# Backdraft Experiments and Large Eddy Simulations in a Scaled Compartment

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# 6 Highlights:

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- Initial conditions for computer fire models are defined from an extensive dataset of backdraft experiments performed at NIST.
  - Chemical composition and heat measurements from these experiments are intended to assess fire models.
  - The Fire Dynamics Simulator with fast chemistry is exercised on some of these backdraft scenarios.
  - For mix-controlled fast reactions, ignition model temperature threshold and ignition procedure are found to play a primary role in backdraft outcomes.
- Abstract: An extensive set of backdraft experiments has been performed at the NIST National
- <sup>16</sup> Fire Research Laboratory. These experiments were conducted in a reduced scale compartment,
- and are part of defining the conditions conducive to backdraft. Further, the detailed chemistry and
- heat measurements are intended to evaluate computational fire models. In this article, we describe
- the modeling effort employing the Fire Dynamics Simulator (FDS) in simulations involving
- 20 different fuels and ignition source locations mirroring a subset of the named experiments. We
- focus on using default simulation parameters and study their effect on the backdraft outcomes. In
- particular, the ignition model's temperature threshold and ignition procedure play a primary role
- 23 in developing a backdraft.

## 24 Keywords:

25 Backdraft Experiments; Fire Simulation; FDS; Large Eddy Simulation;

#### 26 1. Introduction

- 27 Backdrafts pose a life-threatening risk to any firefighter that may encounter them. A backdraft is a
- <sup>28</sup> deflagration resulting from igniting an incoming gravity current of air mixed with a heated,
- <sup>29</sup> fuel-rich, and oxygen-depleted environment [1]. Several experimental works have identified the
- 30 physical mechanisms and conditions conducive to backdrafts. Fleischmann examined the thermal
- and gas compositions within an enclosure's environment influencing the incoming gravity
- <sup>32</sup> current [2, 3]. Gottuk and others [4] estimated critical mass fractions in vitiated atmospheres,
- resulting in backdraft using full-scale steel compartments. Wu and colleagues [5] established a
- 34 correlation between the likelihood of backdraft and compartment configuration by varying

- ventilation conditions, ignition locations, and mass fluxes of gas leakage. Other works have
- 36 studied the probability of backdraft generated from solid fuel combustion within an
- 37 enclosure [6, 7, 8].
- On the other hand, several studies have relied on computational models utilizing large eddy
- simulation (LES), particularly the Fire Dynamics Simulator (FDS), to simulate backdraft. In
- general, these studies use finite-rate chemistry and several-step reactions to capture deflagration
- physics. For example, Weng, Fan, and Hasemi used LES to predict the ignition time for backdraft
- in compartments with different opening geometries [9]. Ferraris and others implemented a
- subgrid-scale model based on mixture fraction to define partially premixed combustion applied to
- backdraft [10]. Park et al. used single and three-step finite rate chemistry in FDS to investigate
- the relationship between the initial fuel mass in a compartment and the resulting backdraft [11].
- 46 Ashok and Echekki used FDS to study the effect of the gravity current magnitude on backdraft
- events [12]. Of particular interest is the work of Myilsamy et al., who used single-step fast
- chemistry to study the fuel fraction required for backdraft in different opening geometries [13].
- 49 Several of these studies have established the critical elements that impact the generation of a
- backdraft. Yet, to date, a constraint has been the limited availability of comprehensive
- 51 experimental datasets to compare simulation results.
- Recently, an extensive series of 500 experiments focused on examining backdraft in a
- reduced-scale enclosure was performed at the NIST National Fire Research Laboratory [14]. The
- 54 experimental series generated an extensive dataset that identifies environmental conditions
- indicative of backdraft under various configurations, including gaseous fuel type, fire size, spark
- location, and compartment opening configuration. The substantial range of comprehensive
- 57 measurements provides detailed information that can be used to evaluate computational models.
- In this work, the wealth of experimental data produced is leveraged to assess the FDS default LES
- 59 combustion models for backdraft, particularly two-step mixing controlled fast reactions in
- 60 development for engineering applications.
- 61 The next section describes experiments and their use in defining initial conditions within the
- 62 compartment for backdraft simulation. Then, the standard FDS combustion model and the
- backdraft model setup are given in section 3. A grid sensitivity study and comparison with
- experiments on representative backdraft cases using methane and propane is shown in section 4,
- and most importantly, effects of the re-ignition model and ignition procedure in deflagration
- outcomes are presented in sections 5 and 6. Finally, some conclusions are provided in section 7.

## 67 2. Experimental Method: Definition of compartment initial conditions

- A schematic of the 1 m by 1 m by 1.5 m reduced-scale enclosure used for all experiments is
- provided in Figure 1. The enclosure was designed to scale, 2/5<sup>th</sup> the dimensions of an ASTM test
- room. Experiments were initiated when a square sand burner, with a 17.8 cm side length, was
- ignited. In these experiments, the sand burner's center was approximately 0.5 m from either side
- wall and 1.25 m from the front opening. After the initial ignition, the fire size increased to a
- predetermined heat release rate and was allowed to burn with the compartment door open for 60 s.

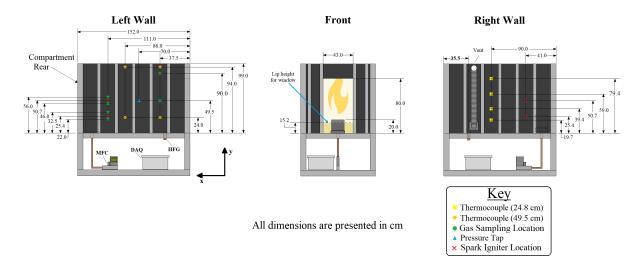


Figure 1. Schematic of the 2/5<sup>th</sup> scale compartment used in backdraft experiments

Afterward, the door was shut. While the doorway was closed, fuel continued to flow through the burner for a predetermined duration, and when reached, the flow was stopped. The fuel flow durations used in this work are shown in Table 1, corresponding to their respective fire sizes. The compartment remained closed for an additional 30.0 s after the fuel flow duration to allow the internal gases to form a more homogenous mixture. Once the door opened, spark ignitors were triggered, providing a continuous ignition source for the residing fuel and incoming gravity current of air.

Table 1: List of fuel flow times for each fire configuration

Fuel	Fire size (kW)	Fuel flow time [FFT] (s)
Methane	$25.0 \pm 1.0 \text{ kW}$	360, 390, 450
	$37.5\pm1.0~\text{kW}$	225, 285
Propane	$16.7\pm1.0~\text{kW}$	255, 270, 315
	$25.0\pm1.0~\text{kW}$	210, 285

The locations of the spark ignitors, thermocouples, and gas sampling locations are displayed in 81 Figure 1. The spark ignitors were positioned either 25.4 cm ('low spark') or 50.7 cm ('mid spark') from the compartment floor. Thermocouples were incorporated on both side walls of the 83 compartment in two different configurations. The first thermocouple configuration used 49.5 cm long, 0.3175 cm diameter sheathed Type K thermocouples configured in a rectangular orientation 85 on the left wall facing the door. The second thermocouple array included four 24.8 cm long, 86 0.3175 cm diameter sheathed Type K thermocouples configured in a line on the right wall facing 87 the door spaced approximately 19.9 cm apart. Gas samples were portioned and analyzed into two 88 gas analyzers and phi meters at one of three location pairs displayed in Figure 1. The gas analyzer 89 included one paramagnetic and two nondispersive infrared sensors to measure oxygen, carbon 90 dioxide, and carbon monoxide, respectively. The phi meter [15, 16] provided equivalence ratio

measurements of the extracted gas sample. A combination of the gas analyzer and phi meter measurements was used to estimate the concentration of major species adjusted to a wet basis. Major gas species included methane/propane, oxygen, carbon dioxide, carbon monoxide, water vapor, and nitrogen. Gas and temperature measurements were monitored using a data acquisition system sampling 1.0 Hz during each experiment. Each experiment was recorded using two external cameras, with one placed nominally 5.5 m from the front and another approximately 4.7 m from the side of the compartment. A detailed description of the compartment is available in Ref. [17].

Initial conditions for backdraft models were determined in two-compartment zones from time-averaged temperature and estimated gas concentration measurements averaged over repeated experiments with the same fuel flow duration and fire size. Zones were divided at the height of 39.25 cm from the species concentration measurements observed to be fairly homogeneous.

Time-averaged measurements were calculated from a 10.0 s time domain immediately before the compartment doorway opening.

An estimated gravity current velocity was computed using Eq. (1), which utilizes a Froude 106 number. The Froude number was determined from the ratio of the observed velocity of the gravity 107 current, and the root of the product of the normalized positive density difference,  $\beta$ , the 108 compartment height, h (1.0 m), and the gravitational constant, q (9.81 m s<sup>-2</sup>). The observed 109 velocity of the gravity current was estimated from the video recordings of the front of the compartment that captured the time to ignition from when the door opened. The density within 111 the compartment before the door opened,  $\rho$ , and the density of the ambient fluid within the gravity current,  $\rho_o$ , estimated to be 1.19 g/L  $\pm$  0.01 from sensors outside of the compartment were used 113 to calculate  $\beta$ . The gas mixture density within the compartment before the door opened was determined from the gas concentrations measured using the gas analyzer and phi meter. The 115 average Froude number was determined to be  $0.2 \pm 0.05$ .

$$U_{est} \approx \frac{1}{5} \sqrt{\beta h g} \quad ; \quad \beta = \frac{\rho_o - \rho}{\rho}$$
 (1)

The uncertainty of the experimental measurements was determined from a combination of the
Type A and B evaluation of uncertainty. The Type A evaluation of uncertainty was estimated from
the variance of the averaged measurements. The Type B evaluation of uncertainty was determined
from the reported instrumentation error. The variance between the averaged measurements was
determined to be the dominant contributor to the estimated uncertainty.

Backdraft intensity was estimated from the total heat release of the exiting flame. During each experiment, the compartment was placed under a canopy hood with a 3.0 MW calorimetry measurement system. In this instance, the total heat release was estimated using carbon dioxide generation calorimetry with a correction for carbon monoxide generation to account for unburned fuel exiting the compartment. The total heat release, *THR*, was calculated using Eq. 2:

$$THR = \sum_{t=0}^{\infty} \left[ \dot{m}_{\rm CO_2}(t) \left( \frac{\rm LHV_F \ MW_F}{\rm x \ MW_{\rm CO_2}} \right) + \dot{m}_{\rm CO}(t) \left( \frac{\rm LHV_F \ MW_F}{\rm x \ MW_{\rm CO}} - \Delta H_{\rm C,CO}^{\rm o} \right) \right] \Delta t \quad (2)$$

Here,  $\dot{m}_{\rm CO_2}(t)$  and  $\dot{m}_{\rm CO}(t)$  represent the mass flow rate of  $\rm CO_2$  and  $\rm CO$  measured in the duct, respectively. The number of carbon atoms and lower heating value of the parent fuel are represented by x and LHV<sub>F</sub>, respectively. The molecular weight of the parent fuel, carbon dioxide, and carbon monoxide is denoted as  $\rm MW_F$ ,  $\rm MW_{\rm CO_2}$ , and  $\rm MW_{\rm CO}$ , respectively. The heat of combustion for carbon monoxide,  $\rm \Delta H^o_{\rm C,CO}$  used in Eq. 2 was 10.10 kJ/g as reported by Ref. [18]. The heat released from the combustion gases trapped in the compartment before opening the door is subtracted from the measured total heat released for a backdraft estimated from scores of experiments that did not ignite and produce a deflagration.

The probability of backdraft for different compartment configurations was determined from a regression fitting. The regression fitting incorporated the number of times a backdraft was observed over the total number of times the experiment was conducted under the same condition as a function of the total chemical energy within the compartment before an anticipated backdraft. Two different regression fittings for the likelihood of backdraft were calculated for different spark locations.

## 3. Numerical Method and Model setup

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The Fire Dynamics Simulator (FDS) is a fire modeling software developed by the Fire Research 142 Division of the National Institute of Standards and Technology, USA. It solves combusting, thermally buoyant flows with application to design and evaluation of fire protection systems, fire 144 forensics studies, and general fire research, among others. Large eddy simulation based on eddy 145 viscosity models, species transport, combustion, and radiation numerical models are at the core of this application. Of primary importance to the present work is the treatment of combustion in 147 FDS, particularly the use of fast chemical reactions, which is essential to simulate realistic engineering problems in a cost-efficient manner. Fire generally involves non-premixed 149 combustion, therefore, species mixing controlled reactions are treated using the Eddy Dissipation Concept (EDC) model [19]. Here, the volumetric mass of fuel consumed by the chemical 151 combustion process is approximated as

$$\dot{m}_f''' = -\rho \frac{\min(Y_f, Y_{ox}/s)}{\tau_{mix}} \tag{3}$$

where  $\rho$  is the local cell density,  $Y_f, Y_{ox}$  are the corresponding fuel and oxidizer mass fractions, s is the stoichiometric ratio of the reaction, and  $\tau_{mix}$  is the mixing process time scale that needs to be modeled [20]. From the volumetric mass consumption (or production) of the chemical species involved in the reaction, the heat release rate per unit volume, (HRRPUV) can be computed as

$$\dot{Q}^{"'} = -\sum_{j} \dot{m}_{j}^{"'} \Delta h_{f,j}^{\circ} \tag{4}$$

where  $\dot{m}_j'''$  volumetric mass rate of change of species j, and  $\Delta h_{f,j}^\circ$  is its enthalpy of formation. EDC and fast kinetics dispense with the need to integrate in-time reaction evolution equations on each computational cell, which generally becomes prohibitively costly for practical engineering



Figure 2. Backdraft computational model: (a) Domain and compartment geometry with ignition and data collection devices, door open. (b) Oxygen mass fraction slice (blue to red, 0.04 to 0.23 kg/kg) for representative run showing the incursion of gravity current into the compartment.

problems. It also holds the unphysical assumption that any amount of fuel and oxidizer that mix will confibust (so-called harders burnt') regardless of their their modynamic state. This leads to the need to define an additional scheme to specify whether the chemical components on a computational cell react. In FDS, two extinction models are present. These models follow empirical rules, based on the definition of critical flame temperature  $T_{CFT}$  [21]:

- EXTINCTION\_1: This model is based on the concept of limiting oxygen concentration [21]  $X_{o_2,lim}$  which is assumed a linear function of the  $T_{CFT}$  up to a "free burn temperature threshold"  $T_{fb}$ . If the cell oxygen concentration is less than  $X_{o_2,lim}$ , combustion is suppressed. The default value of  $T_{fb}$  is 600 °C, consistent with experimental oxygen concentration readings in the upper layer of flashover compartments [22, 23].
- EXTINCTION\_2: This model uses both fuels and oxygen concentrations in a given computational cell. If the potential heat release of a chain of reactions cannot increase the cell temperature over the  $T_{CFT}$ , combustion is suppressed, and initial values of reactants in the cell are restored.

A detailed description of these models and their implementation is given in [20].

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Additionally, commonly in numerical simulations, the species composition in a computational 175 cell is such that combustion can proceed by the extinction model, even though mean cell 176 temperatures are low. Consequently, flames will be seen in regions where gas and air blend at nearly ambient temperature. These spurious flames are called, in practice, "ghost flames". To 178 avoid this unphysical re-ignition, an ignition model has to be added. A simple model is used in FDS, where the cell temperature has to be higher than a threshold value  $T_{TH}$  for the reaction to be 180 allowed to proceed. The value of  $T_{TH}$  is, in general, dependent on the type of problem and grid resolution, with the fuel auto-ignition temperature (AIT) [21] being technically correct for highly 182 resolved direct numerical simulations (DNS). The SFPE Handbook values for different fuels are 183 used as default  $T_{TH}$  in FDS. 184

A computational model was created to simulate with FDS the backdraft experiments given a representative set of species concentration and temperature initial conditions. FDS6.7.9-1409 was compiled from source using the Intel® OneAPI toolchain and used in this work. The Smokeview [24] render of the domain, compartment and ignition/data collection devices can be seen in Figure 2a. In all computations, the domain is of size 4.56 m x 2.22 m x 4.84 m in the x, y, z directions, deemed appropriate to capture the combustion region inside and outside the compartment correctly. The domain is divided into 48 meshes. Computations were done using FDS MPI parallel processing capability. The two-step simple chemistry model for fast reactions [25] was employed to accommodate the different species defined from the experiment

(fuel, oxygen, carbon monoxide, carbon dioxide, water vapor, and soot). Default parameters for the chemistry model were used, and soot yields were set to 0.001 kg/kg for methane and 0.024 195 kg/kg for propane [26]. Different grid sizes were evaluated for the range of cases defined by experimental initial conditions: coarse grid  $\Delta x = 5.7$  cm, medium grid  $\Delta x = 2.8$  cm, and fine 197 grid  $\Delta x = 1.4$  cm. Cells were approximately cubed and kept constant in the domain. Also, for two representative backdraft cases shown below, a very fine grid with  $\Delta x = 0.7$  cm was used to 199 evaluate grid sensitivity. Both the compartment and sand gas burner within were modeled, with 200 the compartment floor at 1 m from the domain floor to mimic the experimental setup. The 201 compartment is defined by adiabatic surfaces, and calculations were done from the door opened 202 condition for 15 s, time sufficient to completely capture the backdraft phenomenon in this 203 problem. Calculations took several minutes to a few days in local clusters at NIST. 204

As described in the previous section, 10 initial conditions were provided for species volume fractions and temperature in two zones within the compartment separated at a height from the 206 compartment floor of 39.25 cm. Together with the option to define a low or mid-ignition location, 207 20 different simulations could be performed per grid resolution. Ambient initial conditions were 208 specified outside of the compartment. A representative slice of oxygen concentration showing the 209 incursion of the gravity current into the back of the compartment can be seen in figure 2b. In this 210 figure, the dot on top of the gravity current front represents the ignition point for the low ignitor 211 configuration. To model the ignition procedure consistently with the re-ignition model, a spark 212 was defined in the corresponding ignitor location (low or mid spark) by placing a SPARK device. 213 The ignition temperature threshold  $T_{TH}$  was therefore set to 0 K in the cell containing the device. 214

### 4. Grid sensitivity and comparisons with experiments

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In Table 2, computed average gravity current height at the door  $H_{avq}$  and velocity  $U_{avq}$ , as well as 216 total heat release, THR, are given as a function of grid size for two representative backdraft cases. These cases are methane, 25 kW, FFT = 450 s, and propane, 25 kW, FFT = 285 s. The fuel 218 load inside the compartment for these was such that a high probability of backdraft was seen in 219 experiments. Both cases exhibited backdraft in calculations. The average gravity current height at 220 the door was computed from the average zero of the normal velocity in the door symmetry plane 221 between 0.5 s and 2 s (about the ignition time). Noise was noted on this function given by the 222 vortical structures formed in the shear layer between hot and cold air regions, and this was more 223 pronounced in finer grids, as expected.  $U_{avg}$  was computed by dividing the distance from the door 224 to the ignition source by the time taken from door opening to ignition. In the simulations, the 225 ignition time was computed as the time taken for the heat release rate HRR to increase over 226 0.5kW. In FDS, THR is calculated by integrating equation (4) in the spatial domain and 227 simulation time interval. It is seen that these global measures of the fluid evolution and backdraft 228 event tend to converge for these benchmark cases as the grid is refined. Differences in  $H_{avg}$  are 229 less than 2% among fine and very fine grids. Similarly, backdraft measures  $U_{avg}$  and THR show differences within 10%. 231

For reference, the total heat release measured in experiments for the methane, 25 kW, FFT = 450s

Case [Fuel, Fire Size, FFT]	Grid	$\Delta x$ (cm)	$H_{avg}$ (cm)	$U_{avg}$ (m/s)	THR (kJ)
Methane, 25 kW, 450 s	coarse	5.7	43.34	0.38	2693
Methane, 25 kW, 450 s	medium	2.8	44.61	0.49	2665
Methane, 25 kW, 450 s	fine	1.4	44.79	0.47	3580
Methane, 25 kW, 450 s	very fine	0.7	44.51	0.47	3619
Propane, 25 kW, 285 s	coarse	5.7	43.61	0.39	2357
Propane, 25 kW, 285 s	medium	2.8	44.41	0.51	2534
Propane, 25 kW, 285 s	fine	1.4	44.41	0.50	2716
Propane, 25 kW, 285 s	very fine	0.7	44.93	0.55	2780

Table 2: Grid sensitivity of two representative backdraft cases using methane and propane as fuels and low ignitor. Cases run in FDS LES mode, EXTINCTION\_2 extinction model and  $T_{TH}=300^{\circ}\mathrm{C}$ .

case was 3470 kJ  $\pm$  520 kJ, and for the propane, 25 kW, FFT = 285 s case, was 2886 kJ  $\pm$  370 kJ. The dominating factor in the total heat release uncertainty attributes to the Type A evaluation of uncertainty, specifically the variance between repeated runs. The average total heat release measurements differ by about 30% and 6% with respect to the corresponding coarse and fine grid calculations. We note that fine grid calculations are within the uncertainty computed for these experimental ensemble averages. Also, the average gravity current height at the door estimated from video analysis of the Propane, 25 kW, FFT = 285 s case is 44.7 cm, differing about 2.5 % and 1 % with respect to coarse and fine grid estimates. Experimental  $U_{obs}$  computed from video analysis of time for ignition is 0.7 m/s and 0.5 m/s, respectively, in agreement with fine grid computation results (0.47 m/s and 0.50 m/s) and analytical estimations [2]. Gravity current estimated velocity values  $U_{est}$  are 0.48 m/s for the methane, 25 kW, FFT = 450 s case, and 0.50 m/s for the propane, 25 kW, FFT = 285 s case.

Experiment pictures and simulation renders of the fireball exit for the Propane, 25 kW, FFT = 285s case are shown in Figure 3. In the simulation, Smokeview rendered pixels are colored where the heat release rate per unit volume is greater than  $200 \text{ kW.m}^{-3}$  and the temperature is higher than  $600^{\circ}\text{C}$ . Experiment and simulation quantities are not the same in this figure, but they provide insight into how the deflagration behaves in both cases. It is seen that the horizontal ejection of the deflagration is slightly weaker in the simulation, consistent with a slower evolution within the compartment. The time taken from door opening to fireball exit was estimated to be  $2.66 \text{ s} \pm 0.7$ , whereas this quantity in the fine grid simulation was 3.9 s. This result was seen using both extinction models, albeit less pronounced for EXTINCTION\_1 (3.1 s), and also in methane simulations. The burning region close to the compartment's floor, as seen in calculations of Figure 3, is consistent with higher density and concentration of propane at door opening time.

Most importantly, in the calculations of Table 2, the ignition threshold  $T_{TH}$  was set to  $300^{\circ}$ C. As shown in the next section, the number assigned to this parameter is critically important to obtain deflagration on numerical simulations of backdraft with fast mixing-controlled chemical reactions.

## 5. Effect of ignition threshold

The re-ignition model  $T_{TH}$  primarily affects backdraft simulations with fast chemistry. To get an 26 estimate for defining if the backdraft event happened in simulations, a test was devised comparing 262 the total heat release THR and total heat release outside of the compartment  $THR_{EXT}$  with 263 respect to the initial total chemical energy, TCE, of the fuel within the compartment. This last 264 value is computed as  $TCE = M_F \Delta H_F^{o}$ , where  $M_F$  is the initial mass of fuel in the compartment, 265 and  $\Delta H_E^{\rm o}$  is the fuel's heat of combustion obtained from Ref. [18]. Therefore, TCE is the 266 maximum amount of thermal energy that can be produced by combustion. Backdraft is assumed 267 to have happened if  $THR > C_1 TCE$  and  $THR_{EXT} > C_2 THR$ . The constants  $C_1$  and  $C_2$  were 268 set to 0.3, a value found to be sufficient to determine the deflagrations seen in the simulations. 269 It was noted initially that using the default fuel auto-ignition temperature for  $T_{TH}$  would lead to 270 having no deflagration for any of the 20 initial condition sets, irrespective of the grid refinement 271 used. AITs in FDS are taken from reference [21] and are for methane and propane 540°C and 272 450°C, respectively. In Figure 4, plots of backdraft events as a function of TCE for propane and 273 different values of  $T_{TH} = 300, 325, 350, 400$ °C are shown, for both "low" and "mid" ignitor 274 positions. Superimposed are the backdraft probability,  $P_{Backdraft}$ , curves computed from the 275 experiment, also as a function of the compartment fuel chemical energy load. It is emphasized 276 that the results of simulations are *deterministic*. Therefore, in order to visualize their progression, 277 "backdraft happened" in the simulations is assigned a value  $P_{Backdraft} = 1$  in these plots, and 278 "backdraft didn't happen" is given a  $P_{Backdraft} = 0$ . In Figure 4a,  $T_{TH} = 300^{\circ}$ C, fine grid results 279 are also shown. Generally, the outcomes were not different between medium and fine refinement 280 grids. 281 It is also observed that the backdraft outcomes are highly dependent on the temperature threshold 282 of the FDS default re-ignition model. Within a small range of 100°C, outcomes range from 283 "deflagration happened" being strongly over-predicted, to practically "no deflagration" simulated. 284 This result is independent of the fuel load in the compartment. A value of  $T_{TH} = 325^{\circ}$ C gives the 285 best approximation to experiment-derived backdraft probability curves. Ideally, a good 286 representative value of  $T_{TH}$  could be found, although it is unclear how that would have to be 287 modified in different scenarios and for other fuels. This poses a great challenge for simulating 288

# 6. Effect of ignition procedure

and more event variability with TCE was seen.

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To test the effect of the ignition procedure in backdraft outcomes, a grid of 8 SPARK devices was defined in a horizontal plane area of 2 cm by 6 cm located near the experiments' ignition spark location. This resulted in a larger mesh volume that would undergo combustion in the ignitor region for each grid resolution, as the corresponding computational cells were assigned  $T_{TH}=0$  K. For comparison, in Figure 5, similar plots to Figure 4 are shown for this larger spark

these transient events using fast chemistry. We also note that analogous behavior was seen using

transition was found to happen at a slightly lower temperature threshold for the ignition model,

the temperature controlled EXTINCTION\_1 method. Similar trends were seen for methane, albeit

299 ignitor.

In terms of backdraft events, the effect of having a larger ignition volume is akin to using a slightly lower  $T_{TH}$  in this problem. A larger ignitor volume where air and fuel can mix and react potentially produces higher heat release and temperatures in computational cells surrounding the ignition region. This leads to a larger chance of prompting deflagration. Similar issues have been found when modeling piloted ignition of burners as observed in Ref. [27]. Although this behavior is expected, it also points to the difficulty of modeling the complex thermo-chemical process of fuel ignition by a spark when using simplified mixing-controlled combustion.

### 7. Conclusions

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In this work, a wealth of experimental data was leveraged to define initial conditions and validation information, which can be used to assess the capability of fire models in simulating backdraft phenomena. Initial condition text data can be obtained by contacting the authors. A model for said experiments was developed within the FDS framework and used to test the simulation software with standard mixing-controlled fast chemical reactions.

Comparison with experiments of global backdraft quantities showed good agreement in relatively 313 fine grids and select high fuel load cases. Nevertheless, it was seen that backdraft simulation 314 presents a great challenge for fast chemistry combustion models used in realistic engineering 315 applications. In particular, simple ignition models based on a temperature threshold require 316 tuning this parameter. It was noted that variations of 50°C on this parameter could significantly 317 change backdraft outcomes across various grids, ignition sources, and initial conditions. Ideally, a 318 variation of the re-ignition model that is not as sensitive to a hand-picked temperature parameter 319 would be available. An option being considered by this group is the use of a test temperature that 320 is not only the mean cell temperature but also contains a subgrid component. 321

On the other hand, it was shown that the ignition procedure based on dropping the temperature 322 threshold at the ignition source location also greatly influences the backdraft simulation outcome. 323 It is unclear how the ignition volume should be defined for backdraft triggered by point sources 324 like sparks, and more study is required on this topic. Further, in these calculations uniform 325 composition and temperature conditions in two zones were used from experiment sensor averages. 326 The effect of this simplification in the results should be tested. One possible way is to actually 327 simulate the fuel loading process in the compartment up to door opening. Another option is to add 328 stochastic disturbances of initial composition fields to assess their effect in backdraft outcomes. 329

Finally, the experiments modeled in this work involved simple hydrocarbon gases, with known chemical and thermal properties. In real fires, gaseous fuels are the product of complex pyrolysis processes and therefore backdraft outcomes are expected to be heavily influenced by their combustion characteristics. Although not touched upon, experiments similar to the ones described section 2 were also performed using wood-cribs as fuel. Their modeling and analysis is left for a future effort.

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## 411 Figure captions

- Figure 1. Schematic of the  $2/5^{\rm th}$  scale compartment used in backdraft experiments
- Figure 2. Backdraft computational model: (a) Domain and compartment geometry with ignition
- and data collection devices, door open. (b) Oxygen mass fraction slice (blue to red, 0.04 to 0.23
- kg/kg) for representative run showing the incursion of gravity current into the compartment.
- Figure 3.Backdraft exit deflagration from the door for propane, 25 kW, FFT = 285s case. (a)
- Experiment photograph at t=2.73 from the door opening, (b) Smokeview HRRPUV render for
- fine grid simulation t = 4.22 s, (c) Experiment t = 2.90, and (d) Simulation t = 4.37 s.
- Figure 4. Simulated Backdraft events as a function of total fuel chemical energy TCE for
- propane. (a) Ignition model  $T_{TH} = 300^{\circ}\text{C}$ , (b)  $T_{TH} = 325^{\circ}\text{C}$ , (c)  $T_{TH} = 350^{\circ}\text{C}$ , and (d)
- 421  $T_{TH} = 400^{\circ}$ C.
- Figure 5. Simulated Backdraft events as a function of total fuel chemical energy TCE for
- propane and 8 spark grid ignitor. (a) Ignition model  $T_{TH}=300^{\circ}\mathrm{C}$ , (b)  $T_{TH}=350^{\circ}\mathrm{C}$ .

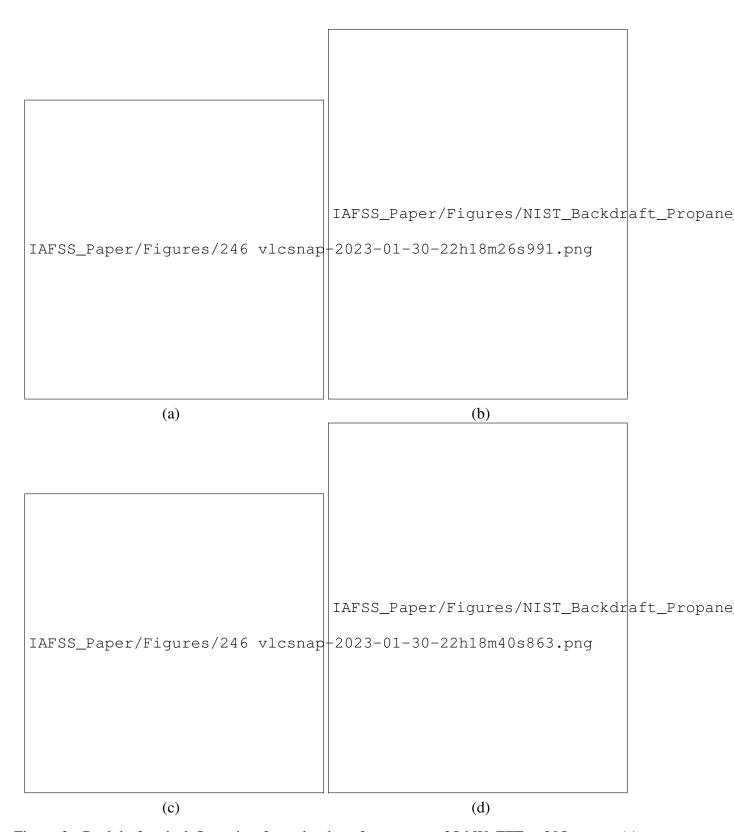


Figure 3. Backdraft exit deflagration from the door for propane, 25 kW, FFT = 285s case. (a) Experiment photograph at t=2.73 from the door opening, (b) Smokeview HRRPUV render for fine grid simulation t=4.22 s, (c) Experiment t=2.90, and (d) Simulation t=4.37 s.

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