

Doctoral Dissertation

**Hybrid Testing Method and Excitation System Design for  
Seismic and Wind-resistant Performance Evaluation of  
Building Structures with Nonlinear Dampers**

by

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## ABSTRACT

# Hybrid Testing Method and Excitation System Design for Seismic and Wind-resistant Performance Evaluation of Building Structures with Nonlinear Dampers

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**Abstract:** The accurate identification of the dynamic characteristics of a building structure excited by input loads such as an earthquake or wind load is essential not only for the evaluation of the safety and serviceability of the building structures but for the verification of an analytical model used in the seismic or wind design. In the field of system identification (SI) which constructs system matrices describing the accurate input and output relationship, it is critical that input should have enough energy to excite fundamental structural modes and a good quality of output containing structural information should be measured. This thesis proposes a hybrid testing method and the controller design of actuators to simulate the responses of a building structure subjected to such dynamic excitations as earthquakes and wind loads. Firstly, The hybrid testing method is a technique to experiment with a non-linear part in an entire structural system as an experimental model and a part that is expected to behave linearly as an analytical model. In this method, the physical unit of the interface between the experimental model and the analytical model should be selected, and an experimental system should be constructed in which the interface of the analytical model is designed to act as the excitation force of the experimental model. In this thesis, a substructure test as part of the hybrid testing method by separating the structural system into the experimental structure and the analytical structure through the shaking table, and the hybrid test of nonlinear vibration control devices, the tuned liquid damper, the tuned liquid column damper and the tuned liquid mass damper as experimental models and building structures as analytical models were designed and implemented to evaluate the performance. Furthermore, seismic performance evaluation of a building with magneto-rheological (MR) damper installed by installing an MR damper, which is a 1-ton inter-story damper with high nonlinearity, on UTM, and the semi-active algorithm applied to the MR damper were experimentally evaluated. This method can provide the possibility and quantitative basis for experimentally finding and applying the optimum algorithm of the control device with unspecific nonlinearity or the performance of the device itself. Secondary, the controller system design that implements the dynamic response is a system that installs a linear mass shaker or an active mass damper on a

particular floor and implements a pseudo-seismic or pseudo wind load response in a full-scale building. In this thesis, to simulate of wind-induced responses, a 76-story benchmark building which has an analytical model and wind load data were numerically analyzed, and to realize the seismic responses, the pseudo-earthquake excitation is implemented by the hybrid mass damper with the real-scaled 5-story building. Also, an MR damper with the capacity of 1-ton was installed to evaluate the semi-active control performance of MR damper under earthquakes load in a real scale building. These techniques can be used to assess the seismic and wind resistant performance of real-scaled building structures and to evaluate the seismic and wind resistance performance of buildings with nonlinear control devices.

**Keywords:** hybrid testing method, substructuring technique, tuned liquid type damper, control-structure interaction, inverse transfer function, active tuned mass damper, hybrid mass damper, design of an actuator, 76-story benchmark problem building, force vibration testing, real-scale building, system identification, finite element model updating, magneto-rheological damper, semi-active control algorithms.

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## CHAPTER 1

# Introduction

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### 1.1 General

Accurate identification of the dynamic response characteristics of building structures excited by external loads such as earthquakes and wind loads is essential not only for the safety and serviceability evaluation of building structures but for the validation of analytical models used for seismic or wind design(Ljung, 1987). In the System Identification (SI) field, which constructs a system matrix that describes precise input and output relationships to build an analytical model of an architectural structure, it is critical that input should have sufficient energy to excite fundamental structural modes and a high quality of output containing structural information should be measured(Alvin and Park, 1994; Madenci and Barut, 1994). However, full-scale dynamic testing is difficult due to size and weight limitations in most large civil engineering structures. Also, in most cases, it is very difficult or impossible to extrapolate the results of small-scale model tests as a result of a full-scale structure that operates non-linearly in most cases. Therefore, it is used as a means to evaluate the dynamic performance of a building or to implement the system identification process by installing an excitation system such as an eccentric exciter or a linear mass shaker which excite the fundamental modes, in the building. However, to evaluate the seismic and the wind resistance performance accurately, it is necessary to fully reflect the characteristics of the seismic load and the wind load.

One method to overcome these limitations is to use the hybrid testing method. This technique for the dynamic testing of large-scale structures, typically involving nonlinearities, does not test the entire structure as a whole. Instead, it often happens that nonlinearity or unpredictable behavior is concentrated in a limited number of local parts of the overall structure. Those important local parts are experimentally tested, while the rest of the structure, which is assumed to behave linearly, in essence, is modeled numerically. Numerical and experimental parts are allowed to interact with each other to produce the response of the entire system. The advantage of the hybrid testing technique over the reduced-scale model testing is that the experimental substructure is tested at full size and the scaling effect can be eliminated.

Another method to overcome above limitations is to design controller of actuators to

simulate an earthquake and wind-induced responses. This method is a kind of forced vibration testing method to implement seismic or wind responses of a building structure by using a vibration control device already installed in a real building such as ATMD or installing additional exciter in specific story of the building. For the successful application of this method, several issues should be considered. First, interaction effects between the actuators and building structure and dynamic characteristics of the actuator itself should be compensated. Second, because this method uses inverse transfer function method, unexpected modal responses and transient responses should be reduced. This method may evaluate the seismic and wind performance not only of the existing building but for of the real scale building equipped with the nonlinear vibration control device.

## 1.2 Objectives and Contents

Figure 1.2 shows the contents of chapters and their relations in this thesis. It can be classified into a hybrid testing method and a pseudo-excitation system for simulating a dynamic load of building structures. Verification methods can be summarized as numerical analysis, shaking table experiment, UTM experiment, and experiment with full-size building with HMD. The objective of this study is to design a hybrid testing method and a controller of an actuator to perform an experimental evaluation of seismic and wind resistance performance of building structures equipped with nonlinear vibration control devices. The contents for these objectives are as follows;

First, in chapter 2, the hybrid testing method is proposed. To be implemented by the hybrid testing technique and accurately represent the whole building structure, the experimental and numerical substructures must operate in parallel with minimal errors at the interface between them. Three factors are essential in the implementation issues of the hybrid testing method. Those issues are the measurement of the interface force between two substructures, the calculation of the numerical substructure, and the loading operation of the experimental substructure with an actuator. The responses of the shared degrees-of-freedom (DOF) are calculated using the numerical model and passed to the controller of actuators or a shaking table. The controller generates signals driving the actuators to impose these responses on the experimental substructure. To be replicated by the behavior of the whole structure by the hybrid testing method, it is important to design the controller of the actuators or the shaking table, so that difference between the expected and actual measured behavior at the interface of the two substructures is minimized. The vibration control effect of a tuned liquid type damper (TLD/TLCD) for a building structure excited by earthquake load is experimentally evaluated through the hybrid testing method. The hybrid testing method doesn't require a physical building structural model in per-

forming the experiment of a TLD/TLCD-structure interaction system, and it only uses a TLD/TLCD which is known to have high nonlinearity dependent on response amplitude, excitation frequency, and depth to length ratio (Yalla, 2001). A new control device, which is called tuned liquid mass damper (TLMD), is discussed in this chapter. The functional characteristic of a TLMD used in this study is that its mass is composed of both a mass of TLCD tank itself and that of liquid within a tank. Natural rubbers were also used to substitute the stiffness of a TMD. Therefore, a TLMD employed in this study operates as a TLCD in one direction and behaves as a TMD in the other orthogonal direction. Regarding the field of structural testing, the hybrid testing method also implemented in this study. As mentioned in the previous sentence, TLD and TLCD has been subjected to many kinds of research and experiments and has been verified by the hybrid testing method in this study. Therefore, as the hybrid testing method of building controlled by TLMD, it is significant to use the hybrid testing method technique by validating the control performance of the newly proposed control device by comparison between the SDOF-actuator experiment method and the hybrid testing method. The structural responses of the interaction system are calculated numerically in real-time using the analytical structural model with the excitations of measured control force, user-defined base earthquake loads, and its state space realization incorporated in the integrated controller of an actuator. Also, to minimize the distortion of the acceleration of the shaking table or an actuator's displacement, the inverse transfer function of the shaking table or the actuator is identified, and its state space realization is implemented in the hybrid testing method controller.

Secondly, in chapter 3, simulation of dynamic responses of a building structure using a linear mass shaker is conducted. For the linear shaker to keep the structure in the target response trajectory, an inverse transfer function of a structural response to the shaker force is obtained using a state space form governing equation of the structure, and the discrete Fourier transform of structural response is performed. Filter and envelope function are used such that the error between wind and actuator induced responses is minimized by preventing the shaker from the excitation, unexpected modal response, and initial transient response. The effectiveness of the proposed method is verified through a numerical example of a 76 story-benchmark building excited by wind load of which deterministic time history is given. The effect of the type of the target structural response on a convergence of the shaker force signal and the error magnitude is investigated. Moreover, in this chapter, forced vibration pseudo-earthquake tests using an HMD(hybrid mass damper) installed on the 4th floor of the real-scale five-story modal testing tower were performed. To experimentally evaluate the seismic performance of a building structure with accuracy, shaking table test should be conducted. However, it is practically impossible and costly to excite a real scale structure using the shaking table. Yu et al. (2005) presented a linear shaker system

to simulate structural seismic response by using inverse transfer function of the structural response induced by the linear shaker of the structure excited by earthquake load. Real-scaled MR damper was installed to verify the effectiveness of the pseudo-earthquake experiment in a real scale building. Under the four historical earthquakes and one filtered artificial earthquake, the performance of the semi-active control algorithms including the passive optimal case is experimentally evaluated by the pseudo-earthquake test method.

Finally, in chapter 4 shows the experimental verification of the hybrid testing method applied to UTM with an MR damper. The MR damper with the capacity of 1-ton was used from chapter 3, the hybrid testing method of the real-scaled building with semi-active controlling MR damper implemented in the chapter 2 was applied by hybrid testing method technique. This study presents the quantitative evaluation of the seismic performance of a building structure installed with an MR damper using the hybrid testing method. A building model is identified from the forced vibration testing results of a full-scale five-story building and is used as the numerical substructure, and an MR damper corresponding to an experimental substructure is physically tested using a universal testing machine (UTM). First, the force required to drive the displacement of the story, at which the MR damper is located, is measured from the load cell attached to the UTM. The measured force is then returned to a control computer to calculate the response of the numerical substructure. Finally, the experimental substructure is excited by the UTM with the calculated response of the numerical substructure. The hybrid testing method implemented in this study is validated because the hybrid testing results obtained by application of sinusoidal and earthquake excitations and the corresponding analytical results obtained using the Bouc-Wen model as the control force of the MR damper with input currents are in good agreement. Also, the results from the hybrid testing method for the passive -on and -off control show that the structural responses did not decrease further by the excessive control force, but decreased due to the increase of the current applied to the MR damper. Also, two semi-active control algorithms (modulated homogeneous friction and the clipped-optimal control algorithms) are applied to the MR damper to control the structural responses optimally. To be compared with the hybrid testing method and numerical results, the Bouc-Wen model parameters should be identified for each input current. The results of the comparison of experimental and numerical responses show that it is more practical to use the hybrid testing method in semi-active devices such as MR dampers. The test results indicate that a control algorithm can be experimentally applied to the MR damper using the hybrid testing method. This study also provides a discussion on each algorithm with respect to the seismic performances.

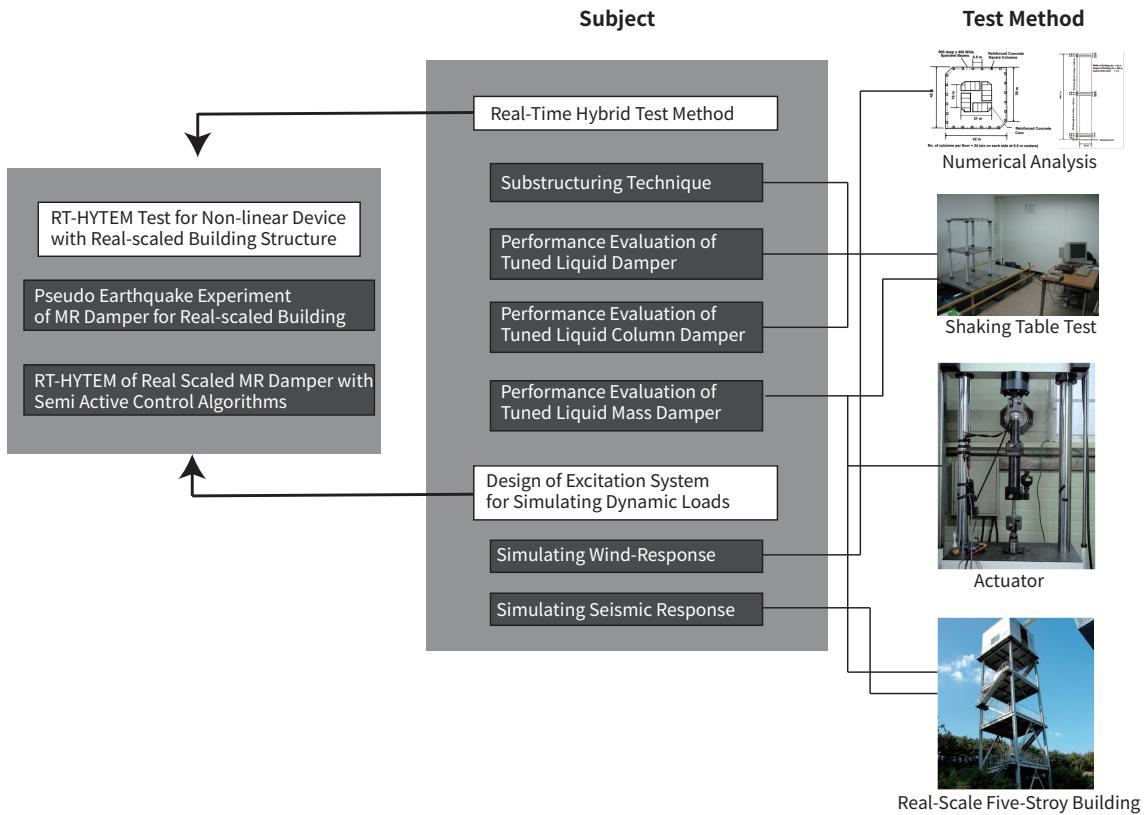


Figure 1.1: Contents of chapters and their relations

## 1.3 Literature Review

### 1.3.1 Hybrid testing method

Since the substructuring technique is first developed for large-scale structures by Nakashima et al. (1992), several pieces of research have been performed on the substructuring technique both experimentally and analytically to overcome difficulties in the test of large-scale structures. Blakeborough et al. (2001); Darby et al. (2001) focused on the control algorithm for operating hydraulic actuators and conducted experiments for a one-story frame with horizontal and rotational DOF and its corresponding numerical substructure, which shows the linear or nonlinear behavior (Darby et al., 2002). Neild et al. (2005) separated a large structural mass of the single DOF system into two parts and selected the smaller one as the experimental substructure and the larger one with attached spring and dashpot as the numerical substructure to conduct a shaking table test (Neild et al., 2005). Nakashima and Pan applied the method to a base-isolated structure consisting of the base-isolation layer as an experimental substructure and the rest of the upper structural model as a numerical substructure (Nakashima and Masaoka, 1999). Besides, they developed a mixed control algorithm using both displacement and force for its implementation (Pan et al.,

2005).

Takanashi et al. (1975, 1980) have firstly developed the pseudo-dynamic testing method, in which only a part of the whole structure, mainly being expected to show nonlinearity, is manufactured and tested while the remainder showing linearity is numerically calculated(Takanashi et al., 1975, 1980). Because there exists propagation of experimental errors in pseudo-dynamic testing methods, the stability, and accuracy of the numerical integration methods were investigated(Shing and Mahin, 1983, 1984). In these pseudo-dynamic testing methods, as implied in the name, the experimental part is not “**dynamically**” but “**statically**” excited under the loading condition which makes the testing part represent identical displacement response to that of the part in whole structure excited by considered dynamic load such as ground acceleration.

Horiuchi et al. (1999) compensated the time-delay effect caused mainly by analytical procedure in the real-time hybrid testing method(Horiuchi et al., 1999). For its experimental verification, a small portion of mass was separated from a mass-spring-dashpot system, and only the small part of the mass was tested considering the effects of the other parts analytically. Iemura et al. (1999); Igarashi et al. (2000) verified the effect of vibration control devices such as a tuned mass damper and an active mass damper installed in a structure excited by ground acceleration, using the hybrid testing method in which the control devices were experimental parts and the remaining structural model was a numerical part. The acceleration signal of the moving mass of the devices was measured and used as input to the numerical modelIemura et al. (1999); Igarashi et al. (2000).

TLD dissipates structural vibratory energy by tuning the frequency of the liquid sloshing to one of the structure(Soong and Dargush, 1997). TLD has been applied to the control of wind-induced acceleration response(Chang and Qu, 1998), and recently, some investigations on the seismic control performance of the TLD have been made(Banerji et al., 2000). In order to describe the behavior of the TLD, linear model based on tuned mass damper analogy(Sun et al., 1995) and linear wave theory, nonlinear stiffness and damping model(Yu et al., 1999), and sloshing-slamping analogy(Yalla, 2001) can be used. However, because any model has the error in capturing the real dynamic characteristics of the control force generated by the TLD and furthermore the error increases for the case of non-stationary excitation such as the earthquake, evaluation of the seismic control performance of the TLD only numerically has the accuracy problem.

Recently, the TLCD has received the attention of researchers as a type of auxiliary mass system(Samali et al., 1998). TLCD has the control characteristics similar to that of tuned mass damper (TMD), which is one of most frequently used dampers for vibration control. Since the viscosity term in the governing equation of motion of TLCD is a function of the absolute value of liquid velocity, the equation is non-linear, and the dynamic characteristics

of TLCD depend on the magnitude and the characteristics of excitation forces and the corresponding structural responses of the floor at which TLCD is installed(Yalla, 2001).

Two or more TMDs and TLCDs have been installed in a building to control bidirectional responses(Xu and Igusa, 1992; Fujino and Sun, 1993; Yamaguchi and Harnpornchai, 1993; Igusa and Xu, 1994; Jangid and Datta, 1997; Li, 2000; LI et al., 2000; Singh et al., 2002) . Since multi-tuned mass dampers (MTLDs) had been proposed by Xu and Igusa (1992); Igusa and Xu (1994), many researchers have devoted their efforts to improve the control performance of MTLDs(Fujino and Sun, 1993; Yamaguchi and Harnpornchai, 1993; Jangid and Datta, 1997; Li, 2000; LI et al., 2000; Singh et al., 2002; ZHANG and ZHONG, 2004). Especially, Jangid and Datta (1997); Singh et al. (2002) have studied the MTMDs by which the torsional response of building structures subjected to bidirectional vibrations can be reduced. They also proposed the optimum parameters such as mass, frequency and damping ratios, in addition to the installation locations of MTMDs, to control the torsional response. Fujino and Sun (1993) have studied about the optimal frequency ratio and the optimal number of MTLDs. However, for applying these multi dampers to building structures, there have been shortcomings such as high installation cost, hard maintenance and additional retrofitting of the story at which they are installed. Additionally, applying the pendulum type TMD used in Taipei 101 building has a limitation that the structural plan should be the similar shape in both weak and strong axes. Also, this type of TMD needs a large installation space(Haskett et al., 2004).

### **1.3.2 Design of Excitation System for Simulating Dynamic Loads**

In the field of earthquake engineering, forced vibration tests for evaluating the dynamic characteristics of structures and evaluating seismic performance have been widely carried out from shaking table, eccentric exciter, and reaction wall to shrinking and real scale structures. Several studies on the system identification to establish the analytical structural model by using the modal characteristics, which are obtained by forced vibration tests, have also been carried out(Halling et al., 2001; Min et al., 2004). The primary purpose of system identification is to construct an analytical model that can simulate the relationship between the excitation load and the measurement response obtained in the forced excitation experiment.

Juang (1994) proposed the system identification method in the frequency domain by using matrix decomposition and that in the time domain by employing Markov parameters and Observer/Kalman filter(Juang, 1994). Dyke et al. (1994) obtained the system matrix of a structure by applying model reduction technique after exciting a 3-story small scale structural model with the shaking table and the active mass driver, and then performing the system identification for the respective excitation loads. However, the analytical model

that is identified by adopting their methods only describes its input and output relations and does not represent physical information such as the mass, viscosity, and stiffness(Dyke et al., 1994). Alvin and Park (1994) extracted the finite element model that has its physical meaning by using common based-normalized system identification (CBSI) technique(Alvin and Park, 1994).

The shaking table test, in which the loads similar to real earthquakes can be easily simulated, is frequently taken to experimentally assess the seismic performance of a structure with high accuracy. It is usually applied to tests on small-scale structure, however, since it is very difficult to excite real-scaled structures with a shaking table. Yu et al. (2005) designed the linear shaker system to simulate earthquake response of a structure by employing the methods both minimizing the error in time histories between the earthquake-induced structural responses and the linear shaker-induced ones and using the inverse transfer function of the responses of a structure loaded by a linear shaker(Yu et al., 2005). However, their study is based on numerical verification, and also has several limitations for being applied to real-scaled structures. First, forced vibration tests are not appropriate for utilizing the vibration source with large amplitude in the experimental structure. Especially, the structure can not be excited to its nonlinear range, since safety may not be ensured during the test and the linear structural model is used for the numerical simulation. Secondly, it is inappropriate for using the vibration source with full bandwidth. Thirdly, sensors may not be sufficiently deployed in field test due to expenses and their installations so that the structural members, in which critical behaviors are expected, are detected. Finally, the process of obtaining the system matrix of a structure by using the system identification technique should be additionally included in the entire research work, since the structure is based on an analytical model. Accordingly, the validity of the proposed pseudo-earthquake excitation test needs to be experimentally investigated under these constraints.

Dyke et al. (1994) obtained controller canonical form state space realization for a small scale three-story building by using both active mass driver and shaking table and measuring the absolute floor acceleration. Juang (1994) proposed Observer/Kalman filter identification using system Markov parameters in time-domain. These studies present the mathematical models which accurately describe input/output relationship but do not provide physical mass, stiffness, and damping matrices, finite element model based SI techniques are developed in the field of health monitoring or damage detection(Van der Auweraer and Peeters, 2003). Yu et al. (2005) performed a series of ambient vibration measurement and force vibration tests on four-story reinforced concrete building by using linear and eccentric mass shakers and updated the analytical finite element model based on the collected dynamic data. Also, Yu et al. (2005) presented a linear shaker system

to simulate linear elastic structural seismic response. However, because it is practically very difficult to excite large-scale civil structures by using artificial actuator-type devices, SI techniques using natural excitation such as the wind, vehicle, and human are investigated. In the most earthquake engineering, forced vibration testing has been implemented for many years to assess dynamic characteristics of a building structure. It is based on the fact that dynamic response is sensitive to changes in mass, damping, or stiffness of a structure. These changes would lead to shifts in the modal parameters, such as natural frequencies and mode shape. The dynamic characteristics in the system to identify the integrity of a structure have also been investigated(Dyke et al., 1994). The primary objective in system identification is to find the relationship between the experimental response data and the analytical model by adjusting the structural parameters in the model.

### **1.3.3 Hybrid Testing Method on a Semi-actively Controlled Building Structure Equipped with Full-scale MR Dampers**

A wide variety of structural control strategies for civil engineering structures such as bridges and buildings have been proposed by structural engineers(Adeli, 1999). Various types of active and semi-active or hybrid control dampers have also been developed to enhance the performance of passive control devices(Kim and Adeli, 2005b,a). Because semiactive control devices such as variable-orifice dampers, variable-friction dampers, and electrorheological/magnetorheological (ER/MR) dampers have the potential to offer the adaptability of active devices without requiring the associated large power sources as well as the inherent stability, it is expected that the application of a semiactive control device is one of the most promising means for mitigating structural response of building structures under the unexpected external loads such as earthquake, wind, blast, impact, and so on.

Analytical and experimental studies on these control devices when applied to MR dampers have been performed in an effort to reduce the structural responses, mainly due to their intrinsic stability and low power consumption(e.g., high dynamic range, mechanical simplicity, environmental robustness, etc.) since mid-1990s (Dyke et al., 1994; Spencer et al., 1997). MR damper-based semiactive control systems require a feedback system consisting of sensors, external power supply and controllers. Semi-active and adaptive control strategies using the MR damper have also been extensively compared and analyzed(Jansen and Dyke, 2000; Kim et al., 2009; Bitaraf et al., 2010). Also a lot of control algorithms for MR damper-based semiactive systems have been proposed and numerically verified for control of civil engineering structures such as buildings and bridges by several researchers (Jansen and Dyke, 2000; Zhou et al., 2003). However, each has emphasized its advantages depending on the specific application and desired effect. In order to demonstrate the pros and cons of the different control algorithms, a series of benchmark structural control prob-

lems have been developed by the ASCE Structural Control Committee and Task Group on Benchmark Problems and International Association of Structural Control and Monitoring (IASCM). Although the benchmark problems may give us a chance to compare various control algorithms directly, they are only numerical simulations which cannot consider practical limitations. In analytical studies, the Bingham, Bouc-Wen, and phenomenological models were proposed as analytical models for describing the hysteretic behavior of the MR damper(Spencer et al., 1997; Yang et al., 2004a). Although these models are useful in the design of the MR damper, they are inappropriate for characterizing the behavior of the MR damper under the excitation of irregular loads such as earthquakes and winds because of the high nonlinearities of the MR damper due to its dependency on the loading rate and the amplitude of excitations. Also, the performance of the MR damper is not guaranteed in accordance with its current providing devices. Moreover, when the MR damper behaves as a semiactive control device, the hysteretic model is unreliable when it varies with the applied current(Lee et al., 2006). For these reasons, there may be disagreement between the corresponding analytical results and the actual responses of a building installed with an MR damper when the semi-active control strategy is applied.

# Hybrid Testing Method

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In this chapter, the hybrid testing method is introduced as a classification according to equipment type. The hybrid testing method is mainly divided into shaking table experiment and UTM experiment by its equipment type. First, the hybrid testing method using the shaking table is split into the substructuring test method which tests the partial structural system separately, and the hybrid shaking table testing method by installing building structure with nonlinear mass typed vibration control devices such as TLD, TLCD, and TLMD. Secondly, in the case of MR damper which is a type of semi-active friction damper, it is very difficult to implement hybrid shaking table testing installed in an actual building. Therefore, the MR damper is installed in a universal testing machine (UTM), and a hybrid testing method is implemented by simulating the MR damper behavior mounted in a real scaled building. This subject will consider with chapter 4.

## 2.1 Substructuring technique

The most substructuring techniques focus on the nonlinear behavior of the lower stories, which are selected as the experimental substructure. Accordingly, the upper stories are considered as the numerical substructure in those conventional substructuring techniques. Differently, from those researches, these studies address the substructuring technique for the shaking table test of a building structure adopting the upper stories as the experimental substructure and the lower stories as the analytical substructure. Figure 2.1 shows the conceptual difference between the conventional and the proposed substructuring techniques as part of a hybrid testing method. The necessity of testing the upper substructure is found in diverse structures. For example, the wind-induced acceleration response of slender building structures is usually tremendous in the top story, and detailed experiment for this upper part of the building structure may be required, since excessive acceleration response may be harmful to human comfort or high precision equipment. Also, in some cases, a light appendage, for example, a penthouse, a small housing for mechanical equipment and an advertising billboard, may vibrate excessively rather than the primary structure. Besides, to suppress excessive vibration induced by winds and earthquakes, a tuned mass damper or tuned liquid damper is employed on the top story of the high-rise building,

and its performance should be verified in the laboratory before installation. Finally, the base isolation system to protect expensive machinery housed in a building structure needs experimental verification for a certain level of earthquake.

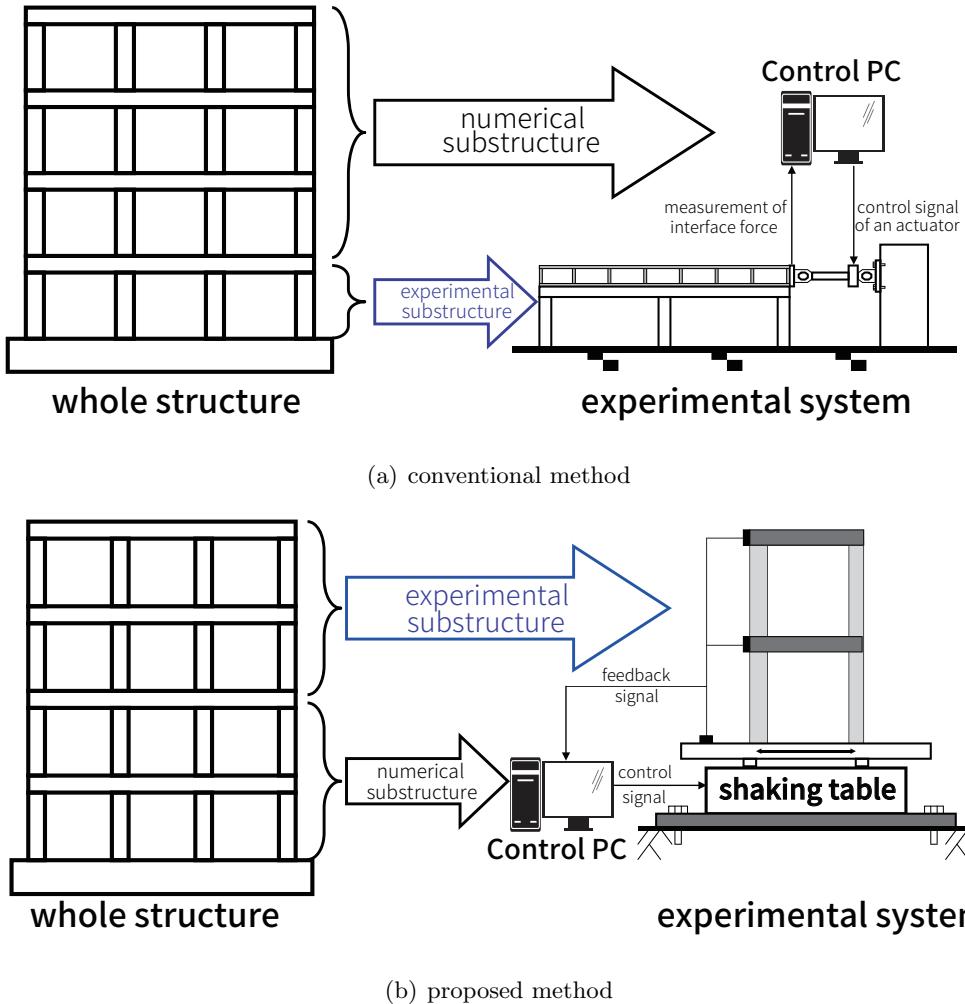


Figure 2.1: Conceptual illustrations of the substructuring methods

On these backgrounds, this subsection proposes a shaking table testing method using the upper part of a building structure as the experimental substructure based on the acceleration feedback from the experimental substructure. Also, the proposed method is verified experimentally through real-time implementation. Figure 2.1 illustrates the schematic diagram of the proposed testing method in this chapter. At first, the interface force acting between the upper experimental and lower numerical substructures is calculated based on the acceleration feedback of the experimental substructure which is mounted on shaking table. Then the interface acceleration required to replicate the dynamic behavior of the whole structure is calculated from the numerical substructure subjected to the interface force and assumed base motion for the entire structure. Finally, shaking table excites the upper experimental substructure according to the command signal from the shaking table

controller. The series of above processes are performed during a single time interval and repeated over the entire duration of the experiment.

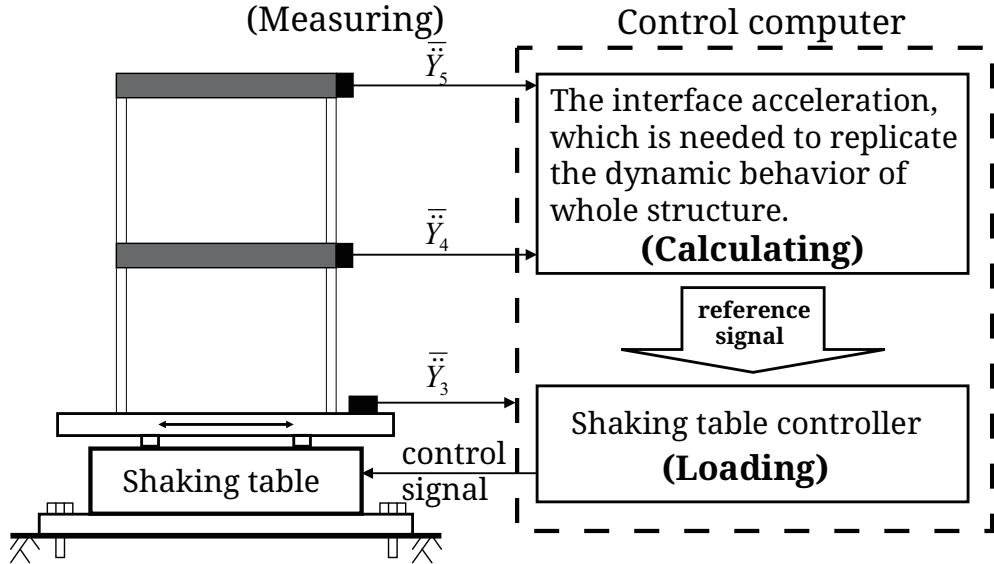


Figure 2.2: Schematic diagram of the proposed method

### 2.1.1 Formulation and Implementation Methodology

#### Experimental and Numerical Substructures

This subsection addresses physical interpretation and formulation of splitting the whole structure into the experimental and numerical substructures. A two-step process is carried out to formulate the equations of motion for those two substructures; separating the experimental substructure from the whole structure and constructing experimental substructure mounted on the shaking table and numerical substructures with external loads on the cutting plane, as shown in Figure 2.3.

The equation of motion of the whole shear-type building structure subjected to the ground acceleration,  $\ddot{Y}_g(t)$ , is represented in Figure 2.3(a) and expressed as

$$\mathbf{M}\ddot{\mathbf{Y}}(t) + \mathbf{C}\dot{\mathbf{Y}}(t) + \mathbf{K}\mathbf{Y}(t) = \mathbf{p}(t) \quad (2.1)$$

where,  $\mathbf{Y}(t)$  is the  $n \times 1$  vector of the absolute floor displacements given by  $\{Y_1, Y_2, \dots, Y_n\}^\top$ , and  $\mathbf{p}(t)$  is the  $n \times 1$  vector of the external force given by  $\{0, \dots, 0, c_1\dot{Y}_g(t) + k_1Y_g(t)\}^\top$ ,  $\mathbf{M}$ ,  $\mathbf{C}$  and  $\mathbf{K}$  are the structural mass, damping and stiffness matrices, respectively, and

expressed as follows.

$$\mathbf{M} = \begin{bmatrix} m_n & m_{n-1} & & \\ & \ddots & & \\ & & m_m & m_{m-1} \\ & & & \ddots \\ & & & & m_1 \end{bmatrix},$$

$$\mathbf{C} = \begin{bmatrix} c_n & -c_n & & \\ -c_n & c_n + c_{n-1} & -c_{n-1} & \\ & \ddots & \ddots & \\ & & -c_{m+1} & c_{m+1} + c_m \\ & & & -c_m & c_m + c_{m-1} & -c_{m-1} \\ & & & & \ddots & \ddots & \ddots \\ & & & & -c_2 & c_2 + c_1 & & \end{bmatrix}, \quad (2.2)$$

$$\mathbf{K} = \begin{bmatrix} k_n & -k_n & & \\ -k_n & k_n + k_{n-1} & -k_{n-1} & \\ & \ddots & \ddots & \\ & & -k_{m+1} & k_{m+1} + k_m \\ & & & -k_m & k_m + k_{m-1} & -k_{m-1} \\ & & & & \ddots & \ddots & \ddots \\ & & & & -k_2 & k_2 + k_1 & & \end{bmatrix}$$

The experimental and numerical substructures are separated from the whole structure by cutting at their interface, as shown in Figure 2.3(b). Mathematically, this operation means separating the equations included in Eq. (2.1) into two groups of which one corresponds to the upper part DOF, and the other corresponds to the lower part DOF. The dotted line in Eq. (2.2) divides those two groups, and the resulting two equations of motions are written as

$$\mathbf{M}_E \ddot{\mathbf{Y}}_E(t) + \mathbf{C}_E(t) \dot{\mathbf{Y}}_E(t) + \mathbf{K}_E \mathbf{Y}_E(t) = \mathbf{p}_E(t) \quad (2.3)$$

$$\mathbf{M}_N \ddot{\mathbf{Y}}_N(t) + \mathbf{C}_N(t) \dot{\mathbf{Y}}_N(t) + \mathbf{K}_N \mathbf{Y}_N(t) = \mathbf{p}_N(t) \quad (2.4)$$

where, subscripts  $E$  and  $N$  denote the experimental and numerical substructures, respectively.  $\mathbf{Y}_E$  and  $\mathbf{Y}_N$  are an  $l \times 1$  vector given by  $\{Y_{E(1)}, Y_{E(2)}, \dots, Y_{E(l)}\}^\top$  and an  $m \times 1$  vector given by  $\{Y_{N(1)}, Y_{N(2)}, \dots, Y_{N(m)}\}^\top$ , respectively. Accordingly,  $n$ , the length of the displacement vector, shown in Eq. (2.1) equals to  $m+l$ . Also,  $Y_{E(1)}$ ,  $Y_{E(l)}$ ,  $Y_{N(1)}$  and  $Y_{N(m)}$  in Eq.'s (2.3) and (2.4) correspond to  $Y_{m+1}$ ,  $Y_n$ ,  $Y_1$  and  $Y_m$  in Eq. (2.1), respectively. The mass, damping and stiffness matrices and the external force vectors in Eq. (2.3) and

(2.4) are expressed as follows.

$$\mathbf{M}_E = \begin{bmatrix} m_{E(l)} & & & \\ & \ddots & & \\ & & m_{E(2)} & \\ & & & m_{E(1)} \end{bmatrix},$$

$$\mathbf{C}_E = \begin{bmatrix} c_{E(l)} & -c_{E(l)} & & \\ -c_{E(l)} & c_{E(l)} + c_{E(l-1)} & -c_{E(l-1)} & \\ & \ddots & \ddots & \ddots \\ & & -c_{E(3)} & c_{E(3)} + c_{E(2)} \\ & & & -c_{E(2)} & c_{E(2)} + c_{E(1)} \end{bmatrix}, \quad (2.5)$$

$$\mathbf{K}_E = \begin{bmatrix} k_{E(l)} & -k_{E(l)} & & \\ -k_{E(l)} & k_{E(l)} + k_{E(l-1)} & -k_{E(l-1)} & \\ & \ddots & \ddots & \ddots \\ & & -k_{E(3)} & k_{E(3)} + k_{E(2)} \\ & & & -k_{E(2)} & k_{E(2)} + k_{E(1)} \end{bmatrix}$$

$$\mathbf{M}_N = \begin{bmatrix} m_{N(m)} & & & \\ & \ddots & & \\ & & m_{N(2)} & \\ & & & m_{N(1)} \end{bmatrix},$$

$$\mathbf{C}_N = \begin{bmatrix} c_{N(m)} & -c_{N(m)} & & \\ -c_{N(m)} & c_{N(m)} + c_{N(m-1)} & -c_{N(m-1)} & \\ & \ddots & \ddots & \ddots \\ & & -c_{N(3)} & c_{N(3)} + c_{N(2)} \\ & & & -c_{N(2)} & c_{N(2)} + c_{N(1)} \end{bmatrix}, \quad (2.6)$$

$$\mathbf{K}_N = \begin{bmatrix} k_{N(m)} & -k_{N(m)} & & \\ -k_{N(m)} & k_{N(m)} + k_{N(m-1)} & -k_{N(m-1)} & \\ & \ddots & \ddots & \ddots \\ & & -k_{N(3)} & k_{N(3)} + k_{N(2)} \\ & & & -k_{N(2)} & k_{N(2)} + k_{N(1)} \end{bmatrix}$$

$$\mathbf{p}_E(t) = \left\{ 0, \dots, 0, c_{E(1)} \dot{Y}_{N(m)} + k_{E(1)} Y_{N(m)} \right\}^\top \quad (2.7)$$

$$\mathbf{p}_N(t) = \left\{ c_{E(1)} (\dot{Y}_{E(1)} - \dot{Y}_{N(m)}) + k_{E(1)} (Y_{E(1)} - Y_{N(m)}), 0, \dots, 0, c_1 \dot{Y}_g(t) + k_1 Y_g(t) \right\}^\top \quad (2.8)$$

It is obvious from Eq. (2.7) that the absolute acceleration of the top story of the numerical substructure acts as the ground acceleration of the experimental substructure given by Eq. (2.3). This condition is physically realized by synchronizing the shaking table motion with the absolute acceleration of the top story of the numerical substructure. Besides, the first element of the external force vector in Eq. (2.8) means the interface force acting on the top story of the numerical substructure. This interface force, which is denoted by  $i(t)$  from now on, is produced by the damping and restoring forces of the first story in the upper experimental substructure and rewritten as follows.

$$i(t) = c_{E(1)} (\dot{Y}_{E(1)} - \dot{Y}_{N(m)}) + k_{E(1)} (Y_{E(1)} - Y_{N(m)}) \quad (2.9)$$

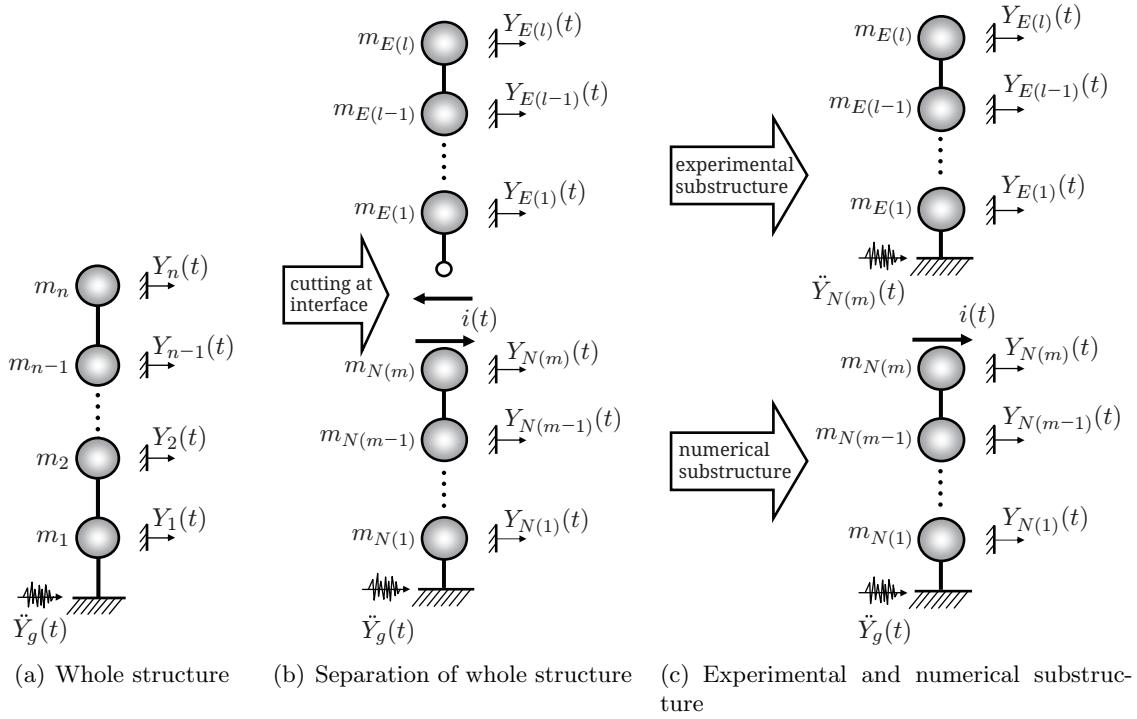


Figure 2.3: Concept of the proposed method

As a result, the numerical substructure, of which equation of motion is given by Eq. (2.4), is subjected to both the interface force at the top story and the ground motion at the base. The boundary condition and external load of each substructure are illustrated in Figure 2.3(c).

### Measurement of the interface force

In this section, the interface force,  $i(t)$ , which should be fed-back to the numerical substructure for calculating the interface acceleration, is indirectly measured by the accelerometers attached on the experimental substructure, as shown in Figure 2.1. Dynamic equilibrium in the experimental substructure is applied to derive the interface force using acceleration response measurement of the experimental substructure. As shown in Figure 2.4, dynamic equilibrium in the experimental substructure is established between the interface force acting at its base and the resultant of its inertial forces, because internal forces between each story are canceled with each other. This relation can be derived in another way. First, the dotted parts are shown in Eq. (2.5) are moved to the right side of Eq. (2.3). Then the relation of Eq. (2.9) is applied to the right side. Finally, the summation of  $l$  simultaneous equations for each side gives following expression of the interface force,  $i(t)$ .

$$i(t) = - \sum_{i=1}^l m_{E(i)} \ddot{Y}_{E(i)} \quad (2.10)$$

Eq. (2.10) physically implies that the interface force, which is required as an input data of the numerical substructure, can be calculated using the measured acceleration responses of the experimental substructure.

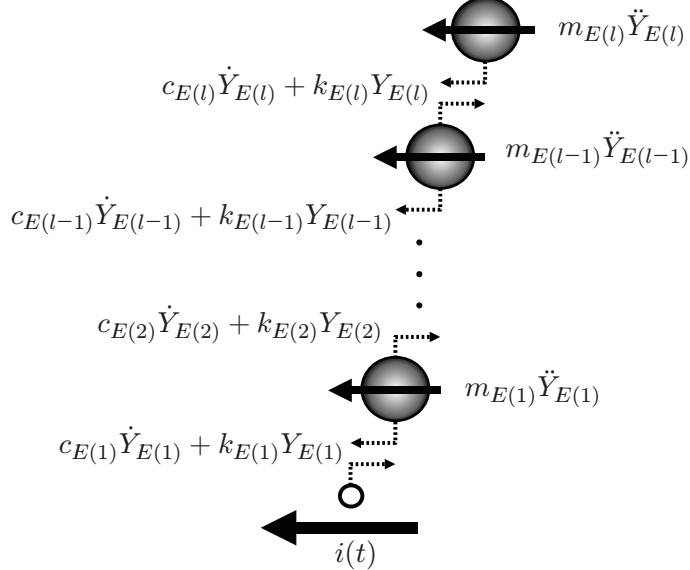


Figure 2.4: Dynamic equilibrium in experimental substructure

### Calculation of the numerical substructure

The calculation of the numerical substructure shown in Figure 2.1 means on-line time history analysis of the Eq. (2.4) that represents the dynamic equilibrium of the numerical substructure shown in Figure 2.3(c). To integrate this analysis procedure with shaking table control process implemented on a digital computer, Eq. (2.4) is transformed into state-space equations. The state-space equations, of which the output is the absolute accelerations of the numerical substructure, are given by the following equations in the continuous time domain.

$$\begin{aligned}\dot{\mathbf{z}}(t) &= \mathbf{A}\mathbf{z}(t) + \mathbf{B}\mathbf{u}(t) \\ \mathbf{O}(t) &= \mathbf{C}\mathbf{z}(t) + \mathbf{D}\mathbf{u}(t)\end{aligned}\tag{2.11}$$

where,  $\mathbf{z}(t)$  is the  $2m \times 1$  vector of state variables composed of the relative displacement and velocity of the numerical substructure.  $\mathbf{u}(t)$  is the  $2 \times 1$  vector of input variables given by  $\{i(t), \ddot{Y}_g(t)\}^\top$ , in which the interface force,  $i(t)$ , is fed-back from the experimental substructure using Eq. (2.10).  $\mathbf{O}(t)$  is the  $m \times 1$  vector of the output composed of the absolute accelerations of the numerical substructure. The system matrices  $\mathbf{A}$ ,  $\mathbf{B}$ ,  $\mathbf{C}$  and  $\mathbf{D}$  have the dimension of  $2m \times 2m$ ,  $2m \times 2$ ,  $m \times 2m$  and  $m \times 2$ , respectively, and are expressed as

$$\mathbf{A} = \begin{bmatrix} \mathbf{0}_{m \times m} & \mathbf{I}_{m \times m} \\ -\mathbf{M}_N^{-1}\mathbf{K}_N & -\mathbf{M}_N^{-1}\mathbf{C}_N \end{bmatrix} \quad (2.12)$$

$$\mathbf{B} = \begin{bmatrix} \mathbf{0} & \mathbf{0} \\ \mathbf{M}_N^{-1}\mathbf{b} & -\mathbf{1} \end{bmatrix} \quad (2.13)$$

$$\mathbf{C} = \begin{bmatrix} -\mathbf{M}_N^{-1}\mathbf{K}_N & -\mathbf{M}_N^{-1}\mathbf{C}_N \end{bmatrix} \quad (2.14)$$

$$\mathbf{D} = \begin{bmatrix} \mathbf{M}_N^{-1}\mathbf{b} & \mathbf{0} \end{bmatrix} \quad (2.15)$$

where,  $\mathbf{0}$  and  $\mathbf{I}$  are  $m \times m$  zero and unit matrices, respectively, and  $\mathbf{0}$  and  $-\mathbf{1}$  are an  $m \times 1$  vector of 0 and an  $m \times 1$  vector of  $-1$ , respectively.  $\mathbf{b}$  is an  $m \times 1$  vector given by  $\{1, 0, \dots, 0\}^\top$ . Eq. (2.11) with the inputs of the interface force and ground motion is incorporated in the ‘Calculating’ part shown in Figure 2.1, and produces the absolute acceleration responses of the numerical substructure on-line. Among those acceleration responses, the one corresponding to the top story is utilized as the reference signal to operate shaking table.

### 2.1.2 Numerical Substructure

For the verification experiment of the proposed methodology, the object structure is assumed to be a five-story shear type building structure, of which two upper stories are assumed to be the experimental substructure as shown in Figure 2.15. As a result, the numerical substructure is the three lower stories of the whole structure. For the construction of the finite element model of the numerical substructure, the inter-story damping and stiffness coefficients of the shear-type experimental substructure are identified based on the measured acceleration responses, and the mass of the two floors are measured directly. The system identification of the experimental substructure is conducted using the command `fmincon` in MATLAB(Coleman et al., 1999), which uses the interior-reflective Newton method to solve the constrained nonlinear minimization problem (Coleman, 1994; Coleman and Li, 1996). Resulting identified parameters can be summarized as follow.

$$\begin{aligned} m_{E(1)} &= m_4 = 2.04(\text{kg}), \quad m_{E(2)} = m_5 = 5.10(\text{kg}) \\ k_{E(1)} &= k_4 = 45.3(\text{N/m}), \quad k_{E(2)} = k_5 = 48.5(\text{N/m}) \\ c_{E(1)} &= c_4 = 0.023(\text{N} \cdot \text{s}/\text{m}), \quad c_{E(2)} = c_5 = 0.015(\text{N} \cdot \text{s}/\text{m}) \end{aligned} \quad (2.16)$$

which correspond to the first and second natural frequencies of 2.5 and 8.6Hz, respectively, as given in Table 2.1. Figure 2.5 compares the experimentally obtained transfer functions with numerically computed ones. As shown in Figure 2.5, measured transfer

functions agree well with those of the corresponding numerical model.

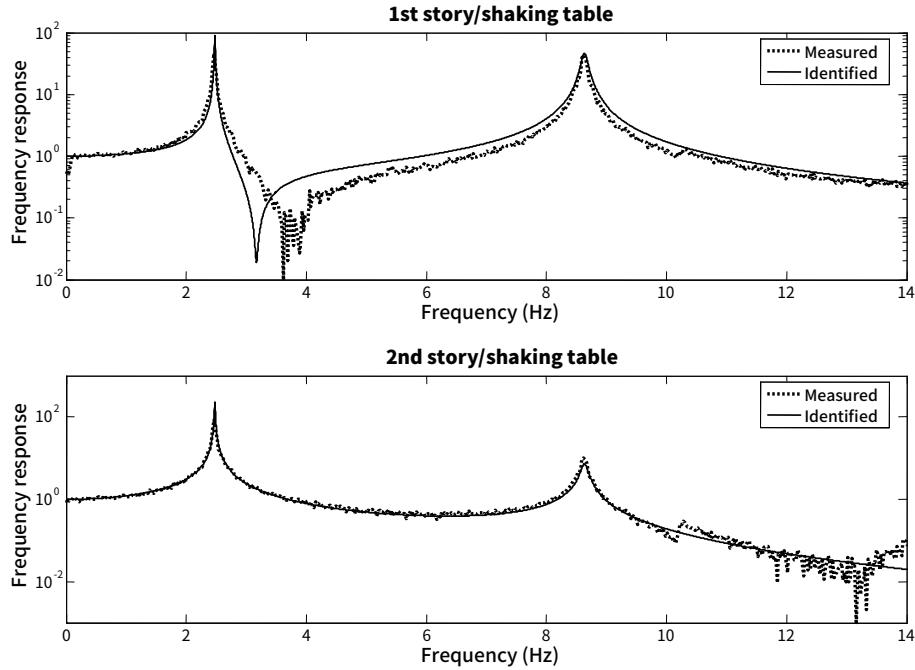


Figure 2.5: Comparison of experimental and approximated transfer functions.

Table 2.1: Frequencies observed in the time records of the experimental substructure from the test without feedback and the natural frequencies of the assumed whole structure

Mode	Frequency components(Hz) observed in the experimental substructure with 2 DOFs	Natural frequencies (Hz) of the whole structure with 5 DOFs
1	1.3	1.3
2	2.5	4.2
3	4.2	7.1
4	7.1	9.4
5	8.6	10.8

The structural parameters for the first story of the experimental structure given in Eq. (2.16) are applied to all stories of the numerical substructure and summarized as the following:

$$i(t) = -m_{E(1)}\ddot{Y}_{E(1)}(t) - m_{E(2)}\ddot{Y}_{E(2)}(t) \quad (2.17)$$

$$\ddot{Y}_{N(3)}(t) = \left[ -c_{N(3)}(\dot{Y}_{N(3)} - \dot{Y}_{N(2)}) - k_{N(3)}(Y_{N(3)} - Y_{N(2)}) + i(t) \right] / m_{N(3)} \quad (2.18)$$

Eq. (2.18) means that the interface force, which is produced by the shaking table, is calculated using the two measured absolute accelerations of the experimental substructure.

Then, the interface acceleration is calculated from the numerical substructure expressed by Eq. (2.11). Figure 2.6 represents the block diagram of the entire substructuring testing system including the numerical substructure marked by the shaded area. In Figure 2.6, the ground acceleration,  $\ddot{Y}_g$ , is not a measured signal but an input data prescribed by a user. Finally, the shaking table motion is controlled using the inverse transfer function of Eq. (2.36) to minimize the errors between the interface acceleration computed as the third story absolute acceleration of the numerical substructure,  $\ddot{Y}_{N(3)}$ , and the actual shaking table acceleration. Therefore, the shaking table itself works as the interface node of Figure 2.3(b) and excites the upper experimental substructure. In the actual implementation of the shaking table test, the continuous filters included in Figure 2.6 are converted into discrete filters with a time step of 0.01 sec.

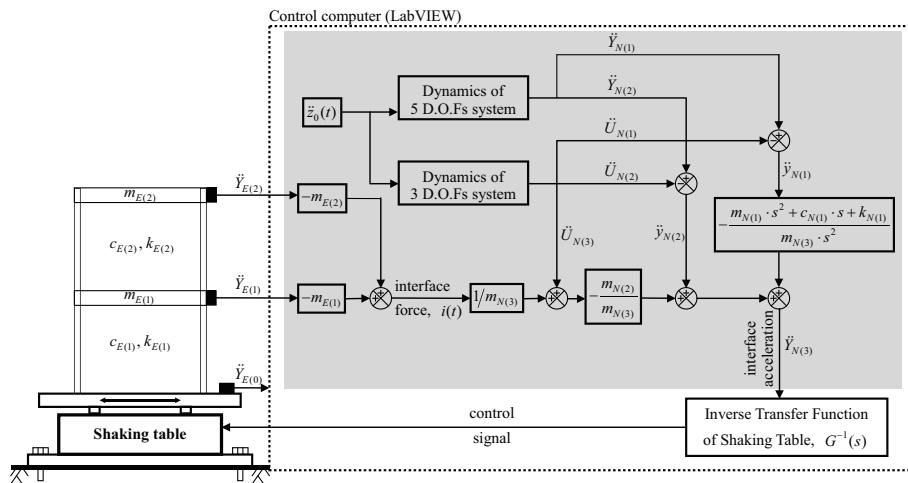


Figure 2.6: Block diagram of the integrated controller for the substructuring testing system.

## 2.2 Hybrid Testing Method for the Performance Evaluation of a Tuned Liquid Damper

In this section, the vibration control effect of a TLD for a building structure excited by earthquake load is experimentally evaluated through the hybrid testing method. The hybrid testing method does not require a physical building structural model in performing the experiment of a TLD–structure interaction system, and it only uses a TLD which is known to have strong nonlinearity dependent on response amplitude, excitation frequency, and depth to length ratio(Yu et al., 1999). The structural responses of the interaction system are calculated numerically in real time using the analytical structural model with the excitations of measured control force, user-defined base earthquake loads, and its state space realization incorporated in the integrated controller of the shaking table. Also, in order to minimize the distortion of the acceleration of the shaking table, the inverse transfer

function of the shaking table is identified, and its state space realization is implemented in the shaking table controller. The motion of shaking table behaves both the controlled and uncontrolled absolute acceleration of the TLD mounted floor by modulating the feedback gain of the shear force measured by the load cell installed between the TLD and the shaking table. Comparison between the structural responses obtained by the hybrid testing method and the conventional shaking table test of a single story steel frame with TLD is made in order to verify the accuracy of the hybrid testing method and the uncontrolled and TLD-controlled structural responses of a three-story structure are obtained by the hybrid testing method in both time and frequency domains.

### 2.2.1 Hybrid Testing Method with TLD

Figure 2.7 depicts the conceptual illustrations of the hybrid testing method for an  $n$ -degrees-of-freedom structural model which is excited by base acceleration and has a tuned liquid damper at its top story. First, the whole control system is separated into the lower part structure, and the upper part TLD and the interaction force between the structure and TLD are considered. The TLD with the interacting force at its bottom is physically tested, and the response of the structure with interacting force at the top floor and the base acceleration is numerically calculated by using the computer controlling the motion of the shaking table. Measurement of interacting force is easily accomplished by installing a shear-type load-cell at the bottom of the TLD, as shown in Figure 2.7. TLD-generated shear force is fed back to the control computer. With this fed-back interacting force, the structural response of the story, where a TLD is installed, is calculated using the numerical part. The shaking table excites the upper TLD according to this calculated response. This process is carried out on real-time.

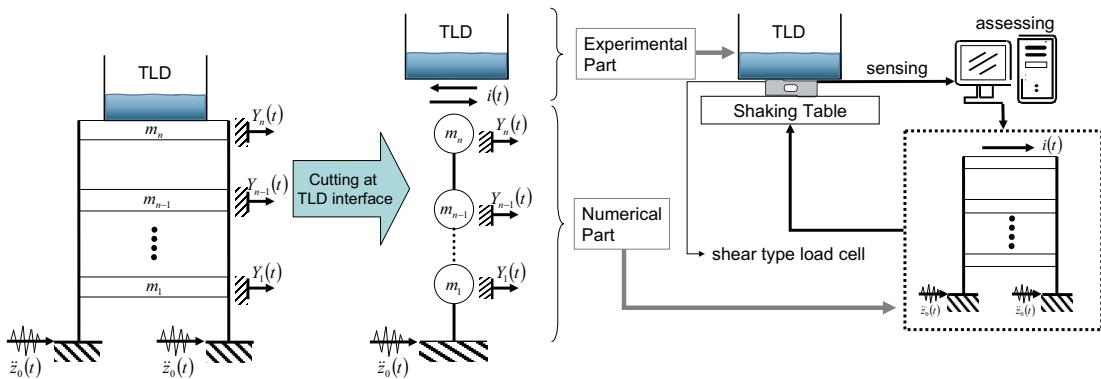


Figure 2.7: Conceptual view of the hybrid shaking table test

The numerical part with  $n$ -DOFs, which is subjected to the excitations of the measured control force,  $i_e(t)$ , and the input acceleration,  $\ddot{z}_0(t)$ , at its top and bottom, respec-

tively, as enclosed in dotted line in Figure 2.7, is calculated by

$$\mathbf{M}\ddot{\mathbf{Y}}_i(t) + \mathbf{C}\dot{\mathbf{Y}}_i(t) + \mathbf{K}\mathbf{Y}_i(t) = \mathbf{p}(t) \quad (2.19)$$

where,  $\mathbf{Y}_i(t)$  is the absolute displacement at the  $i$ th ( $i = 1 \rightarrow n$ ) story, and the location vector of external forces with the length of  $n$ ,  $\mathbf{p}(t)$  equals to  $\{-i_e(t), 0, \dots, 0, c_1\dot{z}_0(t) + k_1z_0(t)\}^\top$ , in which subscript “ $e$ ” denotes the “*experimentally*” measured interacting force. Also, the structural mass, damping and stiffness matrices are represented by

$$\begin{aligned} \mathbf{M} &= \begin{bmatrix} m_n & m_{n-1} & & \\ & \ddots & \ddots & \\ & & \ddots & m_1 \end{bmatrix}, \mathbf{C} = \begin{bmatrix} c_n & -c_n & & \\ -c_n & c_n + c_{n-1} & -c_{n-1} & \\ & \ddots & \ddots & \ddots \\ & & -c_2 & c_2 + c_1 \end{bmatrix}, \\ \mathbf{K} &= \begin{bmatrix} k_n & -k_n & & \\ -k_n & k_n + k_{n-1} & -k_{n-1} & \\ & \ddots & \ddots & \ddots \\ & & -k_2 & k_2 + k_1 \end{bmatrix} \end{aligned} \quad (2.20)$$

To calculate the numerical part such as Eq. (2.20) by a control computer on real-time, it is transformed into the state-space representation given by

$$\begin{aligned} \dot{\mathbf{z}}(t) &= \mathbf{A}_c \mathbf{z}(t) + \mathbf{B}_c \mathbf{u}(t) \\ \mathbf{O}(t) &= \mathbf{C}_c \mathbf{z}(t) + \mathbf{D}_c \mathbf{u}(t) \end{aligned} \quad (2.21)$$

where, the state variable vector,  $\mathbf{z}(t)$ , with the length of  $2n$  comprises the state variables,  $\{\mathbf{y}_i(t), \dot{\mathbf{y}}_i(t)\}^\top$ , in which the structural relative displacement,  $\mathbf{y}_i(t)$ , equals to  $\mathbf{Y}_i(t) - z_0(t)$ . The input vector,  $\mathbf{u}(t)$ , with the length of 2 consists of  $\{-i_e(t), \ddot{z}_0\}^\top$ . The output vector,  $\mathbf{O}(t)$ , with the length of  $n$  corresponds to the structural absolute acceleration,  $\ddot{\mathbf{Y}}_i(t)$ , itself. The matrices  $\mathbf{A}_c$ ,  $\mathbf{B}_c$ ,  $\mathbf{C}_c$  and  $\mathbf{D}_c$  with the sizes of  $2n \times 2n$ ,  $2n \times 2$ ,  $n \times 2n$  and  $n \times 2$ , respectively, are expressed as the following Eqs. (2.22)-(2.25).

$$\mathbf{A}_c = \begin{bmatrix} \mathbf{0}_{m \times n} & \mathbf{I}_{m \times n} \\ -\mathbf{M}^{-1}\mathbf{K} & -\mathbf{M}^{-1}\mathbf{C} \end{bmatrix} \quad (2.22)$$

$$\mathbf{B}_c = \begin{bmatrix} \mathbf{0}_{n \times 1} & \mathbf{0}_{n \times 1} \\ \mathbf{M}^{-1}\mathbf{b} & -\mathbf{1} \end{bmatrix} \quad (2.23)$$

$$\mathbf{C}_c = \begin{bmatrix} -\mathbf{M}^{-1}\mathbf{K} & -\mathbf{M}^{-1}\mathbf{C} \end{bmatrix} \quad (2.24)$$

$$\mathbf{D}_c = \begin{bmatrix} \mathbf{M}^{-1}\mathbf{b} & \mathbf{0}_{m \times 1} \end{bmatrix} \quad (2.25)$$

## 2.3 Hybrid testing method for the Performance Evaluation of a Tuned Liquid Column Damper

The tuned liquid column damper has received the attention of researchers as a type of auxiliary mass system(Samali et al., 1998). TLCD has the control characteristics similar to that of tuned mass damper, which is one of most frequently used dampers for vibration control. Since the viscosity term in the governing equation of motion of TLCD is a function of the absolute value of liquid velocity, the equation is nonlinear, and the dynamic characteristics of TLCD depend on the magnitude and the characteristics of excitation forces and the corresponding structural responses of the floor at which TLCD is installed(Yalla, 2001). In this section, the vibration control effect of a TLCD for a building structure excited by earthquake load is experimentally evaluated through the hybrid testing method. The hybrid testing method does not require a physical building structural model in performing the experiment of a TLCD-structure interaction system, and it only uses a TLCD. The structural responses of the interaction system are calculated numerically in real time using the analytical structural model with the excitations of measured control force, user-defined base earthquake loads, and its state space realization incorporated in the integrated controller of the shaking table. Also, in order to minimize the distortion of the acceleration of the shaking table, the inverse transfer function of the shaking table is identified, and its state space realization is implemented in the shaking table controller. The shaking table behaves as the absolute acceleration of the TLCD mounted floor by calculating the fed back signal of the shear force signal measured by the load cell positioned between the TLCD and plate of the shaking table. Comparison results between the structural responses obtained by the hybrid testing method and the conventional shaking table test of a single story steel frame with TLCD are made to verify the accuracy of the hybrid testing method in both time and frequency domains.

### 2.3.1 Hybrid Testing Method with TLCD

Figure 2.8 shows the conceptual view of the experiment. The whole structural control system, which a TLCD was installed onto the structural model with n-degrees-of-freedom at its top story, is separated, and as the result of that, the force interacts at their interface. The upper part of TLCD with the interacting force at its bottom is physically tested and the lower part of the structural model with the interacting force and the input motion at its top story and base, respectively, is numerically calculated within the computer to control the motion of shaking table. For the experimental implementation, interacting or control force generated by a TLCD, which is observed from a load-cell, is fed back to the control computer. With the fed-back interacting force, the structural response of the story,

where a TLCD is incorporated, is calculated from the numerical part. The shaking table excites the upper TLCD with this calculated response. These processes are carried out on real-time.

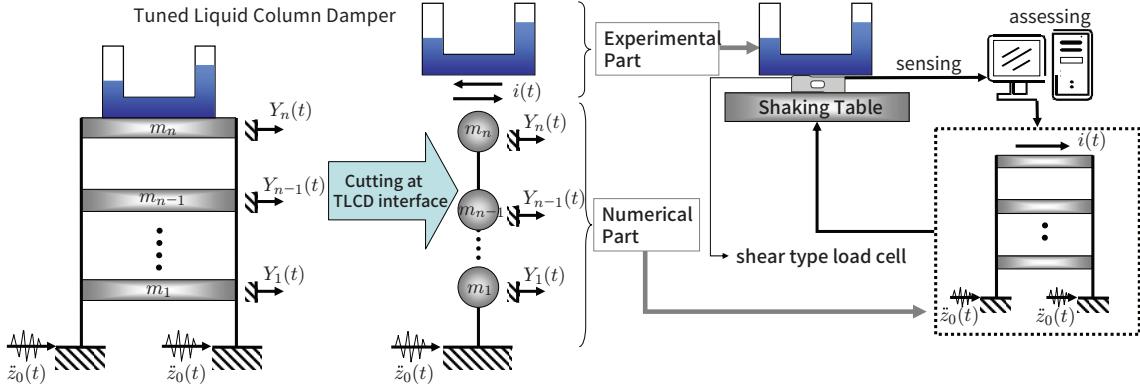


Figure 2.8: Concept of the hybrid testing method (TLCD)

Of these procedures, the numerical part with  $n$ -DOFs, which is subjected to the excitations of the experimentally measured control force,  $i_e(t)$ , and the input acceleration,  $\ddot{z}_0(t)$ , at its top and bottom, respectively, as enclosed in dotted line in Figure 2.8, is calculated by

$$\mathbf{M}\ddot{\mathbf{Y}}_i(t) + \mathbf{C}\dot{\mathbf{Y}}_i(t) + \mathbf{K}\mathbf{Y}_i(t) = \mathbf{p}(t) \quad (2.26)$$

where,  $\mathbf{Y}_i(t)$  is the absolute displacement at the  $i$ th ( $i = 1 \rightarrow n$ ) story, and the location vector of external forces with the length of  $n$ ,  $\mathbf{p}(t)$  equals to  $\{-i_e(t), 0, \dots, 0, c_1\dot{z}_0(t) + k_1z_0(t)\}^\top$ , in which subscript “e” denotes the “experimentally” measured interacting force. Also, the structural mass, damping and stiffness matrices are represented by

$$\begin{aligned} \mathbf{M} &= \begin{bmatrix} m_n & m_{n-1} & & \\ & \ddots & \ddots & \\ & & \ddots & m_1 \end{bmatrix}, \quad \mathbf{C} = \begin{bmatrix} c_n & -c_n & & \\ -c_n & c_n + c_{n-1} & -c_{n-1} & \\ & \ddots & \ddots & \ddots \\ & & -c_2 & c_2 + c_1 \end{bmatrix}, \\ \mathbf{K} &= \begin{bmatrix} k_n & -k_n & & \\ -k_n & k_n + k_{n-1} & -k_{n-1} & \\ & \ddots & \ddots & \ddots \\ & & -k_2 & k_2 + k_1 \end{bmatrix} \end{aligned} \quad (2.27)$$

To calculate the numerical part such as Eq. (2.20) by a control computer on real-time, it is transformed into the state-space representation given by

$$\begin{aligned} \dot{\mathbf{z}}(t) &= \mathbf{A}_c \mathbf{z}(t) + \mathbf{B}_c \mathbf{u}(t) \\ \mathbf{O}(t) &= \mathbf{C}_c \mathbf{z}(t) + \mathbf{D}_c \mathbf{u}(t) \end{aligned} \quad (2.28)$$

where, the state variable vector,  $\mathbf{z}(t)$ , with the length of  $2n$  comprises the state variables,  $\{\mathbf{y}_i(t), \dot{\mathbf{y}}_i(t)\}^\top$ , in which the structural relative displacement,  $\mathbf{y}_i(t)$ , equals to  $\mathbf{Y}_i(t) - z_0(t)$ . The input vector,  $\mathbf{u}(t)$ , with the length of 2 consists of  $\{-i_e(t), \ddot{z}_0(t)\}^\top$ . The output

vector,  $\mathbf{O}(t)$ , with the length of  $n$  corresponds to the structural absolute acceleration,  $\ddot{\mathbf{Y}}_i(t)$ , itself. The matrices  $\mathbf{A}_c$ ,  $\mathbf{B}_c$ ,  $\mathbf{C}_c$  and  $\mathbf{D}_c$  with the sizes of  $2n \times 2n$ ,  $2n \times 2$ ,  $n \times 2n$  and  $n \times 2$ , respectively, are expressed as the following Eqs. (2.29)-(2.32).

$$\mathbf{A}_c = \begin{bmatrix} \mathbf{0}_{m \times n} & \mathbf{I}_{m \times n} \\ -\mathbf{M}^{-1}\mathbf{K} & -\mathbf{M}^{-1}\mathbf{C} \end{bmatrix} \quad (2.29)$$

$$\mathbf{B}_c = \begin{bmatrix} \mathbf{0}_{n \times 1} & \mathbf{0}_{n \times 1} \\ \mathbf{M}^{-1}\mathbf{b} & -\mathbf{1} \end{bmatrix} \quad (2.30)$$

$$\mathbf{C}_c = \begin{bmatrix} -\mathbf{M}^{-1}\mathbf{K} & -\mathbf{M}^{-1}\mathbf{C} \end{bmatrix} \quad (2.31)$$

$$\mathbf{D}_c = \begin{bmatrix} \mathbf{M}^{-1}\mathbf{b} & \mathbf{0}_{m \times 1} \end{bmatrix} \quad (2.32)$$

where,  $\mathbf{0}_{n \times n}$  and  $\mathbf{I}_{n \times n}$  are the zero and unit matrices, respectively, with the size of  $n \times n$ .  $\mathbf{0}_{n \times 1}$  and  $-\mathbf{1}$  are the vector whose components are 0 and  $-1$ , respectively, with the length of  $n \times 1$ .  $\mathbf{b}$  equal to  $\{1, 0, \dots, 0\}^\top$  with the length of  $n \times 1$ .

## 2.4 Hybrid Testing Method for the Performance Evaluation of a Tuned Liquid Mass Damper

In previous study(Heo et al., 2009), a new control device, which is called tuned liquid mass damper (TLMD), was developed and discussed in this section. The dynamic characteristics of a TLMD used in this thesis are that its mass is composed of both a mass of TLCD frame itself and that of liquid in a tank. Natural rubber columns were used to substitute the stiffness of a TMD. Therefore, a TLMD operates as a TLCD in one direction and behaves as a TMD in the other orthogonal direction.

In this section, the control performance of the proposed TLMD for reducing bidirectional responses of building structures is experimentally verified through both a conventional structural testing method and hybrid testing method. First, the control performance of a TLMD is evaluated by forced vibrating sinusoidal signal to an experimental prototype which is composed of both a TLMD and a building structure. Then, the hybrid testing method is performed to evaluate the performance of a TLMD, in which the building structural model is used as a numerical part, and the TLMD is experimentally tested.

### 2.4.1 Developed TLMD Model

Figure 2.9 shows the plan view of the TLMD used in this thesis. The bidirectional TLMD would be installed in the POSCO New Songdo City Tower 1A in Incheon, South Korea with 64 stories and 236 meters in height. An eigenvalue analysis of the tower resulted in

0.182 and  $0.162\text{Hz}$  in the  $x, y$ -directions, respectively, for the first natural frequencies, and 34,000*tons* for the first modal mass. The total mass of the bidirectional TLMD is 600*tons* that results from the effective mass ratio of about 1.76%.

The experimental building and TLMD models were manufactured by applying the scaling factors given in Table 2.2 to the first modal properties of the prototype tower. The building model with the parameters shown in Table 2.3 was made by applying the scaling factors in Table 2.2 to the prototype building. Applying the scale  $1/\sqrt{20}$  in frequency to the natural frequencies of a prototype model, 0.182 and  $0.162\text{Hz}$ , gives 0.82 and  $0.73\text{Hz}$  in the  $x$ - and  $y$ -directions, respectively. Also, the mass of a building structure is reduced by 4,250*kg*. The stiffness of a building model is calculated to be  $4250(\text{kg}) \times (2\pi \times 0.82(\text{Hz}))^2 \cong 110000(\text{N/m})$  in the  $x$ -direction and  $4250(\text{kg}) \times (2\pi \times 0.73(\text{Hz}))^2 \cong 89,000(\text{N/m})$  in the  $y$ -direction, respectively.

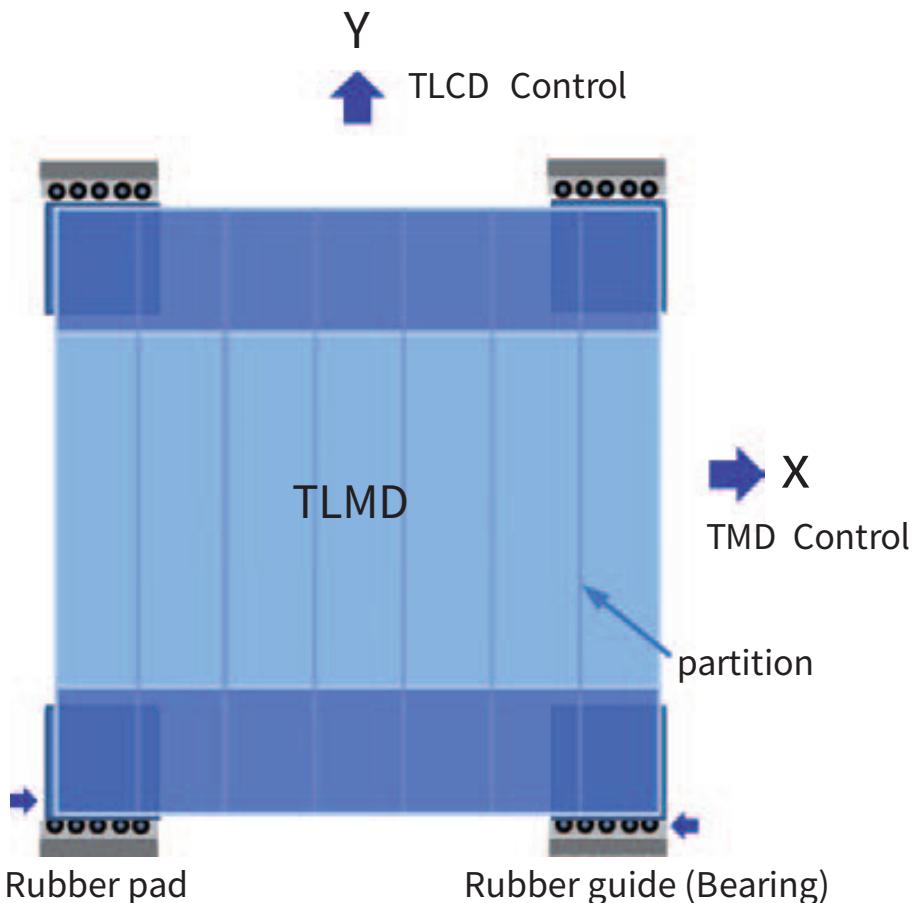


Figure 2.9: Concept of a TLMD

As shown in Table 2.4, it is obvious that the natural frequencies of a TLMD tune to those of a building structure in the TMD( $x$ ) and TLCD( $y$ )-control directions, respectively. The masses of a TLMD with the ratios of 1.8 and 0.8% to the mass of a building model become 75*kg* in the TMD-control and 32*kg* in the TLCD-control directions, respectively.

Table 2.2: Similitude law applied to TLMD model

Quantity	Dimension	Scaling factor
Length	$L$	1 : 20
Mass	$M$	$1 : 20^3 = 1 : 8000$
Frequency (Hz)	$T^{-1}$	$1 : 1/\sqrt{20} = 1 : 0.223$
Acceleration	$LT^{-2}$	1 : 1

Table 2.3: Design parameters of the building model

Design parameter	$y$ -direction	$x$ -direction
Siffness (N/m)	88,000	110,000
Frequency (Hz)	0.73	0.82
Mass (kg)	4,250	4,250

Table 2.4: Design parameters of a TLMD model

Design parameters	TLCD control direction	TMD control direction
Siffness or liquid length	$0.98m$ (liquid length)	$1990N/m$ (stiffness)
Frequency (Hz)	0.73	0.82
Mass (kg)	35	75

The stiffness of rubber columns used in a TMD,  $k$ , and the liquid length of a TLCD,  $L$ , are determined by

$$k = m (2\pi f_m)^2 \quad (2.33)$$

$$L = \frac{2g}{(2\pi f_L)^2} \quad (2.34)$$

where  $m$  and  $f_m$  are the mass and the tuned frequency of a TLCD, respectively, in the TMD-control direction.  $g$  and  $f_L$  are the gravity acceleration and the tuned frequency of the TLCD, respectively, in the TLCD-control direction. With these relations, the final parameters of a TLMD are shown in Table 2.4.

## 2.5 Design of Experimental Controller

### 2.5.1 Experimental System for Substrucrubg Technique

The experimental system is shown in Figure 2.10 and 2.11 was equipped in Seismic Retrofitting & Remodeling Research Center at the Dankook University, Seoul, Korea. The test structure (the experimental substructure) used in this experiment is a two-story steel frame with a single bay. The height and width of the experimental substructure are 1.0 and 0.6 m, respectively. The first floor mass,  $m_{E(1)}$ , and the second floor mass,  $m_{E(2)}$ , are 2.04 and 5.10 kg, respectively. The experimental substructure is excited by a uni-axial shaking table. The accelerometers are attached to each floor of the experimental substructure to measure its absolute floor acceleration responses. Additionally, an accelerometer is placed on the shaking table to monitor its motion. The data acquisition and implementation of the digital shaking table controller are performed using a real-time digital signal processor (DSP). The major task of the data acquisition board is to carry out the analog-to-digital (A/D) conversion of the measured acceleration data and the digital-to-analog (D/A) conversion of the reference signal computed by the shaking table controller, which is programmed using LabVIEW (Bishop, 2007). An 8-channel data acquisition system was employed using an NI PCI-6052E board and an NI SC-2345 BNC cable connector as a signal conditioner.

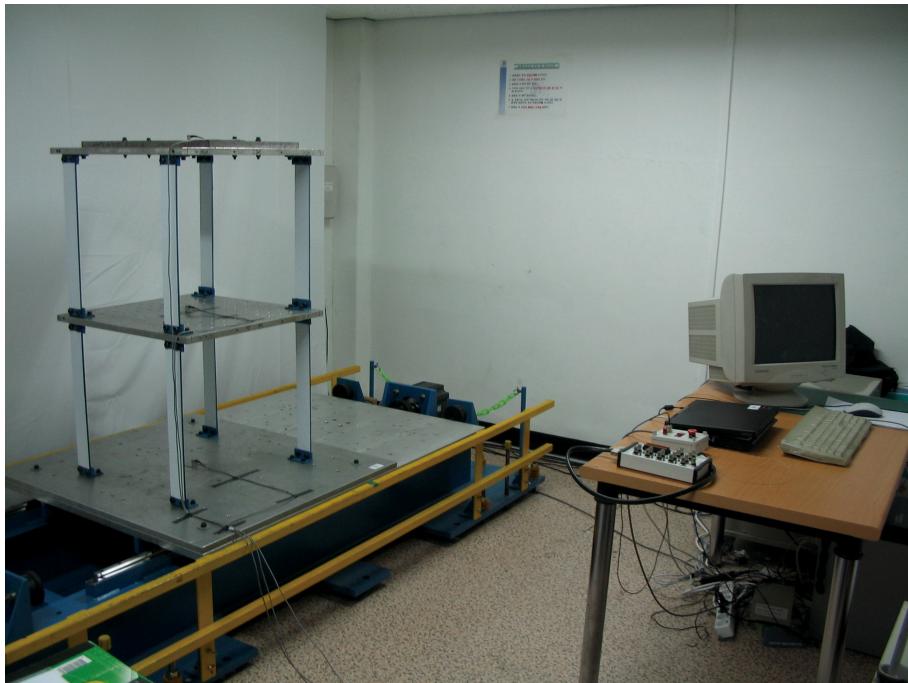


Figure 2.10: Overall view of experimental system

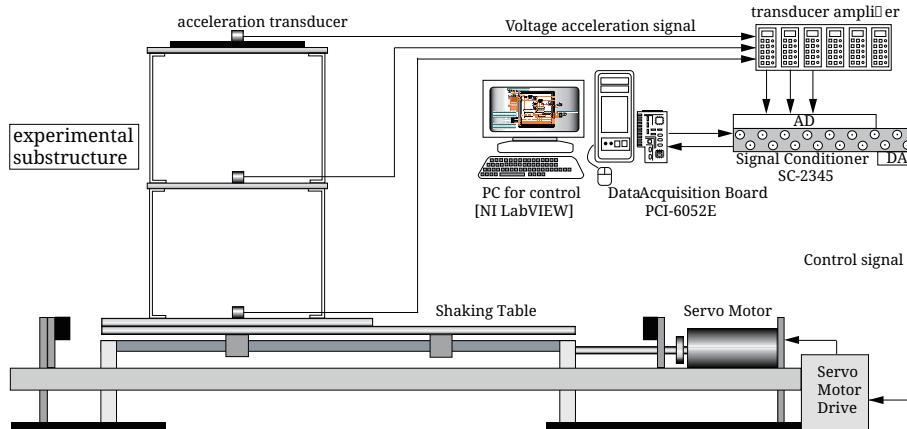


Figure 2.11: Schematic diagram of experimental system

### 2.5.2 Shaking table controller

The composition of the experimental system is illustrated in Figure 2.11. The shaking table shown in Figure 2.11 moves in accordance with the control signal, which is generated by the control computer and sent through D/A conversion board. In almost every case, the target acceleration signal and actual acceleration produced by the shaking table are different in their amplitudes and phases due to the shaking table dynamics. Therefore, in order to compensate the distortion of the actual shaking table acceleration against the shaking table dynamics existing between the reference signal and the actual measured acceleration of the shaking table, the inverse transfer function of the actual acceleration of shaking table with respect to the command signal generated by the control computer is constructed and implemented in the shaking table control computer as shown in Figure 2.15. First, the inverse transfer function, of which amplitude and phase are represented in Figure 2.12 by the dashed line, is obtained experimentally. Then, the experimental inverse transfer function is approximated by a rational function for its implementation in the control computer. In this verification experiment, the inverse transfer function of the shaking table is approximated using the command `invfreqs` in MATLAB (Little and Shure, 1992), which adopts the damped Gauss-Newton method for the iterative search to minimize the sum of the squared error between the measured and the desired frequency response points(Dennis Jr and Schnabel, 1983). The approximation result is given by the following 5-th order linear filter and compared with the experimental one in Figure 2.12.

Inverse transfer function, Eq. (2.35), corresponds to the shaking table controller of Figure 2.1.

$$G^{-1}(s) = \frac{0.6s^5 + 94s^4 + 10746s^3 + 498200s^2 + 167124s + 108216}{s^5 + 204s^4 + 15900s^2 + 8252s^2 + 4676s + 405} \quad (2.35)$$

For the implementation in the digital computer, Eq. (2.35) is realized into the following

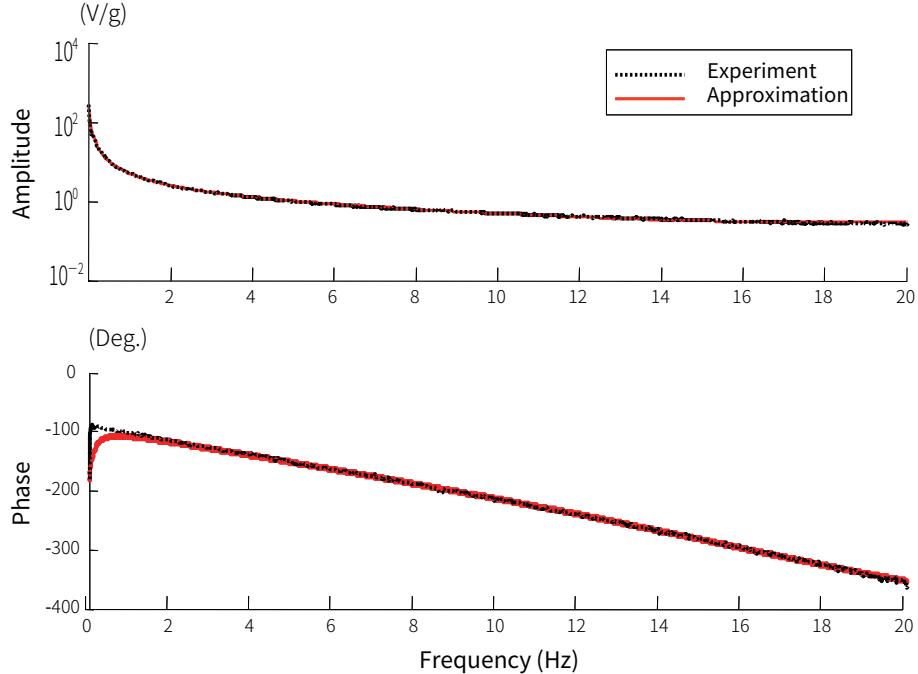


Figure 2.12: Inverse transfer function of shaking table

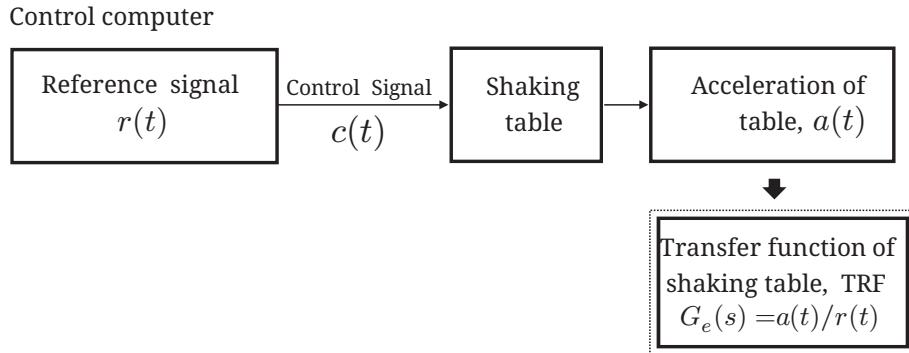


Figure 2.13: Definition of the transfer function of shaking table

state equation.

$$\begin{aligned}\dot{\mathbf{x}}_c &= \mathbf{A}_c \mathbf{x}_c + \mathbf{B}_c r_c \\ y_c &= \mathbf{C}_c \mathbf{x}_c + D_c r_c\end{aligned}\tag{2.36}$$

where,  $\mathbf{x}_c$ ,  $r_c$  and  $y_c$  is the state vector, the reference signal, the control signal of the shaking table controller, respectively.  $\mathbf{A}_c$ ,  $\mathbf{B}_c$ ,  $\mathbf{C}_c$  and  $D_c$  is the  $5 \times 5$  state matrix, the  $5 \times 1$  reference signal influence matrix, the  $1 \times 5$  output matrix and the coupling coefficient between the reference and control signal.

In order to verify the performance of the shaking table controller, a down-scaled El Centro earthquake is input to the inverse transfer function of the shaking table. Then, the corresponding acceleration of the shaking table is measured. Figure 2.16 compares the

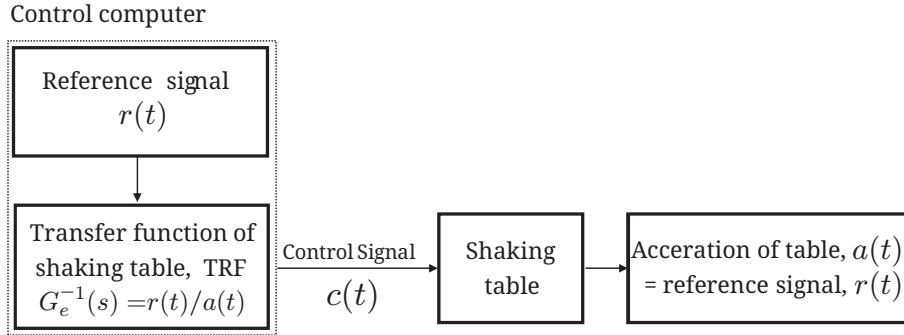


Figure 2.14: Compensation using the inverse transfer function of shaking table

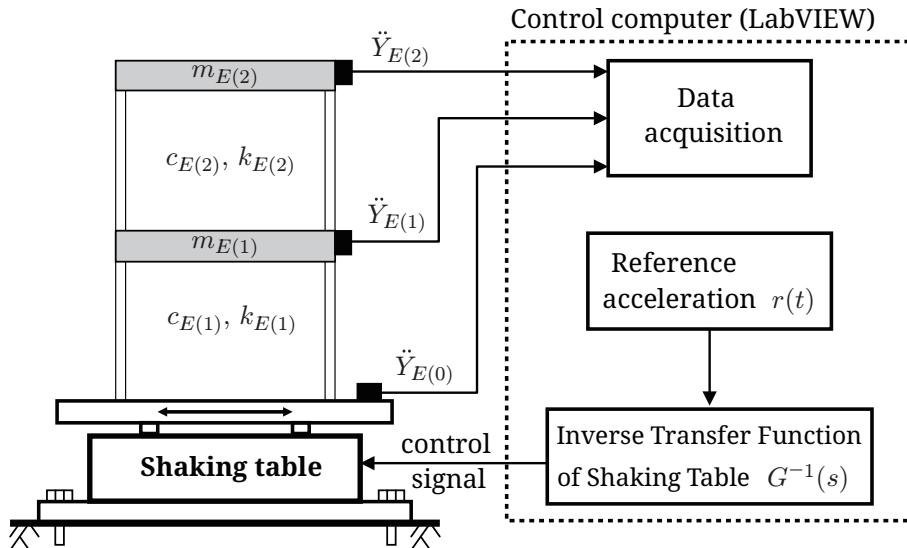


Figure 2.15: Flow chart of the experimental system controller

reference acceleration with the corresponding measured acceleration of the shaking table. It is observed that they agree well with each other.

### 2.5.3 Experimental System for Hybrid Testing of Building with TLD

#### Experimental setup

In order to experimentally verify the hybrid testing method, an experimental system shown in Figure 2.17 was set up in Seismic Retrofitting & Remodeling Research Center at the Dankook University, Seoul, Korea. The TLD was uniaxially excited by the shaking table on which it was mounted. The shear-type load-cell was inserted between the TLD and the shaking table to measure the base shear force yielded by the horizontal motion of the TLD during the test. Also, an accelerometer was attached to the shaking table to monitor its motion.

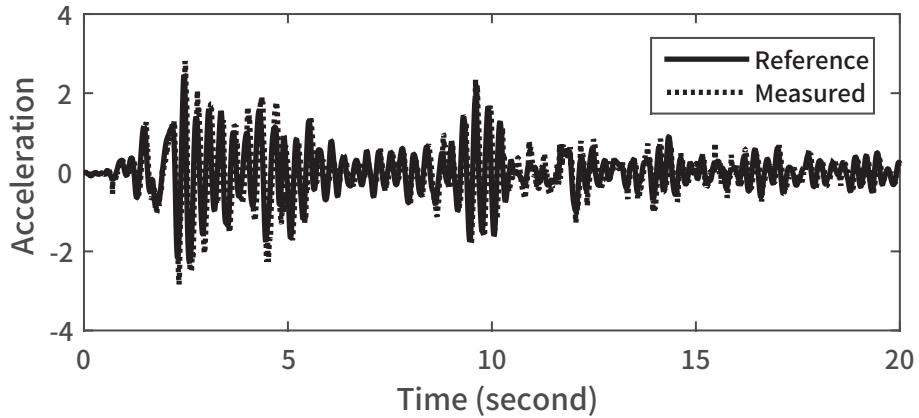


Figure 2.16: Compensation result for the dynamic characteristic of shaking table

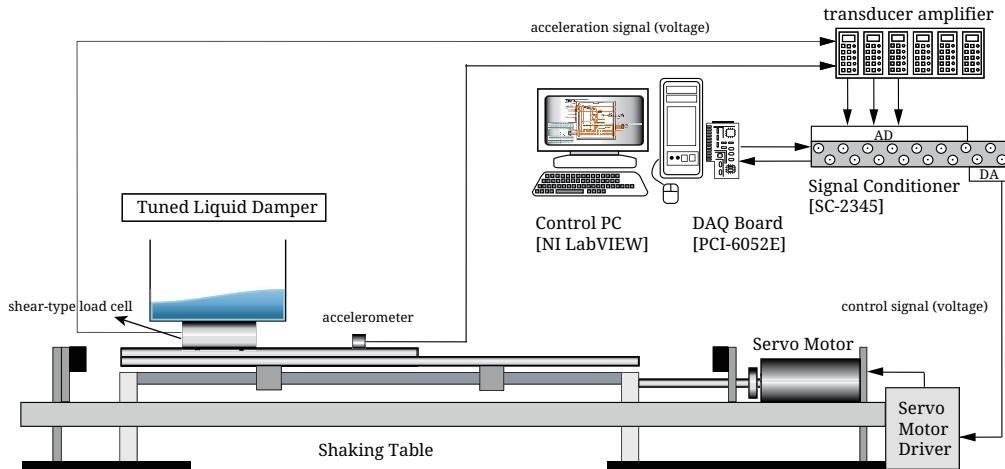


Figure 2.17: Schematic diagram of experimental set-up

### Integrated Controller of the Numerical Structural Model and the Shaking Table

The numerical structural model and the shaking table dynamics discussed in the previous subsections are integrated into the controller to implement the hybrid testing method. Figure 2.18 illustrates the block diagram of the hybrid testing method. In the figure, the absolute acceleration is produced by the numerical structural model of Eq. (2.21) with two inputs of the measured interacting force,  $i_e(t)$ , and not the measured but the prescribed earthquake record signal,  $\ddot{z}_0(t)$ , as marked by the shaded area. The motion of the shaking table is driven by the controller using the inverse transfer function to minimize the error between the absolute acceleration,  $\ddot{Y}_n(t)$ , calculated as the top story response of the structure and the actual shaking table acceleration,  $\ddot{Y}_e(t)$ . Accordingly, the shaking table itself behaves as the top story of the structure, at which a TLD is installed, and excites the upper TLD that should be physically tested.

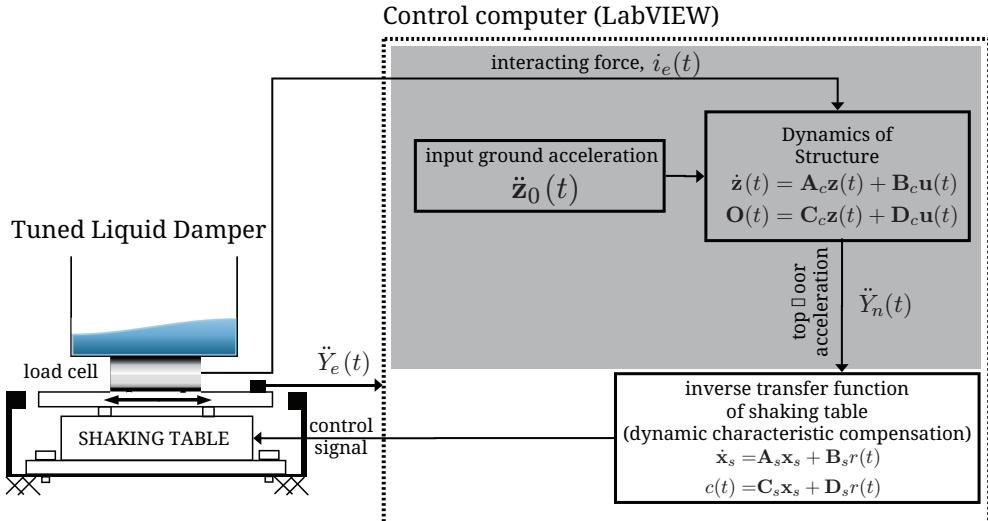


Figure 2.18: Block diagram of the integrated controller for the hybrid testing system.

#### 2.5.4 Experimental System for Hybrid Testing of Building with TLCD

The numerical part and the shaking table controller discussed in previous subsections should be integrated into the controller to implement the hybrid testing shown in Figure 2.8. Figure 2.19 illustrates the block diagram for experimentally implementing the testing method. In the figure, the absolute acceleration is produced by the numerical part such as Eq. (2.28) with two inputs of the measured interacting force,  $i_e(t)$ , and not the measured but the prescribed earthquake record signal,  $\ddot{z}_0(t)$ , by a user in the control computer, as marked by the shaded area. The motion of shaking table is driven by the controller using the inverse transfer function to minimize the error between the controlled absolute acceleration,  $\ddot{Y}_n(t)$ , calculated as the top story response of structure and the actual shaking table acceleration,  $\ddot{Y}_e(t)$ . Accordingly, the shaking table itself behaves as the top story of structure, at which a TLCD is installed, and excites the upper TLCD that should be physically tested. To verify the hybrid testing method, firstly the conventional TLCD-structure interaction model shown in Figure 2.20(a) is experimentally implemented. Then, the hybrid test is shown in Figure 2.20(b), which the structural model in Figure 2.20(a) is incorporated in the numerical calculation of its identified damping and stiffness coefficients and measured mass, is performed for the controlled case. Finally, two results from controlled cases are compared for the experimental verification of the hybrid testing method.

The only shear-type structural model without the upper TLCD shown in Figure 2.20(a) has the  $0.6m$  and  $1.0m$  of width and height and  $169.7kg$  of measured floor mass. Two records of El Centro and Kobe earthquake waves were excited by the shaking table to measure the absolute structural acceleration. The identification was conducted with measured accelerations of the structure model and the shaking table. The identified parameters have slight differences according to input earthquake waves. The averaged damping, and

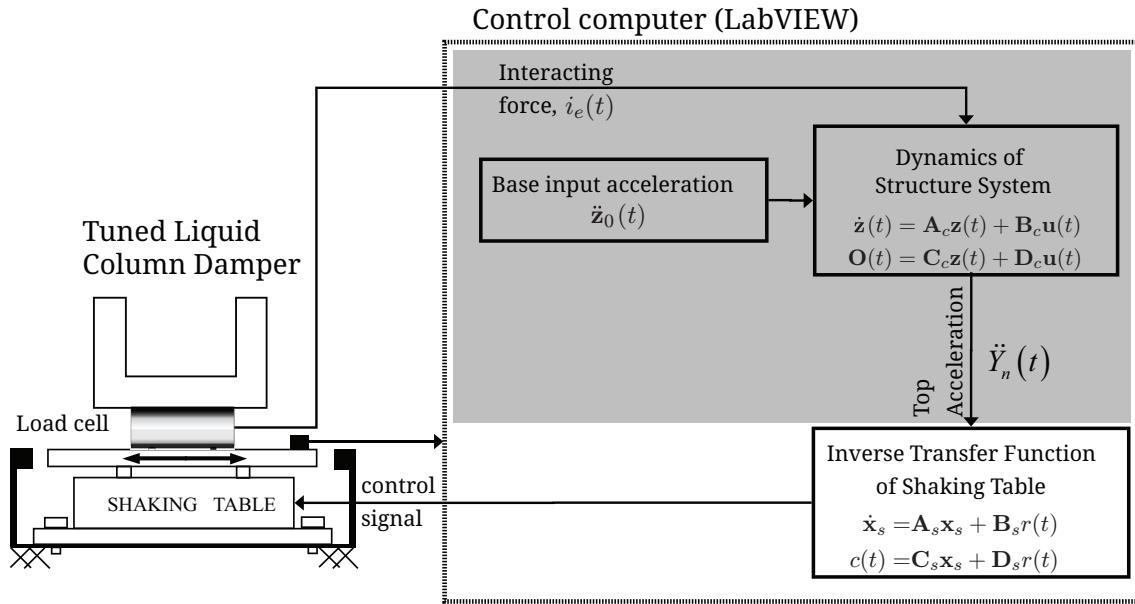


Figure 2.19: Controller for implementing the hybrid testing method

stiffness coefficients were determined by  $14.6 N \cdot s/m$  and  $9914.3 N/m$ , respectively, which correspond to  $1.23 Hz$  of structural natural frequency. The level of water in a TLCD tank was adjusted to sympathize the TLCD frequency to this identified structural one.



(a) conventional testing method



(b) hybrid testing method

Figure 2.20: Experimental view of a building with a TLCD

### 2.5.5 Experimental System for Hybrid Testing of Building with TLMD

The difference between the conventional test method and hybrid testing method is that the upper TLMD is adopted as the experimental part, while the structural model is numerically calculated by a computer in the hybrid testing method, as shown in Figure 2.21. The control force acting between their interfaces is measured with a shear-type load cell which is mounted on the shaking table. Then, the measured force is fed-back to the numerical analysis part. Finally, the shaking table vibrates the upper experimental part with the responses calculated from the numerical analysis part.

In this case, the shaking table is moved by the control signal sent from the control computer through DA channel of DAQ board. The amplitude and phase of the signal values measured at the shaking table, however, are different from the control signal sent from the control computer. In this thesis, the inverse transfer function of a shaking table was designed through the definition of the transfer function of a shaking table, as shown in Figure 2.13. Then, a white-noise excitation test of the shaking table is carried out to compensate the dynamic characteristics between the shaking table and control signals, as shown in Figure 2.14.

Figure 2.12 shows the inverse transfer function of the shaking table which was measured by the acceleration signal as an input data and the command signal as an output data. The measured inverse transfer function of the shaking table could be approximated by using a fifth-order linear filter represented by

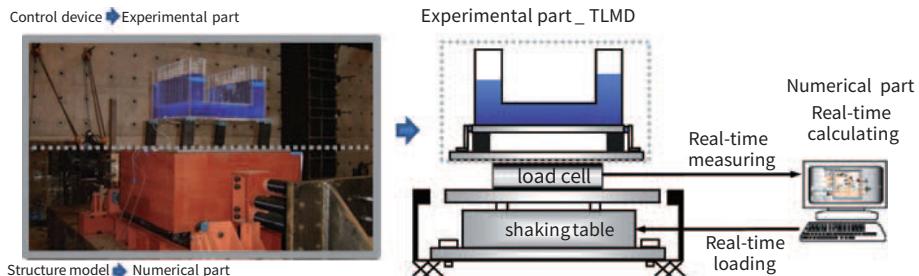


Figure 2.21: Conceptual view of the hybrid testing method of building with TLMD

#### Experimental setup

The SDOF structure subjected to both the external and control forces is given by

$$m\ddot{x} + c\dot{x} + kx = F(t) - i(t) \quad (2.37)$$

where  $m$ ,  $c$  and  $k$  are the mass, damping coefficient and stiffness of the SDOF structure, respectively.  $F(t)$  and  $i(t)$  are the excitation force and TLMD control force measured at the load cell, respectively.

The state-space equation and the output equation for the absolute acceleration of the SDOF structural model such as Eq. (2.37) are represented by

$$\dot{\mathbf{z}} = \mathbf{Az} + \mathbf{Bu} \quad (2.38)$$

$$\bar{y}_1 = \mathbf{Cz} + \mathbf{Du} \quad (2.39)$$

where  $z$  and  $u$  are state variable and input vector, and can be expressed as  $\mathbf{z} = [x, \dot{x}]^\top$  and  $\mathbf{u} = [i(t), F(t)]^\top$ , respectively.  $\bar{y}_1$  is the absolute acceleration of the SDOF structural model, and the matrices  $\mathbf{A}$ ,  $\mathbf{B}$ ,  $\mathbf{C}$  and  $\mathbf{D}$  are follow-up as Eqs. (2.40)-(2.43), respectively.

$$\mathbf{A} = \begin{bmatrix} 0 & 1 \\ -k/m & -c/m \end{bmatrix} \quad (2.40)$$

$$\mathbf{B} = \begin{bmatrix} 0 & 0 \\ -1/m & 1/m \end{bmatrix} \quad (2.41)$$

$$\mathbf{C} = \begin{bmatrix} -k/m & -c/m \end{bmatrix} \quad (2.42)$$

$$\mathbf{D} = \begin{bmatrix} -1/m & 1/m \end{bmatrix} \quad (2.43)$$

Finally, the controller both considering the numerical part of the SDOF structural model and the inverse transfer function of a shaking table is constructed to implement the hybrid testing method as shown in Figure 2.22. Figure 2.23 shows the experimental configuration of the test. The hybrid testing method is fatal to the noise and phase error because the shaking table is vibrated by the numerical part which is calculated in real-time. Accordingly, the hybrid test using the Real-Time Window Target of the MATLAB Simulink with the sampling rate of 1000 Hz was implemented to minimize the calculation and phase error in the actual test. Figure 2.24 is an excitation model of the Real-Time Window Target of the MATLAB Simulink, and it includes the inverse transfer function of the shaking table and numerical analysis of the SDOF structure.

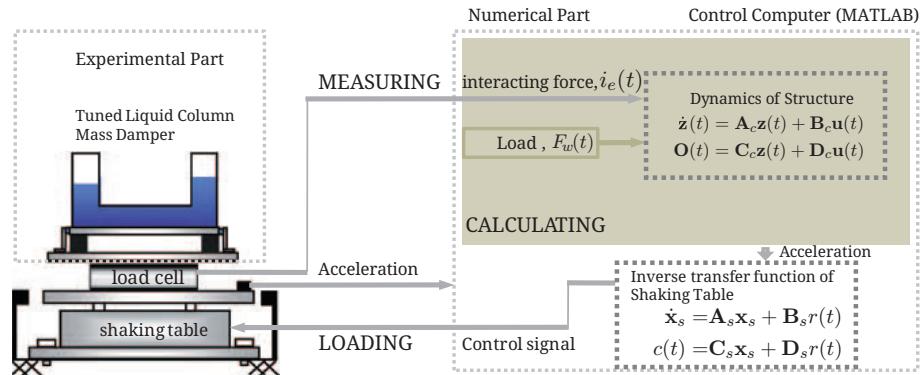


Figure 2.22: Design of the controller for the hybrid testing method

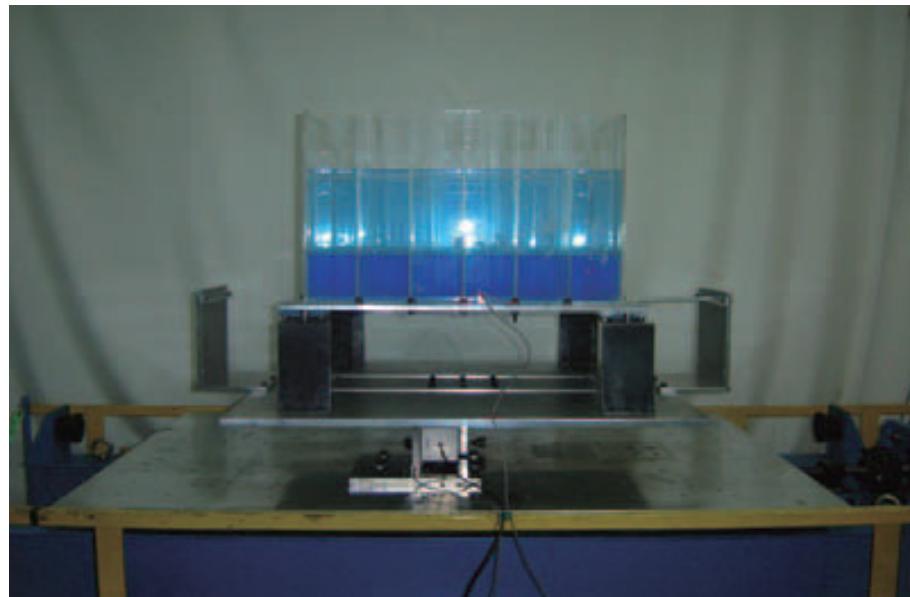


Figure 2.23: Photograph of the hybrid testing

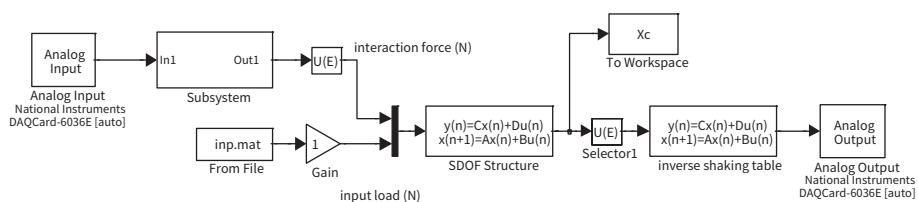


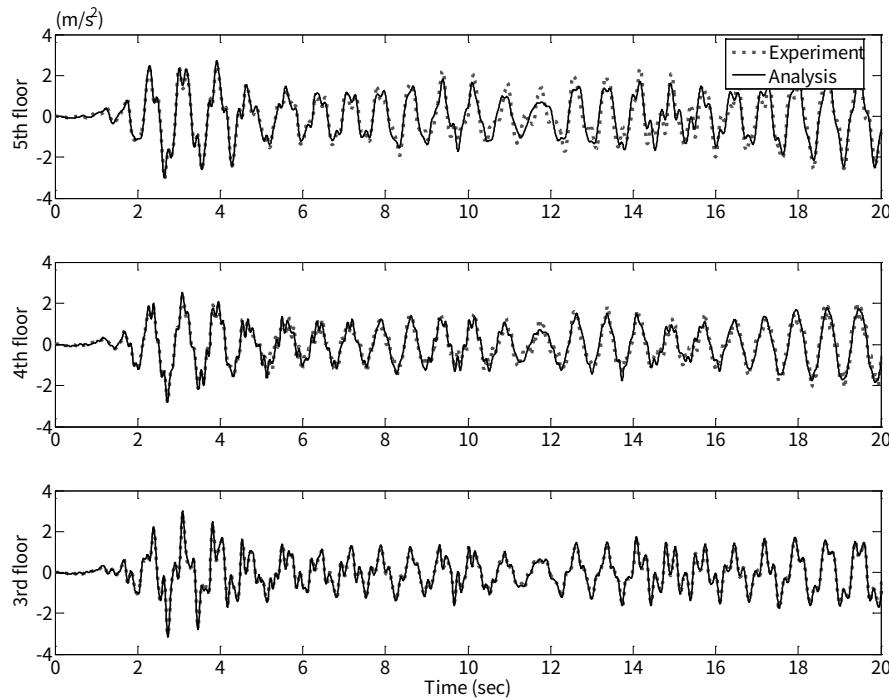
Figure 2.24: Design controller of the hybrid testing method (MATLAB Simulink Real-time Window Target)

## 2.6 Experimental Verification

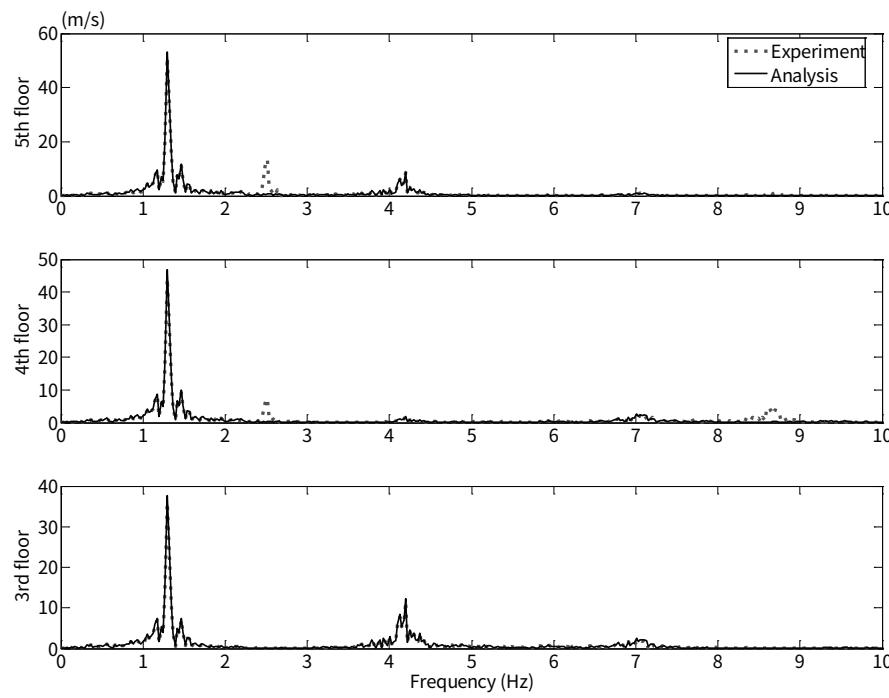
### 2.6.1 Experimental result of Substructuring technique

The validity of the proposed substructuring technique as part of the hybrid testing method is verified for the experimental setting described in the previous sections. First, the shaking table test without feedback loop of accelerations measured from the experimental substructure into the shaking table controller in Figure 2.6 is performed to confirm the validity of the proposed method experimentally. In this case, the third story acceleration of responses pre-calculated from the assumed whole building with structural parameters such as Eqs. (2.16) and (2.17) is used as a reference signal in Figure 2.15. Figure 2.25 compares the time histories and Fourier transform of the acceleration responses, which are measured from the experimental substructure and shaking table, with those calculated from the numerical analysis of the whole assumed structure. In other words, the responses corresponding to the 3rd, 4th and 5th story accelerations are compared. Also, Figures 2.26-2.28 showing the variation of frequency components according to time lapse illustrate the spectrogram and contour plots of experimental and numerical accelerations of the 3rd, 4th and 5th story, respectively. As can be confirmed from Figure 2.25, the experimental accelerations obtained from the shaking table test without their feedback agree well with those obtained from the analysis of the whole assumed structure in both time and frequency domains. However, it can be observed that the small discrepancies are shown in the time histories of 4th and 5th story accelerations as shown from Figure 2.25, while the 3rd story acceleration is identical to the numerical one over the entire time history as known from Figures 2.25-2.26. These differences are caused by the inherent modes of experimental substructure; the frequency components of 2.5 and 8.6 Hz corresponding to the first and second modes of the experimental substructure, respectively, are observed in the measured acceleration of 4th story as known from Figure 2.27, and also the component of 2.5 Hz is expressed in the 5th story experimental acceleration as like Figure 2.28. It is considered that this tendency is especially conspicuous in the case of utilizing a lightly-damped testing model as an experimental substructure. Table 1 shows the frequency components observed in the time records of an experimental substructure from the test without its acceleration feedback and the natural frequencies calculated from the whole assumed structure with five stories. From Table 2.1, it can be noted that the first natural frequency of the experimental substructure is shifted from 2.5 Hz to 1.3 Hz by the dynamics of the three-story numerical substructure added to its base. Then, the substructuring testing expressed as Figure 2.6 was carried out based on the acceleration feedback of the experimental substructure. Figure 2.29 compares the responses measured by the test with the acceleration feedback of the experimental substructure with those

calculated from the numerical analysis of the whole assumed structure with 5 DOFs. Also, Figures 2.30-2.32 express the frequency components according to time lapse, which is observed in the measured responses from the test with feedback and the calculated ones of the 3rd, fourth and fifth story, respectively. As known from Figure 2.29, the inclination in entire time history responses measured from the proposed testing method agrees well with that calculated from the whole assumed structure. These are why the first mode responses of substructured system coincide well with those of the whole supposed system over the entire time range, as shown in Figures 2.30-2.32. However, as can be confirmed from Figures 2.30-2.32, instead of the second and third mode responses of the substructured system, those of the experimental substructure are observed from the testing results in the vicinity of 2.5 and 8.6 Hz. It is considered that in the process of the acceleration feedback of the experimental substructure, its fundamental modes affect to the numerical substructure and then the numerical error occurs in calculating the numerical substructure.



(a) Time domain



(b) Frequency domain

Figure 2.25: Comparisons of results measured from the experiment without feedback and those calculated from numerical analysis.

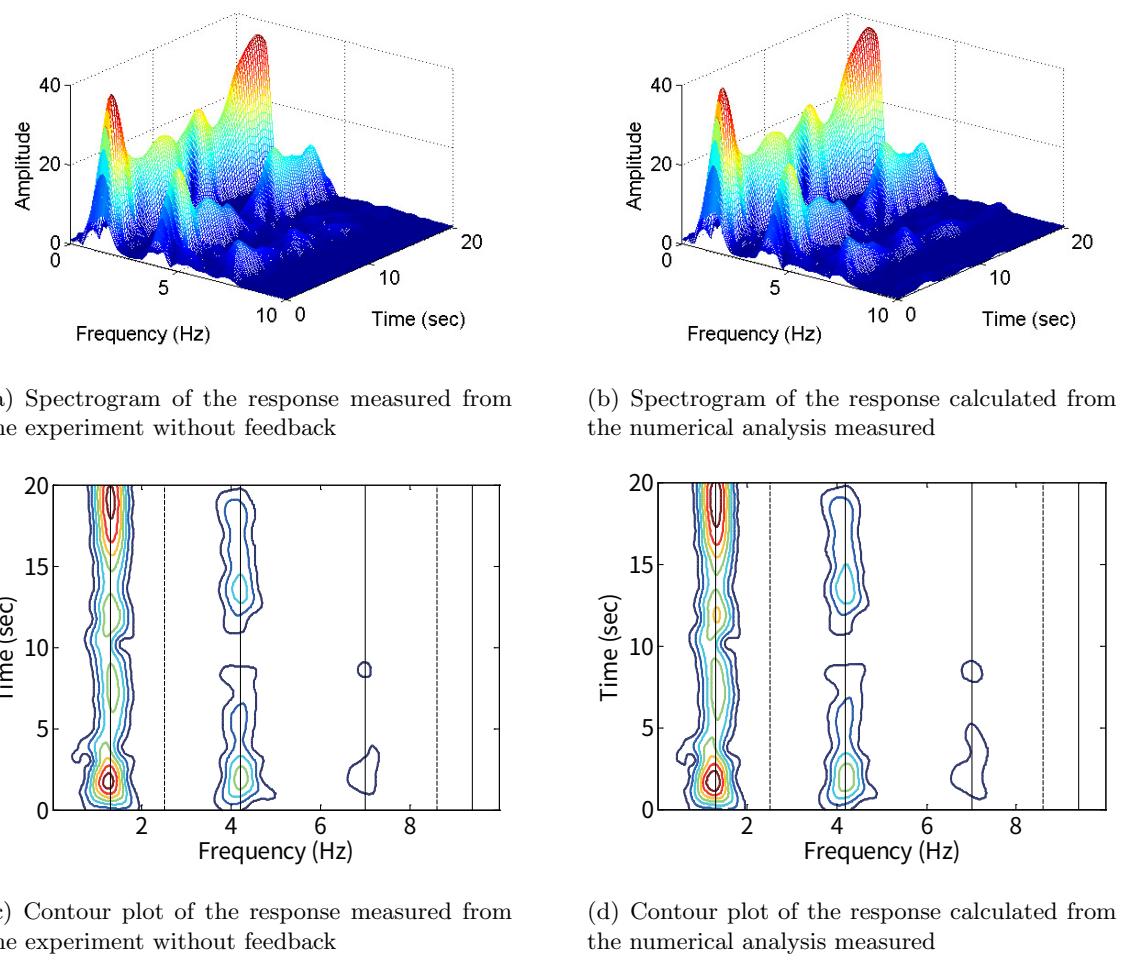


Figure 2.26: Spectrograms and contour plots of the 3rd story acceleration measured from the experiment without feedback and that calculated from the numerical analysis.

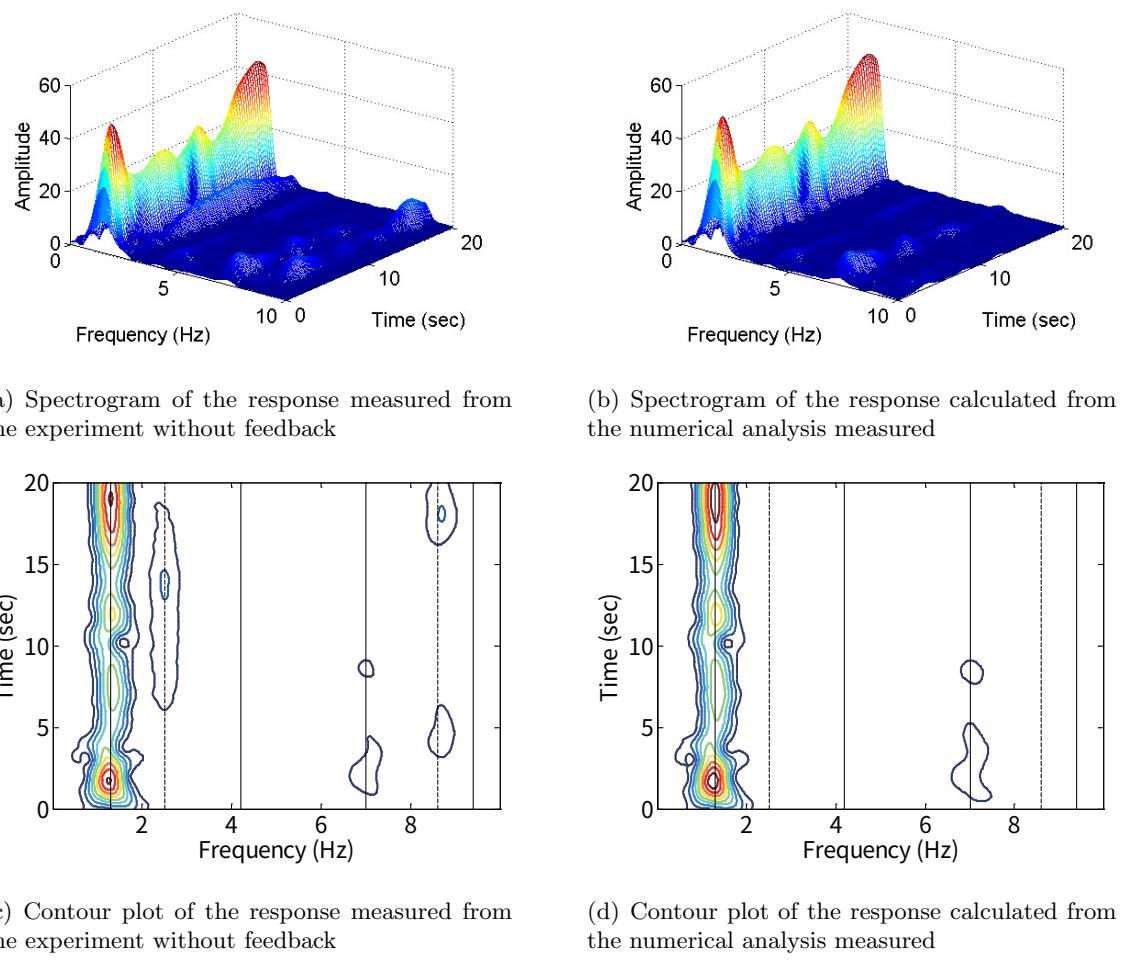


Figure 2.27: Spectrograms and contour plots of the 4th story acceleration measured from the experiment without feedback and that calculated from the numerical analysis.

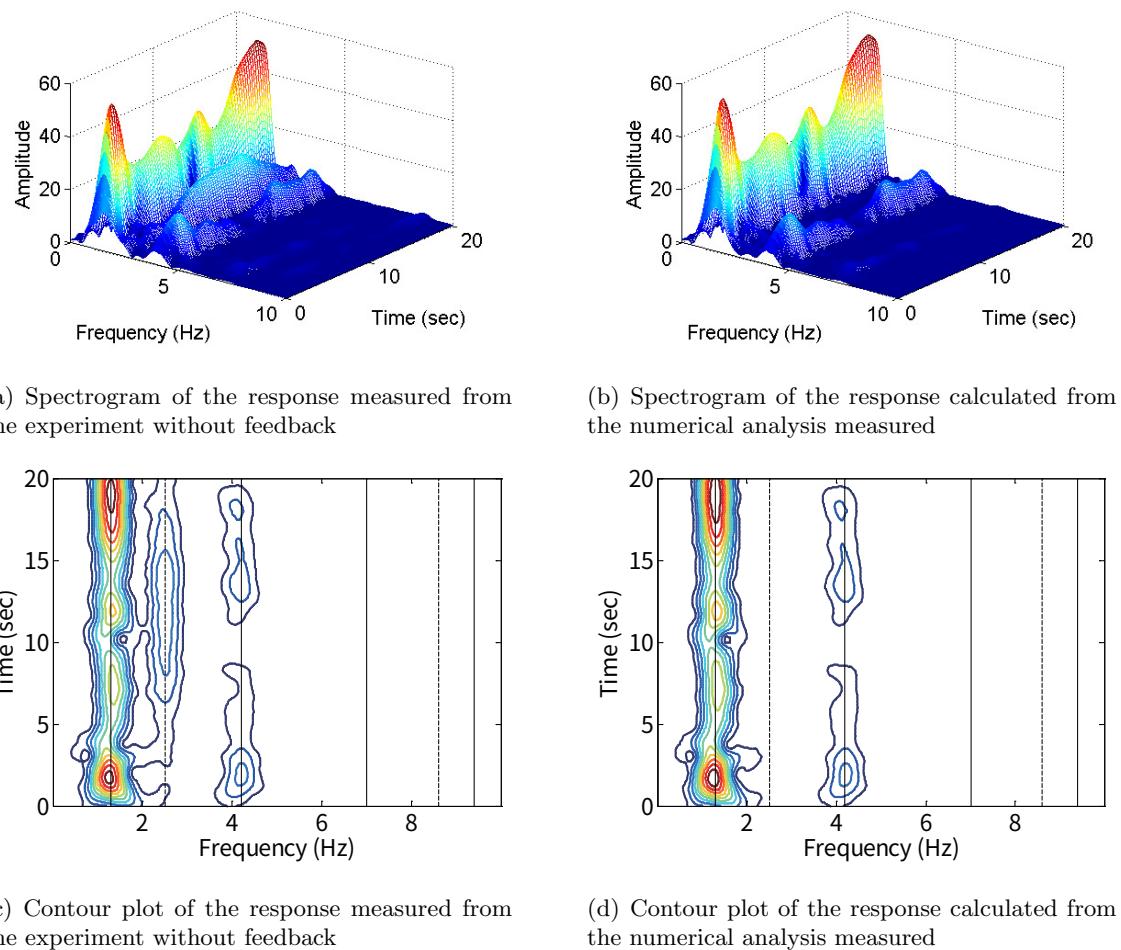
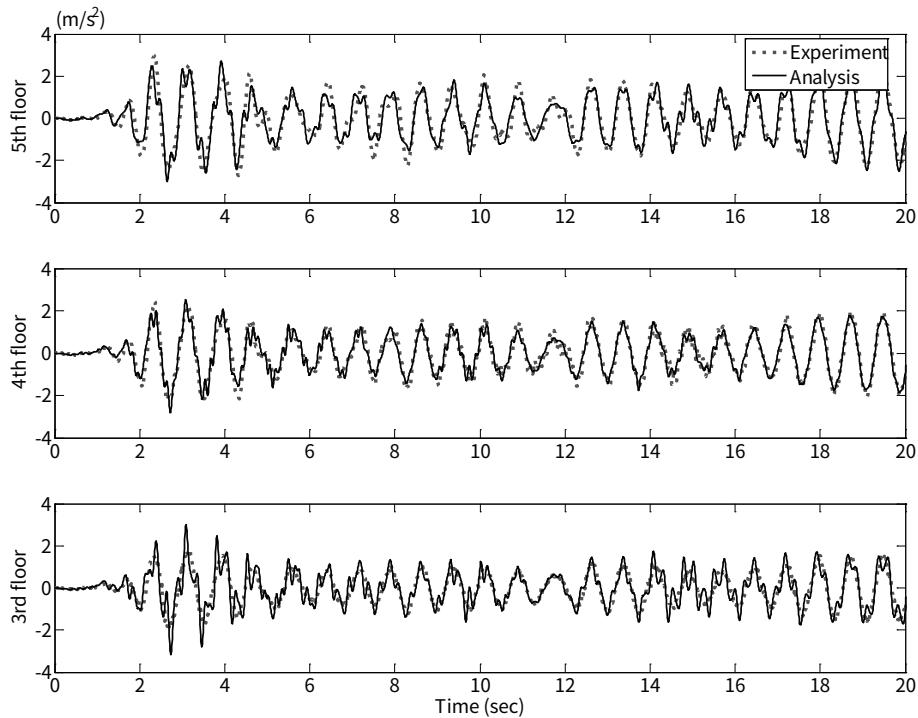
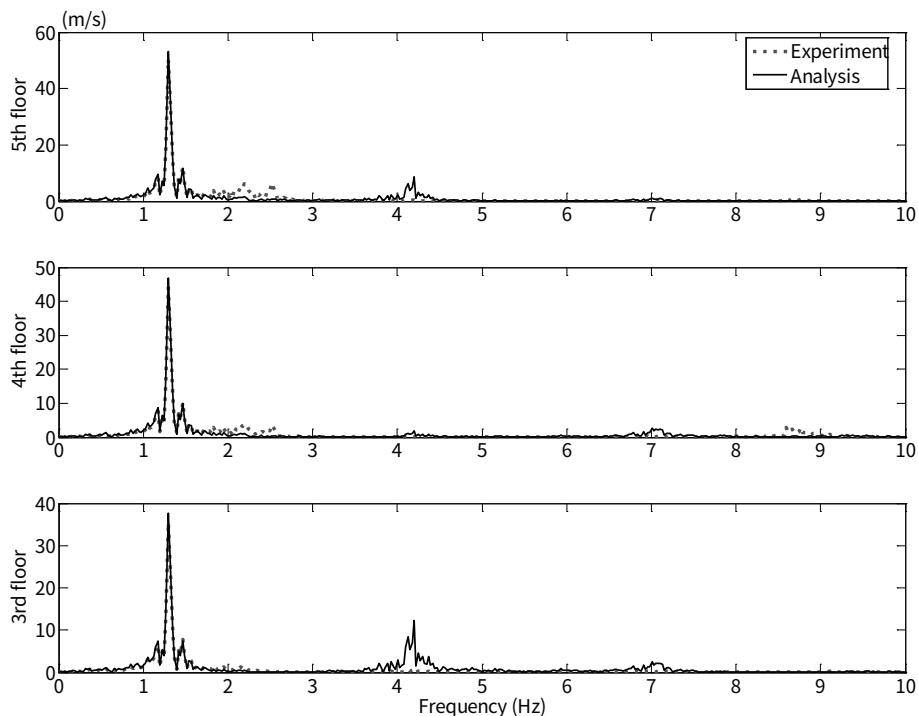


Figure 2.28: Spectrograms and contour plots of the 5th story acceleration measured from the experiment without feedback and that calculated from the numerical analysis.



(a) Time domain



(b) Frequency domain

Figure 2.29: Comparisons between results from the experiment with feedback and those from analysis.

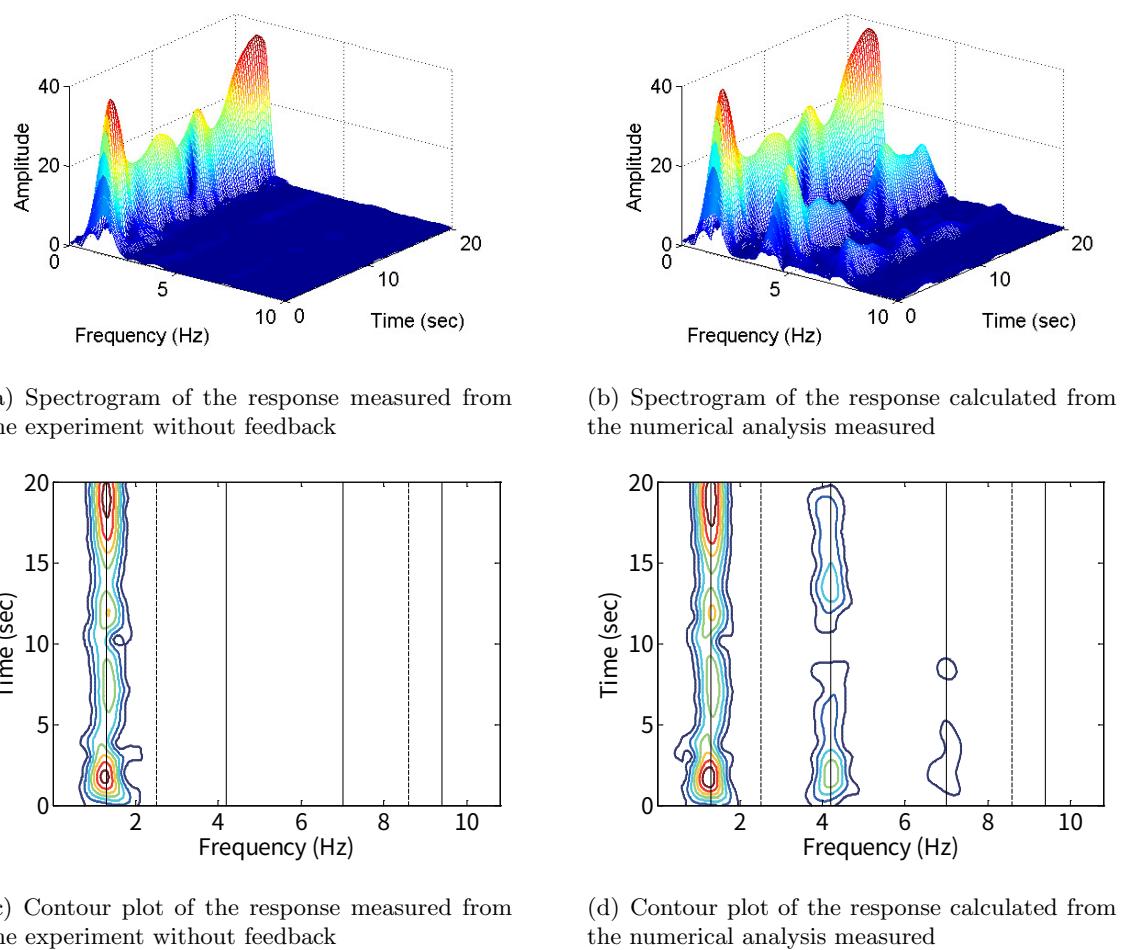


Figure 2.30: Spectrograms and contour plots of the 3rd story acceleration measured from the experiment with feedback and that calculated from the numerical analysis.

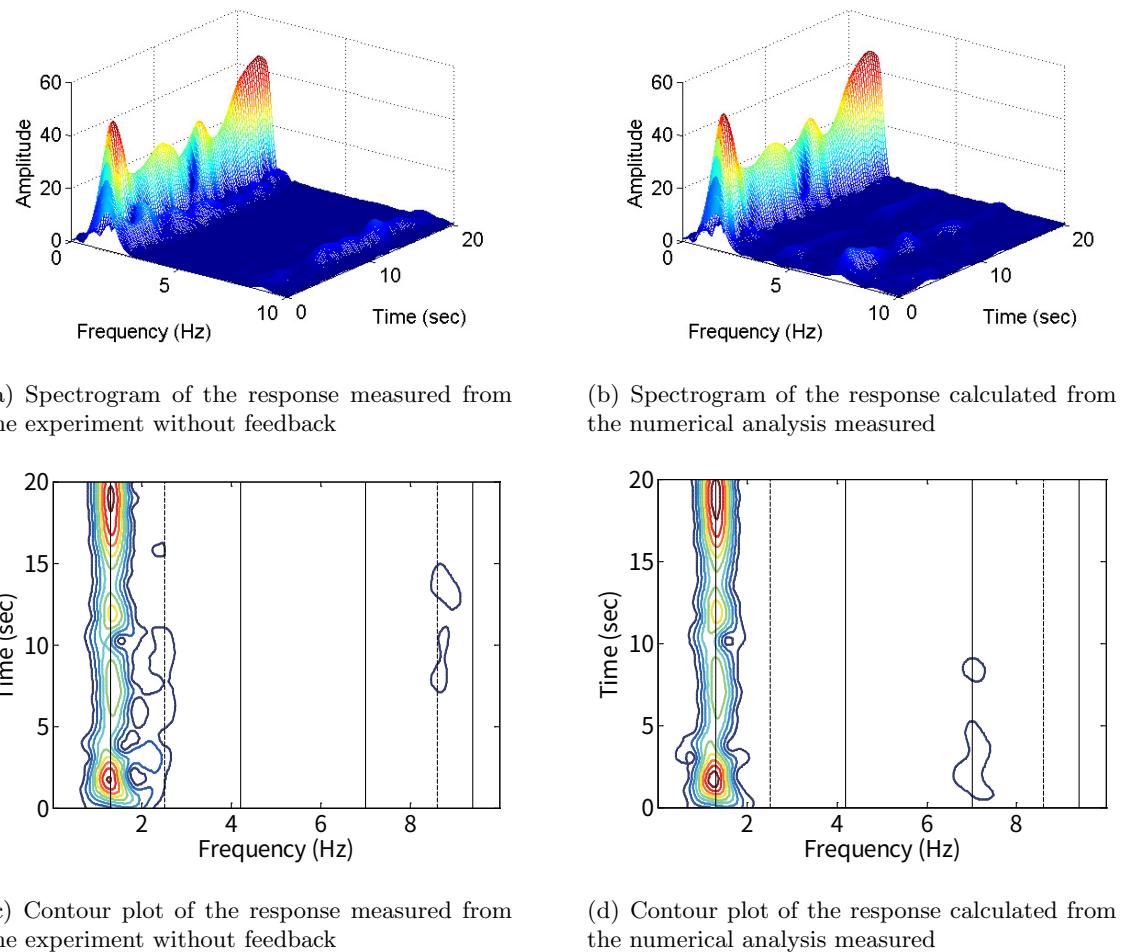


Figure 2.31: Spectrograms and contour plots of the 4th story acceleration measured from the experiment with feedback and that calculated from the numerical analysis.

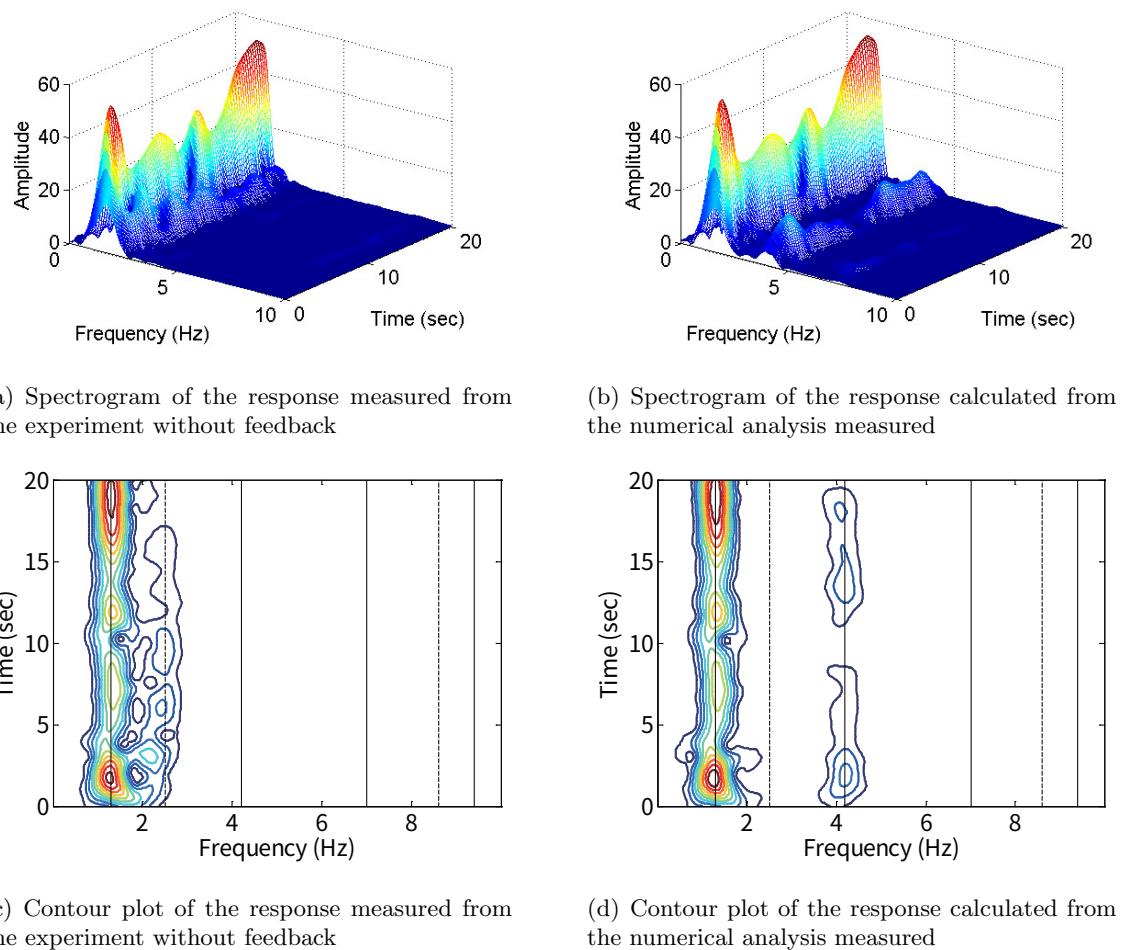


Figure 2.32: Spectrograms and contour plots of the 5th story acceleration measured from the experiment with feedback and that calculated from the numerical analysis.

### 2.6.2 Hybrid Testing Method of a Single Story Strcuture with a TLD

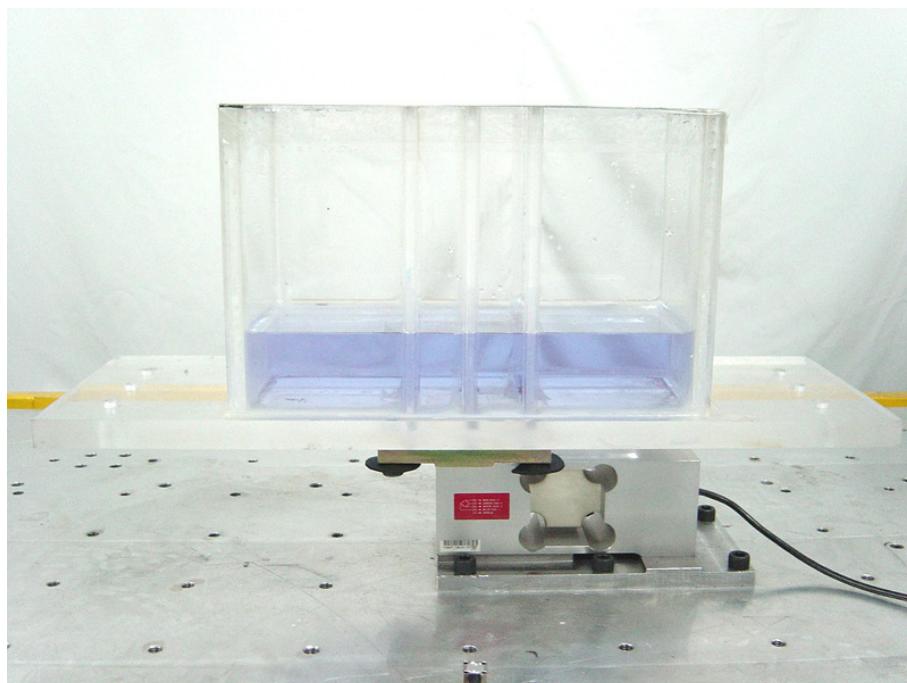
In this section, experimental verification of the hybrid testing method is conducted for a single story steel frame with a TLD. First, the conventional TLD-structure interaction model shown in Figure 2.33(a) is tested. Then, the hybrid testing method shown in Figure 2.33(b), which incorporates the single story steel frame in the numerical calculation, is performed and the results from the two testing methods are compared to each other. For the numerical structural model used in the hybrid testing method, the single story steel frame is assumed to be an SDOF mass-damping-spring system. The structure has  $0.6m$  of width,  $1.0m$  of height and  $169.7kg$  of measured floor mass. El Centro, Hachinohe, Mexico City, and Northridge earthquake waves were realized by the shaking table, and the resulting absolute accelerations of the floor and the shaking table were measured. The system identification was conducted using the measured absolute accelerations. The identified parameters slightly vary according to input earthquake waves. The averaged damping and stiffness coefficients are  $14.6N \cdot s/m$  and  $9914.3N/m$ , respectively, which correspond to  $1.23Hz$  of structural natural frequency. The TLD shown in Figure 2.33 has the size of  $31(cm) \times 14(cm) \times 20(cm)$ . The level of water in the TLD was adjusted to have  $3.4cm$  that is theoretically calculated based on the linear wave theory(Soong and Dargush, 1997) for the TLD to have fundamental sloshing frequency tuned to the identified structural natural frequency. As a result, the mass ratio of the TLD to the structure is about  $1.3\%$ . To confirm whether the numerically calculated frequency of the TLD is modulated to the structural one, the transfer function is shown in Figure 2.34, from the shaking table acceleration to the shear force by the TLD, was obtained by using the white noise excitation. It is observed in Figure 2.34 that the TLD has the sloshing frequency of  $1.25Hz$  which is very close to the structural natural frequency of  $1.23Hz$ .

At first, the conventional shaking table test is shown in Figure 2.33(a) is performed to investigate the seismic response control performance of the TLD. Previously mentioned four earthquake records are scaled to have the peak acceleration of 100 gals and used to excite the TLD-structure system. Figures 2.35 and 2.36 show the measured structural acceleration responses in the time and frequency domains, respectively. It is observed from Figure 2.35 that acceleration in the latter part of the whole response history is significantly reduced. This phenomenon is a common tendency in the structural response controlled by a tuned mass-type control device since it makes an effect when the structural response is governed by the fundamental mode after initial high impulse like component has passed. In response to Mexico-city earthquake excitation, as shown in Figure 2.36(c), the first peak corresponding to the major frequency component of the earthquake itself is not controlled, but the response in the region of the TLD modulation frequency is reduced to nearly zero.

Then, the hybrid testing method is applied with the experimental set-up shown in



(a) Conventional shaking table test



(b) Hybrid shaking table test

Figure 2.33: TLD-structure interaction experimental system.

Figure 2.33(b). For its implementation for the controlled case, the identified structural parameters are reflected in the numerical part expressed by the shaded region in the integrated controller shown in Figure 2.18. The continuous filters are converted into discrete ones with a time interval of 0.01 second. Figures 2.37 and 2.38 compare the controlled

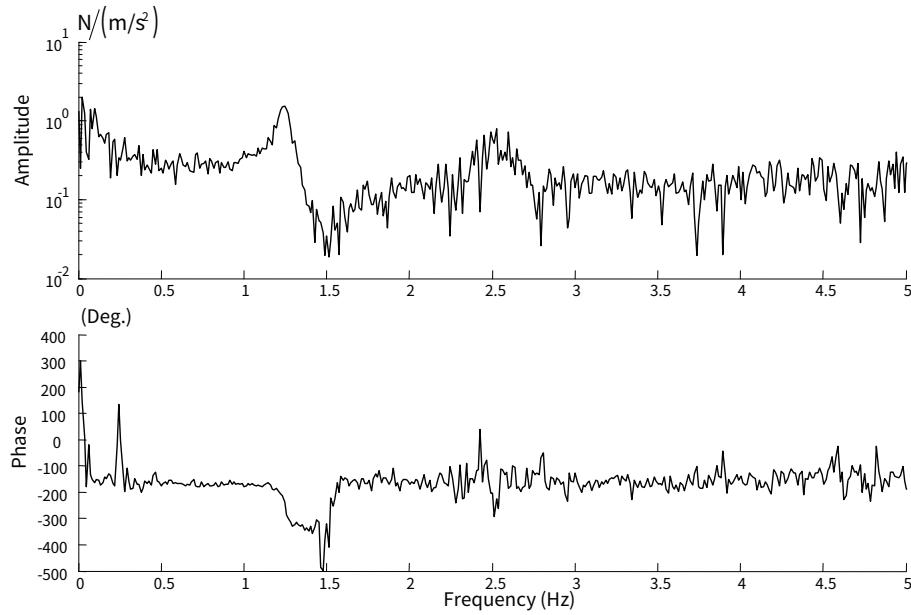


Figure 2.34: TLD transfer function from the table acceleration to the base shear force

accelerations obtained by performing the conventional and the hybrid testing method in time and frequency domains, respectively. The effectiveness of the hybrid testing method is verified by the fact that the experimental results from two methods coincide well with each other on the whole. The small discrepancies existing in the controlled responses subjected to El Centro and Hachinohe earthquakes are considered to result from the underestimation of damping coefficients in the numerical structural model since averaged parameters for the four earthquake data were used.

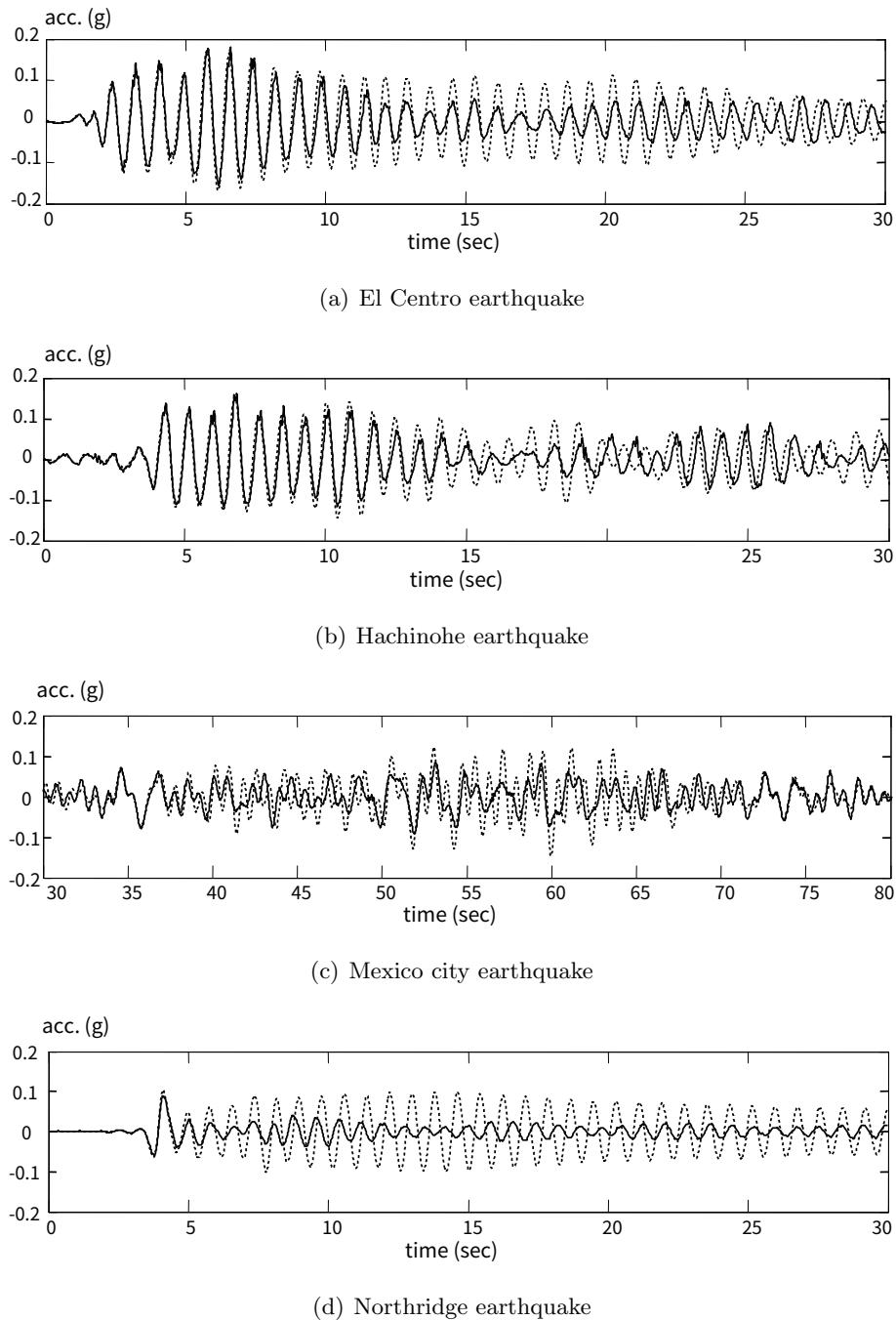


Figure 2.35: Structural acceleration in the time domain measured from the conventional shaking table test of TLD-structure interaction system (dotted line : without control, solid line : with control)

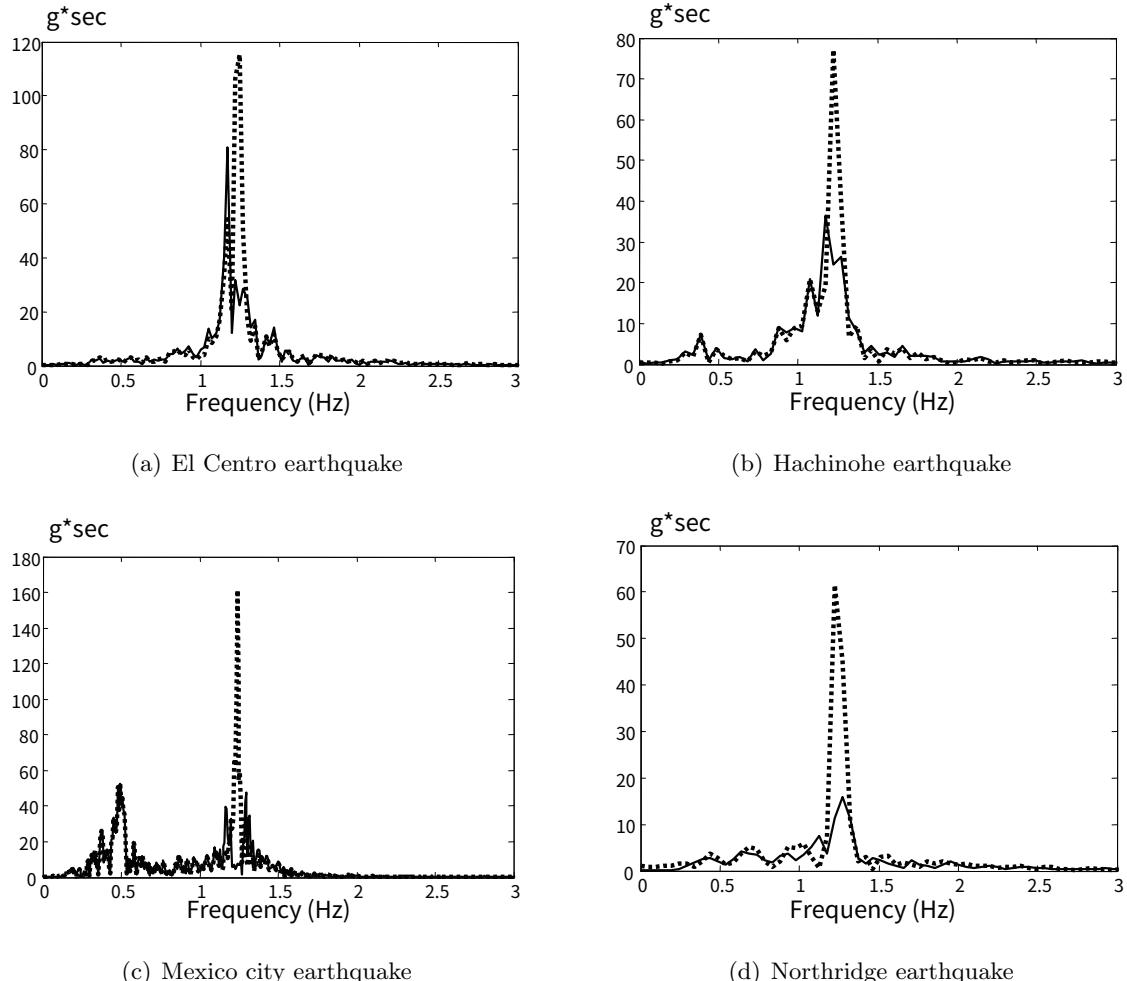


Figure 2.36: Structural acceleration in the frequency domain measured from the conventional shaking table test of TLD-structure interaction system (dotted line : without control, solid line : with control)

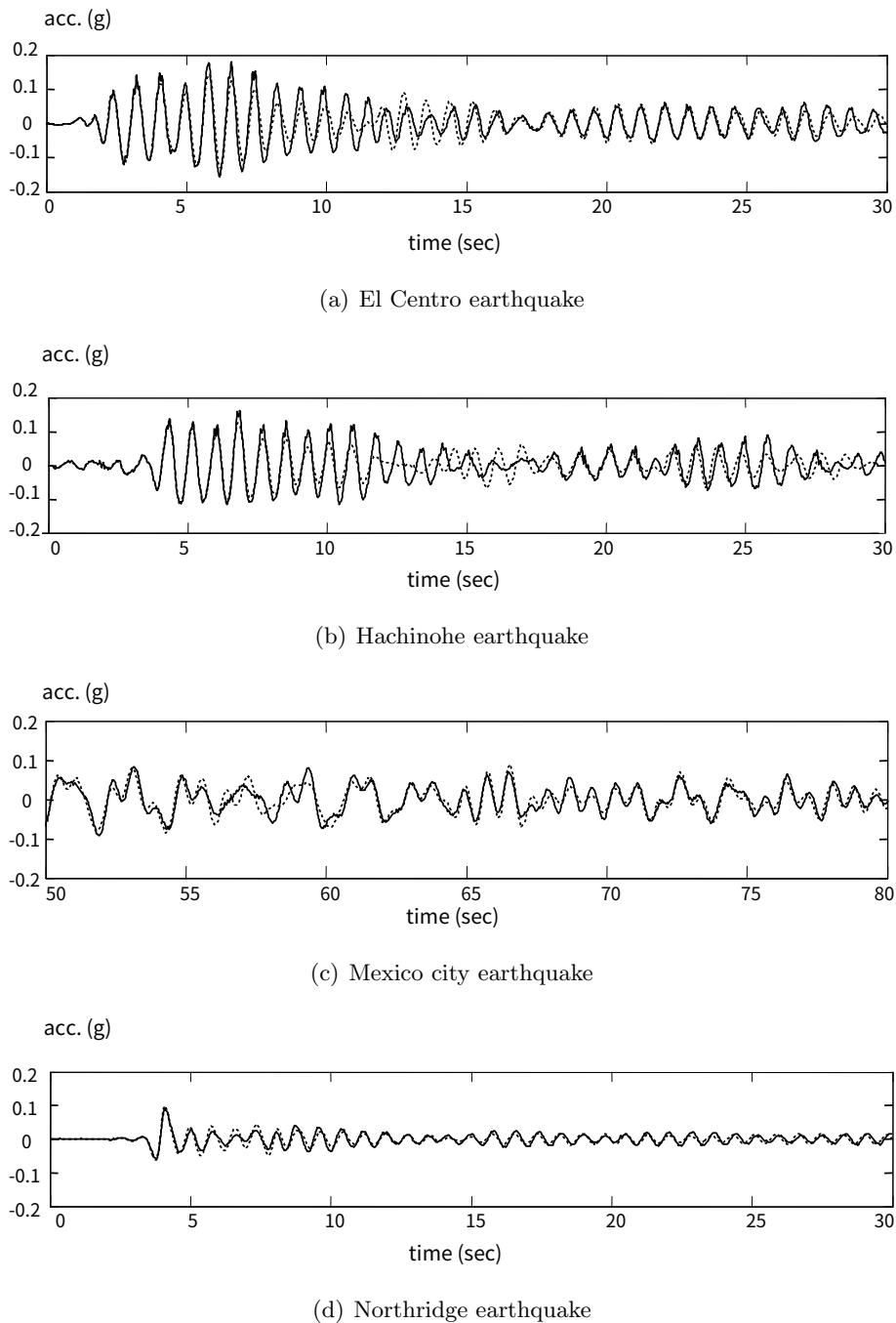


Figure 2.37: Comparisons of controlled structural accelerations in the time domain (dotted line : conventional shaking table test, solid line : hybrid shaking table test)

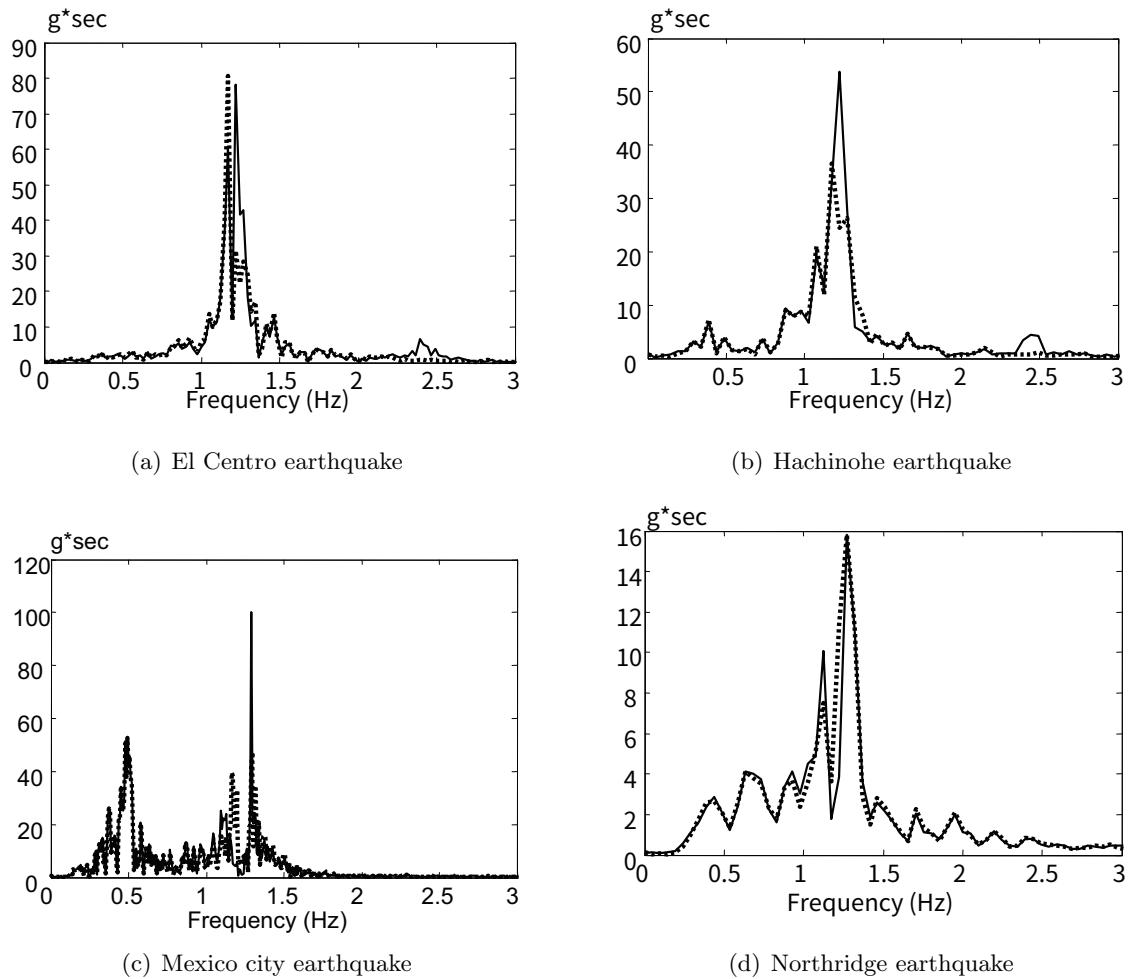


Figure 2.38: Comparisons of controlled structural accelerations in the frequency domain (dotted line : conventional shaking table test, solid line : hybrid shaking table test)

### 2.6.3 Hybrid Testing Method of Three Story Structure with a TLD

The control performance of a TLD installed in a three-story structure is investigated by using the hybrid testing method. The structure is assumed to be a three story shear-type model, which has identical story properties as follows;  $m_i = 128.8kg$ ,  $c_i = 13.52N \cdot s/m$ ,  $k_i = 33908N/m$  for  $i = 1, 2, 3$ . The structure has natural frequencies of  $1.15Hz$ ,  $3.22Hz$  and  $4.65Hz$ . The TLD discussed in the previous section is used, and its water level is modulated to  $4.6cm$  for the TLD to have sloshing frequency of  $1.15Hz$ . As a result, the mass ratio of the TLD to the structure is about 2%. The four earthquake waves used for the excitation of the single story steel frame were scaled to have peak acceleration of  $40gal$ . The uncontrolled structural responses were obtained by removing the feedback loop of the TLD-generated interacting force, which causes the numerical structural model to be excited only by the base earthquake motion. Figures 2.39 and 2.40 compare the uncontrolled and controlled accelerations of the third story in time and frequency domains, respectively, which is realized by the shaking table through the hybrid testing method. It is observed that the structural accelerations are significantly reduced by the TLD, especially in the region of the fundamental frequency. Table 2.5 indicates that the acceleration is reduced by 4 – 30% in peak and by 18 – 60% in RMS responses. It is also identified in Figure 2.40(d) that the TLD lessens the additional second mode response of the structure. Figure 2.41 shows the typical sloshing and slamming behaviors of the water in the TLD tanks during the experiment, which occur in the small and large amplitude of the water motion, respectively(Yalla, 2001).

Table 2.5: Uncontrolled and controlled responses of a combined TLD–MDOF structure system

Responses( $g$ )	El Centro	Hachinohe	Mexico City	Northridge
<i>Peak acceleration</i>				
Uncontrolled	3.85	2.71	2.63	1.34
Controlled	2.69	2.19	252	1.34
<i>RMS acceleration</i>				
Uncontrolled	1.91	1.36	0.54	0.33
Controlled	0.74	0.66	0.45	0.27

### 2.6.4 Hybrid Testing Method of a Single Story Structure with a TLCD

At first, the conventional shaking table test with this TLCD shown in Figure 2.20(a) is performed to reduce the structural response. Two earthquake records with the maximum acceleration of  $100gal$  due to the shaking table performance are used to excite the TLCD-structure system with control case. Then, the hybrid shaking table test is conducted with

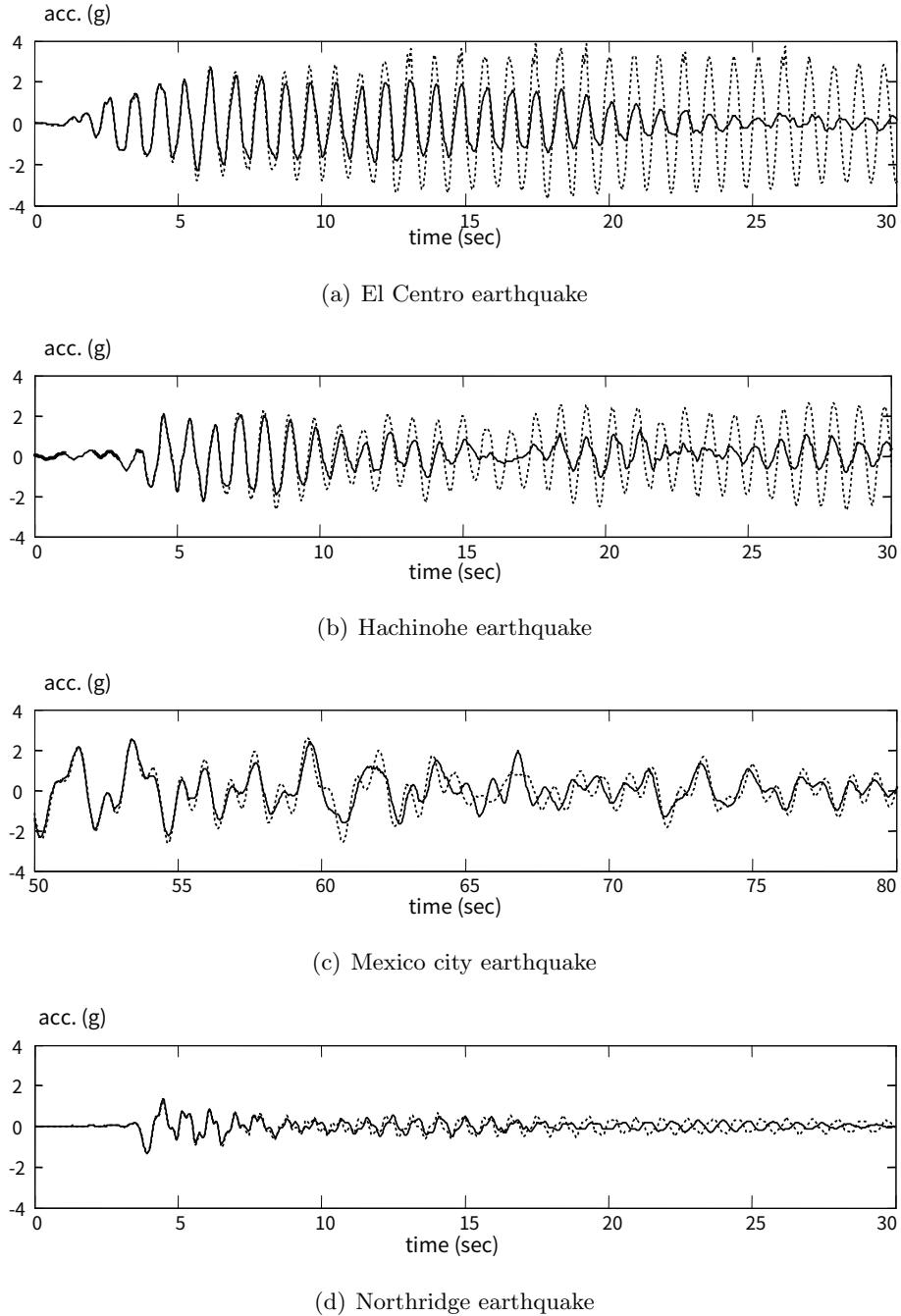
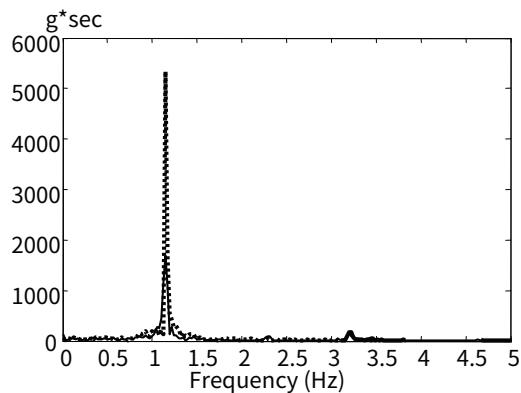
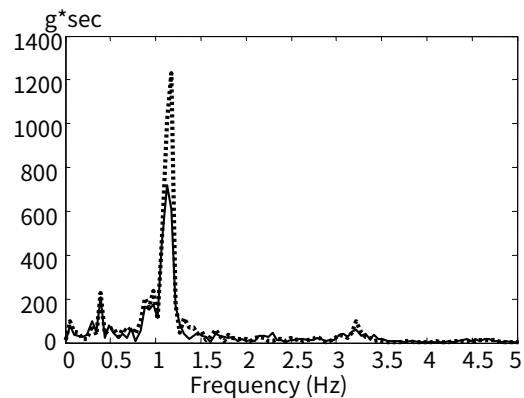


Figure 2.39: Absolute accelerations in the time domain, measured from the top story of MDOF structure with a TLD by the hybrid testing method (dotted line : without control, solid line : with control)

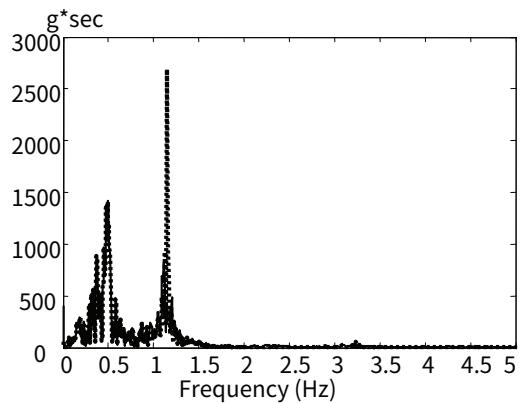
the experimental set-up shown in Figure 2.20(b). For its experimental implementation, the identified structural parameters are reflected in the numerical part expressed by the shaded region in the integrated controller shown in Figure 2.19. The continuous filters in the figure are converted into discrete ones with a time step of 0.01 sec in the actual implementation of the experiment. Figure 2.42 compare the controlled accelerations exper-



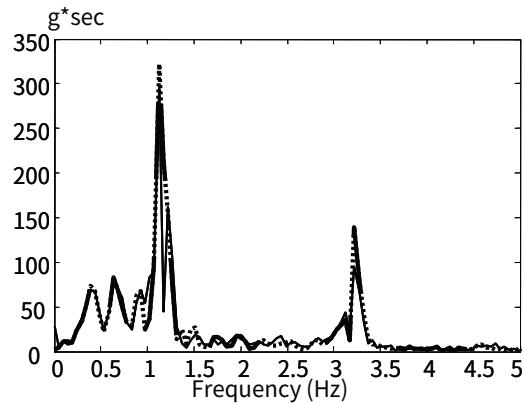
(a) El Centro earthquake



(b) Hachinohe earthquake

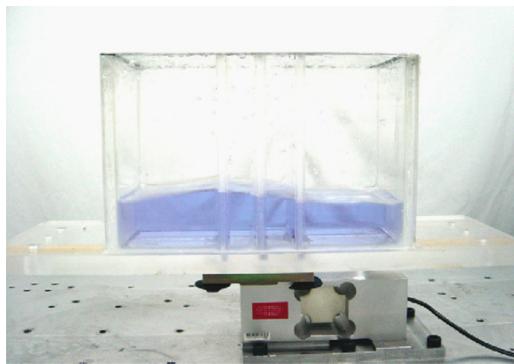


(c) Mexico city earthquake



(d) Northridge earthquake

Figure 2.40: Absolute accelerations in the frequency domain, measured from the top story of MDOF structure with a TLD by the hybrid testing method (dotted line : without control, solid line : with control)



(a) Sloshing of TLD



(b) Slamming of TLD

Figure 2.41: Behaviors of a TLD under the earthquake motion

mentally measured by implementing the conventional and the hybrid testing method in both time and frequency domain, respectively. The validity of the hybrid testing method performed in this thesis is verified from the fact that the experimental results from two

methods well coincide with each other on the whole.

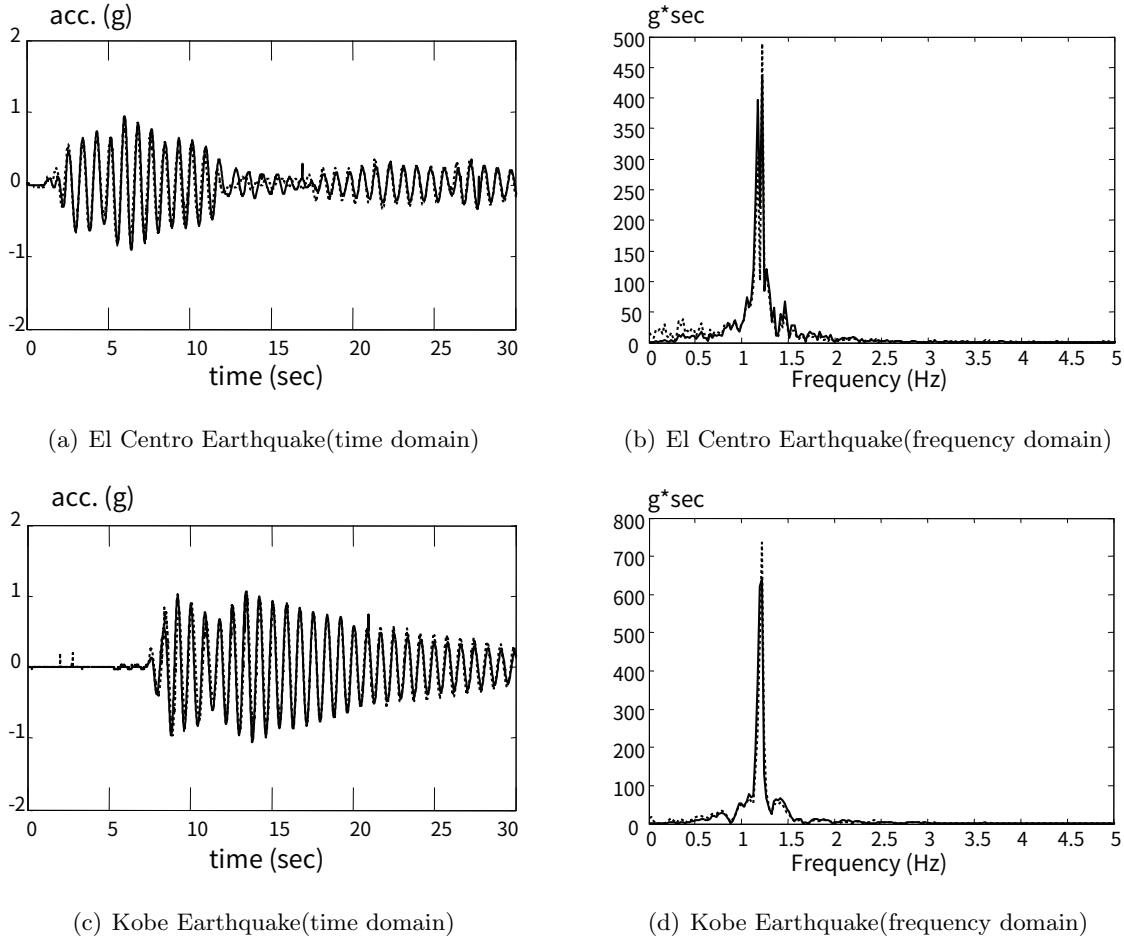


Figure 2.42: Comparisons between the results from the conventional testing method(dotted line) and those from the hybrid testing method(solid line) for the controlled response

## 2.6.5 Conventional Experiment of a Building Controlled by TLMD

### Experimental installation

The test on a scaled-down TLMD building model shown in Figure 2.44 was performed to experimentally verify the control performance of a proposed TLMD.

In order to experimentally describe the dynamic behavior of a building model, the moving mass of  $4250\text{kg}$  was mounted on guide rails. Also, the springs that connect the moving mass to both the actuator and the retaining block were devised to characterize the behavior of the stiffness of a building model, as shown in Figure 2.45. In the figure,  $m_s$ ,  $c_s$  and  $k_1 + k_2 = k_s$  represent the mass, damping coefficient and stiffness of a building model, respectively.  $x_1$ ,  $x_2$  and  $x_3$  denote the displacement and acceleration of a building model, dynamic actuator, and TLMD, respectively. Ten springs with the stiffness of  $11,000\text{N/m}$

for each one were used for the case of a test in the TMD control direction, and eight springs with  $8800N/m$  for each one for the case of a test in the TLCD control direction.

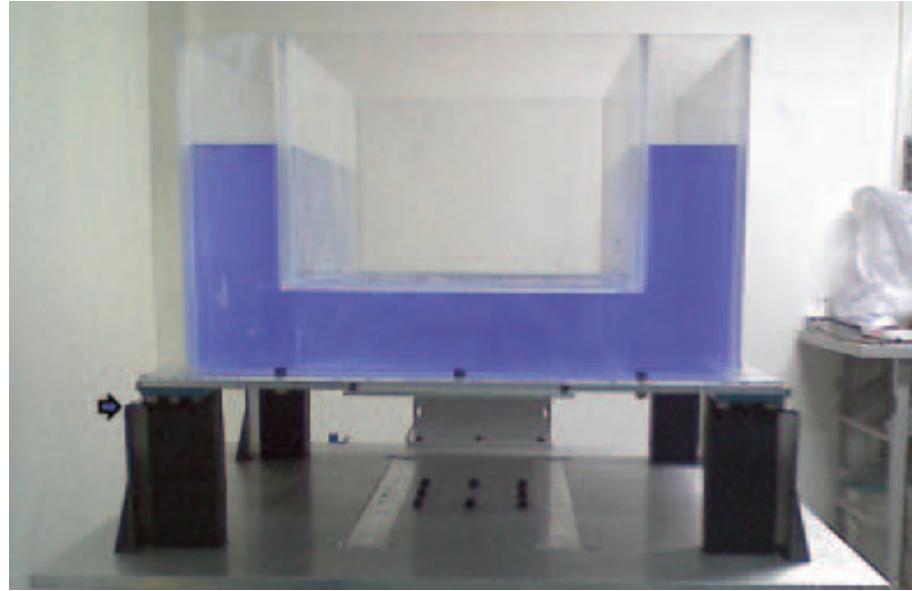


Figure 2.43: Photograph of the manufactured TLMD

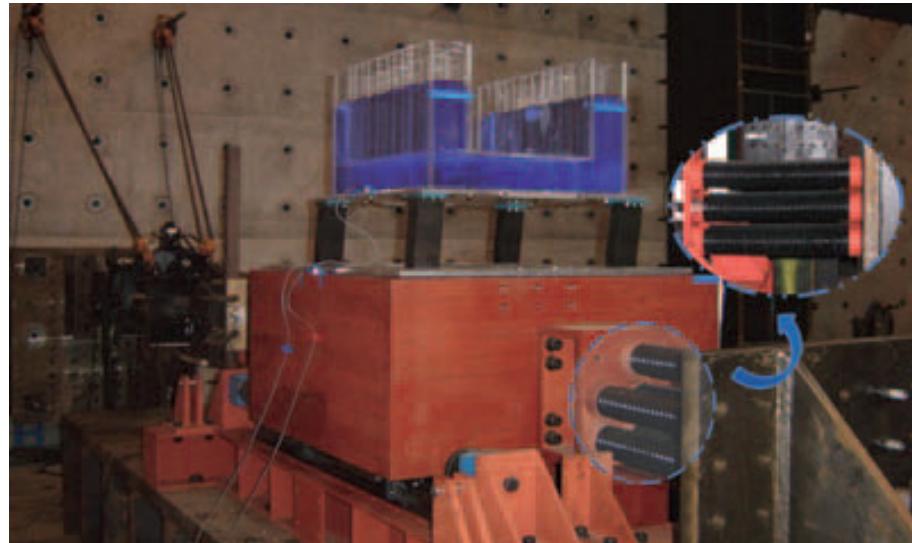


Figure 2.44: Experimental TLMD-building model

It is noted from Figure 2.46 that the mass of a building model is excited by force transmitted by the spring,  $k_2$ , through the displacement of an actuator,  $x_2$ . Accordingly, the motion of a building model is expressed by

$$m_s \ddot{x}_1 + c_s \dot{x}_1 + k_1 x_1 - k_2 (x_2 - x_1) = 0 \quad (2.44)$$

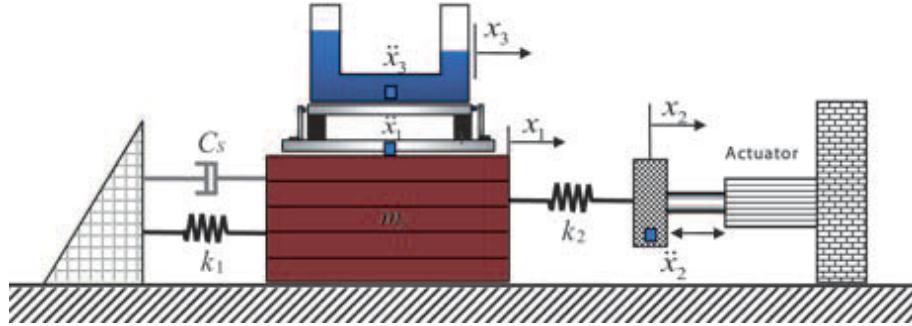


Figure 2.45: Conceptual view of experimental set-up

In this case, the displacement of an actuator,  $x_2$ , is given by

$$x_2 = A \sin(2\pi f_e t) \quad (2.45)$$

where  $A$  and  $f_e$  are the excitation amplitude and frequency of an actuator, respectively.

Finally, the equation of motion of a building model is obtained by substituting Eq. (2.45) into Eq. (2.44).

$$m_s \ddot{x}_1 + c_s \dot{x}_1 + k_s x_1 = k_2 A \sin(2\pi f_e t) \quad (2.46)$$

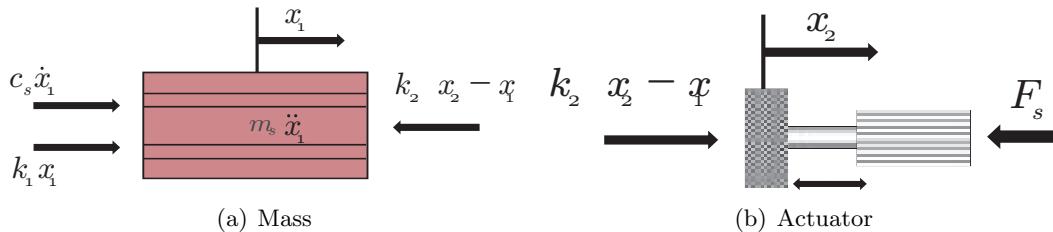


Figure 2.46: Free-body diagram of a building model.

### 2.6.6 Experimental results

In this test, the building model was excited by a dynamic actuator installed on the strong wall. Maximum excitation displacement of the dynamic actuator was set to be 4 mm. Harmonic waves with the frequency interval of 0.05 Hz from 0.1 to 3.0 Hz were imposed on the moving mass of a building model by the actuator. Especially, harmonic waves with the frequency interval of 0.01 Hz were excited to the moving mass in the vicinity of its natural frequencies during 200s. Then, the steady-state response of the moving mass was obtained in each excitation frequency.

First, the test was performed in the TMD control direction. In this case, the frequency of structural model was set to 0.82 Hz by connecting springs with the stiffness of 110,000

N/m to the structural model with the mass of 4,250 kg. Figures 2.47 and 2.48 show the displacement and acceleration response of the SDOF structure in the frequency domain, respectively. It is verified that the displacement response of the SDOF structure was reduced by 82% for the case in the TMD control direction. The displacement control performance index of a TMD is 0.18, as shown in Figure 2.47. Also, the acceleration response of the SDOF structure was reduced by 80%, and the acceleration control performance index of a TMD is 0.20, as shown in Figure 2.48. Figures 2.47 and 2.50 show the displacement and acceleration response of the SDOF structure in the time domain, respectively. Also, the response of the SDOF structure tuned by the TMD is considerably reduced at a resonance frequency of 0.82 Hz, as shown in Figure 2.49.

Then, the test was carried out in the TLCD control direction. In this case, the frequency of structural model was tuned to 0.73 Hz by connecting springs with the stiffness of 88,000 N/m to the moving mass of 4,250 kg. Figures 2.50 and 2.51 show the displacement and acceleration response of the SDOF structure in the frequency domain, respectively. It is observed that the displacement response of the SDOF structure was reduced by 71% for the case in the TLCD control direction. The displacement control performance index of a TLCD is 0.29, as shown in Figure 2.50. Also, the acceleration response of the SDOF structure was reduced by 70%, and the acceleration control performance index of a TLCD is 0.30, as shown in Figure 2.51. Also, the displacement response of the SDOF structure tuned by TLCD is considerably reduced at a resonance frequency of 0.73 Hz, respectively, as shown in Figure 2.52.

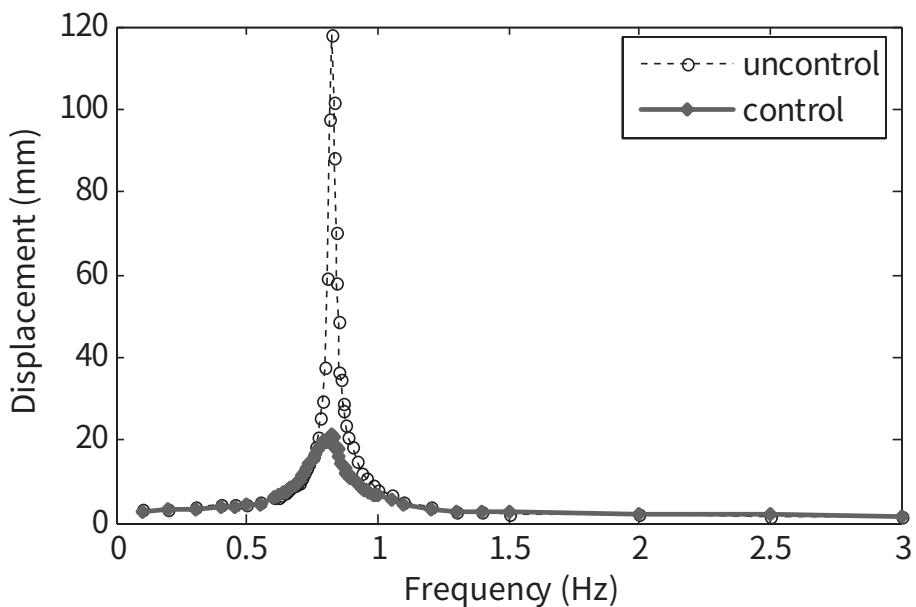


Figure 2.47: Displacement in the frequency domain (TMD direction)

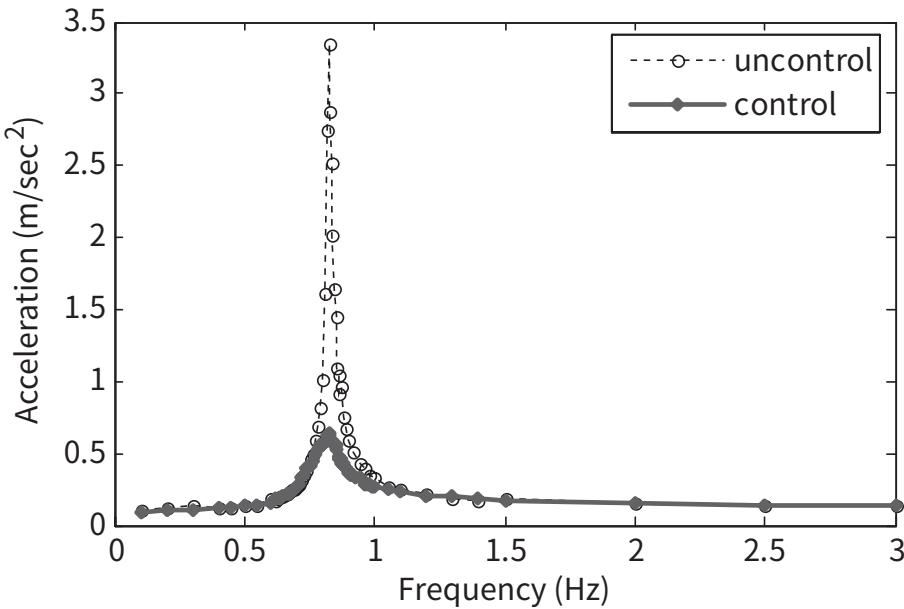


Figure 2.48: Acceleration in the frequency domain (TMD direction)

### 2.6.7 Hybrid Testing Method of a Single Story Building Controlled by TLMD

A TLMD is excited by uniaxial shaking table. Shear type load cell and acceleration sensors are attached on the shaking table to monitor the dynamic characteristic of the shaking table. The vibration control and data acquisition are conducted using a real-time digital signal processor. The main task of the data acquisition board is data conversion; it converts the measured shear force and acceleration to the digital data and converts the reference signal computed by the control program MATLAB to the analog data. An eight-channel data acquisition system was adopted which uses an NI DAQcard-6036E board and a BNC-2110 BNC cable connector. At the hybrid testing method, the control performance results of the TLMD were evaluated. In this test, the checking point is whether the natural frequencies of the TLMD in the TMD and TLCD control direction were seen at 0.82 and 0.73 Hz, respectively. Then, control performance of the bidirectional TLMD will be evaluated.

First, the test was performed in the TMD control direction. Figures 2.53 and 2.54 show the displacement and acceleration response in the frequency domain, respectively. It is observed that the displacement response of the shaking table was reduced by 78% for the case in the TMD control direction. The displacement control performance index of a TMD is 0.22, as shown in Figure 2.53. Also, the acceleration response of the shaking table was reduced by 78%, as shown in Figure 2.54. Figure 2.55 shows the displacement in the time domain. The response of the shaking table is considerably reduced by 78% at

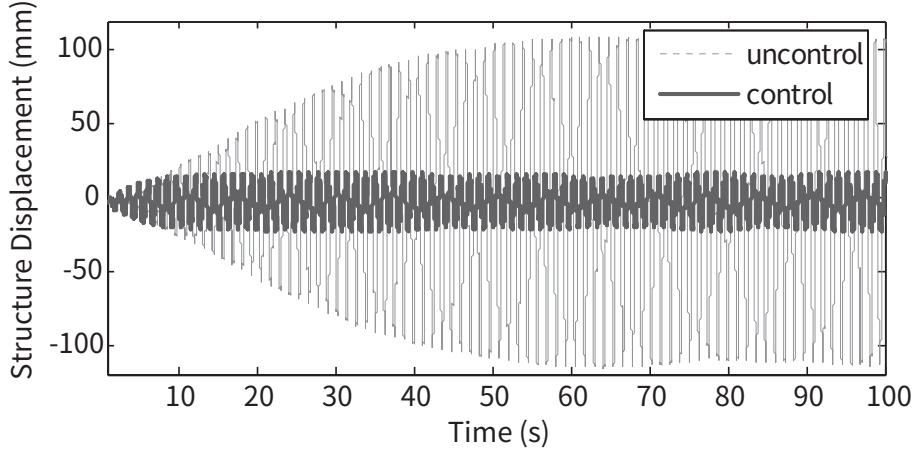


Figure 2.49: Displacement in the time domain (TMD direction, 0.82 Hz)

a resonance frequency of 0.82 Hz.

Then, the test was performed in the TLCD control direction. Figures 2.56 and 2.57 show the displacement and acceleration response of the shaking table in the frequency domain, respectively. It is observed that the displacement response was reduced by 71% for the case in the TLCD control direction, and the acceleration response reduced by 70%. Also, the displacement and acceleration response of the shaking table are considerably reduced at a resonance frequency of 0.73 Hz, respectively, as shown in Figure 2.58.

### Comparisons of conventional experiment and hybrid testing method results

In order to compare the control performances of the TLMD at the hybrid testing method and conventional method,  $J_1$ ,  $J_2$ ,  $J_3$ ,  $J_4$ , are represented by

$$J_1 = \frac{\max[x_c(t)]}{\max[x_u(t)]} \quad (2.47)$$

$$J_2 = \frac{\max[\ddot{x}_c(t)]}{\max[\ddot{x}_u(t)]} \quad (2.48)$$

$$J_3 = \frac{\text{rms}[x_c(t)]}{\text{rms}[x_u(t)]} \quad (2.49)$$

$$J_4 = \frac{\text{rms}[\ddot{x}_c(t)]}{\text{rms}[\ddot{x}_u(t)]} \quad (2.50)$$

where  $x_u$  and  $x_c$  are uncontrolled displacement and controlled displacement, respectively.  $\ddot{x}_u$  and  $\ddot{x}_c$  are uncontrolled acceleration and controlled acceleration, respectively.

Tables 2.6 and 2.7 show the control performance of the TLMD through the conventional test and hybrid test, respectively. According to the test result in the TLCD control direction, the errors of  $J_1$  or  $J_2$  between the results of the conventional test and hybrid

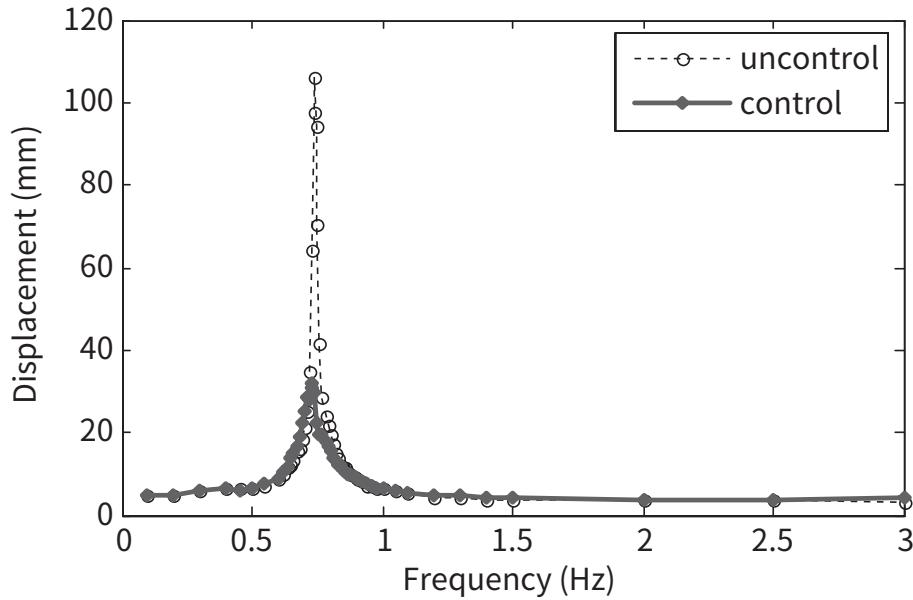


Figure 2.50: Displacement in the frequency domain (TLCD direction)

test were all 0.01, but the errors of  $J_3$  or  $J_4$  between the conventional test and hybrid test were 0.03 and 0.04, respectively. In the TMD control direction, the errors of  $J_1$ ,  $J_2$ ,  $J_3$ ,  $J_4$  between the conventional test and hybrid test were 0.04, 0.02, 0.06, 0.06, respectively.

Table 2.6: Control performance of real-structure test

		TMD	TLCD
Effective mass		1.8%	0.8%
Peak value	$J_1$	0.18	0.29
	$J_2$	0.20	0.30
RMS value	$J_3$	0.17	0.28
	$J_4$	0.17	0.27

Table 2.7: Control performance of hybrid testing method

		TMD	TLCD
Effective mass		1.8%	0.8%
Peak value	$J_1$	0.22	0.30
	$J_2$	0.22	0.29
RMS value	$J_3$	0.23	0.31
	$J_4$	0.23	0.31

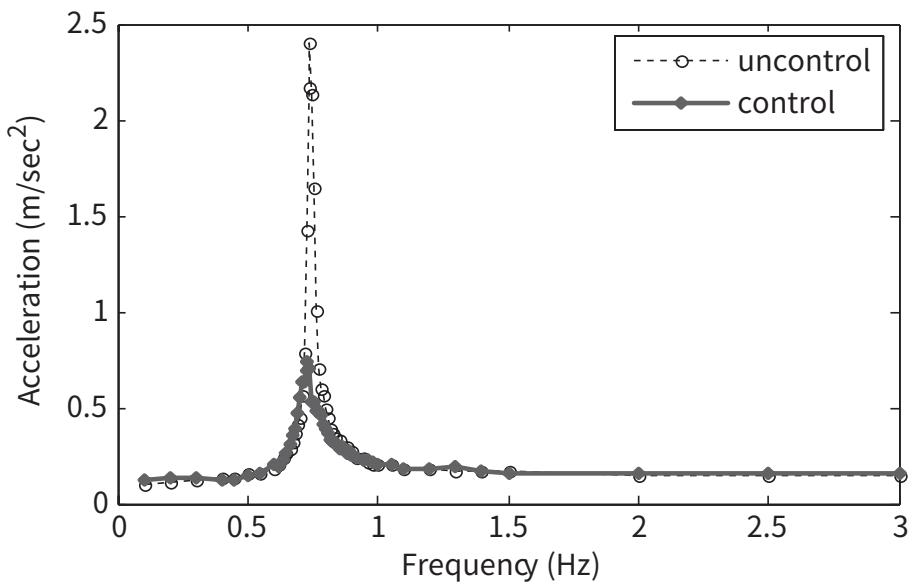


Figure 2.51: Acceleration in the frequency domain (TLCD direction)

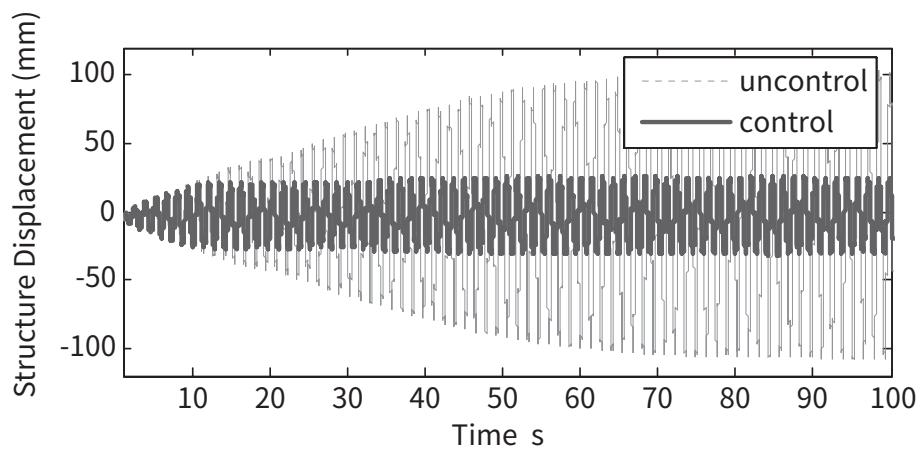


Figure 2.52: Displacement in the time domain (TMD direction, 0 · 73 Hz)

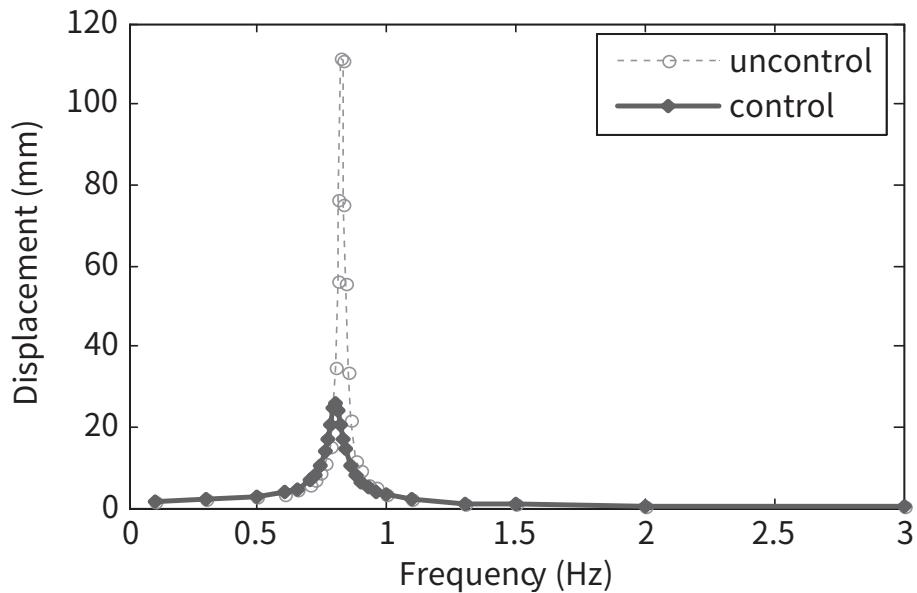


Figure 2.53: Displacement in the frequency domain by the hybrid test (TMD direction)

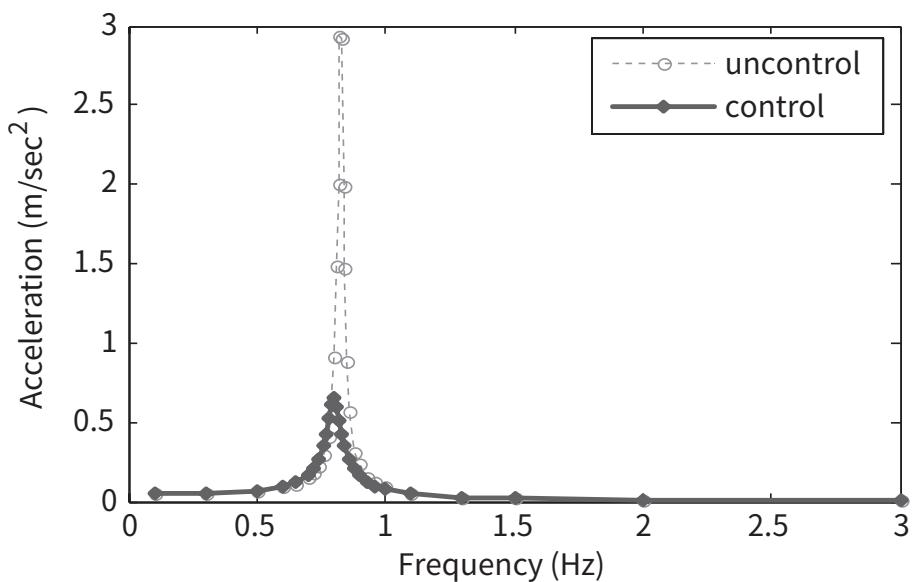


Figure 2.54: Acceleration in the frequency domain by the hybrid test (TMD direction)

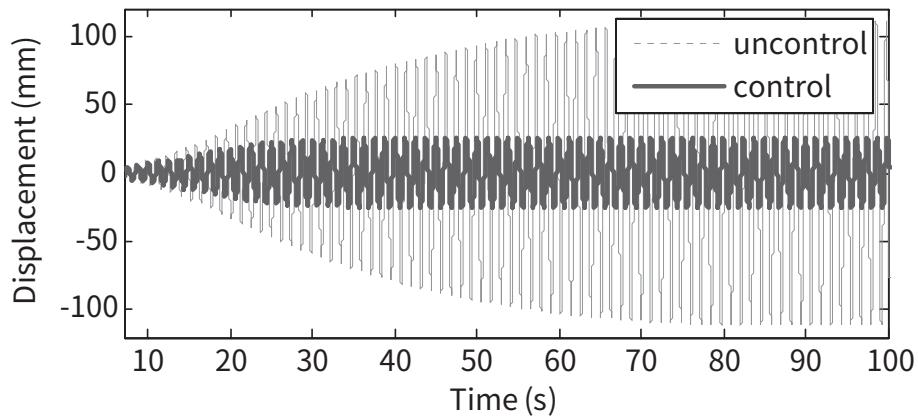


Figure 2.55: Displacement in the time domain by the hybrid test (TMD direction, 0.83 Hz)

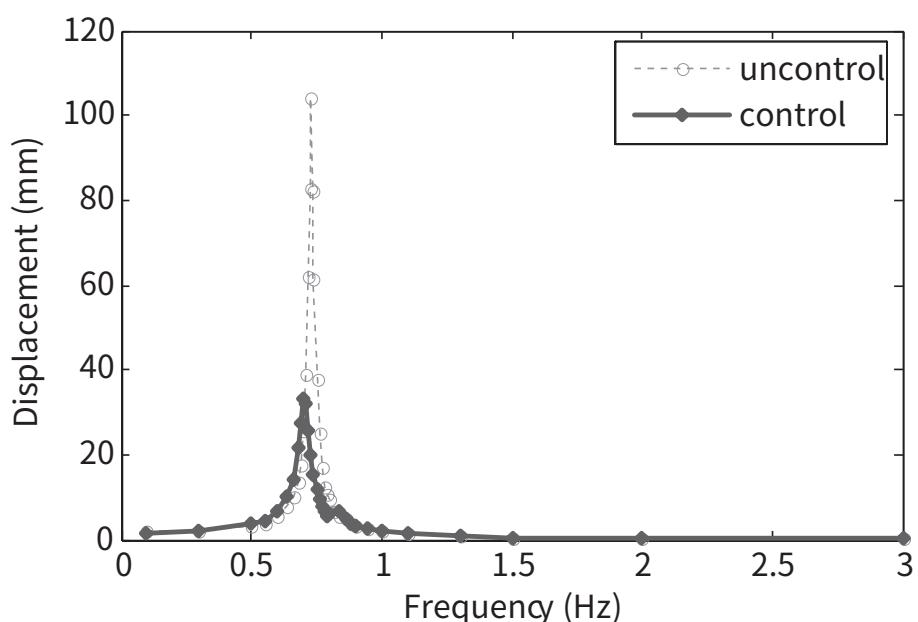


Figure 2.56: Displacement in the frequency domain by the hybrid test (TLCD direction)

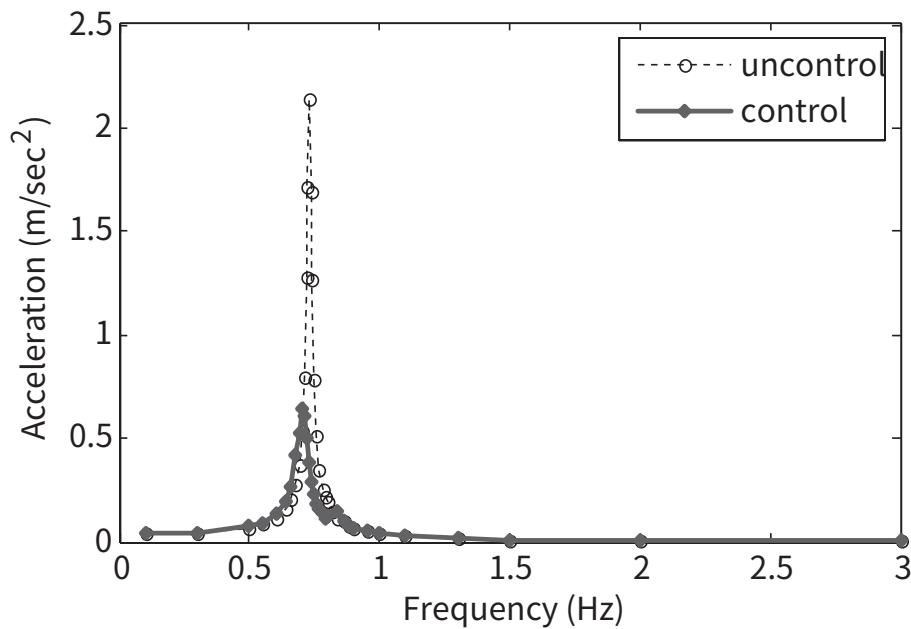


Figure 2.57: Acceleration in the frequency domain by the hybrid test (TLCD direction)

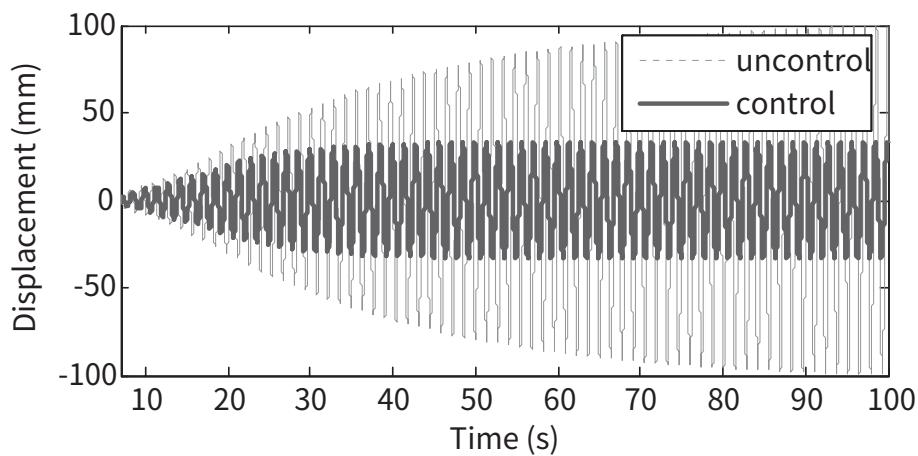


Figure 2.58: Displacement in the time domain by the hybrid test (TLCD direction, 0.73 Hz)

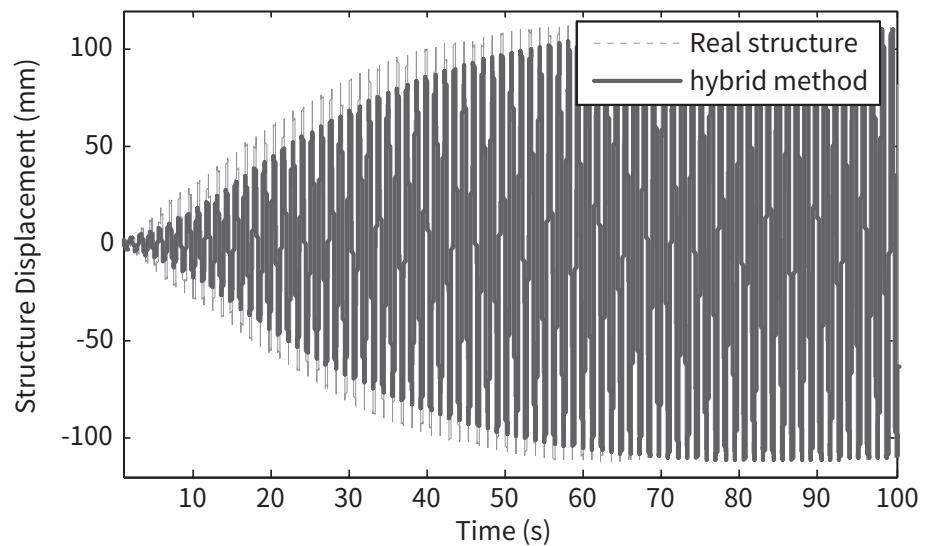


Figure 2.59: Uncontrolled displacement comparison in the time domain between the real-structure test and hybrid testing method (TMD direction)

## 2.7 Summary

In this chapter, a substructuring technique, TLD-, TLCD- and TLMD-building structure hybrid testing method for the shaking table test was proposed. The proposed testing technique adopts the upper part of the whole structure or nonlinear control devices as the experimental substructure, which corresponds to a physical test model. The lower part of the whole structure or whole building structure is modeled numerically. In order to verify the validity and accuracy of the proposed technique, a shaking table test was conducted. The result of the study can be summarized as follows.

1. To reduce the distortion of the interface acceleration, the inverse transfer function of the shaking table was identified, and its state space realization was implemented in the shaking table controller.
2. In this thesis, the linear transfer function approach for controlling the motion of a shaking table was considered to experimentally verify the proposed method for a linear experimental part. However, this approach would be inappropriate in a coupled non-linear system leading to experimental instability. Therefore, in such case, the controller using the inverse transfer function of shaking table, shown in Figure 2.6, would be modified to compensate an experimental instability.
3. The interface force between the experimental and numerical substructures was obtained using only acceleration measurement and mass information so that high-capacity loads cell and installation jigs are not required in the substructuring technique.
4. The proposed method basing the interface force measurement on acceleration measurements from an experimental substructure is partially available only when the mass distribution is discrete - for example, this technique would be applicable to the TMD as an experimental part. Also, the interface force measurement using force transducers is required to perform the proposed method when wind forces are applied to the experimental substructure.
5. Experimental results demonstrate that the proposed substructuring technique can reproduce the dynamic behavior of the assumed whole structure.
6. An unexpected vibration of the experimental substructure can be induced by the feedback of responses including its inherent natural modes and then by the error occurred in calculating the numerical substructure.

7. It is considered that to minimize the effect of natural modes of an experimental substructure on the substructured system, the structural model as heavily-damped as possible would be used as an experimental part.
8. The proposed technique can be extended to the substructuring technique with the middle part of a whole structure in combination with the conventional substructuring technique employing lower part as the experimental substructure.
9. The TLD installed on the top floor of the structure is physically tested, and simultaneously numerical calculation is carried out for the assumed analytical structural model.
10. Comparison between the structural responses obtained by the hybrid testing method and the conventional shaking table test of a single story steel frame with TLD and TLCD indicates that the performance of the TLD and TLCD can be accurately evaluated using the hybrid testing method without the physical structural model.
11. The uncontrolled and TLD-controlled structural responses of a three-story structure are obtained by the hybrid testing method in both time and frequency domains, showing that TLD can effectively mitigate the seismic responses of building structures and the hybrid testing method can reproduce the dynamic behavior of TLD-structure interaction systems for both the uncontrolled and controlled case.
12. The hybrid testing method can also be applied to the performance evaluation of new designed, tuned liquid typed damper which has strong inherent nonlinearity such as TLMD.

The bi-directional control performance of the TLMD was confirmed through the conventional and hybrid testing method. First, resonance frequencies in the TMD and TLCD control direction were confirmed at 0·82 and 0·73 Hz. In the TMD control direction (x-direction), 80% of the uncontrolled peak response of the target structure was removed by a TLMD. Also, in the TLCD control direction (y-direction), 70% was removed by a TLMD. The control performances of the TLMD were checked and compared with the two testing methods. In the TLCD and TMD control direction, the errors of the peak values between the conventional and hybrid testing method were 0·01 and 0·02~0·04, respectively. Also, the errors of RMS values between the conventional testing method and hybrid testing method were up to 0·06. However, tuning error and outside environment are thought as the causes of the differences. The error of the convergence times reaching to the peak value as shown in Figure 2.59. This phenomenon is caused by the minute error of the mass, stiffness and damping ratio of the numerical analysis, which was thought for either reason. As showing the similar results of the two kinds of testing methods, the

hybrid testing method, which does not require the physical structural model but with simple installation, as well as the conventional testing method, can accurately evaluate the control performance of a control device.



# Design of Excitation System for Simulating Dynamic Loads

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## 3.1 Design Controller of Excitation System for Simulating Wind Responses

In this section, simulation of wind-induced responses of a building structure using linear mass shaker (LMS) and active tuned mass damper (ATMD) as an actuator is conducted as shown in Figure 3.1. For the linear actuator to keep the structure in the target response trajectory, an inverse transfer function of a structural response to the shaker force is obtained using a state space form governing equation of the structure, and the discrete Fourier transform of structural response is performed. Filter and envelope function are used such that the error between wind and actuator induced responses is minimized by preventing the shaker from actuating unexpected modal responses and initial transient responses. The effectiveness of the proposed method is verified through a numerical example of a 76 story-benchmark building excited by wind load of which deterministic time history is given. The effect of the type of the target structural response on a convergence of the actuator force signal and the error magnitude is investigated.

### 3.1.1 Force of actuator

The state-space form equation of a structure excited by wind load  $f$  and the shaker generated force  $u$  of size  $r$  is as follows.

$$\begin{aligned}\dot{\mathbf{z}} &= \mathbf{Az} + \mathbf{B}_f f + \mathbf{B}_u u \\ y &= \mathbf{Cz} + \mathbf{D}_f f + \mathbf{D}_u u\end{aligned}\tag{3.1}$$

where  $z$  is the state vector and  $y$  is the output vector of size  $m$ . The output transfer function to  $f$  or  $u$  is given by

$$\begin{aligned}\mathbf{T}_{yf} &= \mathbf{Y}_f(s)\mathbf{F}(s)^{-1} = \mathbf{C}(s\mathbf{I} - \mathbf{A})^{-1}\mathbf{B}_f \\ \mathbf{T}_{yu} &= \mathbf{Y}_u(s)\mathbf{U}(s)^{-1} = \mathbf{C}(s\mathbf{I} - \mathbf{A})^{-1}\mathbf{B}_u\end{aligned}\tag{3.2}$$

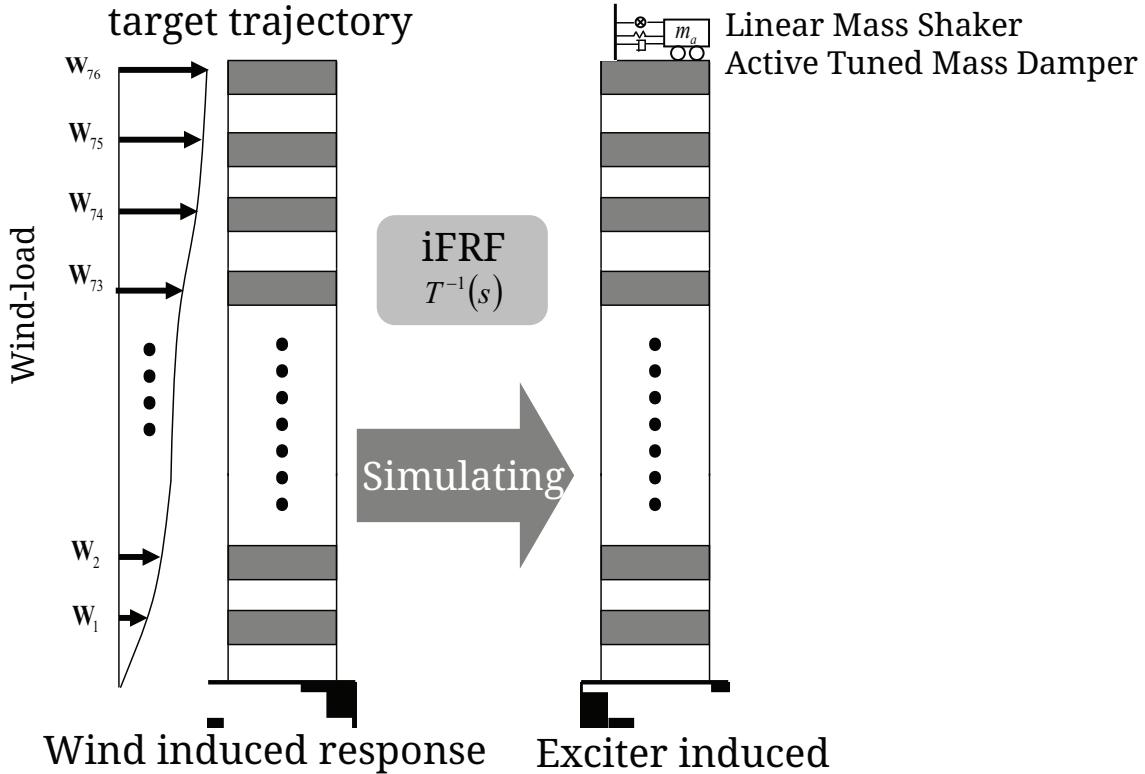


Figure 3.1: Scheme of simulation of wind induced responses using LMS and ATMD

where the scalar  $s$  is a complex variable  $j\omega$ . The inverse of  $\mathbf{T}_{yu}$  exists only if  $r$  equals to  $m$  and the Laplace transform of  $u$  providing the identical output to wind load induced one is determined as

$$\begin{aligned}\mathbf{U}(s) &= \mathbf{T}_{yu}^{-1} \mathbf{Y}_u(s) \\ &= \mathbf{T}_{yu}^{-1} \mathbf{T}_{yf} \mathbf{F}(s)\end{aligned}\quad (3.3)$$

when  $r$  is smaller than  $m$ , the number of structural responses which can be modulated by  $u$  is restricted to  $r$ , and target structural response should be selected. The Laplace transform of input realizing the target response  $\bar{y}$  of size  $r$  is

$$\begin{aligned}\hat{\mathbf{U}}(s) &= \hat{\mathbf{T}}_{yu}^{-1} \hat{\mathbf{Y}}_u(s) \\ &= \hat{\mathbf{T}}_{yu}^{-1} \hat{\mathbf{Y}}_f(s) \\ &= \hat{\mathbf{T}}_{yu}^{-1} \hat{\mathbf{T}}_{yf} F(s)\end{aligned}\quad (3.4)$$

where,  $\hat{\mathbf{T}}_{yu}^{-1}$  is a sub-matrices of  $\mathbf{T}_{yu}^{-1}$  is constructed by extracting the columns in  $\mathbf{T}_{yu}^{-1}$  corresponding to the target response.

## Filter and envelope function

The transfer function of structural response may have frequency intervals in which the magnitude is as small as zero, and in those intervals, the magnitude of the inverse transfer function increases infinitely. Because the input force is calculated by the product of the inverse transfer function and the output signal, significant input force may be calculated in order to realize the small magnitude of output components corresponding to the intervals. This force implies that the shaking system becomes very sensitive to the slight frequency variation of the output signal resulting from measurement noise and spectral leakage which is inevitable in signal processing using discrete Fourier transform, and then unnecessarily high input energy excites unexpected frequency response such that target response may not be induced. Particularly, low-frequency component leads to a large stroke of the shaker. In this chapter, following band-stop filter (BSF) using cosine function is used to prevent the unexpected frequency response from occurring.

$$\hat{\mathbf{U}}_p(\omega) = G(\omega)\hat{\mathbf{U}}(s) \quad (3.5)$$

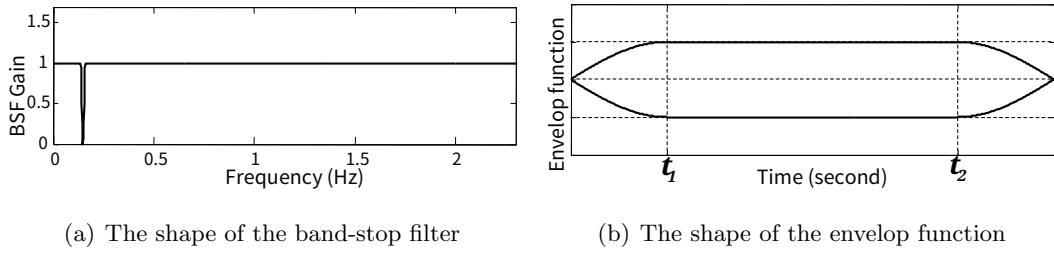
where,

$$G(\omega) = \frac{1 - a_{co}}{2} \cos\left(\frac{2\pi}{\omega_2 - \omega_1}\omega\right) + \frac{1 + a_{co}}{2} \quad (3.6)$$

$$a_{co} = \begin{cases} \omega < \omega_1 & : 1 \\ \omega_1 \leq \omega \leq \omega_2 & : 0 \\ \omega > \omega_2 & : 1 \end{cases} \quad (3.7)$$

where,  $\omega_1$  and  $\omega_2$  are frequencies defining the cut-off frequency interval, and  $a_{co}$  is gain value of the cut-off frequency. Figure 3.2(a) shows the shape of the band-stop filter

In discrete Fourier transform dealing with the finite duration discrete signal as an infinite one multiplied by a rectangular window, the original signal in time domain is distorted especially in initial and final time intervals. This distortion can be reduced by using an envelope function such that the given deterministic wind load has ascending and descending time intervals. Although the envelope function changes the deterministic wind load, the effect of this another distortion would be trivial in evaluating the characteristics of wind load induced response because the grave concern is generally in the intermediate time of the total loading duration when the peak response is expected to occur. Figure 3.2(b) shows the shape of the envelope function used in this thesis.



(a) The shape of the band-stop filter      (b) The shape of the envelop function

Figure 3.2: Exciter gain shape of the band-stop filter and the envelop function.

### 3.1.2 Numerical Example

#### 76 story wind-induced benchmark buildings

The wind-induced response simulating actuator is applied to a 76-story 306 meters office tower benchmark building which has slender with a height to width ratio of  $306.1/42 = 7.3$ . Because the deterministic across-wind load data is given through wind tunnel tests for this benchmark building, the force of the actuator realizes target across-wind induced structural response can be calculated using Eq. (3.4). In order to reduce numerical computation time, a 23 degree of freedom (DOF) state reduced-order system model proposed by Yang et al. (2004b). is used in this thesis. The wind load vector is modeled physically by lumping wind forces on adjacent floors at the locations that correspond to the 23 DOF model. Figures 3.3(a), 3.3(b) shows the plan view and elevation view of the 76th benchmark building and Figures 3.3(c), 3.3(d) show the mode shape of first three mode of the structure and time history of wind-load.

### 3.1.3 Error evaluation criteria

To verify the effectiveness of proposed method through the comparison between the wind and actuator induced structural responses, two error criteria are considered in time and frequency domains, respectively. In time domain, the normalized tracking error is defined as

$$e_t = \frac{1}{n} \sum_{i=0}^{n-1} \sqrt{\{z_a(i\Delta t) - z_f(i\Delta t)\}^2} \Bigg/ \max \left[ \sqrt{z_f(i\Delta t)^2} \right] \quad (3.8)$$

where,  $\Delta t$  denotes time interval,  $n$  denotes the data number.  $x_a(i\Delta t)$  and  $x_f(i\Delta t)$  are, respectively, actuator and wind induced structural response at the  $i$ th time step.

In frequency domain, the normalized tracking error is defined as

$$e_f = \frac{1}{N} \sum_{i=1}^N \sqrt{|Z_a(\omega_i) - Z_f(\omega_i)|^2} \Bigg/ \max \left[ \sqrt{|Z_f(\omega_i)|^2} \right] \quad (3.9)$$

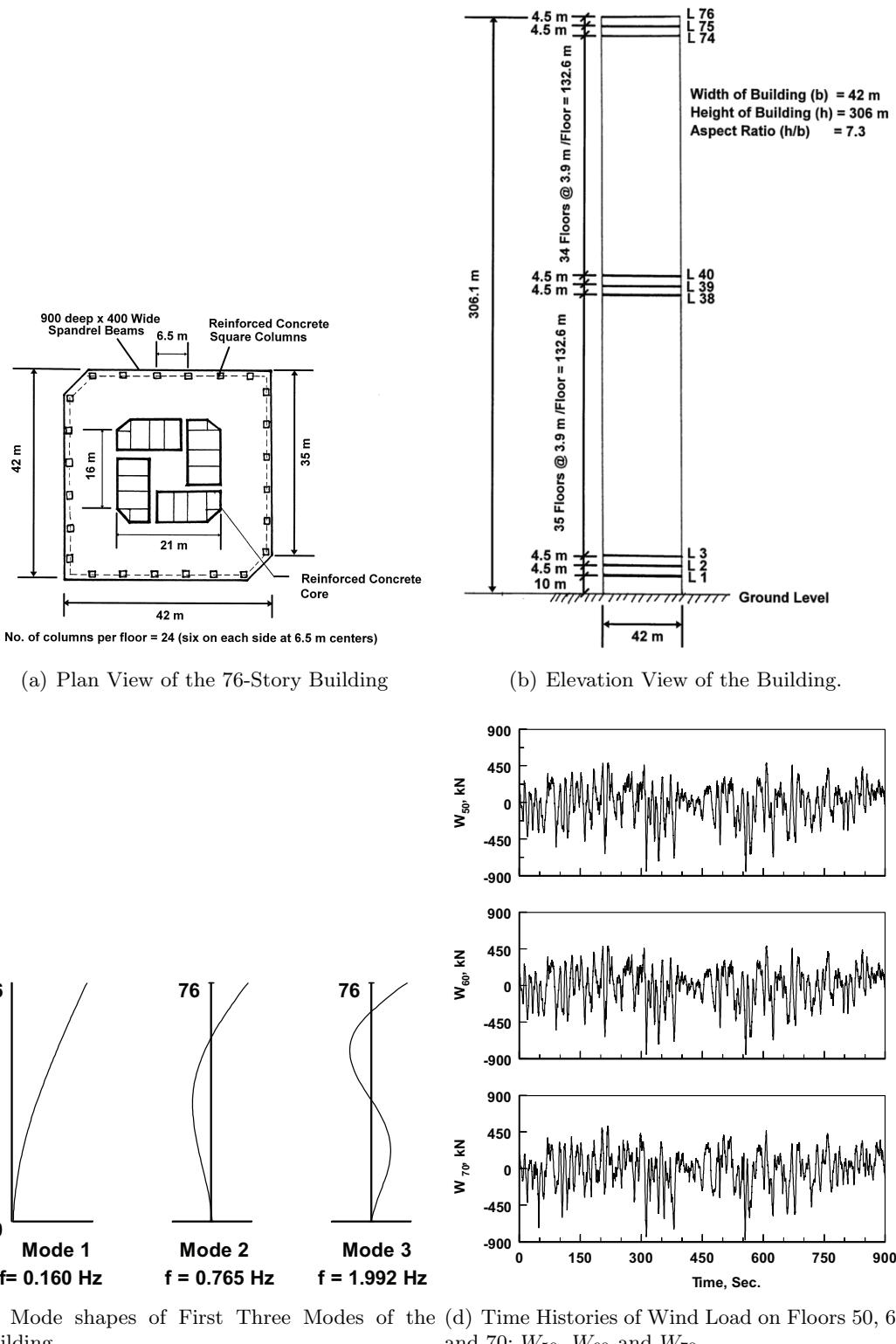


Figure 3.3: 76th story benchmark model.

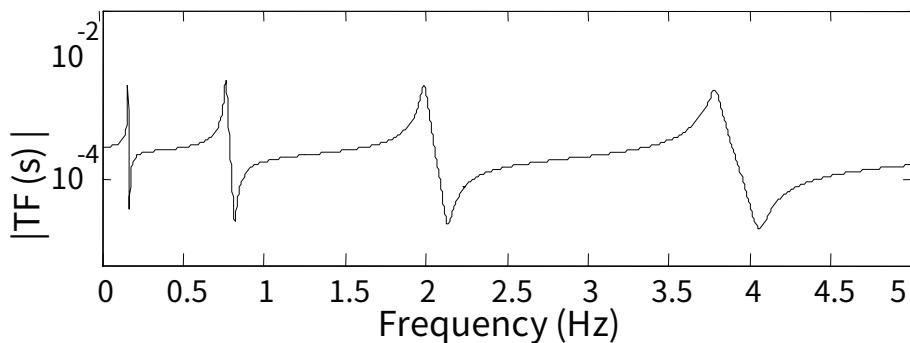
where  $N$  is the number of frequency response data, and  $X_f(\omega)$  and  $X_t(\omega)$  are discrete-Fourier transformation of  $\ddot{x}_f(t)$  and  $\ddot{x}_t(t)$ , which  $E[\epsilon_f(\omega)]$  is normalized mean tracking

error in frequency domain.

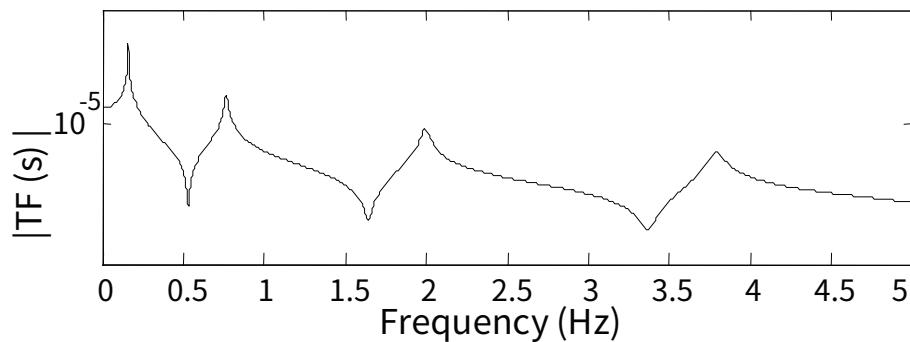
### LMS excitation

In this subsection, an LMS which can produce arbitrary desired force is used as an actuator. LMS is assumed to have a mass of 500 metric ton and be installed at 76th floor. The mass is identical to that of an ATMD used as a vibration dissipation device for the benchmark problem. The mass is about 45% of the top floor mass, which is 0.327% of the total mass of the building.

Figures 3.4(a) and 3.4(b) show the transfer function of the 75th floor acceleration and displacement responses. It is observed that acceleration transfer function in Figure 3.4(a) has zero near the each modal natural frequency with slightly larger value than the corresponding natural frequency. Especially the zero exists near the first modal frequency which is expected to dominate the overall wind-induced structural responses.



(a) 75th story acceleration transfer function.

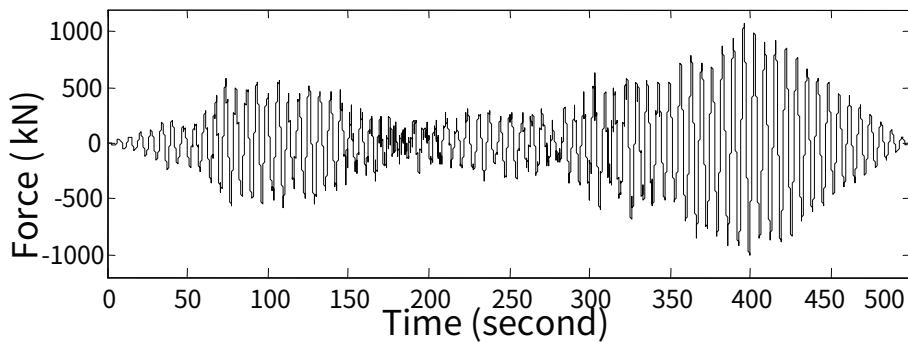


(b) 75th story displacement transfer function.

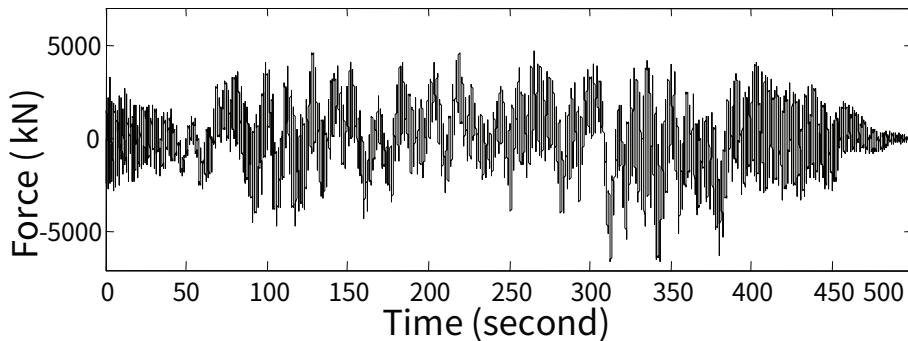
Figure 3.4: Transfer function of 75th story responses to LMS.

Figure 3.5 shows the frequency response function and the time history of the actuator force obtained without using a filter when the target response is acceleration or displacement of the 75th floor. It is known from Figure 3.5 that much larger force is required for the shaker to achieve the target displacement than acceleration response, and further-

more there exist high-frequency components in Figure 3.4(b), which result in high-speed switching of control force as shown in Figure 3.5(b). In practice, hydraulic actuators popular in civil engineering structures are not suitable for this undesirable chattering problem which causes spillover instability in higher modes, and acceleration response concerned with serviceability criteria is more critical for such high-rise building excited by wind load as this benchmark building than displacement. Figure 3.6 shows the comparison between the frequency responses of wind and LMS induced 75th-floor acceleration and displacement when the target response is 75th displacement. It is apparently demonstrated that the wind and LMD induced displacement coincide well with each other while acceleration responses are very different. Based upon the observation in Figure 3.5 and 3.6, only acceleration response is considered as target response for calculating LMD force reproducing wind induced displacement as well as acceleration in this thesis.



(a) LMS force targeting on 75th floor acceleration.



(b) LMS force targeting on 75th floor displacement.

Figure 3.5: LMS force (unfiltered).

Numerical analyses are conducted with/without using the filter and envelope function for different target responses. Table 3.1 lists the cut-off frequencies used for canceling undesirable amplification effect by the zero in the inverse transfer function of each target response as mentioned in previous section. Envelope function with  $t_1 = 100$  second and  $t_2 = 100$  second is applied.

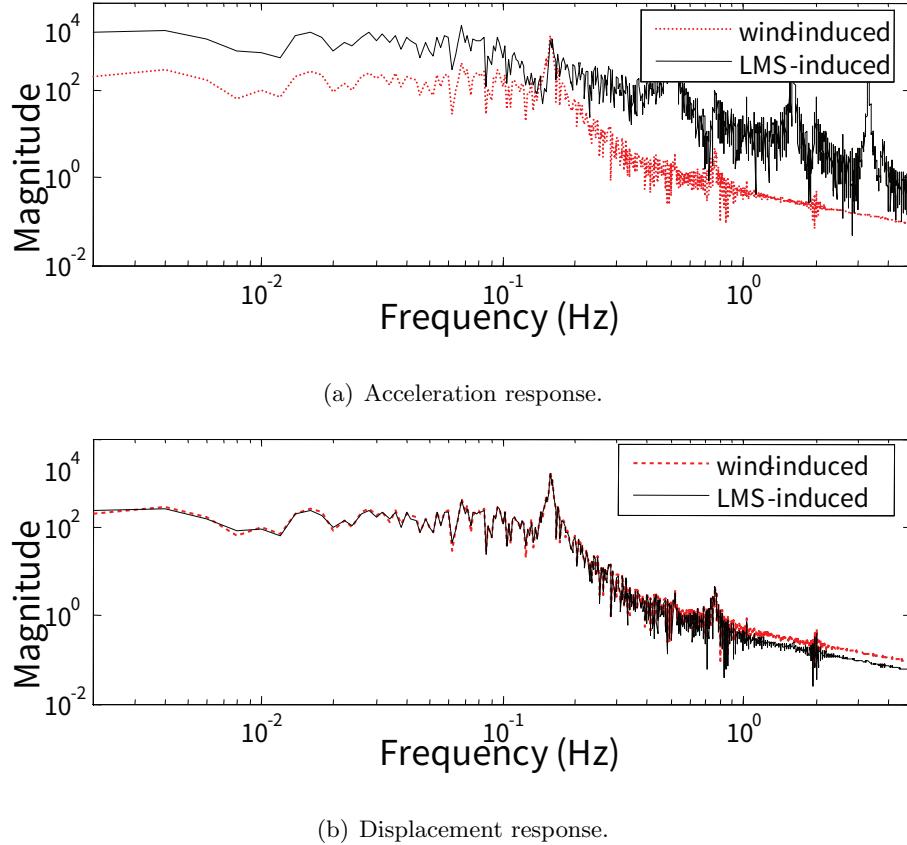


Figure 3.6: Frequency response of 75th floor with LMD force targeting displacement response.

Table 3.1: Cutoff frequency for filter design

Target response	$\omega_1$ (rad/sec)	$\omega_2$ (rad/sec)
75th floor acceleration	1.01	1.13
50th floor acceleration	8.17	8.80
30th floor acceleration	17.59	20.73

Figure 3.7 shows the floor distribution of the time and frequency domain errors defined in the previous subsection, which are obtained with/without using the filter for different target responses. Processed signal denotes one yielded by using the filter. It is identified from Figure 3.7 that the processed signal using filter significantly reduce the magnitude of the tracking error when the target response is 75-floor acceleration of which transfer function has band-stop frequency near the first modal frequency (as observed in Figure 3.4(a)) while the processing effect is trivial when the target response is 30th or 50th floor acceleration of which transfer function has zero away from the first modal one. Also, it is known from observing the error distribution that the targeted responses are almost identical to wind-induced ones with a small magnitude of error while the other non-targeted responses are slightly different with increasing error. Targeting 30th or 50th-

floor acceleration provides greater discrepancy between the wind and LMS induced 75-floor accelerations which are critical in evaluating serviceability. The comparison between the results in Figure 3.7(a) and 3.7(b) indicates that the distribution tendency of  $e_t$  and  $e_f$  is quite different and the magnitude of  $e_t$  is much larger than that of  $e_f$ . When targeting response is 75th floor acceleration, both values of  $e_t$  and  $e_f$  are smallest for the targeted 75th floor acceleration, but the value of  $e_f$  becomes larger for the other story responses while the value of  $e_f$  generally keeps the smallest except for 30th story acceleration. The larger value of  $e_t$  results from the phase difference between the wind and LMS induced responses since  $e_t$  is obtained based on the response difference at the same time step. The phase of the wind-induced response is not the important parameter in evaluating the wind resistance performance of a building structure,  $e_f$  can be said to be the more appropriate index for evaluating the wind response reproducing the performance of LMS.

Figures 3.8 and 3.9 compare the time histories of the wind and LMS induced structural responses for the cases that the target responses are, respectively, 75th and 30th-floor accelerations and filter are applied for the design of LMS. Figure 3.8 shows that the LMS induced acceleration response including targeted 75th-floor acceleration agree well with wind-induced ones while displacement responses at all floors are underestimated by LMS. It is observed in Figure 3.9 that the shaker simulated targeted 30th-floor acceleration as well as displacement responses but overestimates 75th-floor acceleration.

### 3.1.4 ATMD excitation

In the 76th story building benchmark problem, ATMD is used as an example controller, which is composed of spring and viscous damper in addition to the mass and actuator of the LMS. In this subsection, the ATMD is considered as another exciter. The mass of the ATMD is 500 metric ton, and the undamped natural frequency and damping ratio are, respectively, 0.16Hz and 20% (Yang et al., 2004b). Simplification for the control environments have been made, in particular, the actuator dynamics and controller-structure interaction are not considered in the benchmark problem. The equation of motion of the building equipped with an ATMD on the top floor can be expressed as

$$\mathbf{M}\ddot{\mathbf{x}} + \mathbf{C}\dot{\mathbf{x}} + \mathbf{K}\mathbf{x} + \mathbf{H}u = \eta\mathbf{W} \quad (3.10)$$

and considering no wind-load input, the equation of motion of the building can be

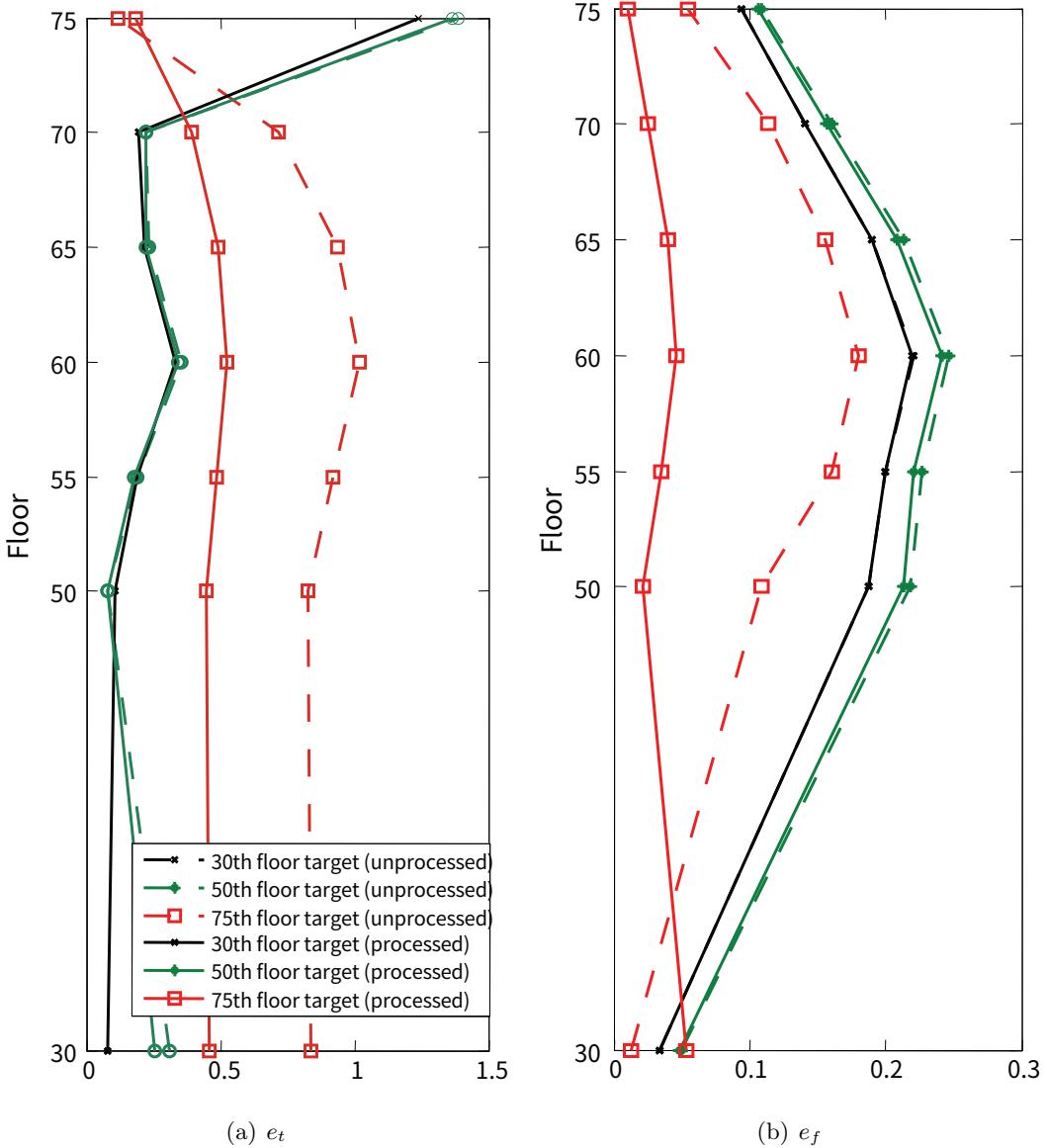


Figure 3.7: Error distribution according to the filter usage and floor of target response.

expressed as

$$\begin{bmatrix} \mathbf{M}_s & 0 \\ \mathbf{0} & m_t \end{bmatrix} \begin{Bmatrix} \ddot{\mathbf{x}}_s \\ \ddot{x}_t \end{Bmatrix} + \begin{bmatrix} \mathbf{C}_s + c_t \mathbf{B}_t \mathbf{B}_t^\top & -c_t \mathbf{B}_t \\ -c_t \mathbf{B}_t^\top & c_t \end{bmatrix} \begin{Bmatrix} \dot{\mathbf{x}}_s \\ \dot{x}_t \end{Bmatrix} + \begin{bmatrix} \mathbf{K}_s + k_t \mathbf{B}_t \mathbf{B}_t^\top & -k_t \mathbf{B}_t \\ -k_t \mathbf{B}_t^\top & k_t \end{bmatrix} \begin{Bmatrix} \mathbf{x}_s \\ x_t \end{Bmatrix} = \begin{bmatrix} -\mathbf{B}_t \\ 1 \end{bmatrix} u \quad (3.11)$$

where  $\mathbf{B}_t$  is position vector of floor installed ATMD,  $\mathbf{M}_s$ ,  $\mathbf{C}_s$  and  $\mathbf{K}_s$  is the mass, damping coefficient and stiffness matrix of the structure and  $m_t$ ,  $c_t$  and  $k_t$  is the mass, damping coefficient and stiffness of the ATMD.

Figure 3.10(a) shows the transfer function of the 75th-floor acceleration to the abso-

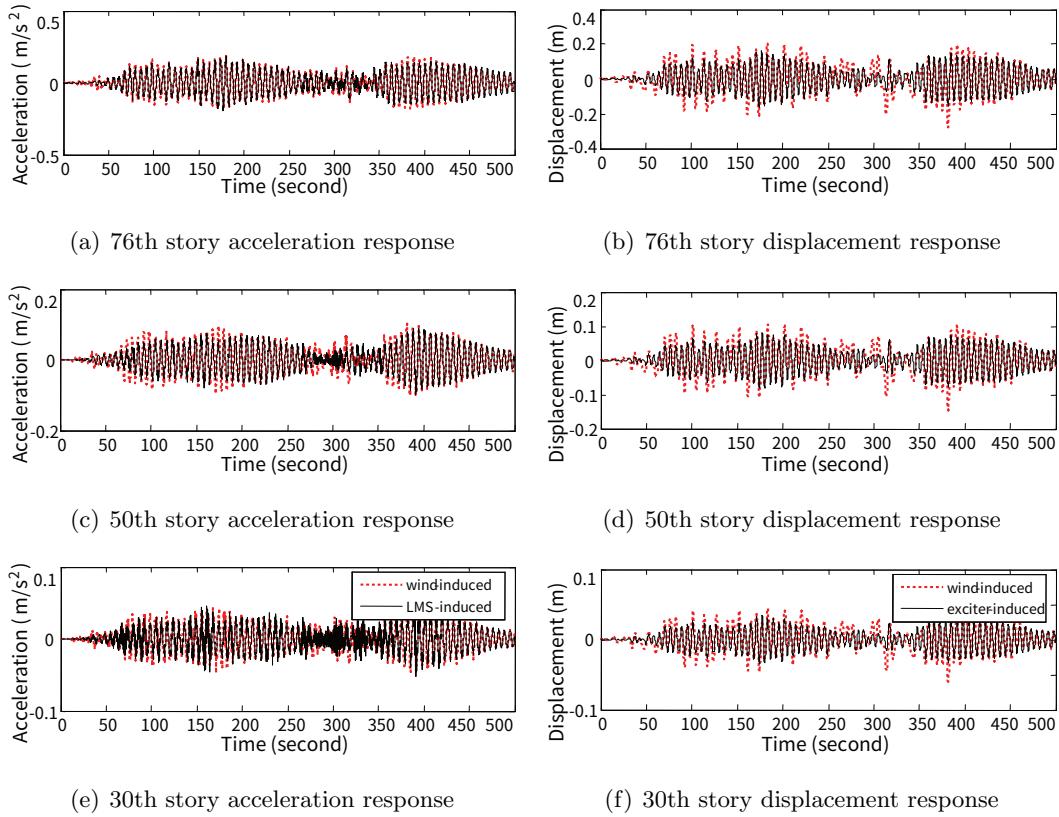


Figure 3.8: Wind and LMS induced acceleration responses (when the target is 75th floor acceleration).

lute acceleration the ATMD. No zero point is observed in the vicinity of the first modal frequency unlike the case for LMS, which indicates that band-stop filter is not required for preventing the ATMD to excite the unexpected frequency response. Figure 3.10(b) shows the frequency response of the actuator force of the ATMD and Figure 3.10(c) and 3.10(d) show the frequency and time responses of the sufficient force applied to the structure by the ATMD.

Figure 3.11 shows the time history comparison between wind induced acceleration and ATMD induced one targeting on 75th story acceleration response.

All acceleration responses coincide well with each other, but in displacement response, there exists slight underestimation and overestimation according to the time ranges.

Figure 3.12 shows the comparison between the frequency responses of 75th-floor acceleration, and it is observed that wind and ATMD induced responses show good agreement overall frequency range.

Figure 3.13 shows the error distribution according to the floor of targeted acceleration response. From Figure 3.13(b) showing that  $e_f$  has the smallest value over floors when the target response is 75th floor acceleration, and the corresponding value ranges only between 1% and 10%, ATMD targeting 75th floor acceleration can be said to provide the

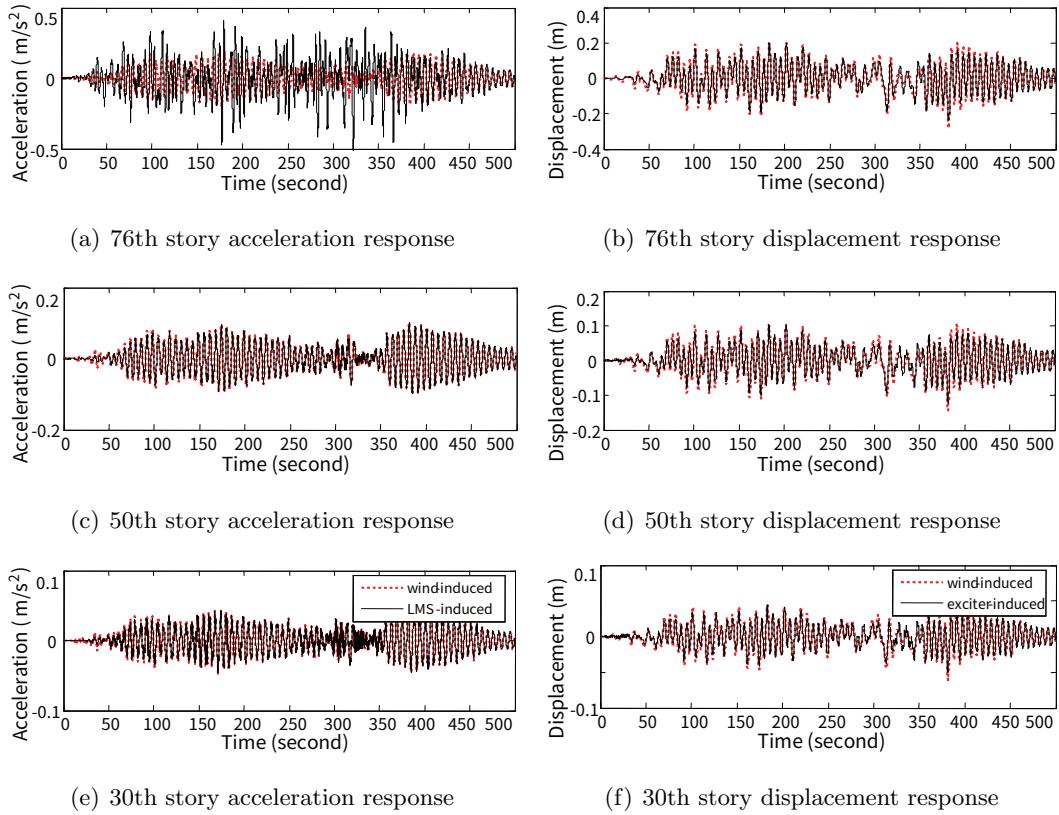


Figure 3.9: Wind and LMS induced acceleration responses (when the target is 30th floor acceleration).

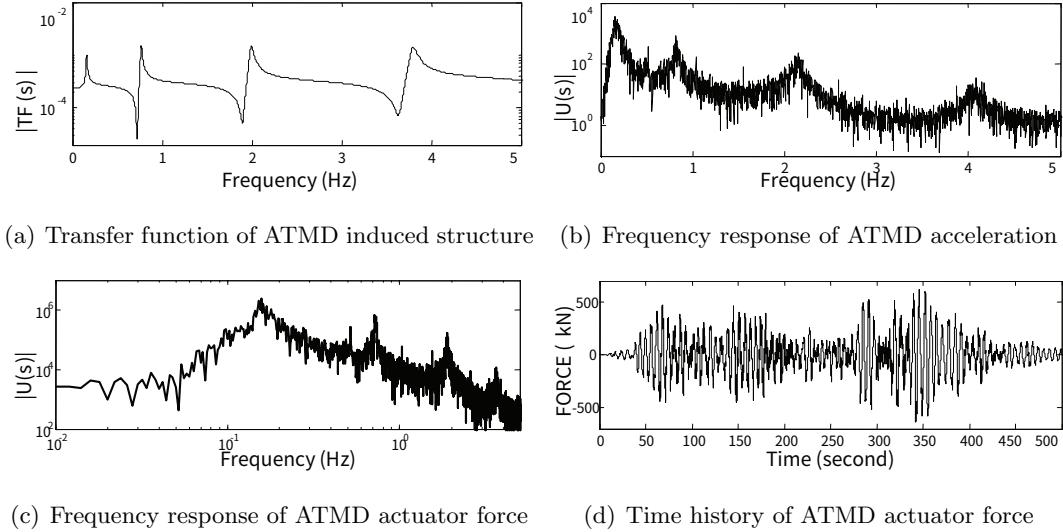


Figure 3.10: ATMD Excitation.

best performance, and it can precisely reproduce the wind-induced acceleration response of all floors including targeted 75th floor.

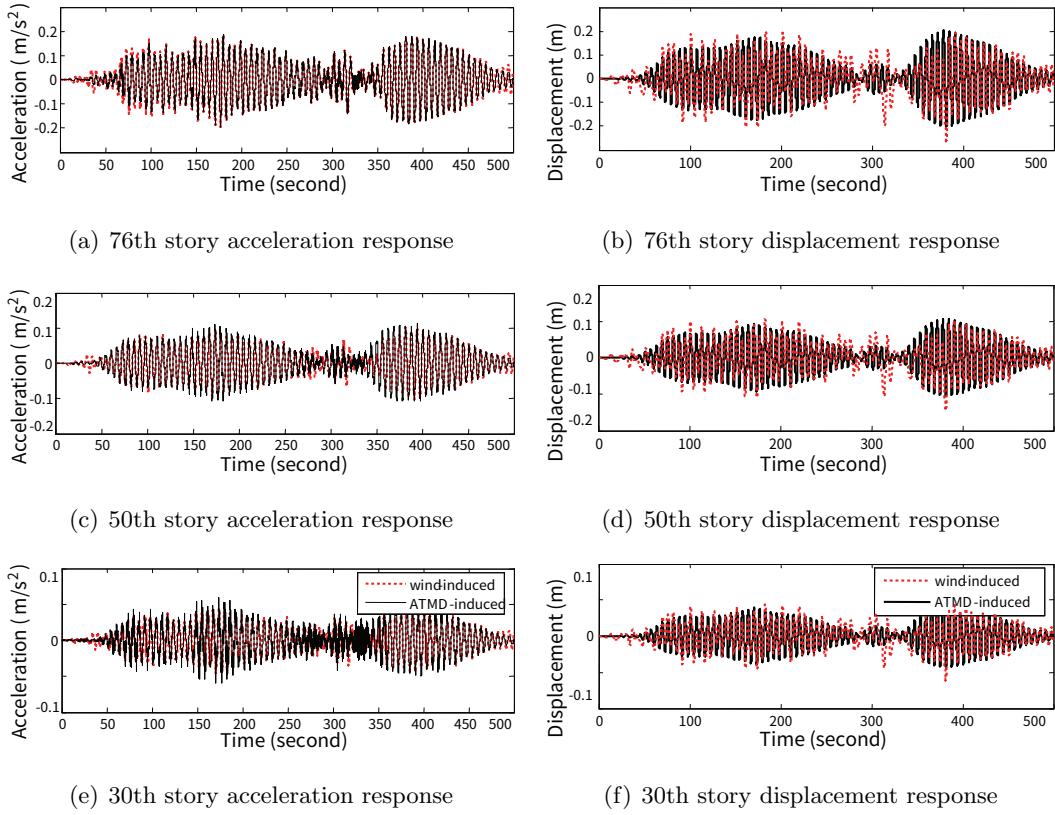


Figure 3.11: Wind and ATMD induced acceleration responses (when the target is 75th floor acceleration).

### 3.1.5 Comparison between LMS and ATMD

Figure 3.14 shows time history of the actuator forces in LMS and ATMD excitation systems. The peak actuator force required for ATMD is larger than that for LMS. Figure 3.15 compares the stroke of LMS with/without filter and ATMD. The stroke of LMS with filter is much smaller than those of LMS without filter and ATMD. This fact implies that ATMD requires massive stroke to show good performance in wind-induced response realization and this stroke requirement should be checked in the design of ATMD.

Table 3.2 shows the numerical values of the error, actuator force, and actuator stroke in LMS with/without the filter, and ATMD systems. The errors are obtained when the target and evaluation responses are identical, and actuator force and stroke are obtained when the target response is 75th-floor acceleration. The facts observed from Figures 3.14 and 3.15 that the performance of LMS can be enhanced using the band-stop filter and ATMD reproduces wind-induced response better than LMS, but ATMD requires larger actuator force and stroke can be identified once more.

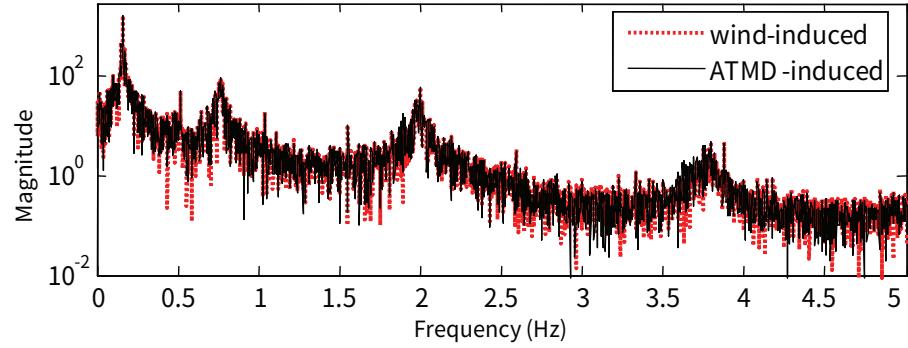


Figure 3.12: Frequency response of wind and ATMD induced 75th floor accelerations

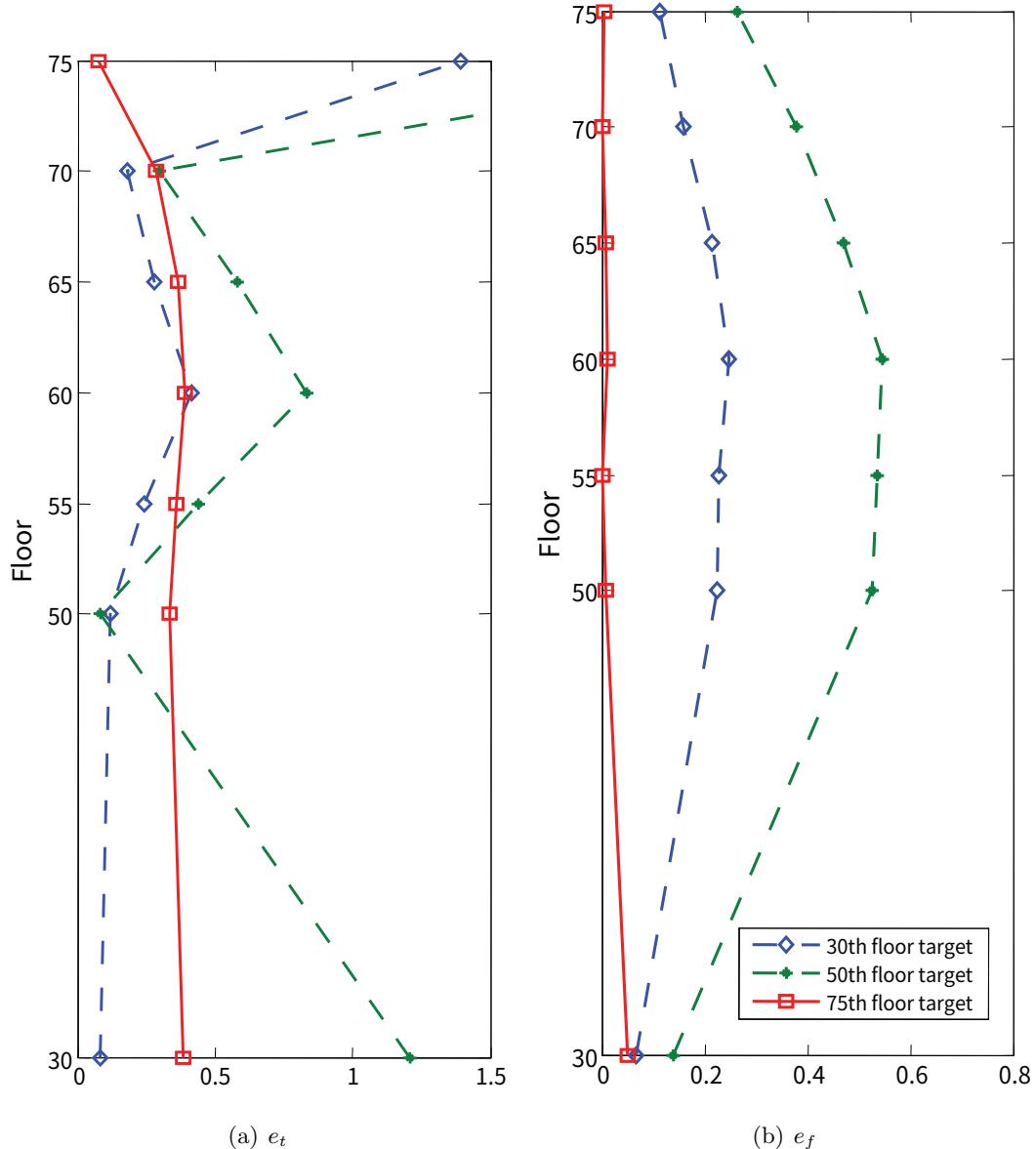


Figure 3.13: Error Distribution with ATMD excitation.

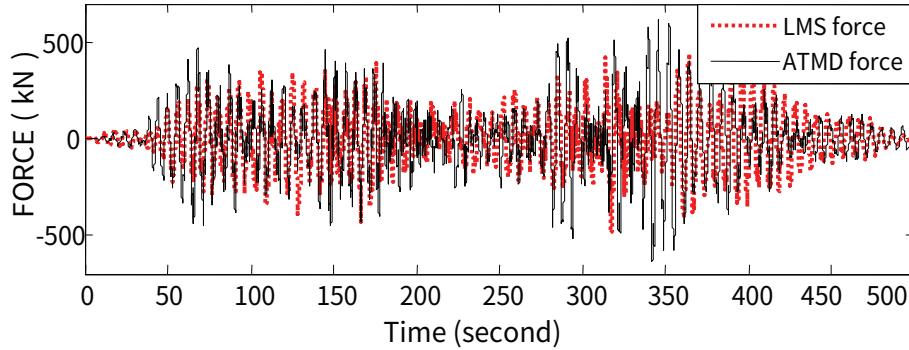


Figure 3.14: Comparison between actuator forces in LMS and ATMD

Table 3.2: Comparison between LMS and ATMD

		LMS (unfil- tered)	LMS (filtered)	ATMD
75th floor acceleration	$e_t$	0.117	0.192	0.074
	$e_f$	0.054	0.011	0.003
50th floor acceleration	$e_t$	0.081	0.081	0.083
	$e_f$	0.219	0.213	0.525
30th floor acceleration	$e_t$	0.082	0.082	0.082
	$e_f$	0.034	0.033	0.065
Stroke	Peak (m)	1.402	0.519	1.265
	RMS (m)	0.436	0.183	0.333
Actuator force	Peak (kN)	1068.22	441.68	655.78
	RMS (kN)	323.92	136.44	170.39

### 3.1.6 Summary

The design of excitation systems for simulating wind-induced responses of a building structure was presented as a preliminary study for evaluating wind-resistance characteristics of building structures. The actuator forces of the LMS and ATMD were obtained using the inverse transfer function of structural responses. Also, band stop filter was used in LMS to remove zero of the transfer function such that undesirable modal excitation is prevented and envelop function was used to reduce the error occurring in transient initial states in both LMS and ATMD. The numerical analyses results from a 76-story benchmark building confirmed that the structural responses of a building structure excited by wind loads acting at all floors could be reproduced by the proposed excitation systems installed at a specific floor. The performances of the excitation systems were dependent on type and position of the target structural response for which acceleration response was suitable because targeting displacement response required large and high-speed changing

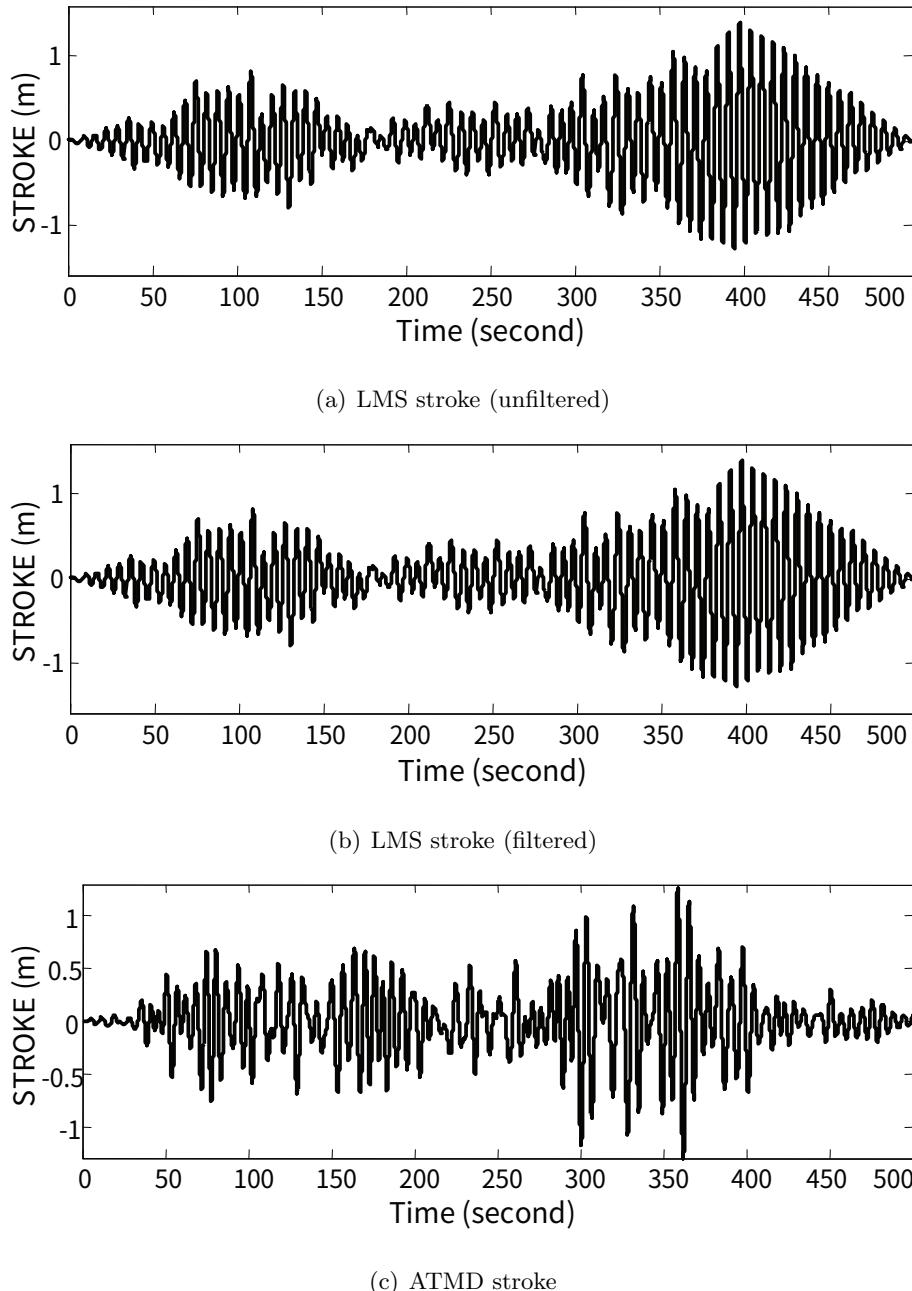


Figure 3.15: Stroke Comparison.

control force.

## 3.2 Forced vibration test of full-scaled five-story building structure simulating eqarthquake responses

### 3.2.1 Overview

In this section, the hybrid mass damper (HMD) controller for the pseudo-earthquake excitation test is designed by employing the method using the inverse transfer function of a structure, which is one of the methodologies proposed by the previous section and is verified for its experimental implementation. First, a real scaled five-story steel frame, in which an HMD installed, is shaken and then transfer functions from the HMD to structural responses of each story are obtained. Then, the FE model numerically calculated from the software for commercial use is renewed based on the modal information extracted from experimentally obtained transfer functions. Also, the earthquake responses based on the renewed structural model are numerically calculated, and the structure is excited by an HMD input signal which is produced to simulate a specific target response of a structure out of these numerically calculated earthquake responses. Finally, the effectiveness of pseudo-earthquake excitation presented in this chapter is verified by comparing numerically calculated seismic responses with experimentally measured ones.

The pseudo-earthquake excitation testing method presented in this chapter has a few importance in an engineering aspect. First, the testing method that is performed for a real scale structure in the field is free from lots of artificial constraints accompanied in a laboratory test. Secondly, valuable data, which are available for the verification of structural seismic performance and the evaluation of the availability of vibration technique required in structural health monitoring, can be acquired from the test. Thirdly, large shaking table is not required to evaluate the seismic response of real-scaled building structures, since in this thesis the earthquake response is simulated by actuating a structure with an HMD installed at the upper story.

### 3.2.2 Experimental Real-Scale Building Model

The experimental model, which is shown in Figures 3.16 and 3.17, is a full-scale five-story steel structure which has the story height of 6m, the plan of  $6m \times 6m$ , and the story mass of 20,000kg. Each floor is composed of four identical wide-flange type steel columns. An HMD using larger scale linear motor damper which was also designed as a passive damper has a moving mass on the fifth floor excited the model structure. Because the columns consist of I-shaped steel, the HMD was installed in minor axis of the structure.

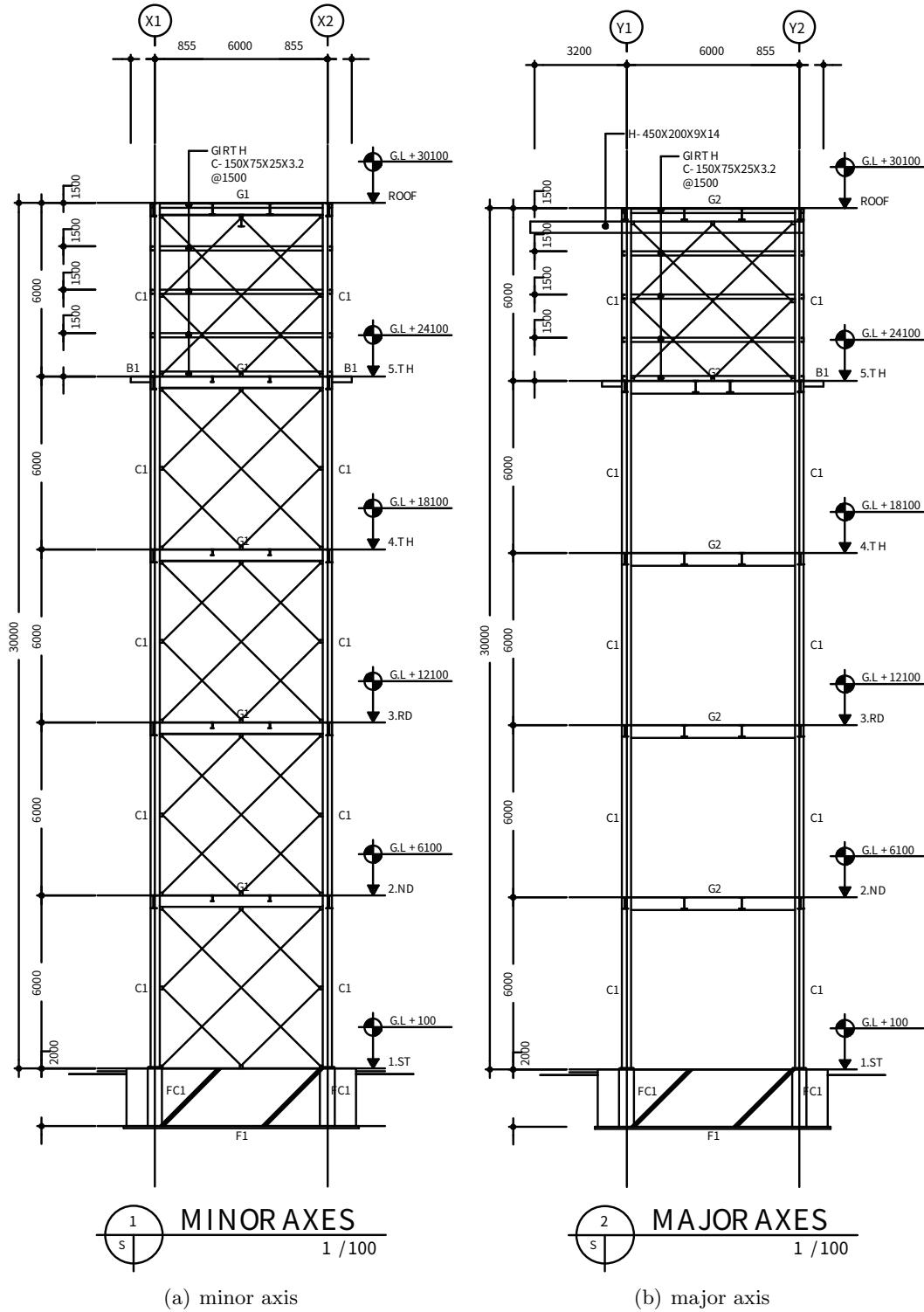
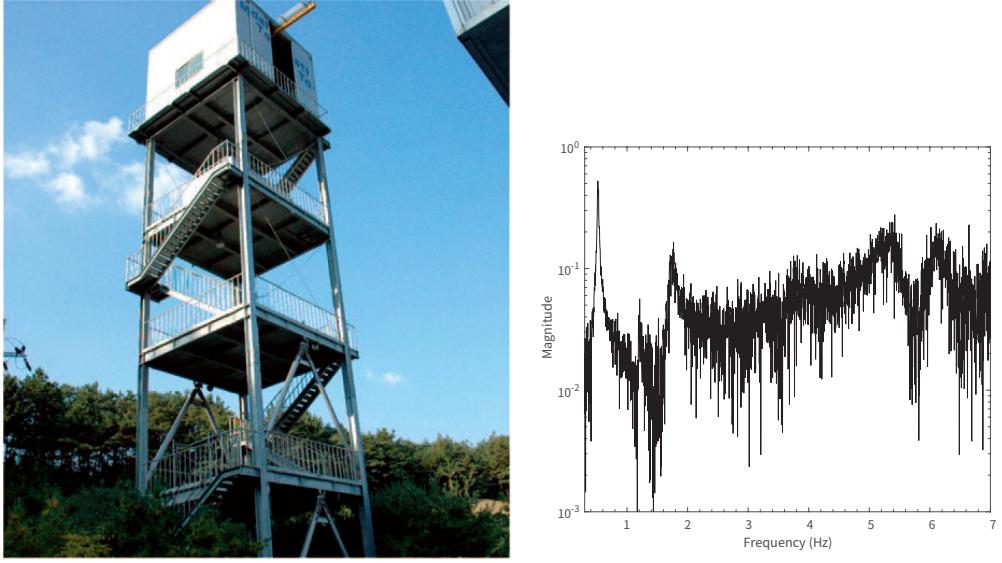


Figure 3.16: Elevation view of the target structure.

### 3.2.3 Field measurement and experimental system

Field measurement and experimental system is shown in Figures 3.18 and 3.19. In order to minimize the latency between the excitation and the measurement, one lap-top PC in



(a) Five-story building structure.

(b) Transfer function of target floor response to the HMD.

Figure 3.17: Photograph and transfer function of the target building structure.

Table 3.3: Member list of the model structure

MEMBER LIST	
C1	H-310×310×20×20
G1	H-400×200×8×13
G2	H-450×200×9×14
B1	H-200×100×5.5×8
B2	H-400×200×8×13
RB1	H-400×200×8×13
FC1	500×500

the experimental system was used for simultaneously implementing the excitation and the measurement. The measurement system has accelerometers, PCB Corp. 393B12 model was installed at the center of the 2nd 5th floor and KYOWA Corp. AS-2GB model was attached to the HMD. PCB Monitran and KYOWA DPM-711B was used to amplify their measured signals. The data cabling connection system has BNC cable with lengths of 25m and 50m. In the data acquisition system, the DAQ board of NI DAQCard-6036E with 16bit-range was used to perform the AD and DA conversion. Also, it was connected to the signal conditioner of NI SCC-2345 by using both the input modules and the output module. In this excitation system, both excitation and measurement signals are voltage signals, and this excitation signal transfers equivalent thrust generated according to the input voltage signal through the inverter of HMD, to the mass of HMD. Accordingly, the excitation signal should be generated to move within the safety range, because the excessive response of the HMD is prevented by its safety device.

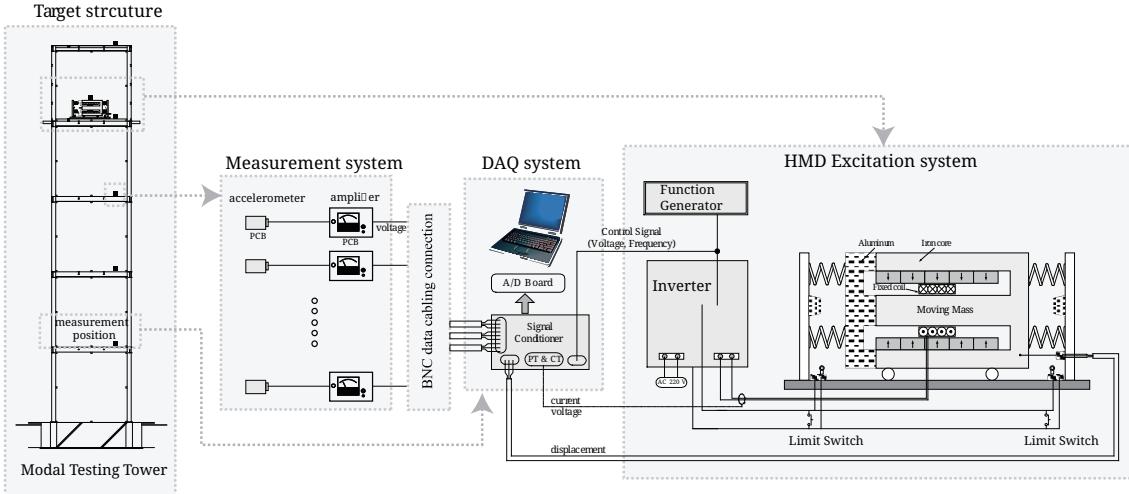


Figure 3.18: Schematic diagram of the field measurement, data acquisition and excitation system.



Figure 3.19: Installation pictures of the measurement, data acquisition and excitation system.

### 3.2.4 System Identification

#### White-noise Test

White noise vibration test was carried out by using broad-band random signals during 410sec. Both the excitation and measurement were performed by constructing a close-loop system to minimize the latency, by which time delay would be induced, and experimental data were acquired with a sampling rate of 100Hz. In order to reduce the unexpected noise in the experiment, a low-pass filter of 30Hz within the amplifier and 25Hz within the signal conditioner module was utilized, and the acquisition system was insulated with the reference signal ended (RSE) method for its grounding. Figure 3.20 shows the transfer function from the absolute acceleration of the HMD to the building accelerations. The lower fundamental modes of the structure are apparently shown.

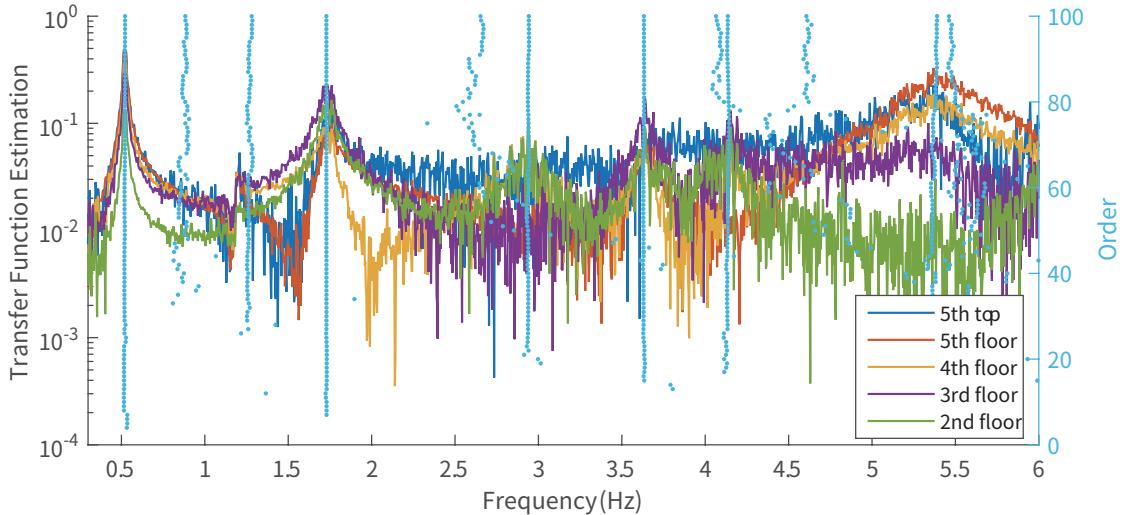


Figure 3.20: The transfer function from the absolute acceleration of the HMD to those of the structure.

Table 3.4: Identified natural frequencies and damping ratios

Modes	Frequency (Hz) & COV (%)	Damping ratio (%) & COV (%)
1	0.52(1.82)	1.46(14.42)
2	1.73(0.13)	2.71(9.40)
3	2.94(0.10)	3.54(1.93)
4	4.14(0.08)	1.72(1.70)
5	5.36(0.23)	3.84(3.56)

### Finite Model Updating

It is difficult to establish the FE model because of limitations in the number of sensors that may be deployed and the difficulty of actuating a structure to higher modes. The process of developing an FE model of a structure relating it to the experimental model is called FE model updating(Bagchi, 2005). In this chapter, an analysis model was established by modeling using ANSYS and model updating was carried out by using the measured modal data.

In this thesis, the optimal matrix update method was used which is based on a constraint optimization problem where the minimum changes in the mass matrix or the stiffness matrix are found subject to constraints such as symmetry, connectivity, and definiteness(Baruch, 1978; Baruch and Bar Itzhack, 1979). The mass matrix is assumed to be exact. There exists a constraint on the corrected stiffness matrix which should reproduce the measured modal data with symmetry. The two independent constraint equations are Eqs. (3.12) and (3.13). The function to be minimized must relate in some way to the difference between the corrected stiffness matrix,  $\mathbf{K}$ , and the analytically derived stiffness

matrix,  $\mathbf{K}_a$ , which is shown in the form of the norm in Eq. (3.14). The expression for the updated mode and stiffness matrix, which is obtained by minimizing Eq. (3.14) satisfying Eqs. (3.12) and (3.13), lead to Eqs. (3.15) and (3.16)(Baruch and Bar Itzhack, 1979; Baruch, 1978).

$$\mathbf{K}\Phi = \mathbf{M}_a\Phi\Lambda, \Phi^\top\mathbf{M}_a\Phi = \mathbf{I} \quad (3.12)$$

$$\mathbf{K}^\top = \mathbf{K} \quad (3.13)$$

$$J = \frac{1}{2} \left\| \mathbf{N}^{-1} (\mathbf{K} - \mathbf{K}_a)^{-1} \right\| \quad (3.14)$$

$$\Phi = \Phi_m \left[ \Phi_m^\top \mathbf{M}_a \Phi_m \right]^{-1/2} \quad (3.15)$$

$$\begin{aligned} \mathbf{K} &= \mathbf{K}_a - \mathbf{K}_a \Phi \Phi^\top \mathbf{M}_a - \mathbf{M}_a \Phi \Phi^\top \mathbf{K}_a \\ &\quad + \mathbf{M}_a \Phi \Phi^\top \mathbf{K}_a \Phi \Phi^\top \mathbf{M}_a + \mathbf{M}_a \Phi \Lambda \Phi^\top \mathbf{M}_a \end{aligned} \quad (3.16)$$

where,  $\mathbf{M}_a$  and  $\mathbf{K}_a$  are the analytically derived mass matrix and stiffness matrix, respectively,  $\Phi_m$  is the measured eigenvector matrix,  $\Lambda$  is a diagonal matrix of the measured eigenvalues and  $\mathbf{N}$  is the appearance of the square root of the mass matrix  $\mathbf{N} = \mathbf{M}_a^{1/2}$ .

In this thesis, as shown in Figure 3.20, the measured eigenvector is extracted from the measured modal data using respectively accurate the first and the second modal information. When the measured eigenvector matrix would be extracted, it could be derived from the transfer function in case of modal resonance using Eq. (3.17) and the sign of the eigenvector are determined from the phase of the transfer function. Reinhorn et al. assumed that resonance responses of the each mode are dependent in case the modal frequencies are not adjacent each other and proposed equation of the transfer function, which is expressed as

$$T_{ai}(\omega_k) = \phi_{ik} H_{ik}(\omega_k) \Gamma_k \quad (3.17)$$

where,  $T_{ai}(\omega_k)$ ,  $H_{ik}(\omega_k)$  are the transfer functions to the  $i$ th floor and the  $k$ th mode of structures in case of resonance and not,  $\Gamma_k$  is the  $k$ th scalar value of  $\Gamma = -\Phi^\top \mathbf{M}_a \mathbf{I}$  and  $\mathbf{I}$  is the influence vector of HMD load. The  $k$ th value of eigenvector by ratio of the  $i$ th floor over the  $j$ th floor can be calculated from amplitude ratio of the transfer function to absolute acceleration responses, which is expressed as

$$\phi_{ik}/\phi_{jk} = T_{ai}(\omega_k)/T_{aj}(\omega_k) \quad (3.18)$$

Table 3.5: Natural frequencies (Hz) for the modal testing tower

	1st mode	2nd mode	3rd mode	4th mode	5th mode
initial	0.5022	1.5623	2.74	3.91	4.83
measured	0.5249	1.7578	2.94	3.67	5.38
updated	0.5249	1.7578	2.95	3.67	5.38

By updating the natural frequencies and modes, the exact results are obtained comparing to those of the initial analysis.

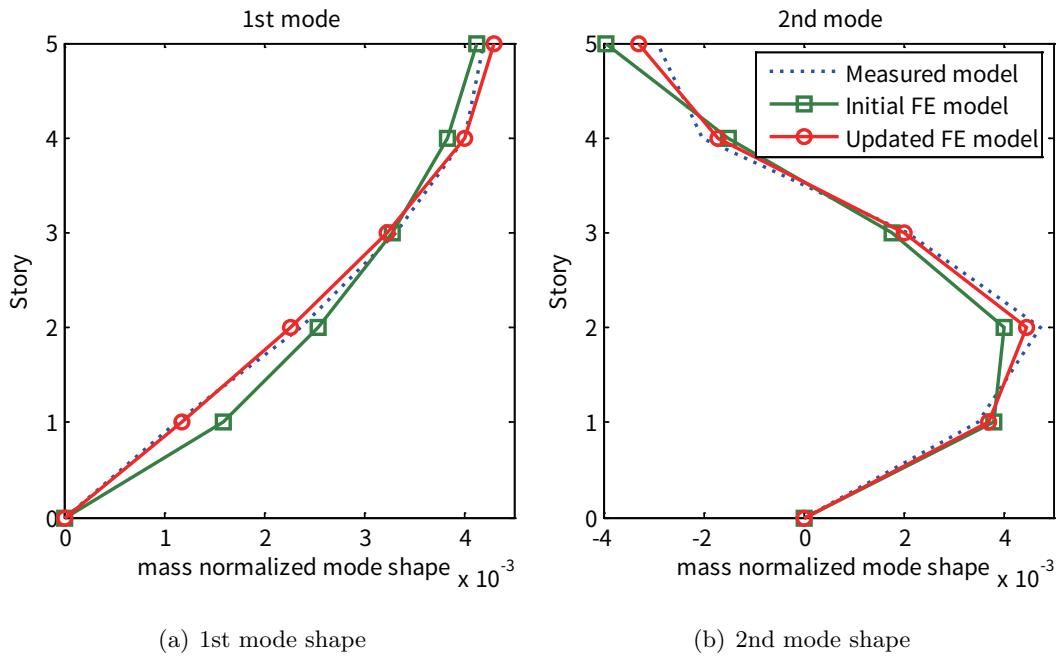


Figure 3.21: The mode shape comparison of initial, the measured and the updated FE models.

### 3.2.5 Design Controller of an Excitation System for Simulating Earthquake Response

#### Generating Input Signal of HMD

In this chapter, the process of generating the HMD input signal which is simulating earthquake response in the elastic range of model structure by HMD is introduced. The state-space form equation of a structure excited by the base acceleration  $\ddot{u}_b$  and HMD acceleration  $\ddot{u}_h$  with size of  $r$  is expressed as follows

$$\begin{aligned} \dot{\mathbf{z}} &= \mathbf{A}\mathbf{z} + \mathbf{B}_b\ddot{u}_b + \mathbf{B}_h\ddot{u}_h \\ \mathbf{y} &= \mathbf{C}\mathbf{z} + \mathbf{D}_b\ddot{u}_b + \mathbf{D}_h\ddot{u}_h \end{aligned} \quad (3.19)$$

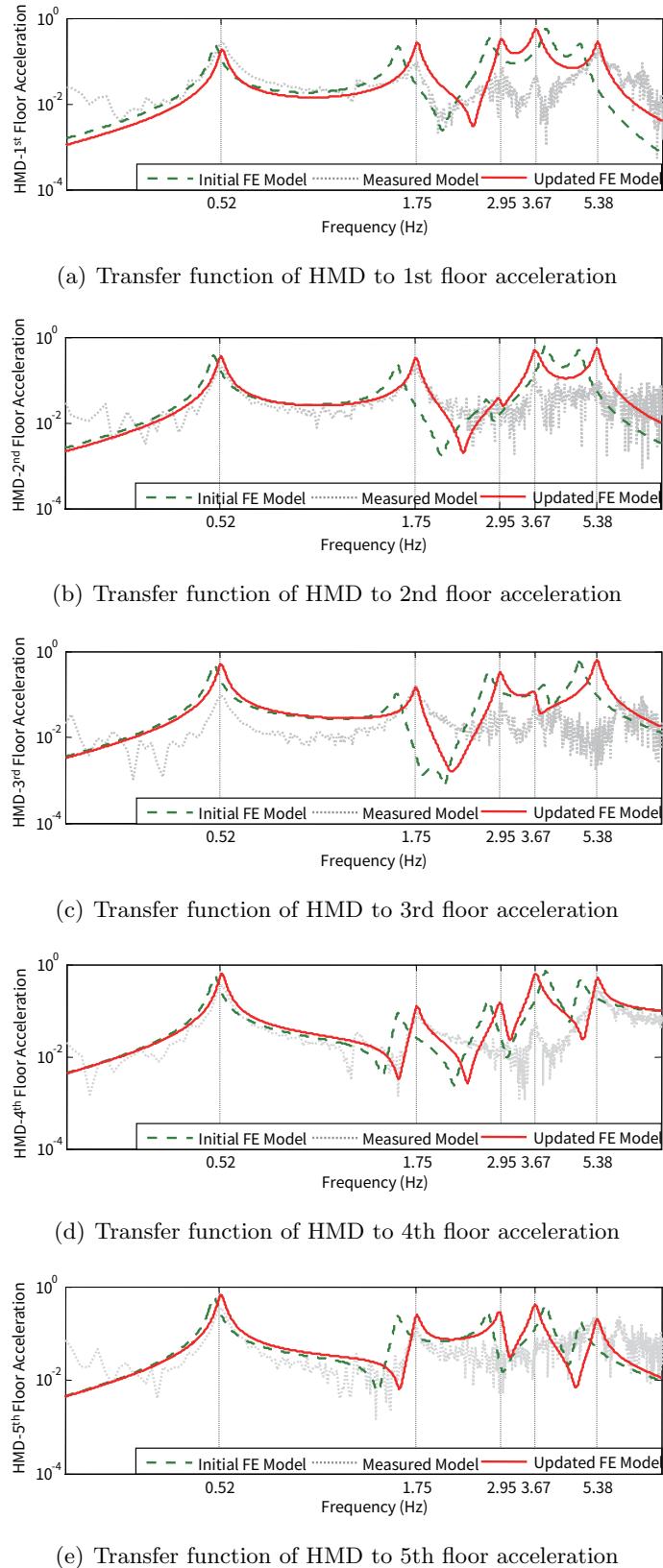


Figure 3.22: The transfer function comparison of the initial, measured, updated FE model.

where  $\mathbf{z}$  is state variable vector size of  $2n \times 1$  and  $\mathbf{y}$  is output vector with size of  $m$ . The output transfer functions to  $\ddot{u}_b$  or  $\ddot{u}_h$  are given by

$$\begin{aligned}\mathbf{T}_h &= \mathbf{Y}_h(s)\mathbf{U}_h(s)^{-1} = \mathbf{C}(s\mathbf{I} - \mathbf{A})^{-1}\mathbf{B}_h + \mathbf{D}_h \\ \mathbf{T}_b &= \mathbf{Y}_b(s)\mathbf{U}_b(s)^{-1} = \mathbf{C}(s\mathbf{I} - \mathbf{A})^{-1}\mathbf{B}_b\end{aligned}\quad (3.20)$$

where the scalar  $s$  is a complex variable  $j\omega$ . The inverse of  $\mathbf{T}_h$  exists only if  $r$  equals to  $m$  and the Laplace transform of  $\ddot{u}_h$  providing the identical output to the base acceleration induced one is determined as

$$\mathbf{U}_h(s) = \mathbf{T}_h^{-1}\mathbf{Y}_h(s) = \mathbf{T}_h^{-1}\mathbf{Y}_b(s) = \mathbf{T}_h^{-1}\mathbf{T}_b\mathbf{U}_b(s) \quad (3.21)$$

when  $r$  is smaller than  $m$ , the number of structural responses which can be modulated by  $\ddot{u}_h$  is restricted to  $r$ , and target structural response should be selected. The Laplace transform of input realizing the target response  $\bar{y}$  with size of  $r$  is

$$\hat{\mathbf{U}}_h(s) = \hat{\mathbf{T}}_h^{-1}\hat{\mathbf{Y}}_h(s) = \hat{\mathbf{T}}_h^{-1}\hat{\mathbf{Y}}_b(s) = \hat{\mathbf{T}}_h^{-1}\hat{\mathbf{T}}_b\hat{\mathbf{U}}_b(s) \quad (3.22)$$

where,  $\hat{\mathbf{T}}_h^{-1}$  is a sub-matrices of  $\mathbf{T}_h^{-1}$ .  $\hat{\mathbf{T}}_h^{-1}$  is constructed by extracting the columns in  $\mathbf{T}_h^{-1}$  corresponding to the target response.

### Design of HMD controller

The excitation force is generated by the HMD to vibrate the structure in the elastic range. In order to control the HMD, the linear oscillating actuator (LOA) using electromagnetic force as an exciter was adopted. The mass of HMD is 1,500kg mass mounted in the 5th floor of the structure. The excitation force is generated by not the HMD acceleration of input voltage signal but 5th-floor absolute acceleration. Therefore, in order to compensate the distortion of the actual HMD acceleration against the HMD dynamics existing between the reference signal and the actual measured acceleration of the HMD, the inverse transfer function of the actual acceleration of HMD with respect to the command signal generated by the control computer is constructed and implemented in the HMD control computer as shown in Figure 3.24. First, the inverse transfer function, of which amplitude and phase are represented in Figure 3.25 by the dashed line, is obtained experimentally. Then, the experimental inverse transfer function is approximated by a rational function for its implementation in the control computer. In this verification experiment, the inverse transfer function of the HMD is approximated using the command `invfreqs` in MATLAB(Coleman et al., 1999), which adopts the damped Gauss-Newton method for the iterative search to minimize the sum of the squared error between the measured and the desired frequency response points(Dennis Jr and Schnabel, 1983). The approximation

result is given by the following linear filter and compared with the experimental one in Figure 3.25.



Figure 3.23: Photograph of the HMD installed in 5th floor.

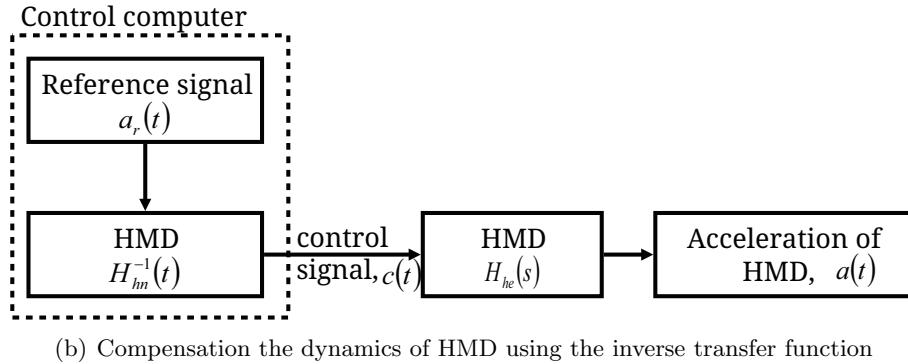
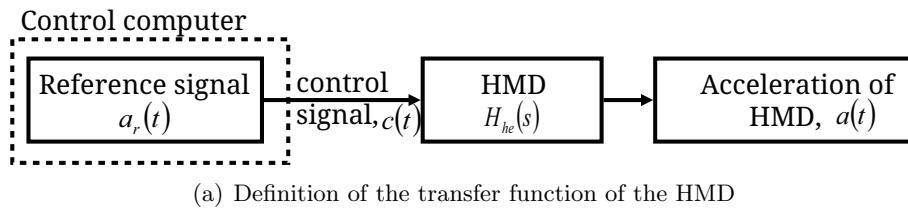


Figure 3.24: Schematic diagram of the HMD controller.

### 3.2.6 Experimental results

#### Pseudo-earthquake excitation result

The pseudo-earthquake forced vibration testing is implemented. The earthquake El Centro(1940, NS) and Hachinohe(1968, NS) earthquake acceleration which PGAs were scaled down to 0.05g for the safety limit device were used in this thesis. The pseudo-earthquake

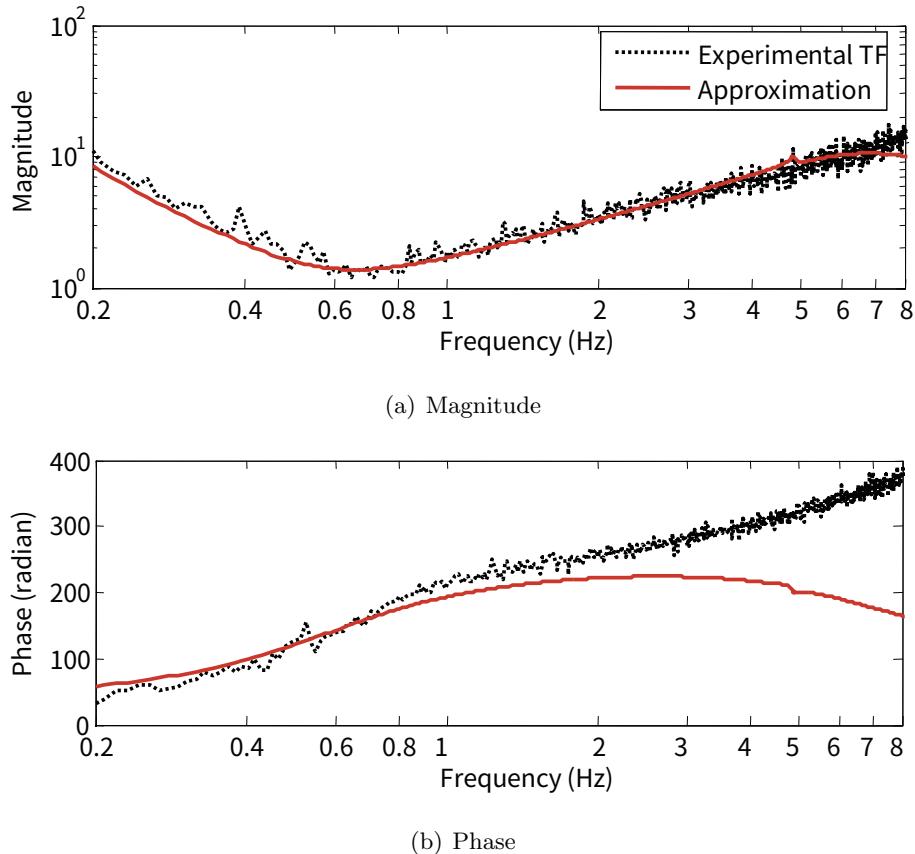


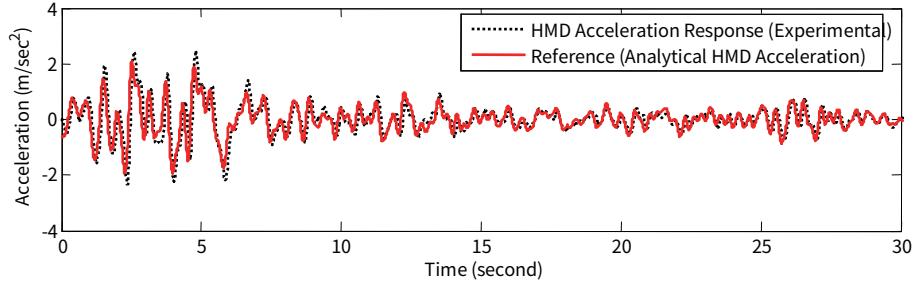
Figure 3.25: Comparison of the measured and the approximated inverse transfer function of the HMD.

excitation signal was generated by the inverse transfer function of structure introduced in the chapter 3.2.5 corresponding to the target responses and the excitation was implemented through the HMD controller introduced in the chapter 3.2.5.

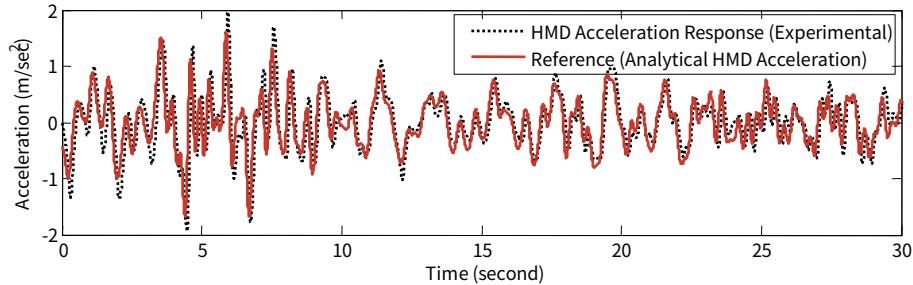
Figure 3.27 compares the result in time history of El Centro earthquake response in the analysis and experimental models. It is observed that both results from FE model and experimental one coincides well each other in the initial time range, but the phase in post time range shows a tendency to be changed over 90 degrees. It is caused that the constraint in curve fitting the transfer function was considered weightily with the amplitude of the response, not the phase since in the seismic performance of the structure the amplitude was laid more weight on than the phase.

Figure 3.28 compares the result in the frequency domain of El Centro earthquake response in analysis model and experimental one. It is observed that both of results from FE model and experimental one coincides well each other, respectively

Figure 3.29 compares the result in the time domain of Hachinohe earthquake response in analysis model and experimental one. It is observed that both of results from FE model and experimental one are good agreement and demonstrate the effectiveness proposed

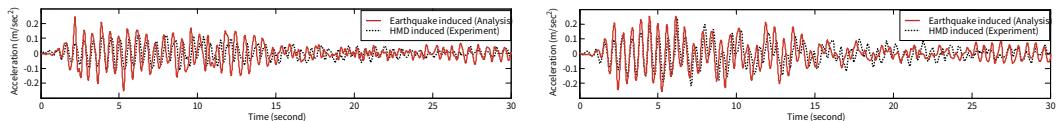


(a) Acceleration of HMD excited by El Centro earthquake

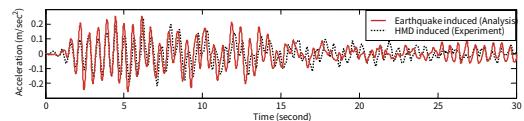


(b) Acceleration of HMD excited by Hachinohe earthquake

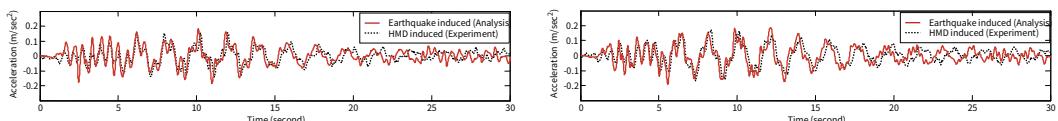
Figure 3.26: Comparison of the reference and measured acceleration time histories when the signals were generated by El Centro and Hachinohe earthquake.



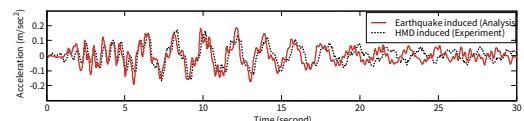
(a) 1st floor acceleration of El Centro earthquake excitation



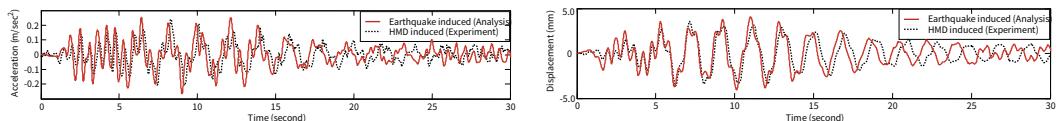
(b) 2nd floor acceleration of El Centro earthquake excitation



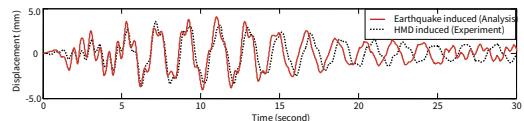
(c) 3rd floor acceleration of El Centro earthquake excitation



(d) 4th floor acceleration of El Centro earthquake excitation



(e) 5th floor acceleration of El Centro earthquake excitation



(f) 1st floor displacement of El Centro earthquake excitation

Figure 3.27: Time history comparison of El Centro earthquake response in the analysis and experimental models. (when the target response is the 5th-floor acceleration).

pseudo-earthquake forced vibration testing method. The discrepancy in amplitude at the second mode frequency is caused by the difficulty of excitation a structure to higher modes limits the modal testing method to capturing only a few lower modes. In order to

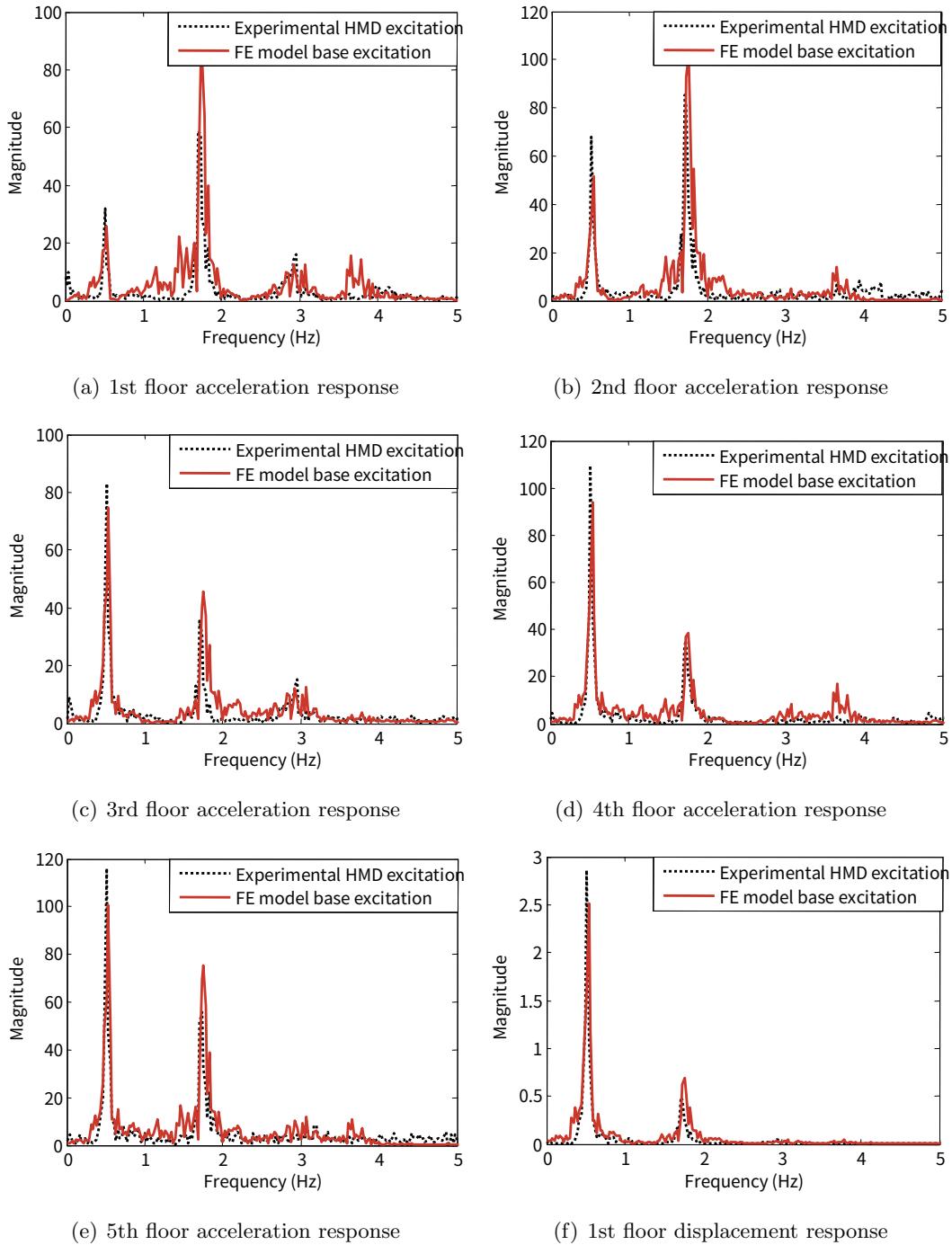


Figure 3.28: Comparison in frequency domain of El Centro earthquake response of the analysis and experimental models. (when the target response is the 5th floor acceleration).

improve this phenomenon, it should be designed to be able to excite higher mode which has fundamental frequencies of a structure.

Figure 3.30 compares the result in the frequency domain of Hachinohe earthquake response in analysis model and experimental one. It is observed that the coincidence of the response at the first mode and second mode frequency was shown, but it would not

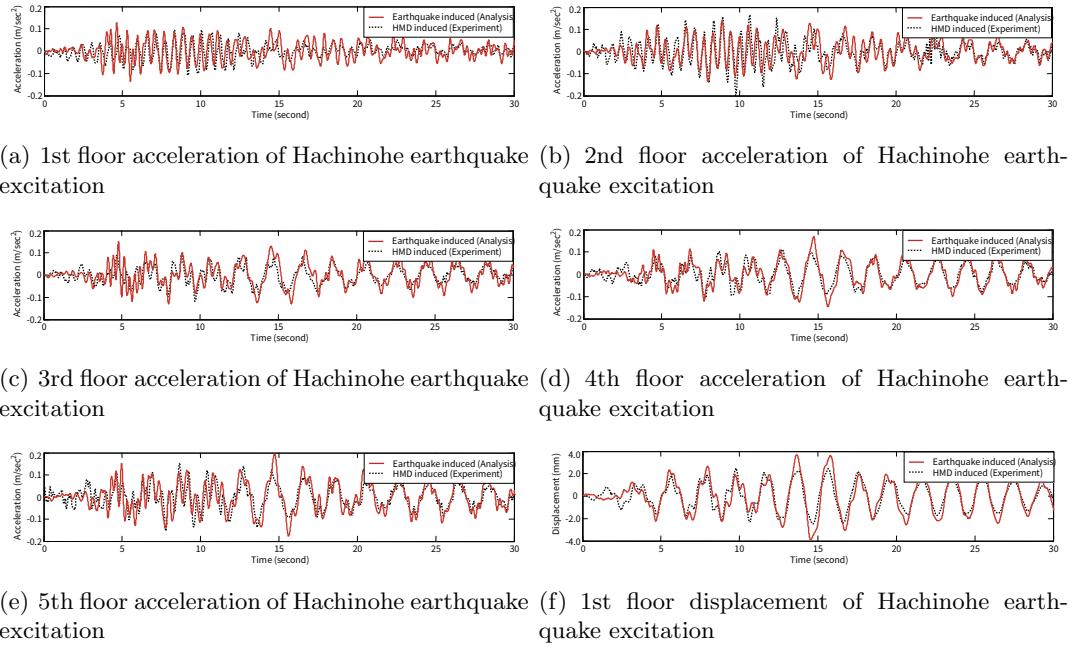


Figure 3.29: Time history comparison of Hachinohe earthquake response in the analysis and experimental models. (when the target response is the 5th-floor acceleration).

reflect the responses at the third mode or higher mode frequency in Figures 3.30(a)-3.30(c).

### 3.2.7 Error Analysis

The normalized RMS error between the absolute acceleration induced by HMD and the base acceleration could be express as Eq. (3.23). Figure 3.31 is shown the floor distribute errors corresponding to the target responses of the analysis FE model. All responses are contained 10-20% error values and the minimal error value presented when the target response consider the 5th-floor acceleration.

$$\text{Normalized RMS Error}(\%) \epsilon = \frac{\sum_{t=0}^T \left\{ \sqrt{\ddot{y}_g(t)^2} \right\} - \sum_{t=0}^T \left\{ \sqrt{\ddot{y}_h(t)^2} \right\}}{\sum_{t=0}^T \left\{ \sqrt{\ddot{y}_g(t)^2} \right\}} \quad (3.23)$$

where  $\ddot{y}_g$ ,  $\ddot{y}_h$  are the structural acceleration responses which are the base acceleration and the HMD acceleration induced.  $T$  is total excitation time of the earthquakes.

Figure 3.32 shows the floor distribute errors of the experimentally measured response corresponding to the target responses. All responses are contained 20-30% error values when the HMD induced.

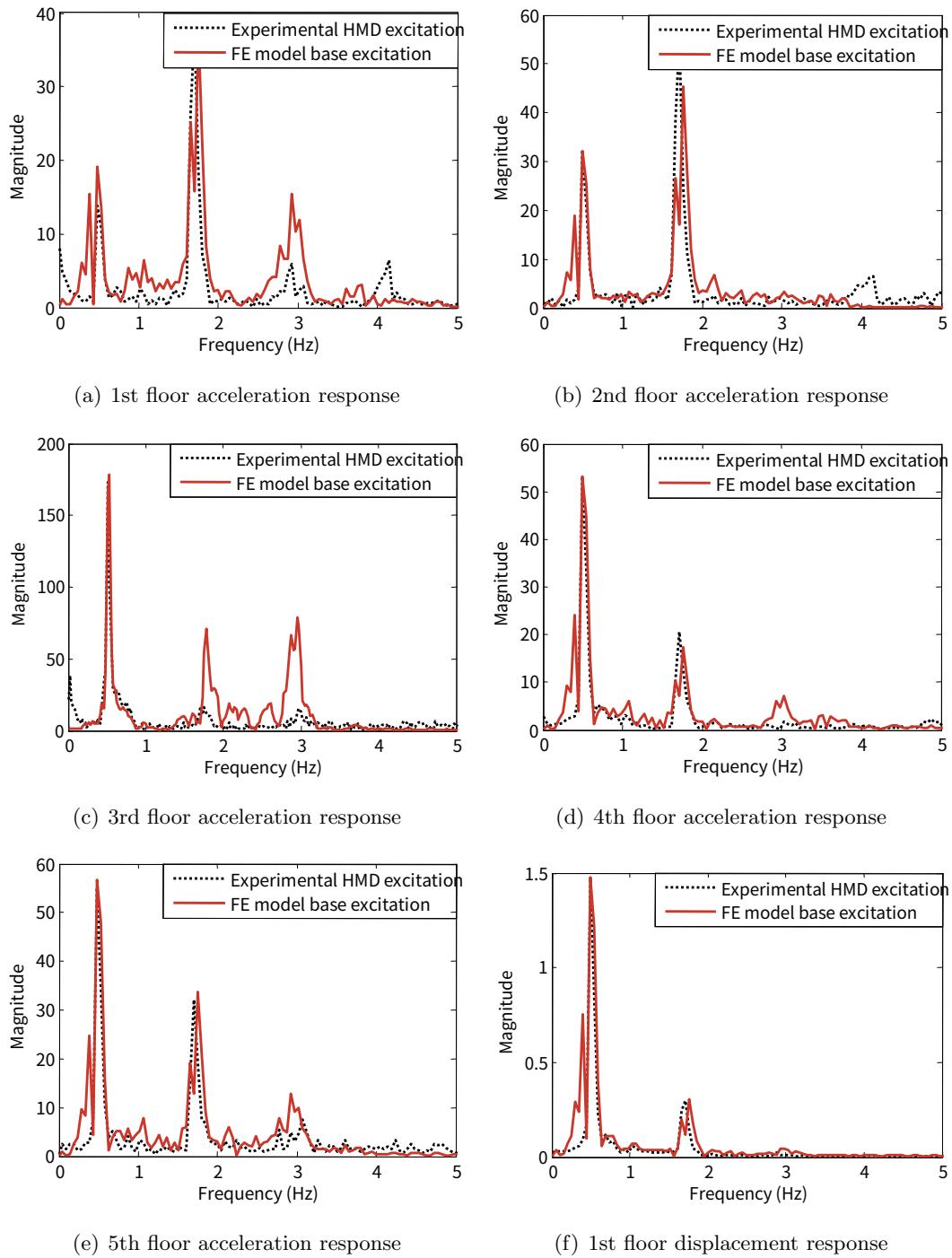


Figure 3.30: Comparison in frequency domain of Hachinohe earthquake response in the analysis and experimental models. (when the target response is the 5th-floor acceleration).

### 3.2.8 Summary

In this section, field measurement system of a full-scale structure and excitation system were established and then forced vibration test was carried out using the HMD to simulate seismic load. System identification of the full-scale structure was carried out through

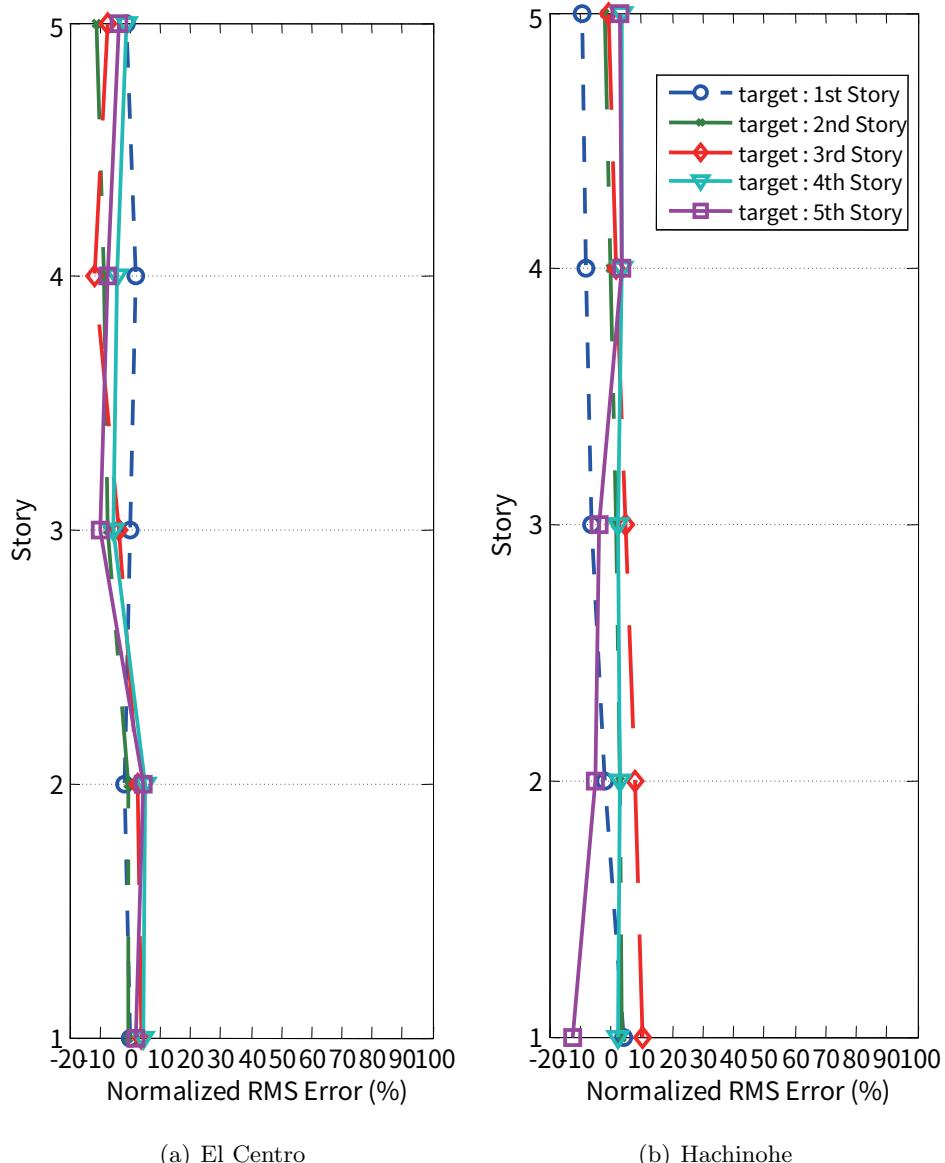


Figure 3.31: Normalized RMS error floor distribution of the analysis FE model corresponding the distributed floor acceleration response.

white noise test, and finite element model is updated. The seismic excitation system was accomplished through inverse transfer functions of structure and HMD by the system identifications. The propriety of seismic excitation system was verified by a comparison between the seismic numerical analysis results of finite element model and the experimental results of HMD excitation.

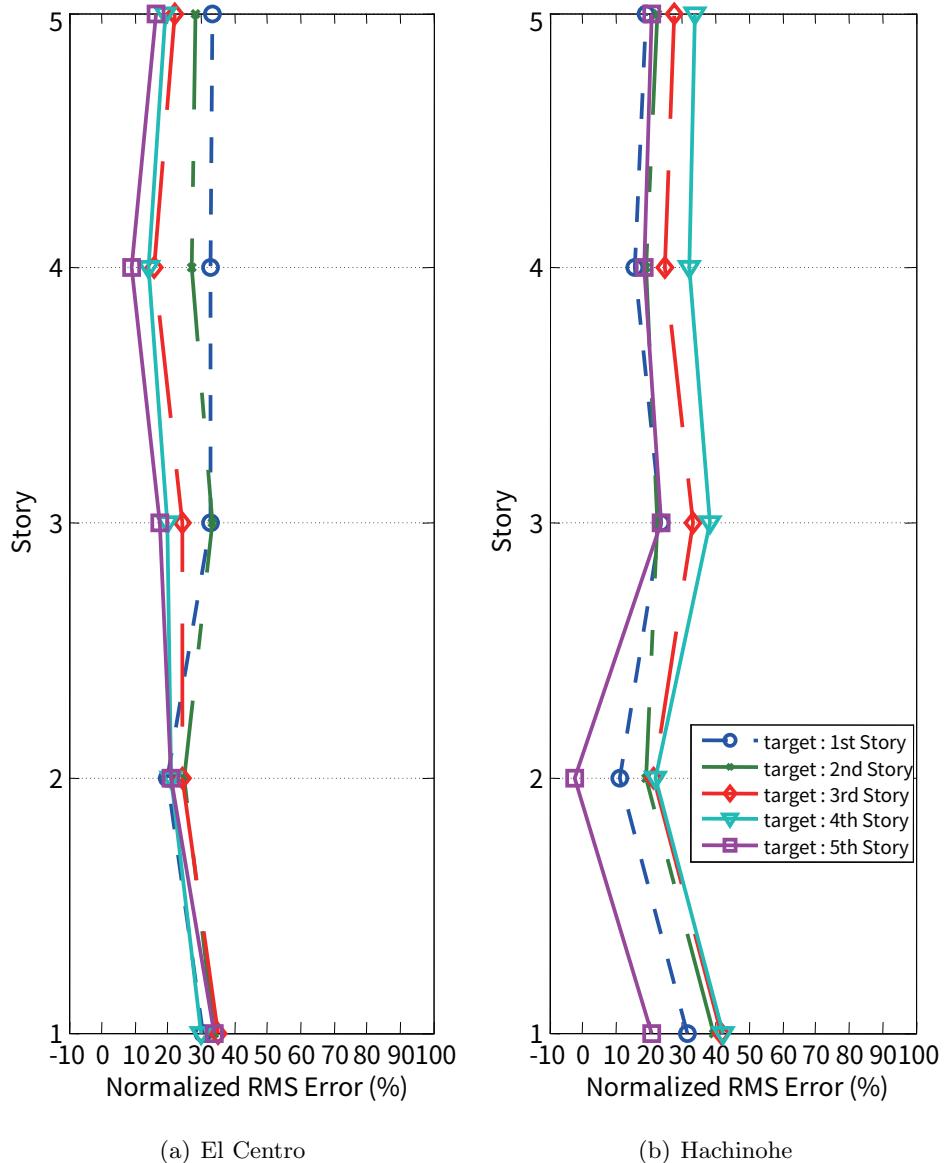


Figure 3.32: Normalized RMS error floor distribution of the experimental results corresponding the distributed floor acceleration response.

### 3.3 Forced vibration test of MR Damper-based Semiactive Control Algorithms for Full-scaled Building

#### 3.3.1 Overview

In this section, the effectiveness of the MR damper-based semiactive control systems for seismic protection of a full-scale five-story steel frame building structure is experimentally verified, when some semi-active control algorithms such as the clipped-optimal control algorithm (CO), the maximum energy dissipation algorithm (MEDA), the Lyapunov stability theory based control algorithm (LYAP) and the neuro-control algorithm are considered.

This may be the first experimental investigation of several semi-active control algorithms using a full-scale test structure. In the experiment, two MR dampers are attached between the ground and the first floor. As described in the previous section, the pseudo-earthquake testing method is used to excite the building structure as if it is subjected to earthquake loading. This method uses filters that modify the displacement of the HMD using both transfer function in the frequency domain and the least squares approximation in the time domain. All the semi-active control algorithms are evaluated under four historical earthquakes (El Centro, Hachinohe, Kobe, and Northridge earthquakes) and one filtered artificial earthquake. The experimental results are compared with the passive optimal case which provides the specific constant voltage to the MR damper.

### 3.3.2 Full scaled MR Damper

Lee et al. (2010) had developed full-scale MR damper, an MR damper is designed by deriving a suboptimal design procedure considering optimization problem and magnetic analysis, and a damper with the capacity of 1.0ton is manufactured.

Table 3.6: Results obtained from the magnetic analysis.

No. of stage	3	3	4
Flux density in steel	< 1.5T	> 1.5T	< 1.5T
Magnetic field intensity	< 150 kAmp/m	>150 kAmp/m	< 150 kAmp/m
Coil turns	20AWG/200 turns	20AWG/230 turns	21AWG/200 turns
Input current	3A	3A	3A

In order to use the designed MR dampers in the experiment, two MR dampers are manufactured as shown in Figure 3.33(a) and 3.33(b) and tested with harmonic dynamic displacements applied by a servo-hydraulic testing system. The result of variable current tests is shown in Figure 3.33(c). As shown in Figure 3.33(c), the total damper force at design velocity (60mm/s) is about 10kN. Among the total force, the uncontrollable force is about 2kN or more and controllable force approximately 8kN or less.

### 3.3.3 Semi-active control algorithms

Semi-active control algorithms determine the most appropriate voltage to change viscous characteristics of the MR damper from the structural response measurements. Since the MR damper force is related to not only command voltage but also the relative velocity and displacement between both ends of the damper, the MR dampers cannot be directly controlled by the command voltage. So far various semi-active control algorithms have been proposed for control of MR dampers (Jansen and Dyke, 2000). In this section, four

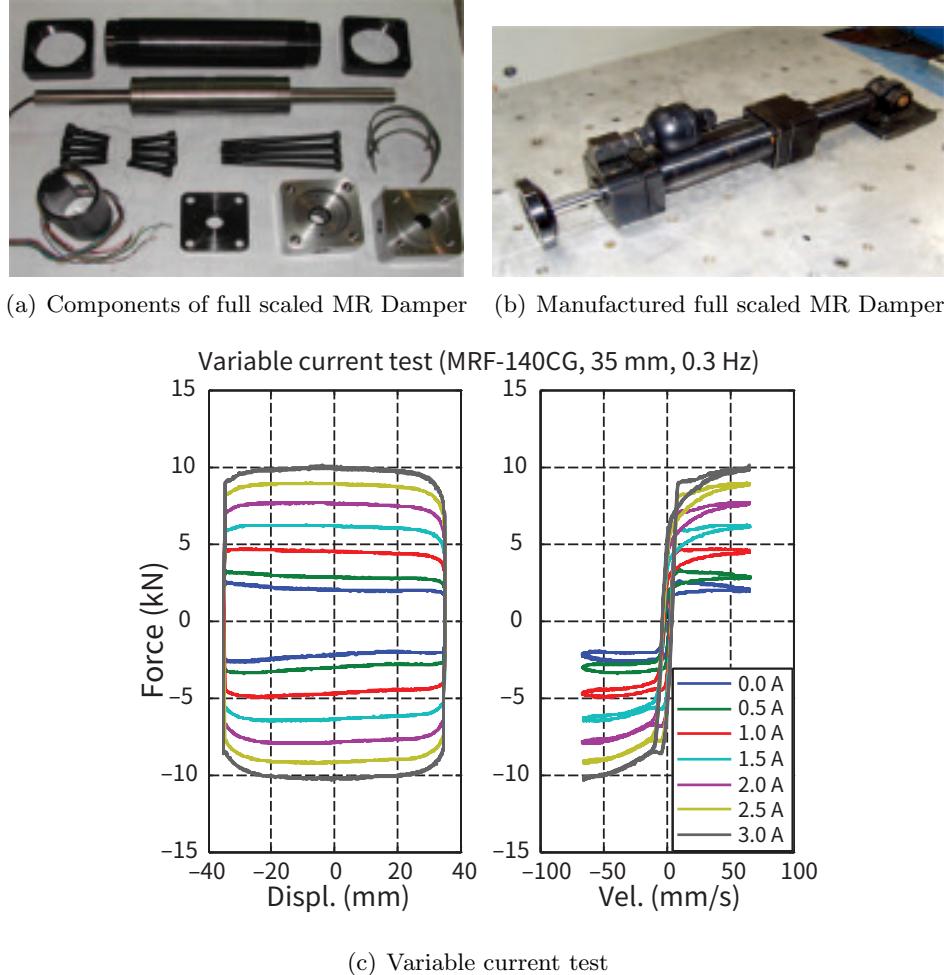


Figure 3.33: Manufactured MR damper and test result.

different control algorithms are considered; CO, LYAP, MEDA, and cost function-based SNC.

### 3.3.4 External excitation loads

In order to evaluate the performance of the MR damper-based semi-active control algorithms and passive control system, a suite of historical ground motions that are intended to encompass both moderate events and severe events is considered as external excitation load. Moreover, a filtered artificial earthquake (Kanai-Tajimi filter) is also adopted to verify the effectiveness of the control algorithms. Each historical ground motion and the artificial earthquake are scaled down in order to bound HMD by limited maximum displacement for safety. Each ground motions used in this investigation are as follows and the acceleration of HMD and its power spectral density are shown in Figure 3.34 and 3.35.

### Moderate events

- El Centro : N-S component of the 1940 Imperial valley, California earthquake (magnitude 7.1) recorded at Imperial Valley Irrigation District substation in El Centro, California, USA (scaled down to 0.303g);
- Hachinohe : N-S component of the 1968 Takochioki(Hachinohe) earthquake (magnitude 7.9) recorded at Hachinohe, Japan (scaled down to 0.265g);

### Severe events

- Kobe: N-S component of the 1995 Hyogo-ken Nanbu(Kobe) earthquake (magnitude 7.2) recorded at Kobe Japanese Meteorological Agency (JMA), Kobe, Japan (scaled down to 0.318g);
- Northridge: N-S component of the 1994 Northridge earthquake (magnitude 6.8) recorded at the Sylmarcounty Hospital parking lot in Sylmar, California, USA (scaled down to 0.227g);

### Artificial eartuquake

- Kanai-Tajimi filtered artificial earthquake (PGA:0.105g)

### 3.3.5 Measurement Devices and Feedback Control System

Measurement devices used in this experiment are displacement sensors, accelerometers, load cells, and a current probe. The displacement sensors are used in measuring the relative displacement between ground and the first floor, which is the displacement of MR damper's both ends. When the MR dampers are installed, two LVDT type potentiometers (Midori America LP-19FB) are used. To the contrary, two laser displacement sensors (Optex CD4-350D) are used, when the MR dampers are not installed. Each floor's acceleration is measured by five accelerometers (PCB 393A03) and that of LAMD by Kyowa AS-2GB accelerometer. The load cells (CAS load cell LS-2T) measuring the MR damper control force are installed between the MR damper and the first floor. The current probes (Tektronix A622) measuring the current into each MR damper are also installed along the MR damper's power cable.

The measured data from the above measurement devices are collected and digitalized through a data acquisition system (NI DAQ-card 6062E). The collected data is selected for each semi-active control algorithm, and finally, determined command voltage is sent to a current generator, which converts the voltage to the corresponding current. Then the MR damper can control the building structure with the generated current.

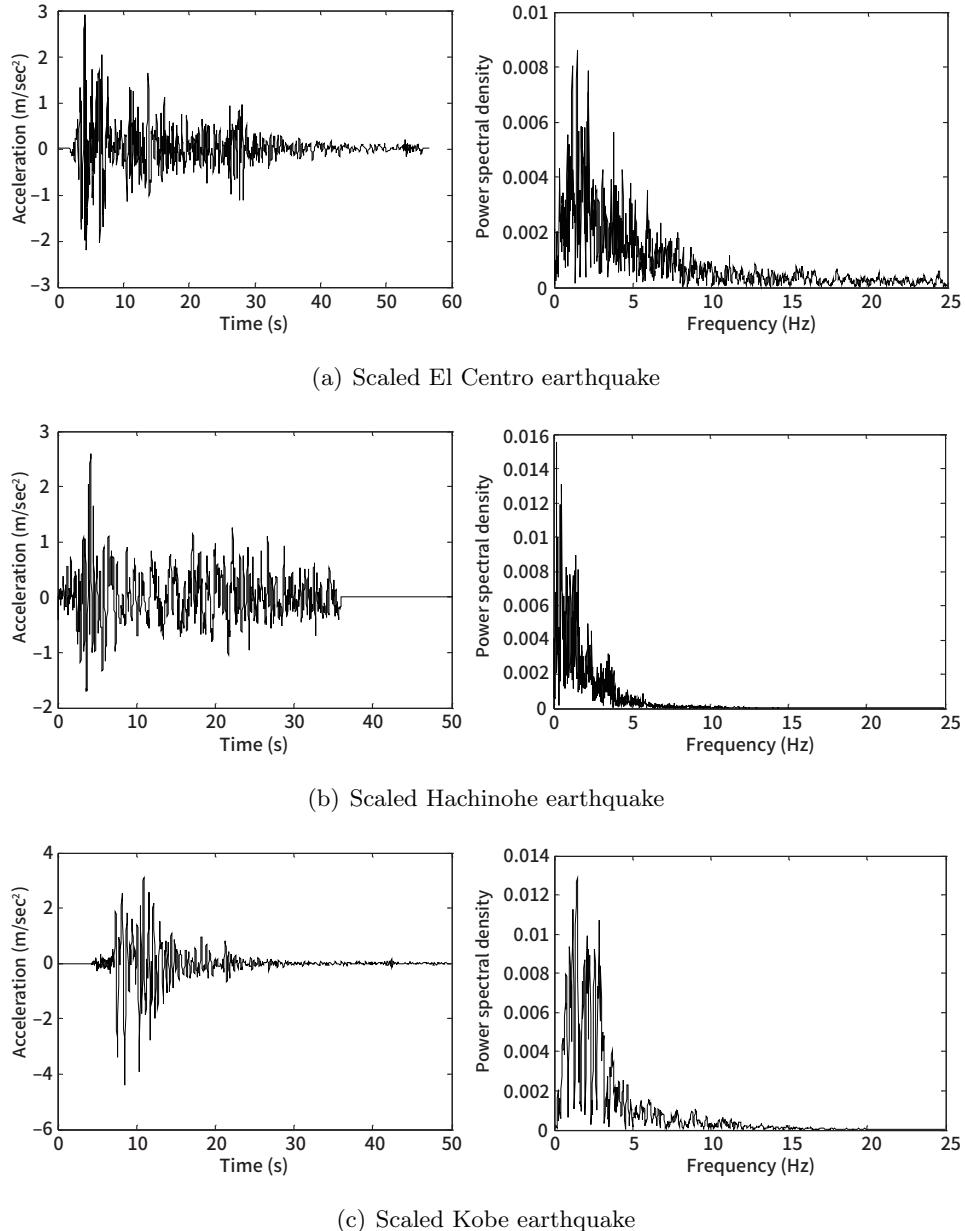


Figure 3.34: Acceleration of HMD and PSD used as external excitation load.

Finally, the MR damper controlled semi-active feedback system, which has 12 inputs and two outputs, can be formed as shown in Figure 3.3.5.

The CO, as well as the cost function-based SNC algorithm, needs full state or the particular state of the structural system to calculate the appropriate desired control force. The control implementation requires full number of sensors for all DOFs. Therefore the Kalman filters as an observer for state estimation is desirable. Considered a controllable and observable system and assumed that full state estimation is required. Then the primary purpose of the general Kalman filter is constructing a state estimate  $\hat{\mathbf{x}}(t)$  that minimizes

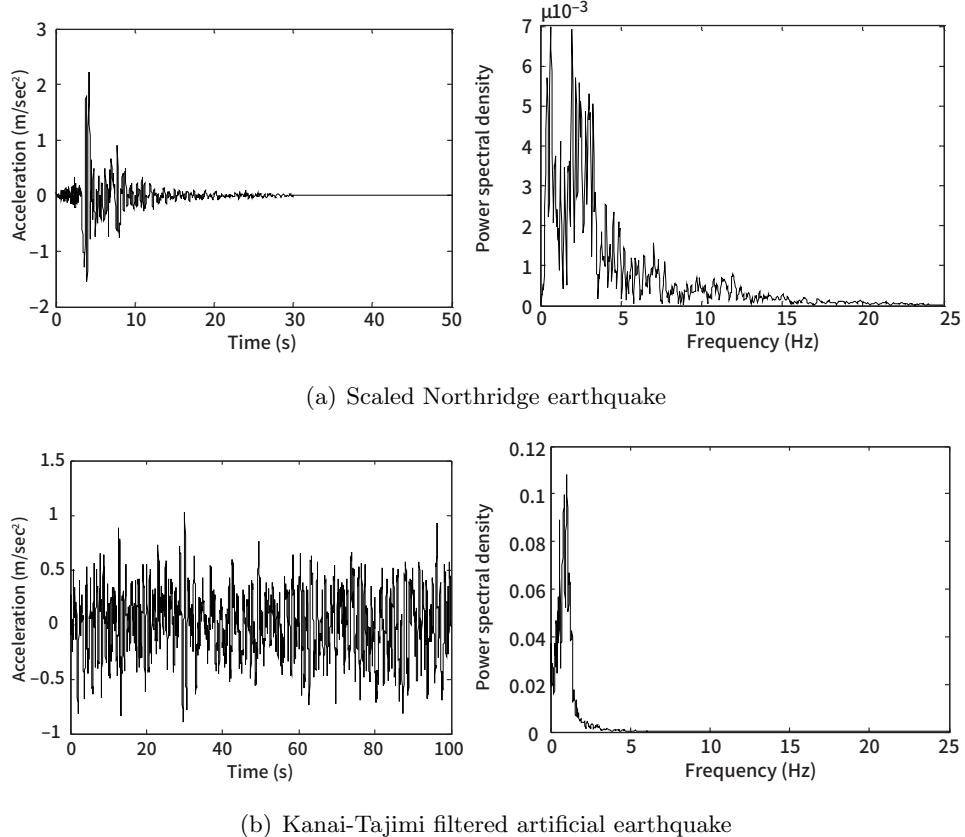


Figure 3.35: Acceleration of HMD and PSD used as external excitation load.

the steady-state error covariance which is defined by:

$$P = \lim_{t \rightarrow \infty} E \left( [\mathbf{x} - \hat{\mathbf{x}}] [\mathbf{x} - \hat{\mathbf{x}}]^T \right) \quad (3.24)$$

The optimal estimator for the state of interest is the Kalman filter which can be expressed as:

$$\begin{aligned} \hat{\mathbf{x}} &= (\mathbf{A} - \mathbf{LC}) \hat{\mathbf{x}} + \begin{bmatrix} \mathbf{L} & \mathbf{B} - \mathbf{LD} \end{bmatrix} \begin{cases} \mathbf{y} \\ \mathbf{u} \end{cases} \\ \begin{cases} \hat{\mathbf{y}} \\ \hat{\mathbf{x}} \end{cases} &= \begin{bmatrix} \mathbf{C} \\ \mathbf{I} \end{bmatrix} \hat{\mathbf{x}} + \begin{bmatrix} \mathbf{0} & \mathbf{D} \\ \mathbf{0} & \mathbf{0} \end{bmatrix} \begin{cases} \mathbf{y} \\ \mathbf{u} \end{cases} \end{aligned} \quad (3.25)$$

where filter gain  $\mathbf{L}$  is determined by solving an algebraic Riccati equation. The Kalman filter uses the known input  $\mathbf{u}$  and the measurements  $\mathbf{y}$  to generate the output and state estimates, that is,  $\hat{\mathbf{y}}$  and  $\hat{\mathbf{x}}$ . Note that  $\hat{\mathbf{y}}$  estimates the true system output without noises.

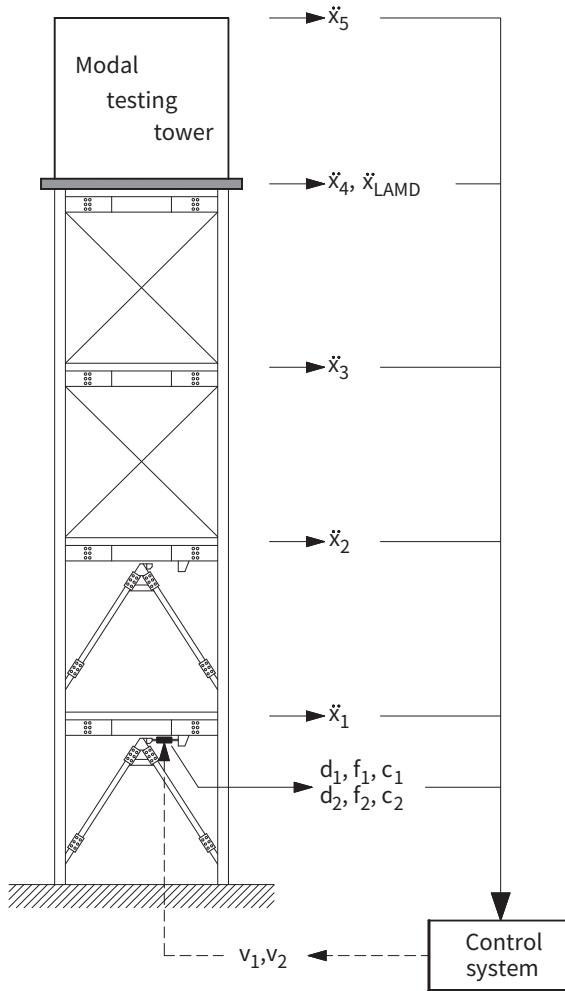


Figure 3.36: MR damper-based semi-active feed back control system.

### 3.3.6 Testing building model

All the stories in this building are 6m in height and are  $6 \times 6m^2$  in the plan. In order to investigate the modal characteristics of the building, a forced vibration test is performed using a hybrid mass damper (HMD) located on the fourth floor in the previous section. The building is excited by a white-noise signal with frequency components of 0-10 Hz, and the corresponding acceleration responses are then measured at each floor. Using these measured acceleration data, transfer functions in the frequency domain are obtained, as shown by the dotted lines in Figure 3.22. Finally, the structural damping and stiffness matrices are identified based on the experimentally obtained transfer functions, as shown by the solid lines in Figure 3.22. The estimated mass and the identified damping and stiffness matrices are given in Eqs. (3.26)-(3.28), respectively, with results of 0.52, 1.76, 2.95, 3.67, and 5.38 Hz for each mode.

$$\mathbf{M}_5 = \begin{bmatrix} 19365.5 & 0 & 0 & 0 & 0 \\ 0 & 19365.5 & 0 & 0 & 0 \\ 0 & 0 & 19365.5 & 0 & 0 \\ 0 & 0 & 0 & 19365.5 & 0 \\ 0 & 0 & 0 & 0 & 19365.5 \end{bmatrix} kg \quad (3.26)$$

$$\mathbf{C}_5 = \begin{bmatrix} 14.184 & -5.015 & 0.084 & -0.789 & 0.101 \\ -5.015 & 14.990 & -6.397 & 0.822 & -1.145 \\ 0.084 & -6.397 & 15.830 & -6.120 & -0.925 \\ -0.789 & 0.822 & -6.120 & 14.065 & -6.383 \\ 0.101 & -1.145 & -0.925 & -6.383 & 9.750 \end{bmatrix} kN \cdot s/m \quad (3.27)$$

$$\mathbf{K}_5 = \begin{bmatrix} 7473.953 & 4860.497 & 1295.606 & -908.262 & 432.162 \\ -4860.497 & 9640.722 & -6637.235 & 2446.225 & -926.951 \\ 1295.606 & -6637.235 & 10861.354 & -6005.863 & 748.607 \\ -908.262 & 2446.225 & -6005.863 & 9132.066 & -4852.783 \\ 432.162 & -926.951 & 748.607 & -4852.783 & 4543.854 \end{bmatrix} kNm \quad (3.28)$$

### 3.3.7 Experimental results

The passive control cases which provide specific constant voltage or current to the MR damper are the simplest means to operate the semiactive control using MR damper, because they need neither control algorithms nor measuring sensors. In this thesis, therefore, the passive optimal case is determined by a series of passively controlled experiments and compared with the performance of the other semi-active control algorithms.

#### Optimal Passive Case

As described above, passive control cases provide specific constant voltage or current to the MR damper. A passive-off case means a case of which voltage is always set to zero, and a passive-on case sets the voltage to the maximum level. Since the maximum current level of the manufactured MR damper used in the experiments is 3.0A, a series of passively controlled tests are conducted as increasing the current level from 0.0A to 3.0A by 0.5A to obtain the passive optimal case. Figures 3.37 and 3.38 show the normalized maximum displacement of the first floor and the normalized maximum acceleration among all the five floors.

It is demonstrated from Figure 3.37 that the performance in the maximum displacement gets better as the current input is increased. The passive-off case (i.e., 0.0A current

input) shows the worst performance. Also, the passive control cases with 1.0 or larger input current show the similar performance. As seen in Figure 3.38, however, there is no trend in the performance of the normalized maximum acceleration. Under the moderate earthquakes and the artificial earthquake, most of the normalized values in maximum acceleration beneath 1.0 which means that the controlled seismic response might be reduced compared with that of the uncontrolled case. Nevertheless, most of the normalized values in acceleration are larger than 1.0 under the severe earthquake such as Kobe and Northridge earthquake. These results indicate that the MR damper-based passive control system is more suitable in moderate earthquake region than severe earthquake region for reducing the structural acceleration. From the observation as mentioned above, the passive control case with 1.0A can be considered as the optimal passive case. The results in cases of the passive optimal (1.0A) and passive on (3.0A) cases are compared with those in the cases of the other semi-active control algorithms.

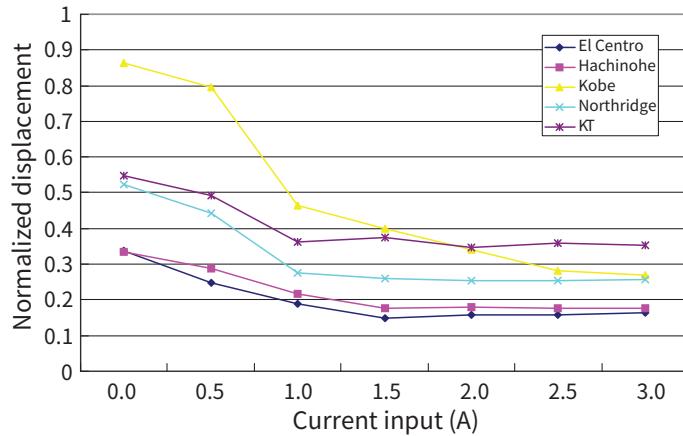


Figure 3.37: Normalized maximum displacement of passive control cases (first floor).

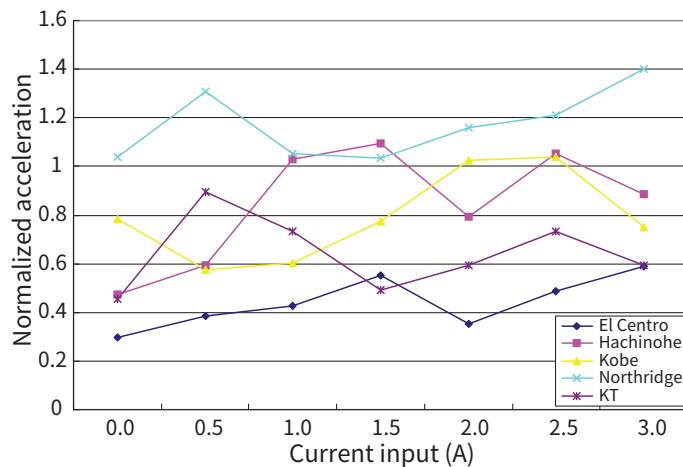


Figure 3.38: Normalized maximum acceleration of passive control cases.

### Semiactive Control Cases

After the optimal passive case is determined, the full-scale five-story steel frame building structure is excited with previously mentioned external excitation loads using LAMD and controlled with various semi-active control algorithms, such as the clipped optimal control algorithm (CO), the LYAP, the MEDA, and the SNC, using MR dampers. The responses of each control case are measured and summarized in Figures 3.39 and 3.40. Figure 3.39 shows the normalized maximum displacement of the first floor with various control algorithms and Figure 3.40 shows the normalized maximum acceleration among all the floors. From the experimental results, one can conclude as follows:

The CO shows good performance in reducing both displacement and acceleration. For severe earthquakes and artificial earthquake, however, the performance becomes slightly worse than that of moderate earthquakes. In order to obtain the better control performance from the CO, it should be designed more appropriately by altering the weighting parameters for the specific purpose of a designer. The Lyapunov algorithm and the SNC algorithm reduce the acceleration response well enough, but they cannot show the good performance in reducing displacement. In these cases, the MR dampers generate less control force than in the clipped optimal algorithm case. The Lyapunov algorithm shows the worst performance among five algorithms in reducing acceleration under the artificial earthquake, and the SNC has a drawback in reducing displacement under the severe earthquakes. Note that among all the semi-active control algorithms, the MEDA shows the best performance in reducing displacement and the worst performance in reducing acceleration. As the MEDA algorithm makes the MR dampers generate larger force than its capacity, the first floor is almost locked with the ground. This phenomenon causes a considerable increase of the acceleration response of the first floor. It is shown in some cases that maximum acceleration with the controller is larger than that of the uncontrolled case. It is considered that in the experiments, the unexpected electrical and the external environmental noises are involved in the feedback of measured responses, which deteriorates the acceleration control performance that is sensitive to those noises. The passive optimal and passive-on cases are one of the best control algorithms in reducing displacement of the structure except for the performance in reducing acceleration. The inter-story drift between the first and second floors can be larger than the displacement of the first floor when the damper force is quite large. Due to the limitation in measuring displacements, the inter-story drift of the second floor could not be measured. In order to verify the above observation, the inter-story drift of the second floor should be measured in the additional tests. The LYAP and SNC algorithms are appropriate in reducing accelerations of the structural system. On the other hand, the passive optimal and passive-on cases and the MEDA show excellent performance in reducing the first-floor displacement.

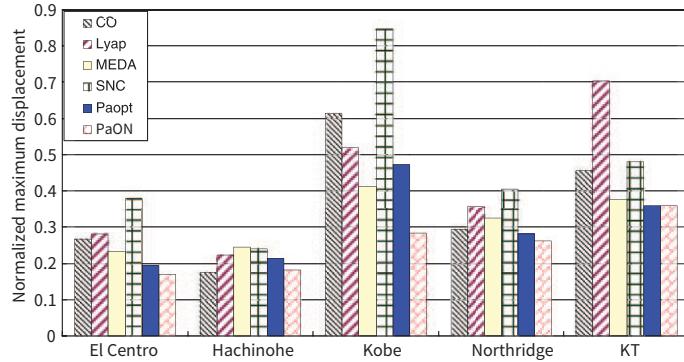


Figure 3.39: Normalized maximum displacement comparison of MR damper-based control systems.

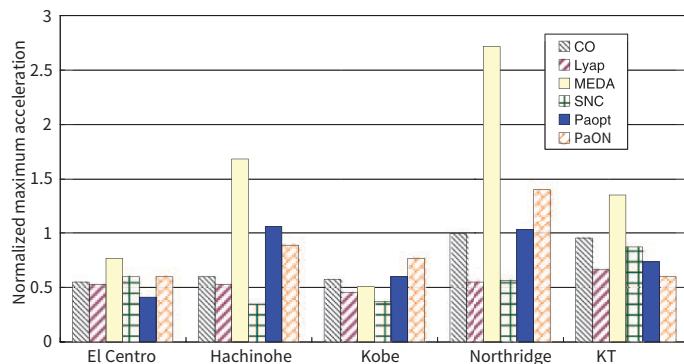


Figure 3.40: Normalized maximum acceleration comparison of MR damper-based control systems.

Figures 3.41 and 3.42 represent time history responses of displacement at first floor and acceleration at second floor, respectively. As shown in the figures, the structural responses in all the control cases are significantly reduced compared to the uncontrolled case results. Especially, some semiactive cases such as the neuro-control algorithm show the good performance in the reduction of the acceleration on the second floor, while the passive cases show good performance in the reduction of the displacement on the first floor.

### 3.3.8 Summary

The effectiveness of the MR damper-based semi-active control systems with various control algorithms for seismic protection of full-scale five-story steel frame building structure is experimentally verified in this investigation. An MR damper is designed by deriving a suboptimal design procedure considering optimization problem and magnetic analysis, and manufactured into 1.0ton MR damper in order to realize semiactive control systems. In the experiments, the pseudo-earthquake testing method is used to excite the building structure as if it is subjected to earthquake loading. Under the four historical earthquakes

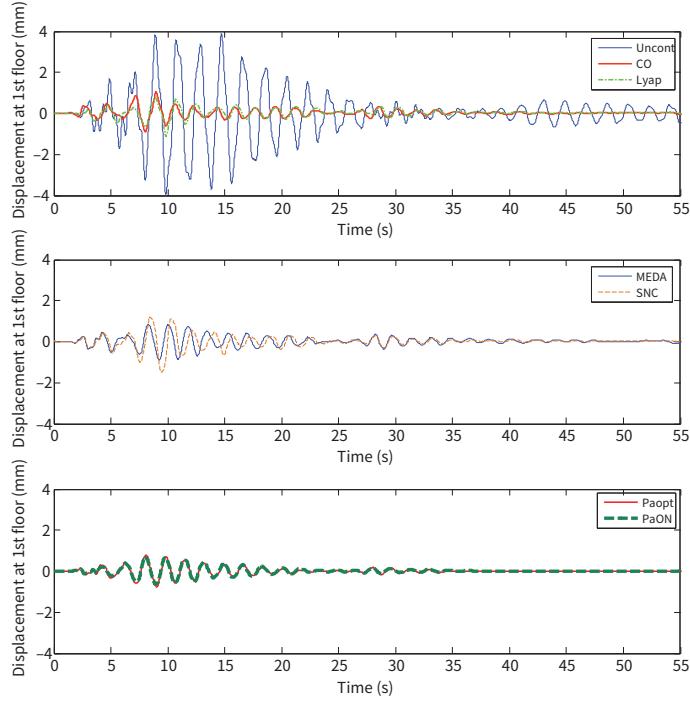


Figure 3.41: Time history responses of displacement at the first floor under the El Centro earthquake.

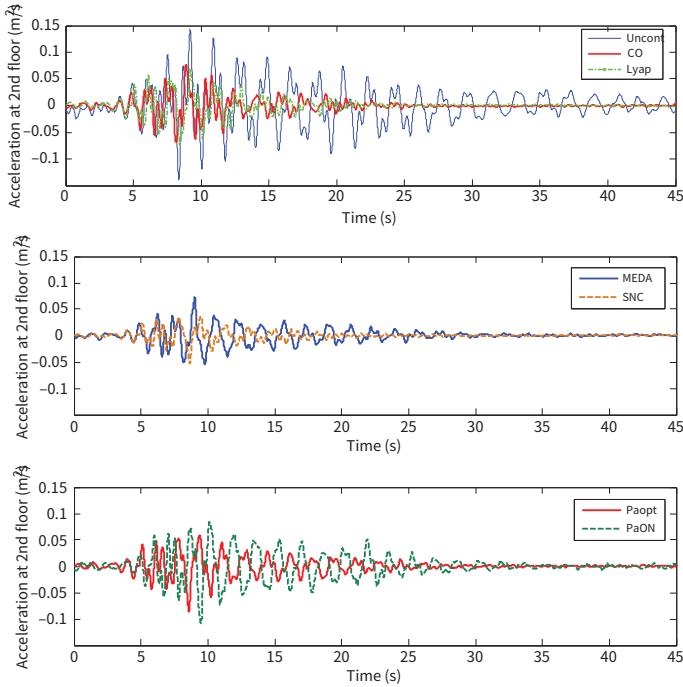


Figure 3.42: Time history responses of acceleration at the second floor under the Kobe earthquake.

and one filtered artificial earthquake, various semi-active control algorithms including the passive optimal case are evaluated and compared with one another. From the series of

passively controlled tests which are conducted as increasing the current level, the passive case with the input current of 1.0A is chosen as the passive optimal system. From the experimental results, one can conclude that LYAP and SNC algorithms are appropriate in reducing accelerations of the structural system, and passive optimal case and MEDA shows excellent performance in reducing the first-floor displacement.



# Hybrid Testing Method on a Semi-actively Controlled Building Structure Equipped with Full-scale MR Dampers

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In this study, the verification of the hysteretic behavior of an MR damper and the quantitative evaluation of the seismic performance of a building structure installed with an MR damper are carried out experimentally using the hybrid testing method. This component should be an electronic control unit or a real engine. The interface between simulated and physical components is achieved through the direct transfer of electrical signals, without actuation devices(Christenson et al., 2008). As described in chapter 2, the hybrid testing method is a structural testing technique, in which the numerical integration of the equation of motion of a numerical substructure and the physical testing of an experimental substructure with severe non-linear characteristics are carried out simultaneously in real-time. In chapter 2, it has been performed both experimentally and analytically on the hybrid testing technique in order to overcome the difficulties in the testing of large-scale structures. This technique is especially useful for evaluating the performance of the non-linear control device itself as well as that of the integrated system incorporated with the nonlinear control device such as MR damper.

## 4.1 Outline of Methodology

Figure 4.1 shows the concept of the hybrid testing method, which is experimentally implemented in this study. As shown in Figure 4.1(a), the structural response of a building model with  $n$ -degrees-of-freedom subjected to base input motion,  $\ddot{x}_g(t)$ , is controlled by an MR damper, which is typically located in the first story to reduce the maximum shear force of a bare structure. The whole system is divided into experimental and numerical substructures (Figure 4.1(b)). The MR damper is used as an experimental substructure because it is very difficult to numerically predict its exact dynamic behavior under seismic

load, due to its strong non-linear characteristic of the dependency of the damper force on the loading rate and amplitude. The remaining parts, that is, the structure without an MR damper, are analytically calculated. As shown in Figure 4.1(c), three procedures including the measurement of force, the numerical calculation of analytical parts, and the loading of the experimental substructure are used to implement the hybrid testing method for the whole structural control system(Blakeborough et al., 2001). First, the force acting at the interface between experimental and numerical substructures (here, the control force generated by the MR damper), is measured by the load cell attached to an actuator. The value of this measured control force is then returned to the control computer for use in the calculation of the displacement constraint condition, which should be satisfied by both the experimental and numerical substructures. Finally, the MR damper physically tested in the laboratory is loaded by an actuator with respect to the displacement response of the story installed with the MR damper. The numerical substructure surrounded by the dotted line in Figure 4.1(c) is calculated based on the equation of motion for the building model equipped with an MR damper that is subjected to ground input acceleration, which is represented by:

$$\mathbf{M}\ddot{\mathbf{x}}(t) + \mathbf{C}\dot{\mathbf{x}}(t) + \mathbf{K}\mathbf{x}(t) = -\mathbf{M}\boldsymbol{\Gamma}\ddot{x}_g(t) + \mathbf{H}f_e(t) \quad (4.1)$$

where,  $\mathbf{M}$ ,  $\mathbf{C}$ , and  $\mathbf{K}$  represent the  $n \times n$  structural mass, damping, and stiffness matrices, respectively;  $\mathbf{x}(t)$  the  $n \times 1$  vector of the relative structural displacement to the ground input motion;  $\boldsymbol{\Gamma}$  the  $n \times 1$  vector of the ground input motion influence coefficients;  $\mathbf{H}$  the  $n \times 1$  vector that represents the location of the MR dampers;  $\ddot{x}_g(t)$  the ground input acceleration; and  $f_e(t)$  the control force exerted by the MR damper on the structure, in which subscript ‘e’ represent ‘*experimentally*’ measured control force.

To perform the hybrid testing method, Eq. (4.1) is transformed into its state-space representation, which positively specifies the relationship between the input and output of the numerical substructure and can be written as:

$$\begin{aligned} \dot{\mathbf{z}}_s(t) &= \mathbf{A}_s \mathbf{z}(t) + \mathbf{B}_s \mathbf{u}(t) \\ \mathbf{Y}_s(t) &= \mathbf{C}_s \mathbf{z}(t) + \mathbf{D}_s \mathbf{u}(t) \end{aligned} \quad (4.2)$$

where  $\mathbf{z}_s(t) = \{\mathbf{x}(t), \dot{\mathbf{x}}\}^\top$  is the  $2n \times 1$  system state vector;  $\mathbf{U}(t) = \{f_e(t), \ddot{x}_g(t)\}^\top$  the  $2 \times 1$  system input vector; and  $\mathbf{Y}_s(t) = \mathbf{x}(t)$  the  $n \times 1$  system output vecor. The  $2n \times 2n$  system state matrix,  $\mathbf{A}_s$ , and the  $2n \times 2$  locaation matrix of system input,  $\mathbf{B}_s$ ,

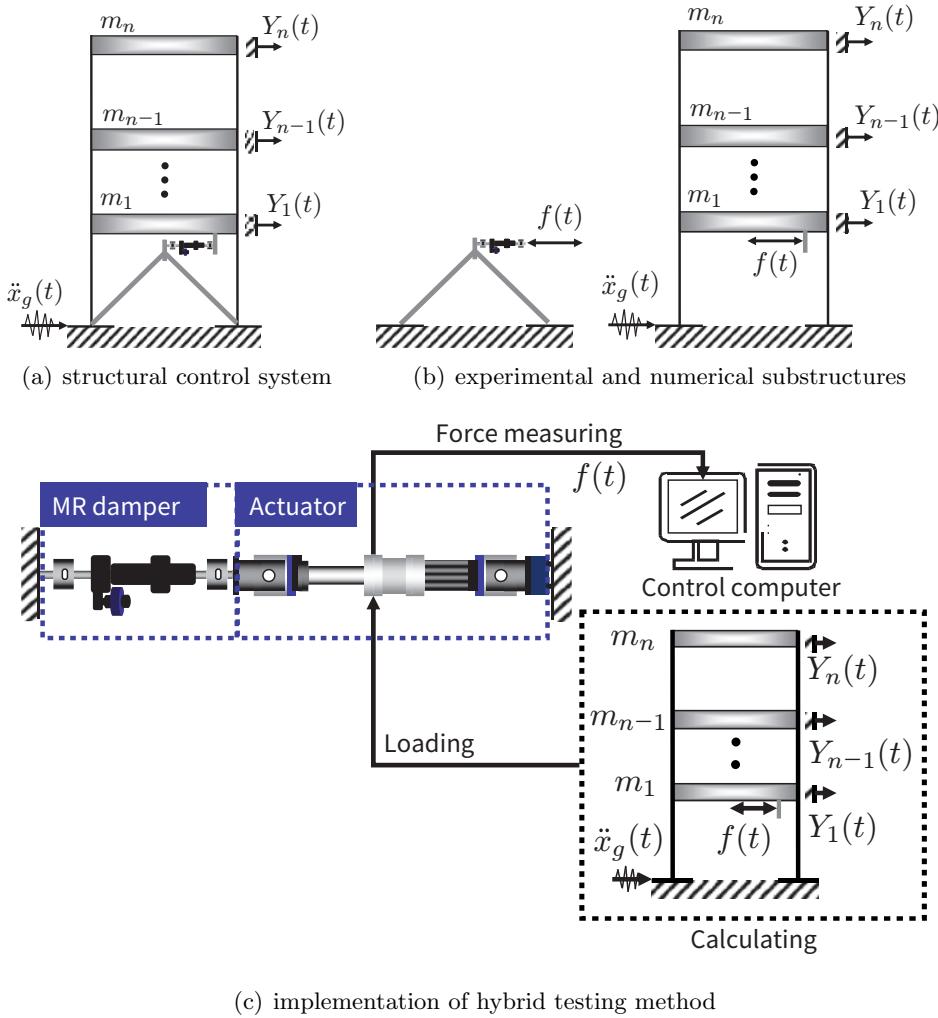


Figure 4.1: Conceptual view of hybrid testing method for a building with an MR damper.

both associated with the system state variables, are represented by:

$$\mathbf{A}_s = \begin{bmatrix} \mathbf{0}_{n \times n} & \mathbf{I} \\ -\mathbf{M}^{-1}\mathbf{K} & -\mathbf{M}^{-1}\mathbf{C} \end{bmatrix} \quad (4.3)$$

$$\mathbf{B}_s = \begin{bmatrix} \mathbf{0}_{n \times 1} & \mathbf{0}_{n \times 1} \\ -\mathbf{M}^{-1}\mathbf{H} & -\mathbf{\Gamma} \end{bmatrix}$$

and the  $n \times n$  system out matrix,  $\mathbf{C}_s$ , and the  $n \times 2$  direct transmission matrix,  $\mathbf{D}_s$ , both related to the system output variables, are given by:

$$\mathbf{C}_s = \mathbf{I}, \mathbf{D}_s = \mathbf{0}_{n \times 1} \quad (4.4)$$

In Eqs. (4.3) and (4.4),  $\mathbf{I}$  is the  $n \times n$  unit matrix, while  $\mathbf{0}_{n \times n}$  and  $\mathbf{0}_{n \times 1}$  are the  $n \times n$  and  $n \times 1$  zero matrices, respectively. Note that, in the actual implementation of hybrid testing method, the system output vector,  $\mathbf{Y}_s(t)$ , in Eq. (4.2) should be designated as the

story drifts where the MR dampers are installed and Eq. (4.4) should also be modified because these story drifts affect the system output vector.

A full-scale five-story steel frame building is considered as the model of the numerical substructure to use in the structural modal testing of this study, as shown in Figure 3.17.

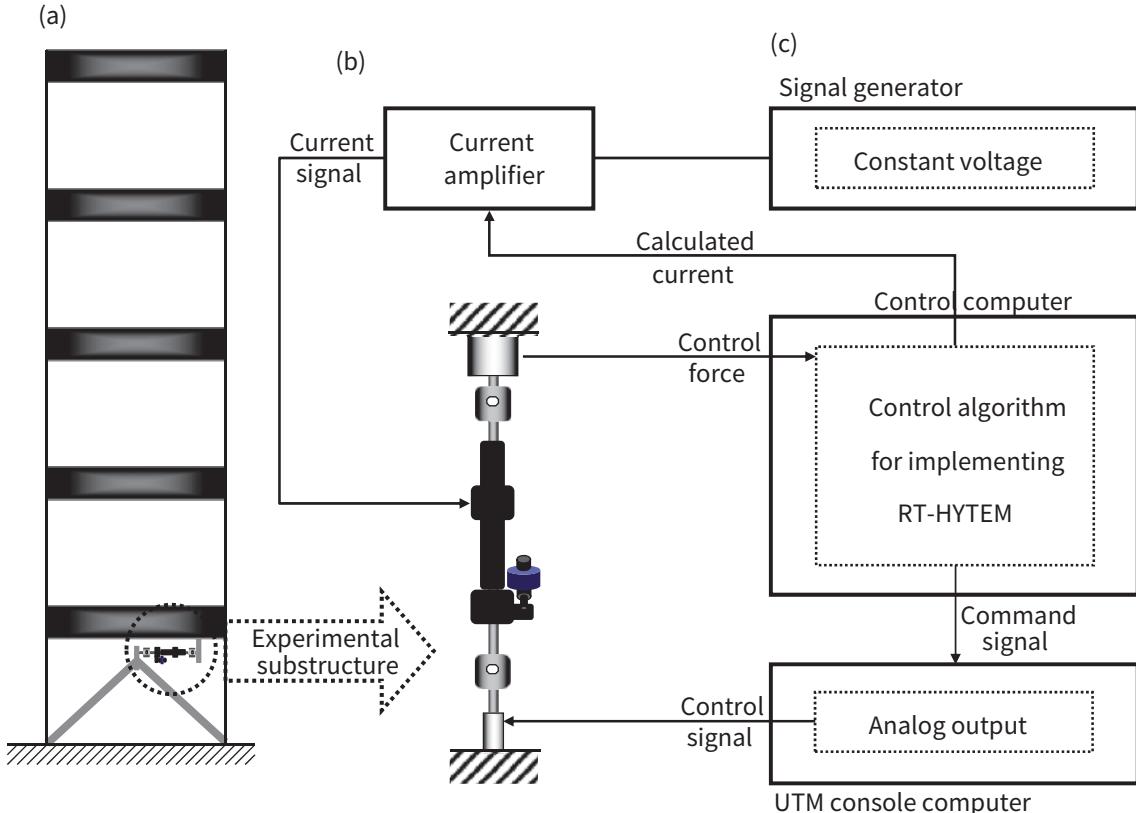


Figure 4.2: Schematic view of experimental set-up: (a) building model installed with an MR damper, (b) UTM installed MR damper, and (c) experimental instrumentation.

When these identified structural properties are incorporated in the numerical substructure, Eqs. (4.2), (4.3) and (4.4) can be rewritten as Eqs. (4.5), (4.6) and (4.7), respectively.

$$\begin{aligned}\dot{\mathbf{z}}_n(t) &= \mathbf{A}_n \mathbf{z}_n(t) + \mathbf{B}_n \mathbf{u}(t) \\ \mathbf{Y}_n(t) &= \mathbf{C}_n \mathbf{z}_n(t) + \mathbf{D}_n \mathbf{u}(t)\end{aligned}\quad (4.5)$$

where,  $\mathbf{Z}_n(t) = \{x_1(t), \dots, x_5(t), \dot{x}_1(t), \dots, \dot{x}_5(t)\}^\top$  is the  $10 \times 1$  system state vector;  $\mathbf{u}(t) = \{f_e(t), \ddot{x}_g(t)\}^\top$  the  $2 \times 1$  system input vector; and  $\mathbf{Y}_n(t) = \{x_1(t), \dots, x_5(t), \dot{x}_1(t), \dots, \dot{x}_5(t), \ddot{x}_1(t), \dots, \ddot{x}_5(t)\}^\top$  the  $15 \times 1$  system output vector. The  $10 \times 10$  system state matrix,

$\mathbf{A}_n$ , and the  $10 \times 2$  location matrix of system input,  $\mathbf{B}_n$ , are represented by Eq. (4.6):

$$\begin{aligned}\mathbf{A}_n &= \begin{bmatrix} \mathbf{0}_{5 \times 5} & \mathbf{I} \\ -\mathbf{M}_5^{-1}\mathbf{K}_5 & -\mathbf{M}_5^{-1}\mathbf{C}_5 \end{bmatrix} \\ \mathbf{B}_n &= \begin{bmatrix} \mathbf{0}_{5 \times 1} & \mathbf{0}_{5 \times 1} \\ -\mathbf{M}_5^{-1}\mathbf{H}_5 & -\mathbf{\Gamma}_5 \end{bmatrix}\end{aligned}\quad (4.6)$$

and the  $15 \times 10$  system out matrix,  $\mathbf{C}_5$ , and the  $15 \times 2$  direct transmission matrix,  $\mathbf{D}_5$  are given by Eq. (4.7)

$$\begin{aligned}\mathbf{C}_n &= \begin{bmatrix} \mathbf{I} & \mathbf{0}_{5 \times 5} \\ \mathbf{0}_{5 \times 5} & \mathbf{I} \\ -\mathbf{M}_5^{-1}\mathbf{K}_5 & -\mathbf{M}_5^{-1}\mathbf{C}_5 \end{bmatrix} \\ \mathbf{D}_n &= \begin{bmatrix} \mathbf{0}_{5 \times 1} & \mathbf{0}_{5 \times 1} \\ \mathbf{0}_{5 \times 1} & \mathbf{0}_{5 \times 1} \\ -\mathbf{M}_5^{-1}\mathbf{H}_5 & -\mathbf{\Gamma}_5 \end{bmatrix}\end{aligned}\quad (4.7)$$

In Eq. (4.6) and (4.7),  $\mathbf{I}$  is the  $5 \times 5$  unit matrix, while  $\mathbf{0}_{5 \times 5}$  and  $\mathbf{0}_{5 \times 1}$  are the  $5 \times 5$  and  $5 \times 1$  zero matrices, respectively.

## 4.2 Controller Design Strategy for the Hybrid Testing Method

### 4.2.1 Experimental Set-up

In general, the successful experimental implementation of the hybrid testing method with high accuracy depends strongly on the dynamic performance of the actuators used to excite the experimental substructure in the test. Namely, the actuators used in the test should promptly react to applied command signals and load the experimental substructure. In this chapter, a universal testing machine (UTM), which is commonly used for the performance test of various structural materials, is utilized as an actuator to load the experimental part. Figure 4.3 shows the experimental set-up using UTM with the maximum loading capacity of 200kN and excitation frequency range of 0-10 Hz (Model name STL-HU10T). An MR damper, which can develop a control force ranging from 3 to 12 kN, is connected to this system. As shown in Figure 4.2, in order to perform the hybrid testing method, three apparatuses with interconnected functions are equipped with the experimental set-up. The main task of the signal generator is to generate the current signal, by which an MR damper is operated, and control force is exerted on the UTM. The control computer with a digital signal processing (DSP) board calculates the responses of the numerical substructure using the input (the control force measured from a load cell of the UTM) and sends the command

signal to the UTM console computer. The primary task of the UTM console computer is to transfer the command signal generated by the control computer, according to the control algorithm for implementing the hybrid testing method, to the UTM. Finally, the UTM excites the experimental part according to this control signal.

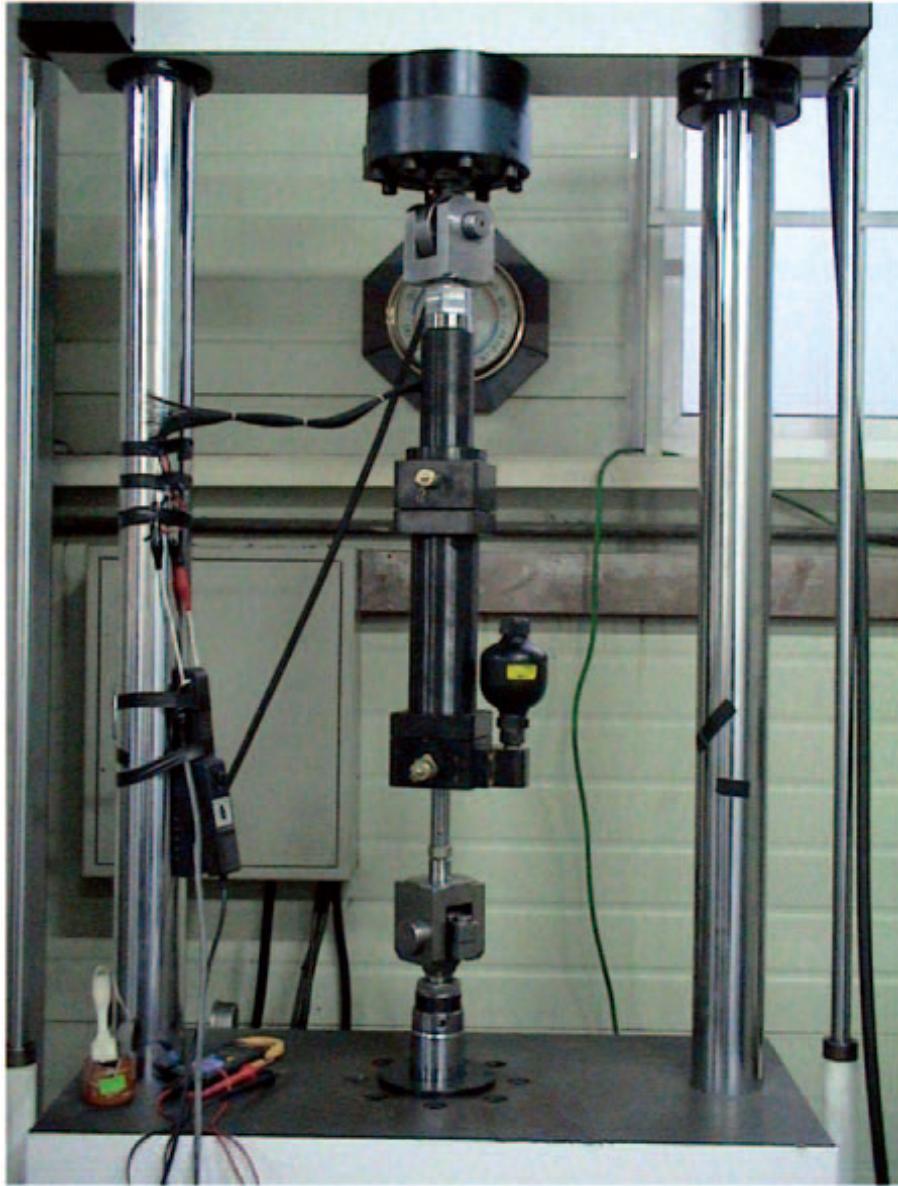


Figure 4.3: Configuration of experimental system.

#### 4.2.2 Controller Design of UTM

In order to accurately load the experimental substructure by an actuator and in accordance with the control algorithm of the hybrid testing method, the dynamic property of the actuator itself between the command signal and the measured response should be appropriately compensated. Without this compensation, problems can occur with the ex-

perimental results such as chattering problems or the operating performance of the testing system can be degraded. In particular, during structural testing associated with vibration control issues, the control force often acts as an external load, and as a result, the experimental system can become unstable.

In order to experimentally measure the dynamics of UTM, white noise with a frequency ranging from 0 to 10Hz is used as the command signal, as shown in Figure 4.2 and the corresponding displacement of UTM that is driven by this signal are then measured. Using these input and output data sets in the time domain, the transfer function in the frequency domain, which corresponds to the dynamics of the UTM itself, is obtained.

The phase delay is compensated using the inverse transfer function (ITRF; Lee et al. (2007a,b)), in which the relationship between the input and output in the transfer function is reversed. Accordingly, the measured ITRF is obtained with the input of the measured displacement of the UTM and the output of the command signal, as shown by the dotted lines in Figure 4.4. This measured ITRF should be incorporated in the control computer as part of the control algorithm shown in Figure 4.2, in order to compensate the phase delay of the UTM and to correctly excite the experimental substructure. The measured ITRF is approximated using the `invfreqs` command in MATLAB(Coleman et al., 1999), which finds the real numerator and denominator coefficient vectors of the approximated transfer function in the form of a fractional expression by adopting the damped Gauss-Newton method for iterative search, which minimizes the sum of the squared error between the measured and the approximated frequency response points(Dennis Jr and Schnabel, 1983). The approximation result is shown by the solid line in Figure 4.4 is given by the following second-order linear analog filter.

$$G^{-1}(s) = \frac{7.1945s^2 + 636.4862s + 27368.6533}{s^2 + 412.4726s + 27758.6756} \quad (4.8)$$

where the Laplace variable,  $s$ , equals  $j\omega$  with the imaginary constant,  $j$ .

Figure 4.4 shows that, at 5 Hz, compensation on phase lag is attained only by 20° with the approximated ITRF, while the phase lag of 60° is observed in the measured ITRF. Further compensation would be possible with the use of other filters. However, further compensation is excluded here since it would have resulted in an unstable ITRF.

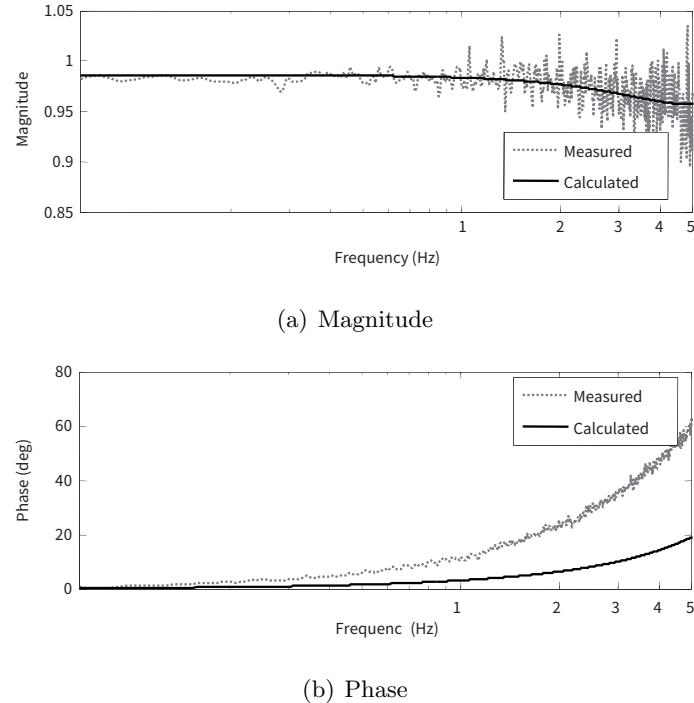


Figure 4.4: Measured and approximated ITRFs from the displacement of UTM to the command signal.

Eq.(4.8) is converted into the following statespace realization for implementation in the control computer.

$$\begin{aligned} \mathbf{y}_s &= \mathbf{A}_i \mathbf{y}_s(t) + \mathbf{B}_i r(t) \\ c(t) &= \mathbf{C}_i \mathbf{y}_s(t) + D_i r(t) \end{aligned} \quad (4.9)$$

where  $\mathbf{y}_s$ ,  $r(t)$  and  $c(t)$  are the  $2 \times 1$  state vector, the  $1 \times 1$  reference signal, and the  $1 \times 1$  command signal of the UTM, respectively; and  $\mathbf{A}_i$ ,  $\mathbf{B}_i$ ,  $\mathbf{C}_i$  and  $\mathbf{D}_i$  are the  $2 \times 2$ ,  $2 \times 1$ ,  $1 \times 2$  and  $1 \times 1$  system matrices, respectively.

### 4.3 Experimental verification

#### 4.3.1 Integrated Controller of the Numerical Substructure and UTM

The numerical substructure and the approximated ITRF explained above are incorporated into the control computer as an integrated controller to implement the hybrid testing method, as shown by the shaded area in Figure 4.5. First, when the constant current sent from the signal generator for the passive control test and the calculated current of semi-active control algorithms sent from the analog output for the semi-active control test are applied to the MR damper, the control force proportional to the magnitude of the current signal is developed and transferred to the UTM. The drift response,  $x_1(t)$ , of

the first story where the MR damper is positioned, is then calculated from the numerical substructure, given by Eq. (4.5), with two inputs: the control force,  $f_e(t)$ , measured from the load cell attached to the UTM and the ground input acceleration,  $\ddot{x}_g(t)$  given by the user. The motion of the UTM is driven by the controller using the ITRF, represented by Eq (4.9) so that the UTM itself operates as the first story where the MR damper is located and excites the MR damper that should be physically tested. In the actual experimental implementation of hybrid testing method, the continuous filters shown in Figure 4.5 are converted into discrete filters with a time interval of 0.01s, while MATLAB Simulink(Simulink, 2009) and Real-Time Windows Target(Target, 2009) are used as the control system.

#### 4.3.2 Experimental Verification for Simple Bouc-Wen Model

The hybrid testing method with a sinusoidal wave excitation is carried out to investigate the hysteretic behavior of the MR damper and to verify the hybrid testing method experimentally implemented in this chapter. A current of 0 A is applied to the MR damper and a sine wave with a frequency of 0.52 Hz, which corresponds to the fundamental frequency of the numerical substructure, and with an amplitude of  $5m/s^2$  is used as the ground input acceleration as shown in Figure 4.5.

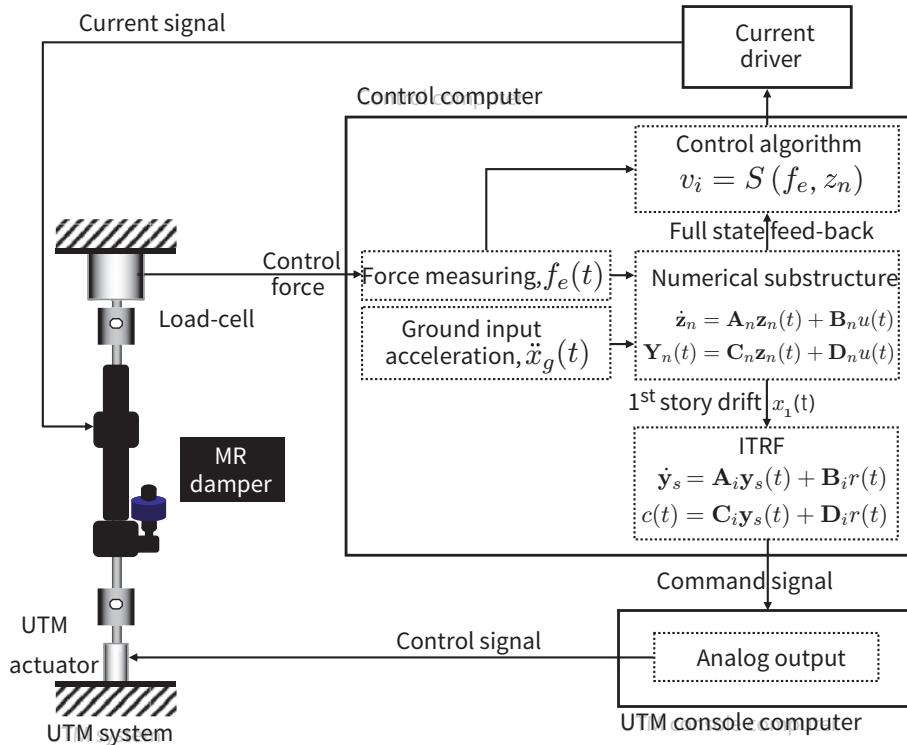


Figure 4.5: Integrated controller for implementing the hybrid testing method.

Parametric identification is then performed to find the numerical model of the MR

damper on the basis of the experimentally measured hysteretic loop. The Bouc-Wen model is widely used for modeling various hysteretic loops(Wen, 1976). The control force by the MR damper can also be specified using this model. The force described by this model is represented by:

$$f_{MR}(t) = \alpha z(t) + c_{MR}\dot{x}_1(t) + k_{MR}x_1(t) + f_0 \quad (4.10)$$

where,  $k_{MR}$  and  $c_{MR}$  are the stiffness and viscosity of the MR damper;  $f_0$  the initial friction force;  $z$  a dimensionless valuable introduced to describe the hysteresis; and  $\alpha$  a variable that regulates the effect of  $z$  on  $f_{MR}(t)$  which depends on the magnetic field;  $z$  is given by the following differential equation.

$$\dot{z}(t) = -\gamma|\dot{x}_1(t)|z(t)|z(t)|^{n-1} - \beta\dot{x}_1(t)|z(t)|^n + A\dot{x}_1(t) \quad (4.11)$$

where  $\gamma$ ,  $\beta$ ,  $A$  and the superscript  $n$  are the coefficients determining the shape of the hysteretic curve.

The eight parameters in Eqs. (4.10) and (4.11) ( $\alpha$ ,  $c_{MR}$ ,  $k_{MR}$ ,  $f_0$ ,  $\gamma$ ,  $n$ ,  $\beta$  and  $A$ ) are identified using least-squares optimization to minimize the performance index defined as:

$$J(\mathbf{p}) = \sum_{k=1}^N [f_e(k \cdot \Delta t) - f_{MR}(\mathbf{p}, k \cdot \Delta t)]^2 \quad (4.12)$$

where  $f_e(k \cdot \Delta t)$  represents the force data measured at the  $k$ -th sampling time;  $\mathbf{p}$  the  $1 \times 8$  vector of the identification parameter,  $\{\alpha, c_{MR}, k_{MR}, f_0, \gamma, n, \beta, A\}$ ; and  $f_{MR}(\mathbf{p}, k \cdot \Delta t)$  the force obtained at the  $k$ -th calculating time from Eq. (4.10) and (4.11) using the parameter  $\mathbf{p}$ .

This procedure is carried out using MATLAB subroutines(Coleman et al., 1999), and the identified parameters are given by:

$$\begin{aligned} \alpha &= 13288.130(N/m), & c_{MR} &= 81418.582(N \cdot s/m), \\ k_{MR} &= 16647.456(N/m), & f_0 &= 6.10(N), \\ n &= 3.0581, & \gamma &= 471409.377(m^{-n}), \\ \beta &= 335518.804(m^{-n}), & A &= 497.295 \end{aligned} \quad (4.13)$$

A comparison between the calculated and experimental responses is provided in Figure 4.6. The displacement and velocity in Figures 4.6(b) and 4.6(c) correspond to the first-story drift and velocity responses, respectively. The Bouc-Wen model represents the force-displacement and the force-velocity hysteretic behaviors of the damper well.

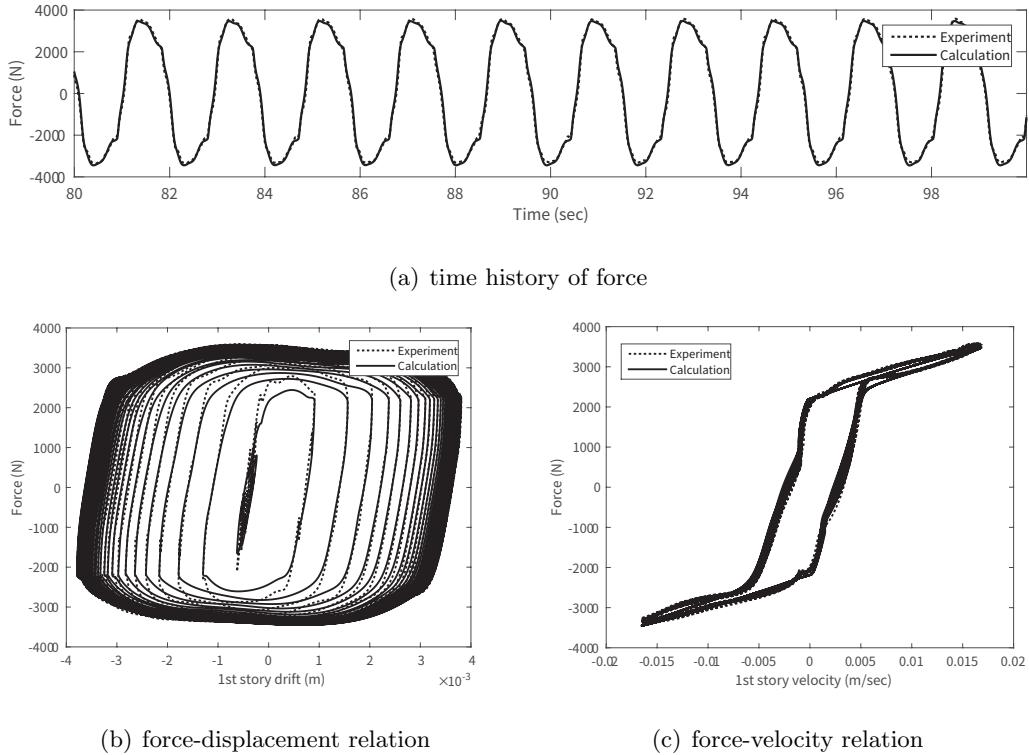


Figure 4.6: Comparison between calculated and experimentally measured responses for the Bouc-Wen model.

The hybrid testing method implemented in this chapter is verified by the experiment using the excitation of historical seismic measurements. Four different records of earthquake acceleration measurements taken at El Centro, Hachinohe, Kobe, and Northridge are used as ground input acceleration in Figure 4.5 by multiplying them by 0.05. Also, the current of 0 [A] is applied to an MR damper in the same manner as the sinusoidal excitation test above. In addition to the measurement of the control force produced by an MR damper, given by Eqs. (4.5)-(4.7), the structural displacement, velocity, and absolute acceleration responses are obtained from the hybrid testing method for earthquake excitations. Figures 4.7 and 4.8 show a comparison of the responses measured by hybrid testing method, as shown in Figure 4.5, with those calculated from the five-story building model installed with an MR damper with the identified parameters such as in Eq. (4.13), in order to compute the control force.

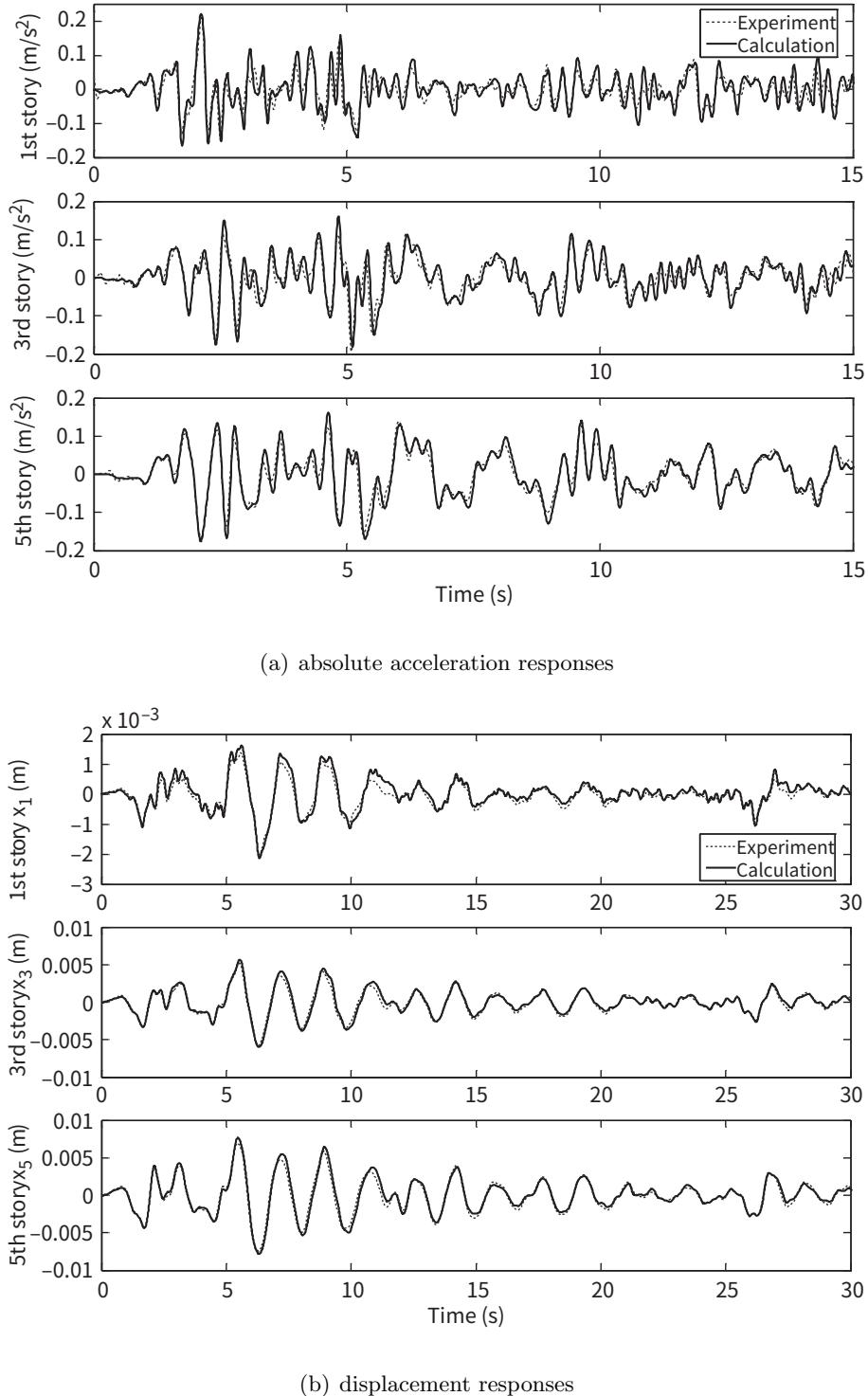


Figure 4.7: Comparison between calculated and experimentally measured responses under El Centro earthquake excitation.

In addition, the measured and calculated hysteretic behaviors, shown in Figures 4.9 and 4.10, are compared for different seismic excitations. Figures 4.7 and 4.8 show that the two results agree well with each other. Figures 4.9 and 4.10 show that the Bouc-Wen model

successfully predicted the overall hysteretic behaviors. In particular, the Bouc-Wen model successfully predicted the hysteretic behavior of an MR damper under the excitation of the seismic measurements of a narrow frequency band, such as those of the Kobe and Northridge earthquakes corresponding to the representative examples of near-fault source ground motion.

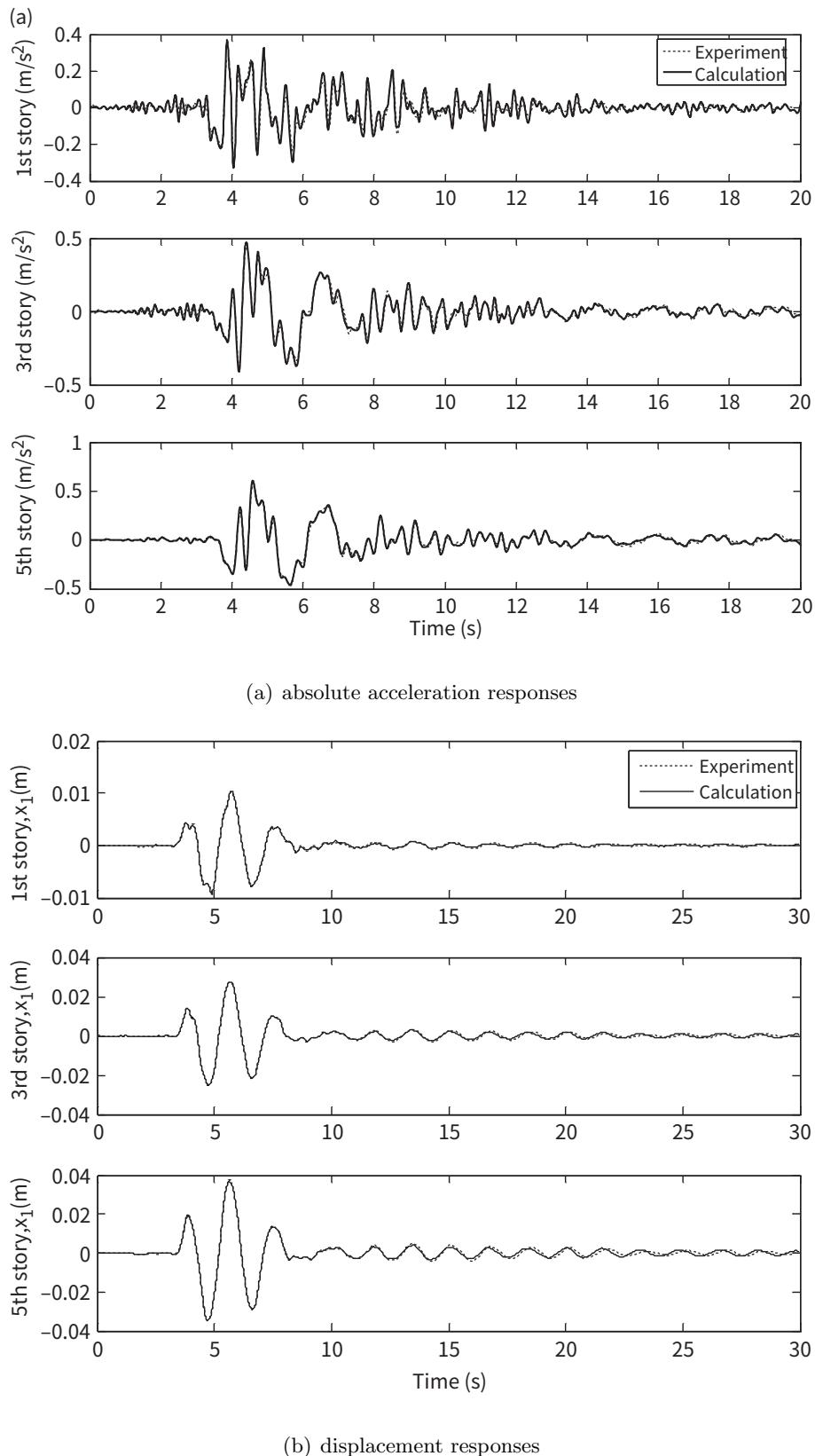


Figure 4.8: Comparison between calculated and experimentally measured responses under Northridge earthquake excitation.

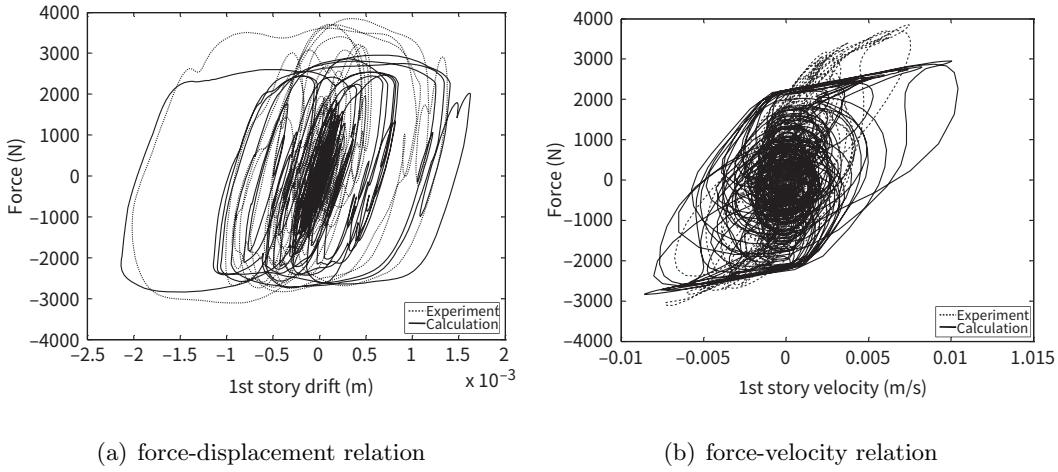


Figure 4.9: Comparison between calculated and experimentally measured hysteresis under El Centro earthquake excitation.

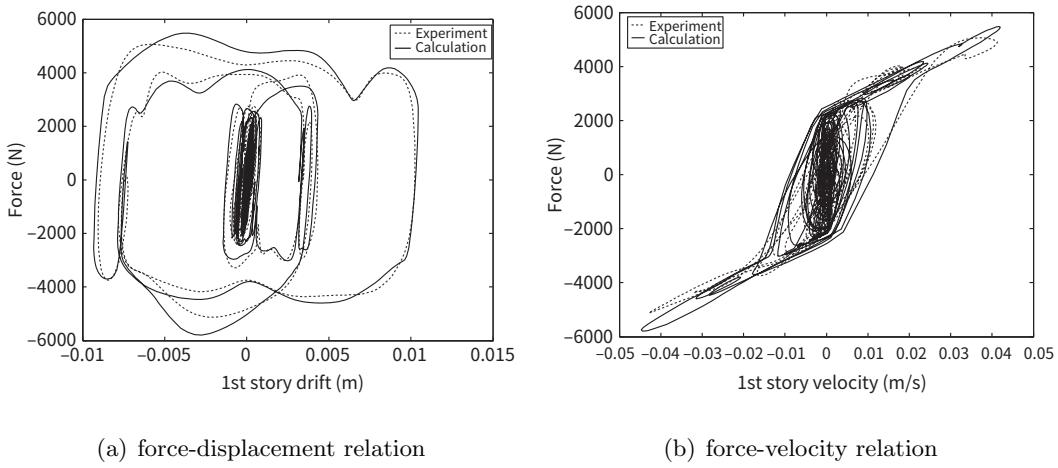


Figure 4.10: Comparison between calculated and experimentally measured hysteresis under Northridge earthquake excitation.

### 4.3.3 Applied Semi-active Control Algorithms

#### OPTIMIZATION OF BOUC-WEN PARAMETER BY INPUT CURRENTS

In order to compare the results of the numerical analysis with the results of the semi-active hybrid testing method, the identification of the Bouc-Wen parameter corresponding to input currents is required. In this study, a sinewave excitation test is implemented, and Bouc-Wen parameters ( $\alpha$ ,  $A$ ,  $c_{MR}$ ) are identified.

Figure 4.11 shows a comparison between the test model and the identified Bouc-Wen model with respect to different input currents. The optimal parameter of the Bouc-Wen model is established by force-velocity and force-displacement relationships, as shown in

Figure 4.11(a) and 4.11(b). For implementing the semi-active control of an MR damper, three parameters which are assumed to be polynomial, exponential functions of the input current are identified using the least square method. Figures 4.11(c)-4.11(e) show the results of each parameter identified and tested.

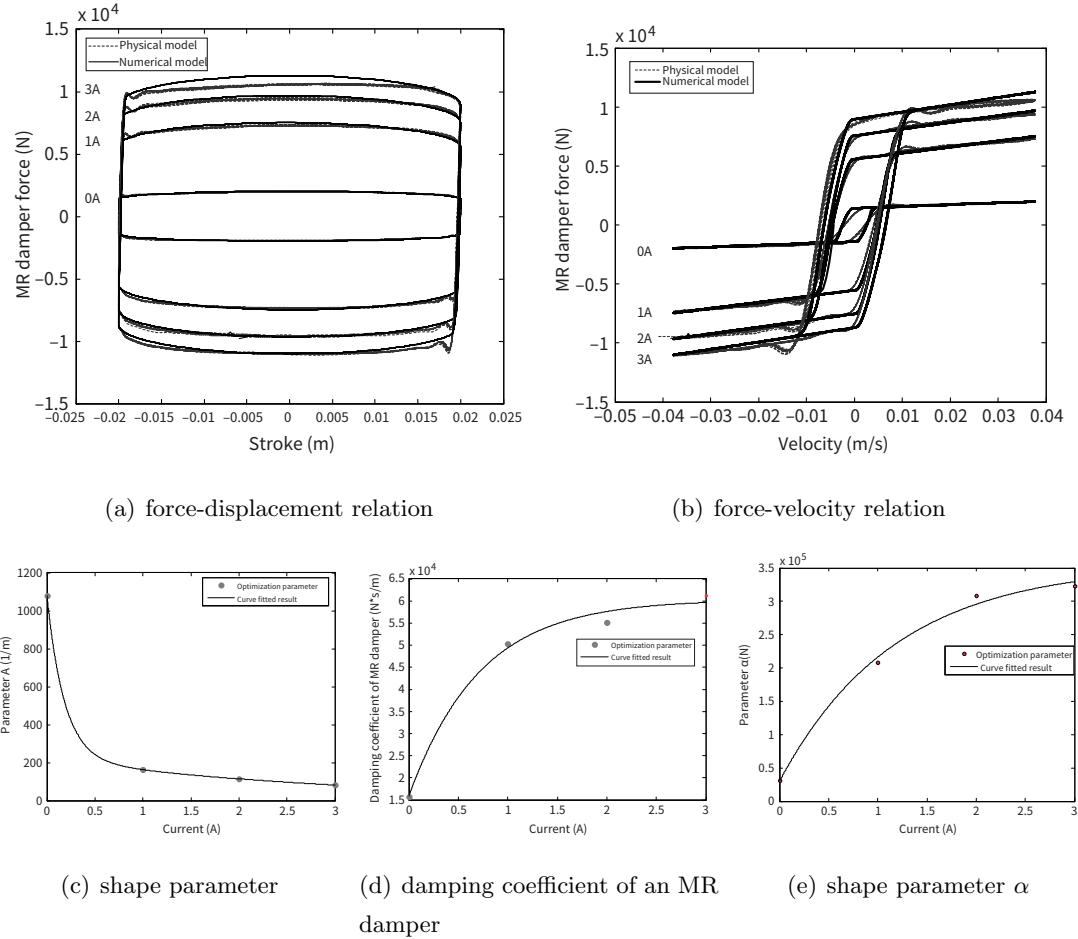


Figure 4.11: Identified Bouc-Wen parameters and the numerical model of an MR damper.

$$\alpha = -324576.03 \exp(-0.85x) + 354777.40 \quad (4.14)$$

$$c_{MR} = -44567.17 \exp(-1.4x) + 60330.96 \quad (4.15)$$

$$A = 854.7 \exp(-5.476x) + 222.5 \exp(-0.3321x) \quad (4.16)$$

where,  $x$  is input current ( $A$ )

### Clipped-optimal Control Algorithm

In terms of implementation, this control strategy seems to be the most direct because it can take advantage of the significant number of experimental and practical studies that

have been conducted on active control strategies. The clipped-optimal algorithm that has been shown to be effective for use with the MR damper has been proposed by Dyke et al. (1996). The clipped-optimal control approach involves designing a linear optimal controller  $\mathbf{K}_c(s)$  that calculates a vector of desired control forces  $\mathbf{f}_c = \{f_{c_1}, f_{c_2}, \dots, f_{c_n}\}^\top$  based on the measured structural responses  $\mathbf{y}$  and the measured control force vector  $\mathbf{f}$  applied to the structure, that is,

$$\mathbf{f}_c = \mathcal{L}^{-1} \left\{ -\mathbf{K}_c(s) \mathcal{L} \{\mathbf{y}, \mathbf{f}\}^\top \right\} \quad (4.17)$$

where,  $\mathcal{L} \{\cdot\}$  is the Laplace transform.

Because the force generated in the MR damper is dependent on the local responses of the structural system, the desired optimal control force  $f_{c_i}$  cannot always be produced by the MR damper. Only the control voltage  $v_i$  can be directly controlled to increase or decrease the force produced by the device. Thus, a force feedback loop is incorporated to induce the MR damper to approximately generate the desired optimal control force  $f_{c_i}$ .

To induce the MR damper to approximately generate the corresponding desired optimal control force  $f_{c_i}$ , the command signal  $v_i$  is selected as follows. When the  $i$ -th MR damper is providing the desired optimal force, the voltage applied to the damper should remain at the present level. If the magnitude of the force produced by the damper is smaller than the magnitude of the desired optimal force and the two forces have the same sign, the voltage applied to the current driver is increased to the maximum level so that the force produced by the damper is increased to match the desired control force. Otherwise, the commanded voltage is set to zero. The algorithm for selecting the command signal for the  $i$ -th MR damper is stated as Eq. (4.18)(Jansen and Dyke, 2000).

$$v_i = V_{\max} H(\{f_{c_i} - f_i\} f_i) \quad (4.18)$$

Although a variety of approaches may be used to design the optimal controller, linear quadratic regulator (LQR) methods are advocated because of their successful application. The optimal gain of a state vector based on the target building model with MR damper is used as follows:

$$\begin{aligned} \mathbf{K}_c = [ & -98301479.1, 110902868.9, \\ & -36659599.6, 24569454.5, -12841877.0, \\ & -4926535.9, -3813412.5, -331115.5, \\ & -1239172.8, -1309155.8 ] \end{aligned}$$

For implementing the hybrid testing method experiment, the integrated MATLAB *Simulink* controller is designed as shown in Figure 4.12. In the field testing application of this control law, it is required that a full structural state vector is obtained using a filter estimation method such as Observer/Kalman. However, because the hybrid testing method uses the structural model as a numerical model, this method has the advantage that the structural state variable is easily used in the experiment.

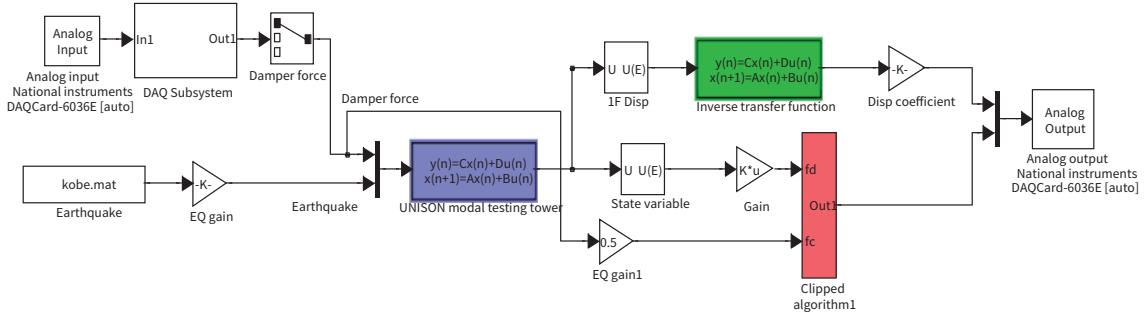


Figure 4.12: Clipped-optimal and hybrid testing method integrated controller.

## MODULATED HOMOGENEOUS FRICTION ALGORITHM (MHF)

This control strategy is originally developed for a variable-friction damper. In this approach, at every occurrence of local extremes in the deformation of the device, the normal force applied to the frictional interface is updated to a new value. At each local minimum or maximum in the deformation, the normal force  $N_i(t)$  is chosen to be proportional to the absolute value of the semi-active damper deformation. The control law is written as Eq. (4.19)(Inaudi, 1997):

$$N_i(t) = g_i |P[\Delta_i(t)]| \quad (4.19)$$

where  $g_i$  is a positive gain and the operator  $P[\cdot]$  as the prior-local-peak operator is defined as  $P[\Delta_i(t)] = \Delta_i(t-s)$ , where  $s = \left\{ \min x \geq 0 : \dot{\Delta}_i(t-x) = 0 \right\}$ , defining  $\Delta_i(t-s)$  as the most recent local extreme in the deformation. Because this algorithm was developed for variable friction devices, the following modifications are needed when applying it to MR dampers.

- (i) There is often no need to check if the force is greater than the static friction because MR dampers have no static friction.
- (ii) A force feedback loop is used to induce the MR damper to produce approximately the frictional force corresponding to the required normal force.

Thus, the goal is to generate a required control force with a magnitude of:

$$f_{n_i} = \mu g_i |P[\Delta_i(t)]| = g_{n_i} |P[\Delta_i(t)]| \quad (4.20)$$

where the proportionality constant  $g_{n_i}$  has a unit of stiffness N/mm.

As with the clipped-optimal control law, because the force produced by the MR damper cannot be directly commanded, a force feedback loop is used. The measured force is compared to the desired force determined by Eq. (4.20), and the resulting control law is:

$$v_i = V_{\max} H(f_{n_i} - |f_i|) \quad (4.21)$$

where  $V_{\max}$  is the maximum current value that can be offered for the MR damper.

An appropriate choice of  $g_{n_i}$  will maintain the force  $f_{n_i}$  within the operating envelope of each MR damper for the majority of the time, allowing the MR damper forces to approximate the desired force of each device closely. However, the optimal value of  $g_{n_i}$  is dependent on the amplitude of the ground acceleration. In this study, the  $g_{n_i}$  value is chosen as 150 kN/m based on the excitation of three earthquakes.

In addition, note that this control law is quite straightforward to implement because it only requires the measurement of the applied force and the relative displacements of the control device.

For applying the hybrid testing method experiment, the integrated MATLAB *Simulink* controller is designed as shown in Figure 4.13.

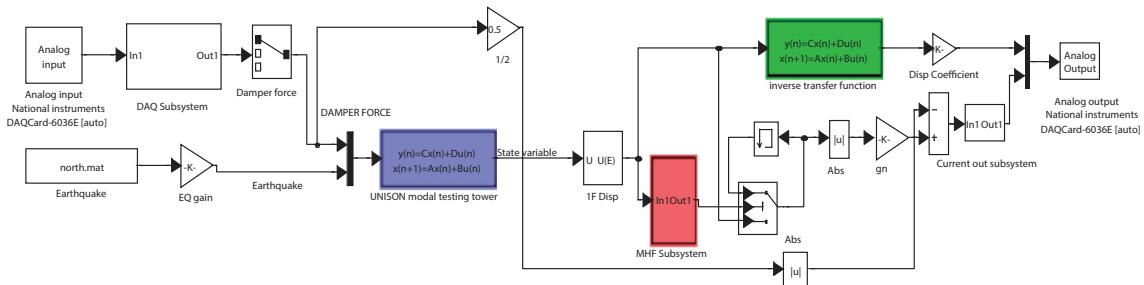


Figure 4.13: MHF and hybrid testing method integrated controller.

## 4.4 Testing Result

### 4.4.1 Passive Control Performance

The hybrid testing method illustrated in Figure 4.5 is implemented to evaluate the seismic performance of the passive-controlled MR damper used in this chapter. Except for the variation of the applied current to the MR damper (in the same manner as that above), four earthquake records are also given as the ground accelerations in Figure 4.5. The tests

for the controlled case are carried out by increasing the current applied to the MR damper from 0 to 3 A. As shown in Figure 4.6, the test for the uncontrolled case is performed by removing the feedback loop of the control force generated by an MR damper into the control computer. Therefore, the numerical substructure is excited only by the ground input acceleration (even though a current of 0A was applied to the MR damper).

Figures 4.14 and 4.15 show comparisons for the selected floors of the time histories of experimentally measured structural displacement responses to the excitation of two earthquake excitations, as the current applied to the MR damper is varied. Significant control effects are observed between the uncontrolled and the passive-off case but are not observed in the other passive-on cases with the increase of the applied current.

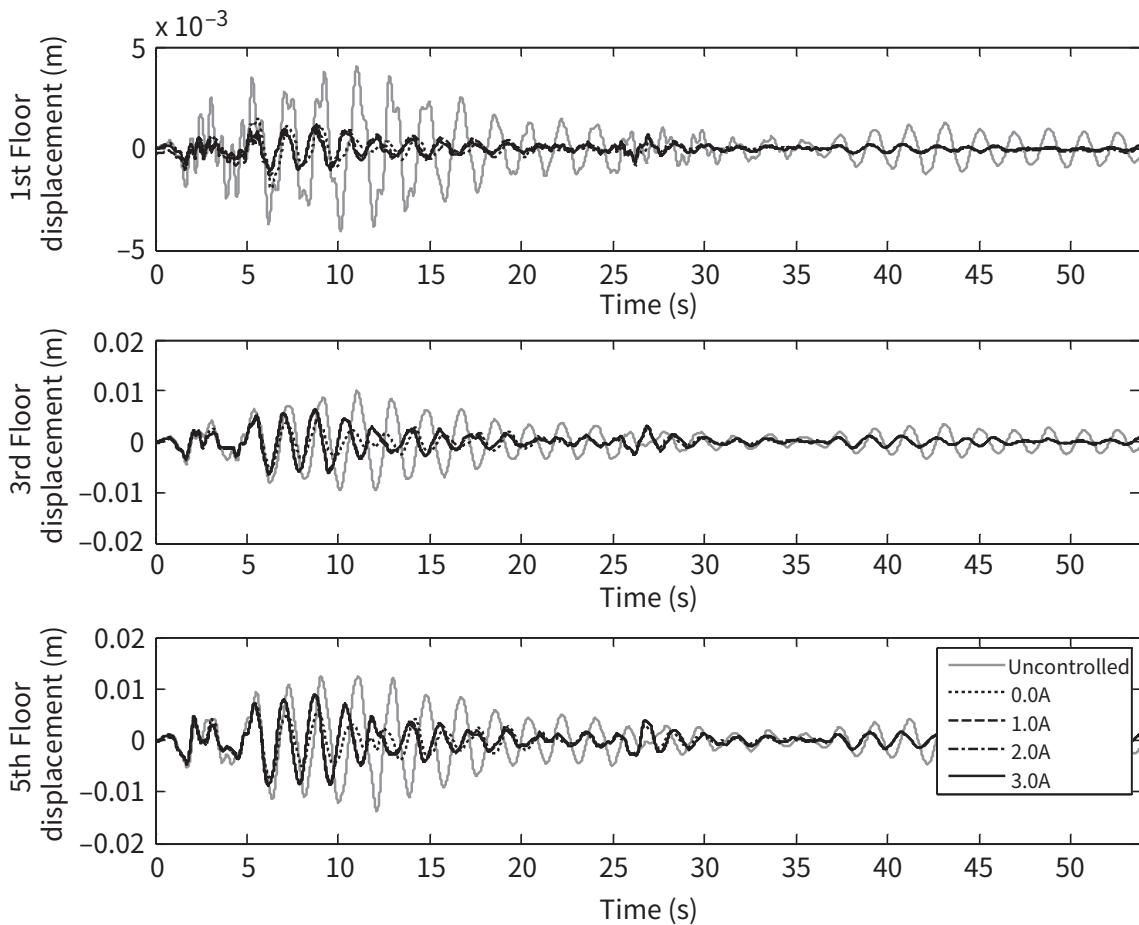


Figure 4.14: Experimental results, in the time domain under El Centro earthquake excitation at different applied currents.

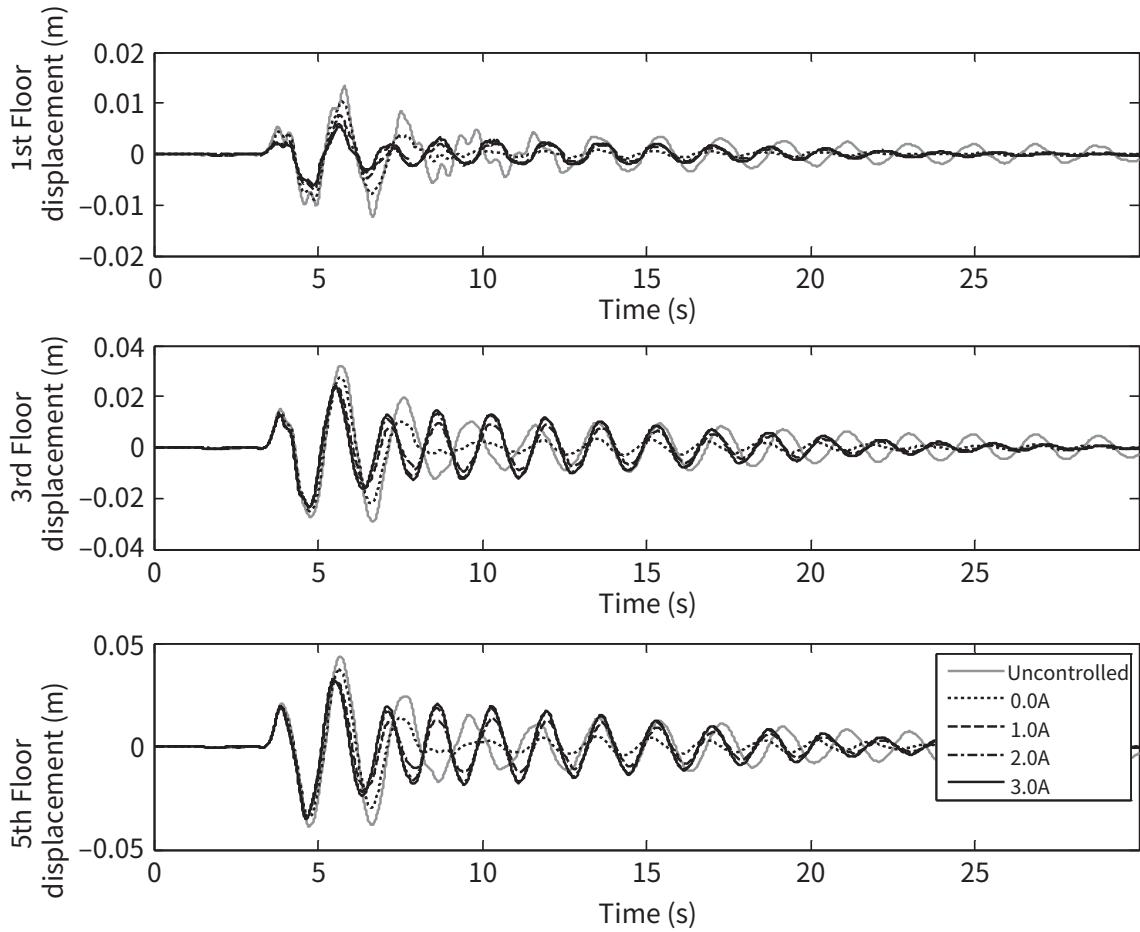


Figure 4.15: Experimental results, in the time domain under Northridge earthquake excitation at different applied currents.

In addition, the displacement responses in the frequency domain to four earthquake excitations at different applied currents are compared to the first floors, as shown in Figure 4.16. The figures show that the peak in the uncontrolled case appears at 0.52 Hz, corresponding to the fundamental frequency of the numerical substructure under consideration, but the peak is shifted to the vicinity of 0.62 Hz as the applied current increases. This small amount of frequency shifting is due to the stiffening effect of the MR damper used in this thesis as the applied current is increased.

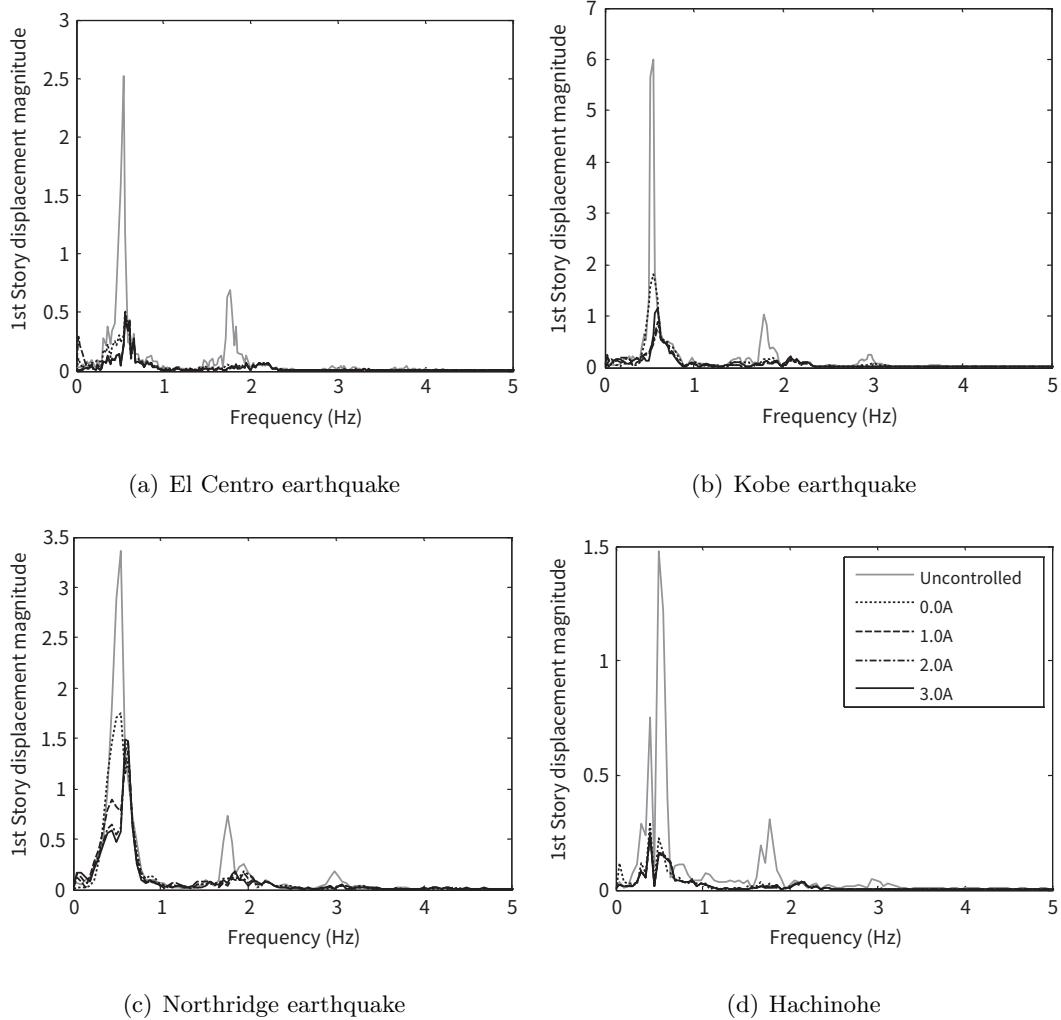


Figure 4.16: First-story displacements in the frequency domain at different applied currents.

#### 4.4.2 Semi-active Control Performance

Figure 4.17 shows a comparison between numerical and experimental results for the clipped-optimal control under El Centro earthquake excitation based on identified Bouc-Wen parameters. As shown in Figure 4.17, the experimental and numerical results show different currencies. Moreover, because the peak responses are very important in evaluating seismic performance, it is critical to have large differences between the experimental and numerical results. These results are due to the nonlinearity of MR dampers which varies in the numerical model corresponding to frequencies of the piston and nonlinear reaction velocity of input currents. Consequently, these results prove that using the hybrid testing method is more practical in non-linear damper models such as MR dampers.

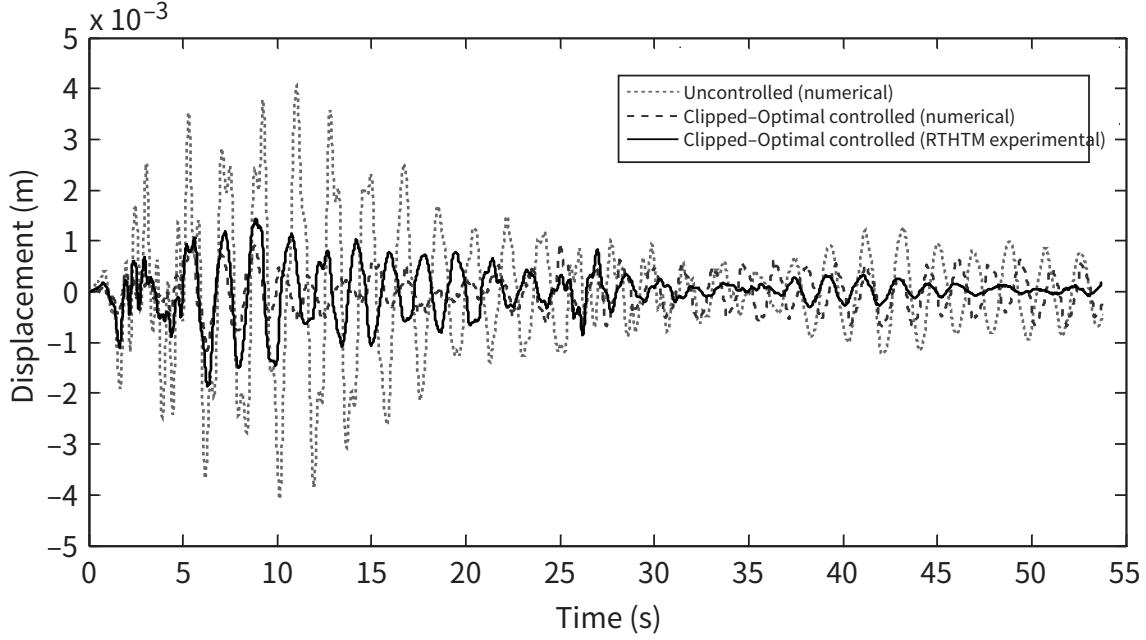


Figure 4.17: Comparison of numerical and experimental results under El Centro earthquake excitation.

Figure 4.18 shows a comparison between the first-floor displacements for the time histories of experimentally measured structural displacement responses to the excitation of three earthquake excitations, as the current applied to the MR damper is varied corresponding to each semi-active control algorithm. Significant control effects are observed between the uncontrolled and the passive control cases but are not observed in each semi-active control case of El Centro, as shown in Figure 4.18(a). In Figure 4.18(c), the clipped optimal control algorithms show remarkable performances in controlling displacement response.

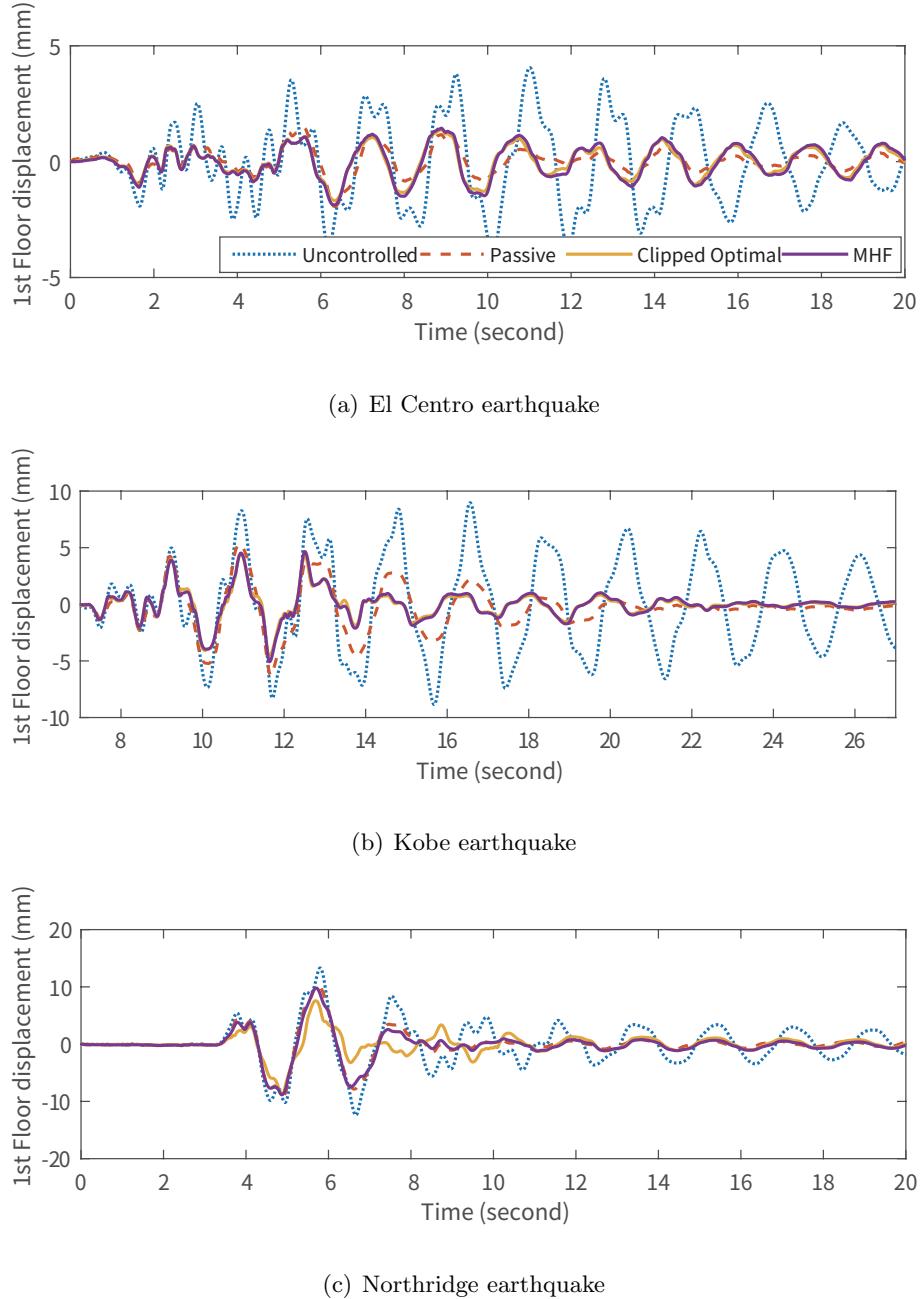


Figure 4.18: Passive and semi-active hybrid testing method experimental results(time domain).

Also, the displacement responses in the frequency domain to three earthquake excitations applied to uncontrolled, passive, and semi-active controlled hybrid testing method are compared to the first floor, as shown in Figure 4.19. The figures show that an overall best control performance is achieved when the clipped optimal control algorithms are applied to the structure. However, all the results show insignificant control performance against the passive control. These results are influenced by the long-period model structure having 0.52 Hz and a scaled-down excitation load due to the UTM capacity. For

the passive result, the peak response is shifted to the vicinity of 0.62 Hz in a semi-active controlled result. This small amount of frequency shifting is due to the stiffening effect of the MR damper used in this thesis as the applied current is increased for the passive control results.

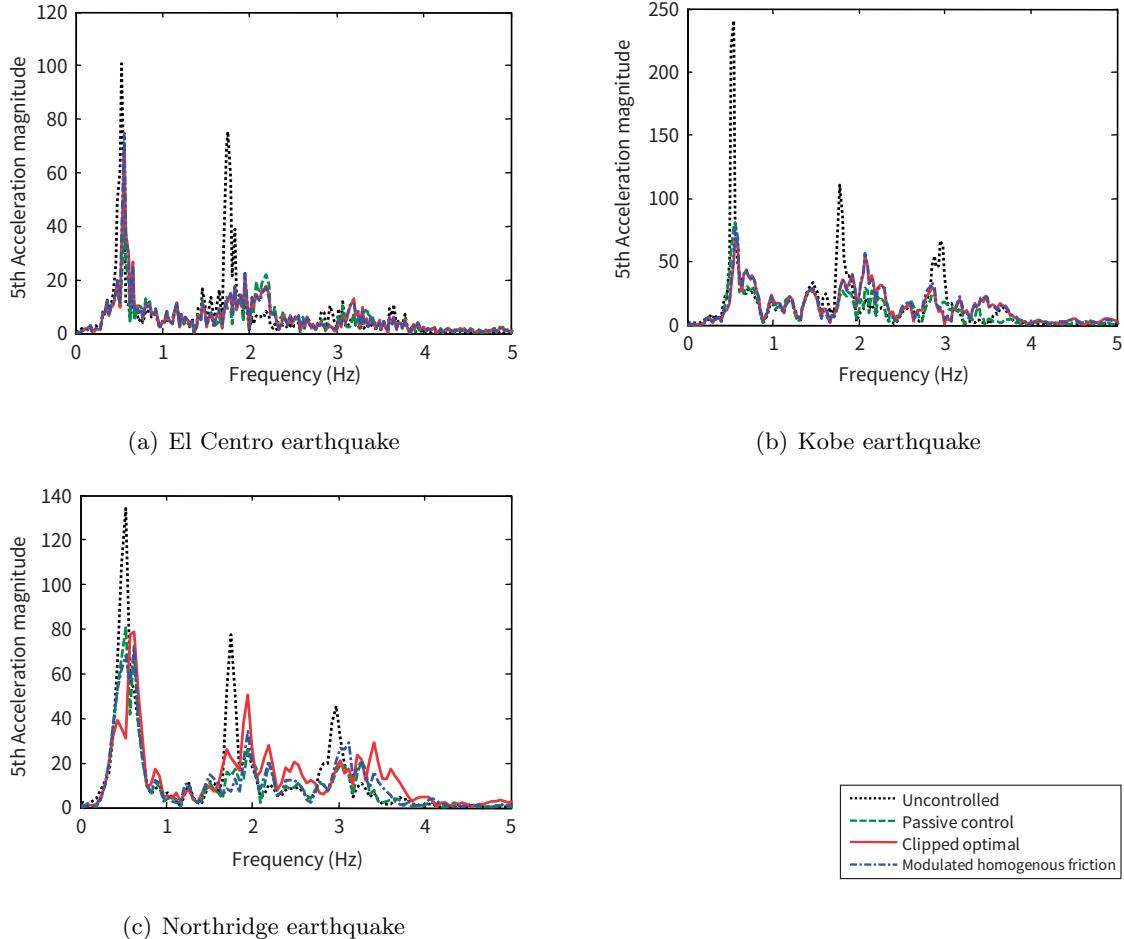


Figure 4.19: Passive and semi-active hybrid testing method experimental results(frequency domain).

## 4.5 Summary

In this chapter, the investigation of the hysteretic behavior of an MR damper and the seismic performance evaluation of a building structure, installed with an MR damper, are experimentally implemented using the hybrid testing method. In the tests, the building model that is identified from the force-vibration test results of a full-scale five-story building is adopted as a numerical substructure in this thesis. Also, an MR damper that corresponds to an experimental substructure is physically tested using a UTM. The Bouc-Wen model is used to calculate the control force of the MR damper used in this thesis, and its parameters are identified based on the experimental results from the hybrid testing

method, which used a sinusoidal wave as the ground input acceleration. The hybrid testing method is validated because the hybrid testing results from the sinusoidal and earthquake excitations and the corresponding analytical results agreed well with each other. In particular, the hybrid testing method is highly reliable with the impulse-like seismic excitation such as that of the Northridge earthquake. In order to compare the results obtained from the hybrid testing method and the numerical analysis, Bouc-Wen model parameters are identified by each input current. The results of this comparison show that the hybrid testing method is more practical than the numerical analysis due to the non-linear variations of the reaction velocity and excitation frequency. The experimental results from the hybrid testing method for the passive control show that structural responses did not decrease further by the excessive control force, but they decreased with the increase of the current applied to the MR damper. In addition, the semi-active controlled result shows an insignificant control performance compared to the passively controlled result. It seems that the passive control forces of the MR damper already reached the optimal friction force by the proportional shear force of the first floor.

# Conclusions

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In this thesis, the hybrid testing method (a substructuring technique, TLD-, TLCD- and TLMD-building structure hybrid testing method) for the shaking table test was proposed. The proposed testing technique adopts the upper part of the whole structure or nonlinear control devices as the experimental substructure, which corresponds to a physical test model. The lower part of the entire structure or whole building structure is modeled numerically. In order to verify the validity and accuracy of the proposed technique, a shaking table test was conducted. Also this thesis presents the design of excitation systems for simulating dynamic loads. The controller design of an actuator is presented to simulate the responses of a building structure subjected to such dynamic excitations as earthquakes and wind loads. The result of the study can be summarized as follows.

In chapter 2, a hybrid testing method is proposed. First, a substructuring test of a building structure using the hybrid testing method was performed by using a shaking table. Selecting a particular story of a shear-type building structure as an interface, the upper part of the target structure is divided into the experimental substructure, and the lower part is divided into numerical analytic substructure. The boundary load at the interface separated by the substructure is calculated in real time of the absolute acceleration response of the experimental model. The acceleration of the interface between two substructures is calculated by time history analysis of the numerical substructure and used as command signal of the shaking table controller. At this time, the shaking table controller that generates the command signal minimizes the error between the generated command signal and the shaking table motion for the excitation of the experimental substructure. The validity and accuracy of the shaking table substructure test of the hybrid testing method were confirmed by the experimental results and the numerical analysis results. Also, a hybrid testing method for evaluating the control performance of a building structure subjected to earthquake excitation of a tuned liquid type damper (TLD, TLCD, and TLMD), which is a nonlinear control device, was experimentally performed. This experiment is performed by using only control devices (TLD, TLCD, and TLMD), load cell and shaking table without physical building structure because this experiment performs divided into nonlinear control device as an experimental model and building model as an analytical model. The structural responses of the interaction system are calculated nu-

merically in real time using an analytical structure model, a given earthquake record, and a shear force generated by the TLD, TLCD, and TLMD, and the shaking table reproduces both the controlled and uncontrolled absolute acceleration of the TLD, TLCD, and TLMD installed floor by manipulating the feedback gain of the shear force signal measured by the load cell positioned between the TLD/TLCD/TLMD and the shaking table. In this thesis, the results of the hybrid test with those of the existing full-scale structural test are compared, and it is present that accurate results of the seismic performance of buildings with nonlinear vibration control system can be obtained by using hybrid test without a physical model. In this thesis, through not only TLD and TLCD which is verified through previous studies but also TLMD which is newly proposed to reduce bidirectional responses of building structures is evaluated by using a hybrid testing method, a hybrid testing method is beneficial for that the recently proposed control device can be assessed experimentally in the design stage.

In chapter 3, a design of an actuator for simulating the wind-induced responses of a building structure is presented. The actuator force is calculated by using the inverse transfer function of a target structural responses to the actuator. Filter and envelope function are used such that the error between wind and actuator induced responses is minimized by preventing the actuator from the exciting, unexpected modal responses and initial transient responses. The analyses result from a 76-story benchmark building problem in which wind load obtained by wind tunnel test is given, indicate that the actuator installed at a particular floor can approximately embody the structural responses induced by the wind load applied to each floor of the structure. The actuator designed by the proposed method can be effectively used for evaluating the wind response characteristics of a building structure in use and for obtaining an accurate analytical model of the building under wind load. Also, a full scale forced vibration test simulating earthquake response is implemented by using a hybrid mass damper. The finite element model of the real-scaled building structure was analytically constructed, and the model was updated using the results experimentally measured by the forced vibration test. Pseudo-earthquake excitation tests showed that the hybrid mass damper induced floor responses coincided with the earthquake induced ones which were numerically calculated based on the updated FE model. Real-scaled MR damper was installed to verify the effectiveness of the pseudo-earthquake experiment in a real scale building. The MR damper-based control systems are realized when an MR damper is designed by deriving a suboptimal design procedure considering optimization problem and magnetic analysis, and then a damper with the capacity of 1.0 ton is manufactured. In the experiments, a linear active mass driver and the linear shaker seismic simulation testing method are used to excite the building structure to match the full-scale building vibrate as if the building undergoes an earthquake. Under the four historical

earthquakes and one filtered artificial earthquake, the performance of the semi-active control algorithms including the passive optimal case is experimentally evaluated. From the experimental results, one can conclude that the Lyapunov and semi-active neuro-control algorithms are appropriate in reducing accelerations of the structural system, and the passive optimal case and the maximum energy dissipation algorithm show the excellent performance in reducing the first-floor displacement.

In chapter 4, The MR damper with the capacity of 1-ton was used from chapter 3, a hybrid testing method of the real-scaled building with semi-active controlling MR damper implemented in the chapter 2 was applied by hybrid testing method technique. This study presents the quantitative evaluation of the seismic performance of a building structure installed with an MR damper using a hybrid testing method as described in chapter 2. A building model is identified from the forced vibration testing results of a full-scale five-story building in chapter 3 and is used as the numerical substructure, and an MR damper as an experimental substructure is physically tested using a universal testing machine (UTM). First, the force required to drive the displacement of the story, at which the MR damper is located, is measured from the load cell attached to the UTM. The measured force is then returned to a control computer to calculate the response of the numerical substructure. Finally, the experimental substructure is excited by the UTM with the calculated response of the numerical substructure. The hybrid testing method implemented in this thesis is validated because the hybrid testing results obtained by application of sinusoidal and earthquake excitations and the corresponding analytical results obtained using the Bouc-Wen model as the control force of the MR damper with respect to input currents are in good agreement. Also, the results from the hybrid testing method for the passive - on and -off control show that the structural responses did not decrease further by the excessive control force, but decreased due to the increase of the current applied to the MR damper. Also, two semi-active control algorithms (modulated homogeneous friction and the clipped-optimal control algorithms) are applied to the MR damper to optimally control the structural responses. To compare the hybrid testing method and numerical results, Bouc-Wen model parameters are identified for each input current. The results of the comparison of experimental and numerical responses show that it is more practical to use the hybrid testing method in semi-active devices such as MR dampers. The test results indicate that a control algorithm can be experimentally applied to the MR damper using the hybrid testing method. This thesis also provides a discussion on each algorithm concerning the seismic performances.

비선형 감쇠장치가 설치된 건축구조물의  
내진 및 내풍 성능 평가를 위한

하이브리드 실험기법 및 가진시스템 설계

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지도교수 : 민 경 원

지진이나 바람하중과 같은 외부하중으로 인해 가진되는 건축구조물의 동적 응답 특성을 정확하게 식별하는 것은 건축구조물의 안전성과 사용성 평가 뿐만 아니라 내진 또는 내풍 설계에 사용되는 해석 모델의 검증에 필수적이다. 건축구조물의 해석 모델을 구축하기 위하여 정확한 입출력 관계를 설명하는 시스템 행렬을 구성하는 시스템 식별 분야에서는 입력이 주요 모드를 가진시키기에 충분한 에너지를 가져야하고 구조물의 정보를 포함하는 양질의 응답인 출력을 측정하는 것이 매우 중요하다. 그러나 내진 성능 평가를 위하여 실물 크기의 건물을 진동대위에 설치하여 실험하기에 시스템을 구축하기에 매우 어려우며, 내풍 성능 평가를 위하여 실물 크기의 고층 건축물의 실험실에서 풍동 테스트하는 방법은 거의 불가능하다. 따라서 일반적으로 비선형 혹은 비탄성거동이 예측되는 부분구조를 나누어 실험하거나 이를 대체하기 위해 편심가진기나 선형질량가진기등의 가진기를 실물 크기의 사용중 건축물에 설치하고 주요 모드를 가진하여 건축물의 동적 성능 평가를 하거나 시스템 식별을 위한 수단으로 사용된다. 그러나 정확한 내진 및 내풍 성능을 실험적으로 평가하기 위해서 지진하중의 특성과 바람하중의 특성이 모두 충분히 실험에 반영되도록 건축구조물을 가진해야 할 필요가 있다.

따라서 본 논문은 구조계를 분할하여 실험하는 방식의 하이브리드 실험법과 실물 크기의 건축구조물을 가진하여 동적특성을 식별하기 위한 가진 시스템 설계 기법을 제안한다. 첫째로 하이브리드 실험법은 전체 구조계에서 비선형성이 강한 파트를 실험모델 그리고 선형거동 할 것이라 예측되는 파트를 해석모델로 두고 실시간으로 그 응답을 받아 실험하는 기법이다. 이러한 실험 방법은 실험모델과 해석모델의 인터페이스를 선정해야 하며 해석모델의 경계면은 실험모델의 가진력으로 작용하도록 설계하는 실험 시스템을 구축하여야 한다. 본 논문은 진동대를 통해 구조계를 실험구조계와 해석구조계로 분리하여 하이브리드 실험을 하는 부분구조실험법과 비선형 진동제어장치인 동조액체감쇠기, 동조액체기동감쇠기 그리고 동조액체질량감쇠기를 실험모델로 실험하고 건축구조물을 해석모델로 계산하는 하이브리드 실험을 설계하고 실험하여 그 성능을 검증하였다. 더 나아가 비선형성이 강한 1톤급 층간형 감쇠기인 MR 감쇠기를 UTM에 장착하여 MR 감쇠기가 설치된 건물의 내진성능 평가를 수행하고 MR감쇠기에 준능동 알고리즘을 적용하여

내진 제어 성능을 실험적으로 평가하였다. 이는 본 실험 방법을 통해 불특정 비선형성을 가진 제어장치의 최적의 알고리즘 혹은 장치 자체의 성능을 실험적으로 찾아서 적용할 수 있는 가능성과 정량적 근거를 제시할 수 있었다. 두 번째로 동적 응답을 구현하는 가진 시스템 설계법은 실물크기 사용 중 건물에서 선형 질량가진기나 능동형 질량감쇠기 등을 특정 층에 설치하고 유사지진 또는 유사 바람하중 응답을 구현하는 시스템이다. 본 논문에서는 풍하중 응답 구현을 위해 건물 모델과 풍하중 데이터가 있는 76층 벤치마크 건물을 대상으로 수치해석적으로 검증하였으며, 지진하중 구현을 위해 실물규모의 5층 건물의 HMD를 가진하여 지진하중을 구현하였다. 또한 이를 제어하는 준능동 감쇠장치인 1톤급 MR 감쇠기를 설치하여 실물규모 건물에서 MR감쇠기의 지진에 대한 준능동 제어 성능을 평가하였다. 이러한 기법은 실물크기 건축구조물의 내진 및 내풍성능 평가와 비선형 제어장치가 설치된 건물의 내진 및 내풍성능 또한 평가하는데 사용될 수 있다.

**주요어:** 하이브리드 실험, 부분구조 실험, 비선형 진동제어 장치, 동조 액체 감쇠기, 동조 액체기 등 감쇠기, 동조 액체 질량 감쇠기, 제어-구조 상호력, 가진기 동특성 보상, 능동 질량 감쇠기, 복합형 질량 감쇠기, 가진 제어기 설계, 76층 벤치마크 건축구조물, 강제 진동 실험, 시스템 식별, 유한요소 모델 업데이트, 자기유변유체 감쇠기, 준능동 제어 알고리즘.

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# Curriculum Vitæ

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### Education

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2005–2007	Master of science, dept. of architectural engineering School of Architecture, Dankook University, Republic of Korea
2007–2017	Ph.D. School of Architecture, Dankook University, Republic of Korea

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## Work Experience

2009.04 - 2015.05 한국유지관리(주) 부설기업연구소 : R&D > U-biz, DAQ developemtn, Measurement of civil infra-structure, Structural health monitoring

2015.08 - 현재 (주)승화기술정책연구소 기술연구소 : R&D > Computer vision/IoT, M2M Platform development/Full stack engineering/Deep learning/Railway projects

## Awards

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2012 The 19th Science and Technology Best Paper Award  
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## Publications

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#### Papers of submission (SCI)

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- Heon Jae Lee, Seok Jun Moon, and **Eun Churn Park**, “Vibration and Control of Smart Material Systems: Experimental Investigation of Semiactive Control Systems

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- **Eunchurn Park**, Yu-Seung Kim, Joong-Yub Lee, Jun-Sung Choi, Yeon-Back Jung, Won-Kyun Seok, Kwang-Soo Jung, Soon-Jeon Park, Joo-Ho Lee, “Study on Vertical Alignment Maintenance Technique using GNSS in Skyscraper”, *2009 International Symposium on GPS/GNSS*; 11/2009
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### Research projects

- 콘크리트궤도 도상/노반결합 상태평가 시스템 개발, 주관:한국철도공사 2016-2017
- 도시 인프라 자산관리 플랫폼과 서비스 모델 개발, 주관:(주)승화기술정책연구소 2015-2016
- M2M 기반 지하공간(지하철) 재난대응 대화형 스마트 네트워크 시스템 개발, 주관:(주)승화기술정책연구소 2015-2017
- 진동특성을 이용한 LCD 생산로봇 및 부품 사전 진단 시스템 개발, 주관:한국유지관리(주) 2011-2012
- 감지형앵커를 이용한 사면 보강 및 모니터링 기술 개발, 주관:한국유지관리(주) 2010-2011

- 교량 및 지반보강을 위한 스마트 텐던 기술의 사업화를 위한 추가기술개발 및 검증,  
주관:한국유지관리(주) 2010
- 철골조 시설물의 붕괴를 방지하는 설치 용이한 경제적인 보강기구 개발, 주관:단국  
대학교 산학협력단 2006
- 다자유도 진동대 및 가력기를 이용한 2방향 동조식 액체형 댐퍼의 풍응답 저감 연구,  
주관:단국대학교 20080901-20110831
- 대형 비탄성구조물 분산 제어시스템 설계기술 개발, 주관:단국대학교 20070301-  
20100228
- 모듈러 유닛 구조물 구조성능 평가 및 운송기술 개발, 주관:(사)대한건축학회 2006-  
2007
- 초대형 구조물의 내풍 및 내진성능향상을 위한 준능동 제어시스템 개발(The Devel-  
opment of a Semi-active Control System for Large-scale Structures under Wind and  
Seismic Loads), 주관:단국대학교 2005-2006

#### Teaching experience

- 2011.09.29      망조정과 외란보정기법의 RTK      서울시  
GNSS를 적용한 고층구조물의 거  
푸집 연직도 관리기술
- 2012.09.01–      공학수치해석 (Numerical Analy-      단국대학교 (Dankook University)  
2013.02.28      sis)

#### Intellectual property

##### - NT

- 망조정과 외란보정기법의 RTK GNSS을 적용한 고층 구조물의 거푸집 연직도 관리  
기술, 고시 : 국토해양부고시 제2011-313호

##### - Patents

- 철도차량의 승차감 분석 시스템(System for ride comfort analysis of Railway vehicle)
- 철도차량의 주행 안정성 분석 시스템(System for driving stability analysis of Railway  
vehicle)
- 이상진단 사전감시 방법(Method for preliminary surveillance of failure diagnosis)
- 유에스엔 기반 지능형 교량 모니터링 및 안전성 평가 시스템(System for intelligent  
monitoring and safety evaluation of bridge based on USN)

## Relevant Skills

### Computer Programming

- Mathworks MATLAB, Mathworks Simulink, National Instruments LabVIEW, Visual Studio (C#, C++)
- Python, Node.js, Javascript, SQL, NoSQL(MongoDB), Shell script of Linux(Debian)

## Referees

**Kyung-Won Min, Ph.D.** : Professor, Advisor

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