

Performance Enhancements for Zero-Flow Simulation of Vapor Compression Cycles

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Motivation & Objective

Motivation:

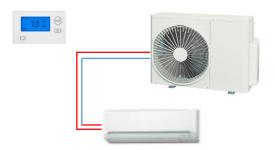
- Numerical simulations are widely employed in the HVAC&R industries
- Low and zero-flow phenomena are often encountered in the operation of vapor compression systems with large transients
- Simulation of system dynamics under low and zeroflow conditions presents numerical challenges
- Reported approaches cannot produce satisfactory results sometimes

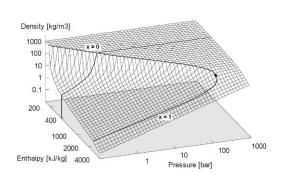
Objective:

 Explore effective techniques to improve the performance of zero-flow simulations, especially focusing on robustness and improvements in the simulation speed, with a goal of achieving faster than real-time dynamic simulation











Vapor Compression Cycle Model

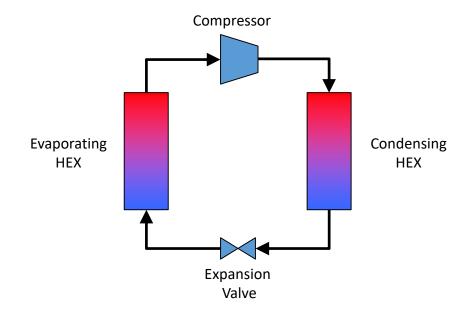
- System dynamics are dominated by heat exchangers
- Refrigerant properties defined via **Equation of State**
 - Pressure and specific enthalpy
- Compressor and valve performance curves defined by the user

Dynamic HEX model

$$\frac{\partial(\rho A)}{\partial t} + \frac{\partial(\rho A v)}{\partial x} = 0$$

$$\frac{\partial(\rho v A)}{\partial t} + \frac{\partial(\rho v^2 A)}{\partial x} = -A \frac{\partial p}{\partial x} - F_f$$

$$\frac{\partial(\rho u A)}{\partial t} + \frac{\partial(\rho v h A)}{\partial x} = vA \frac{\partial p}{\partial x} + vF_f + \frac{\partial Q}{\partial x}$$



Static compressor model

$$\eta_v(f, P_r, \dots) = \frac{\dot{m}_{comp}}{\rho_{suc}Vf}$$

$$\eta_{is}(f, \dots) = \frac{h_{dis,isen} - h_{suc}}{h_{dis} - h_{suc}}$$

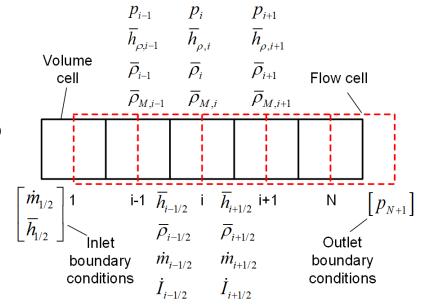
Static valve model

$$\dot{m} = C_v a_v \sqrt{\rho_{in} \Delta P}$$



Staggered Grid Scheme

- Use 1D separated flow model w/o interfacial exchange
- Shift the momentum balance half a cell compared to the mass- and energy balances to obtain a staggered grid
- The staggering tends to decouple the fast pressure from the slow thermal part of the equation system and thereby creates a more robust discretization scheme
- Use the upwind scheme to approximate interface variables



$$\frac{dM_{i}}{dt} = \dot{m}_{i-1/2} - \dot{m}_{i+1/2}$$

$$\frac{dU_{i}}{dt} = \dot{m}_{i-1/2} \bar{h}_{i-1/2} - \dot{m}_{i+1/2} \bar{h}_{i+1/2} + P\Delta z q_{i}''$$

Ignored in our models

Momentum balance

$$\Delta z \frac{d\dot{m}_{i+1/2}}{dt} = \left(\dot{I}_{i} - \dot{I}_{i+1}\right) - A\left(p_{i+1} - p_{i}\right) - \overline{\tau}_{w,i+1/2} P \Delta z - \left[\overline{\rho}_{i+1/2} g A \Delta z \sin \theta\right]$$



Momentum Balance Approximations

Uniform dp/dt:

2N+1 dynamic states

$$\left| \frac{dp_1}{dt} = \dots = \frac{dp_i}{dt} = \dots = \frac{dp_{N+1}}{dt} \right|$$

Linear Δp :

N+3 dynamic states

$$\frac{dp_{i}}{dt} = \left(1 - \frac{i - 1}{N}\right) \frac{dp_{1}}{dt} + \frac{i - 1}{N} \frac{dp_{N+1}}{dt}$$

$$N\Delta z \frac{d\overline{m}}{dt} = \left(\dot{I}_{1} - \dot{I}_{N+1}\right) - A\left(p_{N+1} - p_{1}\right) - P\Delta z \sum_{i=1}^{N} \overline{\tau}_{w,i+1/2}$$

Friction only:

2N dynamic states

$$p_{i+1} = p_i - \frac{P}{A} \overline{\tau}_{w,i+1/2} \Delta z$$



Pressure drop calculation

$$\frac{\Delta p}{\Delta p_0} = K \left(\frac{\dot{m}}{\dot{m}_0}\right)^{\lambda}$$

$$\Rightarrow \dot{m} = \dot{m}_0 \left(\frac{\Delta p}{K \Delta p_0}\right)^{1/\lambda}$$

$$\frac{d\dot{m}}{d(\Delta p)} = \frac{\dot{m}_0^{\lambda}}{\lambda K \Delta p_0} \frac{1}{\dot{m}^{\lambda-1}} \qquad \frac{\dot{m} \to 0}{\Delta m}$$

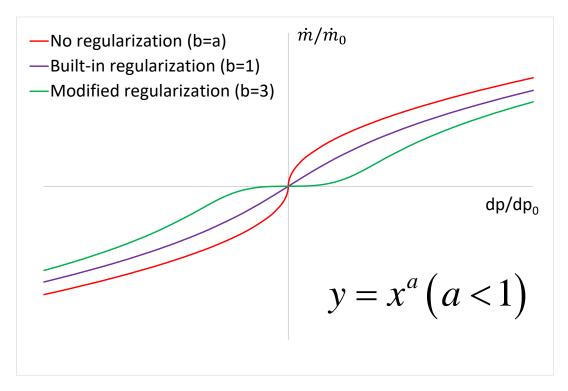
$$\stackrel{\dot{m} \to 0}{\longrightarrow}$$
 Infinity

 $\lambda > 1$

We need to regularize the relation with locally non-singular substitute



Pressure Drop Regularization

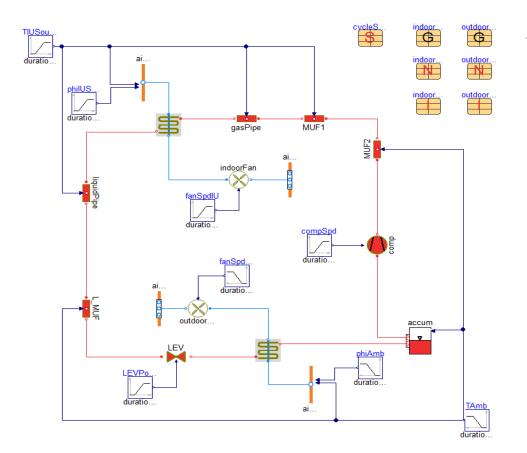


```
function reqPowGen
  extends Modelica. Icons. Function;
  input Real x;
  input Real a;
  input Real delta=0.01;
  input Real b=1;
  output Real y;
algorithm
  y := x^b*(x^*x+delta^*delta)^((a-b)/2);
end reqPowGen;
```

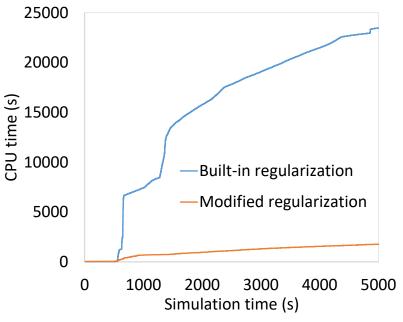
- Original relation without regularization is not Lipschitz continuous
 - Smaller Lipschitz constants around origin are very helpful



Case Study



The DAE has 14626 scalar unknowns and 14626 scalar equations.



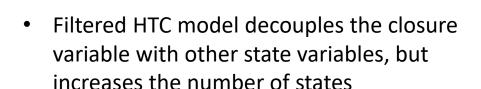
- System was shut down at 500 sec and off-cycle lasted for 4500 sec
- CPU time: 23000 sec with built-in regularization vs. 1700 sec with modified regularization



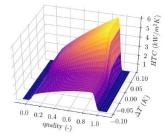
Heat Transfer Model: Static or Filtered?

- Multiple regimes are blended together to form a universal function
- Simplified closure relations can approximate full correlations

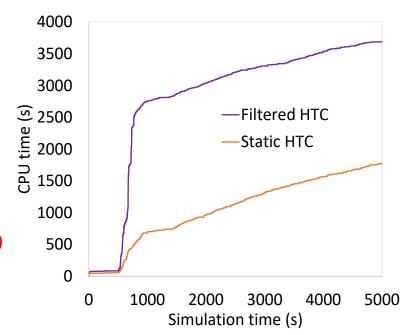
Static HTC
$$\alpha = K\alpha_0 \left(\frac{\dot{m}}{\dot{m}_0}\right)^c$$
 Filtered HTC
$$\frac{d\alpha}{d\alpha} = \frac{1}{1}(\hat{\alpha} - \alpha)$$



- Turning on 'log norm' showed filtered HTCs limited integration time steps
- CPU time: 3700 sec with filtered HTCs vs. 1700 sec with static HTCs



```
6 197 1137 indoorCoil.circuits[1].heatTransfer.alphas[21] (#118)
17 357 1373 indoorCoil.circuits[1].heatTransfer.alphas[22] (#119)
28 315 1666 indoorCoil.circuits[1].heatTransfer.alphas[23] (#120)
32 501 2022 indoorCoil.circuits[1].heatTransfer.alphas[24] (#121)
51 544 2138 indoorCoil.circuits[1].heatTransfer.alphas[25] (#122)
32 333 1674 indoorCoil.circuits[1].heatTransfer.alphas[26] (#123)
19 193 1146 indoorCoil.circuits[1].heatTransfer.alphas[27] (#124)
0 0 2 indoorCoil.header[1].mediums[1].p (#152)
0 0 4 indoorCoil.header[1].mediums[2].p (#154)
13 21 120 indoorCoil.header[2].mediums[1].p (#156)
```

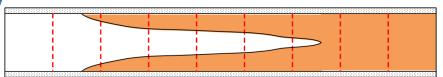




Discretization Methods

Continuity
$$\frac{\partial(\rho A)}{\partial t} + \frac{\partial(\rho A v)}{\partial z} = 0$$
Momentum
$$\frac{\partial(\rho v A)}{\partial t} + \frac{\partial(\rho v^2 A)}{\partial z} = -A \frac{\partial p}{\partial z} - F_f$$
Energy
$$\frac{\partial(\rho u A)}{\partial t} + \frac{\partial(\rho v h A)}{\partial z} = vA \frac{\partial p}{\partial z} + vF_f + \frac{\partial Q}{\partial z}$$

Finite Volume Method



Staggered grid scheme with upwind difference

$$\begin{split} A\Delta z \, \frac{d\,\rho_i}{dt} &= \dot{m}_{i-1/2} - \dot{m}_{i+1/2} \\ \Delta z \, \frac{d\dot{m}_{i+1/2}}{dt} &= -\Delta \dot{I}_i - A\Delta p_i - \overline{\tau}_{w,i+1/2} P\Delta z \\ A\Delta z \, \frac{d\left(\rho_i u_i\right)}{dt} &= \dot{m}_{i-1/2} h_{i-1/2} - \dot{m}_{i+1/2} h_{i+1/2} + P\Delta z q_i'' \end{split}$$

Moving Boundary Method

Vapor Two-phase Liquid

Reynolds transport theorem

$$\frac{d}{dt} \int_{0}^{L} \frac{\partial (\rho A)}{\partial t} dz + \frac{d}{dt} \int_{0}^{L} \frac{\partial \dot{m}}{\partial z} dz = 0$$

$$\frac{d}{dt} \int_{0}^{L} \frac{\partial [A(\rho h - p)]}{\partial t} + \frac{d}{dt} \int_{0}^{L} \frac{\partial (\dot{m}h)}{\partial z} dz = q'$$

Lumped Parameter Method

Single control volume with mean properties

$$AL\frac{d\,\overline{\rho}}{dt} = \dot{m}_{in} - \dot{m}_{out}$$

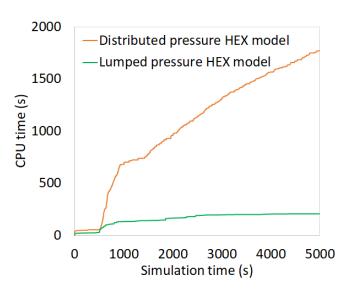
$$AL\frac{d(\overline{\rho h} - p)}{dt} = \dot{m}_{in}h_{in} - \dot{m}_{out}h_{out} + PLq''$$

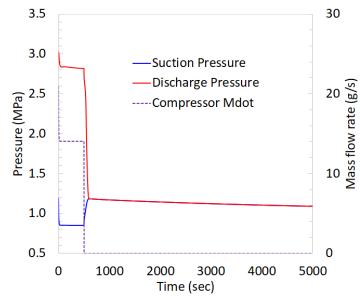


Single-Pressure HEX Model

$$\begin{split} &A\Delta z \Bigg(\frac{\partial \overline{\rho}_{i}}{\partial p}\frac{dp}{dt} + \frac{\partial \overline{\rho}_{i}}{\partial \overline{h}_{\rho,i}}\frac{d\overline{h}_{\rho,i}}{dt}\Bigg) = \dot{m}_{i-1/2} - \dot{m}_{i+1/2} \\ &A\Delta z \Bigg(\overline{\rho}_{i}\frac{d\overline{h}_{\rho,i}}{dt} - \frac{dp}{dt}\Bigg) = \\ &\dot{m}_{i-1/2} \left(\overline{h}_{i-1/2} - \overline{h}_{\rho,i}\right) \overline{h}_{i-1/2} - \dot{m}_{i+1/2} \left(\overline{h}_{i+1/2} - \overline{h}_{\rho,i}\right) + P\Delta z q_{i}'' \end{split}$$

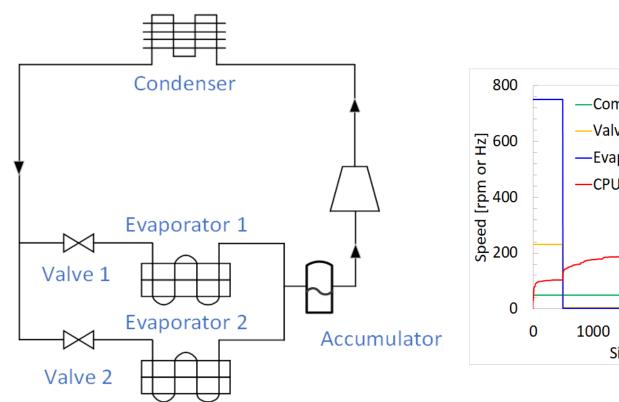
- Dependence of mass flow rates upon pressure difference can be removed
- N+1 states and pressure drop is lumped together and calculated at the inlet or outlet depending on model structure
- CPU time: 1700 sec with distributed HEX model vs. 200 sec with lumped pressure drop model
- 100x speedup with all the enhancements

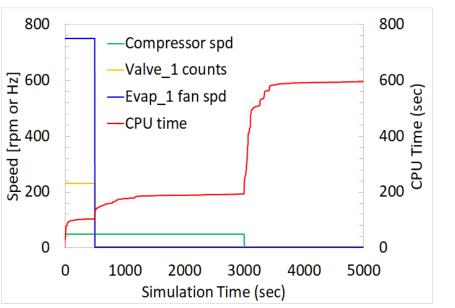






Another Case Study





- System operated for the first 500 sec with two active evaporator branches
- The first branch was off (the associated fan and valve were closed) since 500 sec
- Entire system was turned off at 3000 sec
- 600 sec CPU time with all the enhancements



Conclusions & Future Work

- Reducing the sensitivity of mass flow to pressure differences is a key to accelerating zero-flow simulation
- Static heat transfer model seems more efficient, if no spurious oscillations appear in the simulation
- Lumping pressure drops at the inlet or outlet of HEXs or pipes is helpful for further speeding up simulation
- We would like to evaluate these enhancements with models that can generate Analytical Jacobians

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