

# Uniaxial Tensile Testing

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AE351– Experiments in Aerospace Engineering (Structures)

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*Note: Demonstration and Raw data collection are done by TA Mr. Vipin Chandra and he provided us related data in excel sheet.*

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## Objective

The objective of this test is to analyse the nature of the specimen by stress-strain curves using Young's modulus, yield stress, ultimate tensile stress, toughness and failure stress.

## Introduction and Theory

The uniaxial tension test provide a simple and effective way to characterize a material's response to loading. Therefore, to investigate material mechanics and gain experience in uniaxial testing, we performed tensile test on Aluminium alloy and calculated its Young's modulus, yield stress, ultimate tensile strength, and failure stress.

To find out specimen properties we will use the following equation.

### 1. Engineering strain

$$\varepsilon = \Delta L / L_o = (L_f - L_o) / L_o \quad (1)$$

$\Delta L$  = change in length due to applied load

$L_o$  = initial length of specimen

$L_f$  = final length of specimen

### 2. Engineering Stress

$$\sigma = F / A_o \quad (2)$$

$F$  = Applied load to specimen

$A_o$  = Initial specimen gauge section area normal to the loading direction.

The maximum stress and strain for the uniaxial tension tests were determined by finding the local maximum of the stress-strain curve before fracture of the samples occurred.

### 3. Ultimate Tensile Stress

$$\sigma_{uts} = F_{max} / A_o \quad (3)$$

The material initially behaves in a linear elastic manner: stress and strain are linearly related, and on unloading, the deformation is recoverable. The slope within the linear elastic regime is Young's modulus, or the ratio of the engineering stress to engineering strain in the axis:

### 4. Young's modulus

$$E = (\sigma / \varepsilon)_{\text{linear elastic region}} \quad (4)$$

An example of the engineering stress-strain curve for a typical engineering alloy is shown in Image 1. From it some very important properties can be determined. The elastic modulus, the yield strength, the ultimate tensile strength, and the fracture strain are all clearly exhibited in an accurately constructed stress strain curve.

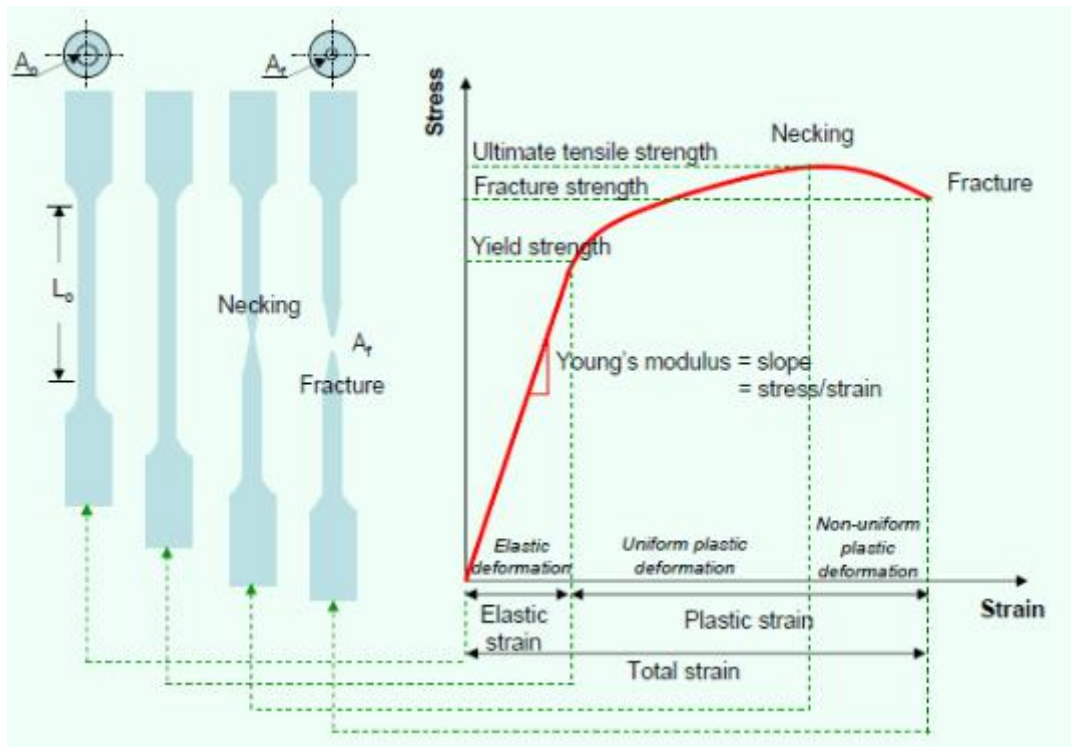


Image 1: stress vs. strain curve

**Elastic region:** The part of the stress-strain curve up to the yielding point. Elastic deformation is recoverable. In the elastic region stress and strain are related to each other linearly.  $E$  is Modulus of Elasticity or Young Modulus which is specific for each type of material.

Hooke's Law:  $\sigma = E\epsilon$

**Plastic region:** The part of the stress-strain diagram after the yielding point. At the yielding point, the plastic deformation starts. Plastic deformation is permanent. At the maximum point of the stress-strain diagram ( $\sigma_{uts}$ ), necking starts.

## Equipments

- Universal Testing Machine (UTM)

Manufactured by Tinius Olsen, the UTM is a machine used for performing stress/strain tests on materials. Although in the experiment we used it for uniaxial testing, it can perform multi-component stress analysis as well.

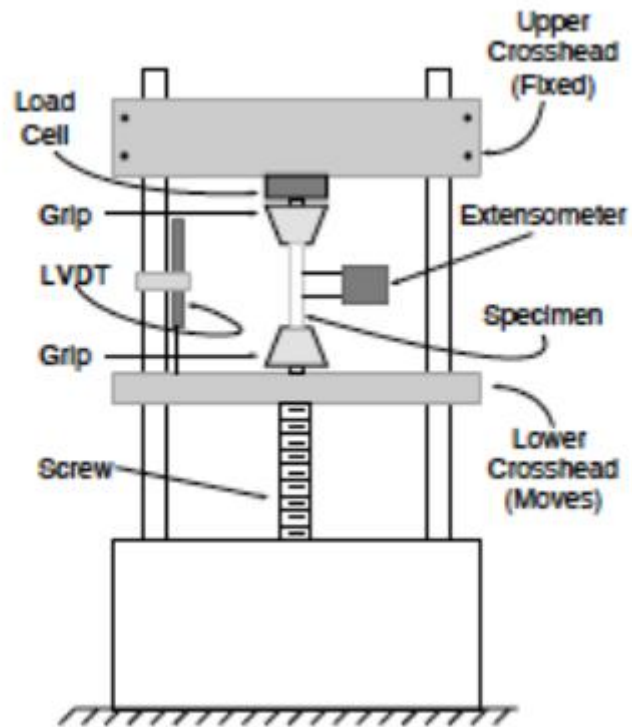


Image 2: Universal Testing Machine

## ● Computer System

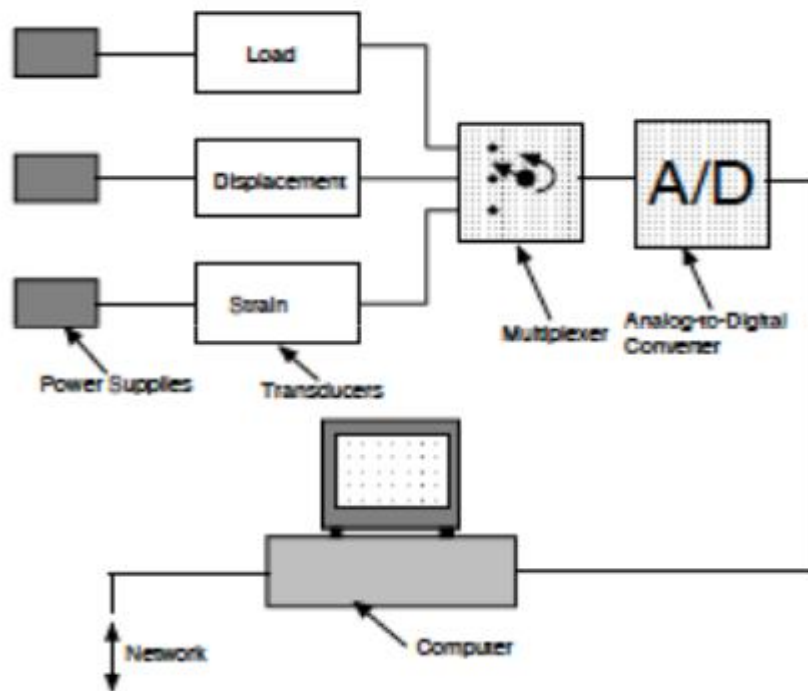
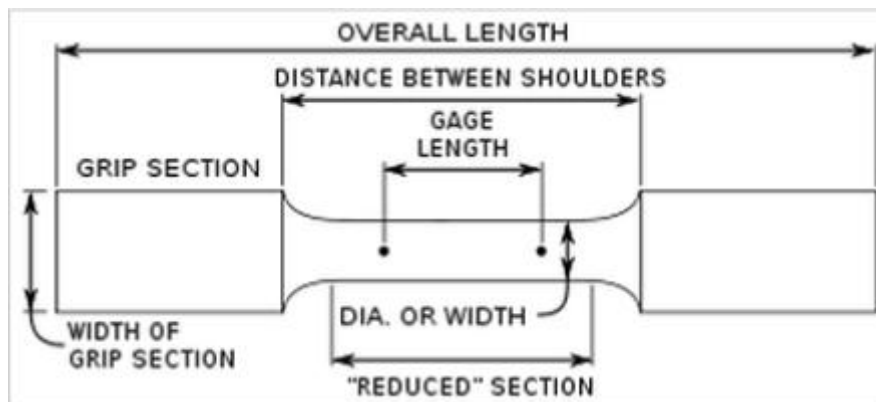


Image 3: Data acquisition model

## ● Extensometer

An extensometer is a device that is used to measure changes in the length of an object. It is useful for stress-strain measurements and tensile tests.

- Dog-bone shape Aluminium alloy (6000 series) specimen



*Image 4: Dog-bone shape specimen*

## Procedure and Measurements

Initial measurement of Dog-bone shape specimen:

Material	initial gauge section diameter ( $D_0$ )	initial gauge section length ( $L_0$ )
Aluminium Alloy	6mm	55mm

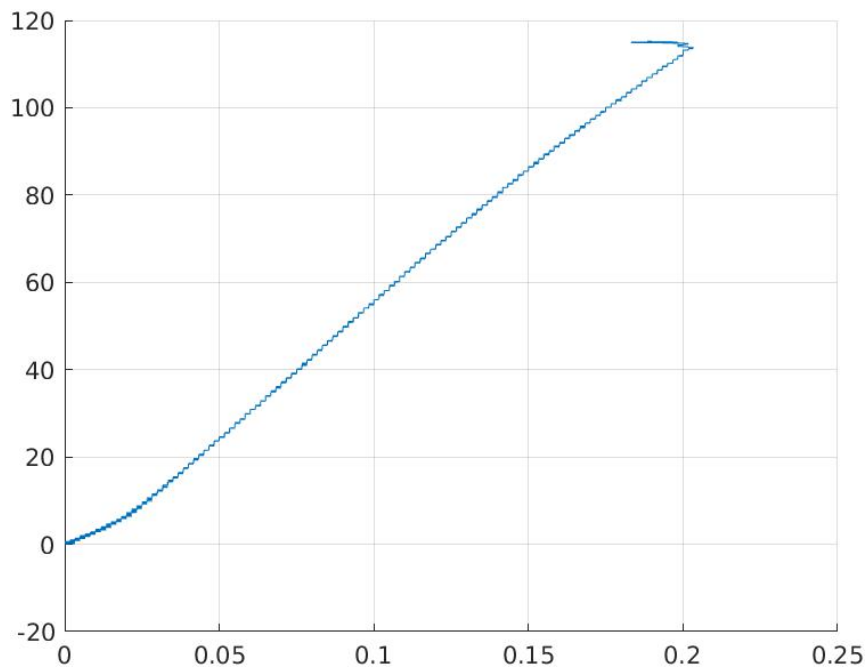
*Table 1: Initial gauge dimension*

Switched on the 15 kN Tinius Olsen universal testing machine (UTM). Then we hold the dog-bone shaped test specimen at the UTM grips and carefully mounted 25 mm extensometer in between the gage length region of the specimen. Loaded the specimen in displacement control mode at the speed of 1mm/min.

Then we recorded data with help of software till the extensometer strain value reached to 0.2%. Then we removed the extensometer and continue loading the specimen until specimen fractured. Specimen broke, but not exactly at middle of length.

# Results & Discussion

## 1. Data Presentation



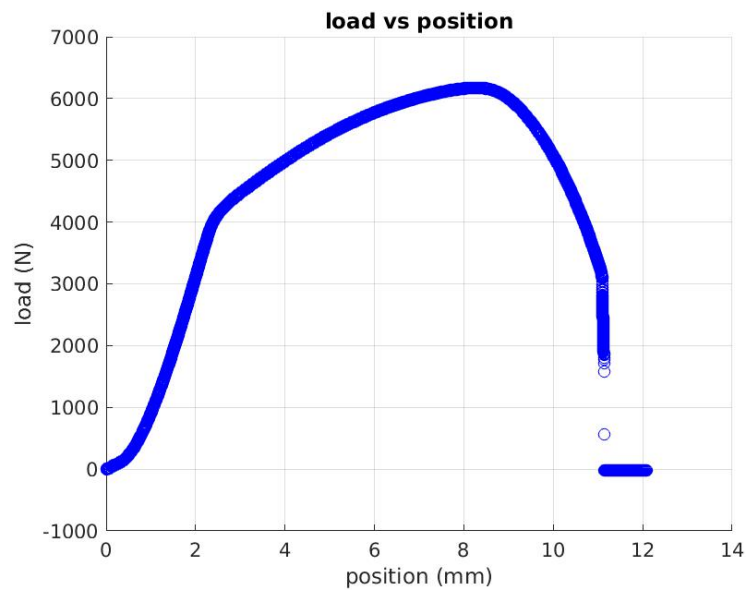
Graph 1 : stress vs extensometer strain

### Experimental Values from stress-strain(Extensometer) curve(graph1)

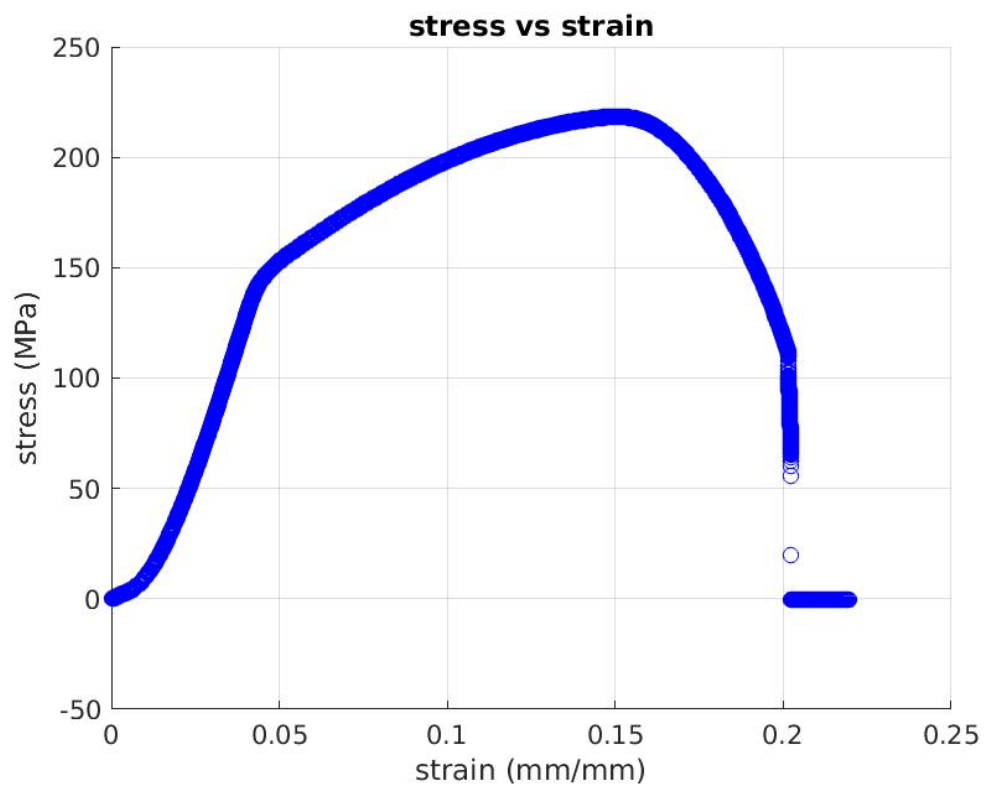
Young's Modulus, E : 62.42GPa

Young's Modulus (E)	Ultimate tensile stress ( $\sigma_{uts}$ )	Yield stress ( $Y_f$ )	Failure stress ( $\sigma_{failure}$ )
62.42GPa	218.4MPa	144MPa	60.41MPa

Table 2 : final specimen properties



Graph 2: load vs position



Graph 3: stress vs strain

### Experimental Values from stress-strain curve(graph3)

Ultimat tensile stress,  $\sigma_{uts}$  : 218.4 MPa

Yield stress,  $Y_f$  : 144Mpa

Failure stress,  $\sigma_{failure}$  :60.41MPa

## **2. Discussion and error analysis**

### **Theoretical Value of Aluminium**

Ultimate tensile stress : 310 MPa

Yield stress : 276 MPa

Failure stress : 96.5MPa

Young's Modulus :70GPa

### **Error Calculation**

**% Error in Young Modulus =  $(70-62.42)/70 = 10.85\%$**

**% Error in Yield stress =  $(276-144)/276 = 47.8\%$**

**% Error in Failure stress =  $(96.15-60.41)/96.15 = 37.17\%$**

**%Error in Ultimate stress=  $(310-218.4)/310 = 29.54\%$**

## **Conclusion**

There is %error in yield stress(47.8%), young modulus(10.85%), ultimate tensile stress(29.54%) and failure stress(37.17%) between theoretical and experimental values, this is due to specimen impurities, temperature variation and some other mechanical handling.

## **Appendix**

[Raw Data](#)