

Last Time

- DDP vs. DIRCC_h
- Quaternions

This Time:

- Optimization with Quaternions

Quaternion Recap

- 4D unit vectors

- Multiplication rule:

$$q_1 * q_2 = \begin{bmatrix} s_1 \\ v_1 \end{bmatrix} * \begin{bmatrix} s_2 \\ v_2 \end{bmatrix} = \begin{bmatrix} s_1 s_2 - v_1^T v_2 \\ s_1 v_2 + s_2 v_1 + v_1 \times v_2 \end{bmatrix}$$

$$L(q) = \begin{bmatrix} s_1 & v_1^T \\ v_1 & s_1 I + \tilde{v}_1 \end{bmatrix} \Rightarrow q_1 * q_2 = L(q_1) q_2$$

- Quaternion Conjugate

$$q^+ = \begin{bmatrix} s \\ -v \end{bmatrix} = Tq, \quad T = \begin{bmatrix} I & 0 \\ 0 & -I \end{bmatrix}$$

- Identity:

$$q^{\pm} = \begin{bmatrix} 1 \\ 0 \end{bmatrix}$$

- "Hat map" for Quaternions

$$\hat{\omega} = \begin{bmatrix} 0 \\ \omega \end{bmatrix} = H\omega, \quad H = \begin{bmatrix} 0 \\ I \end{bmatrix}$$

* Geometry of Quaternions

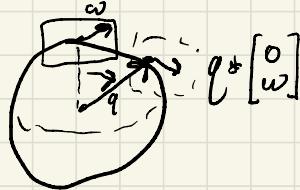


q lives on a sphere in \mathbb{R}^4

\dot{q} lives in the tangent plane to the sphere at q

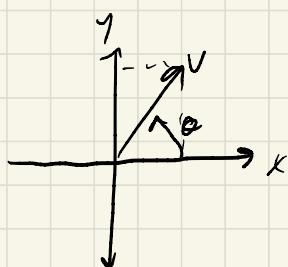
- Kinematics

$$\dot{q} \in \mathbb{R}^4, \quad \omega \in \mathbb{R}^3, \quad \dot{q} = \frac{1}{2} q * \begin{bmatrix} 0 \\ \omega \end{bmatrix} = \frac{1}{2} L(q) H \omega$$



ω is always written down in the tangent plane at the identity then kinematics rotate to the tangent plane at q

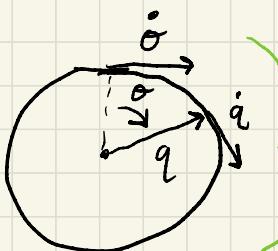
Analogy with unit complex numbers in 2D:



$$q = \underbrace{\cos(\theta)}_{x\text{-component}} + \underbrace{i \sin(\theta)}_{y\text{-component}} \Rightarrow q = \begin{bmatrix} \cos(\theta) \\ \sin(\theta) \end{bmatrix}$$

$$\dot{q} = \frac{\partial q}{\partial \theta} \dot{\theta} = \begin{bmatrix} -\sin(\theta) \\ \cos(\theta) \end{bmatrix} \dot{\theta} = \begin{bmatrix} \cos(\theta) & -\sin(\theta) \\ \sin(\theta) & \cos(\theta) \end{bmatrix} \begin{bmatrix} 0 \\ \dot{\theta} \end{bmatrix}$$

$\underbrace{\begin{bmatrix} \cos(\theta) & -\sin(\theta) \\ \sin(\theta) & \cos(\theta) \end{bmatrix}}_{\text{rotation matrix}}$



2D version of hat map

Kinematics rotates $\dot{\theta}$ from tangent plane at "north pole" to tangent plane at q

* Differentiating Quaternions

- Two key facts:

- 1) Derivatives are really 3D "tangent vectors"
- 2) Rotations compose by multiplication, not addition

- Infinitesimal Rotation

$$S\phi = \begin{bmatrix} \cos(\theta/2) \\ \alpha \sin(\theta/2) \end{bmatrix} \approx \begin{bmatrix} 1 \\ \frac{1}{2}\alpha\theta \end{bmatrix} \approx \begin{bmatrix} 1 \\ \frac{1}{2}\phi \end{bmatrix} \approx \begin{bmatrix} 1 \\ 0 \end{bmatrix} + \frac{1}{2} \begin{bmatrix} 0 \\ \phi \end{bmatrix}$$

axis-angle vector

- Compose with q :

$$\tilde{q} = qS\phi = L(q) \left(\begin{bmatrix} 1 \\ 0 \end{bmatrix} + \frac{1}{2} \begin{bmatrix} 0 \\ \phi \end{bmatrix} \right) = q \cdot \underbrace{\frac{1}{2} L(q) H \phi}_{G(q) \in \mathbb{R}^{3 \times 3}}$$

"attitude Jacobian"

- Note: we can use any 3-parameter rotation representation for ϕ . They all linearize the same up to a constant:

$$q = \begin{bmatrix} \cos(\|\phi\|/2) \\ \frac{\phi}{\|\phi\|} \sin(\|\phi\|/2) \end{bmatrix}, \quad q = \begin{bmatrix} \sqrt{1-\phi^T\phi} \\ \phi \end{bmatrix}, \quad \frac{1}{\sqrt{1+\phi^T\phi}} \begin{bmatrix} 1 \\ \phi \end{bmatrix}$$

axis-angle vector part of q Gibbs/Rodrigues Vector

- We'll use the vector part of q since it is simplest/cheapest.

- This lets us differentiate any quaternion function by simply inserting $G(q)$ in the right places:

$f(q) : \mathbb{H} \rightarrow \mathbb{R}$ (gradient of scalar-valued function)
↑
quaternions
("Hamilton")

$$\nabla f = \frac{\partial f}{\partial q} \frac{\partial q}{\partial \phi} = \frac{\partial f}{\partial q} G(q)$$

$f(q) : \mathbb{H} \rightarrow \mathbb{H}$ (Jacobian of quaternion-valued function)

$$\phi' = [G(f(q))^T \frac{\partial f}{\partial q} G(q)] \phi \quad q' = f(q)$$

$\underbrace{\qquad\qquad\qquad}_{\text{DF } \in \mathbb{R}^{3 \times 3}}$

$q' = \begin{bmatrix} \sqrt{1-\alpha^2}\phi \\ \phi \end{bmatrix} = f\left(q + \begin{bmatrix} \sqrt{1-\alpha^2}\phi \\ \phi \end{bmatrix}\right)$

transform output transform input $\phi \rightarrow \delta q$

$\delta q \rightarrow \phi'$

- Hessian of $f(q) : \mathbb{H} \rightarrow \mathbb{R}$

$$\nabla^2 f(q) = G(q)^T \frac{\partial^2 f}{\partial q^2} G(q) + \underbrace{I \left(\frac{\partial f}{\partial q} q \right)}_{\text{scalar}}$$

comes from $\frac{\partial f}{\partial q}$

- Now we can do Newton's method with quaternions (and DOP and SQP)

Example: Pose Estimation

- Given a bunch of vectors to known features in the environment, determine the robot's attitude
 - Called "Wahba's Problem"
- $$\underset{q}{\operatorname{argmin}} \quad J(q) = \sum_{n=1}^m \| \underbrace{U^N X_n}_{\substack{\text{known vectors} \\ \text{in inertial frame} \\ (\text{e.g. from a map})}} - \underbrace{Q(q)^B X_n}_{\substack{\text{measured vectors} \\ \text{in body frame} \\ (\text{e.g. from camera})}} \|_2^2 = \| r(q) \|_2^2$$

$$r(q) = \underbrace{\begin{bmatrix} {}^N X_1 & - Q^B X_1 \\ {}^N X_2 & - Q^B X_2 \\ \vdots & \end{bmatrix}}_{3m \times 1} \Rightarrow \underbrace{\nabla r(q)}_{3m \times 3} = \underbrace{\frac{\partial r}{\partial q}}_{3m \times 4} \underbrace{G(q)}_{4 \times 3}$$

- Gauss-Newton Method:

do:

$$\nabla r(q) = \frac{\partial F}{\partial q} G(q)$$

$$\phi = - \left[(\nabla r^\top \nabla r)^{-1} \nabla r^\top \right] r(q)$$

$$q \leftarrow q + \begin{bmatrix} \sqrt{1-\phi^2} \\ \phi \end{bmatrix} \quad \text{multiplicative update}$$

while $\| \phi \| > tol$

In general, perform line search here

* Background: Gauss-Newton for least-squares problems

$$\min_x J(x) = \frac{1}{2} \|r(x)\|_2^2 = \frac{1}{2} r(x)^T r(x)$$

$$\frac{\partial J}{\partial x} = \underbrace{r(x)^T \frac{\partial r}{\partial x}}.$$

throw this term out

$$\frac{\partial J}{\partial x^2} = \left(\frac{\partial r}{\partial x} \right)^T \left(\frac{\partial r}{\partial x} \right) + \left(I - r(x)^T \right) \frac{\partial^2 \text{vec}(r)}{\partial x^2}$$

$$\Rightarrow \left(\frac{\partial J}{\partial x^2} \right)^{-1} \nabla J \approx \left[\left(\frac{\partial r}{\partial x} \right)^T \left(\frac{\partial r}{\partial x} \right) \right]^{-1} \frac{\partial r}{\partial x}^T r(x)$$