Normal Procedures





A.- INSIDE INSPECTIONS

Cockpit

1 DE ICE SYSTEM panel	
- All switches	OFF
2 ELT	ARM
3 "NORMAL/MASK" micro inverter	NORMAL
4 Flight control lock	REMOVED / STOWED
5 Flight controls	Deflections checked
6 Parking brake	SET
7 Landing gear control	DN
8 Engine controls	
- "MAN OVRD" control	OFF (Notched)

CAUTION WHEN THE ENGINE IS SHUTDOWN, THE POWER LEVER MUST NOT BE MOVED BEHIND THE FLIGHT IDLE POSITION

- Power lever	IDLE
	(Flight idle stop
- Propeller governor lever	MAX. RPM
- Condition lever	CUT OFF
9 Flaps control	UP
10 Fuel tank selector	L or R



11 Landing gear emergency control	
- Lever	PULLED DOWN
- By-pass selector	PUSHED
- Door	
12 ECS panel	
- "BLEED" switch	OFF
- "AIR COND" switch	OFF
- "DUMP" switch	GUARDED
13 Static Air control knob	PUSHED
14 "RAM AIR" control knob	PUSHED
15 Breakers panel	
- All breakers	ENGAGED
16 "AVIONICS" MASTER switch	OFF
17 "AP TRIMS" MASTER switch	OFF
18 Fuel	
- "FUEL SEL" selector	MAN
- "AUX BP" switch	OFF
19 ENGINE START panel	
- "IGNITION" switch	AUTO or OFF
- "STARTER" switch	OFF
20 ELECTRIC POWER panel	
- CRASH lever	UP
- "GENERATOR" selector	MAIN
- "SOURCE" selector	OFF
21 Access lighting	CHECKED
22 INT LIGHTS paneL	OFF



23 EXT LIGHTS panel - All switches 24 Pilots "OXYGEN" switch 25 "PASSENGERS OXYGEN" switch 26 Emergency lighting	OFF
CAUTION	
BEFORE SELECTING SOURCE	E, CHECK :
27 "IGNITION" switch	AUTO or OFF
28 "STARTER" switch	OFF
29 Landing gear control	DN
30 "SOURCE" selector	BAT or GPU
31 Voltage	
- BAT	
- GPU	8 Volts
32 EXT LIGHTS panel	
- "LTS TEST" push button	PRESS
	(All instrument panel lamps ON
	except on landing gear control panel)
- "L.LDG / TAXI / R.LDG" switches	ON
(3 green lamps ON)	
- "L.LDG / TAXI / R.LDG" switches "STROPE"	
- "STROBE" - "NAV"	
- IV/\V	UN



From outside the airplane, check operation of all lights and the stall warning horn

Reentering the airplane	
33 EXT LIGHTS panel	_ALL SWITCHES OFF
34 DE ICE SYSTEM panel	
- All switches	OFF
- "ICE LIGHT"	ON
35 "AVIONICS" MASTER switch	_START
36 CAS display	CHECK
37 Left and right fuel quantities	CHECK
38 EXT LIGHTS panel	
- "LTS TEST" push button	PRESS
	(red and amber MASTER
	warnings ON)
39 Flaps	LDG
40 Landing gear panel	_Warning lights: 3 GREEN ON
	Test 1, then 2 : RED FLASHING
41 DE ICE SYSTEM panel	
- "PITOT L HTR" switch	_ON
WARNING CAS MESSAGE "PITOT HT L"	_OFF
- "PITOT R & STALL HTR" switch	_ON
WARNING CAS MESSAGE "PITOT HT ON LR" _	_OFF
WARNING CAS MESSAGE "STALL HEAT ON"	_OFF
"PITOT L HTR" switch	OFF
"PITOT R & STALL HTR" switch	OFF



WARNING DO NOT TOUCH PITOTS NOR STALL WARNING VANE. THEY COULD BE HOT ENOUGH TO BURN SKIN

42 "AVIONICS" MASTER switch	OFF
43 "SOURCE" selector	OFF
Cabin II	
1 Cabin fire extinguisher	CHECK
	(Pressure / Attachment)
2 Seats / belts	CHECK
3 Windows	CHECK
	(General condition / No crack
4 Emergency exit	CLOSED / LOCKED
- Antitheft safety	REMOVE / STOW
5 Baggage compartment	STRAPS IN PLACE
6 Partition net	IN PLACE
7 Doors operation	CHECK
8 Stairs condition	CHECK
	(Condition / Play)



B - AIRPLANE OUTSIDE

L.H. wing	
1 Flap	CHECK
	(Condition / Play)
2 Aileron and trim / Spoiler	CHECK
	(Condition / Free movement
	/Deflection)
3 Trailing edge static discharger	CHECK
	(Condition / Attachment)
4 Wing tip / nav. lights /	
Strobe / landing light	Condition CHECK
5 OAT probe	
6 Fuel tank	
7 Fuel tank air vent	Unobstructed CHECK
8 Left pitot	Condition CHECK
9 Wing lower surface	CHECK
	(No leak)
10 Wing deicer boots	CHECK
	(Condition / Attachment)
11 Fuel tank drain (two on each wing)	DRAIN
	(Fuel free of water
	and contamination)
12 L.H. main landing gear	
- Shock absorber / doors / tire / wheel well	CHECK



Fuselage forward section

1 Forward compartment	
- Inside	CONTROLLED
- Door	_CLOSED / LOCKED
2 GPU door	CLOSED
	(If not used)
3 Fuel circuit drain	DRAIN
	(Fuel free of water
	and contamination)
- Filter contamination indicator CHECK	,
4 L.H. exhaust stub	CHECK
	(Condition / No crack)
5 Upper engine cowls	,
For the first flight of the day :	
- Oil cap	CLOSED/LOCKED
- Engine oil level	
- Fuel pipes	
. 40. p.p	(No leak, deterioration,
	wear)
6 Engine cowls	,
o. Engine come	CLOSED / LOCKED
7 Air inlets	020025 / 2001125
- Main	No crack - LINOBSTRUCTED
- Lateral / upper	
8 Propeller and spinner	
o i Topeliei aliu spililiei	
	(No nicks, cracks or
	oil leaks / Attachment)



9 Nose gear	011=017
- Landing light / shock absorber / doors / tire / wheel well	
10 R.H. exhaust stub	_ CHECK
	(Condition / No cracks)
R.H. wing	
1 Fuel tank drain (two on each wing)	DRAIN
3)	(Fuel free of water
	,
	and contamination)
2 Main landing gear	
- Shock absorber / doors / tire / wheel well	_ CHECK
3 Wing deicer boots	CHECK
	(Condition / Attachment)
4 Stall warning	CHECK
<u></u>	(Condition / Deflection)
5 Wing lower surface	,
5 Willig lower surface	
	(No leaks)
6 Fuel tank	
7 Fuel tank air vent	_ Unobstructed CHECK
8 Right pitot	_ Condition CHECK
9 Wing tip / nav. light / strobe / landing light	_ Condition CHECK
10 Trailing edge static discharger	
J 1 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	(Condition / Number
	,
	/ Attachment)



12 Flap	11 Aileron / spoiler	CHECK (Condition / Free movement / Deflection)
- Oxygen cylinder OPEN - Oxygen quantity CHECKED 14 Oxygen pressure CHECK Fuselage rear section / Empennages 1 ELT OFF - ELT door CLOSED/LOCKED 2 Static pressure ports CHECK 3 Ventral fins CHECK (Condition / Attachments) 4 Inspection door under fuselage CLOSED - CHECK (Attachments) 5 Horizontal stabilizer deicer boots (R.H. side) CHECK (Condition / Attachments) 6 Elevator and trim CHECK (Condition / Deflection free movement / Trim position) 7 Static dischargers CHECK (Condition / Attachments) 9 Rudder and trim CHECK (Condition / Attachments) 10 Static dischargers CHECK (Condition / Trim position) 10 Static dischargers CHECK (Condition / Trim position) 11 Tail cone Condition - CHECK	12 Flap	
14 Oxygen pressureCHECK Fuselage rear section / Empennages 1 ELTOFF - ELT doorCLOSED/LOCKED 2 Static pressure portsCHECK 3 Ventral finsCHECK (Condition / Attachments) 4 Inspection door under fuselageCLOSED - CHECK (Attachments) 5 Horizontal stabilizer deicer boots (R.H. side)CHECK (Condition / Attachments) 6 Elevator and trimCHECK (Condition / Deflection free movement / Trim position) 7 Static dischargersCHECK (Condition) 8 Vertical stabilizer deicer bootsCHECK (Condition / Attachments) 9 Rudder and trimCHECK (Condition / Trim position) 10 Static dischargersCHECK (Condition / Trim position) 11 Tail coneCONDITION - CHECK	- Oxygen cylinder	
1 ELT		
- ELT door		OFF
2 Static pressure ports		
4 Inspection door under fuselage		
4 Inspection door under fuselage CLOSED - CHECK (Attachments) 5 Horizontal stabilizer deicer boots (R.H. side) CHECK (Condition / Attachments) 6 Elevator and trim CHECK (Condition / Deflection free movement / Trim position) 7 Static dischargers CHECK (Condition) 8 Vertical stabilizer deicer boots CHECK (Condition / Attachments) 9 Rudder and trim CHECK (Condition / Trim position) 10 Static dischargers CHECK (Condition / Trim position) 11 Tail cone CHECK	3 Ventral fins	CHECK
(Attachments) 5 Horizontal stabilizer deicer boots (R.H. side) 6 Elevator and trim CHECK (Condition / Attachments) CHECK (Condition / Deflection free movement / Trim position) 7 Static dischargers CHECK (Condition) 8 Vertical stabilizer deicer boots (Condition) 9 Rudder and trim CHECK (Condition / Attachments) 9 Rudder and trim CHECK (Condition / Trim position) 10 Static dischargers CHECK (Condition / Trim position) 11 Tail cone Condition - CHECK	4. Inapportion door under fundage	,
5 Horizontal stabilizer deicer boots (R.H. side) CHECK (Condition / Attachments) CHECK (Condition / Deflection free movement / Trim position) 7 Static dischargers CHECK (Condition) 8 Vertical stabilizer deicer boots CHECK (Condition) 9 Rudder and trim CHECK (Condition / Attachments) 9 Rudder and trim CHECK (Condition / Trim position) 10 Static dischargers CHECK (Condition / Trim position) 11 Tail cone Condition - CHECK	4 Inspection door under luserage	
(Condition / Attachments) 6 Elevator and trim CHECK (Condition / Deflection free movement / Trim position) 7 Static dischargers CHECK (Condition) 8 Vertical stabilizer deicer boots CHECK (Condition / Attachments) 9 Rudder and trim CHECK (Condition / Trim position) 10 Static dischargers CHECK (Condition / Trim position) 11 Tail cone Condition - CHECK	5 Horizontal stabilizer deicer boots (R.H. side)	,
(Condition / Deflection free movement / Trim position) 7 Static dischargers	, , , , , , , , , , , , , , , , , , , ,	
free movement / Trim position) 7 Static dischargers CHECK (Condition) 8 Vertical stabilizer deicer boots CHECK (Condition / Attachments) 9 Rudder and trim CHECK (Condition / Trim position) 10 Static dischargers CHECK (Condition / Trim position) 11 Tail cone Condition - CHECK	6 Elevator and trim	
7 Static dischargers CHECK (Condition) 8 Vertical stabilizer deicer boots CHECK (Condition / Attachments) 9 Rudder and trim CHECK (Condition / Trim position) 10 Static dischargers CHECK (Condition) 11 Tail cone Condition - CHECK		•
7 Static dischargers (Condition) 8 Vertical stabilizer deicer boots 9 Rudder and trim CHECK (Condition / Attachments) CHECK (Condition / Trim position) 10 Static dischargers CHECK (Condition / Trim position) 11 Tail cone Condition - CHECK		
8 Vertical stabilizer deicer bootsCHECK (Condition / Attachments) 9 Rudder and trimCHECK (Condition / Trim position) 10 Static dischargersCHECK (Condition) 11 Tail coneCondition - CHECK	7 - Static dischargers	•
8 Vertical stabilizer deicer bootsCHECK (Condition / Attachments) 9 Rudder and trimCHECK (Condition / Trim position) 10 Static dischargersCHECK (Condition) 11 Tail coneCondition - CHECK	7. State destraigers	
9 Rudder and trimCHECK (Condition / Trim position) 10 Static dischargersCHECK (Condition) 11 Tail coneCondition - CHECK	8 Vertical stabilizer deicer boots	,
(Condition / Trim position) 10 Static dischargersCHECK (Condition) 11 Tail coneCondition - CHECK		(Condition / Attachments)
10 Static dischargersCHECK (Condition) 11 Tail coneCondition - CHECK	9 Rudder and trim	
(Condition) 11 Tail cone Condition - CHECK		,
11 Tail cone Condition - CHECK	10 Static dischargers	
	11 - Tail cone	,





BEFORE STARTING ENGINE (1/3)

CAUTION

"BLEED" SWITCH SET TO "AUTO" MAY CAUSE
OVERTEMPERATURE OR ABNORMAL ACCELERATION
AT START

CAUTION

MAKE SURE THAT "MAN OVRD" CONTROL IS OFF TO AVOID OVERTEMPERATURE RISKS AT START

1 Preflight inspection	COMPLETED
2 Cabin access door	_CLOSED / LOCKED
3 "Pilot" door (if installed)	_CLOSED / LOCKED
4 Baggage	STOWED
5 Parking brake	SET
6 Weight and balance	_COMPUTED / CHECKED
7 Pilot seat and R.H. front seat (if occupied)	
- Height adjustment	_Maximum UP
- Fore and aft adjustment	_ADJUST and CHECK LOCKING
- Height adjustment	ADJUST

CAUTION

IT IS MANDATORY TO ADJUST SEAT IN FORE--AFT MOVEMENT WHEN SEAT IS IN MAXIMUM HIGH PERMISSIBLE POSITION, TO AVOID INTERFERENCE BETWEEN SIDE UPHOLSTERY PANEL AND SEAT HOUSING IN LOW AND INTERMEDIATE POSITIONS



BEFORE STARTING ENGINE (2/3)		
8 R.H and L.H. pedals		ADJUSTED
9 Belts and harnesses (Pilot and pas	sengers)	FASTENED
10 "NORMAL/MASK" micro inverter		NORMAL
11 Landing gear control		DN
12 "AVIONICS" MASTER switch		START
13 RADIO VHF		ON ADJUSTED
14 "IGNITION" switch		AUTO or OFF
15 "STARTER" switch		_ OFF
16 "SOURCE" selector		_BAT (or GPU)
17 Authorization for engine starting		
18 Pilots "OXYGEN" switch		ON
19 "PASSENGERS OXYGEN" switch		_ OFF
20 Copilot and pilot masks		Press pushbutton
"PRESS TO TEST":	the blinker shall turn	red momentarily,
	then turns transpare	nt
21 Passengers briefing		_ AS REQUIRED
22 CAS display		
- Access door and (if installed) "pilo	t" door	
WARNING CAS MESS	AGE "DOOR"	OFF
- Oxygen emergency system		
WARNING CAS MESS	AGE "OXYGEN"	OFF



BEFORE STARTING ENGINE (3/3)	
23 Fuel	
- Quantity	CHECKED
- Tank selector	L or R - CHECKED
- "FUEL SEL" switch	AUTO
WARNING CAS MESSAGE "AUTO SEL"	OFF
- "SHIFT" pushbutton	PRESS
	The selector changes tank
	On ground, observe a tank change
	every minute and 15 seconds
24 EXT LIGHTS panel	
- "STROBE"	AS REQUIRED
25 In case of night flight	
- INT LIGHTS panel : "INSTR" + "PANEL" $_{\scriptscriptstyle \perp}$	ADJUSTED
- Navigation lights	ON
- Flashlight (if necessary)	IN PLACE



STARTING ENGINE USING AIRPLANE POWER (1/5)

CAUTION				
BEFORE SELECTING SOURCE, CHECK:				
1 "IGNITION" switch	AUTO or OFF			
2 "STARTER" switch	OFF			
3 "INERT SEP" switch	OFF			
4 Landing gear control				
5 ELECTRIC POWER panel				
·	DAT			
- "SOURCE" selector				
- Voltage				
	² 24.5 Volts			
6 Engine controls				
- "MAN OVRD" control	OFF (Notched)			
CAUTION				
WHEN THE ENGINE IS SHUTDOWN, TI	HE POWER LEVER			
MUST NOT BE MOVED BEHIND TH	E FLIGHT IDLE			
POSITION				
- Power lever	IDLE			
	(Flight idle stop)			
- Propeller governor lever	MAX RPM			
- Condition lever	CUT OFF			



STARTING ENGINE USING AIRPLANE POWER (2/5)

	7 11 11 27 11 27 1 27 1 27 1	
7 Flaps		UP
	WARNING	
	IT IS PROHIBITED TO SET FLAPS CONTROL LEVE	R TO
	"850" POSITION ON GROUND AND FOR TAKEO	
	650 POSITION ON GROUND AND FOR TAKEO	ГГ
0 FUEL 201		
8 FUEL par		
- "AUX BP	" switch	ON
	WARNING CAS MESSAGE "AUX BOOST PMP ON"	ON
	WARNING CAS MESSAGE "FUEL PRESS"	OFF
9 Propeller		_AREA CLEAR
	START panel	
	N" switch	AUTO
	R" switch	
- STARTE	:R Switch	ON
	WARNING CAS MESSAGE "STARTER"	ON
	WARNING CAS MESSAGE "IGNITION"	_ ON
	40.04	
Ng <u>∼</u>		
 Condition 	ı lever	L_LO / IDLE
Monitor ind	crease of :	
- ITT	(max. ITT: ±870°C for 20 seconds max.	
	± 1000°C for 5 seconds max.)	
- Ng	o o o nan)	
0	luro.	
- Oil press		
	WARNING CAS MESSAGE "OIL PRESS"	_OFF



STARTING ENGINE USING AIRPLANE POWER (3/5)

CAUTION

IF 10 SECONDS AFTER HAVING POSITIONED CONDITION LEVER TO "LO / IDLE" THERE IS NO IGNITION OR IF DURING IGNITION SEQUENCE, OVERTEMPERATURE INDICATION APPEARS (MAX. ITT ± 870°C FOR MORE THAN 20 SECONDS -- ± 1000°C FOR MORE THAN 5 SECONDS),

INTERRUPT STARTING PROCEDURE:

Condition lever	CUT OFF
"IGNITION" switch	OFF (or AUTO)
Wait ITT < 850°C, then:	
"STARTER" switch	OFF
BEFORE ANY RESTARTING ATTEMPT, CARRY OUT A MOT	TORING
CONTINUE WITH NORMAL PROCEDURE HEREAFTE	R

CAUTION

IF ENGINE IS SLOW TO START OR STAGNATES, INTERRUPT STARTING PROCEDURE:

Condition lever	CUT OFF
"IGNITION" switch	OFF (or AUTO)
"STARTER" switch	OFF

WAIT FOR 1 MINUTE, THEN TRY TO RESTART



STARTING ENGINE USING AIRPLANE POWER (4/5)

ENGINE START panel - "IGNITION" switch - "STARTER" switch WARNING CAS MESSAGE "STARTER" ON WARNING CAS MESSAGE "IGNITION" ON Ng ≃13 %	
- Condition lever	
Monitor increase of : - ITT (max. ITT : ± 870°C for 20 so	
± 1000°C for 5 s	
- Ng - Oil pressure WARNING CAS MESSAGE "OIL PRESS"	OFF
Ng ≈ 50 %	
- "STARTER" switch	
WARNING CAS MESSAGE "STARTER"	
Engine instruments	
NOTE:	
This behaviour should only be observed with outside low tem (OAT < 0°C), cold engine. This procedure may be used for the first starting of the day. CONTINUE WITH NORMAL PROCEDURE HEREAFTER	perature



STARTING ENGINE USING AIRPLANE POWER (5/5)

11 Condition lever	HI / IDLE
12 Engine instruments	CHECK : Ng \simeq 69 % (± 2 %)
	(Oil pressure / Oil temperature
	/ ITT = green sector)
13 FUEL panel	
- "AUX BP" switch	AUTO
WARNING CAS MESSAGE "AUX BOOST PMP ON"	OFF
14 Generator	
WARNING CAS MESSAGE "MAIN GEN"	OFF
	RESET if necessary
- Ammeter	CHARGE CHECKED
- Voltmeter	VOLTAGE CHECKED
	$(V \sim 28 \text{ Volts})$



STARTING ENGINE USING EXTERNAL POWER (GPU) (1/5)

1 GPU			CONNECTED
	CAU	TION	
	BEFORE SELECTING	G SOURCE, CHE	ECK:
2 "IGNITION" switch		_ AUTO or OFF	
3 "STARTE	ER" switch		_ OFF
4 "INERT S	SEP" switch		_ OFF
5 Landing	gear control		_ DN
6 "SOURCE	E" selector		_ GPU
	WARNING CAS MESSAGE	"GPU DOOR"	ON
	WARNING CAS MESSAGE	"BAT OFF"	ON
- Voltmeter			VOLTAGE CHECKED
			(V ≃28 Volts)
7 Engine co	ontrols		
- "MAN OV	'RD" control		_ (Notched)
	CAU	TION	
	WHEN THE ENGINE IS SHUT	DOWN, THE PO	WER LEVER
	MUST NOT BE MOVED B	,	
POSITION			
- Power lev	/er		- IDLE
			(Flight idle stop)
- Propeller governor lever MAX RPM			
 Condition 	lever		_ CUT OFF



STARTING ENGINE USING EXTERNAL POWER (GPU) (2/5)

8 Flaps		UP
	WARNING	
	IT IS PROHIBITED TO SET FLAPS CONTROL L	EVED TO
	"850" POSITION ON GROUND AND FOR TAK	
	030 TOSTION ON GROUND AND FOR TAI	NLOI I
9 FUEL pa	nel	
- "AUX BP	" switch	ON
	WARNING CAS MESSAGE "AUX BOOST PMP O	ON"ON
	WARNING CAS MESSAGE "FUEL PRESS"	OFF
- Fuel pres	ssure indicator	CHECK
	START panel	
- "IGNITIO	N" switch	AUTO
- "STARTE	ER" switch	ON
	WARNING CAS MESSAGE "STARTER"	ON
	WARNING CAS MESSAGE "IGNITION"	ON
Ng ∼ 13 %		
0	n lever	LO / IDLE
Monitor increa		
- ITT	(max. ITT: ±870°C for 20 seconds max.	
	± 1000°C for 5 seconds max.)	
- Ng	,	
- Oil pressu	ure	
•	WARNING CAS MESSAGE "OIL PRESS"	OFF



STARTING ENGINE USING EXTERNAL POWER (GPU) (3/5)

CAUTION

IF 10 SECONDS AFTER HAVING POSITIONED CONDITION LEVER TO "LO / IDLE" THERE IS NO IGNITION OR IF DURING IGNITION SEQUENCE, OVERTEMPERATURE INDICATION APPEARS (MAX. ITT ± 870°C FOR MORE THAN 20 SECONDS - ± 1000°C FOR MORE THAN 5 SECONDS),

INTERRUPT STARTING PROCEDURE:

CAUTION

CONTINUE WITH NORMAL PROCEDURE HEREAFTER

IF ENGINE IS SLOW TO START OR STAGNATES, INTERRUPT STARTING PROCEDURE:

Condition lever_____CUT OFF
"IGNITION" switch _____OFF (or AUTO)
"STARTER" switch _____OFF

WAIT FOR 1 MINUTE (Refer to Chapter 2.4 "STARTER OPERATION LIMITS"), THEN TRY TO RESTART



STARTING ENGINE USING EXTERNAL POWER (GPU) (4/5)

ENGINE START panel				
- "IGNITION"	" switch	_ AUTO		
- "STARTER	"switch	_ ON		
	WARNING CAS MESSAGE "STARTER"			
	WARNING CAS MESSAGE "IGNITION" _	_ ON		
Ng ≃ 13 %				
- Condition le	2VAT	LO / IDL F		
Monitor increas				
- ITT	(max. ITT: ±870°C for 20 seconds max.			
	± 1000°C for 5 seconds max.)			
- Ng				
- Oil pressur	e			
	WARNING CAS MESSAGE "OIL PRESS"	_ OFF		
Ng ≈ 50 %				
- "STARTER	"switch	OFF		
	WARNING CAS MESSAGE "STARTER"	OFF		
	WARNING CAS MESSAGE "IGNITION"			
	WARNING CAS MESSAGE IGNITION	_011		
Engine instrum	ents	CHECK Ng > 52 %		
5		(Oil pressure / ITT		
		= green sector)		
NOTE:				
This behav	This behaviour should only be observed with outside low temperature			
(OAT < 0°C), cold engine.				
This procedure may be used for the first starting of the day.				
CONTINUE WITH NORMAL PROCEDURE HEREAFTER				



STARTING ENGINE USING EXTERNAL POWER (GPU) (5/5)

12 "SOURCE" selector	
13 Propeller governor lever	
14 GPU	
WARNING CAS MESSAGE "GPU DOOR"	OFF
15 Condition lever	HI / IDLE
16 Propeller governor lever	MAX. RPM
17 Engine instruments	CHECK : Ng ≃ 69 % (± 2 %)
	(Oil pressure / Oil temperature
	/ ITT = green sector)
18 FUEL panel	
- "AUX BP" switch	AUTO
WARNING CAS MESSAGE "AUX BOOST PMP	ON" OFF
19 Generator	
13 Generator	
WARNING CAS MESSAGE "MAIN GEN"	OFF
RESET if necessary	
- Ammeter	CHARGE CHECKED
- Voltmeter	VOLTAGE CHECKED
	(V ≃ 28 Volts)



MOTORING (1/2)

CAUTION

AFTER ANY STARTING INTERRUPT PROCEDURE:
- WAIT FOR ENGINE TOTAL SHUT--DOWN
- WAIT AT LEAST 30 SECONDS BEFORE INITIATING A
MOTORING

1 Engine controls			
- "MAN OVRD" contro		 OFF	(Notched

CAUTION

WHEN THE ENGINE IS SHUTDOWN, THE POWER LEVER
MUST NOT BE MOVED BEHIND THE FLIGHT IDLE
POSITION

- Power lever	IDLE
	(Flight idle stop)
- Propeller governor lever	MAX. RPM
- Condition lever	CUT OFF
2 Fuel	
- Tank selector	LorR
- "AUX BP" switch	ON
WARNING CAS MESSAGE "AUX BOOST PMP ON" -	ON

WARNING CAS MESSAGE "FUEL PRESS" OFF



MOTORING (2/2)

3 "IGNITION" switch	OFF
WARNING CAS MESSAGE "IGNITION"	OFF
To clear fuel and vapor internally trapped:	
4 "STARTER" switch	ON
during 15 sec r	maxi
WARNING CAS MESSAGE "STARTER"	ON
To cool engine following shut-down in high temperature environment:	
4 "STARTER" switch	ON
	during 30 sec
WARNING CAS MESSAGE "STARTER"	ON
5 "STARTER" switch	OFF
WARNING CAS MESSAGE "STARTER"	OFF
6 FUEL panel	
- "AUX BP" switch	OFF
WARNING CAS MESSAGE "AUX BOOST PMP ON"	OFF
WARNING CAS MESSAGE "FUEL PRESS"	ON



MOTORING FOLLOWED BY AN ENGINE START (1/2)

Within starter operating limits (continuous max. 1 minute), it is possible to initiate a starting procedure from a motoring procedure.

- 1.- Engine controls
 - "MAN OVRD" control _____OFF (Notched)

CAUTION

WHEN THE ENGINE IS SHUTDOWN, THE POWER LEVER MUST NOT BE MOVED BEHIND THE FLIGHT IDLE **POSITION**

IDLE

- Power lever	_IDLE
	(Flight idle stop)
- Propeller governor lever	MAX. RPM
- Condition lever	_CUT OFF
2 Fuel	
- Tank selector	_L or R
- "AUX BP" switch	ON
WARNING CAS MESSAGE "AUX BOOST PMP ON"	ON
WARNING CAS MESSAGE "FUEL PRESS"	OFF
3 "IGNITION" switch	OFF
4 "STARTER" switch	ON during 15 sec



MOTORING FOLLOWED BY AN ENGINE START (2/2)

5 After 15 seconds :	
- "IGNITION" switch	AUTO
- Ng	$_{-}$ Check at \simeq 13 $\%$
- Condition lever	LO / IDLE
6 Monitor increase of :	
- ITT	_(max. ITT : ± 870°C for 20
	seconds max.
	± 1000°C for 5 seconds max.)
- Ng	
- Oil pressure	
WARNING MESSAGE "OIL PRESS"	OFF
$Ng \simeq 50 \%$ stable	
- "STARTER" switch	OFF
WARNING CAS MESSAGE "STARTER"	OFF
WARNING CAS MESSAGE "IGNITION"	OFF
7 Engine instruments	CHECK : Ng > 52 %
	(Oil pressure / ITT = green sector)
8 Condition lever	HI / IDLE
9 Engine instruments	_ CHECK : Ng ≃ 69 % (± 2 %)
	(Oil pressure / Oil temperature
	/ ITT = green sector)
10 FUEL panel	
- "AUX BP" switch	AUTO
WARNING CAS MESSAGE "AUX BOOST PMP C	N" OFF
11 Generator	
WARNING CAS MESSAGE "MAIN GEN"	OFF
RESET if necessary	
- Ammeter	_CHARGE CHECKED
- Voltmeter	VOLTAGE CHECKED
	(V ≃ 28 Volts)



AFTER STARTING ENGINE (1/2)

1 "GENERATOR" selector	
- On "MAIN"	_ Voltage and current checked
when current ≤ 50 amps :	
- on "STBY"	_ Voltage and current checked
	(reset if necessary)
- then again on "MAIN"	
2 "AVIONICS" MASTER switch	ON
3 "AP TRIMS" MASTER switch	ON
4 Oxygen supply	_ Available for the planned flight
5 PFD 1, MFD and PFD 2	
- Brightness	_ADJUST
	If necessary
- DISPLAY BACKUP button	CHECK
	then return to NORMAL mode
If ammeter < 100 A :	
0 500	
6 ECS panel	
- "BLEED" switch	AUTO
•	
- "BLEED" switch	AUTO
- "BLEED" switch	AUTO AS REQUIRED
- "BLEED" switch - "AIR COND" switch - "CABIN CTRL" selector	AUTO AS REQUIRED ADJUST
- "BLEED" switch - "AIR COND" switch - "CABIN CTRL" selector - "CABIN TEMP/°C" selectors	AUTO AS REQUIRED ADJUST AS REQUIRED
- "BLEED" switch - "AIR COND" switch - "CABIN CTRL" selector - "CABIN TEMP/°C" selectors - "AIR FLOW" distributor	AUTO AS REQUIRED ADJUST AS REQUIRED AS REQUIRED Airfield altitude
- "BLEED" switch - "AIR COND" switch - "CABIN CTRL" selector - "CABIN TEMP/°C" selectors - "AIR FLOW" distributor - Cabin pressure control panel	AUTO AS REQUIRED ADJUST AS REQUIRED Airfield altitude CHECKED
- "BLEED" switch "AIR COND" switch "CABIN CTRL" selector "CABIN TEMP/°C" selectors "AIR FLOW" distributor Cabin pressure control panel	AUTO AS REQUIRED ADJUST AS REQUIRED Airfield altitude CHECKED CHECKED



AFTER STARTING ENGINE (2/2)

10 VHF/VOR/GPS	_ ADJUSTED - TESTED
- Radar/Stormscope/TAS/TAWS/	
Radio altimeter (if installed)	_ADJUSTED - TESTED
11 MFD flight management	
- Weight computing	_ SET/CHECKED
- FOB synchro	SET
- FPL (if requested)	SET
12 AP / TRIMS	
- "AP TRIMS" MASTER operation	CHECK
- Pitch trim	_ UP / DN, then ADJUSTED
- Yaw trim	_ L / R, then ADJUSTED
- Roll trim	_ L / R, then ADJUSTED
13 DE ICE SYSTEM panel	
- "PROP DE ICE" switch	ON
	Check illumination of the green
	light located above the switch
- "PROP DE ICE" switch	OFF
- "WINDSHIELD" switch	ON
	Check illumination of the green
	lights located above the switch
	(except if hot conditions)
- "WINDSHIELD" switch	OFF
	Increase power so as to get Ng ≥
	80% to check AIRFRAMEDE ICE
- "AIRFRAME DE ICE" switch	ON
	Visually check functioning of
	deicer boots during 1 total cycle and
	illumination of the two green lights
	located above the switch
- "AIRFRAME DE ICE" switch	OFF
- "INERT SEP" switch	ON
WARNING CAS MESSAGE "INERT SEP ON"	ON
after 30 seconds	



CHECK-LIST PROCEDURES TAXIING

1 "TAXI" light	
CHECK WARNING CAS MESSAGE	"INERT SEP ON" ON
3 Passenger briefing 4 Parking brake	
WARNING CAS MESSAGE "PARK E	BRAKE"OFF
5 L.H. and R.H. seats brakes 6 Nose wheel steering 7 Power lever	CHECKED
	ITION RSE DURING TAXIING
8 Flight instruments 9 CAS display 10 Cabin pressurization control panel . Cruis	CHECK



BEFORE TAKEOFF (1/2)

1 Parking brake	_SET
WARNING CAS MESSAGE "PARK BRAKE"	ON
2 Condition lever	_HI / IDLE
	[Ng : 69 % (± 2 %)]
3 Propeller governor lever	_FEATHER twice,
	then MAX. RPM
4 Flaps	_TO
5 DE ICE SYSTEM panel	
- "AIRFRAME DE ICE" switch	_As required
- "PROP DE ICE" switch	_As required
If runway is in good condition, without icing conditions :	
- "INERT SEP" switch	OFF
WARNING CAS MESSAGE "INERT SEP ON	"OFF
If there is standing water or other contamination on the	runway:
- "INERT SEP" switch	_Leave ON
WARNING CAS MESSAGE "INERT SEP ON"	ON
- "WINDSHIELD" switch	_As required
- "PITOT L HTR" switch	
- "PITOT R & STALL HTR" switch	ON
6 Flight controls	_DEFLECTIONS CHECKED
7 Trims	
- Pitch	_ADJUSTED
- Yaw	_ADJUSTED
- Roll	_ADJUSTED



BEFORE TAKEOFF (2/2)

- Passengers's table		REMOVED
9 "STROBE" switch		ON
10CAS display		CHECK
		All messages OFF,
	except "PARK BRAKE"	ON
	and, if used "INERT SEP ON"	ON
11 Fuel		
- Gages : quantity, symmetry		CHECKED
- "FUEL SEL" switch		_CHECK AUTO
- "AUX BP" fuel switch		CHECK AUTO
12 Flight instruments		CHECKED
- Altimeter setting		_ADJUSTED/CHECKED
- "ALT SEL"		ADJUSTED/CHECKED
13 VHF/VOR/GPS/XPDR		ADJUSTED/CHECKED
- Radar/Stormscope/TAS/TA\	WS/ADF	
(if installed)		_ADJUSTED/CHECKED
14 Engine instruments		-CHECK
15 Battery charge		_< 50 Amperes
	CAUTION	
DO NOT TAKE	OFF IF BATTERY CHARGE >	50 Amperes
16 Darking broke		DEL EASED
10 raikiliy blake		-KELEASED

WARNING CAS MESSAGE "PARK BRAKE"____OFF



TAKEOFF (1/2)

WHEN LINED UP

CAUTION - IF HEAVY PRECIPITATION, TURN IGNITION AND INERT SEP ON.

1 Horizon	CHECK altitude \simeq + 2°
2 Heading - HSI - Stand-by compass	CHECK
- Altimeter setting	CHECK
3 Lights	
- "L.LDG / TAXI / R.LDG"	ON
4 Engine instruments	CHECK
	(ITT = green sector)
5 CAS display	CHECK
	All messages OFF,
except "INERT SEP ON"	if used
except "IGNITION"	if used
6 PROP O' SPEED GOVERNOR TEST	
- Increase power until propeller RPM reaches 1900 RPM	
- PROP O' SPEED TEST : Maintain enga	aged
- Observe that propeller RPM decreases by 50 to 250 RPI	M
- PROP O' SPEED	elease
- Check that propeller RPM increases by a minimum of 50	RPM
when compared tominimumvalue during PROP O'SPEEI	D test.



TAKEOFF (2/2)

7 Brakes	RELEASED
8 Power lever	TRQ = 100 %
9 Takeoff	
- Normal takeoff	ATTITUDE: 7°5
- Short takeoff	
. Weight < 6579 lbs (2984 kg)	ATTITUDE: 15°
. Weight ≥ 6579 lbs (2984 kg)	
10 Vertical speed indicator	POSITIVE
11 Brakes	
	(Briefly)
12 Landing gear control (IAS < 128 KIAS)	UP
	At sequence end, check :
	All warning lights OFF
13 Initial climb speed Weight < 6579 lbs (2984 kg)	: 110 KIAS
Weight ≥ 6579 lbs (2984 kg)	:115 KIAS
14 Flaps	UP
Only when flaps are confirmed UP:	
15 Flap control	850
16 Power lever	TRQ =121.4 %
17 Climb speed (recommended)	130 KIAS
- Trims (Pitch, Roll and Yaw) ADJUSTED	
18 "YAW DAMPER" push-button	ON
19 Lights	
- "TAXI"	OFF
- "LLDG / RLDG"	AS REQUIRED



CLIMB

1.- Power lever ______ADJUST

CAUTION OBSERVE TRQ / Ng / Np / ITT / T° AND OIL PRESSURE LIMITATIONS.

2 Climb speed	_ AS REQUIRED
3 ECS panel	
- Cabin pressure control panel	_ Cruise altitude + 1000 feet
- Pressurization	CHECK
- "CABIN TEMP/°C" selectors	ADJUST
4 Fuel tank gages	_ CHECK / CORRECT
	(Quantity / Symmetry)
5 DE ICE SYSTEM	_ As required

CAUTION

IF HEAVY PRECIPITATION, TURN IGNITION

AND INERT SEP ON



CRUISE

1 Power lever	ADJUST
	CAUTION
	OBSERVE TRQ / Ng / Np / ITT / T°
	AND OIL PRESSURE LIMITATIONS.
2 Pressurization	CHECK
3 Fuel	
- Gages	CHECK
REGULARLY CHE	CK:

- consumption
- expected fuel at destination
- tank automatic change (every 10 minutes)
- symmetry [max. dissymmetry 15 us gal (57 Litres)]
- 4.- Cruise parameters / engine data _____CHECK/RECORD
- 5.- DE ICE SYSTEM _____As required

CAUTION

IF HEAVY PRECIPITATION, TURN IGNITION

AND INERT SEP ON



FLAP CONTROL TRANSITION FROM "UP" TO "850"	
1 Flaps	CHECKED UP
2 Propeller RPM	2000
3 Power lever	TRQ ± 100 %
4 Flap control lever	From UP to 850
5 Power lever	As required
	TRQ less than 121.4 %

FLAP CONTROL TRANSITION FROM "850" TO "UP"

ı		
	1 Altitude	At or above 1500 ft AGL
	2 Propeller RPM	2000
	3 Power lever	TRQ ± 100 %
	4 Flap control lever	From 850 to UP
	5 Power lever	As required
		TRQ less than 100 %
		(2000 RPM)



DESCENT

1 Flaps	_UP
2 Altimeter settings	COMPLETE
3 "ALT SEL"	SELECTED
4 ECS panel	
- Cabin pressure control panel	_Airfield altitude
5 DE ICE SYSTEM	_As required

CAUTION IF HEAVY PRECIPITATION, TURN IGNITION AND INERT SEP ON

CAUTION

USE OF CONTROL REVERSE BETA (β) RANGE (BEHIND THE FLIGHT IDLE POSITION) IS PROHIBITED DURING FLIGHT

6 Windshield misting protection system	As required
7 Fuel	
- Gages	CHECK
	(Quantity / Symmetry)
- Fullest tank	SELECT
8 Passengers briefing	As required
9 Seats, belts and harnesses	LOCKED
0 - Passengers's table	REMOVED



BEFORE LANDING

Long final		
1 Altimeters		CHECK
2 Fuel		
- Gages		CHECK
		(Quantity / Symmetry)
- Fullest tank		SELECT
3 "INERT SEP" switch	_(IAS \le 200 KIAS)	ON
4 Propeller lever		MAX RPM
5 Landing gear control	_(IAS ≤ 178 KIAS)	_ DN
- Green indicator lights		ON
- Red warning light		OFF
6 Flaps	_(IAS ≤ 178 KIAS)	ТО
7 Lights		
- "L.LDG / TAXI / R.LDG"		ON
Short final		
8 Autopilot		DISCONNECT
9 Flaps	(IAS ≤ 122 KIAS)	LDG
10 Approach speed		
(Flaps LDG)	Weight < 6250 lbs (2835 kg):	80 KIAS
	Weight ≥ 6250 lbs (2835 kg) :	85 KIAS
(Flaps LDG)	With AP engaged : ≥	85 KIAS
11 "YAW DAMPER" push-but	ton	OFF



LANDING

1 Power lev	erIDLE	
After wheels	touch	
	As required	
To avoid	may be applied as soon as the wheels touch the ground.) ingestion of foreign objects, come out the reverse as speed rakes if necessary for further deceleration.	reduces and
	CAUTION	
	ON SNOWY OR DIRTY RUNWAY, IT IS BETTER NOT TO	
	USE REVERSE BELOW 40 KIAS	
3 Brakes	As required	



GO-AROUND

1 GO AROUND pushbutton	PUSHED
2 Simultaneously	
- Power lever	TRQ = 100 %
- Attitude	_7°5
3 Flaps	
Weight below 6579 lbs (2984 kg)	
If the vertical speed is positive and if IAS is at or above 85 KIAS	:
4 Landing gear control	UP
All warning lights OFF	
If IAS is at or above 110 KIAS :	
5 Flaps	UP
6 Climb speed	
Weight above 6579 lbs (2984 kg)	
If the vertical speed is positive and if IAS is at or above 90 KIAS	
7 Landing gear control	
	All warning lights OFF
If IAS is at or above 115 KIAS :	
8 Flaps	UP
9 Climb speed	
	AS REQUIRED



TOUCH AND GO

After wheels touch

1 Flaps	TO
2 Elevator trim	Green sector
3 Power lever	Display TRQ = 100 %
4 Takeoff	ROTATION
- Normal takeoff	ATTITUDE : 7°5
- Short takeoff	
. Weight < 6579 lbs (2984 kg)	ATTITUDE : 15°
. Weight ≥ 6579 lbs (2984 kg)	ATTITUDE : 12°5



AFTER LANDING

RUNWAY CLEAR -- AIRPLANE STOPPED 1.- DE ICE SYSTEM panel - "AIRFRAME DE ICE" switch _____ OFF - "PROP DE ICE" switch _____OFF - "INERT SEP" switch CHECKED ON - "WINDSHIELD" switch _____ As required - "PITOT L HTR" switch _____ OFF - "PITOT R & STALL HTR" switch _____ OFF - "BLEED" switch ______ As required 2.- Radar (if installed) CHECKED STANDBY 3.- Transponder_____CHECKED SBY 4.- Flaps______**UP** 5.- "STROBE" switch _____ OFF 6.- Lights - "L.LDG / R.LDG" _____ **OFF** - "TAXI" _____ON 7.- "OXYGEN" switch _____ OFF



SHUT-DOWN (1/2)

1 P	arking brake	SET
	WARNING CAS MESSAGE "PARK BRAKE"	ON
2 C	ondition lever	CHECK HI /IDLE
3 P	ower lever	_ IDLE for 1 minute minimum
4 "T	AXI" light	OFF
5 "A	AP TRIMS" MASTER switch	OFF
6 "A	AVIONICS" MASTER switch	START
7 E	CS panel	
- "B	LEED" switch	OFF
- Ch	neck for cabin depressurization	
- "A	IR COND" switch	OFF
	CAUTION	
	IN CASE OF SHUT-DOWN ON A CONTAIN	MINATED
	AREA:	
	- Condition lever	CUT OFF
	- Propeller governor lever	FEATHER
8 P	ropeller governor lever	FEATHER for 15 seconds
9 C	ondition lever	_ CUT OFF
10 "II	NERT SEP" switch	OFF
11 "A	VIONICS" MASTER switch	OFF
12 E	XT LIGHTS panel	
- Al	I switches	OFF



SHUT-DOWN (2/2)

13 INT LIGHTS panel	
- All switches	OFF
14 Fuel	
- "AUX BP" switch	OFF
- "FUEL SEL" switch	MAN
- Tank selector	OFF
15 "GENERATOR" selector	_ OFF
16 "SOURCE" selector	_ OFF
17 CRASH lever	_ PUSHED DOWN
18 Parking brake	As required

CAUTION

IN CASE OF HIGH OAT [ABOVE 35°C (95°F)], IT IS RECOMMENDED TO PERFORM 30 SECONDS DRY MOTORING RUN AFTER SHUT--DOWN TO IMPROVE COOLING OF THE BEARING CAVITIES AND PREVENT OIL COKING (REFER TO PARAGRAPH "MOTORING")