THE COMPUTATION OF WEAKLY IONIZED HYPERSONIC FLOWS IN THERMO-CHEMICAL NONEQUILIBRIUM

A DISSERTATION SUBMITTED TO THE DEPARTMENT OF AERONAUTICS AND ASTRONAUTICS AND THE COMMITTEE ON GRADUATE STUDIES OF STANFORD UNIVERSITY IN PARTIAL FULFILLMENT OF THE REQUIREMENTS FOR THE DEGREE OF DOCTOR OF PHILOSOPHY

By Graham V. Candler June 1988

©Copyright 1988 by Graham V. Candler

I certify that I have read this thesis and in my opinion it is fully adequate, in scope and quality, as a dissertation for the degree of Doctor of Philosophy.
(Principal Adviser)
I certify that I have read this thesis and in my opinion it is fully adequate, in scope and quality, as a dissertation for the degree of Doctor of Philosophy.
I certify that I have read this thesis and in my opinion it is fully adequate, in scope and quality, as a dissertation for the degree of Doctor of Philosophy.
I certify that I have read this thesis and in my opinion it is fully adequate, in scope and quality, as a dissertation for the degree of Doctor of Philosophy.
Approved for the University Committee on Graduate Studies:
Dean of Graduate Studies

ACKNOWLEDGMENTS

This dissertation was made possible with the help of many people. Ellen McGrattan provided help and understanding throughout the process of writing it. My adviser, Professor Bob MacCormack provided insight, encouragement, and direction. Dr. Chul Park and Professor Dean Chapman helped greatly through their technical advice and suggestions. They along with Professor Donald Baganoff had the less than enviable task of reading this thesis in its rough form. My friends and fellow graduate students contributed greatly and made the process more enjoyable. I would particularly like to thank Dr. Ugo Piomelli, Tahir Gökçen, Kurt Fiscko, and Forrest Lumpkin. The use of the computers and the facilities at NASA Ames made this work possible.

This research was supported by the following sponsors. NASA under a Hypersonic Training and Research Grant No. NAGW 965, SDIO/IST managed by the Army Research Office under contract No. DAAl03-86-K-0139, and the Department of the Air Force under contract No. F33615-86-C-3015.

•

THE COMPUTATION OF WEAKLY IONIZED HYPERSONIC FLOWS IN THERMO-CHEMICAL NONEQUILIBRIUM

Graham V. Candler, Ph.D. Stanford University, 1988

Several conceptual designs for vehicles that would fly in the atmosphere at hypersonic speeds have been developed recently. For the proposed flight conditions the air in the shock layer that envelops the body is at a sufficiently high temperature to cause chemical reaction, vibrational excitation, and ionization. However, these processes occur at finite rates which, when coupled with large convection speeds, cause the gas to be removed from thermo-chemical equilibrium. This non-ideal behavior affects the aerothermal loading on the vehicle and has ramifications in its design.

In this dissertation, the differential equations that describe a hypersonic thermochemical nonequilibrium flowfield are derived. A numerical method to solve these equations in two dimensions is discussed and results for some test cases are presented. The state of the gas is represented with seven chemical species, a separate vibrational temperature for each diatomic species, an electron translational temperature, and a mass-averaged translational-rotational temperature for the heavy-particles. The equations for this gas model are solved numerically in a fully coupled fashion using an implicit finite volume time-marching technique. Gauss-Seidel line-relaxation is used to reduce the cost of the solution and flux-dependent differencing is employed to maintain stability.

The numerical method has been tested against several experiments. The calculated bow shock wave detachment on a sphere and two cones was compared to those measured in ground testing facilities. The computed peak electron number density on a sphere-cone was compared to that measured in a flight test. In each case the results from the numerical method were in excellent agreement with experiment. The technique was then used to predict the aerothermal loads on an Aeroassisted Orbital Transfer Vehicle including radiative heating. These results indicate that the current physical model of high temperature air is appropriate and that the numerical algorithm is capable of treating this class of flows.

Approved for publication:

Ву _	
Ů	For Major Department
	<u> </u>
_	
Ву _	
	Dean of Graduate Studies & Research

CONTENTS

	Pa	ige
Abstract	;	vii
List of T	ables	xi
List of F	figures	ciii
Nomenc	lature	vii
Chapter		
I. N	IOTIVATIONS, CONCERNS, AND BACKGROUND	1
I.1 I.2	Introduction	
I.3 I.4	Relevant Physics of a Hypersonic Flow	3
I.5	The Scope of the Current Work $\ \ \ldots \ \ \ldots$	
II. N	IATHEMATICAL FORMULATION	9
II.1 II.2 II.3 II.4 II.5 II.6 II.7 II.8 II.9	Introduction Basic Assumptions The Conservation Equations Simplifications to the Governing Equations Equations of State Shear Stresses, Heat Fluxes and Diffusion Velocities Energy Exchange Mechanisms Chemical Source Terms Boundary Conditions Summary of Governing Equations	9 11 16 18 20 23 29 32 32
III. T	HE NUMERICAL METHOD	35
III.1 III.2 III.3 III.4 III.5	Introduction	$\frac{35}{38}$ $\frac{44}{44}$
	HE COMPARISON OF COMPUTATIONS TO	51
IV.1 IV.2 IV.3	Introduction	

IV.4 Comparison of Shock Detachment Distances on Cones IV.5 Comparison to the RAM-C Flight Experiments IV.6 Conclusions from the Test Cases						62
V. THE COMPUTATION OF AEROTHERMAL LOADS	O	N				
AN AXISYMMETRIC AEROASSISTED ORBITAL						
TRANSFER VEHICLE	•			•	•	73
V.1 Introduction						73
V.2 Radiation Calculations						
V.3 Results for the AFE Vehicle						
V.5 Summary of Axisymmetric AFE Results						86
VI. SUMMARY AND CONCLUSIONS						87
VI.1 Summary						83
VI.2 Conclusions						
VI.3 Future Research						
Appendix A						93
Appendix B						
Appendix C						
Appendix D						
References						
Figures					1	119

LIST OF TABLES

Table		Page
4.1	Conditions for Test Cases 1 and 2	. 54
4.2	Conditions for Test Cases 3 and 4	. 58
4.3	Conditions for RAM-C II Test Cases	. 63
4.1	Conditions for AFE Test Cases	. 77
A.1	Heats of formation	. 93
A.2	Viscosity coefficients for Blottner model	. 93
A.3	Electronic constants	. 94
A.4	Molecular constants for energy exchange models	. 94
A.5	Arrhenius coefficients for forward reaction rates	. 95
A.6	Constants for equilibrium reaction constants	. 96

LIST OF FIGURES

Figure		Page
2.1	Electron-vibration relaxation time as a function of electron translational temperature	. 119
3.1	Finite-volume grid numbering scheme.	. 119
4.1	35×50 grid used for Test Case 1	
4.2	Shock standoff for Test Case 1	. 120
4.3	Mass concentrations on stagnation streamline for Test Case 1	. 121
4.4	Temperatures on stagnation streamline for Test Case 1	. 121
4.5	Mach number contours for Test Case 1	. 122
4.6	Percent O_2 mass concentration contours for Test Case 1	. 122
4.7	Translational-rotational temperature contours for Test Case 1	. 123
4.8	N_2 vibrational temperature contours for Test Case 1	
4.9	Shock standoff for Test Case 2	. 124
4.10	Mass concentrations on stagnation streamline for Test Case 2	. 125
4.11	Temperatures on stagnation streamline for Test Case 2	. 125
4.12	Mach number contours for Test Case 2	. 126
4.13	Percent N_2 mass concentration contours for Test Case 2	. 126
4.14	Percent O_2 mass concentration contours for Test Case 2	. 127
4.15	Translational-rotational temperature contours for Test Case 2	. 127
4.16	N_2 vibrational temperature contours for Test Case 2	. 128
4.17	Electron translational temperature contours for Test Case 2	. 128
4.18	35×50 grid used for Test Case 3	. 129
4.19	Close-up of grid used for Test Case 3 showing stagnation point region	. 129
4.20	Shock standoff for Test Case 3	. 130
4.21	Surface pressure coefficient for Test Case 3	. 130
4.22	Mass concentrations on stagnation streamline for Test Case 3	. 131
4.23	Temperatures on stagnation streamline for Test Case 3	. 131
4.24	Mass concentrations at $x = 1.2 \mathrm{mm}$ for Test Case 3	. 132
4.25	Temperatures at $x = 1.2 \mathrm{mm}$ for Test Case 3	. 132
4.26	Mass concentrations at $x = 15.6 \mathrm{mm}$ for Test Case 3	. 133
4.27	Temperatures at $x = 15.6 \mathrm{mm}$ for Test Case 3	. 133
4.28	Mach number contours for Test Case 3	. 134
4.29	Percent O_2 mass concentration contours for Test Case 3	. 134
4.30	Translational-rotational temperature contours for Test Case 3	. 135
4.31	O_2 vibrational temperature contours for Test Case 3	. 135

4.32	35×50 grid used for Test Case 4	6
4.33	Shock standoff for Test Case 4	7
4.34	Surface pressure coefficient for Test Case 4	7
4.35	Mass concentrations at $x = 1.8 \mathrm{mm}$ for Test Case 4	8
4.36	Temperatures at $x = 1.8 \mathrm{mm}$ for Test Case 4	8
4.37	Mass concentrations at $x = 18.1 \mathrm{mm}$ for Test Case 4	9
4.38	Temperatures at $x = 18.1 \mathrm{mm}$ for Test Case 4	9
4.39	Mach number contours for Test Case 4	0
4.40	Percent O_2 mass concentration contours for Test Case 4	0
4.41	Translational-rotational temperature contours for Test Case 4	1
4.42	O_2 vibrational temperature contours for Test Case 4	1
4.43	35×50 grid used for RAM-C Test Cases	2
4.44	Peak electron number density vs axial distance for RAM-C at 61 km 140	3
4.45	Peak electron number density vs axial distance for RAM-C at 71 km 14	3
4.46	Peak electron number density vs axial distance for RAM-C at 81 km 14	4
4.47	Electron number density vs normal distance for RAM-C at $x/r_n = 8.1$ 14	4
4.48	Mass concentrations on stagnation streamline for RAM-C at 61 km 14	5
4.49	Temperatures on stagnation streamline for RAM-C at 61 km	5
4.50	Mass concentrations at $x/r_n = 1.0$ for RAM-C at 61 km	6
4.51	Temperatures at $x/r_n = 1.0$ for RAM-C at 61 km	6
4.52	Surface pressure coefficient for RAM-C at 71 km	7
4.53	Mass concentrations on stagnation streamline for RAM-C at 71 km 14	8
4.54	Molar concentrations on stagnation streamline for RAM-C at 71 km 14	8
4.55	Temperatures on stagnation streamline for RAM-C at 71 km	9
4.56	Vibrational temperatures on stagnation streamline for RAM-C at $71\mathrm{km}$. 14	9
4.57	Density ratio on stagnation streamline for RAM-C at 71 km	0
4.58	Mass concentrations at $x/r_n=1.0$ for RAM-C at 71 km	1
4.59	Temperatures at $x/r_n=1.0$ for RAM-C at 71 km	1
4.60	Mass concentrations at $x/r_n = 8.1$ for RAM-C at 71 km	2
4.61	Temperatures at $x/r_n = 8.1$ for RAM-C at 71 km	2
4.62	Molar concentrations at $x/r_n = 8.1$ for RAM-C at 71 km	3
4.63	Mach number contours for RAM-C at 71 km	3
4.64	Percent N_2 mass concentration contours for RAM-C at 71 km	4
4.65	Percent O_2 mass concentration contours for RAM-C at 71 km 15	4
4.66	Logarithm (base 10) of electron number density contours for RAM-C at	
	71 km	5
4.67	Translational-rotational temperature contours for RAM-C at $71\mathrm{km}$ 15	5
4.68	N_2 vibrational temperature contours for RAM-C at 71 km	6

4.69	Electron translational temperature contours for RAM-C at 71 km 156
4.70	Mass concentrations on stagnation streamline for RAM-C at $81\mathrm{km}$ 157
4.71	Temperatures on stagnation streamline for RAM-C at 81 km 157
4.72	Density ratio on stagnation streamline for RAM-C at 81 km
4.73	Mass concentrations at $x/r_n = 1.0$ for RAM-C at 81 km
4.74	Temperatures at $x/r_n = 1.0$ for RAM-C at 81 km
5.1	35×50 grid used for AFE at 78 km
5.2	Density ratio on stagnation streamline for AFE at 78 km
5.3	Mass concentrations on stagnation streamline for AFE at 78 km 161
5.4	Molar concentrations on stagnation streamline for AFE at $78\mathrm{km}$ 162
5.5	Electron number density on stagnation streamline for AFE at $78\mathrm{km}$ $162\mathrm{mm}$
5.6	Temperatures on stagnation streamline for AFE at 78 km
5.7	Radiation intensity on stagnation streamline for AFE at 78 km 163
5.8	Convective heat transfer vs distance from nose for AFE at 78 km 164
5.9	Radiative heat transfer vs distance from nose for AFE at 78 km 164
5.10	Pressure contours for AFE at 78 km
5.11	Percent N_2 mass concentration contours for AFE at 78 km 165
5.12	Percent O_2 mass concentration contours for AFE at 78 km 166
5.13	Logarithm (base 10) of electron number density contours for AFE at 78 km
5.14	Translational-rotational temperature contours for AFE at 78 km 167
5.15	N_2 vibrational temperature contours for AFE at 78 km 167
5.16	Electron translational temperature contours for AFE at 78 km 168
5.17	Radiation intensity contours for AFE at 78 km
5.18	Density ratio on stagnation streamline for AFE at 90 km
5.19	Mass concentrations on stagnation streamline for AFE at 90 km 169
5.20	Molar concentrations on stagnation streamline for AFE at 90 km 170
5.21	Electron number density on stagnation streamline for AFE at 90 km 170
5.22	Temperatures on stagnation streamline for AFE at 90 km
5.23	Radiation intensity on stagnation streamline for AFE at 90 km 171
5.24	Convective heat transfer vs distance from nose for AFE at 90 km 172
5.25	Radiative heat transfer vs distance from nose for AFE at 90 km 172
5.26	Pressure contours for AFE at 90 km
5.27	Percent N_2 mass concentration contours for AFE at 90 km
5.28	Percent O_2 mass concentration contours for AFE at 90 km
5.29	Percent NO ⁺ mass concentration contours for AFE at 90 km
5.30	Logarithm (base 10) of electron number density contours for AFE at
	90 km

5.31	Translational-rotational temperature contours for AFE at 90 km	175
5.32	N_2 vibrational temperature contours for AFE at 90 km	176
5.33	Electron translational temperature contours for AFE at 90 km	176
5.34	Streamlines for AFE at 90 km	177
5.35	Radiation intensity contours for AFE at 90 km	177

NOMENCLATURE

Roman Symbols

E

Speed of sound, see (C.8). a \tilde{a} Modified speed of sound, see (C.14). Modified speeds of sound, see (C.16). $a_+, a_ A_{im}$ Constants for use in equilibrium reaction constant, see (A.1). Constant used in Millikan and White vibrational relaxation model, see A_{sr} (2.7.3). A_s, B_s, C_s Constants used in Blottner viscosity model, see (2.6.3). A', B'Jacobians of F' and G', see (3.3.3). A'_{+}, A'_{-} Flux-split Jacobians of F'. B'_{+}, B'_{-} Flux-split Jacobians of G'. $\hat{A}, \ \hat{B}, \dots \hat{E}$ Block matrices defined in (3.4.2). Mass concentration of species s, $c_s = \rho_s/\rho$. c_s Thermal speed of species s, see (2.7.7). $c_{\mathbf{s}}$ Mass-averaged specific heat at constant pressure. c_p Mass-averaged specific heat at constant volume. c_v Translational-rotational specific heat at constant volume of species s. c_{vs} Translational specific heat at constant volume of species s. $c_{v \operatorname{tr} s}$ Rotational specific heat at constant volume of species s. $c_{v \text{rot } s}$ Vibrational specific heat at constant volume of species s. $c_{v \text{vib } s}$ Jacobian of W with respect to U. C $C_A, C_{A'}$ Jacobians used in diagonalization of A', see (3.3.7). C_{f_m} Constant for use in the Arrhenius form of reaction rates, see (2.8.4). \mathcal{D} Diffusion coefficient of uncharged species. \mathcal{D}_s Diffusion coefficient of species s. Dimensionality of the problem. dMolecular diameter of species s. d_{s} Electron charge. Symbol used to denote electrons. e^{-} Electron translational energy per unit mass. e_e Electronic energy of species s per unit mass, see (2.5.8). $e_{\mathrm{el}\,s}$ Vibrational energy per unit mass of species s. e_{vs} Vibrational energy per unit mass of species s at the local T. $e_{vs}^*(T)$

Total energy per unit volume.

E Block matrix for the use in implicit boundary condition treatment, see

(3.5.5).

 E_e Electron translational energy per unit volume. E_{vs} Vibrational energy per unit volume of species s.

 E_s Total energy per unit volume of species s.

 \tilde{E}_i Electric field in the *i* direction.

F, G Flux vectors in the x and y directions.

F', G' Rotated flux vectors in the ξ and η directions, see (3.2.8).

 $F'_{\rm I}, F'_{\rm v}$ Inviscid and viscous components of F'. $G'_{\rm I}, G'_{\rm v}$ Inviscid and viscous components of G'.

 g_{is} Degeneracy of electronic states for state i of species s, see (2.5.8).

 h_s Enthalpy per unit mass of species s, see (2.5.7).

 h_s° Heat of formation of species s.

i, *j* Indices used for vector components and also to denote grid points.

IL, JL Size of grid in the ξ and η directions.

k Boltzmann constant.

 k_{f_m} Forward reaction rate of reaction m, see (2.8.4). k_{b_m} Backward reaction rate of reaction m, see (2.8.4).

K Generic constant.

 K_{eq_m} Equilibrium constant of reaction m, see (2.8.4).

Kn Knudsen number. l Reaction number.

 $\mathcal{L}e$ Lewis number, see (2.6.7).

m Number of diatomic species or reaction number.

M Mass-averaged atomic weight.

 \mathcal{M} Mach number.

 M_s Atomic weight of species s.

 M_{η} Matrix used in the implicit treatment of the viscous terms, see (3.3.21).

n Number of species or time level of computation.

N Jacobian used in the implicit viscous terms, see (3.3.23).

N₂ Symbol used for diatomic nitrogen.
 N Symbol used for atomic nitrogen.
 NO Symbol used for nitric oxide.

NO⁺ Symbol used for ionized nitric oxide.

 N_s Number density of species s,

 \hat{N} Avogadro's number.

O₂ Symbol used for diatomic oxygen.
O Symbol used for atomic nitrogen.

Pressure. p

Species s partial pressure. p_s

 P_{si} Momentum transfer rate of species s in direction i. P_{sr} Probability of vibration energy transfer, see (2.7.13). Translational-rotational heat conduction in direction j. q_j Electron translational heat conduction in direction j.

 q_{ej}

Total heat conduction in direction j. q_{ti}

Vibrational heat conduction of species s in direction j. q_{vsj}

 Q_s Generic energy transfer rate of species s. Vibrational energy transfer rate of species s. Q_{vs}

 Q_{e-vs} Electron-vibration energy transfer rate of species s.

 Q_{T-e} Translation-electron energy transfer rate.

 Q_{T-vs} Translation-vibration energy transfer rate of species s. Q_{v-vs} Vibration-vibration energy transfer rate of species s.

Nose radius. r_n Gas constant. R

 \mathcal{R} Reaction, see (2.8.2). R_A

Rotation matrix.

ReReynolds number based on nose radius or total length.

 Re_{r} Local Reynolds number. R_s Gas constant of species s. \bar{R} Mass-averaged gas constant.

RHSRight hand side of equation, see (3.5.5).

 s'_{ix}, s'_{iy} Direction cosines of surface $i+\frac{1}{2}$, j in the x and y directions, see (B.5). Direction cosines of surface $i, j + \frac{1}{2}$ in the x and y directions, see (B.5).

 $s'_{jx}, \ s'_{jy}$ SJacobian of V with respect to U.

 S_s Exponent for use in the diffusion model of vibrational relaxation, see

(2.7.6).

Time. t

TTranslational-rotational temperature. T_e Electron translational temperature. T_{vs} Vibrational temperature of species s.

Average temperature for use in reaction rates.

 \mathbf{T} Metric transformation matrix, see (B.1).

Mass-averaged velocity components in x and y direction. u, v

Mass-averaged velocity vector. u_j

Species s velocity vector. u_{sj}

Species s diffusion velocity vector. v_{sj}

 u_s, v_s Diffusion velocities of species s in x and y directions.

u', v' Mass-averaged velocity components normal and tangential to a surface.

 \tilde{u}', \tilde{v}' Modified velocity components, see (C.7).

U Vector of conserved quantities. V, \mathbf{V} Vectors of non-conserved variables.

 w_s Mass source term due to reaction of species s.

W Source vector.

x, y Cartesian coordinates.

 x_j Cartesian coordinate vector.

 X_s Molar concentration of species s.

 Z_s Charge number of species s.

 Z_{sr} Number of s-r collisions per unit time.

Greek Symbols

 $\bar{\gamma}$ Effective ratio of specific heats, see (C.10).

 δ Boundary layer thickness.

 δ_{ij} Kronecker symbol.

 $\delta F_{\rm v}, \ \delta G_{\rm v}$ Implicit changes in the Cartesian viscous fluxes. $\delta F_{\rm v}', \ \delta G_{\rm v}'$ Implicit changes in the rotated viscous fluxes.

 δU Implicit change in the vector of conserved quantities.

 Δt Time step.

 ΔU Explicit change in the vector of conserved quantities.

 ε_{vs} Vibrational energy per particle of species s.

 η_m Constant for use in Arrhenius form of reaction rate for reaction m, see

(2.8.4).

 θ_c Cone half-angle.

 θ_{elis} Electronic activation energy of state i for species s, see (2.5.8).

 θ_m Characteristic temperature of reaction for reaction m.

 θ_s Characteristic temperature for use in diffusive model of vibrational relax-

ation.

 θ_{vs} Characteristic temperature of vibration of species s. κ Conductivity of translational-rotational temperature. κ_{e} Conductivity of electron translational temperature. κ_{vs} Conductivity of vibrational temperature of species s.

 λ Second coefficient of viscosity.

 $\Lambda_{A'}$ Diagonal matrix of convection speeds of characteristic variables for A'.

 $\Lambda_{A'+}$, $\Lambda_{A'-}$ Diagonal matrices of positive and negative convection speeds of charac-

teristic variables for A'.

 μ Viscosity of mixture.

 μ_s Species viscosity.

 ξ , η General coordinate directions, ξ along the body and η normal to the body.

 ρ Density.

 ρ_s Density of species s.

 σ_{sr} Collision cross-section for species s and r. (Specific to the process.)

 τ_{cs} Collision limited vibrational relaxation rate of species s, see (2.7.7).

 τ_{es} Electron energy relaxation rate of species s, see (2.7.12).

 τ_{ij} Shear stress tensor, see (2.6.1).

 au_{sij} Species shear stress tensor.

 $\tau_{s_{L-T}}$ Landau-Teller vibrational relaxation rate of species s, see(2.7.2).

 τ_{vs} Vibrational relaxation rate of species s, see (2.7.6). ϕ_s Parameter for use in Wilke mixing model, see (2.6.5).

 Ψ Reactivity of the gas, see (4.2.2).

Difference Operators

$$\frac{D_{+}Z_{ij}}{\Delta\xi} \equiv Z_{i+1j} - Z_{ij}$$
 $\frac{D_{-}Z_{ij}}{\Delta\xi} \equiv Z_{ij} - Z_{i-1j}$

$$\frac{D_{+}Z_{ij}}{\Delta\eta} \equiv Z_{ij+1} - Z_{ij}$$
 $\frac{D_{-}Z_{ij}}{\Delta\eta} \equiv Z_{ij} - Z_{ij-1}$