', \&c.$ , of their perihelions, by means of the equations

$$e^2 = h^2 + l^2 ; \quad e'^2 = h'^2 + l'^2 ; \quad \&c.$$

$$\tan. \varpi = \frac{h}{l} ; \quad \tan. \varpi' = \frac{h'}{l'} ; \quad \&c.$$

thus we shall have

$$\begin{aligned} e^2 &= N^2 + N_1^2 + N_2^2 + \&c. + 2NN_1 \cdot \cos. ((g_1 - g).t + \epsilon_1 - \epsilon)^* \\ &+ 2NN_2 \cdot \cos. ((g_2 - g).t + \epsilon_2 - \epsilon) + 2N_1N_2 \cdot \cos. (g_2 - g_1).t + \epsilon_2 - \epsilon) + \&c. \end{aligned}$$

This quantity is always less than  $(N + N_1 + N_2 + \&c.)^2$ , when

\* By fixing the origin at the epoch when  $h, h', l, l', \&c.$  are known,  $gt$  vanishes, therefore the coefficients of  $N.m.\sqrt{\bar{a}} + N'^2.m'.\sqrt{\bar{a}'} + \&c.$  are  $\sin. \epsilon, \cos. \epsilon$ .

\* The coefficients by which  $2NN$  is multiplied in the values of  $h^2 + l^2$  are  $\sin. (gt + \epsilon), \sin. (gt + \epsilon), \cos. (gt + \epsilon), \cos. (gt + \epsilon)$ , and the sum of these two =  $\cos. (g - g) \cdot t + \epsilon, - \epsilon$ .

the roots  $g, g_1, \&c.$ , are all real and unequal, the quantities  $N, N_1, \&c.$ , being supposed to be positive. In like manner we shall have

$$\tan. \varpi = \frac{N. \sin. (gt + \epsilon) + N_1. \sin. (g_1 t + \epsilon_1) + N_2. \sin. (g_2 t + \epsilon_2) + \&c.}{N. \cos. (gt + \epsilon) + N_1. \cos. (g_1 t + \epsilon_1) + N_2. \cos. (g_2 t + \epsilon_2) + \&c.}.$$

hence it is easy to infer

$$\tan. (\varpi - gt - \epsilon) = \frac{N_1. \sin. ((g_1 - g). t + \epsilon_1 - \epsilon) + N_2. \sin. ((g_2 - g). t + \epsilon_2 - \epsilon) + \&c.}{N + N_1. \cos. ((g_1 - g). t + \epsilon_1 - \epsilon) + N_2. \cos. ((g_2 - g). t + \epsilon_2 - \epsilon) + \&c.}.$$

When the sum  $N_1 + N_2 + \&c.$  of the coefficients of the cosines of this denominator, taken positively, is less than  $N$ ;  $\tan. (\varpi - gt - \epsilon)$  can never become infinite; therefore the angle  $\varpi - gt - \epsilon$  can never attain the fourth part of a circumference; so that in this case, the true mean motion of the perihelion is equal to  $gt$ .

57. From what precedes it follows, that the excentricities of the orbits, and the positions of the greater axes are subject to considerable variations, which change at length the nature of these orbits, and as their periods depend on the roots  $g, g_1, g_2, \&c.$ , they embrace relatively to the planets, a great number of ages. The excentricities may therefore be considered as of variable ellipticities, and the motions of the perihelions as not altogether uniform. These variations are very consider-

$$* \tan. (\varpi - (gt + \epsilon)) = \frac{\tan. \varpi - \tan. (gt + \epsilon)}{1 + \tan. \varpi. \tan. (gt + \epsilon)} = \frac{\frac{h}{l} - \tan. (gt + \epsilon)}{1 + \frac{h}{l}. \tan. (gt + \epsilon)} =$$

$\frac{h. \cos. (gt + \epsilon) - l. \sin. (gt + \epsilon)}{l. \cos. (gt + \epsilon) + h. \sin. (gt + \epsilon)}$ , now by substituting for  $h$  and  $l$  their values, and observing that  $\sin. (gt + \epsilon). \cos. (gt + \epsilon) - \sin. (g_1 t + \epsilon_1). \cos. (gt + \epsilon) = \sin. ((g_1 - g). t + (\epsilon_1 - \epsilon))$ , the numerator of this fraction becomes  $N. \sin. (gt + \epsilon). \cos. (gt + \epsilon) + N_1. \sin. (g_1 t + \epsilon_1). \cos. (gt + \epsilon) + \&c. - N. \sin. (gt + \epsilon). \cos. (gt + \epsilon) - N_1. \sin. (gt + \epsilon_1). \cos. (g_1 t + \epsilon_1) - \&c. = N_1. \sin. ((g_1 - g). t + (\epsilon_1 - \epsilon)) + N_2. \sin. (g_2 t + \epsilon_2) + \&c.$ , and the denominator becomes  $N. \sin^2 (gt + \epsilon) + N. \cos^2 (gt + \epsilon) + N_1. \sin. (gt + \epsilon). \sin. (g_1 t + \epsilon_1) + N_1. \cos. (gt + \epsilon). \cos. (g_1 t + \epsilon_1) + \&c. = N + N_1. \cos. (g_1 - g). t + (\epsilon_1 - \epsilon) + \&c.$

able in the satellites of Jupiter, and we shall see in the sequel that they explain the remarkable inequalities which are observed in the third satellite. But are there limits to the variations of the excentricities, and do the orbits always differ very little from circles? It is of great moment to investigate this question. We have already observed, that if the roots of the equation in  $g$ , are all real and unequal, the excentricity  $e$  of the orbit of  $m$  is always less than the sum  $N + N_1 + N_2 + \&c.$  of the coefficients of the sines of the expression for  $h$ , taken positively; and as these coefficients are supposed to be very small, the value of  $e$  will be always inconsiderable. It is therefore evident, that if we only consider the secular variations, the orbits of the bodies  $m$ ,  $m'$ ,  $m''$ , &c. will undergo slight changes in their compression, deviating inconsiderably from the circular form; but the positions of the greater axes will experience considerable variations. These axes will be always of the same magnitude, and the mean motions which depend on them will be always uniform, as we have seen in N°. 54. The preceding results, which are founded on the small excentricities of the orbits, will invariably subsist, and may be extended to future and past ages; so that we can affirm, that at any assigned period, the orbits of the planets and of the satellites have not been very excentric, at least, if we only consider their mutual action. But this would not be the case if any of the roots  $g$ ,  $g_1$ ,  $g_2$ , &c., were equal or imaginary: the sines and cosines of the expressions of  $h$ ,  $l$ ,  $h'$ ,  $l'$ , &c, corresponding to these roots, will then be changed into arcs of circles, or into exponentials; and as these quantities increase indefinitely with the time, the orbits will eventually become very excentric; the stability of the planetary system will then be destroyed, and the results to which we have arrived will cease to have place. It is therefore very interesting to determine whether the roots  $g$ ,  $g_1$ ,  $g_2$ , &c., are all real and unequal. This may be demonstrated very simply in the case of nature, in which the bodies  $m$ ,  $m'$ ,  $m''$ , &c., revolve in the same direction.

Resuming the equations (A) of N°. 55, and multiplying the first by

$m \cdot \sqrt{a} \cdot h$ ; the second by  $m \cdot \sqrt{a} \cdot l$ ; the third by  $m' \cdot \sqrt{a'} \cdot h'$ ; the fourth by  $m' \cdot \sqrt{a'} \cdot l'$ , &c., and then adding them together; the coefficients of  $h \cdot l$ ,  $h' \cdot l$ ,  $h'' \cdot l''$ , &c., will vanish in this sum; the coefficient of  $h' \cdot l - h \cdot l'$ , will be  $[0, 1] \cdot m \cdot \sqrt{a} - [1, 0] \cdot m' \cdot \sqrt{a'}$ , and it will be equal to nothing, in virtue of the equation  $[0, 1] \cdot m \cdot \sqrt{a} = [1, 0] \cdot m' \cdot \sqrt{a'}$ , which has been found in N°. 55. The coefficients of  $h'' \cdot l - h \cdot l''$ ,  $h'' \cdot l' - h' \cdot l'$  will vanish for the same reason; therefore the sum of the equations (A) thus prepared, will be reduced to the following equation:

$$\frac{(hdh + ldl)}{dt} \cdot m \cdot \sqrt{a} + \frac{(h'dh' + l'dl')}{dt} \cdot m' \cdot \sqrt{a'} + \text{&c.} = 0;$$

and consequently to the following,

$$0 = ede \cdot m \cdot \sqrt{a} + e'de' \cdot m' \cdot \sqrt{a'} + \text{&c.}$$

By integrating this equation, and remarking that by N°. 54, the greater axes  $a$ ,  $a'$ ,  $a''$ , of the orbits are constant, we will have

$$e^2 \cdot m \cdot \sqrt{a} + e'^2 \cdot m' \cdot \sqrt{a'} + e''^2 \cdot m'' \cdot \sqrt{a''} + \text{&c.} = \text{constant}; \quad (u).$$

Now the bodies  $m$ ,  $m'$ ,  $m''$ , &c., being supposed to revolve in the same direction, the radicals  $\sqrt{a}$ ,  $\sqrt{a'}$ ,  $\sqrt{a''}$ ; &c., ought to be taken positively in the preceding equation, as has been observed in N°. 55; therefore all the terms of the first member of this equation are positive, and consequently each of them is less than the constant of the second member; but if we suppose that at any given epoch, the excentricities are very small, this constant quantity will be very small; therefore each of the terms of the equations will always remain very small, and cannot increase indefinitely; consequently, the orbits will be always very nearly circular.

The case which we have now examined, is that of the planets and of the satellites of the solar system; because all these bodies revolve in the same direction, and the excentricities of their orbits are at this present epoch

very inconsiderable. In order to remove every doubt on this important result, it may be observed that if the equation which determines  $g$ , contains imaginary roots, some of the sines and of the cosines of the expressions of  $h, l, h', l', \&c.$ , will be changed into exponentials; thus the expressions for  $h$  will contain a finite number of terms of the form  $P.c^f$ ,  $c$  being the number of which the hyperbolic logarithm is equal to unity, and  $P$  being a real quantity, because  $h$  or  $e. \sin. \pi$  is a real quantity. Let  $Q.c^n, P'.c^{n'}, Q'.c^n, P''.c^{n''}, \&c.$ , be the corresponding terms of  $l, h', l', h'', \&c.$ ;  $Q, P', Q, P'', \&c.$ , being also real quantities: the expression of  $e^z$  will contain the term  $(P^2+Q^2).c^{2f}$ ; the expression of  $e'^z$  will contain the term  $(P'^2+Q'^2).c^{2n'}$ , and so on of the rest: consequently the first member of the equation ( $u$ ) will contain the term

$$(P^2+Q^2).m.\sqrt{a} + (P'^2+Q'^2).m'.\sqrt{a'} + (P''^2+Q''^2).m''.\sqrt{a''} + \&c.).c^{2f}$$

If  $c^{2f}$  be the greatest of the exponentials which  $h, l, h', l', \&c.$ , contain, that is to say, in which  $f$  is the most considerable;  $c^{2n}$  will be the greatest of the exponentials, which the first member of the preceding equation will contain; therefore the preceding term cannot be destroyed by any other term of this first member; consequently in order that this member may be reduced to a constant quantity, it is necessary that the coefficient of  $c^{2f}$  should vanish, which gives

$$0 = (P^2+Q^2).m.\sqrt{a} + (P'^2+Q'^2).m'.\sqrt{a'} + (P''^2+Q''^2).m''.\sqrt{a''} + \&c.$$

When  $\sqrt{a}, \sqrt{a'}, \sqrt{a''}, \&c.$ , have the same sign, or what comes to the same thing, when the bodies  $m, m', m'', \&c.$ , revolve in the same direction, this equation is impossible, unless we suppose  $P=0, Q=0, P'=0, \&c.$ ; hence it follows, that the quantities  $h, l, h', l', \&c.$ , do not contain exponential quantities, and that consequently the equation in  $g$  does not contain imaginary roots.

If this equation have equal roots, the expressions of  $h, l, h', l', \&c.$ , contain, as we know, arcs of circles, and we would have in the expression for  $h$ , a finite number of terms of the form  $P.r$ . Let  $Q.r$ ,

$P't^r, Q't^r, \&c.$ , be the corresponding terms for  $l, h', l', \&c.$ ;  $P, Q, P', \&c.$ , being real quantities; the first member of the equation in  $u$  will contain the term\*

$$(P^2 + Q^2), m.\sqrt{a} + (P'^2 + Q'^2).m'.\sqrt{a'} + (P''^2 + Q''^2).m''.\sqrt{a''}, \&c.) t^{2r}.$$

If  $t^r$  be the highest power of  $t$ , which the values of  $h, l, h', l', \&c.$  contain;  $t^{2r}$  will be the highest power of  $t$ , contained in the first member of the equation ( $u$ ); thus in order that this member may be reduced to a constant quantity, it is necessary that we have

$$0 = (P^2 + Q^2).m.\sqrt{a} + (P'^2 + Q'^2).m'.\sqrt{a'} + \&c.$$

consequently  $P=0, Q=0, P'=0, Q'=0, \&c.$  It follows therefore that the expressions of  $h, l, h', l', \&c.$ , do not contain either exponential quantities, or arcs of circles, and that consequently all the roots of the equation in  $g$  are real and unequal.

The system of the orbits of  $m, m', m'', \&c.$ , is therefore perfectly stable, relatively to their excentricities; these orbits only oscillate about a mean state of ellipticity, from which they deviate a little, the greater axes remaining the same: their excentricities are always subject to this condition, namely, that the sum of their squares multiplied respectively by the masses of the bodies, and by the square roots of their greater axes is constantly the same.

58. When, by what precedes, the values of  $e$ , and of  $\varpi$  shall have been determined; let them be substituted in all the terms of the expressions for  $r$ , and  $\frac{dv}{dt}$ , which are given in the preceding numbers, the terms which contain the time  $t$ , without the signs *sine* and *cosine*,

\* See Lacroix, tom. 2, No. 613, for the truth of the assertion will be immediately apparent, in the first case, if in place of the sines and cosines their imaginary exponentials be substituted, or if in the second, the equal roots be supposed to differ by very small indeterminate quantities.

being effaced. The elliptic part of these expressions will be the same as in the case of the undisturbed orbit; with this sole difference, that the excentricity and the position of the perihelion will be variable; but the period of these variations being very long, on account of the smallness of the masses  $m, m', m''$ , relatively to  $M$ ; we can suppose these variations proportional to the time, for a long interval, which for the planets may be extended to several ages, before and after the epoch which we select for the origin of the time. It is useful, for astronomical purposes, to have under this form the secular variations of the excentricities of the perihelions of their orbits; they can be easily inferred from the preceding formulae. In fact, the equation  $e^2 = h^2 + l^2$ , gives  $ede = hdh + ldl$ ; and if we only consider the action of  $m'$ , we have, by N°. 55,

$$\frac{dh}{dt} = (0, 1).l - [\underline{0, 1}].l' ;$$

$$\frac{dl}{dt} = -(0, 1).h + [\underline{0, 1}].h' ;$$

therefore

$$\frac{ede}{dt} = [\underline{0, 1}] .(h'l - hl') ;$$

but we have  $h'l - hl' = e.e'. \sin.(\varpi' - \varpi)$ ; therefore we shall have

$$\frac{de}{dt} = [\underline{0, 1}] .e'. \sin.(\varpi' - \varpi) ;$$

consequently, if we only take into account the reciprocal action of the bodies  $m', m''$ , &c. we shall have

$$\frac{de}{dt} = [\underline{0, 1}] .e'. \sin.(\varpi' - \varpi) + [\underline{0, 2}] .e''. \sin.(\varpi'' - \varpi) + \&c.$$

$$\frac{de'}{dt} = [\underline{1, 2}] .e. \sin.(\varpi - \varpi') + [\underline{1, 2}] .e''. \sin.(\varpi'' - \varpi') + \&c.$$

\*  $h = e. \sin. \varpi$ ,  $h' = e'. \sin. \varpi'$ ;  $l = e. \cos. \varpi$ ;  $l' = e'. \cos. \varpi'$ ,  $\therefore h'l - hl' = ee'. \sin.(\varpi - \varpi')$ .

$$\frac{d\omega''}{dt} = [\underline{\underline{2,0}}] \cdot e \cdot \sin. (\omega - \omega'') + [\underline{\underline{2,1}}] \cdot e' \cdot \sin. (\omega' - \omega'' + \&c.) ;$$

&c.

The equation  $\tan. \omega = \frac{h}{l}$ , gives by differenceing it

$$e^2 \cdot d\omega = l \cdot dh - h \cdot dl.$$

If the action of  $m'$ , be only considered, by substituting for  $dh$  and  $dl$  their values, we shall have

$$\frac{e^2 d\omega}{dt} = (0, 1) \cdot (h^2 + l^2) - [\underline{\underline{0,1}}] \cdot (hh' + ll') ; *$$

which gives

$$\frac{d\omega}{dt} = (0, 1) - [\underline{\underline{0,1}}] \cdot \frac{e'}{e} \cdot \cos. (\omega' - \omega) ;$$

therefore we shall have, in consequence of the reciprocal actions of the bodies,  $m, m', m'', \&c.$ ;

$$\frac{d\omega}{dt} = (0, 1) + (0, 2) + \&c. - [\underline{\underline{0,1}}] \cdot \frac{e'}{e} \cdot \cos. (\omega' - \omega) -$$

$$[\underline{\underline{0,2}}] \cdot \frac{e''}{e} \cdot \cos. (\omega'' - \omega) - \&c. ;$$

$$\frac{d\omega'}{dt} = (1, 0) + (1, 2) + \&c. - [\underline{\underline{1,0}}] \cdot \frac{e}{e'} \cdot \cos. (\omega - \omega') -$$

$$[\underline{\underline{1,2}}] \cdot \frac{e''}{e'} \cdot \cos. (\omega'' - \omega') - \&c. ;$$

\*  $\frac{d\omega}{\cos. \omega} = \frac{l dh - h dl}{l^2}$ ; but from the equation  $\tan. \omega = \frac{h}{l}$ , we have  $\cos. \omega = \frac{l^2}{h^2 + e^2} = \frac{l^2}{e^2}$ ,  $\therefore$  by substituting we have the expression in the text.

$$\frac{d\varpi''}{dt} = (2, 0) + (2, 1) + \text{&c.} - [\underline{2, 0}] \cdot \frac{e}{e''} \cdot \cos(\varpi - \varpi'') -$$

$$[\underline{2, 1}] \cdot \frac{e'}{e''} \cdot \cos(\varpi' - \varpi'') - \text{&c.}$$

These values of  $\frac{de}{dt}$ ,  $\frac{de'}{dt}$ , &c.;  $\frac{d\varpi}{dt}$ ,  $\frac{d\varpi'}{dt}$ , &c., being multiplied by the time  $t$ , the differential expressions of the secular variations of the excentricities and of the perihelions will be had; and these expressions, which are only rigorously true, when  $t$  is indefinitely small, can however serve for a long interval, relatively to the planets. Their comparison with accurate observations, which are made at considerable intervals from each other, is the most exact means of determining the masses of the planets, which have no satellites. For any time  $t$ , the excentricity  $e$  is equal to  $e + t \cdot \left( \frac{de}{dt} \right) + \frac{t^2}{1.2} \cdot \frac{d^2 e}{dt^2} + \text{&c.}$ ;  $e$ ,  $\frac{de}{dt}$ ,  $\frac{d^2 e}{dt^2}$ , &c. being relative to the origin of the time  $t$ , or to the epoch. The preceding value of  $\frac{de}{dt}$  will give by differencing it, and by observing that  $a$ ,  $a'$ , &c., are constant, the values of  $\frac{d^2 e}{dt^2}$ ,  $\frac{d^3 e}{dt^3}$ , &c., we can therefore continue as far as we please the preceding series, and by a similar process, the series relative to  $\varpi$ ; but in the case of the planets, it will be sufficient, in the comparison of the most ancient observations of which we are in possession, to take into account the square of the time, in the expressions in series of  $e$ ,  $e'$ , &c.,  $\varpi$ ,  $\varpi'$ , &c.

59. Let us now consider the equations relative to the position of the orbits, and for this purpose let the equations (3) and (4) of N°. 53, be resumed,

$$\frac{dp}{dt} = \frac{m'n}{4} \cdot a^2 a' \cdot B^{(1)} \cdot (q - q');$$

$$\frac{dq}{dt} = \frac{m'n}{4} \cdot a^2 a' \cdot B^{(1)} \cdot (p - p').$$

By No. 49, we have,

$$a^2 a' \cdot B^{(1)} = \alpha^2 \cdot \xi_{\frac{3}{2}}^{(1)};$$

and by the same number we have

$$b_{\frac{3}{2}}^{(1)} = - \frac{3 \cdot b_{\frac{1}{2}}^{(1)}}{(1 - \alpha^2)^2};$$

therefore we shall have

$$\frac{m'n}{4} \cdot a^2 a' \cdot B^{(1)} = - \frac{3m' \cdot n \cdot \alpha^2 \xi_{\frac{1}{2}}^{(1)}}{4 \cdot (1 - \alpha^2)^2}.$$

The second member of this equation is that which we have designated by (0, 1) in N°. 55; consequently we shall have

$$\frac{dp}{dt} = (0, 1) \cdot (q' - q);$$

$$\frac{dq}{dt} = (0, 1) \cdot (p' - p);$$

hence it is easy to infer, that the values of  $q, p, q', p', \&c.$ , will be determined by the following system of differential equations,

$$\left. \begin{aligned} \frac{dq}{dt} &= ((0, 1) + (0, 2) + \&c.) \cdot p - (0, 1) \cdot p' - (0, 2) \cdot p'' - \&c. \\ \frac{dp}{dt} &= -((0, 1) + (0, 2) + \&c.) \cdot q + (0, 1) \cdot q' + (0, 2) \cdot q'' + \&c. \\ \frac{dq'}{dt} &= ((1, 0) + (1, 2) + \&c.) \cdot p' - (1, 0) \cdot p - (1, 2) \cdot p'' - \&c. \\ \frac{dp'}{dt} &= -((1, 0) + (1, 2) + \&c.) q' + (1, 0) \cdot q + (1, 2) \cdot q'' + \&c. \\ \frac{dq''}{dt} &= ((2, 0) + (2, 1) + \&c.) \cdot p'' - (2, 0) \cdot p - (2, 1) \cdot p' - \&c. \\ \frac{dp''}{dt} &= -((2, 0) + (2, 1) + \&c.) q'' + (2, 0) \cdot q + (2, 1) \cdot q' + \&c. \\ &\&c. \end{aligned} \right\} \quad (C)$$

This system of equations is similar to that of the equations (A) of N°. 55; it would coincide altogether with it, if in the equations A,  $h$ ,  $l$ ,  $h'$ ,  $l'$ , &c., be changed into  $q$ ,  $p$ ,  $q'$ ,  $p'$ , &c., and if we suppose  $[0, 1] = (0, 1)$ ;  $[1, 0] = (1, 0)$ , &c.; consequently, the analysis which we have employed in N°. 56, in order to integrate the equation (A), is applicable to the equations (C). Therefore let us suppose

$$\begin{aligned} q &= N \cdot \cos. (gt + \epsilon) + N_1 \cdot \cos. (g_1 t + \epsilon_1) + N_2 \cdot \cos. (g_2 t + \epsilon_2) + \&c.; \\ p &= N \cdot \sin. (gt + \epsilon) + N_1 \cdot \sin. (g_1 t + \epsilon_1) + N_2 \cdot \sin. (g_2 t + \epsilon_2) + \&c.; \\ q' &= N' \cdot \cos. (gt + \epsilon) + N'_1 \cdot \cos. (g_1 t + \epsilon_1) + N'_2 \cdot \cos. (g_2 t + \epsilon_2) + \&c.; \\ p' &= N' \cdot \sin. (gt + \epsilon) + N'_1 \cdot \sin. (g_1 t + \epsilon_1) + N'_2 \cdot \sin. (g_2 t + \epsilon_2) + \&c.; \\ &\quad \&c. \end{aligned}$$

and by the method given in N°. 56, an equation in  $g$  of the degree  $i$ , may be obtained, of which the different roots will be  $g$ ,  $g_1$ ,  $g_2$ , &c. It is easy to see that one of these roots vanishes, because the equations (C) will be satisfied by supposing  $p$ ,  $p'$ ,  $p''$ , &c., equal and constant, and also  $q$ ,  $q'$ ,  $q''$ , &c., but this requires that one of the roots of the equation in  $g$  should vanish, and thus the equation is depressed to the degree  $i-1$ . The arbitrary quantities  $N$ ,  $N_1$ ,  $N_2$ , &c.,  $\epsilon$ ,  $\epsilon_1$ ,  $\epsilon_2$ , &c., may be determined by the method detailed in N°. 56. Finally, by an analysis similar to that of No. 57, we shall find

$$\text{const.} = (p^2 + q^2) \cdot m \cdot \sqrt{\bar{a}} + (p'^2 + q'^2) \cdot m' \cdot \sqrt{\bar{a}'} + \&c.;$$

from which may be inferred, as in the above cited N°. that the expressions of  $p$ ,  $q$ ,  $p'$ ,  $q'$ , &c., do not contain either arcs of a circle, or exponential quantities, when the bodies  $m$ ,  $m'$ ,  $m''$ , &c., revolve in the same direction: and that consequently all the roots of the equation in  $g$ , are real and unequal.

Two other integrals of the equations in  $C$  may be obtained. In fact, if the first of these equations be multiplied by  $m \cdot \sqrt{\bar{a}}$ , the third by  $m' \cdot \sqrt{\bar{a}'}$ , the fifth by  $m'' \cdot \sqrt{\bar{a}''}$ , &c., we shall have in consequence of the relations found in N°. 55,

$$0 = \frac{dq}{dt} \cdot m \cdot \sqrt{a} + \frac{dq'}{dt} \cdot m' \cdot \sqrt{a'} + \text{&c.}$$

which being integrated, gives

$$\text{constant} = q \cdot m \cdot \sqrt{a} + q' \cdot m' \cdot \sqrt{a'} + \text{&c.} \quad (1)$$

In the same manner we shall find

$$\text{constant} = p \cdot m \cdot \sqrt{a} + p' \cdot m' \cdot \sqrt{a'} + \text{&c.} \quad (2)$$

Naming  $\phi$  the inclination of the orbit of  $m$ , on the fixed plane, and  $\theta$  the longitude of the ascending node of this orbit on the same plane; the latitude of  $m$  will be very nearly,  $\tan. \phi \cdot \sin. (nt + \epsilon - \theta)$ . By comparing\* this value, with the following,  $q \cdot \sin. (nt + \epsilon) - p \cdot \cos. (nt + \epsilon)$  we will have

$$p = \tan. \phi \cdot \sin. \theta; \quad q = \tan. \phi \cdot \cos. \theta;$$

hence we deduce

$$\tan. \phi = \sqrt{p^2 + q^2}; \quad \tan. \theta = \frac{p}{q};$$

therefore the inclination of the orbit of  $m$ , and the position of its node, may be obtained by means of the value of  $p$  and of  $q$ . If we denote successively by one stroke, two strokes, &c., relatively to  $m'$ ,  $m''$ , &c., the values of  $\tan. \phi$ , and of  $\tan. \theta$ , the inclinations of the orbits of

x x 2

\*  $d \cdot \tan. \varpi \left( = \frac{d\varpi}{\cos. \varpi} \right) = \frac{l dh - h dl}{l^2}, \because \text{as } l^2 = e^2 \cdot \cos. \varpi, \text{ we obtain } e^2 d\varpi = ldh - hdl;$   
 as  $hh' + ll' = ee' \cdot \cos. (\varpi' - \varpi)$ ; if  $\frac{e^2 d\varpi}{dt} = (0, 1) \cdot (h^2 + l^2) - [0, 1] \cdot (hh' + ll')$ , be divided by  
 $e^2 = h^2 + l^2$  we shall have the value of  $\frac{d\varpi}{dt}$  which is given in the text.

\* This is the value of  $s$ , or of the latitude very nearly, when periodic quantities are neglected, in fact the values of  $\phi$  and  $\theta$ , which are derived from a comparison of the two values of  $s$ , are the mean values, only affected with secular inequalities; see N°. 53.

$m'$ ,  $m''$ , &c., and the positions of their nodes, will be had by means of the quantities  $p'$ ,  $q'$ ,  $p''$ ,  $q''$ , &c.

The quantity  $\sqrt{p^2 + q^2}$ , is less than the sum  $N, N_1, N_2, + \&c.$  of the coefficients of the sines of the expression for  $q$ ; therefore these coefficients being very small, because by hypothesis, the orbit is inclined by a very small angle to the fixed plane, its inclination to this plane will be always inconsiderable; hence it follows, that the system of the orbits is always stable relative to their inclinations, as well as relative to their excentricities. The inclinations of the orbits may therefore be considered as variable quantities comprised between determinate limits, and the motions of the nodes as not being altogether uniform. These variations are very sensible in the satellites of Jupiter, and we shall see in the sequel that they explain the singular phenomena, which are observed in the inclination of the orbit of the fourth satellite.

From the preceding expressions for  $p$  and  $q$ , results the following theorem :

That if a circle be conceived, of which the inclination to the fixed plane is  $N$ , and of which  $gt+\epsilon$  is the longitude of its ascending node; and if on this first circle a second be conceived inclined to it by an angle equal to  $N$ ,  $g,t+\epsilon$ , being the longitude of its intersection with the second circle, and so of the rest; the position of the last circle will be that of the orbit of  $m$ .

The same construction being applied to the expressions of  $h$  and of  $l$  of N°. 56; it will appear that the tangent of the inclination of the last circle on the fixed plane, is equal to the excentricity of the orbit of  $m$ , and that the longitude of the intersection of this circle with the same plane, is equal to that of the perihelion of the orbit of  $m$ .

60.\* It is useful for astronomical purposes to obtain the differential variations of the nodes and of the inclinations of the orbits. For this purpose let the equations of the preceding N°. be resumed, namely,

\* It should be observed, that the differential expressions which are given in this N°., are relative to the secular variations of the nodes and of the inclinations of the orbits.

$$\text{tang. } \phi = \sqrt{p^2 + q^2}; \quad \text{tang. } \theta = \frac{p}{q}.$$

By differentiating, there will be obtained,

$$d\phi = dp \cdot \sin. \theta + dq \cdot \cos. \theta; *$$

$$d\theta = \frac{dp \cdot \cos. \theta - dq \cdot \sin. \theta}{\text{tang. } \phi}.$$

Substituting for  $dp$  and  $dq$ , their values, which have been given by the equations (C) of the preceding N°., we will have

$$\frac{d\phi}{dt} = (0, 1) \cdot \text{tang. } \phi' \cdot \sin. (\theta - \theta') + (0, 2) \cdot \text{tang. } \phi'' \cdot \sin. (\theta - \theta'') + \&c.$$

$$\begin{aligned} \frac{d\theta}{dt} = & -((0, 1) + (0, 2) + \&c.) + (0, 1) \cdot \frac{\text{tang. } \phi'}{\text{tang. } \phi} \cdot \cos. (\theta - \theta') \\ & + (0, 2) \cdot \frac{\text{tang. } \phi''}{\text{tang. } \phi} \cdot \cos. (\theta - \theta'') + \&c.; \end{aligned} *$$

\*  $d. \tan. \phi = \frac{d\phi}{\cos. \phi} = \frac{pdः + qdq}{\sqrt{p^2 + q^2}} =$  by substituting for  $p$  and  $q$  their values, and neglecting  $\frac{1}{\cos. \phi}$ , the expression in the text.  $d. \tan. \theta = \frac{d\theta}{\cos. \theta} = d\theta \left( \frac{p^2 + q^2}{q^2} \right) = \frac{q.dp - p.dq}{q^2}$ , which becomes, by substituting for  $p$  and  $q$  their values, and multiplying by  $q^2$ ,  $d\theta. \tan. \theta = dp \cdot \cos. \theta \cdot \tan. \phi - dq \cdot \sin. \theta \cdot \tan. \phi$ .

† When this substitution is made, the first term in the expression for  $dp \cdot \sin. \theta$  becomes equal to the first term of the expression for  $dq \cdot \cos. \theta$ , and affected with a contrary sign, consequently they destroy each other. The second terms of these expressions are respectively  $(0, 1) \cdot \tan. \phi' \cdot \cos. \theta' \cdot \sin. \theta - (0, 1) \cdot \tan. \phi' \cdot \sin. \theta' \cdot \cos. \theta = (0, 1) \cdot \tan. \phi' \cdot \sin. (\theta - \theta')$ ; by a similar process the third and following terms are obtained. The first term in the value of  $dp \cdot \cos. \theta = -((0, 1) + (0, 2) + (0, 3) + (0, 4) + \&c.) \tan. \phi \cdot \cos. \theta$ , and the second term  $= (0, 1) \cdot \tan. \phi' \cdot \cos. \theta' \cdot \cos. \theta$ ; in like manner the first term of the value of  $-dq \cdot \sin. \theta = -((0, 1) + (0, 2) + (0, 3) + \&c.) \tan. \phi \cdot \sin. \theta$ , and the second term  $= -(0, 1) \cdot \tan. \phi' \cdot \sin. \theta' \cdot \sin. \theta$ , &c., by making these terms respectively to coalesce, they become  $-((0, 1) + (0, 2) + (0, 3) + \&c.) + (0, 1) \cdot \frac{\tan. \phi'}{\tan. \phi} \cdot (\cos. \theta \cdot \cos. \theta' + \sin. \theta \cdot \sin. \theta')$ . If there are only

In like manner we will have

$$\begin{aligned}\frac{d\phi'}{dt} &= (1, 0) \cdot \text{tang. } \varphi \cdot \sin. (\theta' - \theta) + (1, 2) \cdot \text{tang. } \varphi'' \cdot \sin. (\theta' - \theta'') + \text{&c.}; \\ \frac{d\theta}{dt} &= -(1, 0) + (1, 2) + \text{&c.} + (1, 0) \cdot \frac{\text{tang. } \varphi}{\text{tang. } \varphi'} \cdot \cos. (\theta' - \theta) \\ &\quad + (1, 2) \cdot \frac{\text{tang. } \varphi''}{\text{tang. } \varphi'} \cdot \cos. (\theta' - \theta'') + \text{&c.} \\ &\text{&c.}\end{aligned}$$

Astronomers refer the celestial motions to the moveable orbit of the earth; in fact, it is from the plane of this orbit that they are observed; it is therefore of consequence to know the variations of the nodes and of the inclinations of the orbits, with respect to the ecliptic. Suppose, therefore, that it were required to determine the differential variations of the nodes, and of the inclinations of the orbits, with respect to the orbit of one of the bodies  $m, m', m'', \text{ &c.}$ , for example, relatively to the orbit of  $m$ . It is evident that  $q \cdot \sin. (n't + \epsilon') - p \cdot \cos. (n't' + \epsilon')$  will be the latitude of  $m'$  above the fixed plane, if it was in motion on the orbit of  $m$ . Its latitude above the same plane, is  $q' \cdot \sin. (n't + \epsilon') - p' \cdot \cos. (n't + \epsilon')$ ; now the difference of those two latitudes, is very nearly the latitude of  $m'$  above the orbit of  $m$ ; therefore  $\varphi'$  representing the inclination,  $\theta'$  being the longitude of the node of the orbit of  $m'$  on the orbit of  $m$ , by what goes before

two bodies  $m, m'$ , the nodes of each of them will regrade on the fixed ecliptic, when  $\frac{\tan. \varphi}{\tan. \varphi'} \cdot \cos. (\theta' - \theta), \frac{\tan. \varphi'}{\tan. \varphi} \cdot \cos. (\theta - \theta')$  are respectively less than unity; if one of them, as, for instance, the first, be greater than unity, this can only arise from  $\tan. \varphi$  being greater than  $\tan. \varphi'$ , therefore the second must be less than unity; consequently, the nodes of one of the orbits must *always* regrade. It appears also from this expression, that if the distance between the ascending nodes of the two planets be greater than  $90^\circ$ , the nodes must regrade. It is likewise evident that if  $\theta - \theta$  is greater than  $180$ , the inclination increases, and that it diminishes when this inclination is less than  $180$ ; the variation is greater according as the distance between the nodes approaches to  $90$ , and according as  $\varphi$  increases. See Princep. Matth. Lib. I. Prop. 66, Cor. 11.

there will be obtained

$$\text{tang. } \phi' = \sqrt{(p'-p)^2 + (q'-q)^2}; \quad \text{tang. } \theta' = \frac{p'-p}{q'-q}.$$
\*

If the fixed plane be assumed to be that of the orbit of  $m$ , at a given epoch; for this epoch  $p$  and  $q$  will be respectively  $= 0$ ; however the differentials  $dp$ ,  $dq$ , will not vanish; thus we shall have

$$d\phi' = (dp' - dp) \cdot \sin. \theta' + (dq' - dq) \cdot \cos. \theta'; \quad \dagger$$

$$d\theta' = \frac{(dp' - dp) \cdot \cos. \theta' - (dq' - dq) \cdot \sin. \theta'}{\text{tang. } \phi'}.$$

By substituting for  $dp$ ,  $dq$ ,  $dp'$ ,  $dq'$ , &c., their values given by the equation (C) of the preceding N°, there will be obtained

$$\frac{d\phi'}{dt} = ((1, 2) - (0, 2)) \cdot \text{tang. } \phi'' \cdot \sin. (\theta' - \theta'') \ddagger$$

\* Neglecting quantities of the second and higher orders, the differences of the expressions for the tangents of these latitudes, which in the present case may be substituted for the latitudes themselves, is equal to  $(q' - q) \cdot \sin. (n't + \epsilon) - (p' - p) \cdot \cos. (n't + \epsilon) = \tan. \phi' \cdot \sin. (n't + \epsilon - \theta')$ ,  $\because q - q' = \tan. \phi' \cdot \cos. \theta'$ ;  $p' - p = \tan. \phi' \cdot \sin. \theta'$ ; hence we get the values of  $\phi'$ , and  $\tan. \theta'$  as before.

†  $d. \tan. \phi' = d\phi' = (\text{as } p \text{ and } q \text{ vanish}) \frac{(dp' - dp) \cdot p' + (dq' - dq) \cdot dq'}{\sqrt{p'^2 + q'^2}}$ , which by substituting for  $p'$  and  $q'$ , their values  $\tan. \phi' \cdot \sin. \theta'$ ,  $\tan. \phi' \cdot \cos. \theta'$ , becomes the expression in the text. Similarly by substituting  $\frac{d\theta'}{\cos. \theta'} = \frac{(dp' - dp) \cdot q' - (dq' - dq) \cdot p'}{q'^2}$ ; but  $\frac{1}{\cos. \theta'} = \frac{q'^2 + p'^2}{q'^2}$ ,  $\therefore d\theta' = \frac{(dp' - dp) \cdot \cos. \theta' - (dq' - dq) \cdot \sin. \theta'}{\tan. \phi'}.$

‡  $dp' \cdot \sin. \theta' = -(1, 0) + (1, 2) + \&c. \tan. \phi' \cdot \cos. \theta' \cdot \sin. \theta' + (1, 2) \cdot \tan. \phi'' \cdot \cos. \theta'' \cdot \sin. \theta' + \&c.; -dp \cdot \sin. \theta' = -(0, 1) \cdot \tan. \phi' \cdot \cos. \theta' \cdot \sin. \theta' - (0, 2) \cdot \tan. \phi'' \cdot \cos. \theta'' \cdot \sin. \theta' - \&c.;$   
 $dq' \cdot \cos. \theta' = (1, 0) + (1, 2) + \&c. \tan. \phi' \cdot \sin. \theta' \cdot \cos. \theta' - (1, 2) \cdot \tan. \phi'' \cdot \sin. \theta'' \cdot \cos. \theta' - \&c.) - dq \cdot \cos. \theta' = (0, 1) \cdot \tan. \phi' \cdot \sin. \theta' \cdot \cos. \theta' + (0, 2) \cdot \tan. \phi'' \cdot \cos. \theta' + \&c.;$  hence, obliterating the terms which destroy each other, and making corresponding factors of  $\tan. \phi''$ ,  $\tan. \phi'''$ , &c., to coalesce, we obtain the expressions which are giveen in the text. Since  $p$  and  $q = 0$ , the coefficients of these terms are neglected in the vhlue of  $\frac{dp}{dt}$ ,  $\frac{dq}{dt}$ ,  $\frac{dp'}{dt}$ , &c.

$$((1, 3)-(0, 3)) \cdot \text{tang. } \phi'' \cdot \sin. (\theta' - \theta'') + \text{ &c.}$$

$$\frac{d\theta'}{dt} = -((1, 0) + (1, 2) + (1, 3) + \text{ &c.}) - (0, 1)$$

$$+ ((1, 2) - (0, 2)) \frac{\text{tang. } \phi''}{\text{tang. } \phi'} \cdot \cos. (\theta' - \theta'')$$

$$+ ((1, 3) - (0, 3)) \frac{\text{tang. } \phi''}{\text{tang. } \phi'} \cdot \cos. (\theta' - \theta'') + \text{ &c.}$$

It is easy to infer from these expressions, the variations of the nodes and of the inclinations of the orbits of the other bodies,  $m'', m''', \text{ &c.}$ , on the moveable orbit of  $m$ .

61. The integrals previously found of the differential equations which determine the variations of the elements of the orbits, are only approximative, and the relations which they indicate between all these elements, have place only on the hypothesis that the excentricities of the orbits and their inclinations are very small. But the integrals (4), (5), (6) and (7), to which we have arrived in N°. 9, give the same relations, whatever may be the excentricities and the inclinations. For this purpose, it may be observed, that  $\frac{xdy - ydx}{dt}$  is double of the area described in the time  $dt$ , by the projection of the radius vector of the planet  $m$ , on the plane of  $x$  and of  $y$ . In the elliptic motion, if the mass of the planet be neglected, relatively to that of the sun, which is assumed equal to unity, we have by N°. 19 and 20, relatively to the plane of the orbit of  $m$ ,

$$\frac{xdy - ydx}{dt} = \sqrt{a \cdot (1 - e^2)^2}.$$

In order to refer the area of the orbit to a fixed plane, it is necessary to multiply it by the cosine of the inclination of  $\phi$ , of the orbit to this plane ; therefore with respect to this plane we will have

$$\frac{xdy-ydx}{dt} = \cos. \varphi. \sqrt{a.(1-e^2)} = \sqrt{\frac{a.(1-e^2)}{1+\tan^2 \varphi}}.$$

In like manner we have

$$\frac{x'dy'-y'dx'}{dt} = \sqrt{\frac{a'.(1-e'^2)}{1+\tan^2 \varphi'}};$$

&c.

These values of  $xdy-ydx$ ,  $x'dy'-y'dx'$ , &c., may be employed, when we do not take into account the inequalities of the motion of the planets, provided that the elements  $e$ ,  $e'$ , &c.,  $\varphi$ ,  $\varphi'$ , &c., are considered as variable, in consequence of the secular inequalities; therefore the equation (4) of N°. (9). will then give

$$c = m. \sqrt{\frac{a.(1-e^2)}{1+\tan^2 \varphi}} + m'. \sqrt{\frac{a'.(1-e'^2)}{1+\tan^2 \varphi'}} + \text{&c.}$$

$$+ \Sigma mm'. \left\{ \frac{(x'-x).(dy'-dy)-(y'-y).(dx'-dx)}{dt} \right\}.$$

This last term, which is always of the order  $mm'$ , being neglected, we shall have

$$c = m. \sqrt{\frac{a.(1-e^2)}{1+\tan^2 \varphi}} + m'. \sqrt{\frac{a'.(1-e'^2)}{1+\tan^2 \varphi'}} + \text{&c.}$$

Therefore, whatever changes may be produced in the progress of time, in the values of  $e$ ,  $e'$ , &c.,  $\varphi$ ,  $\varphi'$ , &c., in consequence of the secular variations; these values ought always to satisfy the preceding equation.

If the very small quantities of the order  $e^4$ ,  $e^2\varphi^2$ , be neglected, this equation will give

$$c = m. \sqrt{a} + m'. \sqrt{a'} + \text{&c.} - \frac{1}{2}m. \sqrt{a.(e^2 + \tan^2 \varphi)}$$

$$- \frac{1}{2}m'. \sqrt{a'.(e'^2 + \tan^2 \varphi')} - \text{&c.};$$

and consequently, if the squares of  $e, e', \varphi, \&c.$ , be neglected, we shall have  $m.\sqrt{a} + m'.\sqrt{a'} + \&c.$ , constant. It has appeared already, that if only the first powers\* of the disturbing force be taken into account,  $a, a', \&c.$ , will be separately constant; the preceding equation will therefore give, when the very small quantities of the order  $e^4$ , or  $e^2\varphi^2$ , are neglected,

$$\text{const.} = m.\sqrt{a}.(e^2 + \text{tang. } {}^2\varphi^2) + m'.\sqrt{a'}.(e'^2 + \text{tang. } {}^2\varphi') + \&c.;$$

on the supposition that the orbits are very nearly circular, and inclined to each other at small angles, the secular variations of the excentricities of the orbits, are by N°. 55, determined by means of differential equations independent of the inclinations, and which are therefore the same as if the orbits existed in the same plane; but on this hypothesis,  $\varphi = 0, \varphi' = 0, \&c.$ ; consequently, the preceding equation becomes

$$\text{const.} = e^2.m.\sqrt{a} + e'^2.m'.\sqrt{a'} + e''^2.m''.\sqrt{a''} + \&c.$$

this equation has been already obtained in N°. 57.

In like manner, the secular variations of the inclinations of the orbits, are by N°. 59, determined by means of differential equations independent of the excentricities, and which are therefore the same as if the orbits were circular; but on this hypothesis,  $e = 0, e' = 0, \&c.$ ; therefore

$$\text{const.} = m.\sqrt{a}.\text{tang. } \varphi^2 + m'.\sqrt{a'}.\text{tang. } \varphi'^2 + m''.\sqrt{a''}.\text{tang. } \varphi''^2 + \&c.$$

which equation has been obtained in N°. 59.

If we suppose, as in this last number, that

$$p = \text{tang. } \varphi. \sin. \theta; \quad q = \text{tang. } \varphi. \cos. \theta;$$

It is easy to be assured, when the inclination of the orbit of  $m$ , on the plane of  $x$  and  $y$ , is  $\varphi$ ,  $\theta$  being the longitude of its ascending node, reckoned from the axis of  $x$ ; that the cosine of the inclination of

\* See No. 54, page 324.

this orbit on the plane of  $x$  and  $z$ , will be

$$\frac{q}{\sqrt{1 + \tan^2 \phi}}.$$

This quantity being multiplied by  $\frac{x dy - y dx}{dt}$ , or by its equivalent value  $\sqrt{a.(1-e^2)}$ , the value of  $\frac{x dz - z dx}{dt}$ , will be obtained; therefore the equation (5) of N°. 9, will give, when quantities of the order  $m^*$  are neglected,

$$c = m.q.\sqrt{\frac{a.(1-e^2)}{1 + \tan^2 \phi}} + m'.q'.\sqrt{\frac{a'.(1-e'^2)}{1 + \tan^2 \phi'}} + \&c.$$

In like manner the equation (6) of N°. 9, will give

$$c'' = mp.\sqrt{\frac{a.(1-e^2)}{1 + \tan^2 \phi}} + m'.p'.\sqrt{\frac{a'.(1-e'^2)}{1 + \tan^2 \phi'}} + \&c.$$

If quantities of the order  $e^3$ , or  $e^2\phi$ , are neglected in these two equations; they become

$$\text{constant} = mq.\sqrt{a} + m'q'.\sqrt{a'} + \&c.†$$

$$\text{constant} = mp.\sqrt{a} + m'p'.\sqrt{a'} + \&c.$$

which equations have been already obtained in N°. 59.

Finally, the equation (7) of No. 9, when quantities of the order

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\*  $\cos. \phi = \frac{1}{\sqrt{1 + \tan^2 \phi}}$ , ∵ by substituting for  $q$  its value  $\tan. \phi. \cos. \theta$ , we obtain

$\frac{q}{\sqrt{1 + \tan^2 \phi}} = \sin. \phi. \cos. \theta$ , which is the cosine of the inclination of the orbit of  $m$  to the plane  $x, z$ .

†  $\tan. \phi, \tan. \phi'$ , being of the order  $e$ , the quantities which are neglected are of the order  $e^3$ , and of higher orders.

$mm'$  are neglected, will give, by remarking that by N°. 18,  $\frac{\mu}{a} = \frac{2\mu}{r} - \frac{(dx^2 + dy^2 + dz^2)}{dt^2}$ ,

$$\text{constant} = \frac{m}{a} + \frac{m'}{a'} + \frac{m''}{a''} + \&c.^*$$

These different equations subsist with respect to those inequalities of very long periods, which may affect the elements of the orbits of  $m$ ,  $m'$ , &c. It has been remarked in N°. 54, that the relation of the mean motions of these bodies may introduce into the expressions of the greater axes of the orbits considered as variable, inequalities of which the arguments being proportional to the time, increase with great slowness, and which as they have for divisors the coefficients of the time  $t$ , may at length become sensible. But it is evident, that if we only take into account the terms which have similar divisors, the orbits being considered as ellipses of which the elements vary in consequence of these terms, the integrals (4), (5), (6) and (7) of N°. 9, will always give the relations which we have found between these elements ; because that the terms of the order  $mm'$  which have been neglected in order to infer these relations, have not for divisors the very small coefficients of which we have spoken ; or at least, they only contain them multiplied by a power of the disturbing force, superior to that which has been taken into account.

62. It has been remarked in N°s. 21 and 22, of the first book, that in the motion of a system of bodies, there exists an invariable plane,

\* When quantities of the order  $mm'$  are neglected, we have ( $M$  being considered as unity)  $h = \Sigma m \cdot \frac{dx^2 + dy^2 + dz^2}{dt^2} - \frac{2\Sigma m}{r}$ , now we have  $\mu = 1 + m$ , ∵ if the expression  $\frac{\mu}{a} = \frac{2\mu}{r} - \left( \frac{dx^2 + dy^2 + dz^2}{dt^2} \right)$  be multiplied by  $m$ , it will give when quantities of the order  $m^2$  are neglected,  $\frac{m}{a} = \frac{2m}{r} - m \cdot \frac{(dx^2 + dy^2 + dz^2)}{dt^2}$ , ∵ by making similar substitutions for the bodies  $m'$ ,  $m''$ , &c., we obtain the expression which is given in the text.

which preserves always a parallel position, and which the following condition enable us to find easily, at all times, namely, that the sum of the masses of the system respectively multiplied by the projections of the areas described by the radii vectores in a given time, is a *maximum*. It is principally in the theory of the solar system, that the investigation of this plane is important, in consequence of the proper motions of the stars, and of the ecliptic, which render the exact determination of the celestial motions a matter of great difficulty to astronomers. Naming  $\gamma$  the inclination of this invariable plane, to the plane of  $x$  and of  $y$ , and  $\Pi$  the longitude of its ascending node, it follows, from what has been demonstrated in Nos. 21 and 22, of the first book, that we will have

$$\tan. \gamma. \sin. \Pi = \frac{c''}{c}; \quad \tan. \gamma. \cos. \Pi = \frac{c'}{c};$$

consequently,

$$\tan. \gamma. \sin. \Pi = \frac{m. \sqrt{a.(1-e^2)}. \sin. \phi. \sin. \theta + m'. \sqrt{a'.(1-e'^2)}. \sin. \phi'. \sin. \theta' + \text{&c.}}{m. \sqrt{a.(1-e^2)}. \cos. \phi + m'. \sqrt{a'.(1-e'^2)}. \cos. \phi' + \text{&c.}} *$$

$$\tan. \gamma. \cos. \Pi = \frac{m. \sqrt{a.(1-e^2)}. \sin. \phi. \cos. \theta + m'. \sqrt{a'.(1-e'^2)}. \sin. \phi'. \cos. \theta' + \text{&c.}}{m. \sqrt{a.(1-e^2)}. \cos. \phi + m'. \sqrt{a'.(1-e'^2)}. \cos. \phi' + \text{&c.}} *$$

The two angles  $\gamma$  and  $\Pi$  may be easily determined, by means of these values. It is evident that in order to determine accurately the invariable plane, it is necessary to know the masses of the comets and the elements of their orbits; fortunately, their masses appear to be very small, so that their action on the planets may be neglected without any sensible error; but time will give us fuller information on this point. It may be remarked here, that with respect to the invariable\* plane, the values of  $p$ ,  $q$ ,  $p'$ ,  $q'$ , &c. do not contain constant terms; for it is

\* Since  $\frac{1}{1+\tan. \gamma. \cos. \Pi} = \cos. \gamma. \cos. \Pi$  we obtain  $c'' = m. \sqrt{a.(1-e^2)}. \sin. \gamma. \cos. \Pi + \text{&c.}$ , by substituting for  $p$ , its value  $\tan. \gamma. \cos. \Pi$ .

evident from the equations (C), of N°. 59, that these terms are the same for  $p, p', p'', \&c.$ , and that they are also the same for  $q, q', q'', \&c.$  and as relatively to the invariable plane, the constant quantities of the first member of the equations (1) and (2) of N°. 59, vanish ; in consequence of these equations, the constant terms must vanish from the expressions of  $p, p', \&c., q, q', \&c.$

Let us now consider the motion of two orbits, which are inclined at any angle to each other, by N°. 61, we will have,

$$c' = \sin. \phi. \cos. \theta. m. \sqrt{a.(1-e^2)} + \sin. \phi'. \cos. \theta'. m'. \sqrt{a'.(1-e'^2)} ;$$

$$c'' = \sin. \phi. \sin. \theta. m. \sqrt{a.(1-e^2)} + \sin. \phi'. \sin. \theta'. m'. \sqrt{a'.(1-e'^2)} ;$$

Let us suppose that the fixed plane to which the motion of the orbits is referred, is the invariable plane of which we have treated, and with respect to which the constant quantities of the first members of these equations vanish, as has been remarked in N°s. 21 and 22 of the first book. The angles  $\phi$  and  $\phi'$  being positive, the preceding equations give the following :

$$m. \sqrt{a.(1-e^2)}. \sin. \phi = m'. \sqrt{a'.(1-e'^2)}. \sin. \phi' ;$$

$$\sin. \theta = - \sin. \theta' ; \quad \cos. \theta = - \cos. \theta' ;$$

hence we infer that  $\theta' = \theta +$  the semicircumference ; consequently the nodes of the orbits are on the same line ; but the ascending node of one coincides with the descending node of the other ; so that the mutual inclination of these two orbits is equal to  $\phi + \phi'$ .

By. N°. 61, we have

$$c = m. \sqrt{a.(1-e^2)}. \cos. \phi + m'. \sqrt{a'.(1-e'^2)}. \cos. \phi' ;$$

this equation being combined with the preceding one between  $\sin. \phi$  and  $\sin. \phi'$ , gives\*

\* These constants must vanish, for they are in fact equal to  $c'$  and  $c''$ , which in the case of the invariable plane are equal to cypher.

$$2mc. \cos. \varphi. \sqrt{a.(1-e^2)} = c^2 + m^2 a.(1-e^2) - m^2 a'.(1-e'^2).$$

If the orbits be supposed to be circular, or at least of such a small eccentricity, that the squares of the excentricities may be neglected, the preceding equations will give  $\varphi$  equal to a constant quantity; therefore the inclinations of the planes of the orbits to the fixed plane, and to each other, will be constant, and these three planes will always have a common intersection. It follows from this, that the mean instantaneous variation of this intersection, is always the same; because it is only a function of those inclinations. When they are very small, it may be easily proved by N°. 60, and in virtue of the relation just found† between  $\sin. \varphi$ , and  $\sin. \varphi'$ , that for the time  $t$ , the motion of this intersection is  $-(0, 1)+(1, 0)). t.$

The position of the invariable plane, to which the motion of the planets has been referred, may be easily determined for any given instant; as it is only requisite to divide the angle of the mutual inclination of the orbits into two angles  $\varphi$ , and  $\varphi'$ , such that the preceding equation may obtain between  $\sin. \varphi$ , and  $\sin. \varphi'$ . Therefore denoting this mutual inclination by  $\varpi$ , we shall have

$$\tan. \varphi = \frac{m'.\sqrt{a'.(1-e'^2)}. \sin. \varpi}{m.\sqrt{a.(1-e^2)} + m'.\sqrt{a'.(1-e'^2)}. \cos. \varpi}. \ddagger$$

\* Multiplying both sides of this equation by  $2m. \sqrt{a.(1-e^2)}. \cos. \varphi$ , we obtain  $2mc. \cos. \varphi. \sqrt{a.(1-e^2)} = 2m^2.a.(1-e^2). \cos^2 \varphi + 2m.m'. \cos. \varphi. \cos. \varphi'. \sqrt{a.(1-e^2)}. \sqrt{a'(1-e'^2)}$ , which will coincide with the second member of this equation, if we substitute for  $e^2$  its value, and observe that  $m^2a.(1-e^2). \sin. {}^2 \varphi = m^2a'.(1-e'^2). \sin. {}^2 \varphi'$ .

† When  $\varphi$  and  $\varphi'$  are very small the nodes must regrade. See page 342.

‡ If one of the angles be  $\varphi$ , then we have  $\sin. \varphi. m.\sqrt{a.(1-e^2)} = \sin. (\varpi - \varphi)$ .  $m'.\sqrt{a'.(1-e'^2)} = (\sin. \varpi. \cos. \varphi - \sin. \varphi. \cos. \varpi). m'.\sqrt{a'.(1-e'^2)}$ , ∵ dividing by  $\cos. \varphi$ .  $\tan. \varphi. (m.\sqrt{a.(1-e^2)} + \cos. \varpi. m'.\sqrt{a'.(1-e'^2)}) = \sin. \varpi. m'.\sqrt{a'.(1-e'^2)}$ .

## CHAPTER VIII.

### *Second method of approximation of the Celestial Motions.*

53. It has been observed in the second chapter, that the coordinates of the heavenly bodies, referred to the foci of the principal forces which actuate them, are determined by differential equations of the second order. These equations have been integrated in the third Chapter, the principal forces being solely taken into account, and it has been shewn that in this case, the orbits are conic sections, of which the elements are the constant arbitrary quantities introduced by the integrations. As the action of the disturbing forces, cause only very small inequalities, to be added to the elliptic motion ; it is natural to endeavour to reduce to the laws of this motion, the disturbed motion of the heavenly bodies. If the method of approximation explained in N°. 45, be applied to the differential equations of elliptic motion, increased by small terms due to the action of the disturbing forces ; we can still consider the celestial motions in the reentrant orbits as being elliptical ; but the elements of this motion will be variable ; and their variations can be obtained by this method. It follows from it that the equations of the motion, being differentials of the second order, not only their finite integrals, but also the indefinitely small integrals of the first order, are the same as in the case of invariable ellipses ; so that we can differentiate the finite equations, the elements of this motion being considered as constant. It likewise results from the same method, that the equations of this motion, which are differentials of the first order, may be differenced, the elements of

the orbits, and the first differences of the coordinates being solely made to vary; provided that in place of the second differences of the coordinates, we only substitute that part of their values, due to the disturbing forces. These results may be immediately inferred from the consideration of elliptic motion.

For this purpose, conceive an ellipse passing through a planet, and through the element of the curve which it describes, the centre of the sun being supposed to exist in one of the foci. This ellipse is that which the planet would invariably describe, if the disturbing forces ceased to act on it. Its elements are constant during the interval of  $dt$ ; but they vary from one instant to another. Let therefore  $V = 0$ , be the finite equation of the invariable ellipse,  $V$  being a function of the rectangular coordinates,  $x, y, z$ , and of the parameters  $c, c', c'', \&c.$ , which last are functions of the elements of elliptic motion. This equation will also obtain in the case of the variable ellipse; but the parameters  $c, c', \&c.$ , will be no longer constant. However, since this ellipse appertains to the element of the curve described by the planet, during the instant  $dt$ ; the equation  $V = 0$ , will also obtain for the first and last point of this element,  $c, c', \&c.$ , being considered as constant.

This equation can therefore be differenced once,  $x, y, z$ , being solely made to vary, which gives

$$0 = \left( \frac{dV}{dx} \right) . dx + \left( \frac{dV}{dy} \right) . dy + \left( \frac{dV}{dz} \right) . dz; \quad (i)^*$$

\* In consequence of the mutual action of the planets on each other, it is necessary to add to the differential equations of their motion, terms which render the accurate integration of the resulting equations impossible in the present state of analysis, we are therefore obliged to have recourse to approximations; fortunately the terms resulting from the action of the disturbing forces are extremely small, for they are multiplied by the masses of the planets, or rather by their ratio to that of the sun; therefore if the differential equations, deprived of these terms, were integrated, the *constant* arbitrary quantities in this case, would only

Hence we see the reason why the finite equations of the invariable ellipse, may, in the case of the variable ellipse, be differenced once, the parameters being considered as constant. For the same reason, every differential equation of the first order, which belongs to the invariable ellipse, obtains equally for the variable ; for let  $V' = 0$ , be an equation of this order,  $V'$  being a function of  $x, y, z, \frac{dx}{dt}, \frac{dy}{dt}, \frac{dz}{dt}$ , and of the parameters  $c, c', &c.$ . It is evident that these quantities are the same for the variable ellipse, as for the invariable ellipse, which coincides with it, during the instant  $dt$ .

Now, if we consider the planet at the end of the time  $dt$ , or at the commencement of the subsequent instant ; the function  $V'$  will not vary from the ellipse relative to the instant  $dt$ , to the consecutive ellipse, except in consequence of the variation of the parameters, since

differ by a very small quantity, from the arbitrary quantities which the integration of the complete equations would furnish, if such integration could be effected ; for since the two equations differ only by these small terms, the difference between the arbitrary quantities must depend on the disturbing force, and therefore must be extremely small ; hence the expressions of the constant arbitrary quantities, which would be furnished by the integration of the imperfect or elliptical equations, may be assumed to express the *variable* arbitrary quantities, provided that the variations of those latter are determined by means of the difference between the two sets of equations ; therefore the elements of elliptic motion, which would be constant if the planet was subject to the sole action of the sun, are liable to small variations ; and although the motion is no longer elliptic, still it may be considered as such, during each indefinitely small portion of time, and the variable ellipse in which the planet may be considered to move during each instant, will be osculatory to the true orbit of the planet ; in fact, since the equation  $V = 0$ , has place for the first and last point of the curves described by the planet during the instant  $dt$ , the expressions for the co-ordinates  $x, y, z$ , will be the same : consequently the curves to which they belong are similar, but in one case the curve is an ellipse ;  $\therefore$  the curve of which  $x, y, z$ , are the co-ordinates when  $c, c', c'', &c.$ , are variable, must be similar to the former, and  $\therefore$  an ellipse, and if the disturbing forces ceased to act, the planet would describe this ellipse ; but as  $c, c', c'', &c.$ , have different values for each subsequent instant, the ellipses which would be respectively described if the disturbing forces ceased to act during these instants, must be different, so that they constitute a series of ellipses of curvature to the orbits of the planets.

the coordinates  $x, y, z$ , relative to the end of the first instant, are the same in the case of the two ellipses; thus the function  $V$  being equal to cypher, we have

$$0 = \left( \frac{dV}{dc} \right) \cdot dc + \left( \frac{dV}{dc'} \right) \cdot dc' + \left( \frac{dV}{dc''} \right) \cdot dc'' + \&c. \quad (i')$$

This equation may be also inferred from the equation  $V = 0$ , by making to vary at once,  $x, y, z, c, c', \&c.$ ; for if the equation  $(i)$  be subtracted from this differential, we shall have the equation  $(i')$ .

By differentiating the equation  $(i)$ , we shall have a new equation in  $dc, dc', \&c.$ , which combined with the equation  $(i')$  will enable us to determine the parameters  $c, c', \&c.$ .

It is thus that the geometers who first occupied themselves with the theory of the celestial motions, have determined the variations of the nodes and of the inclinations of the orbits; but this differentiation may be simplified in the following manner.

Let us consider generally the differential equation of the first order  $V' = 0$ , which equation, as we have seen, appertains equally to the variable ellipse and to the invariable ellipse, which, during the interval  $dt$ , coincides with it. In the following instant, this equation agrees equally to the two ellipses, but with this difference, that  $c, c', \&c.$ , remain the same in the case of the invariable ellipse, whilst they change with the variable ellipse. Let  $V''$  be what  $V'$  becomes, when the ellipse is supposed invariable; let  $V'$  be what this same function becomes, in the case of the variable ellipse. It is evident that in order to obtain  $V''$ , we must change in  $V'$ , the coordinates  $x, y, z$ , which are relative to the commencement of the first instant  $dt$ , into those which are relative to the commencement of the second instant; it is necessary then to increase the first differences  $dx, dy, dz$ , respectively by the quantities  $d^2x, d^2y, d^2z$ , relatively to the invariable ellipse, the element  $dt$  of the time, being supposed constant.

In like manner, in order to obtain  $V'$ , it is necessary to change in  $V'$  the coordinates  $x, y, z$ , into those which are relative to the commencement of the second instant, and which are likewise the same in the two ellipses; it is necessary afterwards to increase  $dx, dy, dz$ , respectively by the quantities  $d^2x, d^2y, d^2z$ ; finally, it is necessary to change the parameters  $c, c', \&c.$ , into  $c+dc, c'+dc', c''+dc'', \&c.$ .

The values of  $d^2x, d^2y, d^2z$ , are not the same in the two ellipses; they are increased in the case of the variable ellipse, by quantities which are due to the action of the disturbing forces. It thus appears that the two functions  $V'$  and  $V''$  only differ in this, that in the second, the parameters  $c, c', \&c.$ , are increased by  $dc, dc', \&c.$ ; and the values of  $d^2x, d^2y, d^2z$ , relative to the invariable ellipse, are increased by the quantities which are due to the disturbing forces.

We shall therefore obtain  $V''-V'$  by differencing on the supposition of  $x, y, z$ , being constant, &c., and of  $dx, dy, dz, c, c', c'', \&c.$ , being variable, provided that in this differential, we substitute for  $d^2x, d^2y, d^2z, \&c.$ , the parts of their values, which arise solely from the action of the disturbing forces.

Now, if in the function  $V''-V'$  we substitute in place of  $d^2x, d^2y, d^2z$ , their values relative to elliptic motions, we shall have a function of  $x, y, z, \frac{dx}{dt}, \frac{dy}{dt}, \frac{dz}{dt}, c, c', \&c.$ , which, in the case of the invariable\* ellipse, is equal to cypher; this function is therefore likewise nothing in

\* When in  $V''-V'$ , the values of  $\frac{d^2x}{dt^2}, \frac{d^2y}{dt^2}, \frac{d^2z}{dt^2}$ , due to the elliptic motion are substituted, the terms of the resulting equation must be identically equal to cypher; but in the case of  $V'-V$ , the values of  $\frac{d^2x}{dt^2}, \frac{d^2y}{dt^2}, \frac{d^2z}{dt^2}$  must be increased by the quantities due to the action of the disturbing forces; so that after substitution, the resulting expression may be resolved into two distinct equations, one of which would obtain, if there were no dis-

the case of the variable ellipse. We have evidently in this last case,  $V, -V' = 0$ ; for this equation is the differential of  $V' = 0$ ; by subtracting from it the equations  $V'' - V' = 0$ , we shall have  $V' - V'' = 0$ ; consequently we can in this case difference the equation  $V' = 0$ ,  $dx, dy, dz, c, c'$ , &c., being solely made to vary, provided that for  $d^2x, d^2y, d^2z$ , be substituted the parts of their values, relative to the disturbing forces. These results are precisely the same as those which we obtained in N°. 45, from considerations purely analytic; but considering their great importance, it was deemed right to deduce them here from the consideration of elliptic motion. This being premised,

64. Let the equations (P) of N°. 46 be resumed,

$$\left. \begin{aligned} 0 &= \frac{d^2x}{dt^2} + \frac{\mu \cdot x}{r^3} + \left( \frac{dR}{dx} \right); \\ 0 &= \frac{d^2y}{dt^2} + \frac{\mu \cdot y}{r^3} + \left( \frac{dR}{dy} \right); \\ 0 &= \frac{d^2z}{dt^2} + \frac{\mu \cdot z}{r^3} + \left( \frac{dR}{dz} \right); \end{aligned} \right\} \quad (P)$$

If we suppose  $R = 0$ , we shall have the equations of elliptic motion, which were integrated in the third chapter.

In N°. 18, the seven following integrals were obtained,

$$\left. \begin{aligned} c &= \frac{xdy - ydx}{dt}; \quad c' = \frac{x dz - z dx}{dt}; \quad c'' = \frac{y dz - z dy}{dt}; \\ 0 &= f + x \cdot \left\{ \frac{\mu}{r} - \left( \frac{dy^2 + dz^2}{dt^2} \right) \right\} + \frac{y dy \cdot dx}{dt^2} + \frac{z dz \cdot dx}{dt^2}; \\ 0 &= f' + y \cdot \left\{ \frac{\mu}{r} - \left( \frac{dx^2 + dz^2}{dt^2} \right) \right\} + \frac{x dx \cdot dy}{dt^2} + \frac{z dz \cdot dy}{dt^2}; \\ 0 &= f'' + z \cdot \left\{ \frac{\mu}{r} - \left( \frac{dx^2 + dy^2}{dt^2} \right) \right\} + \frac{x dx \cdot dz}{dt^2} + \frac{y dy \cdot dz}{dt^2}; \\ 0 &= \frac{\mu}{a} - \frac{2\mu}{r} + \left( \frac{dx^2 + dy^2 + dz^2}{dt^2} \right). \end{aligned} \right\} \quad (p)$$

turbing forces, and by means of the other the variations af the parameter, may be obtained, these equations are respectively equal to  $V''$  and  $V, -V''$ .

These integrals give the arbitrary quantities, in functions of the coordinates and of their first differences ; their form is extremely commodious for determining the variations of these arbitrary quantities. The three first integrals give, by differencing them, and by making the parameters  $c, c', c'', \&c.$ , and the first differences of the coordinates solely to vary,

$$dc = \frac{xd^2y - yd^2x}{dt}; \quad dc' = \frac{xd^2z - zd^2x}{dt}; \quad dc'' = \frac{yd^2z - zd^2y}{dt};$$

By substituting in place of  $d^2x, d^2y, d^2z$ , the parts of their values which are due to the actions of the disturbing forces, and which in virtue of the differential equations (P), are  $-dt^2 \cdot \left( \frac{dR}{dx} \right), -dt^2 \cdot \left( \frac{dR}{dy} \right), -dt^2 \cdot \left( \frac{dR}{dz} \right)$ ; we shall have

$$\begin{aligned} dc &= dt \cdot \left\{ y \cdot \left( \frac{dR}{dx} \right) - x \cdot \left( \frac{dR}{dy} \right) \right\}; \\ dc' &= dt \cdot \left\{ z \cdot \left( \frac{dR}{dx} \right) - x \cdot \left( \frac{dR}{dz} \right) \right\}; \\ dc'' &= dt \cdot \left\{ z \cdot \left( \frac{dR}{dy} \right) - y \cdot \left( \frac{dR}{dz} \right) \right\}. \end{aligned}$$

We have seen in N°s. 18, and 19, that the parameters  $c, c', c'', \&c.$ , determine the three elements of the elliptic orbit, namely,  $\phi$  the inclination of the orbit on the plane of  $x$  and  $y$ , and  $\theta$  the longitude of its node, by means of the equations

$$\tan. \phi = \frac{\sqrt{c'^2 + c''^2}}{c}; \quad \tan. \theta = \frac{c''}{c'},$$

and the semiparameter  $a.(1-e^2)$  of the ellipse, by means of the equation

$$\mu a.(1-e^2) = c^2 + c'^2 + c''^2;$$

These same equations obtain also in the case of the variable ellipse,

provided that  $c, c', c''$ , are determined by means of the preceding differential equations. In this manner the parameter of the variable ellipse, its inclination to the fixed plane of  $x$  and  $y$ , and the position of its node may be obtained.

By means of the three first equations ( $p$ ), we have deduced in N°. 19 the finite integral  $0 = c''x - c'y + cz$ ; this equation, and also its first differential,  $0 = c''dx - c'dy + cdz$ , taken on the supposition that  $c, c', c''$ , are constant, obtain in case of the disturbed ellipse.

If the fourth, the fifth, and sixth of the integrals ( $p$ ) be differenced, the parameters  $f, f', f''$ , and the differences  $dx, dy, dz$ , being considered as the sole variables, and if then we substitute, in place of  $d^2x, d^2y, d^2z$ , the quantities  $-dt^2 \cdot \left(\frac{dR}{dx}\right), -dt^2 \cdot \left(\frac{dR}{dy}\right) - dt^2 \cdot \left(\frac{dR}{dz}\right)$ , we shall have

$$df = dy \cdot \left\{ y \cdot \left(\frac{dR}{dx}\right) - x \cdot \left(\frac{dR}{dy}\right) + dz \cdot \left\{ z \cdot \left(\frac{dR}{dx}\right) - x \cdot \left(\frac{dR}{dz}\right) \right\} \right\} \cdot * \\ + (ydx - xdy) \cdot \left(\frac{dR}{dy}\right) + (zdx - xdz) \cdot \left(\frac{dR}{dz}\right);$$

$$df' = dx \cdot \left\{ x \cdot \left(\frac{dR}{dy}\right) - y \cdot \left(\frac{dR}{dx}\right) \right\} + dz \cdot \left\{ z \cdot \left(\frac{dR}{dy}\right) - y \cdot \left(\frac{dR}{dz}\right) \right\}$$

\* Differentiating under these restrictions we have

$$df = -2x \frac{(dy \cdot d^2y + dz \cdot d^2z)}{dt^2} + ydx \cdot \frac{d^2y}{dt^2} + ydy \cdot \frac{d^2x}{dt^2} + zdz \cdot \frac{d^2x}{dt^2} + zdx \cdot \frac{d^2z}{dt^2},$$

∴ by ordering the terms we have

$$df = dy \cdot \left( y \cdot \frac{d^2x}{dt^2} - x \cdot \frac{d^2y}{dt^2} \right) + dz \cdot \left( z \cdot \frac{d^2x}{dt^2} - x \cdot \frac{d^2z}{dt^2} \right) + (ydx - xdy) \cdot \frac{d^2y}{dt^2} + (zdx - xdz) \cdot \frac{d^2z}{dt^2}, \text{ which becomes the expression in the text, when } \left(\frac{dR}{dx}\right), \left(\frac{dR}{dy}\right), \left(\frac{dR}{dz}\right),$$

are substituted for  $\frac{d^2x}{dt^2}, \frac{d^2y}{dt^2}, \frac{d^2z}{dt^2}$ .

$$+ (xdy - ydx) \cdot \left\{ \frac{dR}{dx} \right\} + (zdy - ydz) \cdot \left\{ \frac{dR}{dz} \right\};$$

$$df'' = dx \cdot \left\{ x \cdot \left\{ \frac{dR}{dz} \right\} - z \cdot \left\{ \frac{dR}{dx} \right\} \right\} + dy \cdot \left\{ y \cdot \left\{ \frac{dR}{dz} \right\} - z \cdot \left\{ \frac{dR}{dy} \right\} \right\} \\ + (xdz - zdx) \cdot \left\{ \frac{dR}{dx} \right\} + (ydz - zdy) \cdot \left\{ \frac{dR}{dy} \right\}.$$

Finally, the seventh of the integrals ( $p$ ), when differenced with the same restrictions, will give the variations of the semiaxis major  $a$ , by means of the equation  $d. \frac{\mu}{a} = 2. dR$ , the differential  $dR$  being\* referred solely to the coordinates  $x, y, z$ , of the body  $m$ .

The values of  $f, f', f''$ , determine the longitude of the projection of the perihelion of the orbit, on the fixed plane, and the ratio of the excentricity to the semiaxis major; for  $I$  being the longitude of this projection, we have by N°. 19,

$$\tan. I = \frac{f}{f'},$$

\* Differentiating the seventh equation under the same restrictions, we obtain  $d. \frac{\mu}{a} = - 2. \left( \frac{d^2x}{dt^2} \cdot dx + \frac{d^2y}{dt^2} \cdot dy + \frac{d^2z}{dt^2} \cdot dz \right) = \frac{dR}{dx} \cdot dx + \frac{dR}{dy} \cdot dy + \frac{dR}{dz} \cdot dz = dR$ . See N°. 46.

By means of this expression, Lagrange ascertained that the mean motions were invariable, if the first power of the disturbing masses be only considered, the approximation being extended to any power of the excentricities and inclinations. From the extreme simplicity of this expression of the differential of the major axis, the determination of the longitude is a very easy problem. In the supplement to the third book, Laplace investigated the simplest form of which the other elements were susceptible, and he has succeeded in assigning such a form to them, that they only depend on partial differences of the same function, taken with respect to these elements, and what is particularly remarkable, the coefficients of these differences do not involve the time, and are solely functions of the elements themselves.

and  $e$  being the ratio of the excentricity to the semiaxis major, we have by the same number

$$\mu e = \sqrt{f^2 + f'^2 + f''^2}.$$

This ratio may also be determined, by dividing the semiparameter  $a(1-e^2)$ , by the semiaxis major  $a$ , and by taking the quotient from unity, the value of  $e^2$  will be obtained.

The integrals ( $p$ ) have given by elimination, in N°. 19, the finite integral,  $0 = \mu r - h^2 + fx + f'y + f''z$ ; this equation obtains also in the case of the disturbed ellipse, and it determines at each instant the nature of the variable ellipse, we can difference it,  $f, f', f''$ , being considered as constant quantities, which gives

$$0 = \mu dr + f dx + f' dy + f'' dz.$$

The semiaxis major  $a$  determines the mean motion of  $m$ , or more accurately, that which in the troubled orbit, corresponds to the mean motion in the invariable orbit; for by N°. 20, we have  $n = a^{-\frac{3}{2}} \cdot \sqrt{\mu}$ ; moreover, if we denote by  $\xi$  the mean motion of  $m$ , we have in the invariable elliptic orbit  $d\xi = ndt$ ; this equation obtains equally for the variable ellipse, since it is a differential of the first order. By differencing, we shall have  $d^2\xi = dn.dt$ ; but we have

$$dn = \frac{3an}{2\mu} \cdot d \cdot \frac{\mu}{a} = \frac{3an.dR}{\mu}, *$$

therefore

$$d^2\xi = \frac{3an.dt.dR}{\mu};$$

$$* dn = -\frac{3}{2}\sqrt{\mu} \cdot \frac{da}{a^{\frac{3}{2}}} = \frac{3}{2} \cdot \sqrt{\frac{\mu}{a}} \cdot d \cdot \frac{1}{a} = \left( \text{as } \sqrt{\frac{\mu}{a}} = na \right) \frac{3an}{2\mu} \cdot d \cdot \frac{\mu}{a}.$$

and by integrating

$$\zeta = \frac{3}{\mu} \cdot \iint a n dt dR.$$

Finally, it has been observed in N°. 18, that the integrals ( $p$ ) are only equivalent to five distinct integrals, and that they furnish between the seven parameters  $c, c', c'', f, f', f''$ , and  $e$  the two following equations of condition,

$$0 = fc'' - f'c' + f''c;$$

$$0 = \frac{\mu}{a} + \frac{f^2 + f'^2 + f''^2 - \mu^2}{c^2 + c'^2 + c''^2};$$

these equations obtain also in the case of the variable ellipse, provided that the parameters be determined by what precedes. We can likewise be assured of this *a posteriori*.

We have thus determined five elements of the disturbed orbit, namely its inclination, the position of its nodes, the semiaxis major (which gives the mean motion), its excentricity, and the position of the perihelion. It now remains for us to determine the sixth element of the elliptic motion, namely, that which in the undisturbed ellipse corresponds to the position of  $m$ , at a given epoch. For this purpose, let the expression for  $dt$  of N°. 18 be resumed,

$$\frac{dt \sqrt{\mu}}{a^{\frac{3}{2}}} = \frac{dv \cdot (1 - e^2)^{\frac{3}{2}}}{(1 + e \cdot \cos(v - \varpi))^2}.$$

This equation being expanded into a series, gave in the number already cited,

$$ndt = dv \cdot (1 + E^{(1)} \cdot \cos(v - \varpi) + E^{(2)} \cdot \cos. 2.(v - \varpi) + \&c.);$$

which being integrated on the supposition that  $e$  and  $\varpi$  are constant, will give

$$\int ndt + \epsilon = v + E^{(1)} \cdot \sin(v - \varpi) + \frac{E^{(2)}}{2} \cdot \sin. 2.(v - \varpi) + \&c.$$

$\epsilon$  being an arbitrary quantity. This integral is relative to the invariable ellipse; in order to extend it to the disturbed ellipse, it is necessary when we make all the terms to vary, even to the arbitrary quantities  $\epsilon$ ,  $e$  and  $\varpi$ , which it contains, that its differential should coincide with the preceding; which gives

$$d\epsilon = de \cdot \left\{ \left\{ \frac{dE^{(1)}}{de} \right\} \cdot (\sin. (v - \varpi) + \frac{1}{2} \cdot \left\{ \frac{dE^{(2)}}{de} \right\} \cdot \sin. 2(v - \varpi) + \text{&c.} \right\} \\ - d\varpi \cdot (E^{(1)} \cdot \cos. (v - \varpi) + E^{(2)} \cdot \cos. 2(v - \varpi) + \text{&c.})$$

$v - \varpi$  is the true anomaly of  $m$ , reckoned on the orbit, and  $\varpi$  is the longitude of the perihelion, also reckoned on the orbit. If the longitude of the projection of the perihelion, on a fixed plane has been already determined; but by N°. 22, we have by changing  $v$  into  $\varpi$ , and  $v$ , into  $I$  in the expression for  $v - \epsilon$  of that N°.,

$$\varpi - \epsilon = I - \theta + \tan. {}^2 \frac{1}{2} \phi. \sin. 2(I - \theta) + \text{&c.}$$

If then  $v$  and  $v$ , be supposed equal to cypher, in this same expression, we have

$$\epsilon = \theta + \tan. {}^2 \frac{1}{2} \phi. \sin. 2\theta + \text{&c.}$$

therefore

$$v = I + \tan. {}^2 \frac{1}{2} \phi. (\sin. 2\theta + \sin. 2(I - \theta)) + \text{&c.};$$

which gives

$$d\varpi = dI(1 + 2 \tan. {}^2 \frac{1}{2} \phi. \cos. 2(I - \theta) + \text{&c.}) \\ + 2d\theta. \tan. {}^2 \frac{1}{2} \phi. (\cos. 2\theta - \cos. 2(I - \theta) + \text{&c.}) \\ + \frac{d\phi. \tan. \frac{1}{2} \phi}{\cos. {}^2 \frac{1}{2} \phi}. (\sin. 2\theta + \sin. 2(I - \theta) + \text{&c.})$$

consequently, the values of  $dI$ ,  $d\theta$  and  $d\varpi$  being determined by what goes before; we shall have that of  $d\epsilon$ , by means of which, the value of  $d\epsilon$  will be obtained.

Hence it follows, that the expressions in series of the radius vector, of its projection on the fixed plane, of the longitude reckoned either on the fixed plane or on that of the orbit, and of the latitude, which have been determined in N°. 22, in the case of the invariable ellipse, obtain equally in the case of the disturbed ellipse, provided that  $nt$  be changed into  $\int ndt$ , and that the elements of the variable ellipse be determined by the preceding formulæ. For, since the finite equations between  $r, v, s, x, y, z$ , and  $\int ndt$ , are the same in the two cases; and since the expressions in series of N°. 22, result from those equations by operations purely analytic, and altogether independent of the constancy or variability of the elements; it is evident that these expressions obtain also in the case of variable elements.

When the ellipses are extremely excentric, as is the case in the orbits of the comets, the preceding analysis should be changed a little. The inclination  $\phi$  of the orbit on the fixed plane,  $\theta$  the longitude of its ascending node, the semiaxis major  $a$ , the semiparameter  $a.(1-e^2)$ , the excentricity  $e$ , and  $I$  the longitude of the perihelion on a fixed plane, may be determined by what goes before. But the values of  $\varpi$ , and of  $d\varpi$  being given in series arranged according to the powers of  $\tan. \frac{1}{2}\phi$ , it is necessary in order to render them convergent, to select the fixed plane, such that  $\tan. \frac{1}{2}\phi$  may be inconsiderable, and the simplest mode of effecting this, is to assume for the fixed plane, that of the orbit of  $m$ , at a given epoch.

The preceding value of  $de$  is expressed in a series which is only convergent, when the excentricity of the orbit is inconsiderable, it cannot therefore be employed in the present case. In order to remedy this, let us resume the equation

$$\frac{dt \cdot \sqrt{\mu}}{a^{\frac{3}{2}}} = \frac{dv \cdot (1-e^2)^{\frac{5}{2}}}{(1+e \cdot \cos(v-\varpi))^2}.$$

If we make  $1-e=\alpha$ , we have by the analysis of N°. 23, in the case of the invariable ellipse,

$$\tau + T = \frac{2\alpha^{\frac{3}{2}}(1-e^2)^{\frac{1}{2}}}{(2-\alpha)^2 \cdot \sqrt{\mu}} \cdot \tan. \frac{1}{2}(v-\omega) \left\{ 1 + \frac{\frac{3}{2}-\alpha}{2-\alpha} \cdot \tan. \frac{1}{2}(v-\omega) + \text{&c.} \right\};$$

$T$  being an arbitrary quantity. In order to extend this equation to the variable ellipse, it is necessary to difference it,  $T$ , the semiparameter  $a.(1-e^2)$ ,  $\alpha$  and  $\omega$  being considered as the sole variables. By this means we shall obtain a differential equation, which will enable us to determine  $T'$ ; and the finite equations which obtain in the case of the invariable ellipse, will likewise subsist in the case of the disturbed ellipse.

65. Let us particularly consider the variations of the elements of the orbit of  $m$ , in the case of the orbits having a small excentricity, and small inclination to each other. In N°. 48, we have shewn how to developpe  $R$  in that case, into a series of sines and cosines of the form  $m'k \cdot \cos. (in't - int + A)$ ,  $k$  and  $A$  being functions of the excentricities and of the inclinations of the orbits, of the positions of their nodes and of their perihelions, of the longitudes of the bodies at a given epoch, and of the greater axes. When the ellipses are variable all these quantities may be supposed to vary agreeably to what precedes, it is necessary moreover, to change in the preceding term the angle  $in't - int$  into  $i\dot{\gamma}n'dt - i\dot{\gamma}ndt$ , or what comes to the same thing, into  $i\dot{\gamma}' - i\dot{\gamma}$ .

Now, by the preceding number we have

$$\frac{\mu}{a} = 2 \int dR;$$

$$\zeta = \int ndt = \frac{3}{\mu} \cdot \iint andt \cdot dR.$$

The difference  $dR$  being taken solely with respect to the coordinates  $x, y, z$ , of the body  $m$ , we should not make to vary in the term  $m'k \cdot \cos. (i\dot{\gamma}' - i\dot{\gamma} + A)$  of the expression for  $R$ , developed into a series, only that part which depends on the motion of this body; besides,  $R$  being a finite function of  $x, y, z, x', y', z'$ , we can by N°. 63 suppose the elements of the orbit constant in the differential  $dR$ , it is therefore

sufficient to make  $\zeta$  to vary in the preceding term, and as the difference of  $\zeta$  is  $ndt$ , we shall have  $i.m'.kn dt \cdot \sin.(i'\zeta' - i\zeta + A)$  for the term of  $dR$ , which corresponds to the preceding term of  $R$ . Thus, by having regard only to this term, we shall have

$$\frac{1}{a} = \frac{2i.m'}{\mu} \cdot \int k n dt \cdot \sin.(i'\zeta' - i\zeta + A);$$

$$\zeta = \frac{3i.m'}{\mu} \cdot \int \int a k n^2 dt^2 \cdot \sin.(i'\zeta' - i\zeta + A).$$

If the squares and products of the disturbing masses be neglected, we can in the integration of these terms, suppose the elements of elliptic motion constant, which changes  $\zeta$  into  $nt$ , and  $\zeta'$  into  $n't$ ; hence we deduce

$$\frac{1}{a} = -\frac{2im'n.k}{\mu.(i'n'-in)} \cdot \cos.(i'n't - int + A);$$

$$\zeta = -\frac{3im'.an^2k}{\mu.(i'n-in)^2} \cdot \sin.(i'n't - int + A).$$

It appears\* from this, that if  $i'n' - in$  does not vanish, the quantities  $a$  and  $\zeta$  only contain periodical inequalities, the approximation being continued as far as the first power of the disturbing force; but as  $i'$  and  $i$  are integral numbers, the equation  $i'n' - in = 0$ , cannot have place when the mean motions† of  $m$  and of  $m'$  are incommensurable,

\* This conclusion which was first shewn by Laplace to be true, when the approximation was continued as far as the first power of the disturbing force, and as far as the products of four dimensions of the excentricities and inclinations, was shewn by Lagrange to be true, taking into account any power of the excentricity and inclination; and it was further extended by Poisson, and afterwards by Laplace and Lagrange, who proved, that even continuing the approximation as far as the squares of the disturbing forces, no inequalities, but those which are periodic affect the major axis; and in general, that the stability of the planetary system is not deranged, when the squares of the masses, and all powers of the excentricities and inclinations, are taken into account. See N°. 54.

† The equation  $i'n' - in = 0$ , would therefore suppose an unique case, among an infinity of others equally possible, besides the disturbing action of  $m'$  is solely considered in

which is the case of the planets, and we may assume in general, since  $n$  and  $n'$  are constant arbitrary quantities susceptible of all possible values, that their exact ratio, number to number, is extremely improbable.

We are consequently brought to this remarkable conclusion, namely, that the greater axes of the orbits of the planets and their mean motions, are only subject to periodic inequalities depending on their mutual configuration; and consequently, if these quantities be neglected, their greater axes are constant, and their mean motions are uniform: which result accords with that which was previously deduced in another manner, in N°. 54.

If the mean motions  $nt$  and  $n't$  without being accurately commensurable, are yet very nearly in the ratio of  $i'$  to  $i$ ; the divisor  $i'n'—in$  will be extremely small, and there would result in  $\zeta$  and  $\zeta'$ , inequalities which increasing with extreme slowness, may give ground to observers, to think that the mean motions of the bodies  $m$  and  $m'$ , are not uniform. We shall see in the theory of Jupiter and Saturn, that this is the case relatively to those two planets; their mean motions are such that twice that of Jupiter, is very nearly equal to five times that of Saturn; so that  $5n'—2n$  is not the seventy-fourth part of  $n$ . The smallness of this divisor renders the term of the expression for  $\zeta$ , which depends on the angle  $5n't—2nt$  extremely sensible, although it is of the order  $i'—i$ , or of the third order,\* with respect to the ex-

this case, but strictly speaking  $R$  is a function of the actions of all the planets  $m'$ ,  $m''$ ,  $m'''$ , &c., ∵ the form of the angle will be  $(in+i'n'+i''n''+\&c.)t+A$ , so that the similar equation of mean motion would suppose  $in+i'n'+i''n''+\&c.=0$ , which is even more improbable than the equation  $i'n'—in=0$ ; besides, if this last equation obtained, when there were only three bodies, it would cease to exist when the action of the other planets was taken into account.

\* As  $i'=5$ , and  $i=2$ ,  $i'-i=3$ , and consequently, the periodic function is multiplied by quantities of the third order, with respect to the excentricities and inclinations. If the axis major is subject to an inequality increasing proportionally to the time, the mean longitude has one increasing proportionally to the square of the time. See N°. 54.

centricities and inclinations of the orbits, as has been observed in N°. 48. The preceding analysis gives the most sensible part of these inequalities; for the variation of the mean longitude depends on two integrations; while the variations of the other elements of elliptic motion depend only on one integration; consequently, the terms of the expression for the mean longitude are those solely, which can have the square of ( $i'n' - in$ ) for a divisor; therefore taking into account these terms solely, which considering the smallness of this divisor, must be the most considerable, it will be sufficient in the expressions for the radius vector, the longitude, and the latitude, to increase by these terms the mean longitude.

When we have inequalities of this kind, which the action of  $m'$  produces in the mean motion of  $m$ ; it is easy to infer the corresponding inequalities produced by the action of  $m$  on the mean motion of  $m'$ . In fact, if we only consider the mutual action of the three bodies  $M$ ,  $m$  and  $m'$ ; the formula (7) of N°. 9. gives

$$\begin{aligned} \text{const.} = m \cdot \frac{(dx^2 + dy^2 + dz^2)}{dt^2} + m' \cdot \frac{(dx'^2 + dy'^2 + dz'^2)}{dt^2} \\ - \frac{((mdx + m'dx')^2 + (mdy + m'dy')^2 + (mdz + m'dz')^2)}{(M + m + m').dt^2} \quad (a) \\ - \frac{2Mm}{\sqrt{x^2 + y^2 + z^2}} - \frac{2Mm'}{\sqrt{x'^2 + y'^2 + z'^2}} - \frac{2mm'}{\sqrt{(x-x)^2 + (y-y)^2 + (z-z)^2}}. \end{aligned}$$

The last of the integrals ( $p$ ) of the preceding number, gives by substituting for  $\frac{t}{a}$ , the integral  $2\int dR$

$$\frac{dx^2 + dy^2 + dz^2}{dt^2} = \frac{2.(M+m)}{\sqrt{x^2 + y^2 + z^2}} - 2\int dR.$$

If we then call  $R'$  what  $R$  becomes, when the action of  $m$  on  $m'$  is considered, we shall have

$$R' = \frac{m.(xx' + yy' + zz')}{(x^2 + y^2 + z^2)^{\frac{3}{2}}} - \frac{m}{\sqrt{(x'-x)^2 + (y'-y)^2 + (z'-z)^2}}$$

$$\frac{dx'^2 + dy'^2 + dz'^2}{dt^2} = \frac{2.(M+m')}{\sqrt{x'^2 + y'^2 + z'^2}} - \int d'R';$$

the differential characteristic  $d'$  only referring to the coordinates  $x', y', z'$ , of the body  $m'$ . By substituting in the equation (*a*) in place of  $\frac{dx^2 + dy^2 + dz^2}{dt^2}$  and of  $\frac{dx'^2 + dy'^2 + dz'^2}{dt^2}$ , these values, we shall have

$$m \int dR + m' \int d'R = \text{const.}$$

$$-\frac{((m.dx + m'dx')^2 + (m.dy + m'dy')^2 + (m.dz + m'dz')^2)}{2.(M + m + m')dt^2}$$

$$+ \frac{m^2}{\sqrt{x^2 + y^2 + z^2}} + \frac{m'^2}{\sqrt{x'^2 + y'^2 + z'^2}}.$$

It is evident that the second member of this equation does not contain terms of the order of the squares and of the products of the masses  $m$  and  $m'$ , which have for a divisor  $i'n' - in$ ; therefore if we only consider such terms, we shall have

$$m \int dR + m' \int d'R' = 0;$$

hence if we only take into account those terms, of which the divisor is  $(i'n' - in)^2$ , we shall have

$$\frac{3 \int d'n' dt d'R'}{M+m'} = - \frac{m.(M+m).a'n'}{m'.(M+m').an} \cdot \frac{3 \int dR dt dR'}{M+m};$$

but we have

$$\zeta = \frac{3 \int dR dt dR}{M+m}; \quad \zeta' = \frac{3 \int d'n' dt d'R'}{M+m};$$

consequently

$$m.(M+m).an\zeta' + m.(M+m).a'n') = 0;$$

moreover

$$n = \frac{\sqrt{M+m}}{a^{\frac{3}{2}}}; \quad n' = \frac{\sqrt{M+m'}}{a'^{\frac{3}{2}}};$$

therefore  $m$  and  $m'$  being neglected in comparison with  $M$ , we shall have

$$m.\sqrt{a}.\zeta' + m'.\sqrt{a'}\zeta' = 0;$$

or

$$\zeta' = - \frac{m.\sqrt{a}}{m'.\sqrt{a'}} \cdot \zeta.$$

Thus the inequalities of  $\zeta$ , which have for a divisor  $(i'n' - in)^2$  will make known those of  $\zeta'$ , which have the same divisor. These inequalities are, as we have seen, affected with contrary signs, if  $n$  and  $n'$  have the same sign, or what comes to the same thing, if the two bodies  $m$  and  $m'$  revolve in the same direction; they are besides in a constant ratio to each other; hence it follows, that if they appear to accelerate the mean motion of  $m$ , they will appear to retard that of  $m'$ , according to the same law, and the apparent acceleration of  $m$ , will be to the apparent retardation of  $m'$ , as  $m'.\sqrt{a'}$  to  $m.\sqrt{a}$ . The acceleration of the mean motion of Jupiter, and the retardation of the mean motion of Saturn, which the comparison of ancient with modern observations made known to Halley, being very nearly in this ratio; I have inferred from the preceding theorem, that they are owing to the mutual action of these two planets; and since it has been demonstrated, that this action cannot produce any change in the mean motions, independent of the configuration of the planets, I did not hesitate to admit that there exists in the theory of Jupiter and Saturn, a great periodic inequality of a very long period. And observing then that five times the mean motion of Saturn, minus twice that of Jupiter, is very nearly equal to cypher, it appeared to me very probable that the cause of the phenomena observed by Halley, was

an inequality depending on this argument. The determination of this inequality verified my conjecture.

The period of the argument ( $i'n't - int$ ), being supposed very long, the elements of the orbits  $m'$  and  $m$  experience in this interval sensible variations, which it is essentially necessary to consider in the double integral  $\iint a k n^2 \cdot dt^2 \cdot \sin. (i'n't - int + A)$ . For this purpose, we shall make the function  $k \cdot \sin. (i'n't - int + A)$  assume the form  $Q \cdot \sin. (i'n't - int + i'e' - ie) + Q' \cdot \cos. (i'n't - int + i'e' - ie)$ ;  $Q$  and  $Q'$  being functions of the elements of the orbits, we shall have consequently

$$\begin{aligned} & \iint a k n^2 \cdot dt^2 \cdot \sin. (i'n't - int + A) = * \\ & - \frac{n^2 a \cdot \sin. (i'n't - int + i'e' - ie)}{(i'n' - in)^2} \left\{ Q - \frac{2dQ'}{(i'n' - in) \cdot dt} - \frac{3d^2 Q}{(i'n' - in)^2 \cdot dt^2} \right. \\ & \quad \left. + \frac{4d^3 Q'}{(i'n' - in)^3 dt^3} + \text{&c.} \right\} \end{aligned}$$

## 3 B 2

\* Substituting for  $A$  its value  $i'e' - i\omega - g\omega - g'\omega - g''\theta - g'''b$ ;  $\sin. (in't - int + A) = \sin. (in't - int + i'e' - ie) \cdot \cos. (g\omega + g'\omega + g''\theta + g'''b) - \cos. (in't - int + i'e' - ie) \cdot \sin. (g\omega + g'\omega + g''\theta + g'''b)$ , hence the value of  $k \cdot \sin. (in't - int + A)$  will be given; calling  $in't - int + i'e' - ie$ ,  $ft + b$ , the quantity to be integrated becomes  $\int dt \cdot \int dt \cdot \sin. (ft + b) Q + \int dt \cdot \int dt \cdot \cos. (ft + b) Q'$ , now one integration gives  $\int dt \cdot \sin. (ft + b) \cdot Q = - \frac{Q}{f} \cdot \cos. (ft + b) + \frac{1}{f} \int \cos. (ft + b) \cdot dt \cdot \frac{dQ}{dt} \left( = \frac{1}{f^2} \cdot \frac{dQ}{dt} \cdot \sin. (ft + b) + \frac{1}{f^2} \cdot f \cdot \sin. (ft + b) \cdot dt \cdot \frac{d^2 Q}{dt^2} + \text{&c.} \right) = \frac{Q}{f} \cdot \cos. (ft + b) - \frac{1}{f^2} \cdot \frac{dQ}{dt} \cdot \sin. (ft + b) + \frac{1}{f^3} \cdot \frac{d^2 Q}{dt^2} \cdot \cos. (ft + b) + \frac{1}{f^4} \cdot \frac{d^3 Q}{dt^3} \cdot \sin. (ft + b) + \text{&c.};$  in like manner we can obtain by partial integration,  $\int dt \cdot \cos. (ft + b) \cdot Q' = \frac{Q'}{f} \cdot \sin. (ft + b) + \frac{1}{f^2} \cdot \frac{dQ'}{dt^2} \cdot \cos. (ft + b) - \frac{1}{f^3} \cdot \frac{d^2 Q'}{dt^2} \cdot \sin. (ft + b) - \text{&c.}$ , in order to obtain the second integrals, i.e.  $\int dt \cdot \int dt \cdot \sin. (ft + b)$ , each of the terms of the preceding series into which the first integrals may be resolved, should be multiplied by  $dt$ , and then integrated in the same manner as  $\int dt \cdot \sin. (ft + b) \cdot Q$ , and if all the factors of  $\sin. (ft + b)$ , and  $\cos. (ft + b)$  be respectively collected, we shall obtain by substituting for  $f$  and  $b$ , the expression given in the text.

$$-\frac{n^2 a \cdot \cos(i'n't - int + i'\epsilon' - i\epsilon)}{(i'n' - in)^2} \left\{ Q' + \frac{2dQ}{(i'n' - in) \cdot dt} - \frac{3d^2 Q'}{(i'n' - in)^2 \cdot dt} \right. \\ \left. - \frac{4d^3 Q}{(i'n' - in)^3 \cdot dt^3} + \text{&c.} \right\}.$$

In consequence of the slowness of the secular variations of the elliptic elements, the terms of these two series decrease with great rapidity. We may therefore only consider the two first terms in each series. If then we substitute in place of the elements of the orbits, their values arranged according to the powers of the time, the first power being the only one which is retained; the preceding double integral may be transformed into one sole term of the form

$$(F + E \cdot t) \cdot \sin(i'n't - int + A + H \cdot t).$$

Relatively to Jupiter and Saturn, this expression will serve for several centuries before and after the instant, which may have been selected for the epoch.

The great inequalities of which we have been speaking, produce some sensible terms among those which depend on the second power of the disturbing masses. In fact, if in the formula

$$\zeta = \frac{3im'}{\mu} \cdot \iint a kn^2 \cdot dt^2 \cdot \sin(i'\zeta' - i\zeta + A);$$

we substitute for  $\zeta$  and  $\zeta'$  their values

$$nt - \frac{3i \cdot m' \cdot a n^2 k}{\mu \cdot (i'n' - in)^2} \cdot \sin(i'n't - int + A); \\ n't' + \frac{3i \cdot m \cdot a n^2 k^2}{\mu \cdot (i'n' - in)^2} \cdot \frac{\sqrt{a}}{\sqrt{a'}} \cdot \sin(i'n't - int + A),$$

there will result among the terms of the order  $m^2$ , the following

$$-\frac{9i^2 \cdot m'^2 \cdot a^2 \cdot n^4 \cdot k^2}{8 \cdot \mu^2 \cdot (i'n' - in)^4} \cdot \frac{(im' \cdot \sqrt{a'} + i'm \cdot \sqrt{a})}{m'^2 \cdot \sqrt{a'}} \cdot \sin 2(i'n't - int + A), *$$

\* Assuming  $p = \frac{3im' \cdot n^2 ak}{\mu \cdot (i'n' - in)^2}$ , and  $p' = \frac{3i \cdot m \cdot a n^2 k}{\mu \cdot (i'n' - in)^2} \cdot \frac{\sqrt{a}}{\sqrt{a'}}$ , the value of  $\zeta = E$ .

There will result in the value of  $\zeta'$  a corresponding term, which is to the preceding in the ratio of  $m.\sqrt{a}$  to  $-m'.\sqrt{a'}$ , it is therefore

$$\frac{9.i^2.m'^2.a^2.n^4.k^2}{8.\mu^2.(i'n'-in)^4} \left\{ im'.\sqrt{a'} + i'.m.\sqrt{a} \right\} \frac{m.\sqrt{a}}{m'^2.a'} \sin. 2.(i'n't - int + A).$$

66. It may happen, that the most sensible inequalities of mean motion occur among the terms of the order of the squares of the disturbing masses. If we suppose three bodies  $m, m', m''$ , to revolve about  $M$ , the expression for  $dR$  relative to terms of this order, will contain inequalities of the form  $k \cdot \sin.(int - i'n't + i''n''t + A)$ , now if the mean motions of  $m, m', m'', \&c.$ , are such that  $in - i'n' + i''n''$ , may be supposed a very small fraction of  $n$ , there will result a very sensible inequality in the value of  $\zeta$ . This inequality may even render rigorously equal to cypher, the quantity  $in - i'n' + i''n''$ , and thus establish an equation of condition between the mean motions and the mean longitudes of the three bodies  $m, m', m''$ ; this remarkable case obtains in the system of the satellites of Jupiter. We proceed to develope the analysis of it.

If we suppose  $M$  to represent the unity of mass, and if  $m, m', m''$ , be neglected in comparison with  $M$ , we shall have

$$n^2 = \frac{1}{a^3}, \quad n'^2 = \frac{1}{a'^3}; \quad n''^2 = \frac{1}{a''^3};$$

$\int \int dt^2 \cdot \sin.(in't - int + (p' - p)) \cdot \sin.(in't - int - A + A) (a)$ , now if  $p', p$  be supposed to be very small, we shall have  $\sin.((p' - p)) \cdot \sin.(in't - int + A) = (p' - p) \cdot \sin.(i'n't - int + A)$ , and the cosine of the same quantity = 1, in each case these expressions are true, for the first power of the disturbing force;  $\therefore$  in the expression (a) a term occurs =  $E \cdot \cos.(i'n't - int + A) (i'p' + ip) \cdot \sin.(in't - int - A) = \frac{E}{2} \cdot (i'p' + ip) \cdot \sin.2.(int - int + A)$ , now  $i'p' + ip = \frac{3ian^2k.(i'm.\sqrt{a} + im'.\sqrt{a'})}{\mu.(i'n' - in)^2.\sqrt{a'}}$ , and  $E = \frac{3im'^2.akn^2}{\mu.m'}$ ,  $\therefore$  the coefficient of  $\sin.2.(n't - nt + A) = \frac{9i^2.m'^2.a^2.n^2.k^2}{2\mu^2.(i'n' - in)^2} \cdot \frac{(i'm.\sqrt{a} + i'm'.\sqrt{a'})}{m'.\sqrt{a'}}$ , and when the double integration is performed, there will result the expression given in the text.

we have also

$$d\zeta = ndt; \quad d\zeta' = n'dt; \quad d\zeta'' = n''dt;$$

consequently

$$\frac{d^2\zeta}{dt^2} = -\frac{3}{2} \cdot n^{\frac{1}{3}} \cdot \frac{da}{a^2}; \quad \frac{d^2\zeta'}{dt^2} = -\frac{3}{2} \cdot n'^{\frac{1}{3}} \cdot \frac{da'}{a'^2}; \quad \frac{d^2\zeta''}{dt^2} = -\frac{3}{2} \cdot n''^{\frac{1}{3}} \cdot \frac{da''}{a''^2}.$$

It has been observed in N°. 61, that if we only consider inequalities which have very long periods, we have

$$\text{constant} = \frac{m}{a} + \frac{m'}{a'} + \frac{m''}{a''};$$

which gives

$$0 = m \cdot \frac{da}{a^2} + m' \cdot \frac{da'}{a'^2} + m'' \cdot \frac{da''}{a''^2}.$$

It has been also observed in the same number, that if the squares of the excentricities and of the inclinations of the orbits be neglected, we have

$$\text{constant} = m \cdot \sqrt{a} + m' \cdot \sqrt{a'} + m'' \cdot \sqrt{a''};$$

which gives

$$0 = \frac{mda}{\sqrt{a}} + \frac{m'.da'}{\sqrt{a'}} + \frac{m''.da''}{\sqrt{a''}}.$$

From these different equations it is easy to infer

$$\begin{aligned} \frac{d^2\zeta}{dt^2} &= -\frac{3}{2} \cdot n^{\frac{1}{3}} \cdot \frac{da}{d^2} & * \\ \frac{d^2\zeta'}{dt^2} &= \frac{3}{2} \cdot \frac{m \cdot n'^{\frac{1}{3}}}{m' \cdot n} \cdot \left( \frac{n-n''}{n'-n''} \right) \cdot \frac{da}{a^2} \\ \frac{d^2\zeta''}{dt^2} &= -\frac{3}{2} \frac{m \cdot n''^{\frac{1}{3}}}{m'' \cdot n} \cdot \left( \frac{n-n'}{n'-n''} \right) \cdot \frac{da}{a^2}. \end{aligned}$$

$$\bullet \quad n^{\frac{2}{3}} = \frac{1}{a}, \quad \therefore -\frac{2}{3} \cdot \frac{dn}{n^{\frac{1}{3}}} = \frac{da}{a^2}, \quad \text{and} \quad \frac{d^2\zeta}{dt^2} = dn = -\frac{3}{2} \cdot n^{\frac{1}{3}} \cdot \frac{da}{a^2}; \quad \text{in like manner}$$

Finally, the equation  $\frac{\mu}{a} = 2 \int dR$ , of N°. 64, gives

$$-\frac{da}{a^2} = 2dR.$$

It is therefore only requisite to determine  $dR$ .

By N°. 46 we have,

$$R = \frac{m' r}{r'^2} \cdot \cos(v' - v) - m' \cdot (r^2 - 2rr' \cdot \cos(v' - v) + r'^2)^{-\frac{1}{2}}$$

$$+ \frac{m'' r}{r''^2} \cdot \cos(v'' - v) - m'' \cdot (r^2 - 2rr'' \cdot \cos(v'' - v) + r''^2)^{-\frac{1}{2}}$$

the squares and the products of the inclinations of the orbits being neglected. If this function be developed into a series arranged according to the powers of the cosines of  $v' - v$ , of  $v'' - v$ , and of their multiples : we shall have an expression of the following form,

$$\begin{aligned} R = & \frac{m'}{2} \cdot (r, r')^{(0)} + m' \cdot (r, r')^{(1)} \cdot \cos(v' - v) + m' \cdot (r, r')^{(2)} \cdot \cos. 2.(v' - v) \\ & + m' \cdot (r, r')^{(3)} \cdot \cos. 3.(v' - v) + \&c. \\ & + \frac{m''}{2} \cdot (r, r'')^{(0)} + m'' \cdot (r, r'')^{(1)} \cdot \cos(v'' - v) + m'' \cdot (r, r'')^{(2)} \cdot \cos. 2.(v'' - v) \\ & + m'' \cdot (r, r'')^{(3)} \cdot \cos. 3.(v'' - v) + \&c. \end{aligned}$$

$$n^{\frac{4}{3}} = n^2 \cdot n^{-\frac{2}{3}} = n^2 \cdot a.$$

$$\frac{mda}{a^2} + \frac{m'da'}{a'^2} = -\frac{m''da''}{a''^2} = \left(\text{as } n = \frac{1}{a^{\frac{3}{2}}}\right) \frac{mn''da}{\sqrt{a}} + \frac{m'n''da'}{\sqrt{a'}} , \text{ therefore multiplying}$$

both sides by  $nn' = a^{-\frac{3}{2}} \cdot a'^{-\frac{3}{2}}$ , we shall obtain  $\frac{m \cdot nn' \cdot da}{a^2} + \frac{m'nn' \cdot da'}{a'^2} = \frac{mn'n'' \cdot da}{a^2} +$

$$\frac{m'nn'' \cdot da'}{a'^2}; \because \frac{da'}{a'^2} = \frac{m \cdot n' \cdot (n - n'')}{m' \cdot n \cdot (n' - n'')} \cdot \frac{da}{a^2}, \therefore \frac{d^2\zeta'}{dt^2} = -\frac{3}{2} \cdot n'^{\frac{1}{3}} \cdot \frac{da'}{a'^2} = \frac{3}{2} \cdot \frac{m \cdot n'^{\frac{4}{3}}}{m' \cdot n}.$$

$\frac{(n - n'')}{(n'' - n')} \cdot \frac{da}{a^2}$ , the expression for  $\frac{d^2\zeta''}{dt^2}$  may be obtained in a similar manner.

hence we obtain

$$dR = \left\{ dr \cdot \left\{ \begin{array}{l} \frac{m'}{2} \cdot \left\{ \frac{d.(r,r')^{(0)}}{dr} \right\} + m' \cdot \left\{ \frac{d.(r,r')^{(1)}}{dr} \right\} \cdot \cos.(v'-v) + m' \cdot \\ \quad \left\{ \frac{d.(r,r')^{(2)}}{dr} \right\} \cdot \cos.2.(v'-v) + \text{etc.} \\ \frac{m''}{2} \cdot \left\{ \frac{d.(r,r'')^{(0)}}{dr} \right\} + m'' \cdot \left\{ \frac{d.(r,r'')^{(1)}}{dr} \right\} \cdot \cos.(v''-v) + m'' \cdot \\ \quad \left\{ \frac{d.(r,r'')^{(2)}}{dr} \right\} \cdot \cos.2.(v''-v) + \text{etc.} \\ + dv \cdot \left\{ \begin{array}{l} m' \cdot (r,r')^{(1)} \cdot \sin.(v'-v) + 2m' \cdot (r,r')^{(2)} \cdot \sin.2.(v'-v) + \text{etc.} \\ + m'' \cdot (r,r'')^{(1)} \cdot \sin.(v''-v) + 2m'' \cdot (r,r'')^{(2)} \cdot \sin.2.(v''-v) + \text{etc.} \end{array} \right\} \end{array} \right\}$$

Suppose agreeably to what is indicated by observations, in the system of the three first satellites of Jupiter, that  $n-2n'$ , and  $n'-2n''$ . are very small fractions of  $n$ , and that their difference  $(n'-2n')-(n'-2n'')$ , or  $n-3n'+2n''$  is incomparably less than each of them. It results from the expressions of  $\frac{\delta r}{a}$ , and of  $\delta v$ , of N°. 50, that the action of  $m'$  produces in the radius vector, and in the longitude of  $m$ , a very sensible inequality, depending on the argument  $2.(n't-nt+\epsilon'-\epsilon)$ . The terms relative to this inequality ,have for a divisor  $4.(n'-n)^2-n^2$ , or  $(n-2n').(3n-2n')$ . and this divisor is extremely small in consequence of the smallness of the factor  $n-2n'$ . It appears also from a consideration of the same expressions, that the action of  $m$  produces in the radius vector, and in the longitude of  $m'$ , an inequality depending on the argument  $(n't-nt+\epsilon'-\epsilon)$ , and which as it has for a divisor  $(n'-n)^2-n^2$  or  $n.(n-2n')$  is extremely sensible. It appears in like manner, that the action of  $m''$  on  $m'$  produces in the same quantities a considerable inequality, depending on the argument  $2.(n''t-n't+\epsilon''-\epsilon)$ . Finally, we may perceive that the action of  $m'$ , produces in the longitude and radius vector of  $m''$  a considerable inequality, depending on the argument  $n't-n't+\epsilon''-\epsilon$ . These inequalities have been recognized by observations, we shall develope them in detail in the theory

of the satellites of Jupiter; their magnitude relative to the other inequalities permits us to neglect the latter in the present question. Let us therefore suppose

$$\begin{aligned}\delta r &= m'.E'. \cos. 2.(n't - nt + \epsilon' - \epsilon); \\ \delta v &= m'.F'. \sin. 2.(n't - nt + \epsilon' - \epsilon); \\ \delta r' &= m''.E''. \cos. 2.(n''t - n't + \epsilon'' - \epsilon') + m.G. \cos. (n't - nt + \epsilon' - \epsilon); \\ \delta v' &= m''.F''. \sin. 2.(n''t - n't + \epsilon'' - \epsilon') + m.H. \sin. (n't - nt + \epsilon' - \epsilon); \\ \delta r'' &= m''.G''. \cos. (n''t - n't + \epsilon'' - \epsilon'); \\ \delta v'' &= m''.H''. \sin. (n''t - n't + \epsilon'' - \epsilon').\end{aligned}$$

It is necessary now to substitute in the preceding expression for  $dR$ , instead of  $r, v, r', v', r'', v''$ , the values of  $a + \delta r, nt + \epsilon + \delta v, a' + \delta r', n't + \epsilon' + \delta v', a'' + \delta r'', n''t + \epsilon'' + \delta v''$ , and only to retain the terms depending on the argument  $nt - 3n't + 2n''t + \epsilon - 3\epsilon' + 2\epsilon''$ , it is easy to see that the substitution of the values of  $\delta r, \delta v, \delta r'', \delta v''$ , cannot produce any such term. Therefore it can only arise from the substitution of the values of  $\delta r'$ , and of  $\delta v'$ ; the term  $m'.(r,r')^{(1)}.dv. \sin. (v' - v)$  of the expression of  $dR$ , produces the following quantity :

$$-\frac{m'.m''.ndt}{2} \cdot \left\{ E''. \frac{d.(a,a')^{(1)}}{da'} - F''.(a,a')^{(1)} \right\} \cdot \sin. nt - 3n't + 2n''t + \epsilon - 3\epsilon' + 2\epsilon''.$$

And it is the only quantity of this kind, which the expression of  $dR$

\*  $(r,r')^{(1)} = (a,a')^{(1)} + \frac{d.(a,a')^{(1)}}{da}. \delta r + \frac{d.(a,a')^{(1)}}{da'} \cdot \delta r' + \text{ &c. } \sin. (v' - v) = \sin. (n't + \epsilon' + \delta v' - nt - \epsilon - \delta v) = \sin. (n't - nt + \epsilon' - \epsilon) + \cos. (n't - nt + \epsilon' - \epsilon). \delta v' \text{ &c. } \text{ by substituting for } \delta r', \text{ we shall have } \frac{d.(a,a')^{(1)}}{da'} \cdot \delta r' = \frac{d.(a,a')^{(1)}}{da'} \cdot m''E'. \cos. (2.(n''t - n't + \epsilon'' - \epsilon')), \text{ which when multiplied into } \sin. (n't - nt + \epsilon' - \epsilon) \text{ gives a term of the form } -m''.E''. \sin. (nt - 3n't + 2n''t + \epsilon - 3\epsilon' + 2\epsilon''), \text{ in like manner by substituting for } \delta v', \text{ we obtain } \cos. (n't - nt + \epsilon' - \epsilon'). \delta v' = m''.F''. \sin. (nt - 3n't + 2n''t + \epsilon - 3\epsilon' + 2\epsilon''), \text{ hence if we substitute for } dv \text{ its value, there will result in the term } m'.(r,r')^{(1)}.dv. \sin. (v' - v), \text{ the expression which is given in the text.}$

contains. The expressions of  $\frac{\delta r}{a}$  and of  $\delta v$  of N°. 50, being applied to the action of  $m''$  on  $m'$ , give, when the terms which have  $n' - 2n''$  for a divisor are retained, and observing that  $n''$  is very nearly equal to  $\frac{n'}{2}$ ,

$$\frac{E''}{a'} = \frac{\frac{1}{2}n'^2 \cdot \left\{ a'^2 \cdot \left\{ \frac{d \cdot (a'a'')^{(2)}}{da'} \right\} + \frac{2n'}{n' - n''} \cdot a' \cdot (a'a'')^{(2)} \right\}}{(n' - 2n'').(3n' - 2n'')}^*$$

$$F'' = \frac{2E''}{a'};$$

therefore we shall have

$$dR = \frac{m' \cdot m'' \cdot ndt}{2} \cdot E'' \cdot \left\{ \frac{2 \cdot (a'a')^{(1)}}{a'} - \frac{(d \cdot (a'a')^{(1)})}{da'} \right\}$$

$$\times \sin. (nt - 3n't + 2n''t + \varepsilon' + 2\varepsilon'') = -\frac{1}{2} \cdot \frac{da}{a^2}.$$

This value of  $\frac{da}{a^2}$  being substituted in the values of  $\frac{d^2\zeta}{dt^2}$ ,  $\frac{d^2\zeta'}{dt^2}$ , and  $\frac{d^2\zeta''}{dt^2}$ , will give, because  $n$  is very nearly equal to  $2n'$ , and  $n'$  is very nearly equal to  $2n''$ ;

\* In page 296, if we substitute for  $\frac{dA^{(2)}}{da'}$ ,  $A^{(2)}$ , their values, the coefficient of cos.

$$2.(n''t - n't + \varepsilon'' - \varepsilon') \text{ becomes } = \frac{m''}{2} \cdot n'^2 \cdot a'^2 \cdot \frac{d(a'a'')^{(2)}}{da'} + \frac{2n'}{n' - n''} \cdot a' \cdot (a'a'')^{(2)} = \frac{m''E''}{a'};$$

in like manner the coefficient of sin.  $2.(n''t - n't + \varepsilon'' - \varepsilon')$  in the expression for  $\delta v'$ , given in page 298,  $= 2n'^3 \cdot a'^2 \cdot \frac{d(a'a'')^{(2)}}{da'} + \frac{2n'}{n' - n''} \cdot a' \cdot (a'a'')^{(2)}$

$$\frac{2n'}{2} \cdot (n' - 2n'').(3n' - 2n'') = F'' = \frac{2E''}{a'}.$$

$$\frac{d^2\zeta}{dt^2} - \frac{3d^2\zeta'}{d^2t} + \frac{2d^2\zeta''}{dt^2} = \epsilon n^2 \cdot \sin. (nt - 2nt' + 2n''t'' + \epsilon - 3\epsilon' + 2\epsilon'') ; *$$

( $\epsilon$  being made, in order to abridge, equal to

$$\frac{3}{2} \cdot E' \cdot \left\{ 2 \cdot (a, a')^{(1)} - a' \cdot \left\{ \frac{d \cdot (a, a')^{(1)}}{da'} \right\} \right\} \cdot \left\{ \frac{a''}{a'} \cdot m' \cdot m'' + \frac{9}{4} \cdot m \cdot m'' + \frac{a''}{4a'} \cdot m \cdot m' \right\} ;$$

Or more accurately,

$$\frac{d^2\zeta}{dt^2} - \frac{3d^2\zeta'}{d^2t} + \frac{2d^2\zeta''}{dt^2} = \epsilon \cdot n^2 \cdot \sin. (\zeta - 3\zeta' + 2\zeta'' + \epsilon - 3\epsilon' + 2\epsilon'') ;$$

so that if we assume

$$V = \zeta - 3\zeta' + 2\zeta'' + \epsilon - 3\epsilon' + 2\epsilon'',$$

we shall have

$$\frac{d^2V}{dt^2} = \epsilon \cdot n^2 \cdot \sin. V.$$

As the mean distances  $a, a', a''$ , and also the quantity  $n$  vary very little, we can in this equation consider  $\epsilon n^2$  as a constant quantity. By integrating it, we obtain

### 3 c 2

\*  $\frac{d^2\zeta}{dt^2} = -\frac{3}{2} \cdot n^{\frac{1}{2}} \frac{da}{a^2}$ , therefore multiplying by  $n$ , we obtain the coefficient of  $\frac{da}{a^2} = -\frac{3}{2} \cdot n^{\frac{4}{2}} = -\frac{3}{2} \cdot n^2 a$ , therefore by substituting for  $\frac{da}{a^2}$ , we obtain  $\frac{d^2\zeta}{dt^2} = \frac{3}{2} \cdot E'' \cdot \left( \frac{2 \cdot (a, a')^{(1)}}{a'} - \frac{d \cdot (a, a')^{(1)}}{da'} \right) \cdot n^2 \cdot m' \cdot m'' \cdot a \cdot \sin. (nt - 2n't + 3n''t + \epsilon - 3\epsilon' - 2\epsilon'')$ ; in like manner the coefficient of  $\frac{da}{a^2}$  in the value of  $\frac{d^2\zeta'}{dt^2} = \frac{3}{2} \cdot \frac{mn'^{\frac{4}{2}}}{m' \cdot n} \cdot \frac{3\frac{n''}{2}}{\frac{n'}{2}} = \frac{9}{2} \cdot \frac{mn'^2 a'}{m' \cdot n}$ ,

which being multiplied into  $m''m'ndt$ , gives (by substituting for  $n'^2$ , its value  $\left(\frac{n}{2}\right)^2$ ) —

$$\frac{3d^2\zeta'}{dt^2} = \frac{3}{2} \cdot E'' \cdot \left( \frac{2 \cdot (a, a')^{(1)}}{a'} - \frac{d \cdot (a, a')^{(1)}}{da'} \right) \frac{9}{4} \cdot n^2 \cdot m \cdot m'' \cdot \sin. (nt - 3n't + 2n''t + \epsilon - 3\epsilon' + 2\epsilon'').$$

The value of  $\frac{2d^2\zeta''}{dt^2}$  may be obtained in a similar manner.

$$dt = \frac{\pm dV}{\sqrt{c - 2\epsilon n^2 \cdot \cos V}},$$

$c$  being a constant arbitrary quantity. From the different values of which this constant is susceptible, the three following cases arise.

If  $c$  be positive and greater than  $\pm 2\epsilon n^2$ , the angle  $V$  will increase continually; and this will be the case, if, at the commencement of the motion  $(n - 3n' + 2n'')^2$  is greater\* than  $\pm 2\epsilon n^2 \cdot (1 \mp \cos V)$ , the superior, or lower signs having place according as  $\epsilon$  is positive or negative. It is easy to be assured, and shall we point it out particularly in the theory of the satellites of Jupiter, that  $\epsilon$  is a positive quantity relative to the three first satellites; therefore, supposing  $\mp \omega = \pi - V$ ,† ( $\pi$  being the semicircumference) we shall have

$$dt = \frac{d\omega}{\sqrt{c + 2\epsilon n^2 \cdot \cos \omega}}.$$

In the interval from  $\omega = 0$ , to  $\omega = \frac{\pi}{2}$ ; the radical  $\sqrt{c + 2\epsilon n^2 \cdot \cos \omega}$  is greater than  $\sqrt{2\epsilon n^2}$ , when  $c$  is equal to or greater than  $2\epsilon n^2$ ; therefore, the time  $t$  in which the angle  $\omega$  passes from zero to a right angle, is less than  $\frac{\pi}{2n\sqrt{2\epsilon}}$ . The value of  $\epsilon$  depends on the masses  $m, m', m''$ . The inequalities which have been observed in the motions of the three first satellites of Jupiter, and which we have already adverted to, assign relations between their masses and that of Jupiter, from which it

\* If  $c$  be positive and greater than  $\pm 2\epsilon n^2$ , the angle  $V$  must always increase, for the quantity under the radical sign can never be equal to cypher;  $c - 2\epsilon n^2 \cdot \cos V = \left(\frac{dV}{dt}\right)^2 = (n - 3n' + 2n'')^2$ , if this quantity be greater than  $\pm 2\epsilon n^2 \cdot (1 \mp \cos V)$ ,  $c - 2\epsilon n^2 \cdot \cos V$  must be greater than  $\pm 2\epsilon n^2 \cdot (1 \mp \cos V)$ ; i. e.  $c$  must be greater than  $2\epsilon n^2$ .

† By making  $\mp \omega = \pi - V$ , we get rid of the ambiguity of sign in the value of  $dt$ .

follows that  $\frac{\pi}{2n\sqrt{2\epsilon}}$  is less than\* two years, as we shall see in the theory of these satellites. Therefore the angle  $\varpi$  passes from zero to a right angle in less than two years; now from observations made on Jupiter's satellites, it appears that since their discovery, the angle  $\varpi$  has been either equal to cypher, or insensible, consequently the case which we have examined, is not that of the three first satellites of Jupiter.

If the constant  $c$  is less than  $\pm 2\epsilon n^2$ , the angle  $V$  will only oscillate, it will never attain to two right angles, if  $\epsilon$  be negative, since then the radical  $\sqrt{c - 2\epsilon n^2} \cdot \cos V$  will become imaginary;† it will be never equal to cypher, if  $\epsilon$  is positive. In the first case, its value will be alternately greater or less than cypher; in the second case, it will be alternately greater or less than two right angles. From all observations made on the three first satellites of Jupiter, it appears that this second case, is that of these stars, therefore the value of  $\epsilon$  ought to be positive relatively to them, and as the theory of gravity assigns a positive value to  $\epsilon$ , we ought to consider this phenomenon as an additional confirmation of this theory.

Since according to observation, the angle  $\varpi$  in the equation

\* As  $n = \frac{2\pi}{P}$ ,  $P$  being the time of revolution of the first satellite, we have  $t \angle \frac{P}{4\sqrt{2\epsilon}}$ ; the value of  $\epsilon$  depends on the masses  $m, m', m''$ , and also on  $n, n', n''$ , these last

are had by knowing the periodic times of the three first satellites, and the first are determined by their effects in producing certain inequalities, and are obtained in the same manner as the masses of Venus, Mercury, and Mars, are determined from certain effects which they produce on the earth's orbit.

† When  $c$  is negative and less than  $\mp 2\epsilon n^2$ , the radical is evidently imaginary when  $V=\pi$ ; ∵  $V$  can never be = to  $\pi$ , and it must be alternately positive and negative, its mean value being equal to *cypher*. If  $\epsilon$  is positive, the radical is evidently imaginary when  $V=0$ ; ∵ in this case  $V$  can never be = 0, its value is therefore periodic, and in its mean state is equal to  $\pi$ .

$$dt = \frac{d\omega}{\sqrt{c+2\epsilon n^2 \cdot \cos \omega}}, *$$

must be always very small, we can suppose  $\cos \omega = 1 - \frac{1}{2}\omega^2$ , and the preceding equation will give by integrating it,

$$\omega = \lambda \cdot \sin(nt\sqrt{\epsilon} + \gamma),$$

$\lambda$  and  $\gamma$  being two constant arbitrary quantities, which can be determined by observation alone. Hitherto, it has not indicated this inequality, which proves that it is extremely small.

From the preceding analysis the following consequences may be inferred. Since the angle  $nt - 3n't + 2n''t + \varepsilon - 3\varepsilon' - 2\varepsilon''$  only oscillates on one side or other of two right angles, its mean value is equal to two right angles; therefore we shall have, if we only consider mean quantities,  $n - 3n' + 2n'' = 0$ ; that is to say, *the mean motion of the first satellite plus twice that of the third, minus three times that of the second, is exactly and constantly equal to cypher*. It is not necessary that this equality should accurately obtain at the commencement, which would be extremely improbable, it is sufficient that it should be nearly the case, and that  $n - 3n' + 2n''$ , should be, abstracting from the sign, less than  $\lambda \cdot n \cdot \sqrt{\epsilon}$ ; and then the mutual attraction of these three satellites would have rendered this relation rigorously exact. We have therefore  $n - 3n' + 2n''$  equal to two right angles; hence, the mean longitude of the first satellite, minus three times the mean longitude of the second, plus twice that of the third is exactly and constantly equal to

\* The equation  $\frac{dt}{\sqrt{c+2\epsilon n^2 \cdot \cos \omega}}$ , is that of a pendulum whose length is  $\frac{2g}{\epsilon i^2}$ , *i.* being the number of seconds in a revolution of the first satellite, the amplitude of the arc of vibration being  $-\frac{c}{2\epsilon i^2}$ .

\* Or in other words, at the origin of the motion, it should be comprised within the limits  $\pm \lambda \cdot n \cdot \sqrt{\epsilon}$ .

two right angles. In consequence of this theorem, the preceding values of  $\delta r'$  and  $\delta v'$  are reduced to the following,

$$\begin{aligned}\delta r' &= (m.G - m'E''). \cos. (n't - nt + \epsilon' - \epsilon);^* \\ \delta v' &= (m.H - m'F''). \sin. (n't - nt + \epsilon' - \epsilon).\end{aligned}$$

The two inequalities in the motion of  $m'$ , arising from the action of  $m$  and of  $m''$ , are consequently confounded into one, and will be always combined. It follows also, that the three first satellites can never be eclipsed together; they cannot be seen together from Jupiter, neither in opposition nor in conjunction with the sun; for it is easy to perceive that the preceding theorems obtain equally for the mean sy-nodic motions, and the mean synodic longitudes of the three satellites. These two theorems likewise obtain, notwithstanding the changes which the mean motions of the satellites may experience, either from a cause similar to that which alters the mean motion of the moon, or from the re-sistance of a very rare medium. It is evident that if these different causes operated it would be merely requisite to add to the value of  $\frac{d^2V}{dt^2}$ , a

quantity of the form  $\frac{d^2\psi}{dt^2}$ , which can only become sensible by inte-grations; supposing therefore  $V = \pi - \omega$ , and  $\omega$  very small, the dif-ferential equation in  $V$  will become

$$0 = \frac{d^2\omega}{dt^2} + \epsilon n^2 \cdot \omega + \frac{d^2\psi}{dt^2}.$$

As the period of the angle  $nt\sqrt{\epsilon}$  embraces but a very few number of years, while the quantities contained† in  $\frac{d^2\psi}{dt^2}$  are either constant or

\* For  $2n''t + 2\epsilon'' - 2n't - 2\epsilon = \pi + n't - ut + \epsilon' - \epsilon$ , ∵  $m''.E'' \cdot \cos. 2.(n''t - n't + \epsilon'' - \epsilon) = -m''.E'' \cdot \cos. (n't - nt + \epsilon' - \epsilon)$ , in a similar manner, for the value of  $m''.F'' \cdot \sin. 2.(n''t - n't + \epsilon'' - \epsilon)$  may be substituted  $-m''.F'' \cdot \sin. (n't - nt + \epsilon' - \epsilon)$ .

† The period of the variation of  $\omega$ , and ∵ of  $V$  will be determined by means of the

extend to several centuries, we shall obtain very nearly, by integrating the preceding equation

$$\varpi = \lambda \cdot \sin. (nt \cdot \sqrt{\epsilon} + \gamma) - \frac{d^2 \psi}{\epsilon n^2 \cdot dt^2}.$$

Thus the value of  $\varpi$  will be always extremely small, and the secular equations of the mean motions of the three first satellites will be co-ordinated by the mutual action of these stars, so that the secular equation of the first plus twice that of the third, minus three times that of the second, is equal to cypher.

The preceding theorems establish between the six constants  $n, n', n'', \epsilon, \epsilon', \epsilon''$ , two equations of condition by means of which these arbitrary quantities are reduced to four. However they are replaced by the two arbitrary quantities  $\lambda$  and  $\gamma$ , of the value of  $\varpi$ . This value is distributed between the three satellites in such a manner, that naming  $p, p', p''$ , the coefficients of  $\sin. (nt \cdot \sqrt{\epsilon} + \gamma)$  in the expressions for  $v, v', v''$ ; those coefficients are in the ratio of the preceding values of  $\frac{d^2 \zeta}{dt^2}, \frac{d^2 \zeta'}{dt^2}, \frac{d^2 \zeta''}{dt^2}$ , and moreover, we have  $p - 3p' + 2p'' = \lambda$ . Hence, results in the mean motions of the three first satellites of Jupiter, an inequality which differs for each of them in the value of its coefficient, and which produces in these motions a species of vibration the extent of which is arbitrary. It appears from observation that it is insensible.

67. Let us now consider the variations of the excentricities and

equation  $nt \cdot \sqrt{\epsilon} = 2\pi$ ,  $\therefore$  as  $nP = 2\pi$ ,  $t = \frac{P}{\sqrt{\epsilon}}$ ; hence the two limits of  $t$  depend on those of  $\epsilon$ .

The integral of the equation  $\frac{d^2 \varpi}{dt^2} + \epsilon n^2 \cdot \varpi = 0$ , is  $\varpi = \lambda' \cdot \sin. (nt \cdot \sqrt{\epsilon} + \gamma)$ ; and in the equation  $\varpi = \lambda \cdot \sin. (nt \cdot \sqrt{\epsilon} + \gamma) - \frac{d^2 \psi}{\epsilon n^2 \cdot dt^2}$ ; the mean value of  $\frac{d\varpi}{dt}$ , and  $\therefore$  of  $n - 3n' + 2n'' = 0$ .

perihelias of the orbits. For this purpose, let the expressions of  $df$ ,  $df'$ ,  $df''$ , found in N°. 64, be resumed : naming  $r$  the radius vector of  $m$ , projected on the plane of  $x$ , and of  $y$ ;  $v$  the angle which this projection makes with the axis of  $x$ , and  $s$  the tangent of latitude of  $m$  above the same plane ; we shall have

$$x = r \cdot \cos. v; \quad y = r \cdot \sin. v; \quad z = rs;$$

hence it is easy to conclude

$$x \cdot \left\{ \frac{dR}{dy} \right\} - y \cdot \left\{ \frac{dR}{dx} \right\} = \left\{ \frac{dR}{dv} \right\};$$

$$x \cdot \left\{ \frac{dR}{dz} \right\} - z \cdot \left\{ \frac{dR}{dx} \right\} = (1+s^2) \cdot \cos. v \cdot \left\{ \frac{dR}{ds} \right\} - rs \cdot \cos. v \cdot \left\{ \frac{dR}{dr} \right\} + s \cdot \sin. v \cdot \left\{ \frac{dR}{dv} \right\};^*$$

$$y \cdot \left\{ \frac{dR}{dz} \right\} - z \cdot \left\{ \frac{dR}{dy} \right\} = (1+s^2) \cdot \sin. v \cdot \left\{ \frac{dR}{ds} \right\} - rs \cdot \sin. v \cdot \left\{ \frac{dR}{dr} \right\} - s \cdot \cos. v \cdot \left\{ \frac{dR}{dv} \right\},$$

moreover, by N°. 64, we have

$$+ \left( \frac{dR}{dx} \right) = \left( \frac{dR}{dr} \right) \cdot \frac{dr}{dx} + \left( \frac{dR}{dv} \right) \cdot \frac{dv}{dx} + \left( \frac{dR}{ds} \right) \cdot \frac{ds}{dx}, \quad \left( \frac{dR}{dy} \right) = \left( \frac{dR}{dr} \right) \cdot \frac{dr}{dy} + \left( \frac{dR}{dv} \right) \cdot$$

$$\frac{dv}{dy} + \left( \frac{dR}{ds} \right) \cdot \frac{ds}{dy}; \quad r = \sqrt{x^2+y^2}, \quad \frac{x}{r} = \cos. v, \quad - \frac{dv}{dx}, \quad \sin. v = \frac{y^2}{r^3}; \quad s =$$

$$\frac{z}{\sqrt{x^2+y^2}}, \quad \therefore \frac{ds}{dx} = - \frac{zx}{r^3}; \quad \frac{dv}{dy} \cdot \cos. v = \frac{x^2}{r^3}; \quad \frac{ds}{dy} = - \frac{zy}{r^3}; \quad \text{hence } x \cdot \frac{dR}{dy}$$

$$- y \cdot \frac{dR}{dx} = \left( \frac{dR}{dr} \right) \cdot \frac{xy-yx}{r} + \left( \frac{dR}{dv} \right) \cdot \left( \frac{x^2+y^2}{r^2} \right) + \left( \frac{dR}{ds} \right) \cdot \left( \frac{xyz-xyz}{r^3} \right) = \left( \frac{dR}{dv} \right), \quad \text{in like manner}$$

$$\text{manner } x \cdot \left( \frac{dR}{dz} \right) - z \cdot \left( \frac{dR}{dx} \right) = \frac{dR}{ds} \left( \frac{r \cdot \cos. v}{r} + \frac{r^2 \cdot s^2 \cdot \cos. v}{r^3} \right) - \frac{dR}{dr} \cdot r \cdot \cos. v \cdot s + \frac{dR}{dv} \cdot$$

$\frac{rs \cdot \sin. v}{r} =$  the expression in the text, and by a similar process the remaining terms may be obtained.

$$xdy - ydx = cdt; \quad xdz - zdx = c'dt, \quad ydz - zdy = c''dt;$$

these differential equations in  $f, f', f''$ , will consequently become

$$\begin{aligned} df &= -dy \cdot \left\{ \frac{dR}{dv} \right\} - dz \cdot \left\{ (1+s^2) \cdot \cos. v. \left\{ \frac{dR}{ds} \right\} - rs. \cos. v. \left\{ \frac{dR}{dr} \right\} + \right. \\ &\quad \left. s. \sin. v. \left\{ \frac{dR}{dv} \right\} \right\} \\ &\quad - cdt. \left\{ \sin. v. \left\{ \frac{dR}{dr} \right\} + \frac{\cos. v}{r} \left\{ \frac{dR}{dv} \right\} - \frac{s. \sin. v. \left\{ \frac{dR}{ds} \right\}}{r} \right\} - \frac{c'. dt}{r} \cdot \left\{ \frac{dR}{ds} \right\}; \\ df' &= dx \cdot \left\{ \frac{dR}{dv} \right\} - dz \cdot \left\{ (1+s^2) \cdot \sin. v. \left\{ \frac{dR}{ds} \right\} - rs. \sin. v. \left\{ \frac{dR}{dr} \right\} - s. \right. \\ &\quad \left. \cos. v. \left\{ \frac{dR}{dv} \right\} \right\} \\ &\quad + cdt. \left\{ \cos. v. \left\{ \frac{dR}{dr} \right\} - \frac{\sin. v}{r} \cdot \left\{ \frac{dR}{dv} \right\} - \frac{s. \cos. v. \left\{ \frac{dR}{ds} \right\}}{r} \right\} - \frac{c''. dt}{r}. \\ &\quad \left\{ \frac{dR}{ds} \right\}; \\ df'' &= dx \cdot \left\{ (1+s^2) \cdot \cos. v. \left\{ \frac{dR}{ds} \right\} - r.s. \cos. v. \left\{ \frac{dR}{dr} \right\} + s. \sin. v. \left\{ \frac{dR}{dv} \right\} \right\} \\ &\quad + dy \cdot \left\{ (1+s^2) \cdot \sin. v. \left\{ \frac{dR}{ds} \right\} - r.s. \sin. v. \left\{ \frac{dR}{dr} \right\} - s. \cos. v. \left\{ \frac{dR}{dv} \right\} \right\} \\ &\quad + c'. dt. \left\{ \cos. v. \left\{ \frac{dR}{dr} \right\} - \frac{\sin. v}{r} \left\{ \frac{dR}{dv} \right\} - \frac{s. \cos. v. \left\{ \frac{dR}{ds} \right\}}{r} \right\} \\ &\quad + c''. dt. \left\{ \sin. v. \left\{ \frac{dR}{dr} \right\} + \frac{\cos. v}{r} \cdot \left\{ \frac{dR}{dv} \right\} - \frac{s. \sin. v. \left\{ \frac{dR}{ds} \right\}}{r} \right\}. \end{aligned}$$

The quantities  $c', c''$  depend, as we have seen in N°. 64, on the inclination of the orbit of  $m$  to the fixed plane, so that these quantities become equal to zero, if this inclination is nothing ; besides it is easy to perceive, from the nature of  $R$ , that  $\left\{ \frac{dR}{ds} \right\}$  is of the order of the inclinations of the orbits ; therefore the products and the squares of the inclinations of the orbits being neglected, the preceding expressions for  $df$ , and  $df'$ , will become

$$df = -dy \cdot \left( \frac{dR}{dv} \right) - cdt \cdot \left\{ \sin. v \cdot \left( \frac{dR}{dr} \right) + \frac{\cos. v}{r} \left( \frac{dR}{dv} \right) \right\};$$

$$df' = dx \cdot \left( \frac{dR}{dv} \right) + cdt \cdot \left\{ \cos. v \cdot \left( \frac{dR}{dr} \right) - \frac{\cos. v}{r} \cdot \left( \frac{dR}{dv} \right) \right\};$$

but we have

$$dx = d.(r. \cos. v); \quad dy = d.(r. \sin. v); \quad cdt = xdy - ydx = r^2 dv;$$

therefore we shall have

$$df = -(dr. \sin. v + 2rdv. \cos. v) \cdot \left( \frac{dR}{dv} \right) - r^2. dv. \sin. v. \left( \frac{dR}{dr} \right); *$$

$$df' = (dr. \cos. v - 2rdv. \sin. v) \cdot \left( \frac{dR}{dv} \right) + r^2. dv. \cos. v. \left( \frac{dR}{dr} \right).$$

These equations will be more exact, if we assume for the fixed plane of  $x$  and  $y$ , that of the orbit of  $m$  at a given epoch; for then  $c'$ ,  $c''$ , and  $s$ , are of the order of the disturbing forces; consequently the quantities which are neglected are of the order of the squares of the disturbing forces multiplied by the square of the respective inclination of the two orbits of  $m$  and of  $m'$ .

The values of  $r$ ,  $dr$ ,  $dv$ ,  $\left( \frac{dR}{dr} \right)$ ,  $\left( \frac{dR}{dv} \right)$  remain evidently the same, whatever be the position of the point from which the longitudes are reckoned; but if  $v$  be diminished by a right angle,  $\sin v$  will be changed into  $-\cos. v$ , and  $\cos. v$  will be changed into  $\sin v$ , consequently the expression for  $df$  will be changed into that of  $df'$ ; hence it follows, that if the value of  $df$  be developed into a series of the sines and cosines of angles increasing proportionally to the time, the value of  $df'$  will

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\*  $\frac{cdt. \cos. v}{r} = rdv. \cos. v$ ,  $\because$  the coefficient of  $\left( \frac{dR}{dv} \right)$  in the value of  $df$  is  $2rdv. \cos. v$ .

be obtained by diminishing in the first, the angles  $\epsilon, \epsilon', \varpi, \varpi', \theta$  and  $\theta'$ , by a right angle.

The quantities  $f$  and  $f'$  determine the position of the perihelion and the eccentricity of the orbit; in fact, we have seen in N°. 64, that

$$\tan. I = \frac{f'}{f},$$

$I$  being the longitude of the perihelion referred to the fixed plane. When this plane is that of the primitive orbit of  $m$ , we have (as far as quantities of the order of the squares of the disturbing forces multiplied by the square of the respective inclination of the orbits)  $I = \varpi$ ,  $\varpi$  being the longitude of the perihelion reckoned on the orbit, therefore we shall then have

$$\text{tang. } \varpi = \frac{f'}{f};$$

which gives,

$$\sin. \varpi = \frac{f'}{\sqrt{f^2 + f'^2}}; \cos. \varpi = \frac{f}{\sqrt{f^2 + f'^2}};$$

hence results, by N°. 64,

$$\mu e = \sqrt{f^2 + f'^2 + f''^2}; f'' = \frac{f'c' - fc''}{c};$$

Thus  $c'$  and  $c''$  being on the preceding hypothesis of the order of the disturbing forces,  $f''$  is of the same order, and neglecting the terms of the square of these forces, we shall have  $\mu e = \sqrt{f^2 + f'^2}$ . If in the expressions of  $\sin. \varpi$ ,  $\cos. \varpi$ , we substitute instead of  $\sqrt{f^2 + f'^2}$ , its value  $\mu e$ , we shall have

$$\mu e. \sin. \varpi = f'; \mu e. \cos. \varpi = f;$$

these two equations will determine the eccentricity and the position of the perihelion, and we can easily infer

$$\mu^2 \cdot ede = f df + f' df' ; \quad \mu^2 \cdot e^2 \cdot d\omega = f df' - f' df .^*$$

By assuming for the plane of  $x$  and of  $y$ , that of the orbit of  $m$ ; we shall have by N°s. 19 and 20, in the case of invariable ellipses,

$$r = \frac{a(1-e^2)}{1+e \cdot \cos(v-\omega)} ; \quad dr = \frac{r^2 \cdot dv \cdot e \cdot \sin(v-\omega)}{a(1-e^2)} ;$$

$$r^2 dv = a^2 ndt \cdot \sqrt{1-e^2} ;$$

and by N°. 63, these equations subsist also in the case of variable ellipses; the expressions of  $df$  and of  $df'$  consequently become

$$df = - \frac{andt}{\sqrt{1-e^2}} \cdot (2 \cdot \cos v + \frac{3}{2} e \cdot \cos \omega + \frac{1}{2} e \cdot \cos(2v-\omega)) \cdot \left( \frac{dR}{dv} \right) +$$

$$- a^2 ndt \cdot \sqrt{1-e^2} \cdot \sin v \cdot \left( \frac{dR}{dr} \right) ;$$

$$df' = - \frac{andt}{\sqrt{1-e^2}} \cdot (2 \cdot \sin v + \frac{3}{2} e \cdot \sin \omega + \frac{1}{2} e \cdot \sin(2v-\omega)) \cdot \left( \frac{dR}{dv} \right)$$

$$+ a^2 ndt \cdot \sqrt{1-e^2} \cdot \cos v \cdot \left( \frac{dR}{dr} \right) ;$$

therefore

$$* \quad \mu^2 \cdot e^2 = f^2 + f'^2 ; \quad \because \mu^2 \cdot ede = f df + f' df' ; \quad \tan \omega = \frac{f'}{f} ; \quad \therefore \frac{d\omega}{\cos^2 \pi} = \frac{f df' - f' df}{f^2} ,$$

$\therefore$  substituting for  $f^2$ , we have  $\mu^2 e^2 \cdot d\omega = f df' - f' df$ .

$$\dagger \text{ Substituting for } dr \text{ and } r^2 dv, \text{ we obtain } df = - \frac{r^2 \cdot dv \cdot e \cdot \sin(v-\omega) \cdot \sin v}{a(1-e^2)} -$$

$$\frac{2a^2 ndt \cdot \sqrt{1-e^2} \cdot (\cos v \cdot (1+e) \cdot \cos(v-\omega))}{a(1-e^2)} = - \frac{andt}{\sqrt{1-e^2}} \cdot (2 \cdot \cos v + e \cdot \sin v \cdot \sin(v-\omega))$$

$$+ 2e \cdot \cos v \cdot \cos(v-\omega) ; \quad e \cdot \sin v \cdot \sin(v-\omega) = e \cdot \sin^2 v \cdot \cos \omega - e \cdot \sin v \cdot \cos v \cdot \sin \omega .$$

$$2e \cdot \cos v \cdot \cos(v-\omega) = 2e \cdot \cos^2 v \cdot \cos \omega + 2e \cdot \sin v \cdot \cos v \cdot \sin \omega = e \cdot (\sin^2 v + \cos^2 v) \cdot \cos \omega$$

$$+ e \cdot \cos v \cdot (\cos v \cdot \cos \omega + \sin v \cdot \sin \omega) = e \cdot \cos v \cdot \cos(v-\omega) = e \cdot \frac{\cos(2v-\omega)}{2} + e \cdot \frac{\cos \omega}{2} ,$$

$\therefore$  by making similar terms to coalesce we obtain the expression given in the text, we can in a similar manner derive the other expressions.

$$\begin{aligned}
 ed\varpi &= -\frac{andt}{\mu\sqrt{1-e^2}} \cdot \sin(v-\varpi) \cdot (2+e \cdot \cos(v-\varpi)) \cdot \left(\frac{dR}{dv}\right)^* \\
 &\quad + \frac{a^2 \cdot ndt \cdot \sqrt{1-e^2}}{\mu} \cdot \cos(v-\varpi) \cdot \left(\frac{dR}{dr}\right); \\
 de &= -\frac{andt}{\mu\sqrt{1-e^2}} \cdot (2 \cdot \cos(v-\varpi) + e + e \cdot \cos^2(v-\varpi)) \cdot \left(\frac{dR}{dv}\right) \\
 &\quad - \frac{a^2 ndt}{\mu} \cdot \sqrt{1-e^2} \cdot \sin(v-\varpi) \cdot \left(\frac{dR}{dr}\right).
 \end{aligned}$$

This expression for  $de$  may be made to assume a form which in several circumstances is more commodious. For this purpose, it may be observed, that  $dr \cdot \left(\frac{dR}{dr}\right) = dR - dv \cdot \left(\frac{dR}{dv}\right)$ , by substituting in place of  $r$ , and  $dr$  their preceding values, we shall have

$$r^2 \cdot dv \cdot e \cdot \sin(v-\varpi) \cdot \left(\frac{dR}{dr}\right) = a \cdot (1-e^2) \cdot dR - a \cdot (1-e^2) \cdot dv \cdot \left(\frac{dR}{dv}\right);$$

but we have

$$r^2 \cdot dv = a^2 ndt \cdot \sqrt{1-e^2}; \quad dv = \frac{ndt \cdot (1+e \cdot \cos(v-\varpi))^2}{(1-e^2)^{\frac{3}{2}}};$$

$$\begin{aligned}
 *fdf' &= -\frac{\mu e \cdot andt}{\sqrt{1-e^2}} \cdot 2 \cdot \sin v \cdot \cos \varpi + \frac{3}{2} \cdot e \cdot \sin \varpi \cdot \cos \varpi + \frac{1}{2} e \cdot \sin(2v-\varpi) \cdot \cos \varpi \cdot \left(\frac{dR}{dv}\right) \\
 &\quad + \mu e a^2 ndt \cdot \sqrt{1-e^2} \cdot \cos v \cdot \cos \varpi \cdot \left(\frac{dR}{dr}\right) \\
 f' df &= -\frac{\mu e \cdot andt}{\sqrt{1-e^2}} \cdot 2 \cdot \cos v \cdot \sin \varpi + \frac{3}{2} \cdot e \cdot \sin \varpi \cdot \cos \varpi + \frac{1}{2} e \cdot \sin \varpi \cdot \cos(2v-\varpi) \cdot \left(\frac{dR}{dv}\right) \\
 &\quad - \mu e a^2 ndt \cdot \sqrt{1-e^2} \cdot \sin v \cdot \sin \varpi \cdot \left(\frac{dR}{dr}\right); \\
 \therefore \mu^2 e^2 d\varpi &= fdf' - f'df = -\frac{\mu e \cdot andt}{\sqrt{1-e^2}} \cdot 2 \cdot \sin(v-\varpi) + \frac{1}{2} e \cdot \sin(2v-2\varpi) \cdot \left(\frac{dR}{dv}\right) \\
 &\quad + \mu e a^2 ndt \cdot \sqrt{1-e^2} \cdot \cos(v-\varpi) \cdot \left(\frac{dR}{dr}\right),
 \end{aligned}$$

therefore,

$$\begin{aligned} & a^2ndt \cdot \sqrt{1-e^2} \cdot \sin(v-\varpi) \cdot \left( \frac{dR}{dr} \right) \\ & = \frac{a \cdot (1-e^2)}{e} \cdot dR - \frac{andt}{e \cdot \sqrt{1-e^2}} \cdot (1+e \cdot \cos(v-\varpi))^2 \cdot \left( \frac{dR}{dv} \right); \end{aligned}$$

therefore the preceding expression for  $dv$  will give

$$ede = \frac{andt \cdot \sqrt{1-e^2}}{\mu} \cdot \left( \frac{dR}{dv} \right) - \frac{a \cdot (1-e^2)}{\mu} \cdot dR.$$

This formula may be also obtained in a very simple manner, by the following method. By N°. 64, we have

$$\frac{dc}{dt} = y \cdot \left( \frac{dR}{dx} \right) - x \cdot \left( \frac{dR}{dy} \right) = - \left( \frac{dR}{dv} \right);$$

but by the same number we have  $c = \sqrt{\mu a \cdot (1-e^2)}$ , which gives

$$dc = - \frac{da \cdot \sqrt{\mu a \cdot (1-e^2)}}{2a} - \frac{ede \cdot \sqrt{\mu a}}{\sqrt{1-e^2}};$$

therefore†

$$ede = \frac{andt \cdot \sqrt{1-e^2}}{\mu} \cdot \left( \frac{dR}{dv} \right) + a \cdot (1-e^2) \cdot \frac{da}{2a^2};$$

then by N°. 64, we have

$$\frac{\mu da}{2a^2} = - dR;$$

which is evidently equal to the expression given in the text, the value of  $de$  may be obtained in a similar manner.

† Dividing both sides by  $\frac{\sqrt{\mu}}{\sqrt{1-e^2}}$ , and observing that  $an = \frac{\sqrt{\mu}}{a^2}$ , and  $\because \sqrt{\mu a} = \frac{\mu}{an}$ , we obtain the value of  $ede$ , which is given in the text.

thus we shall obtain the same expression for  $ede$ , as has been given above.

68. It has been observed in N°. 65, that if the squares of the disturbing forces are neglected, the variations of the greater axis and of the mean motion only contain periodic quantites, depending on the mutual configuration of the bodies  $m, m', m'', \&c.$  This is not the case with respect to the variations of the excentricities and of the inclinations : their differential expressions contain terms which are independent of this configuration, and which if they were rigorously constant would produce by integration terms proportional to the time, which would at length render the orbits extremely excentric, and very much inclined to each other ; consequently, the preceding approximations which depend on the small excentricity and inclination of the orbits, would become inadequate and even erroneous. But the terms which being apparently constant, enter into the differential expressions of the excentricities and inclinations, are functions of the elements of the orbits ; so that in fact they vary with extreme slowness in consequence of the changes which these elements experience. We may conceive therefore that there ought to result from them considerable inequalities, independent of the mutual configuration of the bodies of the system, the periods of which depend on the ratios of the masses  $m, m', m'', \&c.,$  to the mass  $M.$  These inequalities under the denomination of *secular inequalities*, have been already considered in Chapter VII. In order to determine them by this method, let the value of  $df$ , given in the preceding number, be resumed

$$df = -\frac{andt}{\sqrt{1-e^2}} \cdot (2 \cdot \cos. v + \frac{5}{2} \cdot e \cdot \cos. \pi + \frac{1}{2} \cdot e \cdot \cos.(2v-\pi)) \cdot \left( \frac{dR}{dv} \right)$$

$$- a^2ndt\sqrt{1-e^2} \cdot \sin. v \cdot \left( \frac{dR}{dr} \right).$$

In the developement of this equation we shall neglect the squares

and products of the excentricities and of the inclinations of the orbits; and amongst the terms depending on the excentricities and inclinations, we shall only retain those which are constant. Let us then suppose, as in N°. 48,

$$\begin{aligned}r &= a.(1+u), \quad r' = a'.(1+u'), \\v &= nt+\epsilon+v, \quad v' = n't+\epsilon'+v'.\end{aligned}$$

This being premised, if we substitute in place of  $R$ , its value found in N°. 48, observing that by the same N°. we have

$$\left\{ \frac{dR}{dr} \right\} = \frac{a}{r} \cdot \left\{ \frac{dR}{da} \right\} = (1-u) \cdot \left\{ \frac{dR}{da} \right\}$$

finally, if we substitute in place of  $u$ ,  $u'$ ,  $v$ ,  $v'$ , their values

$$\begin{aligned}-e \cdot \cos. (nt+\epsilon-\varpi), \quad -e' \cdot \cos. (n't+\epsilon'-\varpi'), \quad 2e \cdot \sin. (nt+\epsilon-\varpi), \\2e' \cdot \sin. (n't+\epsilon'-\varpi'),\end{aligned}$$

which are given in N°. 22, and if among the terms which depend on the first power of the excentricities, we only retain those which are constant, we shall find (the squares of the inclinations and excentricities being neglected,)

$$\begin{aligned}df &= \frac{am'ndt}{2} \cdot e \cdot \sin. \varpi \cdot \left\{ a \cdot \left\{ \frac{dA^{(0)}}{da} \right\} + a^2 \cdot \left\{ \frac{d^2 A^{(0)}}{da^2} \right\} \right\}^* \\&+ am'ndt \cdot e' \cdot \sin. \varpi' \cdot \left\{ A^{(1)} + \frac{1}{2}a \cdot \left\{ \frac{dA^{(1)}}{da} \right\} + \frac{1}{2}a' \cdot \left\{ \frac{dA^{(1)}}{da'} \right\} + \frac{1}{4}aa' \cdot \left\{ \frac{d^2 A^{(1)}}{da \cdot da'} \right\} \right\} \\&- am'ndt \cdot \Sigma \left\{ i \cdot A^{(i)} + \frac{1}{2}a \cdot \left\{ \frac{dA^{(i)}}{da} \right\} \right\} \cdot \sin. (i \cdot (n't - nt + \epsilon' - \epsilon) + nt + \epsilon);\end{aligned}$$

\*  $\sin. v = \sin. (nt + \epsilon + v) = \sin. (nt + \epsilon) \cdot \cos. v + \cos. (nt + \epsilon) \cdot \sin. v$ , now  $\sin. v = v - \frac{v^3}{3} + \text{&c.}$ ,  $\cos. v = 1 - \frac{v^2}{2} + \text{&c.}$ , hence substituting for  $v$ , its value  $2e \cdot \sin. (nt + \epsilon - \varpi)$ , and neglecting the square of  $e$ , we obtain  $\sin. (nt + \epsilon + v) = \sin. (nt + \epsilon) + \cos. (nt + \epsilon) \cdot$

The integral sign  $\Sigma$  extending in this expression, as in the value of  $R$  of N°. 48, to all entire values of  $i$ , as well positive as negative, the value  $i = 0$  being included.

$2e \cdot \sin.(nt + \epsilon - \varpi)$ , differencing  $R$  with respect to  $a$ , and retaining those terms only in which the first power of the excentricity or inclination can occur, and from which we may

obtain constant quantities, the first term of the expression for  $\frac{dR}{da}$  (differenced under these

restrictions) will give  $\frac{dA^{(0)}}{da}$ , the second term will give  $\frac{dA^{(0)}}{da^2}$ , and also  $a \cdot \frac{d^2A^{(0)}}{da^3}$ ,  $\because (1 -$

$u_i) \cdot \left(\frac{dR}{da}\right) = (1 + e \cdot \cos.(nt + \epsilon - \varpi)) \cdot \left(\left(\frac{dA^{(0)}}{da}\right) - \left(\left(\frac{dA^{(0)}}{da}\right) - a \cdot \left(\frac{d^2A^{(0)}}{da^2}\right)\right) \cdot e \cdot \cos.(nt + \epsilon - \varpi)$ ,

now this quantity should be multiplied into  $\sin. v$ , or into  $\sin.(nt + \epsilon) + \cos.(nt + \epsilon) \cdot 2e \cdot (\sin.(nt + \epsilon - \varpi))$ ; hence, performing this operation, neglecting the square of  $e$ , and we

shall have the coefficient of  $\frac{dA^{(0)}}{da} = 2e \cdot \cos.(nt + \epsilon) \cdot \sin.(nt + \epsilon - \varpi) + e \cdot \cos.(nt + \epsilon - \varpi) \cdot \sin.(nt + \epsilon) - e \cdot \cos.(nt + \epsilon - \varpi) \cdot \sin.(nt + \epsilon) = 2e \cdot \cos.(nt + \epsilon) \cdot \sin.(nt + \epsilon - \varpi) = e \cdot \sin.(2)$

$(nt + \epsilon) - e \cdot \sin.\varpi$ ; in like manner, the coefficient of  $-a \cdot \frac{d^2A}{da^2}$ , is  $e \cdot \sin.(nt + \epsilon) \cdot \cos.(nt + \epsilon - \varpi) = \frac{e}{2} \cdot \sin.2.(nt + \epsilon) - \varpi + \frac{e}{2} \cdot \sin.\varpi$ ; hence, by multiplying by  $\frac{m'}{2}$ , see N°.

48, the constant part of the second term of the value of  $df = \frac{a^2 m' n}{2} \cdot dt \cdot e \cdot \sin.\varpi$ .

$\left(\left(\frac{dA^{(0)}}{da}\right) + \frac{a}{2} \cdot \left(\frac{d^2A^{(0)}}{da^2}\right)\right)$ , in like manner, to obtain the coefficient of  $e \cdot \sin.\varpi'$ , let the

third and fourth terms of the value of  $R$  be differenced with respect to  $a$ , ( $i$  being equal to unity), which will give  $-a' \cdot \left(\frac{d^2A^{(1)}}{da da'}\right) \cdot e' \cdot \cos.(n't - nt + \epsilon' - \epsilon) \cdot \cos.(n't + i' - \varpi') - A^{(1)} \cdot 2e' \cdot \sin.(n't + \epsilon' - \varpi')$ .

$\sin.(nt + \epsilon)$  is the only part of the value of  $\sin.v$ , into which these quantities can be multiplied without introducing powers of  $e$  greater than the first,  $\because$  when for these quantities equivalent expressions are substituted, determined by the equations of the form  $\cos.a \cdot \cos.b = \frac{\cos.(a+b) + \cos.(a-b)}{2}$ ,  $\sin.a \cdot \sin.b = \cos.\frac{(a+b)}{2} - \cos.\frac{(a-b)}{2}$ ;

we shall obtain the second and fourth terms of the second line of the value of  $df$ ; in order to obtain the first and third terms, let the third and fourth terms be differenced, when  $i = 1$ , considering  $\epsilon$  as the variable, for we have  $\left(\frac{dR}{dv}\right) =$

$\left(\frac{dR}{d\epsilon}\right)$ , and then these terms become  $\frac{m'}{2} \cdot u'_i \cdot a' \cdot \frac{dA^{(1)}}{da'} \cdot \sin.(n't - nt + \epsilon' - \epsilon) + \frac{m'}{2} \cdot v'_i A^{(1)}$ .

By the preceding number, the value of  $df'$  will be obtained, if the angles  $\epsilon$ ,  $\epsilon'$ ,  $\varpi$ , and  $\varpi'$ , be diminished by a right angle in that of  $f$ ; hence we deduce

$$\begin{aligned} df' = & -\frac{am'ndt}{2} \cdot e \cdot \cos. \varpi \cdot \left\{ a \cdot \left\{ \frac{dA^{(0)}}{da} \right\} + \frac{1}{2}a^2 \cdot \left\{ \frac{d^2A^{(0)}}{da^2} \right\} \right\} \\ & - am'ndt \cdot e' \cdot \cos. \varpi' \cdot \left\{ A^{(1)} + \frac{1}{2}a \cdot \left\{ \frac{dA^{(1)}}{da} \right\} + \frac{1}{2}a' \cdot \left\{ \frac{dA^{(1)}}{da'} \right\} + \frac{1}{4} \cdot \left\{ \frac{d^2A^{(1)}}{da \cdot da'} \right\} \right\} \\ & + am'ndt \cdot \Sigma \left\{ iA^{(i)} + \frac{1}{2}a \cdot \left\{ \frac{dA^{(i)}}{da} \right\} \right\} \cdot \cos. (i \cdot (n't - nt + \epsilon' - \epsilon) + nt + \epsilon). \end{aligned}$$

Let us name, in order to abridge,  $X$  the part of the expression of  $df$ , contained under the sign  $\Sigma$ , and  $Y$  the part of the expression of  $df'$  contained under the same sign. Moreover, let us make as in N°. 55,

$$\begin{aligned} (0, 1) = & -\frac{m' \cdot n}{2} \cdot \left\{ a^2 \cdot \left\{ \frac{dA^{(0)}}{da} \right\} + \frac{1}{2} \cdot a^3 \cdot \left\{ \frac{d^2A^{(0)}}{da^2} \right\} \right\}; \\ [0, 1] = & \frac{m' \cdot n}{2} \cdot \left\{ aA^{(1)} - a^2 \cdot \left\{ \frac{dA^{(1)}}{da} \right\} - \frac{1}{2}a^3 \cdot \left\{ \frac{d^2A^{(1)}}{da^2} \right\} \right\}. \end{aligned}$$

It should be then observed, that the coefficient of  $e'dt \cdot \sin. \varpi'$ , in the expression of  $df$ , is reduced to  $[0, 1]$ , when we substitute in it, in place of the partial differences of  $A^{(1)}$  in  $a'$ , their values in partial differences relative to  $a$ ; finally, let, as in N°. 50,

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$\cos. (n't - nt + \epsilon' - \epsilon)$ , when we substitute for  $u'$  and  $v'$  their values, and proceed as before, we shall obtain, after the resulting quantities are multiplied by  $2 \cdot \cos. (nt + \epsilon)$ . (the only part of the value of  $\cos. v$  which can be taken into account); the first and third terms of the coefficient of  $e' \cdot \sin. \varpi'$ ; in order to obtain the variable part of the value of  $df$ ,  $e$ ,  $e'$ , do not occur; the first term of the value of  $R$ , must be differentiated with respect to  $v$ , or, what is the same thing with respect to  $\epsilon$ , and then multiplied into  $2 \cdot \cos. (nt + \epsilon)$ , this same term should be also differenced with respect to  $a$ , and then multiplied into  $\sin. (nt + \epsilon)$ .

$$\begin{aligned}e. \sin. \varpi &= h; \quad e'. \sin. \varpi' = h', \\e. \cos. \varpi &= l; \quad e'. \cos. \varpi' = l';\end{aligned}$$

which by the preceding number, gives  $f = \mu l$ ;  $f' = \mu h$ ; or simply,  $f = l$ ;  $f' = h$ , the mass of  $M$  being assumed as unity, and the mass  $m$  being neglected relatively to  $M$ ; we shall have

$$\begin{aligned}\frac{dh}{dt} &= (0, 1).l - [\underline{0, 1}].l' + am'.n.Y; \\ \frac{dl}{dt} &= -(0, 1).h + [\underline{0, 1}].h' - am'.n.X.\end{aligned}$$

Hence it is easy to infer, that if the sum of the terms analogous to  $am'nY$  be named ( $Y$ ), which terms arise from the action of each of the bodies  $m'$ ,  $m''$ , &c. on  $m$ ; if in like manner, the sum of the terms analogous to  $-am'nX$ , arising from the same action, be called ( $X$ ), finally, if we denote by one, two, &c., strokes, what the quantities ( $X$ ), ( $Y$ ),  $h$  and  $l$  become relatively to the bodies  $m'$ ,  $m''$ , &c.; we shall obtain the following system of differential equations :

$$\begin{aligned}\frac{dh}{dt} &= ((0, 1) + (0, 2) + \&c.).l - [\underline{0, 1}].l' - [\underline{0, 2}].l'' - \&c. + (Y); \\ \frac{dl}{dt} &= -((0, 1) + (0, 2) + \&c.).h + [\underline{0, 1}].h' + [\underline{0, 2}].h'' + \&c. + (X); \\ \frac{dh'}{dt} &= ((1, 0) + (1, 2) + \&c.).l' - [\underline{1, 0}].l - [\underline{1, 2}].l'' - \&c. + (Y'); \\ \frac{dl'}{dt} &= -((1, 0) + (1, 2) + \&c.).h' + [\underline{1, 0}].h + [\underline{1, 2}].h'' + \&c. + (X'). \\ &\&c.\end{aligned}$$

In order to integrate these equations, let it be observed that each of the quantities  $h$ ,  $l$ ,  $h'$ ,  $l'$ , &c., is made up of two parts, the one depending on the mutual configuration of the bodies  $m$ ,  $m'$ , &c., the other independent of this configuration, containing the secular variations of these quantities. We shall obtain the first part, if we consider that when we have regard to it solely,  $h$ ,  $l$ ,  $h'$ ,  $l'$ , &c., are of the order of the dis-

turbing masses, and consequently  $(0, 1).h$ ,  $(0, 1).l$ , are of the order of the squares of these masses. Neglecting quantities of this order, we shall have

$$\frac{dh}{dt} = (Y); \quad \frac{dl}{dt} = (X);$$

$$\frac{dh'}{dt} = (Y'); \quad \frac{dl'}{dt} = (X');$$

therefore,

$$h = \int (Y) dt; \quad l = \int (X) dt; \quad h' = \int (Y') dt; \quad l' = \int (X') dt; \quad \text{&c.}$$

If these integrals be taken, the elements of the orbits being considered constant; and if  $Q$  be what  $\int (Q) dt$  then becomes, and if  $\delta Q$  be the variation of  $Q$ , arising from that of the elements, we shall have

$$\int (Y) dt = Q - \int \delta Q;$$

But as  $Q$  is of the order of the perturbating masses, and as the variations of the elements are of the same order,  $\delta Q$  is of the order of the squares of these masses, therefore, if quantities of this order be neglected, we shall have

$$\int (Y) dt = Q.$$

We can therefore take the integrals  $\int (Y) dt$ ,  $\int (X) dt$ ,  $\int (Y') dt$ , &c., on the hypothesis that the elements of the orbits are constant, provided that we consider these elements as variable in the integrals; by this means we shall obtain in a very simple manner, the periodic parts of the expressions of  $h$ ,  $l$ ,  $h'$ , &c.

In order to obtain the parts of these expressions, which contain the secular inequalities, it is to be remarked, that they are furnished by the integration of the preceding differential equations deprived of their last terms  $(Y)$ ,  $(X)$ , &c.; for it is evident that the substitution of the periodic parts of  $h$ ,  $l$ ,  $h'$ , &c., will make these terms to disappear. But if these equations be deprived of their last terms, they will coin-

cide with the differential equations (*A*) of N°. 55, which we have already discussed in detail.

69. It has been observed in N°. 65, that if the mean motions  $nt$  and  $n't$  of the two bodies  $m$  and  $m'$ , are very nearly in the ratio of  $i$  to  $i'$ , so that  $i'n' - int$ , may be a very small quantity, very sensible inequalities may result in the mean motions of these bodies. This ratio of the mean motions may also produce sensible variations in the excentricities of the orbits and in the positions of their perihelia; in order to determine them, let the equation found in N°. 57 be resumed,

$$ede = \frac{andt\sqrt{1-e^2}}{\mu} \cdot \left\{ \frac{dR}{dv} \right\} - \frac{a \cdot (1-e^2)}{\mu} \cdot dR.$$

It follows from what has been stated in N°. 48, that if we assume for the fixed plane, that of the orbit of  $m$  at a given epoch, which permits us to neglect in  $R$ , the inclination  $\phi$  of the orbit of  $m$  on this plane; all the terms of the expression for  $R$  which depend on the angle  $i'n't - int$ , will be comprised in the following form,

$$mk \cdot \cos. (i'n't - int + i'i - i - g\varpi - g'\varpi' - g''\theta');$$

$i, i', g, g', g''$ , being integral numbers, such that we have  $0 = i' - i - g - g' - g''$ . The coefficient  $R$  has for factor  $e^g \cdot e^{g'} \cdot (\tan. \frac{1}{2}\phi')^{g''}$ ,  $g, g', g''$ , being taken positively in these exponents; moreover, if we suppose that  $i'$  and  $i$  are positive, and  $i'$  greater than  $i$ , we have seen in N°. 48, that the terms of  $R$  which depend on the angle  $i'n't - int$  are of the order  $i' - i$ , and or of an order higher by two, by four, &c. unities; if therefore we only consider the terms of the order  $i' - i$ ,  $R$  will be of the form  $e^g \cdot e^{g'} \cdot (\tan. \frac{1}{2}\phi')^{g''} \cdot Q$ ,  $Q$  being a function independent of the excentricities and of the respective inclinations of the orbits. The numbers  $g, g', g''$ , contained under the sign  $\cos.$  are then positive; for if one of them,  $g$ , for example, was negative and equal to  $-f$ ,  $k$  would be of the order  $f + g' + g''$ ; but the equation  $0 = i' - i - g - g' - g''$ , gives  $f + g' + g'' = i' - i + 2f$ ; thus  $k$  would be of an order higher than  $i' - i$ , which is contrary to the hypothesis. This being premised by

N°. 48, we have  $\left\{ \frac{dR}{dv} \right\} = \left\{ \frac{dR}{dt} \right\}$ , provided\* that in this last partial difference we make  $\epsilon - \omega$  equal to a constant quantity, therefore the term of  $\left\{ \frac{dR}{dv} \right\}$  which corresponds to the preceding term of  $R$  is

$$m'.(i+g).k. \sin. (i'n't - int + i'\epsilon' - i\epsilon - g\omega - g'\omega' - g''\theta').$$

The corresponding term of  $dR$  is

$$m'.ink.dt. \sin. (i'n't - int + i'\epsilon' - i\epsilon - g\omega - g'\omega' - g''\theta'),$$

if therefore we only take such terms into account, neglecting the square of  $e$  in comparison to unity, the preceding expression for  $ede$  will give

$$de = \frac{m'.andt}{\mu} \cdot \frac{gk}{e} \cdot \sin. (i'n't - int + i'\epsilon' - i\epsilon - g\omega - g'\omega' - g''\theta').$$

but we have

$$\frac{gk}{e} = ge^{s-1} \cdot e'^{\epsilon'} \cdot (\text{tang. } \frac{1}{2}\phi')^{s''}. Q = \frac{dk}{de};$$

therefore we shall obtain by integrating

$$e = - \frac{m'an}{\mu.(i'n - in)} \cdot \left\{ \frac{dk}{de} \right\} \cos. (i'n't - int + i'\epsilon' - i\epsilon - g\omega - g'\omega' - g''\theta').$$

Now, if the sum of all the terms of  $R$ , which depend on the angles  $i'n't - int$  be represented by the following quantity,

$$m'P. \sin. (i'n't - int + i'\epsilon' - i\epsilon) + m'.P'. \cos. (i'n't - int + i'\epsilon' - i\epsilon);$$

the corresponding part of  $e$  will be

\* Hence  $-\omega = \epsilon - \epsilon$ ,  $\therefore -g\omega = \epsilon - g\epsilon$ , therefore if we substitute this quantity for  $g\omega$ , and then take the value of  $\frac{dR}{dt}$ , we shall obtain the expression for  $\left\{ \frac{dR}{dv} \right\}$ , corresponding to the value of  $R$ .

$$\frac{m'.an}{\mu.(i'n'-in)} \cdot \left\{ \left\{ \frac{dP}{de} \right\} \cdot \sin. \left\{ i'n't - int + i'\epsilon' - i\epsilon \right\} + \left\{ \frac{dP'}{de} \right\} \cdot \cos. \left( i'n't - int + i'\epsilon' - i\epsilon \right) \right\}.$$

This inequality may become extremely sensible if the coefficient  $i'n' - in$  is very small, as is the case in the theory of Jupiter and of Saturn. Indeed, it has for a divisor only the first power  $i'n' - in$ , while the corresponding inequality of the mean motion has for a divisor the second power of this quantity, as has been observed in N°. 65; but  $\left\{ \frac{dP}{de} \right\}$  and  $\left\{ \frac{dP'}{de} \right\}$  being of an order inferior to  $P$  and to  $P'$ , the inequality of the excentricity may be considerable, and even surpass that of mean motion, if the excentricities  $e$  and  $e'$  be very small; we shall see examples of this, in the theory of the satellites of Jupiter.

Let us now determine the corresponding inequality of the motion of the perihelion. For this purpose, let us resume the two equations,

$$ede = \frac{fdf + f'df'}{\mu^2}; \quad e^2 d\omega = \frac{fdf' - f'df}{\mu^2},$$

which were obtained in N°. 67. These equations give

$$df = \mu de \cdot \cos. \omega - \mu e d\omega \cdot \sin. \omega;$$

hence, if we only consider the angle  $i'n't - int + i'\epsilon' - i\epsilon - g\omega - g'\omega' - g''\theta'$ , we shall have

$$df = m'.andt. \left\{ \left\{ \frac{dk}{de} \right\} \cdot \cos. \omega \cdot \sin. (i'n't - int + i'\epsilon' - i\epsilon - g\omega - g'\omega' - g''\theta') - \mu e. d\omega \cdot \sin. \omega. \right\}$$

Let

$$- m'.andt. \left\{ \left\{ \frac{dk}{de} \right\} + k' \right\} \cdot \cos. (i'n't - int + i'\epsilon' - i\epsilon - g\omega - g'\omega' - g''\theta')^*$$

\* By multiplying by  $\sin. \omega$ , we shall have  $df =$

$$- m'.andt. \left( \frac{dk}{de} \right) \cdot (\cos. \omega \cdot \sin. (i'n't - int + i'\epsilon' - i\epsilon - g\omega - g'\omega' - g''\theta') + \sin. \omega \cdot \cos. (i'n't - int +$$

represent the part of  $\mu ed\varpi$ , which depends on the same angle, we shall have

$$df = m'.andt. \left\{ \left\{ \frac{dk}{de} \right\} + \frac{1}{2} k' \right\} \cdot \sin(i'n't - int + i'\epsilon' - i\epsilon - (g-1)\varpi - g'\varpi - g''\theta') \\ - m'. \frac{andt}{2} \cdot k' \cdot \sin(i'n't - int + i'\epsilon' - i\epsilon - (g+1)\varpi - g'\varpi - g''\theta').$$

It is easy to perceive from the last of the expressions of  $df$ , given in N° 67, that the coefficient of this last sine, has for a factor  $e^{g+1} \cdot e'^g \cdot (\tan \frac{1}{2}\phi)^{g''}$ ;  $k'$  is therefore of an order superior by two units, to that of  $\left\{ \frac{dk}{de} \right\}$ ; consequently, if it be neglected in comparison to  $\left\{ \frac{dk}{de} \right\}$ , we shall have

$$- \frac{m'.andt}{\mu} \cdot \left\{ \frac{dk}{de} \right\} \cdot \cos(i'n't - int + i'\epsilon' - i\epsilon - g\varpi - g'\varpi - g''\theta'),$$

for the term of  $ed\varpi$ , which corresponds to the term

$$m'.k. \cos(i'n't - int - int + i'\epsilon' - i\epsilon - g\varpi - g'\varpi - g''\theta'),$$

of the expression of  $R$ . It follows from this, that the part of  $\varpi$  which corresponds to the part of  $R$  expressed by

$$m'.P. \sin(i'n't - int + i'\epsilon' - i\epsilon) + m'.P'. \cos(i'n't - int + i'\epsilon' - i\epsilon),$$

is equal to

$$\frac{m'an}{\mu \cdot (i'n' - in) \cdot e} \cdot \left\{ \left\{ \frac{dP}{de} \right\} \cdot \cos(i'n't - int + i'\epsilon' - i\epsilon) - \left\{ \frac{dP'}{de} \right\} \right. \\ \left. \sin(i'n't - int + i'\epsilon' - i\epsilon) \right\},$$

we shall by this means obtain, in a very simple manner, the variations of

$$i'\epsilon' - i\epsilon - g\varpi - g'\varpi - g''\theta'). \\ - m'andt. k. \sin \varpi \cdot \cos(i'n't - int + i'\epsilon' - i\epsilon - g\varpi - g'\varpi - g''\theta') = \\ - m'andt. \frac{dk}{de} \cdot \sin(i'n't - int - i'\epsilon' - i\epsilon - (g-1)\varpi - g'\varpi - g''\theta'),$$

and the two terms into the value of the coefficient of  $\frac{mandt k}{\mu}$  are obtained from the formula  $\sin a \cdot \cos b = \frac{\sin(a+b) + \sin(a-b)}{2}$ .

the excentricity and of the perihelion, which depend on the angle  $i'n't - int + i'\epsilon' - i\epsilon$ . They are connected with the variation  $\zeta$  of the mean motion, which corresponds to it, in such a manner, that the variation of the excentricity is

$$\frac{1}{3in} \cdot \left\{ \frac{d^2\zeta}{de \cdot dt} \right\}; *$$

and the variation of the longitude of the perihelion is

$$\frac{(i'n' - in)}{3in \cdot e} \cdot \left\{ \frac{d\zeta}{de} \right\}.$$

The corresponding variation of the excentricity of the orbit of  $m'$ , due to the action of  $m$ , will be

$$-\frac{1}{3i'n' \cdot e'} \cdot \left\{ \frac{d^2\zeta}{de' \cdot dt} \right\}.$$

and the variation of the longitude of its perihelion will be

$$-\frac{(i'n' - in)}{3i'n' \cdot e'} \cdot \left\{ \frac{d\zeta}{de'} \right\},$$

And as by N°. 65, we have  $\zeta' = -\frac{m \cdot \sqrt{\bar{a}}}{m' \cdot \sqrt{a'}} \cdot \zeta$  these variations will be

$$\frac{m \cdot \sqrt{\bar{a}}}{3i'n' \cdot m' \cdot \sqrt{a'}} \cdot \left\{ \frac{d^2\zeta}{de' \cdot dt} \right\}, \text{ and } \frac{(i'n' - in) \cdot m \cdot \sqrt{\bar{a}}}{3i'n' \cdot e' \cdot m' \cdot \sqrt{a'}} \cdot \left\{ \frac{d\zeta}{de'} \right\}.$$

When the quantity  $i'n' - in$  is very small, the inequality depending on the angle  $i'n't - int$  produces a sensible one in the expression of the

\*  $\zeta = \frac{3m'an^2i}{(i'n' - in)^2 \mu} \cdot ((P \cdot \cos(i'n't - int + i'\epsilon' - i\epsilon) - P' \cdot \sin(i'n't - int + i'\epsilon' - i\epsilon))$  differencing  $\zeta$ , first with respect to  $e$  and then with respect to  $t$ , the coefficient becomes  $\frac{3m'an^2 \cdot i}{(i'n' - in) \cdot \mu}$ , and the variable part is the same as the variable part of the expression for  $de$ ,

hence the ratio of  $de$  to  $\left( \frac{d^2\zeta}{de \cdot dt} \right)$  is that of 1 to  $3in$ ; in like manner it may be shewn, that the ratio of  $d\pi$  to  $\left( \frac{d^2\zeta}{de} \right)$ , is that of  $\frac{1}{e}$ , to  $\frac{3in}{(i'n' - in)}$ .

mean motion, among the terms depending on the squares of the disturbing masses ; the analysis of them has been given in N°. 65.

This same inequality produces in the expressions of  $de$  and  $d\omega$ , terms of the order of the squares of those masses, which being solely functions of the elements of the orbits, have a sensible influence on the secular variations of these elements. Let us consider for instance, the expression of  $de$  depending on the angle  $i'n't-int$ . By what precedes, we have

$$de = - \frac{m'.an.dt}{\mu} \cdot \left\{ \left\{ \frac{dP}{de} \right\} \cos. (i'n't-int + i'\varepsilon' - i\varepsilon) - \left\{ \left\{ \frac{dP'}{de} \right\} \sin. (i'n't-int + i'\varepsilon' - i\varepsilon) \right\} \right\}.$$

By N°. 65, the mean motion  $nt$  ought to be increased by

$$\frac{3m'.an^2i'}{(i'n'-in)^2.\mu} \cdot \left\{ P. \cos. (i'n't-int + i'\varepsilon' - i\varepsilon - P') \sin. (i'n't-int + i'\varepsilon' - i\varepsilon) \right\},$$

and the mean motion  $n't$  ought to be increased by

$$-\frac{3m'.an^2i'}{(i'n'-in)^2.\mu} \cdot \frac{m.\sqrt{a}}{m'.\sqrt{a'}} \cdot \left\{ P. \cos. (i'n't-int + i'\varepsilon' - i\varepsilon) - P'. \sin. (i'n't-int + i'\varepsilon' - i\varepsilon) \right\}.$$

In consequence of these increments, the value of  $de$  will be increased by the function

$$-\frac{3m'.a^2.in^3.dt}{2\mu^2.\sqrt{a'}.(i'n'-in)^2} \cdot im'.\sqrt{a'} + i'm.\sqrt{a}) \cdot \left\{ P. \left\{ \frac{dP}{de} \right\} - P'. \left\{ \frac{dP}{de} \right\} \right\},$$

and the value of  $d\omega$  will be increased by the function

$$\frac{3m'.a^2.in^3.dt}{2\mu^2.\sqrt{a'}.(i'n'-in)^2.e} \cdot (i.m'.\sqrt{a'} + i'.m.\sqrt{a}) P. \left\{ \frac{dP}{de} \right\} + P'. \left\{ \frac{dP}{de} \right\}.$$

we shall find in like manner, that the value of  $de'$  will be increased by the function

$$-\frac{3ma^2\cdot\sqrt{a}\cdot in^3\cdot dt}{2\mu^2\cdot a'\cdot(i'n'-in)^2}\cdot(im'\cdot\sqrt{a'}+i\cdot m\cdot\sqrt{a})\cdot\left\{P\cdot\left\{\frac{dP'}{de'}\right\}-P'\cdot\left\{\frac{dP}{de'}\right\}\right\};^*$$

and that the value of  $d\varpi'$ , will be increased by the function

$$\frac{3ma^2\cdot\sqrt{a}\cdot in^3\cdot dt}{2\mu^2\cdot a'\cdot(i'n'-in)^2\cdot e'}\cdot(im'\cdot\sqrt{a'}+i\cdot m\cdot\sqrt{a})\cdot\left\{P\cdot\left\{\frac{dP}{de'}\right\}+P'\cdot\left\{\frac{dP'}{de'}\right\}\right\}.$$

These different terms are sensible in the theory of Jupiter and Saturn, and in that of the satellites of Jupiter. The variations of  $e$ ,  $e'$ ,  $\varpi$  and  $\varpi'$ , relative to the angle  $i'n't - int$  may also introduce some constant terms of the order of the squares of the disturbing masses, into the differentials  $de$ ,  $de'$ ,  $d\varpi$ , and  $d\varpi'$ , and depending on the variations of  $e$ ,  $e'$ ,  $\varpi$  and  $\varpi'$  relative to the same angle ; it will be easy by means of the preceding analysis to take them into account. Finally, it will be easy by our analysis to determine the terms of the expressions of  $e$ ,  $\varpi$ ,  $\epsilon'$  and  $\varpi'$ , which depending on the angle  $i'n't - int + i'\epsilon' - ie$  have

\* Let the increment of  $nt = d(nt) = p.(P \cdot \cos. A - P' \cdot \sin. A)$ , and the increment of  $d(n't) = -\frac{m\cdot\sqrt{a}}{m'\cdot\sqrt{a'}}\cdot p.(P \cdot \cos. A - P' \cdot \sin. A)$ , then we have  $d(i'n't - int) = \left(-\frac{i'm\cdot\sqrt{a}}{m'\cdot\sqrt{a'}} - i\right)\cdot p.(P \cdot \cos. A - P' \cdot \sin. A)$ , calling this quantity  $\Delta$ , and substituting it for  $d(i'n't - int)$  in the value of  $de$ , given in this page, we shall have the factor  $\frac{dP}{de} \cdot \cos. (A + \Delta) - \frac{dP'}{de} \cdot \sin. (A + \Delta)$ , then by developing and remarking that  $\sin. \Delta = \Delta$ , and  $\cos. \Delta = 1 q.p.$ , the preceding expression becomes  $\frac{dP}{de} \cdot \cos. A - \frac{dP'}{de}$ .  $\sin. A - \frac{dP}{de} \cdot \sin. A \cdot \left(-\frac{i'm\cdot\sqrt{a}}{m'\cdot\sqrt{a'}} - i\right) \cdot p.(P \cdot \cos. A - P' \cdot \sin. A) + \frac{dP'}{de} \cdot \cos. A \cdot \left(-\frac{i'm\cdot\sqrt{a}}{m'\cdot\sqrt{a'}} - i\right) \cdot p.(P \cdot \cos. A - P' \cdot \sin. A)$ . as  $\sin. A \cdot \cos. A$ , contain only periodic functions, the quantities multiplied by them, or any powers of them, need not be considered at present ; but as  $\sin. {}^2 A = \frac{1}{2} - \frac{1}{2} \cdot \cos. 2A$ ;  $\cos. {}^2 A = \frac{1}{2} + \frac{1}{2} \cdot \cos. 2A$ ; we shall obtain (by substituting for  $\sin. {}^2 A$ ,  $\cos. {}^2 A$ ), two terms which do not involve periodic functions, and which when concinnated, become the quantity by which  $de$  is said in text to be augmented.

not  $i'n'-in$  for a divisor, and those, which depending on the same angle and on double of this angle, are of the order of the square of the disturbing forces. These terms are sufficiently considerable in the theory of Jupiter and of Saturn to induce us to have regard to them; we shall develope them in the requisite detail, when this theory will be more particularly discussed in the 8th Book.

70. In order to determine the variations of the nodes and of the inclinations of the orbits, let the equations of N°. 64, be resumed

$$dc = dt \cdot \left\{ y \cdot \left\{ \frac{dR}{dx} \right\} - x \cdot \left\{ \frac{dR}{dy} \right\} \right\};$$

$$dc' = dt \cdot \left\{ z \cdot \left\{ \frac{dR}{dx} \right\} - x \cdot \left\{ \frac{dR}{dz} \right\} \right\};$$

$$dc'' = dt \cdot \left\{ z \cdot \left\{ \frac{dR}{dy} \right\} - y \cdot \left\{ \frac{dR}{dz} \right\} \right\};$$

If the action of  $m'$  be solely considered, the value of  $R$ , of N°. 46, gives

$$y \cdot \left\{ \frac{dR}{dx} \right\} - x \cdot \left\{ \frac{dR}{dy} \right\} = m' \cdot (x'y - y'x) \cdot \left\{ \frac{1}{(x'^2 + y'^2 + z'^2)^{\frac{3}{2}}} - \frac{1}{((x' - x)^2 + (y' - y)^2 + (z' - z)^2)^{\frac{3}{2}}} \right\}$$

$$z \cdot \left\{ \frac{dR}{dx} \right\} - x \cdot \left\{ \frac{dR}{dz} \right\} = m' \cdot (x'z - z'x) \cdot \left\{ \frac{1}{(x'^2 + y'^2 + z'^2)^{\frac{3}{2}}} - \frac{1}{((x' - x)^2 + (y' - y)^2 + (z' - z)^2)^{\frac{3}{2}}} \right\}$$

$$z \cdot \left\{ \frac{dR}{dy} \right\} - y \cdot \left\{ \frac{dR}{dz} \right\} = m' \cdot (y'z - z'y) \cdot \left\{ \frac{1}{(x'^2 + y'^2 + z'^2)^{\frac{3}{2}}} - \frac{1}{((x' - x)^2 + (y' - y)^2 + (z' - z)^2)^{\frac{3}{2}}} \right\}$$

Let now

$$\frac{c''}{c} = p; \quad \frac{c'}{c} = q;$$

by N°. 64, the two variables  $p$  and  $q$  will determine the tangent of the inclination  $\phi$  of the orbit of  $m$ , and the longitude  $\theta$  of its node, by means of the equations

$$\tan. \phi = \sqrt{p^2 + q^2}; \quad \tan. \phi = \frac{p}{q}.$$

Naming  $p'$ ,  $q'$ ,  $p''$ ,  $q''$ , &c., what  $p$  and  $q$  become relatively to the bodies  $m'$ ,  $m''$ , &c.. we shall have by N°. 64,

$$z = qy - px; \quad z' = q'y' - p'x'; \quad \text{&c.}$$

The preceding value of  $p$  being differenced, gives

$$\frac{dp}{dt} = \frac{1}{c} \cdot \left\{ \frac{dc'' - pdc}{dt} \right\};$$

by substituting in place of  $dc$  and of  $dc''$ , their values, we shall have\*

$$\frac{dp}{dt} = \frac{m'}{c} \cdot ((q - q') \cdot yy' + (p' - p) \cdot x'y) \cdot \left\{ \frac{1}{(x'^2 + y'^2 + z'^2)^{\frac{3}{2}}} - \frac{1}{((x' - x)^2 + (y' - y)^2 + (z' - z)^2)^{\frac{3}{2}}} \right\};$$

\*  $c''x - c'y = -cz$ ,  $\therefore -z = \frac{c''}{c} \cdot x - \frac{c'}{c} \cdot y = px - qy$ ;  $\frac{dp}{dt} = \left( c \cdot \frac{dc''}{dt} - c'' \cdot \frac{dc}{dt} \right) \cdot -\dot{c} =$

$$\frac{1}{c} \cdot \left( \frac{dc''}{dt} - \frac{c'' \cdot dc}{c \cdot dt} \right), \quad \therefore \frac{dp}{dt} = \frac{m'}{c} \cdot ((y'z - z'y) - p(x'y - y'x)) \cdot \left( \frac{1}{(x^2 + y^2 + z^2)^{\frac{3}{2}}} -$$

$\left( \frac{1}{(x' - x)^2 + (y' - y)^2 + (z' - z)^2} - \frac{1}{(x^2 + y^2 + z^2)^{\frac{3}{2}}} \right)$ ; therefore if we substitute for  $z$  and  $z'$  their values, we will obtain by concinnating and obliterating those terms which destroy each other, the expression for  $\frac{dp}{dt}$ , which is given in the text.

in like manner, we shall find

$$\frac{dq}{dt} = \frac{m}{c} \cdot ((p'-p) \cdot xx' + (q-q') \cdot xy') \cdot \left\{ \frac{1}{(x'^2 + y'^2 + z'^2)^{\frac{3}{2}}} \right\}$$

$$- \frac{1}{((x'-x)^2 + (y'-y)^2 + (z'-z)^2)^{\frac{3}{2}}} \cdot$$

If in place of  $x, y, x', y'$ , their values  $r \cdot \cos. v, r \cdot \sin. v, r' \cdot \cos. v', r' \cdot \sin. v'$ , be substituted; we shall have

$$(q-q') \cdot yy' + (p'-p) \cdot x'y = \left\{ \frac{q'-q}{2} \right\} \cdot rr' \cdot ((\cos. v' + v) - \cos. (v' - v)) *$$

$$+ \left\{ \frac{p'-p}{2} \right\} \cdot rr' \cdot (\sin. (v' + v) - \sin. (v' - v));$$

$$(p'-p) \cdot xx' + (q-q') \cdot xy' = \left\{ \frac{p'-p}{2} \right\} \cdot rr' \cdot (\cos. (v' + v) + \cos. (v' - v)).$$

$$+ \left\{ \frac{q'-q}{2} \right\} \cdot rr' \cdot (\sin. (v' + v) + \sin. (v' - v)).$$

The excentricities and inclinations of the orbits being neglected, we have

$$r = a; v = nt + \epsilon; r' = a'; v' = n't + \epsilon';$$

which gives

$$\frac{1}{(x'^2 + y'^2 + z'^2)^{\frac{3}{2}}} - \frac{1}{((x'-x)^2 + (y'-y)^2 + (z'-z)^2)^{\frac{3}{2}}} - \frac{1}{a'^3}$$

$$- \frac{1}{(a'^2 - 2aa' \cdot \cos.(n't - nt + \epsilon' - \epsilon + a^2))^{\frac{3}{2}}}.$$

Moreover, by N°. 48, we have

$$\frac{1}{(a^2 - 2aa' \cdot \cos.(n't - nt + \epsilon' - \epsilon) + a'^2)^{\frac{3}{2}}} = \frac{1}{2} \cdot \Sigma B^{(i)} \cdot \cos. i \cdot (n't - nt + \epsilon' - \epsilon);$$

$$* rr' \cdot \cos. v \cdot \cos. v' = \frac{rr'}{2} \cos. (v + v') + \cos. (v - v), rr' \cdot \sin. v \cdot \cos. v' = \frac{rr'}{2} \cdot (\sin. (v + v') + \sin. (v - v)).$$

the integral sign  $\Sigma$  extending to all entire values of  $i$ , positive as well as negative, the value  $i = 0$  being included; by this means, we shall have, neglecting the terms of the order of the squares and products of the excentricities and inclinations of the orbits,

$$\begin{aligned} \frac{dp}{dt} &= \frac{(q'-q)}{2c} \cdot \frac{m'a}{a'^2} (\cos.(n't+nt+\epsilon'+\epsilon) - \cos.(n't-nt+\epsilon'-\epsilon)) \\ &+ \frac{(p'-p)}{2c} \cdot \frac{m'a}{a'^2} (\sin.(n't+nt+\epsilon'+\epsilon) - \sin.(n't-nt+\epsilon'-\epsilon))^* \\ &+ \frac{(q'-q)}{4c} \cdot m.a.a'.\Sigma B^{(i)}.(\cos.[(i+1).(n't-nt+\epsilon'-\epsilon)] - \cos.(i+1). \\ &\quad (n't-nt+\epsilon'-\epsilon) + 2nt+2\epsilon]) \\ &+ \frac{(p'-p)}{4c} \cdot m.a.a'.\Sigma B^{(i)}.(\sin.[(i+1).(n't-nt+\epsilon'-\epsilon)] - \sin.(i+1). \\ &\quad (n't-nt+\epsilon'-\epsilon) + 2nt+2\epsilon]); \\ \frac{dq}{dt} &= \frac{(p'-p)}{2c} \cdot \frac{m'a}{a'^2} (\cos.(n't+nt+\epsilon'+\epsilon) + \cos.(n't-nt+\epsilon'-\epsilon)) \\ &+ \frac{(q'-q)}{2c} \cdot \frac{m'a}{a'^2} \sin.(n't+nt+\epsilon'+\epsilon) + \sin.(n't-nt+\epsilon'-\epsilon)) \\ &+ \frac{(p-p')}{4c} \cdot m.a'a.\Sigma B^{(i)}.(\cos.[(i+1).(n't-nt+\epsilon'-\epsilon)] + \cos.(i+1). \\ &\quad (n't-nt+\epsilon'-\epsilon) + 2nt+2\epsilon]) \end{aligned}$$

\* The value of the third term in the expression for  $\frac{dp}{dt}$  will be had by observing that

$$\frac{\cos.(\nu \pm v)}{2} \cdot \frac{1}{2} \cdot \Sigma B^{(i)} \cos.i.(n't-nt+\epsilon'-\epsilon) = \frac{\Sigma B^{(i)}}{4} (\cos.i.(n't-nt+\epsilon'-\epsilon) + n't+nt+\epsilon'+\epsilon) + \cos.i.(n't-nt+\epsilon'-\epsilon) - n't-nt-\epsilon'-\epsilon); \text{ therefore, if we concinnate the terms of this expression, we shall obtain by observing that } \cos.i.(n't-nt+\epsilon'-\epsilon) + n't+nt+\epsilon'+\epsilon) = \cos.(i+1)n't-nt+\epsilon'-\epsilon) + 2nt+2\epsilon), \text{ and also that } \cos.i.(n't-nt+\epsilon'-\epsilon) + n't-nt+\epsilon'-\epsilon) = \cos.(i+1).(n't-nt+\epsilon'-\epsilon), \text{ the expression given in the text.}$$

$$+ \frac{(q'-q)}{4c} \cdot m' \cdot aa' \cdot \Sigma B^{(1)} \cdot (\sin[(i+1) \cdot (n't - nt + \varepsilon' - \varepsilon)] + \sin(i+1) \cdot (n't - nt + \varepsilon' - \varepsilon) + 2nt + 2\varepsilon]).$$

The value  $i = -1$  gives in the expression of  $\frac{dp}{dt}$ , the constant quantity  $\frac{(q'-q)}{4c} \cdot m' \cdot aa' \cdot B^{(-1)}$ , all the other terms of the expression of  $\frac{dp}{dt}$  are periodic, if  $P$  represents their sum, we shall have by N°. 48,

$$\frac{dp}{dt} = \frac{q'-q}{4c} \cdot m' \cdot aa' B^{(1)} + P,$$

( $B^{(1)}$  being equal to  $B^{(-1)}$ ).

By the same method, we shall find, that if we denote by  $Q$  the sum of all the periodic terms of the expression of  $\frac{dq}{dt}$ , we shall have

$$\frac{dq}{dt} = \frac{(p-p')}{4c} \cdot m' \cdot aa' \cdot B^{(1)} + Q.$$

If the squares of the excentricities and of the inclinations of the orbits, be neglected, we shall have by N°. 64,  $c = \sqrt{\mu a}$ . If then  $\mu$  be supposed = 1, we have  $n^2 a^3 = 1$ , which gives  $c = \frac{1}{an}$ ; the quantity  $\frac{m' \cdot aa' \cdot B^{(1)}}{4c}$ , thus becomes  $\frac{m' \cdot a^2 a' \cdot n B^{(1)}}{4}$ , which by N°. 59 is equal to (0, 1); hence we shall have

$$\frac{dp}{dt} = (0, 1) \cdot (q'-q) + P;$$

$$\frac{dq}{dt} = (0, 1) \cdot (p-p') + Q.$$

It follows from this, that if ( $P$ ) and ( $Q$ ) denote the sum of all the functions  $P$  and  $Q$ , relative to the action of the different bodies

$m'$ ,  $m''$ , &c., on  $m$ , and if in like manner  $(P')$ ,  $(Q')$ ,  $(P'')$ ,  $(Q'')$ , &c., denote what  $(P)$  and  $(Q)$  become, when the quantities relative to  $m$  are changed into those which refer to  $m'$ ,  $m''$ , &c., and conversely; we shall have for the determination of the variables  $p$ ,  $q$ ,  $p'$ ,  $q'$ ,  $p''$ ,  $q''$ , &c., the following system of differential equations,

$$\frac{dp}{dt} = -((0, 1) + (0, 2) + \&c.)q + (0, 1).q' + (0, 2).q'' + \&c. + (P);$$

$$\frac{dq}{dt} = ((0, 1) + (0, 2) + \&c.).p - (0, 1).p' - (0, 2).p'' - \&c. + (Q);$$

$$\frac{dp'}{dt} = -((1, 0) + (1, 2) + \&c.).q' + (1, 0).q + (1, 2).q'' + \&c. + (P')$$

$$\frac{dq'}{dt} = ((1, 0) + (1, 2) + \&c.).p' - (1, 0).p - (1, 2).p'' - \&c. + (Q'),$$

&c.

From the analysis of N°. 68, it appears that the periodic parts of  $p$ ,  $q$ ,  $p'$ ,  $q'$ , &c., are

$$p = f(P). dt; \quad q = f(Q). dt$$

$$p' = f(P'). dt; \quad q' = f(Q'). dt,$$

we shall afterwards obtain the secular parts of the same quantities, by integrating the preceding differential equations, deprived of their last terms  $(P)$ ,  $(Q)$ ,  $(P')$ , &c.; and then we shall light on the equations  $(C)$  of N°. 59, which have been already discussed with sufficient detail to dispense with our reverting to this object.

71. Let the equations of N°. 64 be resumed, namely,

$$\tan. \phi = \frac{\sqrt{c'^2 + c''^2}}{c}; \quad \tan. \theta = \frac{c''}{c};$$

from them may be deduced

$$\frac{c'}{c} = \tan. \phi. \cos. \theta; \quad \frac{c''}{c} = \tan. \phi. \sin. \theta;$$

differentiating, we shall have

$$d. \tan. \phi = \frac{1}{c} \cdot (dc' \cdot \cos. \theta + dc'' \cdot \sin. \theta - dc \cdot \tan. \phi);^*$$

$$d\theta. \tan. \phi = \frac{1}{c} \cdot (dc'' \cdot \cos. \theta - dc' \cdot \sin. \theta).$$

If in these equations, we substitute in place of  $\frac{dc}{dt}$ ,  $\frac{dc'}{dt}$ ,  $\frac{dc''}{dt}$ , their values  $y \cdot \left\{ \frac{dR}{dx} \right\} - x \cdot \left\{ \frac{dR}{dy} \right\}$ ,  $z \cdot \left\{ \frac{dR}{dx} \right\} - x \cdot \left\{ \frac{dR}{dz} \right\}$ ,  $z \cdot \left\{ \frac{dR}{dy} \right\} - y \cdot \left\{ \frac{dR}{dz} \right\}$ , and in place of these last quantities, their values furnished in N°. 67, and if moreover, we observe that  $s = \tan. \phi \cdot \sin. (v-\theta)$ ; we shall have

$$\begin{aligned} \dagger d. \tan. \phi &= \frac{dt. \tan. \phi. \cos. (v-\theta)}{c} \cdot \left\{ r \cdot \left\{ \frac{dR}{dr} \right\} \cdot \sin. (v-\theta) + \left\{ \frac{dR}{dv} \right\} \cdot \cos. (v-\theta) \right\} \\ &\quad - \frac{(1+s^2).dt}{c} \cdot \cos. (v-\theta) \cdot \left\{ \frac{dR}{ds} \right\}; \end{aligned}$$

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$$* \text{ Hence } \cos. \theta = \frac{c'}{\sqrt{c'^2+c''^2}}, \sin. \theta = \frac{c''}{\sqrt{c'^2+c''^2}}, \therefore d. \tan. \phi = \frac{dc' \cdot c' + dc'' \cdot c''}{c \cdot \sqrt{c'^2+c''^2}} - dc \cdot \frac{\sqrt{c'^2+c''^2}}{c^2}$$

= by substituting  $\cos. \theta$ ,  $\sin. \theta$ , for their respective values, the expression which is given in the text.

$$d. \tan. \theta = \frac{d\theta}{\cos.^2 \theta} = \frac{dc'' \cdot c' - dc' \cdot c''}{c^2}, \therefore \frac{d\theta \cdot \sqrt{c'^2+c''^2}}{c} = \frac{dc''}{c} \cdot \left( \frac{c'}{\sqrt{c'^2+c''^2}} - \frac{dc' \cdot c''}{c \cdot \sqrt{c'^2+c''^2}} \right).$$

† Multiplying the value of  $dc'$ , given in page 385, by  $\cos. \theta$ , and that of  $dc''$  by  $\sin. \theta$ , we shall obtain by adding them together  $(1+s^2) \cdot \cos. (v-\theta) \cdot \left( \frac{dR}{ds} \right) - rs \cdot \cos. (v-\theta) \cdot \left( \frac{dR}{dr} \right) + s \cdot \sin. (v-\theta) \cdot \left( \frac{dR}{dv} \right) - \frac{dR}{dv} \cdot \tan. \phi = d \tan. \phi$ , hence by substituting for  $s$  its value, we shall obtain the expression given in the text.

$$d\theta \cdot \tan. \phi = \frac{dt \cdot \tan. \phi \cdot \sin. (v-\theta)}{c} \cdot \left\{ r \cdot \left\{ \frac{dR}{dr} \right\} \cdot \sin. (v-\theta) + \left\{ \frac{dR}{dv} \right\} \cdot \cos. (v-\theta) \right\}$$

$$- \frac{(1+s^2) \cdot dt}{c} \cdot \sin. (v-\theta) \cdot \left\{ \frac{dR}{ds} \right\}.$$

These two differential equations will determine directly the inclination of the orbit and the motion of the nodes. They give

$$\sin. (v-\theta) \cdot d. \tan. \phi - d\theta \cdot \cos. (v-\theta) \cdot \tan. \phi = 0; *$$

this equation may be also deduced from the equation  $s = \tan. \phi \cdot \sin. (v-\theta)$ ; in fact, as this last equation is finite, we may by N°. 63, difference it, either by considering  $\phi$  and  $\theta$  as constant, or by treating them as if they were variable; so that its differential, taken by making  $\phi$  and  $\theta$  the sole variables, vanishes; hence results the preceding differential equation.

Suppose now, that the inclination of the fixed plane to that of the orbit should be extremely small; so that the square of  $s$  and of the  $\tan. \phi$ , may be neglected, we shall have

$$d. \tan. \phi = - \frac{dt}{c} \cdot \cos. (v-\theta) \cdot \left\{ \frac{dR}{ds} \right\};$$

$$d\theta \cdot \tan. \phi = - \frac{dt}{c} \cdot \sin. (v-\theta) \cdot \left\{ \frac{dR}{ds} \right\};$$

by making, as before,

$$p = \tan. \phi \cdot \sin. \theta; \quad q = \tan. \phi \cdot \cos. \theta;$$

we shall have, in place of the two preceding differential equations, the following

$$dq = - \frac{dt}{c} \cdot \cos. v \cdot \left\{ \frac{dR}{ds} \right\};$$

\* By multiplying the first equation by  $\sin. (v-\theta)$ , and the second by  $\cos. (v-\theta)$ , their second members become identically equal to each other, therefore the first members will be also equal to each other.

$$dp = -\frac{dt}{c} \cdot \sin. v \cdot \left\{ \frac{dR}{ds} \right\};$$

but  $s = q \cdot \sin. v - p \cdot \cos. v$ , hence

$$\left\{ \frac{dR}{ds} \right\} = \frac{1}{\sin. v} \cdot \left\{ \frac{dR}{dq} \right\}, \quad \left\{ \frac{dR}{ds} \right\} = -\frac{1}{\cos. v} \cdot \left\{ \frac{dR}{dp} \right\}; \dagger$$

therefore

$$dq = \frac{dt}{c} \cdot \left\{ \frac{dR}{dp} \right\};$$

$$dp = -\frac{dt}{c} \cdot \left\{ \frac{dR}{dq} \right\}.$$

We have seen in N°. 48, that the function  $R$  is independent of the fixed plane of  $x$  and of  $y$ ; supposing, therefore, that all the angles of this function are referred to the orbit of  $m$ , it is evident that  $R$  will be a function of these angles, and of the respective inclination of these two orbits, which inclination we shall denote by  $\phi'$ . Let  $\theta'$  be the longitude of the node of the orbit of  $m'$  on the orbit of  $m$ , and let us suppose that  $m'k \tan. (\phi')^g \cdot \cos. (i'n't - int + A - g\theta')$  is a term in the value of  $R$ , depending on the angle  $i'n't - int$ : by N°. 60 we shall have

$$\tan. \phi' \cdot \sin. \theta' = p' - p; \quad \tan. \phi' \cdot \cos. \theta' = q' - q,$$

hence we deduce

$$(\tan. \phi')^g \cdot \sin. g\theta' = \frac{((q' - q) + (p' - p) \cdot \sqrt{-1})^g - ((q' - q) - (p' - p) \cdot \sqrt{-1})^g}{2 \cdot \sqrt{-1}};$$

$$* dq = \cos. \theta. d. \tan. \phi - d\theta. \sin. \theta. \tan. \phi = -\frac{dt}{c} \cdot (\cos. v. \cos. \theta + \sin. v. \sin. \theta). \cos. \theta. \left( \frac{dR}{ds} \right) + \frac{dt}{c} \cdot (\sin. v. \cos. \theta - \cos. v. \sin. \theta). \sin. \theta. \left( \frac{dR}{ds} \right) = -\frac{dt}{c} \cdot \cos. v. \left( \frac{dR}{ds} \right); \quad \left( \frac{dR}{ds} \right) = \left( \frac{dR}{dq} \right) \cdot \left( \frac{dq}{ds} \right)$$

but  $\frac{dq}{ds} = \frac{1}{\sin. v}$ ,  $\because \left( \frac{dR}{ds} \right) = \frac{dR}{dq} \cdot \frac{1}{\sin. v}$ .

$$\tan.(\varphi')^g \cos. g\theta' = \frac{((q'-q)+(p'-p)\cdot\sqrt{-1})^g + ((q'-q)-(p'-p)\cdot\sqrt{-1})^g}{2}; *$$

If we only consider the preceding term of the value of  $R$ , we shall have

$$\left\{ \begin{array}{l} \left\{ \frac{dR}{dp} \right\} = -g \cdot (\tan. \varphi')^{g-1} \cdot m'k \cdot \sin. (i'n't - int + A - (g-1)\theta') / ; \\ \left\{ \frac{dR}{dq} \right\} = -g \cdot (\tan. \varphi')^{g-1} \cdot m'k \cdot \cos. (i'n't - int + A - (g-1)\theta') . \end{array} \right.$$

If these values be substituted in the preceding expressions of  $dp$  and of  $dq$ , and if we observe that we have very nearly,  $c = \frac{\mu}{an}$ ; we shall obtain

$$p = \frac{g \cdot m'k \cdot an}{\mu \cdot (i'n' - in)} \cdot (\tan. \varphi')^{g-1} \cdot \sin. (i'n't - int + A - (g-1)\theta') ;$$

$$q = \frac{g \cdot m'k \cdot an}{\mu \cdot (i'n' - in)} \cdot (\tan. \varphi')^{g-1} \cdot \cos. (i'n't - int + A - (g-1)\theta') .$$

and if these values be substituted in the equation  $s = q \cdot \sin. v - p \cdot \cos. v$ , we shall have

$$s = - \frac{g \cdot m'k \cdot an}{\mu \cdot (i'n' - in)} \cdot (\tan. \varphi')^{g-1} \cdot \sin. (i'n't - int + A - (g-1)\theta') .$$

$$* \sin. \theta' = \frac{p' - p}{\sqrt{(p' - p)^2 + (p' - q)^2}}, \cos. \theta' = \frac{q' - q}{\sqrt{(p' - p)^2 + (q' - q)^2}}, \therefore$$

$$\cos. \theta' + \sqrt{-1} \cdot \sin. \theta' \cdot \sqrt{g} = \cos. g\theta' + \sqrt{-1} \cdot \sin. g\theta' = \frac{((q' - q) + \sqrt{-1} \cdot (p' - p))^g}{\sqrt{(p' - p)^2 + (q' - q)^2}}, \text{ hence}$$

we obtain by multiplying by  $\tan. \varphi'^g$ , and its values, the expressions for  $\tan. \varphi'^g \cdot \sin. g\theta'$ ,  $\tan. \varphi'^g \cdot \cos. g\theta'$ , which are given in the text; now  $m'k \cdot (\tan. \varphi')^g \cdot \cos. (i'n't - int + A - g\theta') = m'k \cdot (\tan. \varphi')^g \cdot (\cos. g\theta' \cdot \cos. (i'n't - int + A) + \sin. g\theta' \cdot \sin. (i'n't - int + A))$ , if we substitute for  $(\tan. \varphi')^g \cdot \cos. g\theta'$ ,  $(\tan. \varphi')^g \cdot \sin. g\theta'$ , and their difference with respect to  $p$  and  $q$ , we will obtain the expressions for  $\frac{dR}{dp}$ ,  $\frac{dR}{dq}$ , which are given in the text.

This expression of  $s$ , is the variation of the latitude corresponding to the preceding term of  $R$ , it is evident that it is the same whatever may be the fixed plane to which the motions of  $m$  and of  $m'$  may be referred, provided that its inclination to the plane of the orbits be inconsiderable; therefore we shall by this means obtain that part of the expression for the latitude, which becomes sensible from the smallness of the divisor  $i'n-in$ . Indeed this inequality of the latitude involves only the first power of this divisor, and in this respect it is less sensible than the corresponding inequality of the mean longitude, which contains the square of this divisor; but on the other hand,  $\tan. \phi'$  occurs affected with a power which is less by unity; which remark corresponds to that made in 69, on the corresponding inequality of the excentricities of the orbits. It thus appears that all these inequalities are connected with each other, and the corresponding part of  $R$ , by very simple expressions.

If the preceding expressions of  $p$  and of  $q$  be differenced, and if in the value of  $\frac{dp}{dt}$  and of  $\frac{dq}{dt}$ , which result, the angles  $nt$  and  $n't$  be increased by the inequalities of the mean motions, depending on the angle  $i'n't-int$ ; there will result in these differentials; quantities which are solely functions of the elements of the orbits, and which may sensibly influence the secular variations of the inclinations and of the nodes, although being of the order of the squares of the disturbing masses; which is analogous to what has been stated in N°. 69, relative to the secular variations of the excentricities and aphelia.

72. It remains for us to consider the variation of the longitude  $\epsilon$  of the epoch. By N°. 64, we have

$$d\epsilon = de \cdot \left\{ \left\{ \frac{dE^{(1)}}{de} \right\} \right\} \cdot \sin. (v-\omega) + \frac{1}{2} \cdot \left\{ \left\{ \frac{dE^{(2)}}{de} \right\} \right\} \cdot \sin. 2.(v-\omega) + \&c.$$

$$- d\omega \cdot ((E^{(1)} \cdot \cos. (v-\omega) + E^{(2)} \cdot \cos. 2.(v-\omega) + \&c.);$$

If for  $E^{(1)}$ ,  $E^{(2)}$ , &c. be substituted their values in series arranged ac-

cording to the powers of  $e$ , which series it is easy to infer from the general expression for  $E^{(i)}$ , given in N°. 16, we shall have

$$\begin{aligned} d\epsilon = & -2de \cdot \sin(v-\omega) + 2e \cdot d\omega \cdot \cos(v-\omega) \\ & + ede \cdot (\frac{5}{2} + \frac{1}{2}e^2 + \&c.) \cdot \sin 2(v-\omega) - e^2 \cdot d\omega \cdot (\frac{5}{2} + \frac{1}{4}e^2 + \&c.) \cos 2(v-\omega)^* \\ & - e^2 de \cdot (1 + \&c.) \cdot \sin 3(v-\omega) + e^3 \cdot d\omega \cdot (1 + \&c.) \cdot \cos 3(v-\omega) \\ & + \&c. \end{aligned}$$

If for  $de$ , and  $ed\omega$ , their values, given in N°. 67, be substituted, we shall find, when the approximation is carried as far as quantities of the order  $e^2$  inclusively,

$$\begin{aligned} d\epsilon = & \frac{a^2 \cdot ndt}{\mu} \cdot \sqrt{1-e^2} \cdot (2 - \frac{5}{2}e \cdot \cos(v-\omega) + e^2 \cdot \cos 2(v-\omega)) \cdot \left\{ \frac{dR}{dr} \right\} \\ & - \frac{andt}{\mu \cdot \sqrt{1-e^2}} \cdot e \cdot \sin(v-\omega) \cdot (1 + \frac{e}{2} \cdot \cos(v-\omega)) \cdot \left\{ \frac{dR}{dv} \right\}. \end{aligned}$$

The general expression for  $d\epsilon$  contains terms of the form  $m' \cdot k \cdot ndt \cdot \cos(i'n't - int + A)$ , and consequently the expression for  $\epsilon$  contains terms of the form  $\frac{m' \cdot kn}{i'n' - in} \cdot \sin(i'n't - int + A)$ ; but it is easy to be assured that the coefficient  $k$  in these terms is of the order  $i' - i$ , and that consequently, these terms are of the same order as those of the mean longitude which depend on the same angle. The latter have for divisor the square of  $i'n' - in$ ; we have seen that we can neglect in respect to them, the corresponding terms of  $\epsilon$  when  $i'n' - in$  is a very small quantity.

If in the terms of the expressions of  $d\epsilon$ , which are solely functions

\* By N°. 16,  $E^{(i)} = \pm \frac{2e^i \cdot (1 + e \cdot \sqrt{1-e^2})}{(1 + \sqrt{1-e^2})^i}$ ,  $\therefore \log E^{(i)} = 2i \cdot \log e + \log(1 + e \cdot \sqrt{1-e^2}) - i \cdot \log(1 + \sqrt{1-e^2})$ ,

then by differentiating and substituting for  $E^{(i)}$  its value, we can obtain the expression which is given in the text.

of the elements of the orbits, we substitute in place of these elements the secular parts of their values, it is evident that there will result in them constant terms, and other terms affected with the sines and cosines of the angles on which the secular variations of the excentricities and of the inclinations of the orbits depend. The constant terms will produce in the expression of  $\epsilon$ , terms proportional to the time, which will be confounded with the mean motion of  $m$ . As to the terms affected by the sine and cosine, they will acquire by integration, in the expression of  $\epsilon$ , very small divisors of the same order as the disturbing forces; so that these terms being at once multiplied and divided by these disturbing forces, they may become sensible, although of the order of the squares and products of the excentricities and inclinations. We shall see in the theory of the planets that these terms are insensible, but they are extremely sensible in the theory of the moon and of the satellites of Jupiter, indeed it is on these terms that their secular equations depend.

We have seen in N°. 65, that the mean motion of  $m$  has for expression  $\frac{3}{\mu} \int \int a dt. dR$ , and that if we only consider the first power of the disturbing masses,  $dR$  involves only periodic quantities; but if we take into account the squares and products of these masses, this differential may contain terms which are solely functions of the elements of the orbits. By substituting in place of these elements the secular parts of their values, there will result terms affected with the sines and cosines of angles, on which the secular variations of the orbits depend. These terms will acquire in the expression of the mean motion, by the double integration, very small divisors, which will be of the order of the squares and products of the disturbing masses; so that being simultaneously multiplied and divided by the squares and products of these masses, they may become sensible, although being of the order of the squares and

products of the excentricities and inclinations of the orbits. We shall see that these terms are likewise insensible in the theory of the planets.

73. The elements of the orbit of  $m$ , being determined by what precedes, they should be substituted in the expressions for the radius vector, for the longitude, and latitude, which have been given in N°. 22; the values of these three variables will thus be obtained by means of which astronomers determine the position of the heavenly bodies. By reducing these values into a series of sines and cosines, we shall obtain a series of inequalities, from which tables may be formed, and thus the position of  $m$  at any instant may be computed with great facility. This method, founded on the variation of parameters, is extremely useful in the investigation of those inequalities, which from their relations with the mean motions of the bodies of the system, acquire great divisors, and by this means become very sensible. This species of inequalities affects principally the elliptic elements of the orbits; therefore by determining the variations which result from them in these elements, and by substituting them in the expressions of elliptic motion, we shall obtain in the simplest manner possible, all the inequalities which those divisors render sensible.

The preceding method is likewise useful in the theory of comets; these stars are only perceived for a very small part of their course, and observations solely make known the part of the ellipse, which may be confounded with the arc of the orbit which they describe during their apparition. Therefore, if the nature of the orbit, considered as a variable ellipse, be determined, we shall have the changes which this ellipse undergoes in the interval between two consecutive appearances of the same comet; we can therefore, announce its return, and when it reappears, compare the theory with observations.

After having given the methods and formulæ for determining by successive approximations, the motions of the centres of gravity of

the heavenly bodies, it remains for us to apply them to the different bodies of the solar system; but as the ellipticity of these bodies influences in a sensible manner, the motions of several of them among each other, it is requisite, previously to proceeding to the numerical applications, to treat of the figure of the heavenly bodies, of which the investigation is equally interesting, on its own account, as that of their motions.

END OF THE SECOND BOOK.









