VYSOKÉ UČENÍ TECHNICKÉ V BRNĚ

BRNO UNIVERSITY OF TECHNOLOGY

FAKULTA ELEKTROTECHNIKY A KOMUNIKAČNÍCH TECHNOLOGIÍ ÚSTAV RADIOELEKTRONIKY

FACULTY OF ELECTRICAL ENGINEERING AND COMMUNICATION DEPARTMENT OF RADIO ELECTRONICS

AUGMENTED REALITY APPLICATIONS IN EMBEDDED NAVIGATION DEVICES

SEMESTRÁLNÍ PRÁCE SEMESTRAL THESIS

AUTOR PRÁCE AUTHOR

MARTIN JAROŠ



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BBCE: Examine practical use cases of the proposed navigation device, design applicable user interface.

DOPORUČENÁ LITERATURA:

[1] BIMBER, O.; RASKAR, R. Spatial augmented reality: merging real and virtual worlds. Wellesley: A K Peters, 2005, 369 p. ISBN 15-688-1230-2.

[2] Texas Instruments. OMAP 4460 Multimedia Device [online]. 2012 - [cit. 8. listopadu 2012]. Available: http://www.ti.com/product/omap4460.

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ABSTRAKT

Tato práce se zabývá aplikací rozšířené reality v oblasti navigačních zařízení. Popisuje možnosti zpracování videa za účelem projekce virtuální scény pomocí dat získaných satelitním a inerciálním navigačním sybsystémem. Práce klade důraz na využití moderních hardwarových prostředků pro zpracování videa v mikroprocesorech pomocí grafických akcelerátorů. Součástí je návrh aplikace a realizace prototypu.

KLÍČOVÁ SLOVA

Rozšířená realita, satelitní navigace, inerciální měřící jednotka

ABSTRACT

This work deals with application of augmented reality in navigation devices. It describes possibilities of video processing, rendering a virtual scene by using data measured by satellite and inertial navigation subsystems. Special care is taken into account for use of modern graphic accelerator hardware available in microprocessors. Design of the application is supplemented with prototype realization.

KEYWORDS

Augmented reality, satellite navigation, inertial measurement unit

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Preface

TODO: Introduction [1]

1 Augmented reality

1.1 Project overview

Main goal of this project is to develop a device capable of rendering a real time overlay over the captured video frame. There are three external inputs, the image sensor capable of video capture feeds real-time images to the system. The GPS receiver delivers positional information and inertial sensors supplement it with spatial orientation. Expected setup is that the image and inertial sensors are on a single rack able to freely rotate around, while the GPS receiver is static, relative to the whole moving platform (vehicle for example). The overlay consists of fixed kinematic data such as speed or altitude, reference indicators such a horizon line and dynamic location markers. These will be spatially aligned with real locations visible in the video thus providing navigation information. They work in a way that wherever the camera is pointed to, specific landmarks will label currently visible locations. Complex external navigation system may be also connected. This allows integration with already existing systems, such as moving map systems, PDAs or other specialized hardware. These provides user with classic route navigation and map projection, while this project gives spatial extension to further improve total situational awareness. The analogy are the head up and head down displays, each delivering specific set of information. This project is focuses on visual enhancement instead of a full featured navigation device.

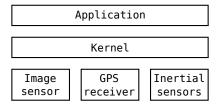


Figure 1: Project overview

1.2 Hardware limitations

Application is designed to run in embedded environments, where power management is very important. While many platforms features multiple symmetrical processor cores - CPUs, application should focus on lowest per-core usage as possible. This can be done by delegating specific tasks to specialized hardware. CPU is specialized in execution of a single thread, integer and bitwise operations, with many branches in its code. With vector and floating point extensions they are also very efficient in computation of difficult mathematical algorithms. However they do not perform well in simple calculations over large amounts of data, where mass parallelization is possible. This is the case in graphics where special graphics processors - GPUs have been deployed. GPU consists of high number (hundreds) of simple cores, which are able to perform operations over large blocks of data. They scale efficiently with the amount of data needed to be processed due to parallelization, however they have problems with nonlinear code with branches. While CPUs have long pipelines for branch optimizations,

GPUs cannot employ those, any branch in their code will be extremely inefficient and should be avoided. Graphics chapter focuses on this area. There are also available specialized subsystems designed and optimized for a single purpose. For example video accelerators, capable of video encoding and decoding, image capture systems or peripheral drivers. They will be mentioned in specific chapters.

Developing an application for an embedded device faces a problem, as there are big differences between these devices it is hard to support the hardware and make the application portable. In order to reuse code and reduce application size, libraries are generally used to create an intermediary layer between application and the hardware. However, to provide enough abstraction some sort of operating system has to be used. Operating systems may be realtime, giving applications full control, behaving just like large libraries. This is favorable approach in embedded systems as it allows precise timings. There are many such systems specially tailored for embedded applications like FreeRTOS or proprietary VxWorks. On the other hand, as recent processors improved greatly in power, efficiency and capabilities, it is possible and quite feasible to run a full featured system like Linux or proprietary Windows CE. Linux kernel is highly portable and configurable, although it does restrict applications from real-time use (Linux RT patches also exist for real-time applications), as all hardware dependent modules which requires full control over the hardware are part of the kernel itself, application does not need to run in real-time at all. Other advantages are free, open and well documented sources, highly standardized and POSIX compliant application interface, large amount of drivers with good manufacturer support. While its disadvantages are very large code base and steep learning curve, which may slow the initial development. Nevertheless Linux kernel has been chosen for the project, more details about its interfaces are in the Linux kernel chapter. While the application is designed to be highly portable depending only on the kernel itself, several devices has been chosen as the reference, they are listed in the hardware chapter.

2 Application

Application is divided into four subsystem, each being standalone component. The video subsystem is responsible for enumeration and control of the video architecture and its devices. It provides the application with raw video buffers and means to configure its format. It is designed to support high range of devices from embedded image sensors to external cameras, while using single application interface and common image format. Depending on the hardware, high definition video output is expected. Video subsystem is optimized for synchronous operation with the graphics subsystem. Graphics subsystem utilizes platform interfaces for its graphic accelerator units to provide optimized video processing and rendering. It is hardware independent through common library support to run on most embedded systems. Its goal is to provide application with efficient methods for rendering primitives, video frames and vector fonts with object oriented interface. These methods combined will create the scene overlay over the source video in real time. Graphic output should be high definition digital, maintaining source quality. Data needed for the overlay creating are provided by satellite and inertial subsystems. They are both designed for asynchronous operation. The satellite navigation provides application with positional and kinematic data. It is responsible for communication with external navigation systems such as GPS receivers and all needed calculations. Its interfaces allows application to access required information asynchronously as needed by the rendering loop. The inertial measurement subsystem utilizes sensors needed for spatial orientation not provided by satellite navigation. As there are many such sensor, common interface is provided. Subsystem handles all initialization, control, data acquisition and required calculations. Its internal state machine provides application the requested data on demand. Application is designed to be modular and highly configurable. All constants used throughout the implementation are defined with a default value and modifiable through the configuration file. This includes for example video setup, device selection or rendering parameters.

2.1 Linux kernel

Programs running in Linux are divided into two groups, kernel-space and user-space. Only kernel and its runtime modules are allowed to execute in kernel-space, they have physical memory access and use CPU in real-time. All other programs runs as processes in user-space, they have virtual memory access, which means their memory addresses are translated to the physical addresses in the background. In Linux each process runs in a sandbox, isolated from the rest of the system. Processes access memory unique to them, they cannot access memory assigned for other processes nor memory managed by the kernel. They may communicate with the outside environment by several means:

- Arguments and environment variables
- Standard input, output and error output
- Virtual File System
- Signals

- Sockets
- Memory mapping

Each process is ran with several arguments in a specific environment with three default file descriptors. For example running

VARIABLE=value ./executable argument1 argument2 <input 1>output 2>error

will execute executable with environment variable VARIABLE of value value with two arguments argument1 and argument2. Standard input will be read from file input while regular output will be written to file output and error output to file error. This process may further communicate by accessing files in the Virtual File System, kernel may expose useful process information for example via procfs file-system usually mounted at /proc. Other types of communication are signals (which may be sent between processes or by kernel) and network sockets. With internal network loop-back device, network style inter process communication is possible using standard protocols (UDP, TCP, ...). Memory mapping is a way to request access to some part of the physical memory.

Process execution is not real-time, but they are assigned restricted processor time by the kernel. They may run in numerous threads, each thread has preemptively scheduled execution. Threads share memory within a process, memory access to these shared resources must done with care to avoid race conditions and data corruption. Kernel provides *mutex* objects to lock threads and avoid simultaneous memory access. Each shared resource should be attached to a *mutex*, which is locked during access to this resource. Thread must not lock *mutex* while still holding lock to this or any other *mutex* in order to avoid dead-locking. Source code on how to use threads is in the threads example appendix.

Linux kernel has monolithic structure, so all device drivers resides in the kernel-space. From application point of view, this means that all peripheral access must be done through the standard library and Virtual File System. Individual devices are accessible as device files defined by major and minor number typically located at /dev. These files could be created automatically by kernel (devtmpfs file-system), by daemon (udev(8))), or manually by mknod(1). Complete kernel device model is exported as sysfs file-system and typically mounted at /sys.

Function name	Access type	Typical usage
<pre>select(), poll()</pre>	event	Synchronization, multiplexing, event handling
ioctl()	structure	Configuration, register access
<pre>read(), write()</pre>	stream	Raw data buffers, byte streams
mmap()	block	High throughput data transfers

Table 1: Available functions for working with device file descriptors

For example let's assume a generic peripheral device connected by the I²C bus. First, to tell kernel there is such a device, the sysfs file-system may be used

```
echo $DEVICE_NAME $DEVICE_ADDRESS > /sys/bus/i2c/devices/i2c-1/new_device
```

This should create a special file in /dev, which should be opened by open() to get a file descriptor for this device. Device driver may export some *ioctl* requests, each request is defined by a number and a structure passed between the application and the kernel. Driver should define requests for controlling the device, maybe accessing its internal registers and configuring a data stream. Each request is called by

```
ioctl(fd, REQNUM, &data);
```

where fd is the file descriptor, REQNUM is the request number defined in the driver header and data is the structure passed to the kernel. This request will be synchronously processed by the kernel and the result stored in the data structure. Let's assume this devices has been configured to stream an integer value every second to the application. To synchronize with this timing application may use

```
struct pollfd fds = {fd, POLLIN};
poll(&fds, 1, -1);
which will block infinitely until there is a value ready to be read. To actually read it,
int buffer[1];
```

ssize t num = read(fd, buffer, sizeof(buffer));

will copy this value to the buffer. Copying causes performance issues if there are very large amounts of data. To access this data directly without copying them, application has to map physical memory used by the driver. This allows for example direct access to a DMA channel, it should be noted that this memory may still be needed by kernel, so there should be some kind of dynamic access restriction, possibly via *ioctl* requests (this would be driver specific).

2.2 Video subsystem

Video support in Linux kernel is maintained by the LinuxTV project, it implements the videodev2 kernel module and defines the V4L2 interface. Modules are part of the mainline kernel at drivers/media/video/* with header linux/videodev2.h. The core module is enabled by the VIDEO_V4L2 configuration option, specific device drivers should be enabled by their respective options. V4L2 is the latest revision and is the most widespread video interface throughout Linux, drives are available from most hardware manufactures and usually mainlined or available as patches. The Linux Media Infrastructure API[2] is a well documented interface shared by all devices. It provides abstraction layer for various device implementations, separating the platform details from the applications. Each video device has its device file and is controlled via ioctl calls. For streaming standard I/O functions are supported, but the memory mapping is preferred, this allows passing only pointers between the application and the kernel, instead of unnecessary copying the data around.

Name	Description
VIDIOC_QUERYCAP	Query device capabilities
VIDIOC_G_FMT	Get the data format
VIDIOC_S_FMT	Set the data format
VIDIOC_REQBUFS	Initiate memory mapping
VIDIOC_QUERYBUF	Query the status of a buffer
VIDIOC_QBUF	Enqueue buffer to the kernel
VIDEOC_DQBUF	Dequeue buffer from the kernel
VIDIOC_STREAMON	Start streaming
VIDIOC_STREAMOFF	Stop streaming

Table 2: V4L2 ioctl calls defined in linux/videodev2.h

Application sets the format first, then requests and maps buffers from the kernel. Buffers are exchanged between the kernel and the application. When the buffer is enqueued, it will be available for the kernel to capture data to it. When the buffer is dequeued, kernel will not access the buffer and application may read the data. After all buffer are enqueued application starts the stream. Polling is used to wait for the kernel until it fills the buffer, buffer should not be accessed simultaneously by the kernel and the application. After processing the buffer, application should return it back to the kernel queue. Note that buffers should be properly unmapped by the application after stopping the stream.

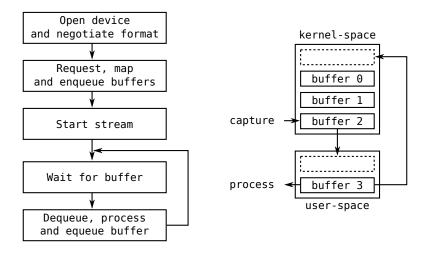


Figure 2: V4L2 capture

Source code for simple video capture is in video capture example appendix. The image format is specified using the little-endian four-character code (FOURCC). V4L2 defines several

formats and provides v412_fourcc() macro to create a format code from four characters. As described later in the graphics subsystem chapter, graphics uses natively the RGB4 format. This format is defined as a single plane with one sample per pixel and four bytes per sample. These bytes represents red, green and blue channel values respectively. Image size is therefore $width \cdot height \cdot 4$ bytes. Many image sensors however support YUV color-space, for example the YU12 format. This one is defined as three planes, the first plane with one luminance sample per pixel and the second and third plane with one chroma sample per four pixels (2 pixels per row, interleaved). Each sample has one byte, this format is also referenced as YUV 4:2:0 and its image size is $width \cdot height \cdot 1.5$ bytes. The luminance and chroma of a pixel is defined as

$$E_Y = W_R \cdot E_R + (1 - W_R - W_B) \cdot E_G + W_B \cdot E_B,$$

$$E_{C_r} = \frac{0.5(E_R - E_Y)}{1 - W_R},$$

$$E_{C_b} = \frac{0.5(E_B - E_Y)}{1 - W_B},$$

where E_R , E_G , E_B are normalized color values and W_R , W_B are their weights. ITU-R Rec. BT.601[3] defines weights as 0.299 and 0.114 respectively, it also defines how they are quantized

$$Y = 219E_Y + 16,$$

$$C_r = 224E_{C_r} + 128,$$

$$C_b = 224E_{C_b} + 128.$$

To calculate R, G, B values from Y, Cr, Cb values, inverse formulas must be used

$$\begin{split} E_Y &= \frac{Y - 16}{219}, \\ E_{C_r} &= \frac{C_r - 128}{224}, \\ E_{C_b} &= \frac{C_b - 128}{224}, \\ E_R &= E_Y + 2E_{C_r}(1 - W_R), \\ E_G &= E_Y - 2E_{C_r}\frac{W_R - {W_R}^2}{W_G} - 2E_{C_b}\frac{W_B - {W_B}^2}{W_G}, \\ E_B &= E_Y + 2E_{C_b}(1 - W_B). \end{split}$$

It should be noted that not all devices may use the BT.601 recommendation, V4L2 refers to it as V4L2_COLORSPACE_SMPTE170M in the VIDIOC_S_FMT request structure. Implementation of the YUV to RGB color-space conversion is most efficient on graphics accelerators, such example is included in colorspace conversion example appendix. It is written in GLSL for fragment processor, see graphics subsystem chapter for further description.

There is a kernel module v412loopback which creates a video loop-back device, similar to network loop-back, allowing piping two video applications together. This is very useful

not only for testing, but also for implementation of intermediate decoders. GStreamer is a powerful multimedia framework widespread in Linux distributions, composed of a core infrastructure and hundreds of plug-ins. This command will create synthetic RGB4 video stream for the application, useful for testing

```
modprobe v412loopback
gst-launch videotestsrc pattern=solid-color foreground-color=0xE0F0E0 ! \
"video/x-raw,format=RGBx,width=800,height=600,framerate=20/1" \
! v412sink device=/dev/video0
```

Texas Instruments distributes a meta package[4] for their OMAP platform featuring all required modules and DSP firmware. This includes kernel modules for SysLink inter-chip communication library, Distributed Codec Engine library and ducati plug-in for GStreamer. With the meta-package installed, it is very easy and efficient to implement mainstream encoded video formats. For example following command will create GStreamer pipeline to receive video payload over a network socket from an IP camera, decode it and push it to the loop-back device for the application. MPEG-4 AVC (H.264) decoder of the IVA 3 is used in this example.

```
modprobe v4l2loopback
gst-launch udpsrc port=5004 caps=\
"application/x-rtp,media=video,payload=96,clock-rate=90000,encoding-name=H264" \
! rtph264depay ! h264parse ! ducatih264dec ! v4l2sink device=/dev/video0
```

On OMAP4460 this would consume only about 15% of the CPU time as the decoding is done by the IVA 3 video accelerator in parallel to the CPU which only passes pointers around and handles synchronization. Output format is NV12 which is similar to YU12 format described earlier, but there is only one chroma plane with two-byte samples, first byte being the U channel and the second byte the V channel, sampling is same 4:2:0. The YUV to RGB color space conversion must take place here, preferably implemented on the GPU as described above.

Cortex-A9 cores on the OMAP4460 also have the NEON co-processor, capable of vector floating point math. Although not very supported by the GCC C compiler, there are many assembly written libraries implementing coders with the NEON acceleration. For example the *libjpeg-turbo* library is implementing the *libjpeg* interface. It is useful for USB cameras, as the USB throughput is not high enough for raw high definition video, but is sufficient with JPEG coding (as most USB cameras supports JPEG, but does not support H.264). 1080p JPEG stream decoded with this library via its GStreamer plug-in will consume about 90% of the single CPU core time (note that there are two CPU cores available). However, comparable to the AVC, JPEG encoding will cause visible quality degradation in the raw stream (video looks grainy).

2.3 Graphics subsystem

TODO: Graphics stack, OpenGL ES 2.0 [5]

2.4 Inertial measurement subsystem

Application needs to know its spatial orientation for rendering, there are three devices which may provide such information, gyroscope, compass (magnetometer) and accelerometer. Hardware details about these devices are in the hardware chapter.

2.4.1 Industrial I/O module

A relatively young kernel module iio has been implemented in recent kernels to provide standardized support for sensors and analog converters typically connected by I²C bus. While many device drivers are still in staging tree, to core module is ready for production code. Subsystem provides device structure mapped in sysfs, typically available at /sys/bus/iio/devices/. Device are implemented usually on top of the i2c-dev driver and registered as /sys/bus/iio/devices/iio:deviceX, where X is the device number and the device name may be obtained by

cat /sys/bus/iio/devices/iio:deviceX/name

There are many possible channels, named by the value type they represents. To read an immediate value, for example from an ADC channel 1

```
cat /sys/bus/iio/devices/iio:deviceX/in_voltage1_raw
cat /sys/bus/iio/devices/iio:deviceX/in_voltage_scale
```

where the result value in volts is $raw \cdot scale$. However, being easy, this is not efficient, buffers have been implemented to stream measured data to the application. Buffer uses device file named after the iio device, e.g. /dev/iio:deviceX. To stream data through the buffer, driver needs to have control over the timing, triggers have been implemented for this purpose. They are accessible as /sys/bus/iio/devices/triggerX, where X is the trigger number and its name may be obtained by

cat /sys/bus/iio/devices/triggerX/name

Software trigger may be created by

```
echo 1 > /sys/bus/iio/iio_sysfs_trigger/add_trigger
```

and triggered by application

```
echo 1 > /sys/bus/iio/trigger0/trigger_now
```

Name of this trigger is sysfstrigX, where X is the trigger number. Hardware triggers are also implemented, both GPIO and timer based triggers. Devices may implement triggers themselves, providing for example the data ready trigger. Device triggers are generally named as name-devX, where name is device name and X is device number. To use trigger with the buffer use

```
echo "triggername" > /sys/bus/iio/devices/iio:deviceX/trigger/current_trigger
```

where triggername is the name of the trigger, for example adc-dev0 will be the device trigger for the ADC. Data are measured in specific channels, they are defined in

/sys/bus/iio/devices/iio:device0/scan_elements. Channels must be enabled for buffering individually, for example

echo 1 > /sys/bus/iio/devices/iio:device0/scan_elements/in_voltage1_en
echo 1 > /sys/bus/iio/devices/iio:device0/scan_elements/in_voltage2_en

will enable ADC channels 1 and 2. Buffer itself can be started by

echo 256 > /sys/bus/iio/devices/iio:deviceX/buffer/length
echo 1 > /sys/bus/iio/devices/iio:deviceX/buffer/enabled

this will start streaming data to the device file. Data are formatted in packets, each packed consists of per-channel values and is terminated by 8 byte time-stamp of the sample. Order of the channels in the buffer can be obtained by

cat /sys/bus/iio/devices/iio:device0/scan_elements/in_voltageX_index

which reads index of the specified channel. Data format of this channel is

cat /sys/bus/iio/devices/iio:device0/scan_elements/in_voltageX_type

which reads encoded string, for example le:u10/16>>0, where le means little-endian, u means unsigned, 10 is the number of relevant bits while 16 is the number of actual bits and 0 is the number of right shifts needed.

Following channels are needed by the application:

- anglvel x
- anglvel_y
- anglvel z
- accel x
- accel y
- accel z
- magn_x
- magn_y
- magn_z

representing measurements from gyroscope, accelerometer and magnetometer respectively.

2.4.2 DCM algorithm

Equations needed to calculate device attitude have been derived from [6]. Gyroscope measures angular speed around device axes, it offers high differential precision and fast sampling rate, however it suffers slight zero offset error. Device attitude can be obtained simply by integrating measured angular rates, provided that initial attitude is known. The angular rate is defined

$$\overrightarrow{\omega_g} = \frac{\mathrm{d}}{\mathrm{d}t} \overrightarrow{\Phi}_{(t)},$$

so the angular displacement between last two samples is

$$[\Phi_x, \Phi_y, \Phi_z] = [\omega_x, \omega_y, \omega_z] \cdot_{\Delta} t.$$

This can be described as a rotation

$$\mathbf{R}_{gyro} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos(\Phi_x) & -\sin(\Phi_x) \\ 0 & \sin(\Phi_x) & \cos(\Phi_x) \end{bmatrix} \times \begin{bmatrix} \cos(\Phi_y) & 0 & \sin(\Phi_y) \\ 0 & 1 & 0 \\ -\sin(\Phi_y) & 0 & \cos(\Phi_y) \end{bmatrix} \times \begin{bmatrix} \cos(\Phi_z) & -\sin(\Phi_y) & 0 \\ \sin(\Phi_y) & \cos(\Phi_y) & 0 \\ 0 & 0 & 1 \end{bmatrix}.$$

With Δt close to zero a small-angle approximation may be used to simplify $\cos(x) = 1$, $\sin(x) = x$

$$\mathbf{R}_{gyro} \doteq \begin{bmatrix} 1 & -\Phi_z & \Phi_y \\ \Phi_x \Phi_y + \Phi_z & 1 - \Phi_x \Phi_y \Phi_z & -\Phi_x \\ \Phi_x \Phi_z - \Phi_y & \Phi_x + \Phi_y \Phi_z & 1 \end{bmatrix}.$$

Let's define the directional cosine matrix describing device attitude

$$\mathbf{DCM} = \begin{bmatrix} \widehat{\mathbf{I}} \cdot \widehat{\mathbf{x}} & \widehat{\mathbf{I}} \cdot \widehat{\mathbf{y}} & \widehat{\mathbf{I}} \cdot \widehat{\mathbf{z}} \\ \widehat{\mathbf{J}} \cdot \widehat{\mathbf{x}} & \widehat{\mathbf{J}} \cdot \widehat{\mathbf{y}} & \widehat{\mathbf{J}} \cdot \widehat{\mathbf{z}} \\ \widehat{\mathbf{K}} \cdot \widehat{\mathbf{x}} & \widehat{\mathbf{K}} \cdot \widehat{\mathbf{y}} & \widehat{\mathbf{K}} \cdot \widehat{\mathbf{z}} \end{bmatrix} = \begin{bmatrix} \widehat{\mathbf{I}}_{xyz} \\ \widehat{\mathbf{J}}_{xyz} \\ \widehat{\mathbf{K}}_{xyz} \end{bmatrix},$$

where $\hat{\mathbf{I}}$ points to the north, $\hat{\mathbf{J}}$ points to the east, $\hat{\mathbf{K}}$ points to the ground and therefore $\hat{\mathbf{I}} = \hat{\mathbf{J}} \times \hat{\mathbf{K}}$. Roll, pitch and yaw angels in this matrix are

$$\gamma = -\arctan_2\left(\frac{\mathbf{DCM}_{23}}{\mathbf{DCM}_{33}}\right),$$

 $\beta = \arcsin(\mathbf{DCM}_{13}),$

$$\alpha = -\arctan_2\left(\frac{\mathbf{DCM}_{12}}{\mathbf{DCM}_{11}}\right).$$

DCM can be computed by applying consecutive rotations over time

$$\mathbf{DCM}_{(t)} = \mathbf{R}_{gyro(t)} \times \mathbf{DCM}_{(t-1)}.$$

If the sampling rate is high enough (over 1kHz at least), this method is very accurate and has good dynamics over short periods of time, but in longer runs errors integrated during processing will cause serious drift (both numerical errors and zero offset errors). To mitigate these problems accelerometer and compass has to be used to provide the initial attitude and to fix the drift over time. Accelerometer measures external mechanical forces applied to the device together with gravitational force. However precision of these devices are generally worse and they have slower sampling rates. If there are no extern forces, it will measure the gravitational vector directly, thus providing the third row of the DCM

$$\overrightarrow{\mathbf{a}}_{acc} = g \ \widehat{\mathbf{K}}_{xyz} + \frac{\overrightarrow{\mathbf{F}}}{m},$$

$$\overrightarrow{\mathbf{F}} = 0 \ \rightarrow \ \widehat{\mathbf{K}}_{xyz} = \frac{\overrightarrow{\mathbf{a}}_{acc}}{|\overrightarrow{\mathbf{a}}_{acc}|}.$$

When there is an external force $\overrightarrow{\mathbf{F}}$ applied, which is not parallel and has significant magnitude relative to gravitational force $m g \widehat{\mathbf{K}}_{xyz}$, measurements will degrade rapidly reaching singularity

during the free fall ($|\overrightarrow{\mathbf{a}}_{acc}| = 0$). This error may be corrected by using device speed measured by satellite navigation system with high sample rate (over 10Hz)

$$\widehat{\mathbf{K}}_{xyz} = \frac{\overrightarrow{\mathbf{a}}_{acc} - \frac{\mathbf{d}}{\mathbf{d}t} \overrightarrow{\mathbf{V}}_{GPS}}{g}.$$

Magnetometer has similar properties, it measures magnetic flux density of the field the device is within. This should ideally result in a vector pointing to the north, therefore providing the first row of the DCM

$$\widehat{\mathbf{I}}_{xyz} = \frac{\overrightarrow{\mathbf{B}}_{corr}}{\left|\overrightarrow{\mathbf{B}}_{corr}\right|}.$$

Magnetometers have even slower sampling rates and far worse precision as the Earth field is distorted by nearby metal objects. This magnetic deviation can be divided into hard-iron and soft-iron effects. Hard-iron distortion is caused by materials that produces magnetic field, that is added to the Earth magnetic field. Vector of this field can be subtracted to compensate this error

$$\overrightarrow{\mathbf{B}}_{corr1} = \overrightarrow{\mathbf{B}}_{mag} - \frac{1}{2} \left[\min(B_x) + \max(B_x), \min(B_y) + \max(B_y), \min(B_z) + \max(B_z) \right].$$

The soft-iron distortion is caused by soft magnetic materials, which reshapes the field in a way that is not simply additive. It may be observed as an ellipse when the device is rotated around and the measured values are plotted. Compensating for these effects is involves remapping this ellipse back to the sphere. This is computation intensive and as soft-iron effects are usually weak (up to few degrees), it may be omitted.

Further more the magnetic field of the Earth itself does not point to the geographic north, but is rotated by an angle specific to the location on the Earth surface. Magnetic inclination is the vertical portion of this rotation causing magnetic vector to incline to the ground, it may be fixed by using measurements from the accelerometer to make the magnetic vector perpendicular to the gravitational vector

$$\overrightarrow{\mathbf{B}}_{corr2} = \widehat{\mathbf{K}}_{xyz} \times \overrightarrow{\mathbf{B}}_{mag} \times \widehat{\mathbf{K}}_{xyz}.$$

Magnetic declination (sometimes referred as magnetic variation) is the horizontal portion of this rotation and is sometimes provided by the satellite navigation systems. To correct for this error, measured values have to be rotated by the inverse angle

$$\overrightarrow{\mathbf{B}}_{corr3} = \overrightarrow{\mathbf{B}}_{mag} \begin{bmatrix} \cos(var) & \sin(var) \\ -\sin(var) & \cos(var) \end{bmatrix}.$$

By combination of the corrected results from accelerometer and magnetometer complete DCM can be calculated. Weighted average should be used, in real-time this yields

$$\mathbf{DCM}_{(t)} = W_{gyro} \left(\mathbf{R}_{gyro} \times \mathbf{DCM}_{(t-1)} \right) + \left(1 - W_{gyro} \right) \begin{bmatrix} \widehat{\mathbf{I}}_{xyz} \\ \widehat{\mathbf{K}}_{xyz} \times \widehat{\mathbf{I}}_{xyz} \end{bmatrix},$$

where $\hat{\mathbf{I}}_{xyz}$ and $\hat{\mathbf{K}}_{xyz}$ are calculated from magnetometer and accelerometer measurements. W_{gyro} is the weight of the gyroscope measurement, it must be estimated by trial and error to mitigate its drift but not add too much noise.

2.5 Satellite navigation subsystem

2.5.1 Communication

Systems such GPS uses numerous satellites orbiting Earth transmitting their position and time. Receiver can measure its position and time with accuracy based on number of satellites in view. GPS receivers usually communicate via serial line, Linux kernel features TTY module originally used for teletypewriters. It handles serial lines, converters and virtual lines with devices generally called /dev/tty*. Serial connections (UART, RS-232) are usually named /dev/ttySn, where n is a device number. Emulated connection are named /dev/ttyUSBn for USB emulators or /dev/ttyACMn for modem emulators. There are also pseudo terminals in /dev/pts/ used for software emulation. This devices allows standard I/O and may be controlled with functions defined in termios.h or by using stty(1) utility. Important physical interface settings are

- baud rate
- parity
- flow control

Usual settings are 4800 or 38 400 baud with no parity and no flow control, without proper interface configuration, no communication will occur Driver further provides line processing before passing data to the application. Important line settings are

- newline handling
- echo mode
- canonical mode
- timeouts and buffer sizes

In canonical mode, driver uses line buffer and allows line editing. Buffer is passed to the application (made readable) after new line is encountered. It may be combined with echo mode, which will automatically send received characters back. This is how standard console line works, however if application needs to be in full control, driver may be switched to raw mode with per-character buffers and echo disabled. Also specific control characters for canonical mode may be configured.

GPS receivers uses NMEA 183 standard[7]. Communication is done in sentences, each message starts with dollar sign followed by source name and sentence type, then there is comma separated list of data fields optionally terminated by a checksum. Each sentence is appended with carriage return and new line. For example in sentence

```
$GPGGA,000000,4914.266,N,01638.583,E,1,8,1,257,M,46.43,M,,*46
```

GP is source name (GPS receiver), GGA is sentence type (fix information), *46 is a checksum and rest are data fields. Checksum is delimited by asterisk and consist of two character hexadecimal number calculated as bitwise XOR of all character between dollar and asterisk. Note that fields may be empty (as last two fields in the example) and numbers may have fractional parts.

Name	Description	Important information
GSA	Satellites (overall)	Visible satellites, dilution of precision
GSV	Satellites (detailed)	Satellite number, elevation, azimuth (per satellite)
GGA	Fix information	Time, latitude, longitude, altitude, fix quality
RMC	Position data	Time, latitude, longitude, speed, track
RMB	Navigation data	Destination, range, bearing

Table 3: Important NMEA 0183 sentences

The **GSA** sentence has following fields:

- 1st field is A for automatic selection of 2D or 3D fix, M for manual
- 2nd field is 3 for 3D fix, 2 for 2D fix or 1 for no fix
- 3-14 fields are identification numbers (PRN) of satellites in view
- 15-17 fields are dilution of precision and its horizontal and vertical parts

The **GSV** sentence has following fields:

- 1-2 fields are number of partial sentences and number of current part
- 3rd field is number of satellites in view
- 4th field is satellite number (PRN)
- 5th field is satellite elevation
- 6th field is satellite azimuth
- 7th field is satellite signal to noise ratio in decibels
- 8-11, 12-15, 16-19 fields are for other satellites info (number, elevation, azimuth, SNR)

The **GGA** sentence has following fields:

- 1st field resembles time of the fix in format hhmmss.ff, where hh are hours, mm are minutes, ss are seconds and ff are fractional seconds.
- 2-3 fields resembles device latitude in format ddmm.ff, [NS], where dd are degrees, mm are minutes, ff are fractional minutes, N means north hemisphere and S means south.
- 4-5 fields resembles device longitude in format dddmm.ff, [EW], where E means east of Greenwich and W means west.
- 6th field is fix type, 1 means GPS fix
- 7th field is number of satellites in view
- 8th field is horizontal dilution of precision

- 9-10 fields resembles device altitude above mean sea level, first field is the value and second field is the unit used
- 11-12 fields resembles height of geoid above WGS84 in the same fashion
- 13-14 fields are usually unused and refers to differential GPS

The **RMC** sentence has following fields:

- 1-5 fields are same as in GGA sentence
- 6th field is speed over the ground in knots
- 7th field is track angle in degrees
- 8th field resembles date in format ddmmyy, where dd is day, mm is month and yy is last two digits of year
- 9-10 fields resembles magnetic variation, first field is value in degrees, second field is E meaning east or W meaning west

The **RMB** sentence has following fields:

- 1st field is status, A means OK
- 2-3 fields resembles cross-track error, first field is the value in nautical miles, second field is E meaning east or W meaning west
- 4th field is origin waypoint name
- 5th field is destination waypoint name
- 6-9 fields are destination waypoint latitude and longitude with same formatting as in GGA sentence
- 10th field is range to destination in nautical miles
- 11th field is bearing to destination in degrees
- 12th field is velocity towards destination in knots
- 13th field is A for arrived or V for not arrived to destination

2.5.2 Navigation

As Earth shape is very complex, there are two layers of approximation used for computing position. Geoid is the equipotential surface, which describes mean ocean level if Earth was fully covered with water. Most recent geoid model is EGM96 which is used together with WGS84 reference ellipsoid[8]. This ellipsoid has semi-major axis of a = 6378137 meters and flattening f = 1/298.257223563. Note that ellipsoid flattening is defined as

$$f = \frac{a-b}{a},$$

where b is the semi-minor axis. The eccentricity is defined as

$$e = 2f - f^2.$$

Geodetic latitude φ is the angle between normal to the reference ellipsoid and the equator, longitude λ is the angle between normal to the reference ellipsoid and the prime meridian.

Because of the flattening, the normal does not intersect ellipsoid center. Geocentric latitude ψ uses line running through the center instead of the normal,

$$\psi = \arctan[(1-e)^2 \tan(\varphi)].$$

Device position measured by GPS is defined by its geodetic latitude, longitude and altitude

$$h_{AMSL} = h_{WGS84} - h_{EGM96}$$

measured as height above mean see level, where h_{WGS84} is the height above reference ellipsoid and h_{EGM96} is the height above geoid. As GPS sensors usually send fix information at low rates, position needs to be interpolated between fixes. If current ground speed v_{GND} and course α_{TRK} are known, the interpolated position in next step will be

$$\varphi_{(t+1)} = \varphi_{(t)} + \frac{\cos(\alpha_{TRK}) \cdot v_{GND} \cdot_{\Delta} t}{R_{(\varphi)}},$$

$$\lambda_{(t+1)} = \lambda_{(t)} + \frac{\sin(\alpha_{TRK}) \cdot v_{GND} \cdot_{\Delta} t}{R_{(\varphi)} \cdot \cos(\varphi)},$$

where R is the ellipsoid radius at given latitude

$$R = \frac{\sqrt{b^4 \sin(\varphi)^2 + a^4 \cos(\varphi)^2}}{\sqrt{b^2 \sin(\varphi)^2 + a^2 \cos(\varphi)^2}}.$$

Application needs to know projections of specific landmarks as normalized horizontal and vertical coordinates (in range of -1 to 1) used for rendering

$$\begin{bmatrix} x \cdot FOV_x \\ y \cdot FOV_y \end{bmatrix} = \begin{bmatrix} \alpha_{proj} - \alpha_{dev} \\ \beta_{proj} - \beta_{dev} \end{bmatrix} \times \begin{bmatrix} \cos(\gamma_{dev}) & -\sin(\gamma_{dev}) \\ \sin(\gamma_{dev}) & \cos(\gamma_{dev}) \end{bmatrix},$$

where FOV is the field of view, dev means device angle (calculated by inertial subsystem) and proj means projection angle (defined later on in this section). Orthodrome (great circle) is the intersection of a sphere and a plane passing though its center. However, because Earth flattening is rather small, it may be used as an approximation for a curve following Earth surface, connecting two points with shortest route. Spherical trigonometry[9] defines basis for orthodrome calculations. Heading changes along the route and its initial value is the horizontal projection angle

$$\alpha_{proj} = \arctan\bigg(\frac{\sin(\lambda - \lambda_0)\cos(\varphi)}{\cos(\varphi_0)\sin(\varphi) - \sin(\varphi_0)\cos(\varphi)\cos(\lambda - \lambda_0)}\bigg),\,$$

the zero index refers to the device coordinate. Angular distance between those two points is $\phi = \arccos(\sin(\varphi_0)\sin(\varphi) + \cos(\varphi_0)\cos(\varphi)\cos(\lambda - \lambda_0)).$

Vertical projection angle is the angle between the horizon (perpendicular to normal) and a line directly connecting the points. Lets construct a triangle connecting the points with the center of the reference ellipsoid, assuming zero flattening, so the normals have intersection in the center $(f = 0 \rightarrow e = 0 \rightarrow \psi = \varphi)$.

$$a_{\Delta} = h_0 + R_{(\varphi_0)},$$

$$b_{\Delta} = h + R_{(\varphi)},$$

$$c_{\Delta} = \sqrt{a_{\Delta}^2 + b_{\Delta}^2 - 2a_{\Delta}b_{\Delta}\cos(\phi)},$$

$$\beta_{proj} = \arcsin\left(\frac{b_{\Delta}}{c_{\Delta}}\sin(\phi)\right) - \frac{\pi}{2},$$

where h is height above reference ellipsoid. These calculations are quite complex and there is plenty of room for approximation. Over short distance, such as typical in visual ranges, orthodromes may be replaced with loxodromes (rhumb lines). Loxodrome is a curve following Earth surface, connecting two points with constant heading. It has similar path to orthodrome, if the points are relatively far from poles a close together. Simplified heading along the line is

$$\alpha_{proj} \doteq \arctan\left(\frac{\lambda\cos(\varphi) - \lambda_0\cos(\varphi_0)}{\varphi - \varphi_0}\right).$$

Note that loxodrome approximation will fail horribly near poles as the curve will run circles around the pole. Angular distance along loxodrome is

$$\phi \doteq \sqrt{(\varphi - \varphi_0)^2 + (\lambda \cos(\varphi) - \lambda_0 \cos(\varphi_0))^2}$$

and a horizontal distance along the arc between those points is

$$d = \phi \cdot R_{(\varphi_0)},$$

assuming flattening difference is close to zero, so R is constant along the curve. With the approximation of local flat Earth surface perpendicular to the normal, the simplified vertical projection angle is

$$\beta_{proj} \doteq \arctan\left(\frac{h-h_0}{d}\right).$$

This approximation will fail at higher altitudes when the visibility range is high enough to make the Earth curvature observable.

2.6 Output creation

The output image example appendix shows the actual output image. In the top left corner is the current speed over ground as reported by the satellite navigation, while in the top right corner is the current altitude above mean sea level. Speed is displayed in kilometers per hour and the altitude in meters. In the top center is the current waypoint name and distance in kilometers, if provided by the NMEA 0183 navigation data sentence (RMB). There is a heading ruler in the bottom, it is scaled by camera field of view and oriented to current heading. Course to any visible object may be deducted directly from the ruler. There are two markers, the arrow marker point to current track angle as reported by the satellite navigation. The circle marker is the course to current waypoint if any. The dashed horizontal line in the center of the image is the horizon line, it follows the horizon as calculated by inertial subsystem by moving up or down and rotating around its center. Orientation is determined by the marginal markers, they points always to the ground. If the horizon is not visible, dashed arrow is shown instead pointing up or down to where the horizon is hidden. Application accepts database of landmarks, following file was used in this example:

```
# Encoding ISO/IEC 8859-1
# Test landmarks (lat[rad], lon[rad], alt[m], name)
25e-5, 1e-5, 0, testA
50e-5, -1e-5, 100, testB
```

The testA landmark label is visible in the image, its location is centered directly above its projection. Projections are calculated in conjunction of both navigation subsystems and graphic acceleration. Large landmark database may be supplemented to provide spatial navigation references, only visible landmarks will be shown. This overlay is rendered in real time over the source video (there is none in the example), all rendering and data gathering methods are provided by the respective subsystems. Upon correct configuration, overlay elements (ruler, horizon and labels) should be aligned with the real visible places in the video frame. System allows free six degrees of freedom movement of the camera while still being able to render the overlay correctly. Waypoint management and route planning is done solely by the external navigation system, it is expected that it will provide its own user interface. This allows connection for example to PDAs or other specialized devices which delivers classic 2D moving map navigation to the user with its own controls.

3 Hardware

TODO: Hardware description

OMAP4460 application processor[10]

- two ARM Cortex-A9 SMP general-purpose processors
- IVA 3 video accelerator, 1080p capable
- image signal processor, 20MP capable
- SGX540 3D graphics accelerator, OpenGL ES 2.0 compatible
- HDMI v1.3 video output

MPU-9150 motion tracking device[11]

- embedded MPU-6050 3-axis gyroscope and accelerometer
- embedded AK8975 3-axis digital compass
- fully programmable, I²C interface

OV5640 image sensor

- 1080p, 5MP resolution
- raw RGB or YUV output

4 Conclusion

TODO: Conclusion

References

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List of Annexes

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A Threads example

In this example two threads share standard input and output, access is restricted by *mutex* so only one thread may use the shared resource at any time.

```
#include <stdio.h>
   #include <pthread.h>
   void *worker(void *arg)
   {
       static int counter = 1;
       pthread_mutex_t *mutex = (pthread_mutex_t*)arg;
       char *buffer;
       // Lock mutex to restrict access to stdin and stdout
10
       pthread mutex lock(mutex);
       printf("This is worker %d, enter something: ", counter++);
12
       scanf("%ms", &buffer);
       pthread mutex unlock(mutex);
14
       return (void*)buffer;
16
   }
17
18
   int main()
19
20
       pthread mutex t mutex;
21
       pthread t thread1, thread2;
       char *retval1, *retval2;
23
       // Initialize two threads with shared mutex, use default parameters
25
       pthread mutex init(&mutex, NULL);
       pthread_create(&thread1, NULL, worker, (void*)&mutex);
       pthread create(&thread2, NULL, worker, (void*)&mutex);
28
29
       // Wait for both threads to finish and display results
       pthread_join(thread1, (void**)&retval1);
31
       pthread join(thread2, (void**)&retval2);
       printf("Thread 1 returned with `%s`.\n", retval1);
33
       printf("Thread 2 returned with `%s`.\n", retval2);
34
35
       pthread mutex destroy(&mutex);
       return 0;
37
   }
38
```

B Video capture example

In this example video device is configured to capture frames using memory mapping. These frames are dumped to standard output, instead of further processing.

```
#include <fcntl.h>
   #include <unistd.h>
   #include <poll.h>
   #include <sys/mman.h>
   #include <sys/ioctl.h>
   #include <linux/videodev2.h>
   int main()
   {
       // Open device
10
       int fd = open("/dev/video0", O_RDWR | O_NONBLOCK);
11
12
       // Set video format
       struct v412_format format =
14
       {
            .type = V4L2_BUF_TYPE_VIDEO_CAPTURE,
16
            .fmt =
            {
18
                .pix =
                {
20
                     .width = 320,
21
                     .height = 240,
22
                     .pixelformat = V4L2_PIX_FMT_RGB32,
23
                     .field = V4L2_FIELD_NONE,
24
                     .colorspace = V4L2 COLORSPACE SMPTE170M,
25
                },
26
            },
27
       };
       ioctl(fd, VIDIOC_S_FMT, &format);
29
30
       // Request buffers
31
       struct v412 requestbuffers requestbuffers =
33
            .type = V4L2_BUF_TYPE_VIDEO_CAPTURE,
34
            .memory = V4L2_MEMORY_MMAP,
35
            .count = 4,
       };
37
       ioctl(fd, VIDIOC_REQBUFS, &requestbuffers);
       void *pbuffers[requestbuffers.count];
39
```

```
// Map and enqueue buffers
41
       int i;
       for(i = 0; i < requestbuffers.count; i++)</pre>
43
            struct v412_buffer buffer =
45
            {
                 .type = V4L2 BUF TYPE VIDEO CAPTURE,
47
                 .memory = V4L2_MEMORY_MMAP,
48
                 .index = i,
49
            };
50
            ioctl(fd, VIDIOC_QUERYBUF, &buffer);
51
            pbuffers[i] = mmap(NULL, buffer.length,
52
                                 PROT_READ | PROT_WRITE, MAP_SHARED,
53
                                 fd, buffer.m.offset);
54
            ioctl(fd, VIDIOC QBUF, &buffer);
55
       }
56
       // Start stream
58
       enum v412 buf type buf type = V4L2 BUF TYPE VIDEO CAPTURE;
       ioctl(fd, VIDIOC STREAMON, &buf type);
60
61
       while(1)
62
       {
63
            // Synchronize
64
            struct pollfd fds =
65
            {
66
                 .fd = fd,
67
                 .events = POLLIN
68
            };
69
            poll(&fds, 1, -1);
70
71
            // Dump buffer to stdout
72
            struct v412_buffer buffer =
73
            {
74
                 .type = V4L2_BUF_TYPE_VIDEO_CAPTURE,
75
                 .memory = V4L2_MEMORY_MMAP,
76
            };
77
            ioctl(fd, VIDIOC DQBUF, &buffer);
            write(1, pbuffers[buffer.index], buffer.bytesused);
79
            ioctl(fd, VIDIOC_QBUF, &buffer);
       }
81
   }
82
```

C Colorspace conversion example

In this example RGB to YUV color-space conversion is implemented in fragment shader. Each input channel has its own texturing unit, texture coordinates are divided by sub-sampling factor 4:2:0.

```
uniform sampler2D texY, texU, texV;
   varying vec2 texCoord;
   void main()
       float y = \text{texture2D(texY, texCoord)}.a * 1.1644 - 0.062745;
       float u = texture2D(texU, texCoord / 2).a - 0.5;
       float v = texture2D(texV, texCoord / 2).a - 0.5;
       gl_FragColor = vec4(
10
           y + 1.596 * v,
11
           y - 0.39176 * v - 0.81297 * u,
           y + 2.0172 * u,
13
           1.0);
14
   }
15
```

D Output image example

This is the actual graphical output rendered by the application, there is no video source (solid white background).

19 km/h TEST01, 0.0 km 102 m

testA

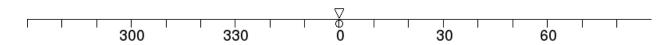


Figure 3: Visible output