

Towards Effective Swarm-Based GPS Spoofing Detection in Disadvantaged Platforms

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- OV-1: Air Launched Effects (ALE)**
- Legend**
- Weapons Launch or ALE Launch (Solid blue line)
 - Friendly Data (Dashed blue line)
 - Enemy Data (Dashed red line)
 - Flight Path (Dashed black line)
- 1 IADS Breach**
- 2 Reconnaissance**
- 3 Attack**
- Aircraft and Systems:**
- FARA: Future Attack Reconnaissance Aircraft
 - FLRAA: Future Long Range Assault Aircraft
 - UAS: Unmanned Aircraft System
 - ALE: Air Launched Effects
- Effects and Data Links:**
- LRPF (Long Range Penetration Flight)
 - LRPM (Long Range Penetration Mission)
 - See (See)
 - Strike (Strike)
 - Illuminate (Illuminate)
 - Decoy
 - Deep Maneuver Area
 - Close Area
 - LZ (Landing Zone)

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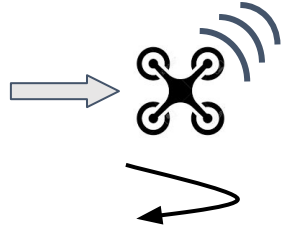
- ## Challenge: ALE sensor suites are limited

- Can we develop intelligent sensor fusion techniques to remediate GPS spoofing on ALE platforms?***

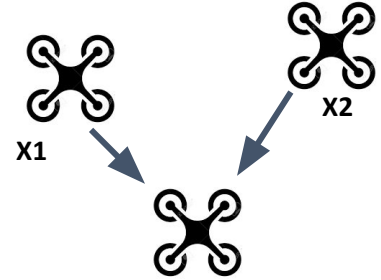
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- Diagram illustrating a GPS spoofing attack on a helicopter:
- GPS Satellite:** i^{th} GPS satellite.
 - Targeted antenna:** The antenna on the helicopter being spoofed.
 - True location:** The actual location of the helicopter, marked with a red 'X' and a red circle with a slash.
 - Simulated location:** The location the helicopter believes it is, marked with a green triangle.
 - Spoofer:** A device (red circle with a slash) that intercepts and retransmits signals to the helicopter.
 - Ground Station (RX, TX):** The receiver (RX) and transmitter (TX) of the spoofer.
 - Signal Types:**
 - Red dashed lines:** Signals to the GPS.
 - Blue solid lines:** Range info used to spoof.
- Legend:
- RX, TX:** spoofer receive and transmit
 - Red dashed line:** signals to the GPS
 - Blue solid line:** range info used to spoof

Example ways we can use sensors

1. IMU: Use **Inertial Measurement Unit** (linear acceleration+angular velocity) and last known good position to perform dead reckoning



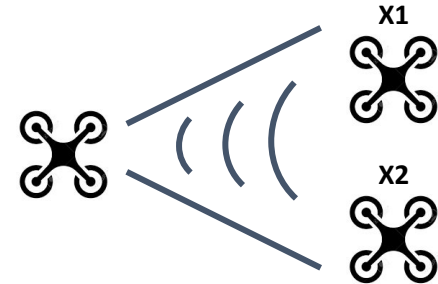
2. Communications: Request GPS coordinates of neighbors, combine them together (e.g., averaging)



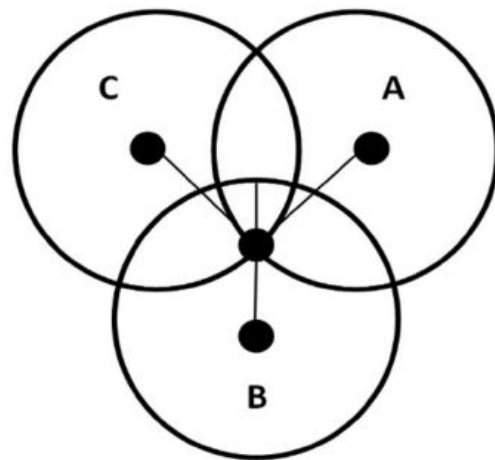
3. RSSI: use **Radio Signal Strength Indicator** to estimate distance and angle to neighboring ALEs, ground stations



4. Camera: use visual odometry to estimate distance, heading, from observing other ALEs and landmark features



- $RSSI = A - 10 \cdot \eta \cdot \log_{10}(d) + \varepsilon_{RSSI}$
- Estimator: $\hat{d} = 10^{\frac{A - RSSI}{10\eta}}$
- $\hat{p} = \operatorname{argmin} \sum_{i=1}^n (\hat{d}_i - ||p - p_i||)^2$
- A: reference distance
- η : path loss exponent
- ε_{RSSI} : white gaussian noise
- \hat{d}_i : RSSI-inferred distance to the i^{th} neighbors
- p_i : neighbours position
- Based on **lognormal shadowing path loss model**[1], distance between transmitter and receiver can be inferred from RSSI, multiple RSSI from neighbors allow us to perform multilateration.
- Solving \hat{p} can be treated as an optimization problem, we solve it by Levenberg-Marquardt method.



State Variables

$$\mathbf{R}_{t+1} = \mathbf{R}_t \exp((\boldsymbol{\omega}_t dt)_{\times})$$

$$\mathbf{v}_{t+1} = \mathbf{v}_t + (\mathbf{R}_t \mathbf{a}_t - \mathbf{g}) dt$$

$$\mathbf{p}_{t+1} = \mathbf{p}_t + \mathbf{v}_t dt + \frac{1}{2} (\mathbf{R}_t \mathbf{a}_t - \mathbf{g}) dt^2$$

R_t : directional rotation matrix

v_t : velocity

p_t : Position

g : gravity

a_t : acceleration read from IMU

ω_t : angular velocity read from IMU

- We develop systems of equations to model sensor properties
 - E.g., state propagation models for IMU dead-reckoning
 - The pose of UAV can be inferred provided continuous reads from IMU.
- Naïve IMU Dead-Reckoning will be erroneous after a few seconds, due to the bias and noise from IMU reading.

$$\omega_m = \omega + \boxed{\omega_b + \omega_n},$$

$$a_m = a + \boxed{a_b + a_n},$$

Error

$$\delta\omega_b = \omega_b^t - \omega_b^{t-1} = \epsilon_r^\omega$$

$$\delta a_b = a_b^t - a_b^{t-1} = \epsilon_r^a$$

ω_m, a_m : Measured value from IMU

ω, a : True value

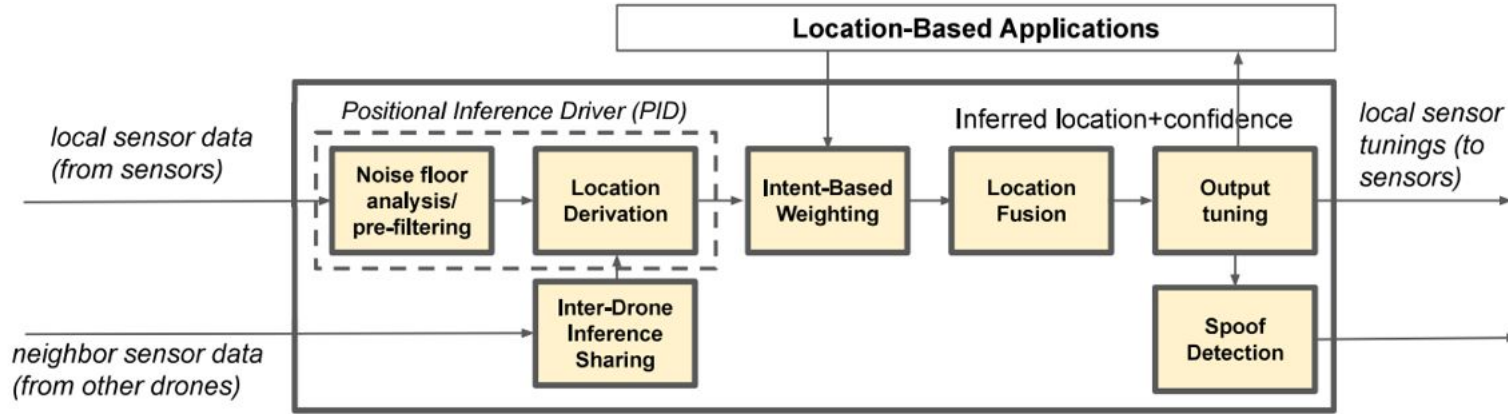
ω_b, a_b : Measurement bias

ω_n, a_n : Measurement gaussian noise

$\epsilon_r^\omega, \epsilon_r^a$: Random-walk gaussian noise

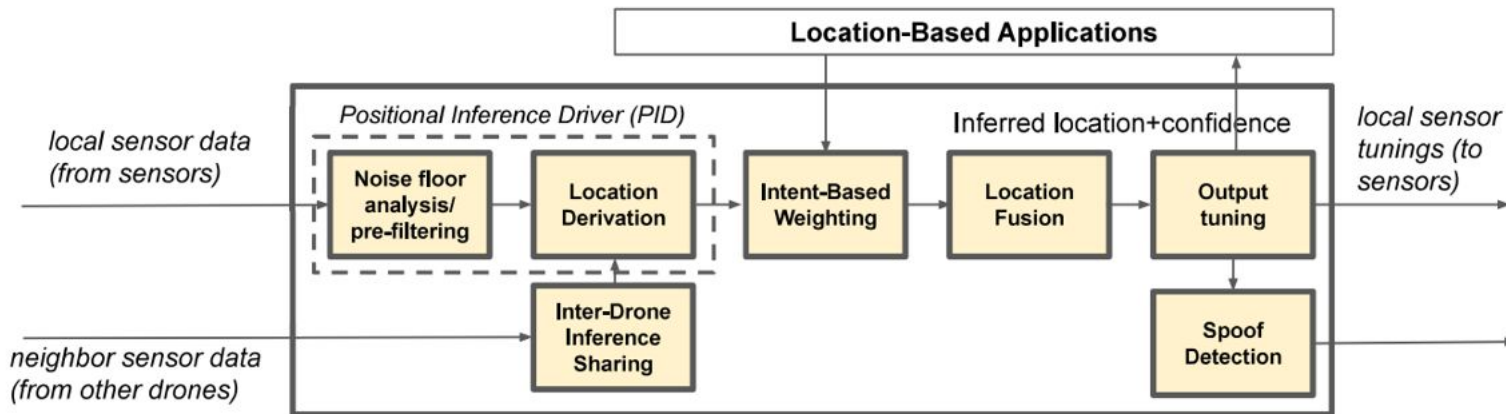
- The bias and measurement noise both contribute to the error between measured value from IMU and the ground truth value.
- The bias is not static, but driven by a random-walk, which controlled by $\epsilon_r^\omega, \epsilon_r^a$.
- Neglecting the bias and noise can drift our pose estimation far away from the ground truth. Hence, we need to model the bias and noise properly.

More on sensor fusion part.



Key idea: Kalman filtering to combine sensors to maximize ability to remediate GPS spoofing; multi-stage pipeline to iteratively improve localization

- Noise floor analysis removes background noise and attacker-introduce randomness from sensor inputs, Sensor-specific weightings compute heuristic curve to weight based on parameter, combining readings across sensors with Kalman filter



Input is non-positional sensor inputs, output is inferred location with estimation confidence (feedback to sensors, GPS-leveraging applications)

- Approach: convert sensor data into location information, weighting by confidence, then fusing locations into single estimation
- Key questions: how to convert non-positional sensor data into locations, how to determine weights of sensors?

○ State Propagation (IMU)

$$\delta x \leftarrow f(x, \delta x, u_m) = F_x(x, u_m) \cdot \delta x + G_x(x) \cdot w$$
$$x \leftarrow F(x, u_m)$$

$$\bullet \delta \mathbf{x} \triangleq [\delta p, \delta v, \delta \theta, a_b, \omega_b]^T$$

$$\bullet \mathbf{x} \triangleq [p, v, q]^T$$

$$\bullet u_m \triangleq [a_m, \omega_m]^T$$

$$\bullet w \triangleq [a_n, \omega_n, \delta a_b, \delta \omega_b]^T$$

○ State Update (RSSI)

$$y = h(x) + v$$

$$h(x) = A - 10 \cdot \eta \cdot \log_{10}(\|p - p_i\|)$$

y are RSSI measurements

Algorithm 1 ES-EKF Algorithm

Input: $x_{initial}$, $\delta x_{initial}$, $P_{initial}$, u_m , V , Q

Output: \hat{x} , $\delta \hat{x}$, P

loop

$$\hat{u} = CORRECTION_{bias}(u_m, \delta \hat{x})$$

$$\hat{x} \leftarrow F(\hat{x}, u)$$

$$P \leftarrow F_x P F_x^T + G_x Q G_x^T$$

if RSSI measurement available **then**

$$H = H_x \cdot X_{\delta x}$$

$$K \leftarrow P H^T (H P H^T + V)^{-1}$$

$$P \leftarrow (I - K H) P$$

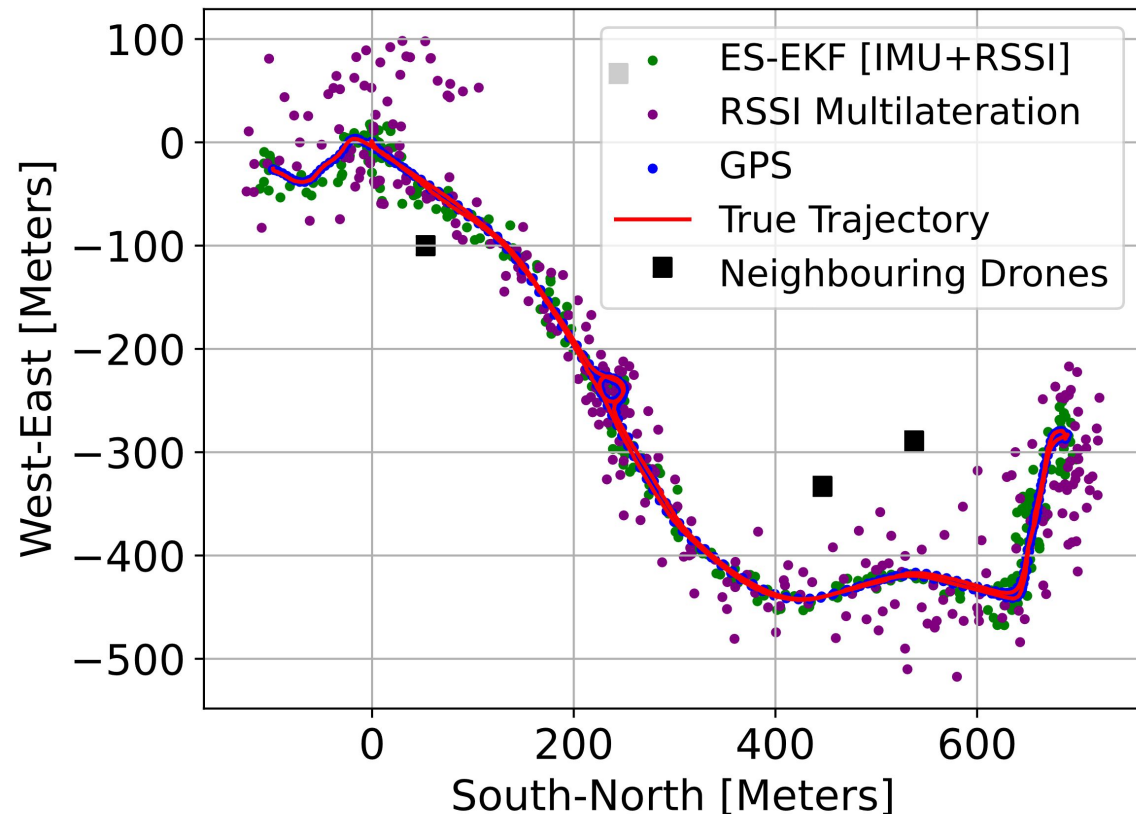
$$\delta \hat{x} \leftarrow K(y - h(\hat{x}))$$

$$\hat{x} = CORRECTION_{perturbation}(\hat{x}, \delta \hat{x})$$

end

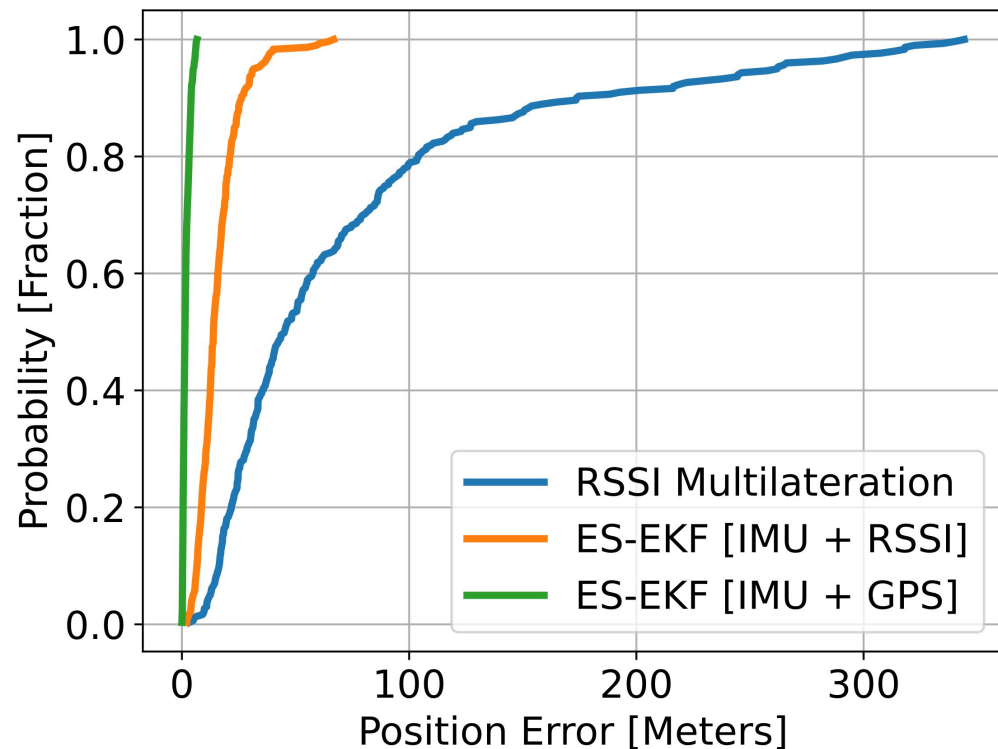
end loop

Compensate



- True trajectory is based on valid GPS data.
- RSSI Multilateration is prone to have outliers, also deviates more from true trajectory.
- Sensor fusion with IMU+RSSI removes most of outliers, closely align with the true trajectory.

Distribution of positional error



RSSI Multilateration:

- High positioning error.
- Long-tailing effect.
- Worst 10% deviations exceed 174 meters.

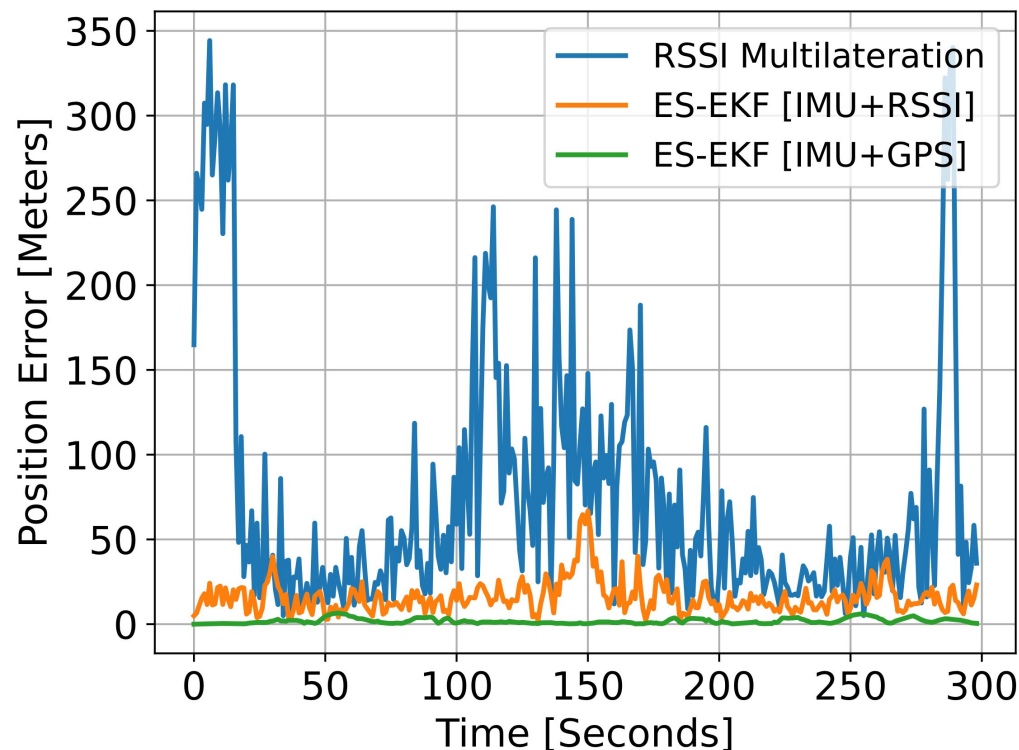
ES-EKF Fusion of IMU and RSSI:

- Alleviates long-tailing effect.
- Increases positioning accuracy.
- Worst 10% deviations only exceed 26 meters.

Improvement with ES-EKF Fusion:

- Nearly 80% improvement in positioning accuracy over RSSI multilateration.

Time-series variation in position error



ES-EKF vs. RSSI Multilateration:

- Similar error peaks at the same time periods.
- Magnitude of error peaks significantly lower in ES-EKF.
- Sensor fusion significantly attenuates RSSI error.

Variance Reduction:

- Similar patterns observed among different algorithms.
- ES-EKF fusion eliminates variance in position uncertainty.
- Achieves a two-magnitude reduction in positioning error variance compared to RSSI multilateration.

GPS spoofing can cause significant damage to ALE assets

- Strategic adversaries can amplify power of these attacks across time and space

Leveraging common, low-cost sensing infrastructures can offer substantial protection

- Kalman-filtering based combination outperforms individual and statically weighted combinations
- ES-EKF prevents introduction of non-linear errors and boosts performance compared to sole reliance on individual sensors

Future work: leverage deep learning to fuse sensor inputs, develop real-time navigation algorithm robust to GPS attacks, develop flight algorithms that change flight plan to maximize ability to correct coordinates in presence of spoofing.