

# NEW RUNGE–KUTTA ALGORITHMS FOR NUMERICAL SIMULATION IN DYNAMICAL ASTRONOMY

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**Abstract.** Some new Runge–Kutta and Runge–Kutta–Nystrom algorithms are presented for the solution of ordinary differential equations of the initial value type. The methods are compared with others in integrating the equations of motion of the two body problem and are shown to offer advantages in efficiency. It is also demonstrated that the new methods can be ‘tuned’ to achieve some measure of global error control.

## 1. Introduction

Runge–Kutta embedding is now a popular method for the numerical solution of ordinary differential equations. Being a single step procedure it is relatively stable and hence particularly suitable for the simulation of long-period evolution in dynamical astronomy. We present here some formulae of the Runge–Kutta (RK) and Runge–Kutta–Nystrom (RKN) types which seem to have advantages in efficiency over those currently in use.

For the first order initial value problem

$$\dot{x} = f(t, x), \quad x(t_0) = x_0, \quad (1)$$

the embedded RK  $p(p+1)$  (or RK  $p+1(p)$ ) algorithm has the form

$$\left. \begin{aligned} X(t_0 + h) &= x_0 + \sum_{i=1}^{v_p} b_i k_i, \\ \hat{X}(t_0 + h) &= x_0 + \sum_{i=1}^{v_{p+1}} \hat{b}_i k_i, \end{aligned} \right\} \quad (2)$$

where

$$k_i = hf(t_0 + c_i h, x_0 + \sum_{j=1}^{i-1} a_{ij} k_j), \quad i = 1, 2, \dots, v_{p+1}.$$

If  $x(t_0 + h)$  represents the true value of  $x$  after step  $h$ , then

$$x(t_0 + h) = X + O(h^{p+1}) = \hat{X} + O(h^{p+2}), \quad (3)$$

and so the difference  $(\hat{X} - X)$  represents an approximation for the leading error term of the lower order formula which can be used to estimate a new or subsequent step size for local error control. Fehlberg (1968, 1969) has established a number of formulae of type (2) ( $p \leq 8$ ). The coefficients  $b_i$ ,  $\hat{b}_i$ ,  $a_{ij}$ , and  $c_i$  must satisfy equations of condition that may be obtained from Taylor expansions; these are given by

Butcher (1964) for  $p \leq 8$ . To obtain embedded formulae of type (2) one must solve two such sets of equations simultaneously.

Fehlberg presents his formulae as  $p$ th order methods, i.e., the value of  $x(t)$  is estimated by  $X$ . We prefer to use  $\hat{X}$  to estimate  $x(t)$  hence implementing the formulae in  $(p + 1)$ th order mode where possible.

Fehlberg's formulae have been developed to give very small leading truncation error terms for the  $p$ th order formula. This can result in an underestimate of the actual truncation error since the higher order terms could be dominant for moderate  $h$ . For example the error of the lower order formula is given by

$$E = h^{p+1} \sum_{i=1}^{n_{p+1}} \alpha_i^{(p+1)} F_i^{(p+1)} + h^{p+2} \sum_{i=1}^{n_{p+2}} \alpha_i^{(p+2)} F_i^{(p+2)} + O(h^{p+3})$$

where the  $F_i^{(p+1)}$  are the elementary differentials of order  $p + 1$  of  $f$  and  $\alpha_i^{(p+1)}$  are the leading truncation coefficients (Stetter, 1971). If the  $\alpha_i^{(p+1)}$  are minimized then the  $\alpha_i^{(p+2)}$  could become dominant if  $h$  is not small. To take a specific case, the Fehlberg RKF4 (Fehlberg, 1969) where  $p = 4$ , we have

$$\|\alpha_t^{(6)}\|_2 / \|\alpha^{(5)}\|_2 = 3.15.$$

Assuming that the elementary differentials  $F_i^{(5)}$  and  $F_i^{(6)}$  do not differ in magnitude then unless  $h < 1/3.15$  the principal error term is not dominant.

It seems more appropriate to minimize the leading truncation terms for the higher order formula thus making  $\hat{X}$  a better estimate of  $x(t)$ . This may not, however, make  $(\hat{X} - X)$  a more realistic error estimate.

In order to provide flexibility in the new methods the coefficient  $b_{\nu_p}$  has been left free. The leading error term of the lower order formula can be made proportional to this coefficient (see (6)) and therefore the value of  $(\hat{X} - X)$  can be tuned to give reliable step-size and error control.

The above comments are also applicable to RKN formulae which may be used directly to solve the second order equation

$$\ddot{x} = f(t, x), \quad \text{with} \quad x(t_0) = x_0 \quad \text{and} \quad \dot{x}(t_0) = \dot{x}_0. \quad (4)$$

The embedded RKN  $p + 1(p)$  algorithm may be written in the form

$$\left. \begin{aligned} X(t_0 + h) &= x_0 + \dot{x}_0 h + h^2 \sum_{i=1}^{\nu_p} b_i f_i \\ \hat{X}(t_0 + h) &= x_0 + \dot{x}_0 h + h^2 \sum_{i=1}^{\nu_{p+1}} \hat{b}_i f_i \\ \hat{\dot{X}}(t_0 + h) &= \dot{x}_0 + h \sum_{i=1}^{\nu_{p+1}} \hat{b}_i f_i \end{aligned} \right\} \quad (5)$$

where

$$f_i = f(t_0 + c_i h, x_0 + \dot{x}_0 c_i h + h^2 \sum_{j=1}^{i-1} a_{ij} f_j), \quad i = 1, \dots, \nu_{p+1}.$$

Using (3) we note that  $(\hat{X} - X)$  can be used for error estimation and step-size control. In addition to (5) (or alternatively) the embedding could be achieved for  $\dot{x}$ . The equations of condition for formulae of type (5) have been presented by Fehlberg (1972) who has also derived similar formulae up to order 8.

2. Some New RK and RKN Formulae

In Tables I and II we reproduce fourth and fifth order embedded RK formulae. The parameter  $\lambda$  may be chosen arbitrarily but suitable values in this case are found

TABLE I  
Runge-Kutta 4(3)*T*

$c_i$					$a_{ij}$	$\hat{b}_i$	$b_i$
0						$\frac{1}{6}$	$\frac{1}{6}$
$\frac{1}{2}$	$\frac{1}{2}$					$\frac{1}{3}$	$\frac{1}{3}$
$\frac{1}{2}$	0	$\frac{1}{2}$				$\frac{1}{3}$	$\frac{1}{3}$
1	0	0	1			$\frac{1}{6}$	$\frac{1}{6}-\lambda$
1	$\frac{1}{6}$	$\frac{1}{3}$	$\frac{1}{3}$	$\frac{1}{6}$		0	$\lambda$

TABLE II  
Runge-Kutta 5(4)*T*

$c_i$					$a_{ij}$	$\hat{b}_i$	$b_i$
0						$\frac{17}{192}$	$\frac{17}{192}$
$\frac{1}{8}$	$\frac{1}{8}$					0	0
$\frac{1}{4}$	0	$\frac{1}{4}$				$\frac{64}{231}$	$\frac{64}{231}$
$\frac{4}{9}$	$\frac{196}{729}$	$-\frac{320}{729}$	$\frac{448}{729}$			$\frac{2187}{8960}$	$\frac{2187}{8960}$
$\frac{4}{5}$	$\frac{836}{2875}$	$\frac{64}{575}$	$-\frac{13\,376}{20\,125}$	$\frac{21\,384}{20\,125}$		$\frac{2875}{8448}$	$\frac{2875}{8448}$
1	$-\frac{73}{48}$	0	$\frac{1312}{231}$	$-\frac{2025}{448}$	$\frac{2875}{2112}$	$\frac{1}{20}$	$\frac{1}{20}-\lambda$
1	$\frac{17}{192}$	0	$\frac{64}{231}$	$\frac{2187}{8960}$	$\frac{2875}{8448}$	$\frac{1}{02}$	$\lambda$

TABLE III  
Runge-Kutta-Nystrom 7(6)*T*

$c_i$	$a_{ij}$						$\hat{b}_i$	$b_i$	$\hat{\hat{b}}_i$	
0							$\frac{1}{20}$	$\frac{1}{20}$	$\frac{1}{20}$	
$\frac{1}{10}$	$\frac{1}{200}$						0	0	0	
$\frac{1}{5}$	$\frac{1}{150}$	$\frac{1}{75}$					0	0	0	
$\frac{3}{8}$	$\frac{171}{8192}$	$\frac{45}{4096}$	$\frac{315}{8192}$				0	0	0	
$\frac{1}{2}$	$\frac{5}{288}$	$\frac{25}{528}$	$\frac{25}{672}$	$\frac{16}{693}$			$\frac{8}{45}$	$\frac{8}{45}$	$\frac{16}{45}$	
	$\frac{7-\sqrt{21}}{14}$	$\frac{1003-205\sqrt{21}}{12348}$	$\frac{-25}{90552}^{(751}$ $-173\sqrt{21})$	$\frac{25}{43218}^{(624}$ $-137\sqrt{21})$	$\frac{-128}{237699}^{(361}$ $-79\sqrt{21})$	$\frac{3411-745\sqrt{21}}{24696}$	$\frac{7}{360}^{(7+\sqrt{21})}$	$\frac{7}{360}^{(7+\sqrt{21})}$	$\frac{49}{180}$	
	$\frac{7+\sqrt{21}}{14}$	$\frac{793+187\sqrt{21}}{12348}$	$\frac{-25}{90552}^{(331}$ $+113\sqrt{21})$	$\frac{25}{43218}^{(1044}$ $+247\sqrt{21})$	$\frac{-128}{9745659}^{(14885}$ $+3779\sqrt{21})$	$\frac{3327+797\sqrt{21}}{24696}$ $\frac{-(581+127\sqrt{21})}{1722}$	$\frac{7}{360}^{(7-21)}$	$\frac{7}{360}^{(7-\sqrt{21})}$	$\frac{49}{180}$	
1	$\frac{-(157-3\sqrt{21})}{378}$	$\frac{25(143-10\sqrt{21})}{2772}$	$\frac{-25}{3969}^{(876}$ $+55\sqrt{21})$	$\frac{1280}{596673}^{(913}$ $+18\sqrt{21})$	$\frac{-1}{2268}^{(1353}$ $+26\sqrt{21})$	$\frac{7}{4428}^{(1777}$ $+377\sqrt{21})$	$\frac{7(5-\sqrt{21})}{36}$	0	$-\lambda$	$\frac{1}{20}$
1	$\frac{1}{20}$	0	0	0	$\frac{8}{45}$	$\frac{7(7+\sqrt{21})}{360}$	$\frac{7(7-\sqrt{21})}{360}$	0 0	$+\lambda$	0

to be  $1/10$  for the RK4(3) $T$  and  $1/60$  for the RK5(4) $T$ . In each case the final function evaluation at the  $n$ th step is the same as the first at the  $(n + 1)$ th step. Thus the effective number of function evaluations per step are four and six respectively. The basis of the RK4(3) $T$  is the familiar classic RK4 of Kutta. This idea has been used by Fehlberg (1972) and Bettis (1973) with regard to RKN methods. Zonneveld (1970) also uses an extra function evaluation in this manner for calculation of the last Taylor series term which is then used for step-size control.

Apart from  $\lambda$ , the RK5(4) $T$  can be expressed in terms of  $c_2$  and  $c_3$ , and these have been chosen to give small error terms for  $\hat{x}$ . Thus for  $p = 5$ ,  $\|\alpha^{(6)}\|_2 = 1.12 \times 10^{-3}$  compared with  $3.36 \times 10^{-3}$  for the Fehlberg RKF4. For  $p = 4$ ,  $\|\alpha^{(6)}\|_2/\|\alpha^{(5)}\|_2 = 1.05$  and so the principal error term in the lower order formula will generally be dominant.

Table III shows a seventh order RKN formula;  $\lambda = 1/20$  meets our requirements in this case. The leading local truncation error estimates in the lower order formulae are as follows:

$$\left. \begin{aligned} \text{RK4(3)}T: \quad T &= \lambda(k_4 - k_5) \\ \text{RK5(4)}T: \quad T &= \lambda(k_6 - k_7) \\ \text{RKN7(6)}T: \quad T &= \lambda h^2(f_8 - f_9). \end{aligned} \right\} \quad (6)$$

Following Hull *et al.* (1972) the  $n$ th step-size  $h_n$  is given by

$$h_n = 0.9h_{n-1}(\varepsilon/T_{\max})^{1/p}, \quad (7)$$

where  $p$  is the order of the lower order formula,  $\varepsilon$  is the tolerated local error per *unit step*, and  $T_{\max}$  is given by (6). If it is required to control error per *step* the exponent in (7) should be  $1/(p + 1)$ . The step would be rejected if  $T_{\max} > \varepsilon$  and recomputed with  $h$  given by (7).

### 3. Some Numerical Comparisons

As a suitable test problem we have considered the equations of two-body motion

$$\left. \begin{aligned} \ddot{x} &= -x/r^3, & x(0) &= 1 - e, & \dot{x}(0) &= 0, \\ \ddot{y} &= -y/r^3, & y(0) &= 0, & \dot{y}(0) &= \{(1 + e)/(1 - e)\}^{1/2}, \end{aligned} \right\} \quad (8)$$

where

$$r^2 = x^2 + y^2,$$

which are satisfied by an elliptic orbit of eccentricity  $e$  and major semi axis unity. Five values of  $e$  have been used, namely 0.1 (0.2) 0.9, which will be referred to as cases 1 to 5. For the application of RK methods equations (8) were reduced to four first order equations. In every case equations (8) were integrated from  $t = 0$  to



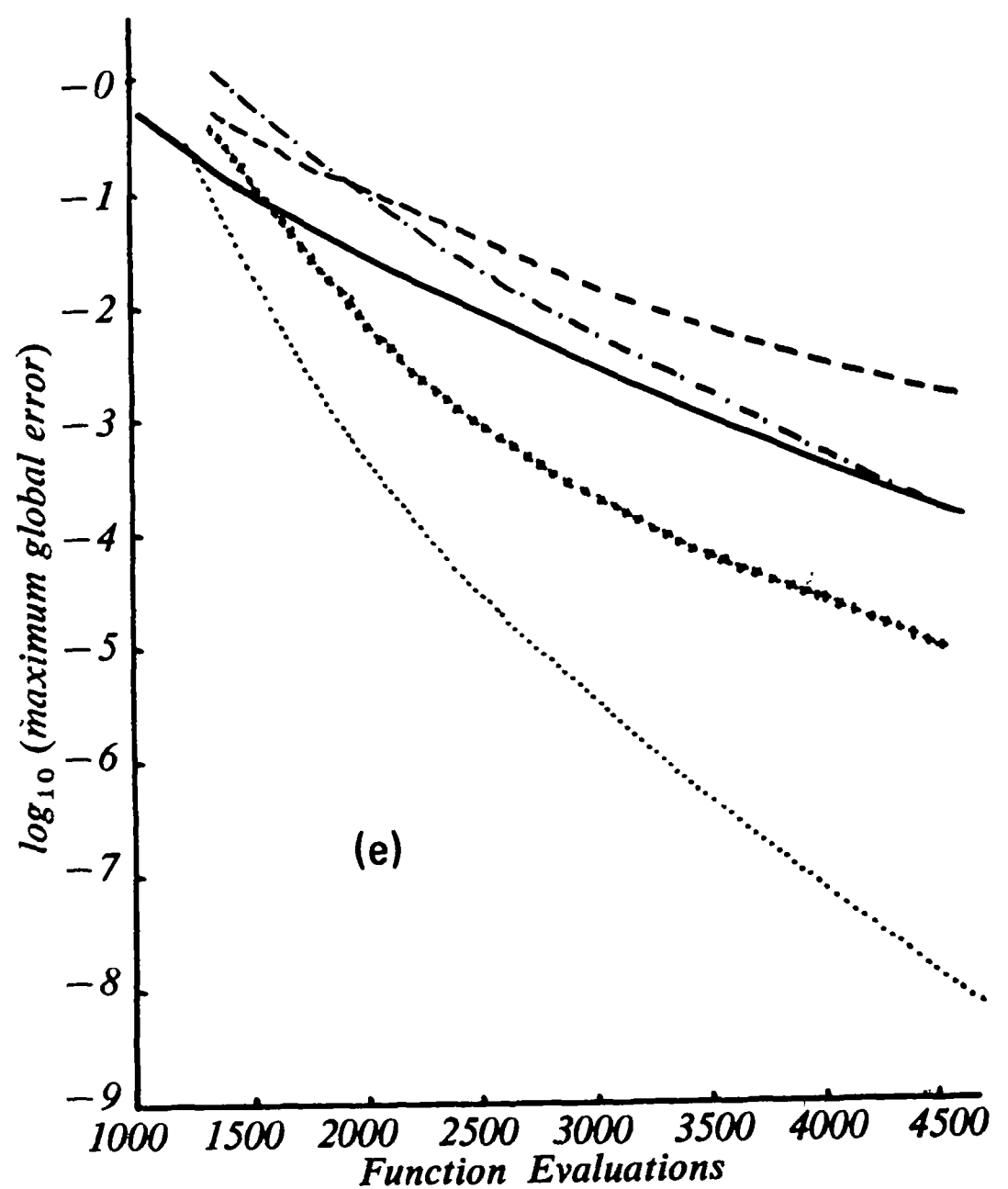
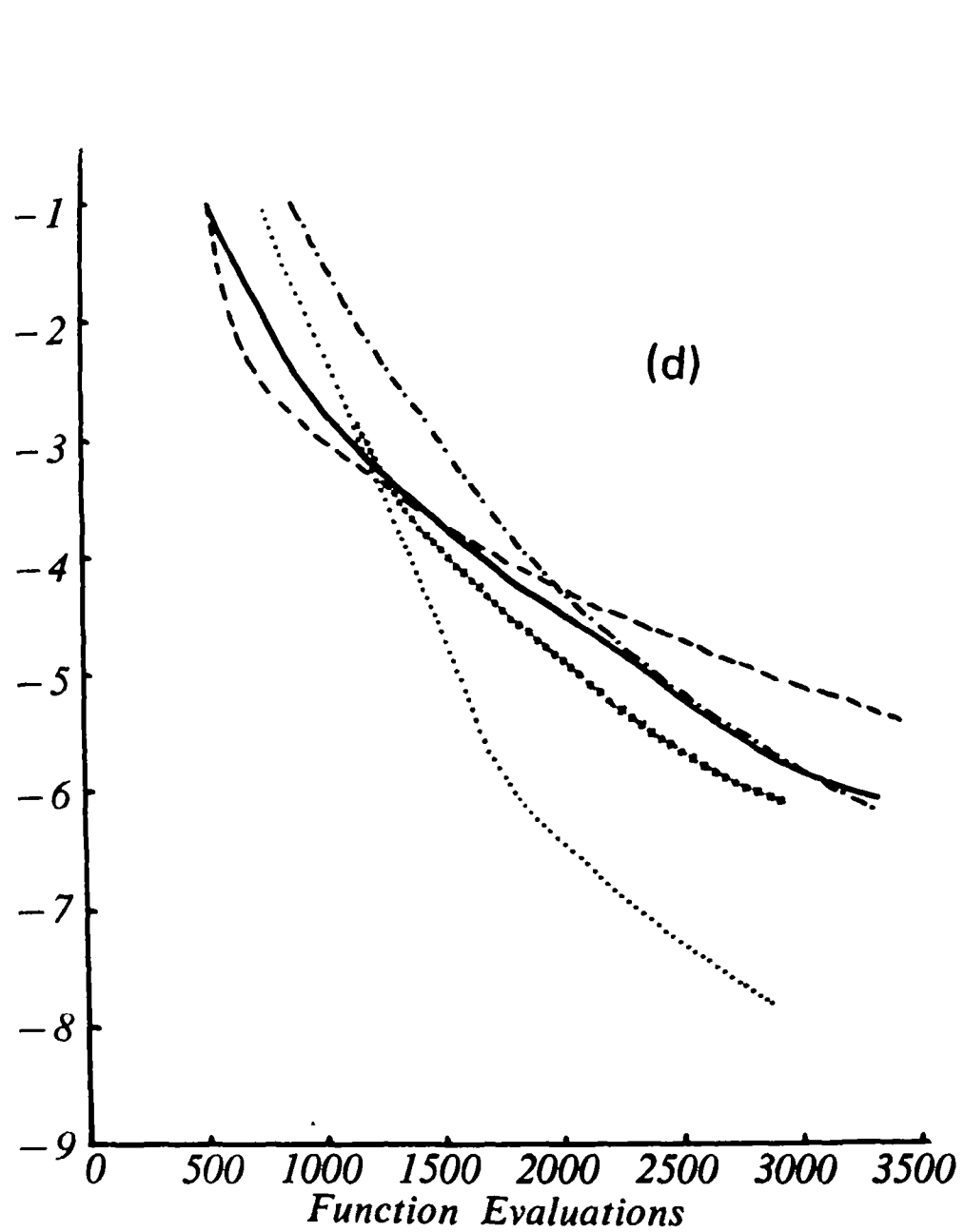
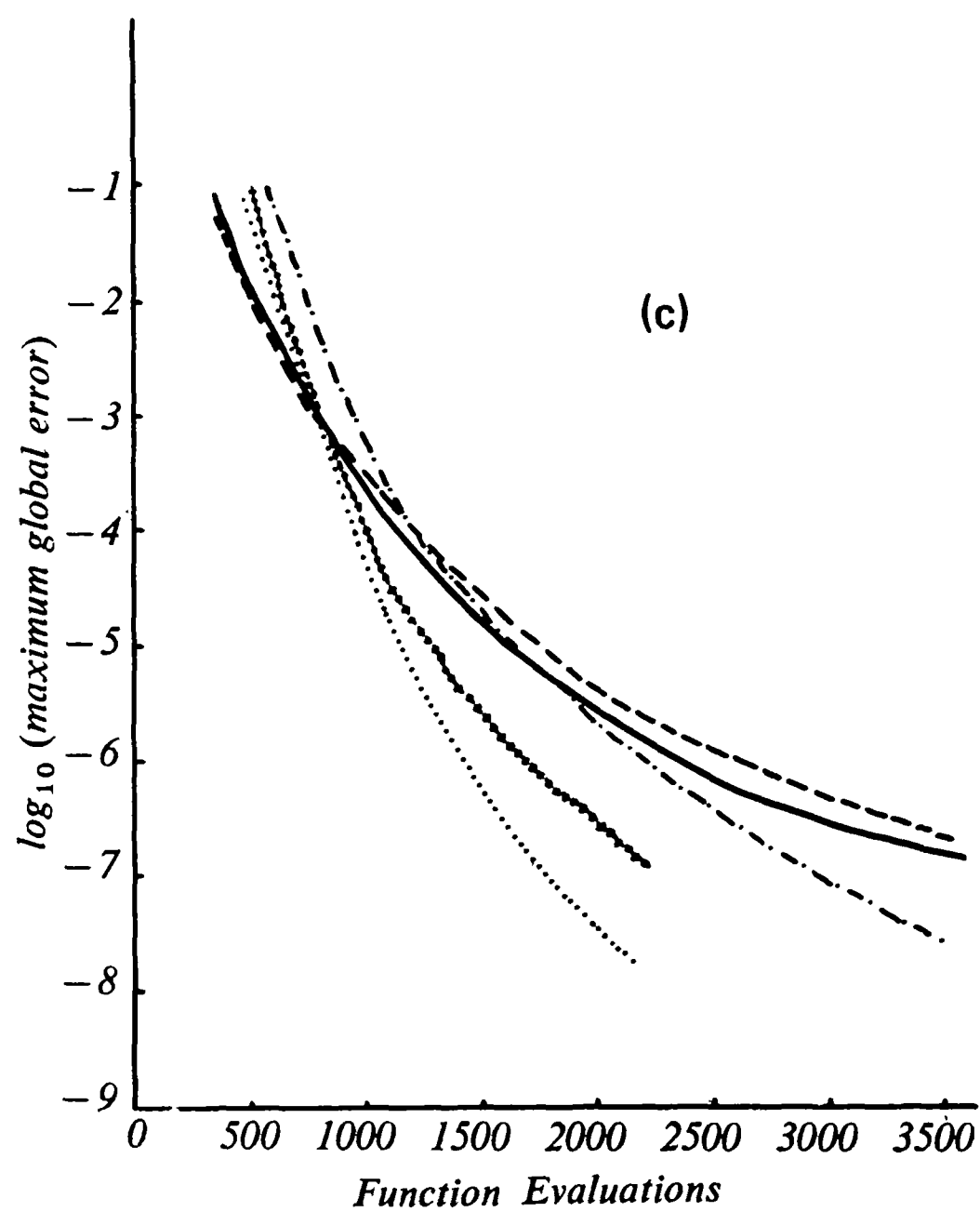


Fig. 1c-e.

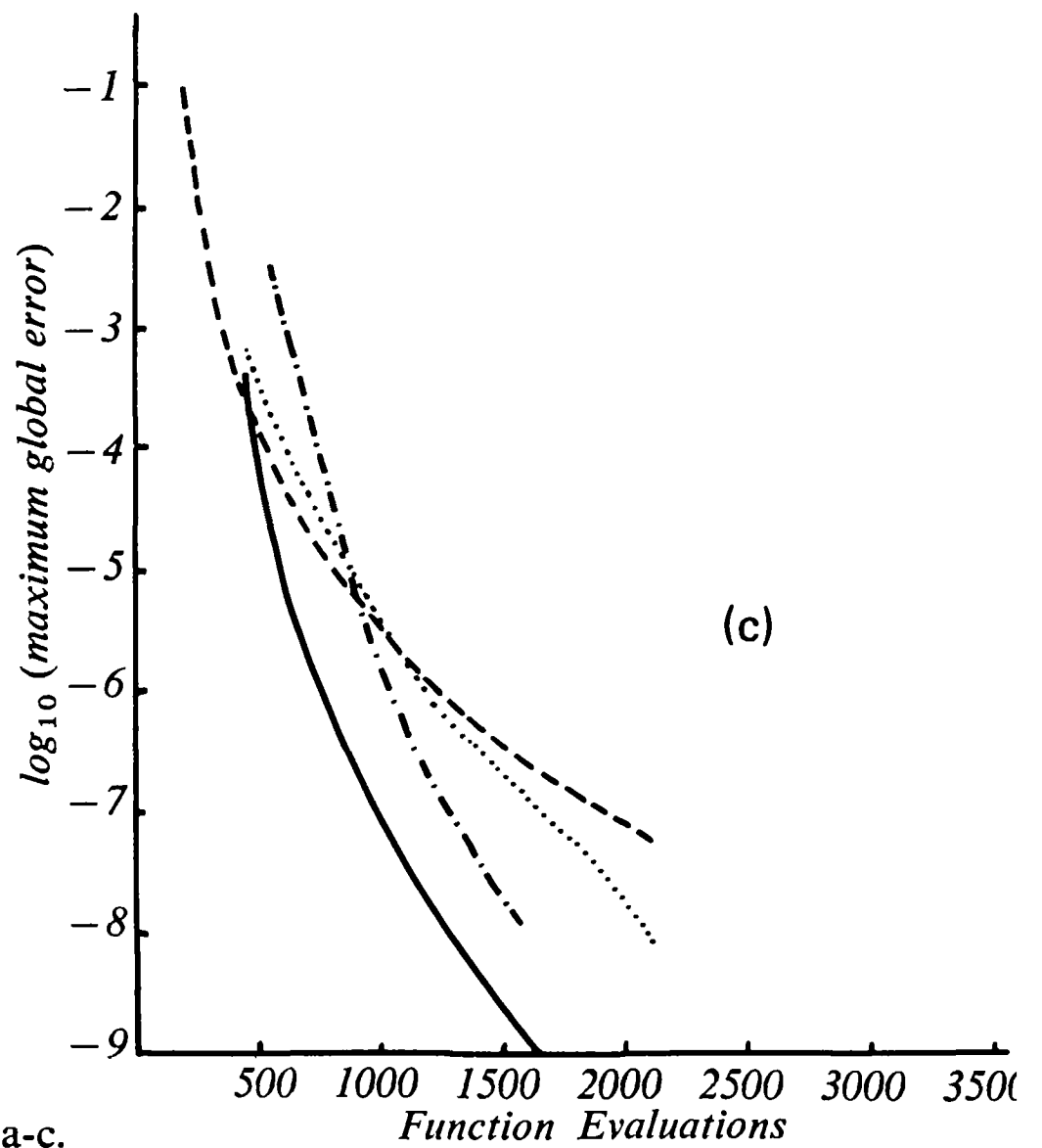
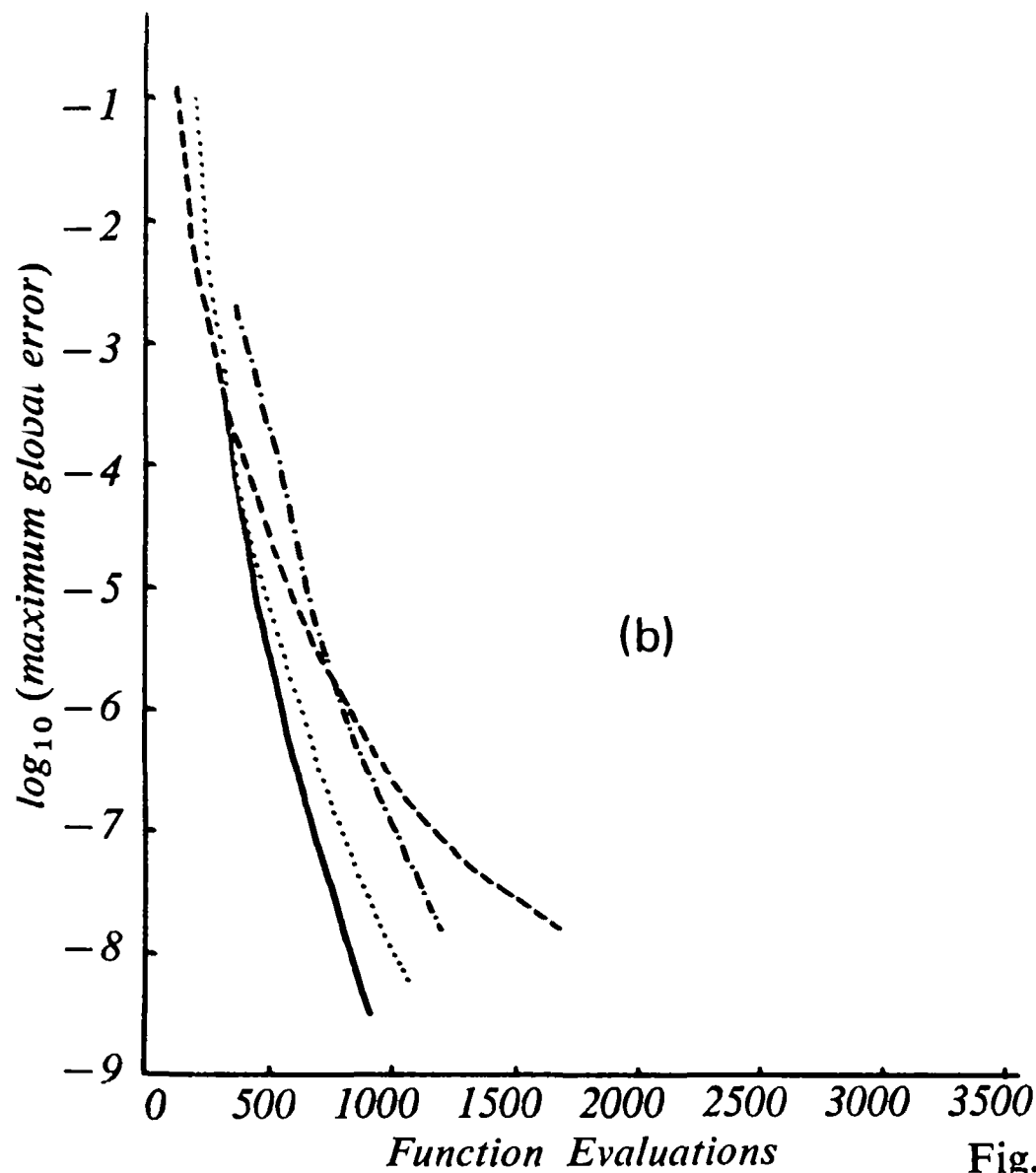
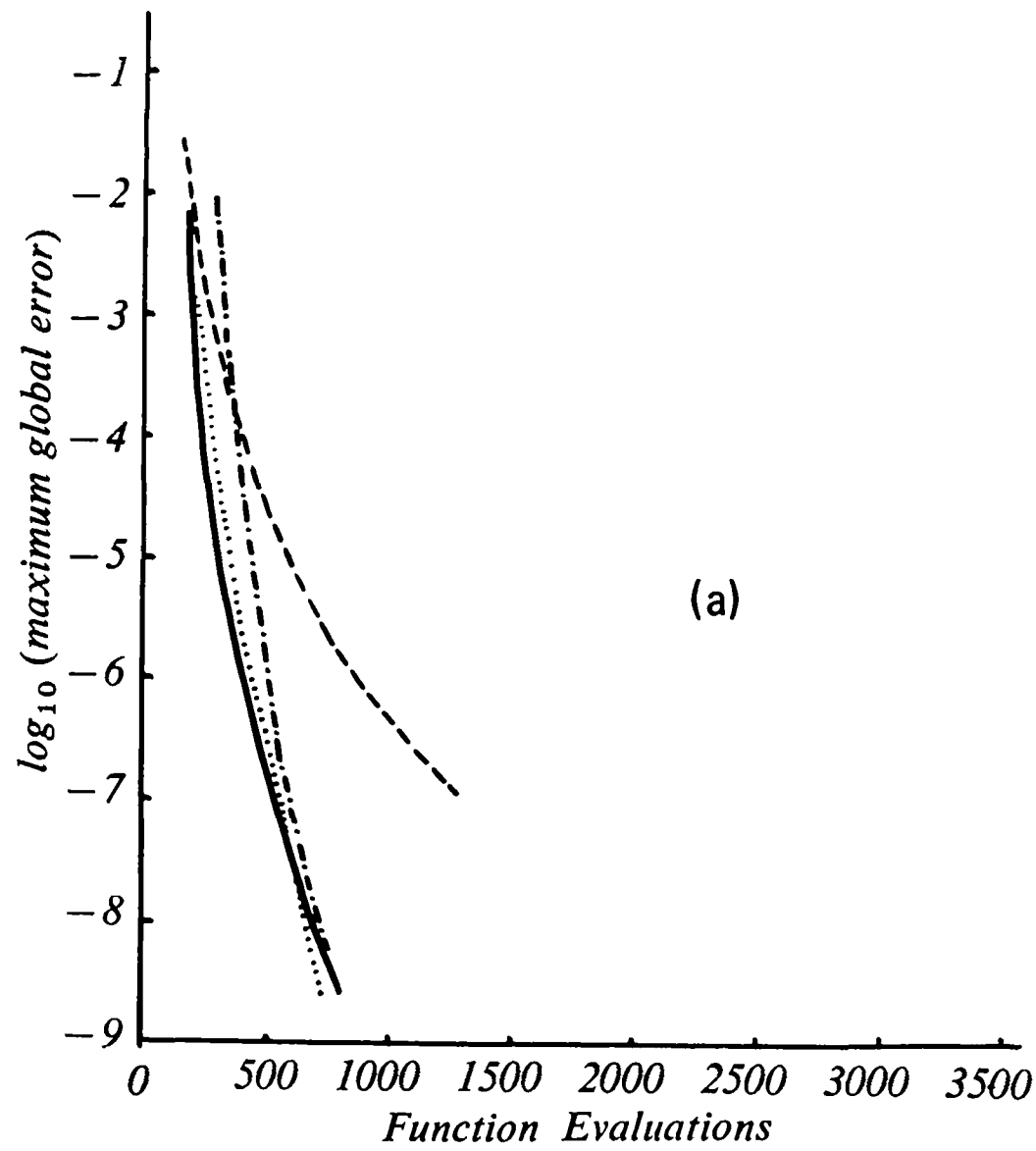


Fig. 2a-c.

Fig. 2. Efficiency curves for Runge-Kutta-Nyström methods applied to the two-body problem.

(a)  $e = 0.1$ , (b)  $e = 0.3$ , (c)  $e = 0.5$ , (d)  $e = 0.7$ , (e)  $e = 0.9$ .

Key: ————— RKN7(6)T      - - - - - RKN5(4)  
- . - . - . RKNF8      ..... RKNF7



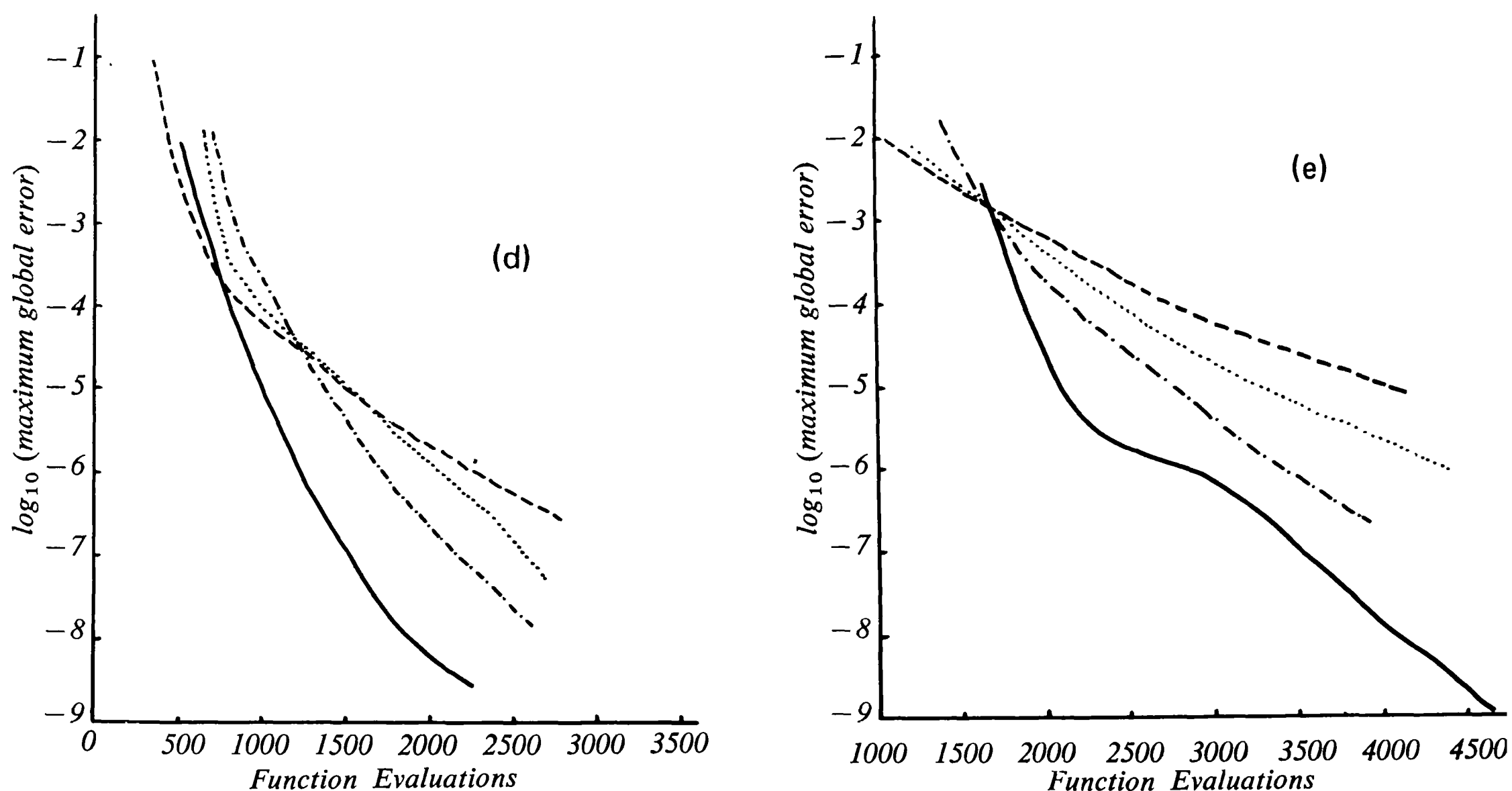


Fig. 2d-e.

It is clear from Figure 1 that the higher order mode gives substantially better accuracy in the case of RKF7. Our results consistent with those of Enright *et al.* (1974) indicate that the order of the optimum formula should be chosen with regard to the required accuracy. For low accuracy, global error  $> 10^{-3}$  approximately, low order methods are to be preferred over high order methods which can suffer from absolute instability due to the very large step sizes encountered when using high tolerances. These comments apply equally to RK and RKN formulae.

An examination of Figure 1 indicates that RKF7 in eighth order mode should be preferred to the others for global error less than  $10^{-3}$  approximately otherwise either RKF4 (in fifth order mode) or RK5(4)*T* should be used. The intermediate formula RKF5 does not seem to offer advantages for any accuracy. A comparison of Figures 1 and 2 indicates that RKN methods should be preferred wherever possible. This conclusion was reached by Fehlberg (1972) with regard to a different problem. The RKN5(4) of Bettis is most efficient for low accuracy and though it is far inferior to the RKN7(6)*T* for high accuracy it is as efficient as the RKF7 even for an accuracy of  $10^{-5}$  on problem 4. For high accuracy the RKN7(6)*T* is easily the best formula.

#### 4. Discussion

We have presented only results for the integration of the two-body problem but it will be clear that these are applicable to many-body problems; the different cases

(1 to 5) give an indication of numerical behaviour in the simulation of interplay in  $N$ -body systems.

In view of the rather small global error range in which low order methods are useful, it may seem reasonable to disregard these formulae in celestial mechanical calculations. However we feel that they should be used in some cases when only a few integration steps are required. The resisting medium calculations of Dormand and Woolfson (1977) use a combination of a fourth order method and RKF7. In this, and many other, cosmogonical computations, the initial conditions are somewhat speculative and therefore the application of low tolerances is unreasonable.

It will be seen that in cases 4 and 5, especially the latter, the global error for a given number of function evaluations is considerably larger than in the other cases, thus illustrating the need for regularization (Bettis and Szebehely, 1971) during close encounters in the numerical analysis of the  $N$ -body problem.

The computations described above were performed in FORTRAN using 11 or 20 significant figure precision on the ICL 1905E of the Teesside Polytechnic computer unit. We are grateful to a referee for his valuable comments on an earlier version of this paper.

### References

- Bettis, D. G.: 1973, *Celest. Mech.* **8**, 229–233.  
 Bettis, D. G. and Szebehely, V.: 1971, *Astrophys. Space Sc.* **14**, 133–150.  
 Butcher, J. C.: 1964, *J. Austral. Math Soc.* **4**, 179–194.  
 Dormand, J. R. and Woolfson, M. M.: 1977, *Monthly Notices Roy. Astron. Soc.* **180**, 243–279.  
 Enright, W. H., Bedet, R., Farkas, I., and Hull, T. E.: 1974, Technical Report No. 68, Dept of Computer Science, University of Toronto.  
 Fehlberg, E.: 1968, NASA TR R287.  
 Fehlberg, E.: 1969, NASA TR R315.  
 Fehlberg, E.: 1972, NASA TR R381.  
 Hull, T. E., Enright, W. H., Fellen, B. M., and Sedgwick, A. E.: 1972, *SIAM J. Num. An.* **9**, 603–637.  
 Stetter, H. J.: 1971, *SIAM J. Num. An.* **8**, 512–523.  
 Zonneveld, J. A.: 1970, *Automatic Numerical Integration*, 2nd edition, Mathematisch Centrum, Amsterdam.