

8 Digital or HDTV (High Definition TV) or IDTV (Improved Definition TV)

Digital TV Principles

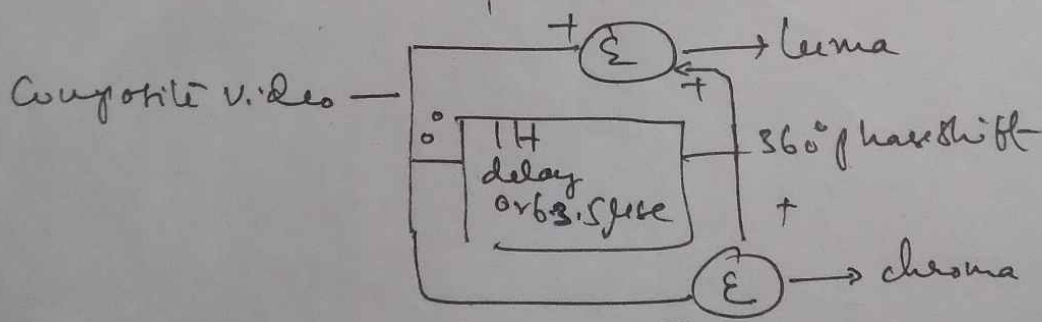
Important deficiencies of NTSC analog colour Txn are MW3-18

- (1) Luminance resolution is inadequate for large screen
- (2) chrominance \rightarrow is good along vertical direction & insufficient along horizontal direction which causes colour smearing
- (3) 4:3 aspect ratio is insufficient with commercial motion pictures where aspect ratio is 16:9
- (4) Interleaving of luminance & chrominance signals in 6 MHz b/w may conserve b/w but there may be interference or cross luminance (spurious colours in motion pictures)
- (5) Interlaced scanning conserves b/w & avoids flicker but not suitable for alphanumeric / medical displays.

Hence analog system must be converted to digital system.

Remedial measures in digital TV are:-

- (1) with COMB filtering (Combinational) interference between luminance & chrominance signals at Rx can be reduced.



- (2) Greater V resolution is obtained by deriving additional scanlines.
- (3) using more elaborate signal processing methods.

100IRE white level

luminance level

chrominance signal

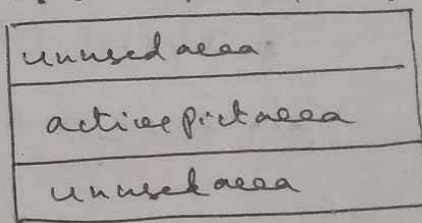
- * digitize the signal to reduce noise, distortion, interference etc.
- * use compression techniques JPEG, MPEG, Dolby methods to reduce b/w requirement & better resolution/clarity
- * compatibility with computers (storing), mobile units
- * sharper edges of digitized pic by convolution & correlation to reduce noise/distortion/interference effects.

(4) picture in picture is another possibility

(5) store video in memory in a compressed format & then use progressive scanning

(6) To get wider aspect ratio use fewer active lines by a technique called letter boxing. used in sports/news/movies etc.

emblem of broadcast station



commercial/advertisements

(7) separate luma/chroma TX. improves resolution used in short distance comm. like VCR/DVD use separate wires for audio/video/sound etc.

II Based on quality:-

There are 3 stds in digital TV:-

	Horizontal resolution	Vertical resolution	Aspect ratio	Pixel shape	frame rate / scan
SDTV (Standard Defn TV) - 1980	704	480	4:3	Square	24p, 30p, 30i, 60p
EDTV (Enhanced DTV) - 1985	1280	720	16:9	-	24p, 30p, 60p
HDTV (High DTV or) - 1996	1920	1080	16:9	rectangle	24p, 30p, 30i

ITV Improved - 11-

Pict-quality

Bitrate

moderate

225 mbps

better -

922 - 11-

best -

922 - 11- or 1.5 Gbps.

- progressive
- Interlaced

18 * Digital TV broadcasts pic & audio signals as digital packets or bursts of data.

* Higher resolution & better quality possible.

* 3 ITN standards. Absolutely:
based on ITN methods.

(1) DVB: Digital Video Broadcasting (used in India, Europe)

* DVB-S (Satellite) This is an

* DVB-T (Terrestrial)

* DVB-C (Cable)

* DVB-H (Handheld)

(2) ATSC: Advanced TV Syst Committee: US, Canada, Mexico, South Korea.

(3) ISDB: Integrated Services Digital Broadcasting used in Japan.

India: - DVB-T (8MHz b.w) H.262 / MPEG-2 Video

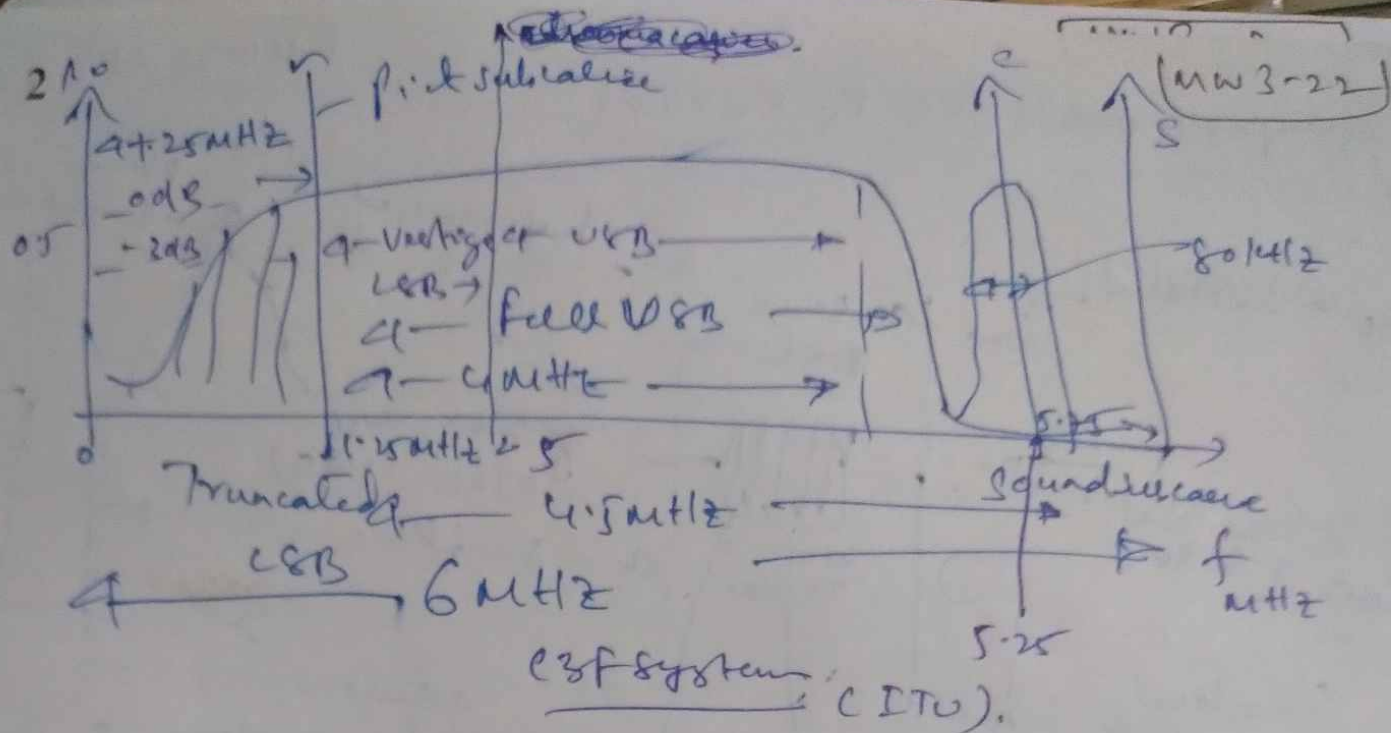
* heavy Video-Audio compression

* Theater quality pic & CD quality sound.

* For cable TV (DTH or DVB-C) they use ~~6~~ COFDM (Coded OFDM) or 256 QAM (data rate

19.39 Mbps) or DCT (Discrete Cosine

* Immune to multipath interference. ^{Transform} Algorithm



Advanced Radars:-

DVB: Digital Video Broadcasting

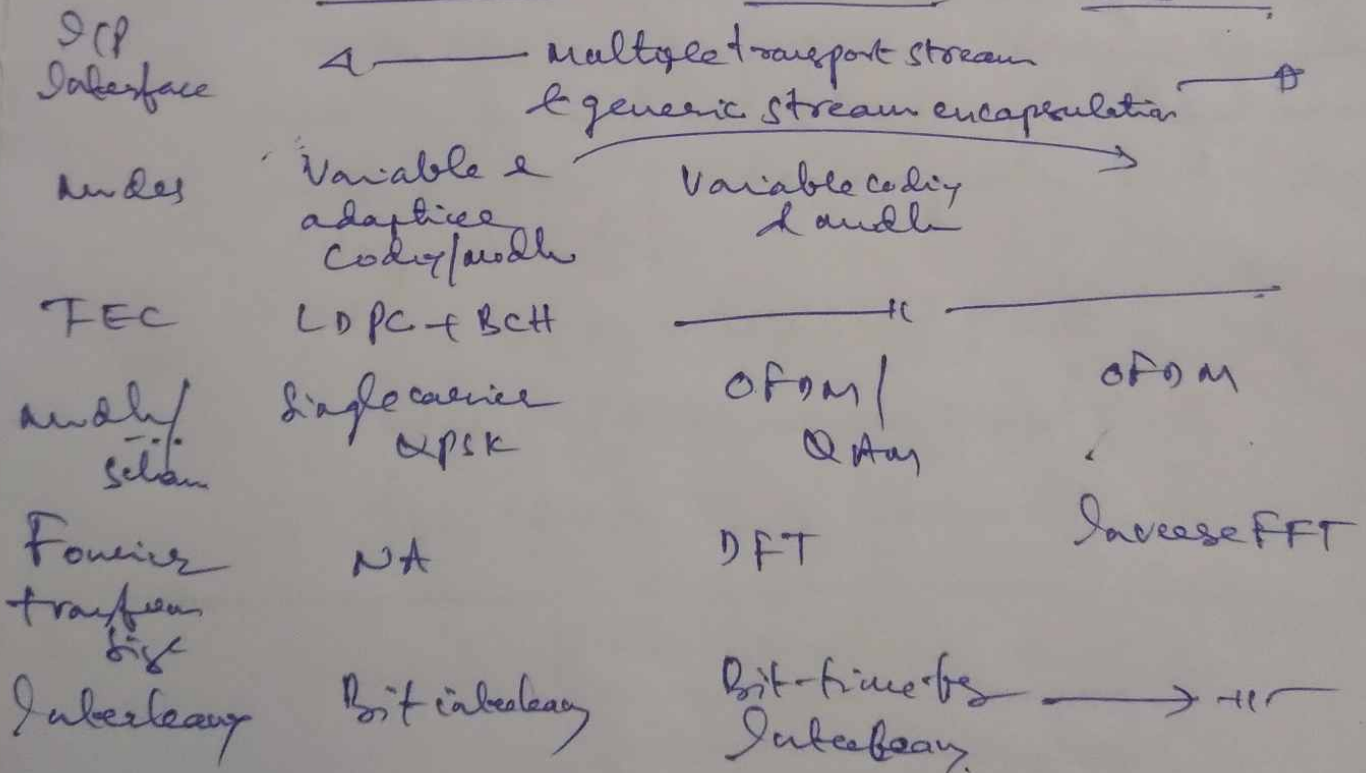
UNIT 3 - 23

- open standard for digital TV
- ETSI - European Telecomm. Std. Institute maintains the std.
- DVB-S, T, C, H meet the std. defined by the Physical & data link layers of OSI model
- & vice, ~~are connected to~~ ^{interfaced with} physical layer the SPI (Synchronous Parallel Interface) or ASI (Asynchronous Serial Interface).
- All data are TX in MPEG streams.
- DVB-S uses 8-QPSK, 16QAM
- DVB-C - 16, 32, 64, 128, 256 QAM
- DVB-T - 16 or 64 QAM (QPSK), COFDM

DVB-S2

DVB-T2

DVB-C2



HDTV Principles:-

UNIT 3-24

- Eye is more tolerant to q noise (8 to 10 bits/sample also used)
- In digital audio with 14 to 18 bits/sample - poorer tolerance of ear.

$$\text{No of bits/sample } m = \frac{f_b}{N_{PT} R_f}$$

where f_b = bit rate, N_{PT} = No of pixels/frame including luma & chroma signals, R_f = frame repetition rate
of N_L = No of luminance pixels/frame
 $= N_V \times N_H$
No of vertical & horizontal resolution pixels
hence $N_{PT} = 1.5 N_L$

eg: Consider a digital video resolution of 640×480 pixels with $R_f = 30 \text{ Hz}$ using p-scan. Assume luminance signal is sampled using 8 bits/sample. Find approximate bit rate of video neglecting sync, correlation, compression etc.

Sol:-

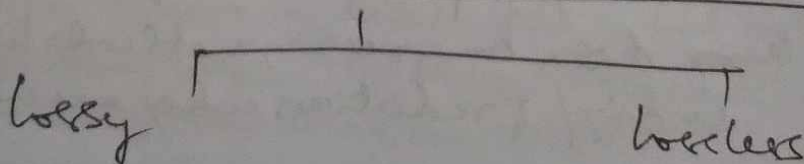
$$N_L = N_H N_V = 307.2 \times 10^3 \text{ pixels}$$

$$m = 8$$
$$R_f = 30$$

$$N_{PT} = 1.5 N_L = 460.8 \times 10^3 \text{ pixels}$$

$$f_b = N_{PT} m R_f = 110.6 \text{ Mbps.}$$

Video Compression methods



• JPEG: Joint Picture Expert Group - ^(UNIT 3-25) used for
(lossless) Compressing Still pictures

• MPEG: Motion Picture Experts Group
(lossy) (Audio compression is also done).

↳ MPEG 1, 2, 3, 4 stds.

MPEG 1: deals with low data rate Video like CD-Roms,
Internet etc. (1D, 2D, 3D are common)

MPEG 2: uses VCT to reduce data rate to 19.39 Mbps
for Terrestrial broadcasting in 6 MHz band.
Signal is converted from 't' to 'f' domain,
digital filtering is used to reduce data rate

Other Video Compression methods:-

I lossless { Huffman coding } H.W.
Run length coding

II lossy { Spatial coding (lossy)
Temporal coding (lossy)

• Spatial: Redundancies within a frame is used.
used for still images. This is the basic of JPEG

• Temporal: This includes std video codes such as
H.261, H.263, H.264, MPEG 1 to 4

This is based on 4 fundamental redundancy
reduction principles.

✓ to reduce redundancy among pixels within a picture
(similarity of pixels in a frame is looked into)
using data compression methods like transform
coding / predictive coding are used

- ✓ To remove similarities between successive pictures, we code difference between them.
- Tree static part of image sequence, difference may be close to zero & hence not-coded. This ↑ bit saving.
- ✓ use variable length coding for compressed data symbols. This reduces redundancy.

- ✓ Transform coding is used to remove spatial redundancy in images by mapping the pixels into a transform domain for data reduction. The starting of this method is, energy of most of the natural scenes is mainly concentrated in low freq region & hence we can transform them into few coeffs. They can be quantized to remove insignificant coeffs.

Audio:- Compression is used to save bits.

- Storage capacity of files can be improved.
- Theoretically opt. compression possible but not suitable for real time audio
- Pattern recognition, LPC method are used.

H.264 (MPEG-4 AVC (Advanced Video Coding) (lossy method)

- Most common format for recording, compression & distributing video content.
- A block oriented compression video std. developed by ITU-T-VCEG (Video Coding expert group) with MPEG. They are together called JVT (Joint Video Team)
- Used in Blu-ray disc (digital optical disc)
- Storage used for HD video. Resolution 1080 pixels
- Means less base or video tapes used to record disc
- Video, YouTube, iTunes, HDTV-T, C, S use this

Analog TV

- 1) AM, FM signal modulator
- 2) Y, I, Q signals
- 3) AM-VSB
- 4) 6MHz + Audio ch
- 5) EIRP - 45 to 75 kW
- 6) Noise & multipath interference more
- 7) phosphor, LED screen
- 8) Aspect ratio 4:3
- 9) poor quality

Digital TV

(UNIT 3-28)

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Digital Pixels

Y, B-G, R-G signals

8 VSB-TCM, MPEG-2, AVC-Advanced Video coding

6MHz ch + multicasting / Multiprogram Viewing

5 to 7.5 kW

Nil

LED, plasma screen

16:9

Best quality

UNIT 4:- RADAR Principles:-

UNIT 4-1

("Radio Detection And Ranging")

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Introduction:- "Gathering information about distant objects or targets by sending EM waves and analyzing echoes"
These words were coined by US Navy in 1942 during world war.

- * Radar can improve human vision during darkness / fog / haze / rain / snow etc to detect targets (airplanes / ships etc)
- * Frequency bands are similar to that of microwave band.
VHF (30-300 MHz), UHF (300-1 GHz), L, S, C, X, Ku, K, Ka are used. Power will be from 0.25 MW to few MW.

Applications:- Used on ground, space, sea.

- a) Ground based: Tsunami early warning, weather, tracking of targets
 - b) Air / Space based: Detect aircraft, ships, land vehicles movement, storm avoidance, navigation, remote sensing, guidance of spacecraft etc.
 - c) Sea based: Navigation of ships, locating shore, rescue operations etc.
- d) Military applications.

Major areas:-

- 1) ATC - Air traffic control: GCA - Ground Control Approach
ILS - Instrument Landing System, Radar beacons
- 2) Satellite tracking, missile guidance, Space station docking, outer space missions etc, remote sensing applications.

Principles of Radar

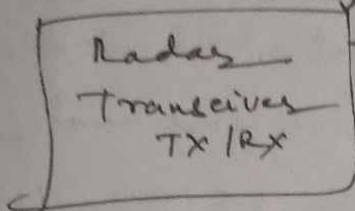
1) Pulse Radar

Sec 1

Unit 4-2

Basic principle of Radar:-

Ground based Radar



Here RF pulse train of specific PRF,

pulse width, power, direction are transmitted towards the target. Echo from the target is analyzed for parameters like

- Range • direction • Velocity • Direction of motion etc.

- If T_R is the round trip delay for RF pulse, r the range then $r = \frac{c T_R}{2}$ $c \rightarrow$ vel of EM wave $3 \times 10^8 \text{ m/sec}$ (Two way motion)

- r can be in km or miles or Nautical miles (NM)
1 NM = 1.854 km or 1854 meters.

- T_R is function of f_p (pulse repetition rate)

$$T_R = 1/f_p \quad \therefore r = \frac{c}{2f_p} \text{ as } f_p \uparrow \text{ or } T_R \uparrow \quad r \uparrow$$

- MUR: (max Unambiguous Range)

For correct estimation of range of a target, correct detection of echo from target is necessary. Hence sufficient time must be given to Rx for detection of echoes from desired target. Hence MUR is important.

$$\text{MUR (miles)} \leq \frac{\text{PRT}}{12.2}$$

PRT \rightarrow pulse repetition time or f_p

$$\text{But } \text{PRT} = \frac{\text{pulse width}}{\text{Duty cycle}}$$

Classification of Radars

- 1) pulse Radars — [Search Radars
Tracking — II —
Search while Tracking radars
- 2) Continuous wave Radars
- 3) Doppler Radars
- 4) Radar beacons
- 5) MTI - moving target indicators
- 6) ILS - Instrument Landing System
- 7) Advanced Radars.

Radar Range equation: (Unambiguous range)

- correct equation for range can be developed by incorporating parameters like transmitted power (P_t), gain antenna (A_p), antenna aperture (A_o), wavelength transmission (λ), c/s area of target etc.
- if P_t is the peak transmitted pulse power, according to Inverse square law power density at a distance 'r' from an isotropic antenna in (forward direction)

$$P_1 = \frac{P_t}{4\pi r^2} \quad \text{--- (1)}$$

Surface area of sphere

- we use directional antennas for TX & reception. Then power density is $P_2 = A_p P_1$ where A_p is the power gain of TX. ant
- $$= \frac{A_p P_t}{4\pi r^2} \quad \text{--- (2)}$$

- power intercepted by a target with a c/s area 'S' is

$$P_3 = P_2 S = \frac{A_p P_t S}{4\pi r^2} \quad \text{--- (3)}$$

$k = \text{Boltzmann's constant} = 1.38 \times 10^{-23} \text{ J/K}$

UNIT 4-5

- A portion of the signal will be reflected back in the form of an echo, hence power density at the radar receiver is $P_4 = \frac{P_3}{4\pi r^2}$ → Inverse square law in reverse direction

$$= \frac{A_p P_t S}{(4\pi r^2)^2} \quad \text{--- (4)}$$

- If A_0 is the capture or aperture area of receiving antenna at the RX, then power intercepted is

$$P_5 = P_4 A_0 = \frac{A_0 A_p P_t S}{(4\pi r^2)^2} \quad \text{--- (5)}$$

- For better efficiency same antenna is used for both TX & RX. A_p & A_0 are related by $A_p = 4\pi \frac{A_0^2}{\lambda^2}$ --- (6)

where $\lambda \rightarrow \text{wavelength of TX RF pulse}$

eg (6) in (5) gives $P_5 = \frac{P_t A_0^2 S}{4\pi r^4 \lambda^2}$ or $\frac{P_t A_p^2 \lambda^2 S}{(4\pi)^3 r^4}$ --- (7)

- except 'r' & 'P₅' all other parameters are constants.

For 'r' to be maximum, 'P₅' must be minimum (i.e. radar receives weakest echo from farthest target)

i.e. as $P_5 \rightarrow P_{\min}$ then $r \rightarrow r_{\max}$

∴ $r_{\max}(\lambda) = \left(\frac{P_t A_0^2 S}{4\pi \lambda^2 P_{\min}} \right)^{1/4}$ or $\left(\frac{P_t A_p^2 \lambda^2 S}{(4\pi)^3 P_{\min}} \right)^{1/4}$ --- (8)

→ Ideal radar range equation

Note: - No effect of noise, interference is considered.

- Most of the RX work under threshold, i.e. threshold limit is thermal noise limit.

$$\therefore P_{\min} = k T_0 \Delta f (F-1) \quad \text{--- (9)}$$

$k = \text{Boltzmann's constant} = 1.38 \times 10^{-23} \text{ J/K}$

CONST 4.5

$T_0 = 273 + t^\circ \text{C} \rightarrow \text{Temp in degree kelvin.}$

$\delta f = \text{bw of the RX, } F \rightarrow \text{Noise figure ratio} = \frac{(S/N)_{\text{out}}}{(S/N)_{\text{in}}}$

eq (9) in (8) gives

$$r_{\text{max}} = \left(\frac{P_t A_0^2 S}{4\pi \lambda^2 k T_0 \delta f (F-1)} \right)^{1/4} \quad \text{--- (10)}$$

Most of the time we use dish antenna, whose aperture area is given by $A_0 = \eta \frac{\pi D^2}{4}$ where η is the aperture efficiency ≈ 0.65 (typically)

$D = \text{mouth diameter of ant.}$
 $\rightarrow [A_0 = \eta A \rightarrow \text{physical area of the mouth}]$
 $\text{i.e. } \pi R^2 \text{ or } \frac{\pi D^2}{4}$

Assume $T_0 = 300 \text{ K}$ (room temp), substituting these & simplifying

$$r_{\text{max}} = 4.8 \left[\frac{P_t \overset{\text{watts}}{D^4} \overset{\text{m}}{S} \rightarrow \text{m}^2}{\underset{\text{Hz}}{\delta f} \underset{\text{m}}{\lambda^2} \underset{\text{ratio}}{(F-1)}} \right]^{1/4} \quad \text{--- (10A)}$$

\rightarrow Noise equation considering noise effects.

Factors affecting the range:- $r_{\text{max}}(10A)$

1. $r_{\text{max}} \propto \sqrt[4]{P_t} \rightarrow$ To double the range P_t must be \uparrow by 16 fold, uneconomical not suggested.

2. $r_{\text{max}} \propto \frac{1}{\sqrt[4]{P_{\text{min}}}} \rightarrow$ By $\downarrow P_{\text{min}}$, optimizing max possible

But $\downarrow P_{\text{min}}$ may result in less sensitivity of RX & RX may promote jamming & interference.

3. T for $\downarrow \lambda$

3. (a) $r_{max} \propto D$ (b) $r_{max} \propto \frac{1}{\sqrt{\lambda}}$

(UNIT 4-6)

- As $D \uparrow$, r_{max} also \uparrow \rightarrow this is better economical
Also as $\lambda \downarrow$ or $f \uparrow$ then also $r_{max} \uparrow$
- Here too via consequence, as Half power beam width of an antenna (Dirh) is $= \frac{70\lambda}{D}$, then

✓ as $D \uparrow$, B.W \downarrow - advantage in tracking radars
but - disadvantage in search \rightarrow
(Radar takes more time for search)

✓ as $\lambda \downarrow$ then also B.W \downarrow

- Hence compromise is necessary between D & λ for optimizing r_{max} for a specific appl.

4. $r_{max} \propto \sqrt{S}$ \rightarrow c/s area ^{of target} viewed from the ant is assumed to be constant, no control here

5. Ground interference (magnetic effect of ground) affects side lobe of the antenna, if ant is not properly elevated

Hence ✓ target detection becomes difficult - which may

be used as an advantage in military appl.. Aircrafts can fly at very low altitude without being detected by enemy radar - F16, Mirage, Stealth bomber etc.

✓ range will be affected.

Radar Beacons:-

UNIT 4-7

- This is active target detection.
- Beacon is a ~~small~~ target that contains a small radar set and an omnidirectional antenna.
- Signal processing is possible in beacon radar, hence range optimization possible.

Applications: * Satellite transponders - work under UHFOL

* In aircraft - Beacons - for comm with ATC

Altitudes - continuous monitoring of height of aircraft during flight. This is necessary to maintain constant pressure/humidity/temp/oxygen etc inside the craft at high altitudes.

* Military: IFF appn

(Identification of Friend or Foe)

In border areas there will be air zone intrusions from enemy aircraft. ATC sends a coded message to the target during the interrogation line. If it is a friend aircraft it replies, otherwise enemy aircraft does not reply. ATC takes suitable action.

Beacon range equation: on similar lines as we have derived radar range eqn., here also we can derive.

• Interrogation link range: $\left(\begin{array}{l} \text{Ground based RR is } TX_g \\ \text{Beacon radar is } RX_g \end{array} \right)$

Power intercepted by beacon antenna is

$$P_R = \frac{A_{PT} P_{ET} A_{OB}}{4\pi r^2} \quad (1)$$

Subscript T \rightarrow ground based RR, B \rightarrow Beacon radar

A for v

Radar Beacons:-

[UNIT 4-7]

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Beacon range equation: on similar lines as we have derived radar range eqn, here also we can derive.

• Interrogation line range: (Ground based RR is TXg, Beacon radar is RXg)

Power intercepted by beacon antenna is

$$P_B = \frac{A_{PR} P_{T AOB}}{4\pi r^2} \quad (1)$$

Subscript T \rightarrow ground based RR, B \rightarrow Beacon radar

3. \uparrow for v

UNIT 4-8

4) Subrogation,

- ↓

$$kT_0 \delta f \lambda^2 (F_B - 1)$$

- RA

$$kT_0\delta + \lambda^2(F_T - 1)$$

- $$\lambda \rightarrow \lambda_0$$

To \rightarrow τ_{avg} calculation $\propto \lambda$, $\delta f \propto \frac{1}{\lambda} \propto \frac{1}{\tau_{\text{avg}}}$

 $\lambda \rightarrow \lambda_D$

- $$\gamma_{\max, I} \gg \gamma_{\max, R}$$

beaver range.

Problem: In Moon mission, following data are available

UNIT 4-9

$$f = 2.5 \text{ GHz}, T_0 = 29^\circ \text{K}, \delta f = 5 \text{ kHz}$$

Ground Radar

Beacon radar

$$P_t = 0.5 \text{ MW (P}_{tT}\text{)}$$

$$50 \text{ W (P}_{tB}\text{)}$$

$$\text{Ant dia} = 64 \text{ m}$$

$$1 \text{ m}$$

$$13 \text{ dB (ratio 20)} (F_B)$$

$$\text{Noise Fg (F)} = 1.1 \text{ (ratio)} (F_T)$$

Assume $\eta = 0.65$ for dish antenna

- Is tracking possible?
- How range can be optimized
- What is the range of beacon?

Sol.

$$A_{OT} = \frac{(0.65)(\pi)(64)^2}{4} \text{ m}^2$$

$$A_{OB} = \frac{(0.65)(\pi)(1)^2}{4} \text{ m}^2$$

Substituting & simplifying

$$r_{\text{max}, I} =$$

$$\text{km}$$

$$r_{\text{max}, B} =$$

$$\text{km}$$

Inference:-

- $r_{\text{max}, I}$ extends beyond our solar system

(≈ 1.5 times that the dia of our solar system)

Hence tracking a beacon on moon, which is at a shorter distance is easily possible

- Beacon range is $r_{\text{max}, B} =$ km

Range optimization:

1. Increase ant dia - This is easily possible in ground based radar, but difficult in beacon.

Hence use folded umbrella type ant. on beacon.

~~Ant~~ Ant will be in folded form during launch & after it is in orbit, ant can be defolded.

2. Increase sensitivity of Rx

3. \uparrow for $\downarrow \lambda$

same period.

$$0.1 < \tau < 1 \mu\text{sec}, \quad T \gg \tau$$

T - Time

$$\boxed{\text{UNIT 4-10}}$$

Pulse Radar:

- Emits short & powerful pulses & receives echotypals during silent period.

UNITY-II 31

$0.1 < \tau < 1 \mu\text{sec}$, $T \gg \tau$.

T - Time period of pulses

τ - pulse width

$T \geq 1 \text{ msec}$

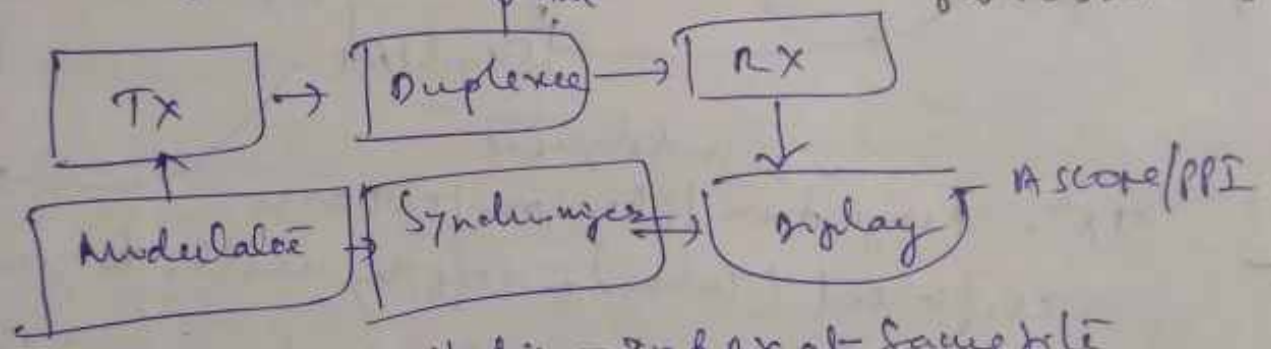
* PRF, T & τ have influence on the range i.e. minimal measuring range & MRR.

* Range resolution $\Delta R = 0.5 \tau c$

* B.W of radar $B = \frac{1}{\tau}$

$1 \text{ NM} = 1.852 \text{ km}$

$0 \sim 1.852 \text{ meters}$



* Pulse radar - Monostatic - Tx & Rx at same place
Bistatic - Tx & Rx at different place

* Appl:-
Duplexer has TR/ATR Switch

✓ Long distance comm

✓ ATC

✓ weather monitoring

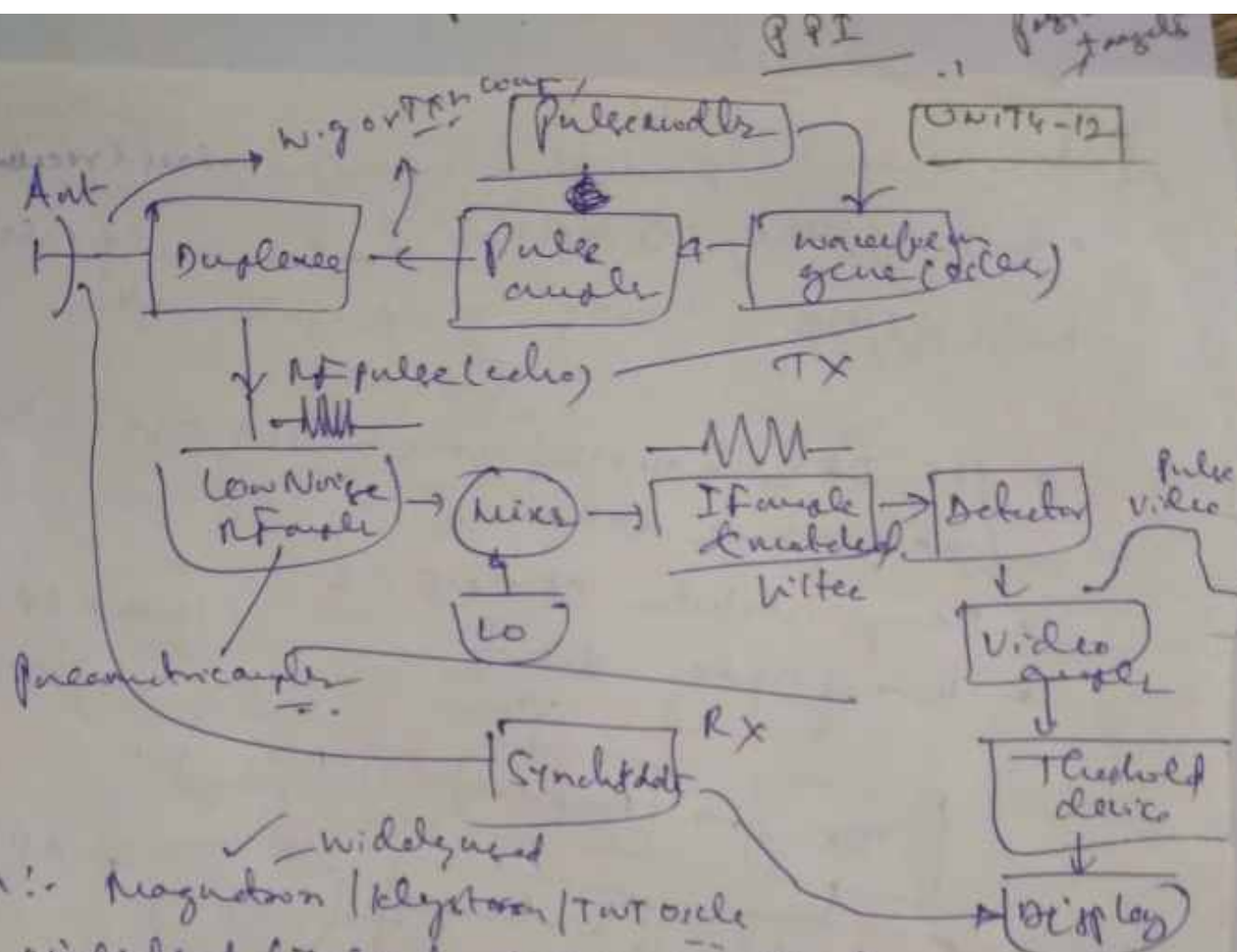
✓ Remote sensing

Isolation during TX of a pulse
TR Switch connect Tx to ant
& isolate Rx from stray signal
& during Rx mode, ATR
Switch deactivate & connect
echo to Rx isolating
Tx & Rx.

* If T is the round trip delay then

$R (\text{km}) \leq 0.15 TR (\mu\text{sec})$

or $R (\text{Nmi}) \leq 0.081 TR / \mu\text{sec}$



TX: - Magnetron / klystron / TWT or Cle
 circ pulsed (Turned on/off) by needles to generate a
 train of pulse. For $R \geq 100$ to 2000 mi in ATC requires
 1 MW peak power. For air surveillance IF of 100 MHz
 will be 300 to 600 MHz & B.W ≥ 1 MHz.

Ant will be a reflector / phased array ant.
 Prescribed reflector: $A_p \geq 10$ dB, $H_p B_w = 11$ to 19°

$\tau = 60$ sec, $R_{min} = 2$ NM to 2000 NM

$PRF = 300$ to 1200 Hz $N.F = 4.2$

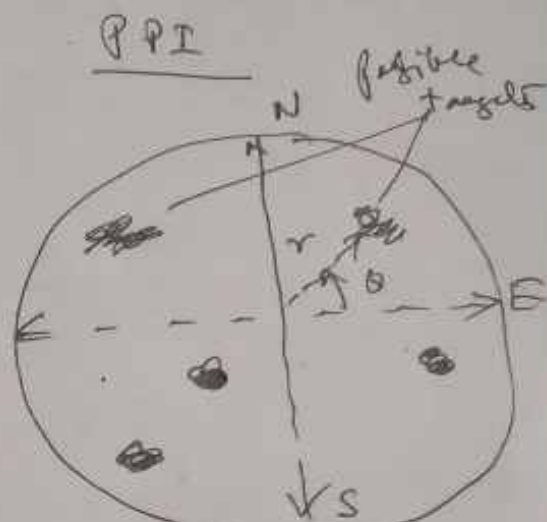
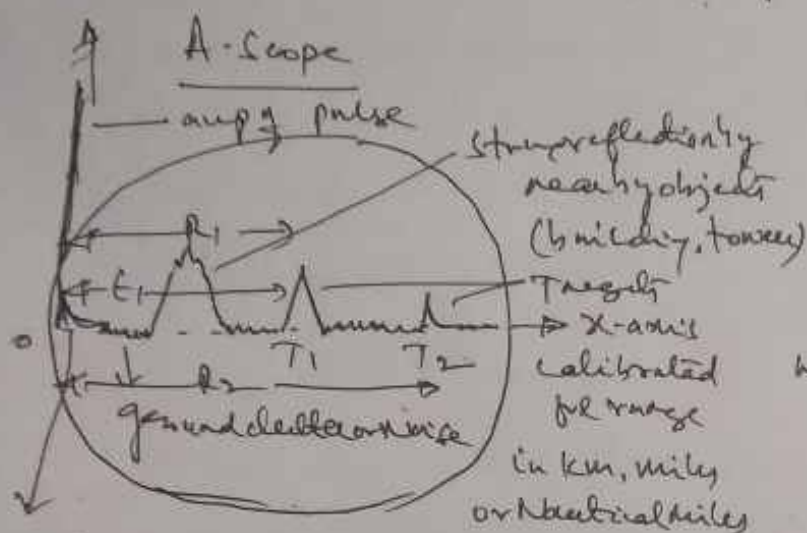
Advanced Radars:-

Display in Radars:-

A-Scope

PPI - plan position indicator

UNITY-13



Ref. blip (pulse) at t_1 in TX
 $t_1 \rightarrow$ sound trip delay of Target Comparison

1. Deflection Modulated CRT (Electrostatic defl.)
2. Range only
3. Spiral scanning
4. Short / long persistence phosphor coated screen
5. Used in tracking radars
6. X axis calibrated for range
 $y \rightarrow$ displays amplitude of echoes / pulses
7. Rectangular coordinates (distance only)

1. Intensity Modulated CRT (Magnetic defl.)
2. Range & direction
3. Conical scanning
4. Long persistence...
5. Used in search radars
6. X & Y axis for direction
7. Polar coordinates r/θ
 $r \rightarrow$ distance
 $\theta \rightarrow$ direction

... 012 fl-

[UNIT 4-14]

CW Radar (Continuous Wave Radar)

OPETE-15



Detects objects and able to measure radial vel of target using Doppler shift principle. (Austrian mathematician)

+ Doppler principle: Doppler shift in freq due to relative motion of

the target is
$$f_d = \frac{2 V_r \cos \theta}{\lambda} = \frac{2 V_r \cos \theta}{c} f_t$$

V_r — radial vel, $c = 3 \times 10^8$ /sec, f_t = Tx frequency

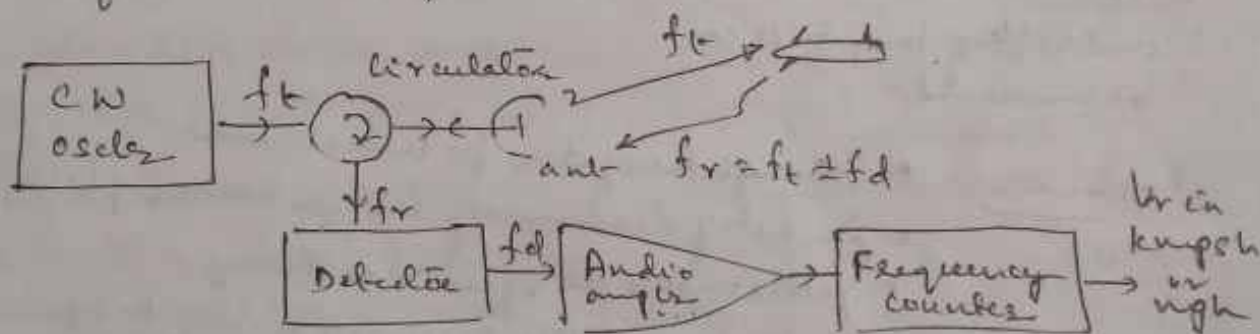
* for approaching targets $f_r = f_t + f_d$ & hence $f_r > f_t$

i.e Rx receives a higher freq

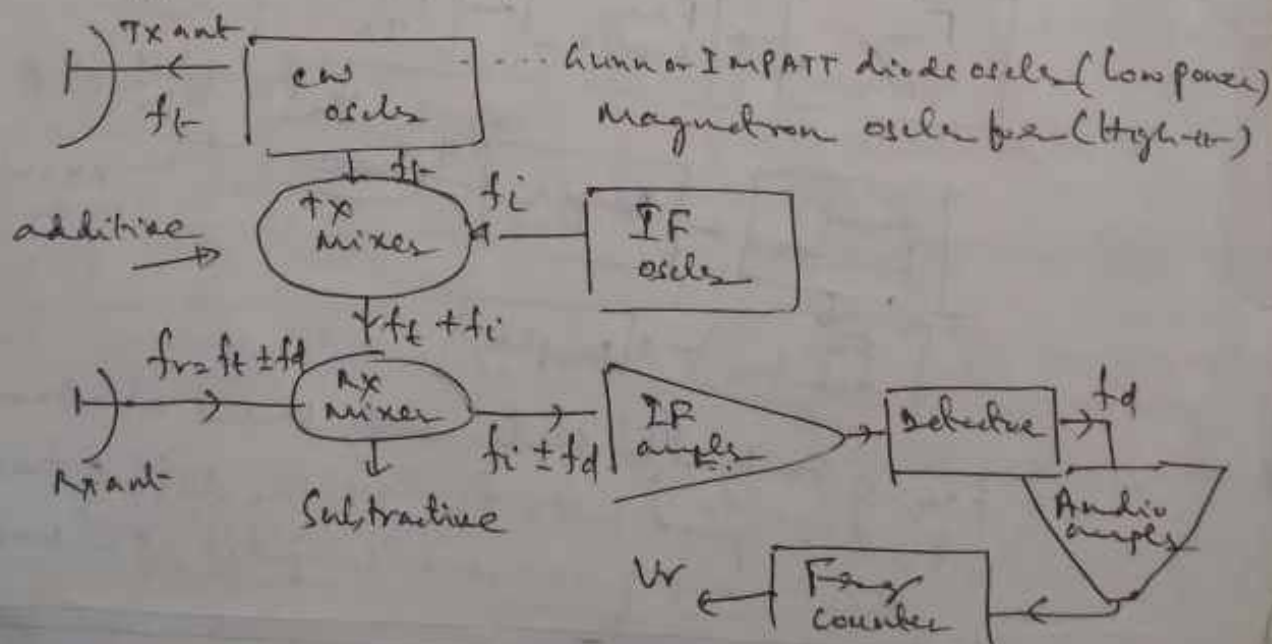
* for receding ——— $f_r = f_t - f_d$ ——— $f_r < f_t$

* if $\theta = 0 \rightarrow$ inline reception $f_{dmax} = \frac{2 V_r}{\lambda}$

if $\theta = 90 \rightarrow$ orthogonal ——— $f_d = 0$ No Doppler shift is observed

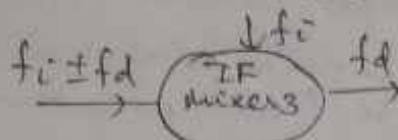


FM
Direct F_a noise of detector Rx is less sensitive. Hence we use 2 antennas, one for Tx & another for Rx, Also CHA Rx.



Instead of detector we can use 3rd mixer

UNIT 6 (2)



f_t will be around 300 MHz. Since TX is continuous & variable aimed.

Advantages

- * Accurate measurement is possible
- * Low TX power
- * Improved sensitivity
- * It can work in 2 ranges
- * If we preserve polarity of f_d we can determine direction of motion.
- * Elevation of target 'B' can be measured.

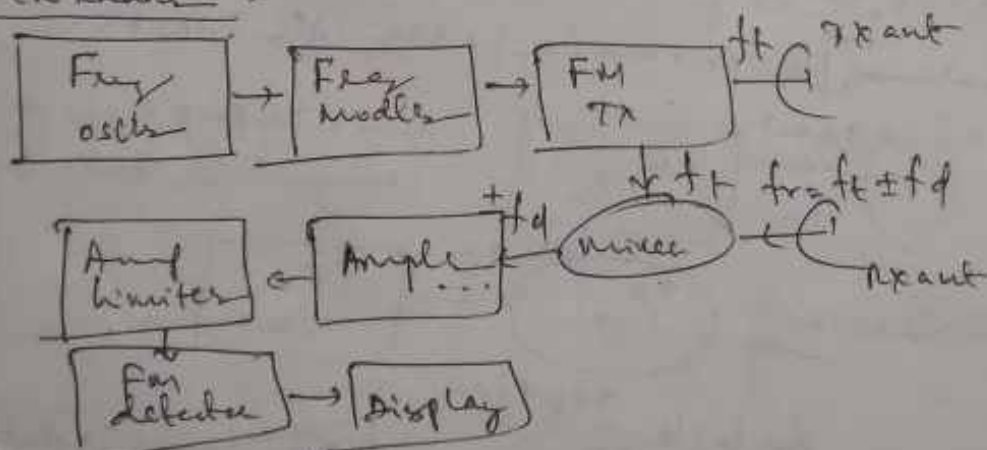
Disadvantages

- * As TX power is limited, range is also limited
- * Radar is confused if there are large no. of targets in the vicinity
- * Range determination not possible

Applications:-

- * Aircraft navigation for measurement
- * In Navy ships making vertical takeoff planes/helicopters
- * Police radars for speed monitoring
- * To indicate the presence of moving targets.

FM-CW Radar :-



FM modulator

- * ~~low~~ ^{low} power
- * low
- * as well as
- * dual

... f_{osc} of FM is continuously varied at a known rate & reflected by in comparison with f_t . By the diff of two frequencies slant range of target is computed at an instant where an echo is received

Advanced Radars:-

UNIT-25

MTI Radar principles:-

(Moving Target Indicators)

- Moving targets can be distinguished from stationary targets using this principle.
- Both A scope & PPI displays can be used.

(i) A-Scope display:-

- By observing Video/p (amp vs range) for successive sweep intervals.
- echoes from fixed targets appear at fixed time distance whereas moving target echoes amplitude & position vary in the window based on the Δt
- If we superimpose the off together, due to butterfly effect moving targets can be detected.

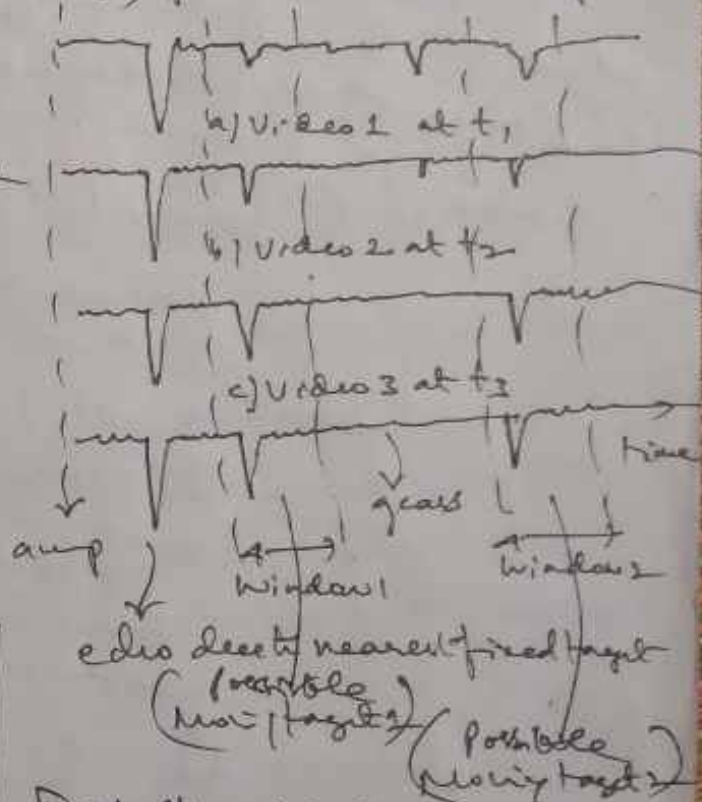


Fig 1- Phase detection for four successive time intervals.

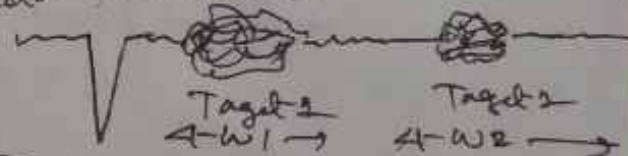
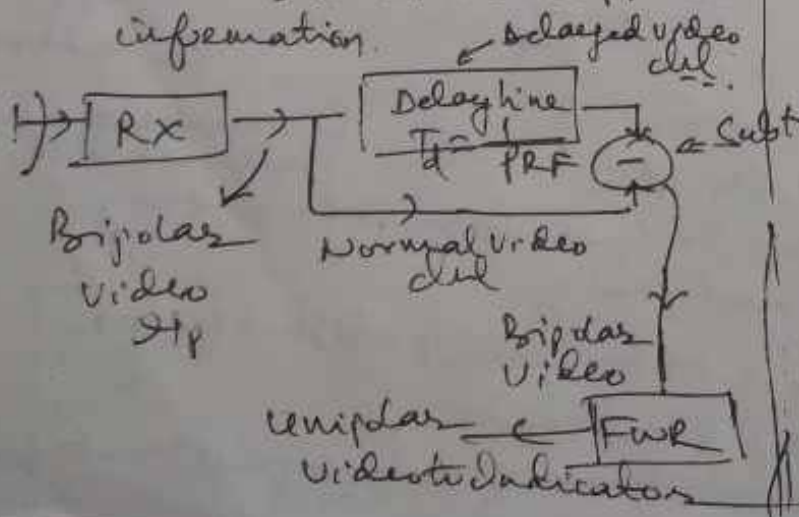


Fig 2- Superposition of many sweeps w/ butterfly effect.

(ii) PPI display:-

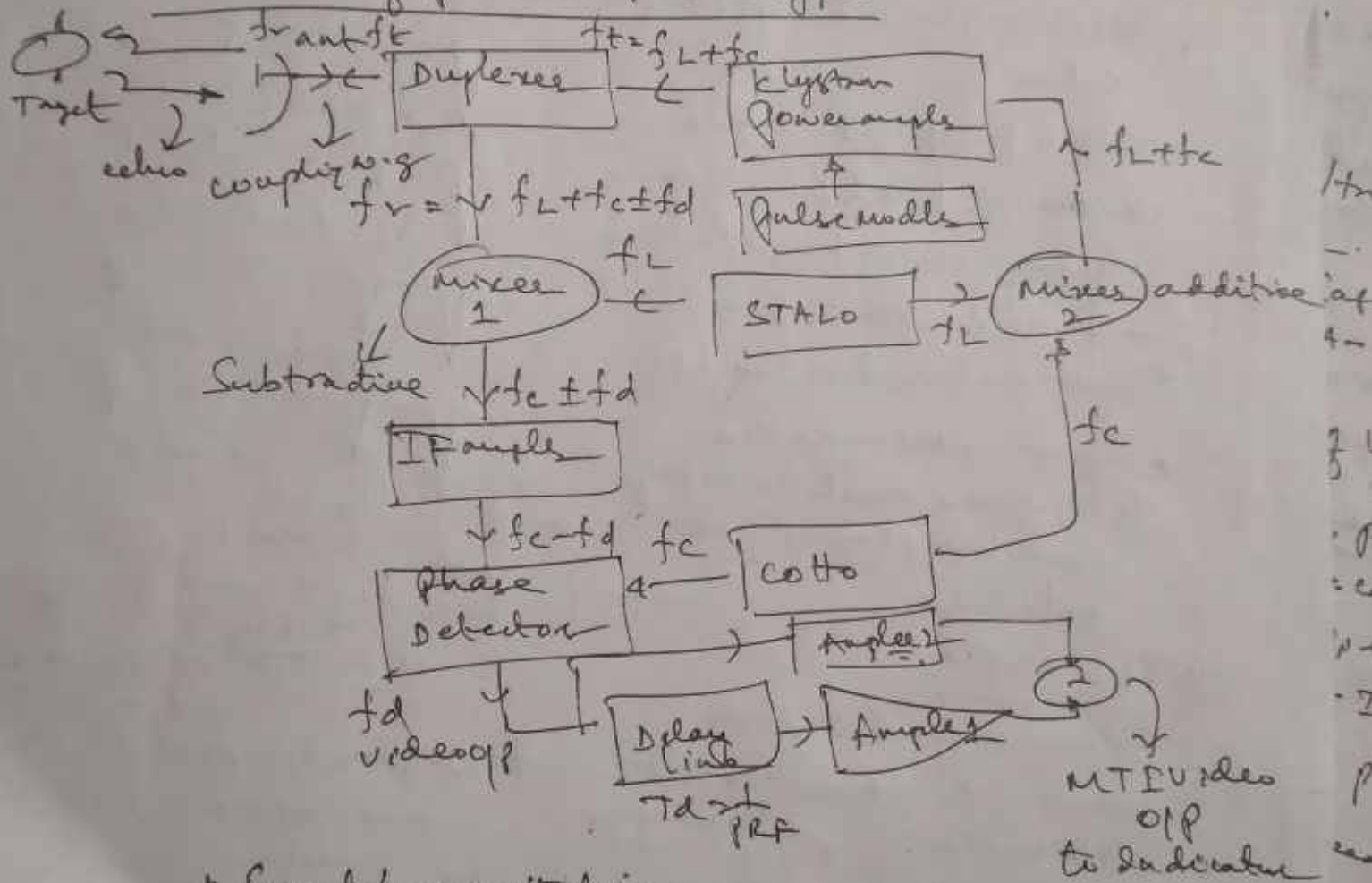
- Here we use delay line canceller to extract Doppler information.



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- Delay lines act as filters to block dc components of ρ from fixed targets & passes components of moving targets.
- In the o/p of Subtractor, echoes from fixed targets are cancelled, but gives moving targets residue.
- Fwd. converts bipolar to unipolar, which can intensity modulate the PPI screen.

MTI Radar of power Amplifier type:-



* Signal transmitted is

$$V_t = A_1 \sin 2\pi f_c t \quad (1) \quad \text{Since wave is sinusoidal and pulse modulated.}$$

↳ Includes losses in coupling

* Strengthening echo signal is

$$V_{echo} = A_2 \sin \left[2\pi (f_c \pm f_d) t - \frac{4\pi f_c R_0}{c} \right]$$

$R_0 \rightarrow$ Range of the target
 $A_2 \rightarrow$ amplitude of echo signal from target

Advanced Radars:-

UNIT-25

UNIT-19

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After heterodyning in mixer 1,

$$V_{diff} = A_3 \sin \left(2\pi f_d t - \frac{4\pi f_c R_0}{c} \right)$$

- * For stationary targets $f_d = 0$, $\Rightarrow V_{diff}$ does not vary w.r.t time, A_3 assumes values from $+A_3$ to $-A_3$ continuously.
- * For moving targets $f_d \neq 0$, V_{diff} varies w.r.t time which can be detected & displayed.

operation:-

1. Klystron power amplifier :- It will be a CFA / triode / tetrode TWT device amplifier. It is a stable amplifier with sufficient gain. It is also called MOPA - Master Oscillator Power Amplifier.

STATLO - stable local oscillator :- cascaded stages of varactor diodes which generate the RF freq.

COTLO - coherent oscillator. operates at IF, provides coherence between TX & RX signals for noise cancellation.

Mixer 1 & 2 are identical, since phase between I & Q are preserved. This helps in the detection circuit at IF & then RF.

Phase detector :- This is phase to amp converter, where phase difference between $f_1 = f_c$ & $f_2 = f_c - f_d$ are compared.

For stationary targets: $f_d = 0$, $f_1 = f_2$, hence difference is zero. If $f_1 > f_2$ or $f_1 < f_2$ then bipolar output is produced.

Blind speeds:- If V_r is such that it produces a phase change 2π radians between successive pulses (i.e. no phase change) then target appears as stationary & echoes from true target are blindly cancelled by MTI action. The radial vel corresponding to this situation is called blind speeds & must be avoided.

n th order blind speed is given by

$$V_{r_n} = \frac{PRF \cdot n \cdot \lambda}{2} \text{ where } n = 1, 2, 3, \dots$$

Remedy:- As λ is constant, for given order PRF is varied for successive pulses such that phase change produced by $V_r \neq 0$

Characteristics:-

- It is a form of pulsed radar

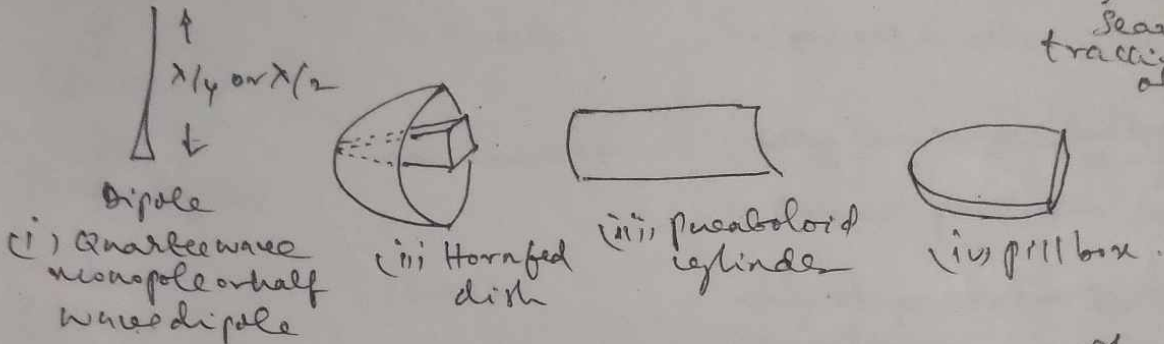
- Low PRF is used & hence no range ambiguity
- $R_{unambiguous} \approx \frac{c}{PRF}$
- V_r can be determined from fd.
- Eliminates clutter, echoes from stationary targets
- Identifies moving targets



Scanning principles:-

Scanning is nothing but searching a portion of a sky for the presence or absence of a target. Scanning depends on type of application like search radar and tracking radar. Search RR. has to locate the target & tracking radar establishes the different antennas are used for scanning.

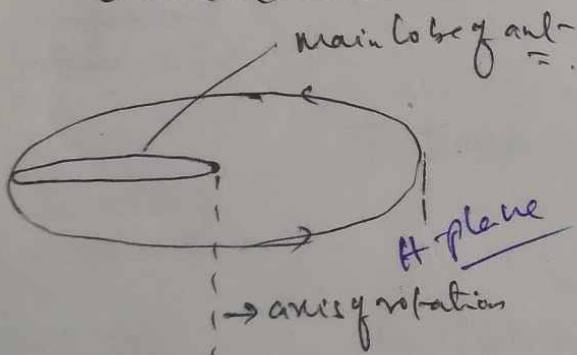
Search while tracking is other option.



- Adv:-
- * Reduced ant size & weight
 - * reduced wind loading
 - * Smaller drive motors etc

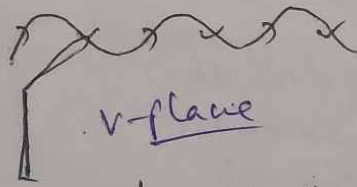
- * Steerable ant
- * reduced lossy

(a) Horizontal scan:-



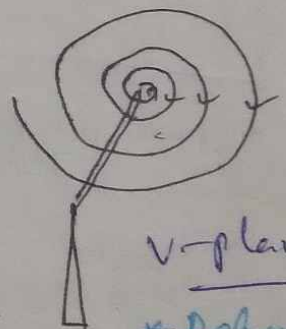
- * only one plane
- * used in ship to ship.

(b) nodding scan



- * tracking of slow moving bodies
- * airplane navigation

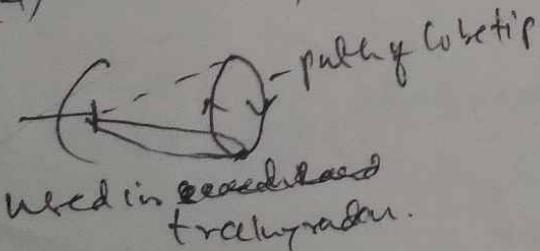
(c) Spiral scan



- * Detects both moving & stationary targets

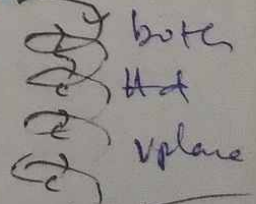
(a) to (d) are used in search RR.

(d) conical scanning



(d) Helical scan

- * search over complete hemisphere
- * Elevation ant in slowly raised while it rotates more rapidly in azimuth
- * 6 rpm, rise rate 20/min



above earth

10km-50km

0km-10km

UNIT 4-23

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~~Proffman~~

Radar Frequency bands and applications

Band	Freq	Appl
HF (High freq)	3-30 MHz	<ul style="list-style-type: none"> Coastal Radar OTH - over the horizon
P (Previous)	30-300 MHz	<ul style="list-style-type: none"> Conventional Radar
VHF (Ultra High freq)	300-1000 MHz	<ul style="list-style-type: none"> Very long range (Ballistic Missile early warning) Ground penetration (Remotely sensing)
L (Long)	1-2 GHz	<ul style="list-style-type: none"> Long range air to air control Surveillance
S (Short)	2-4 GHz	<ul style="list-style-type: none"> ATC terminal Long range weather Marine studies
C (Communication between X & S)	4-8 GHz	<ul style="list-style-type: none"> Satellite transponders (beacon) Weather
X (Secret) (military use)	8-12 GHz	<ul style="list-style-type: none"> Missile guidance Marine, weather, mapping Ground surveillance Airport Radar
Ku (under K band)	12-18 GHz	<ul style="list-style-type: none"> High resolution mapping Satellite telemetry
K	18-27 GHz	<ul style="list-style-type: none"> Meteorology - cumulonimbus clouds for storm prediction Police - speed monitoring Astronomy
Ka (above K band)	27-40 GHz	<ul style="list-style-type: none"> Mapping - flood, rain etc Surveillance Photoradar by NATO to detect submarine flares

Troposphere Mesosphere Stratosphere & Troposphere

↓ ↓ ↓

0 km - 10 km 10 km - 50 km 0 km - 10 km

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Q. 6.
~~max max~~ 40-200 Hz

V 60-75 MHz.

W 75-1104H2

UNIT 4-24

dictate communications

Atmospheric monitoring Co, ozone,
temperature profiles.

High resolution microscopical
observation, imaging, automation
appl.

Advanced Radars:-

UNIT-25



1) MST-Radar (Mesosphere Stratosphere & Troposphere Radar)

↓ ↓ ↓

50 km to 100 km 10 km - 50 km 0 km - 10 km

above earth

- * Designed to measure wind speeds & other atmospheric parameters like snow, rain, hail ^{clouds} etc.
- * mainly used for weather forecasting
- * High power TX are used to penetrate different layers of atmosphere (Pz 2.5 MW). Large aperture dish antennas used ($100m < D < 300m$). ~~Phase~~ ^{phased array} & 3 element Yagi-uda ^{dipoles} ant are also used
- * PRF is 62.5 Hz to 8 KHz

2) SAR (Synthetic Aperture Radar):

- * Dynamically move beam of antenna a/c an area to synthesise the target. provides excellent ^{image} L (elevation, range, az area & high resolution - Two dimensional
- * Can work with both stationary & moving targets
- * Inverse SAR (ISAR) is used to penetrate aircraft parameters from ground or ship.
- * can work under multiple reflection echoes.
- * used in weapon targeting, geological & mineral explorations; ^{missile guidance}
- * It can work in adverse weather conditions

3) LIDAR: Light Detection And Ranging):

- * used for measuring precise distance of targets in reference to it (can carry out surgical attacks)
- * uses large gain & low antenna
- * gives 3 dimensional image of targets

- # Provides fd on a extended scale to that we can recognize glowning targets
- # Antennae highly directional & devoid of side lobes

UNIT 4-26

2

4) ILS (Instrument Landing System):-

- # used in aerodromes for safe take off ^{or} landing of aircraft in adverse weather conditions.

- # It consists of three parts namely (i) localizer (ii) glide slope (iii) marker beacons

- (i) localizer: provides azimuth angle information to pilot & properly guides aircraft to runway
- (ii) glide slope: - usually placed on the side of runway.

It provides signal to the aircraft regarding vertical altitude during landing or take off. This is necessary to control vel of aircraft

- (iii) marker beacons: - provides information at the time movement either for defolding of wheels during landing or folding of wheels during take off.