Normal Procedures





AIRSPEEDS FOR NORMAL OPERATION

Unless otherwise noted, the following speeds are based on a maximum weight of 8750 pounds for takeoff and 8500 pounds for landing and may be used for any lesser weight. However, to achieve the performance specified for takeoff distance, climb performance, and landing distance, the speed appropriate to the particular weight must be used.

- 1 /	4 1 1	 , _	_	_

Short Field Take	Flaps 20°eoff, Flaps 20°, Speed at 50 Feetor Type IV Anti-ice Fluid Takeoff (Flaps UP)	86 KIAS
Cruise Climb Best Rate of Clin Best Rate of Clin	mb, FLAPS UP: mb, Sea Level to 10,000 Feet mb, 20,000 Feet. imb, Sea Level to 20,000 Feet	
Normal Approac	ROACH: ch, Flaps UP ch, Flaps FULL roach, Flaps FULL	75-85 KIAS



AIRSPEEDS FOR NORMAL OPERATION (Continued)

BALKED LANDING: Takeoff Power, Flaps 20°
MAXIMUM RECOMMENDED TURBULENT AIR PENETRATION SPEED: 8750 Pounds .148 KIAS 7500 Pounds .137 KIAS 6250 Pounds .125 KIAS 5000 Pounds .112 KIAS
MAXIMUM DEMONSTRATED CROSSWIND VELOCITY: Takeoff or Landing



CABIN

1. PITOT/ STATIC Tube Covers	
2. Pilot's Operating Handbook	ACCESSIBLE TO PILOT
3. Garmin G1000 CRG	ACCESSIBLE TO PILOT
4. Control Locks	REMOVE (disengage RUDDER LOCK)
5. PARKING BRAKE	SET



CABIN (Continued)

6. All Switches	OFF
7. Circuit Breakers	
8. ALT STATIC AIR	
9. INERTIAL SEPARATOR T-Handle	
10. STBY FLAP MOTOR Switch	
11. OXYGEN SUPPLY PRESSURE (if installed)	
12. Oxygen Masks (if installed)	
13. FUEL TANK SELECTOR Valves	
44 VENTU ATION FANCYAID CONDITIONING (IS in a	(feel against stop)
14. VENTILATION FANS/ AIR CONDITIONING (if ins	
15. BLEED AIR HEAT Switch	
16. EMERGENCY POWER Lever	
17. TRIM Controls	
18. FUEL SHUTOFF Knob	
19. CABIN HEAT FIREWALL SHUTOFF Control	
20. BATTERY Switch	
	n fans audible and airflow from each fan)
21. AVIONICS No. 1 Switch	
22. PFD 1	,
23. AVIONICS No. 2 Switch	
24. PFD 2 and MFD	
25. FUEL QTY	
26. ENGINE Softkey	
27. SYSTEM Softkey	
Reset Fuel Totalizer if desired. Select ENGINE S	, ,
28. WING FLAPS Handle	FULL DOWN
29. PITOT/STATIC and	
STALL HEAT Switches	ON FOR 30 SECONDS;
	THEN OFF
30. AVIONICS No. 1 and No. 2 Switches	OFF
31. BATTERY Switch	OFF

LEFT SIDE

NOTE

Properly dispose of samples from all fuel drains. Aviation turbine fuel will deteriorate asphalt surfaces.

4. Inboard Fuel Tank Sump and External SumpQuick-Drain Valves DRAIN (using fuel sampler) Drain to check for water, sediment, and proper fuel before each flight and after each refueling. If water is observed, take additional samples until clear. Take repeated samples from all fuel drain points until all contamination has been removed.

LEFT WING Leading Edge

WARNING

• It is essential in cold weather to remove even the smallest accumulations of frost, ice, snow, or slush from the wing and control surfaces. To assure complete removal of contamination, conduct a visual and tactile inspection up to two feet behind the protected surfaces at one location along the wing span as a minimum. Also, make sure the control surfaces contain no internal accumulations of ice or debris. If these requirements are not performed, aircraft performance will be degraded to a point where a safe takeoff and climb may not be possible.



LEFT WING Leading Edge (Continued)

WARNING

• Prior to any flight in known or forecast icing conditions, check that PITOT/STATIC tube(s) and STALL warning heaters are warm to touch after turning PITOT/STATIC and STALL HEAT switches ON for 30 seconds, then OFF. Make sure the pitot covers are removed prior to turning PITOT/ STATIC HEAT ON.

1. Wing Tie-Down
(verify freedom of movement, audible warning and warmth)
Ensure the elevator control is off the forward stop in order to check audible warning. 3. PITOT/ STATIC Tube
(verify security, openings for stoppage and warmth)
4. LDG and TAXI/RECOG Lights
5. Fuel Quantity
6. Fuel Filler Cap SECURE
7. Outboard Fuel Tank Sump
Quick-Drain Valve DRAIN (using fuel sampler) Drain to check for water, sediment and proper fuel before each flight and after each refueling. If water is observed, take additional samples until clear. Take repeated samples from all fuel drain points until all contamination has been removed.
8. NAV and STROBE Lights
(verify condition and cleanliness)



LEFT WING Trailing Edge

1. Fuel Tank Vent	
	(verify no obstructions)
2. Aileron and Servo Tab	
	(verify condition and security)
3. Static Wicks (4 total)	CHECK
	(verify condition)
4. Spoiler	
·	(verify condition and security)
5. Flap Leading Edge Vortex Generators	` ,
	(verify condition and security)
6. Flap	
	(verify condition and security)



EMPENNAGE

WARNING

It is essential in cold weather to remove even the smallest accumulations of frost, ice, snow, or slush from the tail and control surfaces. Exercise caution to avoid distorting the vortex generators on horizontal stabilizer while deicing. To assure complete removal of contamination, conduct a visual and tactile inspection of all critical surfaces. Also, make sure the control surfaces contain no internal accumulations of ice or debris. If these requirements are not performed, aircraft performance will be degraded to a point where a safe takeoff and climb may not be possible.

Baggage	CLOSED and LATCHED
Verify condition, security, and verify 18 vortex g horizontal stabilizer.	
4. Control Surfaces and Elevator Trim Tabs Verify condition, security, freedom of movement	
5. Static Wicks (14 total)	
6. Rudder Gust Lock	DISENGAGE
7. NAV Light	CHECK
	(verify condition and cleanliness)
8. Tail Tie-Down	
9. Oxygen Filler Door (if installed)	SECURE
10. Passenger Entry Door (if installed)	
	(condition and security)

RIGHT WING Trailing Edge	
1. Flap	
2. Spoiler	(verify condition and security)
Flap Leading Edge Vortex Generators	(verify condition and security)
4. Aileron and Trim Tab	(verify condition and security)
	(verify condition and security)
5. Static Wicks (4 total)	(verify condition)
6. Fuel Tank Vent	

RIGHT WING Leading Edge

WARNING

- It is essential in cold weather to remove even the smallest accumulations of frost, ice, snow, or slush from the wing and control surfaces. To assure complete removal of contamination, conduct a visual and tactile inspection up to two feet behind the protected surfaces at one location along the wing span as a minimum. Also, make sure the control surfaces contain no internal accumulations of ice or debris. If these requirements are not performed, aircraft performance will be degraded to a point where a safe takeoff and climb may not be possible.
- Prior to any flight in known or forecast icing conditions, check that PITOT/ STATIC tube(s) and STALL warning heaters are warm to touch after turning PITOT/STATIC and STALL HEAT switches ON for 30 seconds, then OFF. Make sure the pitot covers are removed prior to turning PITOT/STATIC HEAT ON.

RIGHT WING Leading Edge (Continued)

1. NAV and STROBE Lights
3. Fuel Filler Cap
4. Outboard Fuel Tank Sump Quick-Drain Valve (if airplane parked with one wing low on a sloping ramp) DRAIN (using fuel sampler) Drain to check for water,
sediment and proper fuel before each flight and after each refueling. If water is observed, take additional samples until clear. Take repeated samples from all fuel drain points until all contamination has been removed.
5. LND and TAXI/ RECOG Lights
(verify condition and cleanliness) 6. PITOT/ STATIC Tube
7. Radome (if installed)
(verify condition and security) 8. Wing Tie-Down
Quick-Drain Valves DRAIN (using fuel sampler) Drain to check for water, sediment, and proper fuel before each flight and after each refueling. If water is observed, take additional samples until clear. Take repeated samples from all fuel drain points until all contamination has been removed.
10. Main Landing Gear
(check proper tire inflation and condition of gear)



NOSE

WARNING

It is essential in cold weather to remove even the smallest accumulations of frost, ice, snow, or slush from the propeller blades and spinner, and the air inlets (starter/generator, oil cooler and engine inlets). To assure complete removal of contamination, conduct a visual and tactile inspection of all critical surfaces. If these requirements are not performed, aircraft performance will be degraded to a point where a safe takeoff and climb may not be possible.

1. Right Crew Door
WARNING
Avoid touching the output connectors or coupling nuts or ignition excitor with bare hands.
5. Battery
6. Exhaust System
(verify condition, security, cracks, distortion and damage)
7. Cowling
8. Propeller Anchor
9. Air Inlet Covers
10. Air Inlets
Check starter/generator blast tube opening and oil cooler inlet (right) and engine induction air inlet (left) for condition, restrictions, and debris.

NOSE (Continued)

11 . Propeller
12. Propeller Spinner
13. Nose Wheel Strut and Tire
14. Cowling
(left side of upper cowling for access and check condition and security) 15. Engine (left side)
(verify general condition, security, fuel, no oil leakage, and no damage to any components) 16. INERTIAL SEPARATOR Bypass Outlet
17. Oil Dipstick/Filler Cap
1 1/2 quarts of MAX HOT or MAX COLD (as appropriate) on dipstick. Markings indicate U.S. quarts low if oil is hot.
WARNING
Make sure the oil dipstick cap is securely latched down. Operating the engine with less than the recommended oil level and with the dipstick cap unlatched will result in excessive oil loss and eventual engine stoppage.
18. Electrical Power Box Circuit Breakers and Diodes
(for proper location - flush)

NOSE (Continued)

21. Brake Fluid Reservoir
BEFORE STARTING ENGINE 1. Preflight Inspection
WARNING Failure to correctly use seat belts and shoulder harnesses could result in serious or fatal injury in the event of an accident. 9. Switches

(Continued Next Page)

BEFORE STARTING ENGINE (Continued)

CAUTION

Leaving the BLEED AIR HEAT Switch ON (up) can result in a hot start or abnormal acceleration to idle.

15. CABIN HEAT MIXING AIR Control 16. EMERGENCY POWER Lever. 17. POWER Lever. 18. PROP RPM Lever 19. FUEL CONDITION Lever 20. FUEL SHUTOFF Knob 21. BATTERY Switch 22. WING FLAPS Handle 23. NO SMOKE/SEAT BELT Switches (if installed) 24. TEST SWITCH	
STARTING ENGINE (Battery Start) 1. BATTERY Switch 2. BCN Switch 3. AVIONICS No. 1 Switch 4. EIS 5. BUS VOLTS 6. EMERGENCY POWER Lever.	ONCHECK PARAMETERS (verify no red X's)CHECK (24 volts minimum)

CAUTION

Make sure that the EMERGENCY POWER Lever is in the NORMAL (full aft) position or an over-temperature condition will result during engine start.

STARTING ENGINE (Battery Start) (Continued)

7. Propeller Area
CAUTION
 If ITT climbs rapidly towards 1090°C, be prepared to return the FUEL CONDITION Lever to CUTOFF. Under hot OAT and/or high ground elevation conditions, idle ITT can exceed maximum idle ITT limitation of 685°C. Increase Ng and/or reduce accessory load to maintain ITT within limits.
TITI IIIIII lation of 665 C. Increase by and/or reduce accessory load to maintain ITT within limits.
c. Ng
Defrosting Controls

STARTING ENGINE (External Power Start)

(24-28 Volt. Minimum 800 Amp and Maximum 1700 Amp Capacity)

(24-28 Voit, Minimum 800 Amp and Maximum 1700 Amp Capacity)
1. BATTERY SwitchON 2. AVIONICS No. 1 SwitchON 3. EIS
CAUTION
Make sure that the EMERGENCY POWER Lever is in the NORMAL position or an over-temperature condition will result during engine start.
10. BATTERY Switch ON 11. BCN Switch ON 12. AVIONICS No. 1 Switch ON 13. BUS VOLTS CHECK (MIN 20 Volts) 14. EXTERNAL POWER Switch STARTER 15. EMERGENCY POWER Lever. NORMAL (verify EMERG PWR LVR CAS MSG - OFF) 16. Propeller Area CLEAR 17. FUEL BOOST Switch ON a. FUEL BOOST ON CAS MSG ON b. FUEL PRESS LOW CAS MSG OFF c. FFLOW PPH ZERO
CAUTION If the external power unit drops off the line, initiate engine shutdown.
18. STARTER Switch

(Continued Next Page)

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STARTING ENGINE (External Power Start) (Continued)

CAUT	ION
 If ITT climbs rapidly towards 1090°C, be prepa CUTOFF. 	red to return the FUEL CONDITION Lever to
 Under hot OAT and/or high ground elevation of ITT limitation of 685°C. Increase Ng and/or red limits. 	
	52% MINIMUM
20. STARTER Switch	(verify STARTER ON CAS MSG - OFF)
21. EIS	
22. EXTERNAL POWER Switch	
23. External Power Unit	
	FF CAS MSG OFF and BAT AMPS charging)
25. FUEL BOOST Switch	
	(verify FUEL BOOST ON CAS MSG - OFF)
26. AVIONICS No. 2 Switch	
27. NAV LIGHTS	AS REQUIRED
28. Cabin Heating, Ventilating and	AS DESIDED
Deliosting Controls	

TAXIING

1. Brakes
NOTE Propeller BETA range can be used during taxi with minimum blade erosion up to the point where Ngincreases (against beta range spring) to control taxi speed and improve brake life.
2. Flight Instruments
BEFORE TAKEOFF 1. PARKING BRAKE
WARNING Failure to correctly use seat belts and shoulder harnesses can result in serious or fatal injury in the event of an accident.
3. Flight Controls FREE and CORRECT 4. Flight Instruments CHECK 5. Altimeters: a. PFD 1 and 2 BARO SET b. Standby Altimeter SET 6. ALT SEL SET 7. Standby Flight Instruments CHECK 8. FUEL BOOST Switch NORM 9. FUEL TANK SELECTORS BOTH ON 10. FUEL QTY CHECK 11. FUEL SHUTOFF Knob FULLY ON 12. ELEVATOR, AILERON, and RUD TRIM Controls 3 SET
13. POWER Lever

BEFORE TAKEOFF (Continued)

a. PROP RPM Lever b. OVERSPEED GOVERNOR TEST Button c. POWER Lever d. POWER Lever e. OVERSPEED GOVERNOR TEST 15. Quadrant Friction Lock 16. Standby Power (first flight of the data a. ENGINE Softkey. b. STBY ALT PWR Switch c. GEN AMPS Generator load can be increased 60 amps. d. ALT AMPS e. GENERATOR Switch	(first flight of the day and after maintenance)
g. BUS VOLTS	
A fully charged battery will carry part of the tor to standby alternator power because of	electrical load when initially switching from generathe generator's higher voltage regulation.
h. STBY PWR ON CAS MSG	
	(verify GENERATOR OFF CAS MSG ON)RESET (verify GENERATOR OFF CAS MSG OFF)
I. STBY ALT PWR Switch	ON

(Continued Next Page)

NOT INTENTED FOR REAL FLIGHTS.



BEFORE TAKEOFF (Continued)

17. Manual Electric Pitch Trim (MEPT)	switch NOSE DOWN (verify correct trim wheel and pointer ISC/TRIM INTER Switch (verify trim wheel stops moving). switch NOSE UP (verify correct trim wheel and pointer ISC/TRIM INTER Switch (verify trim wheel stops moving).
d. Set trim as required within TAKEOFF band.	in TAKEOFF band.
18. Known Icing System (if installed)	
(see Systems Checks prior to any flight in icing conditions)	
19. Ice Protection (if installed)	
a. PITOT/STATIC HEAT ON (when OAT is below 5°C (41°F)	
b. STALL HEAT	
c. PROP HEAT	
20. INERTIAL SEPARATOR	
21. Avionics and Radar	
22. Nav Source	SET FOR DEPARTURE
23. XPDR	
24. STROBE Lights	ON
25. CAS MSG(s)	
26. WING FLAPS Handle	
27. CABIN HEAT MIXING AIR Control	
28. Window	
29. Brakes	
30. FUEL CONDITION Lever	



BEFORE TAKEOFF (Continued)

WARNING

- When ground icing conditions are present, a pre-takeoff visual and tactile check should be conducted by the pilot in command within five minutes of takeoff, preferably just prior to taxiing onto the active runway.
- Takeoff is prohibited with any frost, ice, snow, or slush adhering to the wings, tail, control surfaces, propeller blades, or engine air inlets.
- Even small amounts of frost, ice, snow, or slush on the wing can adversely change lift and drag. Failure to remove these contaminants will degrade airplane performance to a point where a safe takeoff and climb may not be possible.
- Make sure that the anti-ice fluid (if applied) is still protecting the airplane.

TAKEOFF

6. WING FLAPS Handle

NORMAL TAKEOFF	
1. WING FLAPS Handle	UP or TO/APR
	TO/APR Recomended
2. POWER Lever	
	(observe Takeoff ITT and Nglimits)
3. CAS MSG(s)	
4. Rotate	
5. Airspeed	85-95 KIAS

RETRACT to UP (after reaching 95 KIAS)

SHORT FIELD TAKEOFF 1. WING FLAPS Handle
TO/APR Recomended 2. Brakes
3. POWER Lever
4. CAS MSG(s)
7. Airspeed 86 KIAS (until all obstacles are cleared) 8. WING FLAPS Handle
RETRACT to 10° (after reaching 85 KIAS) RETRACT to UP (after reaching 95 KIAS)
TYPE II, TYPE III OR TYPE IV ANTI-ICE FLUID TAKEOFF
1. WING FLAPS Handle
4. Rotate .83 KIAS 5. Airspeed .104 KIAS
ENROUTE CLIMB
CRUISE CLIMB 1. Ice Protection (if installed)
4. PROP RPM Lever

(Continued Next Page)

NOTE

To achieve maximum flat rated horsepower, use a minimum of 1800 RPM.

6. POWER Lever
(observe Maximum Climb ITT and Nglimits)
NOTE Engine operations which exceed 740°C ITT can reduce engine life.
CAUTION
For every 10° below -30°C ambient temperature, reduce maximum allowable Ngby 2.2%.
MAXIMUM PERFORMANCE CLIMB 1. Ice Protection (if installed)
CAUTION

• For every 10° below -30°C ambient temperature, reduce maximum allowable Ngby 2.2%.

• Engine operations which exceed 740°C ITT can reduce engine life.



CRUISE

1. Ice Protection (if installed)
a. PITOT/STATIC HEAT . ON (when OAT is below 5°C (41°F)
b. STALL HEAT ON (when OAT is below 5°C (41°F)
c. PROP HEAT ON (when OAT is below 5°C (41°F)
2. INERTIAL SEPARATOR
3. PROP RPM Lever
4. POWER Lever
(observe Maximum Cruise ITT and Nglimits)
Refer to Cruise Performance and/or Cruise Maximum Torque charts
5. Fuel Balance
(maximum 200 pounds imbalance)

NOTE

Engine operations which exceed 740°C ITT can reduce engine life.

CAUTION

For every 10° below -30°C ambient temperature, reduce maximum allowable Ngby 2.2%.

DESCENT

1. Ice Protection (if installed) AS REQUIRED
a. PITOT/STATIC HEAT . ON (when OAT is below 5°C (41°F)
b. STALL HEAT ON (when OAT is below 5°C (41°F)
c. PROP HEAT ON (when OAT is below 5°C (41°F)
2. INERTIAL SEPARATOR
3. NO SMOKE/SEAT BELT SIGN Switches (if installed)
4. Altimeters
5. NAV Source

NOTE

The overspeed warning horn and MAXSPD annunciation will activate when either PFD1 or PFD2 airspeed reaches greater than 175 KIAS. In addition, the overspeed warning horn and MAXSPD annunciation may appear prior to 175 KIAS if the aircraft is accelerating at a rate that will rapidly exceed VMO.

DESCENT (Continued)

CAUTION Set PROP RPM Lever at 1900 RPM prior to beginning any instrument approach procedure.			
6. POWER Lever			
BEFORE LANDING			
1. Seats, Seat Belts, Shoulder Harnesses			
WARNING Failure to correctly use seat belts and shoulder harnesses could result in serious o fatal injury in the event of an accident.			
2. FUEL TANK SELECTORS			
7. WING FLAPS Handle			
NORMAL LANDING 1. WING FLAPS HandleFULL 2. Airspeed75-85 KIAS 3. TouchdownMAIN WHEELS FIRST 4. POWER LeverBETA RANGE AFTER TOUCHDOWN			



SHORT FIELD LANDING

1. WING FLAPS HandleLAND2. Airspeed.78 KIAS3. POWER Lever.REDUCE to IDLE (after clearing obstacles)4. Touchdown.MAIN WHEELS FIRST5. POWER Lever.BETA RANGE AFTER TOUCHDOWN				
NOTE				
Use of reverse thrust will reduce the landing roll by approximately 10%				
6. Brakes				
BALKED LANDING				
1. POWER Lever				



AFTER LANDING

1. WING FLAPS Handle	UE
2. Ice Protection (if installed)	
a. PITOT/STATIC HEAT	OFF
b. STALL HEAT	OFF
c. PROP HEAT	
3. STBY ALT PWR	OFF
4. STROBE Lights	
5. LDG and TAXI/RECOG Lights	
6. FUEL CONDITION Lever	
	(when clear of the runway

CAUTION

If the FUEL CONDITION Lever is moved past the LOW IDLE position and the engine Ng falls below 53%, moving the lever back to the LOW IDLE position can cause an ITT over-temperature condition. If the engine has started to shutdown in this situation, allow the engine to complete its shutdown sequence, and proceed to do a normal engine start using the "Starting Engine" checklist.

SHUTDOWN AND SECURING AIRPLANE

1. PARKING BRAKE	
· · · · · · · · · · · · · · · · · · ·	OFF
3. POWER Lever.	
4. ITT	
	(at minimum temperature for one minute)
5. PROP RPM Lever	FEATHER
6. FUEL CONDITION Lever	
7. LIGHTS	OFF
8. FUEL BOOST Switch	OFF
9. AVIONICS No 1 and No 2 Switches	
10. BATTERY Switch	
11. Control Lock	
12. OXYGEN SUPPLY Control Lever (if installed) .	
13. FUEL TANK SELECTORS	
Turn high wing tank off if parked on a sloping su	
14. Tie-Downs and Chocks	
15. External Covers	
16. Fuel Filter	
17 Oil Breathar Drain Con	(for proper location - flush)
17. Oil Breather Drain Can	DRAIN (until empty)

NOTE

Possible delays of subsequent flights, or even missed flights, are often eliminated by routinely conducting a brief postflight inspection. Usually, a visual check of the airplane for condition, security, leakage, and tire inflation will alert the operator to potential problems, and is therefore recommended.