



# 1 Problem Description

## 2 Similitude

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# Initial Conditions & Constraints

## Inlet conditions

- $P_{T0} = 1\text{bar}$
- $T_{T0} = 300\text{K}$

## Constraints

- $r_{max} = 0.45\text{m}$
- $\beta_{TT} = 1.45$
- $\dot{Q} = 100 \frac{\text{kg}}{\text{s}}$
- $\max \eta$

Due to the **course track** and **preference**, the turbomachinery design will be on an **axial** compressor.



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Main Design Quantities

$V_{t_{mean}}$ ,  $V_{a_{mean}}$ ,  $U_{mean}$  & velocity triangles

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# Procedural Steps

## Hypothesis

It has been chosen to not use an **inlet guide vane** for simplicity of design and so  $V_{t0} = 0 \frac{m}{s}$  and  $\chi$  dictate the behaviour of  $\lambda$ . Another initial design choice is to keep, in the similarity/adimensional analysis of the compressor,  $V_a$  **constant** and to **avoid** using a **flaring based** approach.

### Main procedural steps:

- $\lambda$  and  $\psi$  computation from  $\chi$  and  $V_{t0}$
- $\phi$  and  $\eta$  computation
- $V_a$  and  $L_{eu}$  computation from  $\phi$ ,  $\beta_{TT}$  and  $\eta$
- computing **mean** velocity triangles, using the above hypothesis
- computing **blade height**

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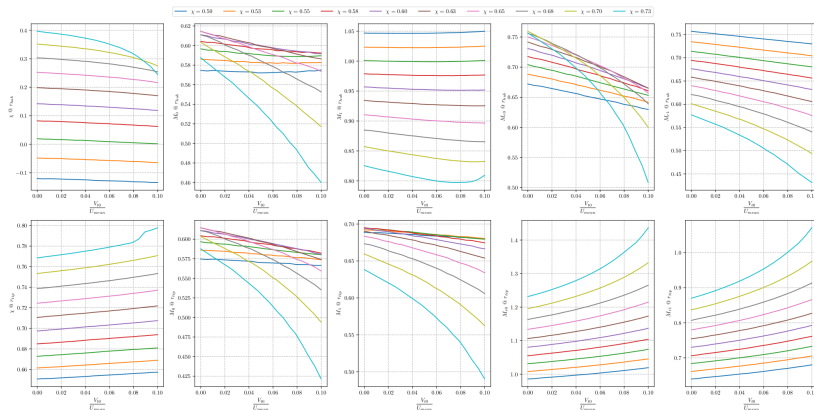
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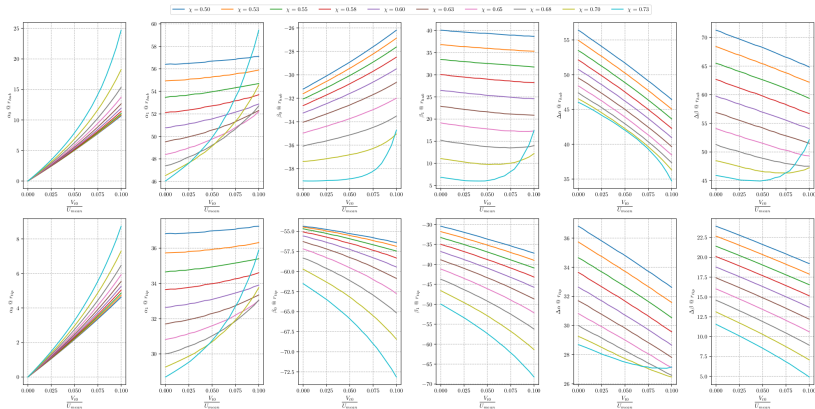
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# Graph Analysis: $\chi$ & $M$





# Graph Analysis: $\alpha$ & $\beta$



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# $\lambda$ & $\psi$

From the previous graphs:

- $\chi = 0.55$
- $r_{mean} = 0.325m$
- $\frac{V_t}{U_{mean}} = 0$

Taking into account the previous modeling hypothesis:

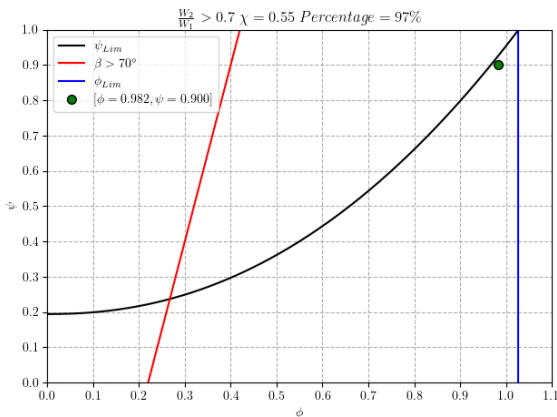
$$\lambda = \left(1 - \chi - \frac{V_t}{U_{mean}}\right) \cdot 4 \quad (1)$$

$$\psi = \frac{\lambda}{2} \quad (2)$$



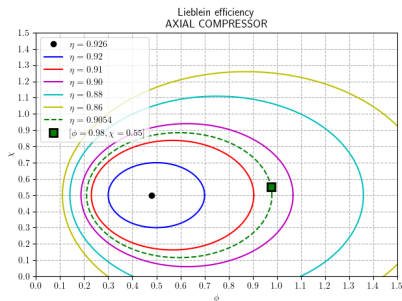
$$\phi(\psi)$$

From [?, Sec. 10.4] it is imposed that  $\frac{W_2}{W_1} \geq 0.7$  with a *safety* margin of 2%.



$\eta$  &  $L_{eu}$ 

$\eta$  is computed from an **Lieblein** efficiency chart<sup>1</sup> given  $\phi$  and  $\chi$ . This parameter will be used for the computation of  $L_{eu}$  given the  $\beta_{TT}$  target.



$$L_{is} = \frac{\gamma R}{\gamma - 1} T_{in} \left( \beta_{TT}^{\frac{\gamma-1}{\gamma}} - 1 \right) \quad (3)$$

$$L_{eu} = \frac{L_{is}}{\eta} \quad (4)$$

<sup>1</sup>This chart has been interpolated from the course slides charts.



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$$U_{mean} = \frac{L_{eu}}{\psi} \quad (5)$$

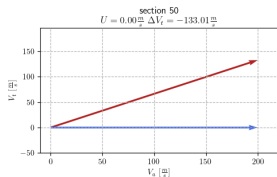
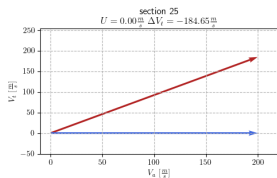
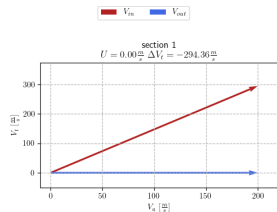
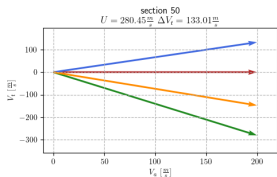
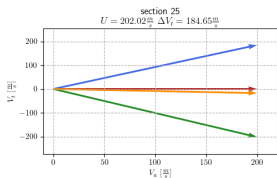
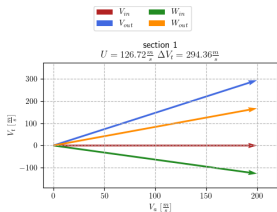
$$V_{a_{mean}} = \phi U_{mean} \quad (6)$$

$$L_{eu} = U_1 V_{t1} - U_0 V_{t0} \quad (7)$$

$$= U_{1_{mean}} V_{t1_{mean}} - U_{0_{mean}} V_{t0_{mean}} = U_{mean} \Delta V_{t_{mean}} \quad (8)$$

$\Delta V_t$  computation allows to get a *first sketch* of the **velocity triangles**. The first analysis results are stored in `compressor_0.55_0.325_28_28.txt`.





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Losses Modeling

Radial Equilibrium

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## Losses Modeling

Profile Losses

Compressibility Losses

Shock Losses

Tip Leakage Losses

Radial Equilibrium

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# Profile Losses

The profile losses used are related to the **Leiblein modeling** approach, [?, Ch. 6].

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Losses Modeling

Radial Equilibrium

General Formulation

Vortex Model

NISRE Equilibrium Results

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**turboLIB**

Section Analysis & Optimization

.stl & .scad Generation

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