## Tutorial for the simulation of wall-bounded turbulent channel flow

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### 1. Introduction

This tutorial illustrates the step by step usage of Fourth-order **Channel3d** for wall-bounded turbulent channel flow at  $Re_{\tau} \approx 180$ . Here, periodic boundary conditions are used in streamwise (x) and spanwise (z) directions, while no-slip conditions are adopted for top and bottom walls in normal direction (y). The computational sketch is shown is Fig.1.

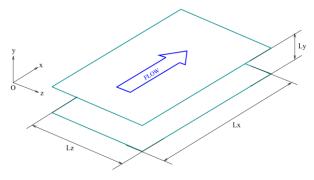


Fig.1. Sketch of computational domain for wall-bounded turbulent flows

The simulation has been initiated with a laminar parabolic Poiseuille velocity profile, plus a divergence-free large-amplitude sinusoidal perturbation:

$$u_0 = r_d u_b \overline{y} \exp\left(-\frac{9}{2} \overline{y}^2\right) \cos\left(\frac{\pi z}{100h} \operatorname{Re}_\tau\right) + 3u_b \left(\overline{y} - \frac{1}{2} \overline{y}^2\right)$$
 (1a)

$$w_0 = r_d u_b \overline{y} \exp\left(-\frac{9}{2} \overline{y}^2\right) \sin\left(\frac{\pi x}{250h} \operatorname{Re}_{\tau}\right)$$
 (1b)

$$v_0 = 0 \tag{1c}$$

where  $\overline{y}$  is the dimensionless y coordinate defined as  $\overline{y} = \min(y, 2h - y)/h$ , and h denotes half channel height.  $r_d$  is a random number, ranging from 0.9 to 1.1, to further disturb the initial flow. The flow is driven by a uniform pressure gradient, which varies in time to maintain a constant mass flow rate. After several time unit, the laminar profiles transit to turbulent ones, and the pressure gradient begins to fluctuate near an almost constant value.

#### 2. Preparation

The friction Reynolds number is defined as  $Re_{\tau} = u_{\tau}h/\nu$ , where  $\nu$  is the fluid kinematic viscosity, and  $u_{\tau}$  is friction velocity:  $u_{\tau} = \sqrt{(\nu \partial \overline{u}/\partial y)_{wall}} = \sqrt{(-dp/dx)h}$ .  $Re_b$  denotes the bulk Reynolds number:  $Re_b = u_b h/\nu$ , where  $u_b$  is the bulk velocity. For wall-bounded turbulence,  $Re_b$  and  $Re_{\tau}$  have the following empirical relationship:

$$Re_{\tau} = 0.1538 Re_{b}^{0.8877} \tag{2}$$

Eq.(2) is regressed from the data provided by Lee & Moser [1], and the regression result is also shown in Fig.2. If we want to simulate the case  $Re_{\tau} \approx 180$ , we can use Eq.(2) to get an approximate value for  $Re_b$ , i.e.  $Re_b \approx 2860$ . If we further set  $u_b = 2/3$  and h = 1, we can get the corresponding viscosity:  $v = (u_b h)/Re_b = 2.3310 \times 10^{-4}$ .

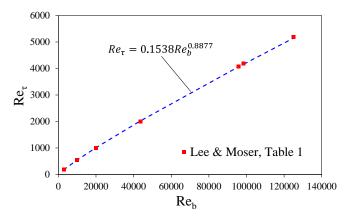


Fig.2. Relationship between bulk and friction Reynolds number

### 3. Parameter setting

The computational domain sizes are set to  $4\pi h$ , 2h,  $2\pi h$  in x-, y- and z-directions respectively. Details about the flow parameters and grid mesh are listed in Table 1. The input parameter text can be found in file ./doc/TurbulentChannel\_4th\_direct.prm. The illustrations for every input parameters can be found in file ./doc/channel4th\_prm.md.

#### Table 1

List of parameters for turbulent channel flow cases. Here  $N_x$ ,  $N_y$ , and  $N_z$  are the number of grid points in streamwise, wall-normal and spanwise directions, respectively.  $\Delta x^+$  and  $\Delta z^+$  denote grid spacings in wall-parallel directions.  $\Delta y_w^+$  and  $\Delta y_c^+$  represent the grid spacing at the wall and the centerline in wall-normal direction respectively.  $Tu_\tau/h$  is the time averaging window without transition.

Flow Case	$Re_{ au}$	Re <sub>b</sub>	$N_x$	$N_y$	$N_z$	$\Delta x^+$	$\Delta y_w^+$	$\Delta y_c^+$	$\Delta z^+$	$Tu_{\tau}/h$
Cha180	182.3	2860	384	192	256	5.97	0.0244	2.98	4.48	20.40

# 4. Running the code

If the executable file ./channel4th and ./Tool/interpolateField/interpolateField has not been compiled, you should compile it firstly, following the guidance written in ./README.md, Installation part. The simulation results and restart files will be stored in the folder ./CFD/Results/and ./CFD/Restart, respectively.

Now, you can run this test case directly by typing:

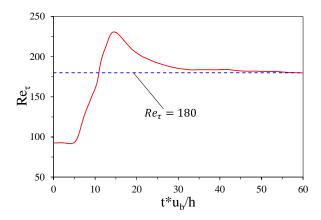
- 1 chmod a+x ./channel4th
- 2 mpirun –n 8 ./channel4th ./Input/TurbulentChannel\_4th\_direct.prm

Alternatively, we can run the case on a coarse grid firstly, and then interpolate the resulting flow field to the fine mesh. If so, we can do as follows.

### 4.1 Running on the coarse mesh

Firstly, we run on a coarse grid where  $N_x \times N_y \times N_z = 180 \times 132 \times 120$ . Here we choose an arbitrary coarse mesh configuration. The evolution of  $Re_{\tau}$  for coarse mesh is shown in Fig.3.

1 mpirun –n 8 ./channel4th ./Input/TurbulentChannel\_4th\_1.prm



**Fig.3.** Time evolution of the friction Reynolds number  $Re_{\tau}$  for coarse mesh

#### 4.2 Interpolating flow field from coarse mesh to fine mesh

Now, we need to interpolate flow field from coarse mesh to fine mesh, using the executable file ./Tool/interpolateField/interpolateField:

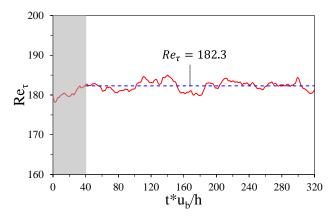
1 cd./Tool/interpolateField/
2 chmod a+x interpolateField
3 mpirun –n 8 interpolateField interp1.prm
4 cd../../

After that, New restart file (./CFD/RestartForCha180\_2\_0000003000) will be generated.

#### **4.3 Running on the fine mesh**

Lastly, we run on the fine mesh where  $N_x \times N_y \times N_z = 384 \times 192 \times 256$ . The evolution of  $Re_{\tau}$  for fine mesh is shown in Fig.4.

### 1 mpirun –n 8 ./channel4th ./Input/TurbulentChannel\_4th\_2.prm



**Fig.4.** Time evolution of the friction Reynolds number  $Re_{\tau}$  for fine mesh. Gray area indicates the adjustment stage from coarse mesh to fine mesh, and the corresponding time interval will be excluded from the statistic calculation.

#### 5. Visualization

One can use Paraview for visualization. An interface file ended with the suffix .xmf is included in the folder ./CFD/Results/ (./CFD/Results/VisuForCha180\_2\_.xmf here), and this file can be read by Paraview directly.

### 6. Post-processing

Post-processing might be the most interesting part for wall-bounded turbulence. Here I will illustrate how to get the desired statistic data using **Channel3d**.

#### **6.1 Data storage**

The canonical statistic data include mean velocity profile, turbulent intensity, Reynolds stress, vorticity correlation, skewness, flatness, turbulent budget, energy spectra and so on.

If a, b, c, and d denote four variables. The second-order correlation can be defined as:

$$\langle a'b' \rangle = \langle \lceil a - \langle a \rangle \rceil \lceil b - \langle b \rangle \rceil \rangle \tag{3}$$

While Eq.(3) can not be used directly, due to the fact that the exact  $\langle a \rangle$  and  $\langle b \rangle$  are not available until end of the simulation. So alternately, we can use the following expression instead:

$$\langle a'b'\rangle = \langle ab\rangle - \langle a\rangle\langle b\rangle \tag{4}$$

Similarly, the third-order and fourth-order correlations read:

$$\langle a'b'c'\rangle = \langle abc\rangle - \langle a\rangle\langle bc\rangle - \langle b\rangle\langle ac\rangle - \langle c\rangle\langle ab\rangle + 2\langle a\rangle\langle b\rangle\langle c\rangle$$
(5)

$$\langle a'b'c'd' \rangle = \langle abcd \rangle - 3\langle a \rangle \langle b \rangle \langle c \rangle \langle d \rangle + \langle a \rangle \langle b \rangle \langle cd \rangle + \langle b \rangle \langle c \rangle \langle da \rangle + \langle c \rangle \langle d \rangle \langle ab \rangle + \langle d \rangle \langle a \rangle \langle bc \rangle + \langle a \rangle \langle c \rangle \langle bd \rangle + \langle b \rangle \langle d \rangle \langle ac \rangle - \langle a \rangle \langle bcd \rangle - \langle b \rangle \langle cda \rangle - \langle c \rangle \langle dab \rangle - \langle d \rangle \langle abc \rangle$$

$$(6)$$

For instance, if we want to determine pressure strain term  $\pi_{1,2} = \langle p'(\partial u'/\partial y + \partial v'/\partial x) \rangle$ , we can further expand it as follows:

$$\pi_{1,2} = \left\langle p \left( \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) \right\rangle + \left\langle p \right\rangle \left[ \frac{\partial \left\langle u \right\rangle}{\partial y} + \frac{\partial \left\langle v \right\rangle}{\partial x} \right] = \left\langle p \left( \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) \right\rangle + \left\langle p \right\rangle \frac{\partial \left\langle u \right\rangle}{\partial y} \tag{7}$$

So we need to calculate and store  $\langle p(\partial u/\partial y + \partial v/\partial x) \rangle$ ,  $\langle p \rangle$ , and  $\langle u \rangle$  in advance. Table 2 lists the raw averaged date contained in file ./CFD/Results/stats\*. It is also worth nothing that due to the use of fully staggered grids, so averaged variables (e.g.  $\langle u \rangle$ ) are placed in the y-center layer, while others (e.g.  $\langle v \rangle$ ) are stored in the y-node layer. The definition of turbulent budget terms are listed in Appendix A.

The raw averaged 1D energy spectra data are also stored in the file ./CFD/Results/Spec\*.

#### **6.2 Statistic results calculation**

The final statistic results can be calculated by the post-processing Matlab code, ./Tool/readStat\_CH.m. If Matlab is not available, one can use Octave instead:

- 1 cd./Tool/
- 2 octave readStat CH.m
- 3 cd ...

After running this Matlab code, the final statistic results will be generated in the folder ./StatOut/.

**Table 2**List of the raw averaged date contained in file ./CFD/Results/stats\*. Here 'location' column indicates where the variable is placed. 'C' means 'Center', and 'N' stands for 'Node'.

id	variable name	location	id	variable name	location
1	<u></u>	С	19	<uuv></uuv>	С
2	<v></v>	N	20	<www></www>	С
3	<w></w>	С	21	<uuuv></uuuv>	С
4		С	22	<vvvv></vvvv>	N
5	<uu></uu>	С	23	<www>&gt;</www>	С
6	<vv></vv>	N	24	$\langle p \partial u / \partial x \rangle$	С
7	<ww></ww>	С	25	$\langle p \partial v / \partial y \rangle$	С
8	<pp></pp>	С	26	$\langle p \partial w / \partial z \rangle$	С
9	<uv></uv>	С	27	$\langle p(\partial u/\partial y + \partial v/\partial x) \rangle$	N
10	<vw></vw>	С	28	$< u \nabla^2 u >$	С
11	<uw></uw>	С	29	$< v \nabla^2 v >$	N
12	<up></up>	С	30	$< w \nabla^2 w >$	С
13	<vp></vp>	С	21	$\partial u \partial v \partial u \partial v$	С
14	<wp></wp>	С	31	$<\frac{\partial u}{\partial x}\frac{\partial v}{\partial x} + \frac{\partial u}{\partial y}\frac{\partial v}{\partial y}>$	
15	<uuv></uuv>	С	32	$<(\partial u/\partial z)(\partial v/\partial z)>$	N
16	<vvv></vvv>	N	33	$<\omega_{\rm x}\omega_{\rm x}>$	N
17	<wwv></wwv>	С	34	$<\omega_{\rm y}\omega_{\rm y}>$	С
18	<uvv></uvv>	С	35	$<\omega_z\omega_z>$	N

# 7. Statistic results validation

This section compare the statistic results with Lee & Moser [1], and Vreman & Kuerten [2] to for validation.

#### 7.1 Basic profiles

Fig.5 shows some turbulent statistic profiles, i.e. mean velocity, velocity fluctuations, Reynolds stress, vorticity fluctuations, mean pressure, pressure fluctuation.

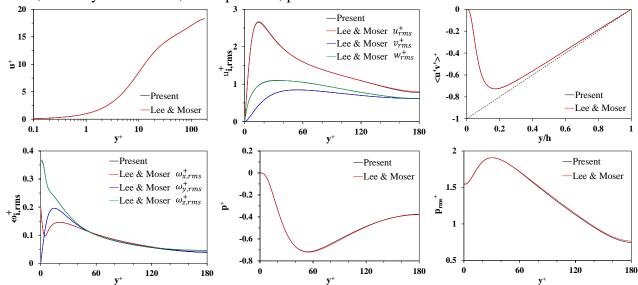


Fig.5. Basic turbulent statistic profiles

### 7.2 Budget profiles for TKE

Fig.6 shows the budget profiles for TKE.

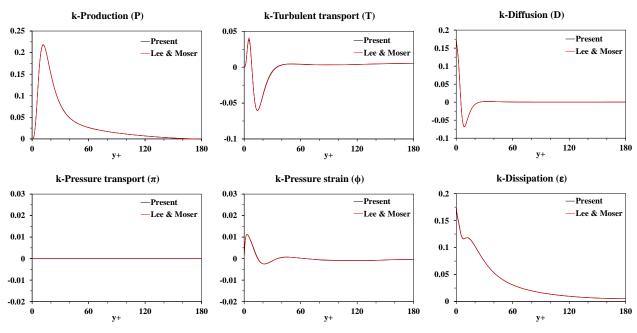
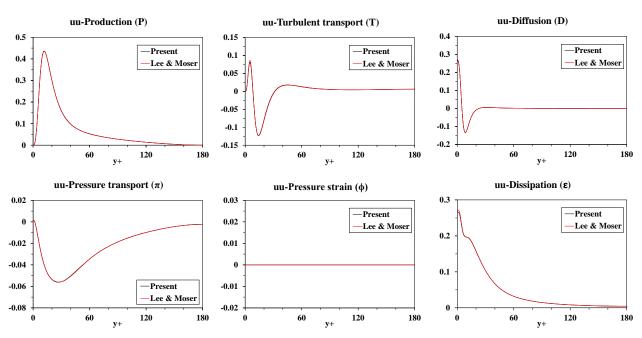


Fig.6. Budget profiles for TKE

## 7.3 Budget profiles for $\langle u'u' \rangle$

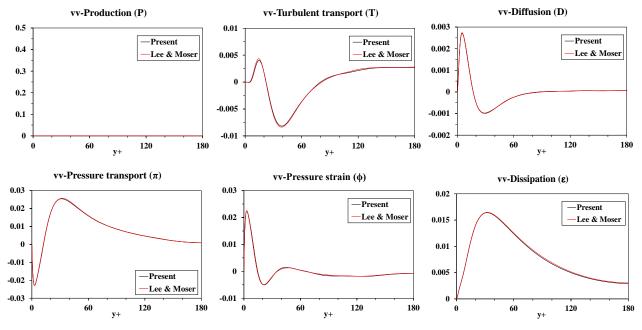
Fig. 7 shows the budget profiles for  $\langle u'u' \rangle$ .



**Fig.7.** Budget profiles for  $\langle u'u' \rangle$ 

### 7.4 Budget profiles for $\langle v'v' \rangle$

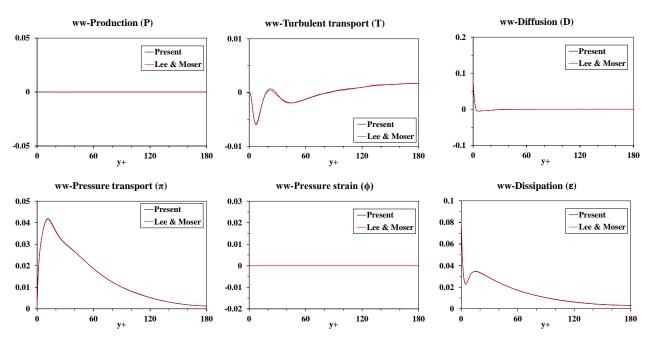
Fig.8 shows the budget profiles for  $\langle v'v' \rangle$ .



**Fig.8.** Budget profiles for  $\langle v'v' \rangle$ 

### 7.5 Budget profiles for $\langle w'w' \rangle$

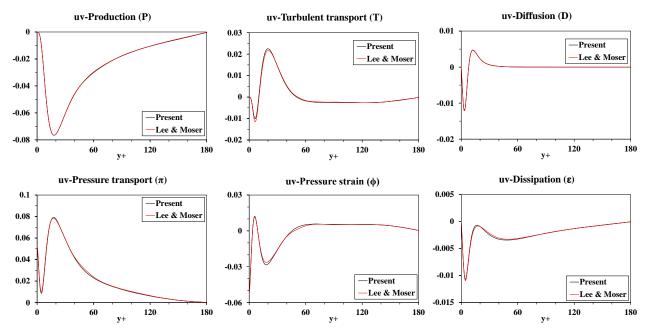
Fig.9 shows the budget profiles for  $\langle w'w' \rangle$ .



**Fig.9.** Budget profiles for  $\langle w'w' \rangle$ 

### 7.6 Budget profiles for $\langle u'v' \rangle$

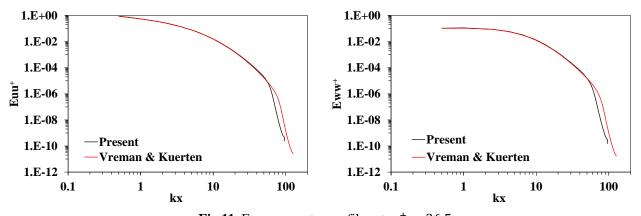
Fig. 10 shows the budget profiles for  $\langle u'v' \rangle$ .



**Fig.10.** Budget profiles for  $\langle u'v' \rangle$ 

#### 7.7 Energy spectra profiles

Fig.11 shows the energy spectra profiles at  $y^+ \approx 36.5$ .



**Fig.11.** Energy spectra profiles at  $y^+ \approx 36.5$ 

### **Reference:**

- [1] Lee M, Moser RD. Direct numerical simulation of turbulent channel flow up to Re  $\tau \approx 5200$ . J Fluid Mech.2015;774:395-415. doi:10.1017/jfm.2015.268.
- [2] Vreman AW, Kuerten JGM. Comparison of direct numerical simulation databases of turbulent channel flow at Re $\tau$  = 180. Phys Fluids. 2014;26(1):1-21. doi:10.1063/1.4861064.

### Appendix A. Definition of turbulent budget terms for wall-bounded turbulence

The general Reynolds stress transport equation reads:

$$\frac{\partial \left\langle u_{i}'u_{j}'\right\rangle}{\partial t} + \left\langle u_{k}\right\rangle \frac{\partial \left\langle u_{i}'u_{j}'\right\rangle}{\partial x_{k}} = P_{i,j} + T_{i,j} + \phi_{i,j} + \pi_{i,j} + D_{i,j} - \varepsilon_{i,j} \tag{A.1}$$

where  $P_{i,j}$ ,  $T_{i,j}$ ,  $\phi_{i,j}$ ,  $\pi_{i,j}$ ,  $D_{i,j}$ , and  $\varepsilon_{i,j}$  stand for production term, turbulent transport term, pressure transport term, pressure strain term, viscous transport (diffusion) term, and viscous dissipation term respectively:

$$P_{i,j} = -\left\langle u_i' u_k' \right\rangle \frac{\partial \left\langle u_j \right\rangle}{\partial x_{\nu}} - \left\langle u_j' u_k' \right\rangle \frac{\partial \left\langle u_i \right\rangle}{\partial x_{\nu}} \tag{A.2a}$$

$$T_{i,j} = -\frac{\partial \left\langle u_i' u_j' u_k' \right\rangle}{\partial x_k} \tag{A.2b}$$

$$\phi_{i,j} = -\frac{\partial \left\langle p' u_i' \right\rangle}{\partial x_j} - \frac{\partial \left\langle p' u_j' \right\rangle}{\partial x_i}$$
(A.2c)

$$\pi_{i,j} = \left\langle p' \left( \frac{\partial u_i'}{\partial x_j} + \frac{\partial u_j'}{\partial x_i} \right) \right\rangle \tag{A.2d}$$

$$D_{i,j} = \nu \frac{\partial^2 \left\langle u_i' u_j' \right\rangle}{\partial x_{\iota} \partial x_{\iota}} \tag{A.2e}$$

$$\varepsilon_{i,j} = 2\nu \left\langle \frac{\partial u_i'}{\partial x_k} \frac{\partial u_j'}{\partial x_k} \right\rangle \tag{A.2f}$$

Specifically, for wall-bounded turbulent flow, where periodic boundary conditions are applied in streamwise and spanwise directions. So Eq.(A.2) can be further simplified as follows:

$$P_{i,j} = -\langle u'_{i}v' \rangle \frac{\partial \langle u_{j} \rangle}{\partial y} - \langle u'_{j}v' \rangle \frac{\partial \langle u_{i} \rangle}{\partial y}, \qquad T_{i,j} = -\frac{\partial \langle u'_{i}u'_{j}v' \rangle}{\partial y}$$

$$\phi_{i,j} = -\frac{\partial \langle p'u'_{i} \rangle}{\partial x_{j}} - \frac{\partial \langle p'u'_{j} \rangle}{\partial x_{i}}, \qquad \qquad \pi_{i,j} = \left\langle p' \left( \frac{\partial u'_{i}}{\partial x_{j}} + \frac{\partial u'_{j}}{\partial x_{i}} \right) \right\rangle$$

$$D_{i,j} = v \frac{\partial^{2} \langle u'_{i}u'_{j} \rangle}{\partial y^{2}}, \qquad \qquad \varepsilon_{i,j} = 2v \left\langle \frac{\partial u'_{i}}{\partial x_{k}} \frac{\partial u'_{j}}{\partial x_{k}} \right\rangle$$
(A.3)

And for  $\langle u'u' \rangle$ :

$$P_{1,1} = -2\left\langle u'v'\right\rangle \frac{\partial \left\langle u\right\rangle}{\partial y}, \quad T_{1,1} = -\frac{\partial \left\langle u'u'v'\right\rangle}{\partial y}, \phi_{1,1} = 0$$

$$\pi_{1,1} = 2\left\langle p'\frac{\partial u'}{\partial x}\right\rangle, \quad D_{1,1} = v\frac{\partial^2 \left\langle u'u'\right\rangle}{\partial y^2}, \quad \varepsilon_{1,1} = 2v\left\langle \frac{\partial u'}{\partial x_k}\frac{\partial u'}{\partial x_k}\right\rangle$$
(A.4)

For  $\langle v'v' \rangle$ :

$$P_{2,2} = -2\left\langle v'v'\right\rangle \frac{\partial \left\langle v\right\rangle}{\partial y} = 0, \quad T_{2,2} = -\frac{\partial \left\langle v'v'v'\right\rangle}{\partial y}, \quad \phi_{2,2} = -2\frac{\partial \left\langle p'v'\right\rangle}{\partial y}$$

$$\pi_{2,2} = 2\left\langle p'\frac{\partial v'}{\partial y}\right\rangle, \quad D_{2,2} = v\frac{\partial^2 \left\langle v'v'\right\rangle}{\partial y^2}, \quad \varepsilon_{2,2} = 2v\left\langle \frac{\partial v'}{\partial x_k}\frac{\partial v'}{\partial x_k}\right\rangle$$
(A.5)

For  $\langle w'w' \rangle$ :

$$P_{3,3} = -2\left\langle v'w'\right\rangle \frac{\partial \left\langle w\right\rangle}{\partial y}, \quad T_{3,3} = -\frac{\partial \left\langle w'w'v'\right\rangle}{\partial y}, \quad \phi_{3,3} = 0$$

$$\pi_{3,3} = 2\left\langle p'\frac{\partial w'}{\partial z}\right\rangle, \quad D_{3,3} = v\frac{\partial^2 \left\langle w'w'\right\rangle}{\partial y^2}, \quad \varepsilon_{3,3} = 2v\left\langle \frac{\partial w'}{\partial x_k}\frac{\partial w'}{\partial x_k}\right\rangle$$
(A.6)

For  $\langle u'v' \rangle$ :

$$P_{1,2} = -\langle v'v' \rangle \frac{\partial \langle u \rangle}{\partial y}, \quad T_{1,2} = -\frac{\partial \langle u'v'v' \rangle}{\partial y}, \quad \phi_{1,2} = -\frac{\partial \langle p'u' \rangle}{\partial y}$$

$$\pi_{1,2} = \left\langle p' \left( \frac{\partial u'}{\partial y} + \frac{\partial v'}{\partial x} \right) \right\rangle, \quad D_{1,2} = v \frac{\partial^2 \langle u'v' \rangle}{\partial y^2}, \quad \varepsilon_{1,2} = 2v \left\langle \frac{\partial u'}{\partial x_k} \frac{\partial v'}{\partial x_k} \right\rangle$$
(A.7)