Theory of Elasticity

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upcoming schedule

- Oct 7 Exam 2 Review
- Oct 8 Homework 4 Self-grade Due, Homework 5 Due
- (Oct 12) Fall Break (No Class)
- Oct 14 Exam 2
- Oct 19 Strain Energy
- Oct 21 Virtual Work

outline

- exam
- group problems
- stress and equilibrium
- material behavior
- problem formulation

exam

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exam format

- Similar format to last exam
- Three problems
- Focus on organizing your work clearly to maximize partial credit

group problems

problem one - thermoelasticity

As a first-order model of the problem of freezing water in a glass bottle, we treat water as a thermoelastic solid and the glass as a fixed boundary. Find the stress and strain field in the water as a function of the elastic properties (E, ν) and the coefficient of thermal expansion (α) .

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problem two - inverse solution

Consider the stress field

$$\sigma = \begin{bmatrix} Ay & 0 & 0 & 0 & 0 & 0 & 0 & 0 \end{bmatrix}$$

Show that this is a valid solution to an elasticity problem. What problem does it solve?

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problem three - semi-inverse

To solve the problem of torsion in prismatic bars we consider the displacement field

$$u = -\alpha yz$$
, $v = -\alpha xz$, $w = w(x, y)$

Solve this problem using the boundary conditions for a solid square cross-section.

stress and equilibrium

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topics

- Traction
- Stress transformation
- Principal stress
- Equilibrium

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derivations

- Cauchy's stress theorem
- Max shear stress for plane stress
- Mohr's circle

stress tensor

 To simplify the notation, we introduce the stress tensor

$$\sigma_{ij} = t_j^{(\hat{\mathbf{e}}_i)}$$

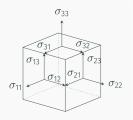


Figure 1: stress tensor illustrated on a cube

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traction

 We can find some interesting information about the traction vector by considering an arbitrary tetrahedron with some traction t
 ⁽ⁿ⁾ applied to the surface • If we consider the balance of forces in the x_1 -direction

$$t_1 dA - \sigma_{11} dA_1 - \sigma_{21} dA_2 - \sigma_{31} dA_3 + b_1 \rho dV = 0$$

• The area components are:

$$dA_1 = n_1 dA$$
$$dA_2 = n_2 dA$$
$$dA_3 = n_3 dA$$

• And $dV = \frac{1}{3}hdA$.

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traction

$$t_1 dA - \sigma_{11} n_1 dA - \sigma_{21} n_2 dA - \sigma_{31} n_3 dA + b_1 \rho \frac{1}{3} h dA = 0$$

• If we let $h \to 0$ and divide by dA

$$t_1 = \sigma_{11}n_1 + \sigma_{21}n_2 + \sigma_{31}n_3$$

• We can write this in index notation as

$$t_1 = \sigma_{i1} n_i$$

traction

• We find, similarly

$$t_2 = \sigma_{i2}n_i$$

$$t_3 = \sigma_{i3}n_i$$

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traction

• We can further combine these results in index notation as

$$t_i = \sigma_{ii} n_i$$

• This means with knowledge of the nine components of σ_{ij} , we can find the traction vector at any point on any surface

maximum shear stress

- For plane stress problems, we can also use the stress transformation equations to find the maximum shear stress
- We desire to maximize this equation:

$$\tau_{xy}' = \frac{\sigma_y - \sigma_x}{2} \sin 2\theta + \tau_{xy} \cos 2\theta$$

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maximum shear stress

• Taking the derivative with respect to θ gives

$$\frac{\partial}{\partial \theta}(\tau'_{xy}) = (\sigma_y - \sigma_x)\cos 2\theta - 2\tau_{xy}\sin 2\theta = 0$$

• Which we can use to find 2θ

$$2\theta = \tan^{-1}\left(\frac{(\sigma_y - \sigma_x)}{2\tau_{xy}}\right)$$

maximum shear stress

Substituting back into the original equation gives

$$\tau_{\textit{max}}' = \frac{\sigma_{\textit{y}} - \sigma_{\textit{x}}}{2} \sin \left[\tan^{-1} \left(\frac{(\sigma_{\textit{y}} - \sigma_{\textit{x}})}{2\tau_{\textit{xy}}} \right) \right] + \tau_{\textit{xy}} \cos \left[\tan^{-1} \left(\frac{(\sigma_{\textit{y}} - \sigma_{\textit{x}})}{2\tau_{\textit{xy}}} \right) \right]$$

Note that

$$\sin(\tan^{-1}(x)) = \frac{x}{\sqrt{1+x^2}}$$
$$\cos(\tan^{-1}(x)) = \frac{1}{\sqrt{1+x^2}}$$

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maximum shear stress

We note that

$$\sqrt{1 + \left(\frac{\sigma_y - \sigma_x}{2\tau_{xy}}\right)^2} = \frac{\sqrt{(\sigma_y - \sigma_x)^2 + 4\tau_{xy}^2}}{2\tau_{xy}}$$

And thus we find

$$\tau_{\text{max}} = \frac{(\sigma_{\text{y}} - \sigma_{\text{x}})^2}{2\sqrt{(\sigma_{\text{y}} - \sigma_{\text{x}})^2 + 4\tau_{\text{xy}}^2}} + \frac{4\tau_{\text{xy}}^2}{2\sqrt{(\sigma_{\text{y}} - \sigma_{\text{x}})^2 + 4\tau_{\text{xy}}^2}}$$

maximum shear stress

· Adding the terms and simplifying, we find

$$\tau_{\text{max}} = \sqrt{\left(\frac{\sigma_{\text{y}} - \sigma_{\text{x}}}{2}\right)^2 + \tau_{\text{xy}}^2}$$

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tractions

- We can use what we know about principal values to find some interesting things about the tractions
- Consider the traction vector on an arbitrary internal face, and decompose into Normal and Shear components.

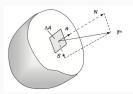


Figure 2: arbitrary body with arbitrary loading applied

tractions

• The normal component can be found using the dot product

$$N = \hat{T}^n \cdot \hat{n}$$

The shear component can be found using the Pythagorean theorem

$$S^2 = |\hat{T}^n|^2 - N^2$$

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tractions

 We now use the stress tensor in the principal direction to simplify the calculations

$$N = \hat{T}^n \cdot \hat{n}$$

$$= T_i^n n_i$$

$$= \sigma_{ji} n_j n_i$$

$$= \sigma_1 n_1^2 + \sigma_2 n_2^2 + \sigma_3 n_3^2$$

We also know that

$$|\hat{T}^n|^2 = \hat{T}^n \cdot \hat{T}^n$$

$$= T_i^n T_i^n$$

$$= \sigma_{ji} n_j \sigma_{ki} n_k$$

$$= \sigma_i^2 n_1^2 + \sigma_2^2 n_2^2 + \sigma_3^2 n_3^2$$

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mohr's circle

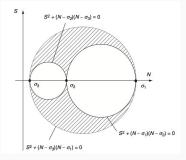
 If we constrain the normal vector to be a unit vector we can formulate the following inequalities

$$S^{2} + (N - \sigma_{2})(N - \sigma_{3}) \ge 0$$

 $S^{2} + (N - \sigma_{3})(N - \sigma_{1}) \le 0$
 $S^{2} + (N - \sigma_{1})(N - \sigma_{2}) \ge 0$

These inequalities form what is known as Mohr's circle

mohr's circle



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material behavior

topics

- Hooke's Law
- Physical meaning of elastic constants
- Thermal expansion

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hooke's law

• Can be written in terms of strain

$$\epsilon_{ij} = \frac{1+\nu}{E}\sigma_{ij} - \frac{\nu}{E}\sigma_{kk}\delta_{ij}$$

Or stress

$$\sigma_{ij} = \lambda \epsilon_{kk} \delta_{ij} + 2\mu \epsilon_{ij}$$

physical meaning

- Young's modulus
- Poisson's ratio
- Shear modulus
- Bulk modulus

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thermoelasticity

• Separate strain into mechanical and thermal components

$$\epsilon_{ij} = \epsilon^{M}_{ij} + \epsilon^{T}_{ij}$$

• For isotropic materials:

$$\epsilon_{ij} = \alpha (T - T_0) \delta_{ij}$$

• We can combine this with Hooke's Law to find

$$\epsilon_{ij} = \frac{1+\nu}{F}\sigma_{ij} - \frac{\nu}{F}\sigma_{kk}\delta_{ij} + \alpha(T-T_0)\delta_{ij}$$

• Or formulated in terms of stress (and Lamé constants)

$$\sigma_{ij} = \lambda \epsilon_{kk} \delta_{ij} + 2\mu \epsilon_{ij} - (3\lambda + 2\mu)\alpha (T - T_0)\delta_{ij}$$

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problem formulation

topics

- Boundary conditions
- Compatibility
- Beltrami-Michell
- Navier's Equations
- Superposition

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boundary conditions

- Traction
- Displacement
- Mixed

compatibility

- If continuous, single-valued displacements are specified, differentiation will result in well-behaved strain field
- The inverse relationship, integration of a strain field to find displacement, may not always be true
- There are cases where we can integrate a strain field to find a set of discontinuous displacements

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compatibility

- The compatibility equations enforce continuity of displacements to prevent this from occurring
- To enforce this condition we consider the strain-displacement relations:

$$\epsilon_{ij} = \frac{1}{2}(u_{i,j} + u_{j,i})$$

• and differentiate with respect to x_k and x_l

$$\epsilon_{ij,kl} = \frac{1}{2}(u_{i,jkl} + u_{j,ikl})$$

Or

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compatibility

 We can eliminate the displacement terms from the equation by interchanging the indexes to generate new equations

$$2\epsilon_{ik,jl} = u_{i,jkl} + u_{k,ijl}$$
$$2\epsilon_{il,ik} = u_{i,ikl} + u_{l,iik}$$

• Solving for $u_{i,ikl}$ and $u_{i,ikl}$

$$u_{i,jkl} = 2\epsilon_{ik,jl} - u_{k,ijl}$$

$$u_{j,ikl} = 2\epsilon_{jl,ik} - u_{l,ijk}$$

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compatibility

Substituting these values into the equations gives

$$2\epsilon_{ij,kl} = 2\epsilon_{ik,jl} - u_{k,ijl} + 2\epsilon_{il,ik} - u_{l,ijk}$$

• We now consider one more change of index equation

$$2\epsilon_{kl,ij} = u_{k,ijl} + u_{l,ijk}$$

and substituting this result gives

$$2\epsilon_{ij,kl} = 2\epsilon_{ik,jl} + 2\epsilon_{jl,ik} - 2\epsilon_{kl,ij}$$

Or, simplified

beltrami-michell

- When working with stress functions, it is convenient to check compatibility of the stress function directly
- Using Hooke's Law, we can formulate compatibility in terms of stress
- These are known as the Beltrami-Michell equations

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navier's equations

- Similarly, we can write the equilibrium equations in terms of displacement
- This is convenient when dealing with displacement boundary conditions
- Known as Navier's equations