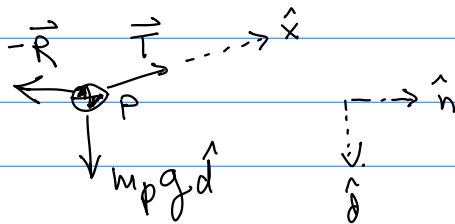


# Rocket Equations of Motion w/ Moving Center of Mass

## Assumptions.

1. Rocket motor is treated as a point mass.
2. Simple variable mass rocket model:  $\vec{T} = \dot{m} u e \quad \Sigma \vec{F} = m \vec{a}$

### FBD Motor



$$\vec{T} = \dot{m}_p u e \hat{x}$$

$$m_p \frac{d}{dt} \vec{V}^P = m_p \left( \frac{d}{dt} \vec{V}^P + \vec{\omega} \times \vec{V}^P \right) = \dot{m}_p u e \hat{x} + m_p g \hat{d} - \vec{R}$$

$$\vec{r}^{OP} = \vec{r}^{Oc} + \vec{r}^{cP}$$

$$\frac{d}{dt} \vec{r}^{OP} \equiv \vec{V}^P = \vec{V}^c + \vec{\omega} \times \vec{r}^{cP}$$

$$\frac{d}{dt} \vec{V}^P = \frac{d}{dt} \vec{V}^c + \vec{\omega} \times \vec{V}^{cP} + \vec{\omega} \times \vec{r}^{cP}$$

Choose  $b$  as working frame.

$$\frac{d}{dt} \vec{V}_b^P = \frac{d}{dt} \vec{V}_b^c + \vec{\omega}_b \times \vec{V}_b^{cP}$$

$$m_p \left( \frac{d}{dt} \vec{V}_b^c + \vec{\omega}_b \times \vec{V}_b^{cP} + \vec{\omega}_b \times \vec{V}_b^c + \vec{\omega}_b \times \vec{\omega}_b \times \vec{r}_b^{cP} \right) = \dot{m}_p u e \hat{x} + m_p g \hat{d} - \vec{R}_b$$

*small, ignore* (pointing to  $\vec{\omega}_b \times \vec{V}_b^{cP}$ )

*small, ignore* (pointing to  $\vec{\omega}_b \times \vec{\omega}_b \times \vec{r}_b^{cP}$ )

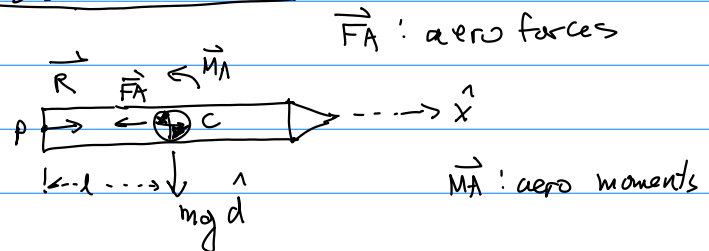
$\vec{\omega}_b$  would cause a coupling to be more difficult to solve EOM's  $R_b$  dominated by  $\dot{m}_p u e + m_p g \hat{d}$

$$m \left( \frac{d}{dt} \vec{V}_b^c + \vec{\omega}_b \times \vec{V}_b^c \right) = \vec{R}_b + m g \hat{d} + \vec{F}_A$$

$$\vec{J} \frac{d}{dt} \vec{\omega}_b + \vec{\omega}_b \times \vec{J} \vec{\omega}_b = \vec{M}_A + (-l \hat{x}) \times \vec{R}_b$$

↓  
solve for  $\vec{\omega}_b$  &  $\frac{d}{dt} \vec{V}_b^c$

### FBD Rocket



$$m \frac{d}{dt} \vec{V}^c = m \left( \frac{d}{dt} \vec{V}^c + \vec{\omega} \times \vec{V}^c \right) = \vec{R} + m g \hat{d} + \vec{F}_A$$

$$\frac{d}{dt} (\vec{J} \vec{\omega}_b) = \vec{J} \frac{d}{dt} \vec{\omega}_b + \vec{\omega} \times \vec{J} \vec{\omega}_b = \vec{M}_A + (-l \hat{x}) \times \vec{R}_b$$

note  $\vec{J} = \vec{J}_b^{C \leftarrow @ C}$   
 $\leftarrow$  in frame  $b$

$$\vec{R}_b \approx m_p u_e \hat{x} + m_p g \hat{d} - m_p (\vec{v}_b^c + \vec{\omega}_b \times \vec{v}_b^c)$$

$$m(\vec{v}_b^c + \vec{\omega}_b \times \vec{v}_b^c) = (m+m_p)g \hat{d} + m_p u_e \hat{x} - m_p (\vec{v}_b^c + \vec{\omega}_b \times \vec{v}_b^c) + \vec{F}_{AB}$$

$$m_T \vec{v}_b^c = (m+m_p)g \hat{d} + m_p u_e \hat{x} + \vec{F}_{AB}$$

$$\vec{v}_b^c = g \hat{d} + \frac{m_p u_e \hat{x}}{m_T} + \frac{\vec{F}_{AB}}{m_T} - \vec{\omega}_b \times \vec{v}_b^c$$

$$\vec{\omega}_b = J^{-1} \left[ \vec{M}_{AB} + \left[ m_p u_e \hat{x} + m_p g \hat{d} - m_p (\vec{v}_b^c + \vec{\omega}_b \times \vec{v}_b^c) \right] \times (r_{\hat{x}}) - \vec{\omega}_b \times J \vec{\omega}_b \right]$$

$$\vec{\omega}_b = J^{-1} \left[ \vec{M}_{AB} - \frac{m_p \vec{F}_{AB}}{(m+m_p)} \times (r_{\hat{x}}) - \vec{\omega}_b \times J \vec{\omega}_b \right]$$

↑  
makes sense, gravity never rotates vehicle,  
but aerodynamics can after c.g. moves.