
MatRTKLIB Manual

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1 Introduction

MatRTKLIB provides a MATLAB wrapper for RTKLIB, an open source GNSS data processing library, and also provides various processes required for actual GNSS analysis and research in its own MATLAB classes.

This porting facilitates the use of RTKLIB from within the MATLAB environment. In addition to the porting, MatRTKLIB offers editing functions for GNSS observations and visualization methods, which are frequently necessary in the processing of GNSS analysis. MATLAB, a programming language, has been particularly well-suited to educational and research purposes, particularly for novices, due to its ease of use for vector and matrix calculations, as well as its extensive visualization functions for calculation results. However, there has been no complete porting of RTKLIB to MATLAB. The arrival of MatRTKLIB is expected to greatly reduce the cost of entry for novices in GNSS studies and development. The features of the developed MatRTKLIB are as follows.

1. **MatRTKLIB** uses the RTKLIB source code as a submodule and compiles each RTKLIB function as an MEX function in C to provide calls in MATLAB, improving processing speed and allowing for immediate reflection of any feature additions to RTKLIB (e.g., support for new satellites/signals and support for new RINEX/RTCM versions).
2. Single-input/output RTKLIB functions were adapted to the vector input/output to suit MATLAB's unique vector/matrix processing; this allows for batch processing of the GNSS data analysis without iterative processing, improving execution speed and code readability.
3. Providing a unique MATLAB class called **GT**, which not only ports RTKLIB but also provides useful tools related to GNSS data processing, such as allowing easy editing of GNSS observation data, and computing linear combinations commonly used in GNSS data processing. GT also provides methods for visualizing various GNSS-related data using the visualization capabilities of MATLAB.
4. The source code for various concrete examples of GNSS data processing is provided. Many sample implementations will facilitate understanding of GNSS data processing, such as step-by-step implementations of linear combination generation, residual evaluation, single-point positioning, and PPK using the double differences of the GNSS carrier phases.

2 GNSS Observation Model

As MatRTKLIB is based on RTKLIB, the GNSS observation model conforms to the observation model adopted in RTKLIB manual. However, GT provides methods for calculating GNSS pseudorange and carrier phase residuals for positioning calculations and GNSS analysis. In this section, the pseudorange and carrier phase residual models provided by MatRTKLIB are explained.

Let ρ denote the GNSS pseudorange observation, and the residual model v_ρ is expressed by the following equation.

where r_r^s is the geometric distance between the antenna and the satellite, corrected for the Sagnac effect, and is calculated using `rtklib.geodist()`. The satellite position can be calculated using `rtklib.satposs()`, using broadcast ephemeris or precise ephemeris. c is the speed of light, and t_r is the receiver clock error. I is the ionosphere delay, and can be calculated using the Klobuchar model, etc., using `rtklib.ionomodel()`. T is the tropospheric delay, and it can be calculated using the Saastamoinen model with `rtklib.tropmodel()`. In GT, it is possible to calculate the pseudorange residuals shown in equation (1) all at once using the method for calculating the pseudorange observation residuals (`gobs.residuals()`).

Let ϕ denote the GNSS carrier phase observation, and the residual model v_ϕ is expressed by the following equation.

where, λ is the wavelength of the carrier wave, B is the bias of the carrier phase including the initial phase, and $\delta\phi$ is a correction term that includes the antenna phase center offset, the earth tide, and the phase wind-up effect. The carrier phase and pseudorange residuals can be used to estimate the position of the receiver antenna and for various other analyses.

Furthermore, GT provides `gobs.singleDifference()`, which calculates the difference between GNSS observations between reference stations, and `gobs.doubleDifference()`, which calculates the difference between observations between satellites. Using MatRTKLIB, it is possible to analyze these GNSS observation residuals and easily construct positioning methods, and it is expected to be used for various educational and research purposes.

3 Requirement

MatRTKLIB was tested and compiled on the following

- MATLAB 2024a
- OS: Windows 11, 64bit, Compiler: Microsoft Visual Studio 2022
- OS: Ubuntu 20.04, 64bit, Compiler: GCC

MatRTKLIB supports MATLAB 2023a and higher and does not require a special toolbox. The recompiled MEX files are provided, they do not need to be recompiled, except in special cases.

4 Installation

If you do not want to compile **MatRTKLIB** yourself, you can download a pre-compiled package:

```
1 git clone https://github.com/taroz/MatRTKLIB.git
```

To install **MatRTKLIB**, simply add its folder path to your MATLAB path list in MATLAB comand window:

```
1 addpath('/path/to/MatRTKLIB');
```

5 Compile

For 32-bit systems, you will need to recompile **MatRTKLIB** yourself. When compiling, clone including submodules.

```
1 git clone --recursive https://github.com/taroz/MatRTKLIB.git
```

Or, if you have already cloned

```
1 git submodule update --init --recursive
```

The compilation procedure is as follows. 1. In MATLAB, enter `mex -setup` to see if compiler is configured 2. Run `compile.m`

Note: If you are syncing directories via OneDrive or Dropbox, the compilation may fail. If this happens, please pause the synchronization.

6 Usage

6.1 Call RTKLIB functions

Use `rtklib.***` to call RTKLIB wrapper functions in MATLAB. Vector I/O is supported for many functions; see Appendix 1 for RTKLIB function support status and vector I/O support. You can also use the `doc rtklib.***'orhelp rtklib.***'` commands to get help on each function.

6.2 GT

To create a GT object in MATLAB, use `gt.***`. See Appendix 2 for more information on GT class types and methods. You can also use the `doc gt.***'orhelp gt.***'` commands to get help on each function.

6.3 Run examples

MatRTKLIB contains over 30 examples, all of which exist as MATLAB script files in the `/examples` folder. You can learn how to use the GT class by running the examples. See Appendix 3 for a description of the types and contents of the examples.

The examples folder contains a GNSS observation dataset for testing. Thus, all examples could be run without additional datasets.

The following are three example problems and their results.

6.3.1 Example 1: Visualization of observation data

This example is provided as `Example1_visualization_RINEX_observation.m` in the examples folder. The specific source code for visualizing RINEX observations is shown in Fig. 1-1.

```
1| gobs = gt.Gobs("rover.obs"); % Read RINEX observation
2| gobs.plot           % Plot observation
3| gobs.plotNSat       % Plot number of observation
4| gnav = gt.Gnav("rover.nav"); % Read RINEX navigation
5| gobs.plotSky(gnav);  % Plot satellite
```

Figure 1: Source code of example 1: Reading and visualizing RINEX observations.

As shown in Fig. 1, L1 creates a Gobs object by passing the RINEX observation file directly to the constructor of the Gobs class. L2 plots the satellite signals contained in the observation data, and the figure displayed in MATLAB is shown in Fig. 2. Similarly, L3 visualizes the number of satellite systems in the observation data (Fig. 3); L4 reads the RINEX navigation file and generates a Gnav object. L5 visualizes satellite constellations in the observation data by calculating the satellite positions using the receiver's approximate position in the RINEX observation file and navigation data (Fig. 4). Thus, MatRTKLIB can be used to concisely visualize signals in GNSS observations, changes in the number of satellites, and satellite constellations.

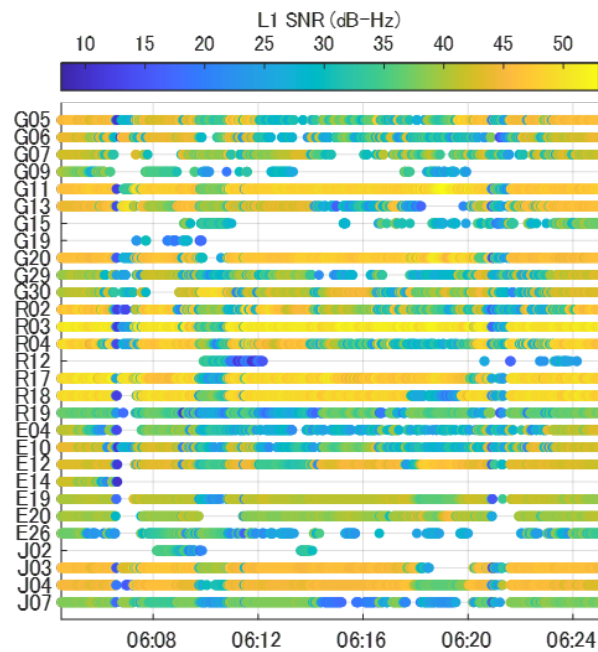


Figure 2: Visualizing the availability of the GNSS observations contained in RINEX files. The plot represents the epoch at which the observations were obtained, and the color of the plot represents the signal strength.

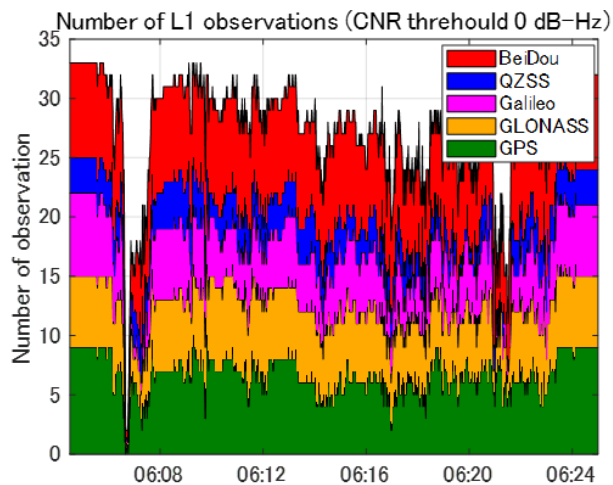


Figure 3: Visualize the change in the number of GNSS observations in the RINEX file. The colors represent the satellite system per GNSS.

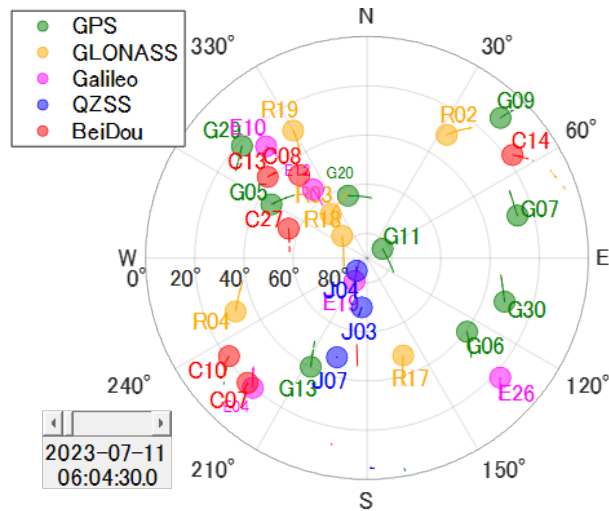


Figure 4: Visualize the GNSS constellations contained in the RINEX file. The color of the plot represents each satellite system, and the slider can be used to change the time.

6.3.2 Example 2: Post-processing kinematic (PPK) analysis

This example is provided as `Example2_PPK_analysis.m` in the examples folder. The source code for PPK positioning using GNSS base station observations is shown in Fig. 5.

```

1| gobsr = gt.Gobs("rover.obs"); % Rover RINEX observation
2| gobsb = gt.Gobs("base.obs"); % Base RINEX observation
3| gnav = gt.Gnav("rover.nav"); % RINEX navigation
4| gopt = gt.Gopt("rtk.conf"); % Process options
5| grtk = gt.Grtk(gopt); % RTK object
6| [grtk, gsol] = ... % Call rtkpos function
7|     gt.Gfun.rtkpos(grtk, gobsr, gnav, gopt, gobsb);
8| gsol.plot(); % Plot solution
9| gsol.outSol("rover.pos"); % Save solution

```

Figure 5: Source code of example 2: Post-processing kinematic (PPK) analysis using GNSS base station observation.

L1–L3 reads the RINEX observations and navigation data from the rover and reference station. L4 reads the configuration file, which is a processing option for RTKLIB, that contains the base station coordinates and analysis settings. L6–L7 calls the RTKPOS function of RTKLIB to perform PPK and generate a Gsol object, a class of positioning solutions. As shown on Figure 6, L8 represents the positioning solution. The green dots represent the FIX solution with resolved carrier-phase ambiguity and yellow dots

represent the float solution. L9 outputs the computed solution to a file in RTKLIB solution format. Thus, MatRTKLIB facilitates performing positioning calculations and visualizing positioning solutions.

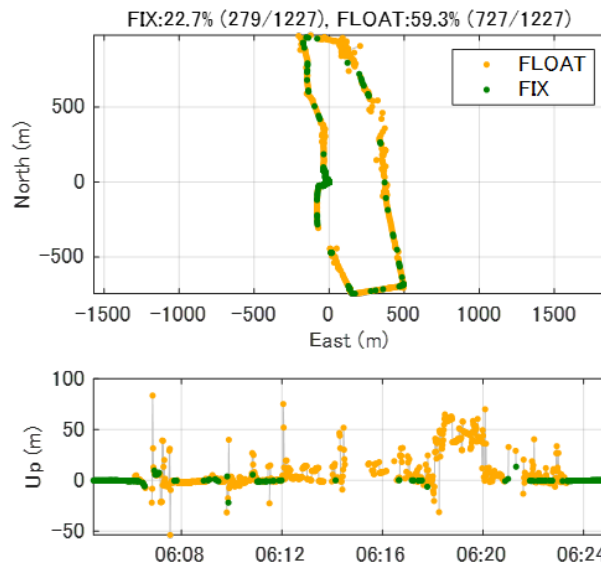


Figure 6: Visualize the position and status of GNSS positioning solutions. The color of the plot indicates whether carrier phase ambiguity is solved.

6.3.3 Example 3: GNSS position error analysis

This example is provided as `Example3_Positioining_error_analysis.m` in the examples folder. The source code for analyzing the errors in the positioning solution in Example 3 is shown in Figure 7.

```
1| gsol = gt.Gsol("rover.pos"); % Read position solution
2| gsol = gsol.fixedInterval(); % Constant time interval
3| ref = readmatrix("ref.csv"); % Read reference position
4| gpos = gt.Gpos(ref, "llh"); % Create Gpos object
5| gerr = gsol-gpos;           % Create Gerr object
6| gerr.plotCDF3D              % Plot CDF of 3D error
```

Figure 7: Source code of example 3: Positioning error analysis.

L1 creates a Gsol object by passing a positioning solution file through a Gsol class constructor. L2 is a process that makes the time interval of the solution constant. L3–L4 read the reference location from the CSV file to create the Gpos object. L5 created an object of class Gerr by subtracting the

Gsol and Gpos objects. Several GT classes support intuitive subtraction. L6 shows the cumulative distribution function of the third-dimensional position error plotted using the Gerr class methods (Figure 8). The Gerr class also provides other useful methods for analyzing GNSS positioning errors, including calculating statistics and plotting various errors.

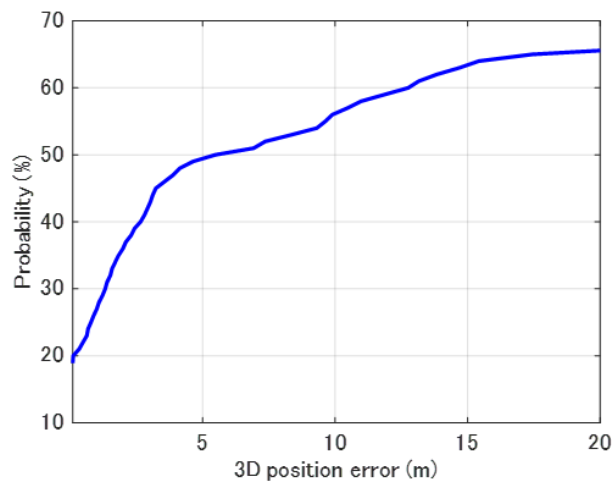


Figure 8: Visualize the error of GNSS positioning solutions. This figure shows the cumulative distribution function of 3D position error.

7 Appendix 1: Supported RTKLIB functions

7.1 Satellites, Systems, and Codes functions

RTKLIB function name	Function	Ported	Vector input support	Note
satno	Convert satellite system+prn/slot number to satellite number	☑	☑	
satsys	Convert satellite number to satellite system	☑	☑	
satid2no	Convert satellite id to satellite number	☑	☑	
satno2id	Convert satellite number to satellite id	☑	☑	
obs2code	Convert obs code string to obs code	☑	☑	
code2obs	Convert obs code to obs code string	☑	☑	
code2freq	Convert system and obs code to carrier frequency	☑	☑	
sat2freq	Convert satellite and obs code to frequency	☑	☑	
code2idx	Convert system and obs code to frequency index	☑	☑	

7.2 Time and String functions

RTKLIB function name	Function	Ported	Vector input support	Note
tow2epoch	Convert GPS time of week to calendar day/time	☒	☒	Function change from gpst2time
epoch2tow	Convert calendar day/time to GPS time of week	☒	☒	Function change from time2gpst
gsttow2epoch	Convert Galileo time of week to calendar day/time	☒	☒	Function change from gst2time
epoch2gsttow	Convert calendar day/time to Galileo time of week	☒	☒	Function change from time2gst
bdttow2epoch	Convert BeiDou time of week to calendar day/time	☒	☒	Function change from bdt2time
epoch2bdttow	Convert calendar day/time to BeiDou time of week	☒	☒	Function change from time2bdt
gpst2utc	Convert GPST epoch to UTC epoch	☒	☒	
utc2gpst	Convert UTC epoch to GPST epoch	☒	☒	

RTKLIB function name	Function	Ported	Vector input support	Note
gpst2bdt	Convert GPST epoch to BDT epoch	☒	☒	
bdt2gpst	Convert BDT epoch to GPST epoch	☒	☒	
epoch2doy	Convert calendar day/time to day of year	☒	☒	Function change from time2doy
tow2doy	Convert GPS time of week to to day of year	☒	☒	Function change from time2doy
utc2gmst	Convert utc to GMST (Greenwich Mean Sidereal Time)	☒	☒	
adjgpsweek	Adjust GPS week number using cpu time	☒	☒	
reppath	Replace keywords in file path	☒		

7.3 Coordinates transformation

RTKLIB function name	Function	Ported	Vector input support	Note
xyz2llh	Transform ECEF position to geodetic position	☒	☒	Function change from ecef2pos

RTKLIB function name	Function	Ported	Vector input support	Note
llh2xyz	Transform geodetic position to ECEF position	☒	☒	Function change from pos2ecef
xyz2enu	Transform ECEF position to local ENU position	☒	☒	New development function
enu2xyz	Transform local ENU position to ECEF position	☒	☒	New development function
enu2llh	Transform local ENU position to geodetic position	☒	☒	New development function
llh2enu	Transform geodetic position to local ENU position	☒	☒	New development function
ecef2enu	Transform ECEF “vector” to local tangential coordinate	☒	☒	Function change from ecef2enu
enu2ecef	Transform ENU “vector” to ECEF coordinate	☒	☒	Function change from enu2ecef
covenu	Transform xyz-ECEF covariance to local ENU coordinate	☒	☒	

RTKLIB function name	Function	Ported	Vector input support	Note
covenusol	Transform xyz-ECEF covariance to local ENU coordinate	☒	☒	New development function
covecef	Transform local ENU covariance to xyz-ECEF coordinate	☒	☒	
covecefsol	Transform local ENU covariance to xyz-ECEF coordinate	☒	☒	New development function
eci2ecef	Compute ECI to ECEF transformation matrix	☒	☒	
deg2dms	Convert degree to degree-minute-second	☒	☒	
dms2deg	Convert degree-minute-second to degree	☒	☒	

7.4 Input and Output functions

RTKLIB function name	Function	Ported	Vector input support	Note
readpos	Read positions from station position file	☒	☒	

RTKLIB function name	Function	Ported	Vector input support	Note
readblq	Read BLQ ocean tide loading parameters	☒	☒	
readerp	Read earth rotation parameters	☒	☒	
geterp	Get earth rotation parameter values	☒	☒	

7.5 Platform dependent functions

RTKLIB function name	Function	Ported	Vector input support	Note
expath	Expand file path with wild-card (*) in file	☒		

7.6 Positioning models

RTKLIB function name	Function	Ported	Vector input support	Note
satazel	Compute satellite azimuth/elevation angle	☒	☒	
geodist	Compute geometric distance and receiver-to-satellite unit vector	☒	☒	

RTKLIB function name	Function	Ported	Vector input support	Note
dops	Compute DOP (dilution of precision) from azimuth and elevation	☑	☑	

7.7 Atmosphere models

RTKLIB function name	Function	Ported	Vector input support	Note
ionmodel	Compute ionospheric delay by broadcast ionosphere model (klobuchar model)	☑	☑	
ionmapf	Compute ionospheric delay mapping function by single layer model	☑	☑	
ionppp	Compute ionospheric pierce point (ipp) position and slant factor	☑	☑	
tropmodel	Compute tropospheric delay by standard atmosphere and saastamoinen model	☑	☑	

RTKLIB function name	Function	Ported	Vector input support	Note
tropmapf	Compute tropospheric mapping function by NMF	☑	☑	
iontec		WIP		
readtec		WIP		
ionocorr	Compute ionospheric correction	☑	☑	
tropcorr	Compute tropospheric correction	☑	☑	

7.8 Antenna models

RTKLIB function name	Function	Ported	Vector input support	Note
readpcv	Read antenna parameters	☑		
searchpcv	Search antenna parameter	☑		
antmodel	Compute receiver antenna offset by antenna phase center parameters	☑	☑	
antmodel_s	Compute satellite antenna offset by antenna phase center parameters	☑	☑	

7.9 Earth tide models

RTKLIB function name	Function	Ported	Vector input support	Note
sunmoonpos	Get sun and moon position in ECEF	☑	☑	
tidedisp	Compute displacements by earth tides	☑	☑	

7.10 Geoid models

RTKLIB function name	Function	Ported	Vector input support	Note
geoidh	Get geoid height from geoid model	☑	☑	

7.11 Datum transformation

RTKLIB function name	Function	Ported	Vector input support	Note
tokyo2jgd	Transform position in Tokyo datum to JGD2000 datum	☑	☑	
jgd2tokyo	Transform position in JGD2000 datum to Tokyo datum	☑	☑	

7.12 RINEX functions

RTKLIB function name	Function	Ported	Vector input support	Note
readrnxobs	Read RINEX observation file	☒		Function change from readrnx
readrnxnav	Read RINEX navigation file	☒		Function change from readrnx
outrnxobs	Output RINEX observation file	☒		outrnxobsb+outrnxobsb
outrnxnav	Output RINEX navigation file		☒	
readrnxcl	Read RINEX clock files	☒		
convrnx		WIP		

7.13 Ephemeris and clock functions

RTKLIB function name	Function	Ported	Vector input support	Note
eph2clk	Compute satellite clock bias with broadcast ephemeris (GPS, GAL, QZS, BDS, IRN)	☒	☒	
geph2clk	Compute satellite clock bias with GLONASS ephemeris	☒	☒	
seph2clk		WIP		

RTKLIB function name	Function	Ported	Vector input support	Note
eph2pos	Compute satellite position and clock bias with broadcast ephemeris (GPS, GAL, QZS, BDS, IRN)	☑	☑	
geph2pos	Compute satellite position and clock bias with GLONASS ephemeris	☑	☑	
seph2pos		WIP		
peph2pos	Compute satellite position/clock with precise ephemeris/clock	☑	☑	
satantoff	Compute satellite antenna phase center offset in ECEF coordinate			
☑	☑			
satpos	Compute satellite position, velocity and clock	☑	☑	
satposs	Compute satellite position, velocity and clock	☑	☑	
readsp3	Read SP3 file	☑		
readsap	Read satellite antenna parameters	☑		

RTKLIB function name	Function	Ported	Vector input support	Note
readdcb	Read differential code bias (DCB) parameters	☒		
alm2pos		WIP		
tle_read		WIP		
tle_name_read		WIP		
tle_pos		WIP		

7.14 RTCM functions

RTKLIB function name	Function	Ported	Vector input support	Note
gen_rtc2		WIP		
gen_rtc3		WIP		

7.15 Solution functions

RTKLIB function name	Function	Ported	Vector input support	Note
readsol	Read rtklib solution file	☒		
readsolstat	Read rtklib solution status file	☒		
outsol	Output rtklib solution file	☒		
outsolex		WIP		
outnmea_rmc		WIP		

RTKLIB function name	Function	Ported	Vector input support	Note
outnmea_gga		WIP		
outnmea_gsv		WIP		

7.16 Google earth kml/gpx converter

RTKLIB function name	Function	Ported	Vector input support	Note
convkml	Convert from solution files to Google Earth KML files	☒		
convgpx	Convert from solution files to GPX files	☒		

7.17 SBAS functions

RTKLIB function name	Function	Ported	Vector input support	Note
sbsreadmsg		WIP		
sbssatcorr		WIP		
sbsioncorr		WIP		
sbstropcorr		WIP		

7.18 Options functions

RTKLIB function name	Function	Ported	Vector input support	Note
loadopts	Load option struct	☒		
saveopts	Save option struct	☒		

7.19 Integer ambiguity resolution

RTKLIB function name	Function	Ported	Vector input support	Note
lambda	Integer least-square estimation	☒	WIP	

7.20 Standard positioning

RTKLIB function name	Function	Ported	Vector input support	Note
pntpos	Compute receiver position, velocity, clock bias by single-point positioning	☒	☒	

7.21 Precise positioning

RTKLIB function name	Function	Ported	Vector input support	Note
rtkinit	Initialize RTK control struct	☒		

RTKLIB function name	Function	Ported	Vector input support	Note
rtkpos	Compute rover position by precise positioning	☑	☑	

7.22 Precise point positioning

RTKLIB function name	Function	Ported	Vector input support	Note
pppos		WIP		

8 Appendix 2: GT Classes

Class name	Function
Gobs	GNSS observation: read/edit/write/visualization
Gnav	GNSS navigation: read/edit/write/visualization
Gtime	GPS time: time system conversion
Gpos	Geodetic position: coordinate system conversion
Gvel	Velocity: coordinate system conversion
Gcov	Covariance: coordinate system conversion
Gsat	Satellite-related data: edit/visualization
Gsol	Position solution: read/edit/write/visualization
Gstat	Position status: read/edit/write/visualization
Grtk	RTK control class
Gopt	Process option: read/edit/write
Gfun	Wrapper for positioning function
C	Define constants

9 Appendix 3: Examples

File	Description
compute_double_difference.m	Compute double-differenced GNSS observation
compute_fixrate.m	Compute ambiguity fixed rate from RTK-GNSS solution
compute_float_ambiguity.m	Compute double-differenced float carrier phase ambiguity
compute_geoid.m	Compute Geoid and orthometric height
compute_mean_position.m	Compute mean position from solution file
compute_residuals_doppler.m	Compute Doppler residuals
compute_residuals_pseudorange.m	Compute pseudorange residuals
convert_coordinate.m	Convert LLH, ECEF and ENU position to each other
convert_solution_to_kml.m	Convert positioning solution to Google Earth KML file
convert_time.m	Convert GPS time, calendar time and UTC time to each other
edit_rinex_observation1.m	Read and write RINEX observation
edit_rinex_observation2.m	Trim RINEX observation using time span
edit_rinex_observation3.m	Modify RINEX observation interval
edit_rinex_observation4.m	Exclude satellites from RINEX observation
edit_solution.m	Read position solution file and trim solution
estimate_position_rtk.m	RTK-GNSS positioning using RTKLIB
estimate_position_rtk_step_by_step.m	Step by step example of RTK-GNSS positioning
estimate_position_spp.m	Single point positioning using RTKLIB
estimate_position_spp1_step_by_step.m	Step by step example of single point positioning
estimate_position_spp2_step_by_step.m	Step by step example of single point positioning
estimate_velocity_doppler_step_by_step.m	Step by step example of velocity estimation by Doppler

File	Description
estimate_velocity_tdcg_step_by_step.m	Step by step example of velocity estimation by TDCP
evaluate_position_error.m	Evaluate positioning accuracy and plot error
evaluate_velocity_error.m	Evaluate velocity accuracy and plot error
Example1_visualization_RINEX_observation.m	Visualization of RINEX observation in GPS solutions paper
Example2_PPK_analysis.m	Post-processing kinematic example in GPS solutions paper
Example3_Positioning_error_analysis.m	Positioning error analysis in GPS solutions paper
generate_configuration_file.m	Generate RTKLIB configuration file
generate_solution_file.m	Generate RTKLIB solution file
plot_observation1.m	Show observation status and number of satellite
plot_observation2.m	Show raw GNSS measurements
plot_position.m	Show position on map
plot_satellite_constellation1.m	Show satellite constellation
plot_satellite_constellation2.m	Show satellite elevation and azimuth angles
plot_solution.m	Show RTK position solutions
