

# ENGG1300

Notes for HKU · Spring 2024

**Author:** Jax

**Contact:** [enhanjax@connect.hku.hk](mailto:enhanjax@connect.hku.hk)

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# 1 Vectors

## Basics

Vector can be represented by  $\vec{F}$  or  $\mathbf{F}$ . The magnitude can be represented by  $|F|$  or  $F$ .

## Unit vectors

They are vectors with magnitude 1.  $\hat{A} = \frac{\vec{A}}{|A|}$

## 1.1 Coplanar vectors

### Cartesian vector notation

In two dimensions, the Cartesian unit vectors  $\mathbf{i}, \mathbf{j}$  are used to designate the directions of the x and y axes respectively.

$$\mathbf{F} = F_x \mathbf{i} + F_y \mathbf{j}$$

Where  $F_{x/y}$  is the x/y component of  $F$ . And to find the x/y components we can use trigonometry:

$$F_x = F \cos \theta, \quad F_y = F \sin \theta$$

Where the angle  $\theta$  is the angle between  $F$  and the x-axis.

### Resultant force

The resultant force  $F_R$  can be found by the sum of the components of  $F$ :

$$F_R = \sum F$$

In Cartesian form, it's the same as adding all the terms together:  $F_R = (F_{x1} + F_{x2})\mathbf{i} + (F_{y1} + F_{y2})\mathbf{j}$

### Orientation of vector

We always consider the angle between  $F$  &  $F_x$ . It can be found by  $\theta = \tan^{-1} \frac{F_y}{F_x}$ .

### Magnitude of forces

The magnitude will simply be the square root of the sum of squared components of the force:

$$|F| = \sqrt{F_x^2 + F_y^2 + \dots}$$

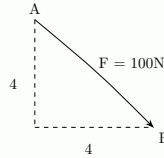
### Moving vectors by their position vectors

We can move a force  $F$  to a point using its position vector  $r$  (pointing to their tail), using the following definition:

$$\vec{F} = |F| \times \hat{F} = |F| \times \frac{r}{|r|}$$

The position vectors are in Cartesian form, so the moved vectors will also be in Cartesian form.

Consider the following graph:



First, we consider  $r_{AB}$ . From the graph:

$$r_{AB} = 4\mathbf{i} + 4\mathbf{j}$$

Then, we find the magnitude of  $r_{AB}$ :

$$|r_{AB}| = \sqrt{4^2 + 4^2} = 5.65 \dots$$

Finally, we apply our formula for  $F$ :

$$\begin{aligned}\vec{F} &= |F| \times \frac{r_{AB}}{|r_{AB}|} = 100 \times \left( \frac{r_x}{5.65} \mathbf{i} + \frac{r_y}{5.65} \mathbf{j} \right) \\ &= 100 \times \left( \frac{4}{5.65} \mathbf{i} + \frac{4}{5.65} \mathbf{j} \right) \\ &= 70.7\mathbf{i} + 70.7\mathbf{j} \text{ N}\end{aligned}$$

## 1.2 Vectors in 3D

The concepts above can be extended to 3D simply by adding another variable to the system.

## 2 Moment of forces

### Definition of moment

The moment of a force is a measure of its *tendency* to cause a body to rotate about a specific point. The moment about a point  $O$ , when  $F$  is applied a distance  $d$  from the point is:

$$M_O = F \times d$$

Keep in mind that *positive* moment is *anti-clockwise*.

### 2.1 Coplanar / 2D moment

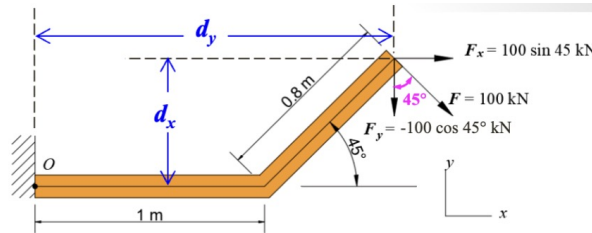
#### Resultant moments

The resultant moment is the **sum** of all moments present on the point, given by:

$$M_R = \sum M_O$$

#### 2.1.1 The moment of a non-linearly attached force

One simple way is to find the *components* of the force, and sum their individual moments together. The following is a simple example:



After finding the component forces of  $F$ , we can deduce the resultant moment to be:

$$\begin{aligned} M_O &= F_x \times 0.8 \sin 45 \text{ deg} + F_y \times (1 + 0.8 \cos 45 \text{ deg}) \\ &= -150.7 \text{ kNm} \end{aligned}$$

### 2.2 Non-coplanar / 3D moment

#### Moments in a 3D system

Consider position vector  $\vec{r}$  drawn from  $O$  to any point on the *line of action* of  $F$ . The moment can hence be given by:

$$M_O = \vec{r} \times \vec{F}$$

### Finding the moment via cross products Cartesian vectors

The cross product  $C$  given by  $A$  and  $B$  is:

$$A \times B = \begin{vmatrix} i & j & k \\ A_x & A_y & A_z \\ B_x & B_y & B_z \end{vmatrix} = C$$

The cross-product for vectors going in the *same direction* is 0. (i.e.  $n\mathbf{k} \times m\mathbf{k} = 0$ )

### Resultant moments

The resultant moment is simply the **sum** of *couple moments* and moments of forces:

$$(M_R)_O = \sum M_O + \sum M$$

You can interpret  $(M_R)_O$  as the resultant moment about point  $O$ .

## 2.3 Couple moments

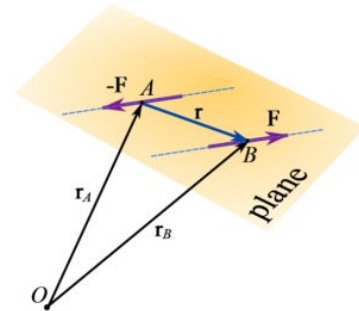
Couples are *two parallel forces* that have the same magnitude but have *opposite directions*, separated by a *perpendicular distance*  $d$ . The magnitude of the moment is given by:

$$M = Fd$$

Notice that there's no point mentioned so far. For couple moment, it is **always the same about any point**. Let's assume for any point  $O$  (refer to graph), the moment is:

$$\begin{aligned} M_O &= r_B \times F + r_A \times -F \\ &= (r_B - r_A) \times F \\ &= r \times F \quad \text{which is independent of } O \end{aligned}$$

Hence, we can say that couple moments are **free vectors**.



### 3 Axially loaded members

Axial loading refers to the application of a force along the axis of the member.

#### 3.1 Stress and strain

##### Axial stress

Axial stress is the stress that is *parallel* to the cross-sectional area of the member. It is given by:

$$\sigma = \frac{F}{A} \quad (Nm^{-2})$$

Where  $F$  is the force applied, and  $A$  is the cross-sectional area of the member. Note that  $1Pa = 1Nm^{-2}$ .

##### Eccentric loading and stress

When a force is applied *off-centre* to the member, the stress at each end is given by:

$$\sigma = \frac{\sum F}{wd} \pm \frac{6F \times e}{d \times w^2}$$

Where  $w$  is the width of the member,  $d$  is the depth of the member, and  $e$  is the eccentric distance from the centroid of the member to the point of application of the force.

The  $\pm$  sign is used to denote the *maximum* and *minimum* stress on opposite sides.

##### Axial strain

Axial strain is the ratio of the change in length to the original length of the member. It is given by:

$$\epsilon = \frac{\Delta x}{x} \quad (\text{Ratio})$$

Where  $\Delta x$  is the change in length, and  $L$  is the original length of the member.

#### 3.2 Materials

##### Strength

Material strength is defined as the **maximum stress** that can be resisted by the material.

##### Young's Modulus

Young's modulus is the ratio of stress to strain, given by:

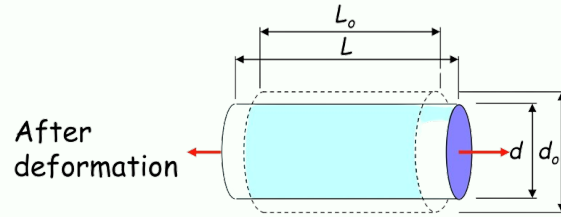
$$E = \frac{\sigma}{\epsilon} = \frac{Fx}{A\Delta x} \quad (Pa, Nm^{-2})$$



### Poisson's ratio

Poisson's ratio is the ratio of lateral strain  $\epsilon_l$  to axial strain  $\epsilon$ , given by:

$$\nu = \frac{\epsilon_l}{\epsilon} \quad (\text{Ratio})$$



The lateral strain  $\epsilon_l$  is  $\frac{\Delta d}{d_o}$ .

## 3.3 Hydrostatic pressure

### Hydrostatic/water pressure

The water pressure acting on any surface is *always perpendicular* to the surface, and the pressure is given by:

$$p = \rho gh \quad (Pa, Nm^{-2})$$

Where  $\rho$  is the density of water, and  $h$  is the depth of the water.

Hence, we can see that the water pressure **increases linearly with depth**.

### Water pressure load on slanted surface

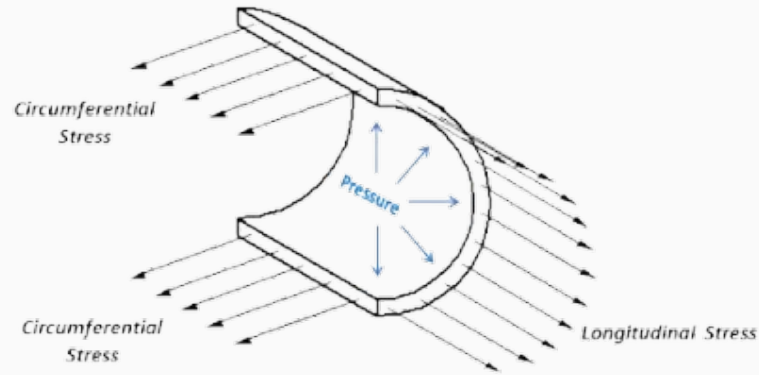
The load exerted on a slanted surface by water pressure  $F$  is given by and located at:

$$F = \frac{\rho g d w L}{2} \quad @ \frac{1}{3} L / \frac{2}{3} d$$

Where  $d$  is the depth of water,  $w$  is the width of the volume, and  $L$  is the length of the surface.

### Water pipe

1. **Internal pressure** refers to the pressure inside the pipe.
2. **Internal force** refers to the pressure's effect onto a side of the pipe.
3. **Internal stress** refers to the stress caused by the internal force, acting along the *circumferential direction* of the pipe. This is also called the **hoop stress**.



## 4 Statics

### 4.1 Equilibrium of rigid bodies

#### The 3 equations of equilibrium

A rigid body is in equilibrium if:

$$\sum F_x = 0, \quad \sum F_y = 0, \quad \sum M_O = 0$$

Where  $O$  is any point.

#### Static equilibrium

A system is in **static equilibrium** if it experiences no acceleration when loads are applied to it.

#### Free body diagrams (FBD)

Free body diagrams are diagrams that show all the forces acting on a body. They are useful in determining the forces that cause the body to be in equilibrium.

### 4.2 Members

A member is a *straight* and *long* structural element that is subjected to axial forces. The following are the types of members:

### 4.3 Static systems

#### Supports and reaction forces

Each support will restrict the movement of the rigid body in a certain way. If a certain degree of freedom is restricted, a reaction force will be present to counteract the force that would have caused the movement.

Support type	Restricted movement and reaction forces	Shape
Fixed	x, y, r (moment)	Flat
Pinned	x, y	Triangle
Roller	y	Circle

#### Stable structures

A structure is said *stable* if all members remain in place under any loading conditions.

#### Static conditions of a system

A system's state of equilibrium can be determined by the number of restraints present:

1. Insufficient restraints → **non-static system** (contains mechanisms)
2. Sufficient restraints → **static system**

A static system is said to be **statically indeterminate** if the number of unknowns is larger than the number of equations of equilibrium.

#### Degree of indeterminacy (Trusses)

$$I = m + r - 2j$$

Where  $m$  is the number of members,  $r$  is the number of reaction forces,  $j$  is the number of joints.

#### Degree of indeterminacy (Frames)

$$I = r - 3 + 3n$$

Where  $n$  is the number of members that we can cut through for the frame to be statically determinate.

## 4.4 Loading

#### Types of loading

1. **Concentrated load** is a force applied at a single point. ( $N$ )
2. **Distributed load** is a force applied over a length. ( $N/m$ )

#### Distributed load

A *distributed load* ( $w$ ) is a force that is distributed over a length ( $l$ ), that has the unit  $N/m$ .

The **resultant force** is the supposed area of the load. To consider the **moment** by a distributed load, we can treat the load as a *single force* acting at the **centroid** of the load.

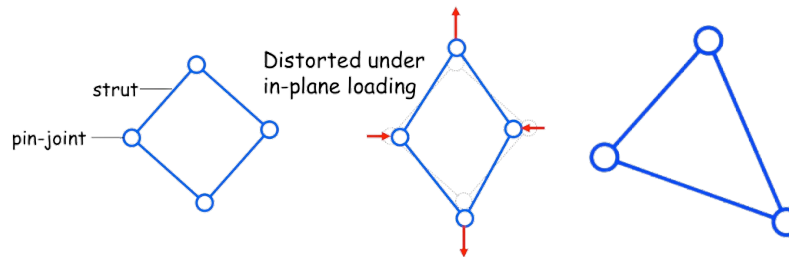
- Unifrom →  $F = wl$  @  $\frac{1}{2}l$
- Triangular →  $F = \frac{1}{2}wl$  @  $\frac{1}{3}l$  (from the base of the load)

## 5 Structural analysis

### 5.1 Pin-jointed trusses

A **truss** is a structure that consists of *straight members* connected at their ends by *pin joints*.

A triangular truss is a statically determinate structure, but a quadrangular truss is not.



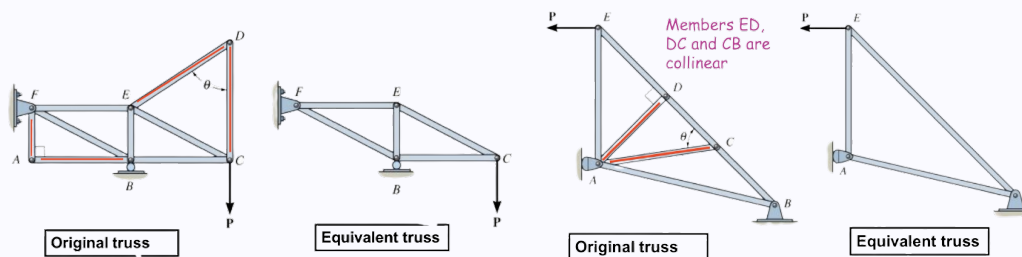
#### Zero-force members

The following are zero-force members given that **no external load or support reaction is applied to the joint**.

- **2 non-collinear members** of a two-member-joint
- **3rd member** of a three-member-joint, where the other two are collinear.

Additionally, members that provide **no structural support against the applied load** are also considered to be zero-force members.

Removing zero-force members allows us to simplify a truss structure.



### 5.2 Statically determinate trusses

To construct an *internally* statically determinate truss, we begin with a triangular pin-jointed truss and then successively adding two new members with a new joint.

We say the truss is only *internally* statically determinate because the external reactions are not yet known.

### 5.3 Truss analysis

When we analyze a truss, we assume that forces are only applied at joints.

### Internal vs. External forces

- Forces acting on the members → **Internal forces**
- Forces acting on joints → **External forces**

### Sign convention of analysis

As we focus on the analysis of **internal forces acting on joints**, we always draw:

- Tension forces → **pointing away** from the joint  $\{\leftarrow \circ \boxed{\rightarrow T \leftarrow} \circ \rightarrow\}$
- Compression forces → **pointing towards** the joint  $\{\rightarrow \circ \boxed{\leftarrow C \rightarrow} \circ \leftarrow\}$

During analysis, we always draw forces **pointing towards** the joint  $\{\rightarrow \circ \leftarrow\}$ , which means we treat **compression as positive**.

We also treat **downward forces as positive**  $\{+ \downarrow\}$ .

## 5.4 Equilibrium sections

If we cut out any section of an equilibrium system, the cut will be in equilibrium. That is, the **external forces** are balanced by **internal forces**.

Therefore, to solve for the forces in a system, we simply cut out a section that we deem solvable (by practice!) and draw its FBD. The forces includes *all external forces in the cut* as well as the *internal forces of the cut members*.

## 5.5 Cable analysis

### Cable forces

Cables are **always in tension**. The tension force is always in the direction of the cable.

Some characteristics of cable structures:

- Supports are always 2 inverted pinned supports
- Applied forces all point downwards due to gravity
- Horizontal reaction forces at supports are equal in magnitude

Hence, we can use the following steps to solve for cable forces:

1. Find reaction forces by considering joint equilibrium
2. Tension in cables attached to a support can be found by  $\sqrt{(R_x)^2 + (R_y)^2}$
3. Tension in cables not attached to a support can be found by  $\sqrt{(R_x)^2 + (R_y - \sum y)^2}$

## 6 Bending forces and stress

### 6.1 Internal force in beams

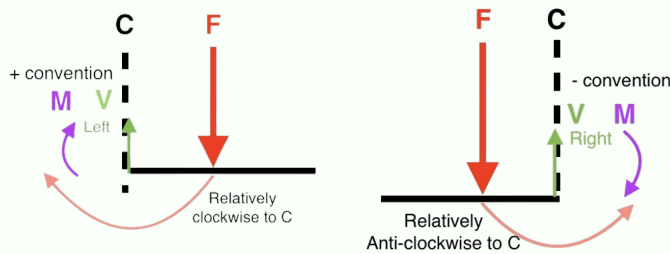
When we consider the internal forces of a beam, we take a *cross section* and consider the forces acting on it. The internal forces are:

- **Shear force**  $V$  is the force acting parallel to the cross-section.
- **Bending moment**  $M$  is the moment generated by the variation of forces acting perpendicularly to the cross-section.

#### Sign convention of internal forces

{+ ◯} The idea is that if the acting force is relatively **clockwise** to the cross-section  $C$ , it is **positive**. The following is the signs of internal forces for a *positive* moment:

- **V**: ◯ (L ↑ R ↓)
- **R**: ◯ ◯ (sagging moment is *positive*)



Couple moments can be treated as a rotational moment at clock-direction the arrows are pointing to.

#### Finding internal forces by side

To find the internal forces at cross-section  $C$ , we consider the *left* and *right* side of the section:

- **V**:  $\sum F$  on either side
- **M**:  $\sum M_C$  on either side about  $C$

Note that the sums are all **signed by convention**. (e.g. A force  $F$  is positive if it points upwards on the left side.)

This also gives us the fact that  $\sum F_L = \sum F_R$  and  $\sum M_L = \sum M_R$ , signed by convention on both sides.

### 6.2 Diagramming internal forces

#### Diagramming internal forces

The two internal force diagrams are  $V(x) - x$  and  $M(x) - x$  diagrams.  $V(x)$  gives the shear force  $V$  at distance  $x$  from the *left side* of the beam.

### Relation between internal forces

The following is the relation between the internal forces:

$$\frac{dM}{dx} = V \quad \frac{dV}{dx} = -w$$

Where  $w$  is the distributed load acting on the beam.

- This tells us that the slope of the  $M(x)$  diagram at an interval is the value of  $V$  at that interval.
- This tells us that the slope of the  $V(x)$  diagram at an interval is the value of  $-w$  at that interval.

Steps to follow:

1. Solve for reaction forces
2. Draw the FBD of a cut from the leftmost side to a location with distance  $x$  from the leftmost force *not analysed*
3. Analyse the FBD of the cut (solve for  $V$  and  $M$ )
4. Draw the graphs in the cuts and repeat for all cuts

*Figurative steps:*

1.  $\rightarrow R_A, R_B \dots$
2.  $\left\{ \overset{R}{\uparrow} \underbrace{\text{=====}}_{\text{=====}} \downarrow \overset{V}{\downarrow} \overset{M}{\circ} \dots \right\}$
3.  $\rightarrow V, M^x$
4.  $\rightarrow V(x), M(x)$

To analyse a distributed load  $w$ , consider the FBD of the whole system on the left, then let  $x$  be the distance from the leftmost force *analysed*:

$$\left\{ \overset{F}{\downarrow} \underbrace{\text{=====}}_{\text{=====}} \downarrow \overset{w \cdot x}{\downarrow} \downarrow \overset{V}{\downarrow} \overset{M}{\circ} \dots \right\}$$

## 6.3 Bending stress



## 7 Centroids and moment of inertia

### 7.1 Centroids

#### Expression of centroids

We use  $\bar{x}$  and  $\bar{y}$  to denote the *centroid* of a shape. They are the horizontal and vertical distance about a given **reference axis**.

The centroid is the **centre of mass** of the shape. For an axis that the shape is *symmetric about*, the centroid will be **on the axis**.

#### Centroid of composite shapes

The centroid of a shape can be found by:

$$\bar{x} = \frac{\sum(A_i x_i)}{\sum A_i}, \quad \bar{y} = \frac{\sum(A_i y_i)}{\sum A_i}$$

Where  $A_i$  is the area of the composition shape, and  $x_i, y_i$  are the distances of the composition shape's centroid from the reference axes.

### 7.2 Moment of inertia

#### The reference axis for moment of inertia

We measure the moment of inertia about a **reference axis**. This is because at different points, the moment of inertia will be different. (*Unlike the use of reference axes in centroids, where the reference axes is just relative*).

For finding the moment of inertia at the centroid, we simply set the reference axis at the centroid.

#### Moment of inertia

The moment of inertia is a measure of an object's resistance to changes in its rotation, about a reference axis. It is given by:

$$I_x = \int y^2 dA, \quad I_y = \int x^2 dA$$

Where  $x$  and  $y$  are the distances from the axis.

### Moment of inertia for rectangles

The moment of inertia for a rectangle is given by:

$$I_x = \frac{bh^3}{12}, \quad I_y = \frac{hb^3}{12}$$

Where  $b$  is the width of the rectangle, and  $h$  is the height of the rectangle. This can be derived by the formulas above.

### Parallel axis theorem

The moment of inertia about an axis parallel to a reference axis for a shape is given by:

$$I_{x2} = I_{x1} + Ad^2$$

Where  $A$  is the area of the shape, and  $d$  is the distance between the **reference axis** and the **parallel axis**.

### Moment of inertia for composite shapes

The moment of inertia for a composite shape is simply the sum of the moments of inertia of the individual shapes **about the same axis**.

To find  $I_c$  of the composite shape, we *translate*  $I_{c_i}$  (moment of inertia the centroid of the individual shapes) to the centroid of the composite shape using the **parallel axis theorem**, and then sum them up.