



THE FINITE ELEMENT METHOD

L. Amerio

DATA:

- Mass per unit length $m = 9.75 \text{ kg/m}$
- Flexural rigidity $EJ = 1.34 \cdot 10^4 \text{ Nm}^2$
- Axial stiffness $EA = 2.57 \cdot 10^7 \text{ N}$
- Length $L = 2 \text{ m}$
- Structural Damping Matrix $[C_s] = \alpha[M] + \beta[K]$ ($\alpha = 0.2 \text{ s}^{-1}$; $\beta = 1 \cdot 10^{-4} \text{ s}$)

We want to calculate:

- **Natural frequencies;**
- **Mode shapes.**



The length of the beam finite element influences the maximum frequency of the structure analysis.

The beam finite element must have a quasi-static behaviour (the considered shape functions are the static ones) during vibration of the structure.

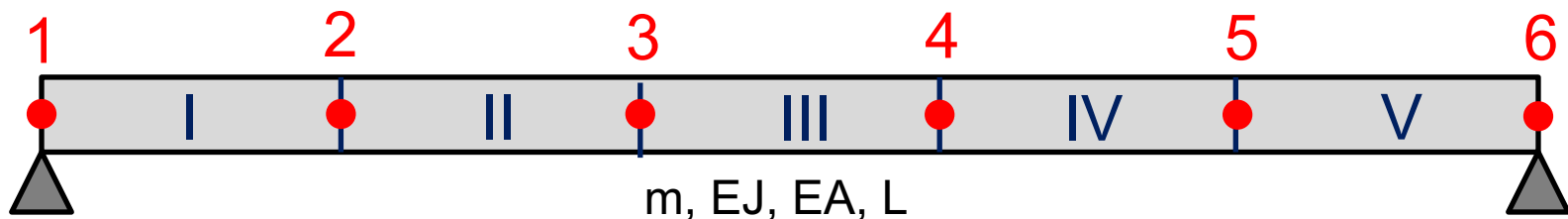


$$\omega_{fe,i} = \left(\frac{\pi}{l_{fe,i}} \right)^2 \sqrt{\frac{EJ}{m}} \gg \Omega_{max} \Rightarrow l_{fe,i,max} \ll \sqrt{\frac{\pi^2}{\Omega_{max}}} \cdot \sqrt{\frac{EJ}{m}}$$



For the considered structure:

f_{\max} [Hz]	Ω_{\max} [rad/s]	$l_{fe,i,\max}$ [m]
5	31.4	3.41
10	62.8	2.41
20	125.7	1.71
50	314.2	1.08
100	628.3	0.76
200	1256.6	0.54
300	1885.0	0.44



```
! FEM(1)
! 1st Exercise
! -----
! list of nodes :
*NODES
! n. of node - constraint code (x,y,theta) - x coordinate- y coordinate.
1      1 1 0      0.0      0.0
2      0 0 0      0.4      0.0
3      0 0 0      0.8      0.0
4      0 0 0      1.2      0.0
5      0 0 0      1.6      0.0
6      1 1 0      2.0      0.0
! end card *NODES
*ENDNODES
! -----
! list of elements :
*BEAMS
! n. of elem. - n. of input node - n. of output node - m, EA, EJ
1      1      2      9.75      2.57e7      1.34e4
2      2      3      9.75      2.57e7      1.34e4
3      3      4      9.75      2.57e7      1.34e4
4      4      5      9.75      2.57e7      1.34e4
5      5      6      9.75      2.57e7      1.34e4
*ENDBEAMS
```

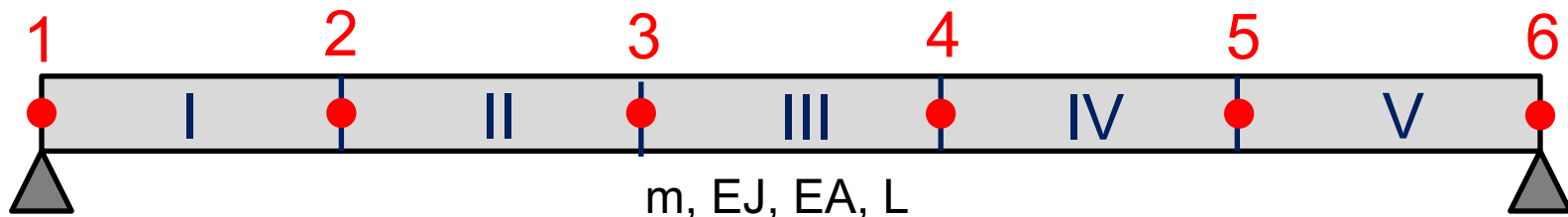
! = symbol for comment

*NODES = start definition of nodes

*ENDNODES = end definition of nodes

*BEAMS = start definition of beam finite element

*ENDBEAMS = end definition of beam finite element



Definition of input file (*.inp)

6

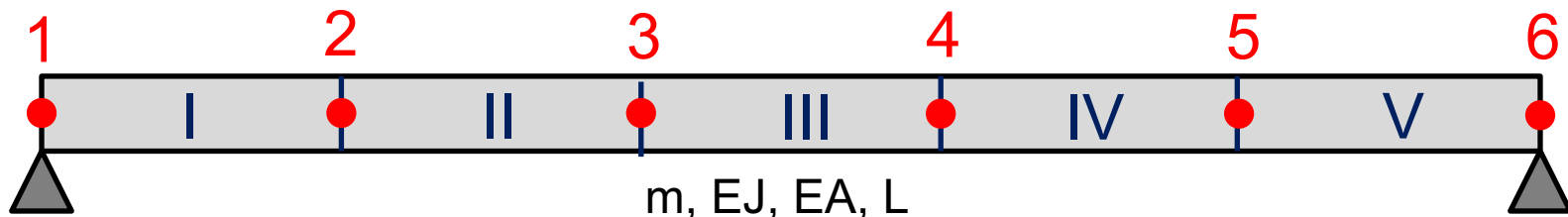
```
! FEM(1)
! 1st Exercise
!
! list of nodes :
*NODES
! n. of node - constraint code (x,y,theta) - x coordinate- y coordinate.
1      1 1 0      0.0      0.0
2      0 0 0      0.4      0.0
3      0 0 0      0.8      0.0
4      0 0 0      1.2      0.0
5      0 0 0      1.6      0.0
6      1 1 0      2.0      0.0
! end card *NODES
*ENDNODES
!
! list of elements :
*BEAMS
! n. of elem. - n. of input node - n. of output node - m, EA, EJ
1      1      2      9.75      2.57e7      1.34e4
2      2      3      9.75      2.57e7      1.34e4
3      3      4      9.75      2.57e7      1.34e4
4      4      5      9.75      2.57e7      1.34e4
5      5      6      9.75      2.57e7      1.34e4
*ENDBEAMS
```

Constraint code:
0 = free d.o.f.
1 = constrained d.o.f.

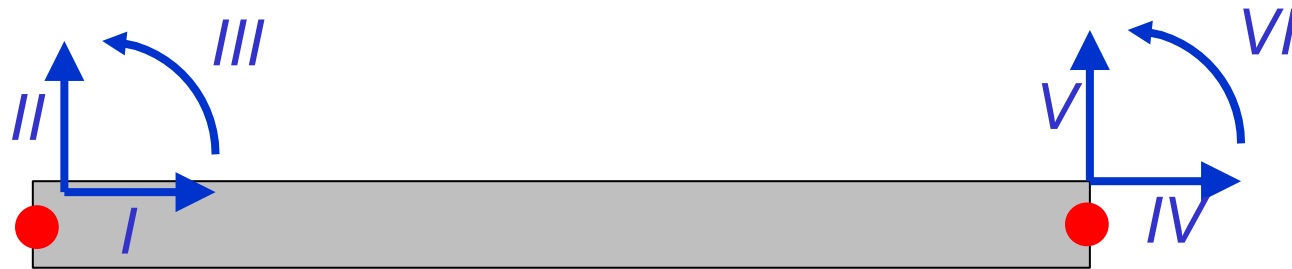
Coordinates of node:
x [m], y [m]

Left and right nodes of the beam finite element

Properties of the beam finite element:
m [kg/m], EA [N], EJ [Nm²]

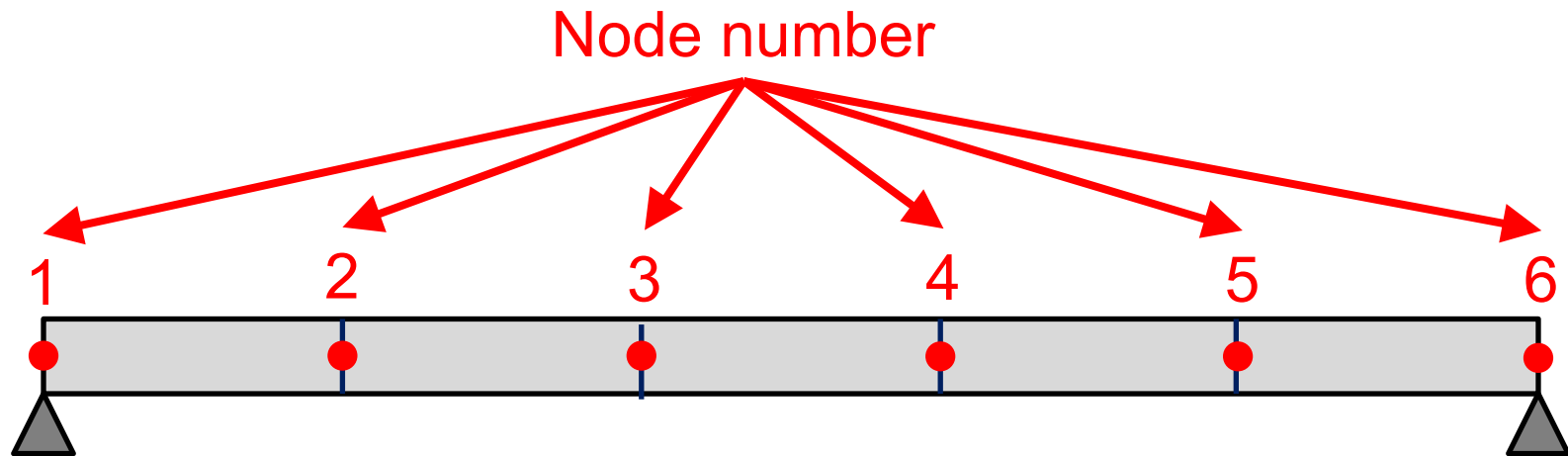


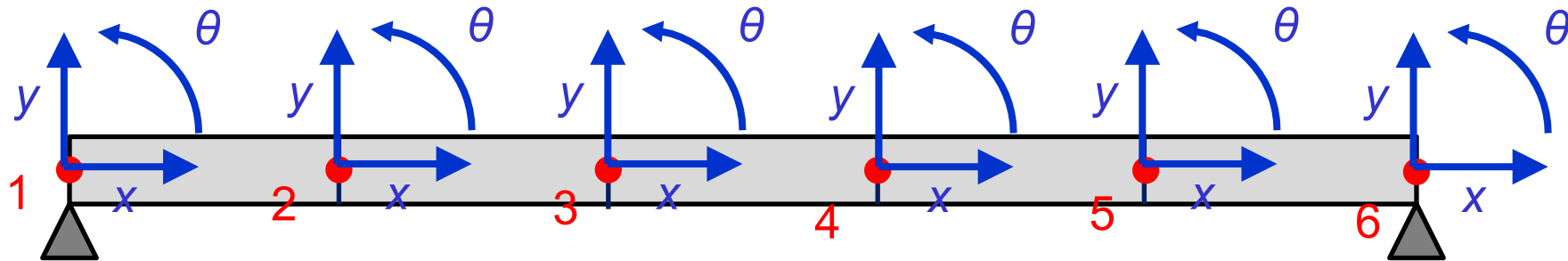
Each element has it's own **mass** and **stiffness matrixes** depending on the type of the element (string, beam, shell, brick...)

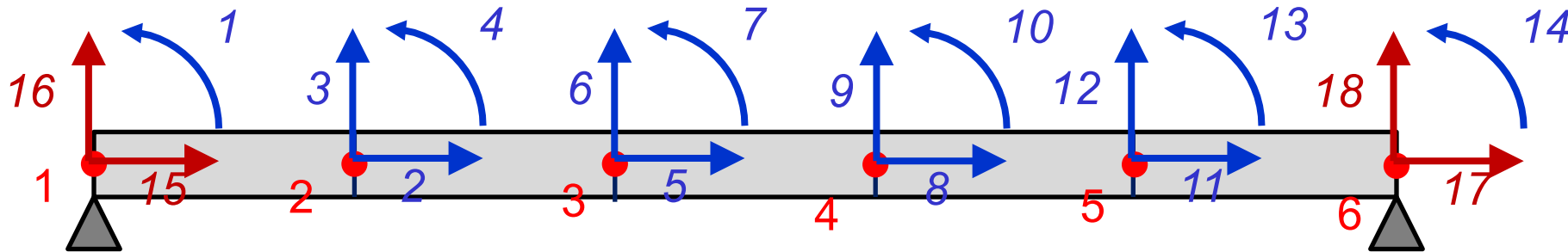


$$\mathbf{K}_{el} = \frac{EJ}{L^3} \begin{bmatrix} 12 & 6L & -12 & 6L \\ 6L & 4L^2 & 6L & 2L^2 \\ -12 & -6L & 12 & -6L \\ 6L & 2L^2 & -6L & 4L^2 \end{bmatrix} \quad \mathbf{M}_{el} = mL \begin{bmatrix} \frac{12}{35} & -\frac{11}{210}L & \frac{9}{70} & \frac{13}{420}L \\ \frac{11}{210}L & \frac{1}{105}L^2 & -\frac{13}{420}L & -\frac{1}{140}L^2 \\ \frac{9}{70} & -\frac{13}{420}L & \frac{13}{35} & \frac{11}{210}L \\ \frac{13}{420}L & -\frac{1}{140}L^2 & \frac{11}{210}L & \frac{1}{105}L^2 \end{bmatrix}$$

How to obtain the system matrixes?

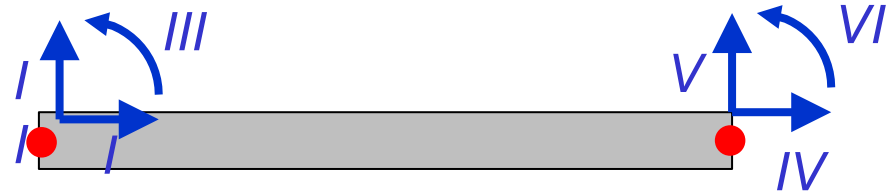
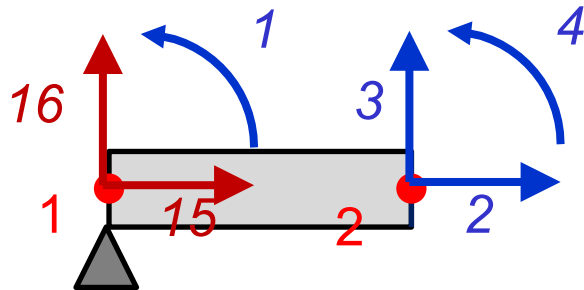






idf matrix

Node	X DoF	Y DoF	Rotation
1	15	16	1
2	2	3	4
3	5	6	7
4	8	9	10
5	11	12	13
6	17	18	14

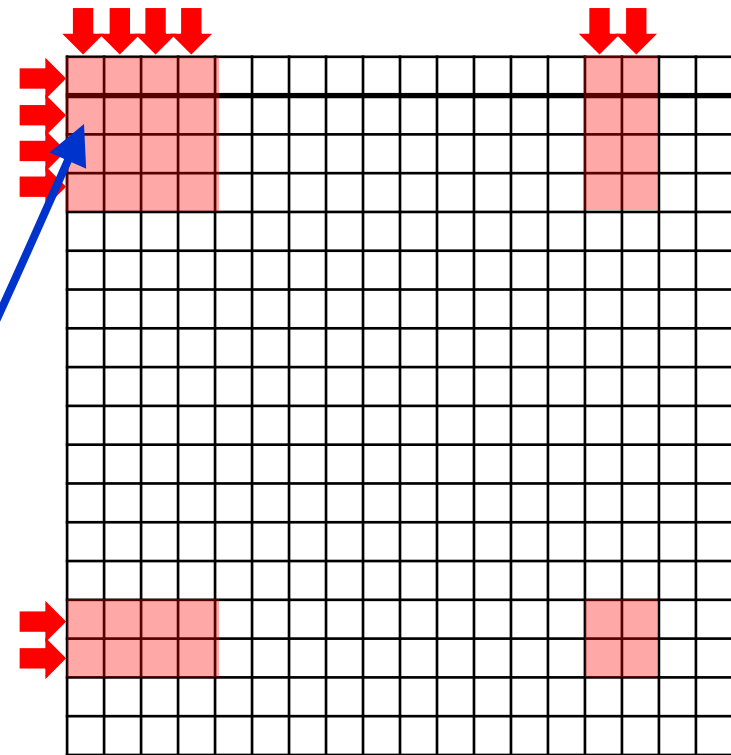


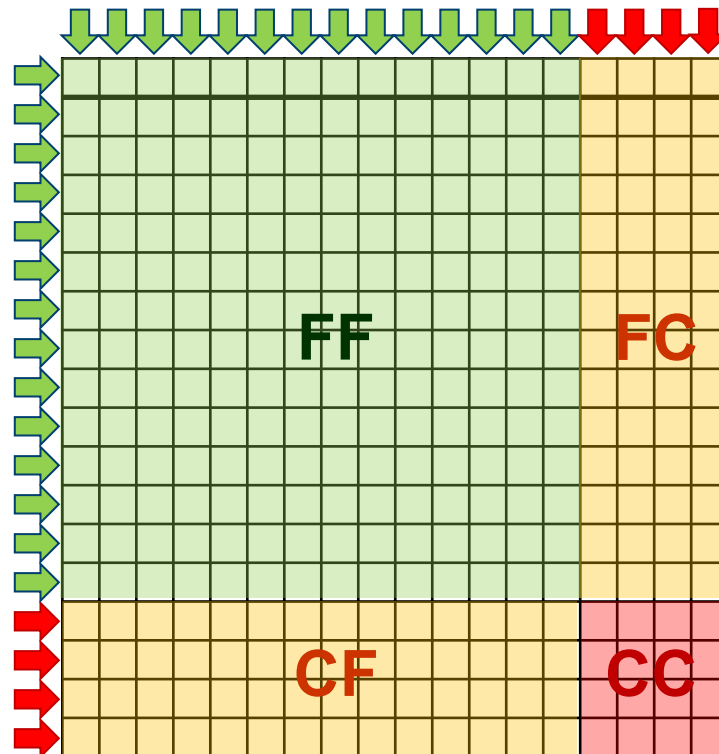
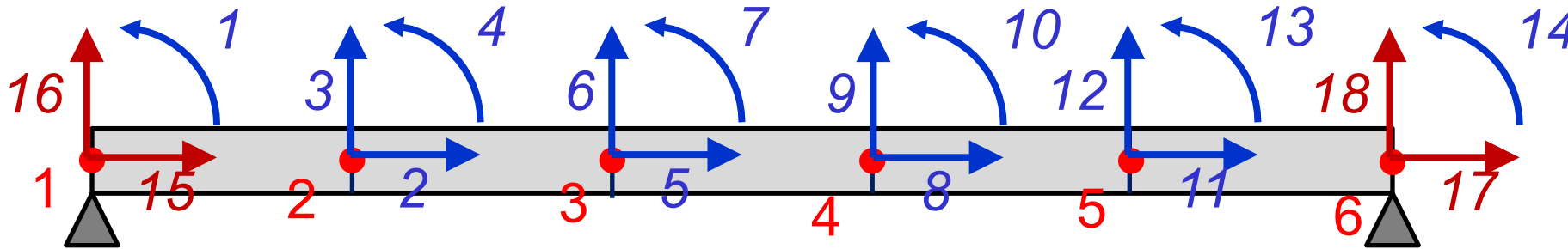
Incidence matrix

Element	I	II	III	IV	V	VI
1	15	16	1	2	3	4
2	2	3	4	5	6	7
3	5	6	7	8	9	10
4	8	9	10	11	12	13
5	11	12	13	17	18	14

Element	I	II	III	IV	V	VI
1	15	16	1	2	3	4
2	2	3	4	5	6	7

$$\mathbf{M}_{el} = mL \begin{bmatrix} \frac{12}{35} & -\frac{11}{210}L & \frac{9}{70} & \frac{13}{420}L \\ -\frac{11}{210}L & \frac{1}{105}L^2 & -\frac{13}{420}L & -\frac{1}{140}L^2 \\ \frac{9}{70} & -\frac{13}{420}L & \frac{13}{35} & \frac{11}{210}L \\ \frac{13}{420}L & -\frac{1}{140}L^2 & \frac{11}{210}L & \frac{1}{105}L^2 \end{bmatrix}$$



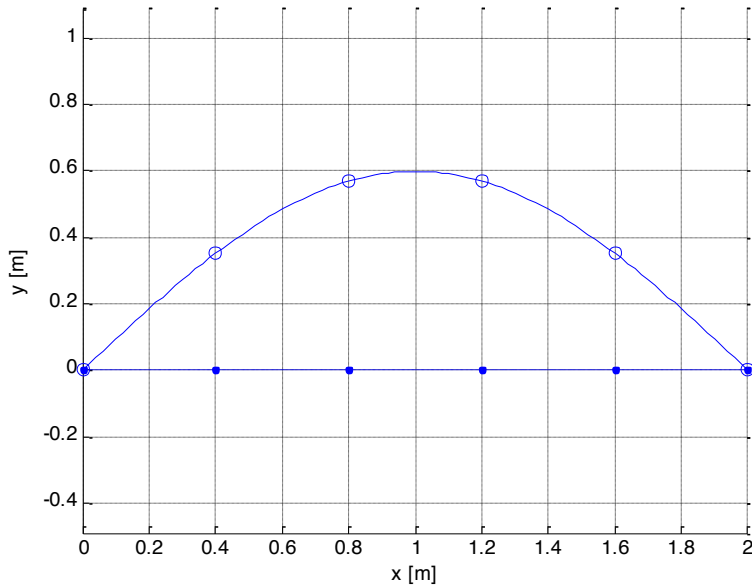


If we divide the structure in 5 beam finite elements (6 nodes – 3 d.o.f. for each node = 18 d.o.f.):

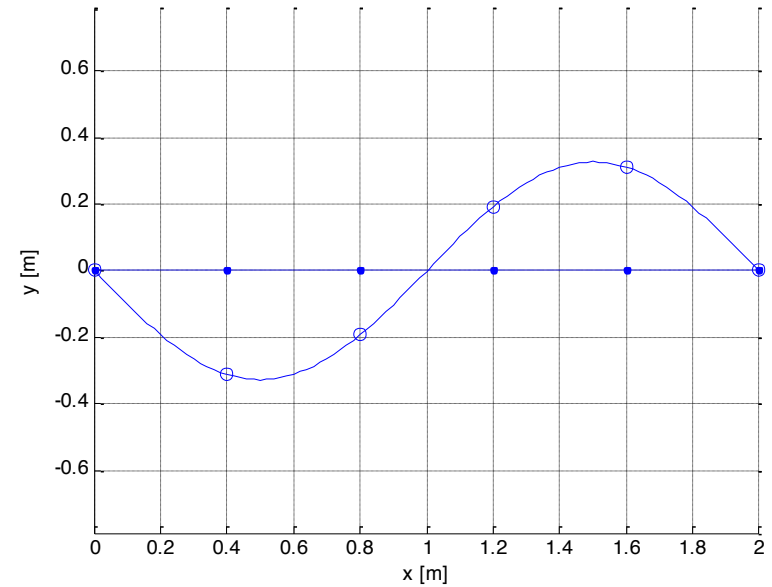
$$[M]_{18 \times 18} = \begin{bmatrix} [M_{FF}]_{14 \times 14} & [M_{FC}]_{14 \times 4} \\ [M_{CF}]_{4 \times 14} & [M_{CC}]_{4 \times 4} \end{bmatrix}$$

$$[K]_{18 \times 18} = \begin{bmatrix} [K_{FF}]_{14 \times 14} & [K_{FC}]_{14 \times 4} \\ [K_{CF}]_{4 \times 14} & [K_{CC}]_{4 \times 4} \end{bmatrix}$$

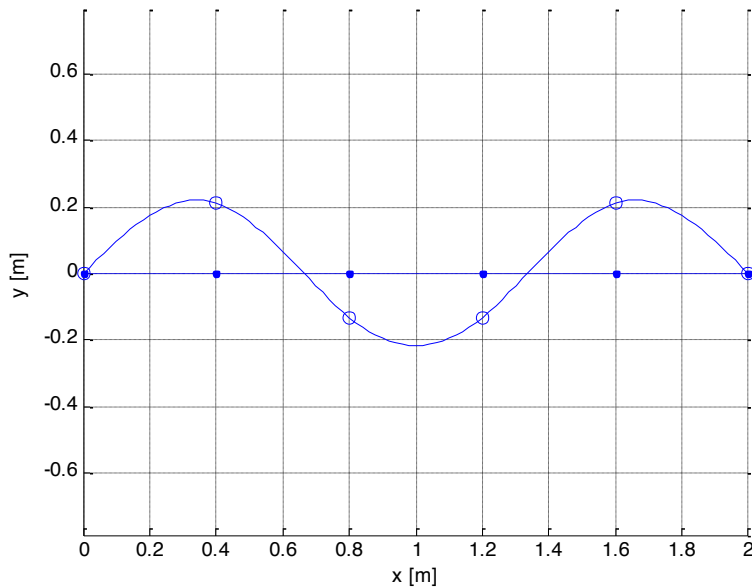
Mode 1: Freq [Hz]=14.5598



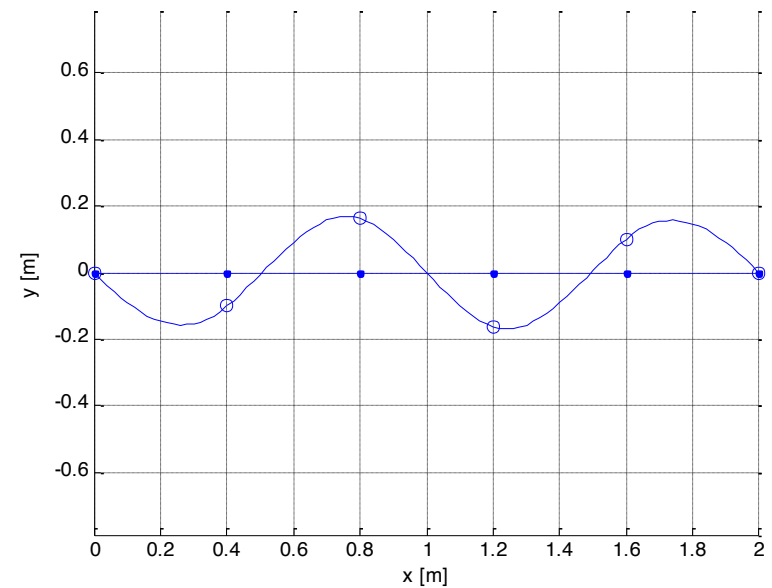
Mode 2: Freq [Hz]=58.3296



Mode 3: Freq [Hz]=132.065



Mode 4: Freq [Hz]=238.2984



Ridurre il numero di elementi aumenta la rigidezza "apparente" del sistema

	$l_{fe} = 1\text{m}$	$l_{fe} = 0.5\text{m}$	$l_{fe} = 0.4\text{m}$	$l_{fe} = 0.1\text{m}$
	$\omega_n = 58 \text{ Hz}$	$\omega_n = 233$	$\omega_n = 364$	$\omega_n = 5823$
$f_{\text{real}} [\text{Hz}]$	$f_{\text{fem}} [\text{Hz}]$	$f_{\text{fem}} [\text{Hz}]$	$f_{\text{fem}} [\text{Hz}]$	$f_{\text{fem}} [\text{Hz}]$
14.56	14.62	14.56	14.56	14.56
58.23	64.63	58.46	58.33	58.23
131.02	162.46	133.42	132.06	131.03
232.93	296.19	258.54	238.3	232.96

$$f_{\text{real},k} = \frac{1}{2\pi} \left(\frac{k\pi}{L} \right)^2 \sqrt{\frac{EJ}{m}}$$

The equation of motion for the system is

$$M\ddot{x} + R\dot{x} + Kx = F_0 e^{i\Omega t}$$

To solve the ODE we assume

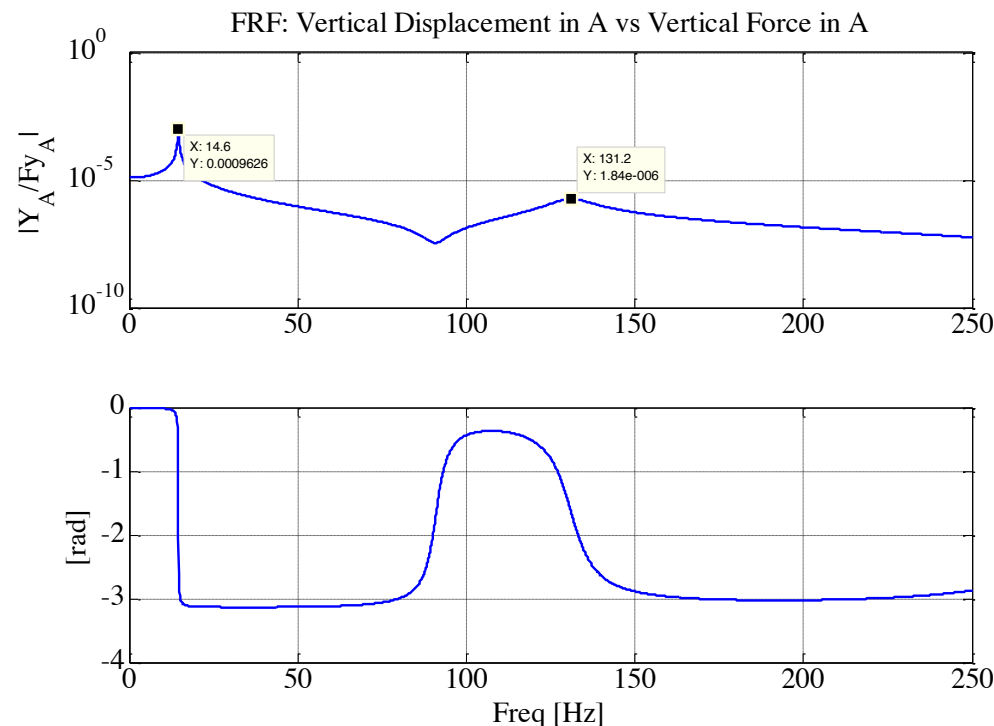
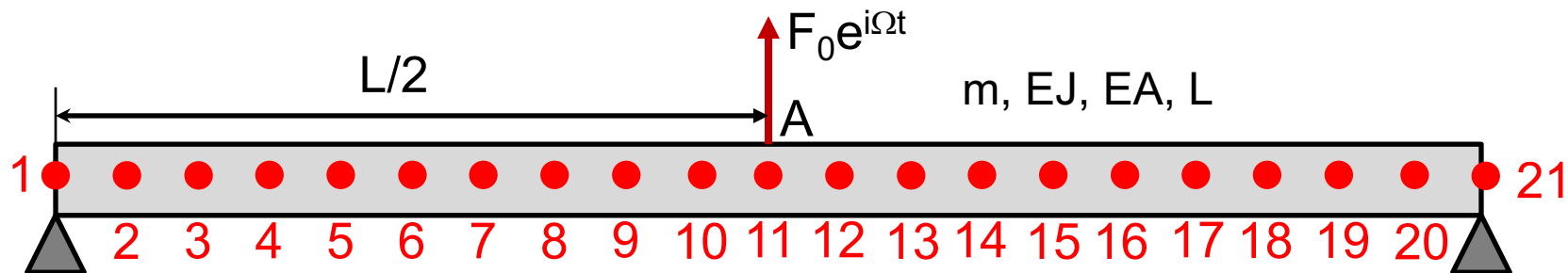
$$x = x_0 e^{i\Omega t}$$

Therefore

$$(-\Omega^2 M + i\Omega R + K) x_0 = F_0$$

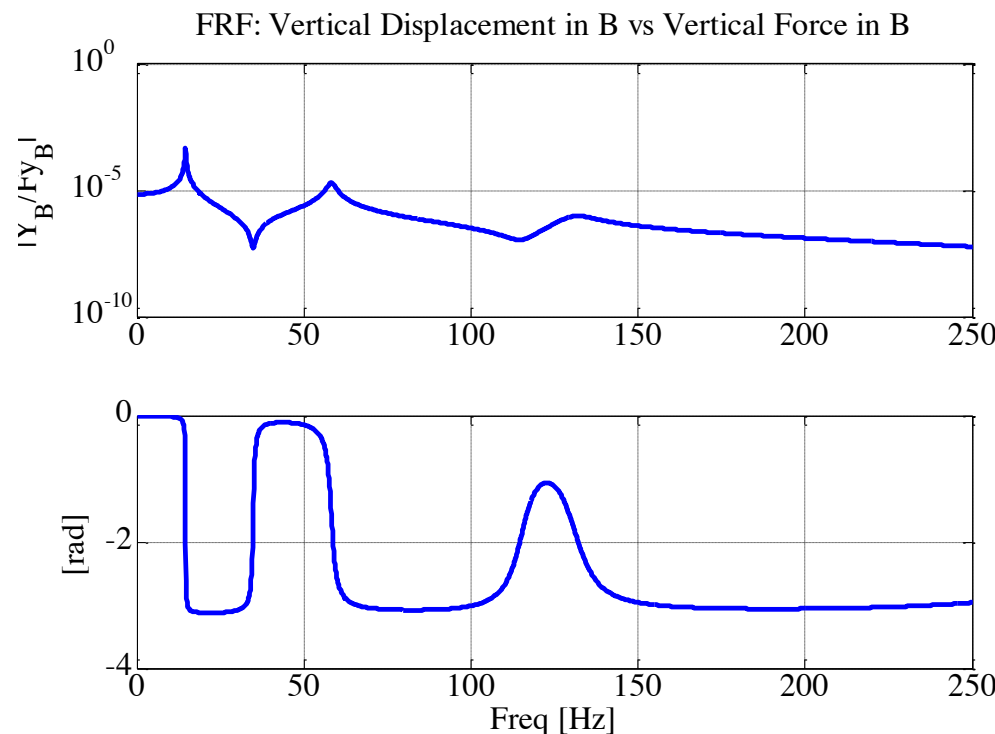
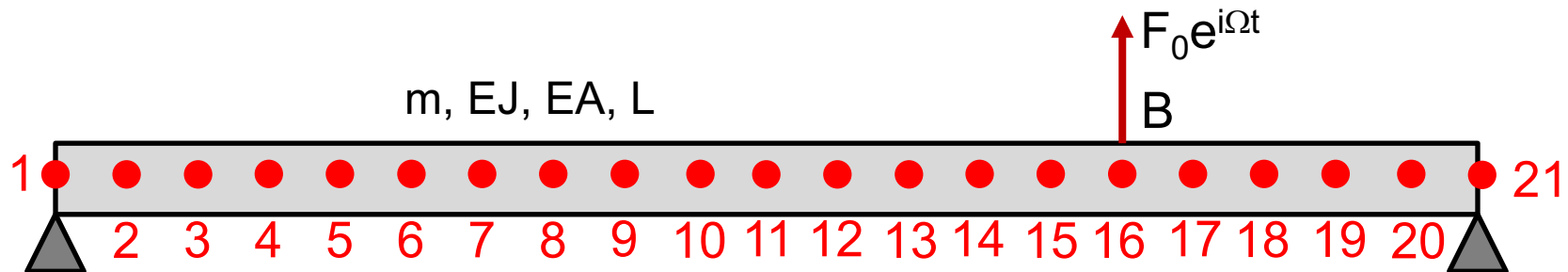
$$x_0(\Omega) = (-\Omega^2 M + i\Omega R + K)^{-1} F_0$$

A harmonic force $F=F_0e^{i\Omega t}$ is applied in A: we want to calculate the FRF of the system ($l_{fe} = 0.1m$).

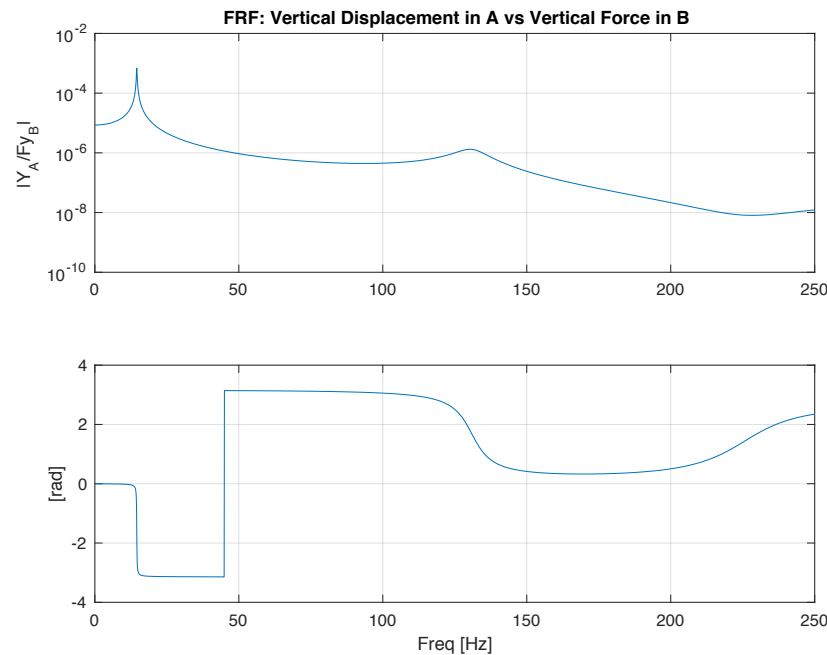
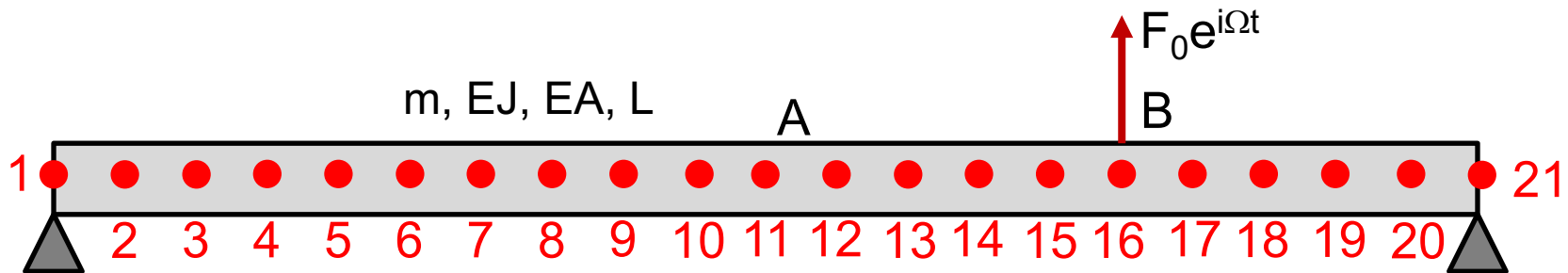


Now it's your turn...

A harmonic force $F=F_0e^{i\Omega t}$ is applied in B: we want to calculate the FRF of the system for the displacement in B ($l_{fe} = 0.1m$).

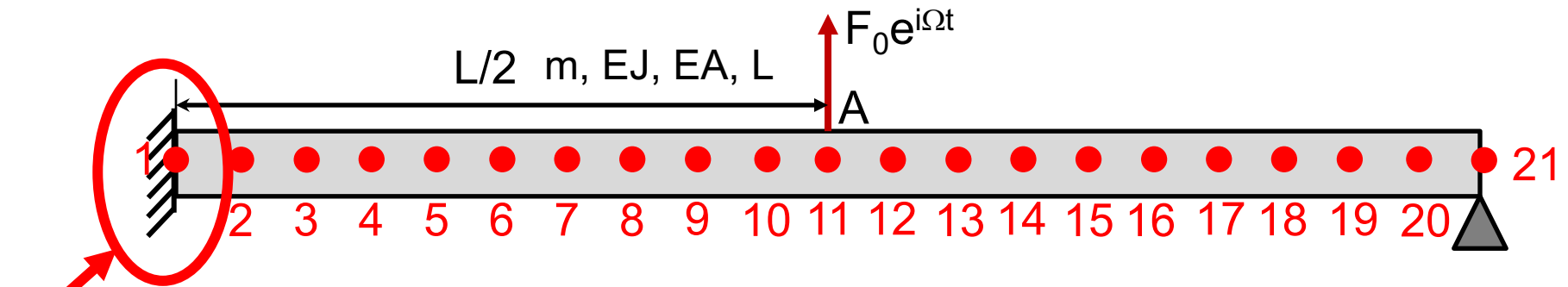


A harmonic force $F=F_0e^{i\Omega t}$ is applied in B: we want to calculate the FRF of the system for the displacement in A (i.e. $l_{fe} = 0.1\text{m}$).

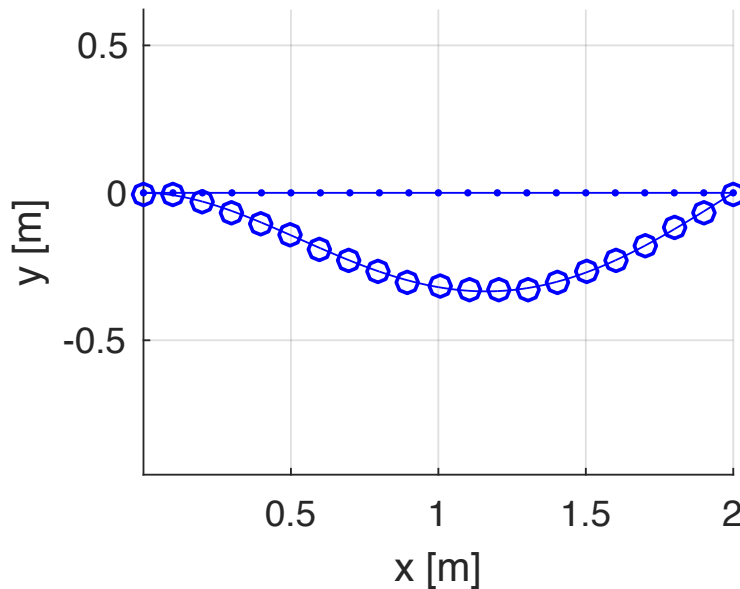


Modal shapes and Frequency Response Function

22



Mode 1: Freq [Hz]=22.7428



FRF: Vertical Displacement in A vs Vertical Force in A

