16.3 (a) The electron in an elliptic orbit has a regative energy, and the bending energy is the negative of that. Therefore, the secular change in energy is given by

$$\frac{dE}{dt} = \left\langle -\frac{dE}{dt} \right\rangle = \frac{\tau}{m} \left\langle \left( \frac{dV}{dr} \right)^2 \right\rangle = \frac{\tau}{m} \cdot \frac{1}{7} \int_0^7 \left( \frac{dV}{dr} \right)^2 dt,$$

Where  $T = \pi Ze^{\gamma} \int_{2E^{\gamma}}^{m}$  is the period of the orbit. (See Landau & Lifshieg, Methanic, (15.5)). The time integral over one period can be replaced by an integral in angele,  $\int_{0}^{7} dt \rightarrow \int_{0}^{2\pi} d\theta \frac{mr^{\gamma}}{L}$ . Then,  $\frac{dE}{dt} = \frac{T}{M} \cdot \frac{1}{T} \int_{0}^{2\pi} \frac{mr^{\gamma}}{L} \frac{Z^{2}e^{4}}{r^{4}} d\theta = \frac{2e^{\gamma}}{2m^{\gamma}} \frac{Z^{1/2}}{r^{2}} \frac{Z^{1/2}}{L} \frac{Z^{2}e^{4}}{\pi^{2}} \int_{0}^{2\pi} \frac{d\theta}{r^{\gamma}} d\theta$ 

$$= \frac{2e^{2}}{3m^{2}c^{3}} \frac{2''^{2} \varepsilon^{3k}}{m'^{2}} \frac{1}{\pi ze^{2}} \frac{m}{L} z^{2}e^{4} \frac{z^{2}e^{4}m^{2}}{L^{4}} \left[ 2\pi + \pi \left( 1 - \frac{z \varepsilon L^{2}}{z^{2}e^{4}m} \right) \right]$$

$$= \frac{z^{3k}}{3} \frac{z^{3}e^{8}m'^{k}}{C^{3}} \frac{\varepsilon^{3/2}}{C^{3}} \left( 3 - \frac{z \varepsilon L^{2}}{z^{2}e^{4}m} \right)$$

Similarly,  $\frac{dL}{dt} = -\frac{7}{m} \cdot \frac{1}{7} \int_{0}^{7} \frac{1}{r} \frac{dV}{dr} dt \cdot L$   $= -\frac{2e^{2}}{3m^{2}c^{3}} \frac{2^{1/2}}{m^{1/2}} \frac{2^{1/2}}{\pi^{2/2}} \frac{m}{\pi^{2/2}} \cdot \frac{2e^{2}}{L} \cdot \frac{1}{2e^{2}} \cdot \frac{dV}{r}$   $= -\frac{2e^{2}}{3m^{2}c^{3}} \frac{2^{1/2}}{m^{1/2}} \frac{2^{1/2}}{\pi^{2/2}} \frac{e^{3/2}}{L^{2}} \frac{1}{\pi^{2/2}} \frac{2e^{4}}{L^{2}} \cdot \frac{e^{3/2}}{L^{2}}$   $= -\frac{2^{5/2}}{3} \frac{2e^{4}}{\pi^{2/2}} \frac{e^{3/2}}{L^{2}}$ 

(b) From part (a), we know

$$\frac{d\xi}{dL} = -\frac{1}{2} \frac{Z^2 e^4 m}{L^3} \left( 3 - \frac{2 \xi L^2}{Z^2 e^4 m} \right) = -\frac{3}{2} \frac{Z^2 e^4 m}{L^3} + \frac{\xi}{L}$$

Then 
$$\frac{d}{dL}\left(\frac{\varepsilon}{L}\right) = \frac{1}{L}\frac{d\varepsilon}{dL} - \frac{\varepsilon}{L^2} = \frac{1}{L}\left(\frac{d\varepsilon}{dL} - \frac{\varepsilon}{L}\right) = -\frac{3}{2}\frac{2^2e^4m}{L^4}$$

Integrating both sides w.r.t. +0 L, we will arrive at

$$\frac{\mathcal{E}(L)}{L} - \frac{\mathcal{E}(L_0)}{L_0} = \frac{\mathcal{Z}^2 e^4 m}{l} \left( \frac{1}{L^3} - \frac{1}{L_0^3} \right), \text{ or } \mathcal{E}(L) = \frac{\mathcal{Z}^2 e^4 m}{2L^2} \left[ 1 - \left( \frac{L}{L_0} \right)^3 \right] + \frac{\mathcal{E}_0}{L_0} L,$$
where  $\mathcal{E}(L_0) = \mathcal{E}_0$ .

The eccentricity of the elliptic orbit is given by, for some argular momentum,  $e = \left(1 - \frac{2EL^2}{Z^2e^4m}\right)^{1/2}.$  Then, the charge in argular momentum leads to  $e(L) = \left(1 - \left(\frac{L}{L_0}\right)^3\right) - \frac{2L^2}{Z^2e^4m} \frac{E_0}{L_0} L\right)^{1/2}$   $= \left(\left(\frac{L}{L_0}\right)^3 - \frac{2E.L_0^3}{Z^2e^4m} \left(\frac{L}{L_0}\right)^3\right)^{1/2} = \left(\frac{L}{L_0}\right)^{3/2} \left[1 - \frac{2E.L_0^3}{Z^2e^4m}\right]^{1/2}$   $= e(L_0) \left(\frac{L}{L_0}\right)^{3/2},$ 

Where the eccentricity documents as  $(L/k_0)^{3h}$ . Since circular bibts has an eccentricity of v, as time passes by, with decreasing eccentricity, the orbit win become more circular. This can also be understood as the effect of increasing birding energy. For attractive potential, the effective potential  $\left(-\frac{2e^{2}}{r} + \frac{L^{2}}{r^{2}}\right)$  has a minimum, which corresponds to a manimum triving energy and a circular probit.