

Optimisation et quantification d'incertitudes en dynamique non-linéaire des structures

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Collaborators

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Main outlines

- Personal background
- Inria presentation
- Scientific seminar
 - Surrogate modelling for uncertainty quantification – Application to rotordynamics
 - Topology optimisation based on EGO algorithm for nonlinear resonances

Personal background



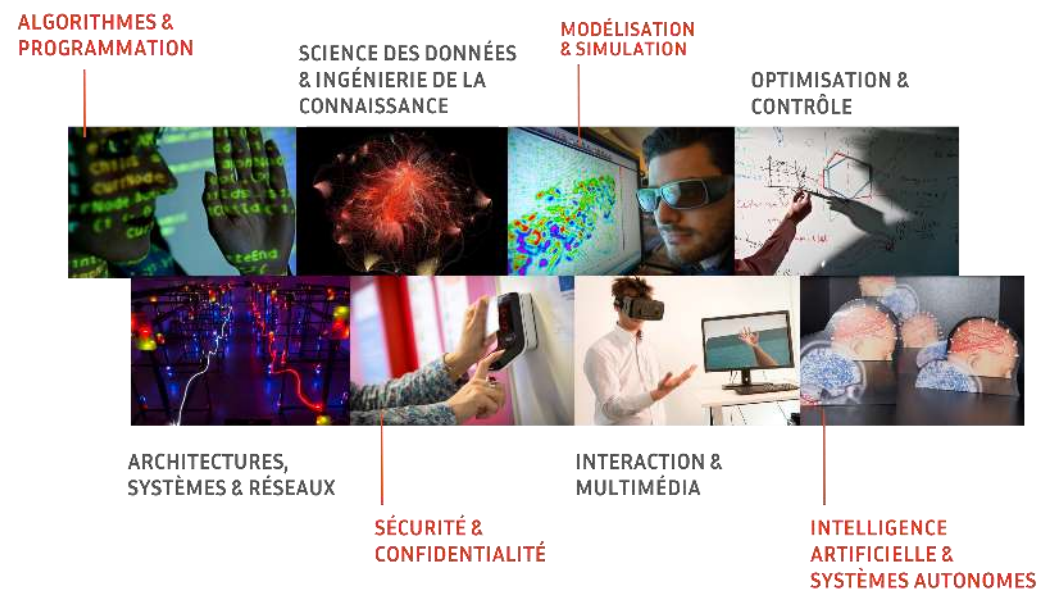
Curriculum

- Master degree in mechanical engineering and structural dynamics
- PhD in structural dynamics for squeal noise
- Postdoc in structural dynamics for nonlinear resonance mitigation
- Researcher I4S team, Inria, Rennes
- Researcher PLATON team, Inria, Saclay

Research activity and topics

- Nonlinear structural dynamics (contact, friction, crack)
- Structural optimisation (topology, shape, parametric, model updating)
- Uncertainty propagation for large nonlinear dynamic systems
- Structural Health Monitoring

Inria - Institut National de la recherche en sciences du numérique



De nombreuses thématiques de recherche...

... pour de nombreux secteurs d'applications



SANTÉ



ÉNERGIE



SÉCURITÉ
& RÉSILIENCE



ENVIRONNEMENT



CLIMAT



TRANSPORT



CULTURE &
DIVERTISSEMENT



ÉCONOMIE

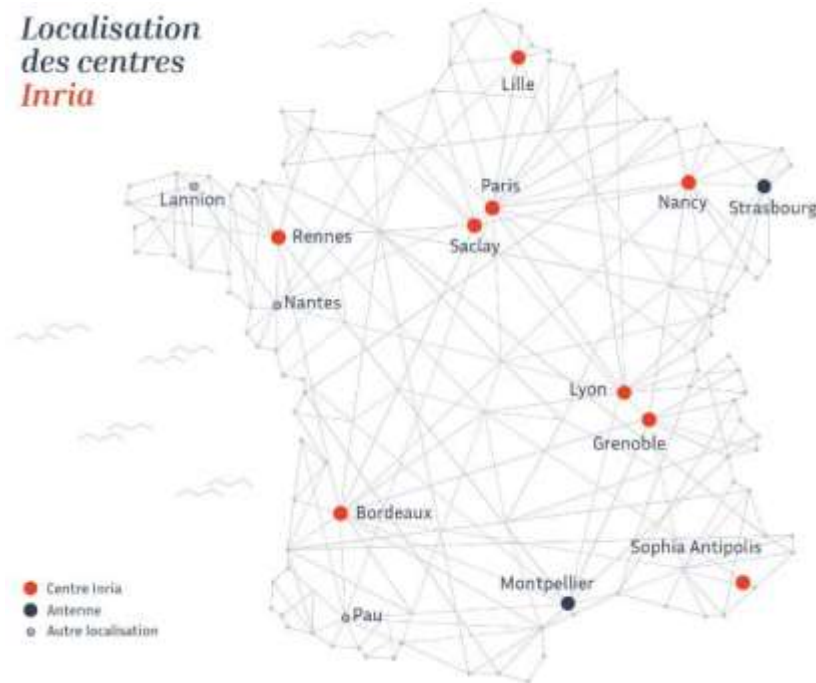


FINANCE



ALIMENTATION &
AGRICULTURE

Répartition en 9 centre de recherche sur le territoire



Structuration de l'institut

- 3900 scientifiques – 665 supports
 - Inria: 2700 personnes
 - Partenaires: 1700
- Structuration sous forme d'Equipe-Projet
 - 10 à 30 personnes
 - Feuille de route scientifique précise
 - Autonomie scientifique et financière
 - Évaluation internationale tous les 4 ans
 - Durée de vie maximale de 12 ans
 - Des équipes en coopération industrielle et internationale
- En 2020
 - 200 Equipes Projets
 - 80% en collaborations

- Partenariats industriels



- Laboratoires communs avec des PME



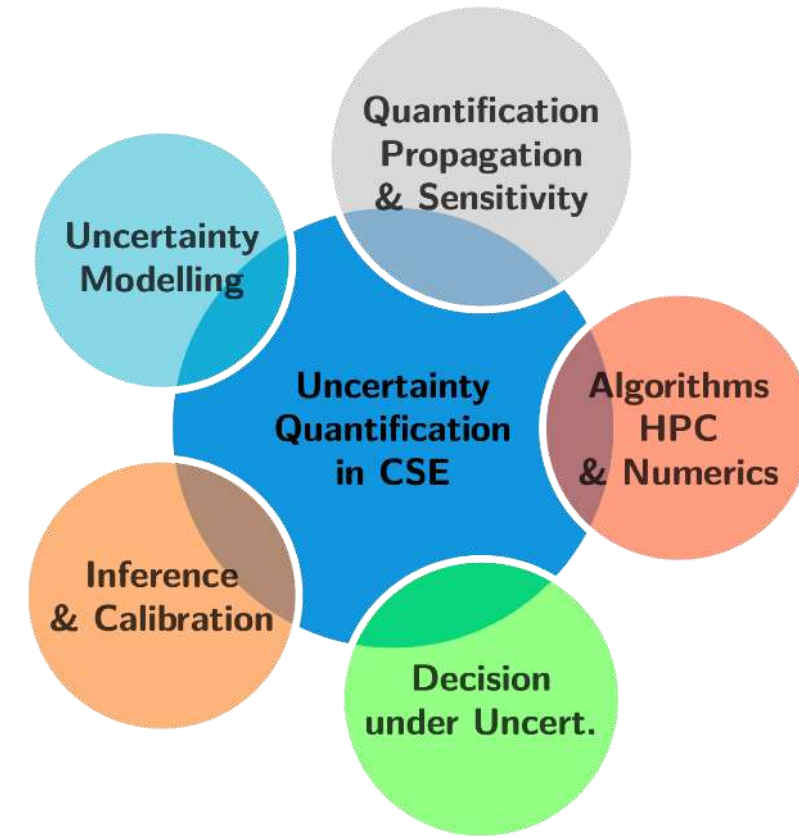
- Des logiciels



- Uncertainty Quantification in computation Science and Engineering
- What is PLATON ?
 - Join reserach group: Inria, CNRS and Ecole Polytechnique
 - Located in Saclay
- PLATON's collaborations
 - Industries: ArianeGroup, Dassault systems, Framatome etc
 - EPICS: CEA, Onera, IFPEN etc
 - Academics: Stanford, Polytecnico di Milano, VKI, U-Grenoble, ICL etc

Key features of **PLATON**:

- Created in 2020
- Core activities on UQ
- Exhaustive in scope
- From methodology to implementation
- Applications with Partners
- Factorize developments
- Cross-fertilization between application domains



General context



Large industrial structures

- Civil engineering
- Energy production (rotor, wind turbines etc)
- Transport (aircraft, train, car etc)
- Spatial

Complex vibration behaviour

- Nonlinear vibrations
- Large and complex models
- Expensive solver

Current challenges

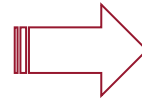
- Uncertainties
- Optimisation
- Design
- Etc...

General context

Nonlinear vibrations

*joints, friction, contact,
large deformation, crack, etc...*

$$\mathbf{M}\ddot{\mathbf{x}} + \mathbf{C}\dot{\mathbf{x}} + \mathbf{K}\mathbf{x} + \boxed{\mathbf{F}_{nl}(\mathbf{x}, \dot{\mathbf{x}})} = \mathbf{F}_e$$



- Dedicated solvers: Harmonic Balance Method, Shooting method, etc
- Methods developed for linear structural not adapted anymore
 - Uncertainty Quantification
 - Optimisation
 - etc

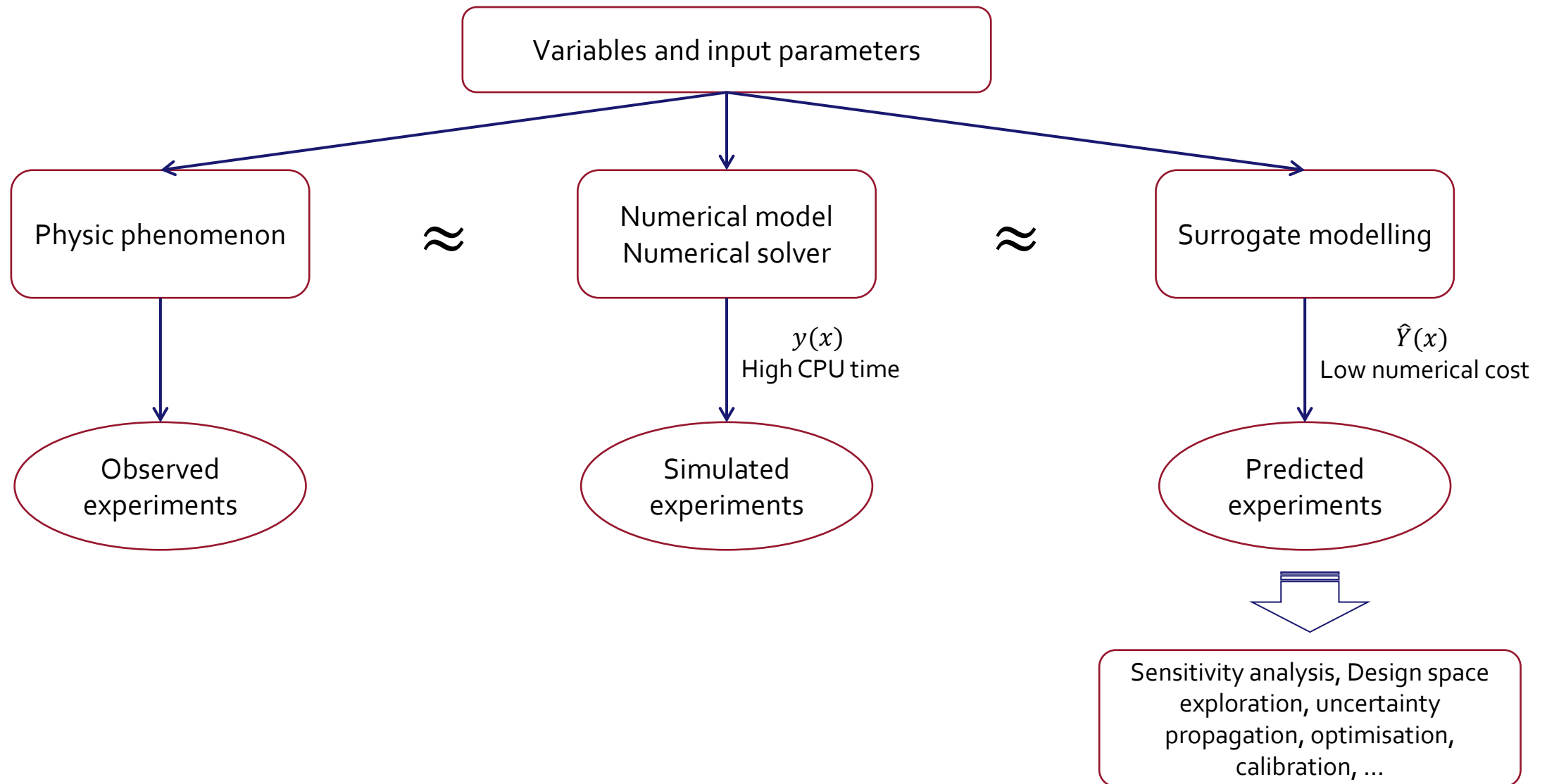
Limitations & current challenges

- Large computational time
 - Large models
 - Nonlinear behaviour
- Robust design of nonlinear structures
 - Uncertainties
 - Optimisation technics
 - Many parameters
 - New manufacturing technologies

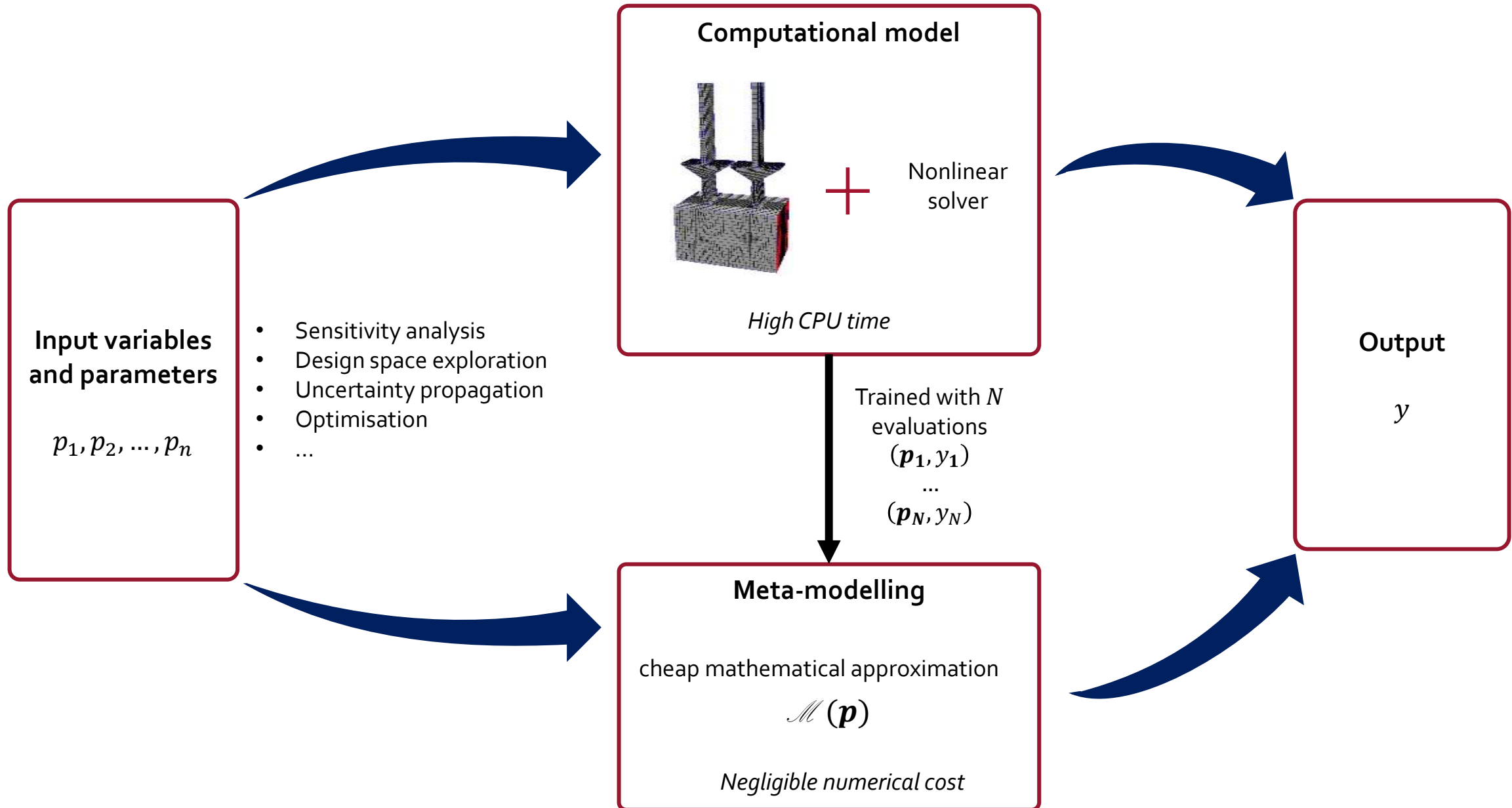
Contributions

- Exploitation of surrogate modelling technics
 - Uncertainty quantification
 - Structural optimization
 - Robust Structural Health Monitoring

Surrogate modelling: why?



Surrogate modelling – general presentation



Outlines

- Surrogate modelling for uncertainty quantification – Application to rotordynamics
 - Coupling PCE and kriging
 - Include physical knowledge in kriging
 - Prediction of dynamic features
 - Extension for rotor SHM in crack detection and localisation
- Topology optimisation based on EGO algorithm for nonlinear resonances
 - MMC framework
 - EGO algorithm
 - Application for friction ring dampers

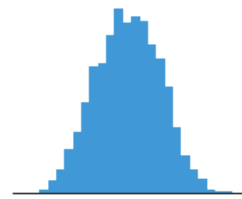
Rotor modelling

- **Vibrations**

- Critical speeds
- Amplitudes

- **Real conditions**

- Environmental variations
- Manufacturing tolerances
- Wear



→ Numerous sources of uncertainties!

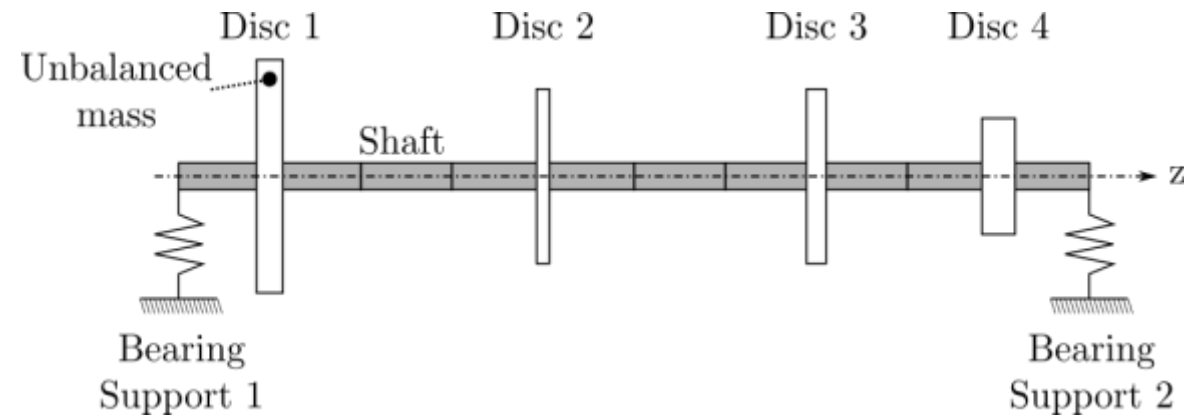
- **Design**

- Few parameters
- Robust design

Parametric studies

$$p \in [p_{min}, p_{max}]$$

Crucial to design robust rotors by taking into consideration uncertainties



Academic rotor under study

Shaft

- 10 Euler-beam finite elements
- mass, stiffness and gyroscopic matrices
- $\mathbf{C} = \alpha \mathbf{M}_S + \beta \mathbf{K}_S$ damping matrix

Bearings

- One on each side
- Stiffness in horizontal and vertical direction

Discs

- 4 discs at different locations
- Mass and gyroscopic matrices

Unbalanced mass on Disc 1

Rotor modelling

Equation of motion

$$\mathbf{M}\ddot{\mathbf{x}} + (\mathbf{C} + \omega\mathbf{G})\dot{\mathbf{x}} + \mathbf{K}\mathbf{x} = \mathbf{f}_1$$

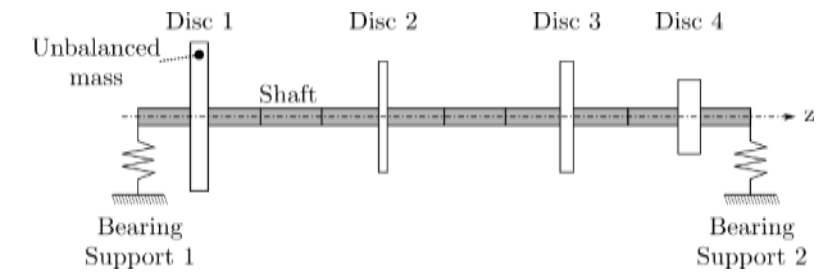
With $\mathbf{M} = \mathbf{M}_S + \mathbf{M}_{d1} + \mathbf{M}_{d2} + \mathbf{M}_{d3} + \mathbf{M}_{d4}$

$\mathbf{G} = \mathbf{G}_S + \mathbf{G}_{d1} + \mathbf{G}_{d2} + \mathbf{G}_{d3} + \mathbf{G}_{d4}$

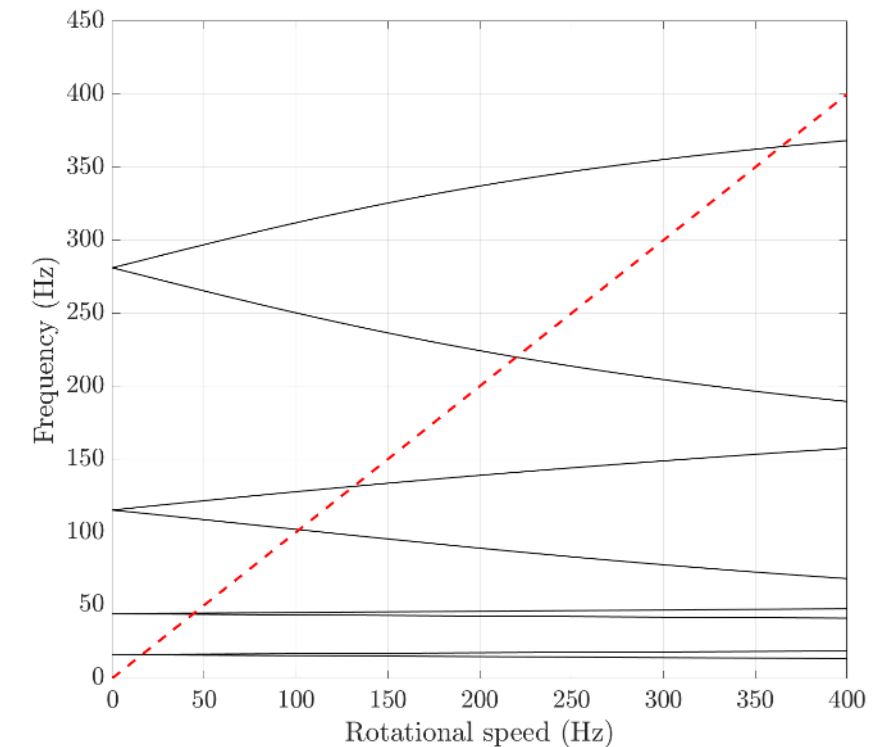
$\mathbf{K} = \mathbf{K}_S + \mathbf{K}_{b1} + \mathbf{K}_{b2}$

\mathbf{x} dof displacements

- Eigenvalues and modeshapes depend on the rotational speed $\omega \rightarrow$ Campbell diagram
- Critical speeds:** crossing between engine order (red dotted line) and natural frequency $\rightarrow f^{(i)}$ *Parameters to predict*
- Vibration amplitude** at critical speed $i \rightarrow a^{(i)}$



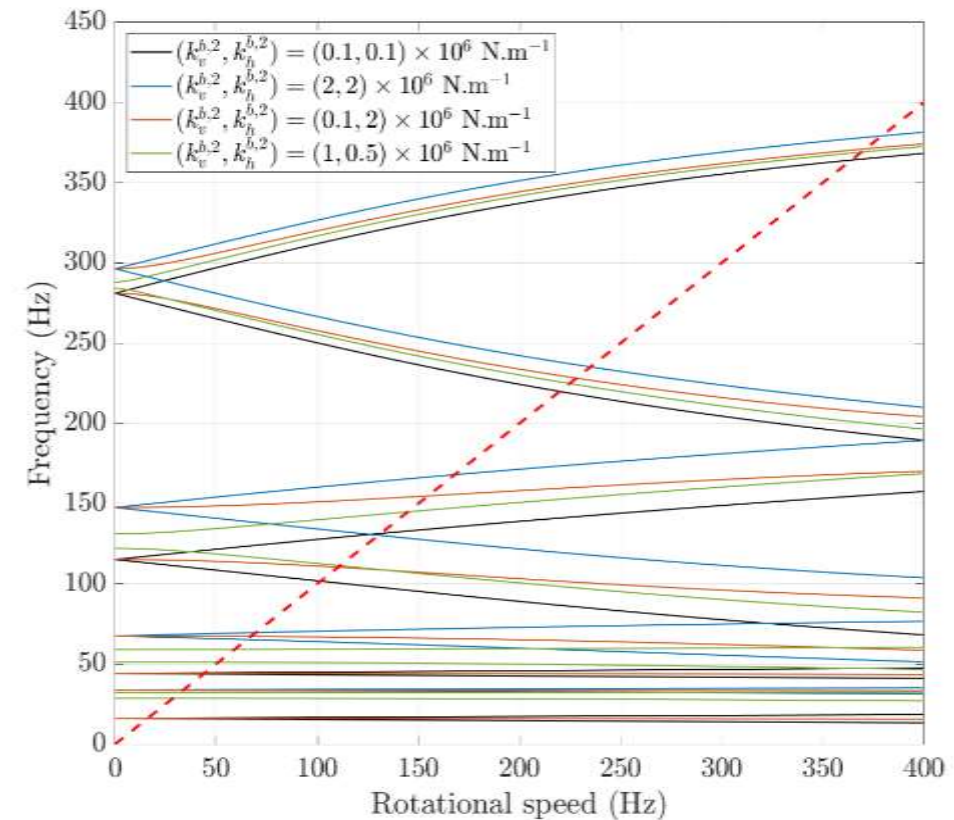
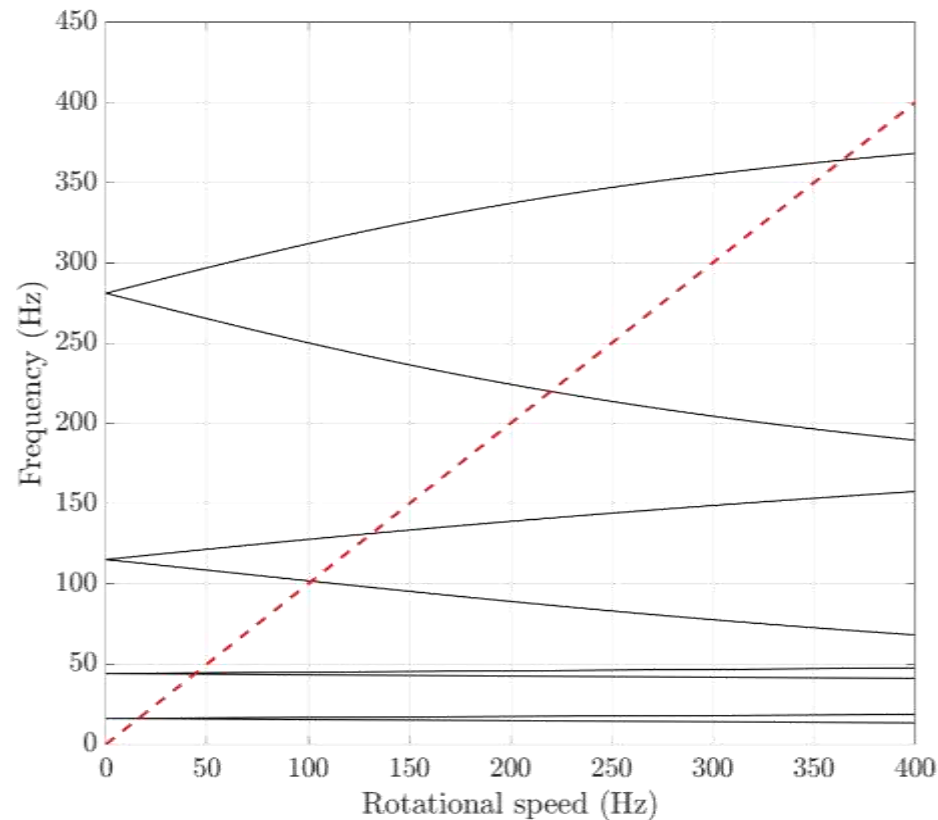
Academic rotor under study



Campbell diagram of the rotor

Rotor modelling

- **Critical speeds:** crossing between engine order (red dotted line) and natural frequency $\rightarrow f^{(i)}$ *Parameters to predict*
- **Vibration amplitude** at critical speed $i \rightarrow a^{(i)}$

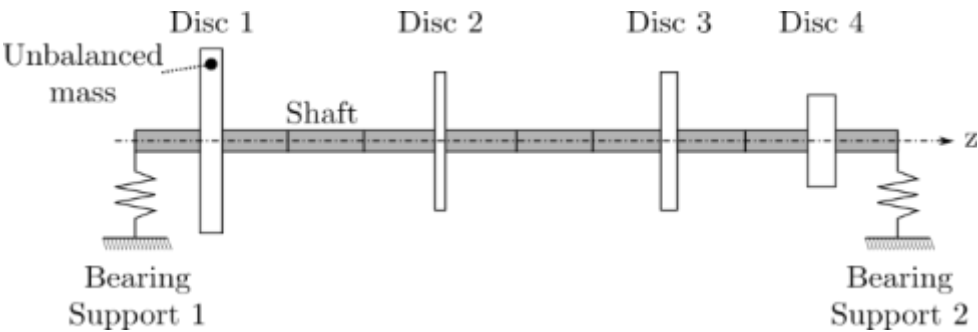


Campbell diagram of the rotor – impact of bearing stiffness

Uncertain parameters in the model

Uncertain parameters

- Manufacturing, environment, wear etc
- Modelled as random (**PDF**)
- Rotor response = stochastic
- Indicators = stochastic
- 7 uncertain parameters with different contributions



Academic rotor under study

Design parameters

- 2 design parameters
- $k_{x,2}$ and $k_{y,2} \in [0.1; 2]10^6 N/m$.
- Scanning

Problem

Scanning + Monte Carlo = too expensive !

Hybrid formulation

- PCE for stochastic
- Kriging for scanning

Parameter	Notation	% variation	Law	Contribution
Young modulus – shaft	E	$\pm 5\%$	Uniform	\mathbf{K}_s
Thickness - disc 1	t_1	$\pm 10\%$	Uniform	$\mathbf{G}_{d1}, \mathbf{M}_{d1}$
Thickness - disc 2	t_2	$\pm 10\%$	Uniform	$\mathbf{G}_{d2}, \mathbf{M}_{d2}$
Thickness – disc 3	t_3	$\pm 10\%$	Uniform	$\mathbf{G}_{d3}, \mathbf{M}_{d3}$
Thickness – disc 4	t_4	$\pm 10\%$	Uniform	$\mathbf{G}_{d4}, \mathbf{M}_{d4}$
Stiffness – right support – vert.	$k_{x,1}$	$\pm 10\%$	Uniform	\mathbf{K}_{b2}
Stiffness – right support – hor.	$k_{y,1}$	$\pm 10\%$	Uniform	\mathbf{K}_{b2}

Uncertain parameters

Polynomial Chaos – Mathematical background

Approximation:

$$\lambda(\xi) \approx \sum_{i=0}^m a_i \Psi_i(\xi)$$

With

ξ Random parameter

$f(\xi)$ Probability density function

$(\Psi_i)_{i \in 1, \dots, m}$ Basis of orthogonal polynomials of degree i (m is the chaos order)

Inner Product

$$\langle \Psi_j, \Psi_k \rangle = \int_D \Psi_j(x) \Psi_k(x) f(x) dx$$

Given by the Askey scheme

$\hat{\mathbf{a}} = (a_i)_{i=0, \dots, m}$ Unknown deterministic coefficients

To be determined

Regression method:

- Determination of an experimental design of N samples

$$\Xi = \{\xi^{(1)}, \dots, \xi^{(N)}\}$$

- Computation of their evaluations

$$\mathbf{Y} = \{\lambda(\xi^{(1)}), \dots, \lambda(\xi^{(N)})\}$$

- Least squares minimization

$$\hat{\mathbf{a}} = (\mathbf{W}^T \mathbf{W})^{-1} \mathbf{W}^T \mathbf{Y}$$

$$\text{with } W_{ij} = \Psi_j(\xi^{(i)})$$

Post processing of the coefficients

- Coefficients $(a_i), i \in [0, m]$ directly related to the stochastic properties of λ

$$E[\lambda] = a_0$$

$$\sigma_\lambda^2 = \sum_{k=1}^m a_k^2 \|\Psi_k\|^2$$

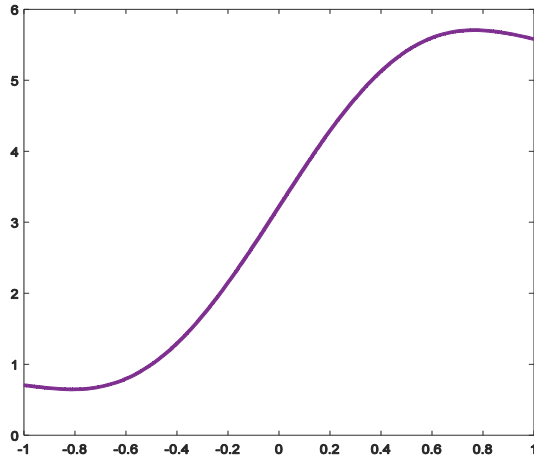
- Directly related to the Sobol indices

$$S_i = \frac{V_i}{\sigma_\lambda^2}$$

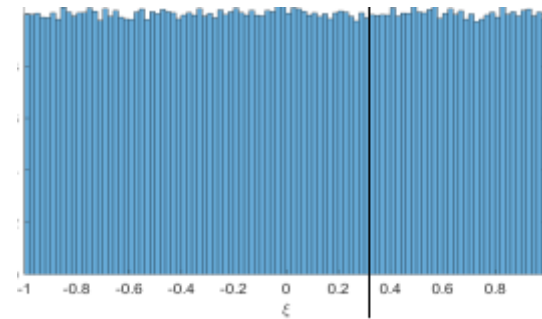
$$\text{With } V_i = \sum_{j \in v_i} a_j^2 \|\Psi_k\|^2$$

Polynomial Chaos - Example

Function f



ξ a uniform random parameter



Experimental Design

7 samples

$$(x_1, \dots, x_7)$$

and their evaluations

$$\mathbf{Y} = (f(x_1), \dots, f(x_7))$$

Creation of the basis
Chaos of order 2

Legendre polynomial

$$\Psi_0(\xi) = 1$$

$$\Psi_1(\xi) = \xi$$

$$\Psi_2(\xi) = (3\xi^2 - 1) / 2$$

Computation of the
coordinates in the bases

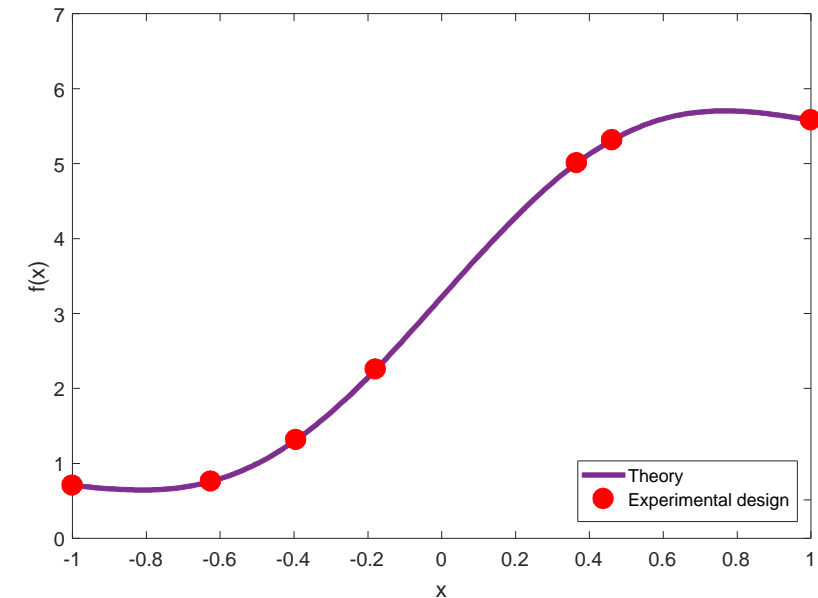
Computation of \mathbf{W}

$$\mathbf{W}_{ij} = \begin{pmatrix} \Psi_0(x_1) & \Psi_1(x_1) & \Psi_2(x_1) \\ \vdots & \vdots & \vdots \\ \Psi_0(x_7) & \Psi_1(x_7) & \Psi_2(x_7) \end{pmatrix}$$

Computation of \mathbf{a}

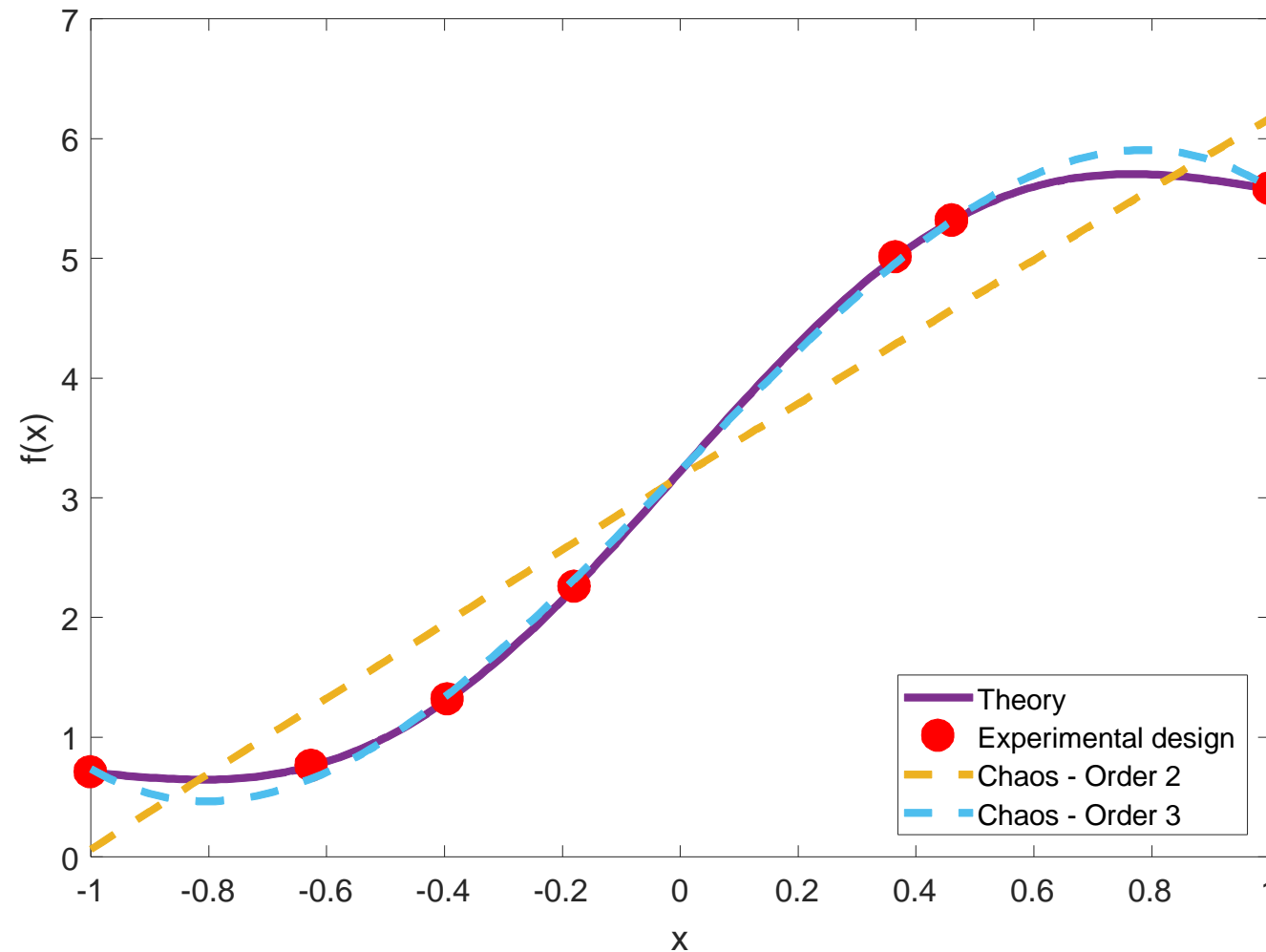
$$\hat{\mathbf{a}} = (\mathbf{W}^T \mathbf{W})^{-1} \mathbf{W}^T \mathbf{Y}$$

$$\hat{\mathbf{a}} = (a_0, a_1, a_2)$$



Polynomial Chaos - Example

Reconstruction of the function with polynomial chaos



Kriging – Mathematical background

Approximation:

$$\lambda(\mathbf{p}) = \sum_{i=1}^m \beta_i f_i(\mathbf{p}) + Z(\mathbf{p})$$

with

$\mathbf{p} \in \mathbf{R}^k$ parameters vector

N samples $\mathbf{S} = (\mathbf{s}^{(1)}, \dots, \mathbf{s}^{(N)})$

and their evaluations

$$\mathbf{Y}_S = (\lambda(\mathbf{s}^{(1)}), \dots, \lambda(\mathbf{s}^{(N)}))$$

$(f_i)_{i=1, \dots, m}$ Regression functions
 m polynomial basis

$(\beta_i)_{i=1, \dots, m}$ Regression coefficients
solution of a least square problem

Global behaviour

$Z(\cdot)$ Zero-mean Gaussian process

Covariance $E[Z(\mathbf{s}), Z(\mathbf{x})] = \sigma^2 \mathbf{R}(\boldsymbol{\theta}, \mathbf{s}, \mathbf{x})$

Process variance σ^2

Correlation function $\mathbf{R}(\boldsymbol{\theta}, \mathbf{s}, \mathbf{x}) \in [0, 1]$

Depends on the distance between \mathbf{s} and \mathbf{x}

Zero distance \rightarrow Correlation = 1

Infinite distance \rightarrow Correlation = 0

$\boldsymbol{\theta}$ characterizes the distance of influence

Solution of a maximum likelihood problem

Local behaviour

Characteristics to choose

- **Regression function**
 - Polynomials
- **Correlation function**
 - Exponential, Gaussian, linear, spherical, matern...
 - Depends of the regularity of the phenomenon
- **Experimental Design**
 - Input/output set
 - Learning points
 - Smallest as possible

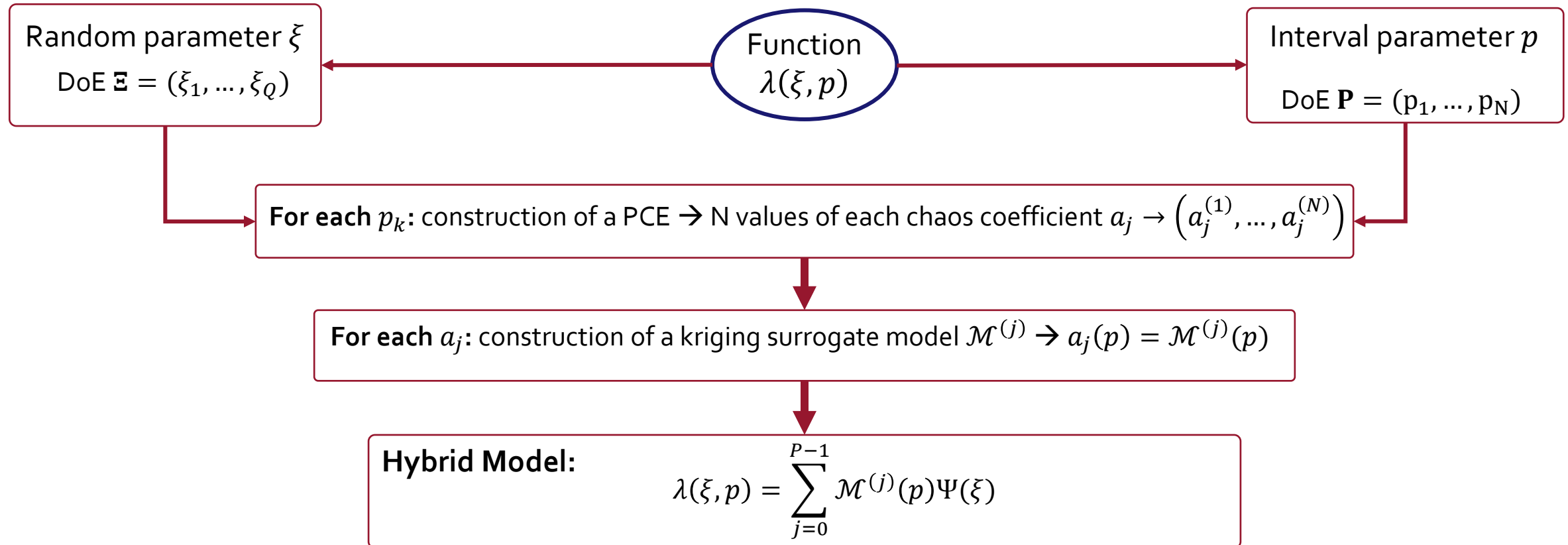
Hybrid formulation

Formulation

Main idea: express the PCE coefficients as a function of the design parameters

$$\lambda(\xi, p) = \sum_{k=0}^m a_k(p) \Psi(\xi) = \sum_{k=0}^m \mathcal{M}^{(k)}(p) \Psi(\xi)$$

Construction



Advanced DoE for the kriging

- **Problem knowledge:** symmetry w.r.t. $k_v^{(b,2)} = k_h^{(b,2)}$ (design variables)
- **3 strategies to integrate this knowledge**
 - *Classic strategy:* regression g , correlation \mathcal{R} , no sampling strategy
 - *Half-design space:* kriging construction only on $k_v^{(b,2)} \leq k_h^{(b,2)}$, reconstruction by symmetry
 - *Symmetric regression*
$$\mathbf{g}_s(x_1, x_2) = \begin{cases} \beta_0 + \beta_1 x_1 + \beta_2 x_2 + \beta_3 x_1 x_2 + \beta_4 x_1^2 + \beta_5 x_2^2 & \text{if } x_1 \leq x_2 \\ \mathbf{g}_s(x_2, x_1) & \text{otherwise} \end{cases}$$
- **Evaluation of the performances**
 - Different DoE size pour kriging: [20,40,60,80]
 - 4000 validation points

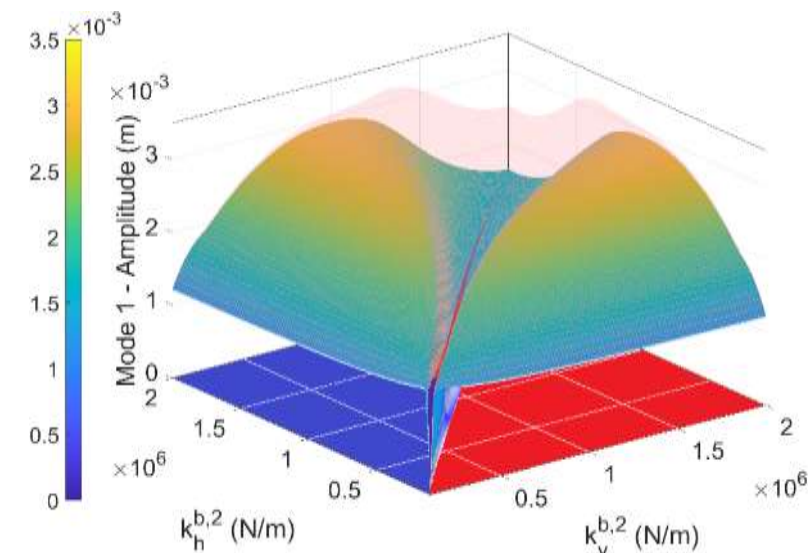


Fig.: Vib. amplitude at first critical speed versus the two design parameters – Average values (coloured surface) – Average \pm standard deviation (red and blue surfaces)

➡ Symmetry in the DoE

➡ Symmetry in the Kriging regression

DoE strategies comparison

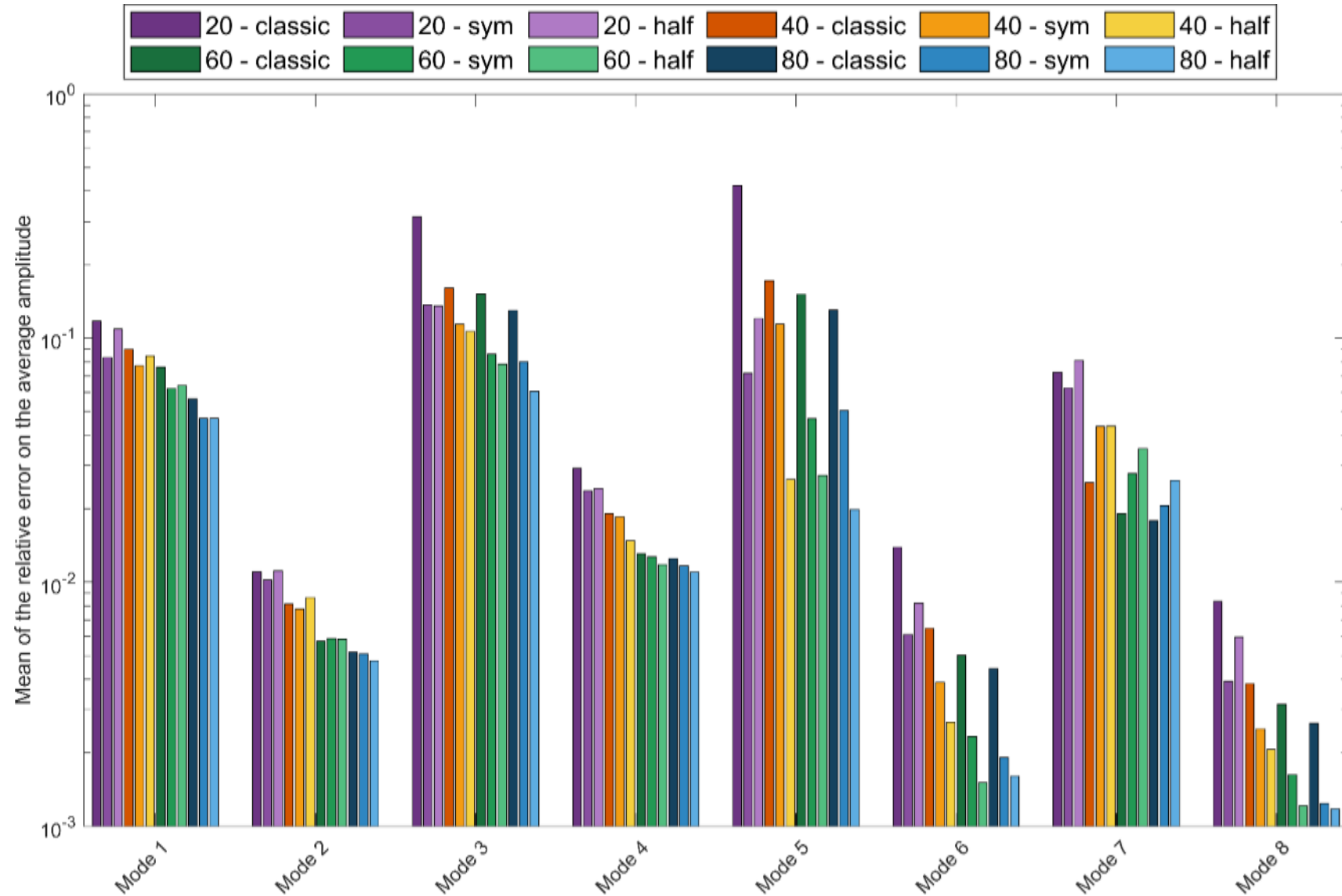


Fig.: Mean on the relative error on the average amplitude

- Error lower for forward modes (even number) than backward modes (odd number)
- When DoE size increases, error decreases
- Classic kriging: worst strategy
- Half: best strategy

Integrating the symmetry property in the DoE is the best strategy here

Evolution of the critical speeds and vibration amplitudes

MCS + Scanning	VS	Hybrid
10^7	VS	7500

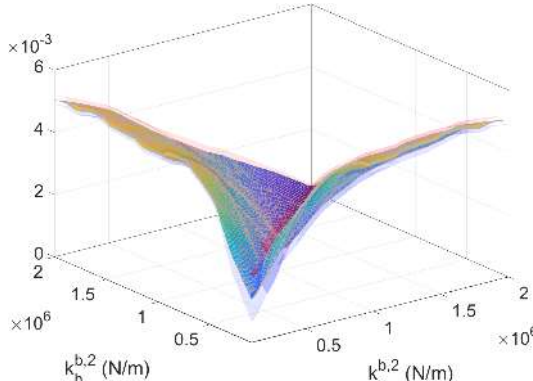
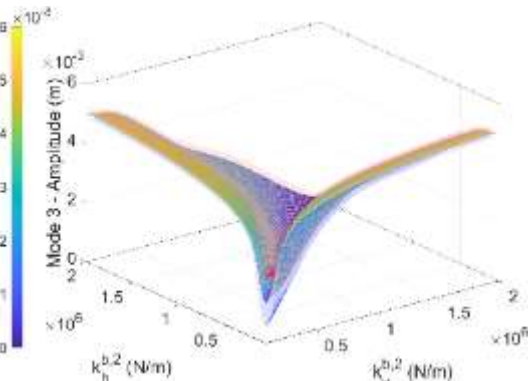
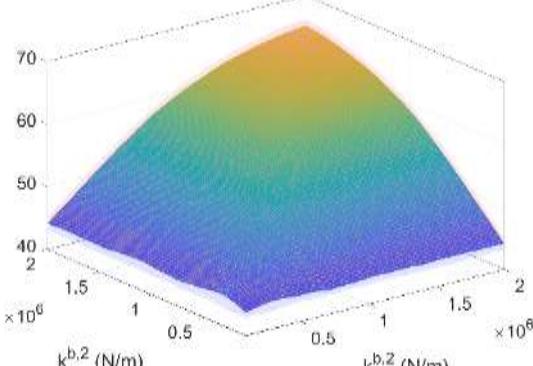
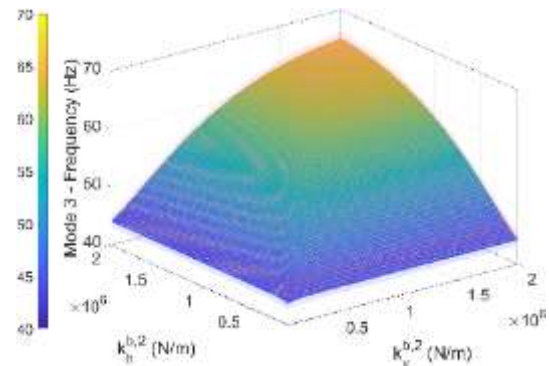
Freq. [Hz]

Ampl. [m]

Mode 3

Reference

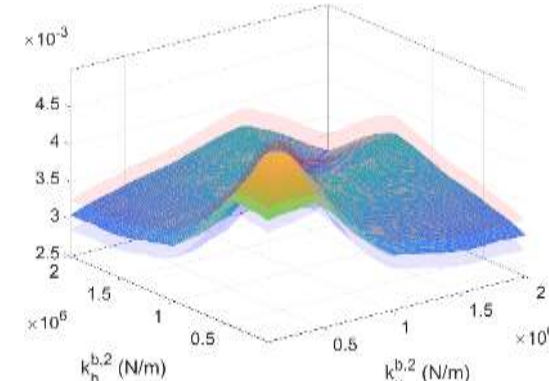
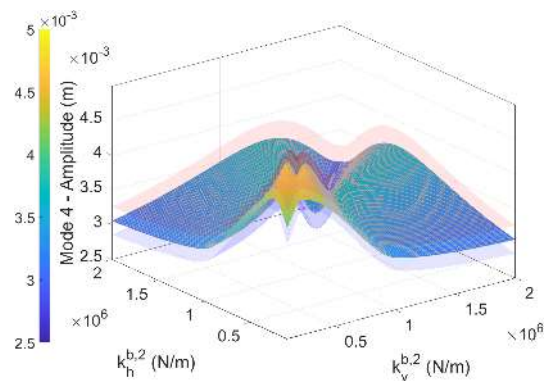
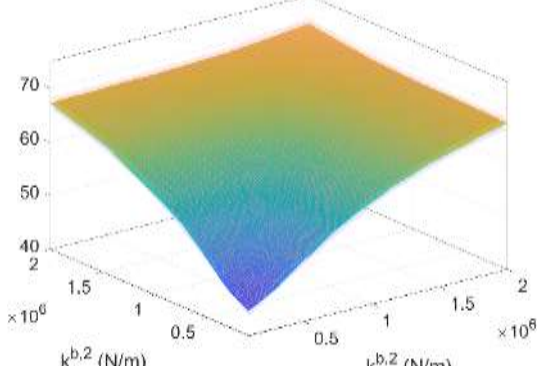
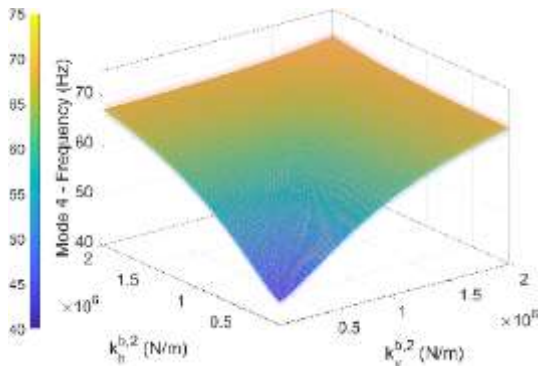
Prediction



Mode 4

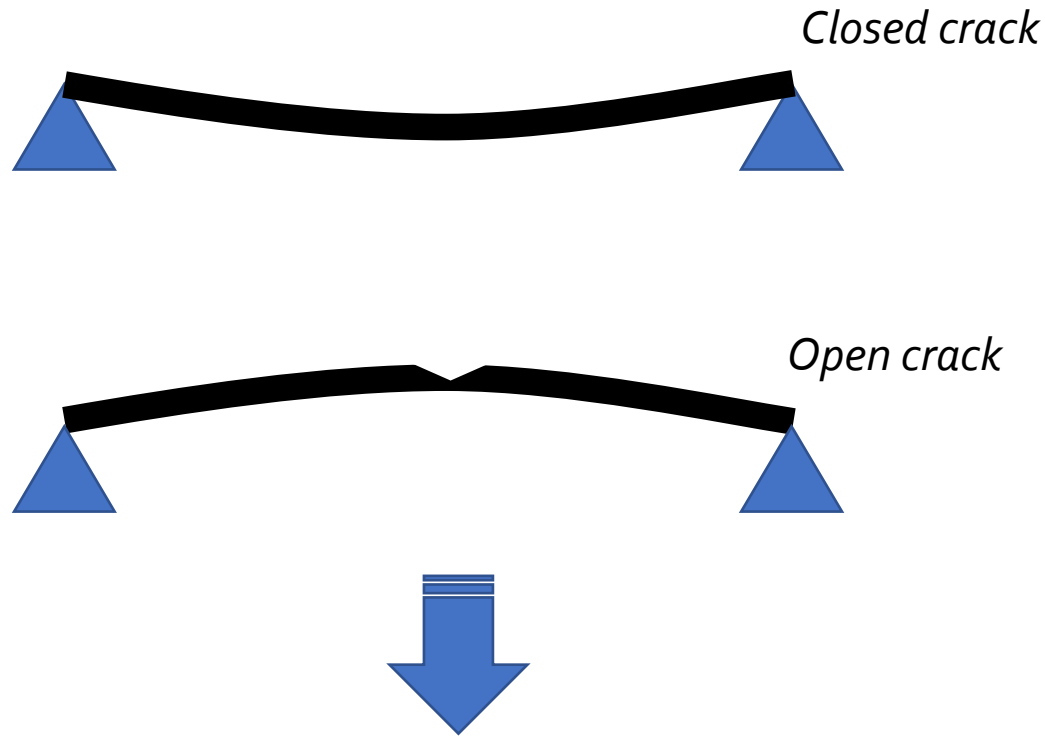
Reference

Prediction



Application for the SHM of cracked rotor

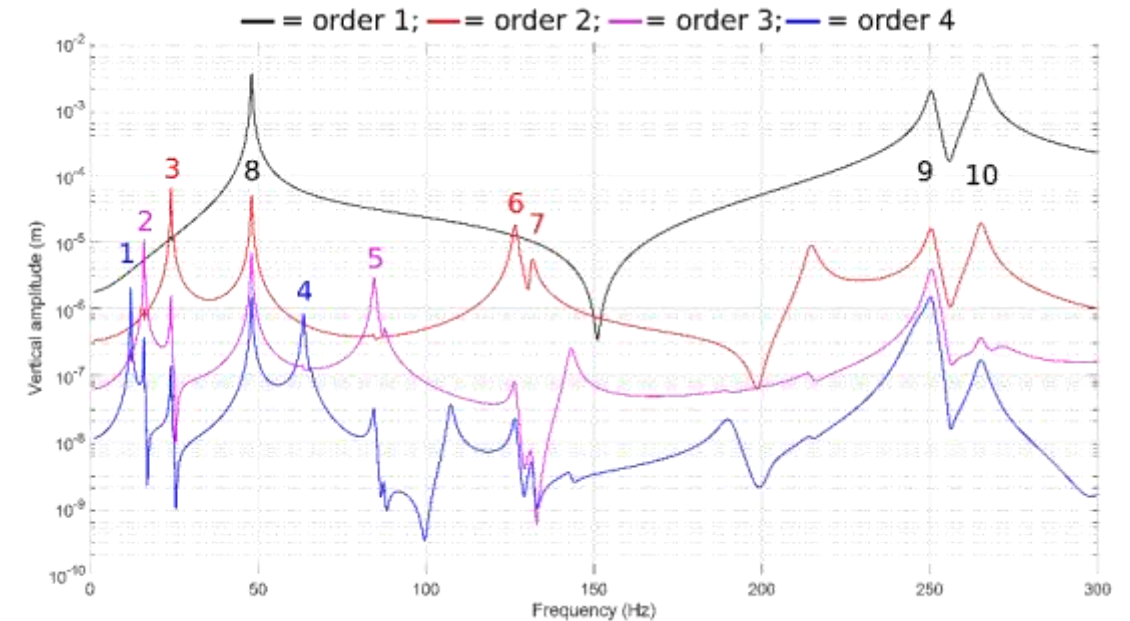
Deformation of the shaft during rotation



Breathing mechanism of the crack

Nonlinear dynamic response

Nonlinear FRF

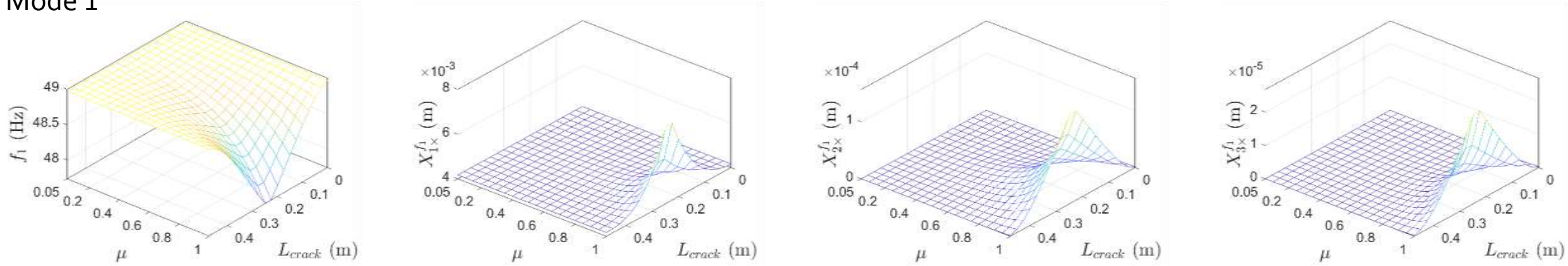


Objectif

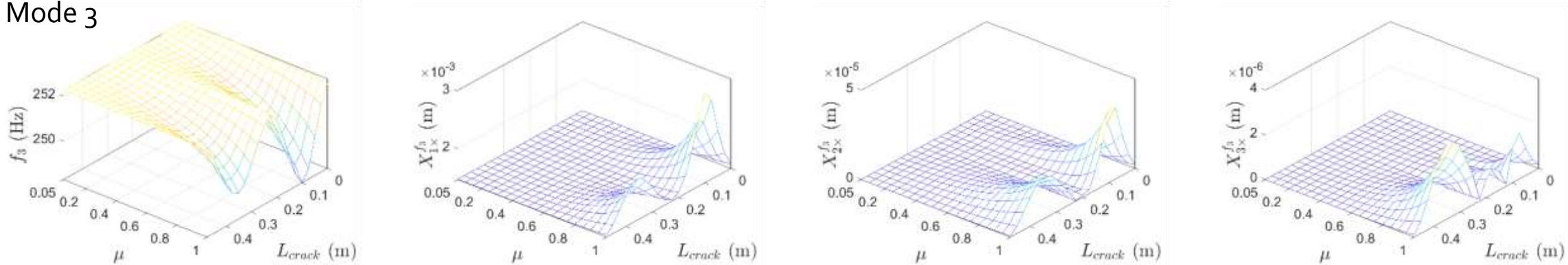
Prediction of sub-critical and critical speeds
and corresponding vibration amplitudes

Impact of the crack on the deterministic response

Mode 1

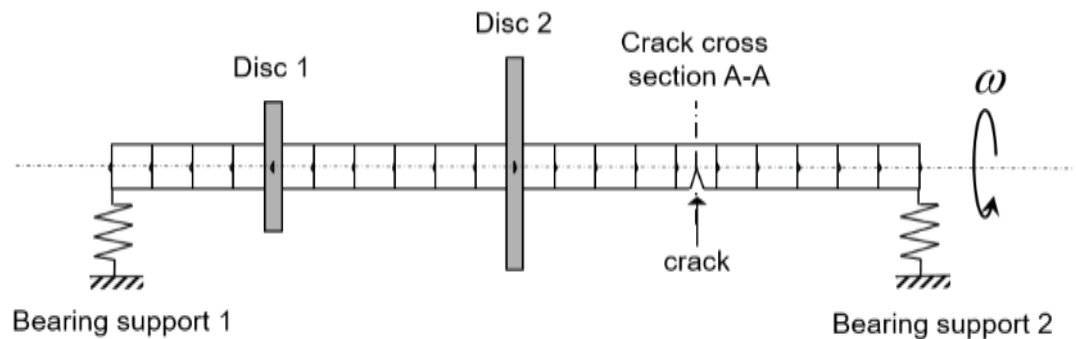


Mode 3



But what happens when the system is uncertain ?

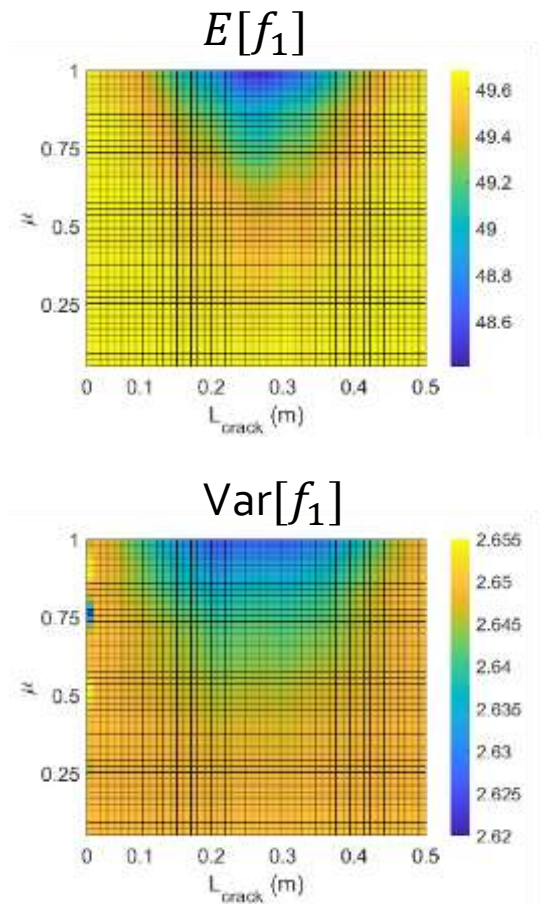
Uncertain parameters in the model



Parameter	Notation	% variation	Law	Contribution
Young modulus – shaft	E	$\pm 5\%$	Uniform	\mathbf{K}_s
Density – shaft	ρ	$\pm 5\%$	Uniform	\mathbf{M}_s and \mathbf{q}^f
Thickness - disc 1	t_1	$\pm 10\%$	Uniform	$\mathbf{G}_{d1}, \mathbf{M}_{d1}$ and \mathbf{q}
Thickness – disc 2	t_2	$\pm 10\%$	Uniform	$\mathbf{G}_{d2}, \mathbf{M}_{d2}$ and \mathbf{q}
Stiffness – right support – vert.	$k_{x,2}$	$\pm 10\%$	Uniform	\mathbf{K}_{b2}
Stiffness – right support – hor.	$k_{y,2}$	$\pm 10\%$	Uniform	\mathbf{K}_{b2}



Prediction of the resonance frequencies and amplitudes

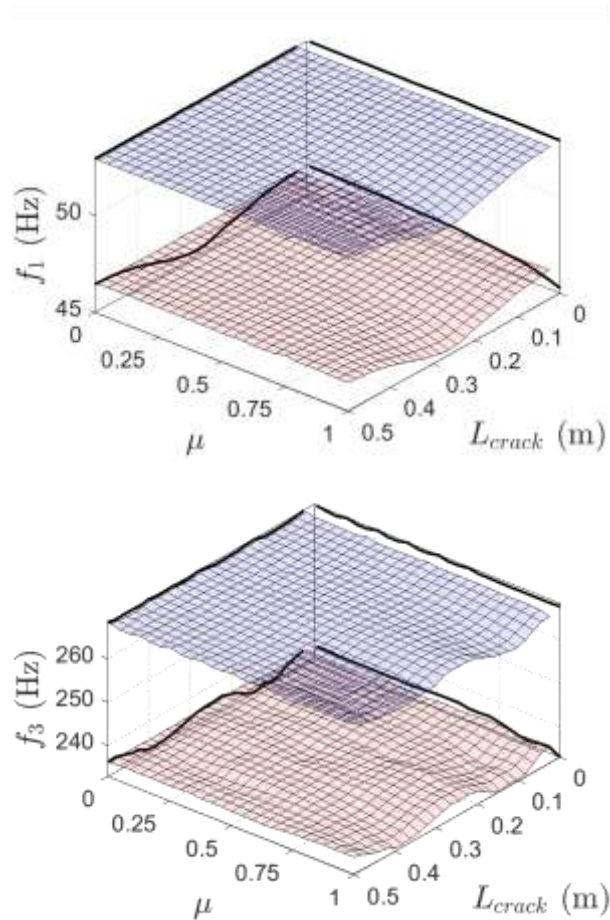


Analysis of the 95% confidence interval

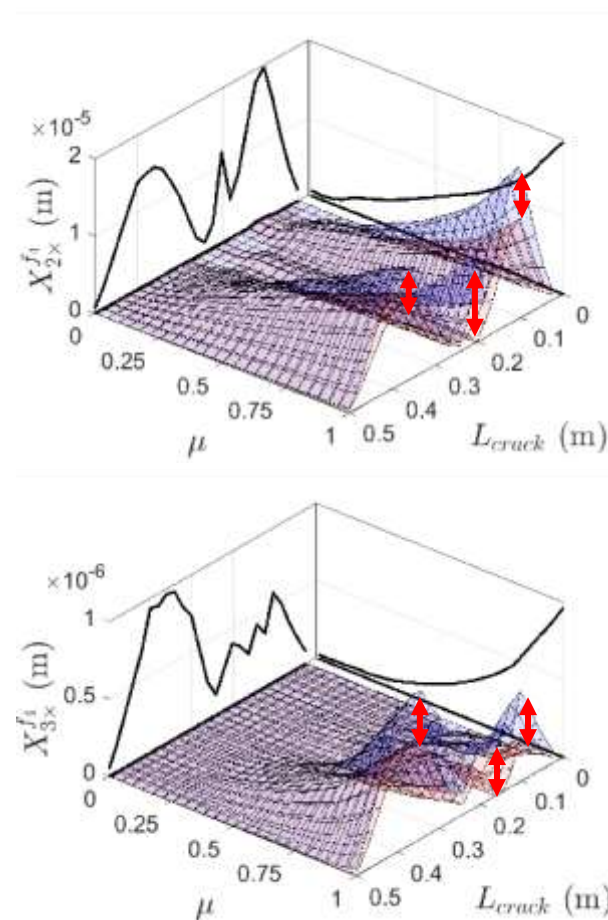
- Determination of the 95% confidence interval over the map:
 - $CI_a(\mu, L_c)$: Confidence interval of the parameter a .
- Three categories:
 - **Category 1**: the variation of the minimum and maximum bounds are too small compared to the width of the CI → not an indicator
 - **Category 2**: the variation of the minimum and the maximum bounds are predominant compared to the width of the CI, but for small variations of the crack parameters the bounds remain in the interval due to uncertainties → partial indicators
 - **Category 3**: impact of uncertainties is negligible → robust indicators

Illustration of the indicators

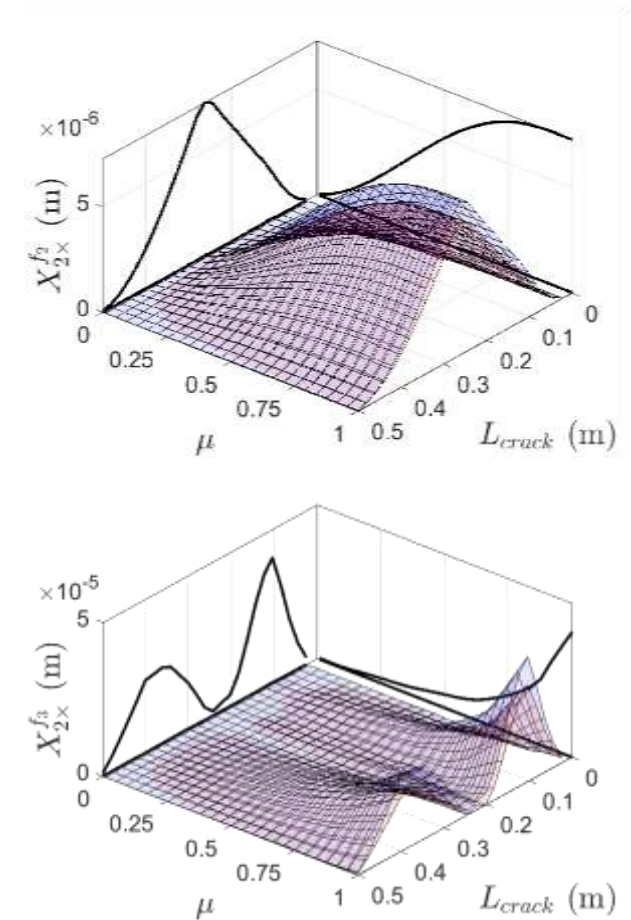
Caterogy 1



Caterogy 2



Caterogy 3

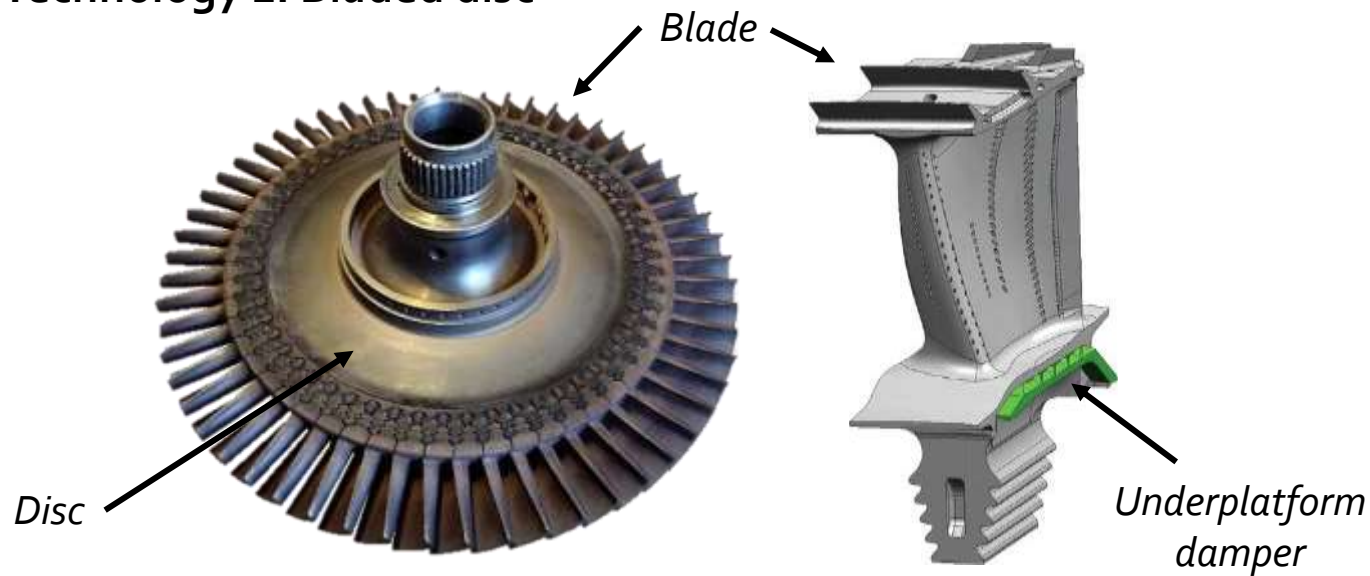


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 - Include physical knowledge in kriging
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 - Application for friction ring dampers

Bladed discs and friction dampers

Technology 1: Bladed disc



Technology 2: Integrally bladed disc (blisk)



Control of vibrations

- Introduction of damping
- Dampers are small metal pieces
- Use of friction to dissipate vibration energy

Dynamic equation of the system

$$\mathbf{M}\ddot{\mathbf{U}} + \mathbf{C}\dot{\mathbf{U}} + \mathbf{K}\mathbf{U} + \mathbf{F}_{nl}(\mathbf{U}) = \mathbf{F}(t)$$

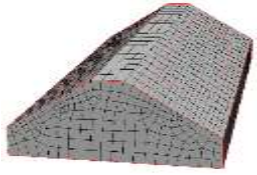
Non-linear equation solved with specific solvers

→ Numerically expensive !

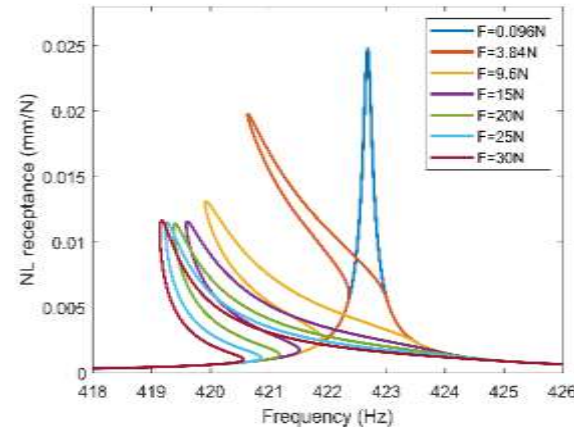
⇒ Full nonlinear dynamic response highly sensitive to the contact interface

Impact of damper geometry

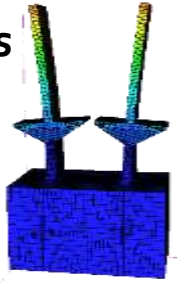
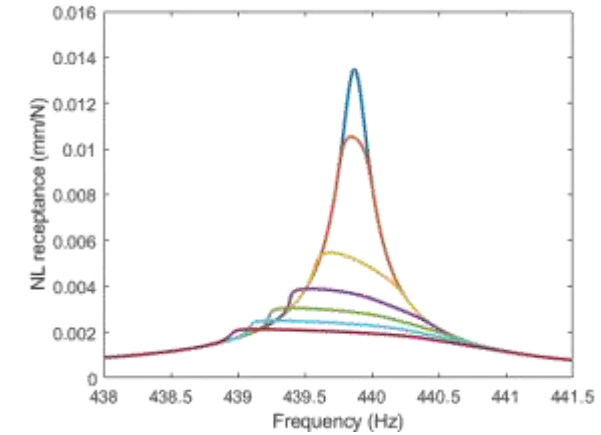
Wedge damper



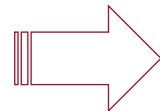
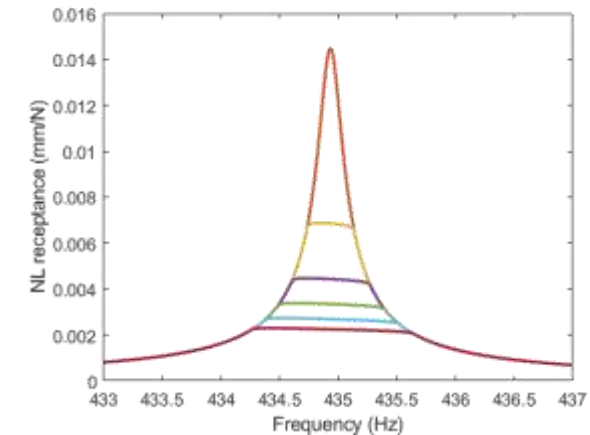
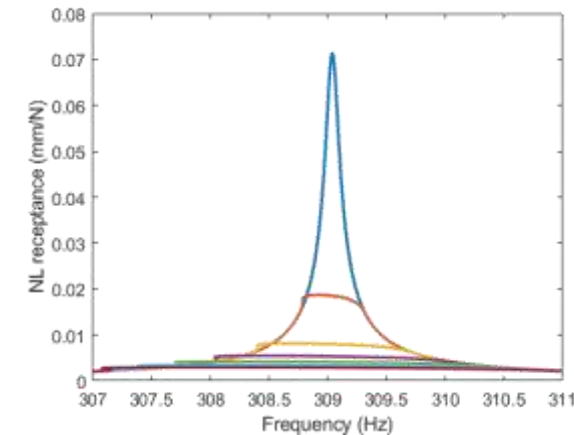
In-phase mode of the blades



Out-of-phase mode of the blades



Conical damper



Damper shape highly influent: optimisation of the topology

Topology optimisation – general presentation

- Objective: over a design domain Ω , find the best material domain Ω_m to minimise the objective function $f_{obj}(\Omega_m)$ under the constraints $g(\Omega_m) \leq 0$

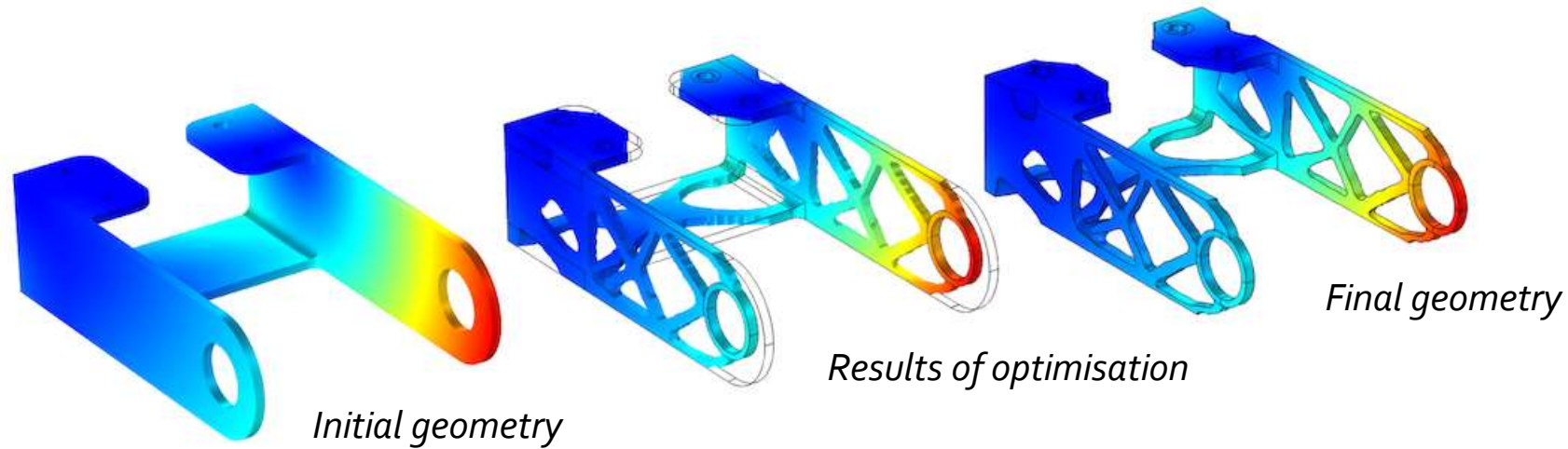


Fig.: Illustration of the topological optimisation process (Comsol website)

- Interests
 - Lighter and more efficient structures
 - Identify new geometries
 - 3D printing: open the possibilities

Topological optimisation – general presentation

- Objective: over a design domain Ω , find the best material domain Ω_m to minimise the objective function $f_{obj}(\Omega_m)$ under the constraints $g(\Omega_m) \leq 0$
- Classical approaches: density-based and Level-Set Method
 - Compliance, deflection, natural frequencies, stress, mass etc
 - But: require the sensitivities of the objective function w.r.t to the densities or shape function
→ Not adapted for nonlinear vibrations
- Global optimisation
 - No need for gradient
 - Well adapted to non-convex optimisation problems
 - But: require a parametrisation of the problem

Presentation of the mechanical system

- **2D-geometry**
 - Reduction computational time
 - Reduction of the number of optimization parameters
 - Problem can be approximated as 2D
- **Friction contact elements**
 - 3 status: stick, slip, separated
 - Uniform contact pressure

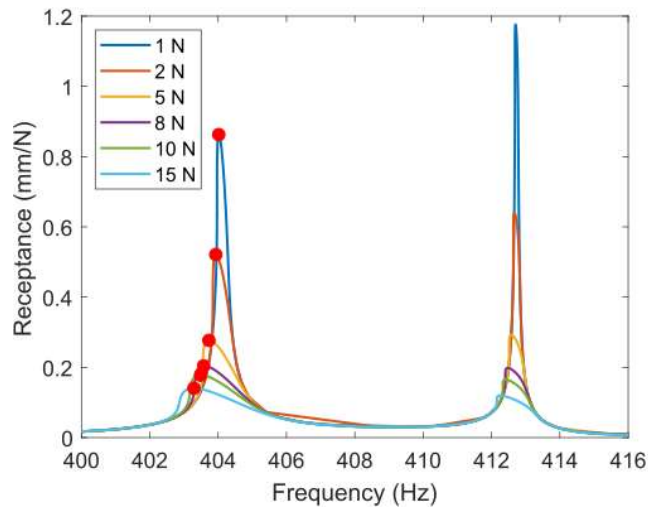


Fig.: Receptances FRF for different excitation amplitudes

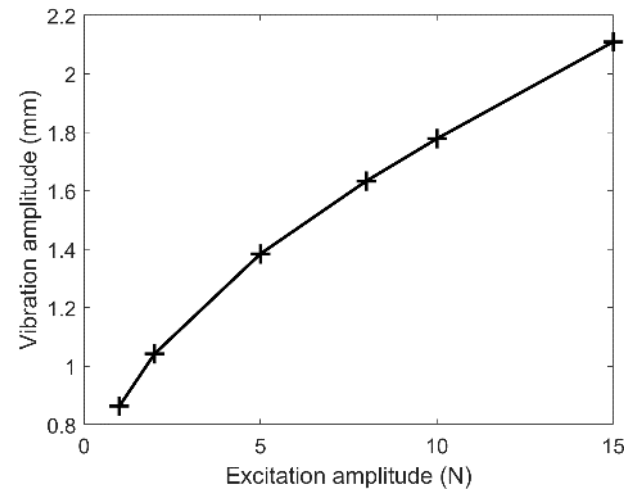


Fig.: Evolution of the amplitude at resonance for different excitation amplitudes

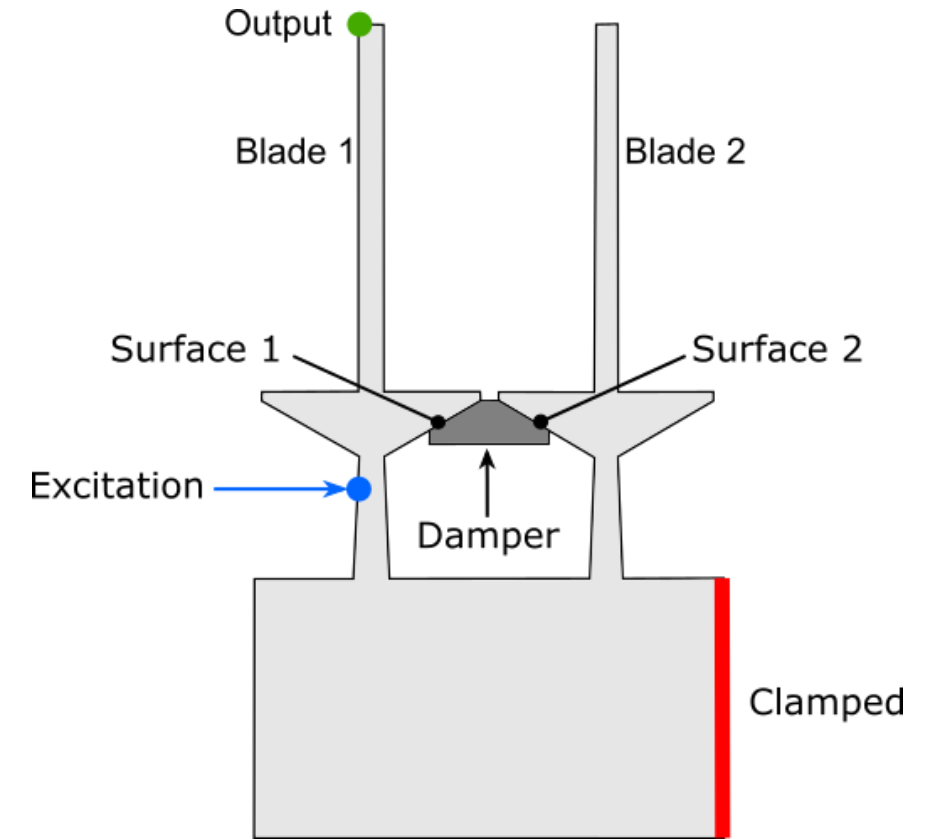
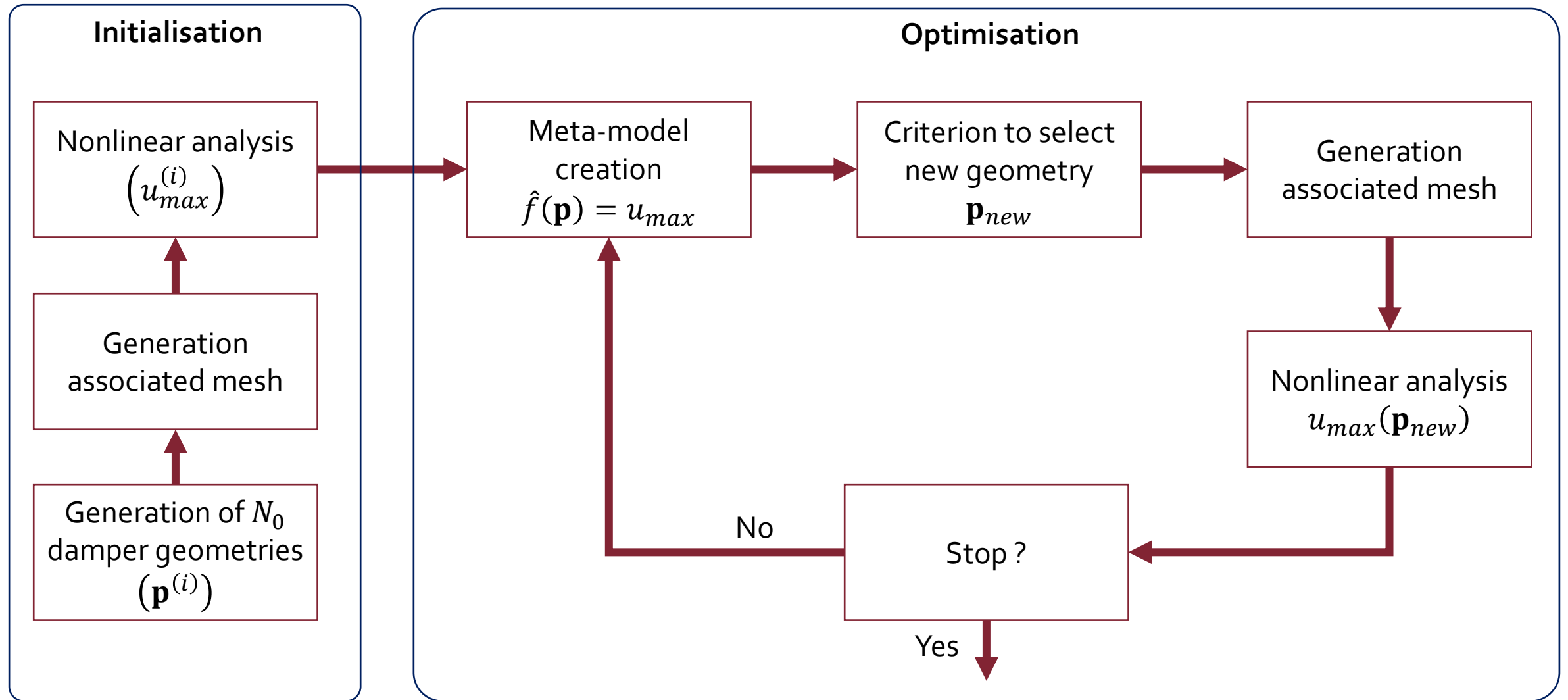
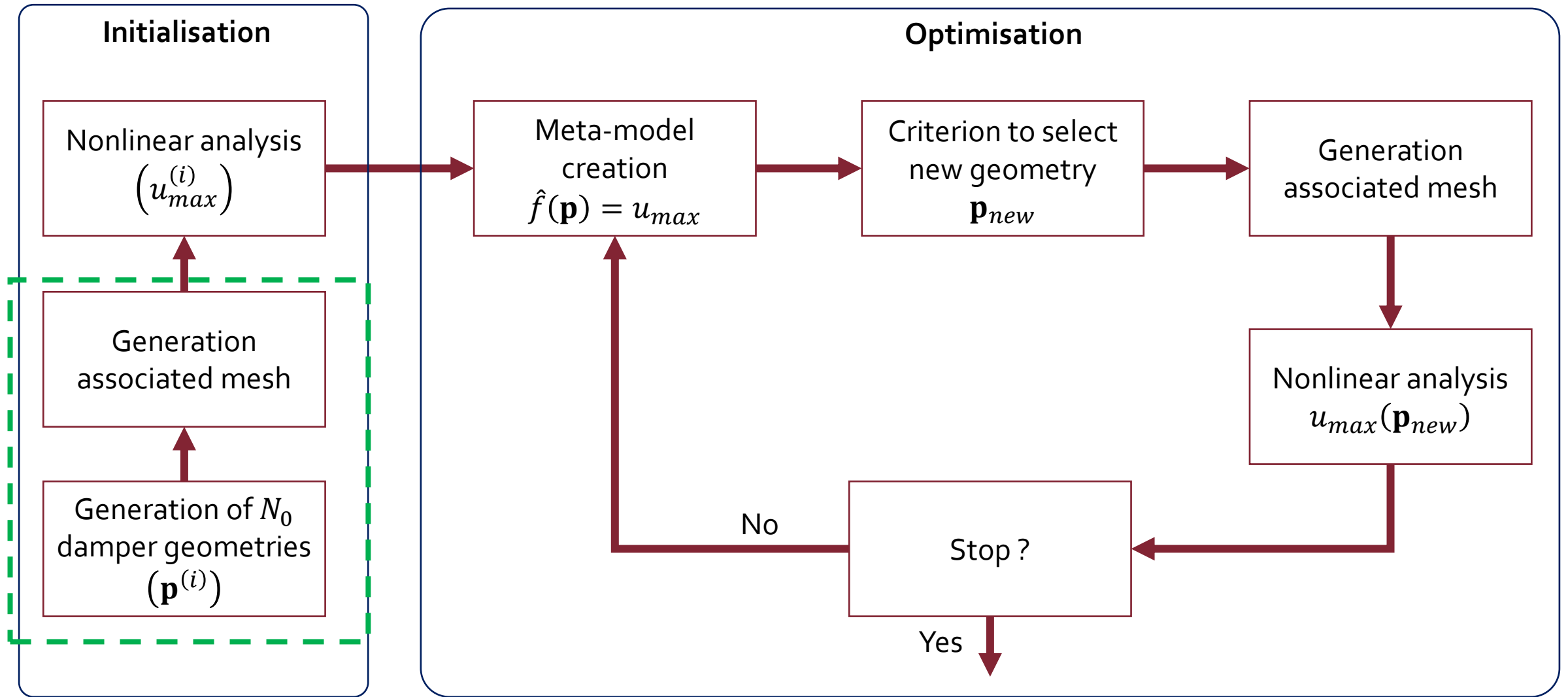


Fig.: Studied mechanical system

Surrogate modelling – optimisation process



Surrogate modelling – optimisation process



Damper parametrisation

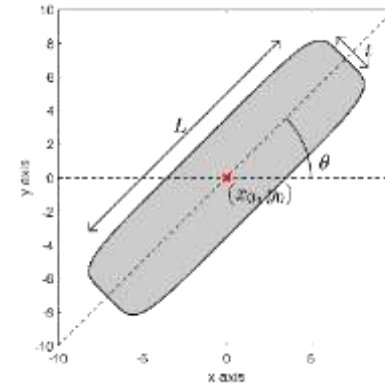
Moving Morphable Components (MMC) framework :

- Structure = assembly of elementary components
 - 1 component = 5 parameters
 - location (x_0, y_0) , inclination θ , length L and thickness t
- Moving, shrinking, expanding components => complex topologies
- Translated into a Level-Set Function (LSF) $\Psi(\mathbf{u})$:

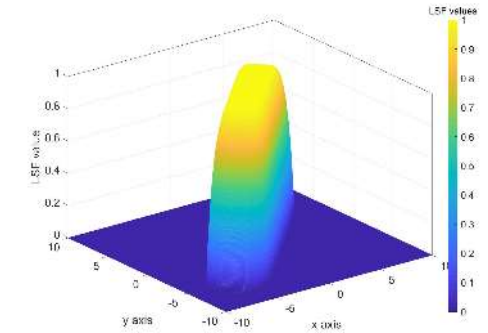
$$\begin{cases} \Psi(\mathbf{u}) > 0, \mathbf{u} \text{ is in the material domain} \\ \Psi(\mathbf{u}) < 0, \mathbf{u} \text{ is in the void domain} \\ \Psi(\mathbf{u}) = 0, \mathbf{u} \text{ is at the boundary} \end{cases}$$
- 1 component = 1 unitary LSF Ψ_i
- Combination of the elementary LSF => global LSF

$$\Psi(\mathbf{u}) = \max_i \Psi_i(\mathbf{u})$$

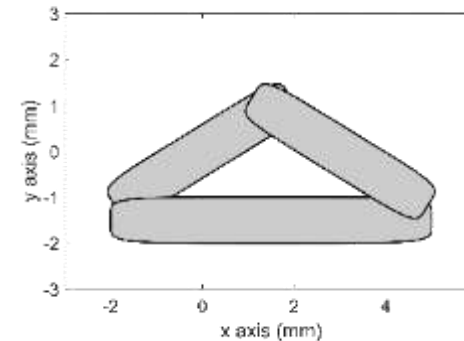
Illustration:



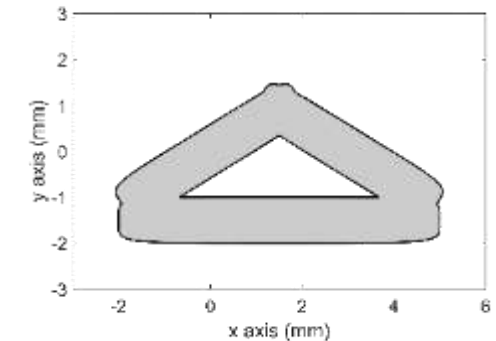
(a) Unitary component



(b) LSF of a unitary component



(c) Assembly of different components



(c) Final geometry

Fig.: Illustration of the MMC framework
Grey = material domain

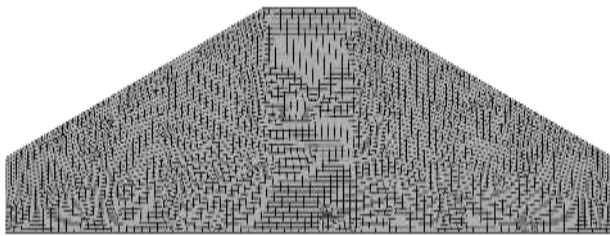
Damper parametrisation – mesh creation

Mesh mapping

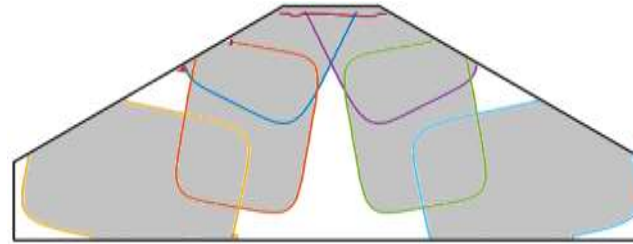
- Initial damper mesh
- Discretisation of the LSF in the center of each element
- Void elements removed

Geometrical constraints

- 1 component set to the top: seal the platforms
- Damper symmetric
- 1 center fixed to the contact line



Initial mesh



*LSF computation
(color line = component limits, grey = material domain)*



Selection of elements to keep

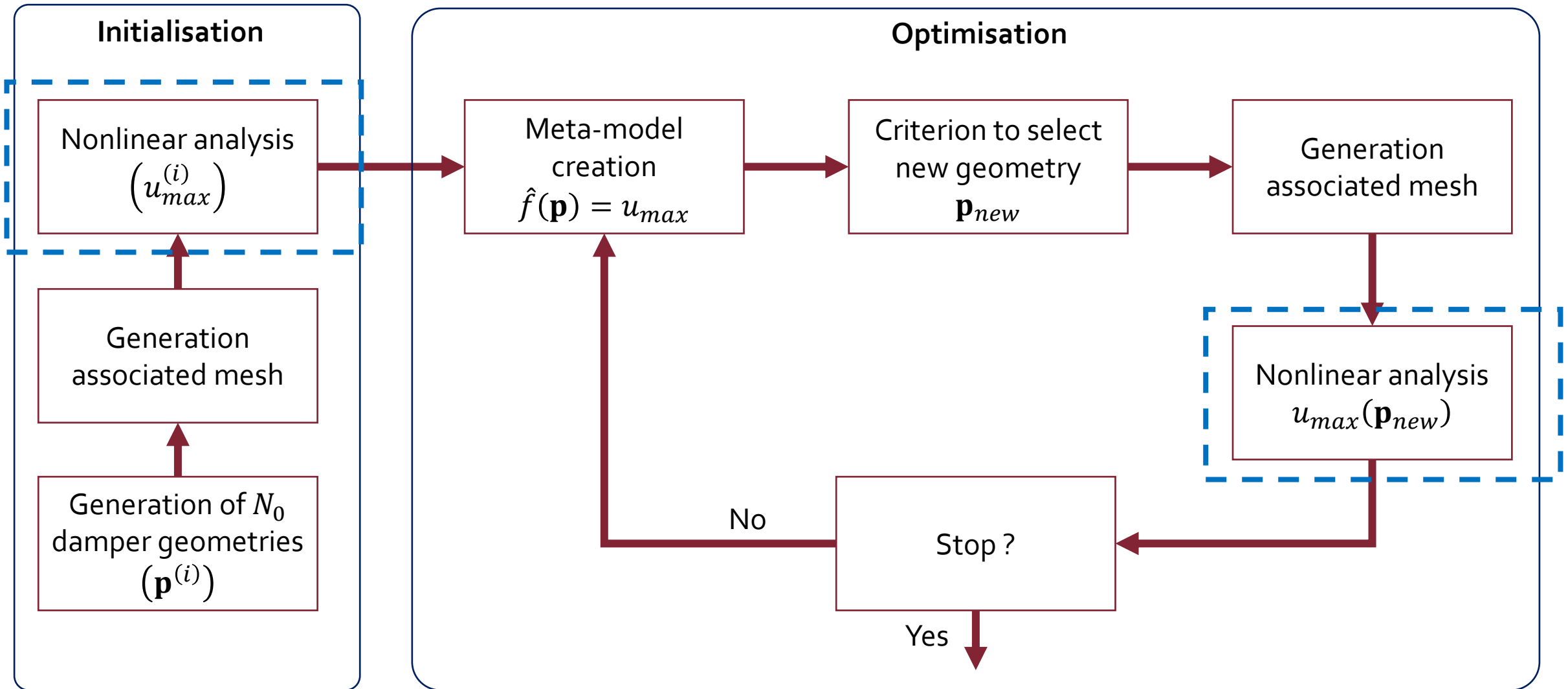
Fig.: General process for damper mesh creation

Computation of the new damper characteristics

- Structural matrices $\mathbf{M}_{\text{damper}}$, $\mathbf{C}_{\text{damper}}$, $\mathbf{K}_{\text{damper}}$ from Craig-Bampton reduction
- Contact parameters updating (normal pressure, contact stiffness)

Intermediary conclusion – Damper geometry parametrized: depends on a few parameters p

Surrogate modelling – optimisation process



Computation of the non-linear response

Equation of motion

$$\mathbf{M}\ddot{\mathbf{X}} + \mathbf{C}\dot{\mathbf{X}} + \mathbf{K}\mathbf{X} + \mathbf{F}_{nl}(\mathbf{X}, \dot{\mathbf{X}}) = \mathbf{F}_e$$

\mathbf{M} , \mathbf{C} and \mathbf{K} are the global mass, damping and stiffness matrices

Harmonic Balance Formulation

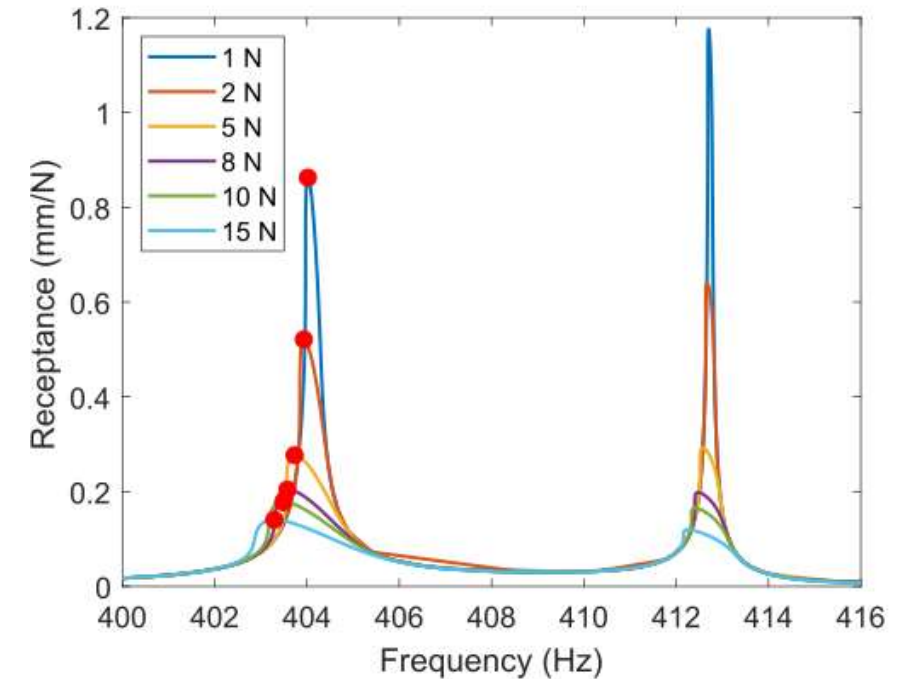
$$\mathbf{J}_1(\mathbf{Q}, \omega) = \mathbf{Z}(\omega)\mathbf{Q} + \tilde{\mathbf{F}}_{nl}(\mathbf{Q}) - \tilde{\mathbf{F}}_{ex} = \mathbf{0}$$

\mathbf{Q} vector of sine and cosine coefficients

Phase quadrature

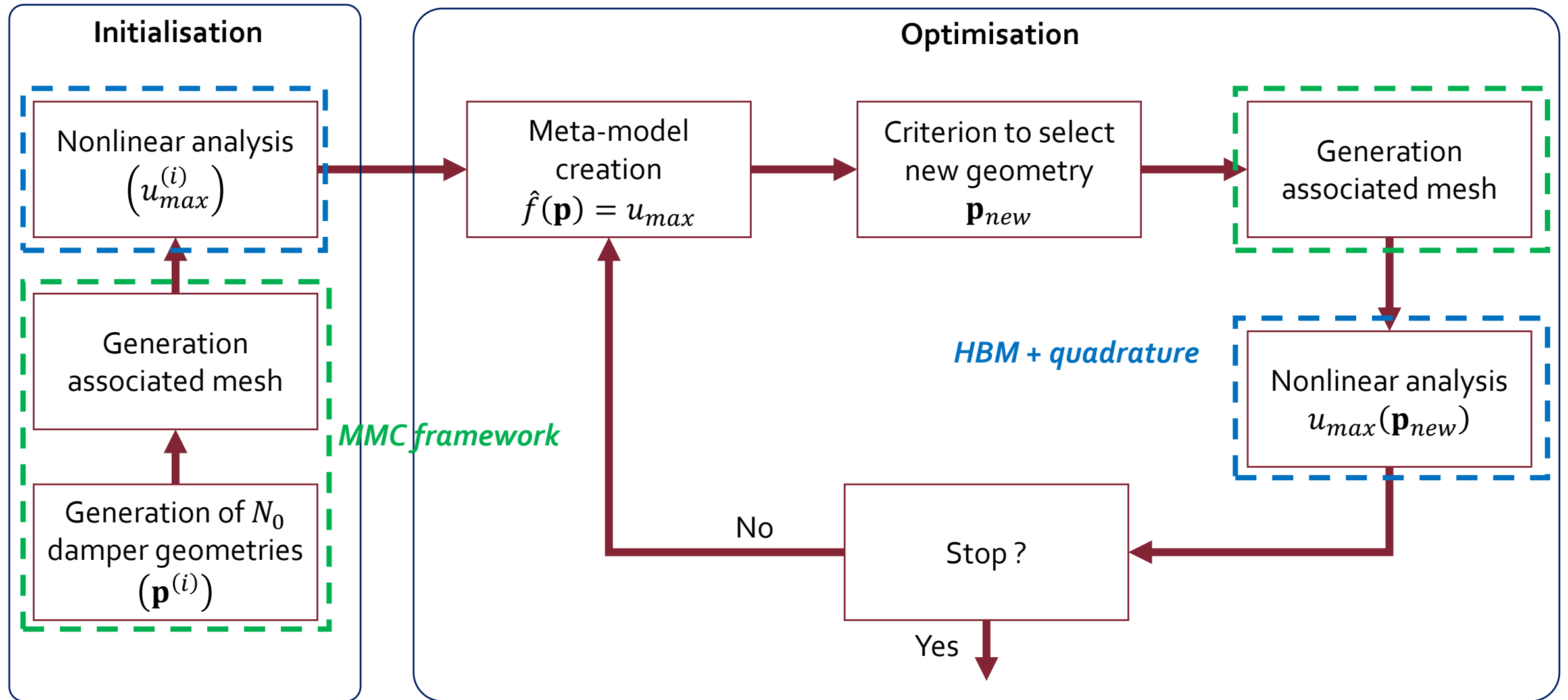
$$\mathbf{J}_2(\mathbf{Q}, \omega) = \phi - \frac{\pi}{2} = 0 \quad \begin{array}{l} \phi \text{ phase between excitation and response} \\ \rightarrow \text{directly on the resonance peak} \end{array}$$

Finding \mathbf{Q} and ω so that $[\mathbf{J}_1(\mathbf{Q}, \omega); \mathbf{J}_2(\mathbf{Q}, \omega)] = \mathbf{0}$

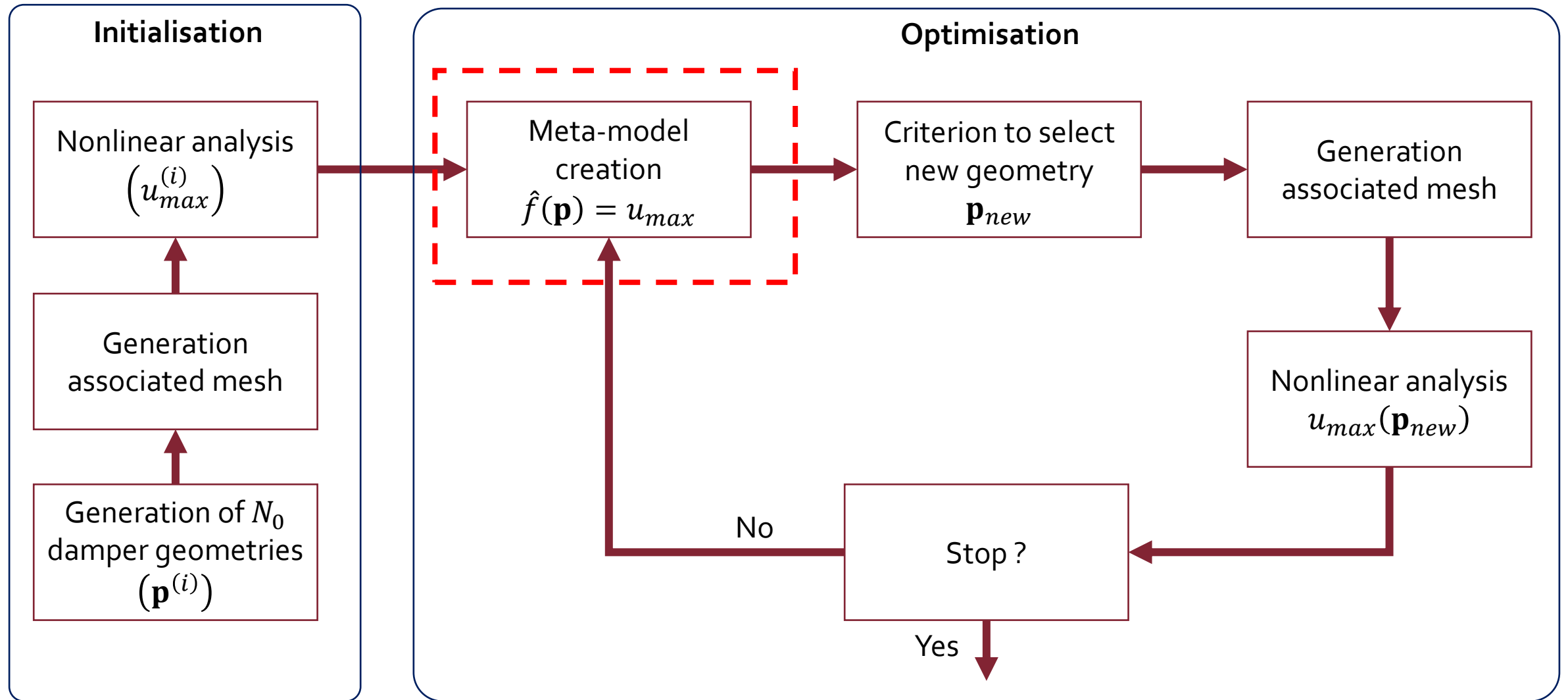


*Fig.: Full FRF versus Phase quadrature criterion
Vibration at tip blade
(●): phase quadrature solution*

Surrogate modelling – optimisation process



Surrogate modelling – optimisation process



Kriging – general formulation

Approximation :

$$f(\mathbf{p}) \sim \hat{f}(\mathbf{p}) = \underbrace{\sum_{i=1}^m \beta_i g_i(\mathbf{p})}_{\substack{\text{Regressive part} \\ \text{Global behaviour}}} + \underbrace{Z(\mathbf{p})}_{\substack{\text{Gaussian process} \\ \text{Local behaviour}}}$$

with

$\mathbf{p} \in \mathbf{R}^k$ Parameters vector (input)

N samples $\mathbf{P} = (\mathbf{p}^{(1)}, \dots, \mathbf{p}^{(N)})$
and their evaluations $\mathbf{F} = (f(\mathbf{p}^{(1)}), \dots, f(\mathbf{p}^{(N)}))$ *Called Design of Experiments (DoE)*

$(g_i)_{i=1,\dots,m}$ Regression functions

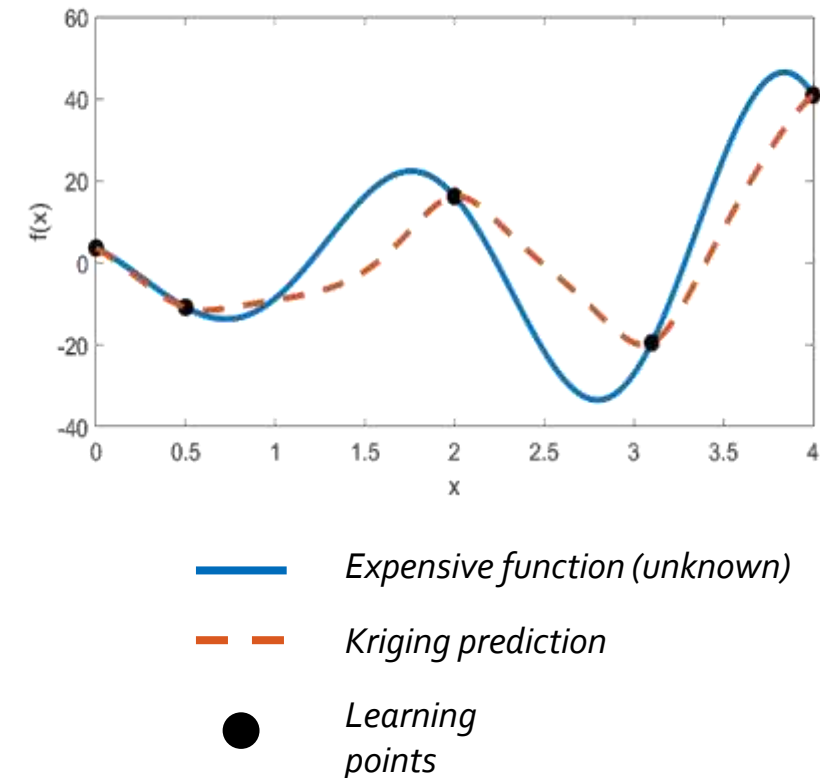
$(\beta_i)_{i=1,\dots,m}$ Regression coefficients

$Z(\cdot)$ Zero-mean Gaussian process

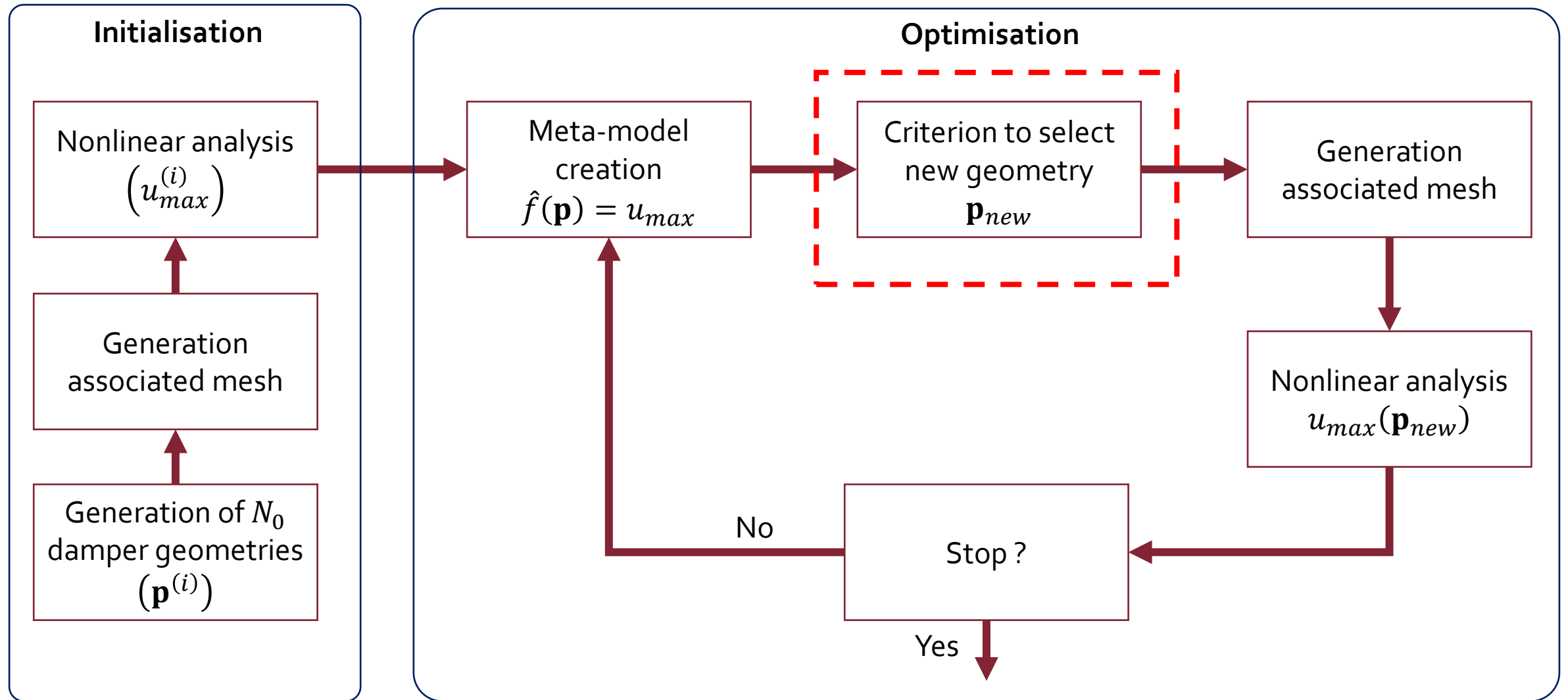
Covariance $\mathbb{E}[Z(\mathbf{s}), Z(\mathbf{x})] = \sigma^2 \mathcal{R}(\boldsymbol{\theta}, \mathbf{s}, \mathbf{x})$

Correlation function $\mathcal{R}(\boldsymbol{\theta}, \mathbf{s}, \mathbf{x})$

Illustration – 1d case :



Surrogate modelling – optimisation process



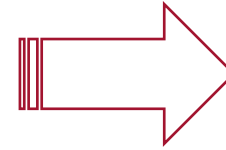
Expected improvement criterion

Modified Expected Improvement criterion:

$$EI(\mathbf{p}) = \begin{cases} (f_{min} - \hat{f}(\mathbf{p})) \Phi\left(\frac{f_{min} - \hat{f}(\mathbf{p})}{\hat{s}(\mathbf{p})}\right) + \hat{s}(\mathbf{p}) \phi\left(\frac{f_{min} - \hat{f}(\mathbf{p})}{\hat{s}(\mathbf{p})}\right) \\ 0 \\ \text{Penalty} < 0 \quad \text{if infeasible (disconnected)} \end{cases}$$

where

- $\hat{f}(\mathbf{p})$ kriging prediction at \mathbf{p}
- $\hat{s}(\mathbf{p})$ standard error of the kriging at \mathbf{p} (known)
- f_{min} current minimum observed so far
- Φ cumulative distribution function of the normal law
- ϕ probability density function of the normal law



New point

$$\mathbf{p}_{new} = \operatorname{argmax}_{\mathbf{p} \in \mathbb{R}^p} EI(\mathbf{p})$$

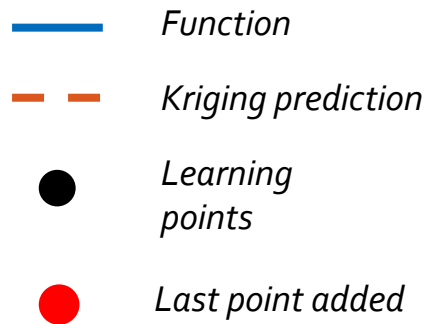
Resolution with genetic algorithm

Good balance between global and local search:

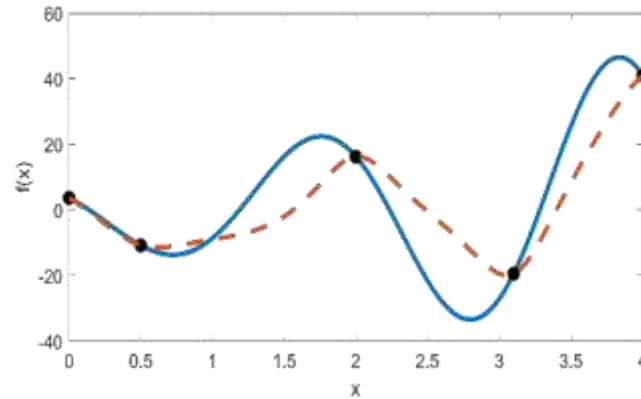
- Exploration of the space
 - area where no information is available
 - far from learning points
- Improvement of the minimum found
 - area where the probability to find the minimum of the objective function is high

Illustration of the process on a 1d case

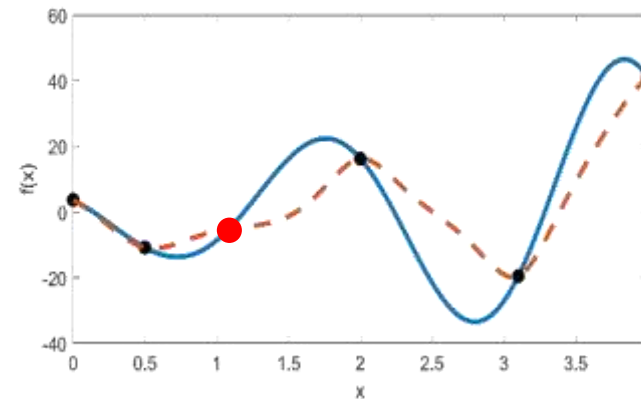
Function



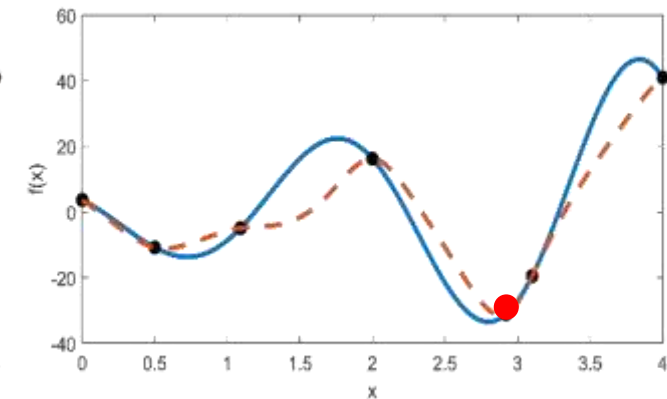
Step 1



Step 2



Step 3



Expected Improvement

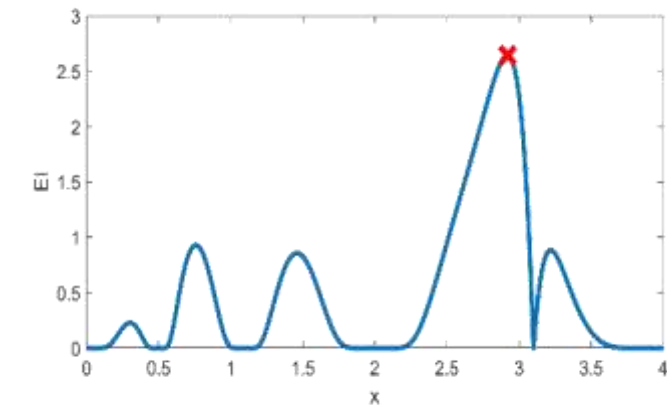
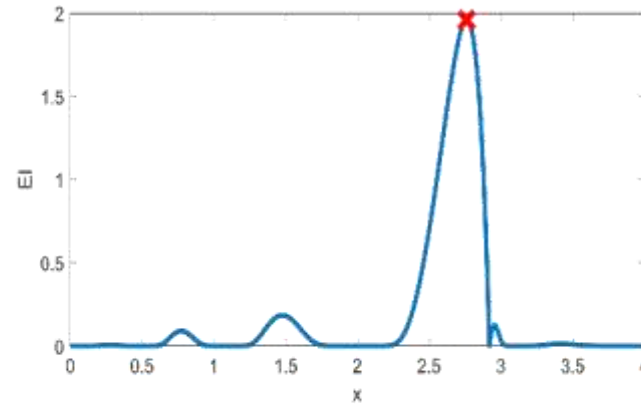
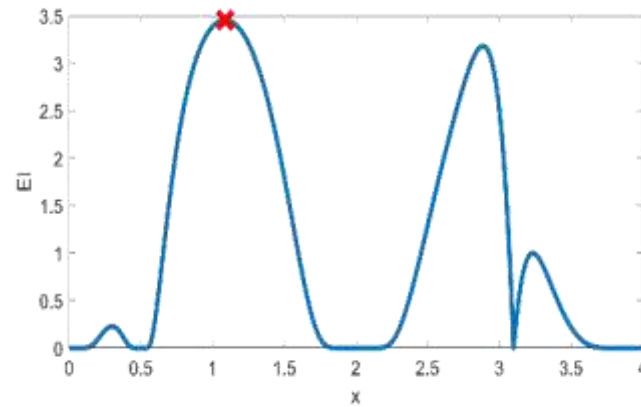
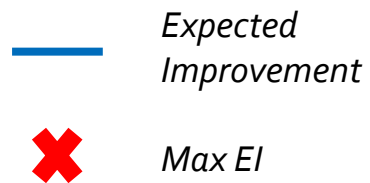
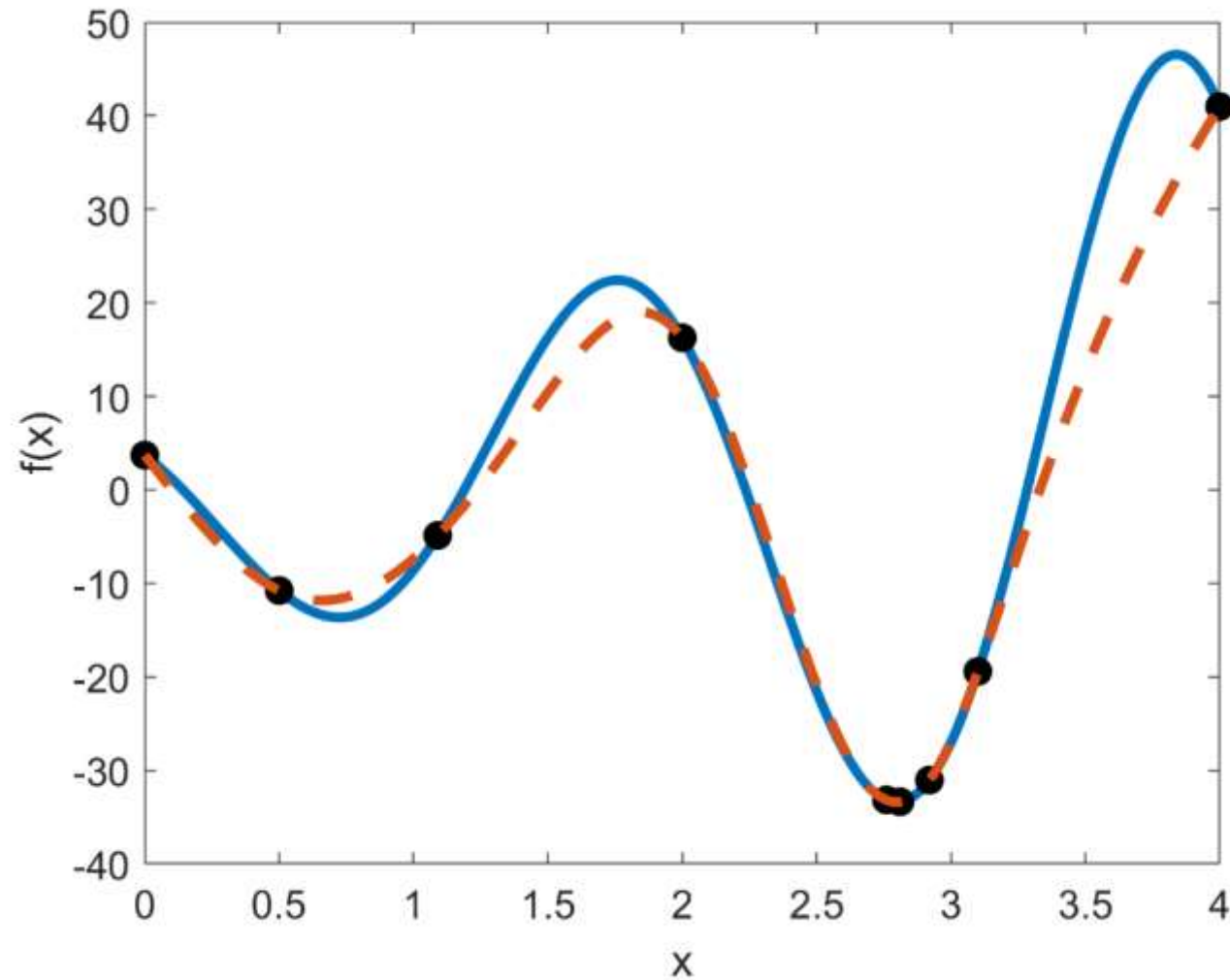
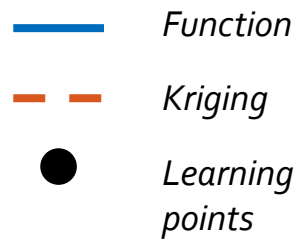
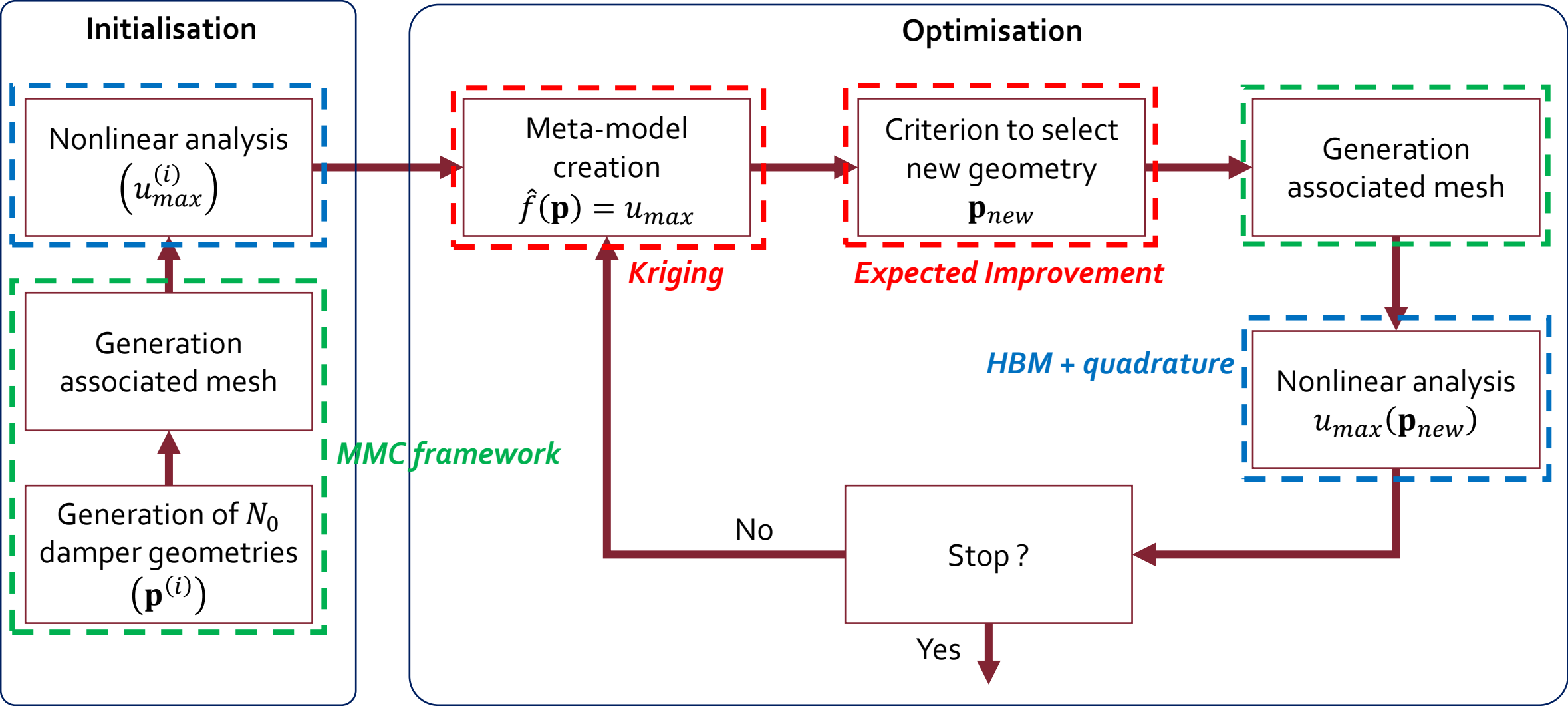


Illustration of the process on a 1d case

Step 5



Surrogate modelling – optimisation process



Results – general overview

Optimisation parameters:

- Optimisation of $-1/u_{max}$ instead of u_{max}
- 9 optimisation parameters
- 55 initial points
- 200 iterations

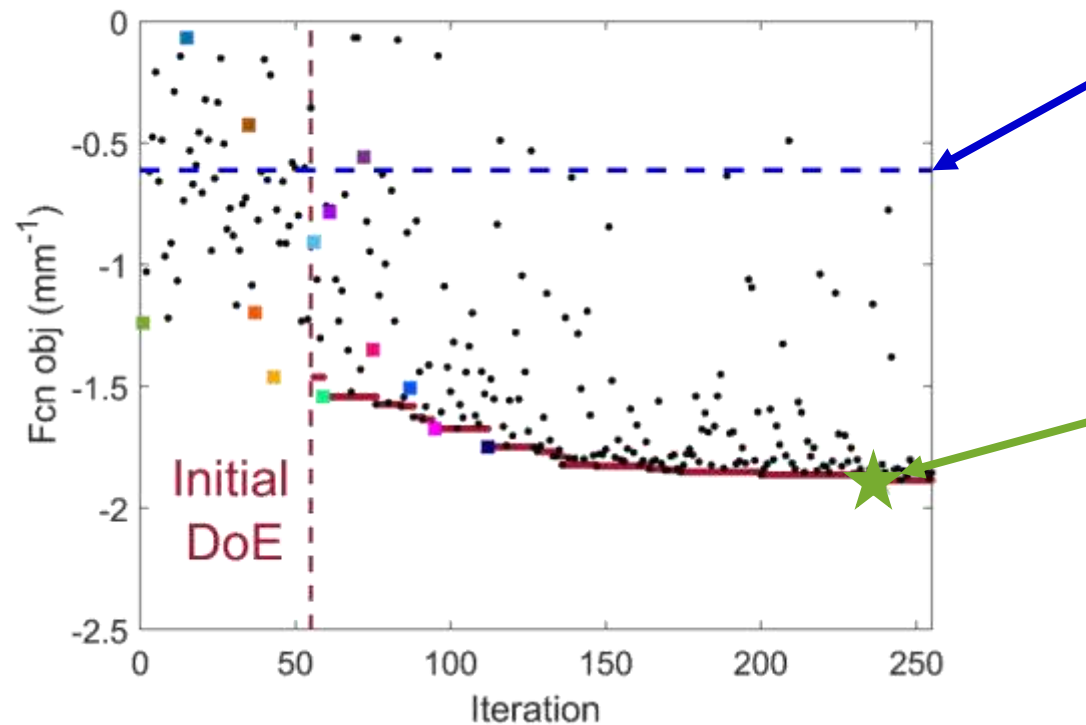


Fig.1: Evolution of the objective function vs iteration number
(—): current minimum

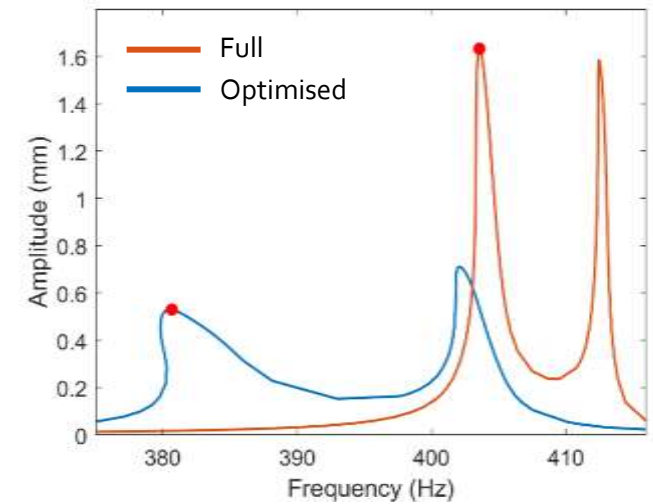
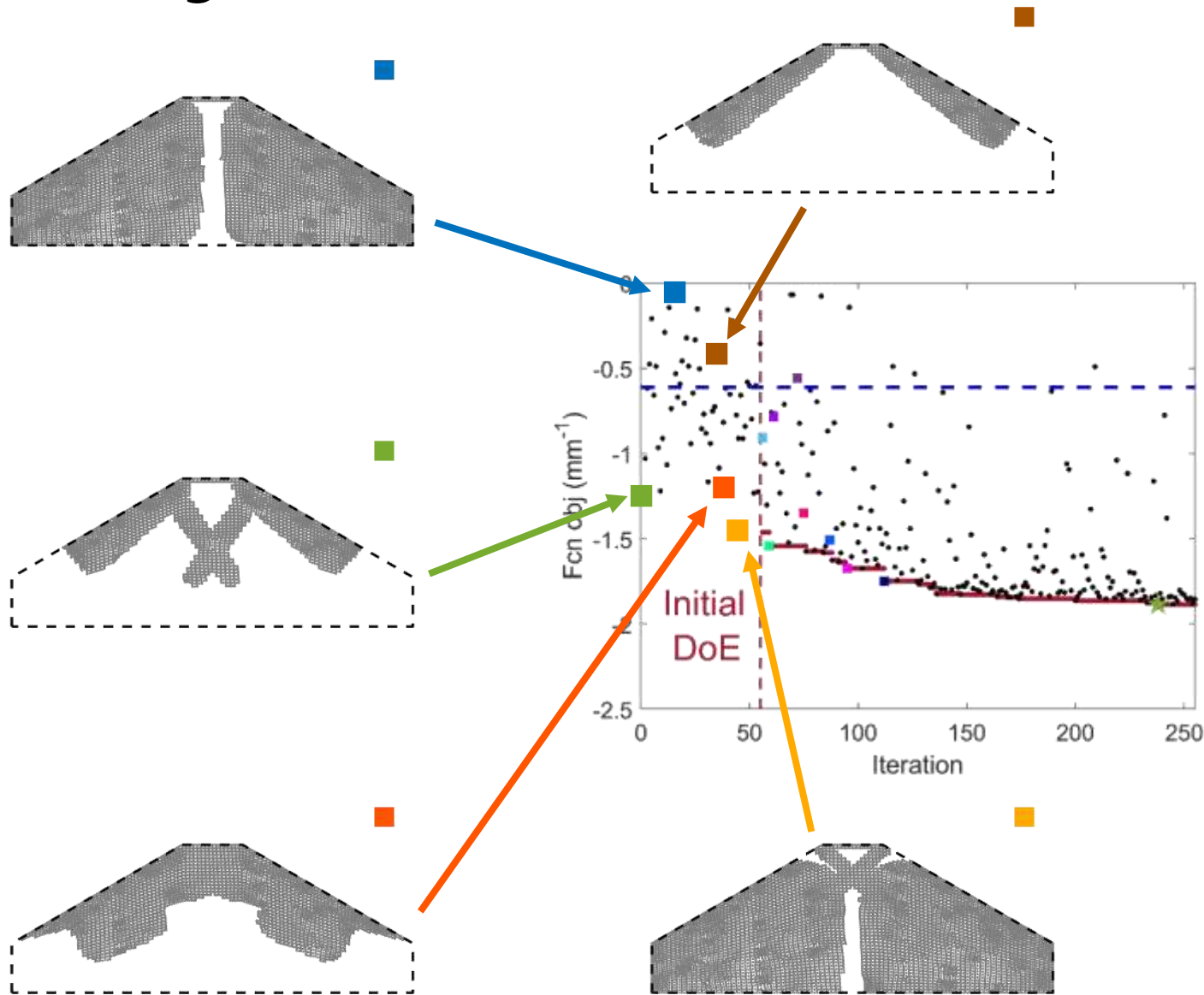


Fig.2: FRFs of the optimised (blue) and full damper (orange) dampers

Comparison:

- Vibrations divided by 3
- Frequency shift of 20 Hz
- Mass reduced by 87%

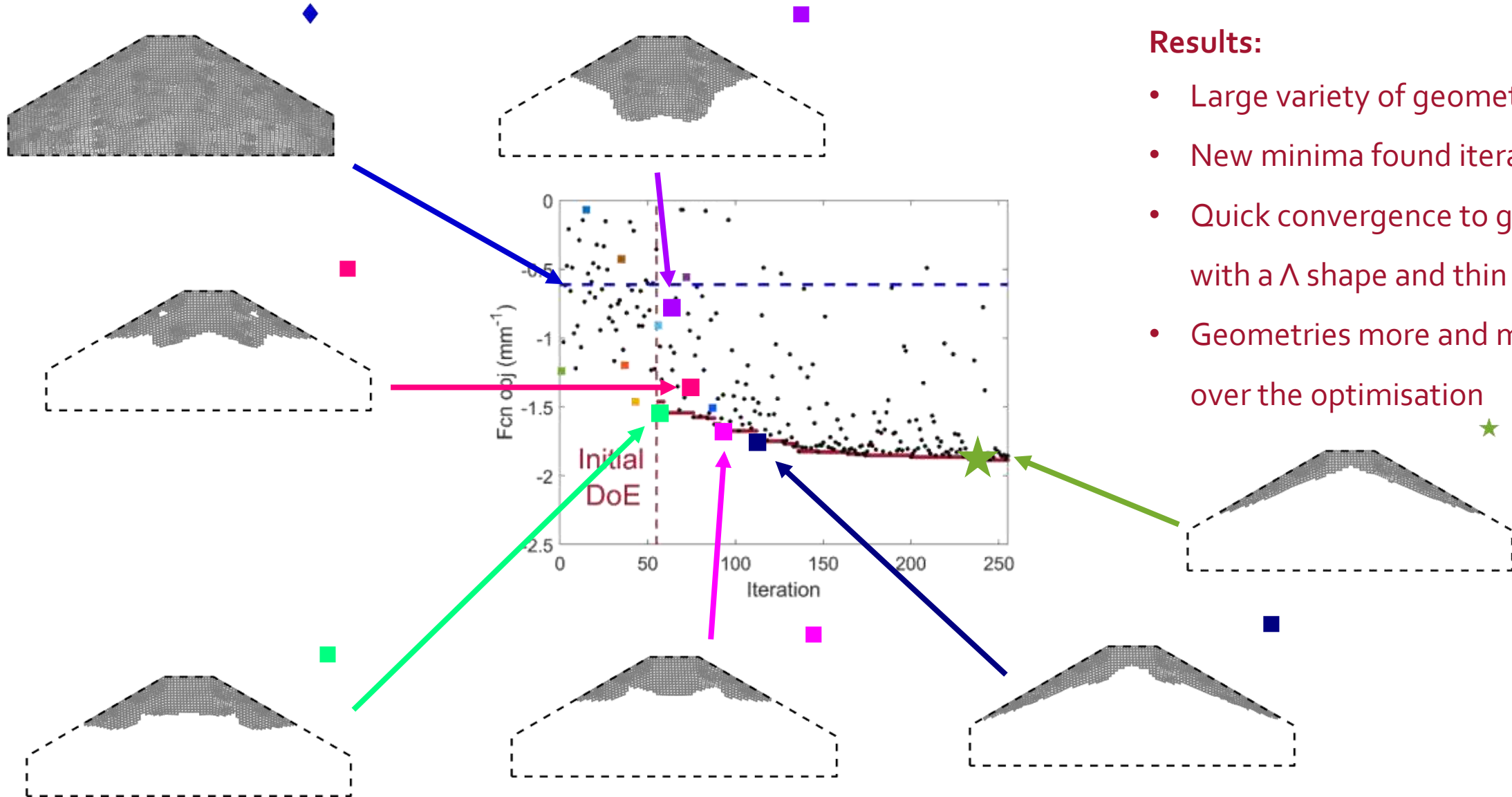
Results – general overview



Results:

- Large variety of geometries
- New minima found iteratively
- Quick convergence to geometry with a Λ shape and thin branches
- Geometries more and more similar over the optimisation

Results – general overview



Results:

- Large variety of geometries
- New minima found iteratively
- Quick convergence to geometry with a Λ shape and thin branches
- Geometries more and more similar over the optimisation

Results – general properties of the tested geometries

(●) : initial points – (★) : best geometry – (◆) : full damper

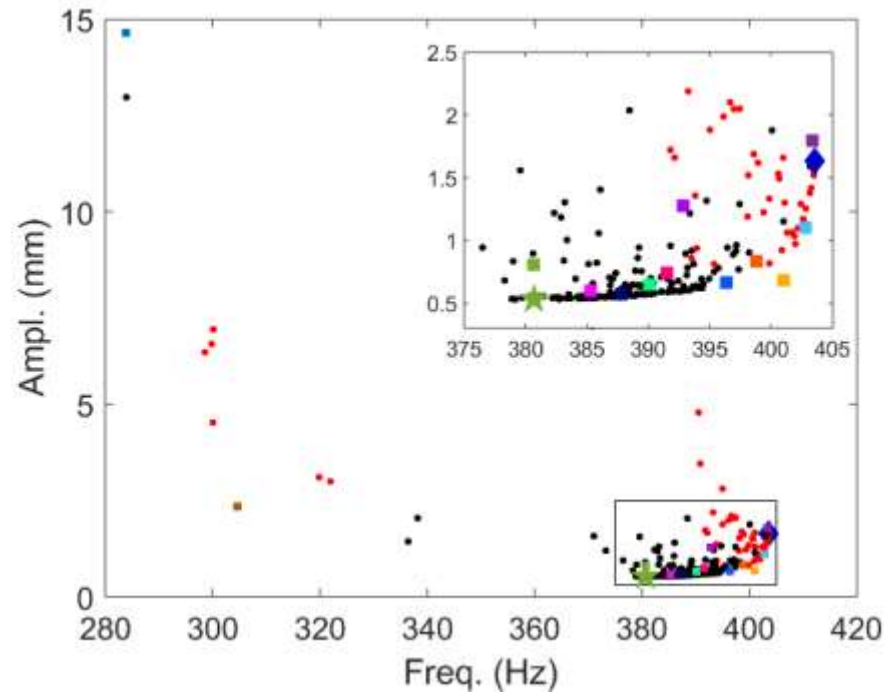


Fig.1: amplitude vs resonance frequency

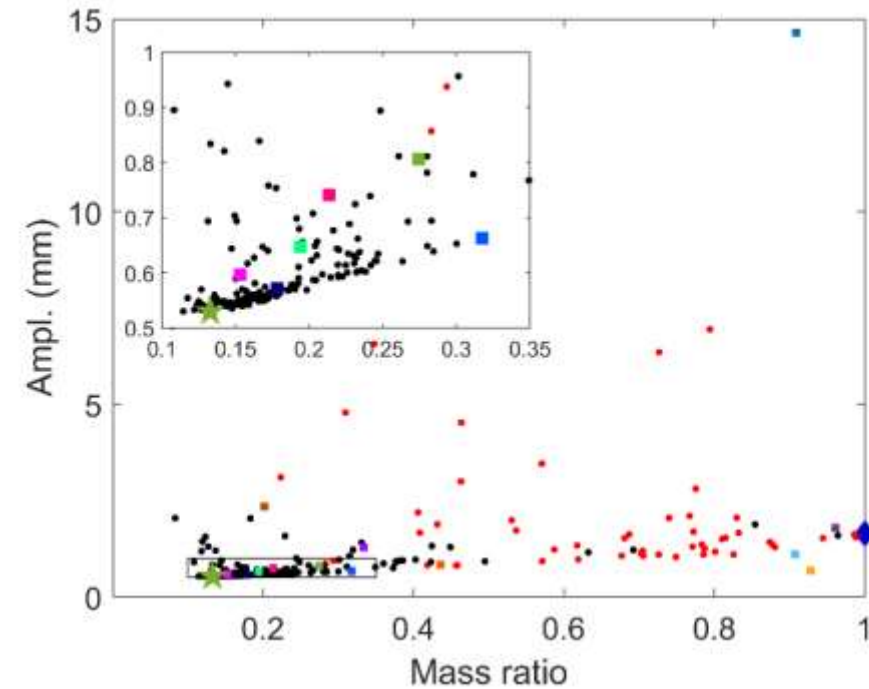


Fig.2: amplitude vs mass ratio

Results:

- Reduction of amplitude => reduction of resonance frequency
- Reduction of mass => reduction the vibration amplitude

Results – geometry clustering

Idea: group similar geometries to get general properties related to the geometry

Distance definition $d(\mathbf{p}^{(1)}, \mathbf{p}^{(2)})$:

comparison of elements in common in the geometry $\mathbf{p}^{(1)}$ and in the geometry $\mathbf{p}^{(2)}$

- $d = 1$: all elements are the same (void and material at the same location)
- $d = 0$: all elements are different

Results:

- Geometries different from each other in the initial set
- Geometries more and more similar over the optimisation

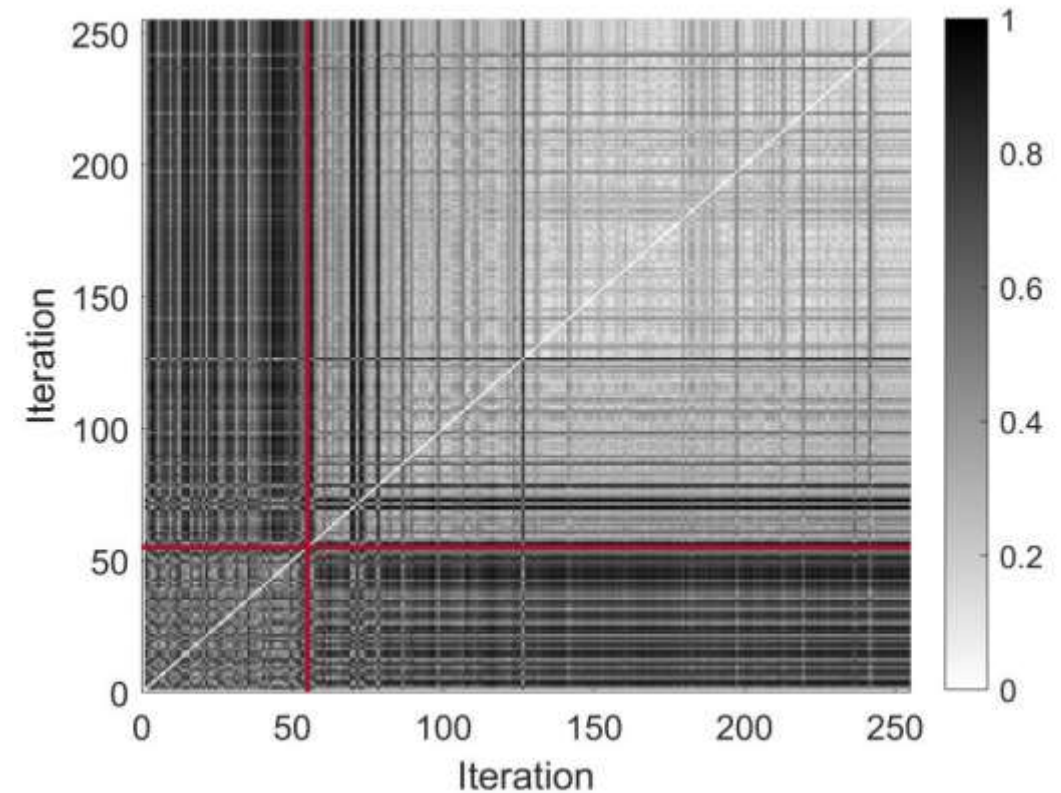


Fig.: matrix of the distances $d_{i,j} = d(\mathbf{p}^{(1)}, \mathbf{p}^{(2)})$

Results – geometry clustering

Idea: group similar geometries to get general properties related to the geometry

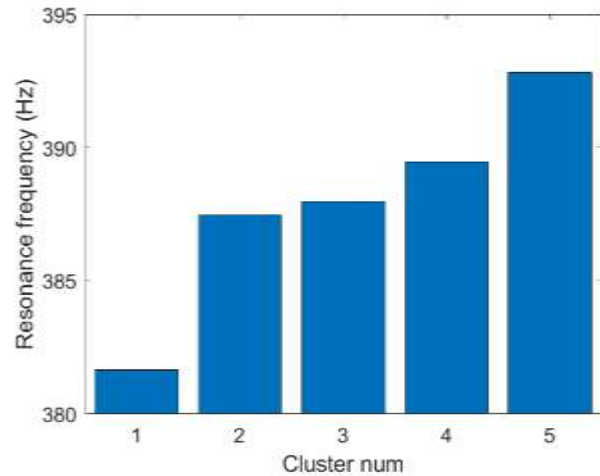
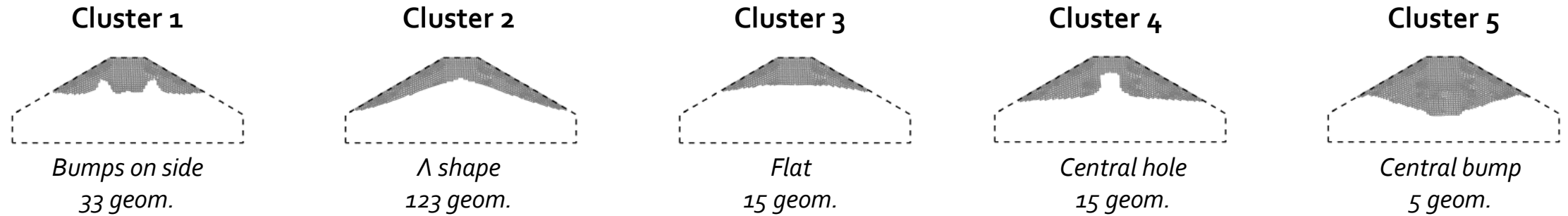


Fig.: Resonance frequency on average

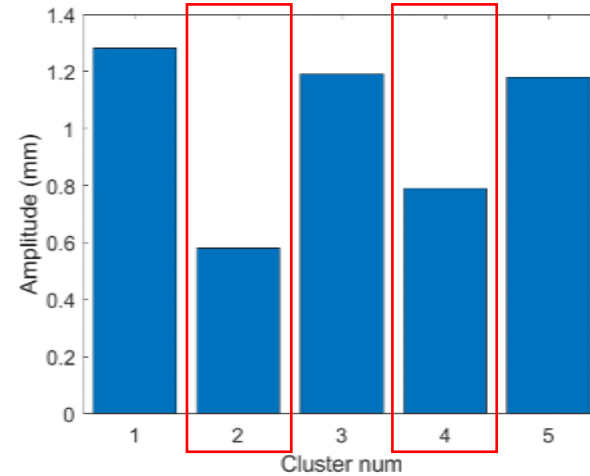


Fig.: Vib. amplitude on average

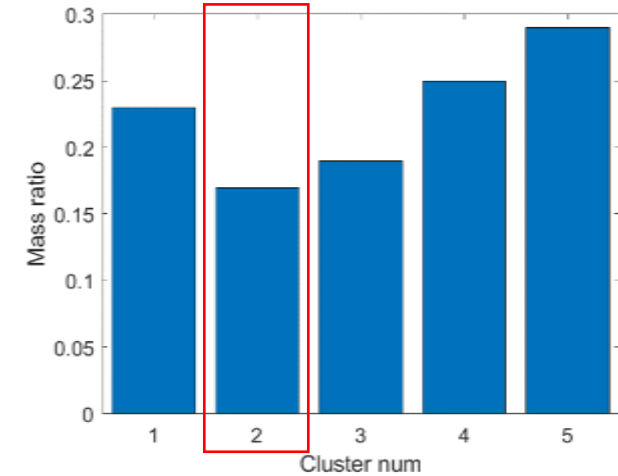
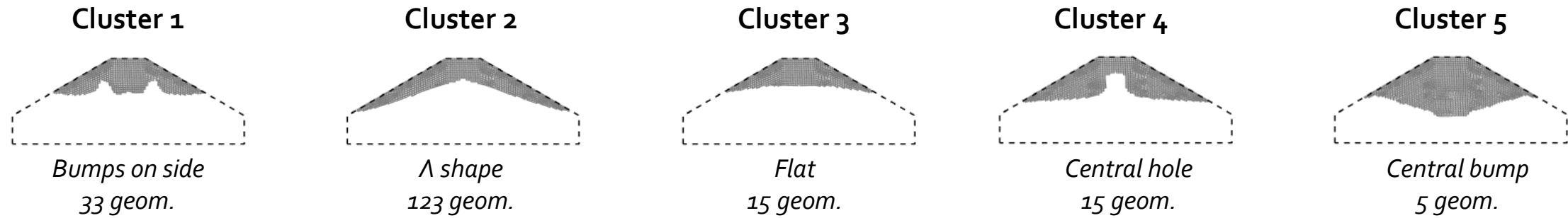


Fig.: Mass ratio on average

Results – geometry clustering



$$N = \text{constant} \cdot m/A$$

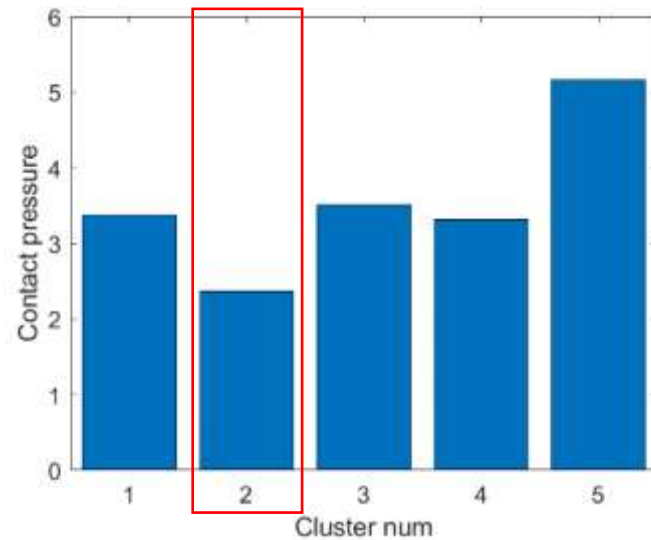


Fig.: Initial contact pressure on average

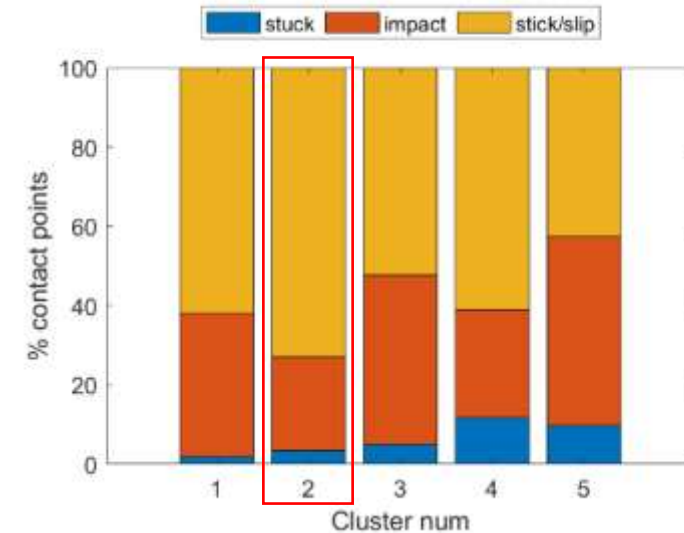


Fig.: Contact status distribution on average

Best geometries tend to minimise the initial contact pressure to minimise stuck points and maximise stick/slip
→ low mass and large contact area

Conclusion and perspectives

- Design of nonlinear dynamic systems = challenging
- Limitations due to computational time and realistic modelling of manufacturing, operating condition etc
- Surrogate modelling = promising tool
 - Physical properties integrated in DoE or regression
 - Prediction of QoI at reduced numerical cost
 - Deep understanding in the dynamic behaviour of complex systems
 - Topology and structural optimisation for nonlinear resonance mitigation
- Perspectives
 - Optimisation: robustness, 3D, experimental validation
 - Include more advanced nonlinear dynamic features (bifurcation)

Thanks!

Enora Denimal Goy

Inria, Saclay, France - Platon research team

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Related publications

- E. Denimal, J-J. Sinou, *Advanced kriging-based surrogate modelling and sensitivity analysis for rotordynamics with uncertainties*, European Journal of Mechanics - A/Solids, 2021, 90:1-20 [\[DOI\]](#) [\[HAL\]](#).
- E. Denimal, J-J. Sinou, *Efficient parametric study of a stochastic airfoil system based on hybrid surrogate modelling with advanced automatic kriging construction*, European Journal of Mechanics - A/Solids, 104926 (2023) [\[DOI\]](#).
- E. Denimal, L. Renson, C. Wong, L. Salles, *Topology optimisation of friction under-platform dampers using Moving Morphable Components and the Efficient Global Optimization algorithm*, Structural and Multidisciplinary Optimization, 65, 56 (2022), [\[DOI\]](#).