

# Simulation of starting/stopping vortices for a lifting airfoil

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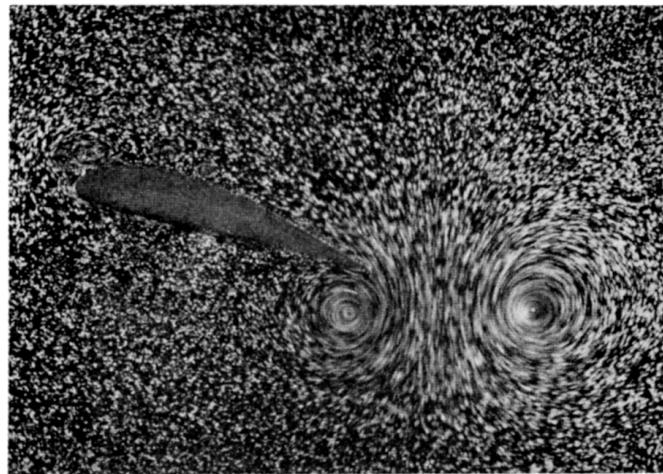


FIG. 55.—After formation of the starting vortex the airfoil was stopped and then the picture was taken.

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## Prandtl's experiments

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FIG. 48.—Streamlines round an airfoil the very first moment after starting.

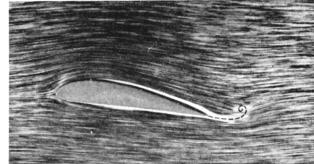


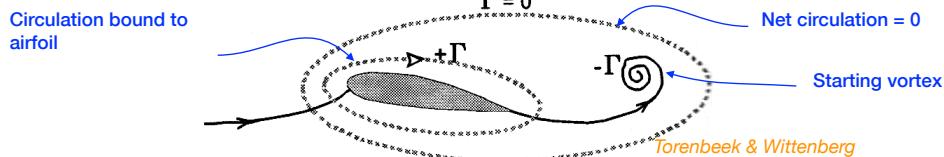
FIG. 49.—Formation of the starting vortex which is washed away with the fluid.



FIG. 50.—Growth of the starting vortex.



FIG. 51.—Taken somewhat later than Fig. 50.



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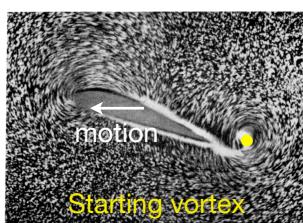


FIG. 54.—Like Fig. 52, but with greater angle of attack and consequently stronger starting vortex. Also shorter exposure of plate.

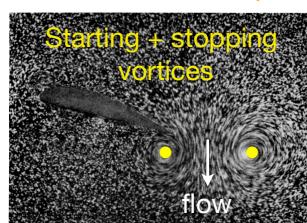
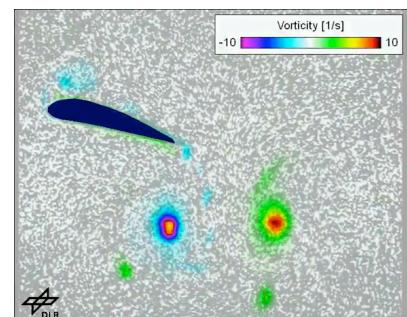


FIG. 55.—After formation of the starting vortex the airfoil was stopped and then the picture was taken.

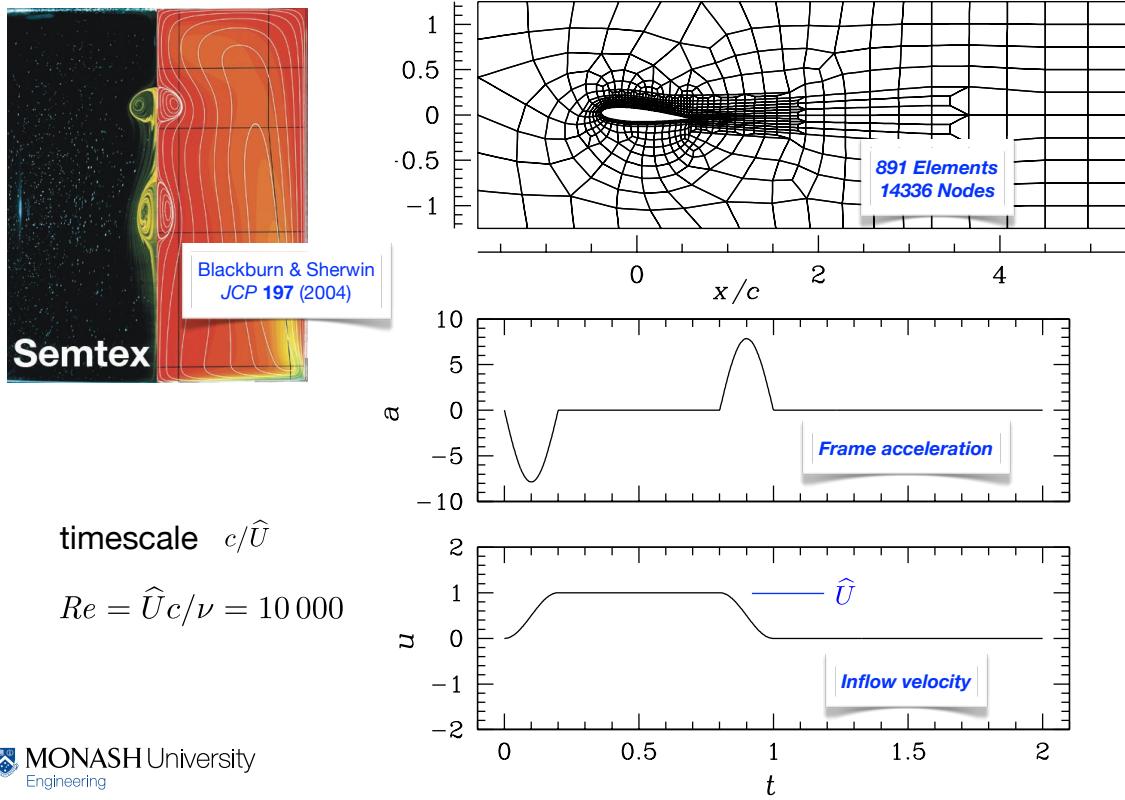


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## Spectral element DNS

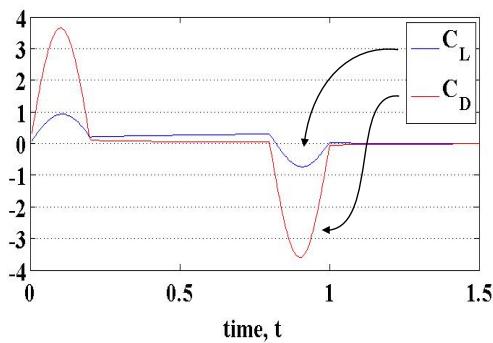
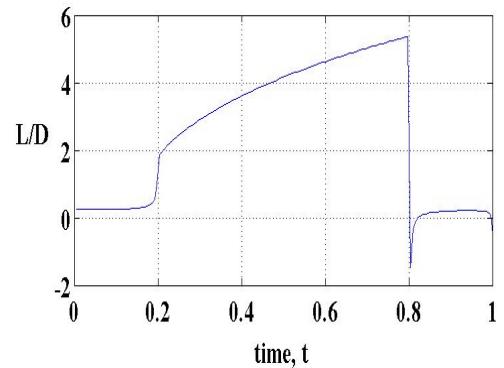
$$\partial_t \mathbf{u} + \mathbf{u} \cdot \nabla \mathbf{u} + \boxed{\mathbf{a}} = -\rho^{-1} \nabla p + Re^{-1} \nabla^2 \mathbf{u} \quad \nabla \cdot \mathbf{u} = 0$$



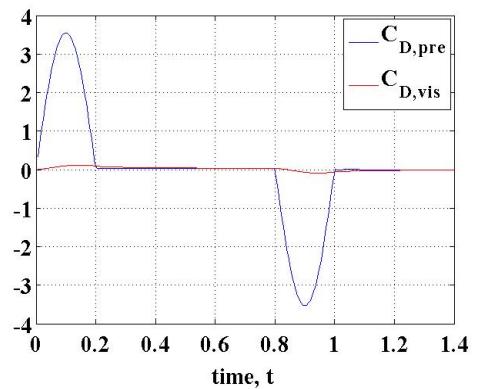
## Time evolution



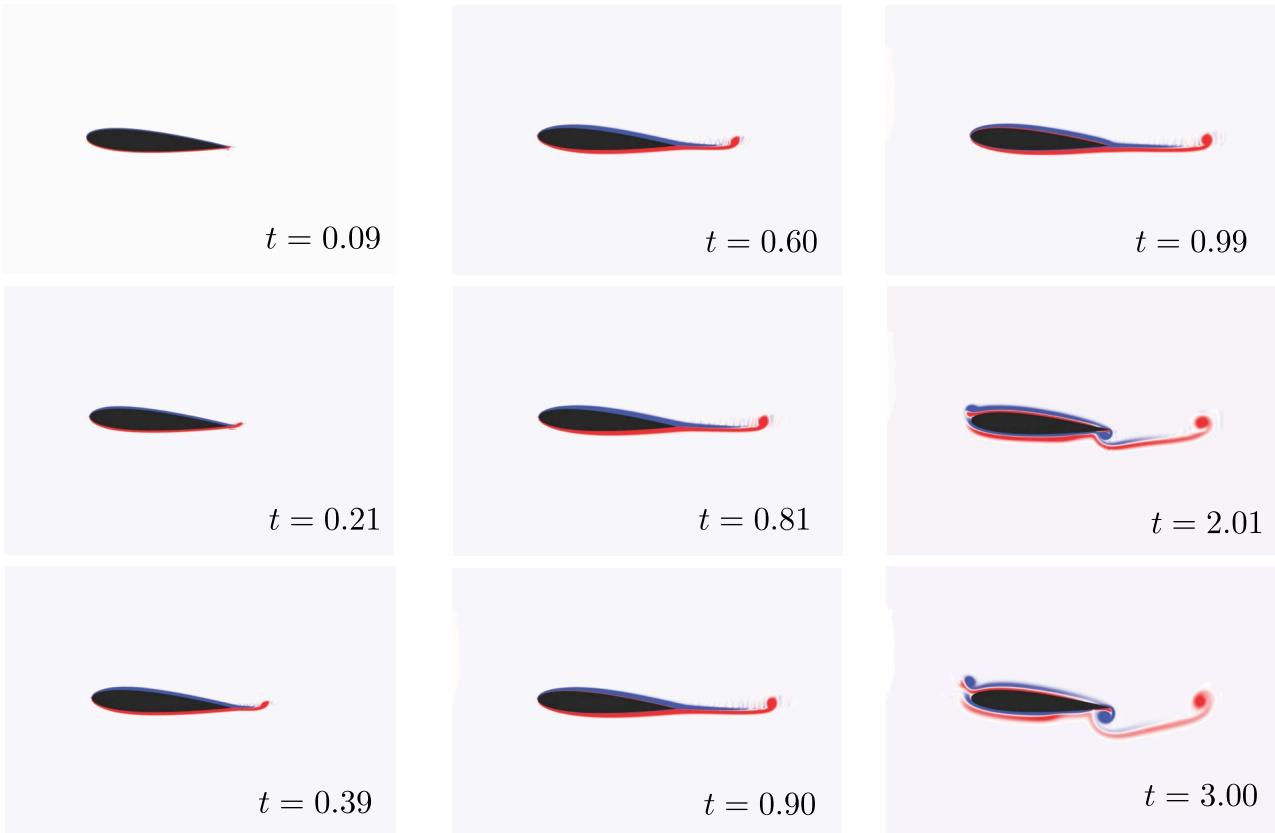
$$C_l = L' / \left( \frac{1}{2} \rho \hat{U}^2 c \right) \quad C_d = D' / \left( \frac{1}{2} \rho \hat{U}^2 c \right)$$



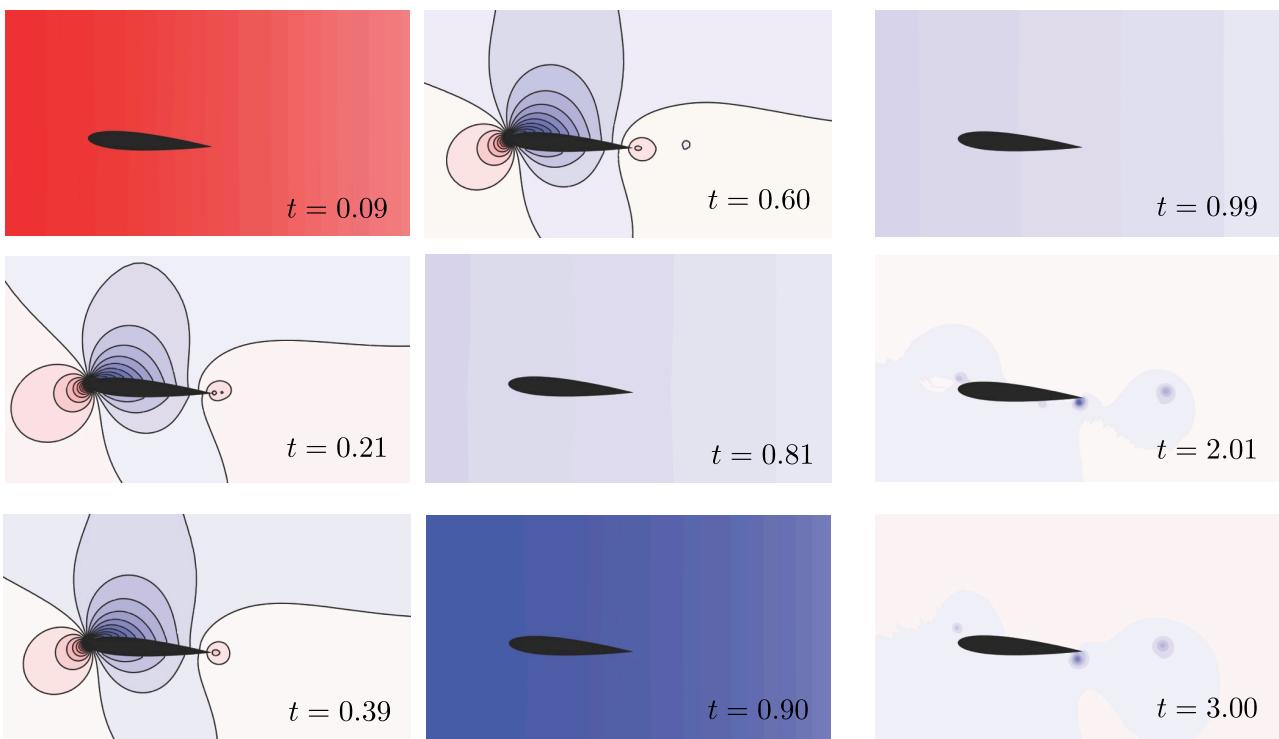
For  $\frac{\partial C_l}{d\alpha} = 2\pi$  steady-state  $C_l = 0.438$



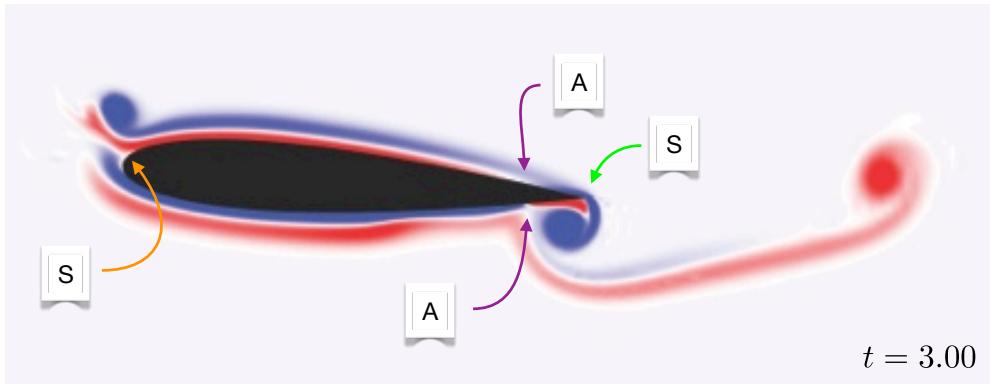
## Vorticity



## Pressure



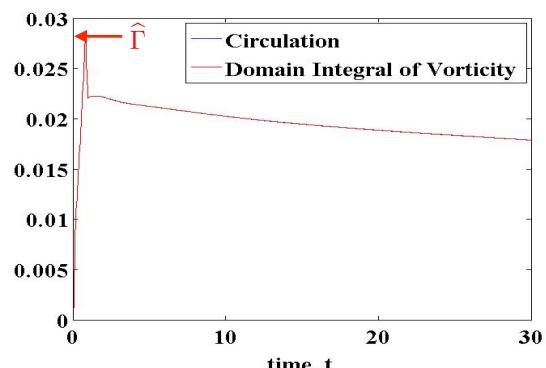
## Flow topology



## Circulation and vorticity

### Kutta–Joukowsky theorem

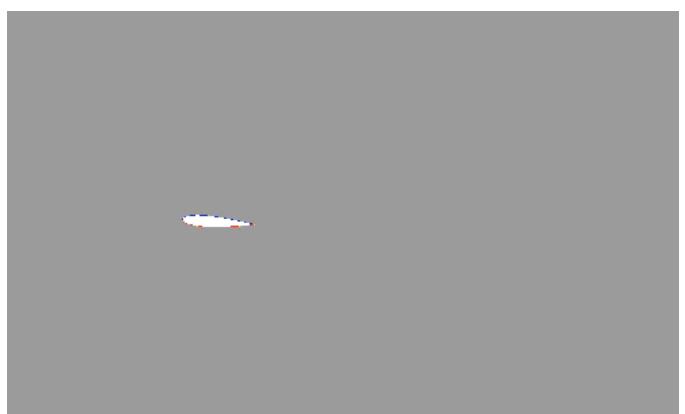
$$\Gamma = \oint \mathbf{u} \cdot d\mathbf{l} = \int \mathbf{n} \cdot \boldsymbol{\omega} dS$$



### Kelvin–Helmholtz theorem

$$A = \Delta x \times \Delta y = 60.0c^2 \quad \hat{\Gamma} = 0.0183$$

$$A = \Delta x \times \Delta y = 17.5c^2 \quad \hat{\Gamma} = 0.0281$$



## Summary



1. Very similar to Prandtl's experiment
2. Additional detail: generation of vortex pair at leading edge
3. Global circulation approaches zero as domain size increases
4. Forces dominated by added mass effects