## FACULTY OF APPLIED SCIENCE AND ENGINEERING UNIVERSITY OF TORONTO

MIE 404F Final Exam December 14, 2001

Examiner: Prof. R. Ben Mrad

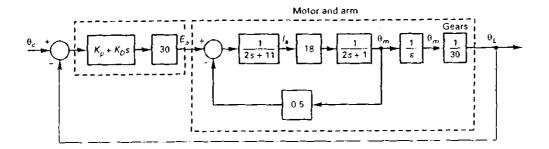
## General Comments:

- 1. You have 2.5 hours to complete the exam.
- 2. The maximum number of points you can get on the exam is 100 points.
- 3. Write your name and student number on the front page to ensure proper identification.
- 4. Calculators are allowed. The exam is open textbook, and open lecture and tutorial notes and handouts. No additional material is allowed.
- 5. The exam contains 9 pages.
- 6. Note that a quadratic equation of the form  $s^2 + bs + c = 0$  admits, if  $b^2 4c > 0$ , two solutions of the form:  $s_{1,2} = \frac{-b \pm \sqrt{b^2 4c}}{2}$

Name:
Student Number:
Signature:

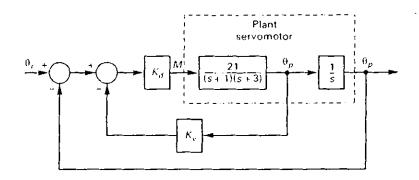
Problem 1 (25%): Shown below is the control system for one joint of a robot arm. The controller is a PD controller.

- (a) Determine the plant transfer function  $\frac{\Theta_L(s)}{E_a(s)}$ .
- (b) Find conditions on the controller gains  $K_p$  and  $K_D$ , with these gains positive, such that the closed-loop system is stable.
- (c) Let  $K_D = 1$ . Fing  $K_p$  such that the system will have a steady-state oscillation once a step input is applied to it, and find the period of such an oscillation.

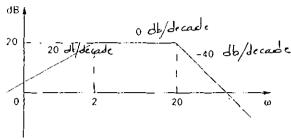


Problem 2 (25%): The block diagram below represents the servo control system for one of the axes of a digital plotter.

- (a) Let  $K_v = 0$ . Sketch the root locus for this system as the positive gain  $K_d$  varies from zero to infinity. In drawing the root locus make sure you determine:
  - 1. arrival and departure angles if there are any.
  - 2. angles of the asymptotes if there are any.
  - 3. imaginary axis crossing points if there are any.
  - 4. break-away/break-in points if there are any.
- (b) Remove the rate feedback path  $(K_v = 0)$  and replace  $K_d$  with a PD controller, with the transfer function  $G_c(s) = K_p + K_D s$ . Calculate the controller gains  $K_p$  and  $K_D$  to give a characteristic equation of the overall closed loop PD controlled system to have roots at  $s = -1 \pm j$ .

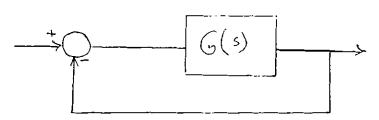


Problem 3 (15%): Consider the magnitude plot of the straight line Bode diagram for the transfer function G(s). Find G(s).



Problem 4 (25%): The unity feedback system shown below has a plant transfer function:

$$G(s) = \frac{Ks}{(0.1s+1)(s+2)}$$



- 1. Draw the Bode plot of the function G(s) with K=30.
- 2. What are the approximate values of the gain and phase crossover frequencies?
- 3. What are the approximate values of the gain and phase margins of the system? Is the closed-loop system stable?
- 4. Is the closed-loop system stable for the value of K=100 ? (Explain)

Problem 5 (10%): Shown below is the block diagram of the lateral control system of an aircraft landing system. The output Y(s) is the aircraft lateral position, and the input  $Y_c(s)$  is the desired aircraft position. The aircraft position is determined by radar, which is modeled as unity gain with an added noise signal  $D_r(s)$ . This noise signal represents the inaccuracies of the radar and will be neglected in the following analysis  $(D_r(s) = 0)$ . The signal  $D_w(s)$  represents the wind disturbance on the aircraft.

The transfer functions  $G_p(s)$  and  $G_d(s)$  each have two poles at s=0 in the actual aircraft models. Suppose that the wind on the aircraft is constant, such that  $d_w(t)$  is modeled as a constant signal. A system design criterion requires that steady-state effects of the wind disturbance alone on the aircraft are zero. Assuming the system is stable, what form should  $G_c(s)$  take such that this criterion is satisfied?

